# Optimum Loft Angle for Greatest Carry Distance

# Group D

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# 1 Response

The loft angle of the golf club which maximises the range of the golf ball trajectory is referred to as the optimum loft angle. This investigation consisted of two main parts: the impact of the club, and the flight of the golf ball. The golf ball considered is the Titleist Pro V1x, with characteristics outlined under (2.1) Assumptions.

### 1.0.1 Atmospheric conditions at different golf courses

Table 1: Atmospheric conditions at each golf course

|                           | 1                 |              | 0            |                        |
|---------------------------|-------------------|--------------|--------------|------------------------|
| Location                  | Temperature $(K)$ | Humidity (%) | Altitude (m) | Air density $(kg/m^3)$ |
| Renaissance, East Lothian | 287               | 85           | 84           | 1.128                  |
| La Paz, Bolivia           | 278               | 60           | 3342         | 0.861                  |
| Sentosa, Singapore        | 307               | 75           | 33           | 1.154                  |

The tournament in Singapore took place in January 17th-20th 2019. The tournament in East Lothian is scheduled to take place in July, 8th-14th (2019) and the Bolivian tournament is scheduled for August (2019). The values for temperature and humidity are average values for the given time of year.

#### 1.1 Results

The following graphs are for the conditions at Renasissance, East Lothian.

Figure 1: Effect of loft angle on carry distance

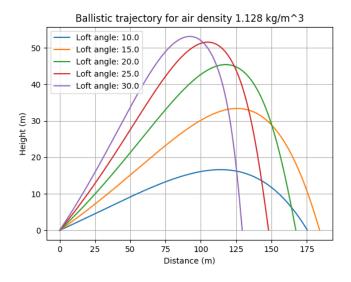
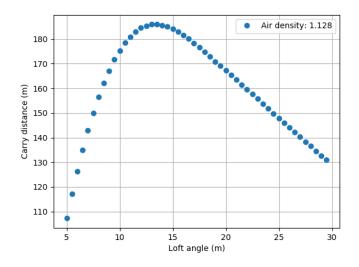


Figure 2: Loft angle vs carry distance to obtain optimum loft angle



From the investigation, the recommended loft angle to maximise the carry distance based on a club speed of  $51.4~ms^-1$  for the Scottish Open at the Renaissance Club in East Lothian is  $13.4^{\circ}$  which produces a flight distance of  $186~\mathrm{m}$ .

The following graphs are for the conditions at La Paz, Bolivia.

Figure 3: Effect of loft angle on carry distance

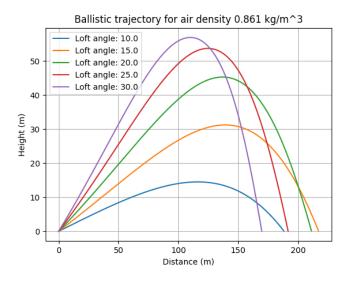
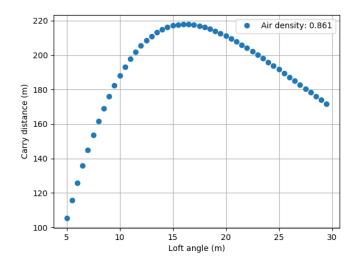


Figure 4: Loft angle vs carry distance to obtain optimum loft angle



The recommended loft angle to maximise the flight distance based on a club speed of  $51.4\,\mathrm{m\,s^{-1}}$  for the La Paz golf club in Bolivia is  $16.2^\circ$  which produces a carry distance of  $218\,\mathrm{m}$ .

The following graphs are for the conditions at Sentosa, Singapore.

Figure 5: Effect of loft angle on carry distance

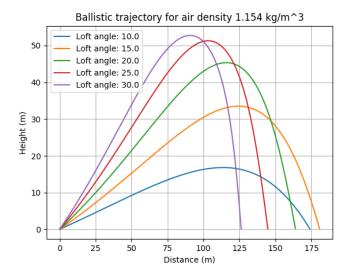
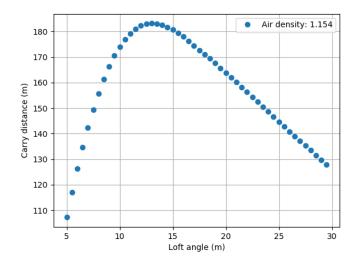
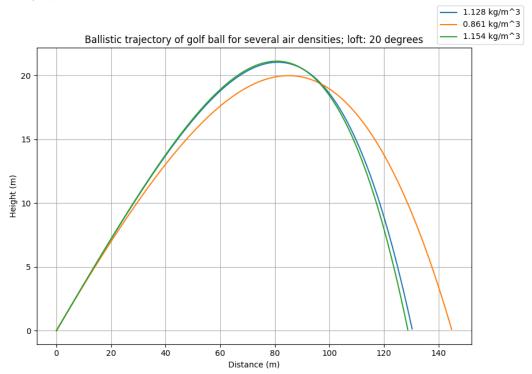


Figure 6: Loft angle vs carry distance to obtain optimum loft angle



The recommended loft angle to maximise the flight distance based on a club speed of  $51.4\,\mathrm{m\,s^{-1}}$  for the Sentosa, is  $13.2^\circ$  which produces a carry distance of  $183\,\mathrm{m}$ .

Figure 7: Effect of club speed on carry distance and optimum loft angle,  $\omega_{spin}=300\,\mathrm{rad}\,\mathrm{s}^{-1},\,v_i=50\,\mathrm{m}\,\mathrm{s}^{-1}$ 



# 2 Theory

# 2.1 Assumptions

The following assumptions are made throughout the report and model:

- · golf course is level and has no effect on trajectory;
- · height of the tee is negligible;
- gravitational field strength is constant  $(9.81 \,\mathrm{m/s^2})$  and does not flucuate with height;
- · driver is roughly a flat plate and strikes the ball precisely at the center, with no draw or fade;
- · the mass of the club head is significantly greater than that of the shaft, so we consider the shaft's influence to be neglible;

· the golf ball is a Titleist Pro V1x, with a mass of  $45.93\,\mathrm{g}$ , diameter of  $42.67\,\mathrm{mm}$ , a moment of inertia of  $0.009\,145\,\mathrm{g}\cdot\mathrm{m}^2$ , and 352 circular dimples.

## 2.2 Impact

#### 2.2.1 Conservation of energy

The collision between the club head and golf ball is inelastic, meaning the kinetic energy of the system is not conserved, but rather some of the energy is converted into different forms. One such form is elastic energy in the deformation of the golf ball. This has significant effect on the initial velocity of the golf ball, as the more the ball is deformed, the less kinetic energy the ball will have after the collision. The coefficient of restitution e is the ratio of the final and initial velocities between the golf ball and club head after the collision and it is introduced in calculations. Further energy is lost as heat and sound.

#### 2.2.2 Conservation of momentum

The law of conservation of momentum was applied and the velocities were resolved into components.

$$Mv_{cfn} + mv_{bfn} = Mv_{ci}\cos\theta \tag{1}$$

$$Mv_{cfp} + mv_{bfp} = -Mv_{ci}\sin\theta \tag{2}$$

 $v_{bfn}$  and  $v_{bfb}$  can be expressed in terms of  $v_{ci}$ , the initial club speed, as equations (3-4).

$$v_{bfn} = (1+e)v_{ci}\frac{\cos\theta}{1+\frac{m}{M}}\tag{3}$$

$$v_{bfp} = -v_{ci} \frac{\sin \theta}{1 + \frac{m}{M} + \frac{mr^2}{I}} \tag{4}$$

These two vector components can be added to yield  $v_{bo}$  and  $\phi_{bo}$ , the velocity and its direction for the ball on departure; equations (5-6)[5]

$$v_{bo} = v_{bf} = \sqrt{v_{bfn}^2 + v_{bfp}^2} \tag{5}$$

$$\phi_{bo} = \theta + \tan^{-1} \frac{v_{bfp}}{v_{bfn}} \tag{6}$$

Lieberman and Johnson give values for e decreasing from approximately 0.76 for impact speeds of  $37 \,\mathrm{m \, s^{-1}}$  to values of around 0.72 for impact speeds of  $50 \,\mathrm{m \, s^{-1}}$ . Applying a linear fit gives the empirical equation (7)[5].

$$e = 0.86 - 0.0029v_{impact}\cos\theta\tag{7}$$

1600 spin
1400
1200
1200
1000
400
200
0
200
400
600
80

Figure 8: Loft angle and spin

## 2.3 Flight conditions

The main atmospheric factor affecting the golf balls flight is air density which is dependent on air pressure, temperature and humidity. When air density increases, there is increased resistance against the ball during its flight, thus the maximum carry distance is less. Air molecules have greater kinetic energy as the temperature of the air increases, causing them to occupy a larger volume. This results in a decrease in air density. When the air pressure is increased, air density also increases as more collisions between particles occur. The relative humidity of air is a measure of the water vapour relative to temperature and is the percentage of water vapour that could potentially be held in the air at that given temperature. The air density of humid air is less than dry air. Relative humidity is given by equation (8):

Loft angle (degrees)

$$\phi = \frac{P_w}{P_w'} \times 100 \tag{8}$$

where  $P_w$  is the pressure for water vapour and  $P'_w$  is the equilibrium vapour pressure, that is the maximum pressure it could be in that given temperature value. This value of  $P'_w$  can be obtained by Antoine's equation (9) [6]:

$$P_w' = e^{\frac{A-B}{C+T}} \tag{9}$$

where T is the temperature in K and A, B, and C are are component specific constants for the given medium. Through the use of Dalton's law for partial pressures, the molar

fraction of elements which compose the atmosphere can be calculated using equation (10):

$$x_w = \frac{P_w}{P_T} \tag{10}$$

where  $x_w$  is the molar fraction of water vapour and  $n_T$  is the total number of moles which is obtained using the ideal gas law (11):

$$P_T V = n_T R T \tag{11}$$

where  $P_T$  is the total pressure, which is found using the barometric formula (12) [3]:

$$P_T = P_0 e^{\frac{-Mg}{RT_0}h} \tag{12}$$

where  $P_0$  and  $T_0$  are, respectively, pressure and temperature at sea level (103 125 Pa, 288.15 K), g is the gravitational field strength, L is the temperature lapse rate (0.0065 K m<sup>-1</sup>). At higher altitudes, the air molecules can spread out further resulting in a decrease in air density.

The molar fraction of water vapour can be obtained through equation (13)

$$x_w = \frac{n_w}{n_T} \tag{13}$$

From here, the number of moles for each major component of the atmosphere nitrogen, oxygen, argon, and water vapour - was obtained using equation (14):

$$n_T - n_w = n_O + n_N + n_{Ar} (14)$$

and considering the fraction of each component of air:

$$n_N = 0.7808(n_T - n_w) \tag{15}$$

$$n_O = 0.20195(n_T - n_w) (16)$$

$$n_{Ar} = 0.0093(n_T - n_w) (17)$$

By the consideration of the molar masses, the density of air can be calculated by the equation (18):

$$\rho = \frac{((M_N P_N) + (M_O P_O) + (M_{Ar} P_{Ar}))}{RT}$$
(18)

where P is the partial pressure of an element, found by equation (19):

$$P = \frac{n_x RT}{V} \tag{19}$$

where  $n_x$  is the number of moles for a given element. Consequently, this allowed the air density to be found for each course based on environmental conditions. [4]

# 2.4 Flight

# 2.4.1 Simple golf ball

A simple golf ball experiencing only weight may be modelled by the following system of differential equations;

$$a_x = \frac{\partial v_x}{\partial t} = 0 \tag{20}$$

$$a_y = \frac{\partial v_y}{\partial t} = -g \tag{21}$$

where q is the gravitional field strength.

Assuming the initial velocity is  $v_0$  and the launch angle is  $\theta$ , equations (20-21) can be solved to give equations (22-23):

$$v_x = v_0 \cos \theta \tag{22}$$

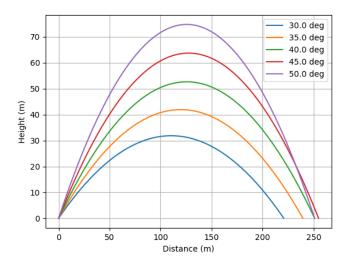
$$v_y = v_0 \sin \theta - gt \tag{23}$$

Integrating equations (22-23) with respect to time yields the displacement as a function of time to give equations (24-25) [5]:

$$x = v_0 t \cos \theta \tag{24}$$

$$y = v_0 t \sin \theta - \frac{1}{2} g t^2 \tag{25}$$

Figure 9: Golf ball trajectory, no drag or lift



Range against loft angle, with an initial velocity of 50 m/s

255

250

245

230

225

230

225

220

30

35

40

45

50

55

Loft angle (deg)

Figure 10: Range as a function of loft angle, no drag or lift

Figures (1-2) show the maximum range is when the loft angle at  $45.0^{\circ}$ , as predicted by the equations of projectile motion.

#### 2.4.2 Smooth golf ball experiencing drag

The drag equation (26)

$$\vec{F}_d = \frac{1}{2} A C_d \rho_{air} |\vec{v}| \vec{v} \tag{26}$$

where  $\rho_{air}$  is the density of air; A is the reference area, which in the case of a smooth sphere of radius r, is the cross-sectional area  $\pi r^2$ ;  $C_d$  is the coefficient of drag, which is dependent on the Reynolds number; and  $\vec{v}$  is the flow velocity relative to the golf ball. In this case, we assume the air is stationary and the golf ball is moving through the air with velocity  $\vec{v}$ .

Applying equation (26) to equations (20-21), we get

$$\frac{\partial v_x}{\partial t} = -k|v_x|v_x \tag{27}$$

$$\frac{\partial v_y}{\partial t} = -g - k|v_y|v_y \tag{28}$$

where  $k = \frac{1}{2}AC_d\rho_{air}$ . These equations already do not have a closed-form solution and require numerical methods.

Figure 11: Golf ball trajectory when experiencing drag but no lift,  $C_d = 0.5$ 

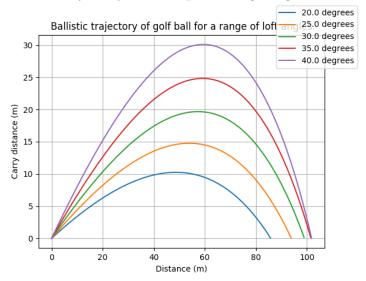


Figure 12: Range as a function of loft angle,  $C_d = 0.5$ 

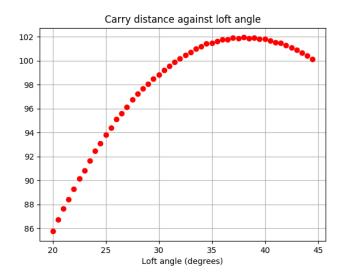


Figure (4) shows that the maximum range is achieved at  $40.2^{\circ}$ , for an initial velocity of  $50\,\mathrm{m\,s^{-1}}$ . Additionally, the range is significantly decreased when drag was added to the model. The greatest range with drag is about  $114\,\mathrm{m}$  versus the previous range of  $255\,\mathrm{m}$  at  $45.0^{\circ}$ . However,  $C_d$  is not constant and depends on the Reynolds number, which is proportional to the velocity of the golf ball. The Reynolds number

is given by equation (29)[1]:

$$R = \frac{2vr}{\nu} \tag{29}$$

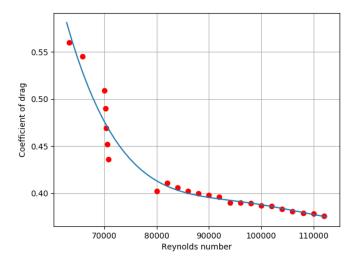
where  $\nu$  is the kinematic viscosity of air.

Using data from Alam 2011[1], Python was used to construct a quartic fit of drag coefficient as a function of Reynolds Number, as shown in Figure (5).

A quartic approximation is best as any greater degree would begin to oscillate too much at the edges (Runge's phenomenon). Likewise, a lesser degree would be too linear to give any meaningful relation.

One of the limitions of this approach is that there is no data above Reynolds number  $112,000~(38.39\,\mathrm{m\,s^{-1}})$  or below  $63,300~(21.70\,\mathrm{m\,s^{-1}})$ , so the approximation does not hold for those conditions, and may in fact be drastically worse due to the chaotic behaviour of the polynomial at the aforementioned values. To overcome this, the drag coefficient was fixed at 0.8 for Reynolds numbers less than 53,000 and fixed at 0.37 for Reynolds numbers greater than 120,000.

Figure 13: Drag coefficient as a function of Reynolds number with curve of best fit



Applying the approximation for the coefficient of drag, the optimum loft angle decreases down to about 34.5°, shown in figures (6-7).

Figure 14: Trajectory of golf ball with drag considering Reynolds number,  $v_i = 50\,\mathrm{m\,s^{-1}}$ 

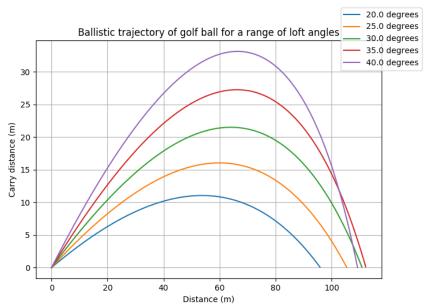


Figure 15: Range against loft angle for golf ball with drag considering Reynolds number,  $v_i = 50\,\mathrm{m\,s^{-1}}$ 



#### 2.4.3 Smooth golf ball experiencing lift

Lift on a golf ball is caused by the Magnus effect, which is dependent on the backspin of the ball. The equation is similar to the drag equation, however the direction of the force is perpendicular to both angular velocity and translational velocity.

$$F_l = \frac{1}{2} \rho_{air} C_l |v|^2 (\hat{\omega} \times \hat{v})$$
(30)

where  $C_l$  is dependent on the spin parameter of the ball according to [2].

$$C_l = -3.25S^2 + 1.99S (31)$$

The spin parameter is given by the ratio of the magnitude of tangential velocity to the magnitude of translational velocity.

$$S = \frac{r|\omega|}{|v|} \tag{32}$$

# References

- [1] Firoz Alam, Tom Steiner, Harun Chowdhury, Hazim Moria, Iftekhar Khan, Fayez Aldawi, and Aleksandar Subic. A study of golf ball aerodynamic drag. *Procedia Engineering*, 13:226–231, 2011.
- [2] P W Bearman and J K Harvey. Golf ball aerodynamics. *Aeronautical Quarterly*, 27(02):112–122, May 1976.
- [3] Mário N. Berberan-Santos, Evgeny N. Bodunov, and Lionello Pogliani. On the barometric formula. *American Journal of Physics*, 65(5):404–412, May 1997.
- [4] Roger Legg. Properties of humid air. In Air Conditioning System Design, pages 1–28. Elsevier, 2017.
- [5] A. Raymond Penner. The physics of golf: The optimum loft of a driver. *American Journal of Physics*, 69(5):563–568, May 2001.
- [6] Denis Roizard. Antoine equation. In *Encyclopedia of Membranes*, pages 1–3. Springer Berlin Heidelberg, 2014.

The model was written using Python version 3.8, on a Linux machine, although earlier versions (i.e. 3.5) should work. It requires the following Python packages: NumPy, SciPy, and Matplotlib. Use ./model3d.py -h for a list of parameters. It also supports 3d plots, though these are experimental. The source code is also hosted as a git repository at https://github.com/s-ballantyne/gdp.

Listing 1: Python model)

```
#!/usr/bin/env python3
      import numpy as np
      import argparse
       {\bf from} \ \ {\bf scipy.integrate} \ \ {\bf import} \ \ {\bf odeint} \ \ {\bf as} \ \ {\bf integrate} 
      from matplotlib import pyplot as plot
from numpy.linalg import norm
from mpl_toolkits.mplot3d import Axes3D
      parser = argparse.ArgumentParser()
      # Ball parameters
13
      constants = parser.add_argument_group("Constants")
constants.add_argument("-m", "--mass", default=0.04593, help="Mass of ball (kg)")
constants.add_argument("-r", "--radius", default=0.04267/2, help="Radius of ball (m)")
17
      # Intital parameters initial parameters = parser.add_argument_group("Initial parameters")  
#initialparams.add_argument("-vi", "--velocity", type=float, default=50, help="Initial velocity (m/s)")  
initialparams.add_argument("-yi", "--height", type=float, default=0, help="Initial height (m)")
26
     30
      # Lost united parser.add_argument("-li", "--loftinitial", type=float, default=10, help="Loft angle (initial)") parser.add_argument("-lf", "--loftfinal", type=float, default=35, help="Loft angle (final)") parser.add_argument("-st", "--step", type=float, default=5, help="Loft angle (step)")
32
      parser.add_argument("-v", "--verbose", action="store_true")
38
      # Ball speed catculations
parser.add_argument("--clubmass", type=float, default=0.2, help="Mass of club head (kg)")
parser.add_argument("--vclub", type=float, default=51.4, help="Club speed (m/s)")
parser.add_argument("--inertia", type=float, default=9.145e-6, help="Inertia of golf ball")
40
42
      # Parse args
44
      args = parser.parse_args()
46
      \# \ Input \ validation
      # Input variation
assert args.loftfinal > args.loftinitial, "Final loft angle must be gretaer than initial loft angle!"
assert args.step != 0, "Step must be non-zero!"
assert ((args.loftfinal - args.loftinitial) / args.step).is_integer(), "Step size must divide the change in loft angle
48
50
      assert args.mass != 0, "Mass must be non-zero."
      assert args.radius!= 0, "Radius must be non-zero." assert args.viscosity!= 0, "Kinematic viscosity must be non-zero." assert args.density!= 0, "Density of air must be non-zero."
      g = args.gravity
58
      density = args.density
59
      # Ball speed from club speed and loft angle
      def ball-speed(theta):
    theta = np.radians(theta)
    e = 0.86 - 0.0029 * args.vclub * np.cos(theta)
63
            bfn = (1 + e) * args.vclub * np.cos(theta) / (1 + args.mass / args.clubmass)
            65
67
```

# Spin

```
def ball_spin(theta):
    theta = np.radians(theta)
    bfp = args.vclub * np.sin(theta) / (1 + args.mass / args.clubmass + (args.mass * args.radius**2 / args.inertia))
 70
 72
 73
            return args.mass * bfp * args.radius / args.inertia
 74
      # Coefficient of drag from Reynolds number, based on degree four polynomial.
 76
      def re_to_cd(re):
# Clamp output value as it is only an approximation
if re > 120000:
 77
78
 79
            return 0.370 elif re < 53000:
 80
 81
 82
                return 0.8
 83
            \# Array of coefficients
 84
           # Array 6, 5-2, coeffs = np.array([ 9.46410458e-20, -3.80736984e-14, 5.72048806e-09, -3.81337408e-04,
85
86
 87
                          9.92620188\,\mathrm{e}{+00}
 88
                       1)
 89
 90
           # Return value of polynomial approximation return np.polyval(coeffs, re)
 91
 93
      95
 97
 99
      # Linear velocity to drag coefficient
def sphere_cd(velocity, radius):
    cd = re_to_cd(reynolds(velocity, radius))
101
103
            return cd
105
     107
109
110
111
      \# Lift equation \# F_l = 1/2 * air density * ref. area * coefficient * |v|^2 * (what x vhat) def lift(density, area, cl, velocity, rvelocity):
112
113
114
115
            if cl == 0:
116
                 \textbf{return} \ \text{np.array} \left( \left[ \begin{array}{cc} 0 \ , & 0 \ , & 0 \end{array} \right] \right)
117
           S = 0.5 * density * area * cl
119
           \# Cross product of angular velocity and linear velocity, for direction of spin rxv=np.\,cross\,(rvelocity\,,\,velocity\,)
120
122
           rxv /= norm(rxv)
123
           # Magnitude of spin is considered in coefficient of lift return S * norm(velocity)**2 * rxv
124
125
126
      \# Simple golfball, no drag, no lift, smooth
      class BasicGolfball:
def __init__(self):
# Properties
128
130
                 self.mass = args.mass
131
132
                  self.radius = args.radius
                 # Coordinates
134
                 self.x = 0

self.y = args.height

self.z = 0
136
138
139
                  self.vx = 0
140
                 self.vy = 0
self.vz = 0
142
143
                 # Rotational velocities

self.rvx = 0

self.rvy = 0

144
145
146
                  self.rvz = 0
147
            \#\ Reference\ area\ ,\ for\ a\ sphere\ this\ is\ the\ cross-section\ .
148
            def area(self):
    return np.pi * self.radius**2
149
150
```

```
151
            # Set initial velocity
def set_velocity(self, v, theta):
    self.vx = v * np.cos(np.radians(theta))
    self.vy = v * np.sin(np.radians(theta))
153
154
155
             # Set spin
157
            def set_spin (self, spin):
    self.rvx, self.rvy, self.rvz = spin
158
159
160
             # Get all coordinates def coords (self):
161
162
                  return np.array([self.x, self.y, self.z, self.vx, self.vy, self.vz, self.rvx, self.rvx, self.rvz])
163
164
             # Set all coordinates [x, y, z, vx, vy, vz, rvx, rvy, rvz]

def set_coords(self, coords):
    self.x, self.y, self.z, self.vx, self.vy, self.vz, self.rvx, self.rvx, self.rvz = coords
165
166
167
168
            # Returns numpy array of position coordinates def position(self):
169
170
                  return np.array([self.x, self.y, self.z])
171
172
             # Returns numpy array of velocity at the current position
def velocity(self):
    return np.array([self.vx, self.vy, self.vz])
173
174
175
176
              EReturns numpy array of acceleration at the current position
             def acceleration (self):
178
179
                  return np.array([0, -g, 0])
180
            \# Returns numpy array of rotational velocity (spin) at the current position \operatorname{\mathbf{def}} rvelocity(self):
182
                  return np.array([self.rvx, self.rvy, self.rvz])
184
             \# Returns numpy array of rotational acceleration at the current position def racceleration (self):    return np.array([0, 0, 0])
186
188
            \# Returns numpy array of differential eqns to be solved by odeint def differentials (self): d = np.zeros(9)
190
192
                  d[0:3] = self.velocity()

d[3:6] = self.acceleration()
193
194
195
196
                  d[6:9] = self.racceleration()
197
198
                  return d
\frac{199}{200}
             \# (Internal) Updates coordinates and returns list of equations to solve (for odeint)
                  __eqns(self, t, coords):
self.set_coords(coords)
201
             def
203
                   \begin{array}{ll} \textbf{if} & \texttt{args.verbose:} \\ & \textbf{print}(\texttt{t}, \texttt{self.velocity()}, \texttt{self.rvelocity()}, \texttt{self.acceleration()}, \texttt{self.racceleration())} \end{array} 
204
205
206
                  return self.differentials()
207
            209
211
213
                  out = np.array([e for e in res if e[1] >= 0])
                  return out
215
       # Simple golf ball but with drag class DragGolfball(BasicGolfball):
217
219
            def __init__(self):
    BasicGolfball.__init__(self)
220
221
              4 Coefficient of drag from velocity & radius
            # Coejjic...
def cd(self):
223
224
225
                  return sphere_cd(norm(self.velocity()), self.radius)
226
             \mathbf{def} acceleration (self):
                  return BasicGolfball.acceleration(self) + fd / self.mass
227
228
229
230
      # Golfball with lift and drag
```

```
class LiftGolfball(DragGolfball):
    def __init__(self):
        DragGolfball.__init__(self)
232
234
235
                  # Returns spin factor
def spinf(self):
    v = norm(self.velocity())
    w = self.radius * norm(self.rvelocity())
    return w / v
236
237
238
239
240
241
                  # Returns coefficient of lift based on spin factor
def cl(self):
    s = self.spinf()
    return -3.25 * s**2 + 1.99 * s
242
243
244
245
246
\frac{247}{248}
                   def acceleration(self):
    fl = lift(density, self.area(), self.cl(), self.velocity(), self.rvelocity())
    return DragGolfball.acceleration(self) + fl / self.mass
249
250
                  # Spin decreases by about 1% every second def racceleration(self):
return -0.01 * self.rvelocity()
251
252
253
254
255
         if __name__ == "__main__":
    # Initial conditions
    for density in [1.128, 0.861, 1.154]:
        plot.figure()
        for theta in np.arange(args.loftinitial, args.loftfinal, args.step):
            ball = LiftGolfball()
            ball.set_velocity(ball.speed(theta), theta)
            ball.set_spin([0, 0, ball_spin(theta)])
256
257
259
260
261
263
                                   \begin{array}{lll} \mathrm{res} &=& \mathrm{ball.solve} \left( 0 \,, & 10 \right) \\ \mathrm{x} \,, & \mathrm{y} \,, & \mathrm{z} \,=\, \mathrm{res} \,. \mathrm{T} \end{array}
265
266
267
268
                                   plot.plot(x, y, label="Loft angle: " + format(theta, ".1f"))
269
270
                           plot.legend()
                           plot.regend()
plot.grid(True)
plot.xlabel("Distance (m)")
plot.ylabel("Height (m)")
plot.ylabel("Height (m)")
plot.title("Ballistic trajectory for air density " + format(density, ".3f") + " kg/m^3")
271
272
273
274
275
276
                           plot.figure()
277
                           xdata = []
ydata = []
for theta in np.arange(5, 30, 0.5):
278
279
\frac{280}{281}
                                   ball = LiftGolfball()
ball.set_velocity(ball_speed(theta), theta)
282
                                    ball.set_spin([0, 0, ball_spin(theta)])
283
284
                                   \mathtt{res} \; = \; \mathtt{ball.solve} \, (0 \, , \; \; 10)
285
                                  x, y, z = res.T
286
287
                                   xdata.append(theta)
288
                                   ydata. append (x[-1])
                           plot.plot(xdata, ydata, 'o', label="Air density: " + format(density, ".3f"))
plot.legend()
290
                           plot.grid(True)
plot.xlabel("Loft angle (m)")
plot.ylabel("Carry distance (m)")
292
294
                   plot.show()
296
```