

# دانشکده مهندسی مکانیک

## درس مكانيك كامپوزيت پيشرفته

تمرین:تمرین دوم

نام و نام خانوادگی: صبا عباسزاده منتظری

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### 1 Question 1

1. Given two pieces of a unidirectional composite material joined in a manner shown in Figure below determine the deformed shape under uniaxial stress. Is deformed shape a,b or c? what principle is involved?

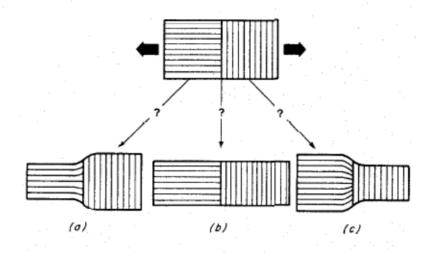


Figure 1: Possible deformed shapes of a 0/90 composite

For a 0 degree composite in on axis coordinate:

$$\varepsilon_x = \frac{1}{E_x} \sigma_x$$

$$\varepsilon_y = -\frac{\nu_x}{E_x} \sigma_x = -\nu_x \varepsilon_x$$
(1)

For a 90 degree composite in on axis coordinate: he strain equations in the y-direction are:

$$\varepsilon_{y} = \frac{1}{E_{y}} \sigma_{y}$$

$$\varepsilon_{x} = -\frac{\nu_{y}}{E_{y}} \sigma_{y} = -\nu_{y} \varepsilon_{y}$$
(2)

Now, in the global coordinate system along the y-direction, the strain for the intersection line is calculated as:

$$\varepsilon_{y0,\text{global}} = -\frac{\nu_x}{E_x} \sigma 
\varepsilon_{y90,\text{global}} = -\frac{\nu_y}{E_y} \sigma$$
(3)

The compliance matrix in 2D (without shear loading) is:

$$\begin{bmatrix} \varepsilon_x \\ \varepsilon_y \end{bmatrix} = \begin{bmatrix} \frac{1}{E_x} & -\frac{\nu_{xy}}{E_x} \\ -\frac{\nu_{yx}}{E_y} & \frac{1}{E_y} \end{bmatrix} \begin{bmatrix} \sigma_x \\ \sigma_y \end{bmatrix}$$
(4)

By symmetry of the compliance matrix:



$$\frac{\nu_x}{E_x} = \frac{\nu_y}{E_y}$$

From this, it can be concluded that the strains along the global y-direction are the same. Therefore, the correct deformed shape is (b).

#### 2 Question 2

2. Write a computer program to calculate the stiffness and compliance matrices of T300/5208 unidirectional ply.

First we compute stiffness(Q) and compliance(S) matrices for Transversely isotropic materials and save it into a file to load it whenever we want:

Then we substitute the numbers given in the table below into the matrices:

Type	Material	$E_x$ (GPa)	$E_y$ (GPa)	$\nu_x$	$E_s$ (GPa)	$v_f$	Specific gravity
T300/5208	Graphite /Epoxy	181	10.3	0.28	7.17	0.70	1.6
B (4)/5505	Boron /Epoxy	204	18.5	0.23	5.59	0.5	2.0
AS/3501	Graphite /Epoxy	138	8.96	0.30	7.1	0.66	1.6
Scotchply 1002	Glass /Epoxy	38.6	8.27	0.26	4.14	0.45	1.8
Kevlar 49 /Epoxy	Aramid /Epoxy	76	5.5	0.34	2.3	0.60	1.46

Table 1: Properties of composite materials

Now we should load the code and substitute:

```
1 load('compliance')
 load('stiffness')
 %%properties of the composite
 T300_5208=struct('Ex',181,...
      'Ey',10.3,...
      'Vx',0.28,...
      'Es',7.17);
 % Compute Vy dynamically and add it to the structure. It is computed by
 % supposing symmetricity.
 T300_5208.Vy = T300_5208.Vx * (T300_5208.Ey / T300_5208.Ex);
12 S_T300_5208=subs(S,fieldnames(T300_5208), struct2cell(T300_5208));
Q_{T300}_{5208} = subs(Q_{fieldnames}(T300_{5208}), struct2cell(T300_{5208}));
disp('Numerical Compliance Matrix (S):');
16 disp(vpa(S_T300_5208, 4));
17 %%Show Q
18 disp('Numerical Stiffness Matrix (Q):');
```



```
19 disp(vpa(Q_T300_5208, 4));
```

Answers:

$$S = \begin{bmatrix} 0.005525 & -0.001547 & 0 \\ -0.001547 & 0.09709 & 0 \\ 0 & 0 & 0.1395 \end{bmatrix}$$

$$Q = \begin{bmatrix} 181.8 & 2.897 & 0 \\ 2.897 & 10.35 & 0 \\ 0 & 0 & 7.17 \end{bmatrix}$$

#### 3 Question 3

3. Write a computer program to find the stress field based on the following strain field for T300/5208 unidirectional ply.

$$\left\{ \begin{array}{c} \varepsilon_x \\ \varepsilon_y \\ \varepsilon_s \end{array} \right\} = \left\{ \begin{array}{c} 0.001 \\ 0.003 \\ 0.002 \end{array} \right\}$$

We multiply Q matrix and given strain:

```
%%define strain
epsilon=[0.001;0.003;0.002]
%%constitutive law
sigma=Q_T300_5208*epsilon,
disp(vpa(sigma, 4));
```

Answer:

$$\left\{ \begin{array}{c} \sigma_x \\ \sigma_y \\ \sigma_s \end{array} \right\} = \left\{ \begin{array}{c} 0.1905 \\ 0.03394 \\ 0.01434 \end{array} \right\}$$