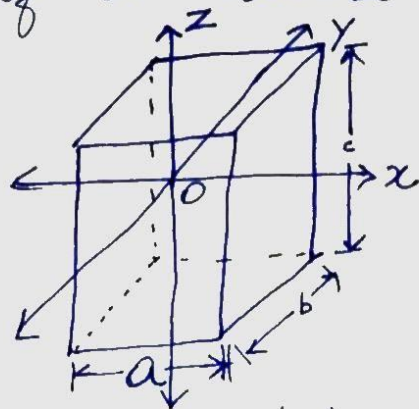


Theory for Problem - 4:

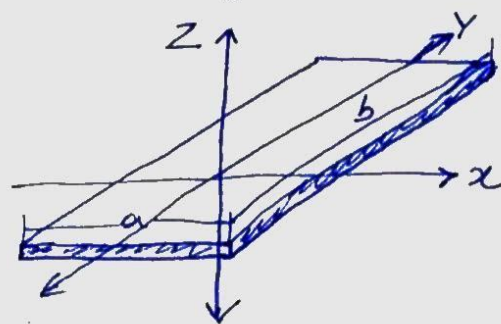
The moment of Inertia of a cuboid

$$I_{\text{cuboid}} = \frac{M}{12} \begin{bmatrix} b^2 + c^2 & 0 & 0 \\ 0 & c^2 + a^2 & 0 \\ 0 & 0 & a^2 + b^2 \end{bmatrix}$$



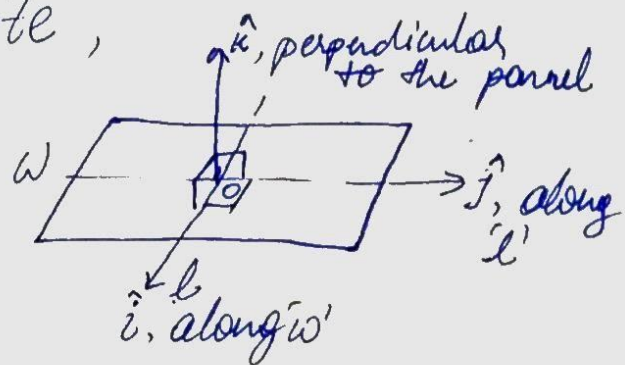
If $c \ll a$ & $c \ll b$, a thin plate,
the c^2 is insignificant compared to a^2 & b^2
So the moment of Inertia is as follows

$$I_{\text{plate}} = \frac{M}{12} \begin{bmatrix} b^2 & 0 & 0 \\ 0 & a^2 & 0 \\ 0 & 0 & a^2 + b^2 \end{bmatrix}, \quad c^2 \approx 0$$



In the context of the satellite,

$$I_{\text{panel}} = \frac{m}{12} \begin{bmatrix} l^2 & 0 & 0 \\ 0 & w^2 & 0 \\ 0 & 0 & l^2 + w^2 \end{bmatrix}$$



$$I_{\text{box}} = \frac{M}{12} \begin{bmatrix} b^2 + c^2 & 0 & 0 \\ 0 & c^2 + a^2 & 0 \\ 0 & 0 & a^2 + b^2 \end{bmatrix}$$

where the center of this is assumed to be the origin of the whole satellite.

In order to Express I_{panel} along the main Co-ord-Syst along the box, we can write this Relation

$$I_{\text{panel}}^{\text{in box}} = R_2(\theta) I_{\text{panel}} R_2(-\theta)^T$$

Proof:

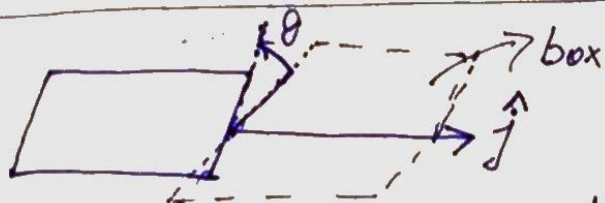
I_{panel} is the Principle MoI Matrix & we know the

$$\underbrace{I_{\text{MOI}}^c}_{\text{Principle}} = R_{pr}^T I_{\text{MOI}}^b R_{pr}$$

$$R_{pr} I_{\text{MOI}}^c = \underbrace{R_{pr} R_{pr}^T}_{I_{3 \times 3}} I_{\text{MOI}}^b R_{pr} \quad (\text{multiplying with } R_{pr})$$

$$\boxed{R_{pr} I_{\text{MOI}}^c R_{pr}^T = I_{\text{MOI}}^b}$$

(post multiplying with R_{pr}^T)



because the panel is rotated by $+\theta$, a rotation of $-\theta$ of the principle axis will change to the box (Q) main Co-ord System

$$\therefore \boxed{R_{pr} = R_2(-\theta)}, \text{ where } R_2(\theta) = \begin{bmatrix} \cos(\theta) & 0 & -\sin(\theta) \\ 0 & 1 & 0 \\ \sin(\theta) & 0 & \cos(\theta) \end{bmatrix}$$

So, the moment of Inertias of the three components are

$$\left. \begin{array}{ccc} I_{\text{box}}^{\text{in box}}, & I_{\text{panel}}^{\text{in box}}, & I_{\text{panel}}^{\text{in box}} \\ \downarrow & \downarrow & \downarrow \\ \text{box} & \text{Right}, & \text{left} \end{array} \right\} \text{all expressed in the main co-ord system}$$

the \vec{r}_{cm} for the components are

$$\vec{r}_{\text{cm}, \text{box}} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}; \quad \vec{r}_{\text{cm}, \text{left panel}} = \begin{bmatrix} 0 \\ -\frac{b}{2} - \frac{d}{2} \\ \frac{c}{2} \end{bmatrix}; \quad \vec{r}_{\text{cm}, \text{right panel}} = \begin{bmatrix} 0 \\ \frac{b}{2} + \frac{d}{2} \\ \frac{c}{2} \end{bmatrix}$$

The Following Eq (also coded in the function)

$$M_{\text{tot}} = M_{\text{box}} + 2 M_{\text{panel}} = M + 2m$$

$$\vec{r}_{\text{cm}, \text{tot}} = \frac{1}{M_{\text{tot}}} \left[\vec{r}_{\text{cm}, \text{box}} M_{\text{box}} + \vec{r}_{\text{cm}, \text{left panel}} M_{\text{panel}} + \dots + \vec{r}_{\text{cm}, \text{right panel}} M_{\text{panel}} \right]$$

$$I_{\text{MOI}, \text{tot}}^{\text{in box}} = \sum_{i=1}^3 \left(I_{\text{MOI}, i}^{\text{in box}} + M_i \left[\left(\vec{r}_{\text{cm}, i} - \vec{r}_{\text{cm}, \text{tot}} \right)^T \left(\vec{r}_{\text{cm}, i} - \vec{r}_{\text{cm}, \text{tot}} \right) I_{3 \times 3} \dots - \left(\vec{r}_{\text{cm}, i} - \vec{r}_{\text{cm}, \text{tot}} \right) \left(\vec{r}_{\text{cm}, i} - \vec{r}_{\text{cm}, \text{tot}} \right)^T \right] \right)$$

$i = 1, 2, 3$ are the three components box, left panel & right panel