

General Information

Prof. Antonella Ferrara

<https://scholar.google.com/citations?user=r5JuMskAAAAJ&hl=en>

Course Teaching Material:

KIRO UNIPV <https://elearning.unipv.it/>

- [504462 - PROCESS CONTROL 2025-26 - PROF.SSA FERRARA ANTONELLA](#)

Lecture Time-table:

<http://www-3.unipv.it/ingserv/orario2526/1sem/insegnamenti/ProCont.html>

Exams:

<https://studentionline.unipv.it/esse3/Home.do>

<https://kirotesting.unipv.it/>



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Introduction

- The course and its objectives
- Kiro UNIPV
- Teaching material
- The preparatory course
- The exam
- Office hours
- The Intelligent Robotics Lab
- Theses and research
- The role of Ph.D. Students and PostDocs



Introduction

- Program of the course:

Advanced SISO control schemes:

Pre-filters and parallel compensators, two degrees of freedom control schemes, compensation of measurable disturbances, systems with delays and Smith Predictor, Padé approximation, decoupling in the frequency domain, control of open loop unstable systems.

Advanced MIMO control schemes:

Decoupling based control schemes, decentralized control, relative gain array.

PID controllers:

Features and properties. Rules for the empirical calibration. Wind-up and anti wind-up schemes.

Digital control:

Discrete-time systems. The concept of equilibrium for discrete-time systems. Stability of linear time-invariant discrete-time systems. Jury test. Digital control schemes. Zeta transform and its properties. Transfer functions in the z domain. Sampling and aliasing. Choice of the sampling time. Zero-order-Hold. Discretization of continuous-time controllers. Bilinear transformation, Euler and Tustin methods.

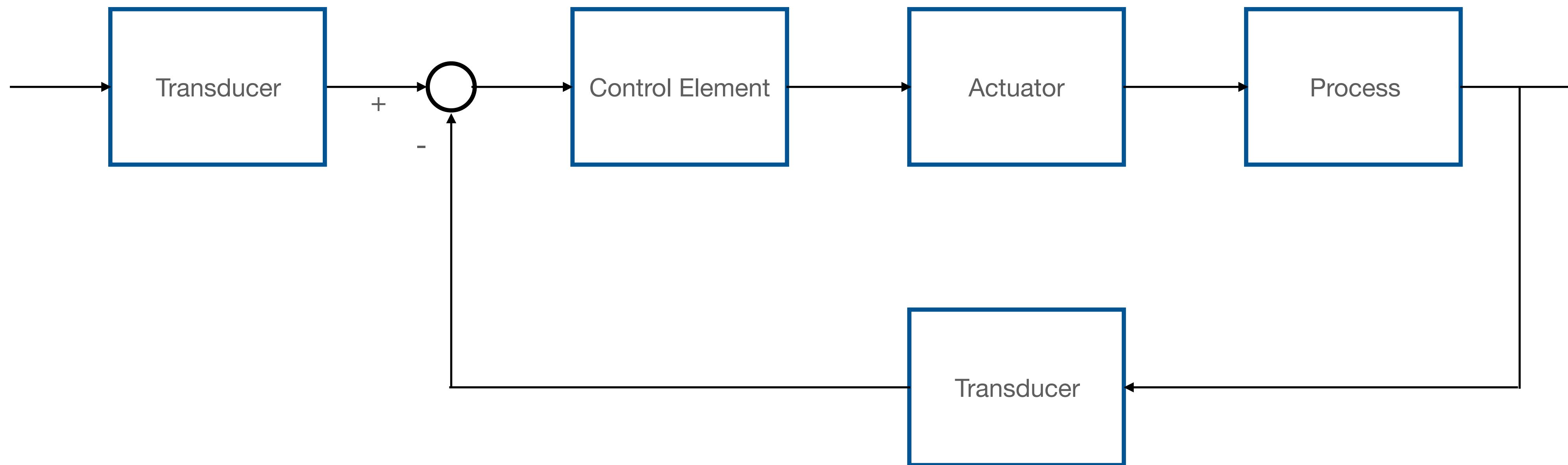


Introduction

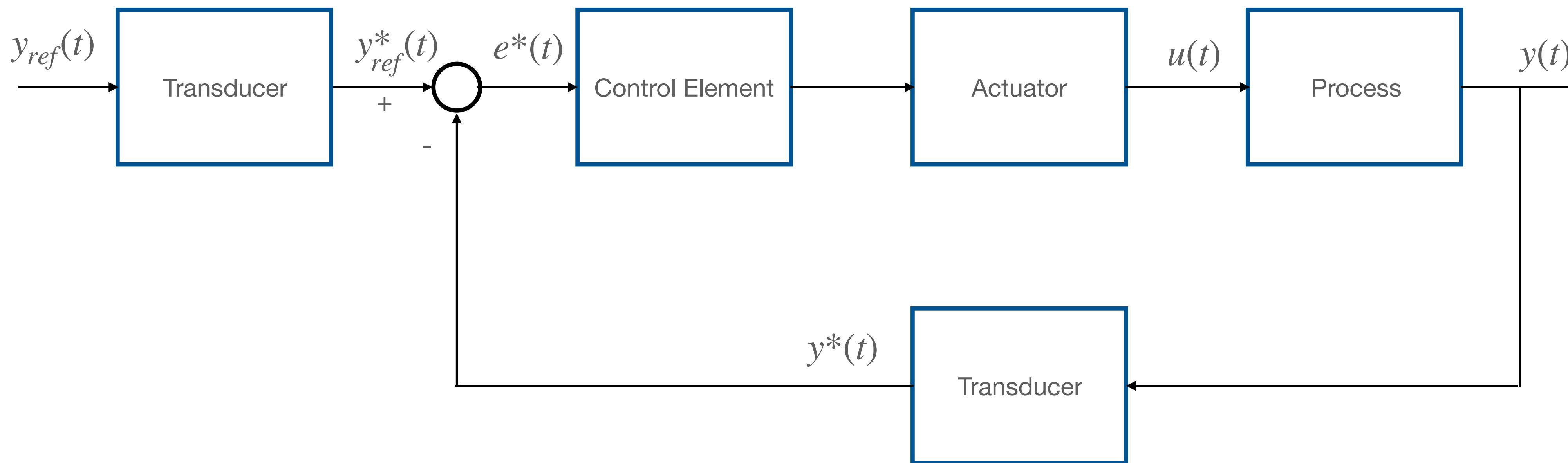
- Some of the figures in these slides, kindly provided by McGraw-Hill, are those of the Textbook:



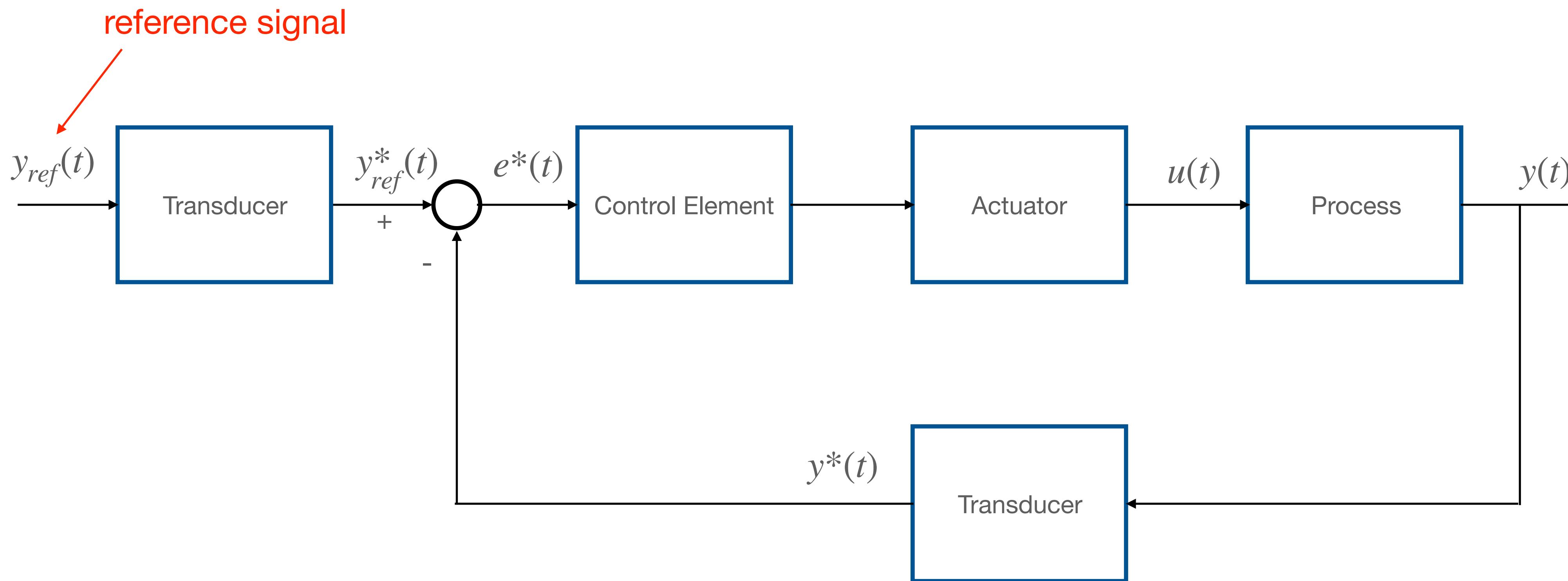
Basic Control Scheme



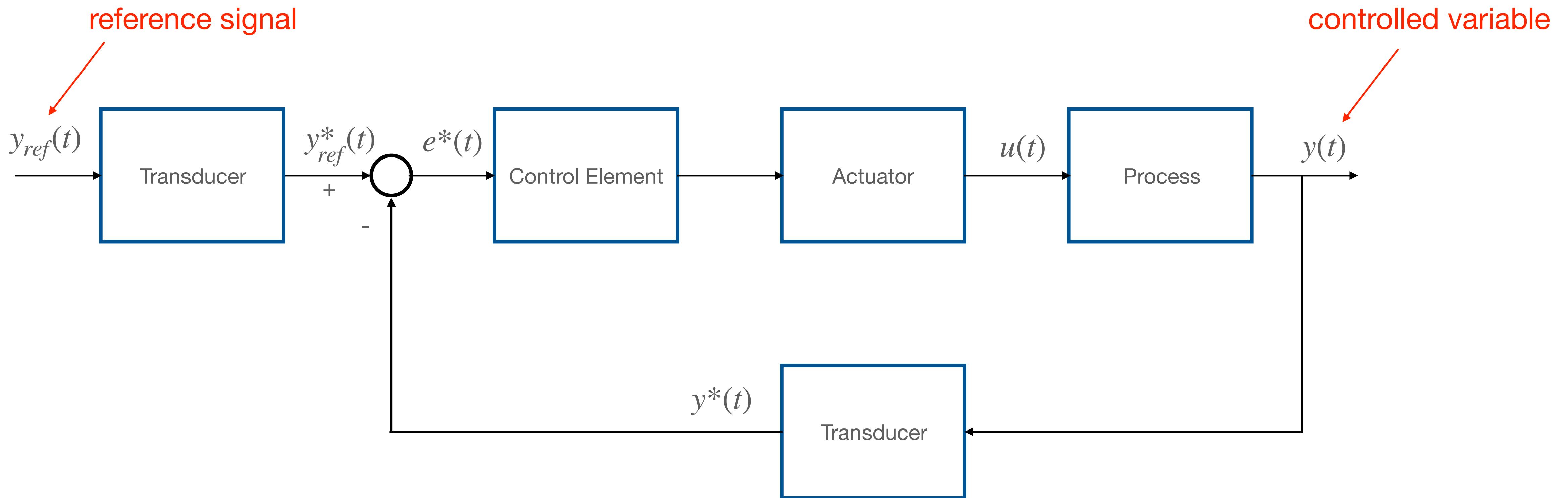
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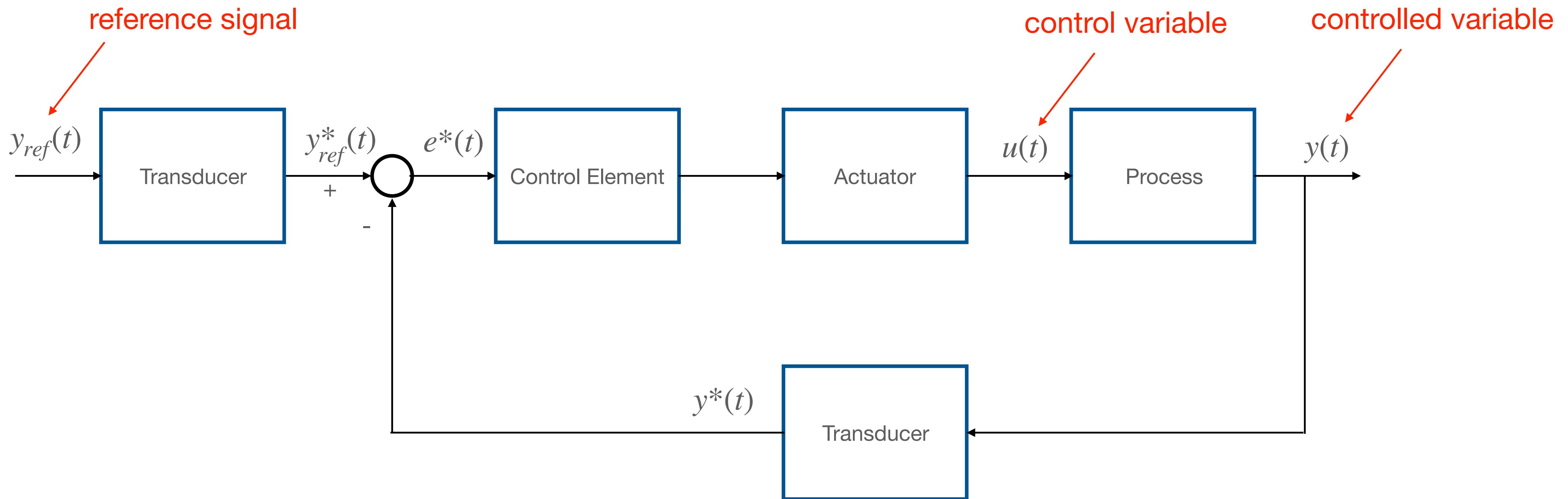
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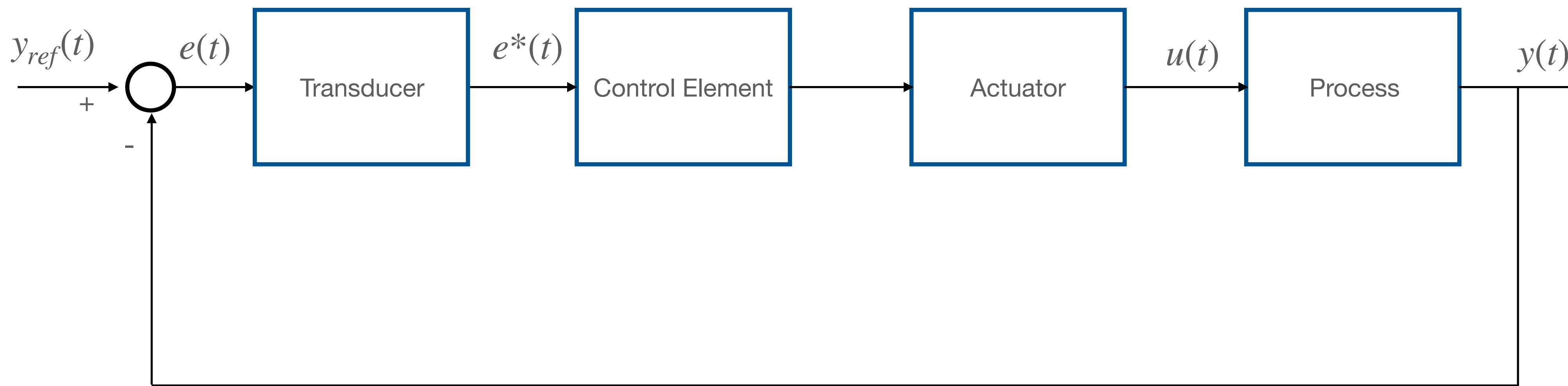
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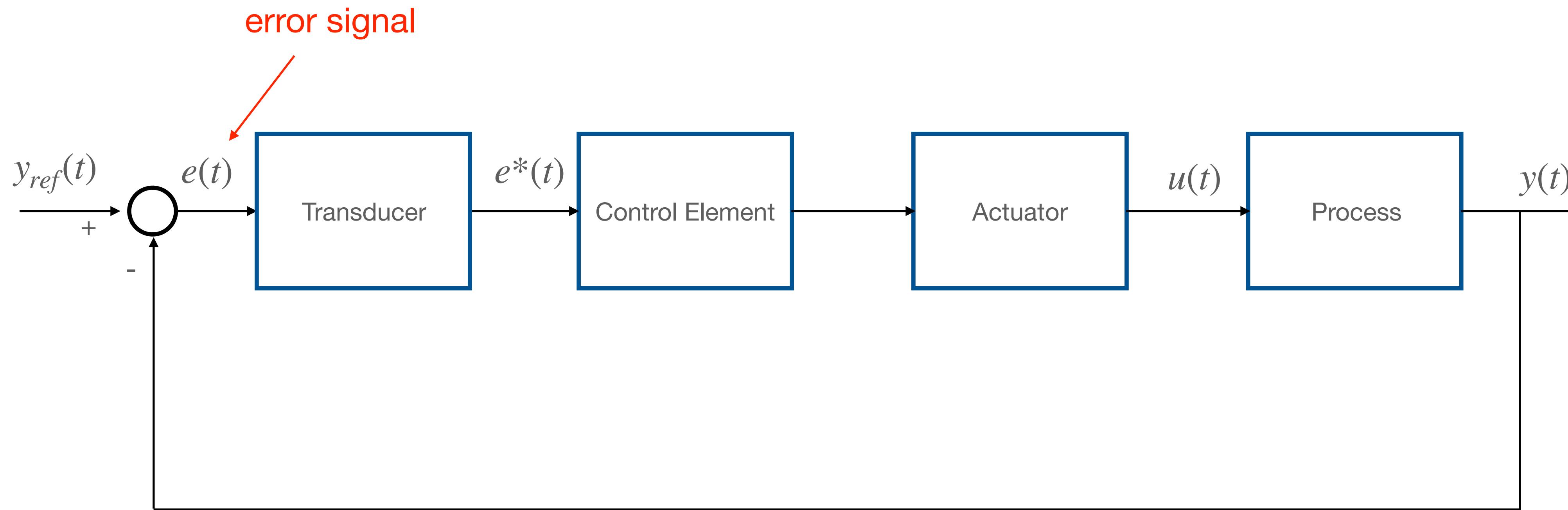
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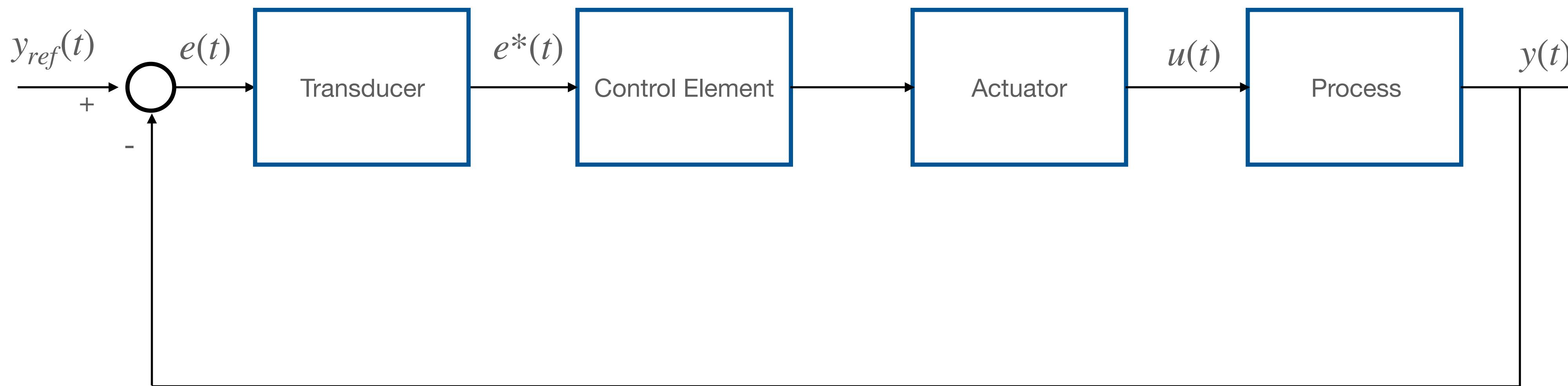
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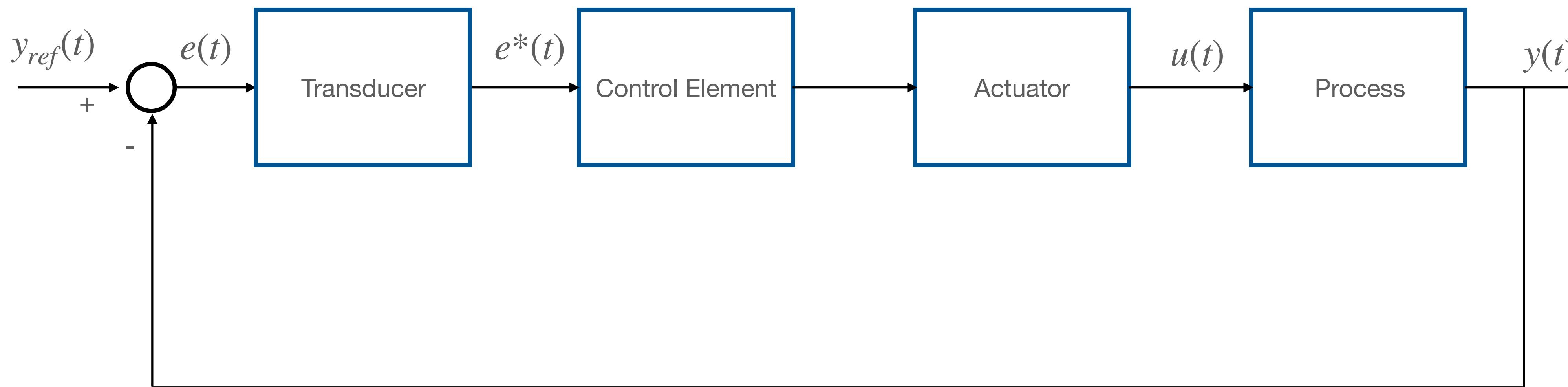
Basic Control Scheme



Assumption: Linear Time-Invariant (LTI) Dynamic Systems



Basic Control Scheme

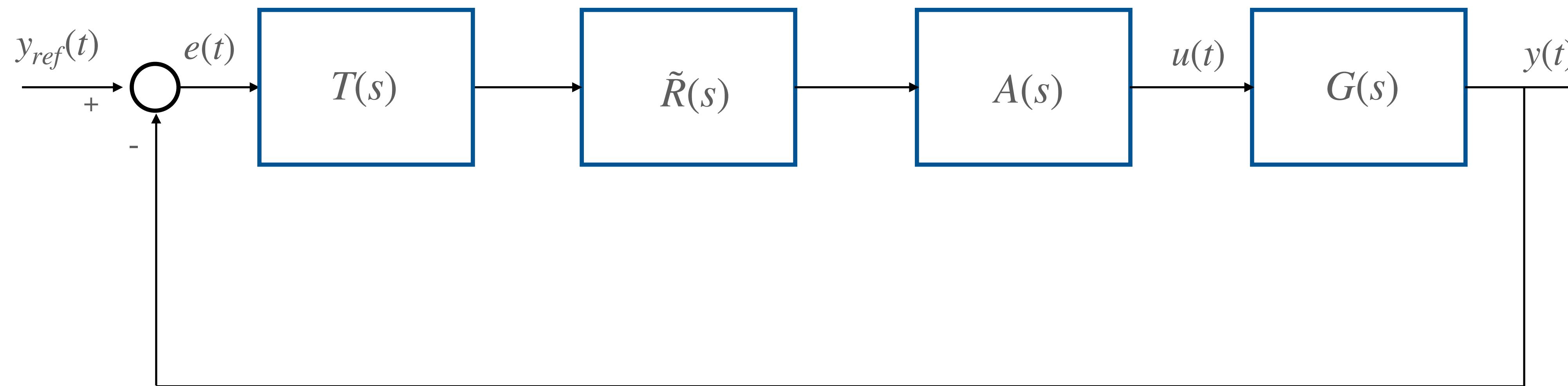


Assumption: Linear Time-Invariant (LTI) Dynamic Systems

They can be modelled via Transfer Functions



Basic Control Scheme

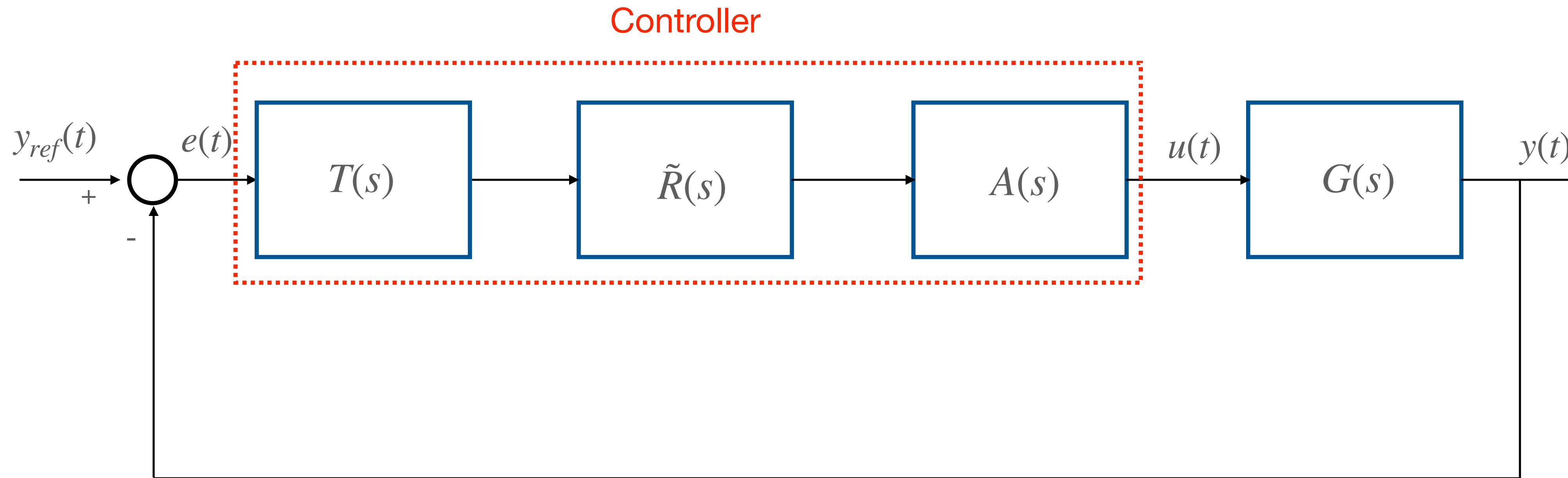


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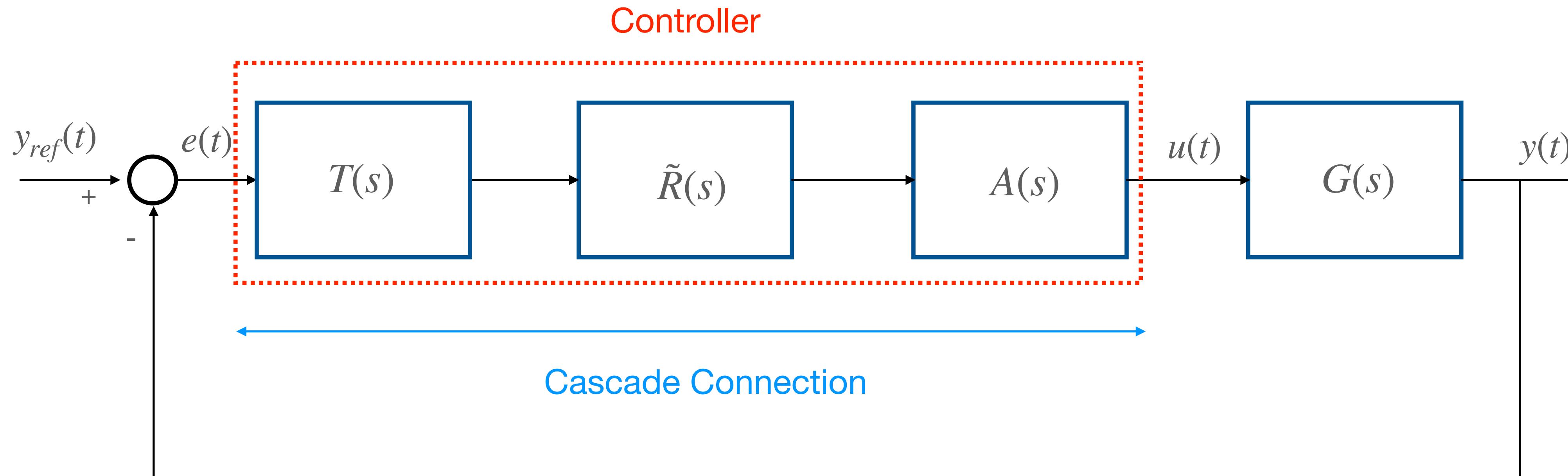
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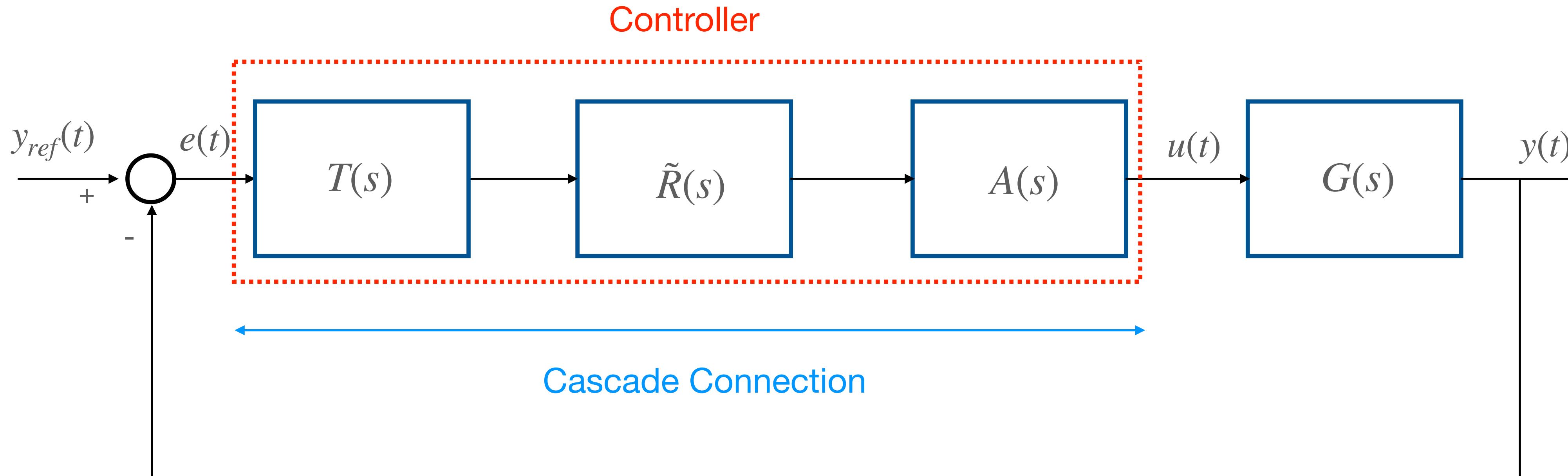
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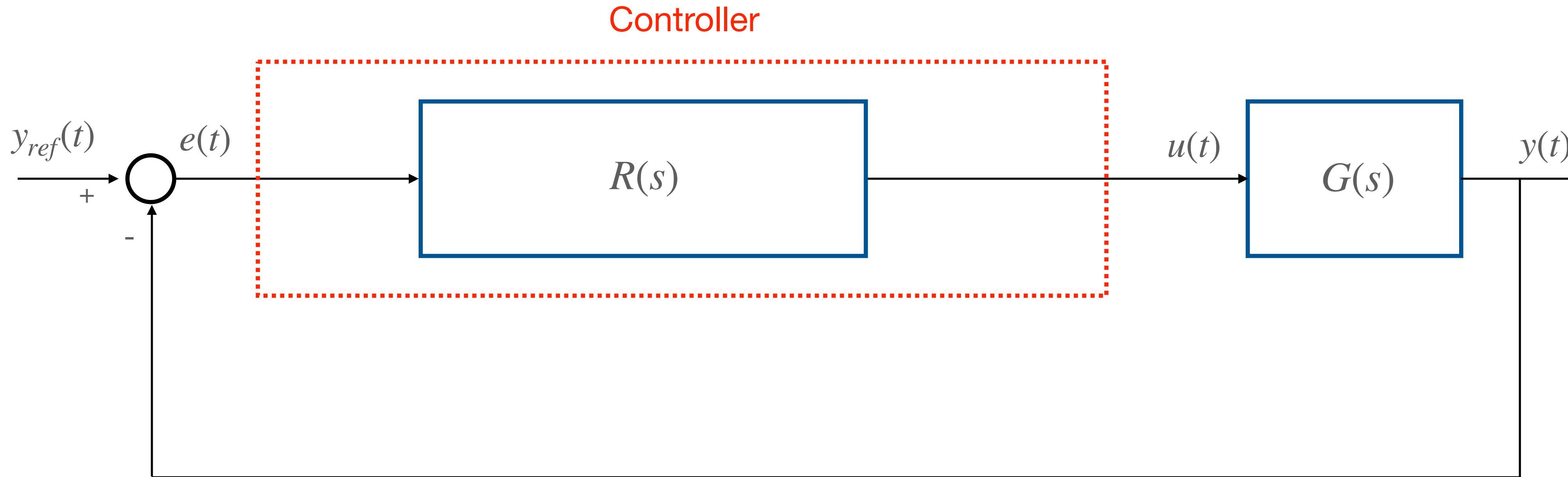


Basic Control Scheme



Block Algebra Rule: $T(s) \cdot \tilde{R}(s) \cdot A(s) = R(s)$

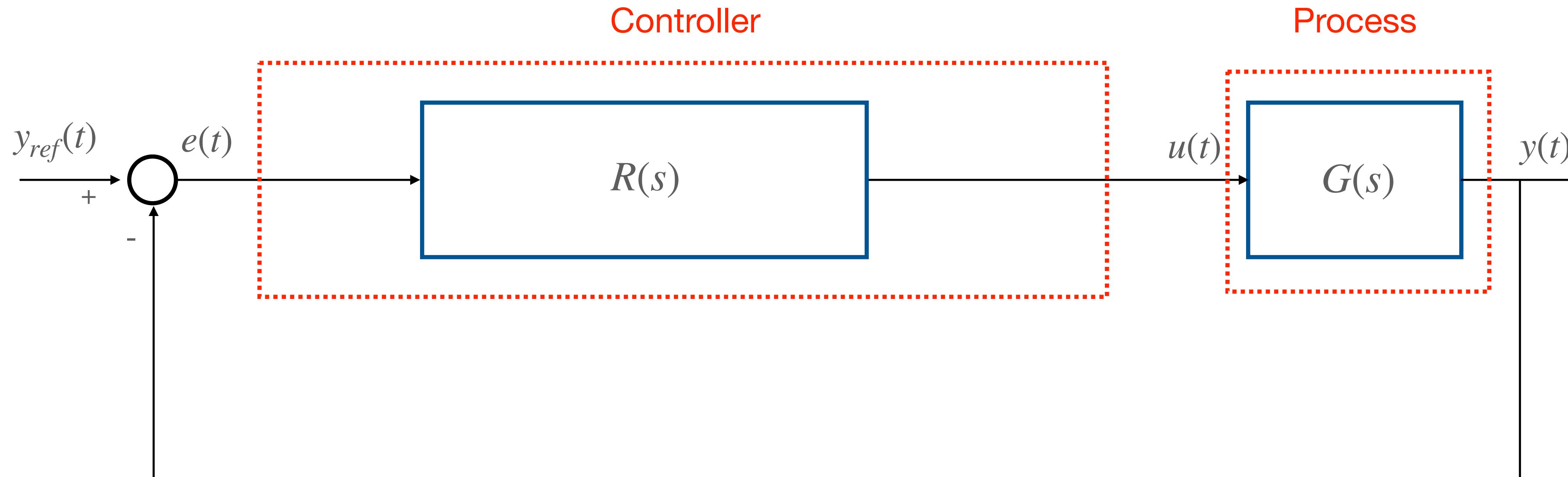
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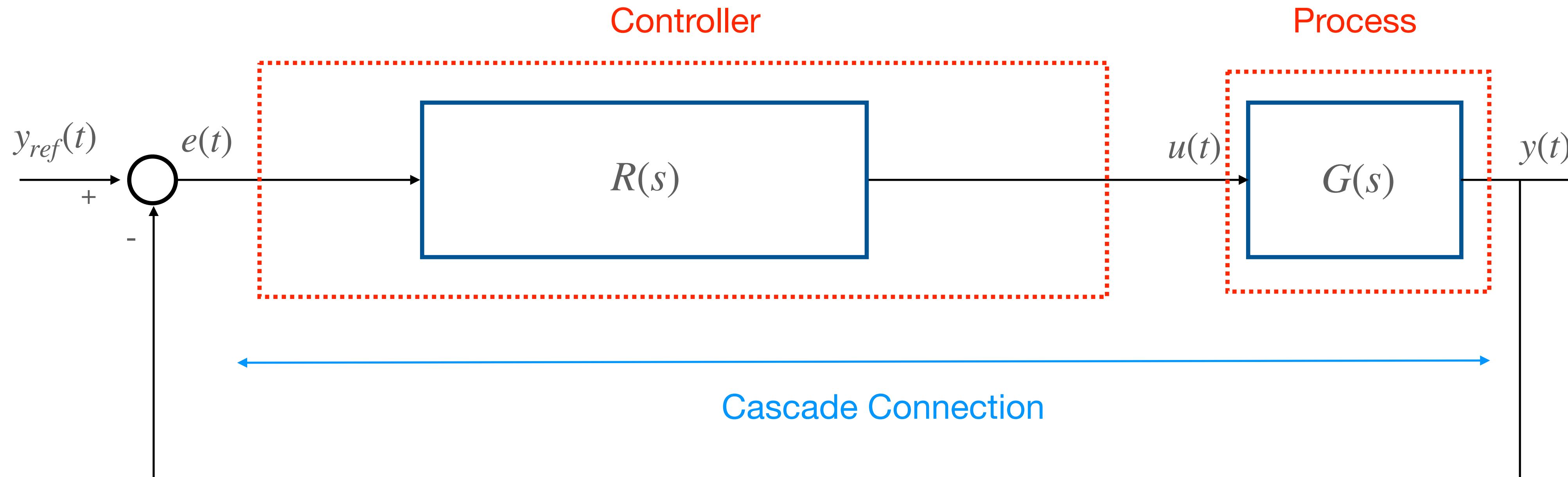
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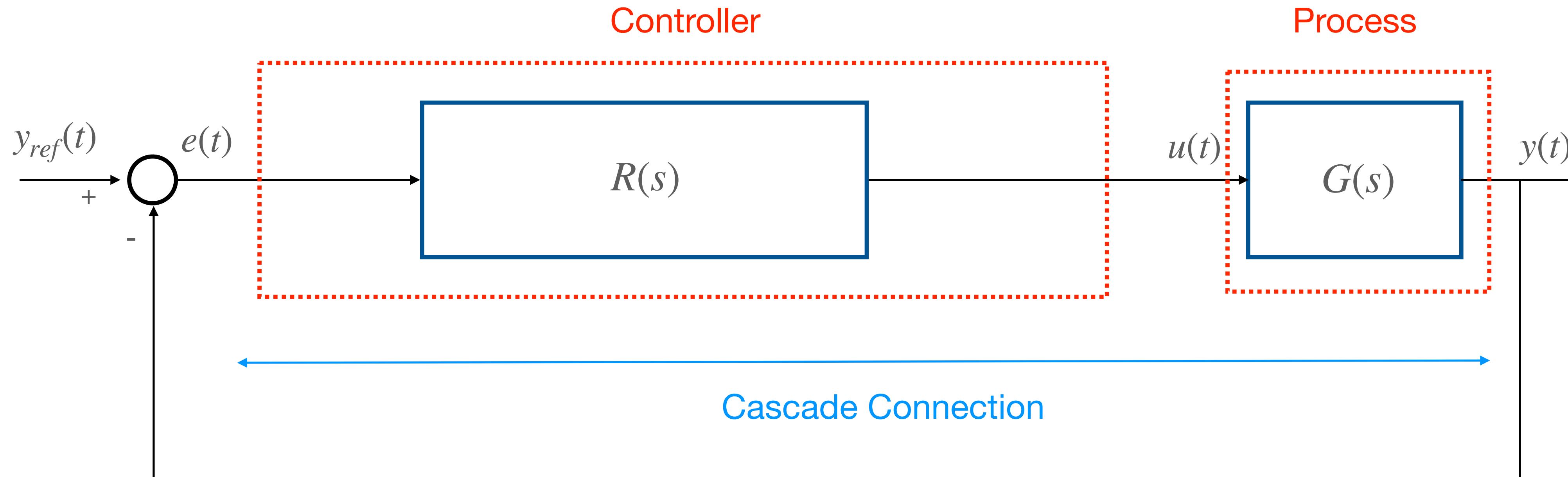
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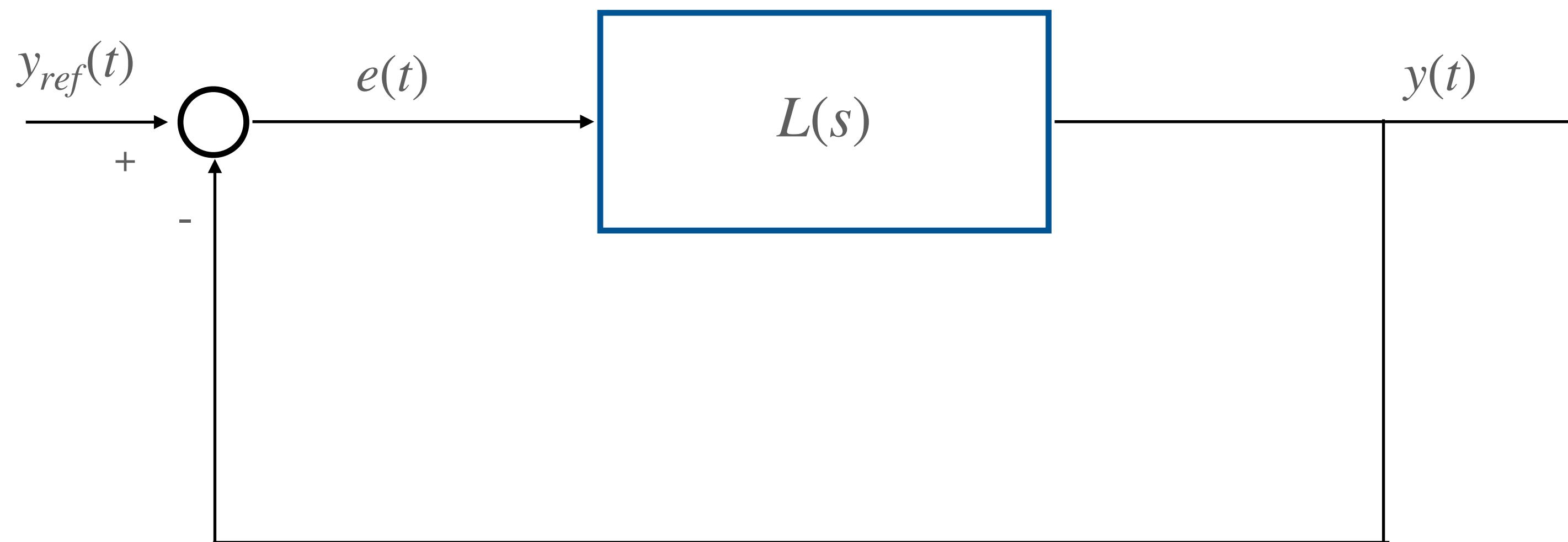


Basic Control Scheme



Block Algebra Rule: $R(s) \cdot G(s) = L(s)$

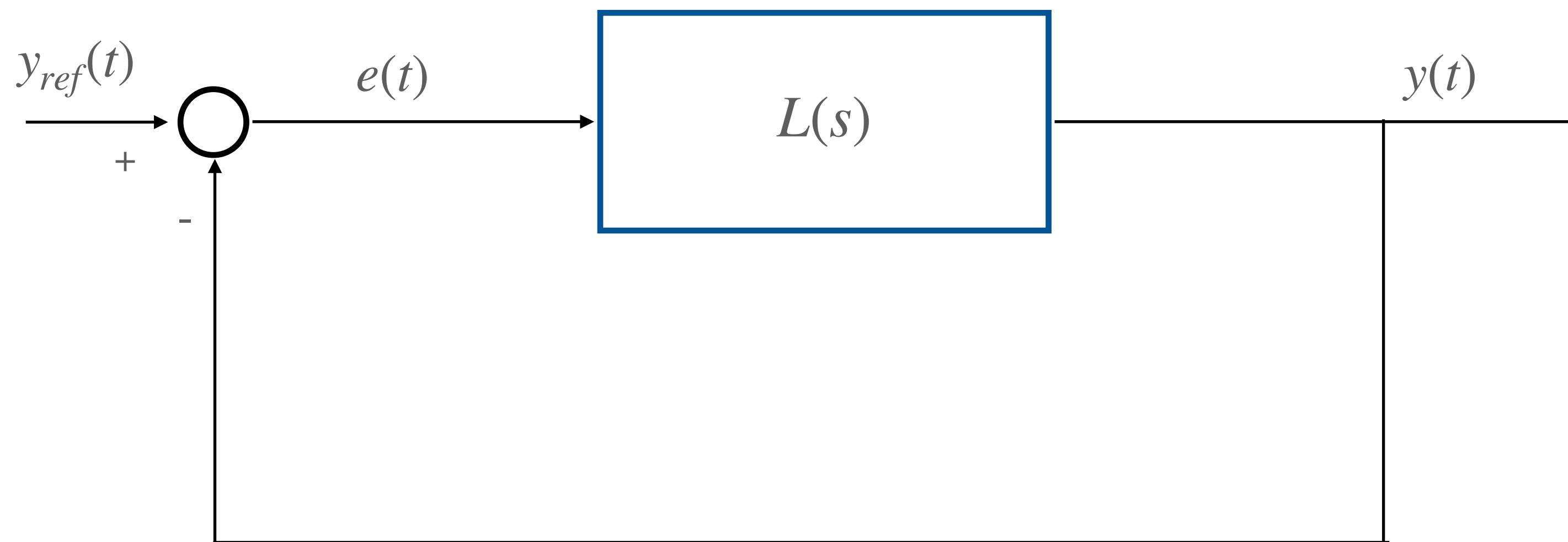
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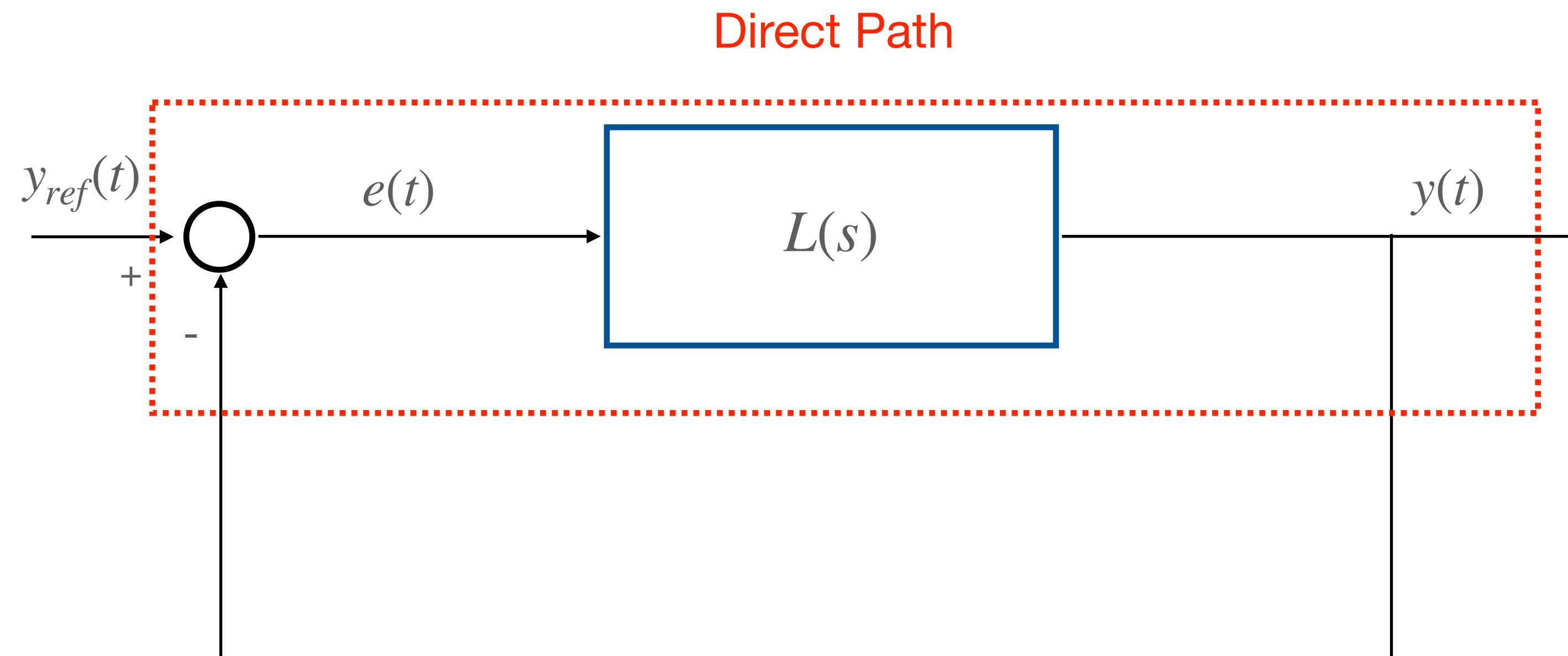


Unitary Feedback Control Scheme



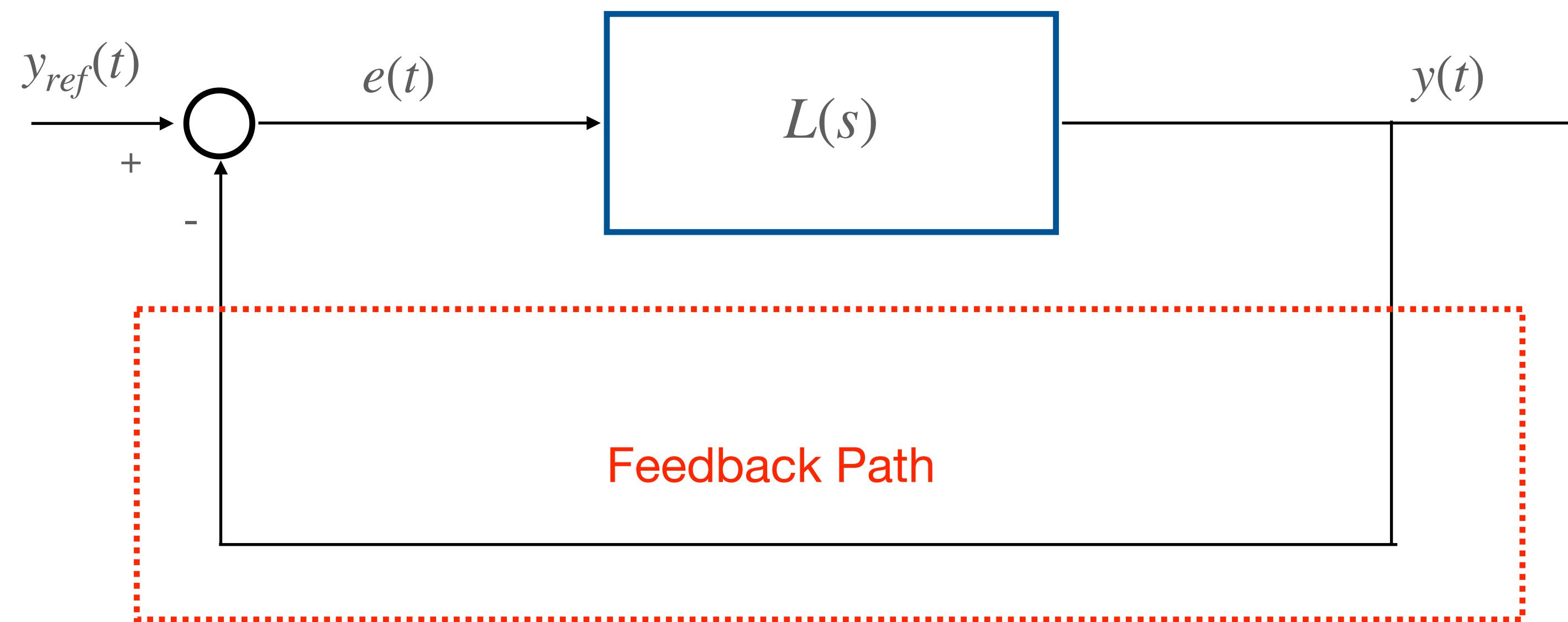
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Unitary Feedback Control Scheme



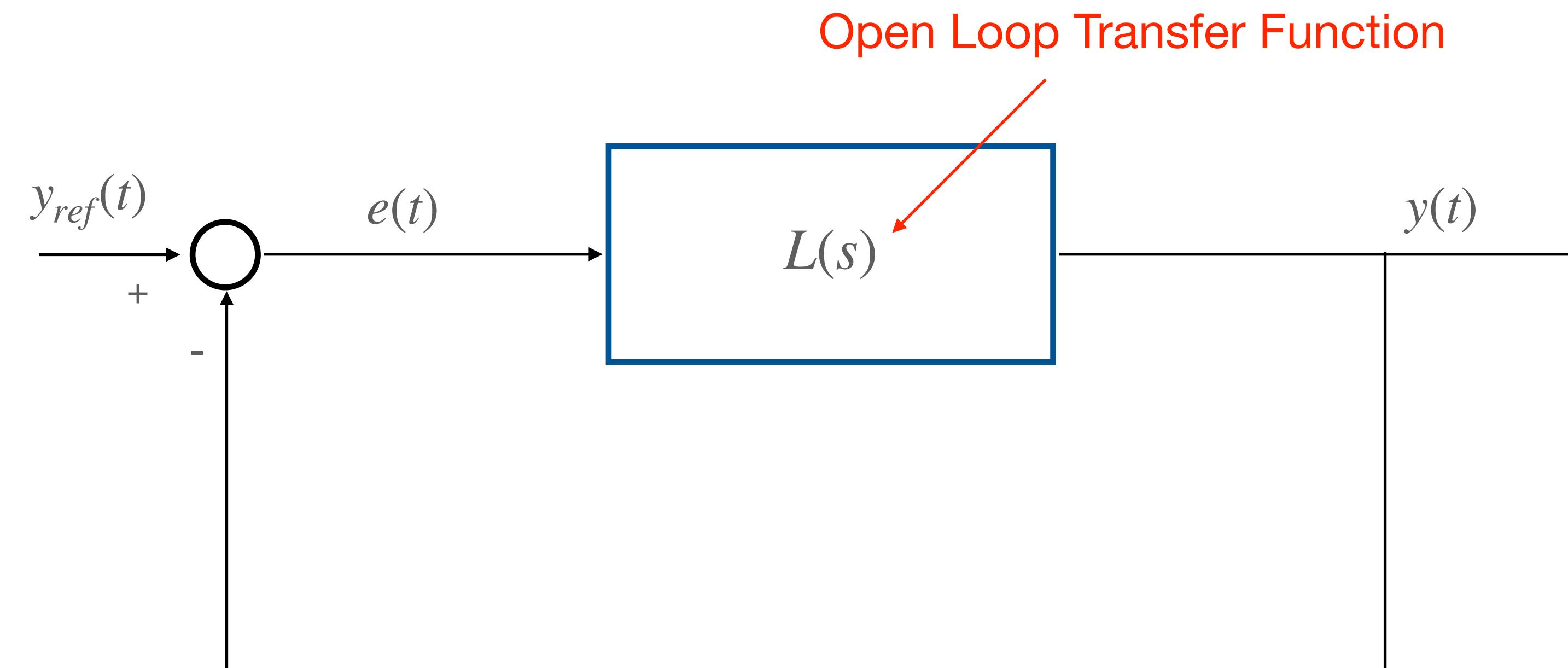
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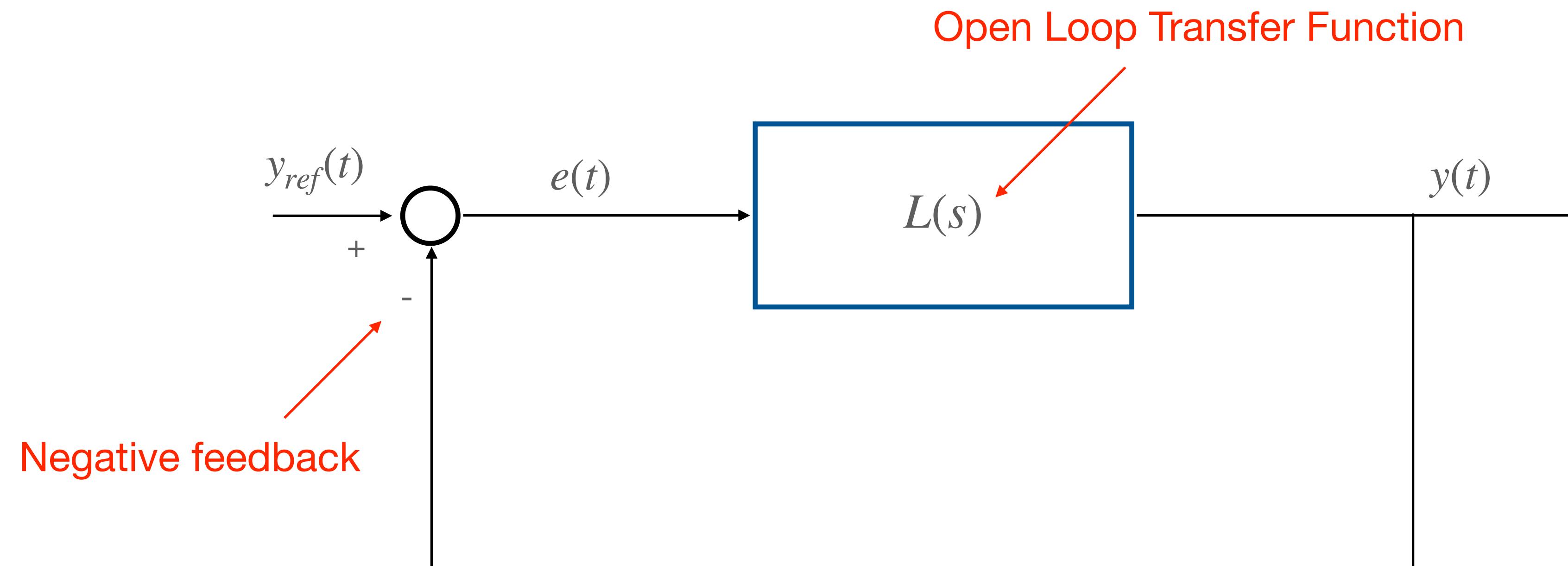
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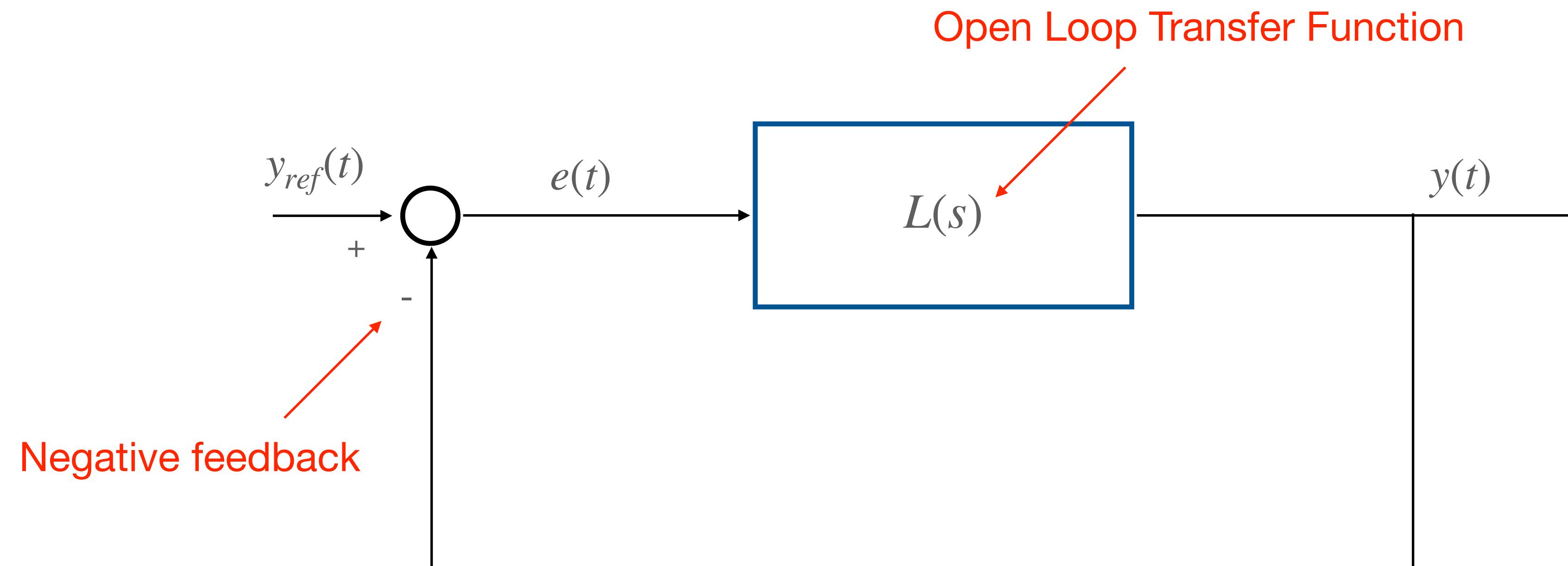
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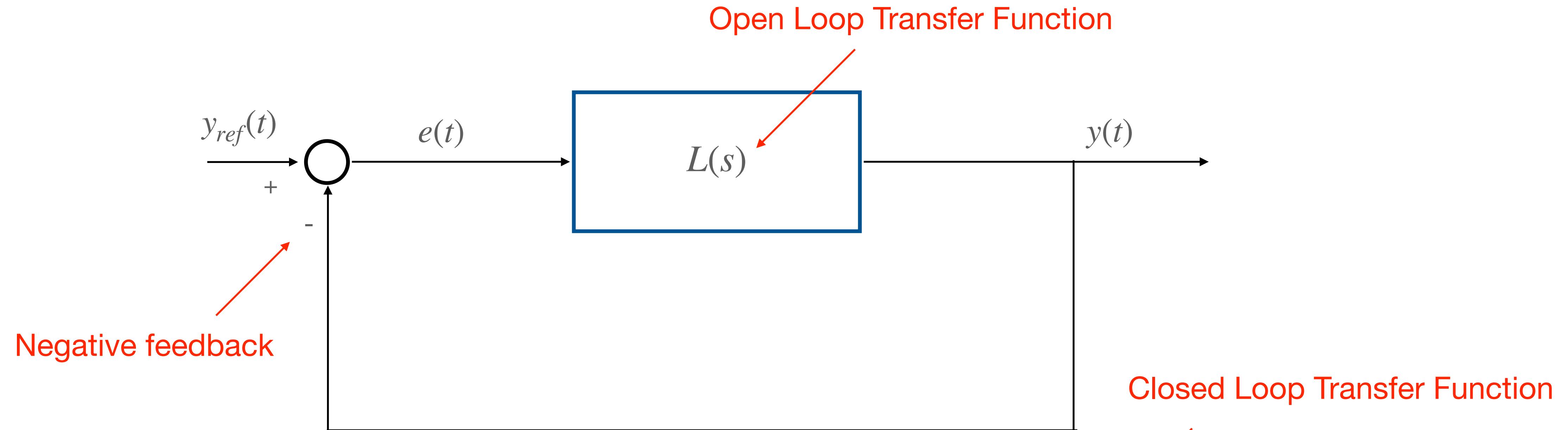


Block Algebra Rule:

$$F(s) = \frac{R(s) \cdot G(s)}{1 + R(s) \cdot G(s)} = \frac{L(s)}{1 + L(s)}$$



Unitary Feedback Control Scheme



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Limitations of the Basic Unitary Feedback Control Scheme

- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays
- No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
- Possible difficulties in enforcing steady-state precision with step reference signal via the classical approach



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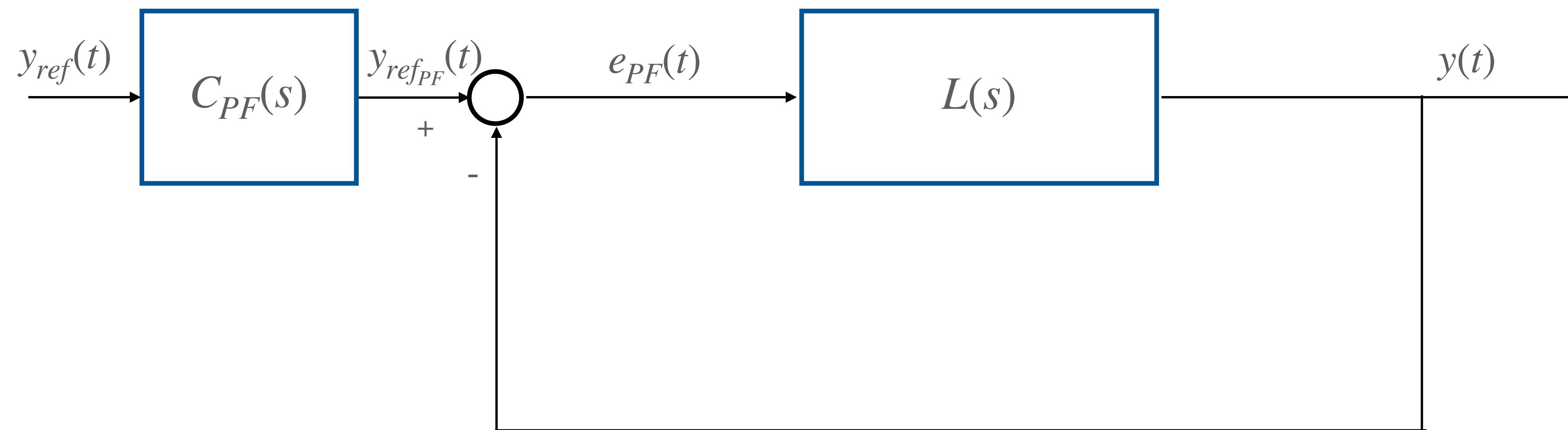
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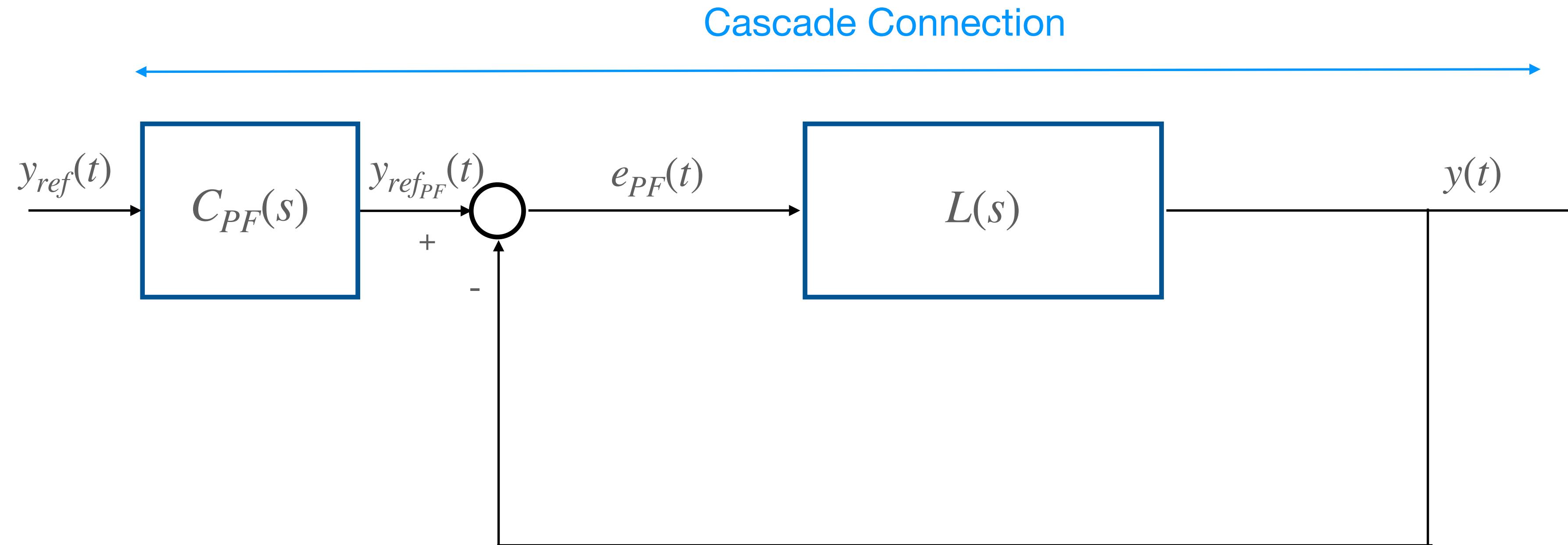
Possible solution: add a Pre-Filter to the basic unitary feedback control scheme



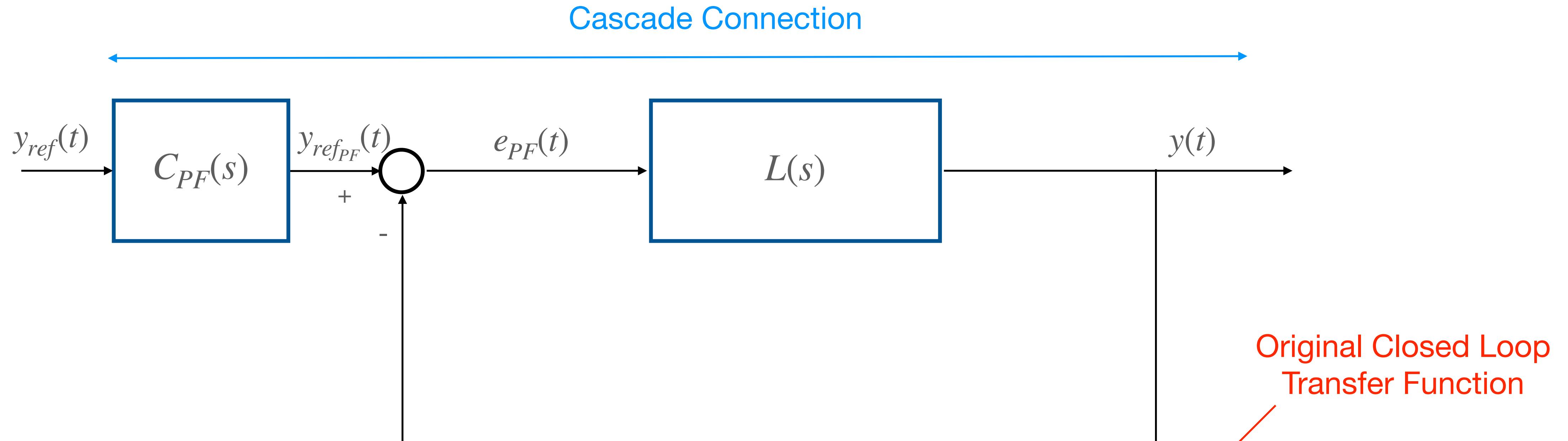
Pre-filter Based Control Scheme



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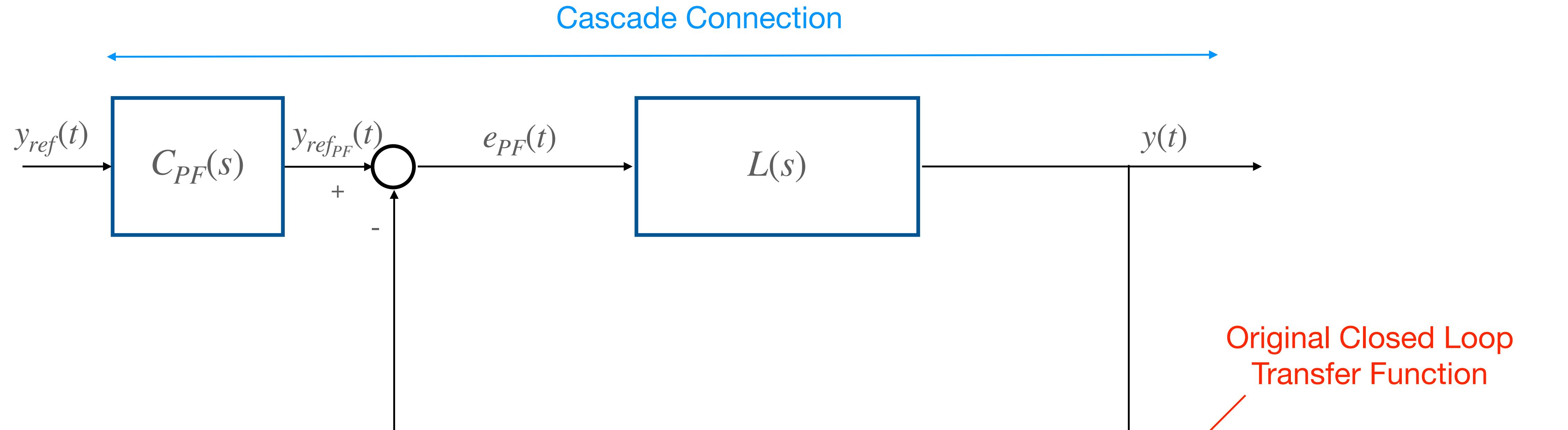


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Pre-filter Based Control Scheme



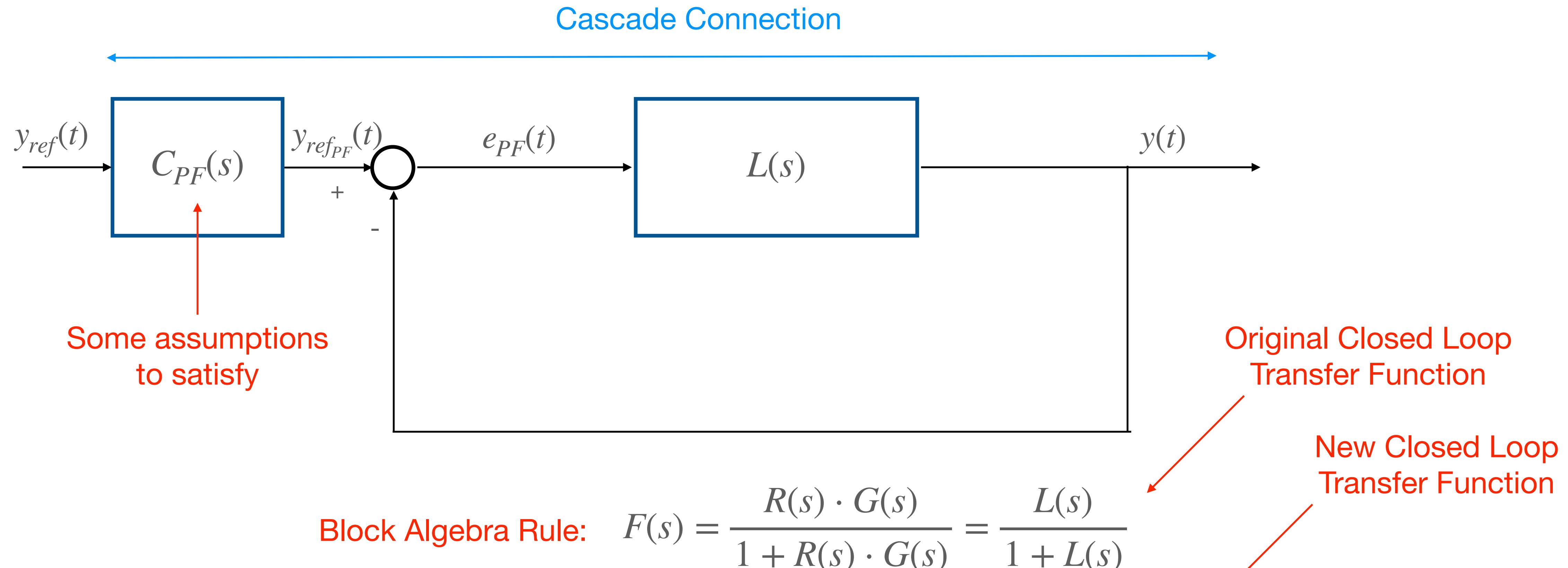
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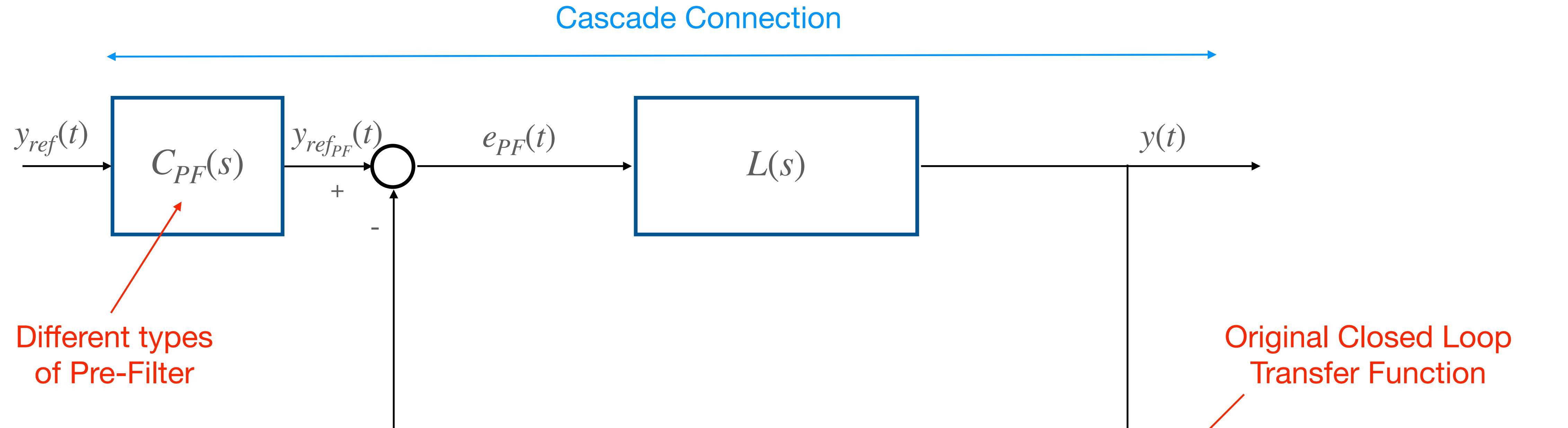
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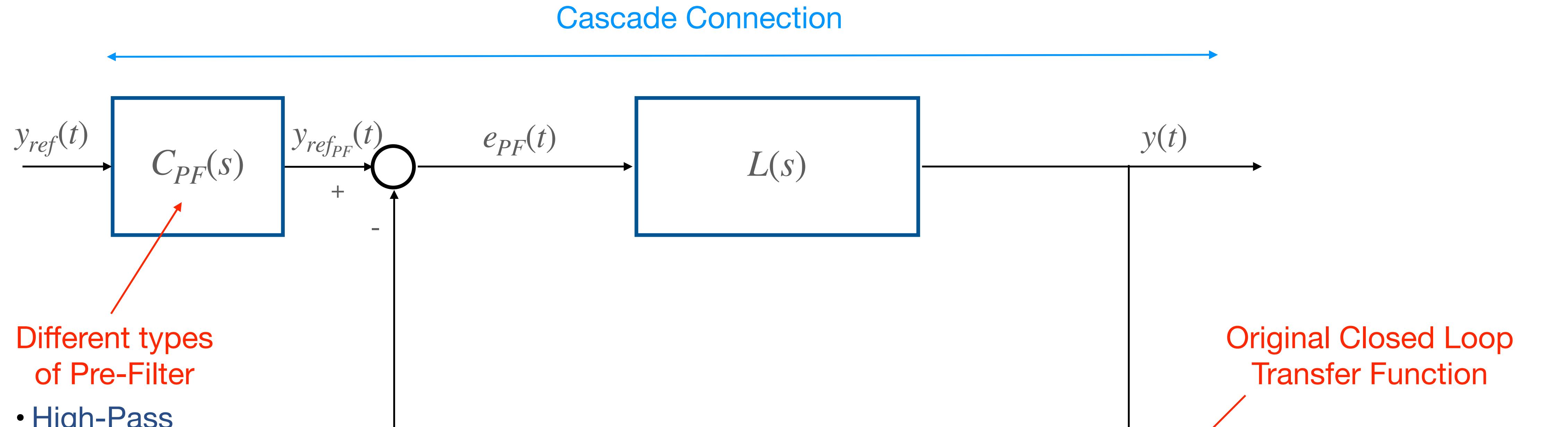
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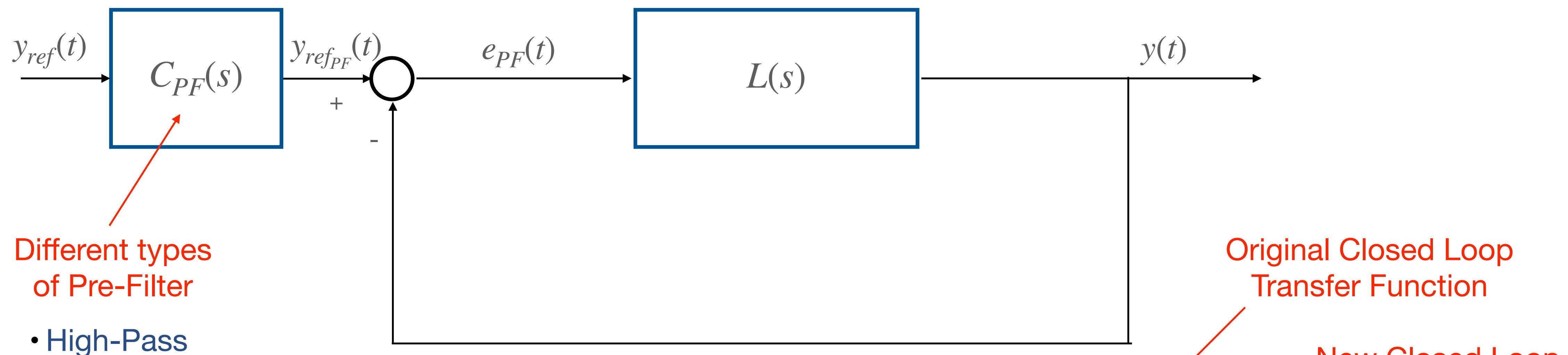
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Pre-filter Based Control Scheme: Case 1



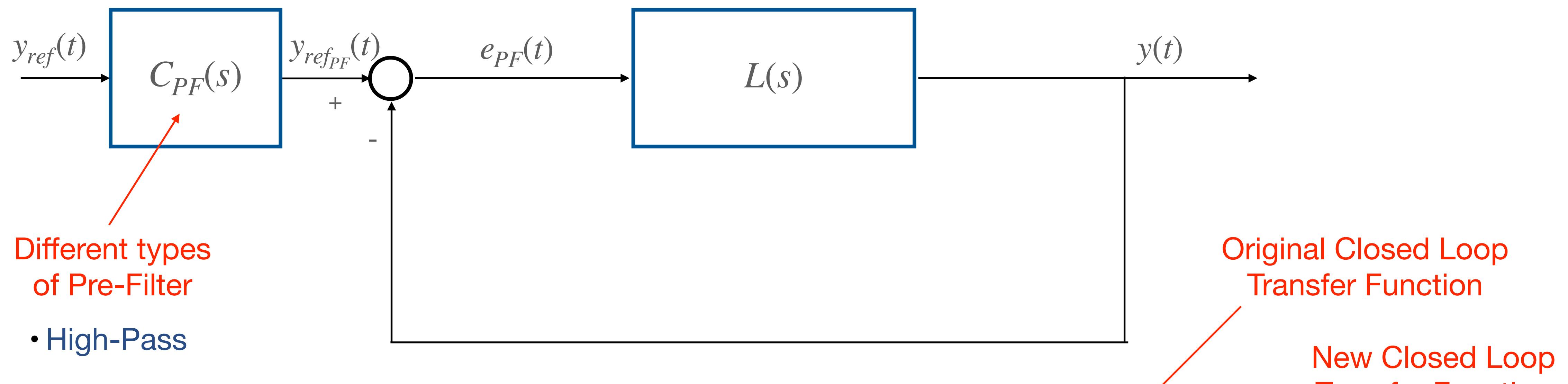
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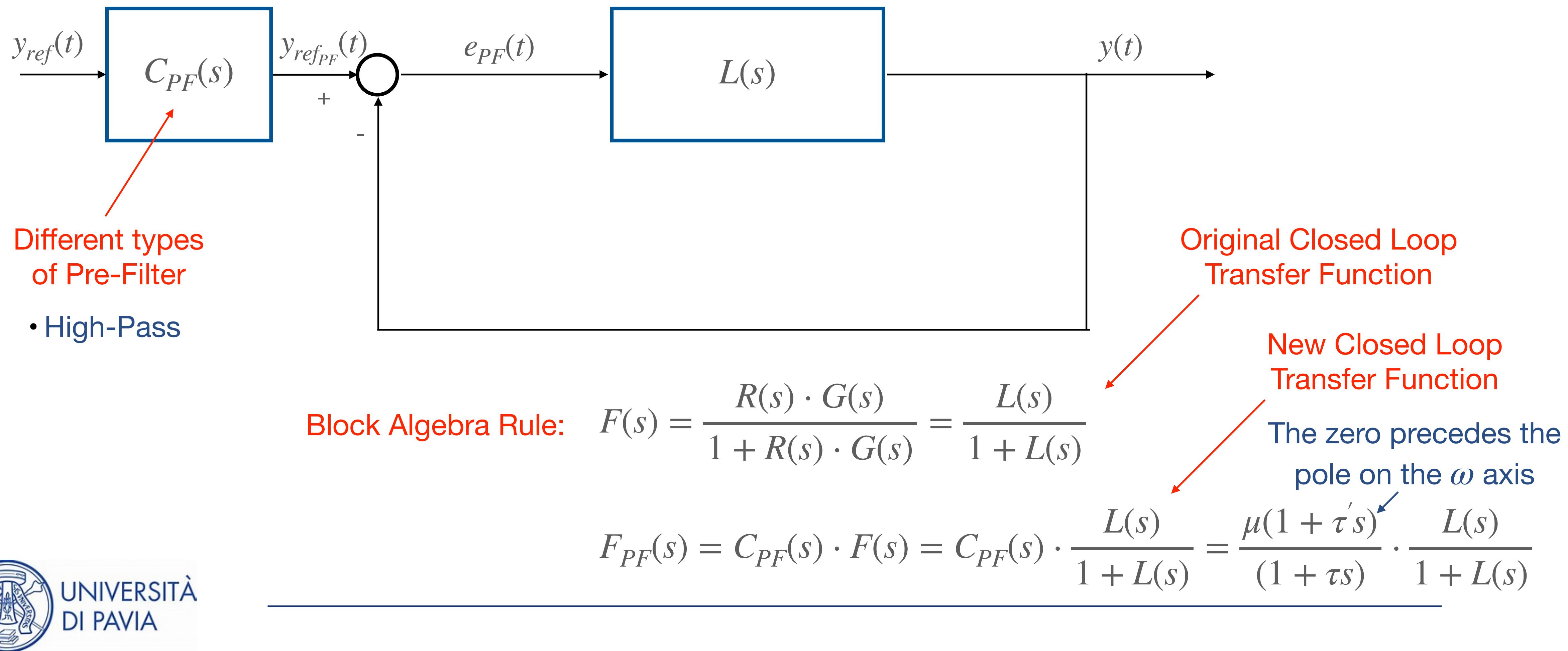
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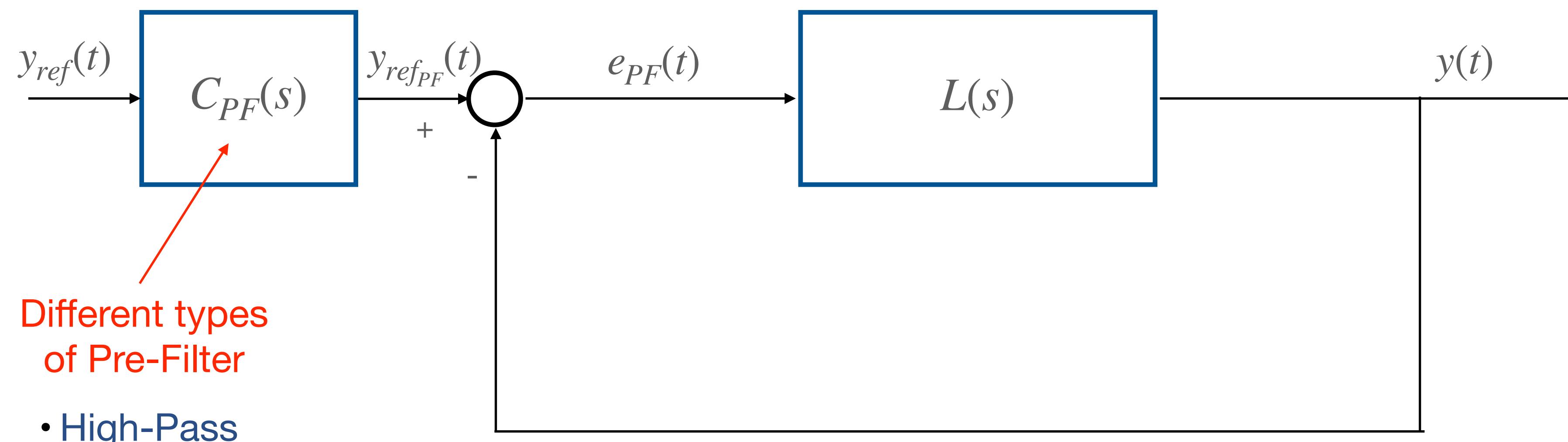
$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



Pre-filter Based Control Scheme: Case 1



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Assumptions:

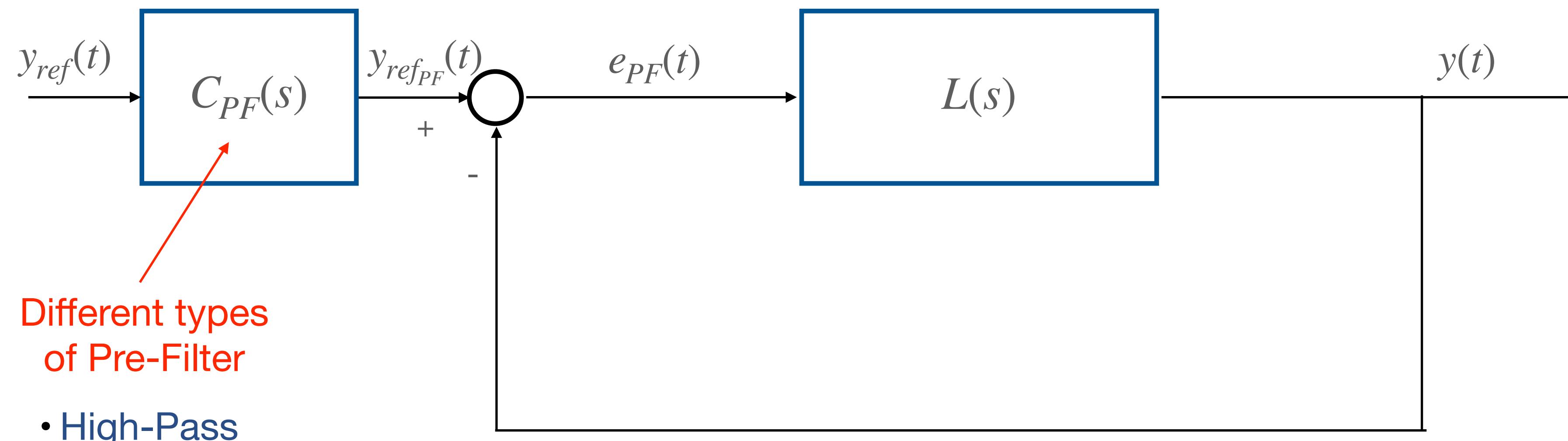
- $C_{PF}(s)$ As. Stable
- Proper
- Unitary gain

The zero precedes the pole on the ω axis

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Pre-filter Based Control Scheme: Case 1

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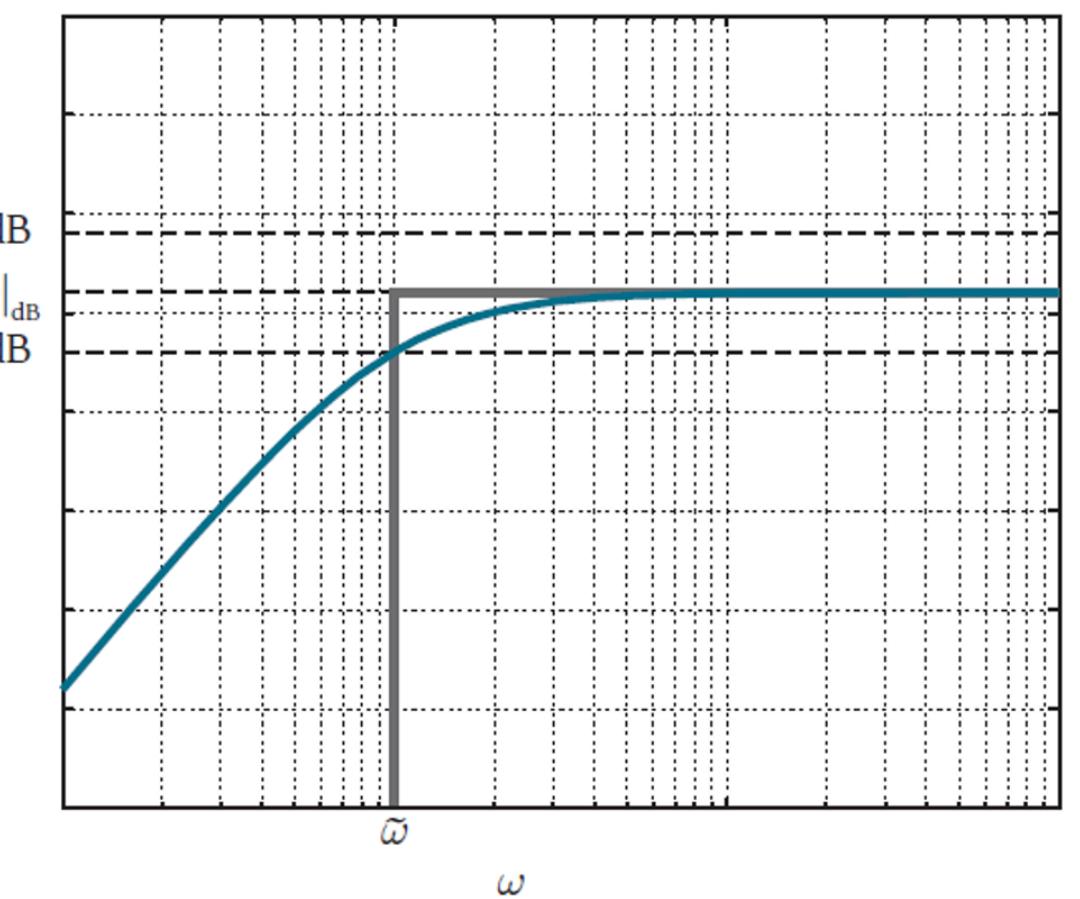
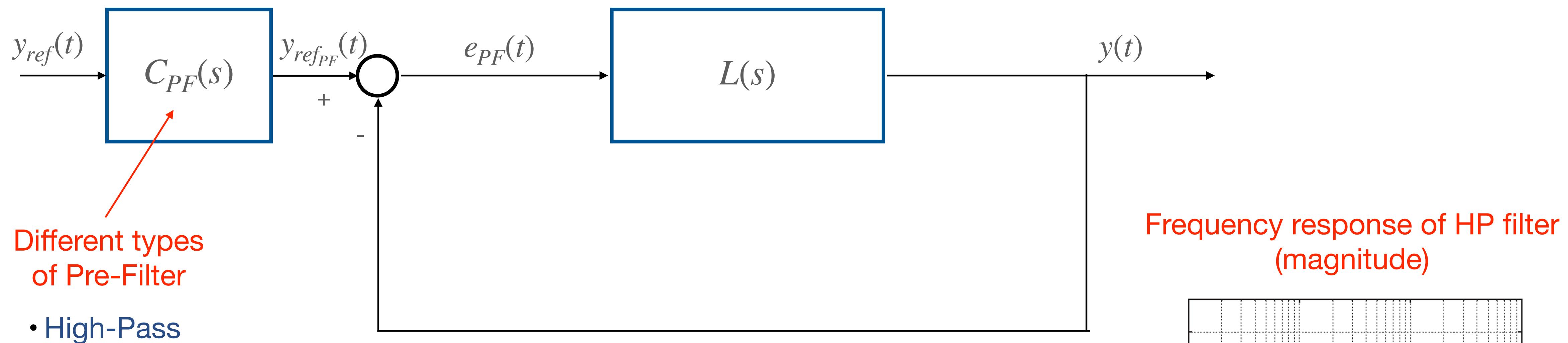
$$\lim_{s \rightarrow 0} \frac{\mu(1 + \tau' s)}{1 + \tau s} = \mu = 1$$

The zero precedes the pole on the ω axis

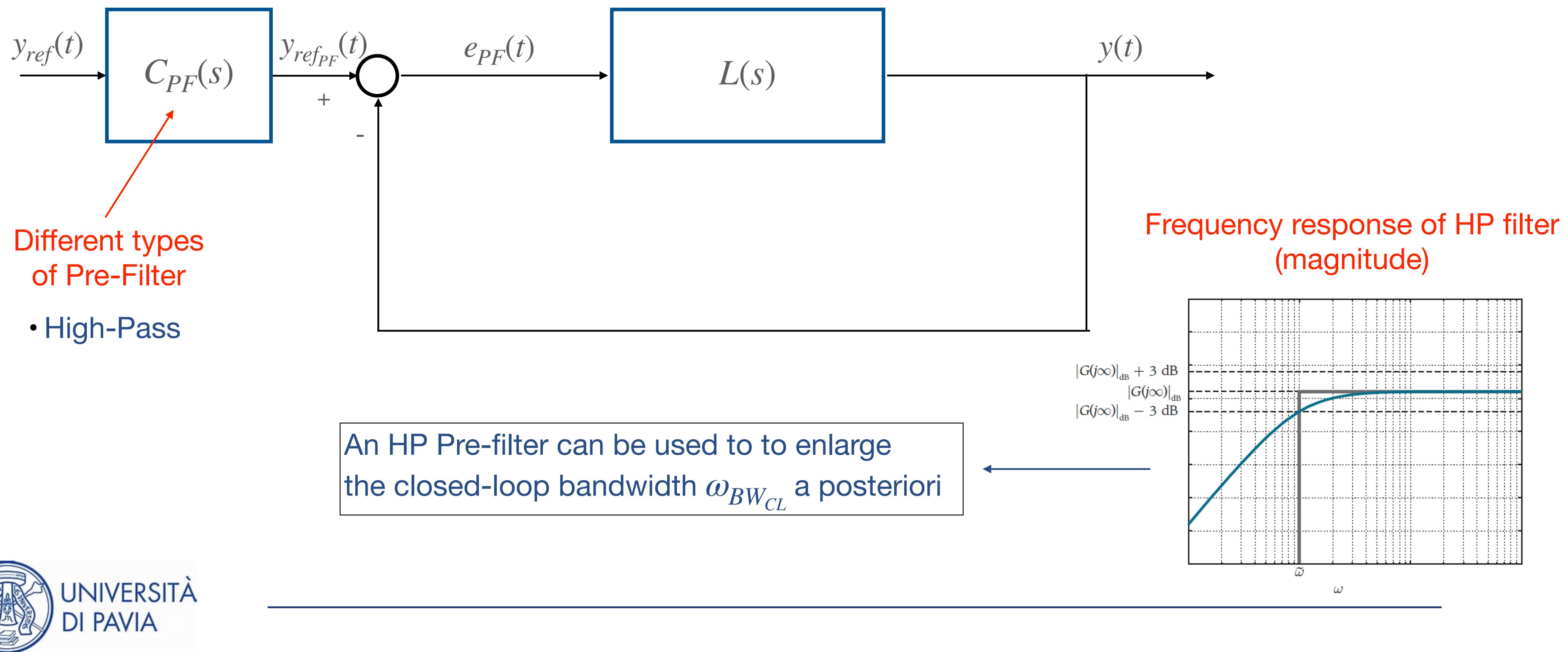
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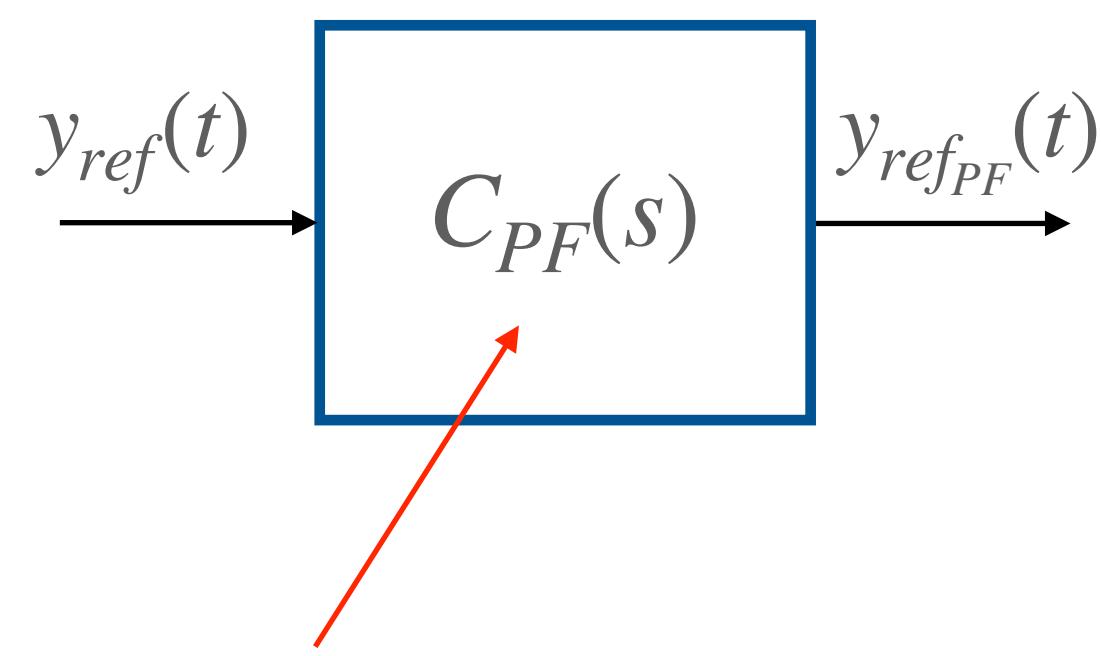
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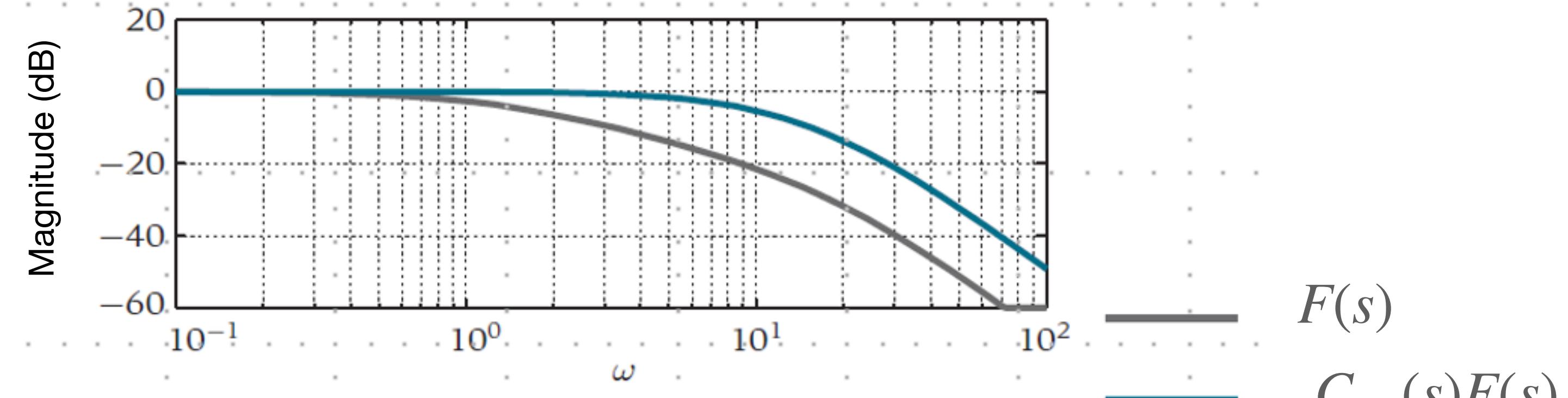
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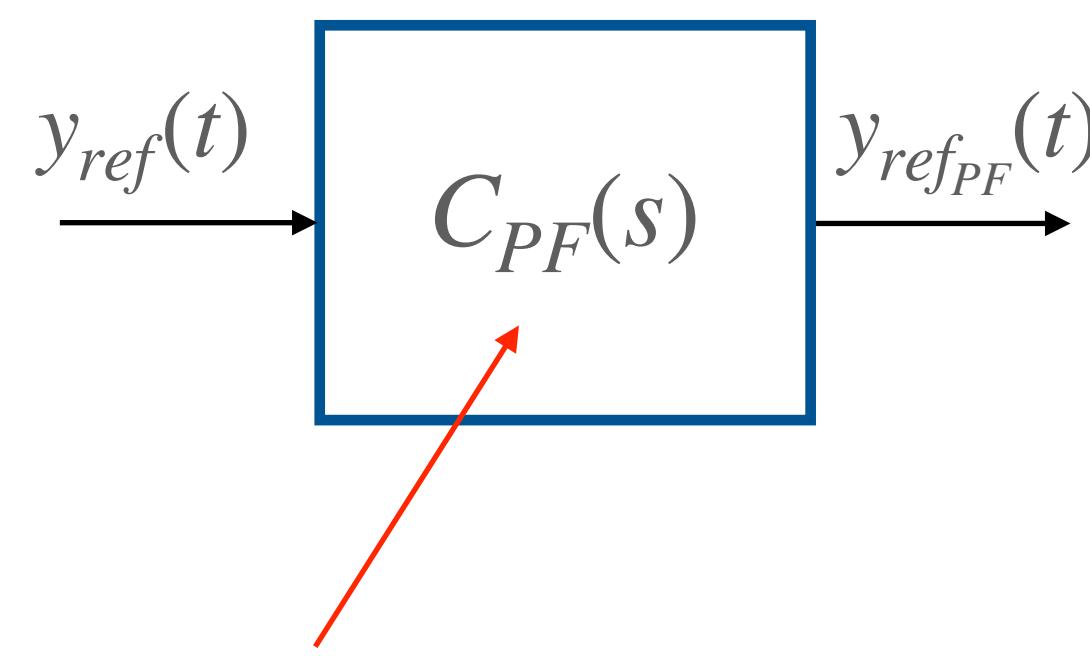
Different types
of Pre-Filter

- High-Pass

Frequency domain



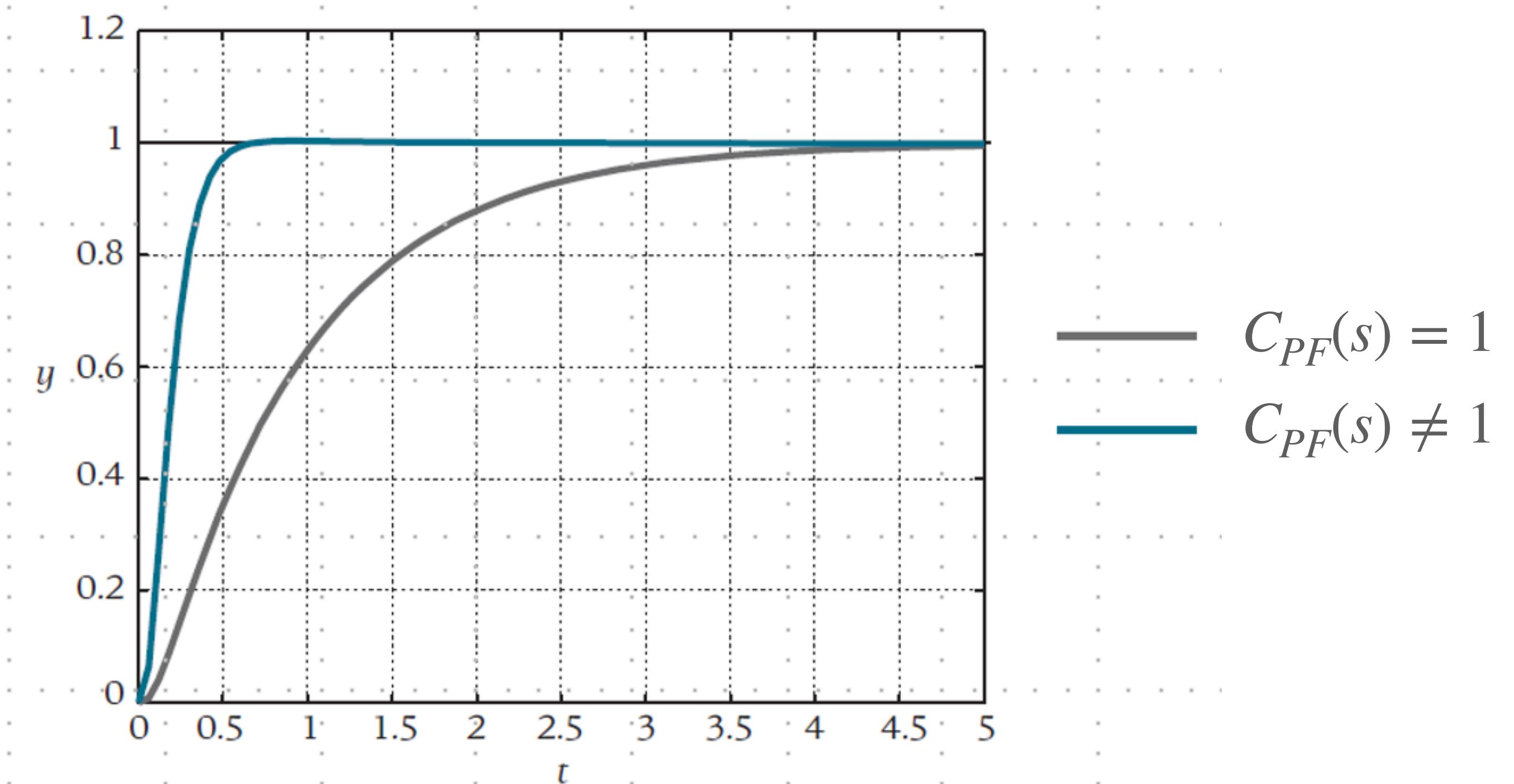
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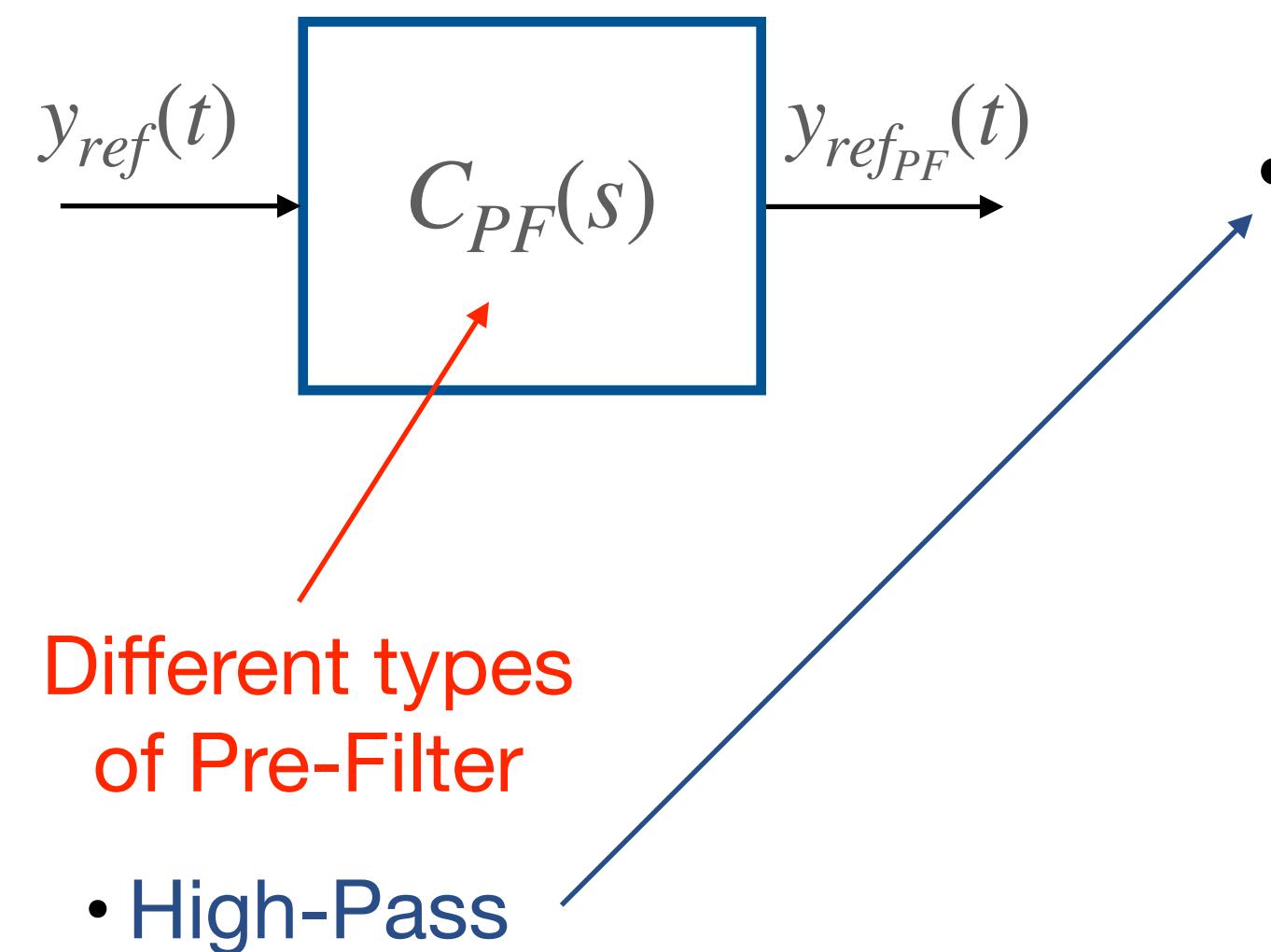
Different types
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Time domain



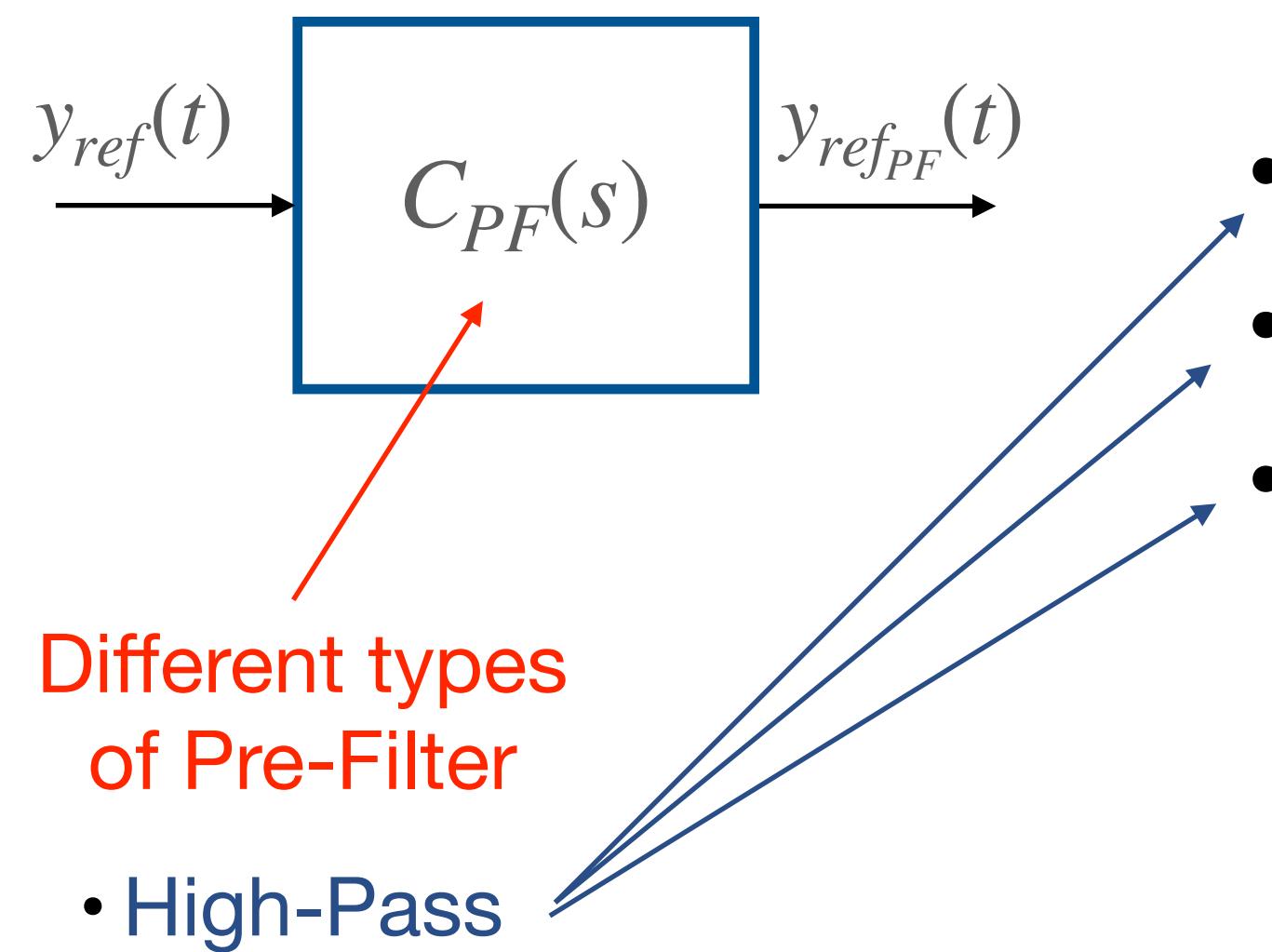
Pre-filter Based Control Scheme: Case 1



- Limited bandwidth in case of measurement disturbances

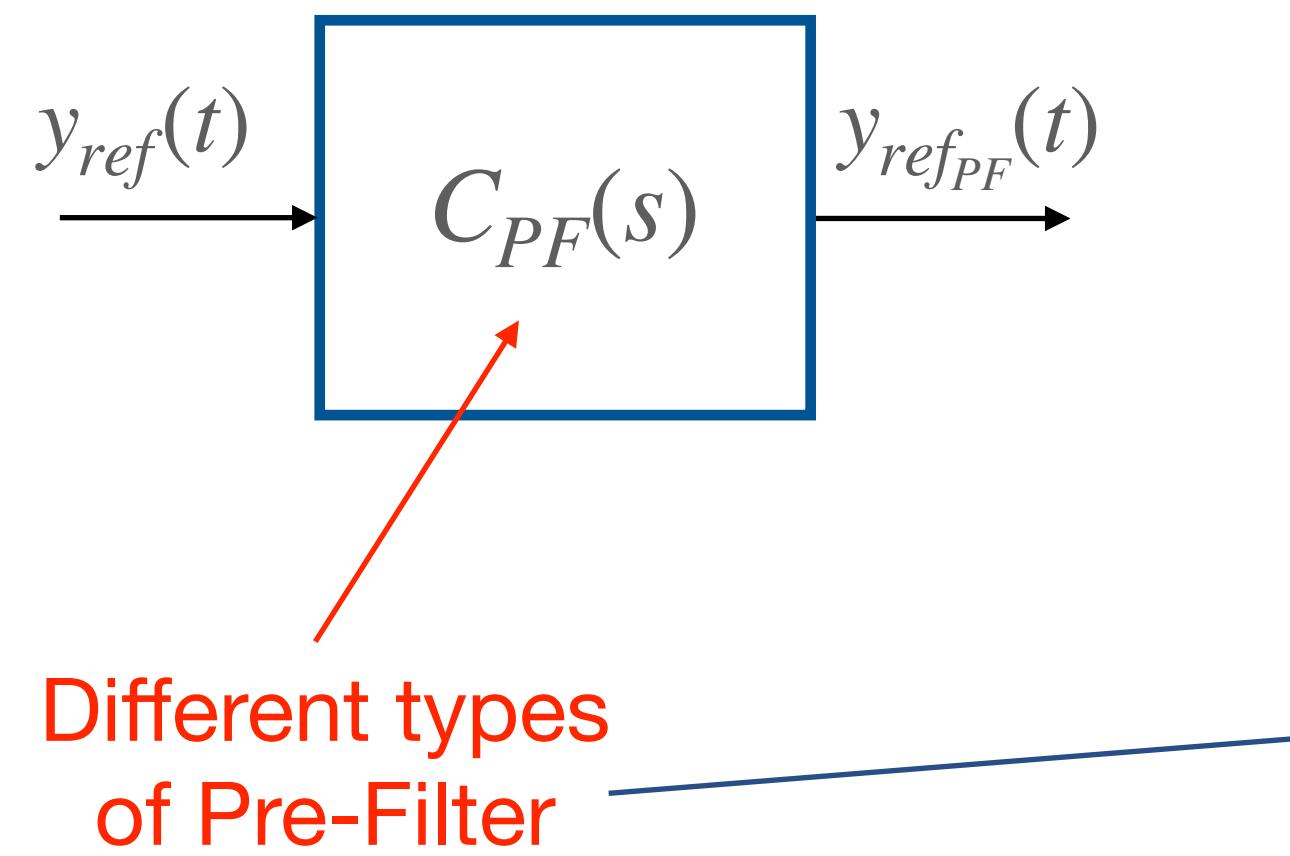


Pre-filter Based Control Scheme: Case 1



- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays

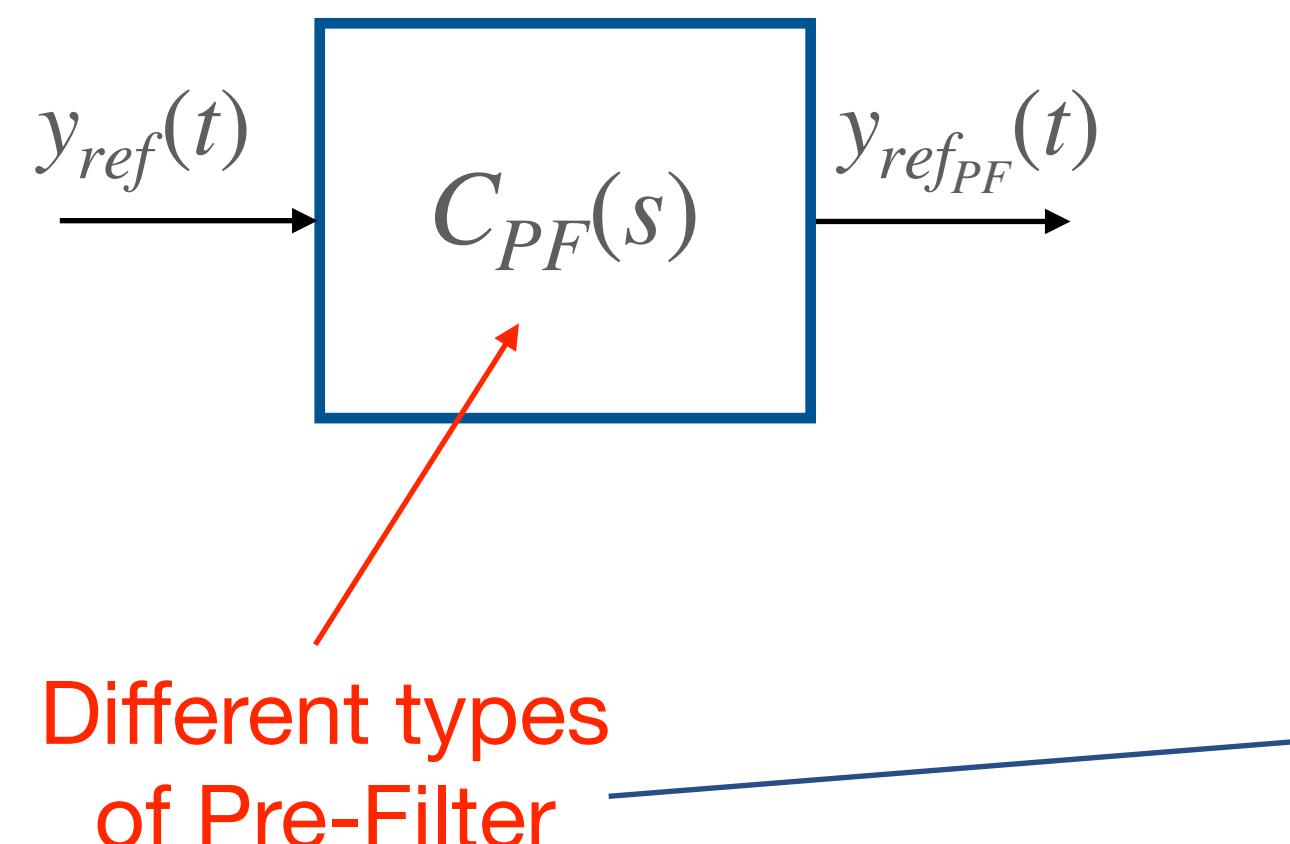
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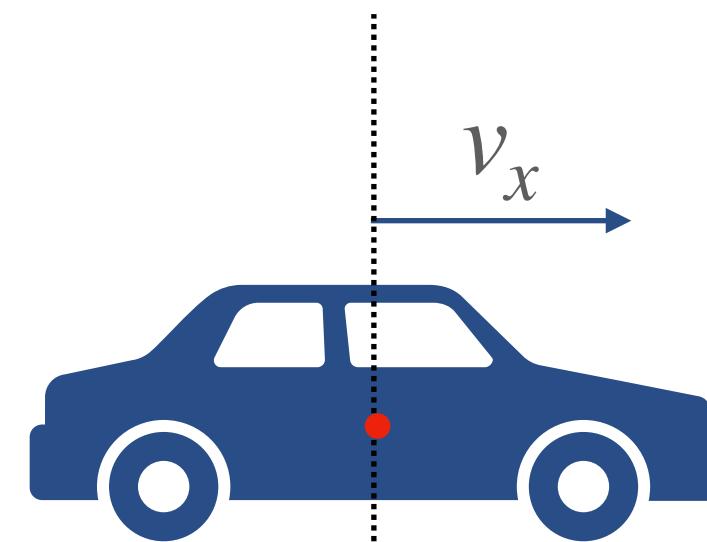


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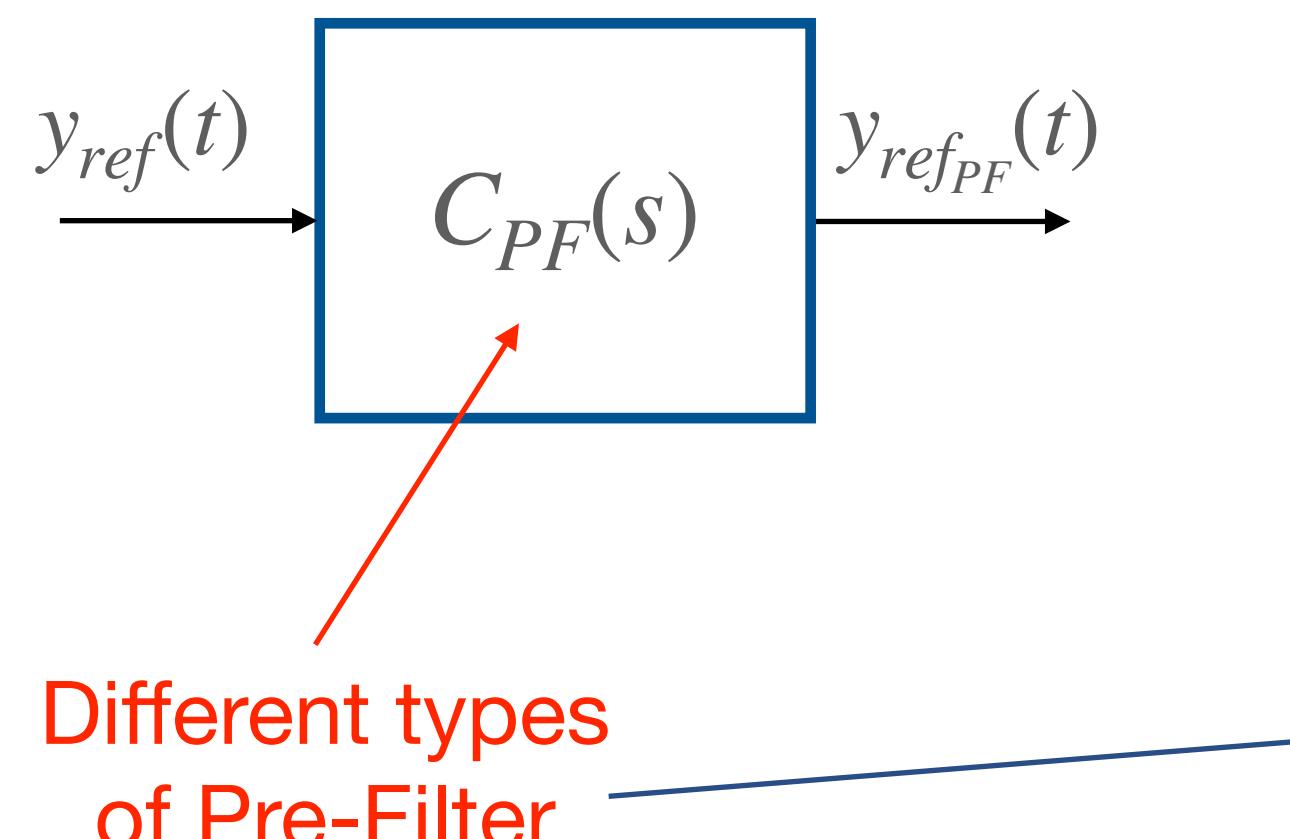


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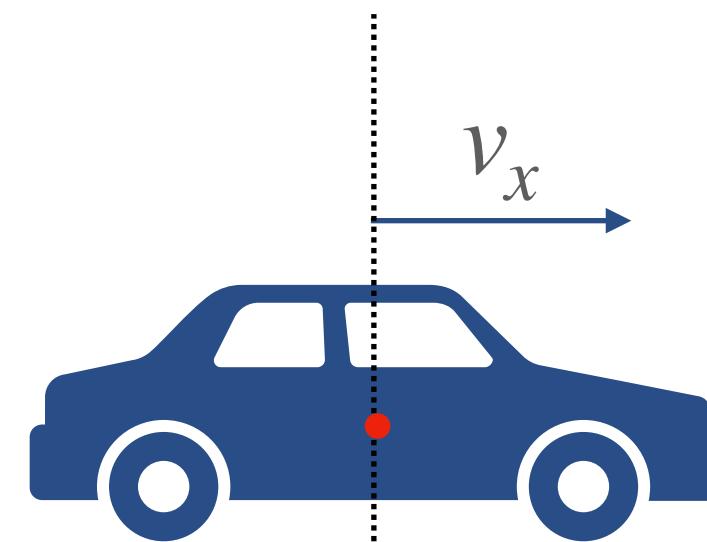


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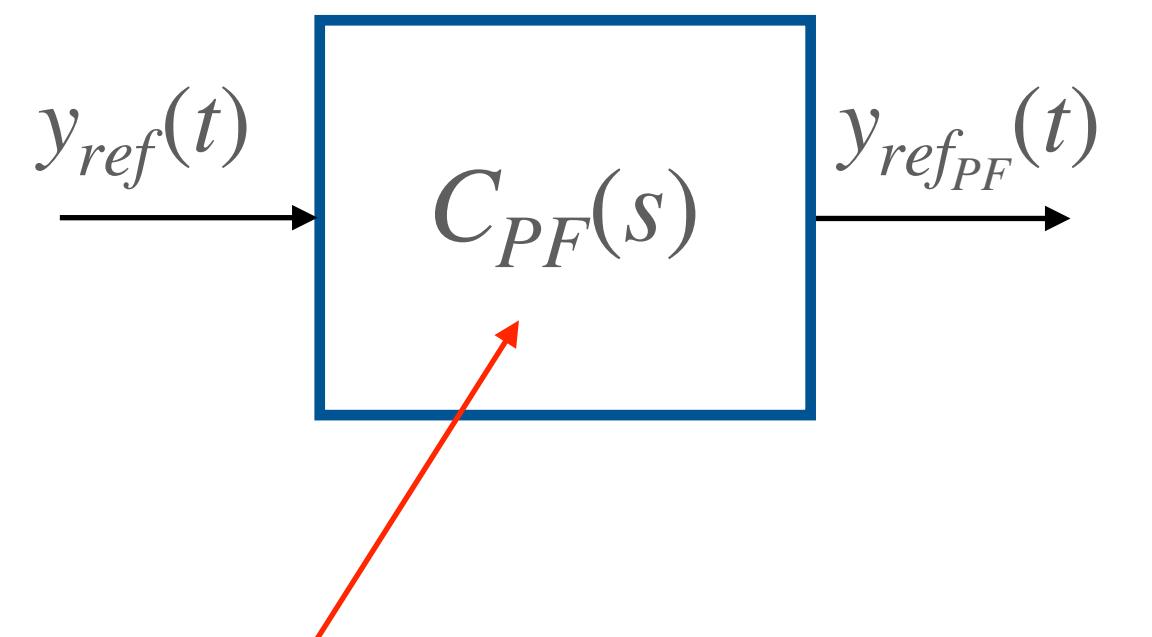
Example:



$$\lambda := \frac{\omega r - v_x}{\max(\omega r, v_x)}$$



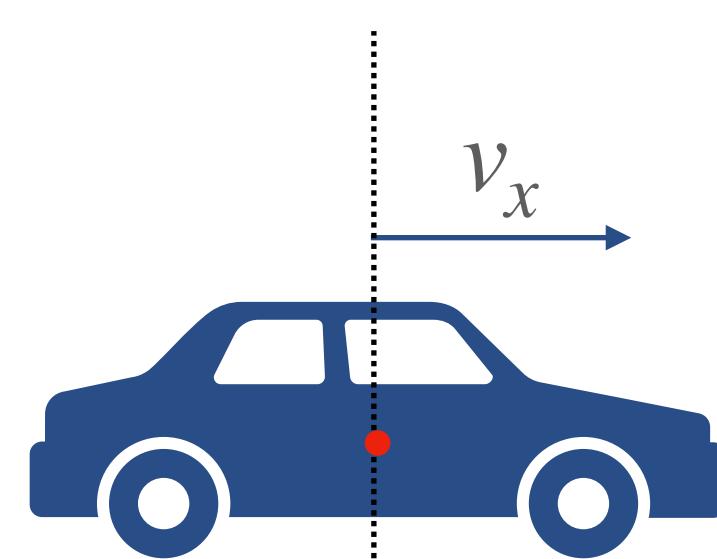
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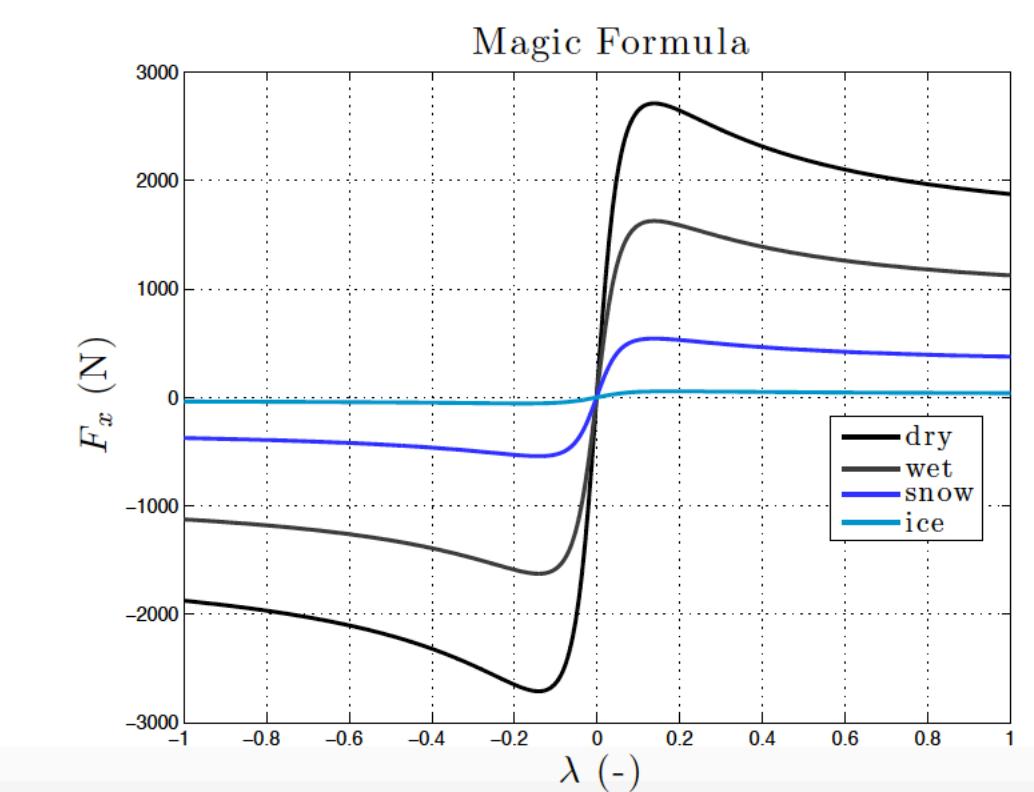
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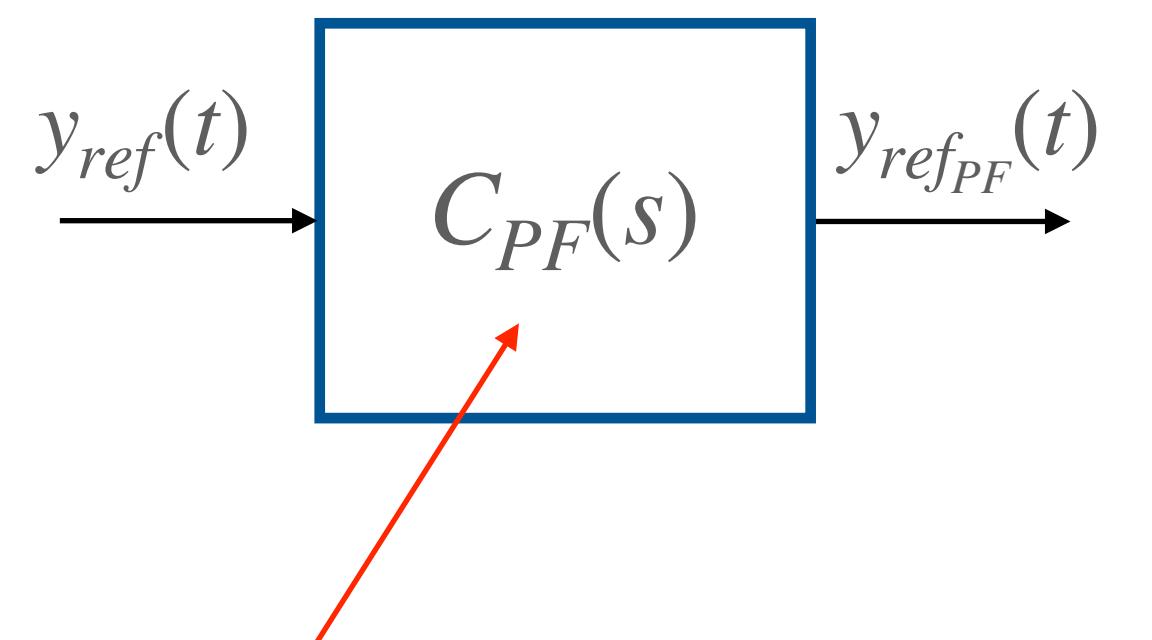
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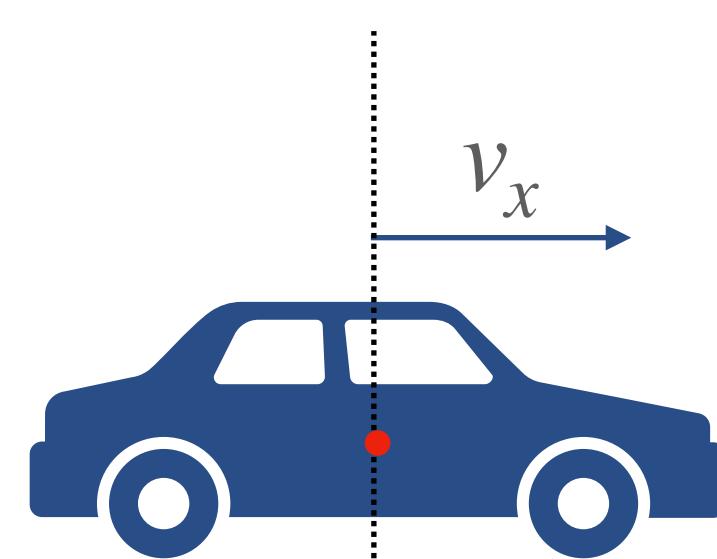
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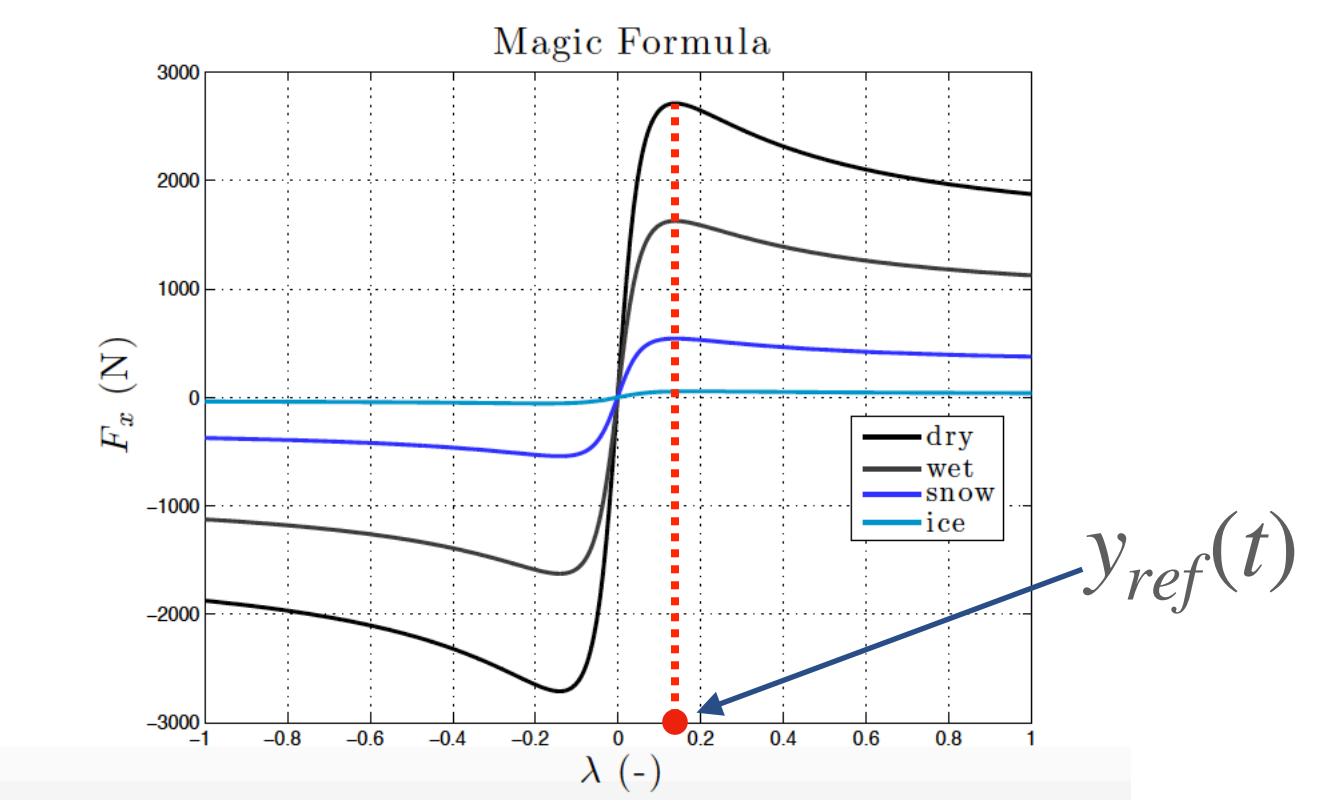
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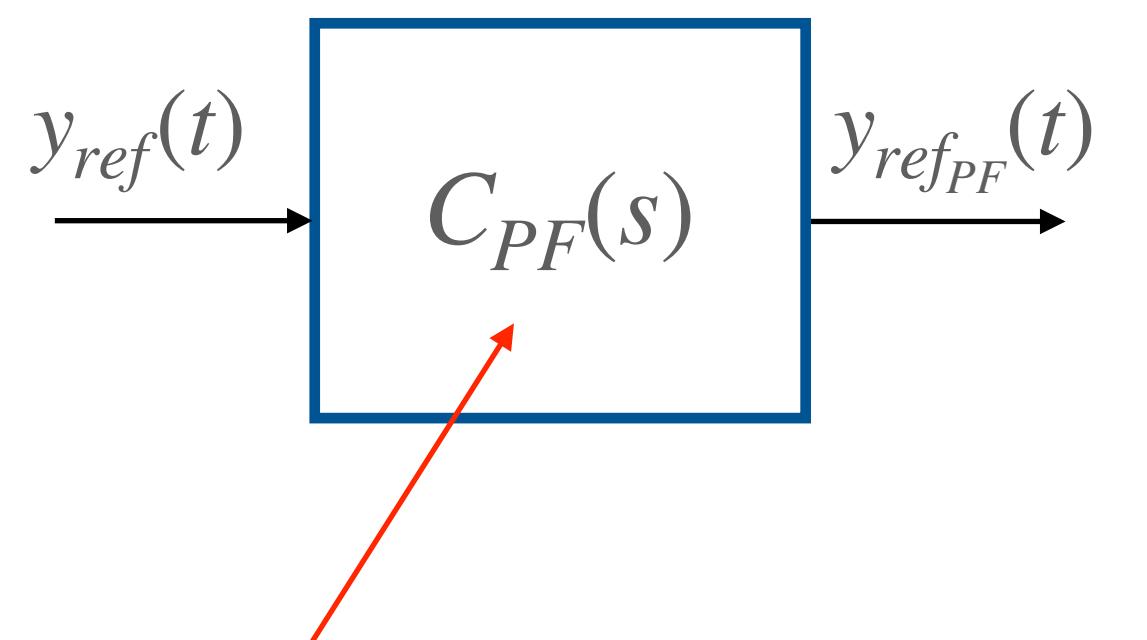
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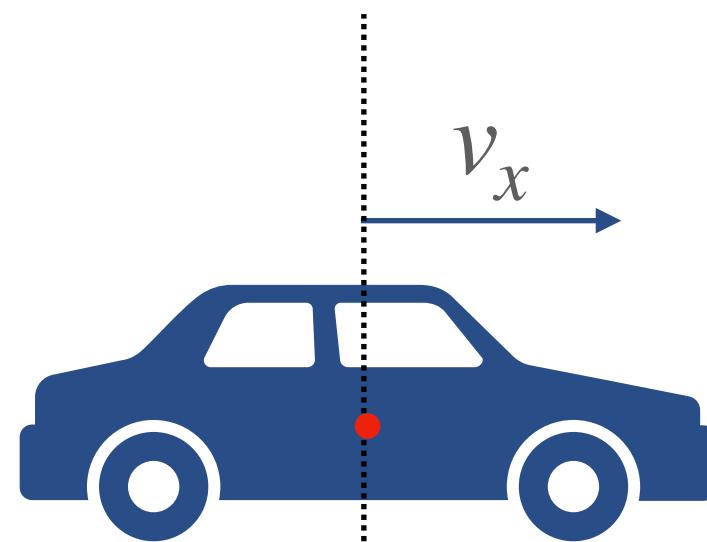
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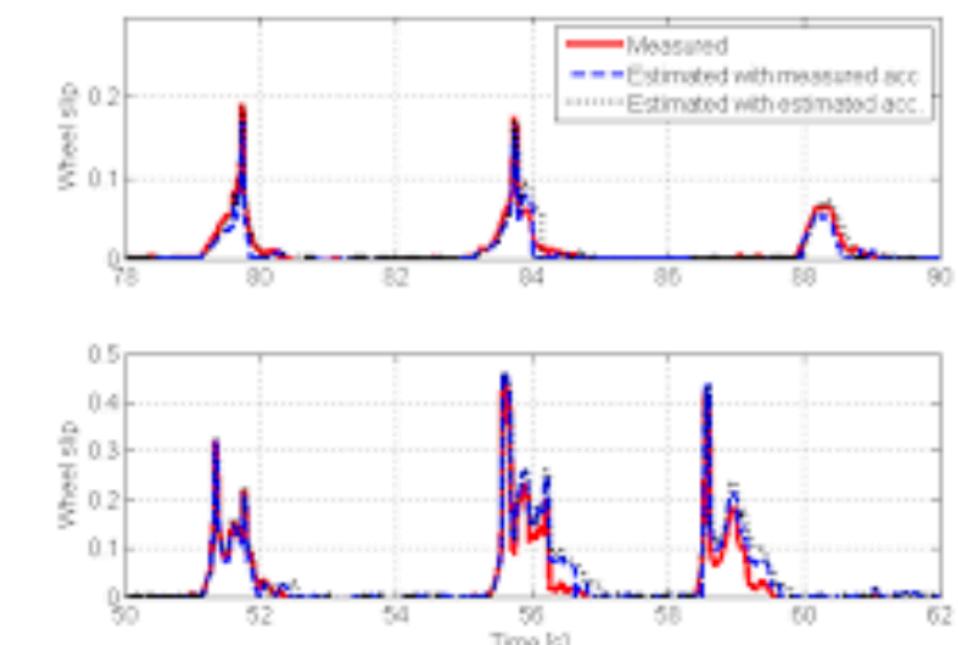
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Example:



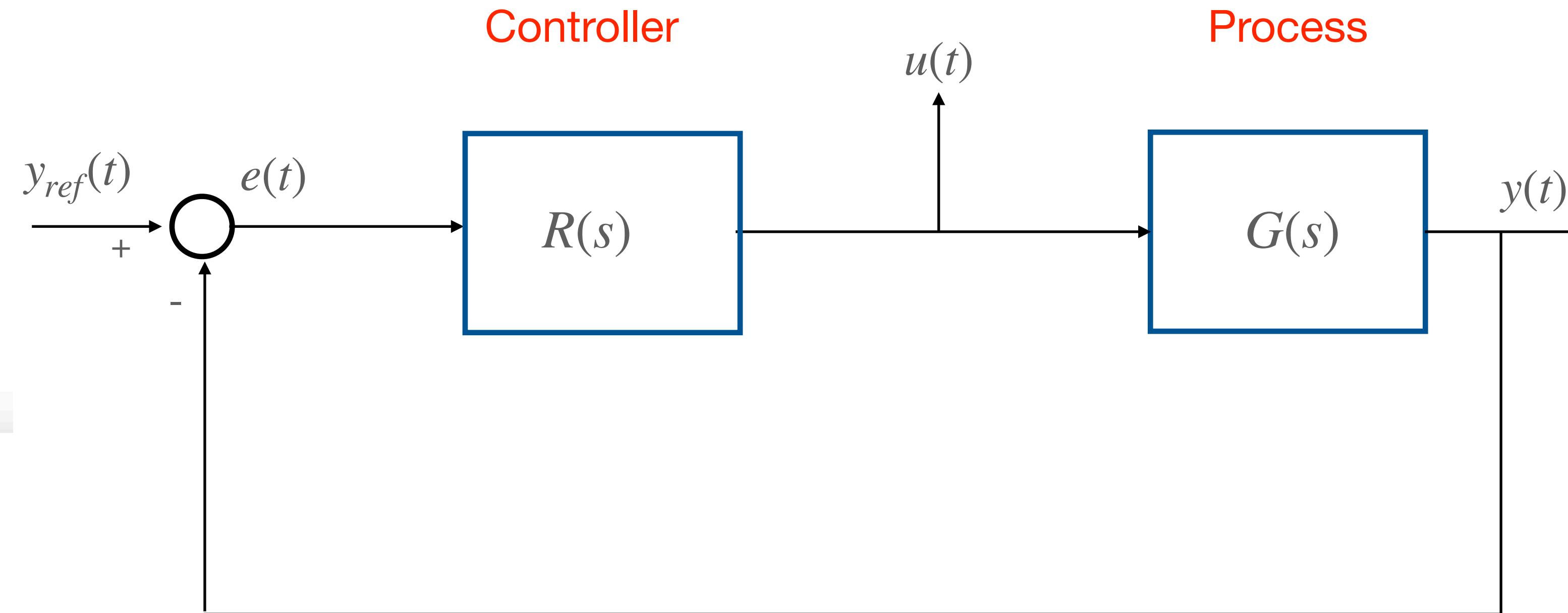
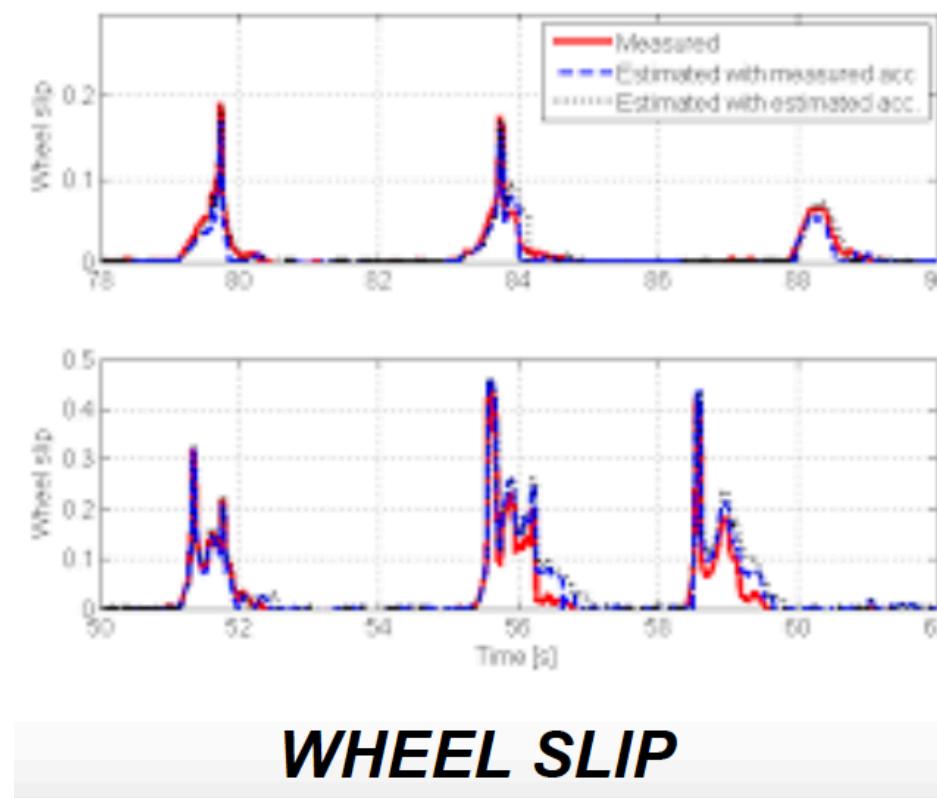
$$\lambda := \frac{\omega r - v_x}{\max(\omega r, v_x)}$$



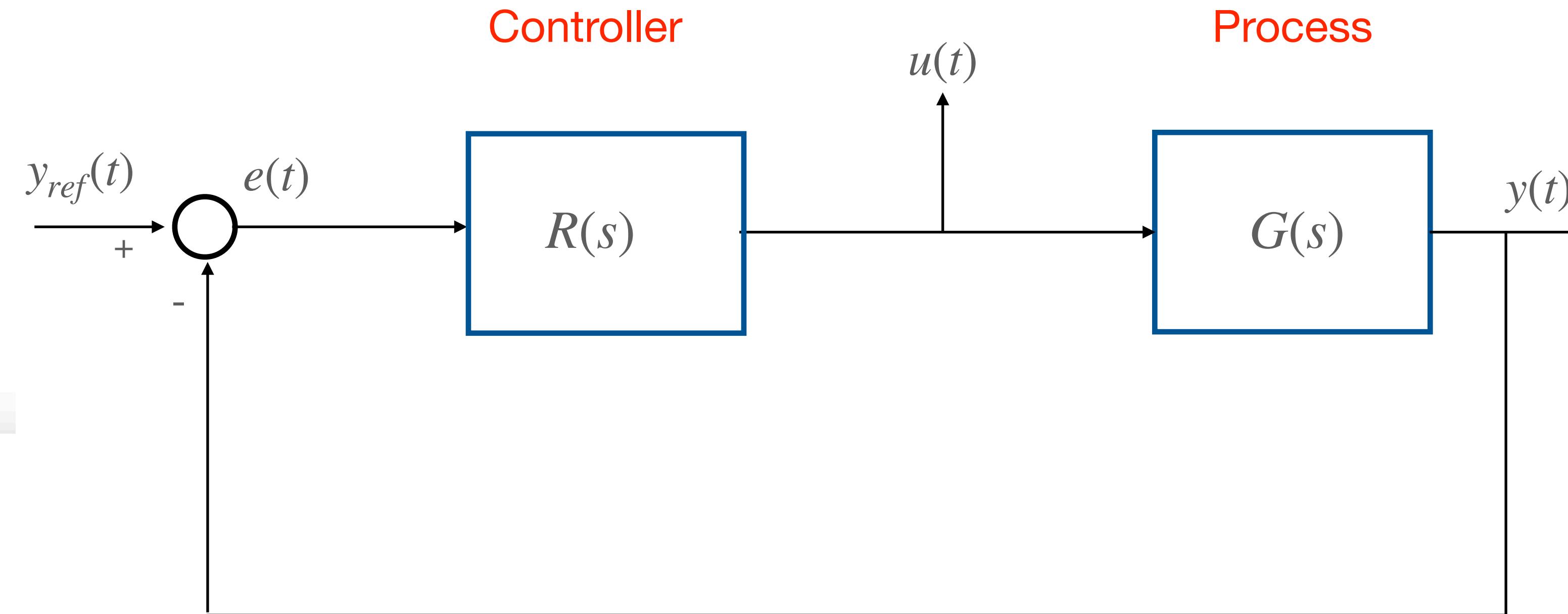
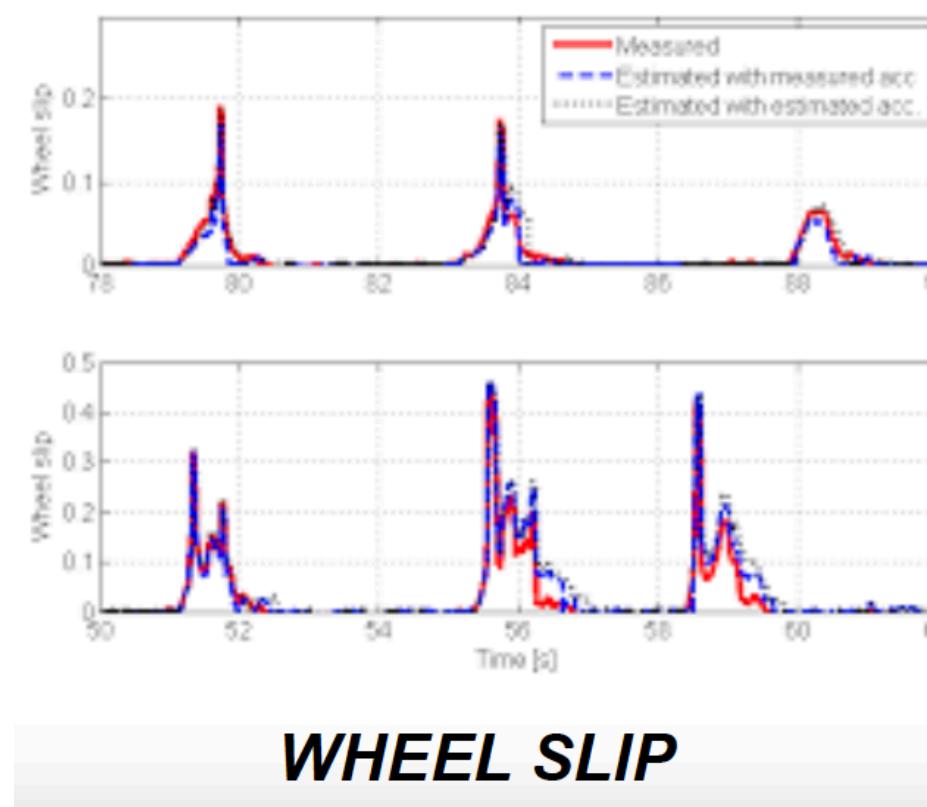
WHEEL SLIP



Basic Control Scheme



Basic Control Scheme

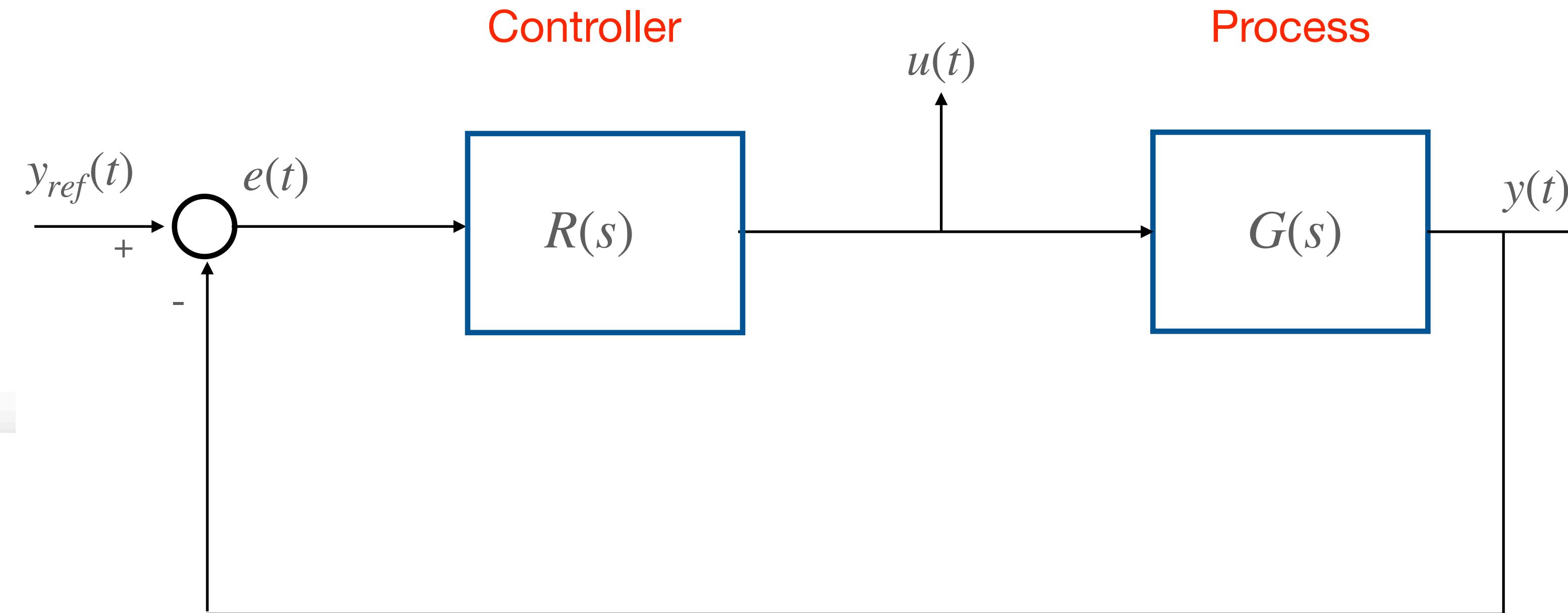
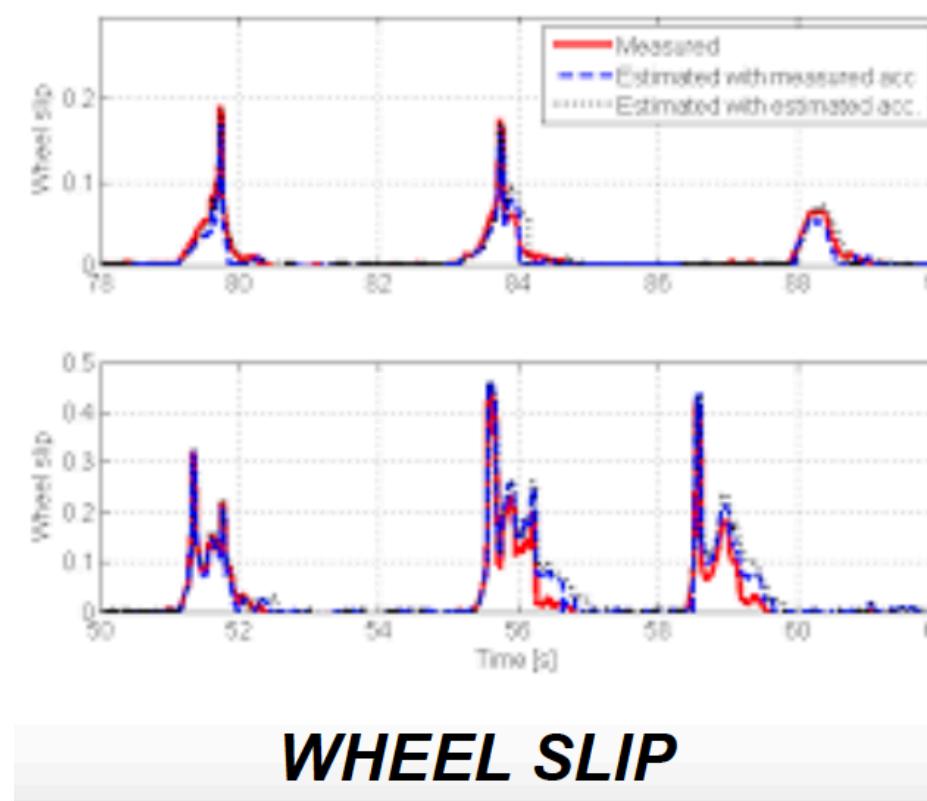


Block Algebra Rule:

$$y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$$



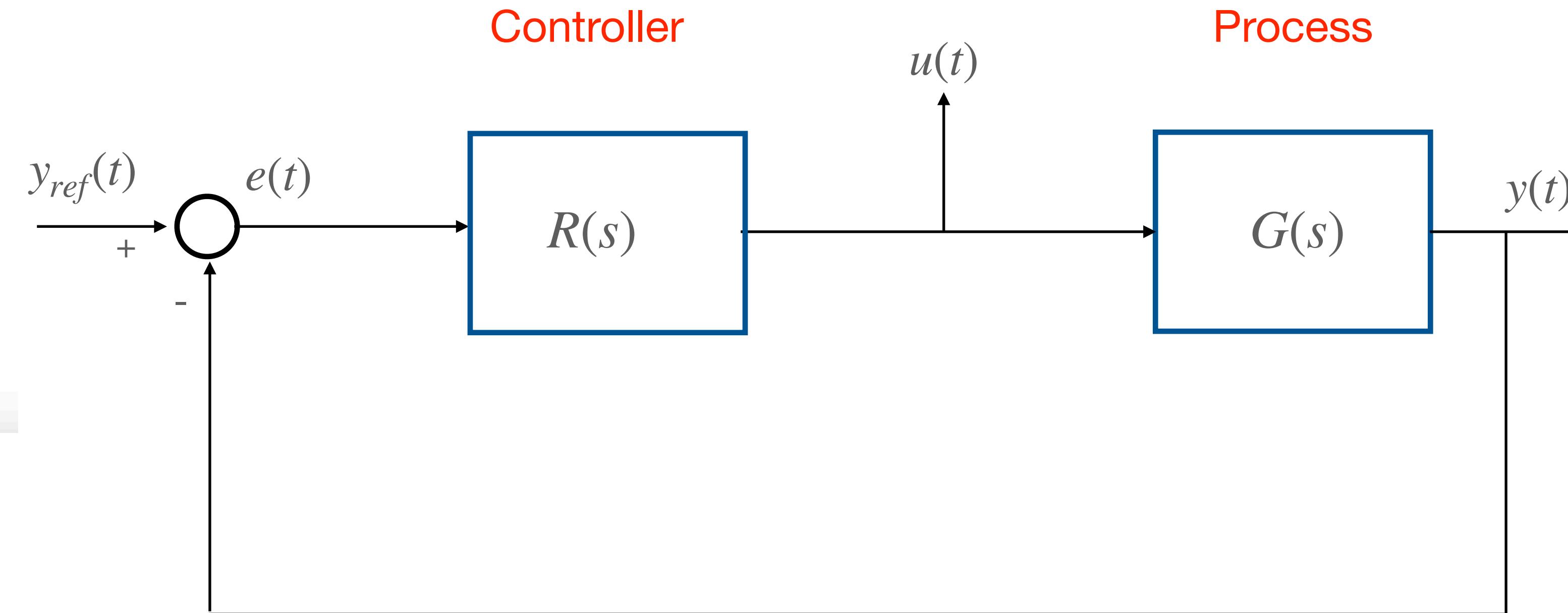
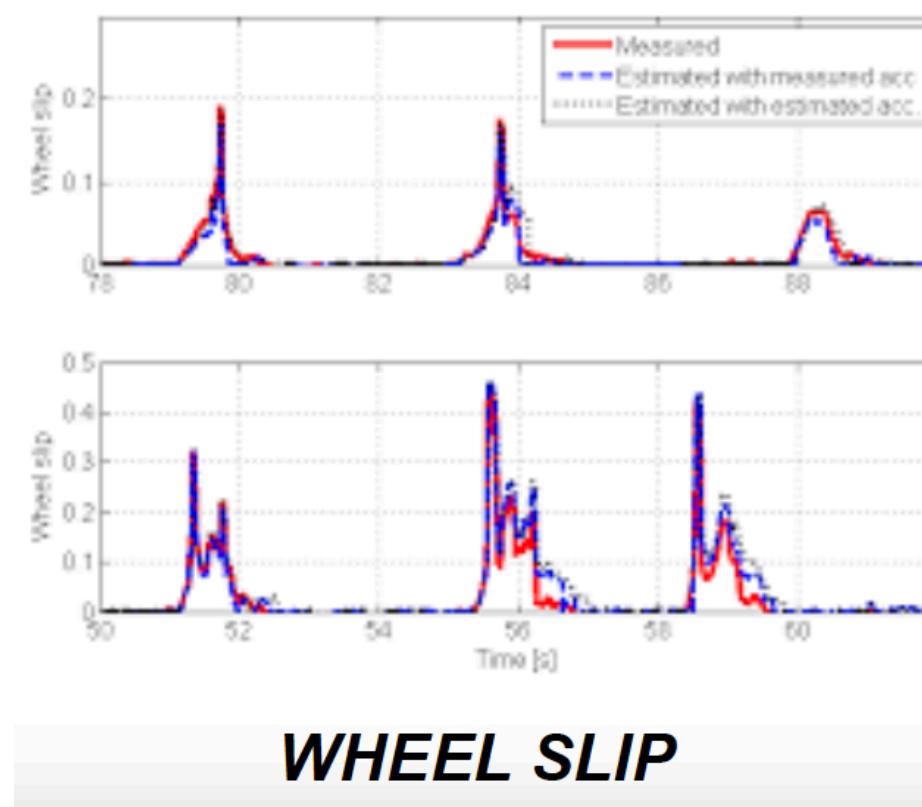
Basic Control Scheme



Block Algebra Rule: $y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$ Control Sensitivity Function



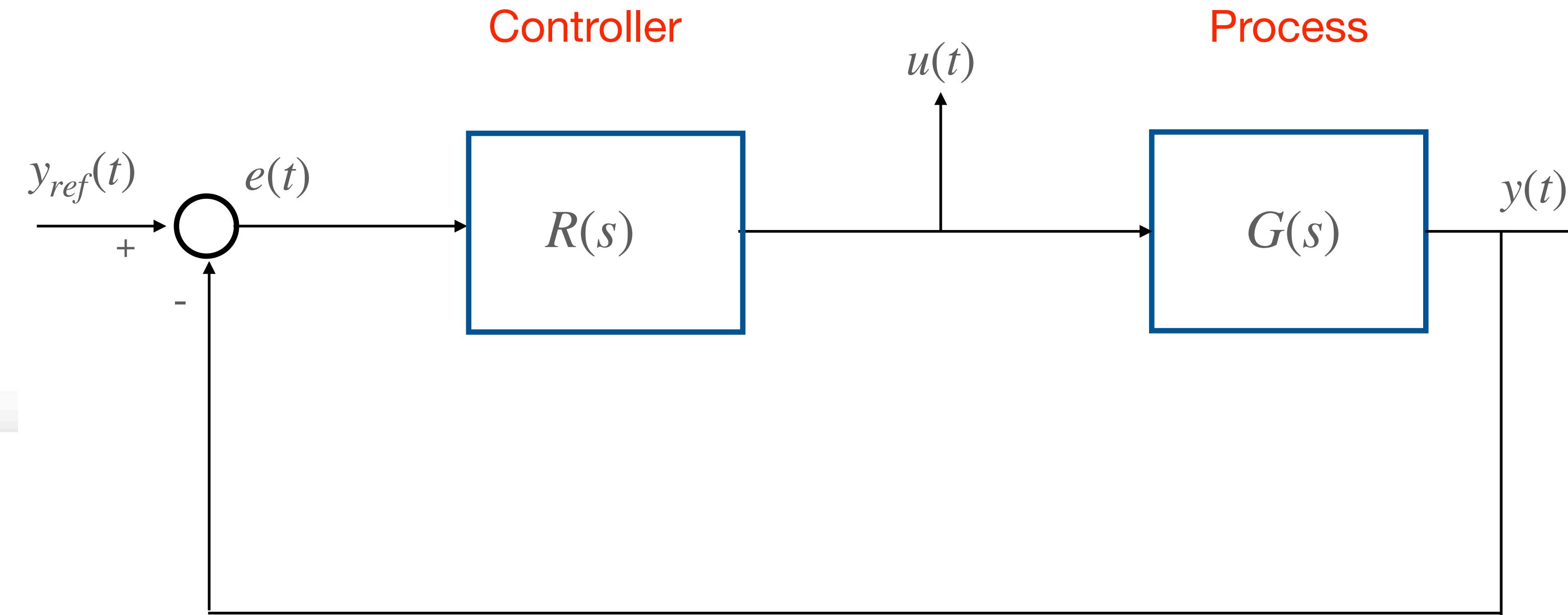
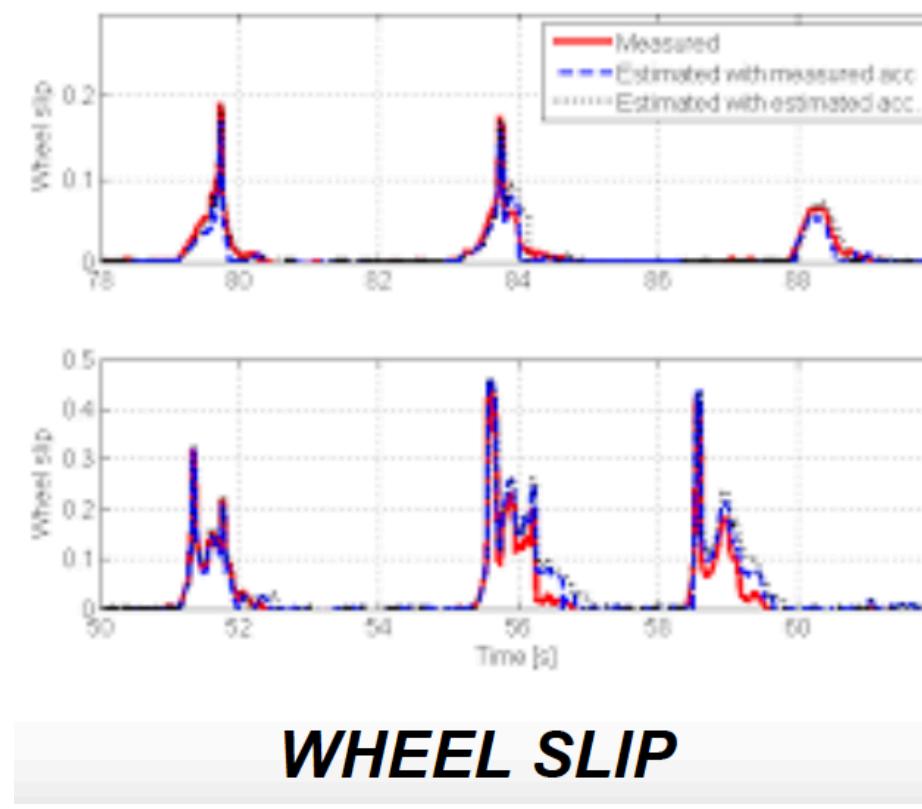
Basic Control Scheme



Block Algebra Rule: $y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$ Control Sensitivity Function

Recommendation: $|Q(j\omega)|$ sufficiently low $\forall \omega \in \mathbb{R}_+ \cup \{0\}$

Basic Control Scheme



Block Algebra Rule:

$$y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$$

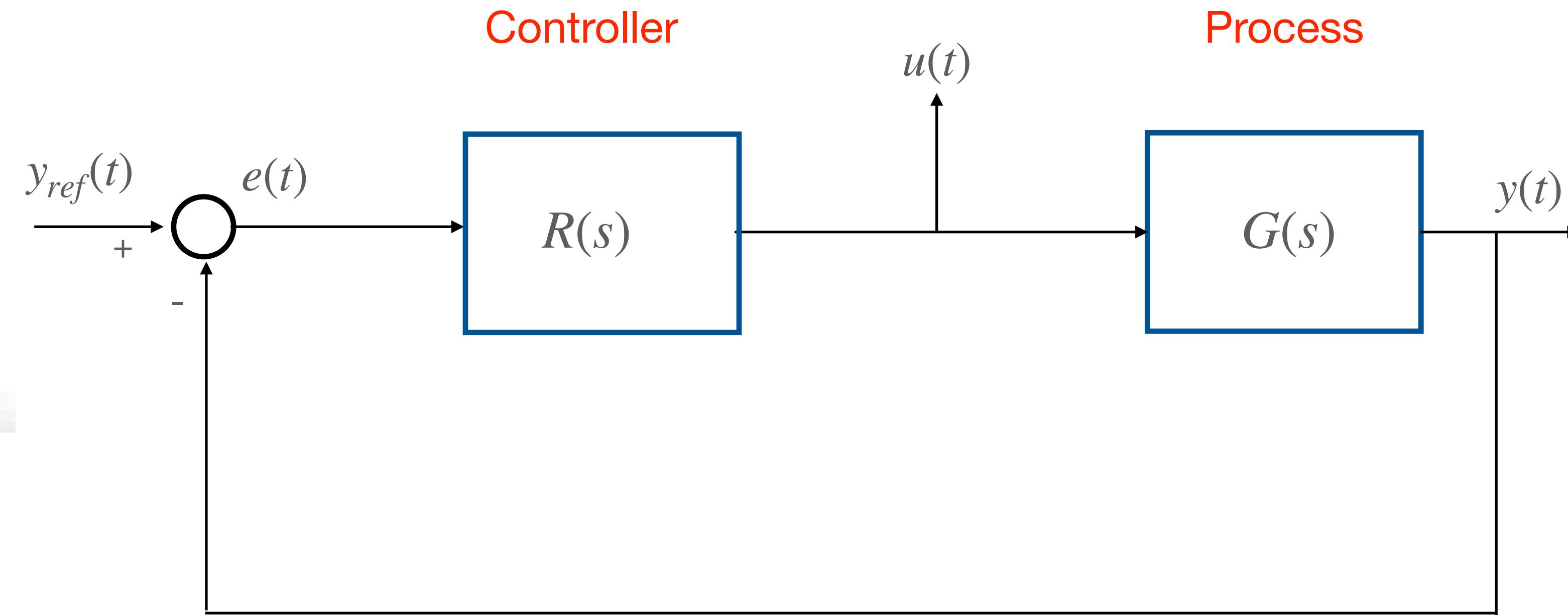
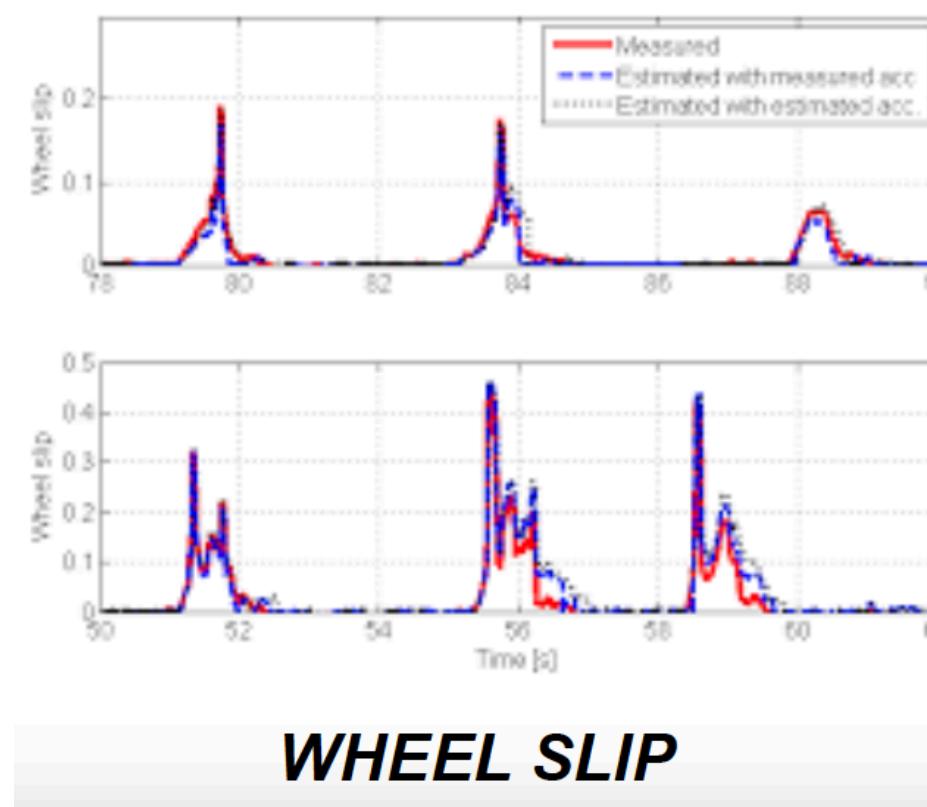
Control Sensitivity Function

Recommendation: $|Q(j\omega)|$ sufficiently low $\forall \omega \in \mathbb{R}_{+} \cup \{0\}$

LF: $|Q(j\omega)| \rightarrow \left| \frac{1}{G(j\omega)} \right|$



Basic Control Scheme



Block Algebra Rule: $y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$ **Control Sensitivity Function**

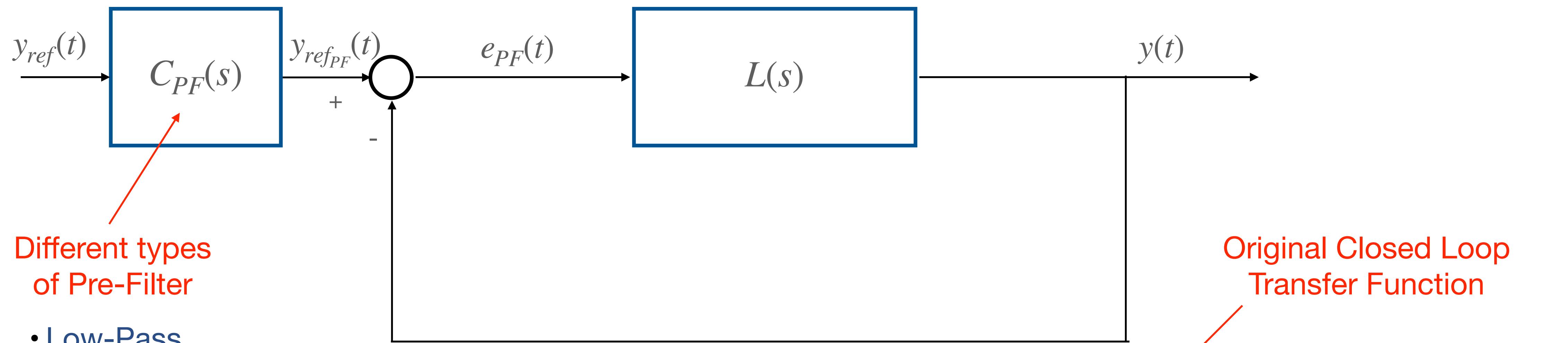
Recommendation: $|Q(j\omega)|$ sufficiently low $\forall \omega \in \mathbb{R}_{+} \cup \{0\}$

LF: $|Q(j\omega)| \rightarrow \left| \frac{1}{G(j\omega)} \right|$

HF: $|Q(j\omega)| \rightarrow |R(j\omega)|$



Pre-filter Based Control Scheme: Case 2



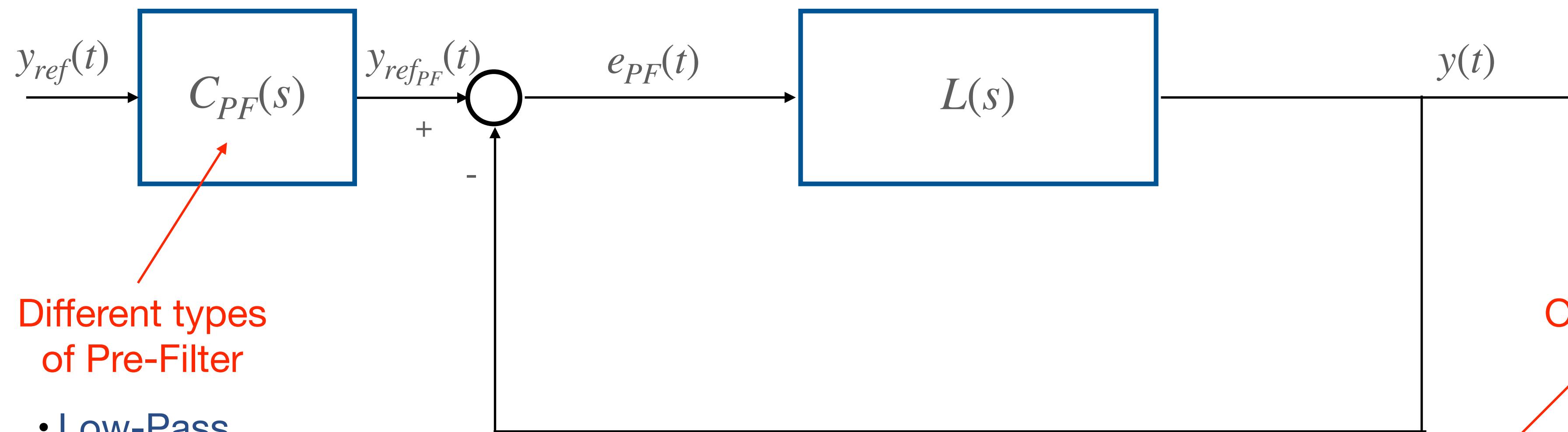
Block Algebra Rule:

$$F(s) = \frac{R(s) \cdot G(s)}{1 + R(s) \cdot G(s)} = \frac{L(s)}{1 + L(s)}$$

$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



Pre-filter Based Control Scheme: Case 2



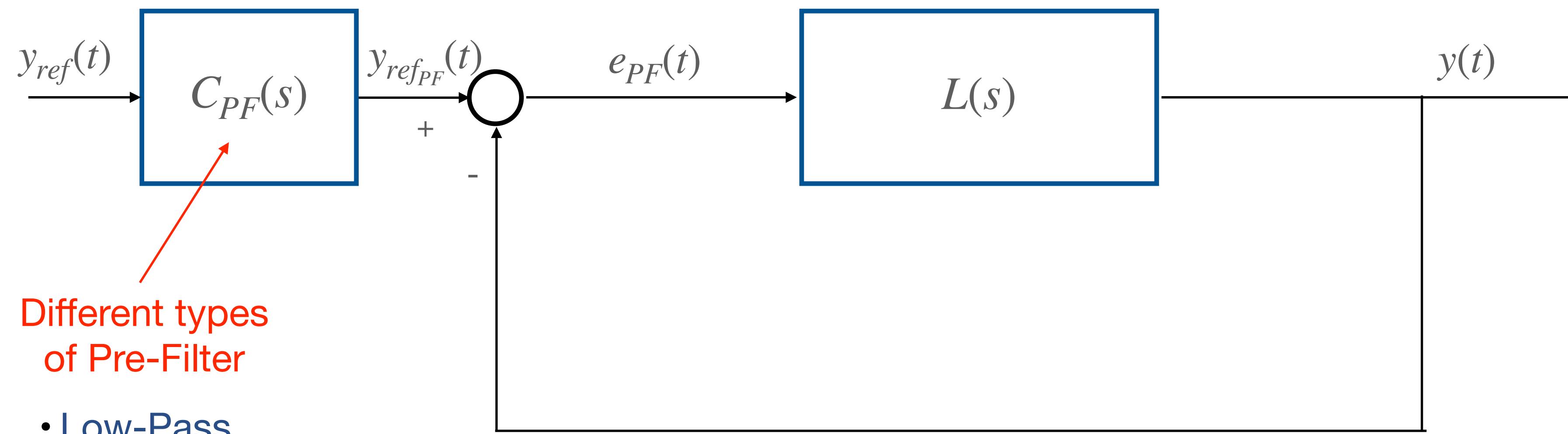
Block Algebra Rule:

$$F(s) = \frac{R(s) \cdot G(s)}{1 + R(s) \cdot G(s)} = \frac{L(s)}{1 + L(s)}$$

$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



Pre-filter Based Control Scheme: Case 2



Assumptions:

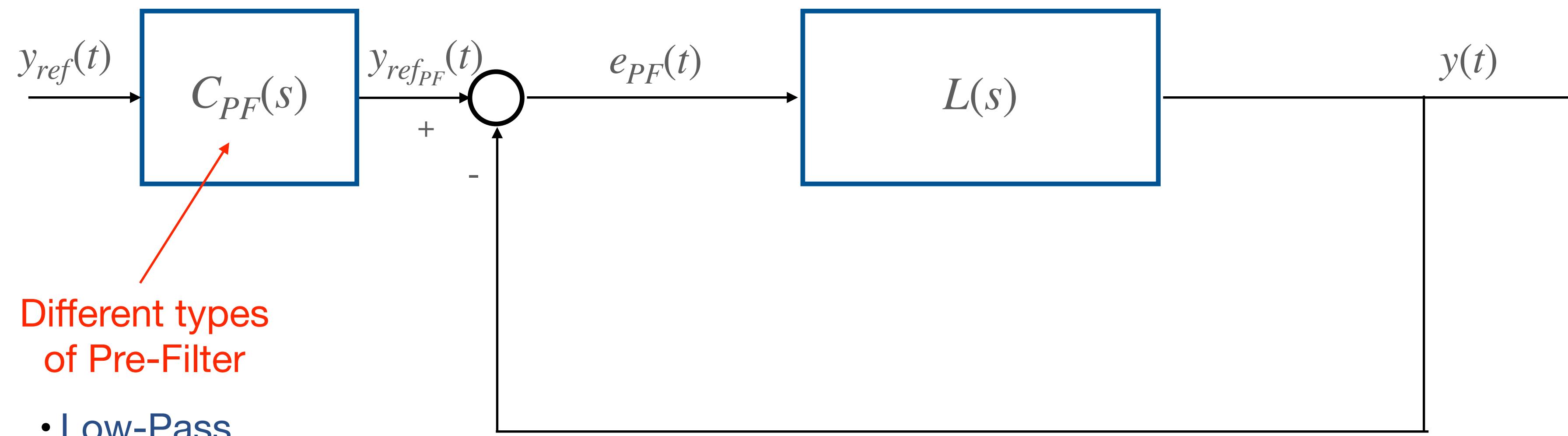
- $C_{PF}(s)$ As. Stable
- Proper
- Unitary gain

The pole precedes the zero on the ω axis

$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



Pre-filter Based Control Scheme: Case 2

**Assumptions:**

- $C_{PF}(s)$ As. Stable
- Proper
- Unitary gain

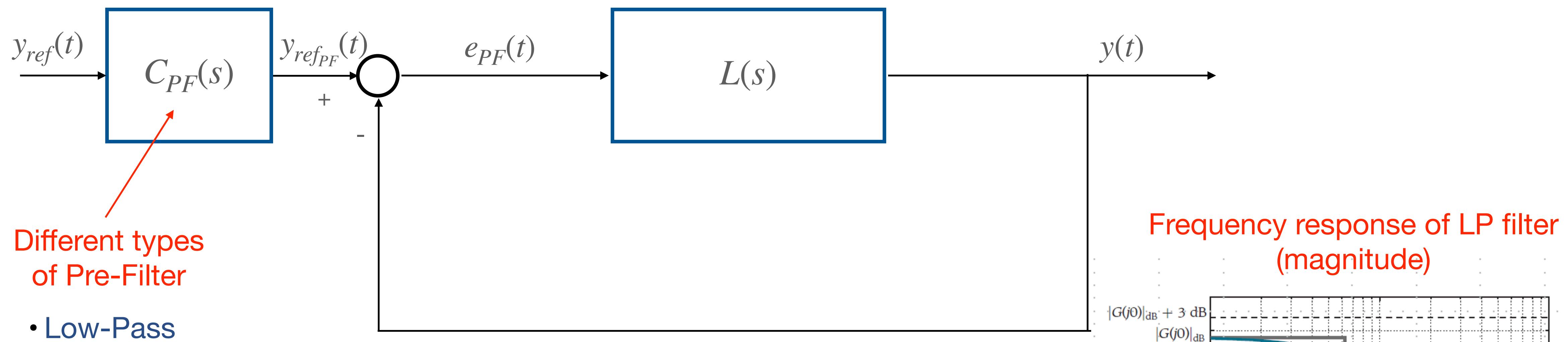
$$\lim_{s \rightarrow 0} \frac{\mu(1 + \tau' s)}{1 + \tau s} = \mu = 1$$

The pole precedes the zero on the ω axis

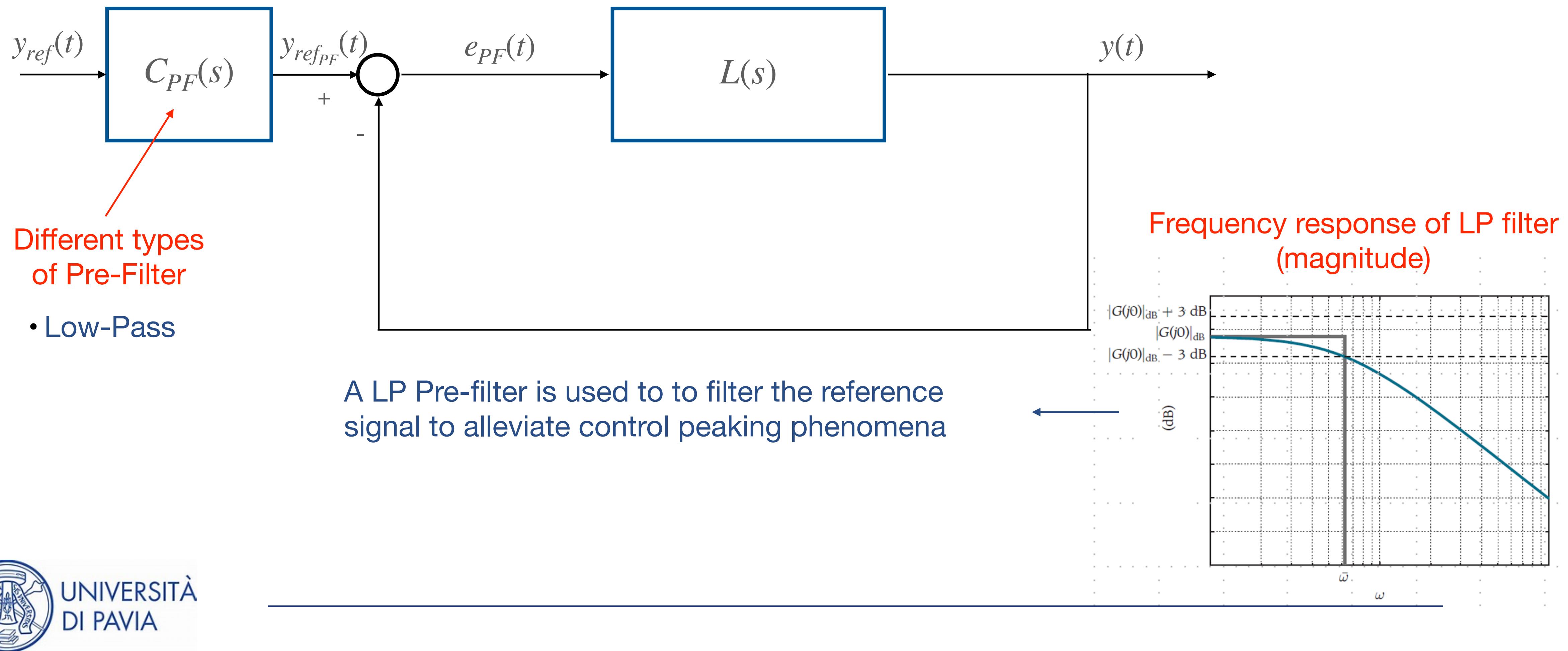
$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



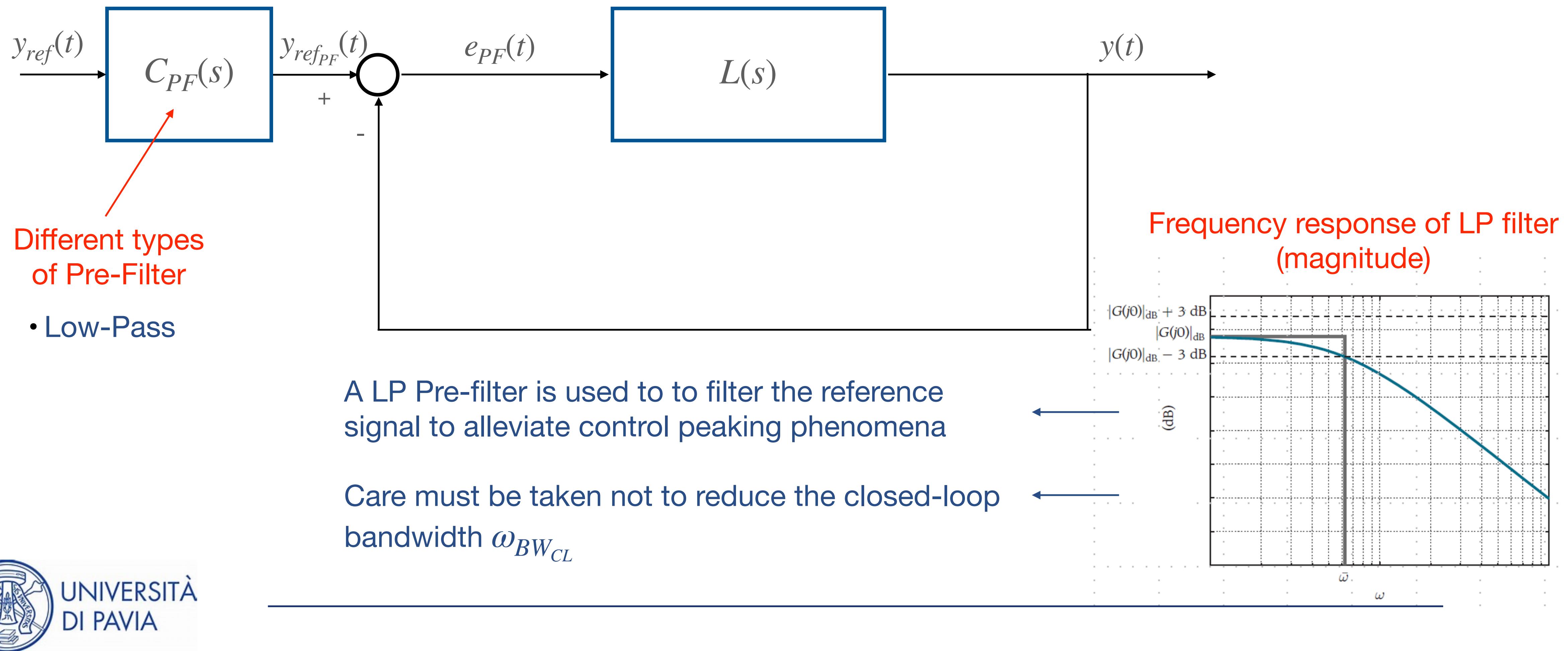
Pre-filter Based Control Scheme: Case 2



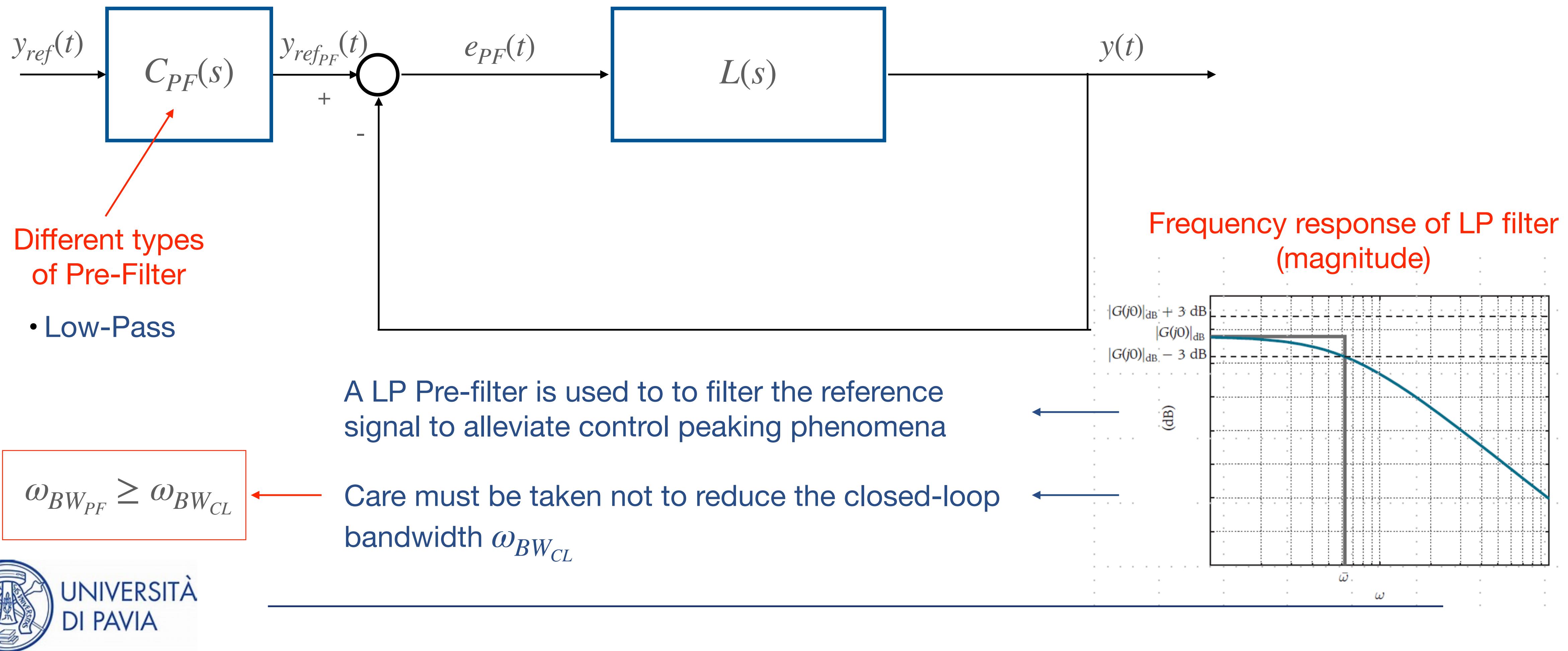
Pre-filter Based Control Scheme: Case 2



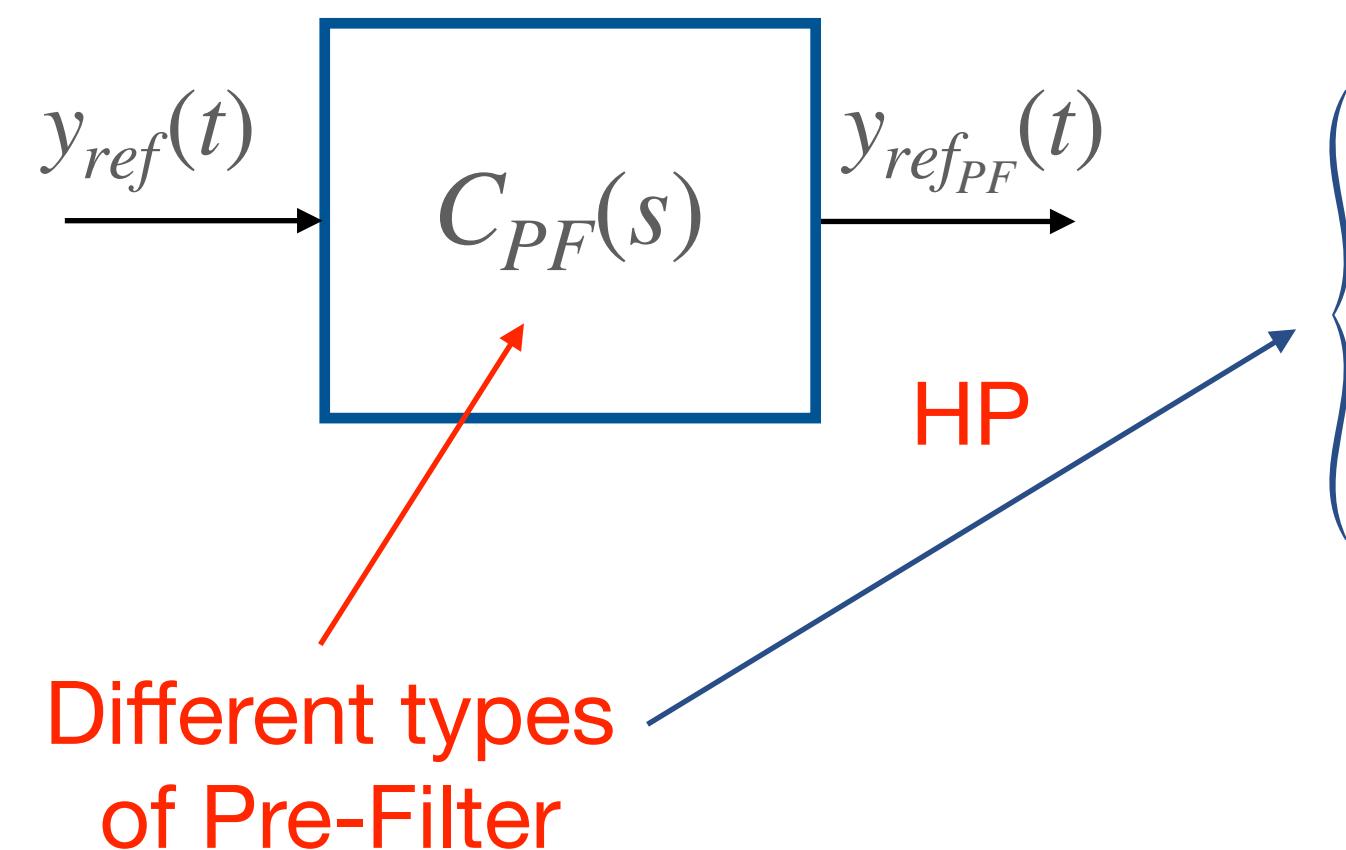
Pre-filter Based Control Scheme: Case 2



Pre-filter Based Control Scheme: Case 2



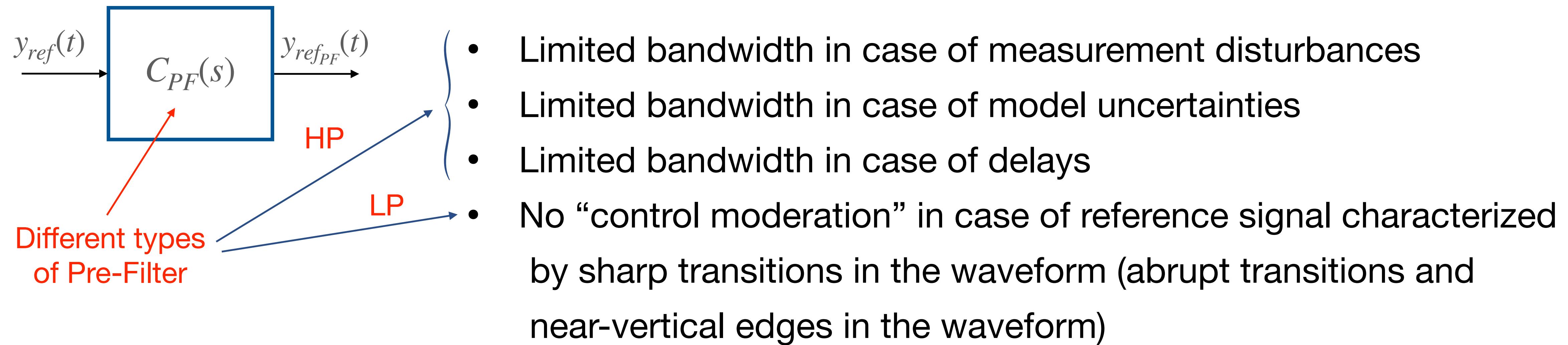
Pre-filter Based Control Scheme: Case 3



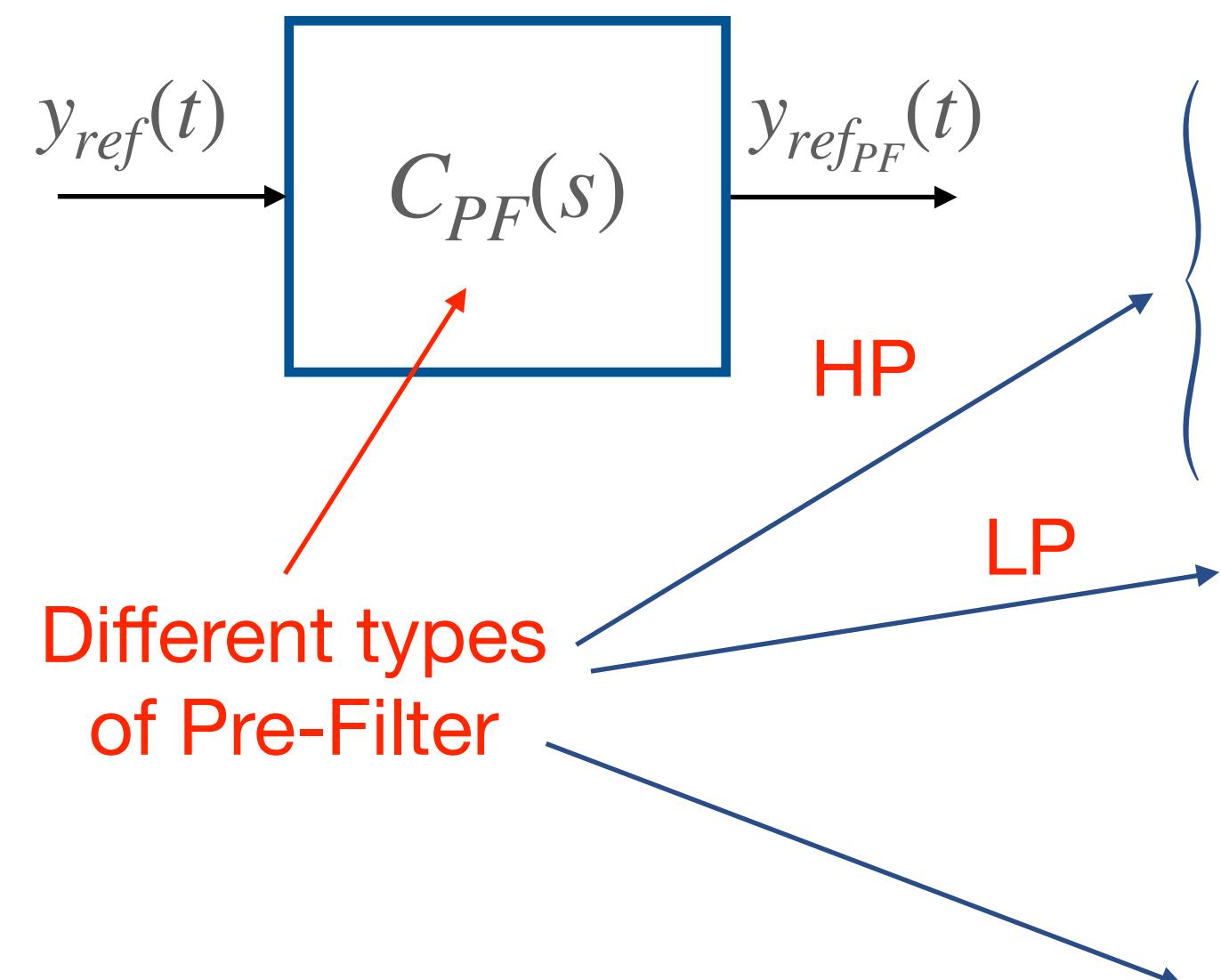
- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays
- No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)



Pre-filter Based Control Scheme: Case 3



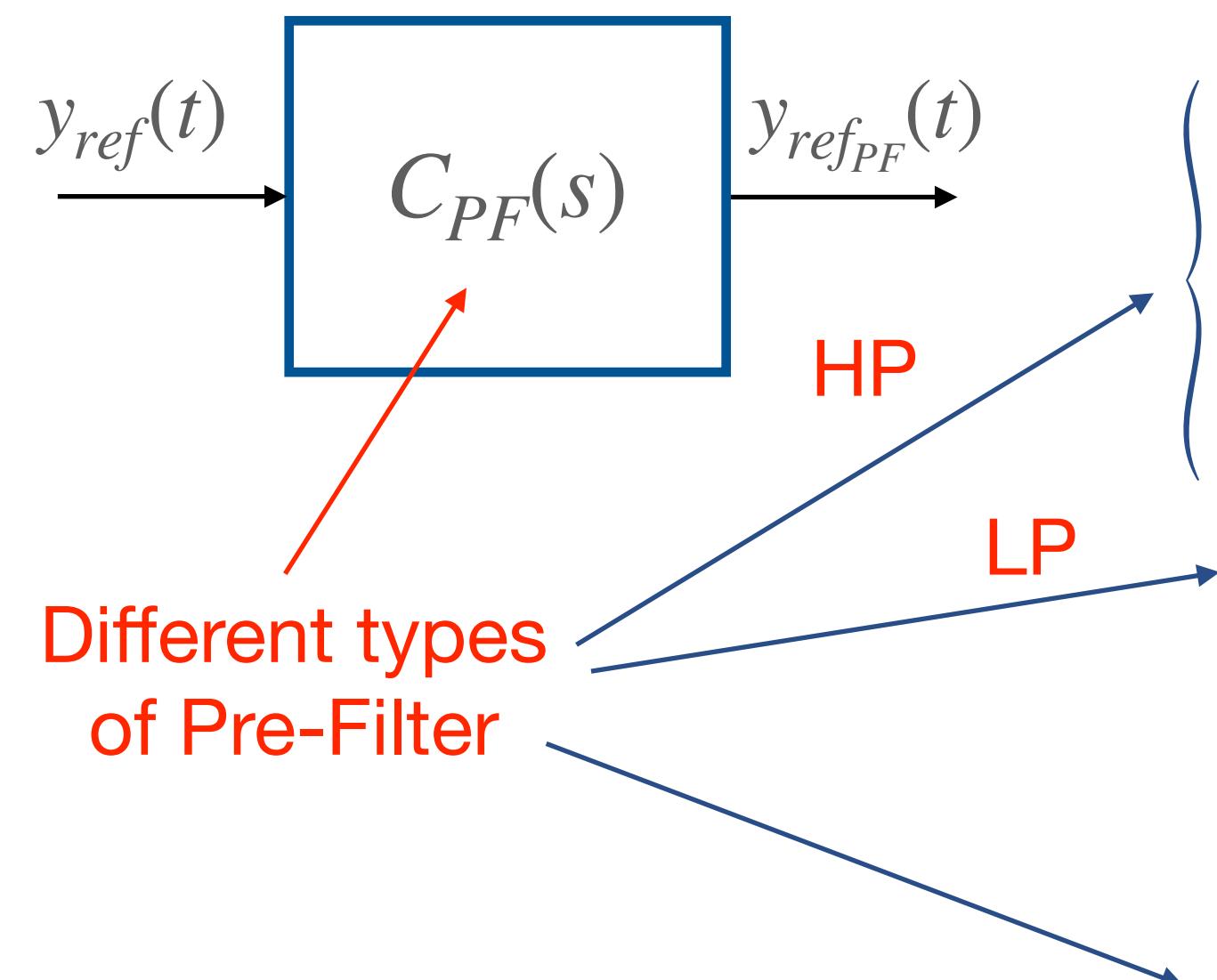
Pre-filter Based Control Scheme: Case 3



- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays
- No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
- Possible difficulties in enforcing steady-state precision with step reference signal via the classical approach



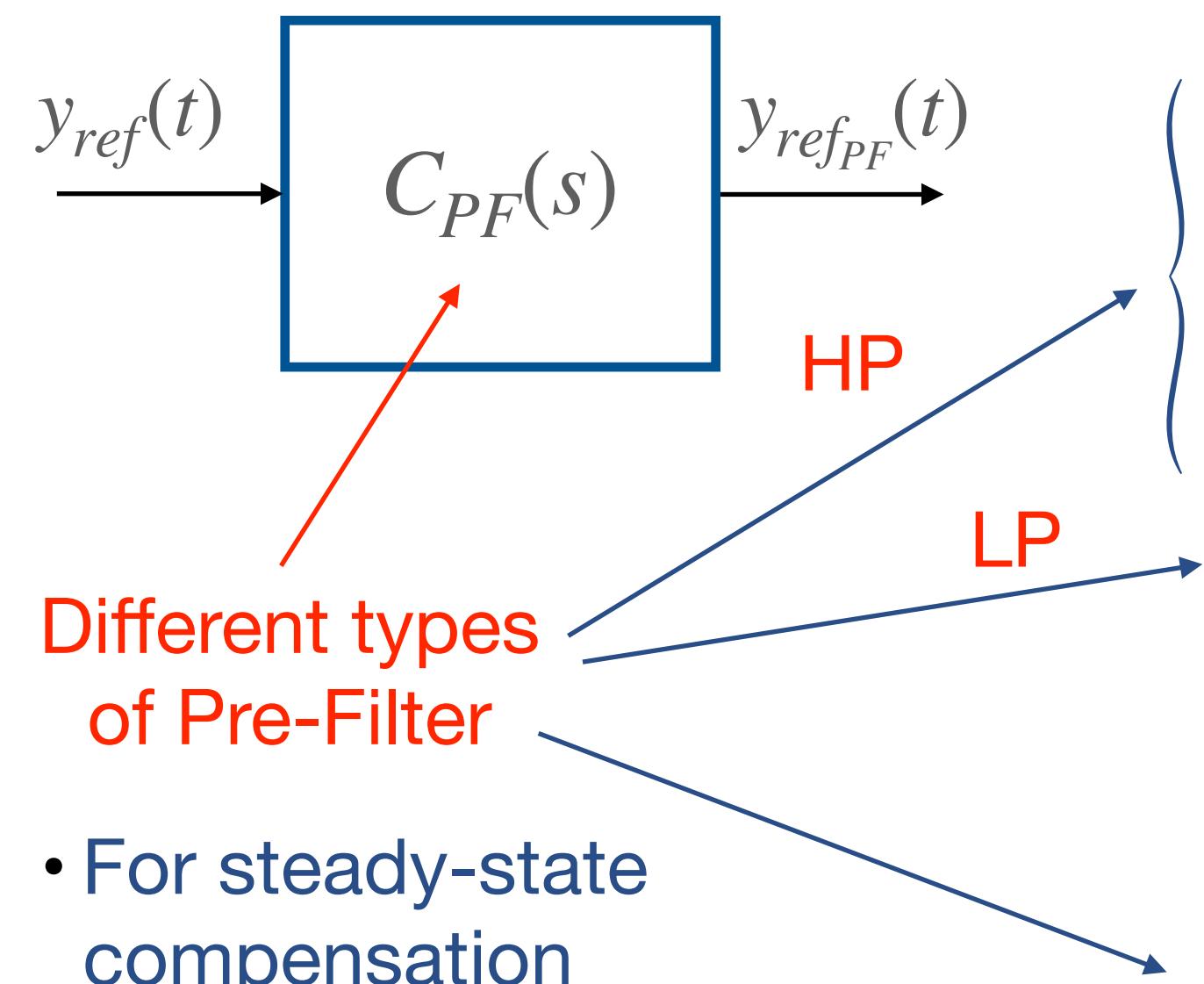
Pre-filter Based Control Scheme: Case 3



- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays
- No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
- Possible difficulties in enforcing steady-state precision with step reference signal via the classical approach
(by introducing integrators in the direct path)



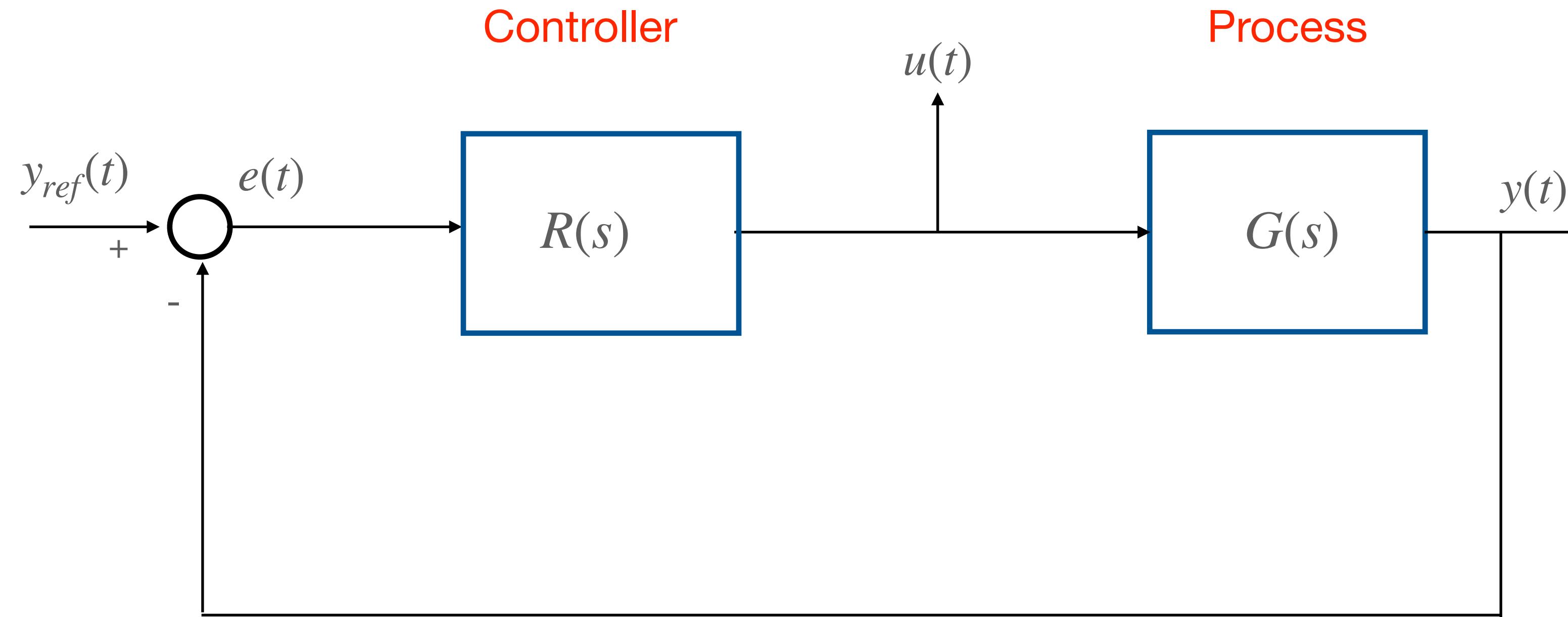
Pre-filter Based Control Scheme: Case 3



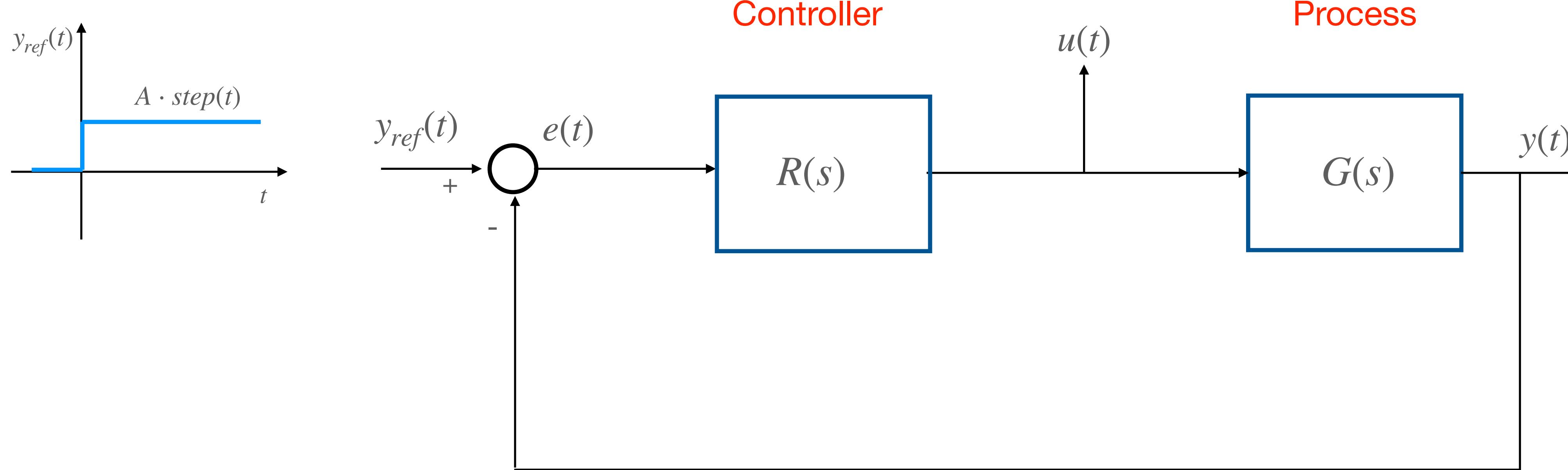
- Limited bandwidth in case of measurement disturbances
 - Limited bandwidth in case of model uncertainties
 - Limited bandwidth in case of delays
 - No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
 - Possible difficulties in enforcing steady-state precision with step reference signal via the classical approach (by introducing integrators in the direct path)
- For steady-state compensation



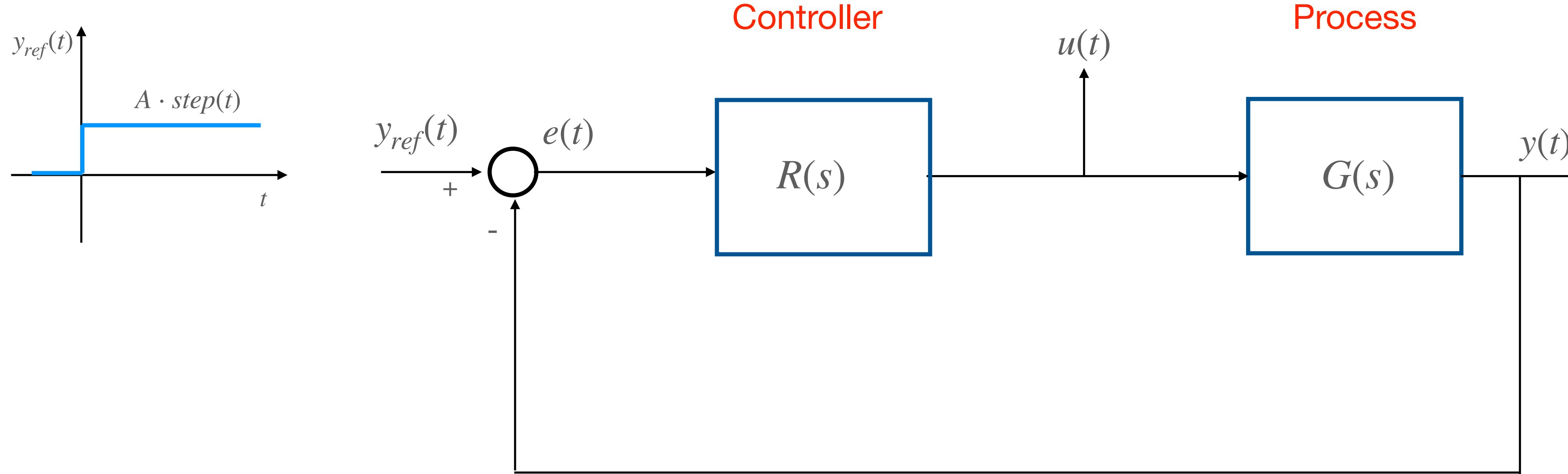
Basic Control Scheme



Basic Control Scheme



Basic Control Scheme

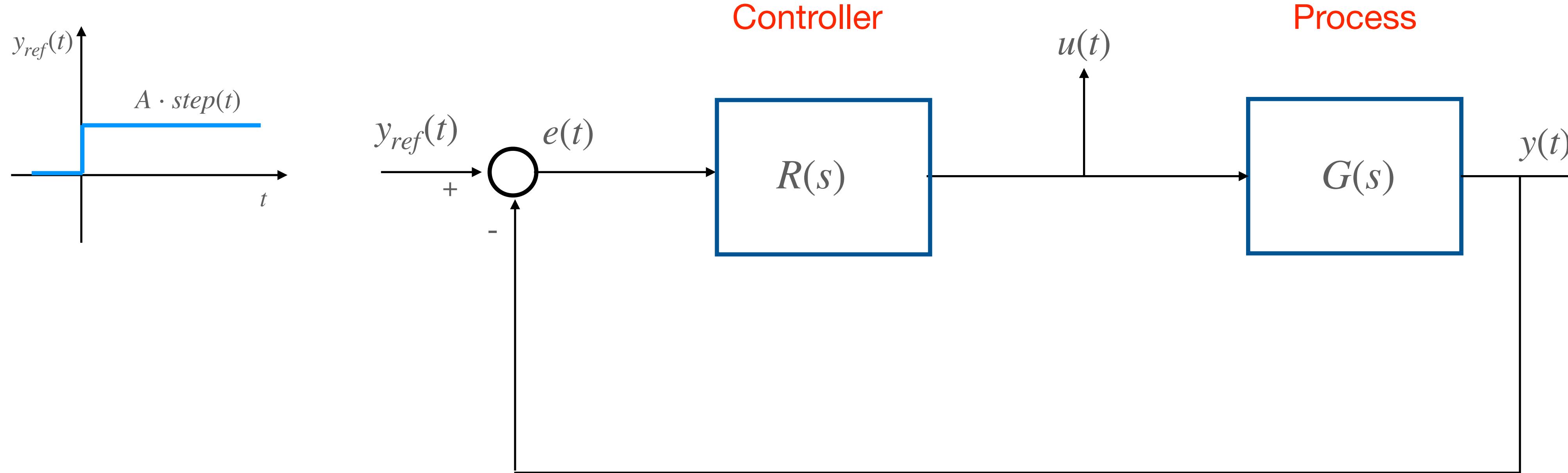


Block Algebra Rule: $y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$

$$y \approx y_{ref} : F(0) = 1$$



Basic Control Scheme

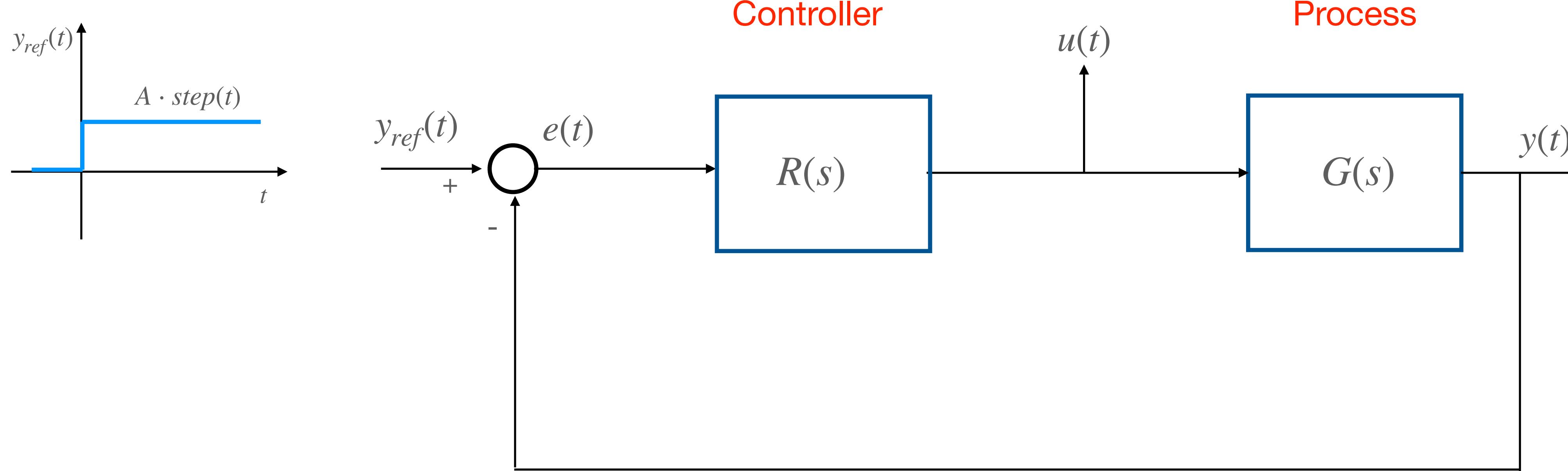


Block Algebra Rule: $y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$

$y \approx y_{ref} : F(0) = 1 \longrightarrow \text{Impossible if } L(s) \text{ is not at least a TYPE 1 transfer function}$



Basic Control Scheme: Example



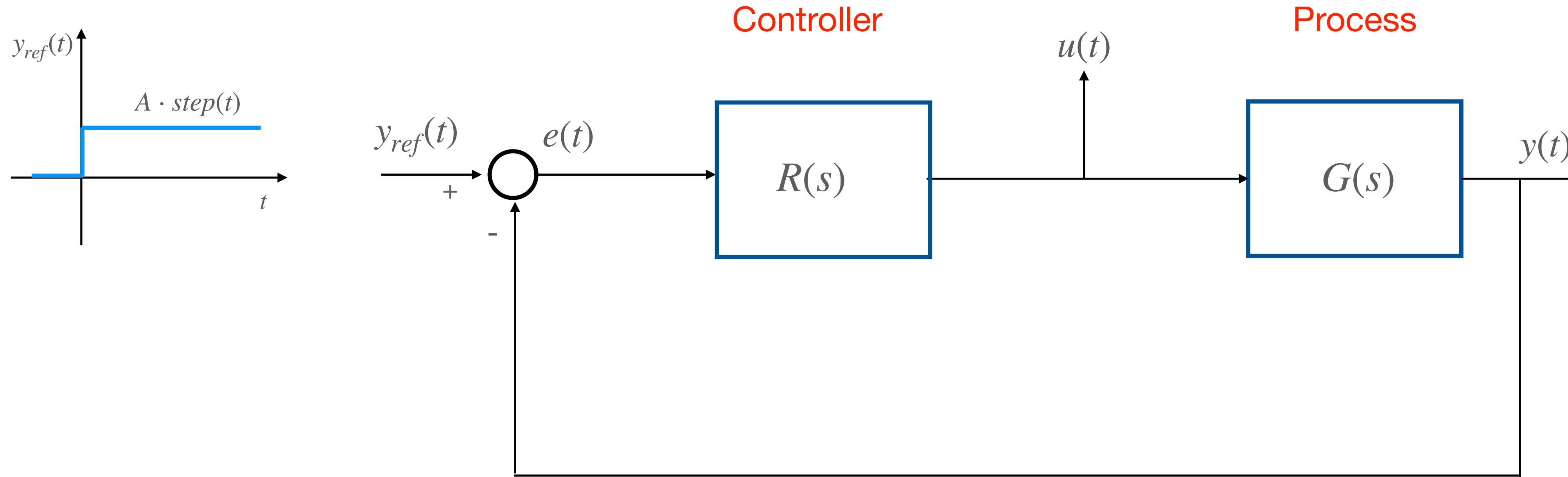
$$L(s) = \frac{100}{s + 10} \rightarrow F(s) = \frac{\frac{100}{s + 10}}{1 + \frac{100}{s + 10}} = \frac{100}{s + 110}$$

$$y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$$

$y \approx y_{ref} : F(0) = 1 \quad \longrightarrow \text{Impossible if } L(s) \text{ is not at least a TYPE 1 transfer function}$



Basic Control Scheme: Example



$$L(s) = \frac{100}{s + 10} \rightarrow F(s) = \frac{\frac{100}{s + 10}}{1 + \frac{100}{s + 10}} = \frac{100}{s + 110}$$

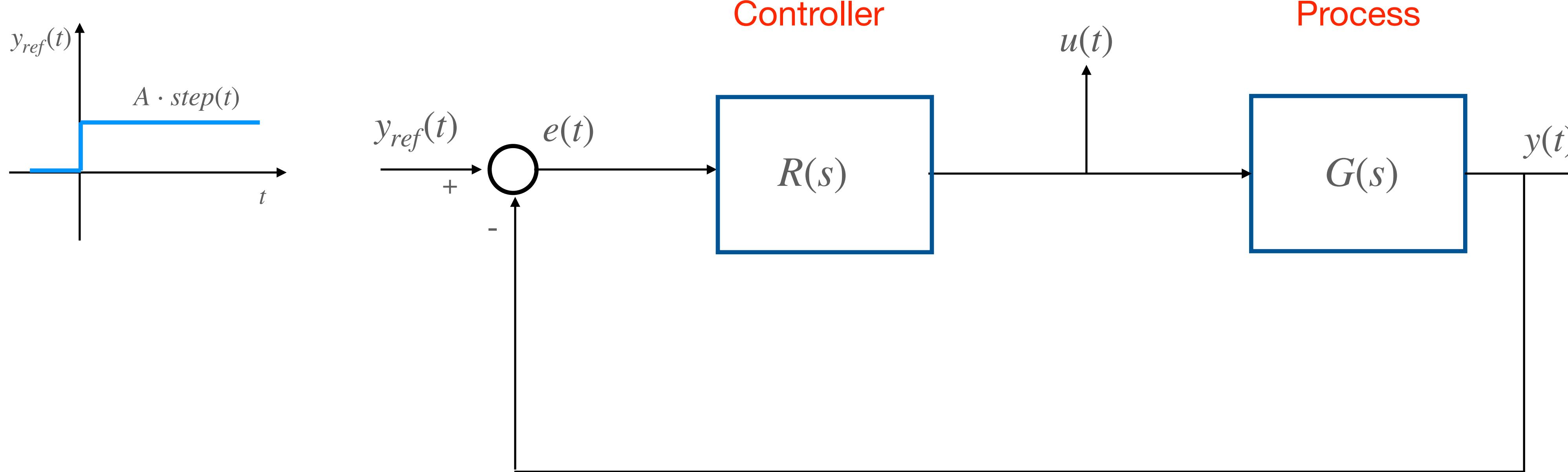
$$\lim_{s \rightarrow 0} F(s) = \frac{100}{110} = \frac{10}{11} \neq 1$$

$$y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$$

$$y \approx y_{ref} : F(0) = 1 \quad \longrightarrow \text{Impossible if } L(s) \text{ is not at least a TYPE 1 transfer function}$$



Basic Control Scheme: Example



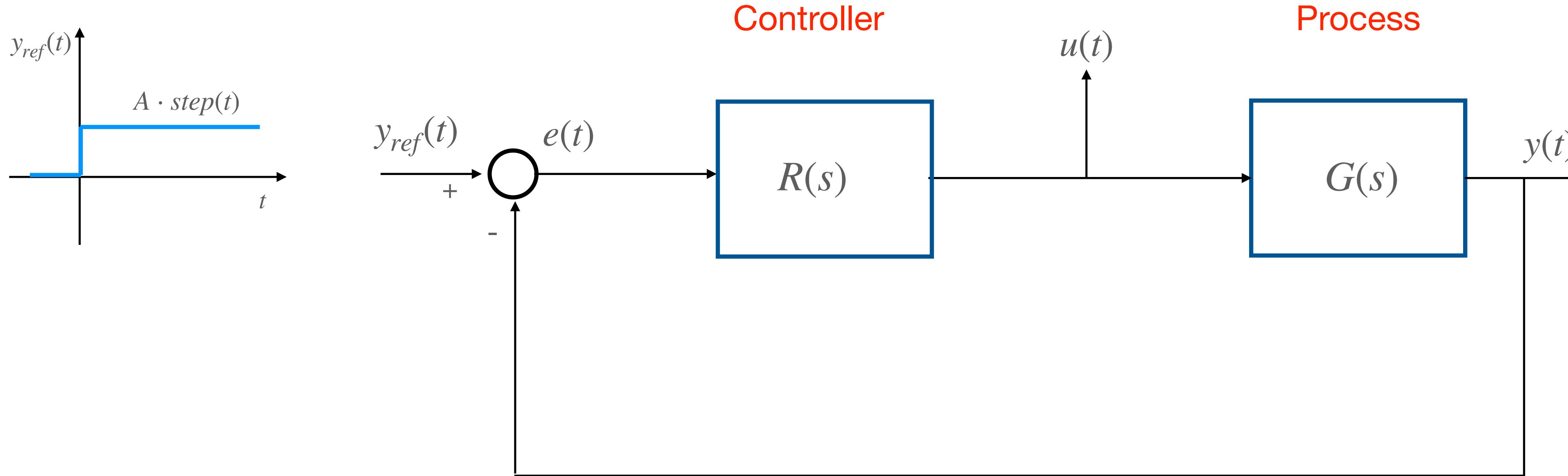
$$L(s) = \frac{100}{s(s+10)} \rightarrow F(s) = \frac{\frac{100}{s(s+10)}}{1 + \frac{100}{s(s+10)}} = \frac{100}{s(s+10)+100}$$

$$y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$$

$y \approx y_{ref} : F(0) = 1 \longrightarrow$ Impossible if $L(s)$ is not at least a TYPE 1 transfer function



Basic Control Scheme: Example



$$L(s) = \frac{100}{s(s+10)} \rightarrow F(s) = \frac{\frac{100}{s(s+10)}}{1 + \frac{100}{s(s+10)}} = \frac{100}{s(s+10)+100}$$

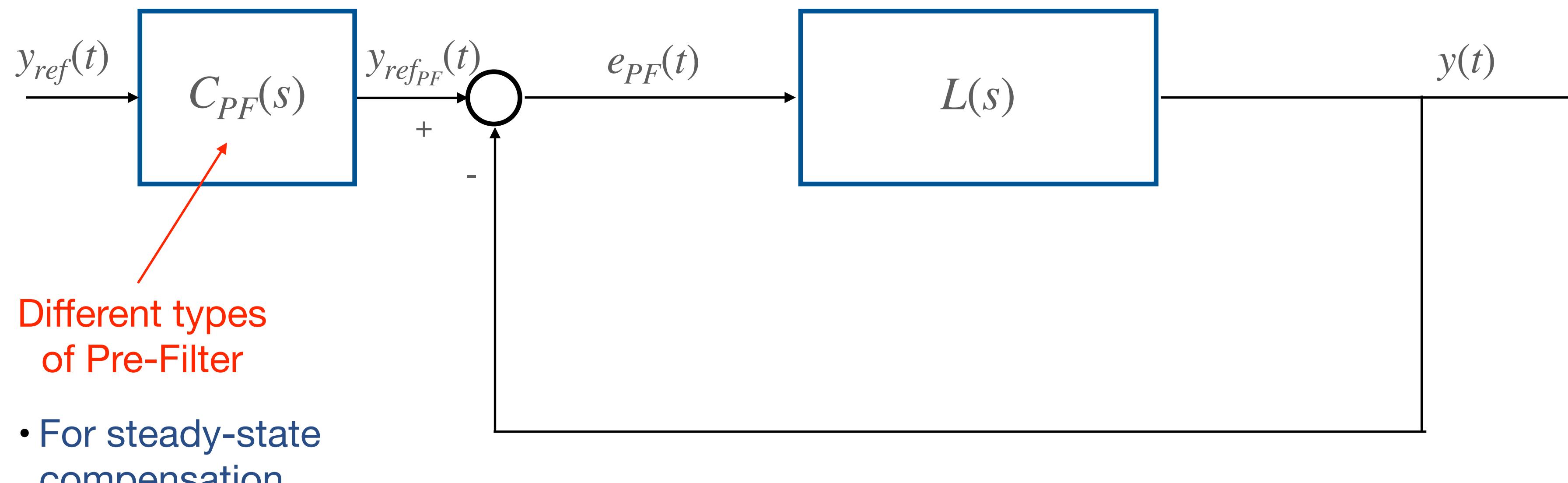
$$\lim_{s \rightarrow 0} F(s) = \frac{100}{100} = 1$$

$$y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$$

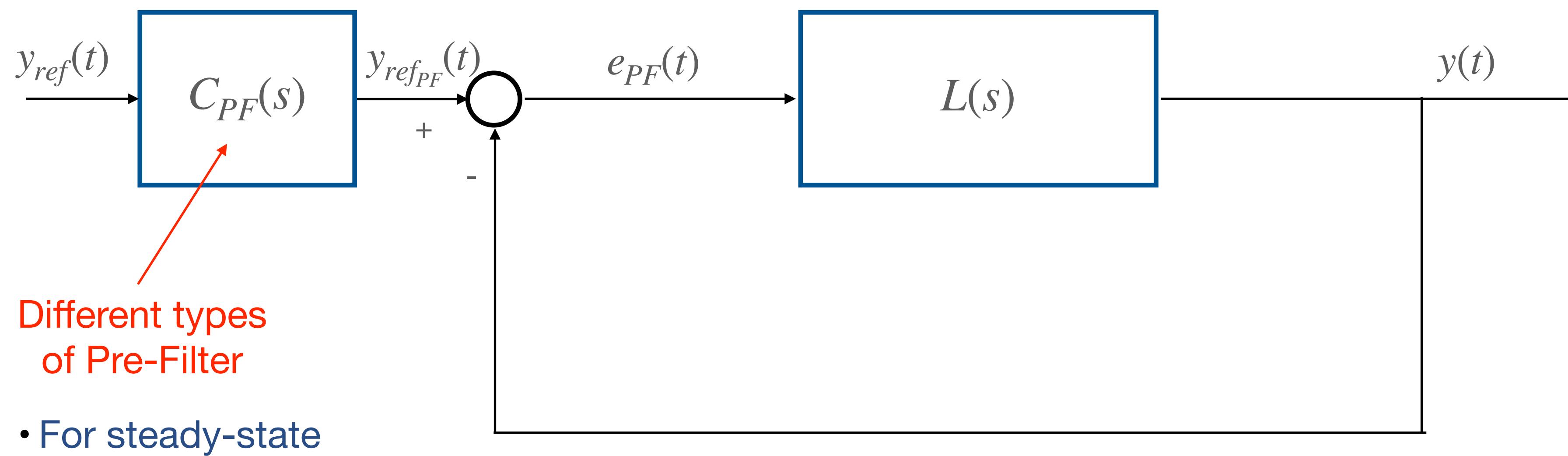
$y \approx y_{ref} : F(0) = 1 \longrightarrow$ Impossible if $L(s)$ is not at least a TYPE 1 transfer function



Pre-filter Based Control Scheme: Case 3



Pre-filter Based Control Scheme: Case 3

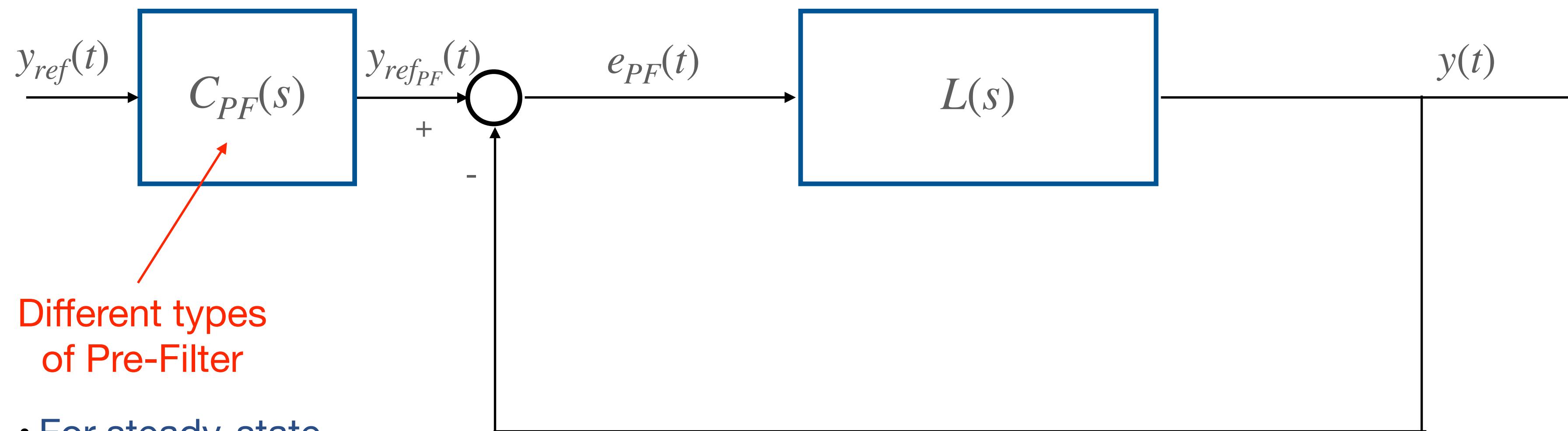


Assumptions:

- $C_{PF}(s)$ As. Stable
- Proper
- ~~Unitary gain~~



Pre-filter Based Control Scheme: Case 3



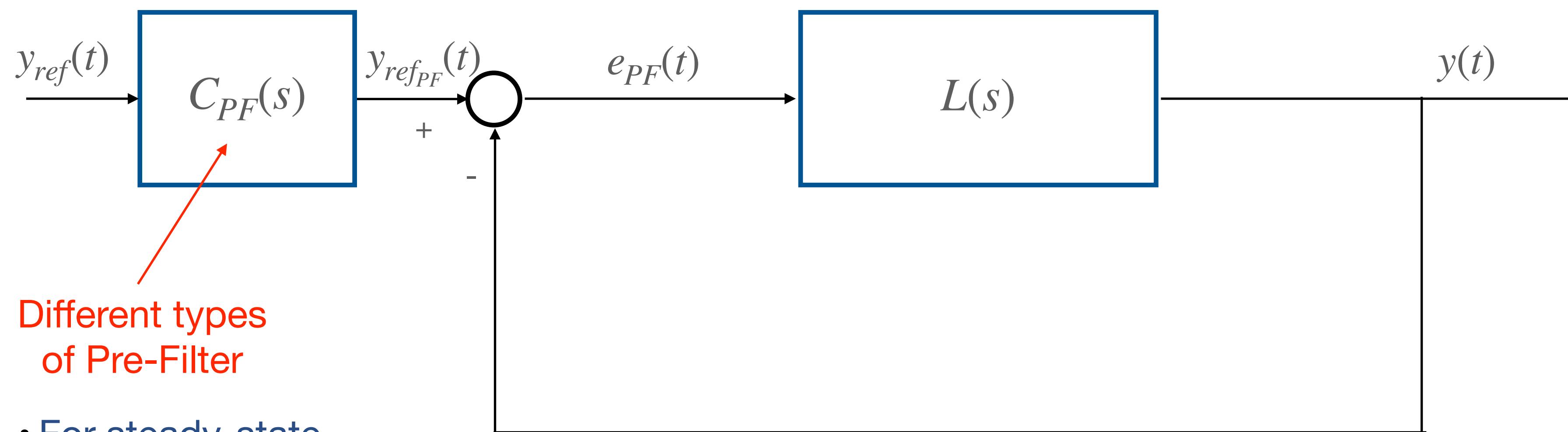
Assumptions:

- $C_{PF}(s)$ As. Stable
- Proper
- ~~Unitary gain~~

Block Algebra Rule: $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$



Pre-filter Based Control Scheme: Case 3



Assumptions:

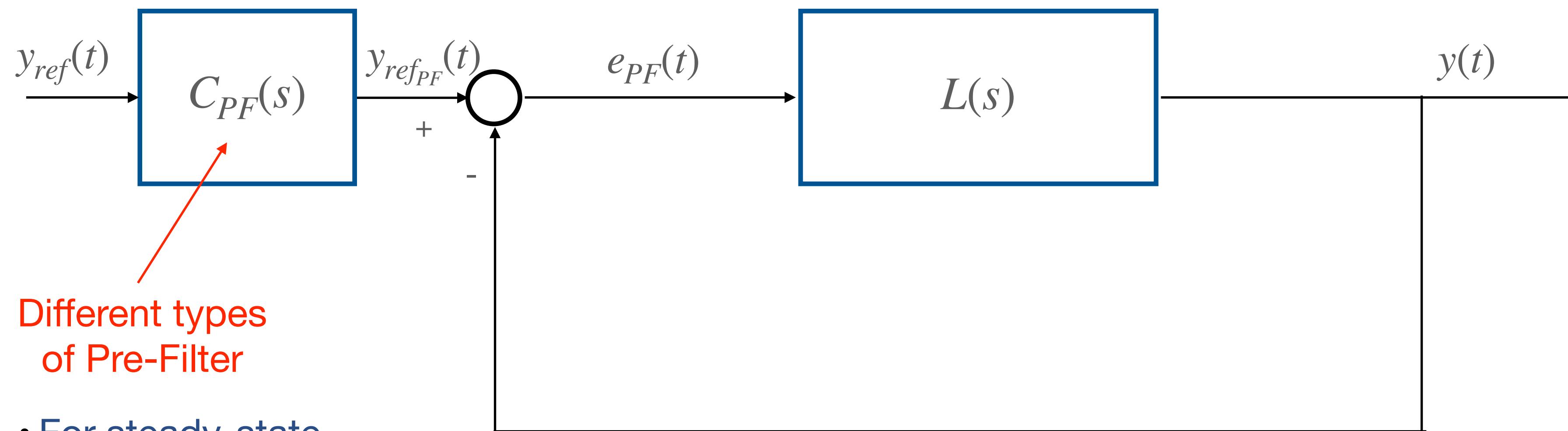
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$$y \approx y_{ref} : C_{PF}(0) \cdot F(0) = 1$$



Pre-filter Based Control Scheme: Case 3



Assumptions:

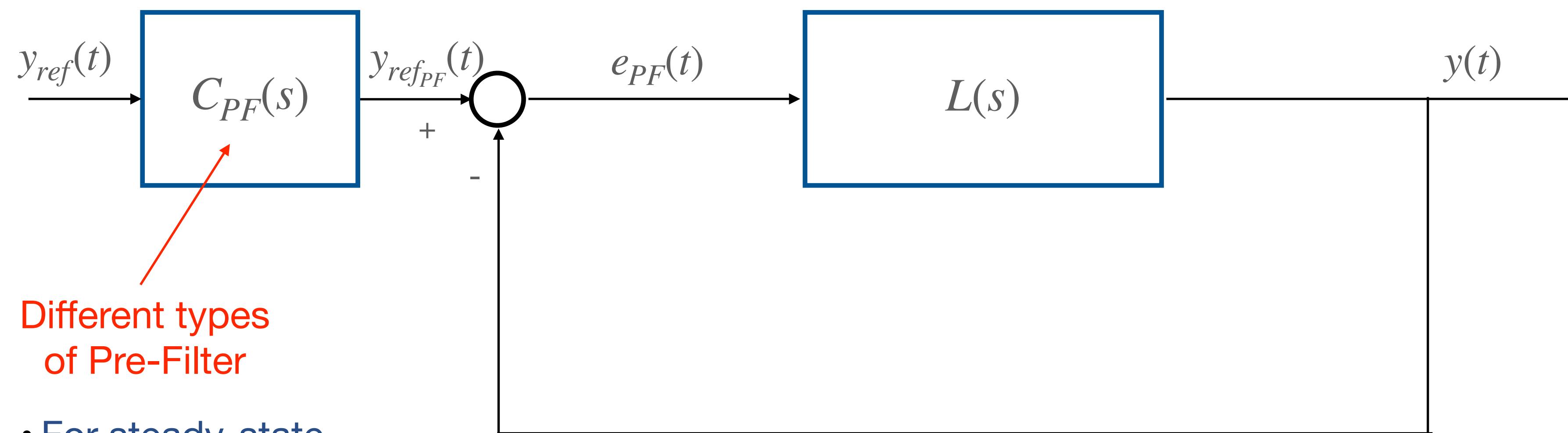
- $C_{PF}(s)$ As. Stable
- Proper
- ~~Unitary gain~~

Block Algebra Rule: $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$

$$y \approx y_{ref} : C_{PF}(0) \cdot F(0) = 1 \longrightarrow C_{PF}(0) = \frac{1}{F(0)}$$



Pre-filter Based Control Scheme: Case 3



Assumptions:

- $C_{PF}(s)$ As. Stable
- Proper
- ~~Unitary gain~~

The filter could be just a constant gain

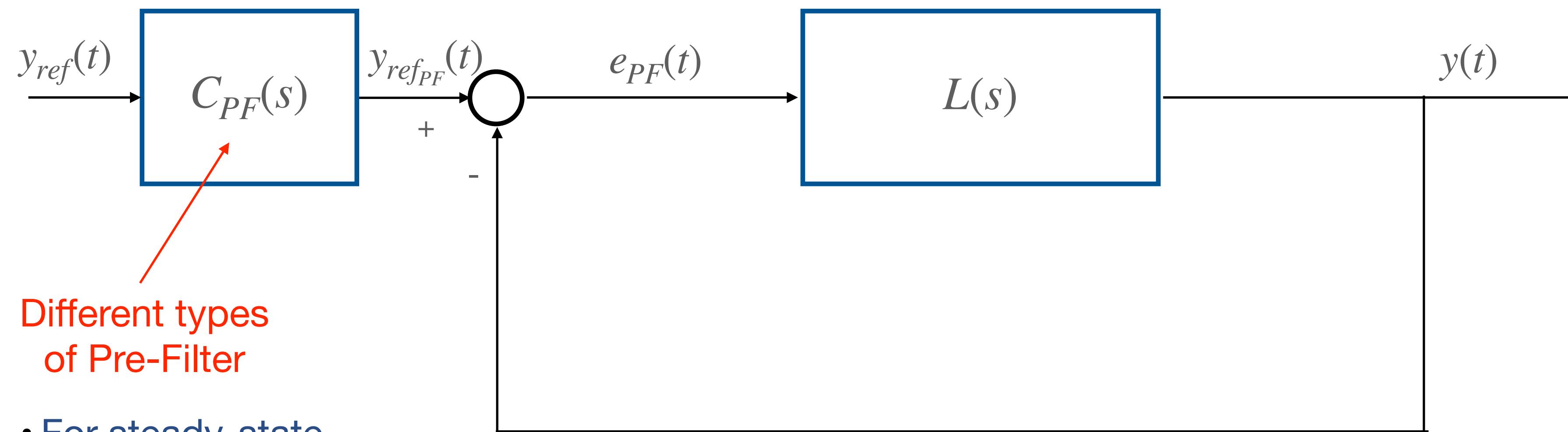
Block Algebra Rule: $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$

$$y \approx y_{ref} : C_{PF}(0) \cdot F(0) = 1$$

$$C_{PF}(0) = \frac{1}{F(0)}$$



Pre-filter Based Control Scheme: Case 3



Assumptions:

- $C_{PF}(s)$ As. Stable
- Proper
- ~~Unitary gain~~

Warning:
Use only when strictly necessary!

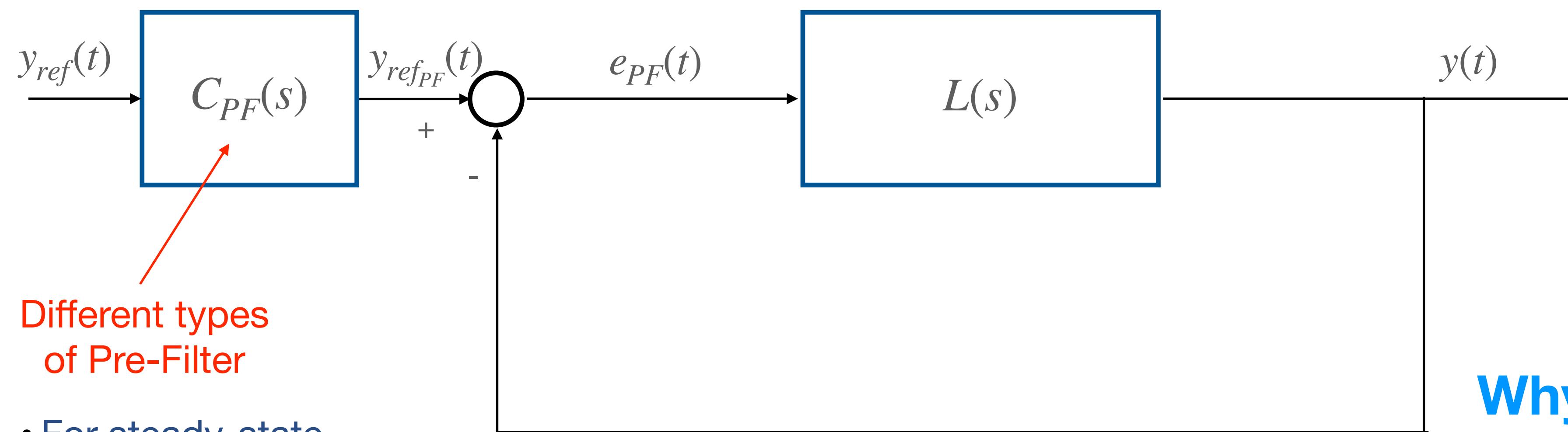
Block Algebra Rule: $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$

$$y \approx y_{ref} : C_{PF}(0) \cdot F(0) = 1$$

$$C_{PF}(0) = \frac{1}{F(0)}$$



Pre-filter Based Control Scheme: Case 3



Assumptions:

- $C_{PF}(s)$ As. Stable
- Proper
- ~~Unitary gain~~

Why?

Warning:
Use only when strictly necessary!

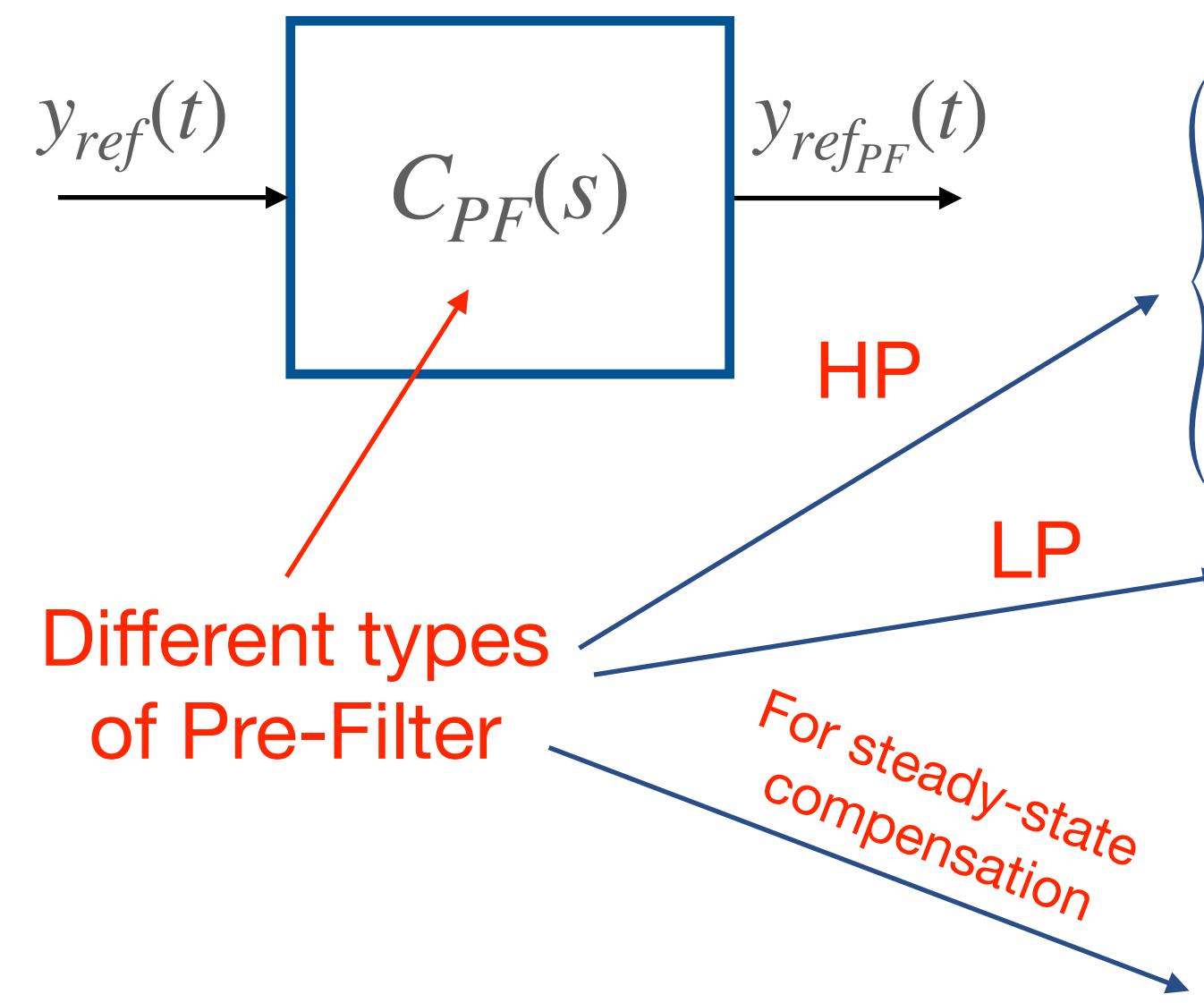
Block Algebra Rule: $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$

$$y \approx y_{ref} : \quad C_{PF}(0) \cdot F(0) = 1$$

$$C_{PF}(0) = \frac{1}{F(0)}$$



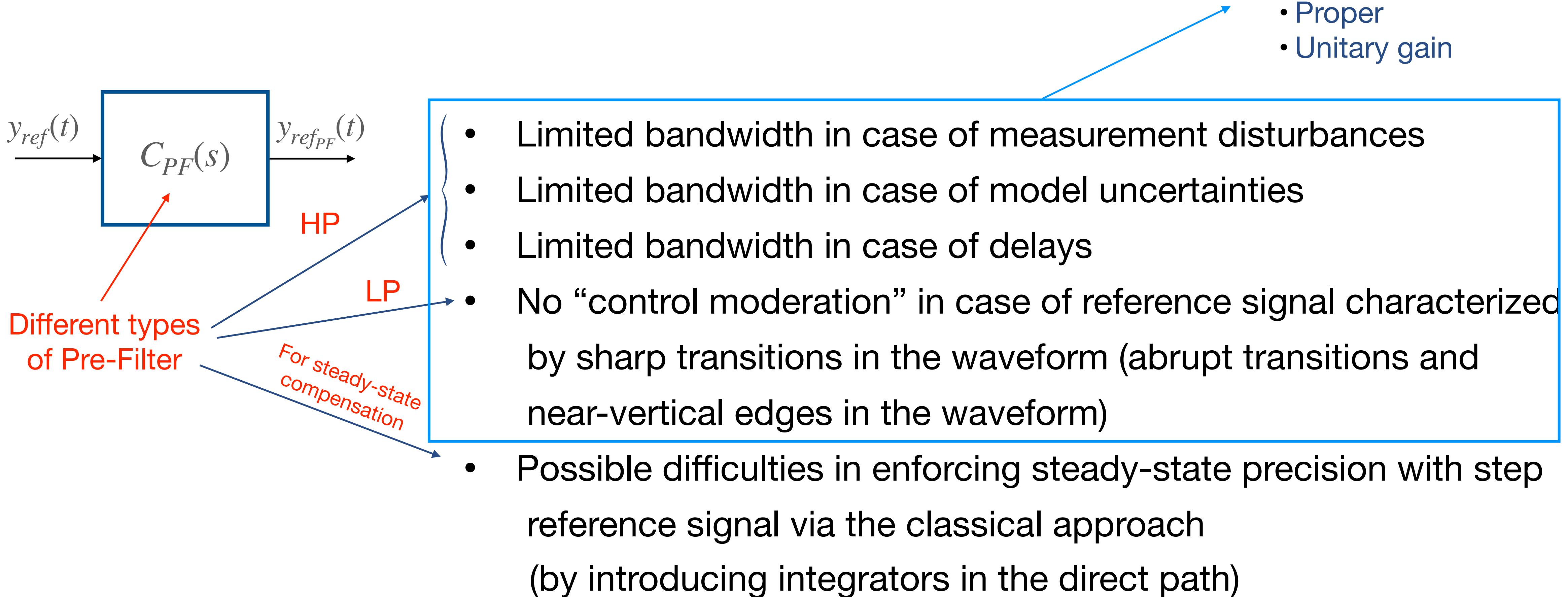
Pre-filter Based Control Scheme: Summary



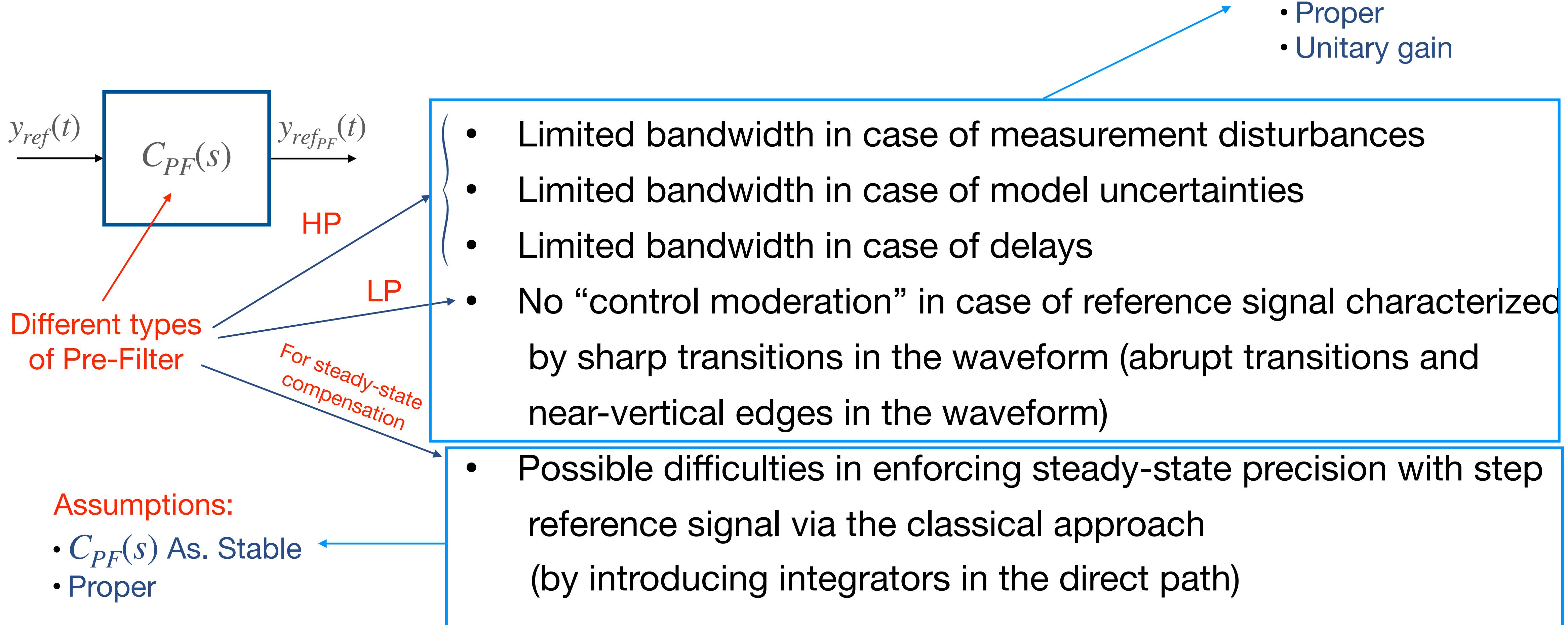
- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays
- No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
- Possible difficulties in enforcing steady-state precision with step reference signal via the classical approach
(by introducing integrators in the direct path)



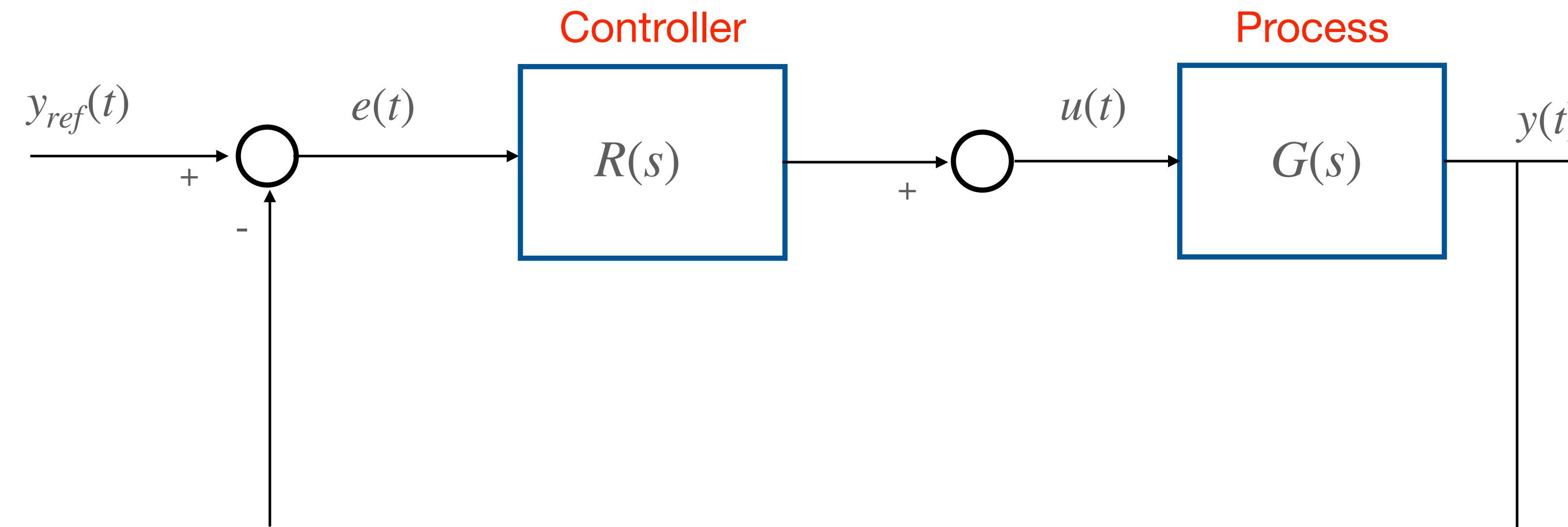
Pre-filter Based Control Scheme: Summary



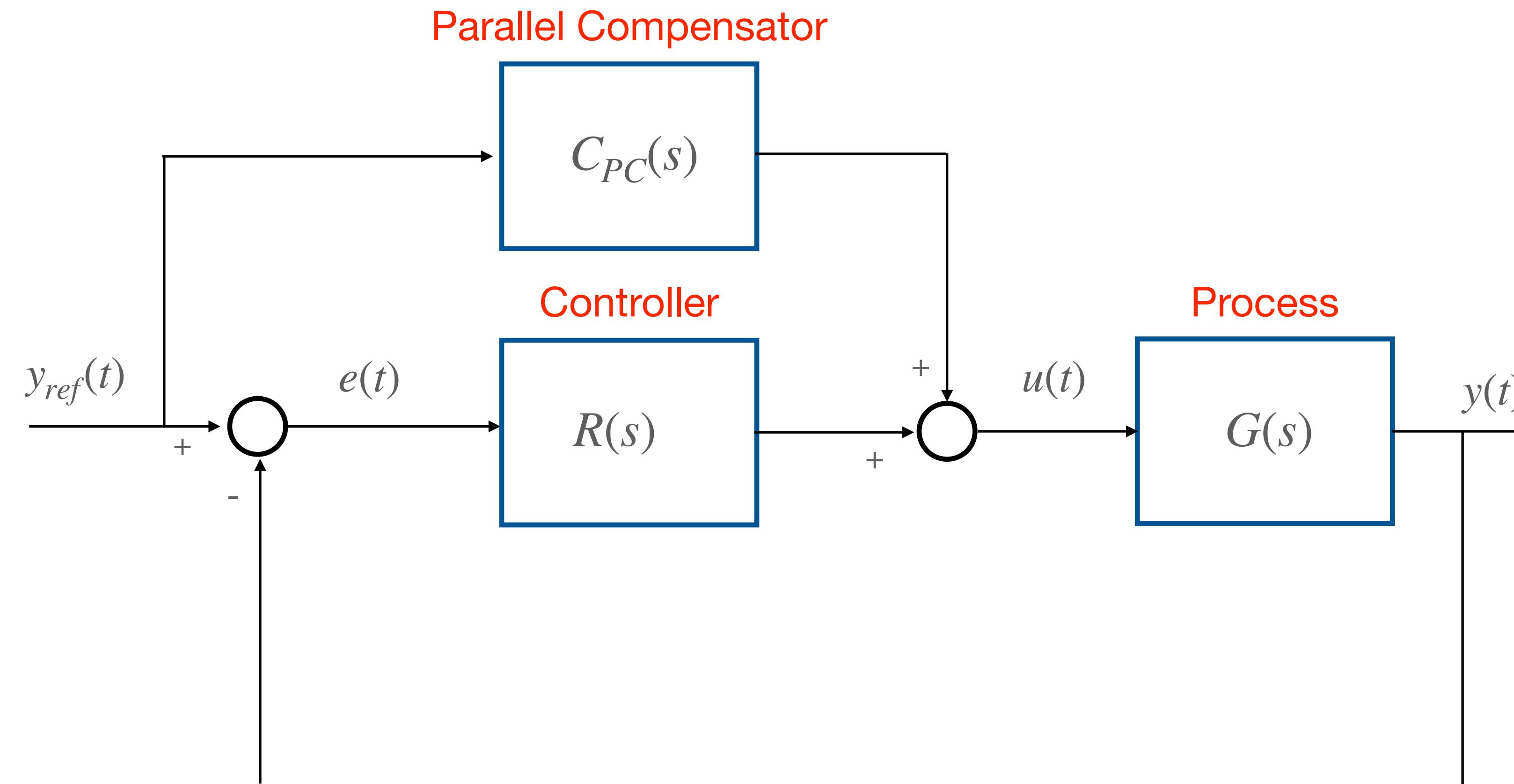
Pre-filter Based Control Scheme: Summary



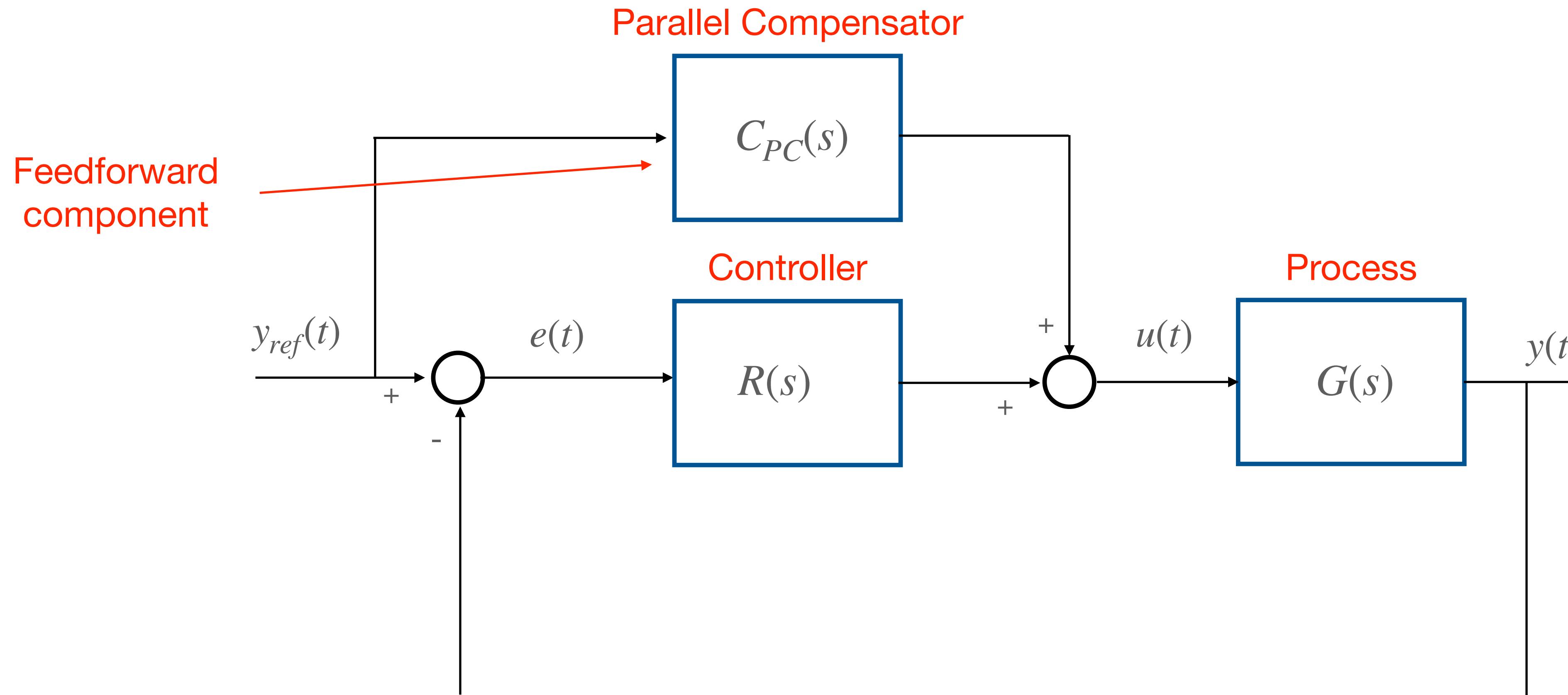
Parallel-compensator Based Control Scheme



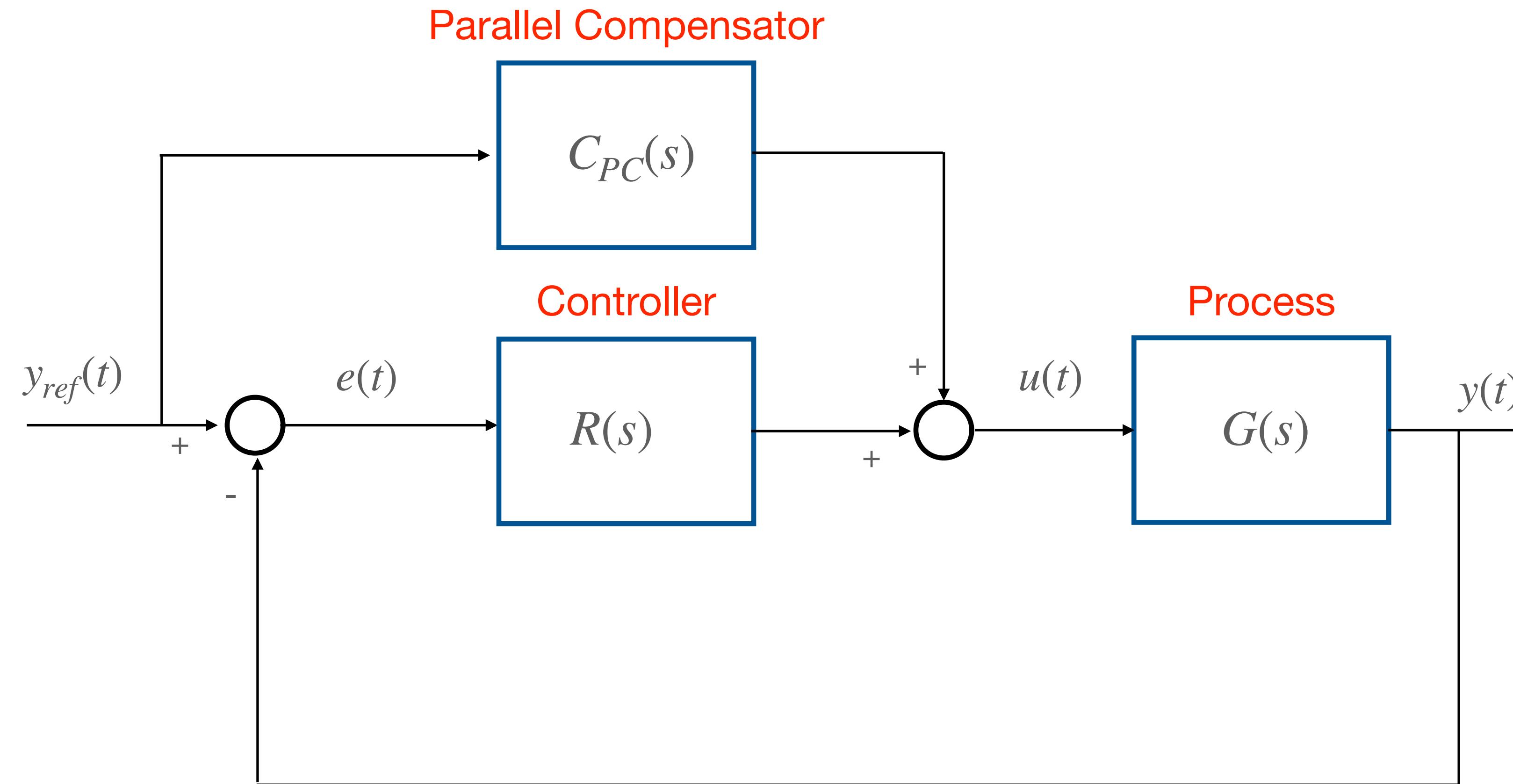
Parallel-compensator Based Control Scheme



Parallel-compensator Based Control Scheme



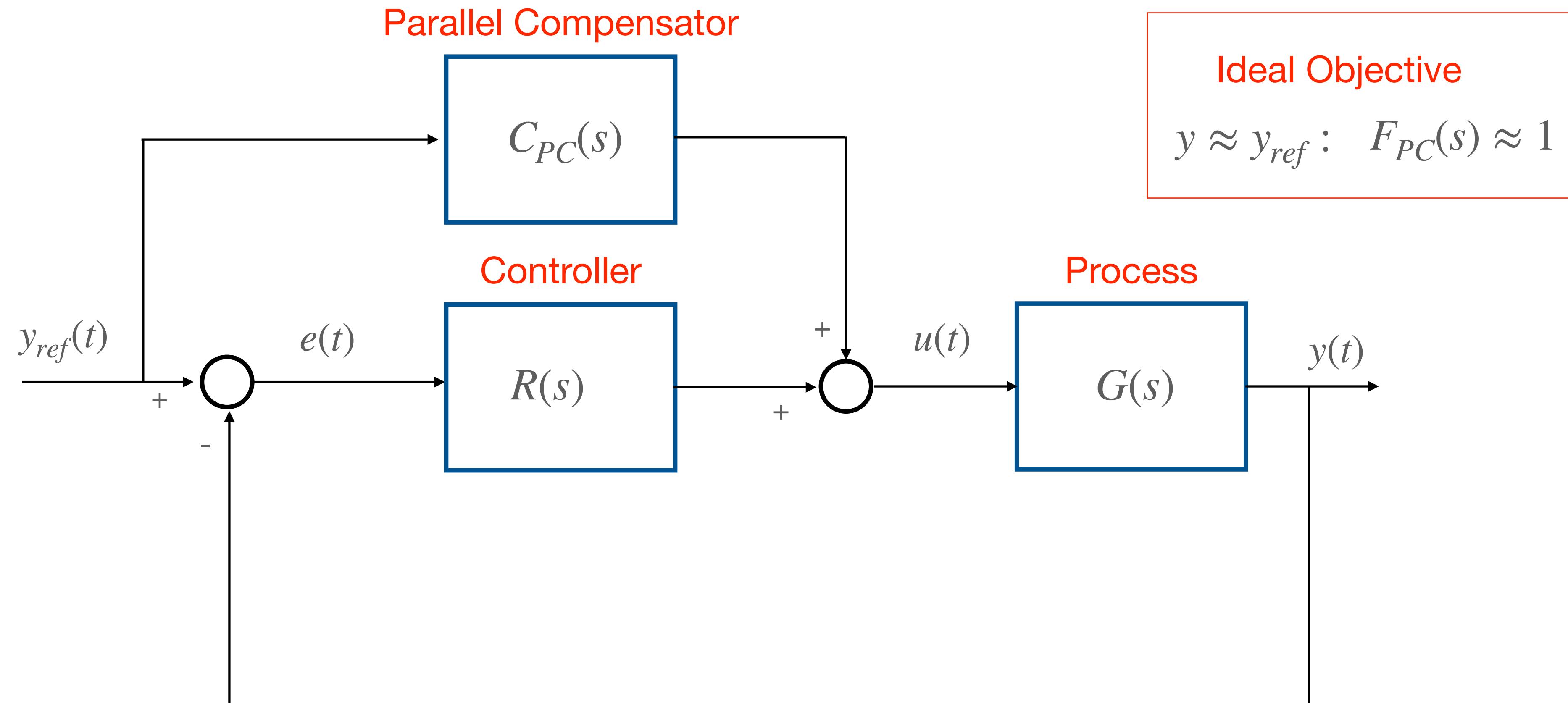
Parallel-compensator Based Control Scheme



$$y_{ref} \rightarrow y : F_{PC}(s) = \frac{C_{PC}(s)G(s) + R(s)G(s)}{1 + R(s)G(s)}$$



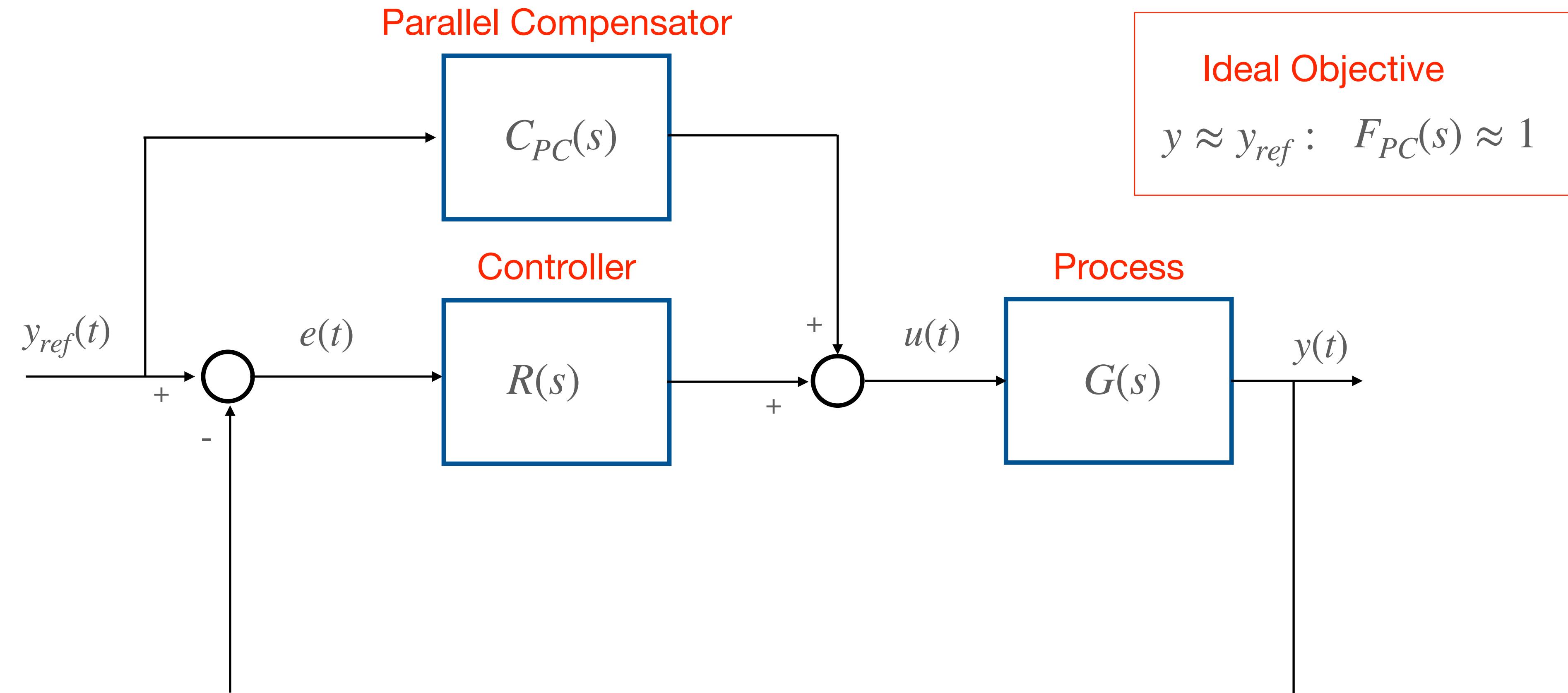
Parallel-compensator Based Control Scheme



$$y_{ref} \rightarrow y : F_{PC}(s) = \frac{C_{PC}(s)G(s) + R(s)G(s)}{1 + R(s)G(s)}$$



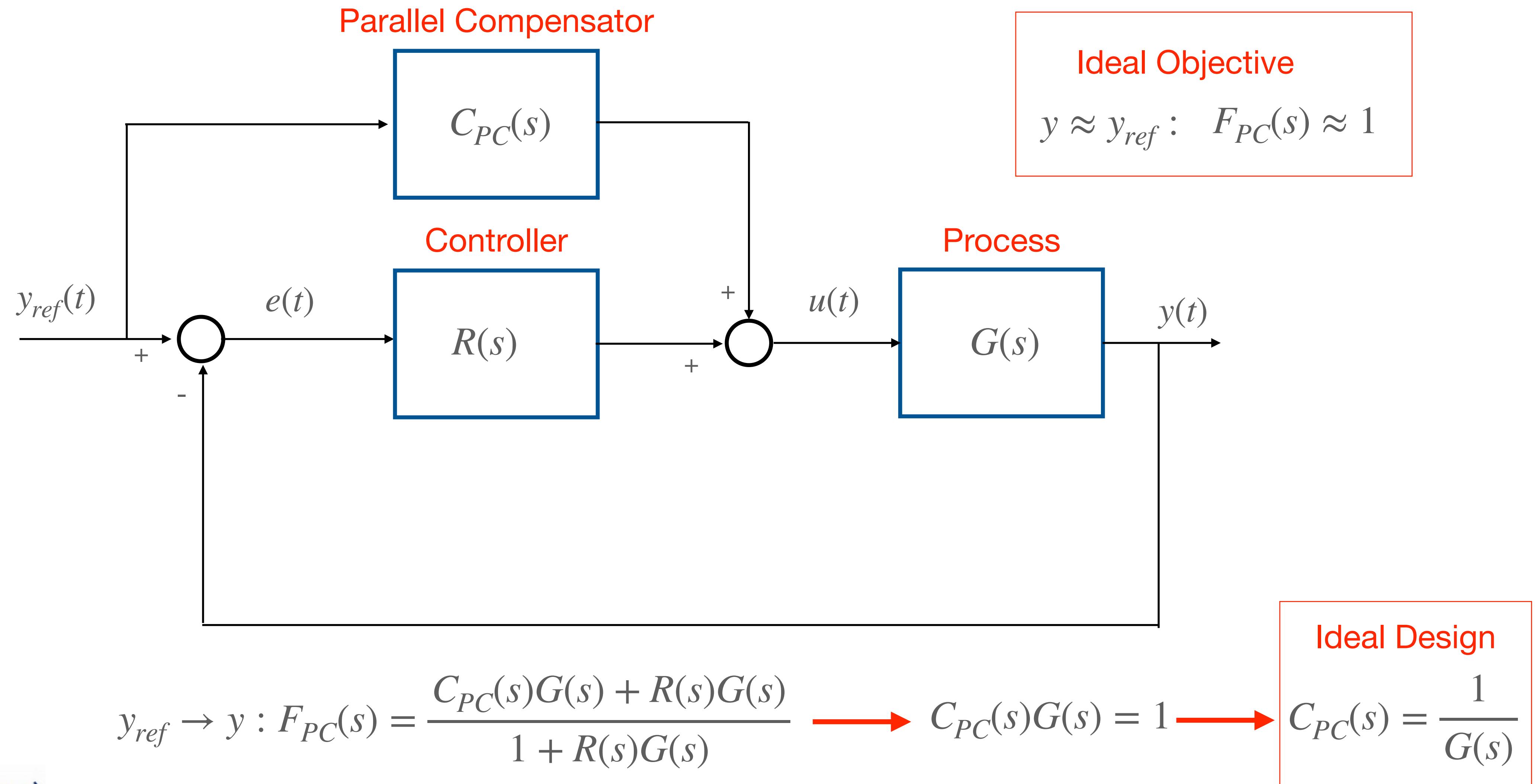
Parallel-compensator Based Control Scheme



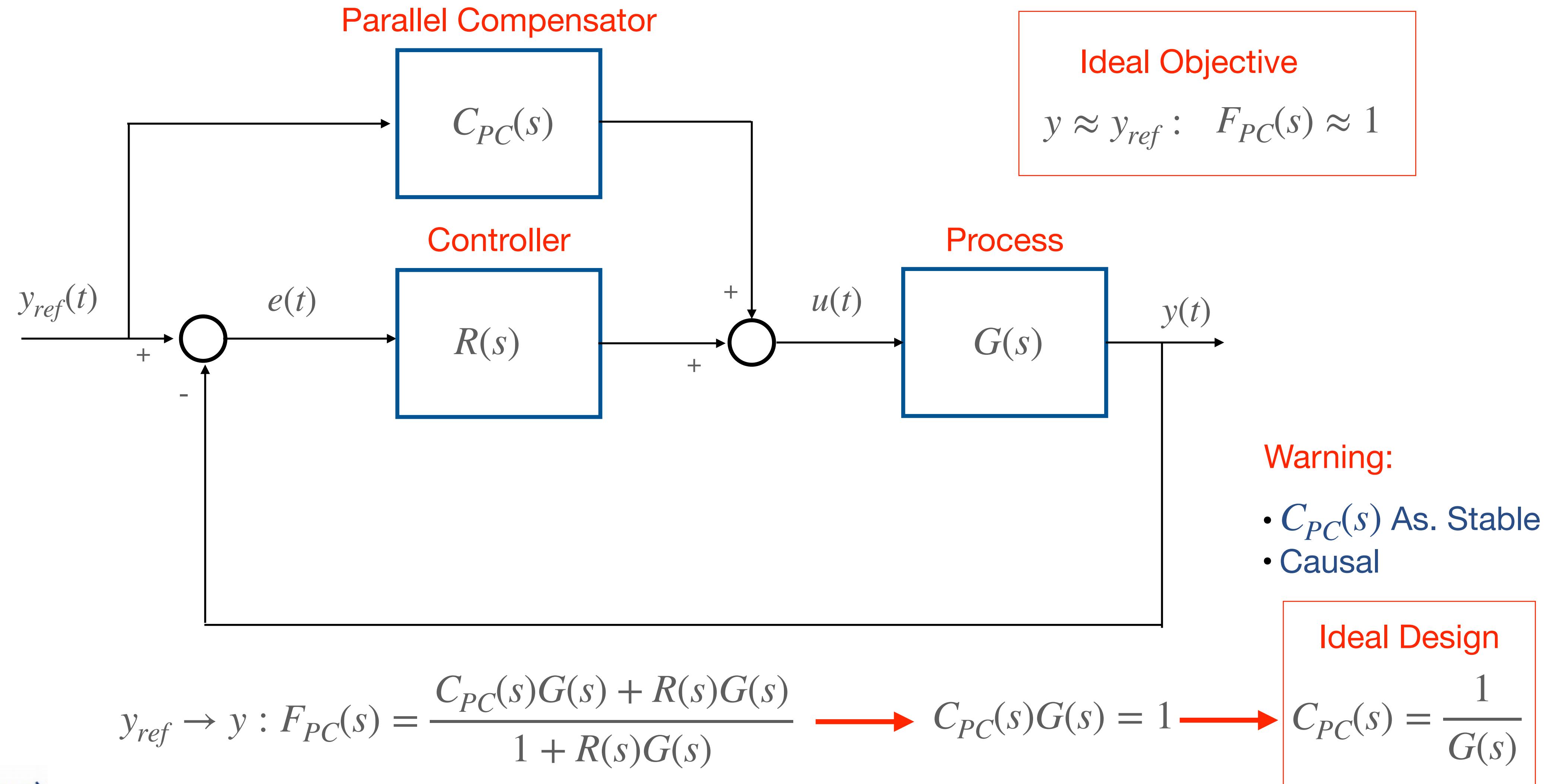
$$y_{ref} \rightarrow y : F_{PC}(s) = \frac{C_{PC}(s)G(s) + R(s)G(s)}{1 + R(s)G(s)} \longrightarrow C_{PC}(s)G(s) = 1$$



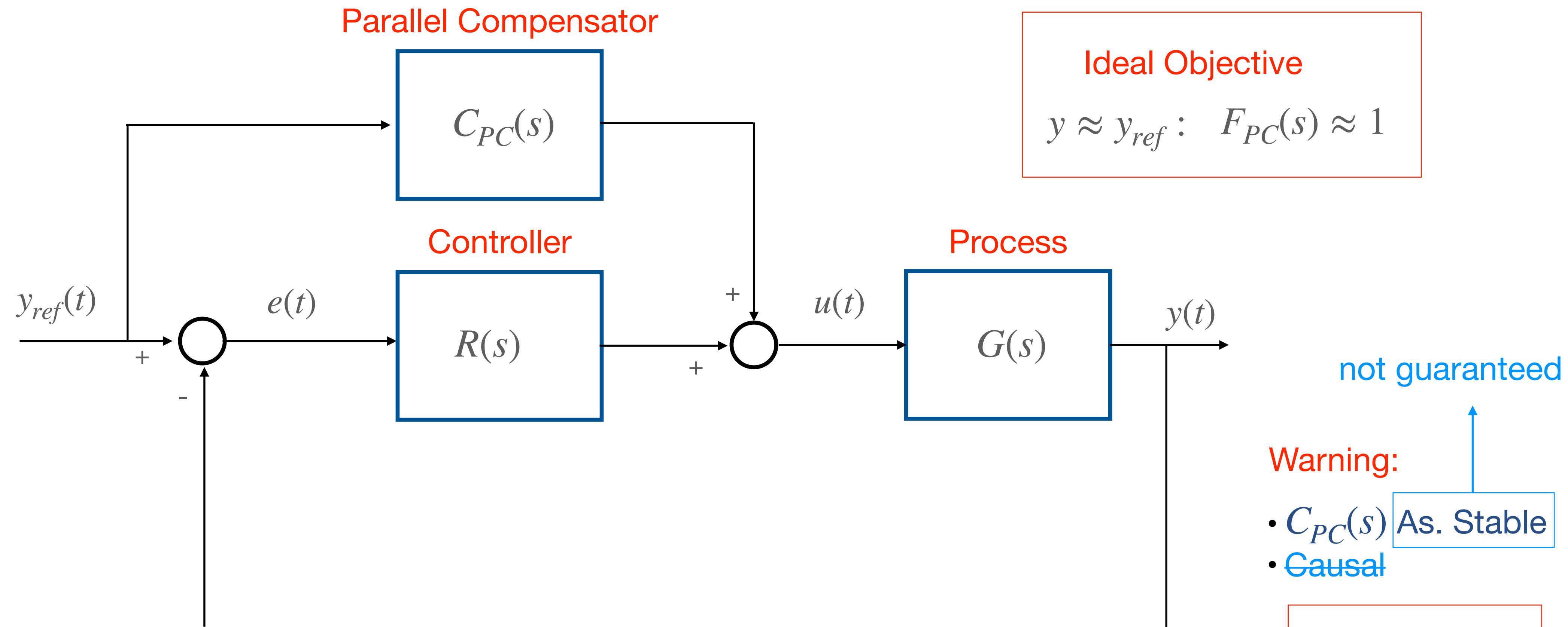
Parallel-compensator Based Control Scheme



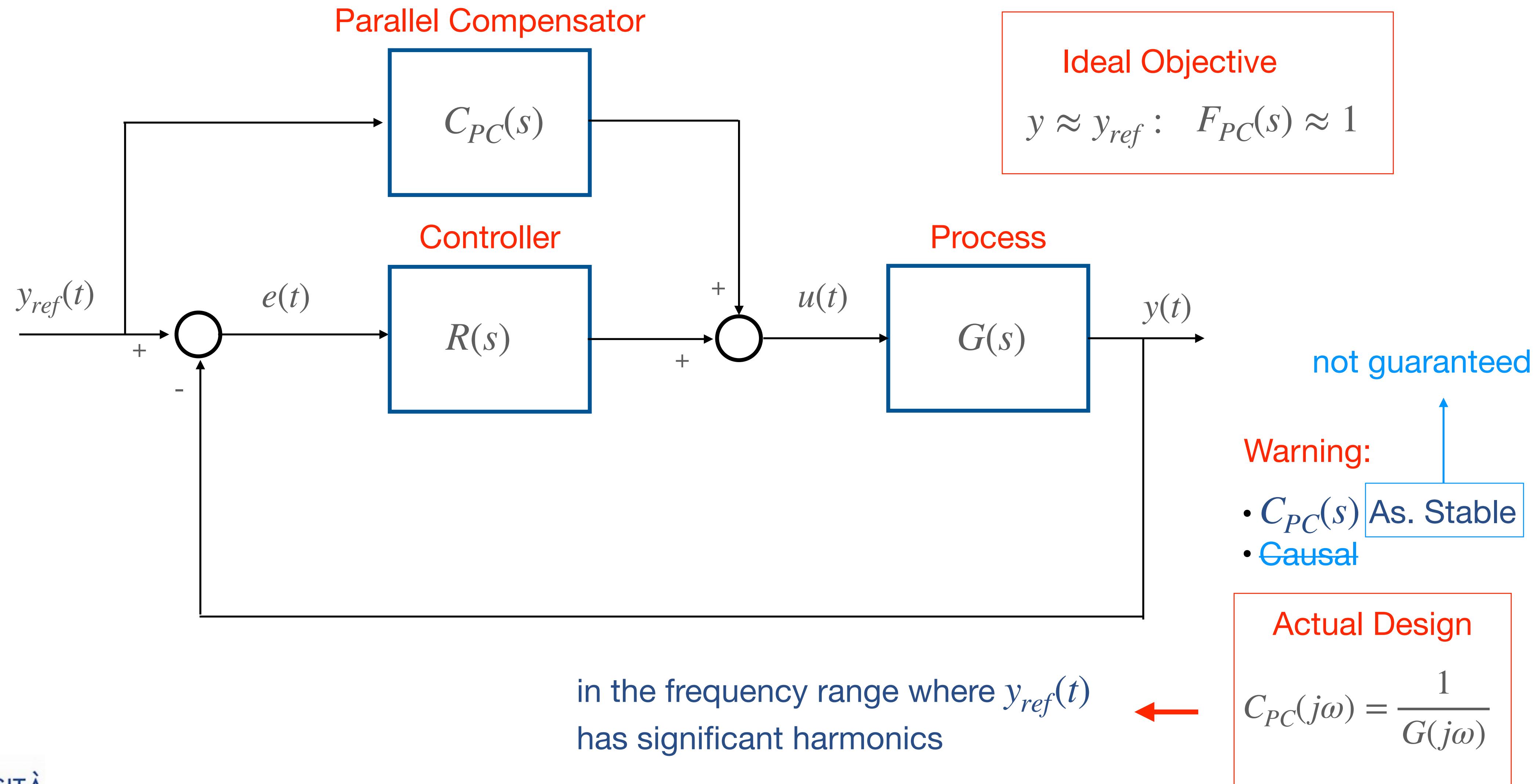
Parallel-compensator Based Control Scheme



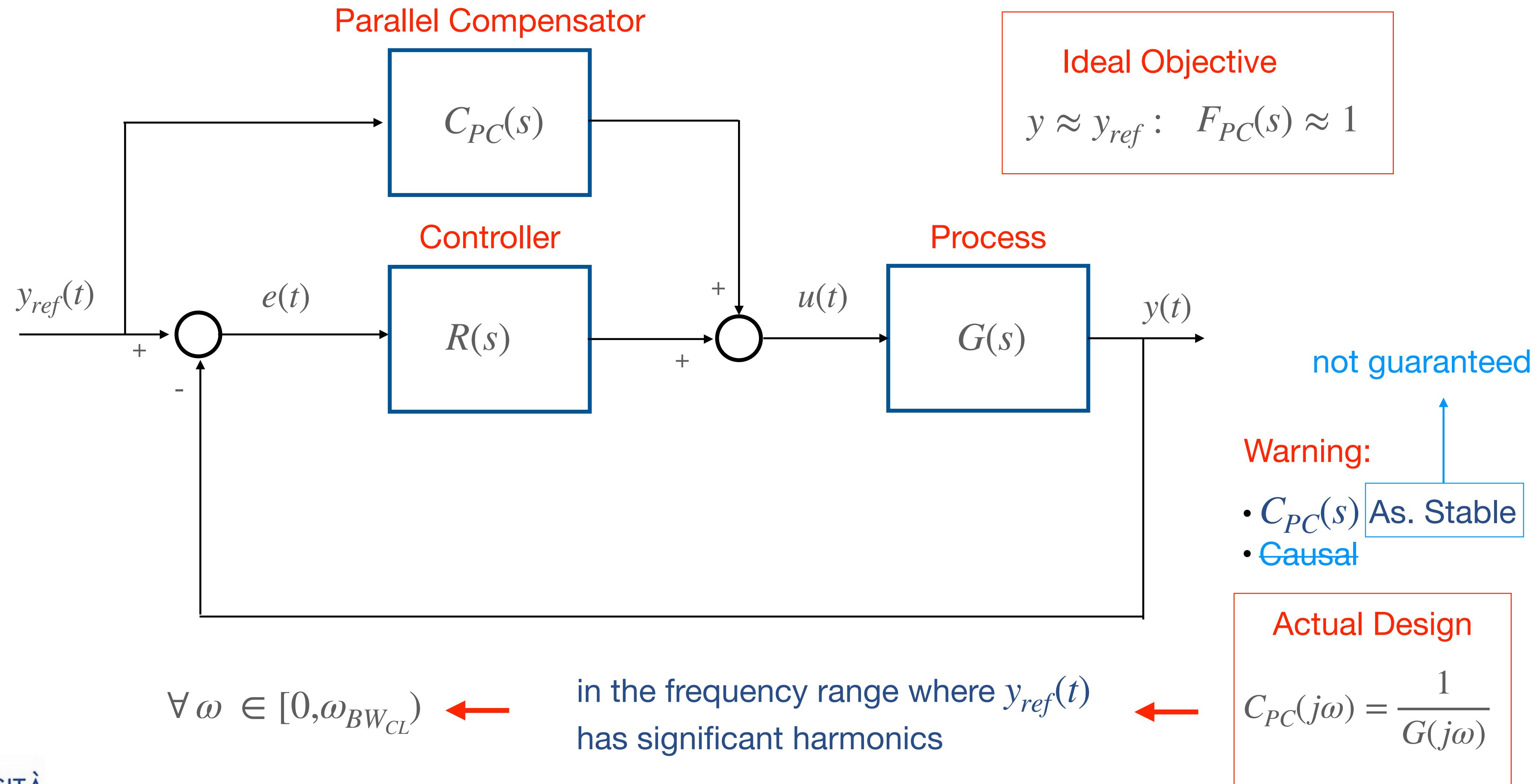
Parallel-compensator Based Control Scheme



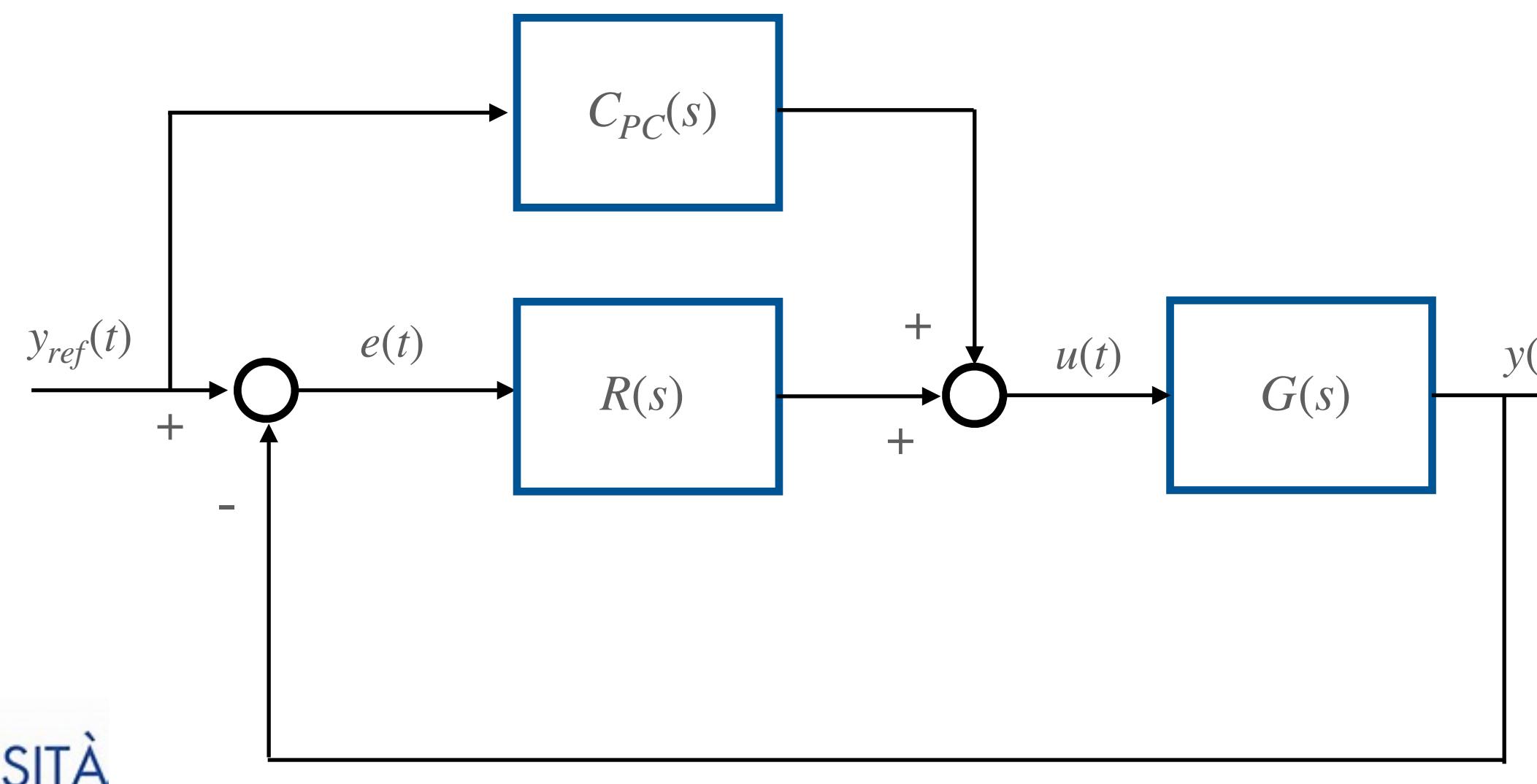
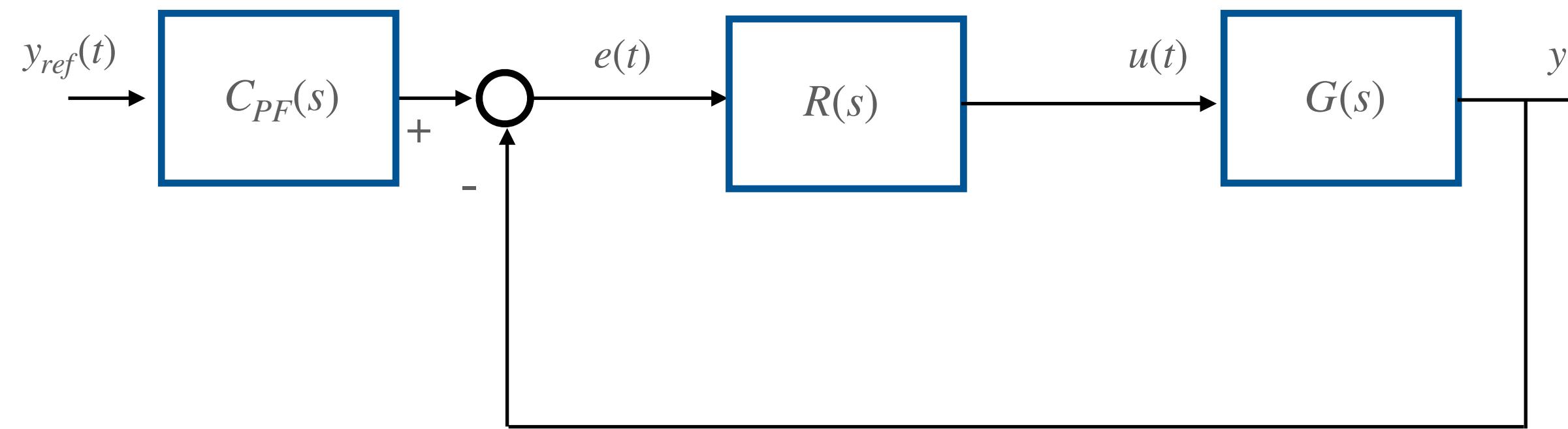
Parallel-compensator Based Control Scheme



Parallel-compensator Based Control Scheme

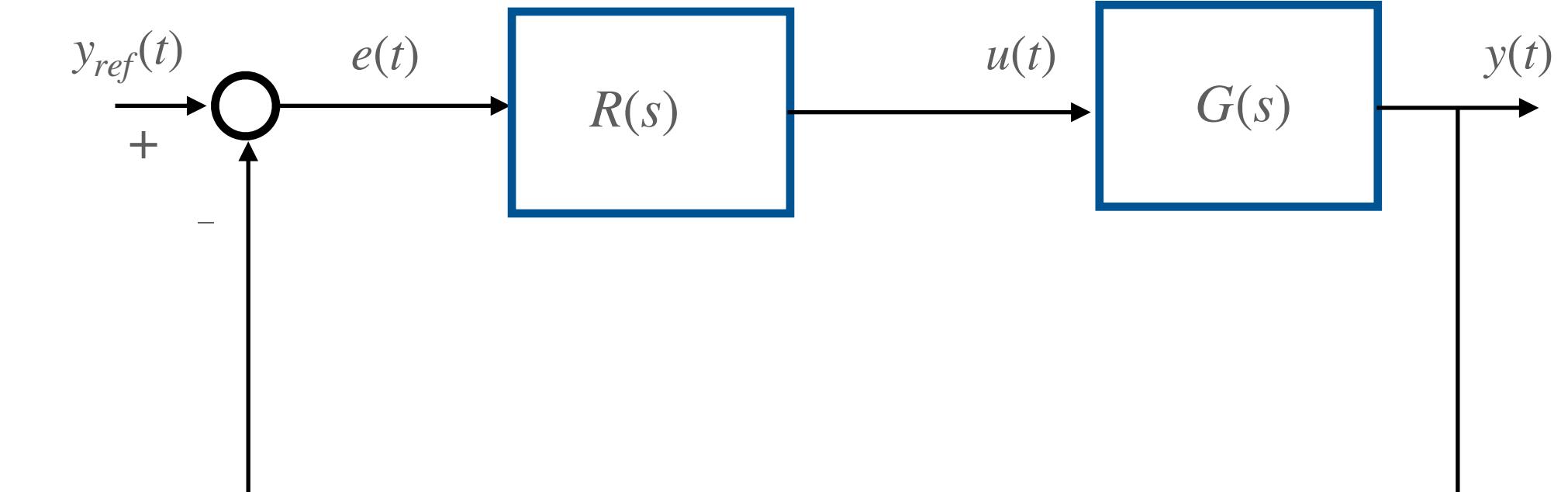
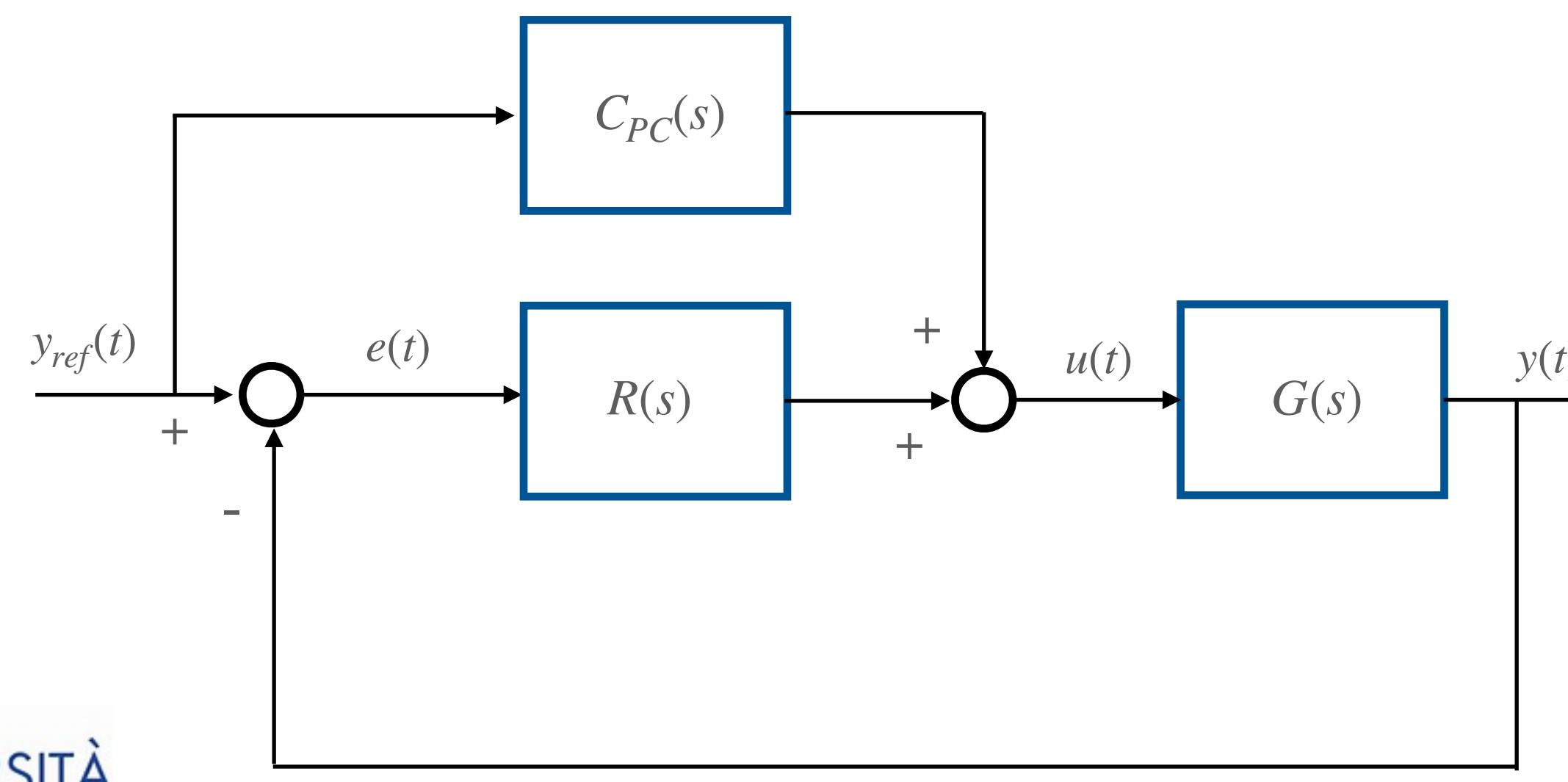
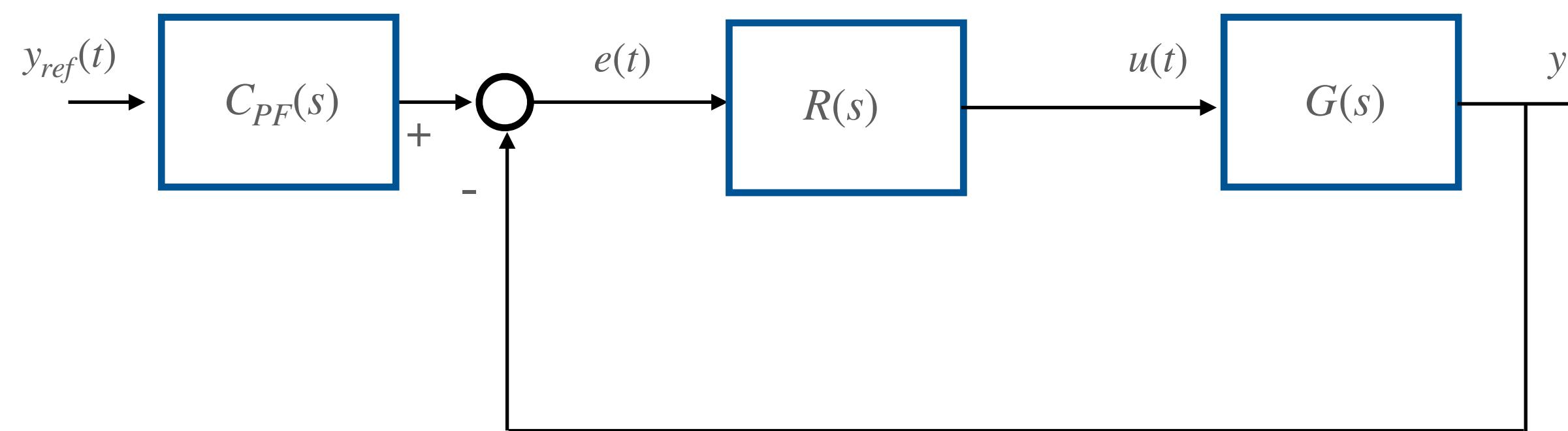


Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes

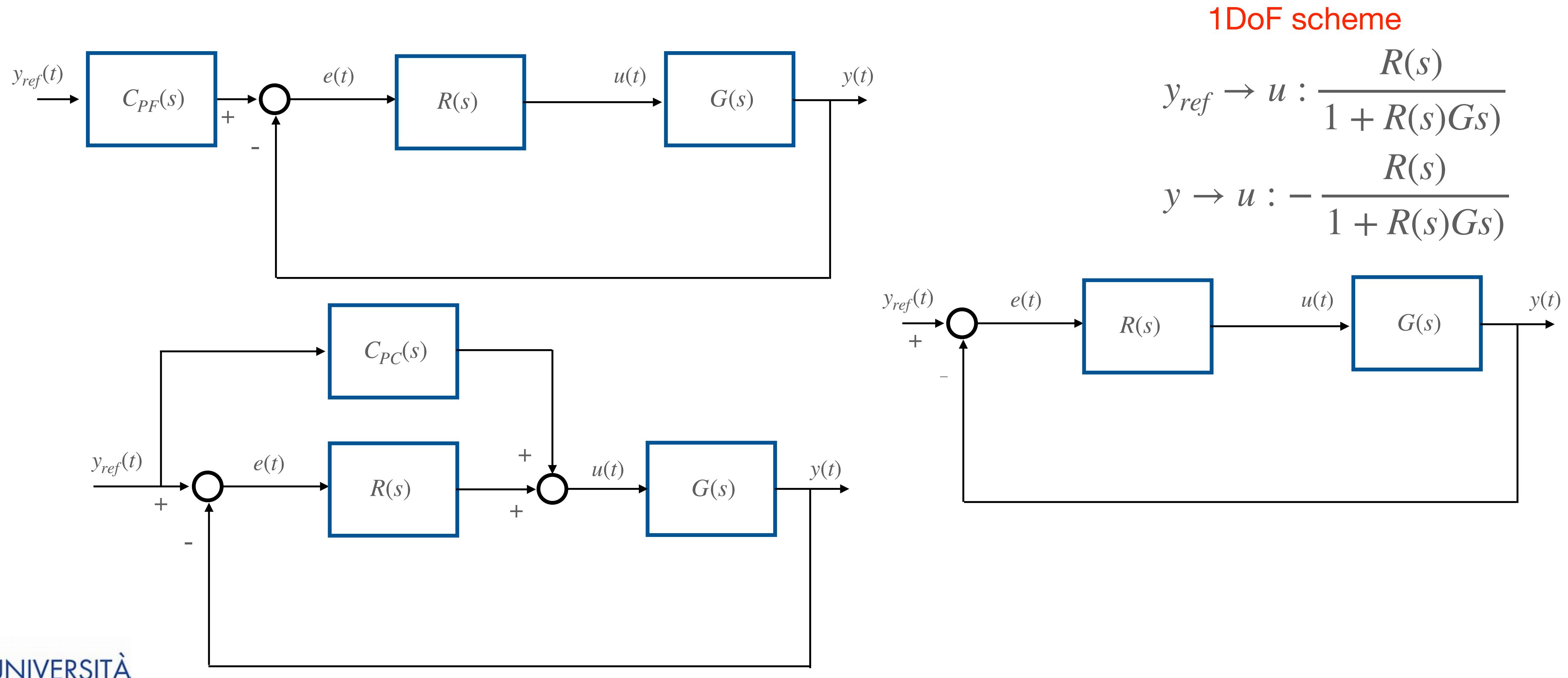


Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes

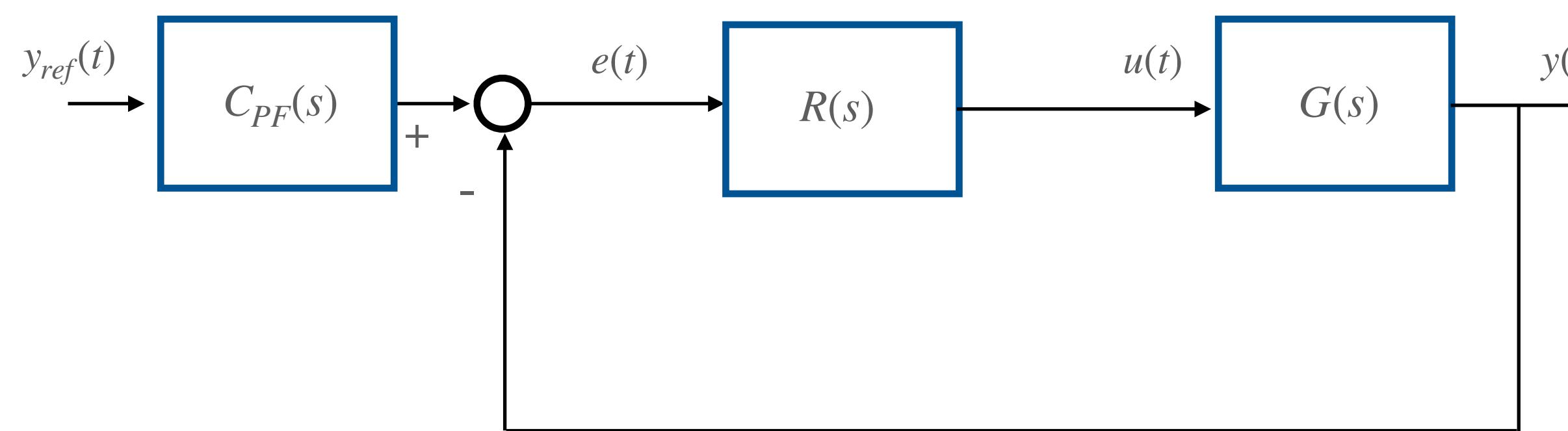
1DoF scheme



Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



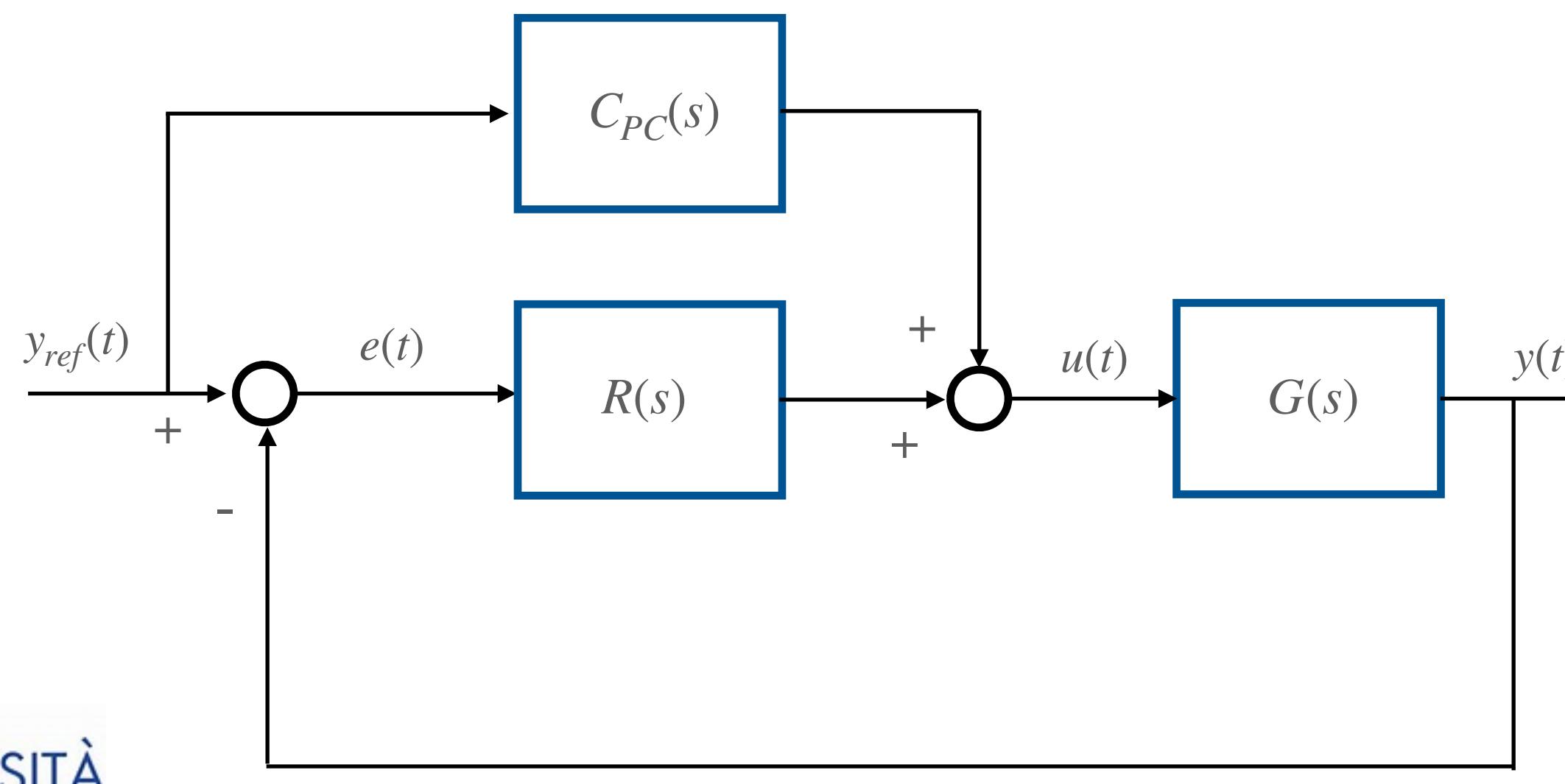
Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



2DoF scheme

$$y_{ref} \rightarrow u : \frac{C_{PF}(s)R(s)}{1 + R(s)Gs}$$

$$y \rightarrow u : -\frac{R(s)}{1 + R(s)Gs}$$



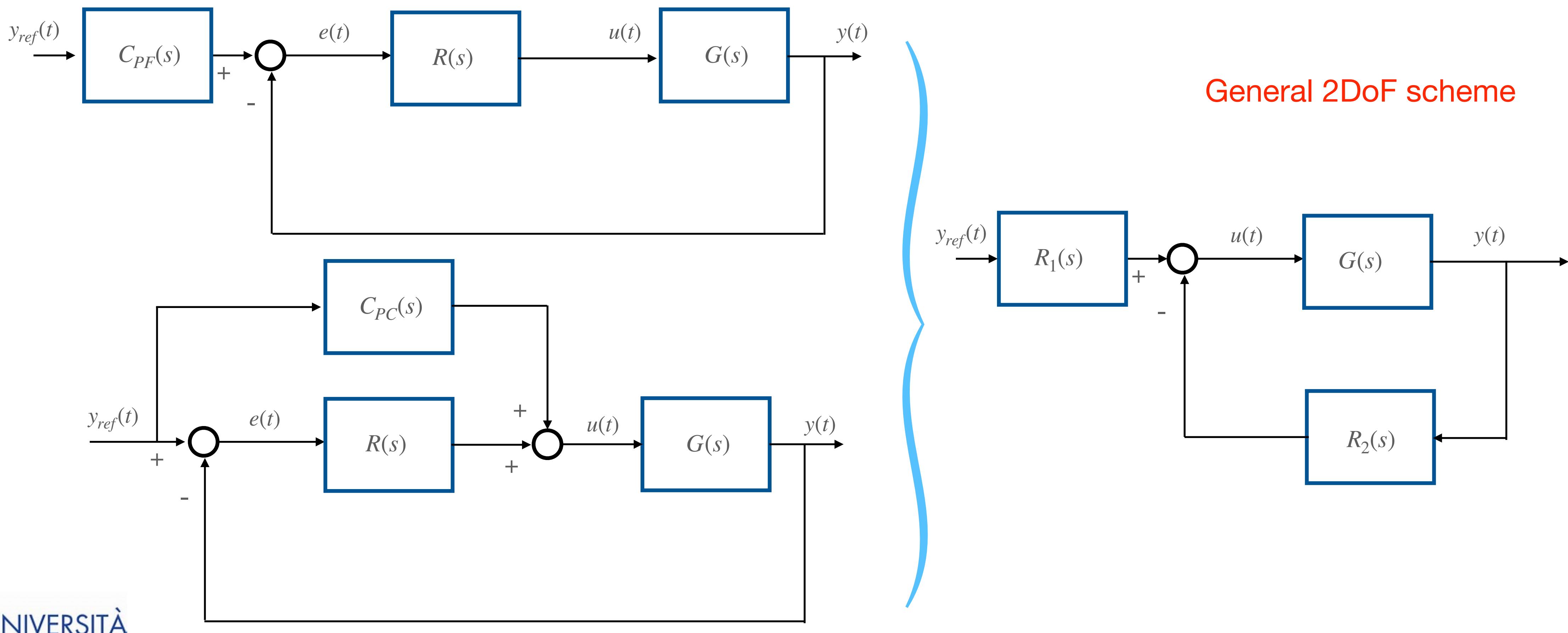
2DoF scheme

$$y_{ref} \rightarrow u : \frac{C_{PC}(s) + R(s)}{1 + R(s)Gs}$$

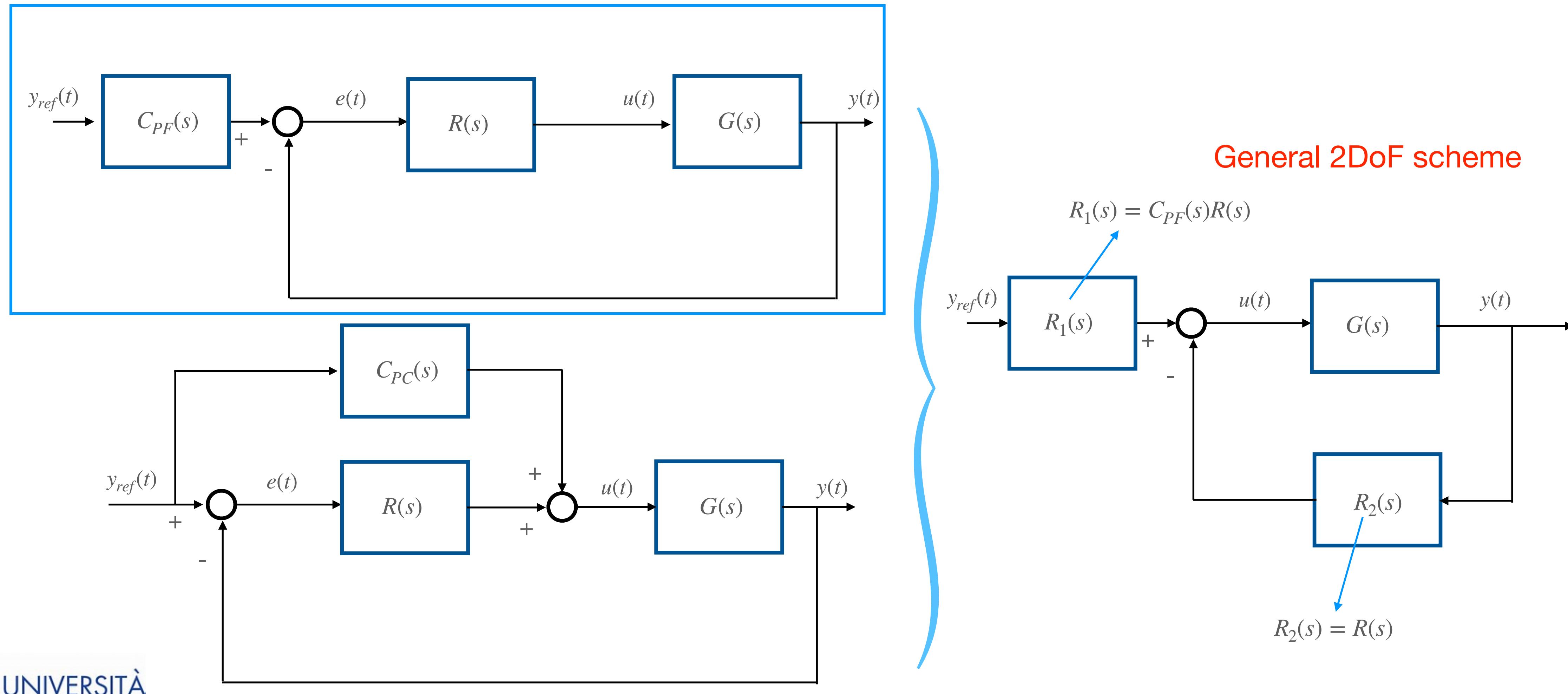
$$y \rightarrow u : -\frac{R(s)}{1 + R(s)Gs}$$



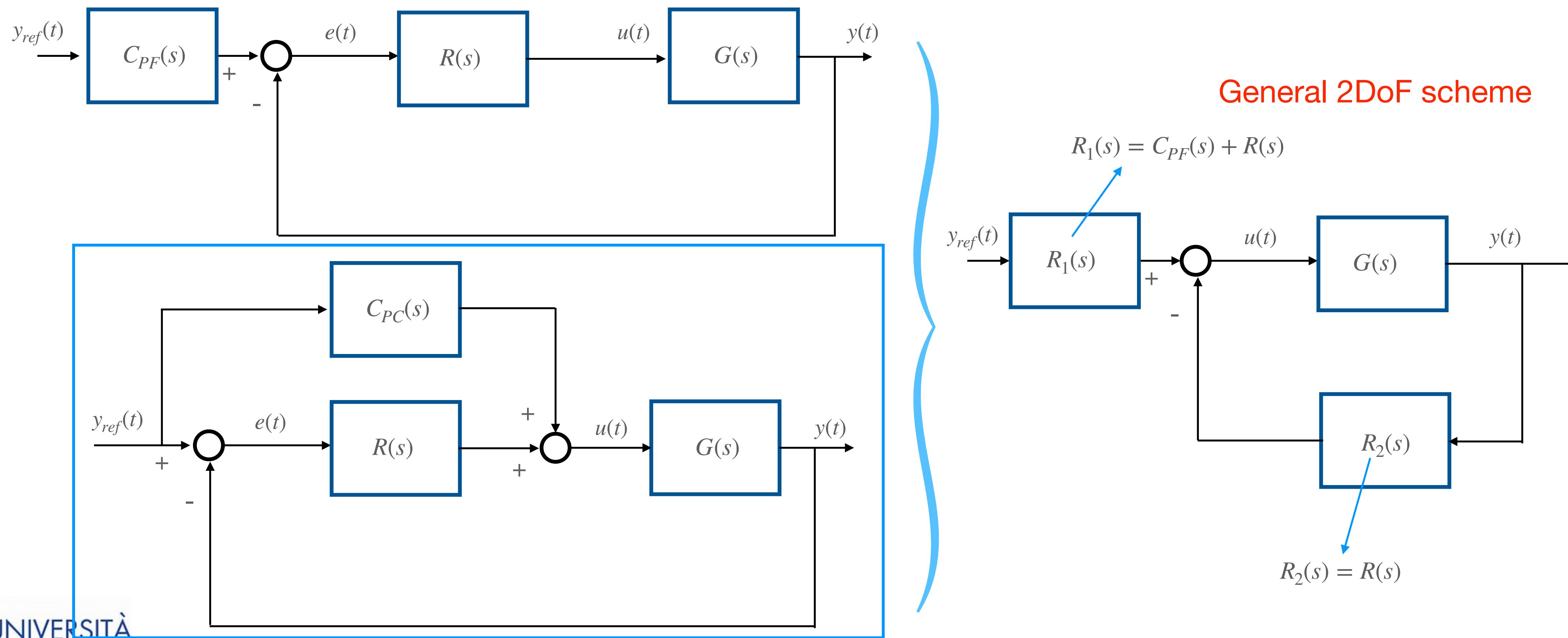
Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



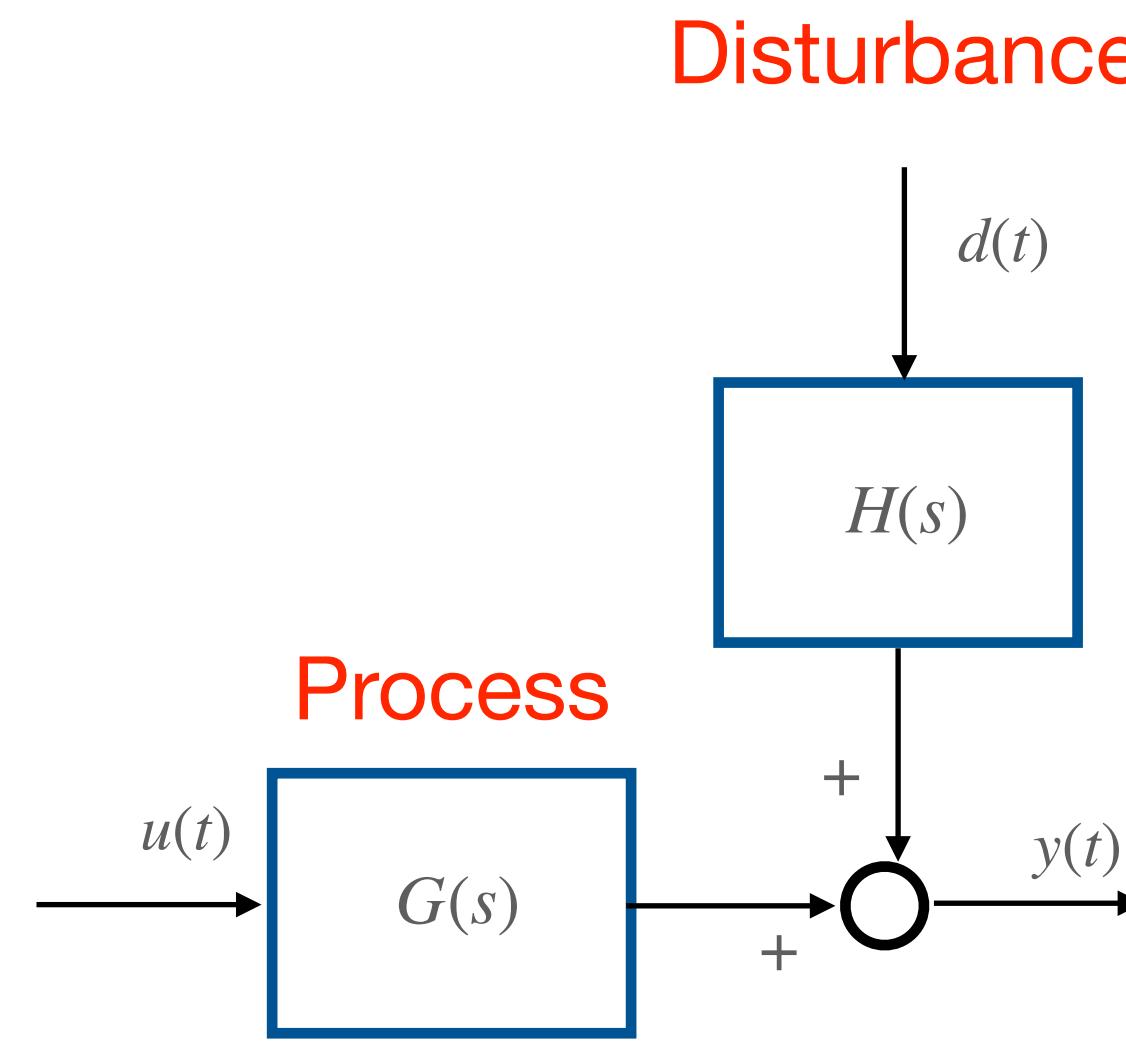
Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



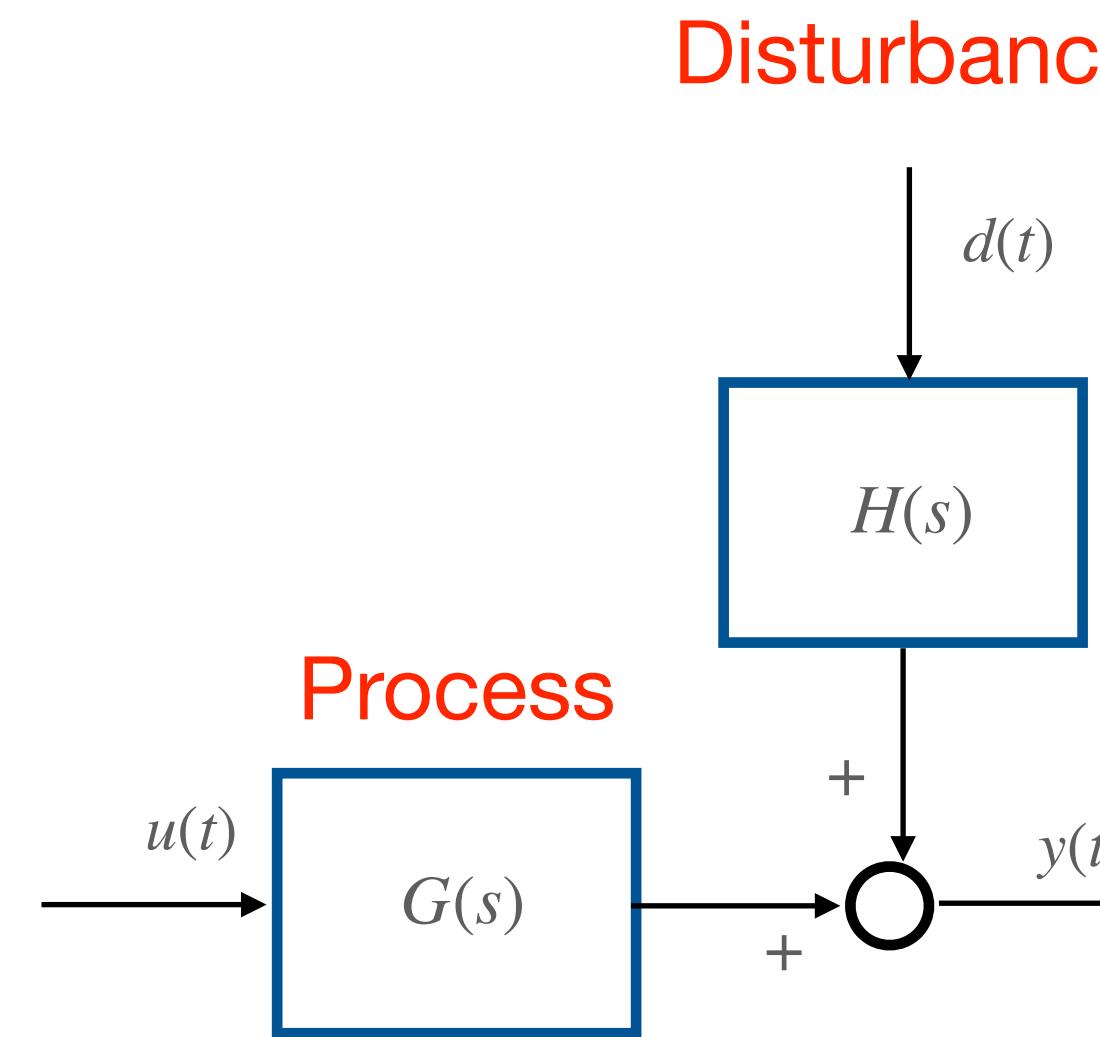
Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



Control Scheme with Measurable Disturbance Compensation



Control Scheme with Measurable Disturbance Compensation



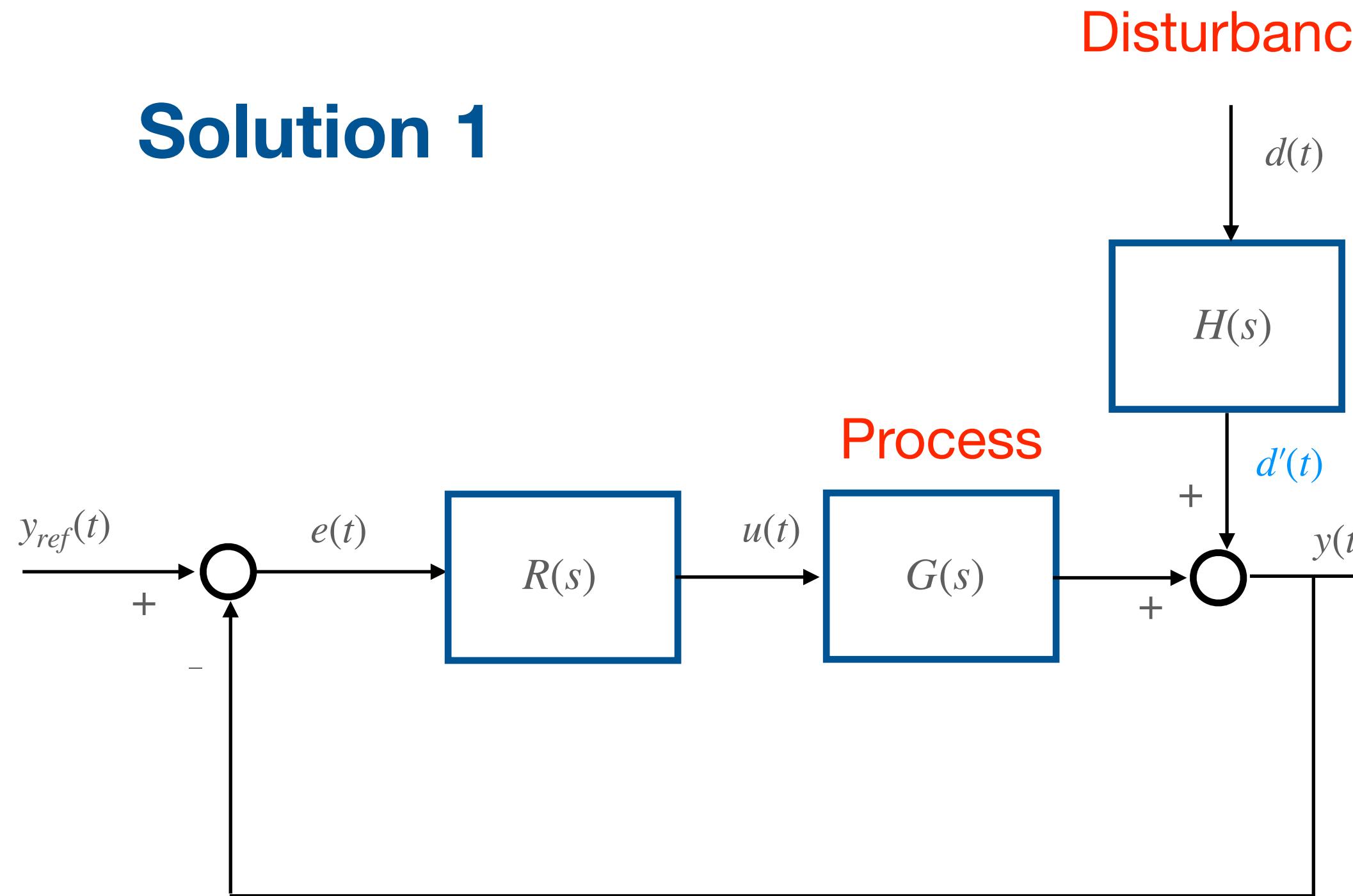
Assumptions:

- $d(t)$ measurable
- $H(s)$ known (possibly 1)



Control Scheme with Measurable Disturbance Compensation

Solution 1

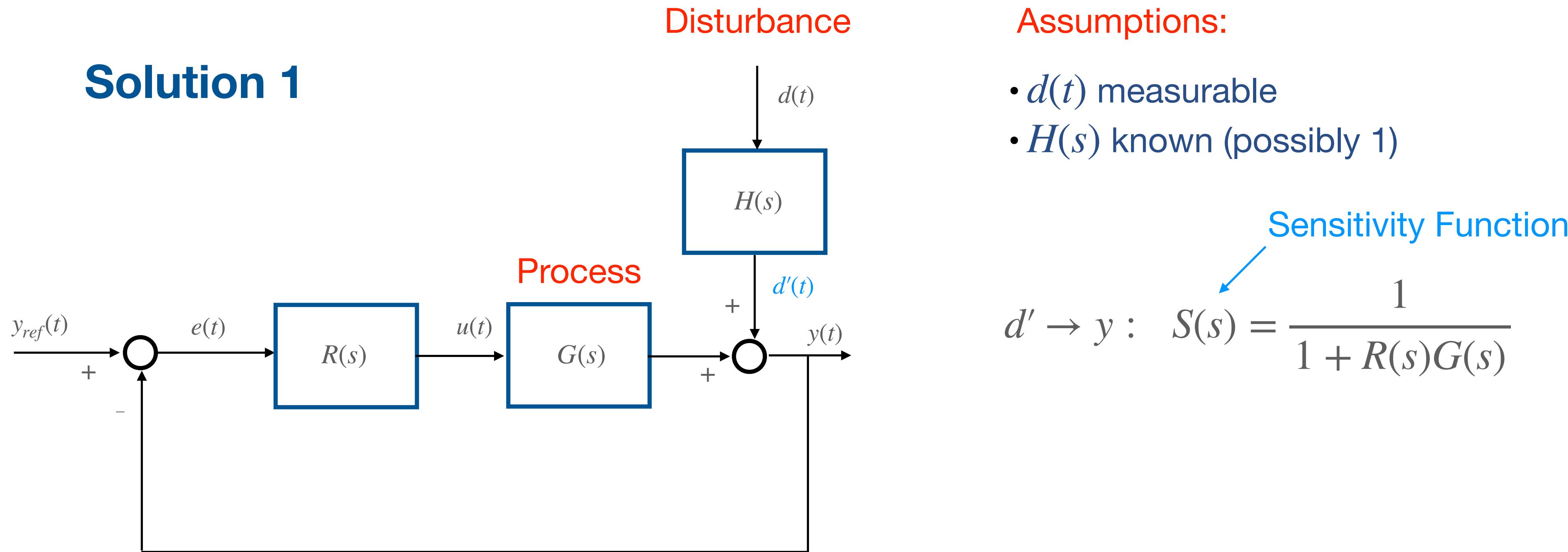


Assumptions:

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Control Scheme with Measurable Disturbance Compensation

Solution 1



Assumptions:

- $d(t)$ measurable
- $H(s)$ known (possibly 1)

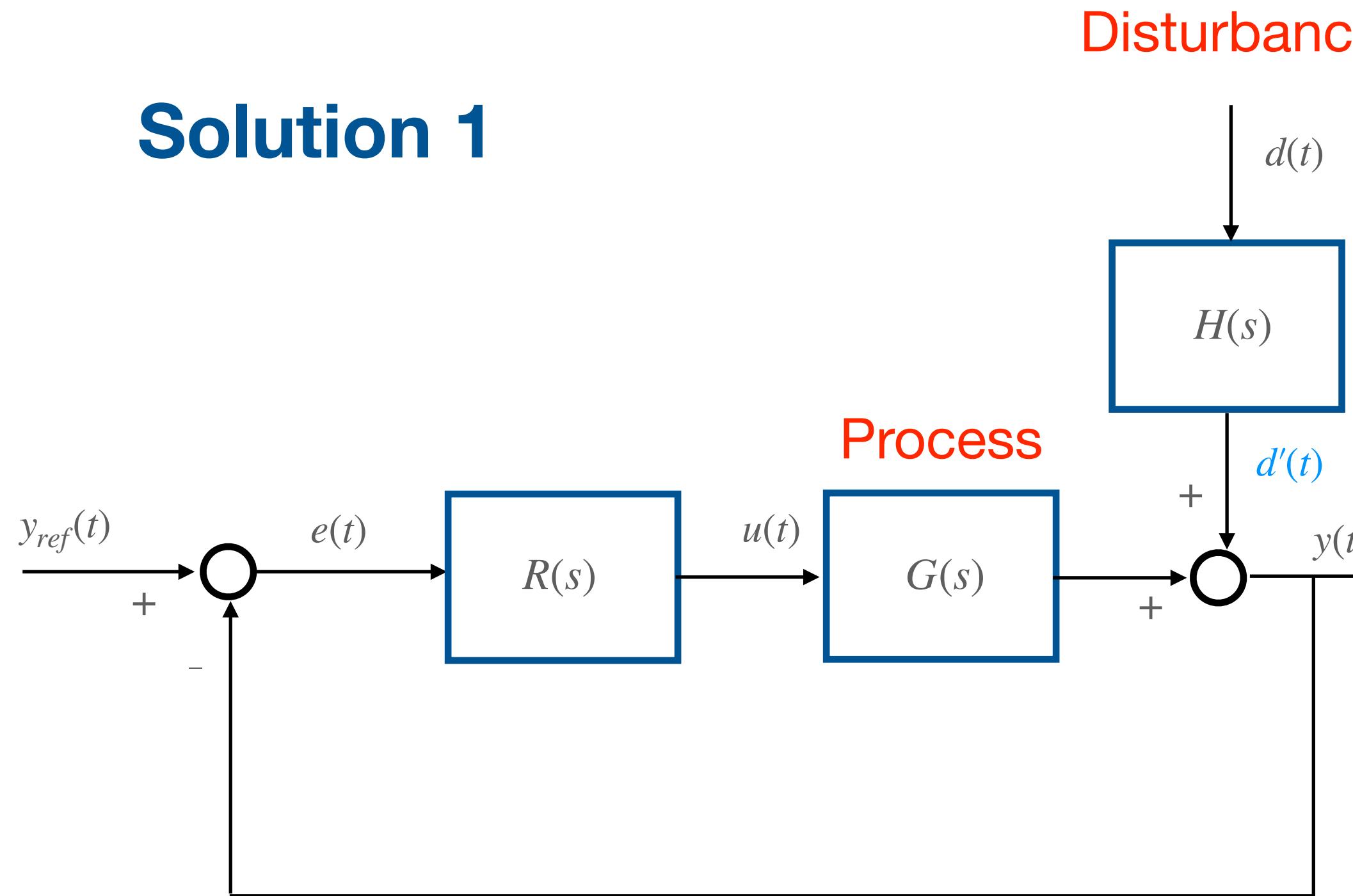
Sensitivity Function

$$d' \rightarrow y : S(s) = \frac{1}{1 + R(s)G(s)}$$



Control Scheme with Measurable Disturbance Compensation

Solution 1



Disturbance

Assumptions:

- $d(t)$ measurable
- $H(s)$ known (possibly 1)

Sensitivity Function

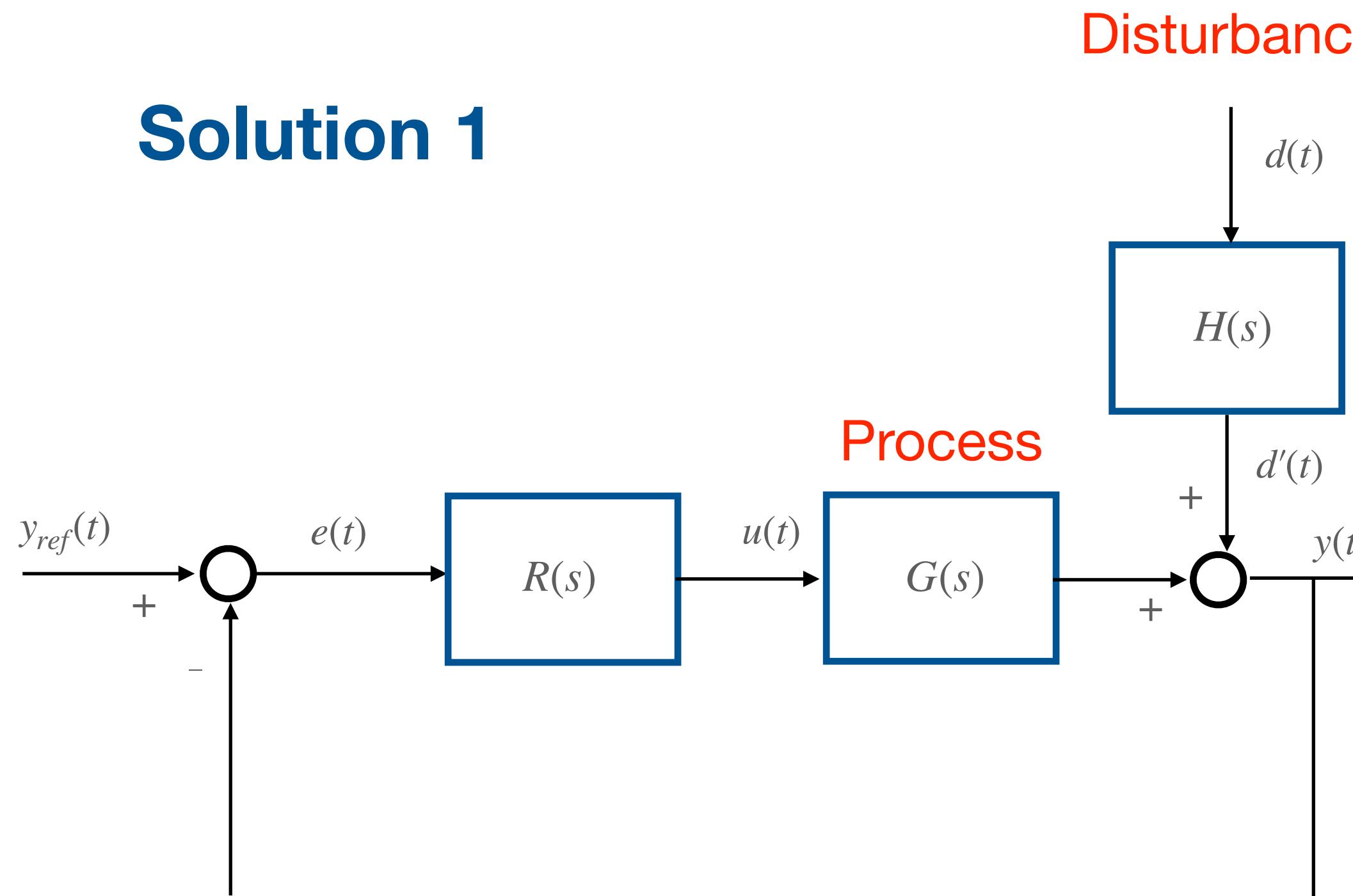
$$d' \rightarrow y : S(s) = \frac{1}{1 + R(s)G(s)}$$

Recommendation: $|S(j\omega)|$ sufficiently low $\forall \omega$
in the frequency range where $d'(t)$
has significant harmonics



Control Scheme with Measurable Disturbance Compensation

Solution 1



Disturbance

Assumptions:

- $d(t)$ measurable
- $H(s)$ known (possibly 1)

Sensitivity Function

$$d' \rightarrow y : S(s) = \frac{1}{1 + R(s)G(s)}$$

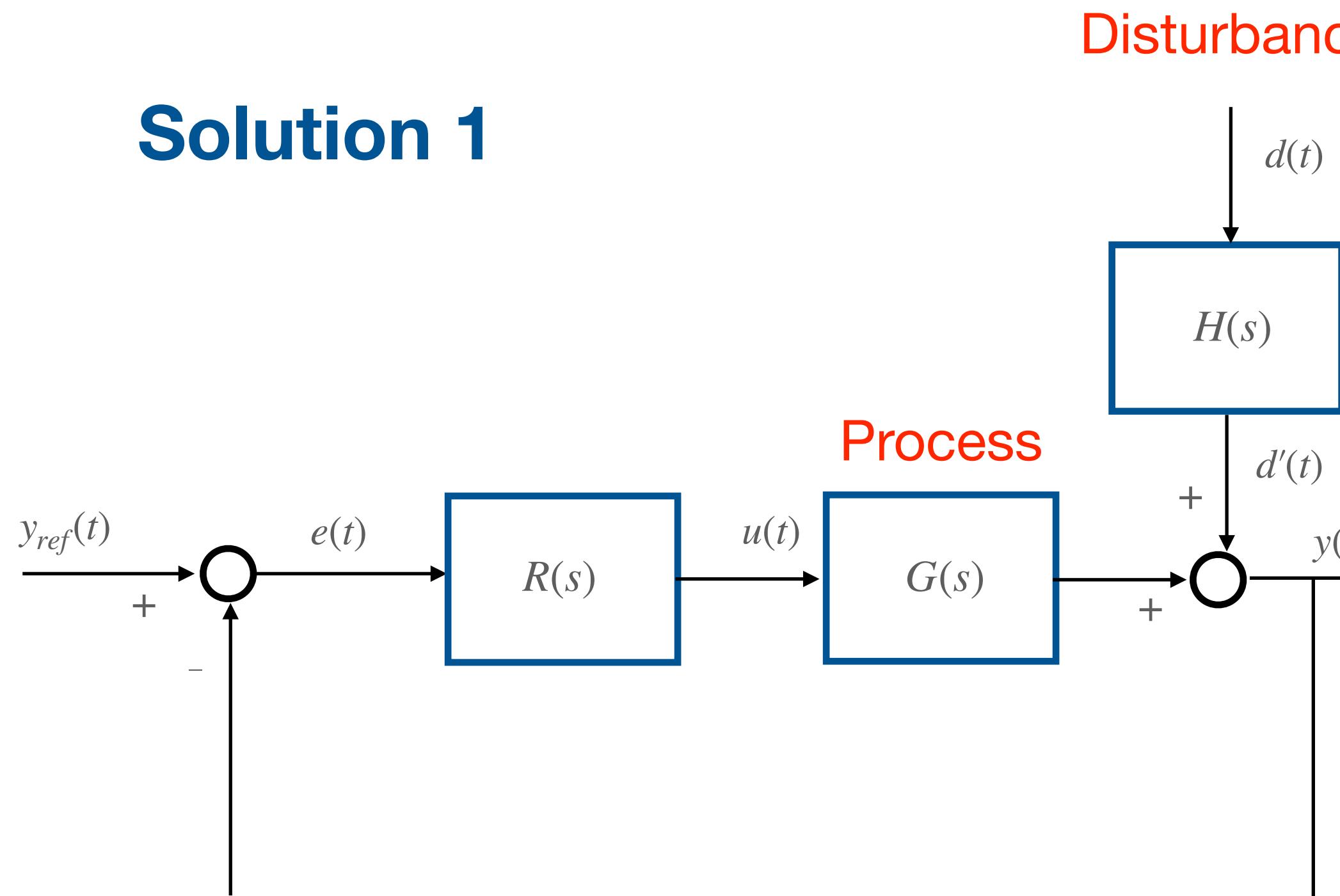
Recommendation: $|S(j\omega)|$ sufficiently low $\forall \omega$
in the frequency range where $d'(t)$
has significant harmonics

$$\omega_{d'} \in [0, \omega_{BW_{CL}})$$



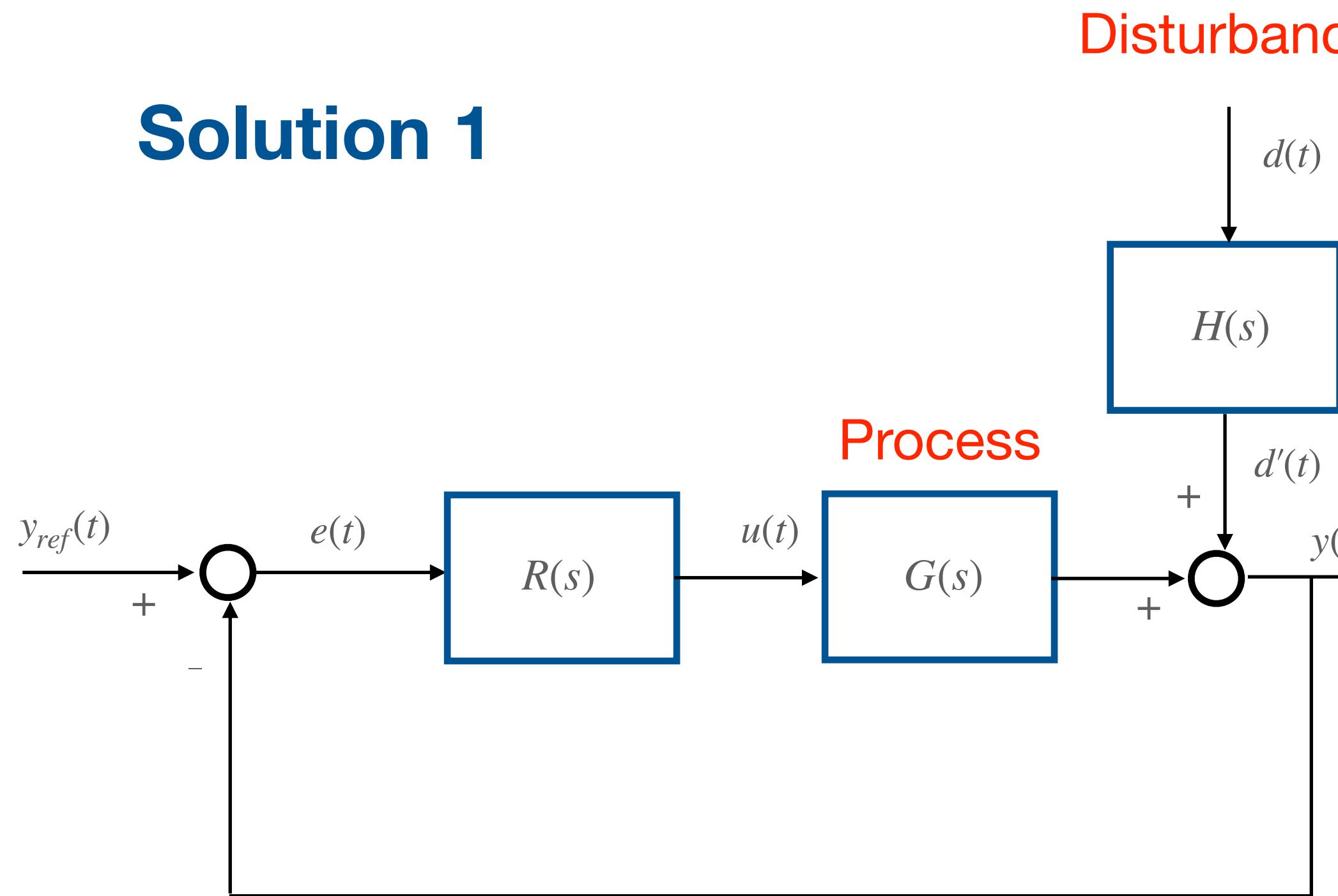
Control Scheme with Measurable Disturbance Compensation

Solution 1



Control Scheme with Measurable Disturbance Compensation

Solution 1



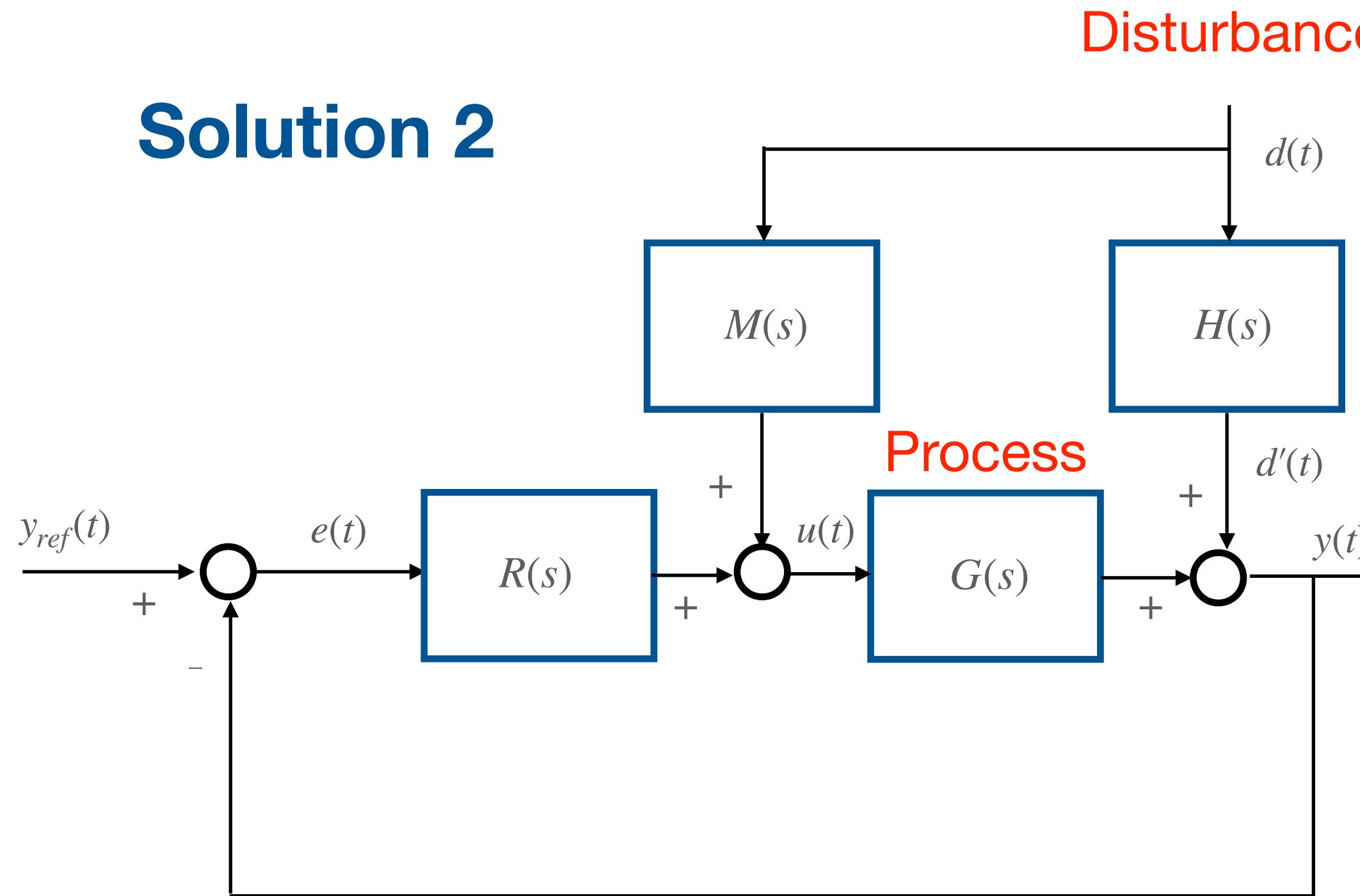
$$|S(j\omega)| \xrightarrow{\text{LF:}} |S(j\omega)| \rightarrow \left| \frac{1}{L(j\omega)} \right|$$

$$|S(j\omega)| \xrightarrow{\text{HF:}} |S(j\omega)| \rightarrow 1$$



Control Scheme with Measurable Disturbance Compensation

Solution 2



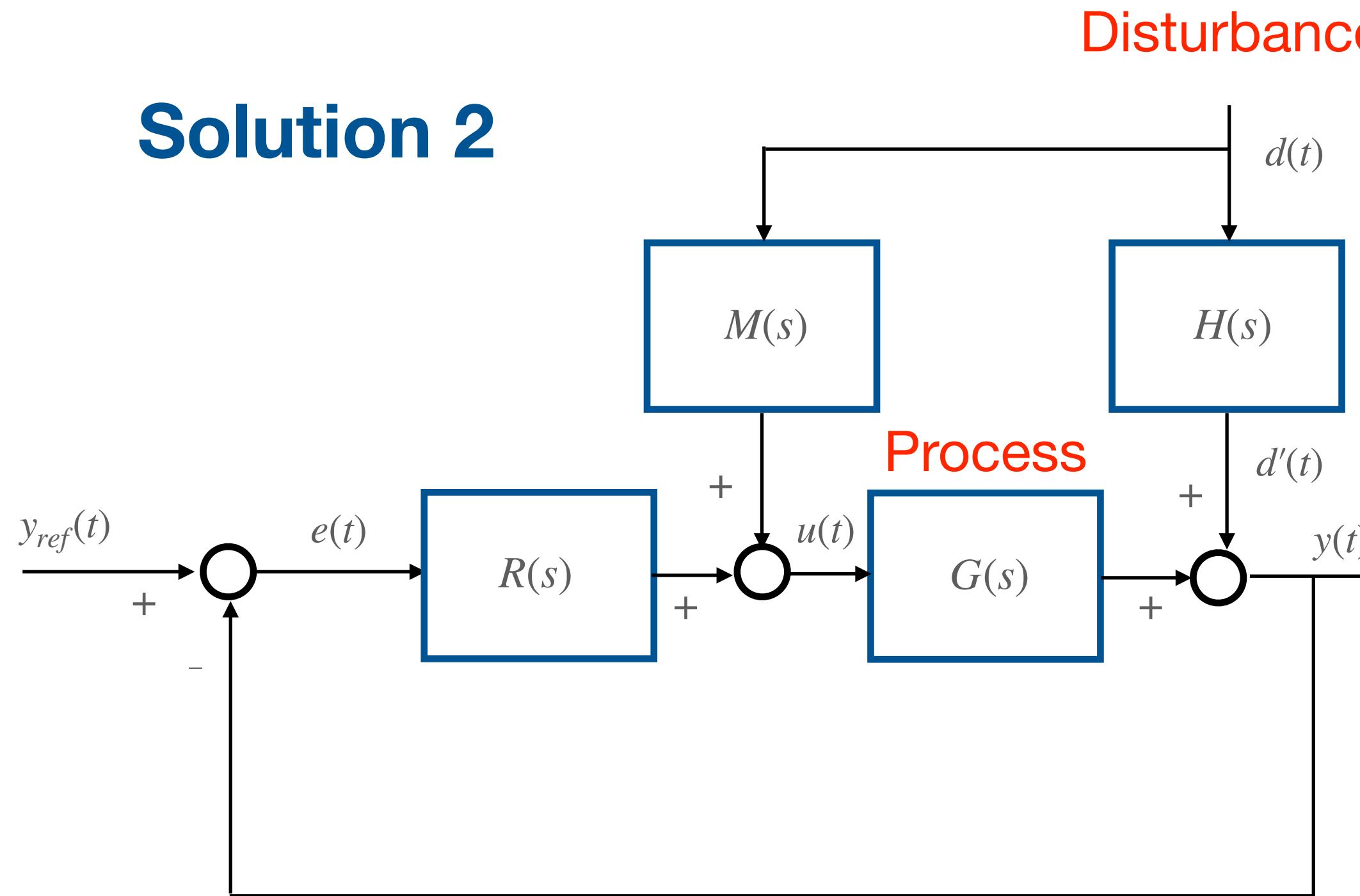
Assumptions:

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Control Scheme with Measurable Disturbance Compensation

Solution 2



Disturbance

Assumptions:

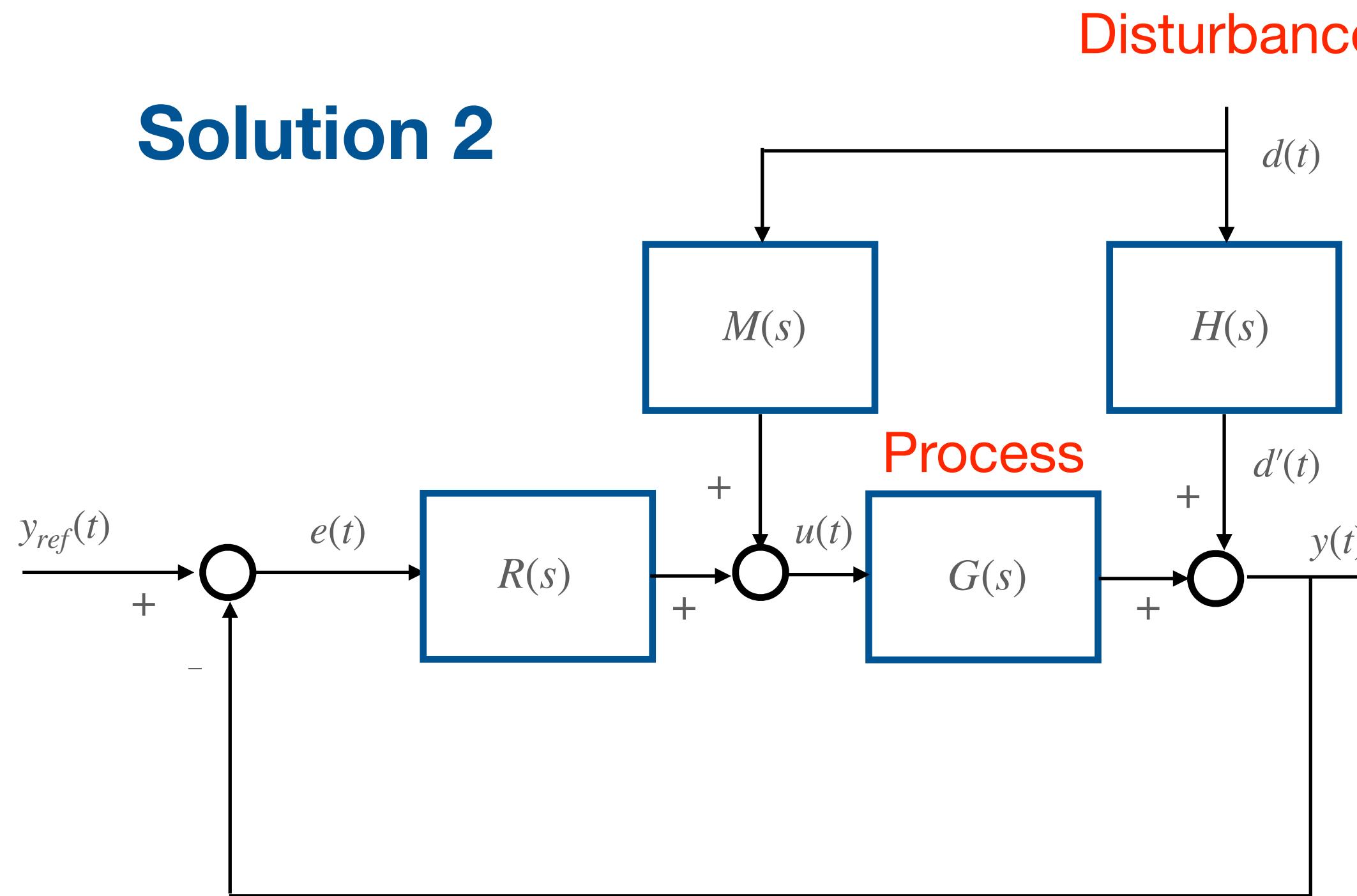
- $d(t)$ measurable
- $H(s)$ known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)}$$



Control Scheme with Measurable Disturbance Compensation

Solution 2



Disturbance

Assumptions:

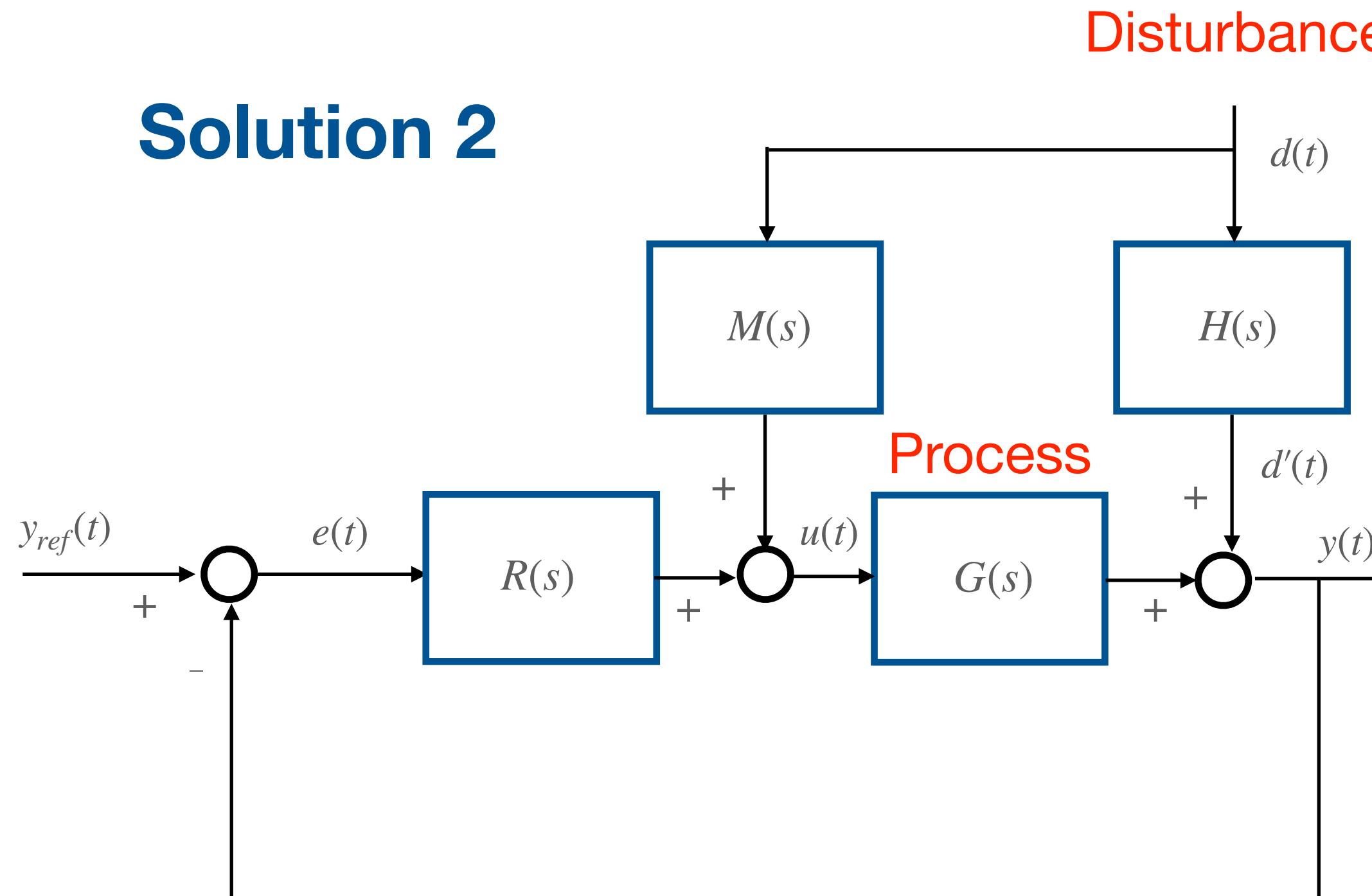
- $d(t)$ measurable
- $H(s)$ known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$



Control Scheme with Measurable Disturbance Compensation

Solution 2



Disturbance

Assumptions:

- $d(t)$ measurable
- $H(s)$ known (possibly 1)

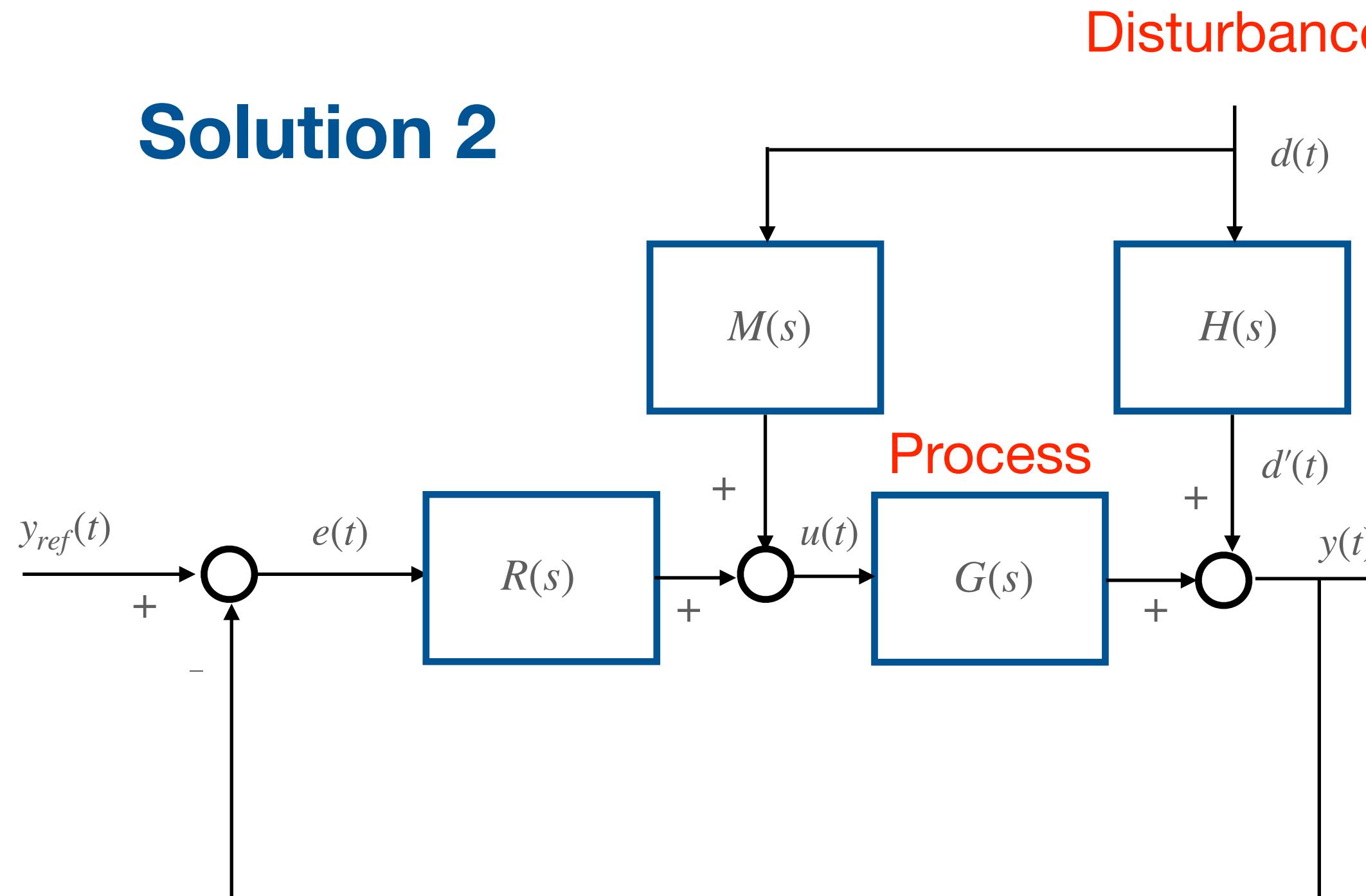
$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

$$M(s) = -\frac{H(s)}{G(s)}$$



Control Scheme with Measurable Disturbance Compensation

Solution 2



Disturbance

Assumptions:

- $d(t)$ measurable
- $H(s)$ known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

Warning:

- $M(s)$ As. Stable
- Causal

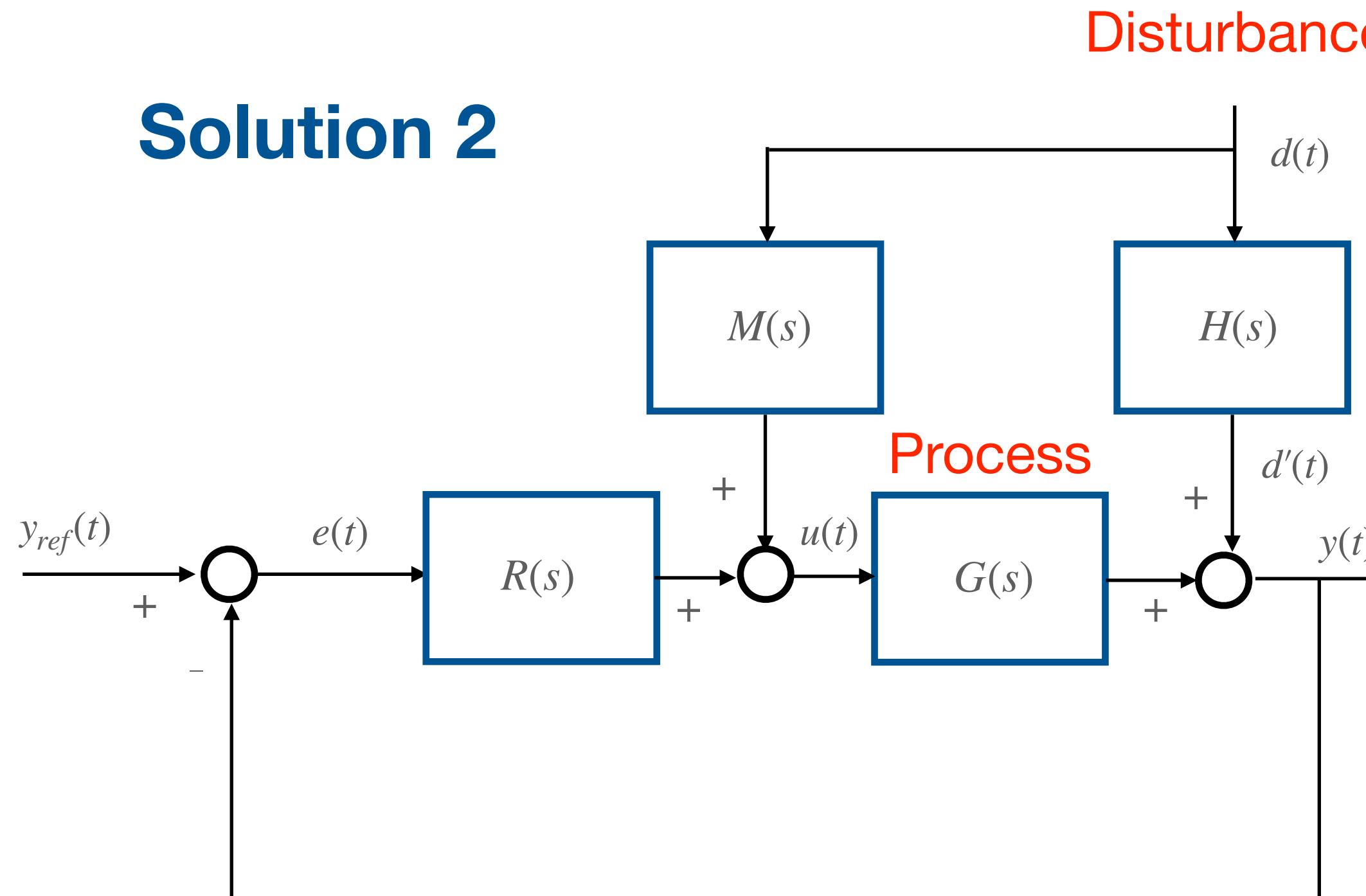
Ideal Design

$$M(s) = -\frac{H(s)}{G(s)}$$



Control Scheme with Measurable Disturbance Compensation

Solution 2



Assumptions:

- $d(t)$ measurable
- $H(s)$ known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

Ideal Design

$$M(s) = -\frac{H(s)}{G(s)}$$

not guaranteed

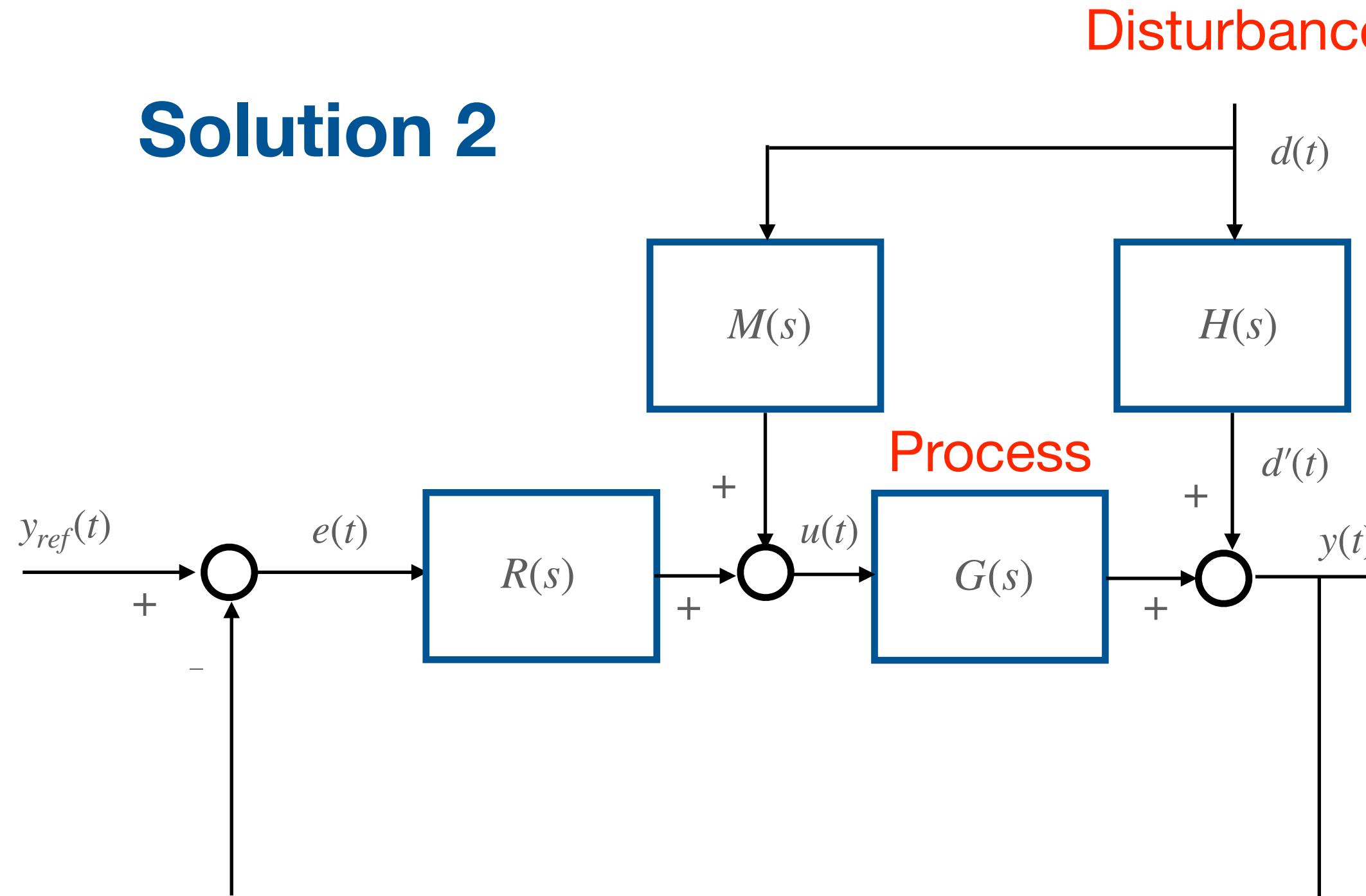
Warning:

- $M(s)$ As. Stable
- Causal



Control Scheme with Measurable Disturbance Compensation

Solution 2



Disturbance

Assumptions:

- $d(t)$ measurable
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$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

not guaranteed

Warning:

- $M(s)$ As. Stable
- Causal

Actual Design

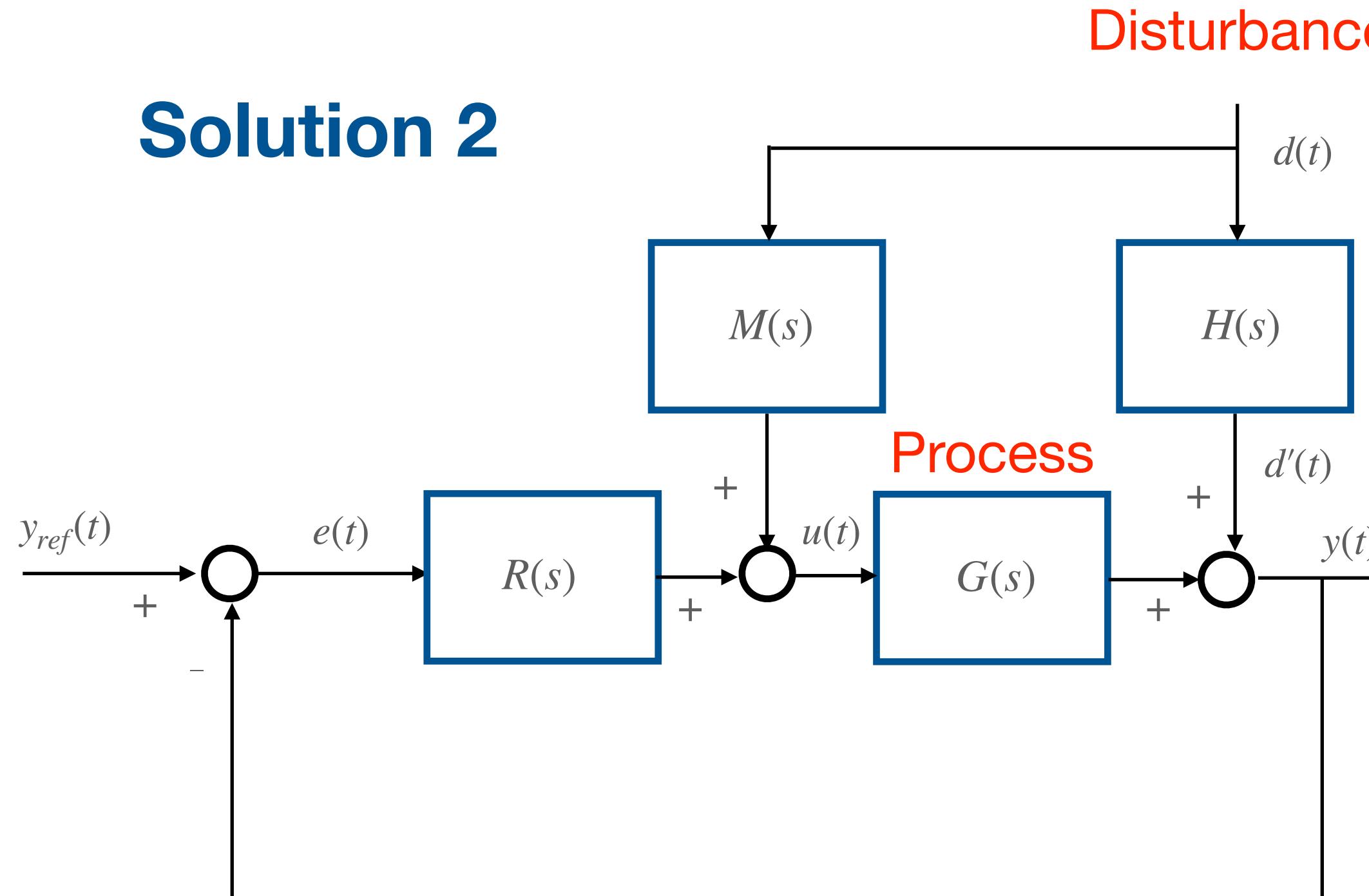
$$M(j\omega) = -\frac{H(j\omega)}{G(j\omega)}$$

in the frequency range where $d(t)$ has significant harmonics



Control Scheme with Measurable Disturbance Compensation

Solution 2



Assumptions:

- $d(t)$ measurable
- $H(s)$ known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

not guaranteed

Warning:

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Actual Design

$$M(j\omega) = -\frac{H(j\omega)}{G(j\omega)}$$

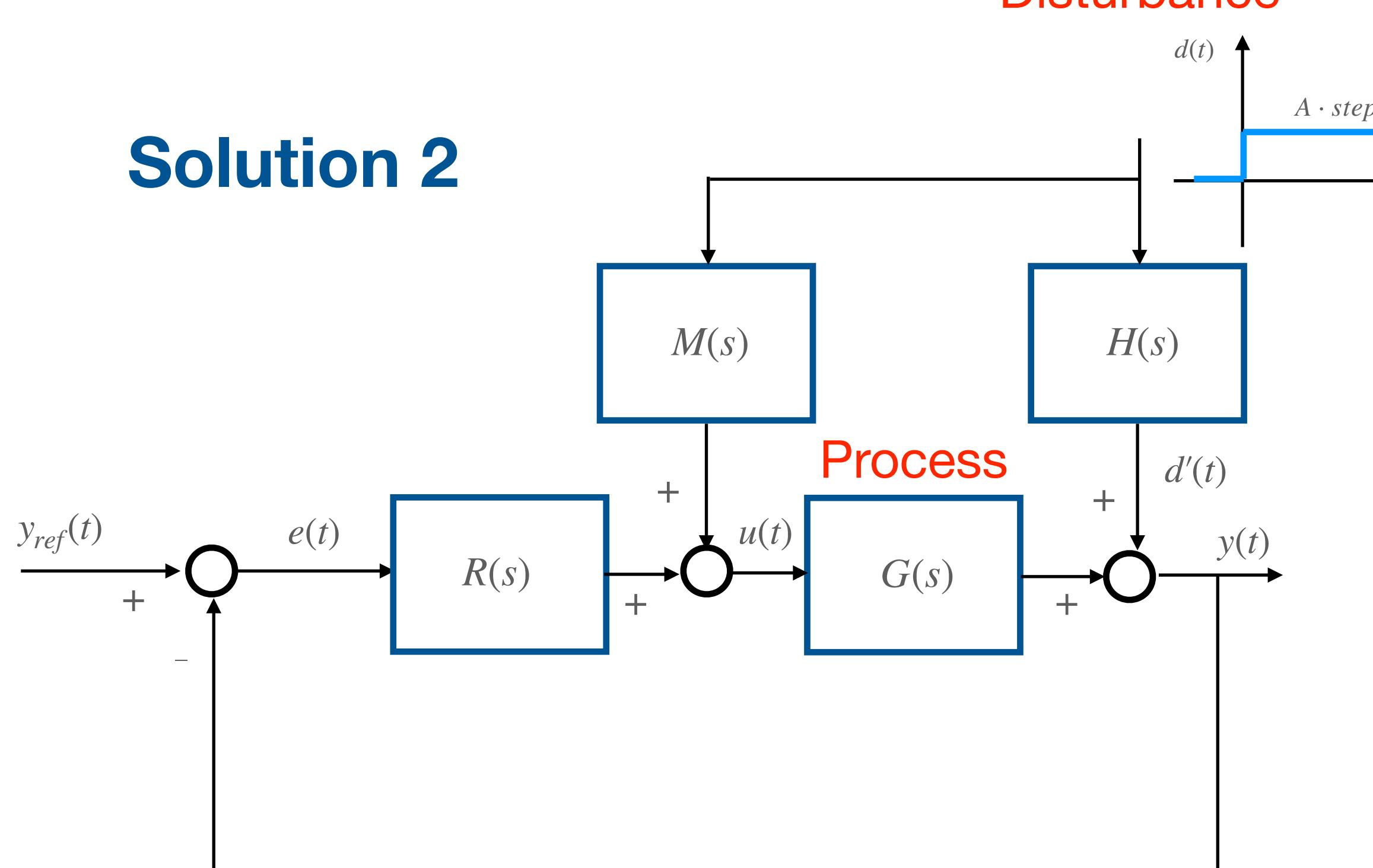
Example 1:

$$d(t) = \sin(\hat{\omega} t) \longrightarrow M(s) \text{ s.t. } M(j\omega) \Big|_{\omega=\hat{\omega}} = -\frac{H(j\hat{\omega})}{G(j\hat{\omega})}$$



Control Scheme with Measurable Disturbance Compensation

Solution 2



Assumptions:

- $d(t)$ measurable
- $H(s)$ known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

not guaranteed

Warning:

- $M(s)$ As. Stable
- Causal

Actual Design

$$M(j\omega) = -\frac{H(j\omega)}{G(j\omega)}$$

Example 2:
static case

$$\hat{\omega} = 0$$

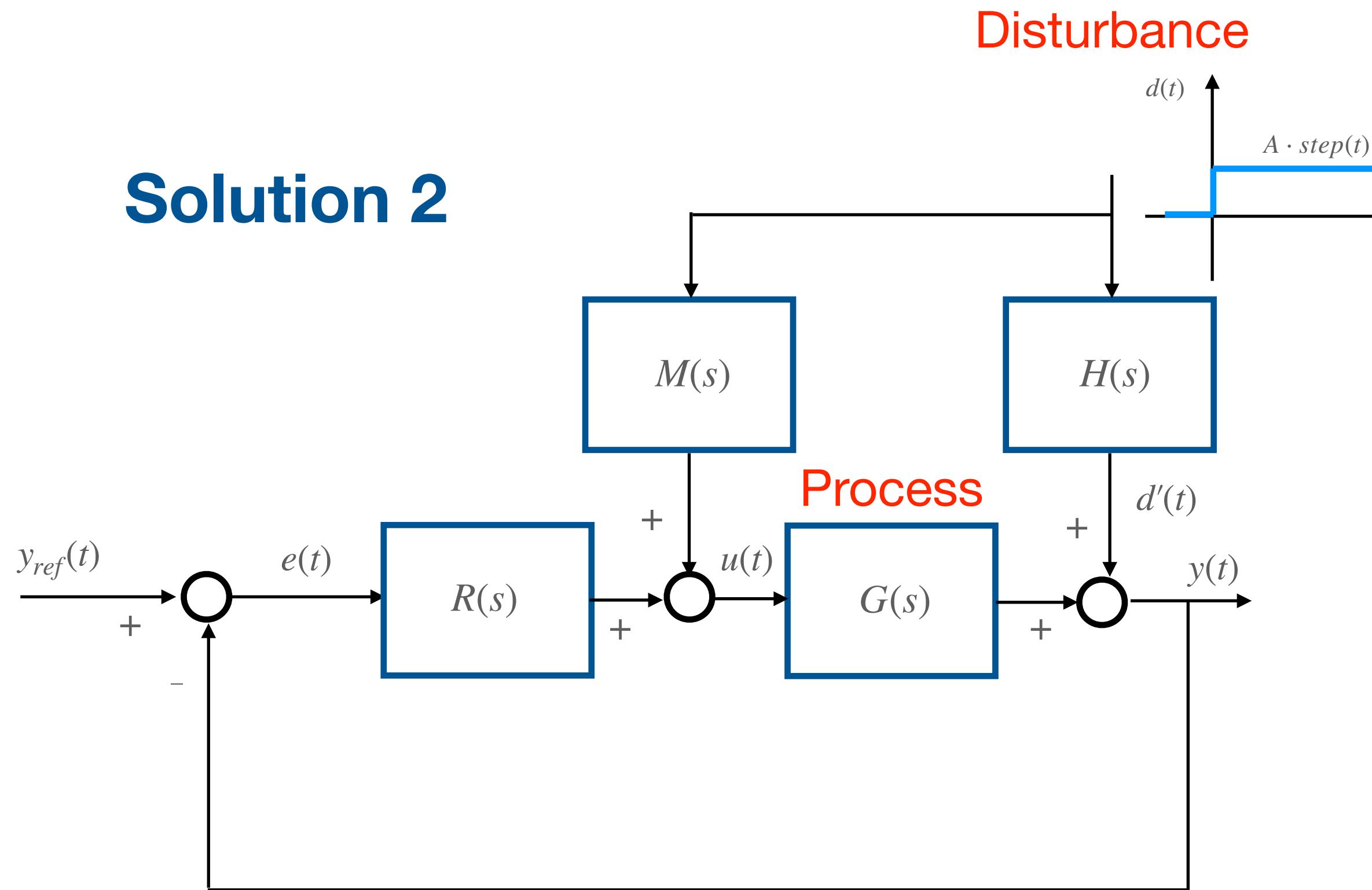
$M(s)$ s.t.

$$M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

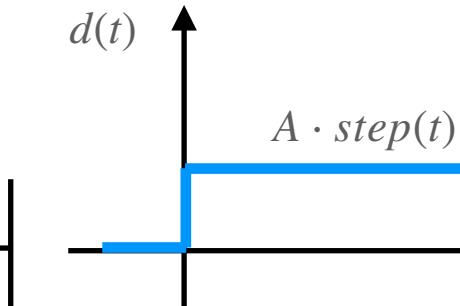


Control Scheme with Measurable Disturbance Compensation

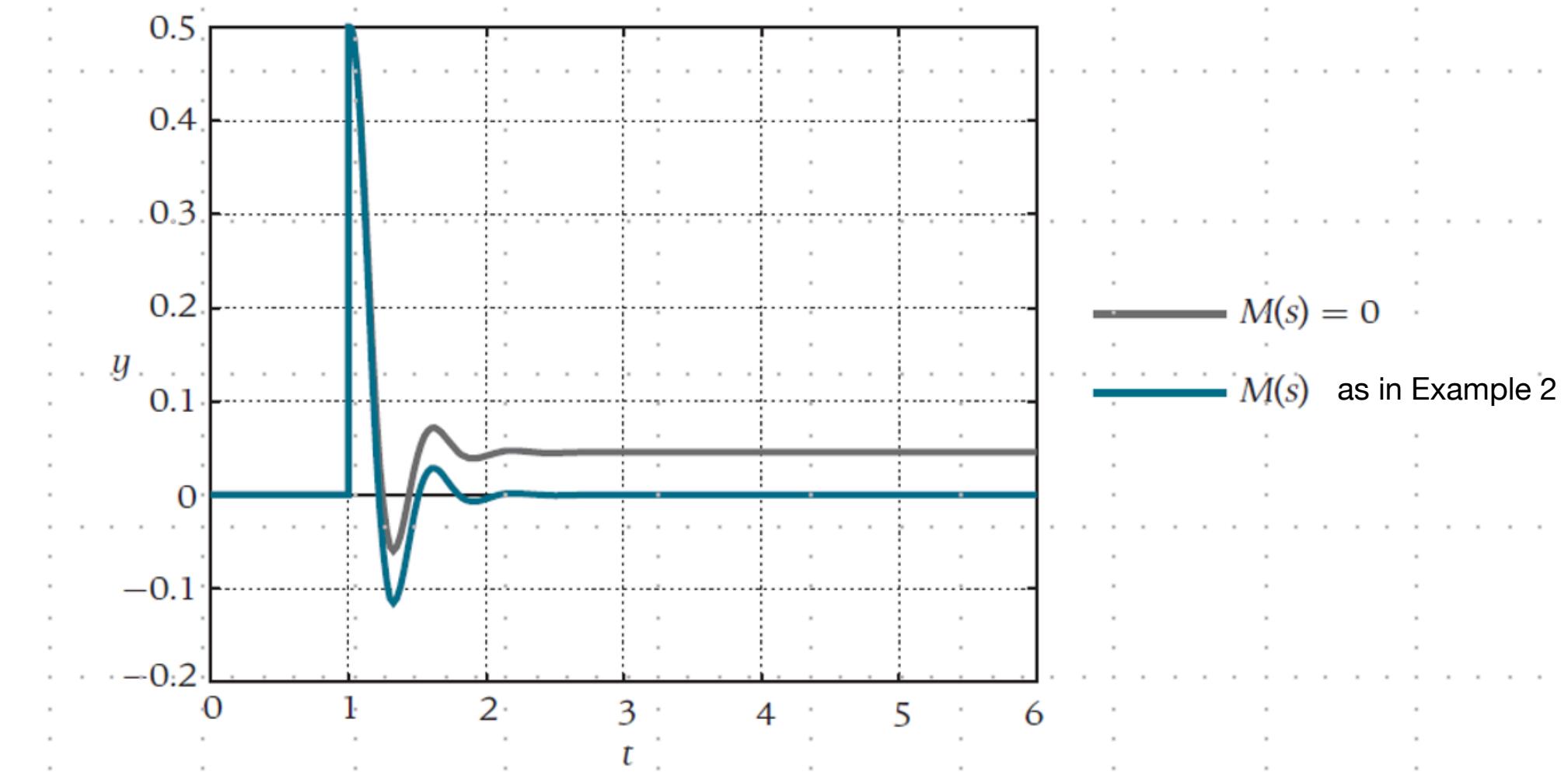
Solution 2



Disturbance



Response to a step disturbance



Example 2:
static case

$$\hat{\omega} = 0$$

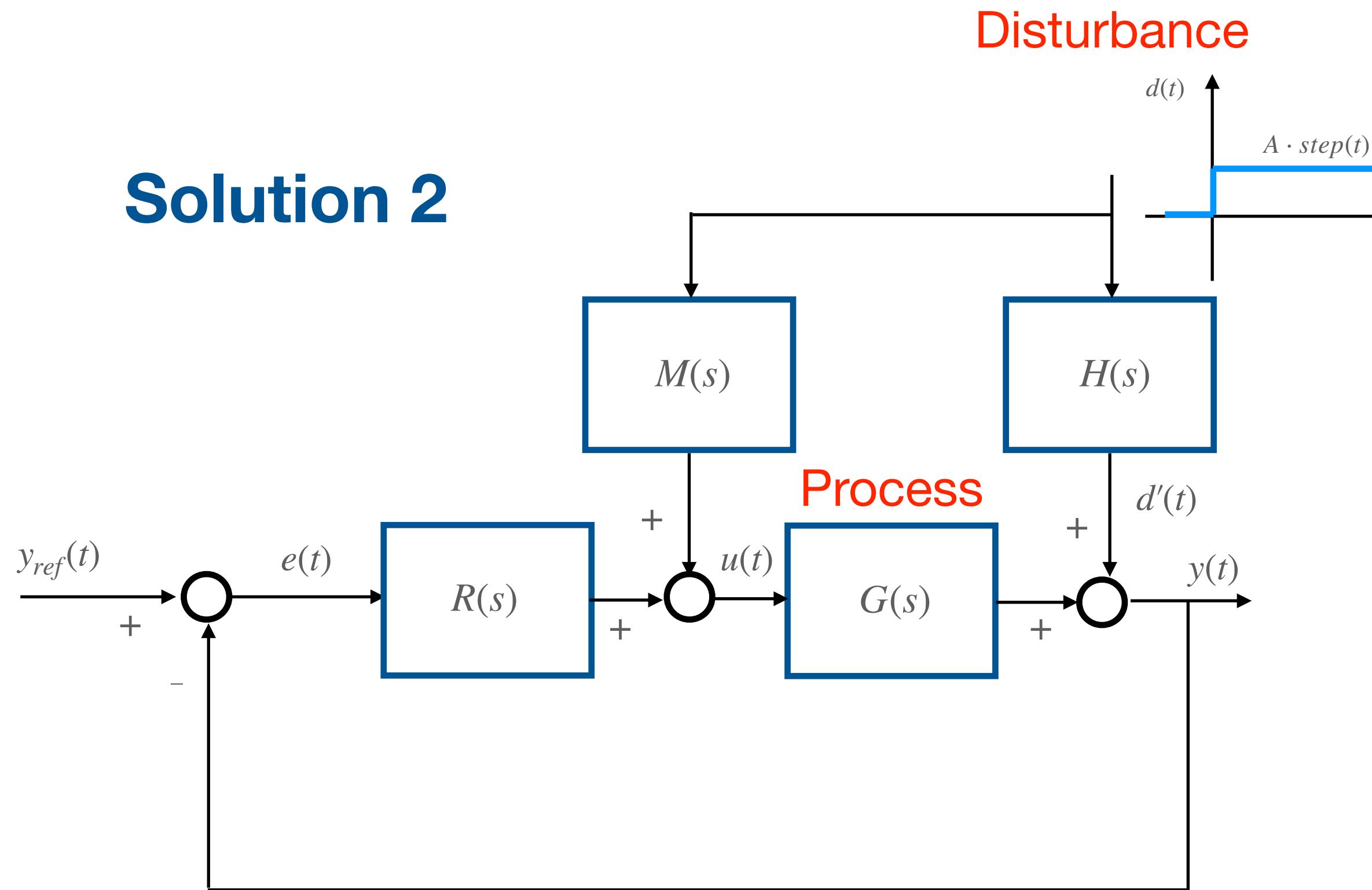
$\rightarrow M(s)$ s.t.

$$M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

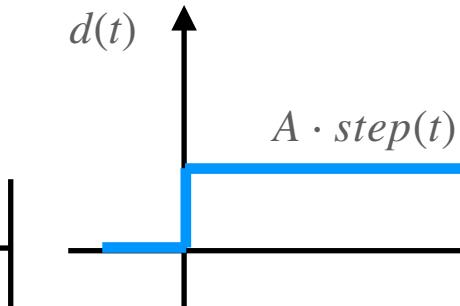


Control Scheme with Measurable Disturbance Compensation

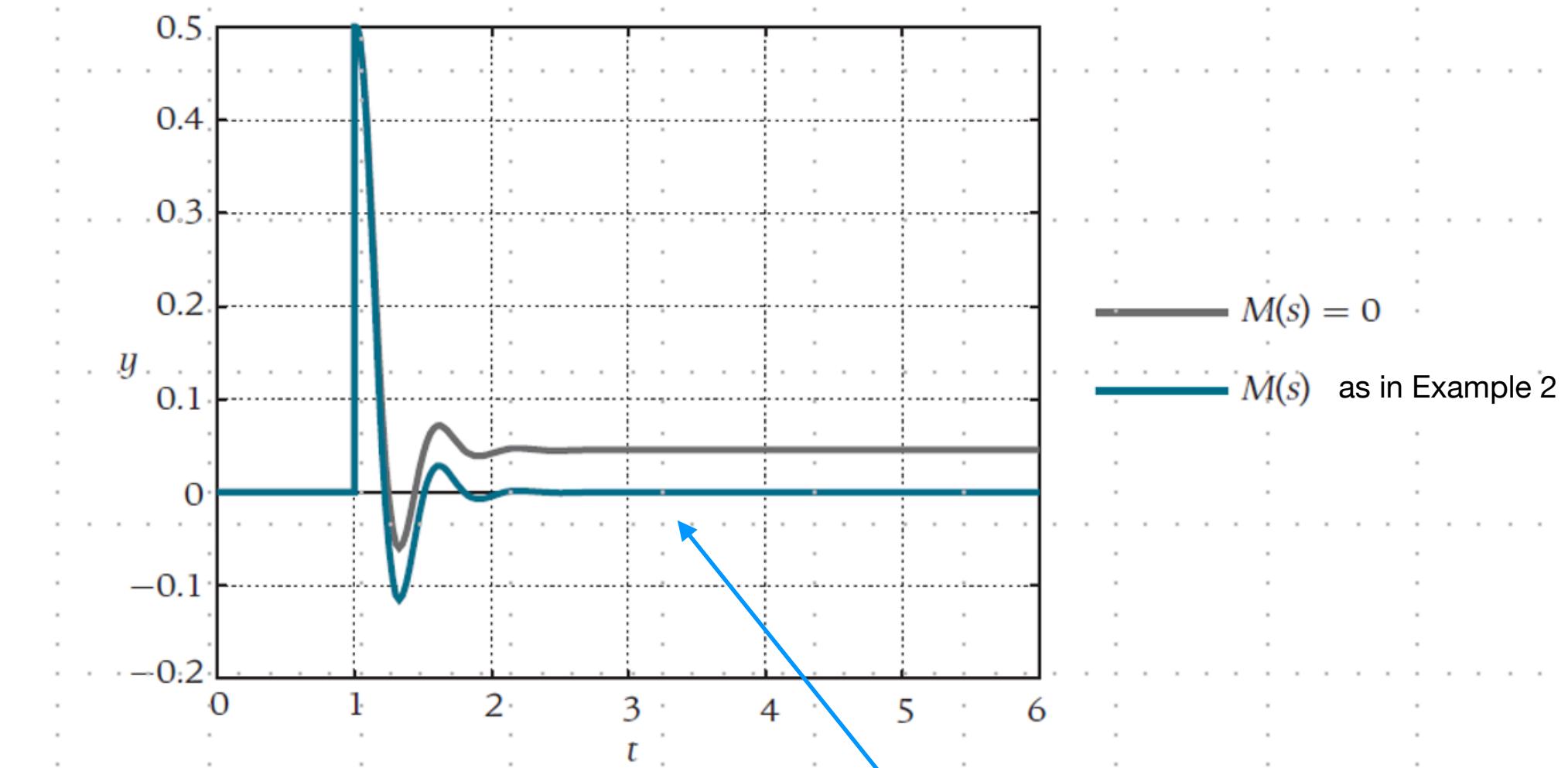
Solution 2



Disturbance



Response to a step disturbance



Example 2:
static case

$$\hat{\omega} = 0$$

$M(s)$ s.t.

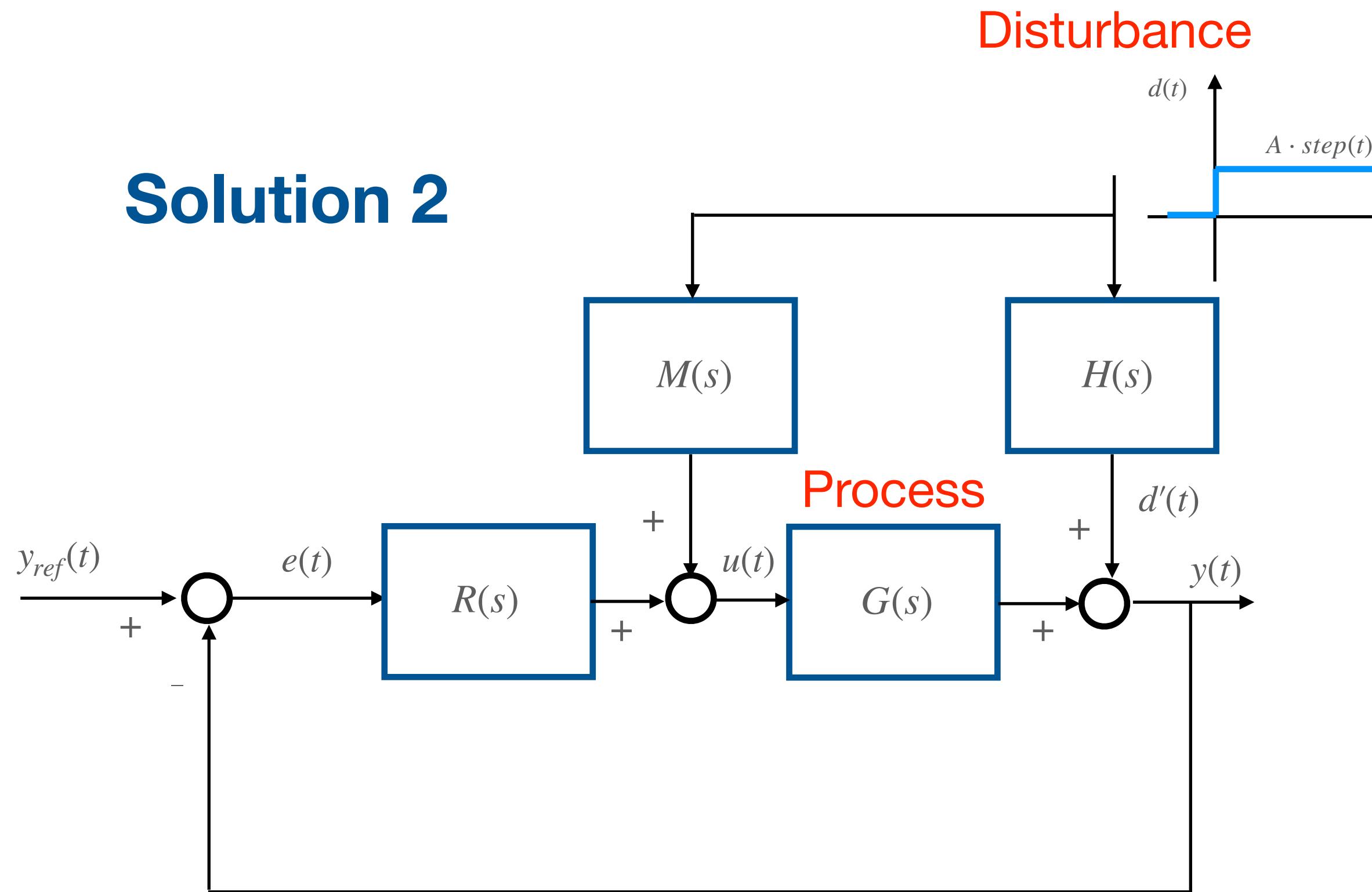
$$M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

the disturbance is rejected
in steady-state

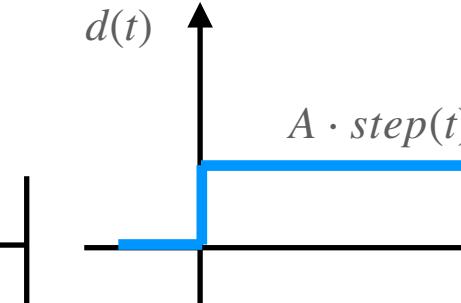


Control Scheme with Measurable Disturbance Compensation

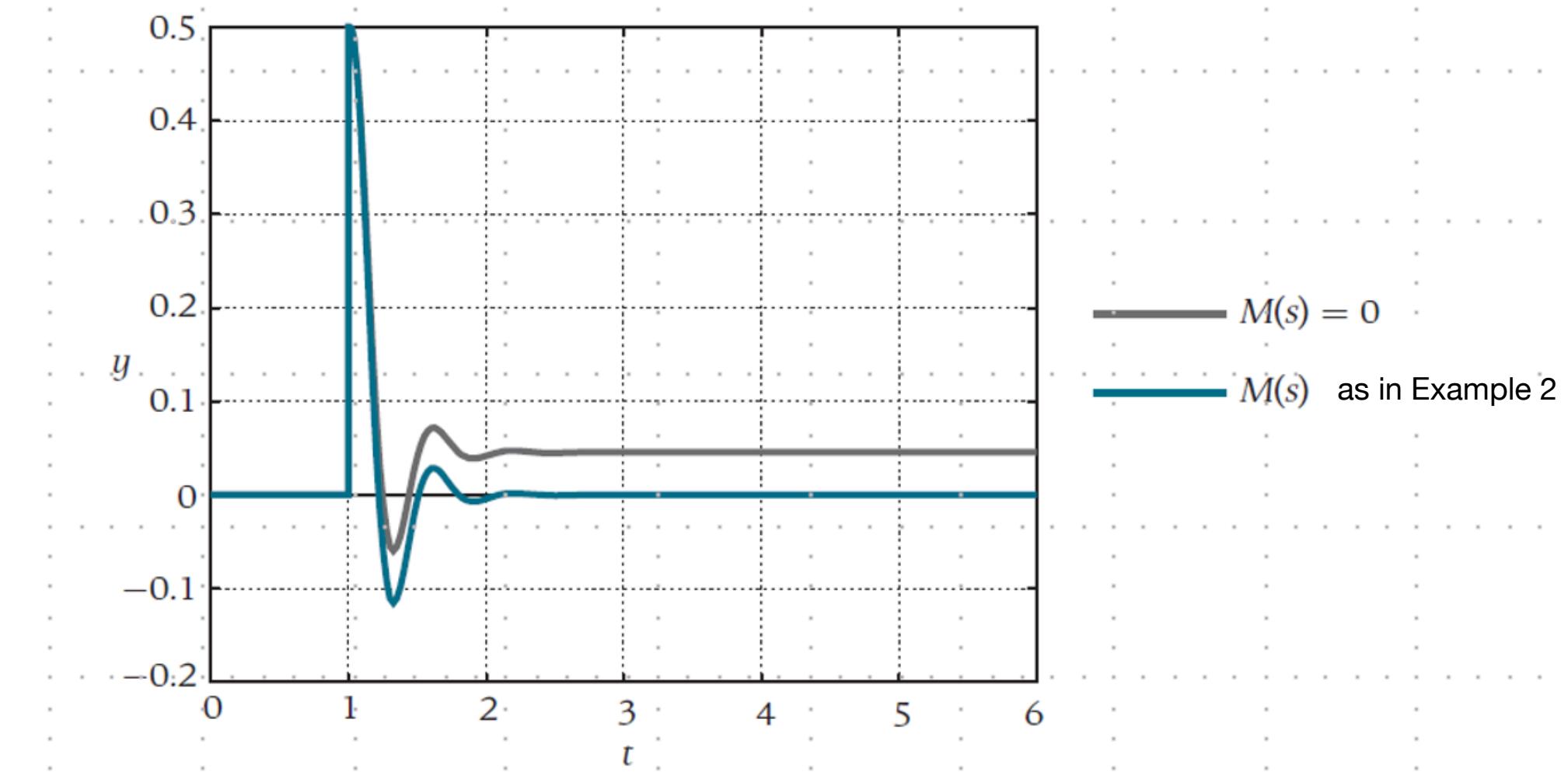
Solution 2



Disturbance



Response to a step disturbance



Example 2:
static case

$$\hat{\omega} = 0$$

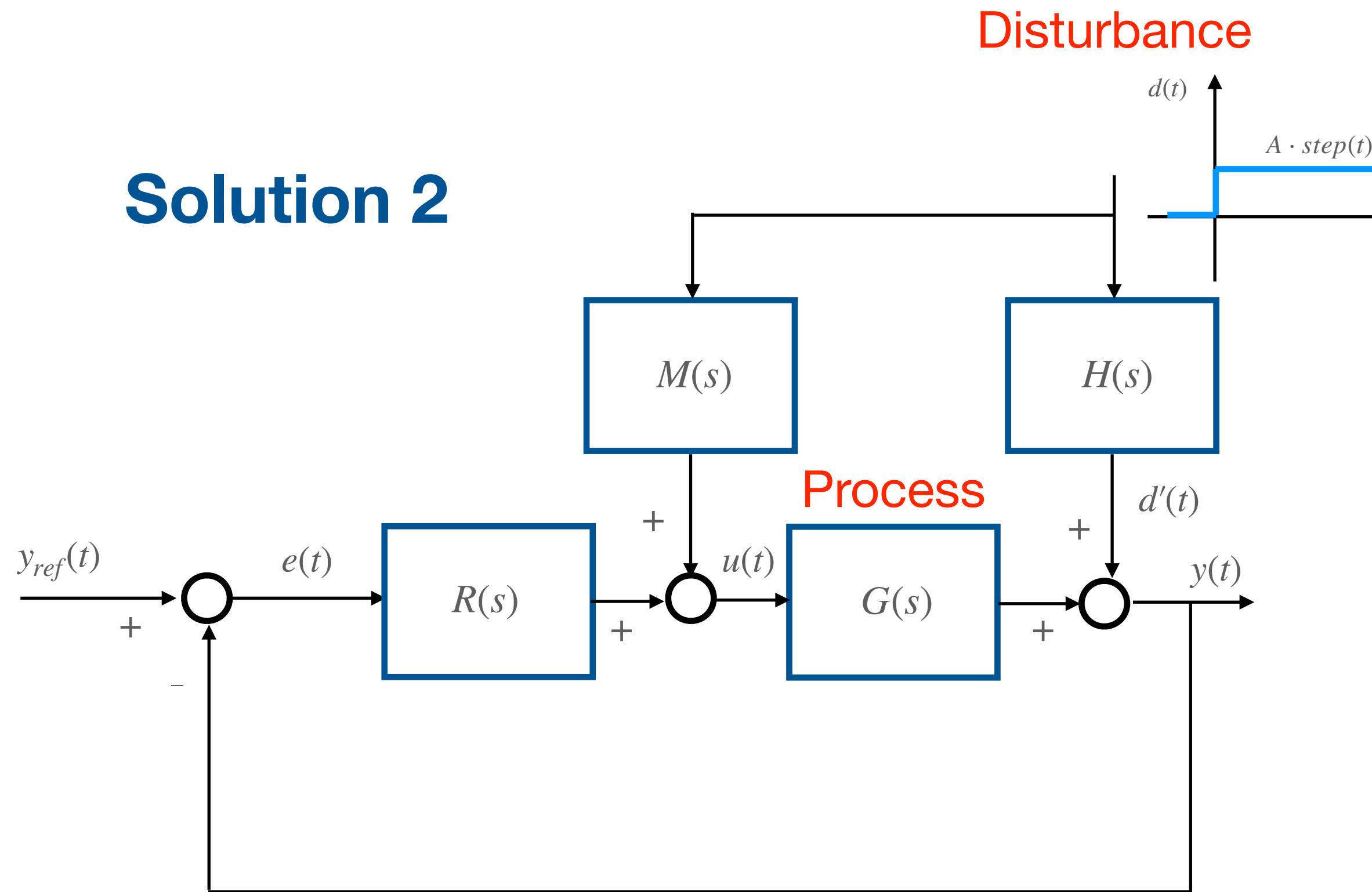
$$\xrightarrow{\quad M(s) \text{ s.t. } \quad} M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

Is this static
compensation
robust?

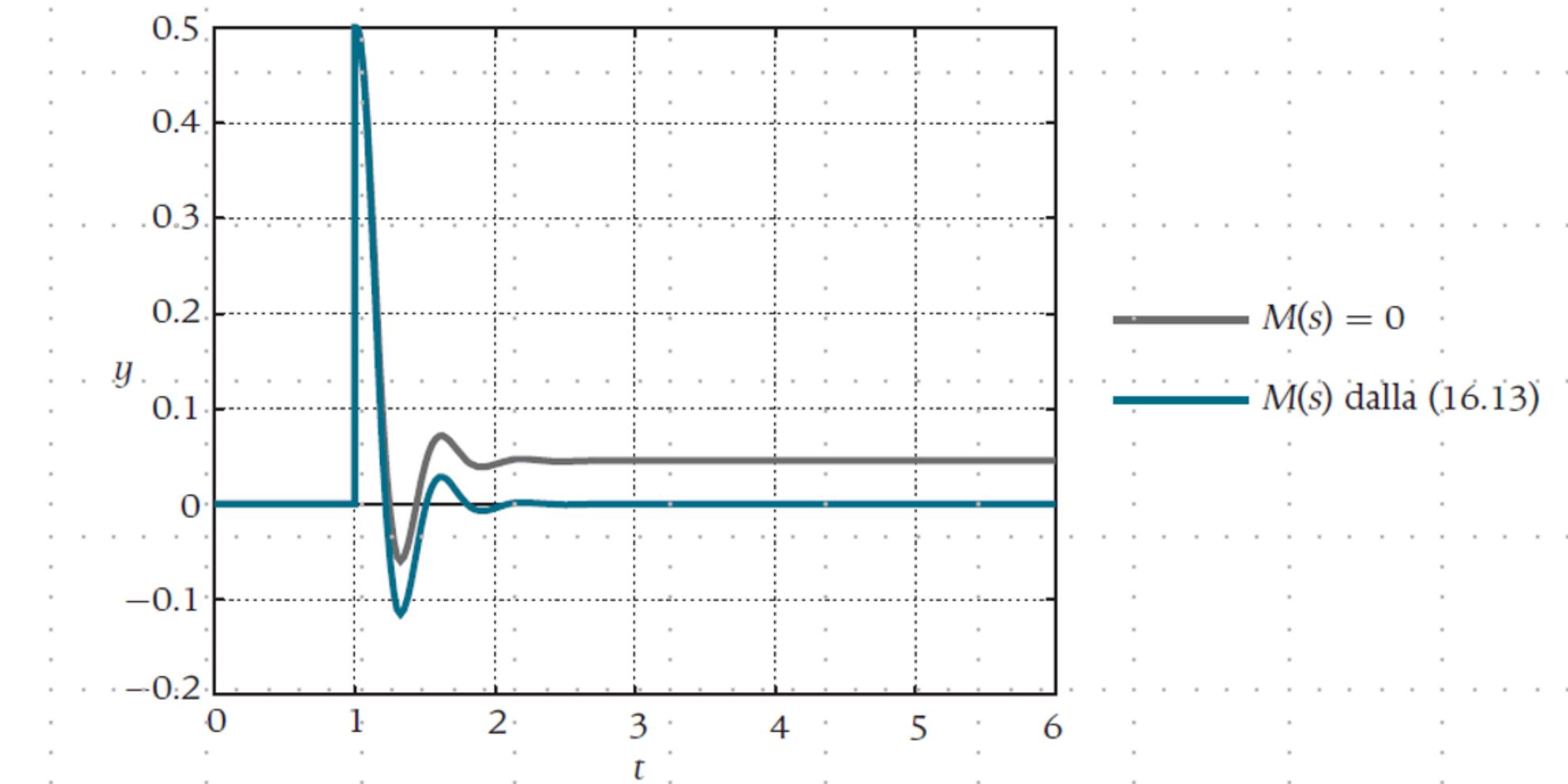


Control Scheme with Measurable Disturbance Compensation

Solution 2



Response to a step disturbance



Example 2:
static case

$$\hat{\omega} = 0$$

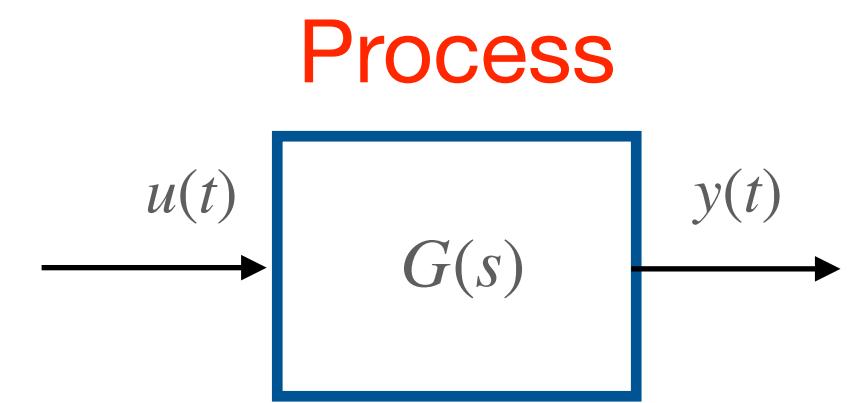
$$\xrightarrow{M(s) \text{ s.t.}} M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

Is this static
compensation
robust?

Is there a
better
alternative?



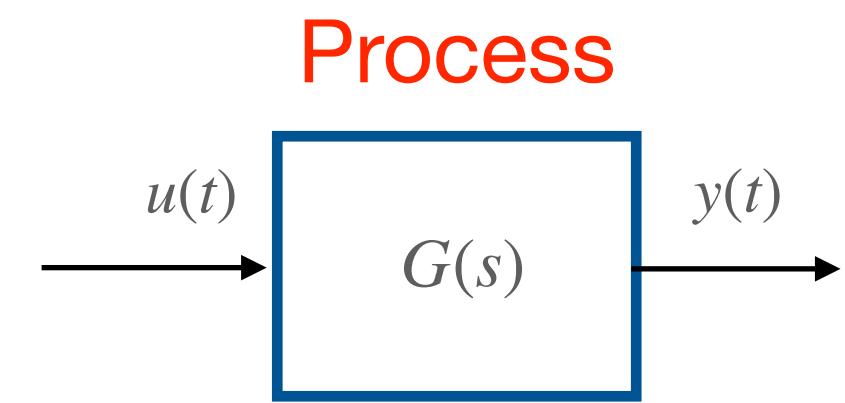
Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$



Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

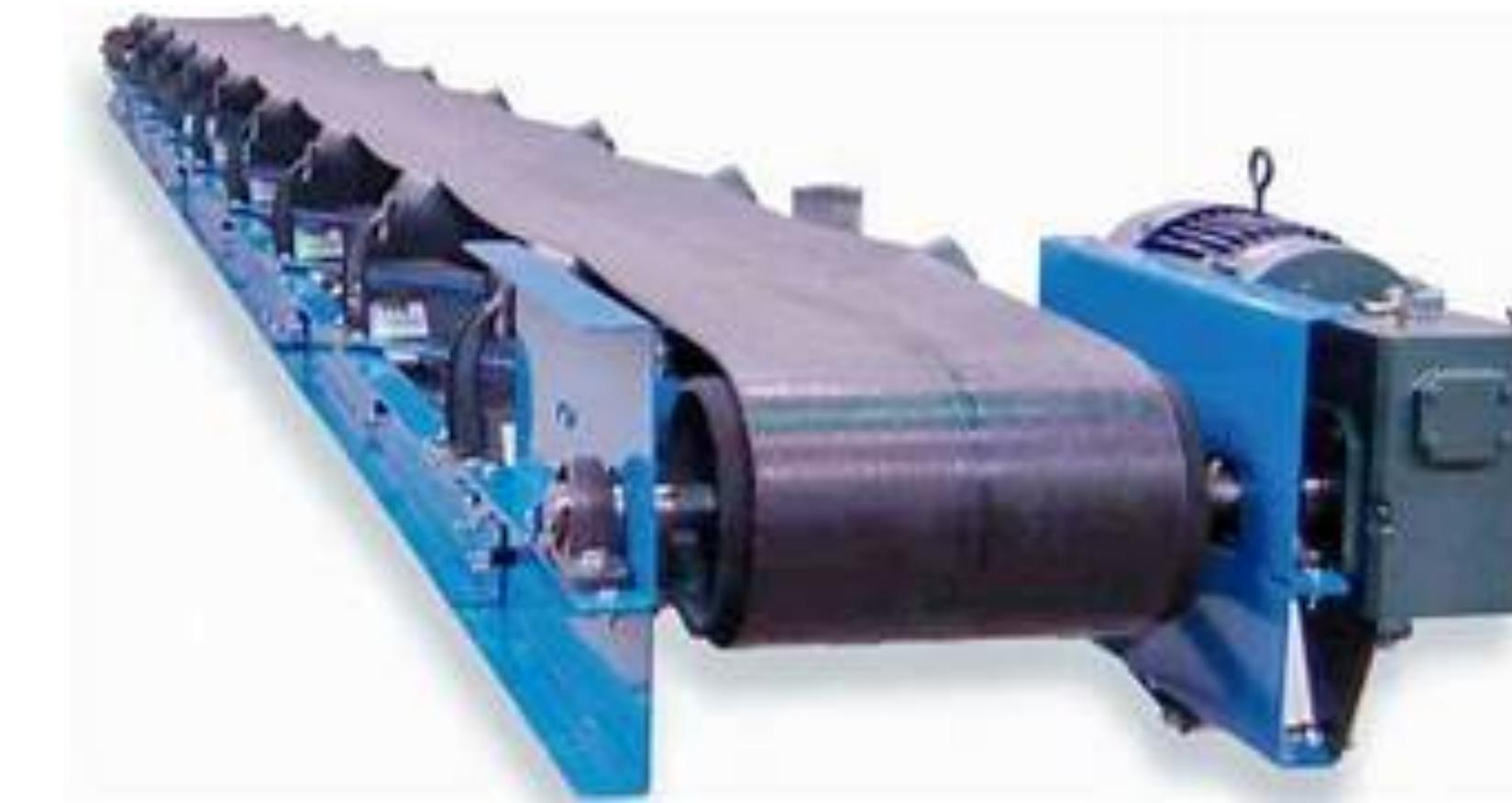
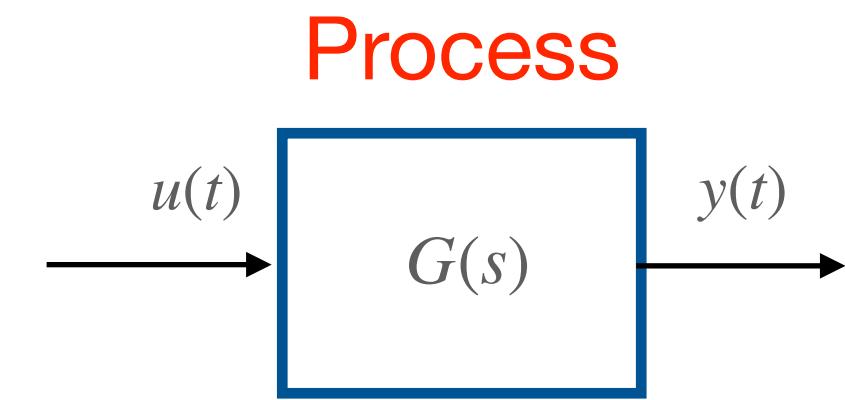


Image credit to: <https://www.gramconveyor.com/how-to-install-conveyor-belt/>

Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

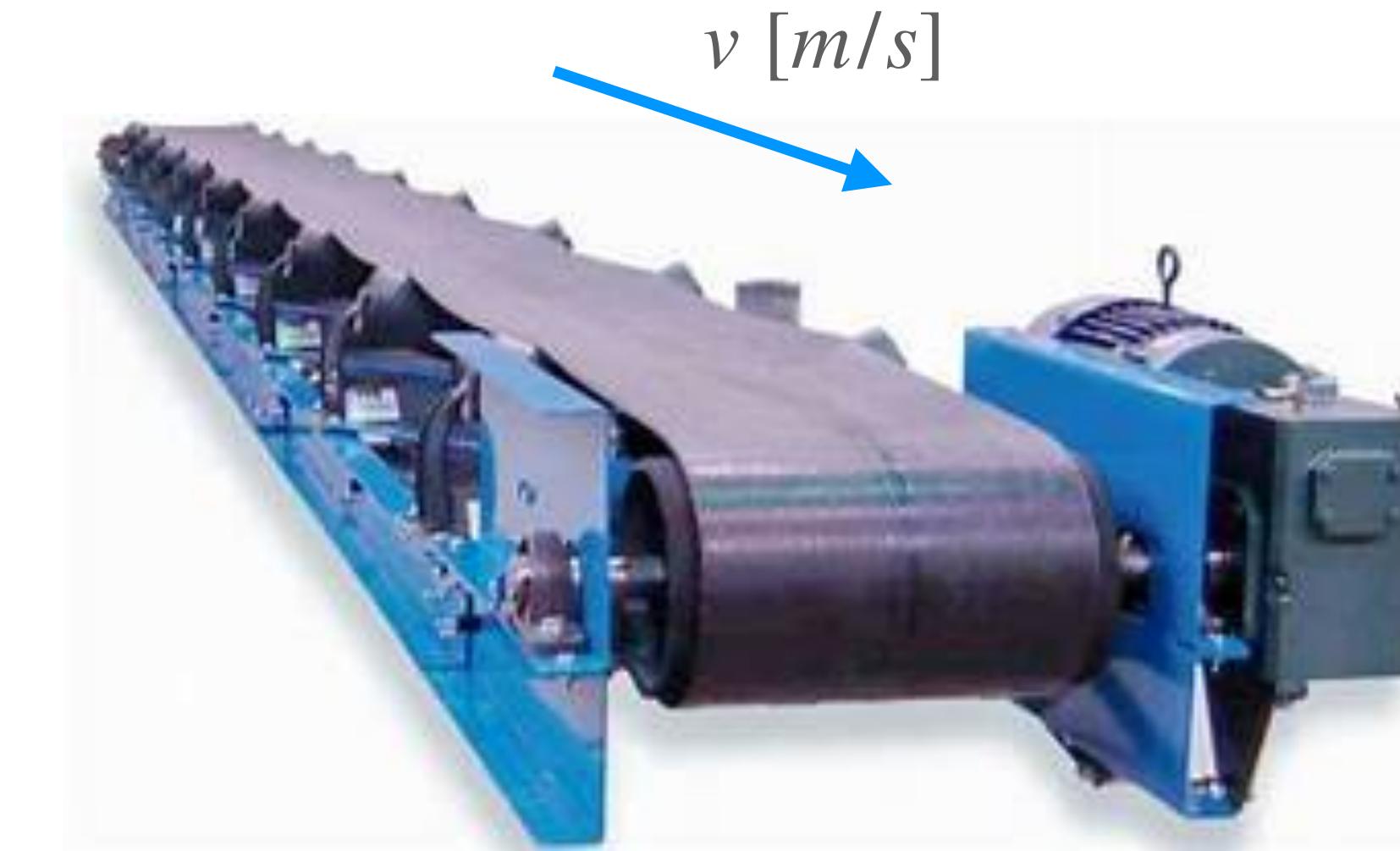
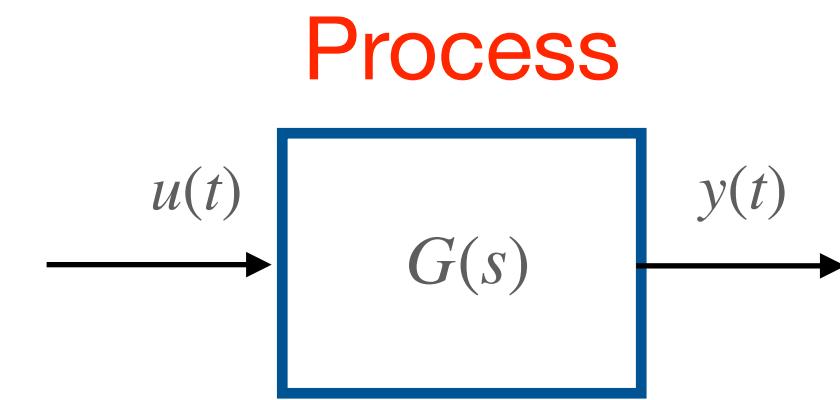


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Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

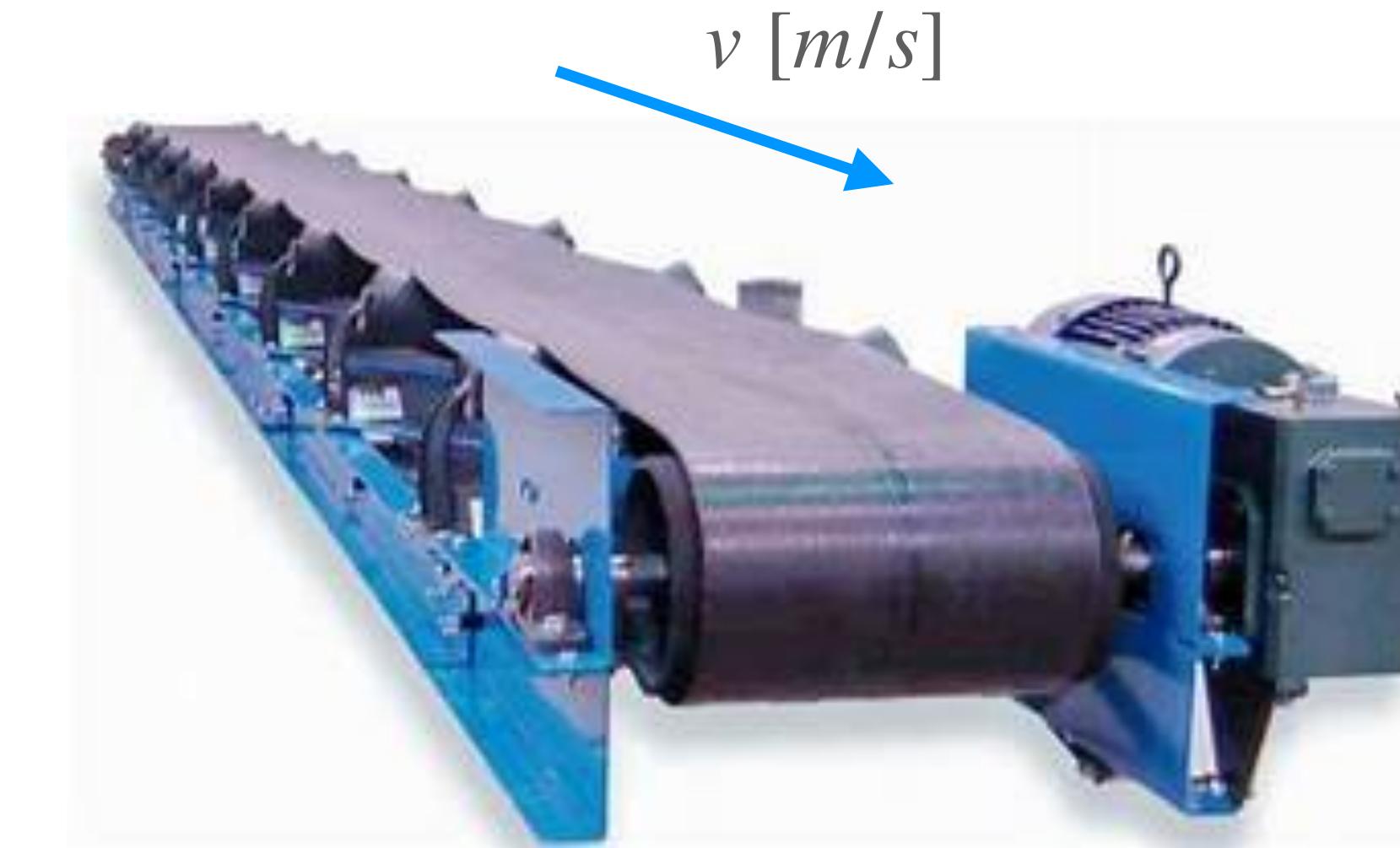
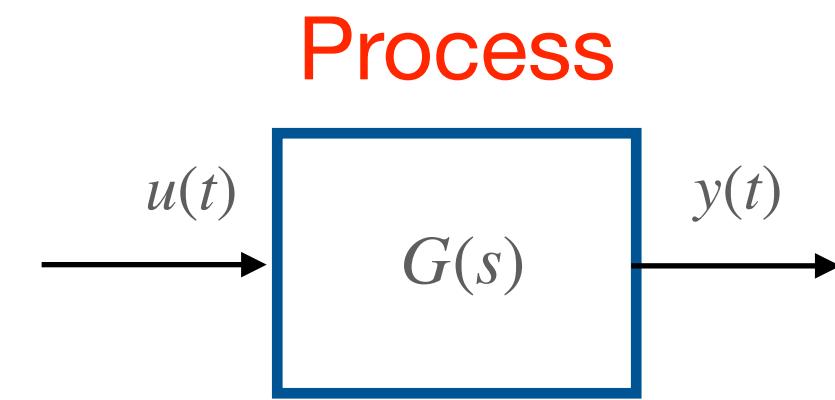


Image credit to: <https://www.gramconveyor.com/how-to-install-conveyor-belt/>

$$q_{OUT}(t) = q_{IN}(t - \tau) \quad \tau = \frac{L}{v}$$



Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

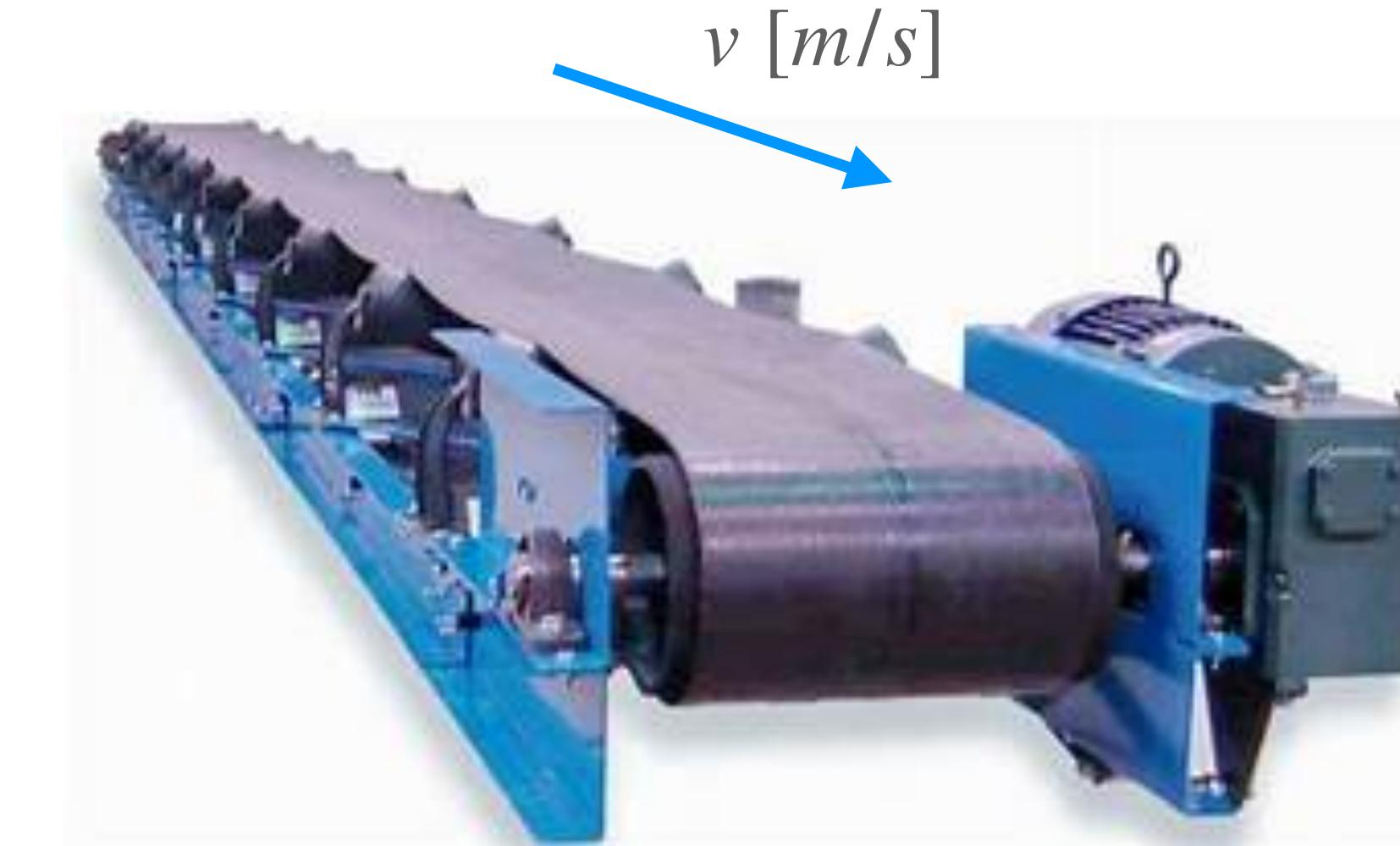
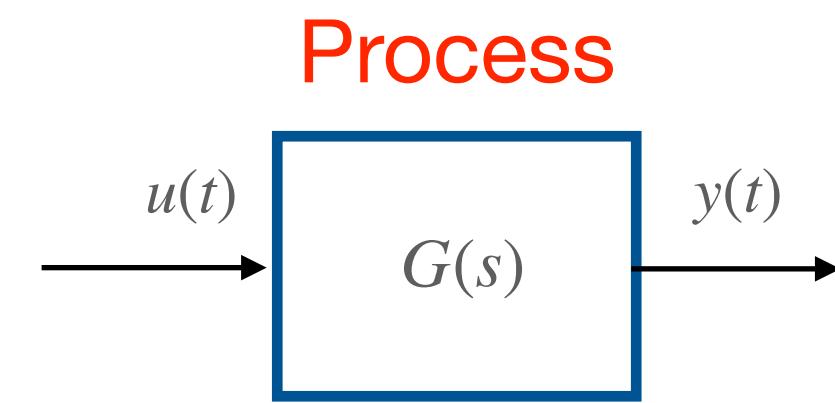


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$$\mathcal{L} \leftarrow q_{OUT}(t) = q_{IN}(t - \tau) \quad \tau = \frac{L}{v}$$
$$Q_{OUT}(s) = e^{-\tau s} Q_{IN}(s)$$



Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

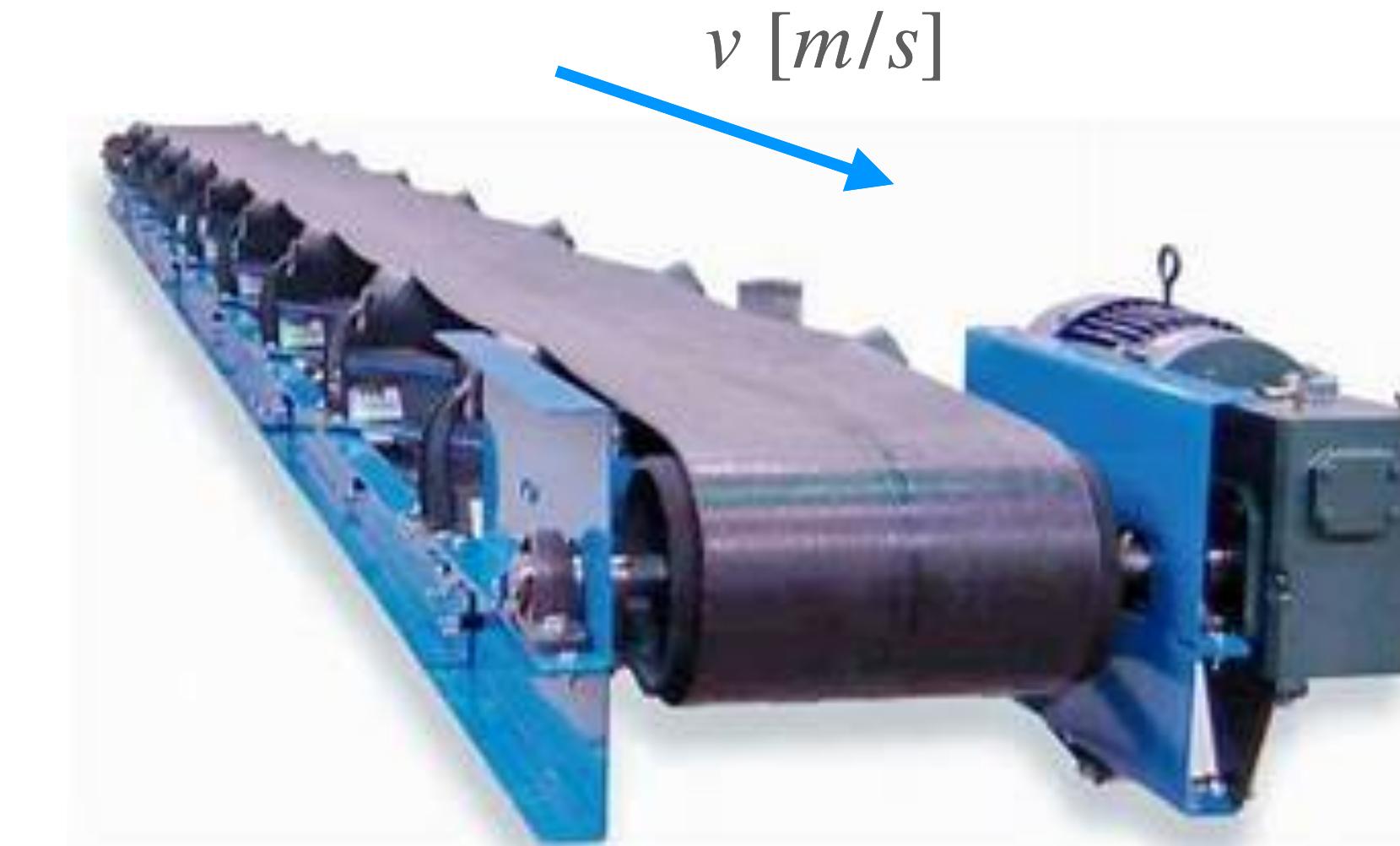
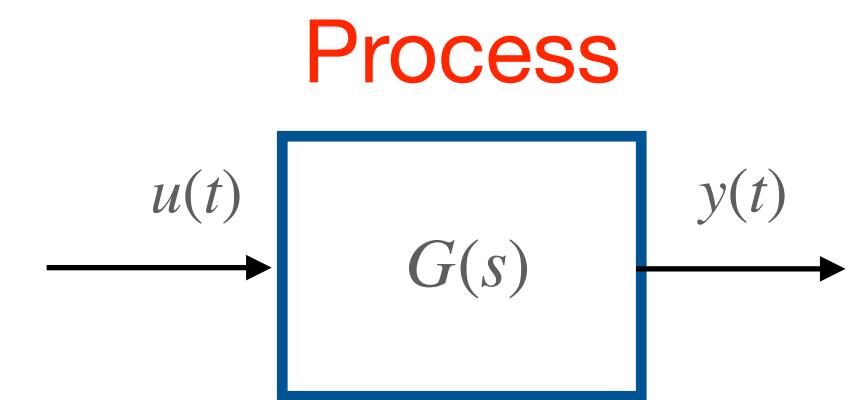


Image credit to: <https://www.gramconveyor.com/how-to-install-conveyor-belt/>

$$G(s) = e^{-\tau s} \quad \leftarrow \quad \mathcal{L} \quad q_{OUT}(t) = q_{IN}(t - \tau) \quad \tau = \frac{L}{v}$$
$$Q_{OUT}(s) = e^{-\tau s} Q_{IN}(s)$$

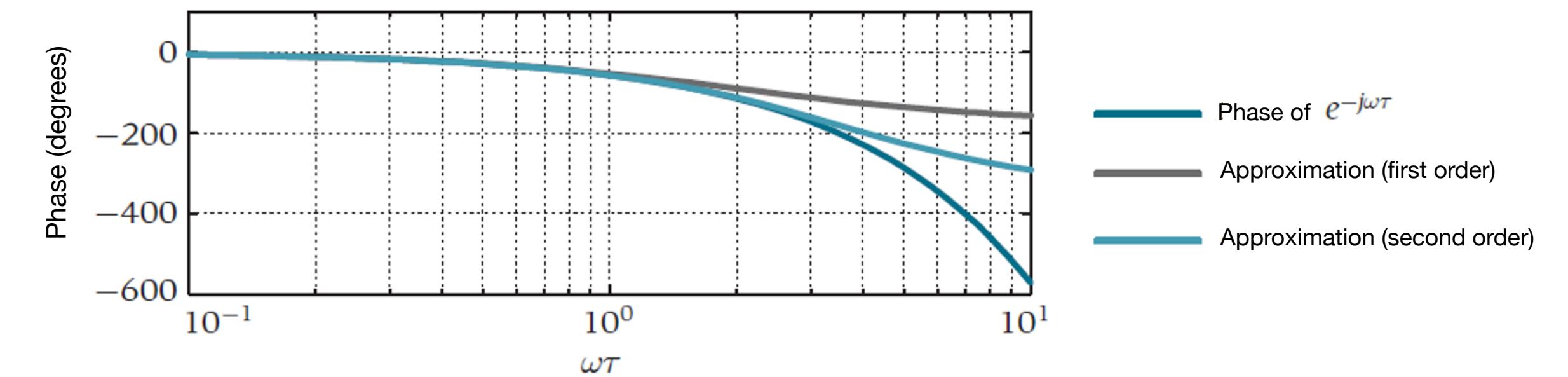


Control of LTI Systems with Delays

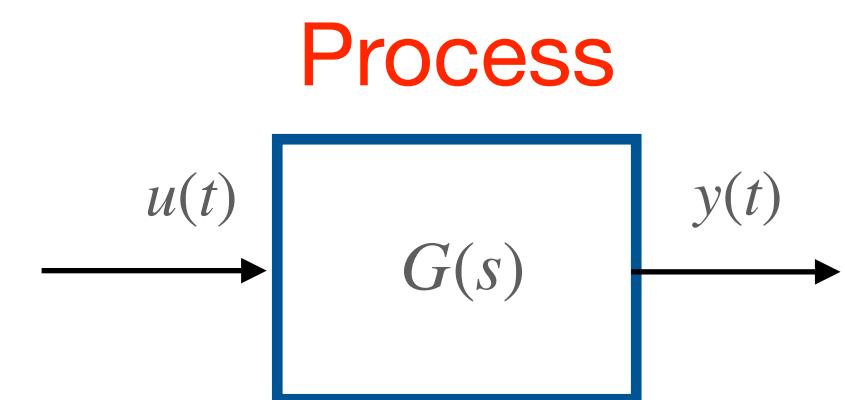


$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Phase Lag due to the delay term

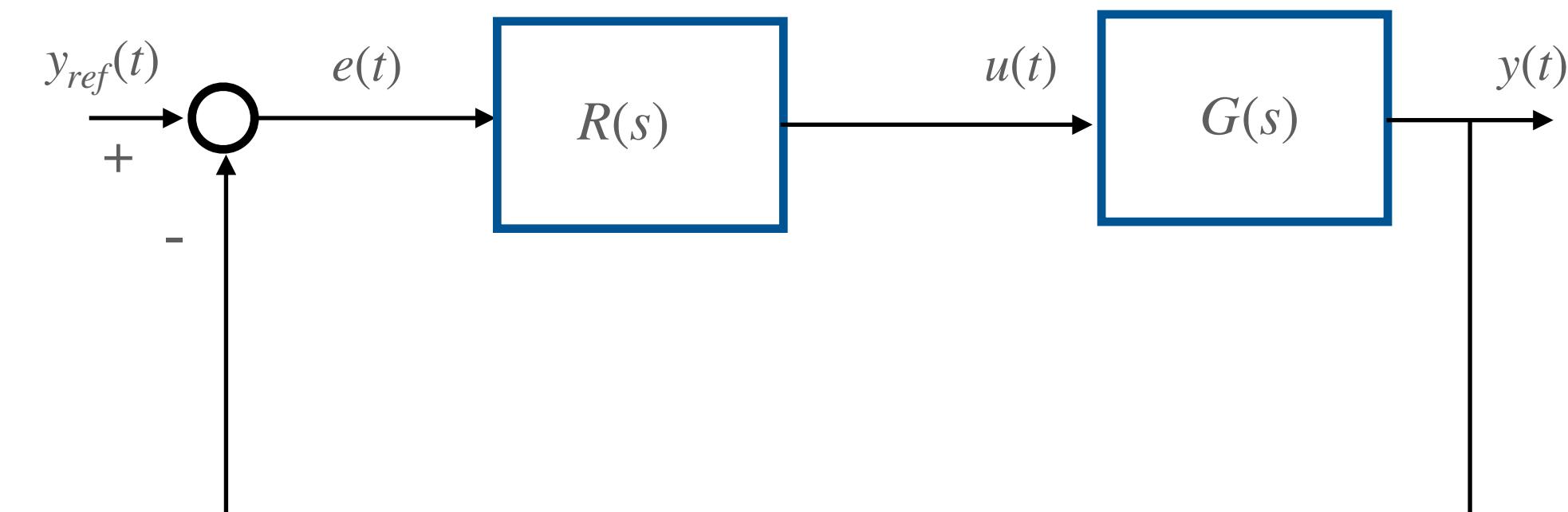
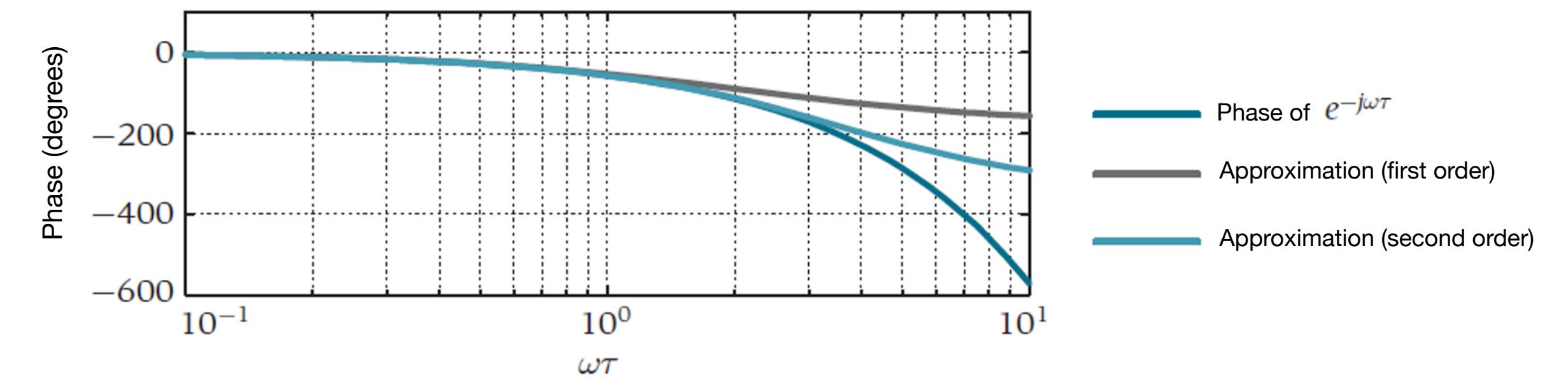


Control of LTI Systems with Delays

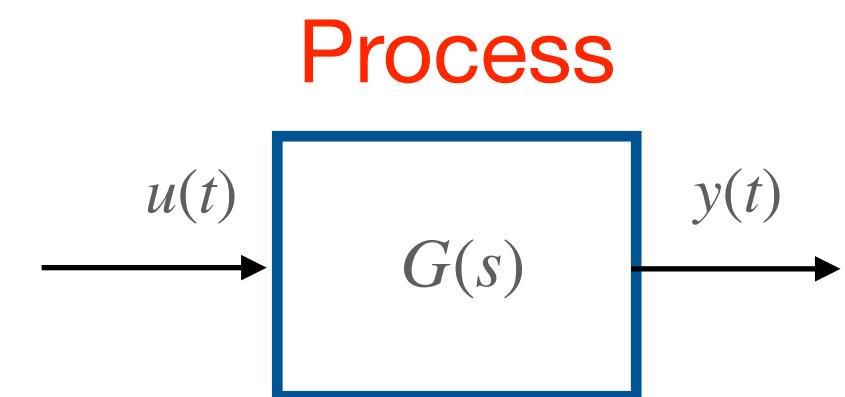


$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Phase Lag due to the delay term



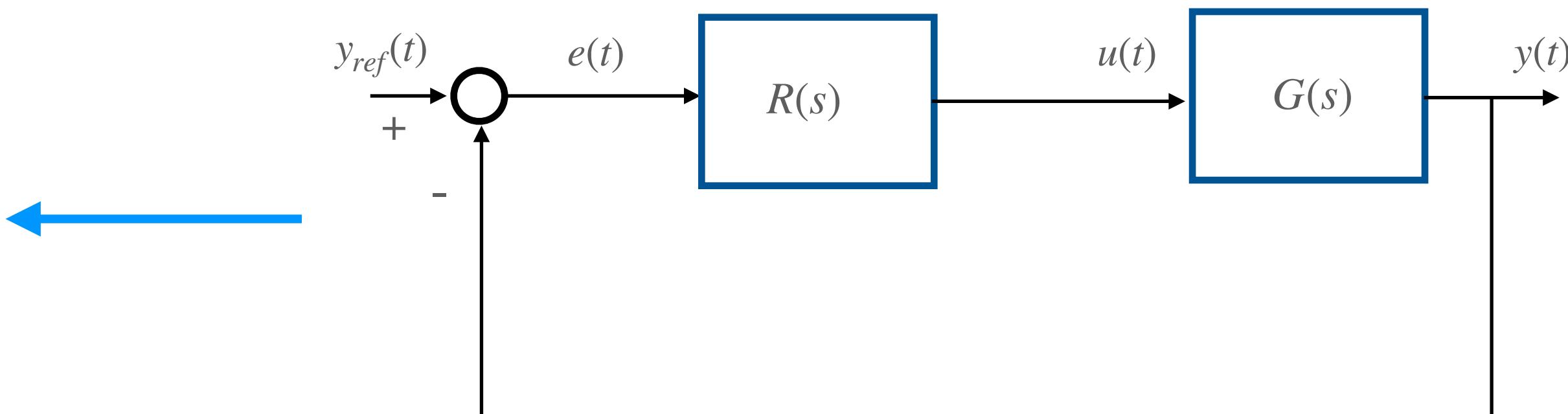
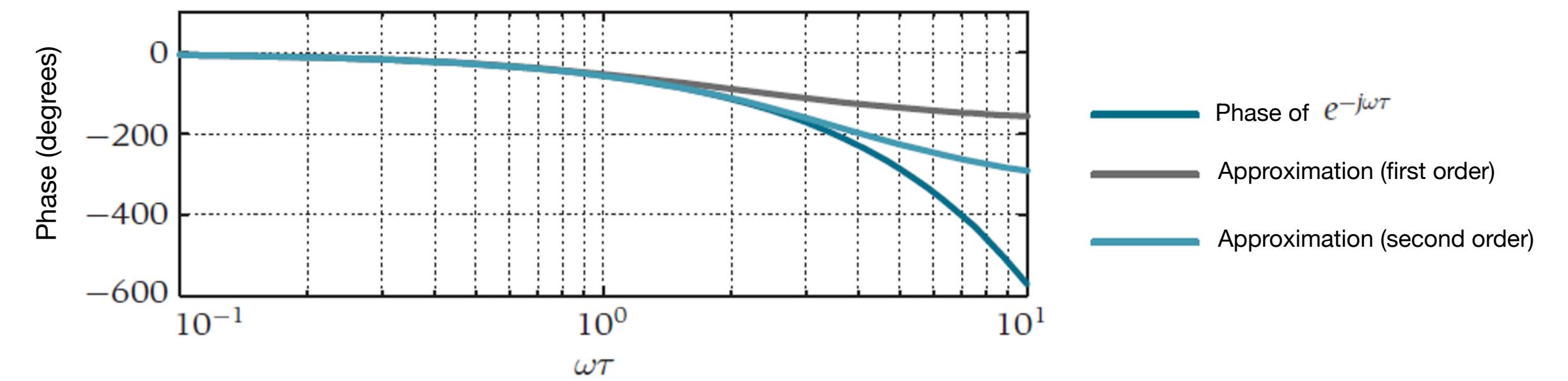
Control of LTI Systems with Delays



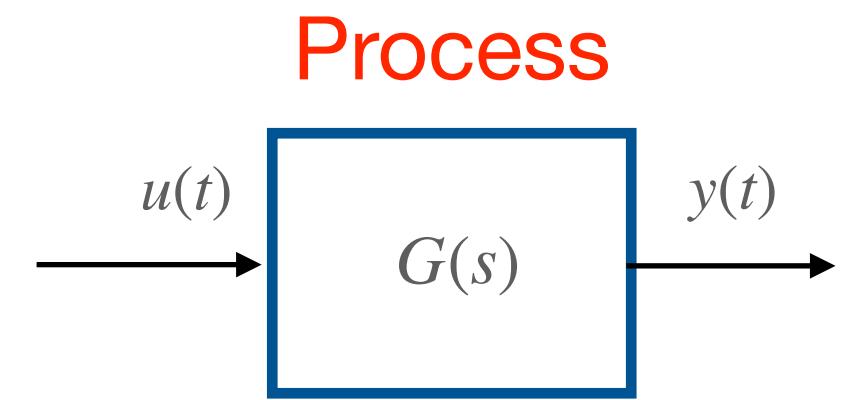
$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Recommendation: $\omega_{BW_{CL}}$ sufficiently narrow

Phase Lag due to the delay term



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme



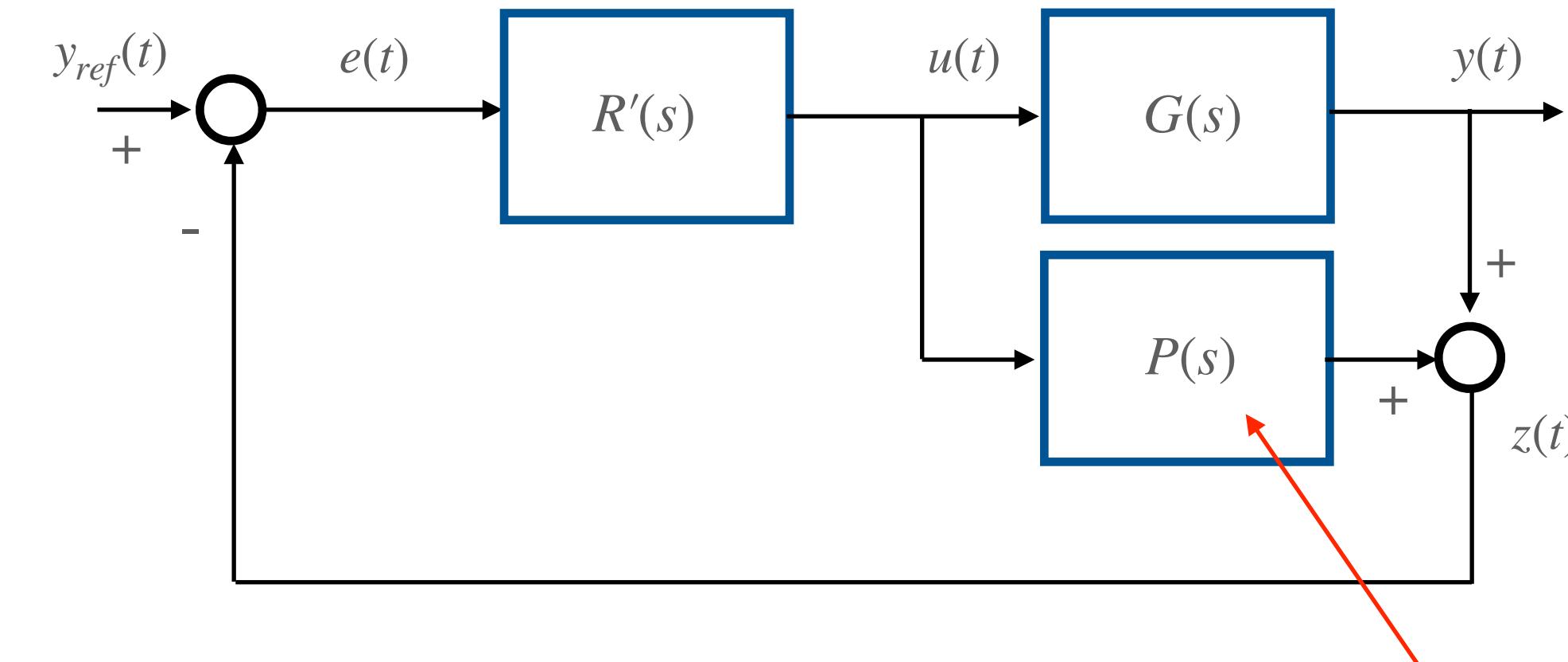
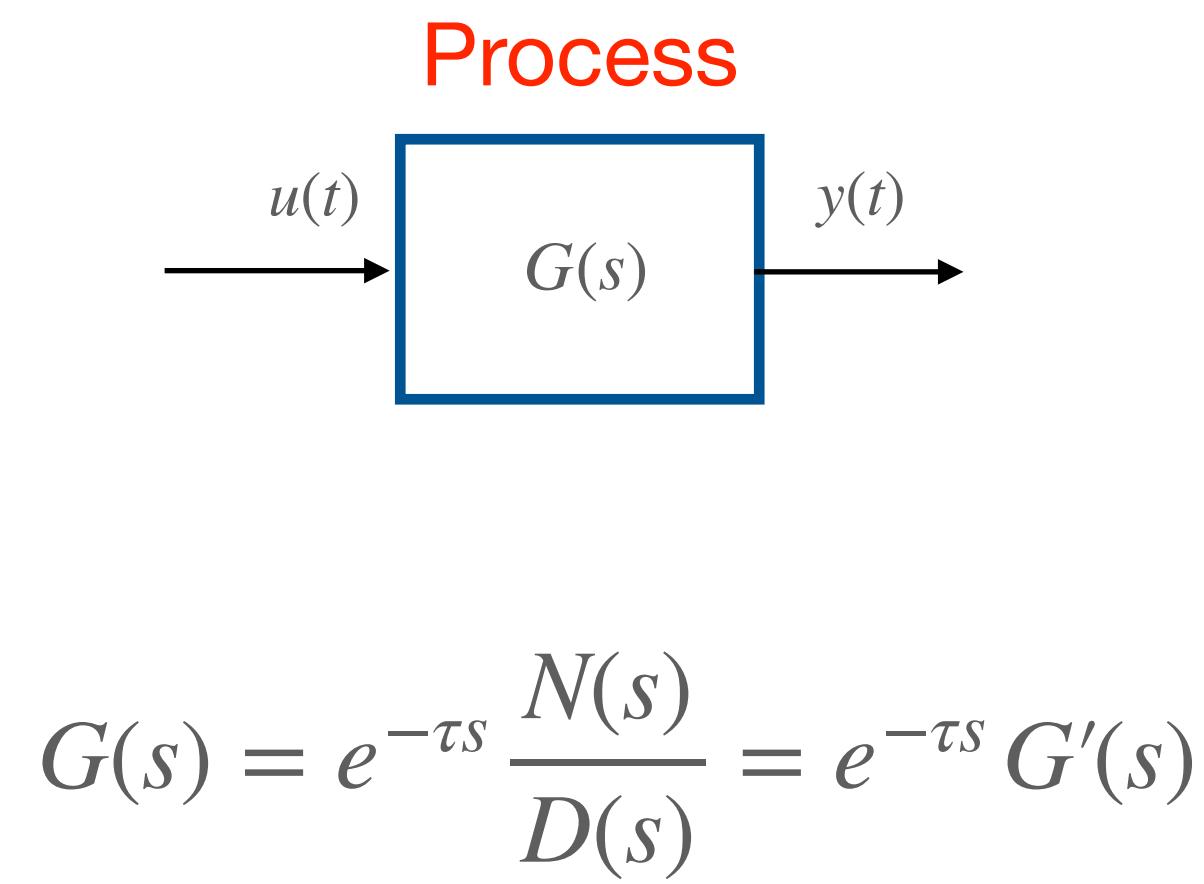
$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Assumptions:

- $G'(s)$ As. Stable
- τ known



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme



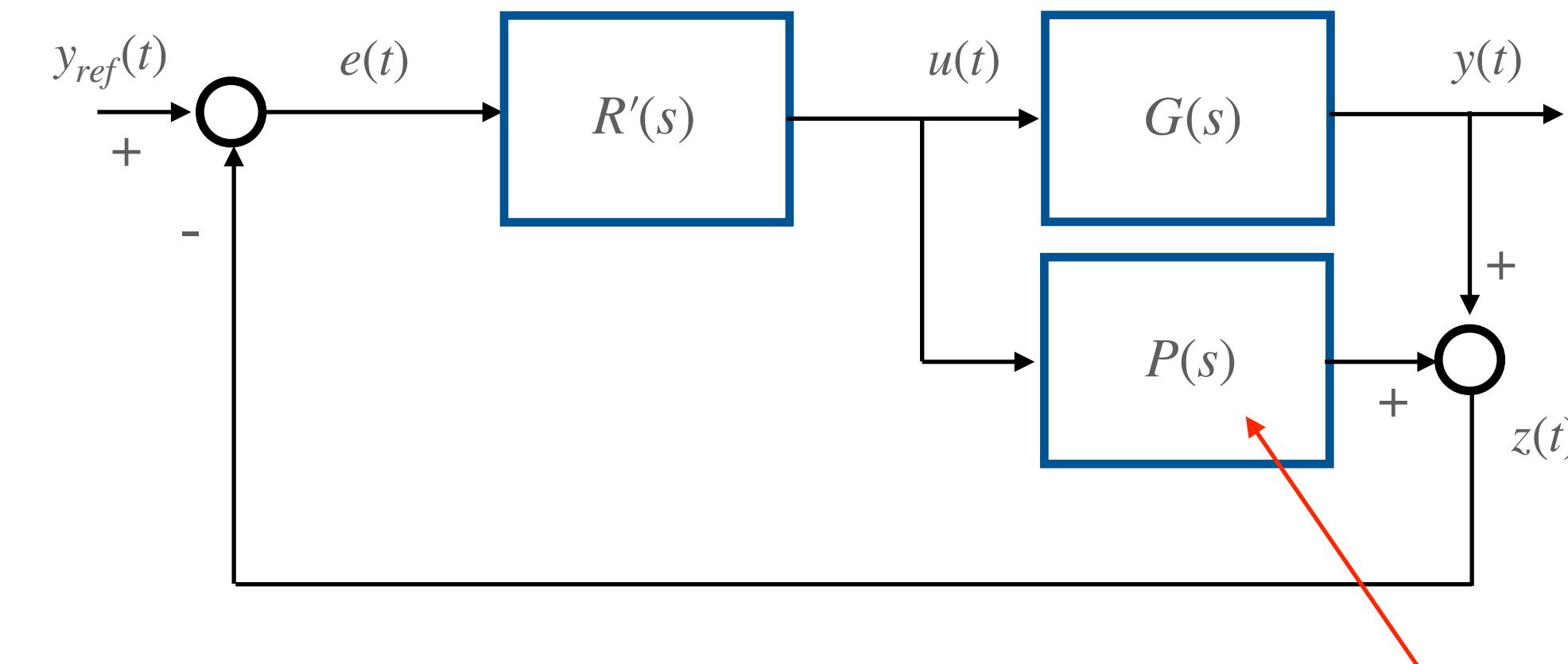
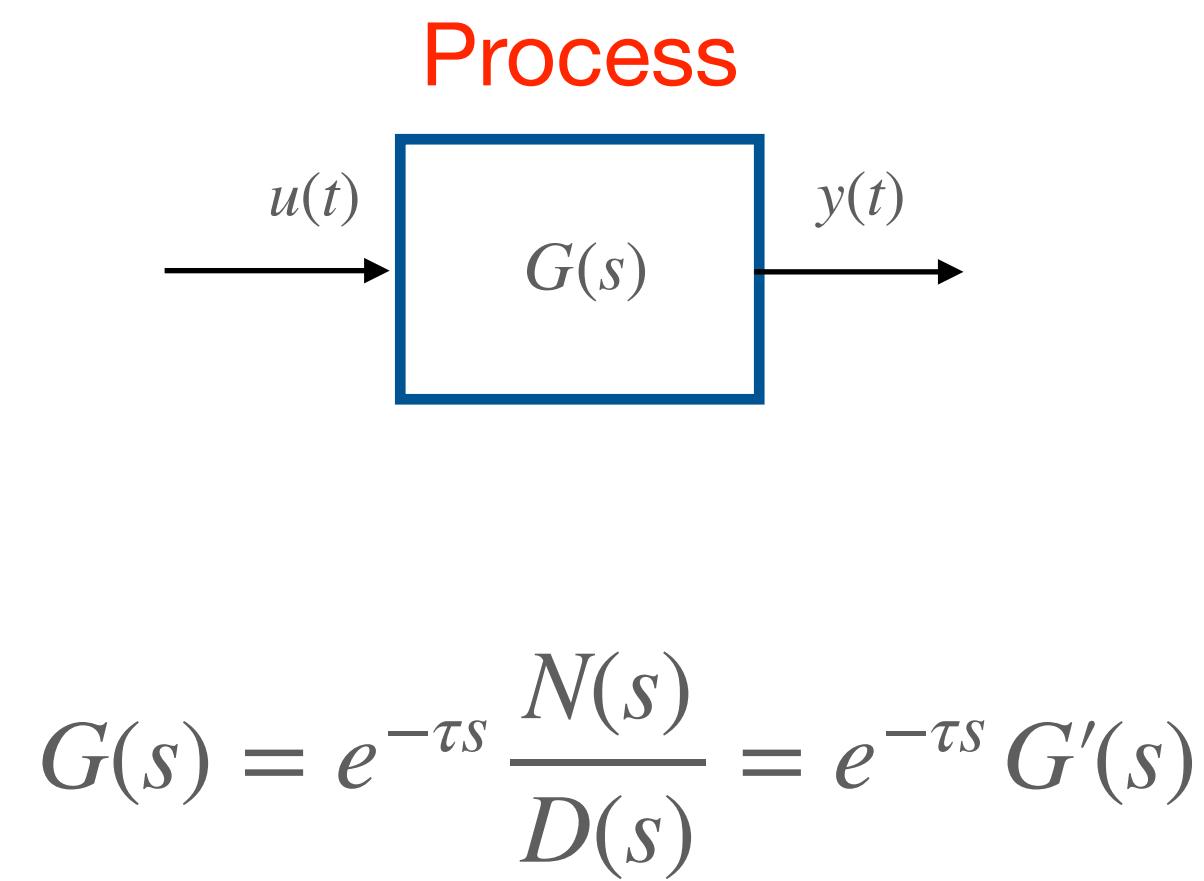
Smith Predictor

Assumptions:

- $G'(s)$ As. Stable
- τ known



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme



Smith Predictor

Assumptions:

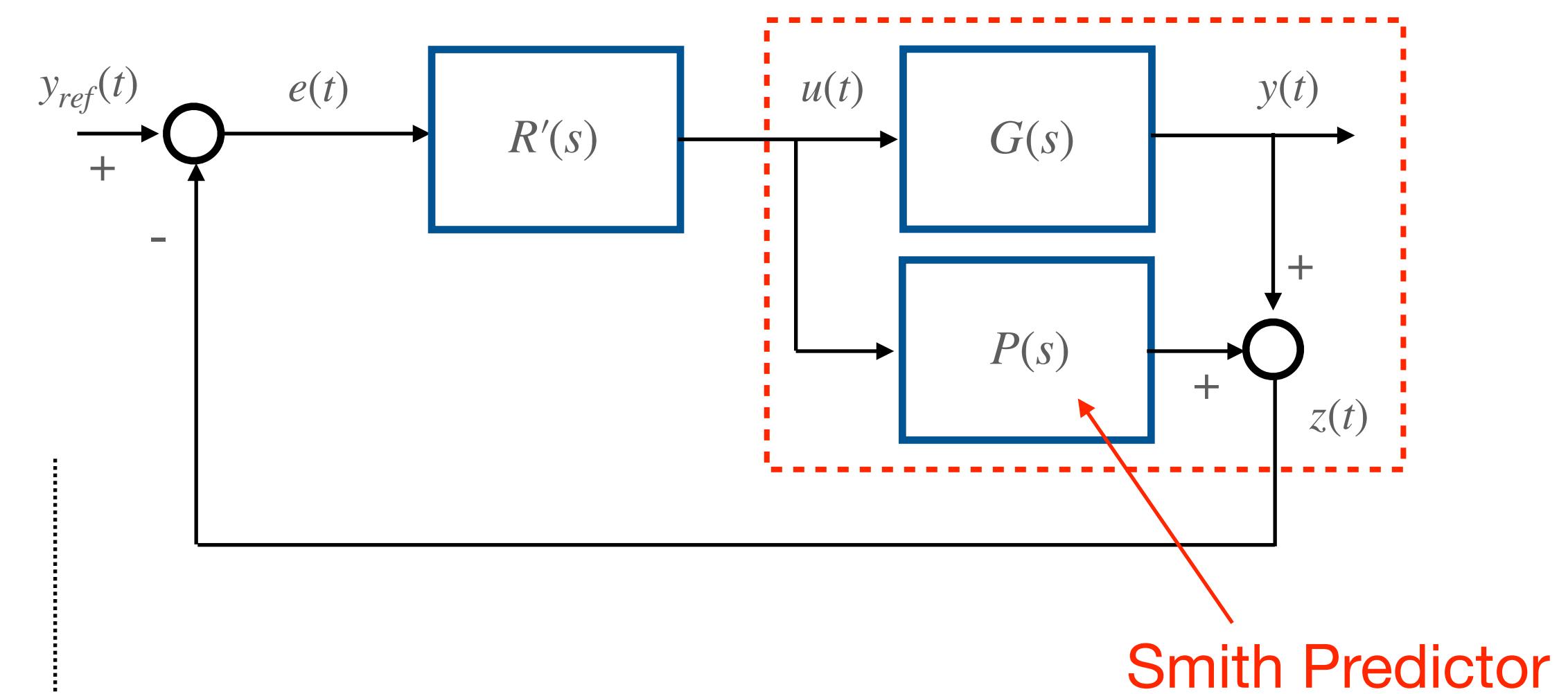
- $G'(s)$ As. Stable
- τ known

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$$\begin{aligned}
 u \rightarrow z : \quad G(s) + P(s) &= e^{-\tau s} G'(s) + (1 - e^{-\tau s}) G'(s) \\
 &= e^{-\tau s} G'(s) + G'(s) - e^{-\tau s} G'(s) \\
 &= G'(s)
 \end{aligned}$$

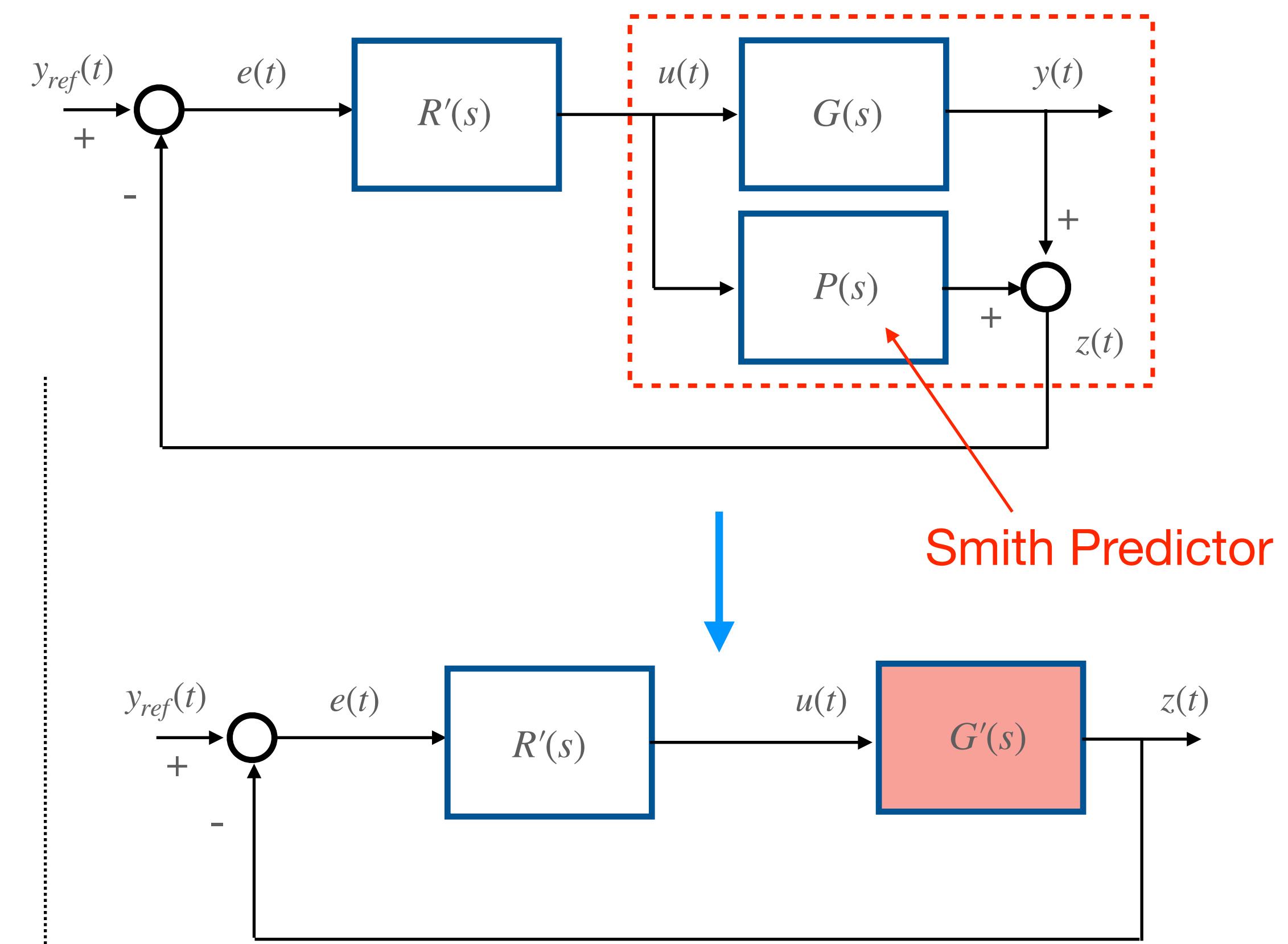


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

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 \end{aligned}$$



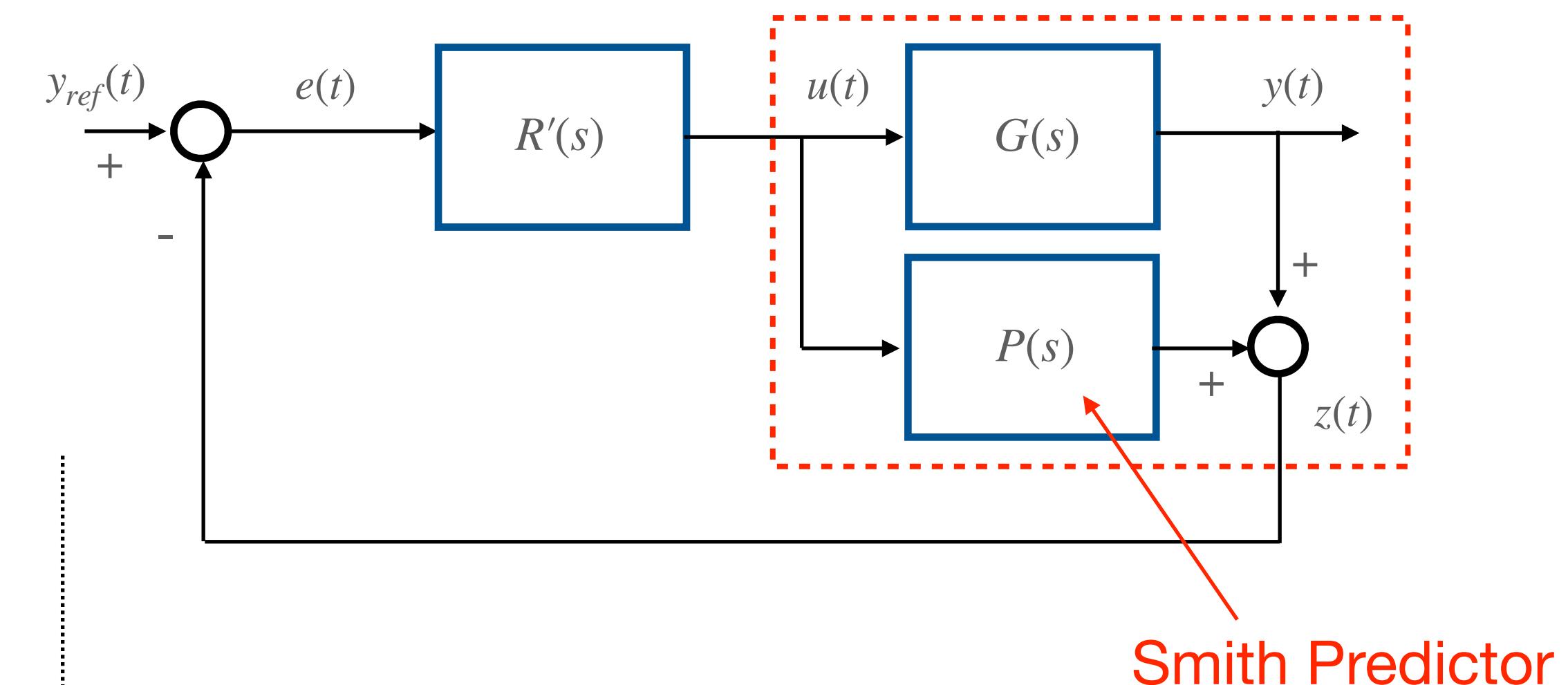
Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$$\begin{aligned}
 u \rightarrow z : \quad G(s) + P(s) &= e^{-\tau s} G'(s) + (1 - e^{-\tau s}) G'(s) \\
 &= e^{-\tau s} G'(s) + G'(s) - e^{-\tau s} G'(s) \\
 &= G'(s)
 \end{aligned}$$



Warning:

$$z(t) \neq y(t)$$

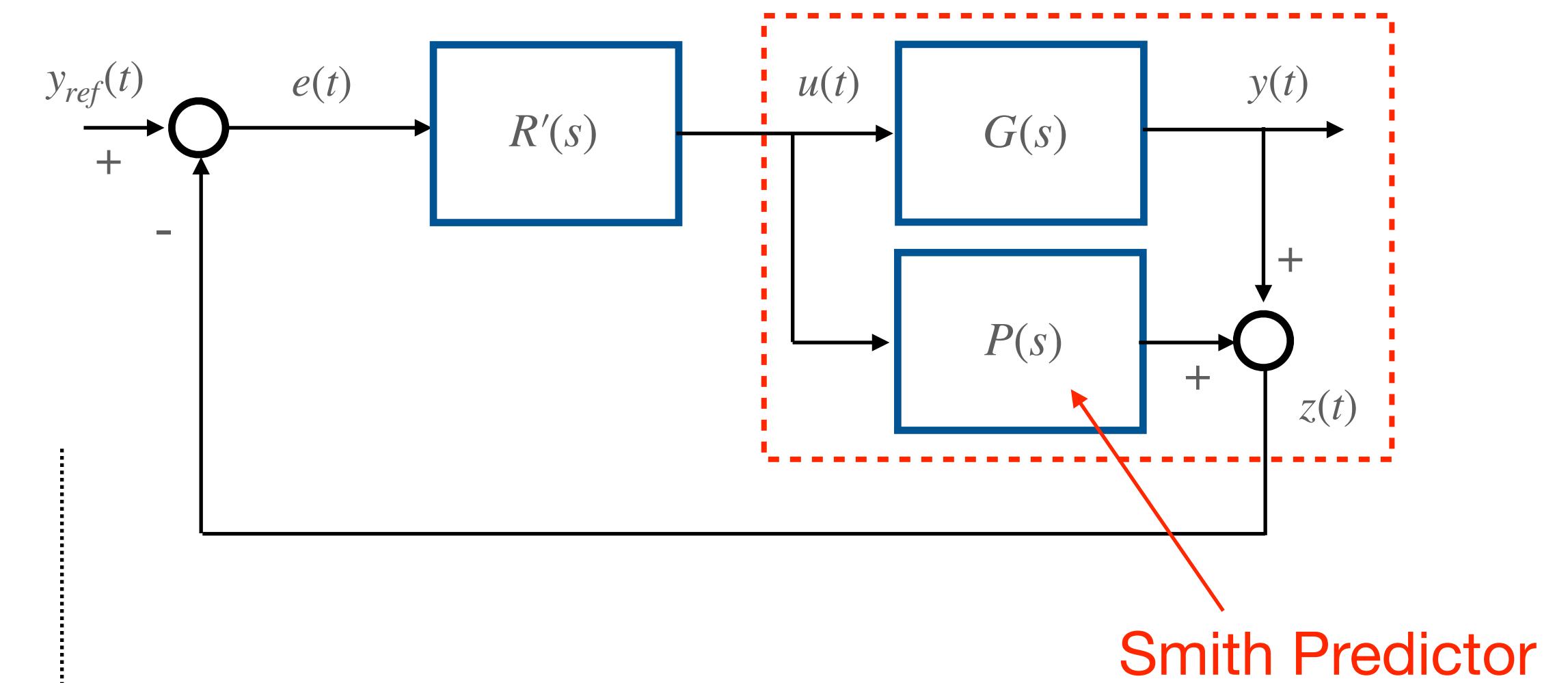


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$z(t) \neq y(t)$ → $z(t) ?$



$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$z(t) \neq y(t)$

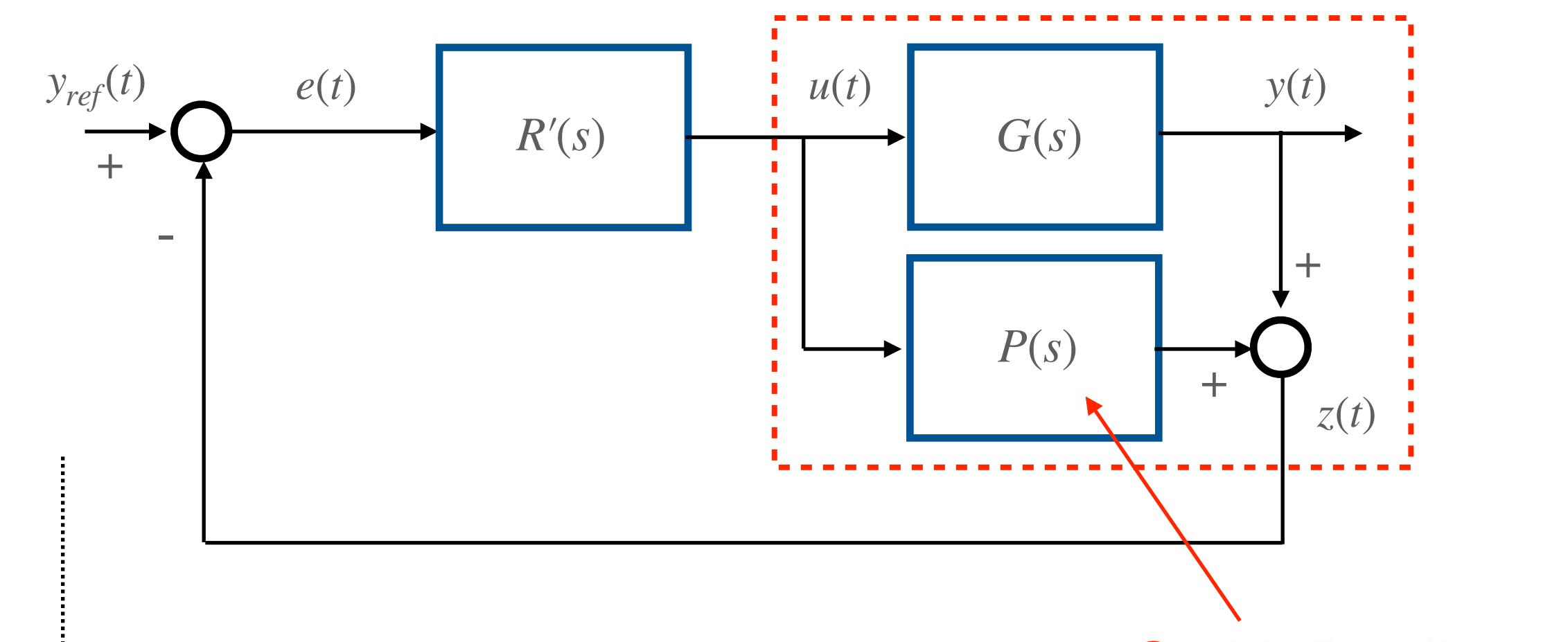


$z(t) ?$

In open loop:

$$u \rightarrow z : G'(s)$$

$$u \rightarrow y : e^{-\tau s} G'(s)$$



$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$z(t) \neq y(t)$



$z(t) ?$

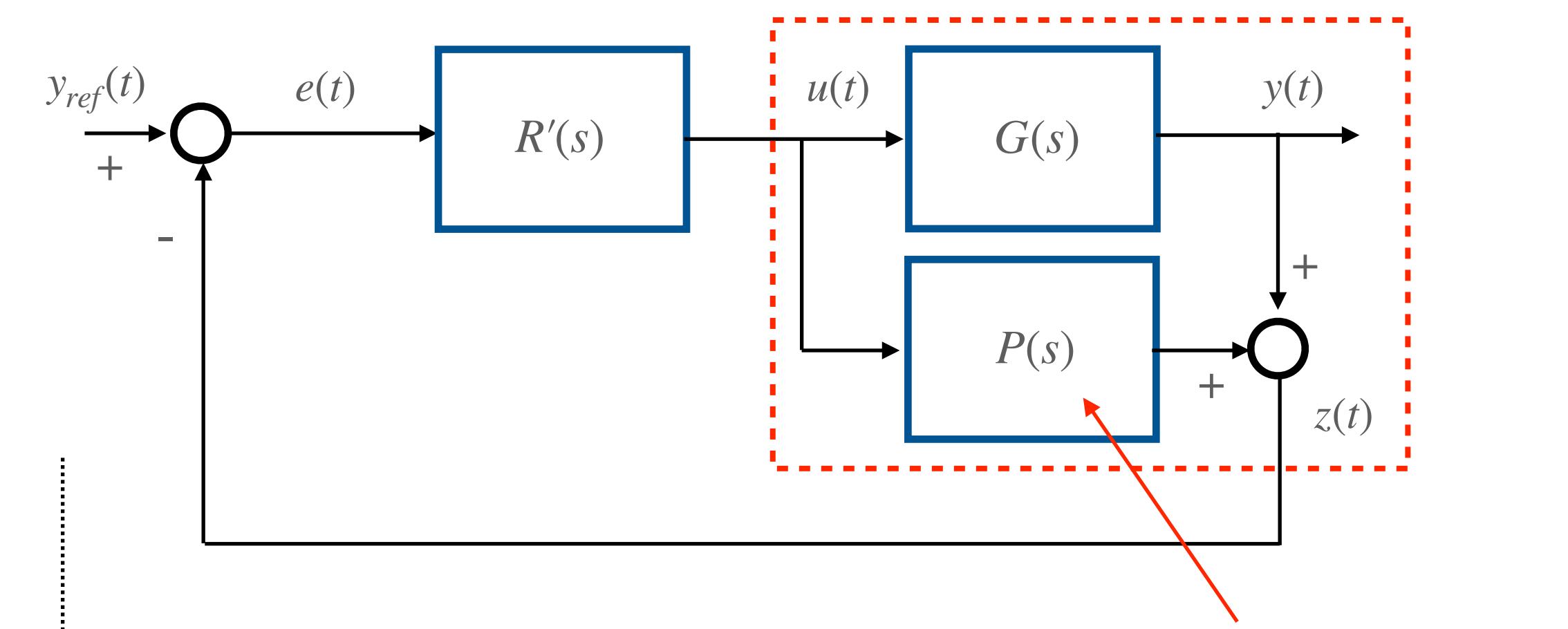
In open loop:

$$u \rightarrow z : G'(s)$$

$$u \rightarrow y : e^{-\tau s} G'(s)$$

\mathcal{L} ↘ $Z(s) = G'(s) U(s)$

$$Y(s) = e^{-\tau s} G'(s) U(s)$$



$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$$z(t) \neq y(t)$$



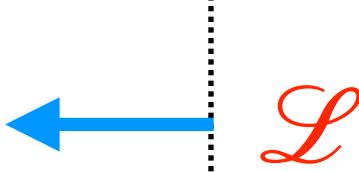
$z(t)$?

In open loop:

$$u \rightarrow z : G'(s)$$

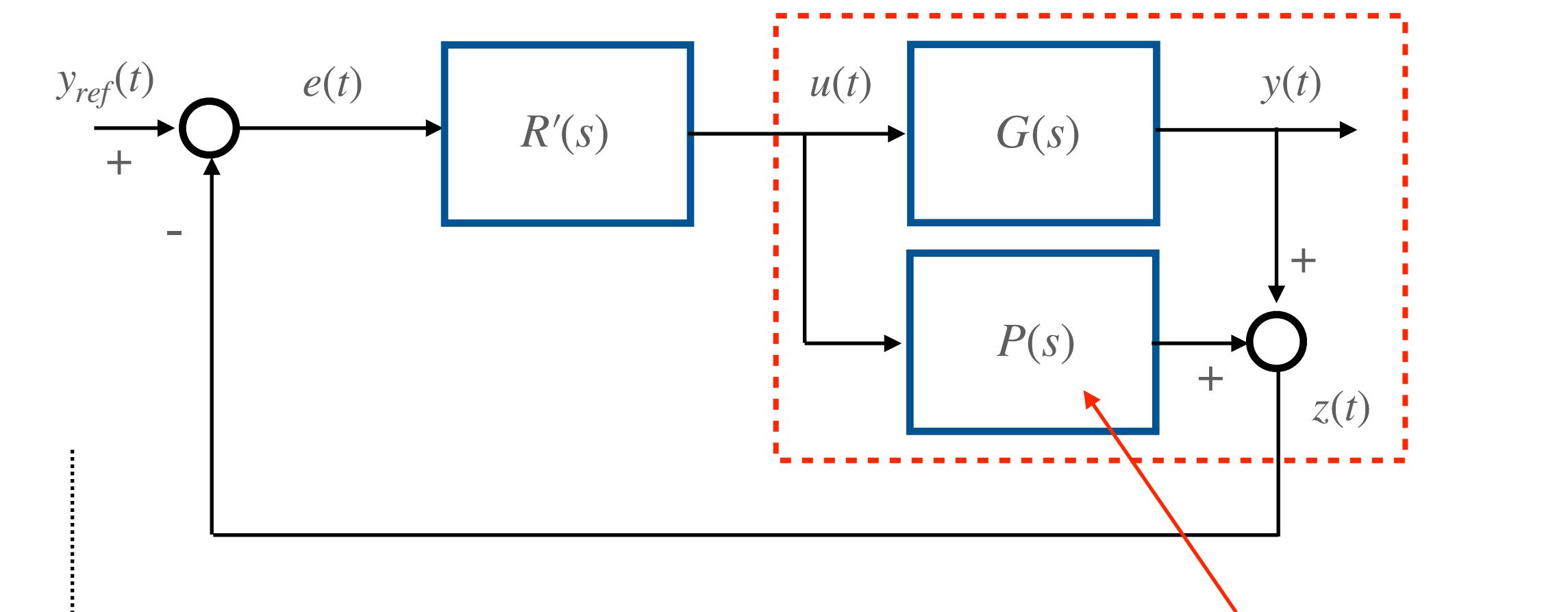
$$u \rightarrow y : e^{-\tau s} G'(s)$$

$$Z(s) = e^{\tau s} Y(s)$$



$$Z(s) = G'(s) U(s)$$

$$Y(s) = e^{-\tau s} G'(s) U(s)$$

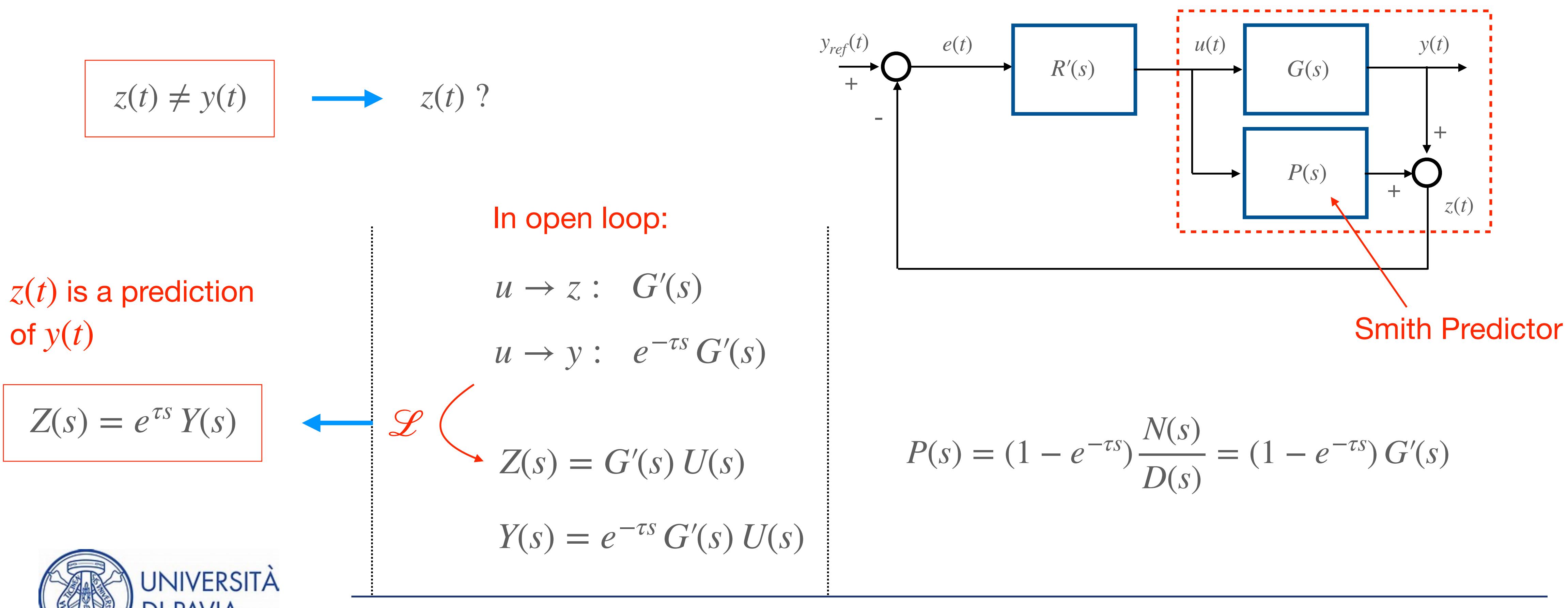


Smith Predictor

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

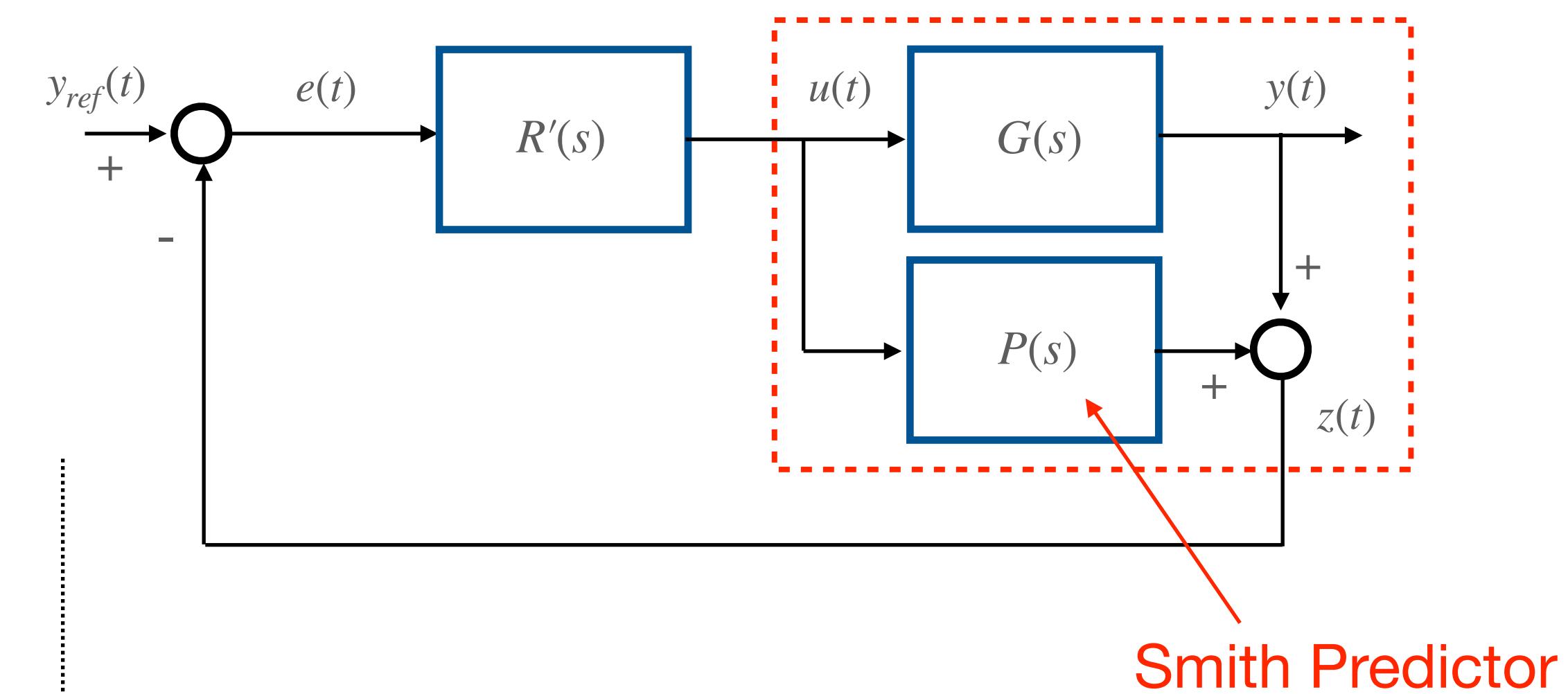


Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 1:

In the static case

$$P(0) = (1 - e^{-\tau_0}) G'(0) = 0$$



$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

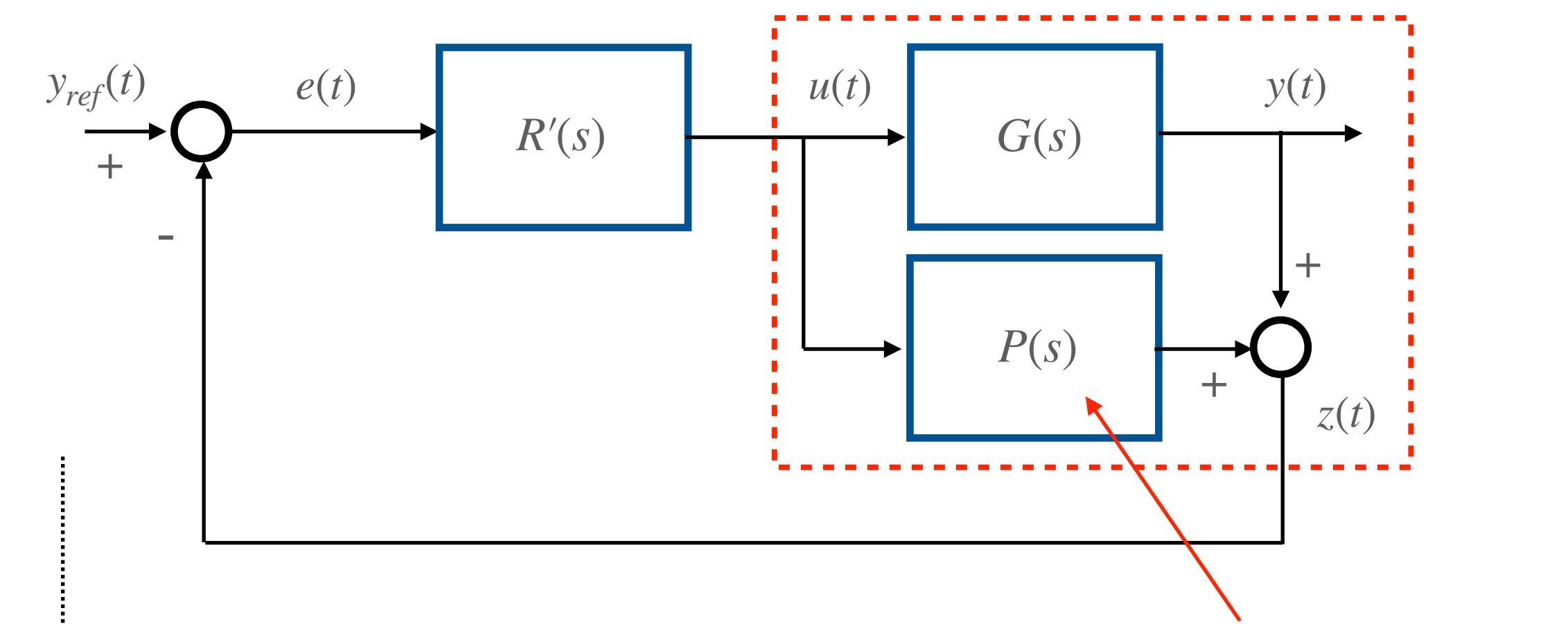
Observation 1:

In the static case

$$P(0) = (1 - e^{-\tau_0}) G'(0) = 0$$

Then, if the closed loop system is As. Stable
and $y_{ref}(t) = A \text{ step}(t)$:

$$z(t) \rightarrow y(t) \quad \text{for} \quad t \rightarrow \infty$$

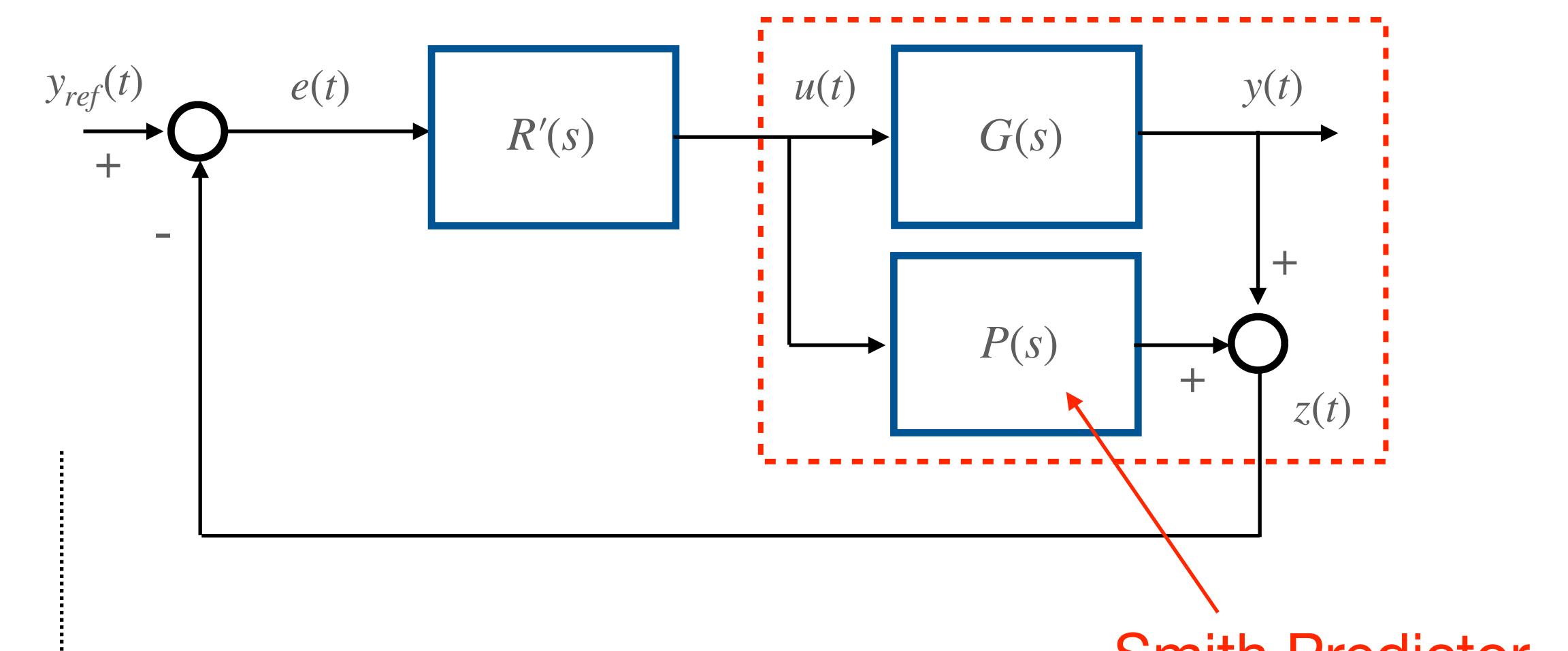
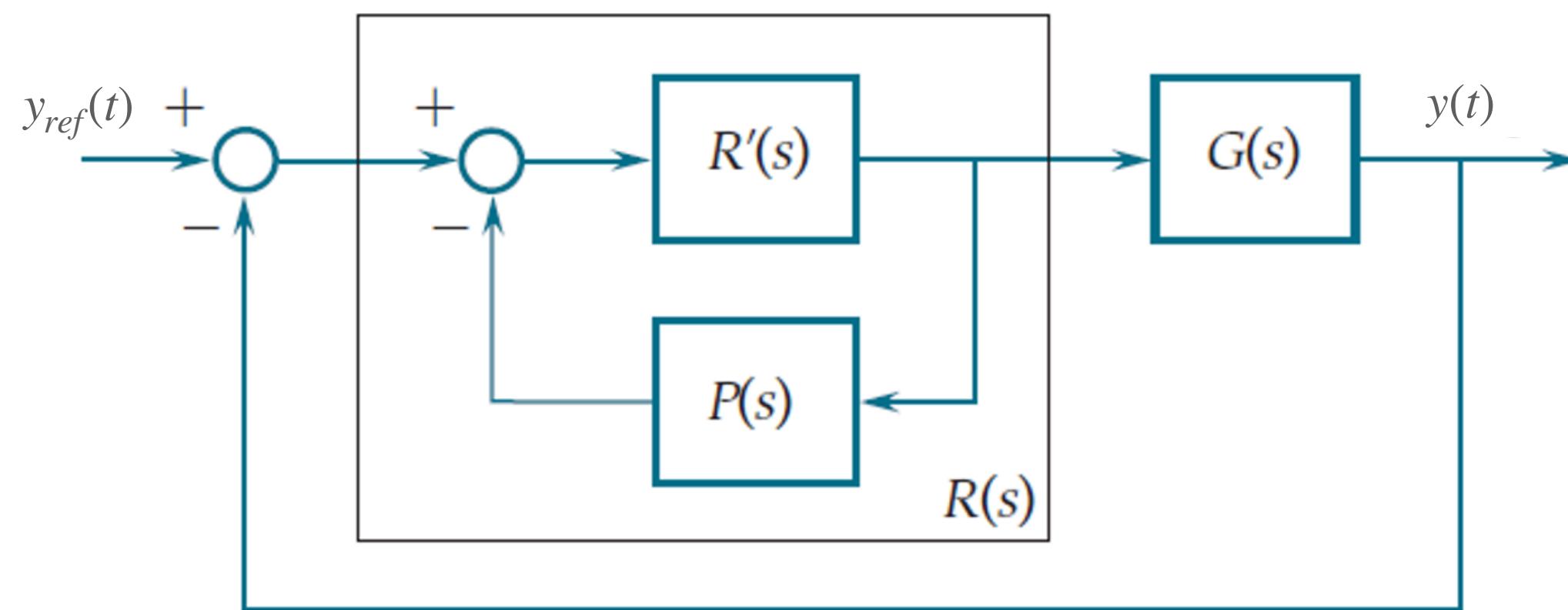


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation

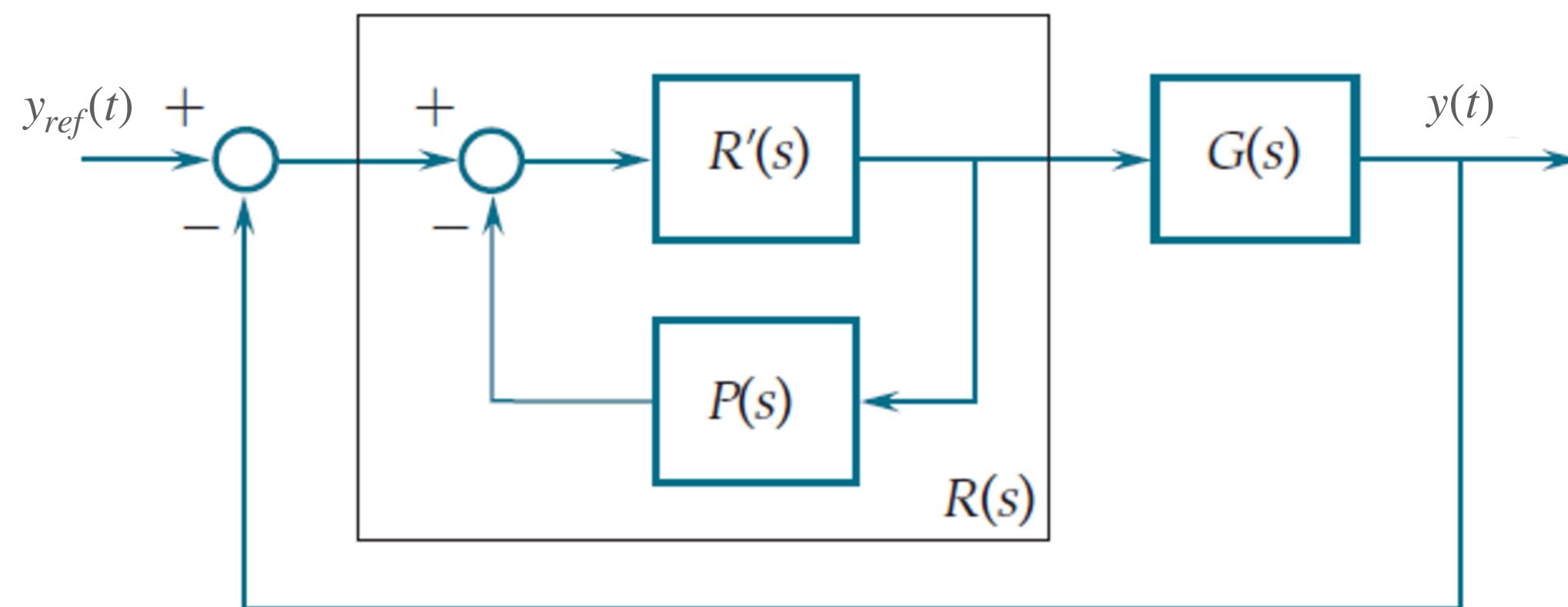


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

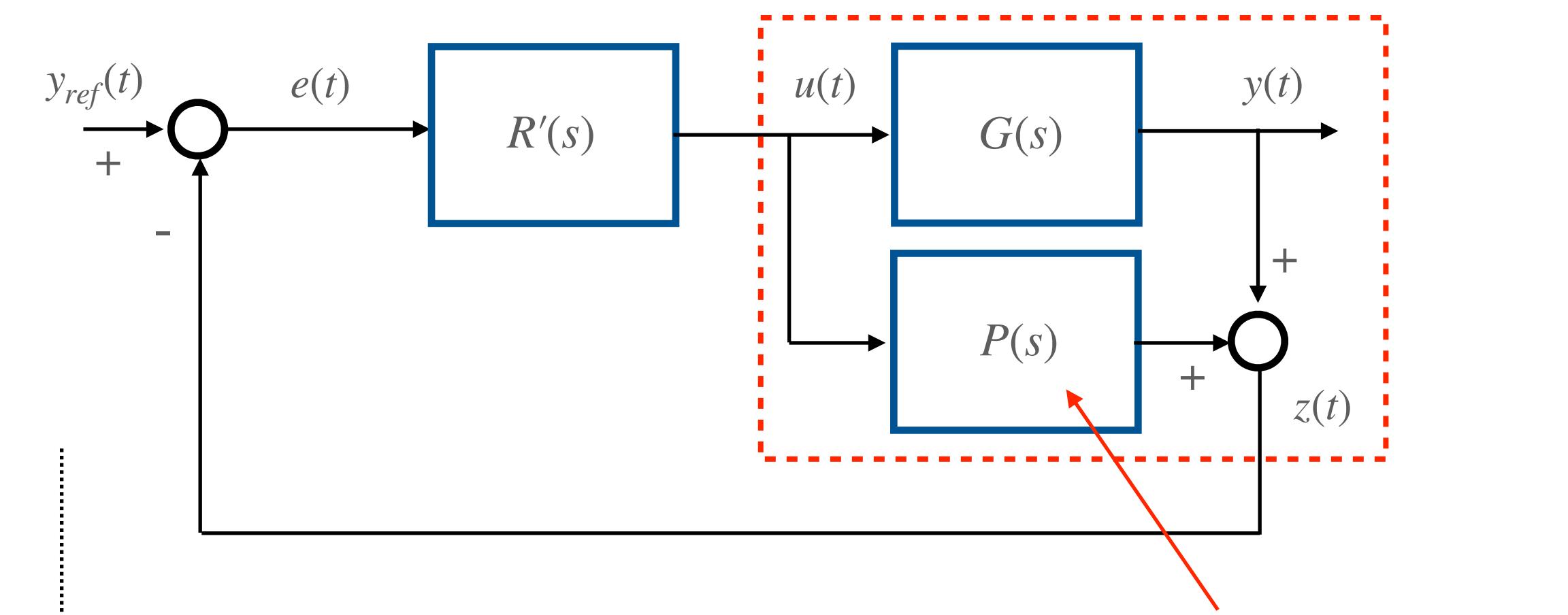


Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation



$$R(s) = \frac{R'(s)}{1 + R'(s)P(s)}$$



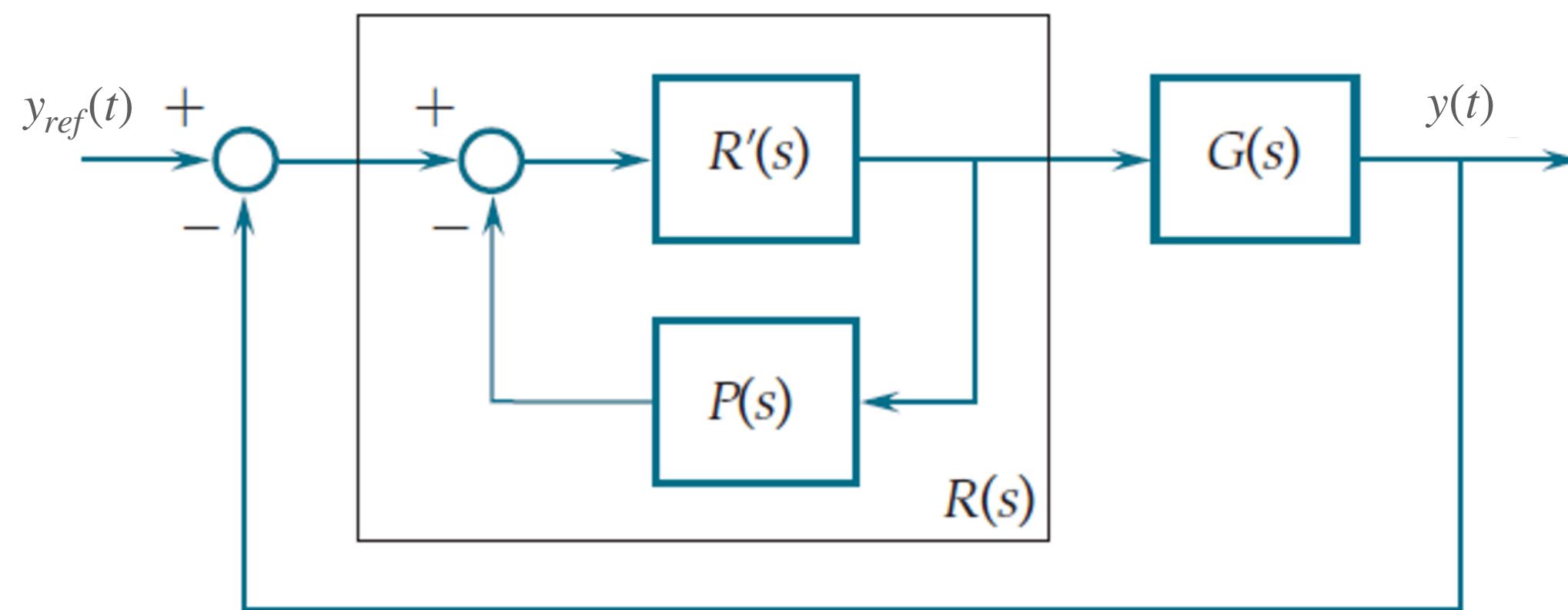
Smith Predictor

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

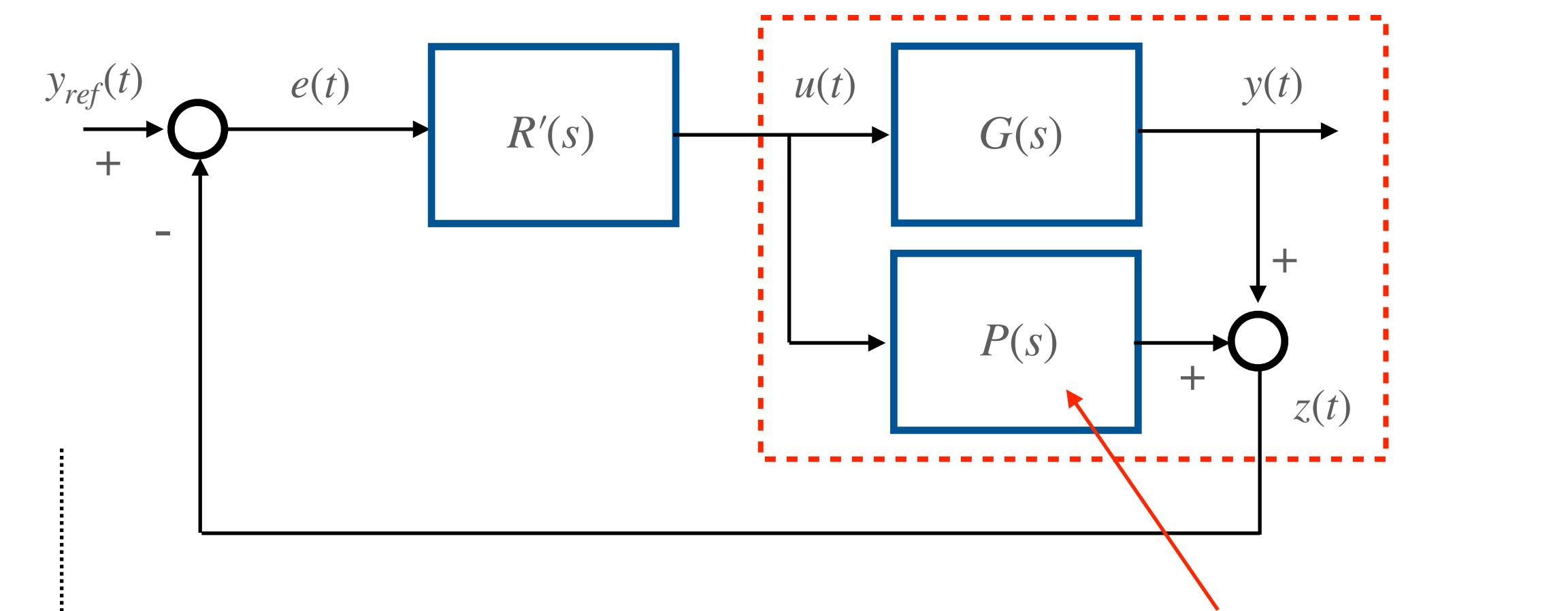


Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation



$$\begin{aligned} R(s) &= \frac{R'(s)}{1 + R'(s)P(s)} \\ &= \frac{N_{R'}(s)/D_{R'}(s)}{1 + (N_{R'}(s)/D_{R'}(s))(1 - e^{-\tau s})(N(s)/D(s))} \end{aligned}$$

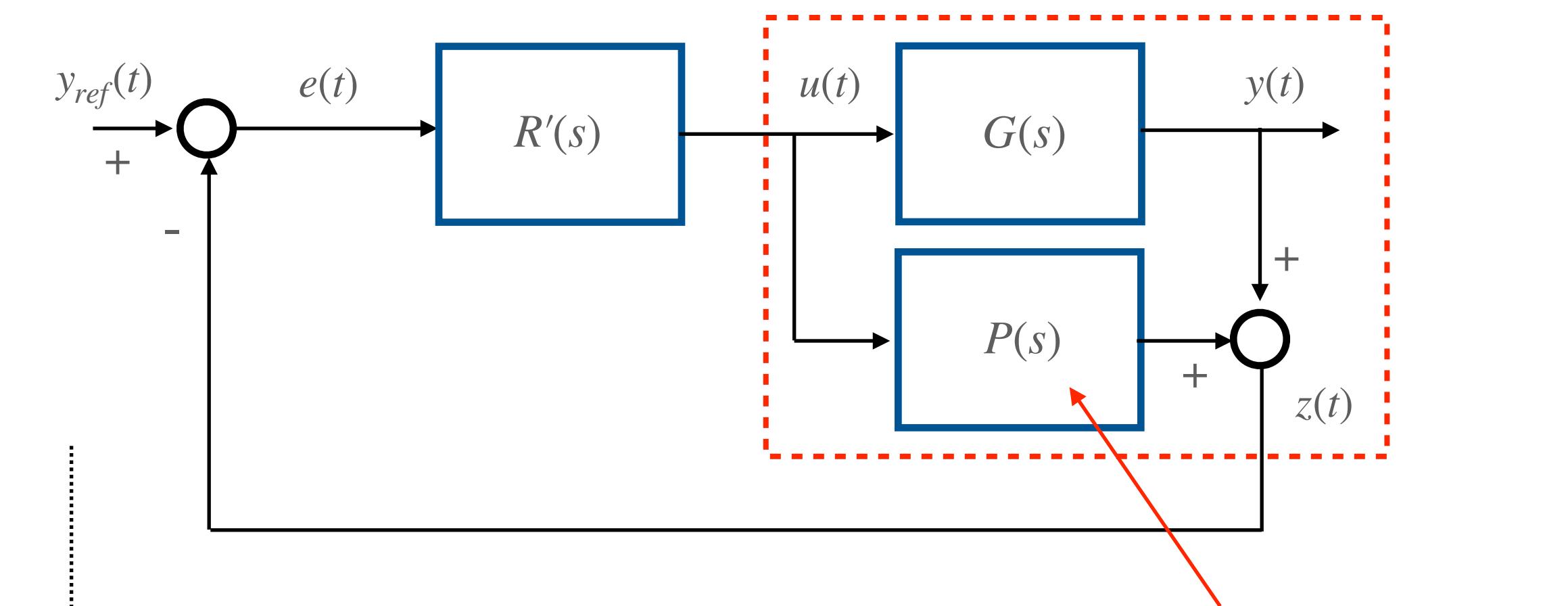
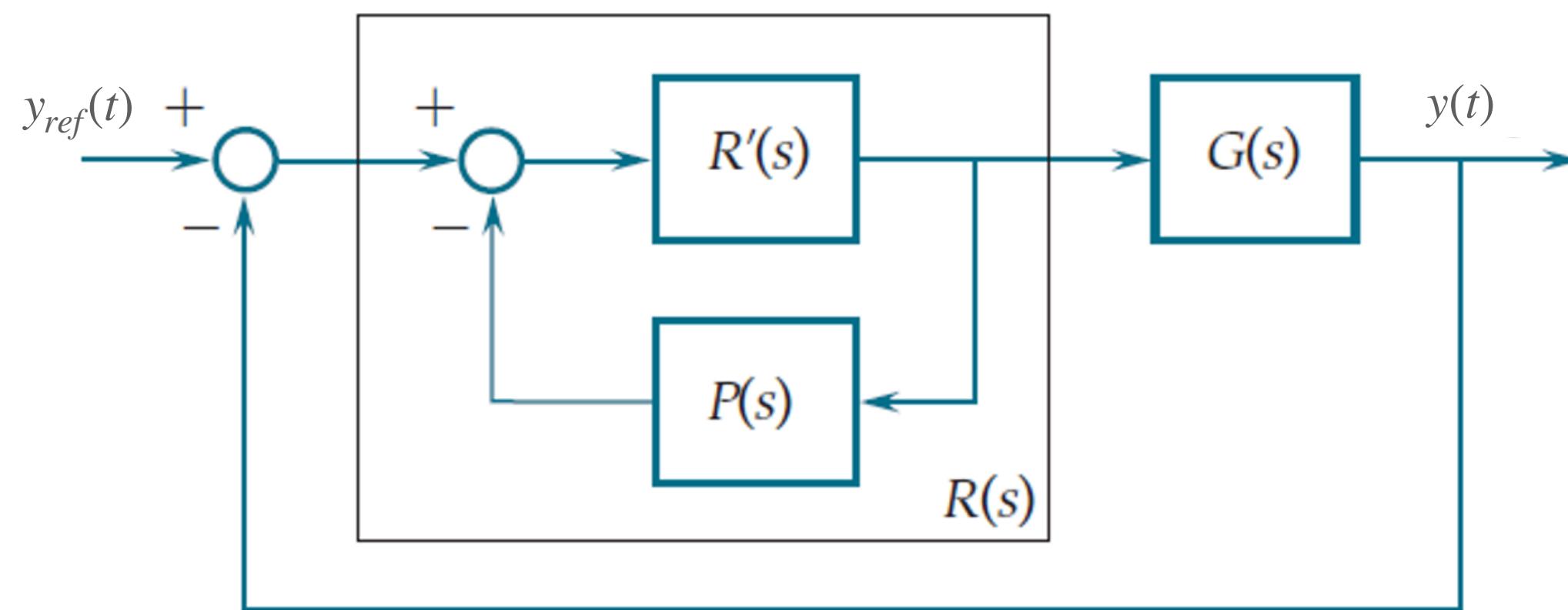


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation



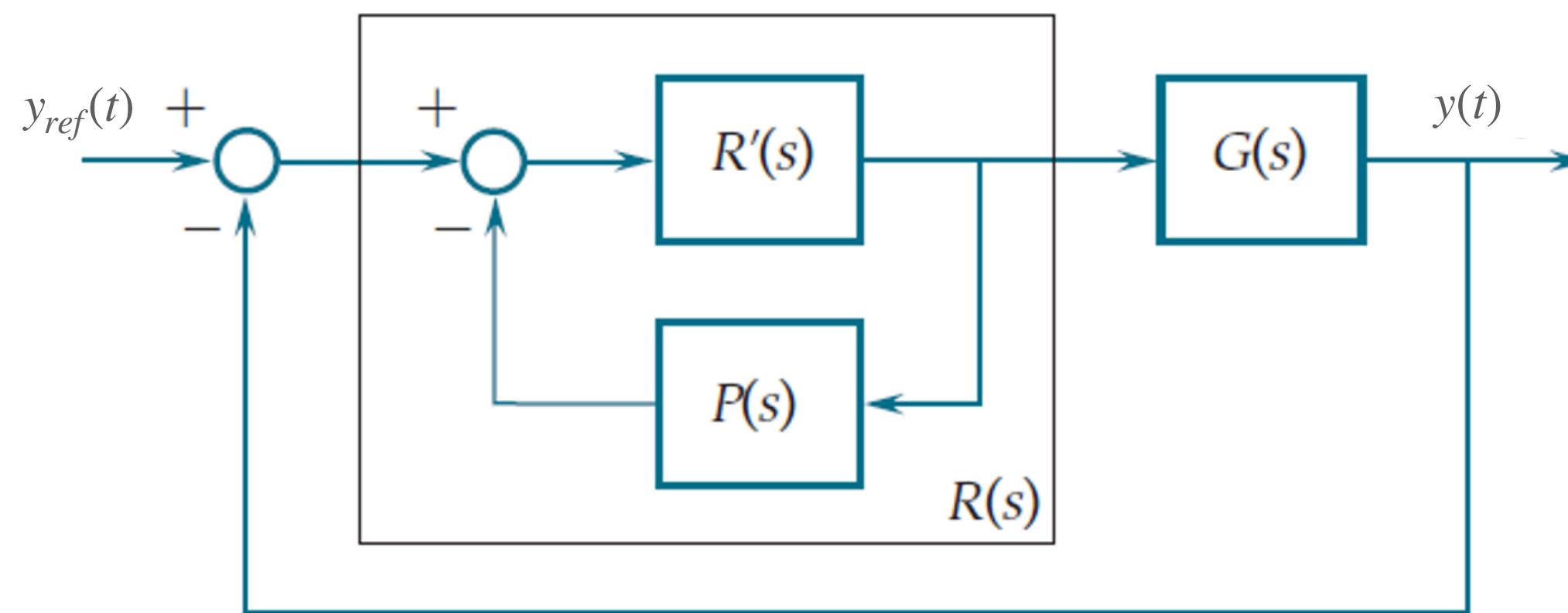
$$\begin{aligned}
 R(s) &= \frac{R'(s)}{1 + R'(s)P(s)} \\
 &= \frac{N_{R'}(s)/D_{R'}(s)}{1 + (N_{R'}(s)/D_{R'}(s))(1 - e^{-\tau s})(N(s)/D(s))} \\
 &= \frac{N_{R'}(s)D(s)}{D_{R'}(s)D(s) + N_{R'}(s)N(s)(1 - e^{-\tau s})}
 \end{aligned}$$

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

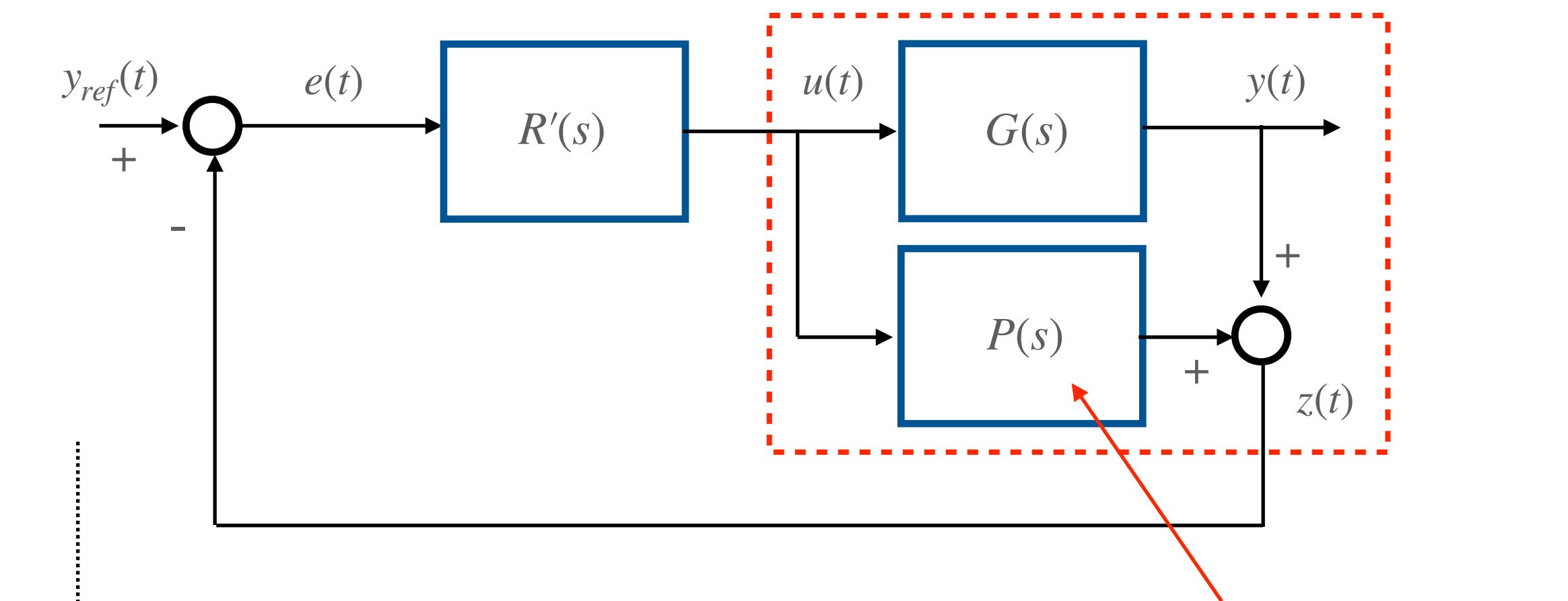


Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation



$$\begin{aligned}
 R(s) &= \frac{R'(s)}{1 + R'(s)P(s)} \\
 &= \frac{N_{R'}(s)/D_{R'}(s)}{1 + (N_{R'}(s)/D_{R'}(s))(1 - e^{-\tau s})(N(s)/D(s))} \\
 &= \frac{N_{R'}(s)D(s)}{D_{R'}(s)D(s) + N_{R'}(s)N(s)(1 - e^{-\tau s})}
 \end{aligned}$$



Smith Predictor

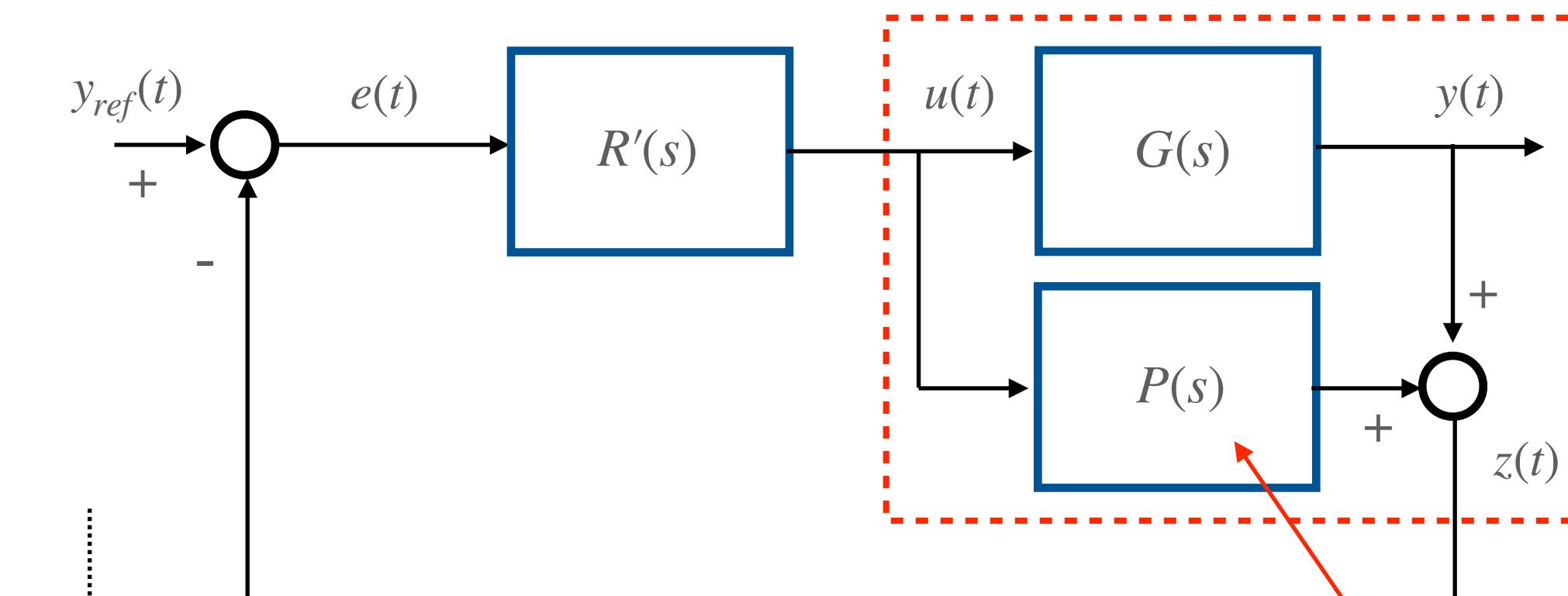
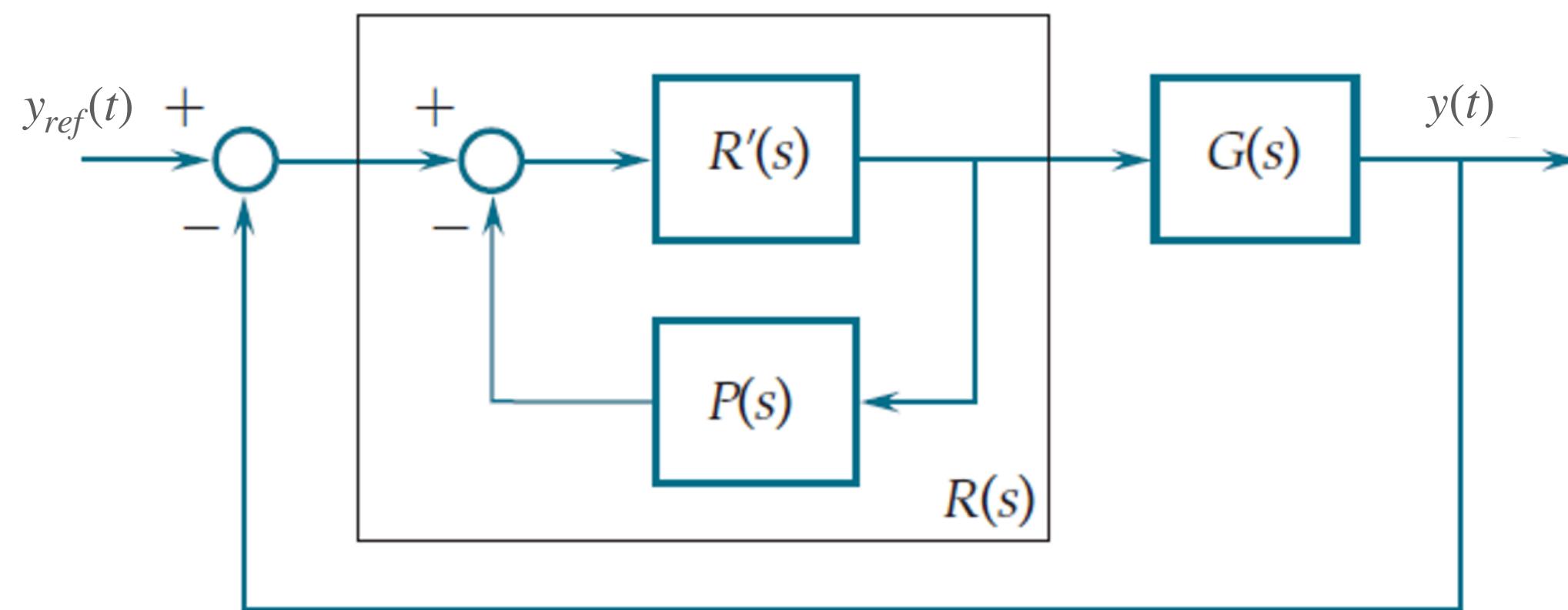
$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

synthesis by cancellation



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 3:



$$\begin{aligned} R(s) &= \frac{R'(s)}{1 + R'(s)P(s)} \\ &= \frac{N_{R'}(s)/D_{R'}(s)}{1 + (N_{R'}(s)/D_{R'}(s))(1 - e^{-\tau s})(N(s)/D(s))} \\ &= \frac{N_{R'}(s)D(s)}{D_{R'}(s)D(s) + N_{R'}(s)N(s)(1 - e^{-\tau s})} \end{aligned}$$

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

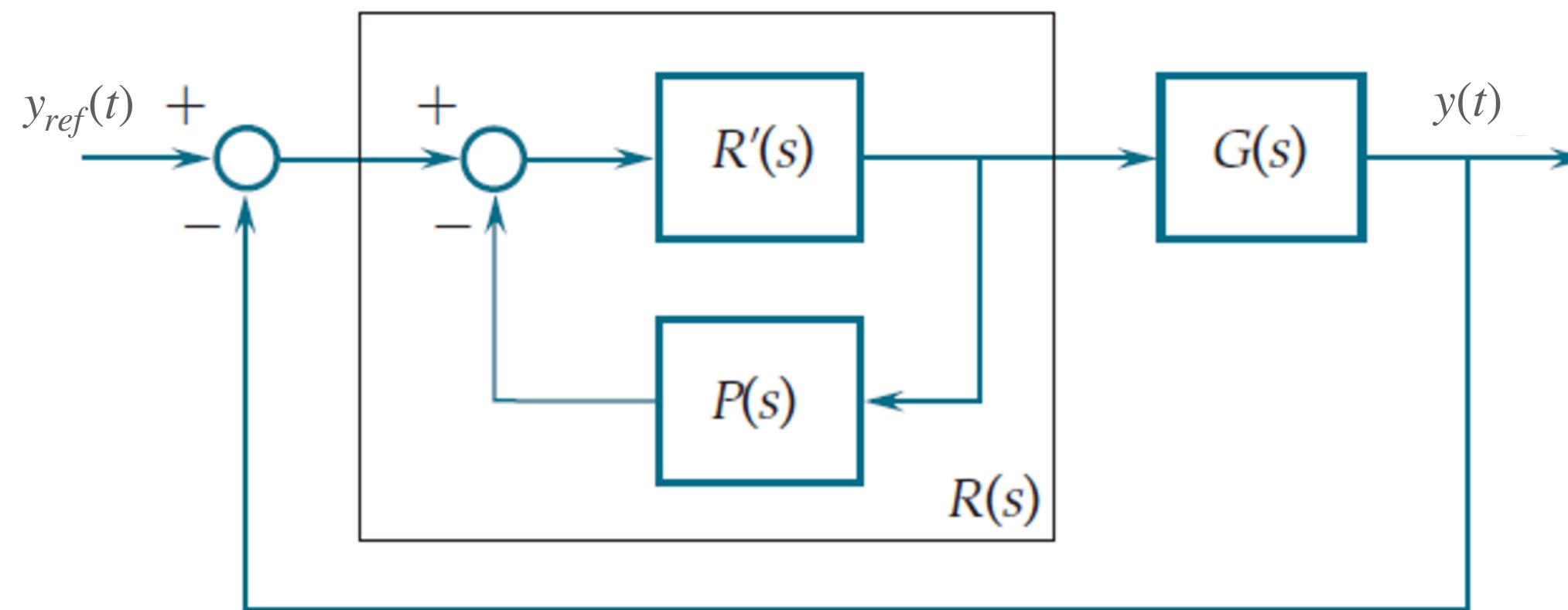
$R(s)$ is a controller with a non-rational transfer function



Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Summary:

Alternative scheme



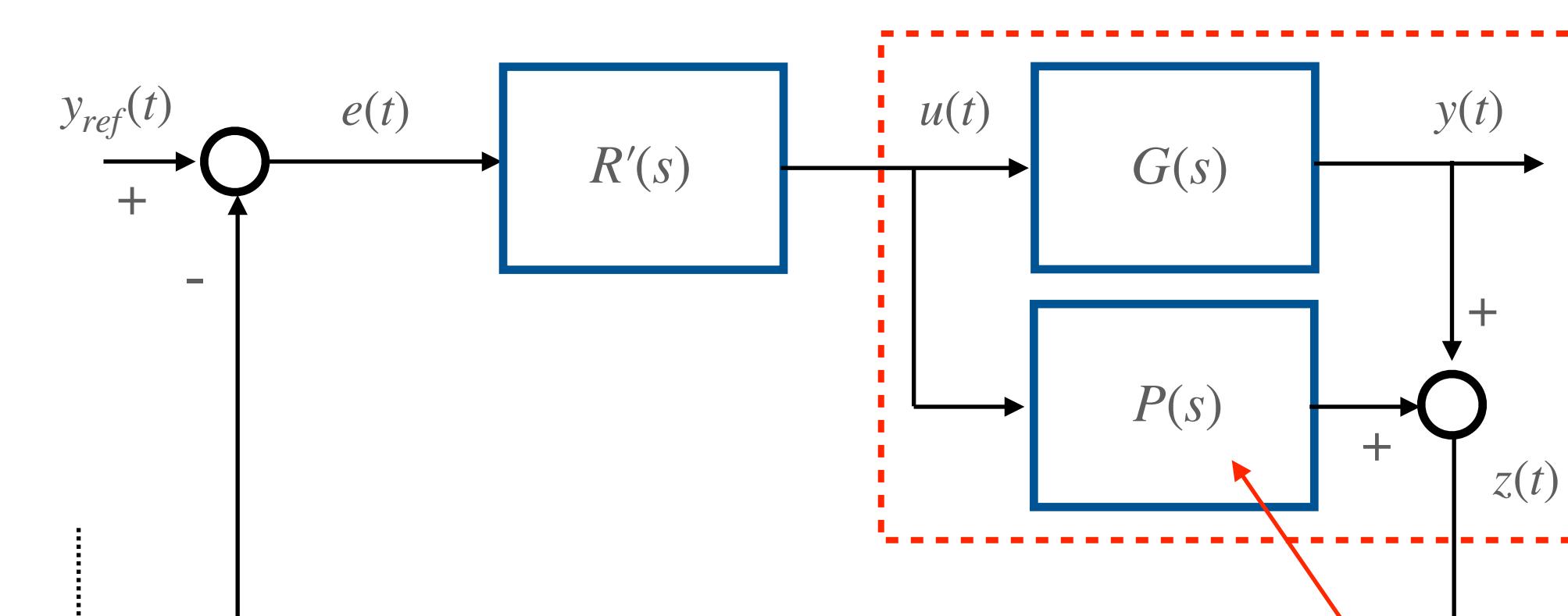
$z(t)$ is a prediction
of $y(t)$

$$Z(s) = e^{\tau s} Y(s)$$

$y(t)$ is $z(t)$ delayed of τ seconds

$$Y(s) = e^{-\tau s} Z(s)$$

Classical scheme



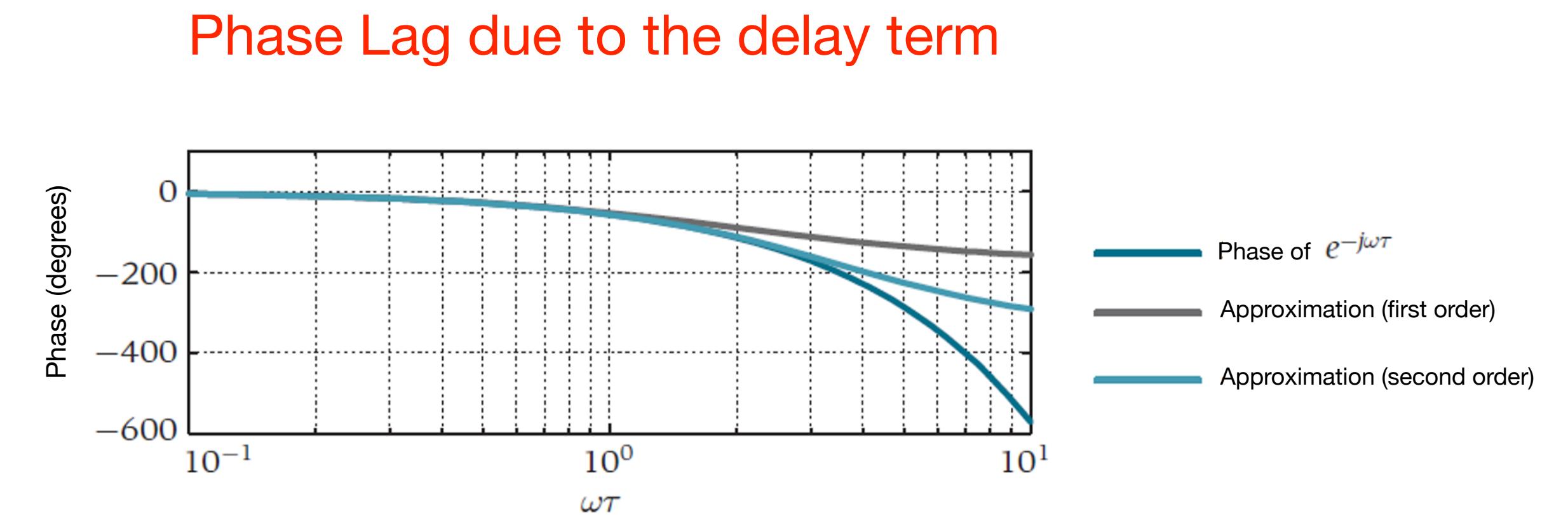
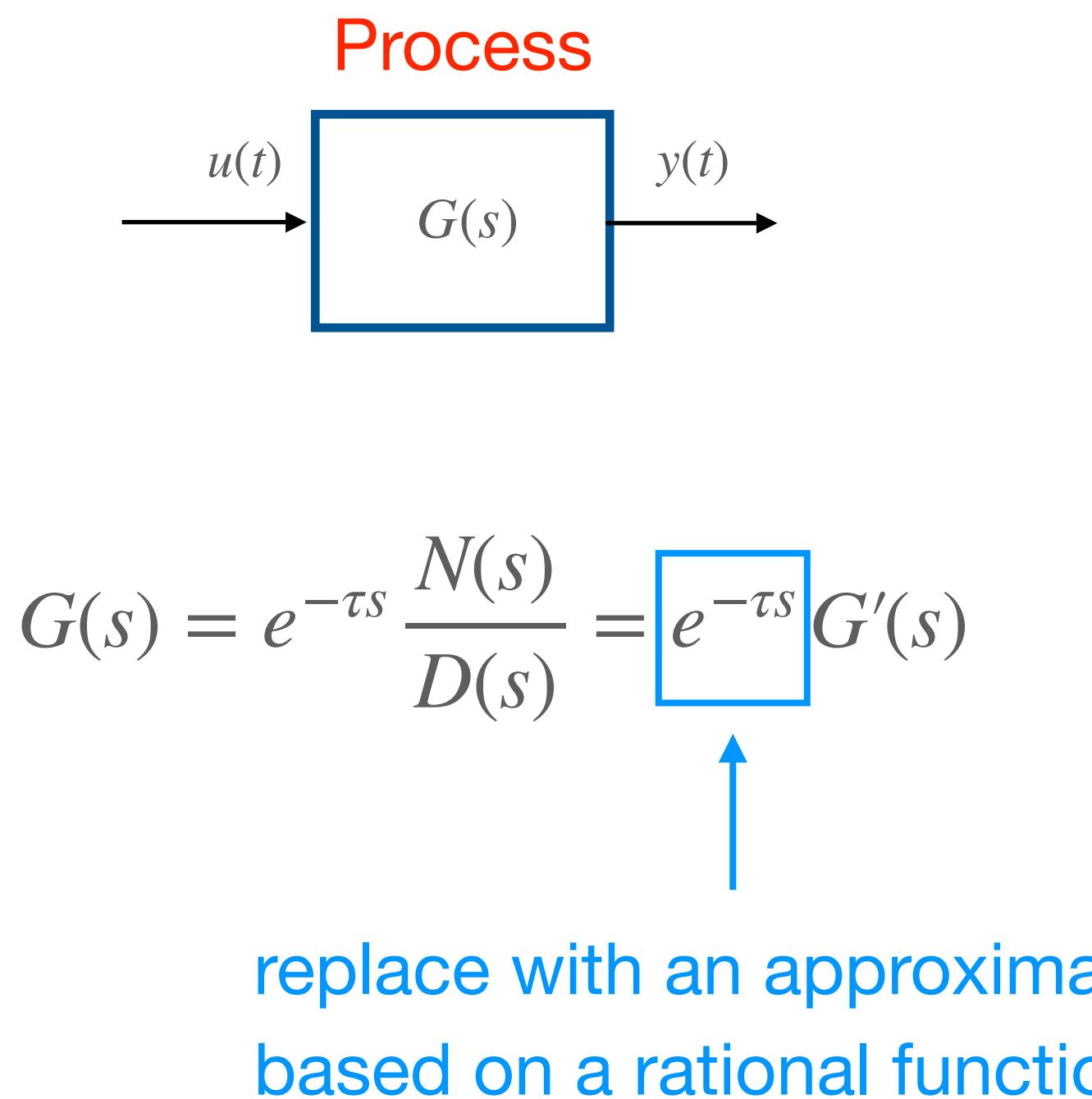
Smith Predictor

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

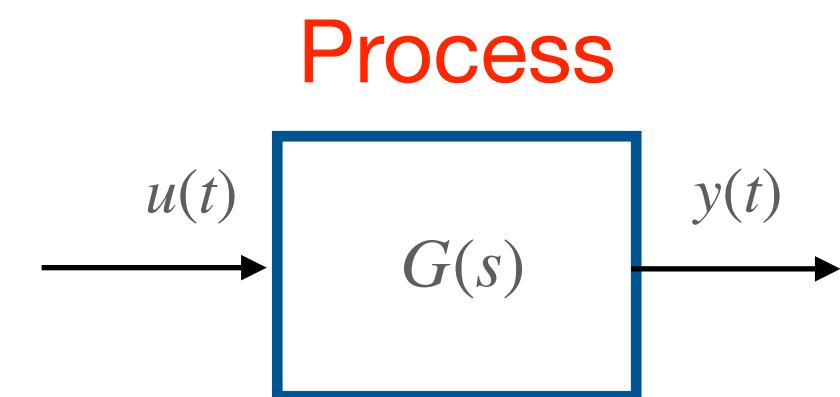
the delay has not disappeared!



Control of LTI Systems with Delays: Padé Approximation



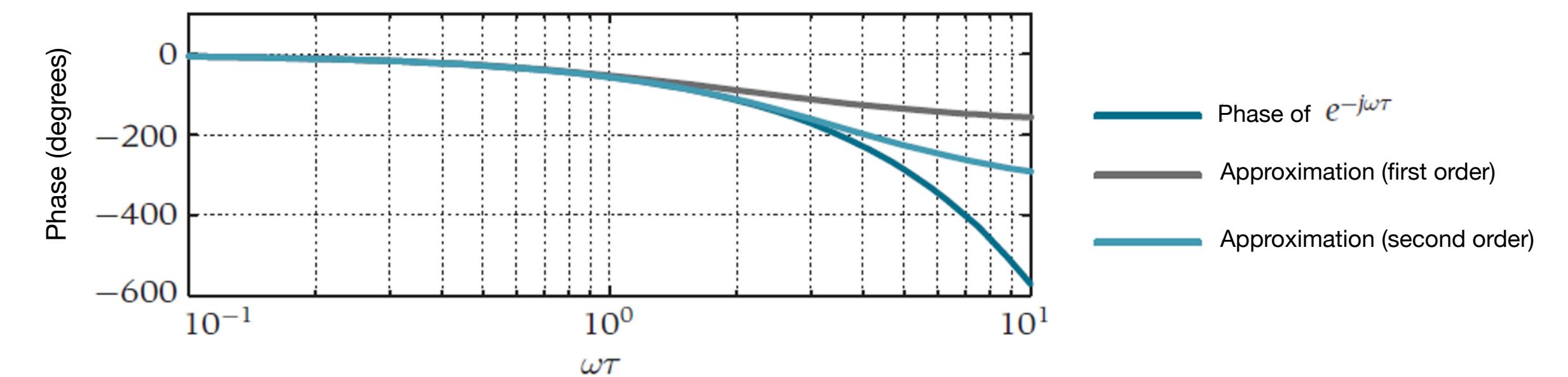
Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Function to approximate: $e^{-\tau s}$

Phase Lag due to the delay term

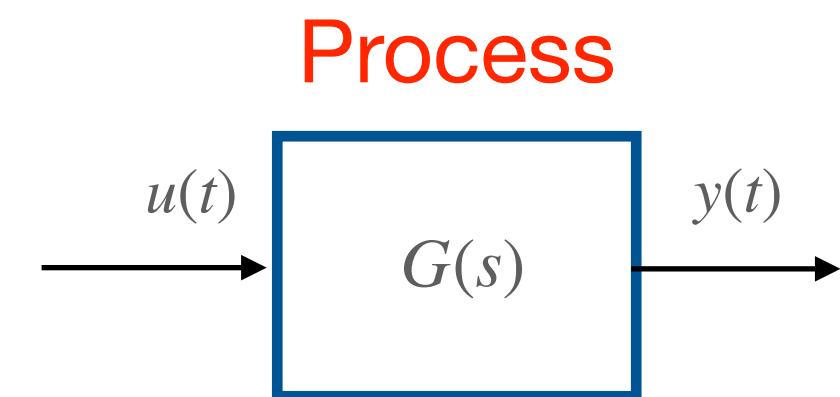


First order approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

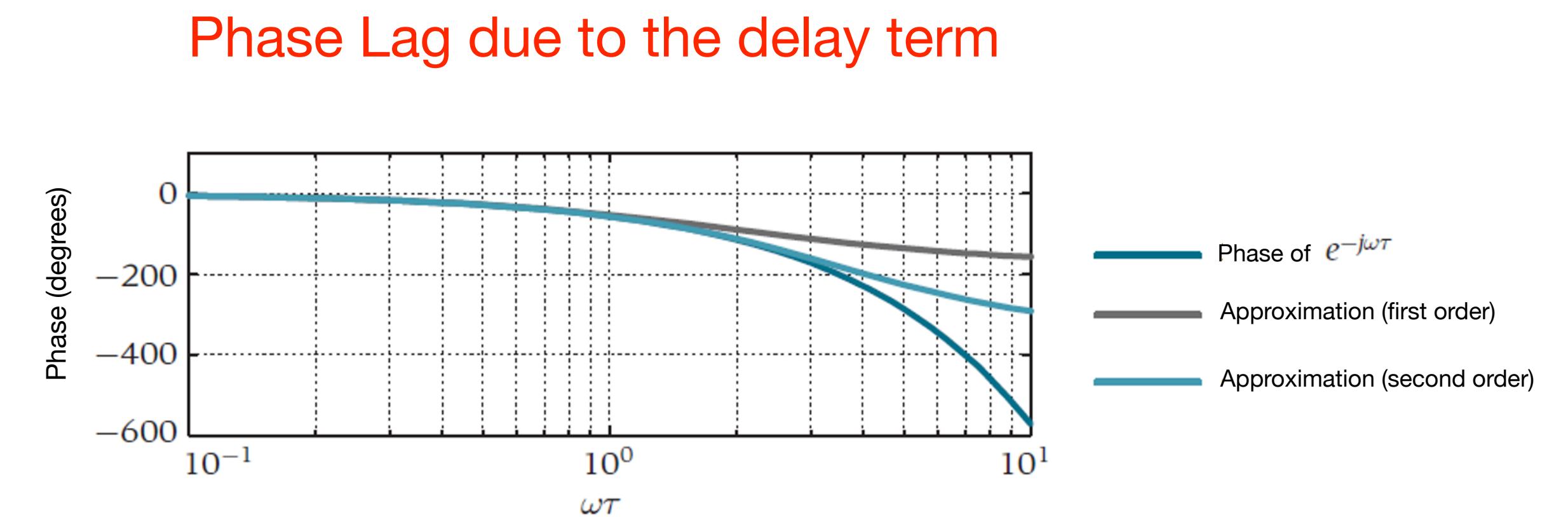


Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Function to approximate: $e^{-\tau s}$



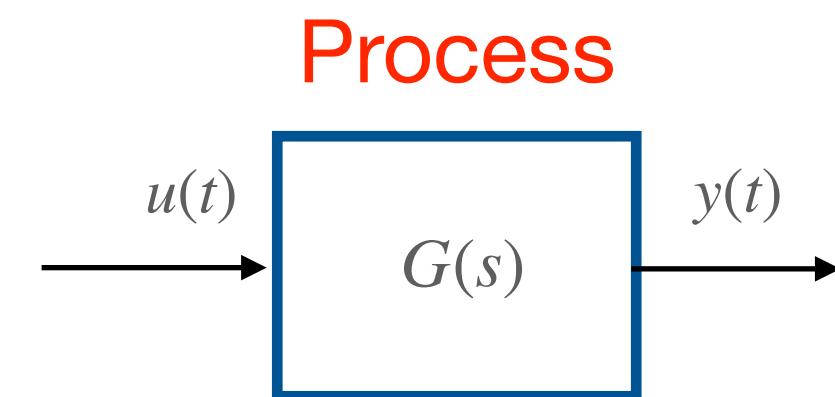
First order approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

How to select the parameters?



Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

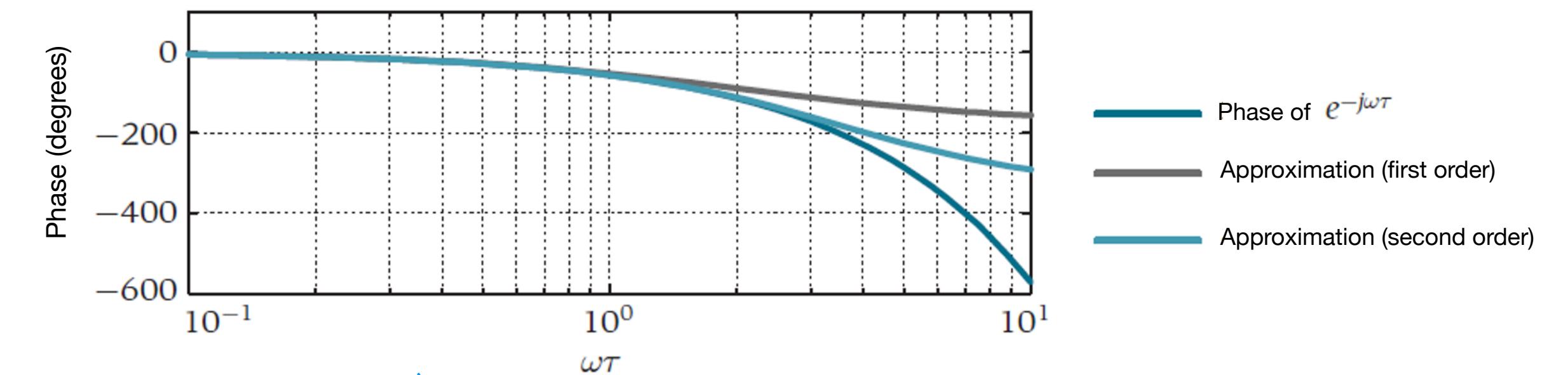
Function to approximate:

$$e^{-\tau s}$$

First order approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

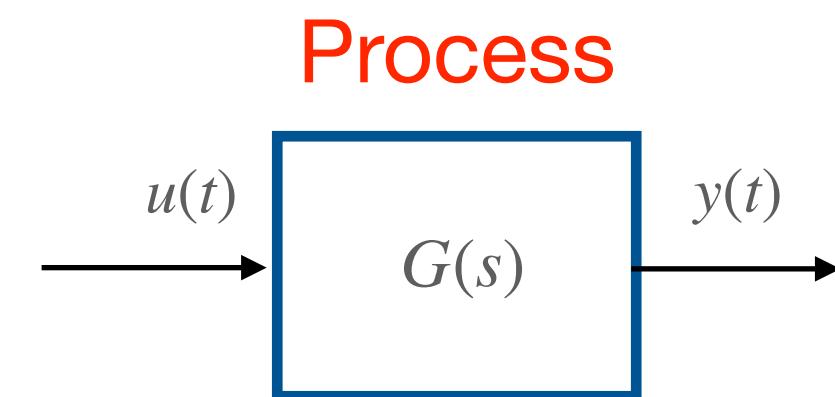
Phase Lag due to the delay term



Use MacLaurin series expansion



Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

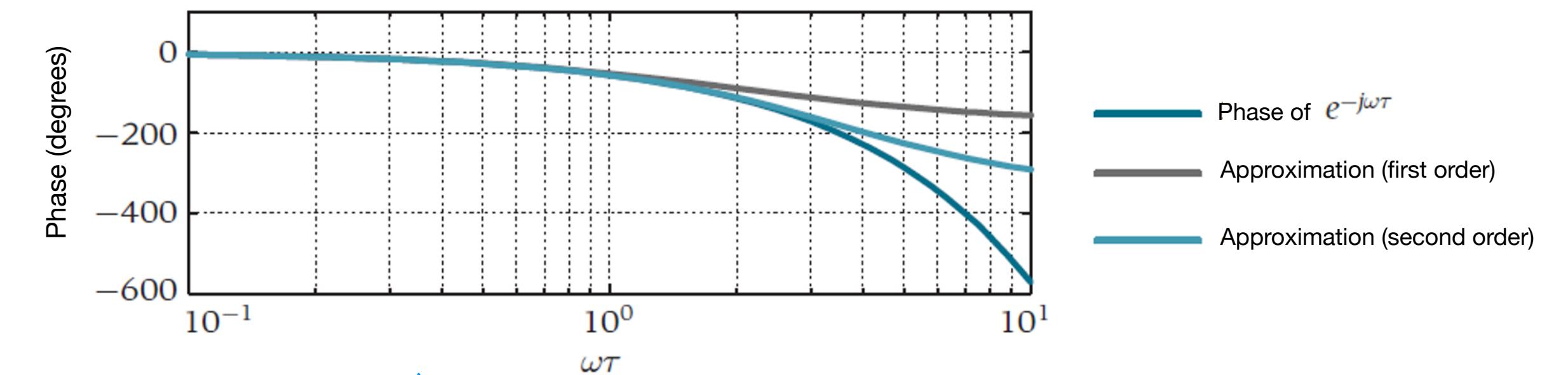
Function to approximate:

$$e^{-\tau s}$$

First order approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

Phase Lag due to the delay term

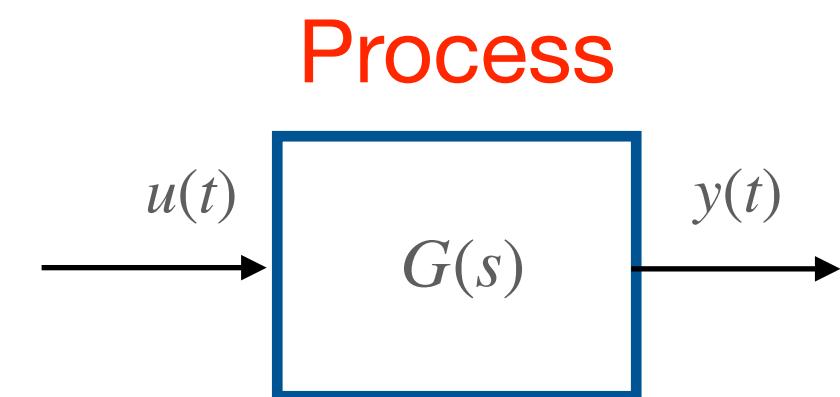


Use MacLaurin series expansion

$$f(s) = f(0) + \frac{df}{ds}(0)s + \frac{1}{2!} \frac{d^2f}{ds^2}(0)s^2 + \frac{1}{3!} \frac{d^3f}{ds^3}(0)s^3 + \mathcal{O}(s^4)$$



Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

Function to approximate:

McLaurin approximation:

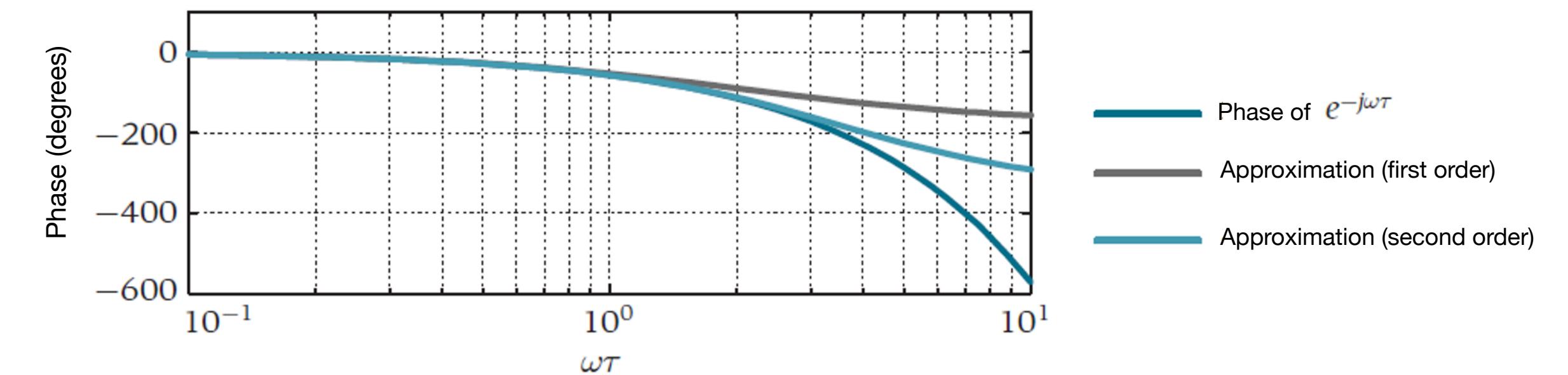
First order approximation:

$$e^{-\tau s}$$

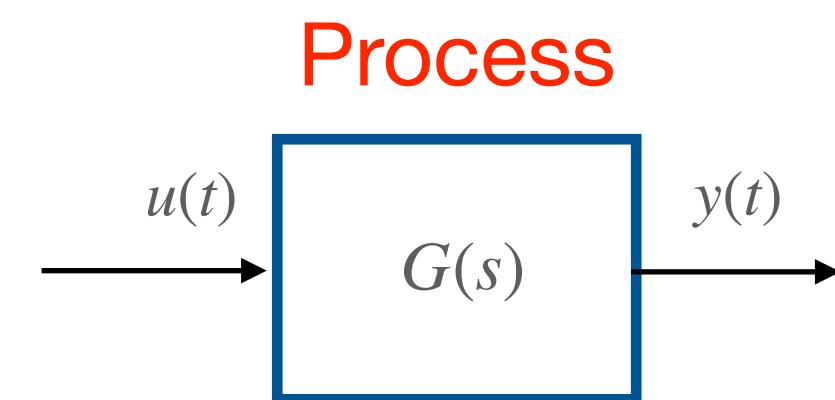
$$e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$$

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

Phase Lag due to the delay term

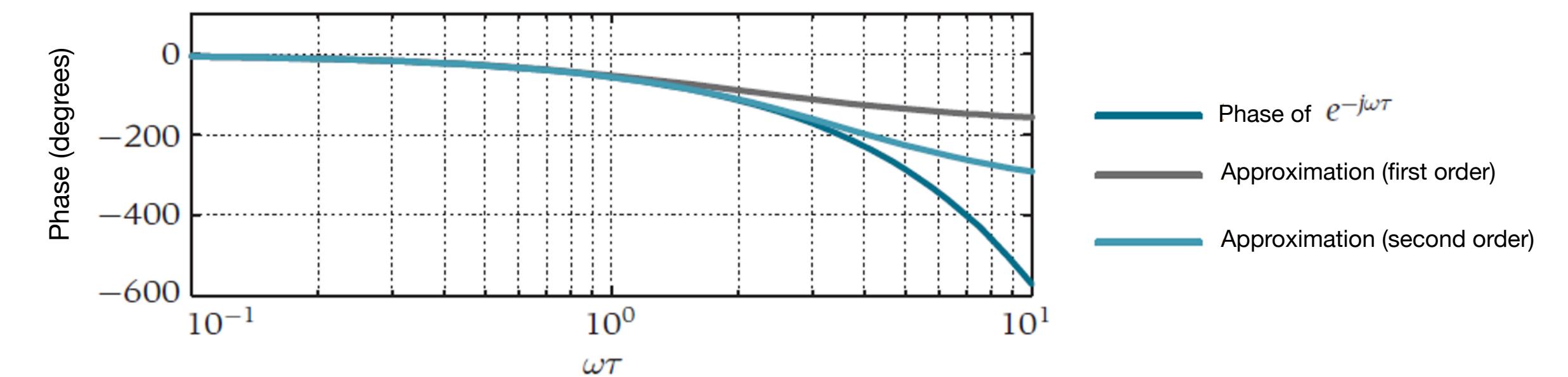


Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

Phase Lag due to the delay term



Function to approximate:

$$e^{-\tau s}$$

McLaurin approximation:

$$e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$$

First order approximation:

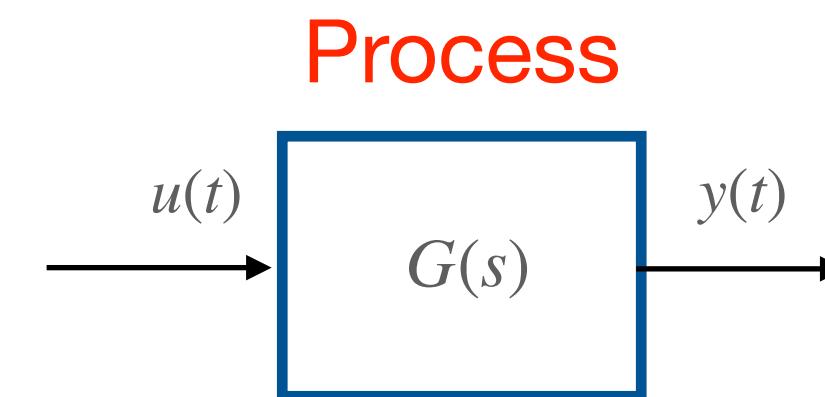
$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

McLaurin approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs} = \approx \mu + \mu(a - b)s - \mu b(a - b)s^2$$



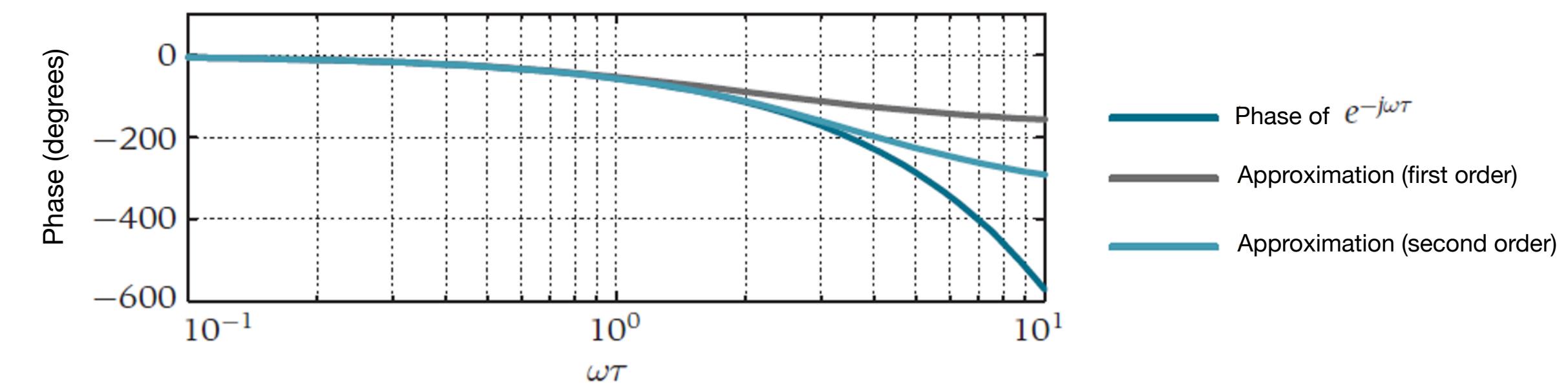
Control of LTI Systems with Delays: Padé Approximation



$$f(s) = f(0) + \frac{df}{ds}(0)s + \frac{1}{2!} \frac{d^2f}{ds^2}(0)s^2 + \frac{1}{3!} \frac{d^3f}{ds^3}(0)s^3 + \mathcal{O}(s^4)$$

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

Phase Lag due to the delay term



$$\mu = 1$$

$$\mu(a-b) = -\tau$$

$$-\mu b(a-b) = \frac{\tau^2}{2}$$

Function to approximate:

McLaurin approximation:

First order approximation:

McLaurin approximation:

$$e^{-\tau s}$$

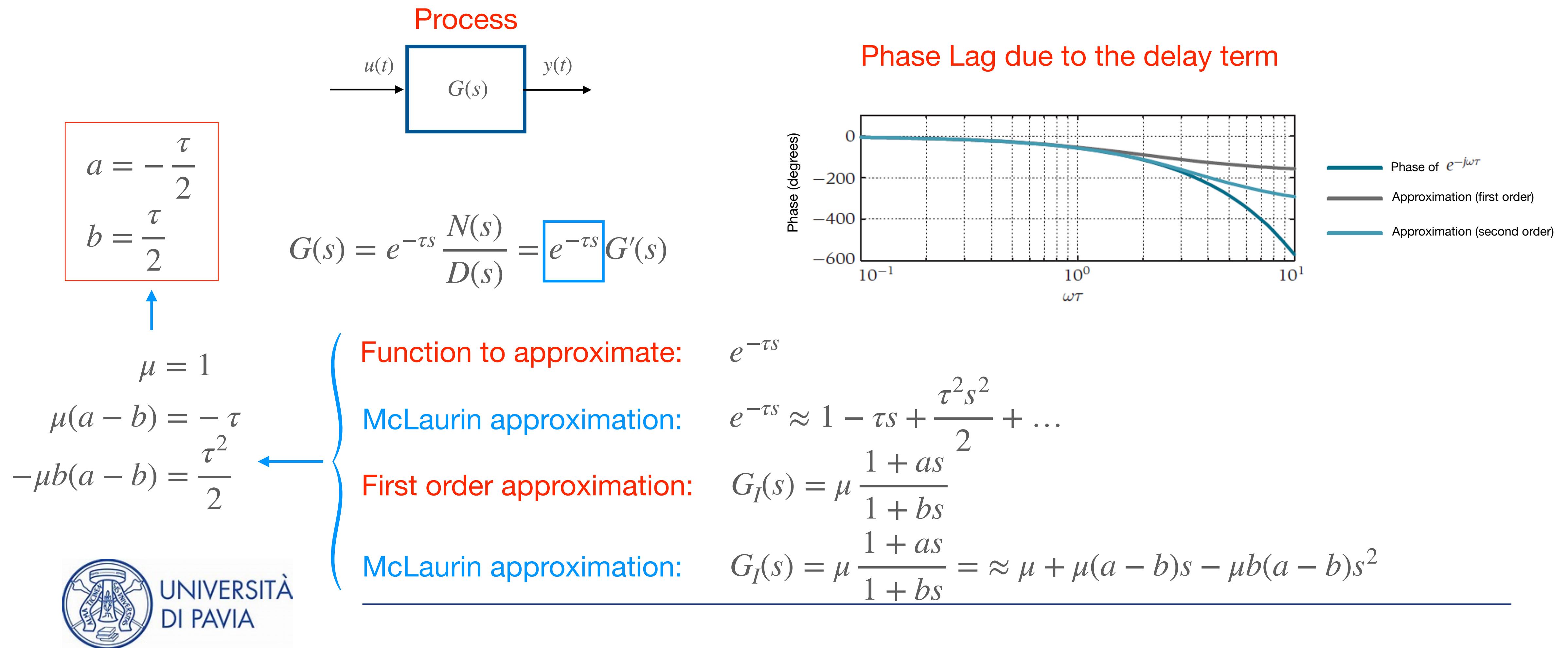
$$e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$$

$$G_I(s) = \mu \frac{1+as}{1+bs}$$

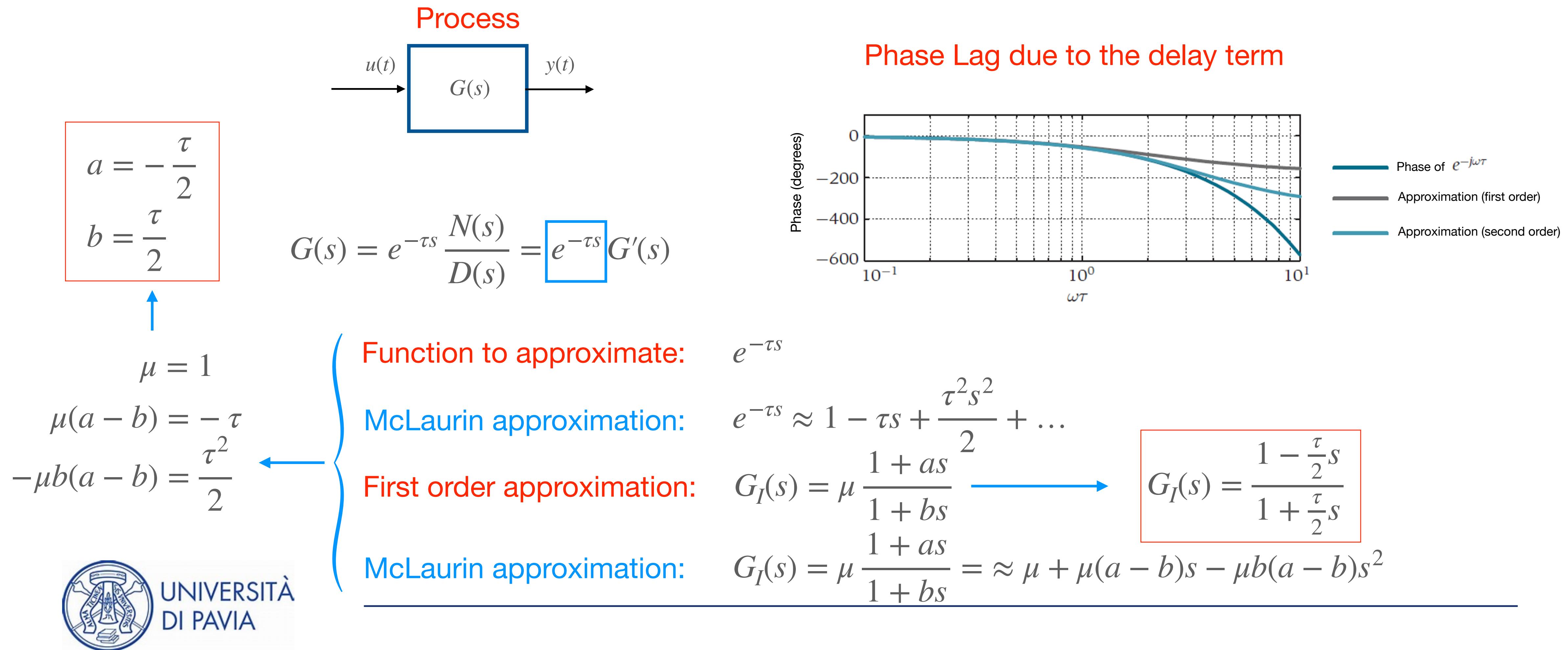
$$G_I(s) = \mu \frac{1+as}{1+bs} \approx \mu + \mu(a-b)s - \mu b(a-b)s^2$$



Control of LTI Systems with Delays: Padé Approximation



Control of LTI Systems with Delays: Padé Approximation



Control of LTI Systems with Delays: Padé Approximation

Process

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Phase Lag due to the delay term

Function to approximate: $e^{-\tau s}$

McLaurin approximation: $e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$

First order approximation: $G_I(s) = \mu \frac{1 + as}{1 + bs}$

McLaurin approximation: $G_I(s) = \mu \frac{1 + as}{1 + bs} = \approx \mu + \mu(a - b)s - \mu b(a - b)s^2$

Magnitude and phase ?

$a = -\frac{\tau}{2}$

$b = \frac{\tau}{2}$

$\mu = 1$

$\mu(a - b) = -\tau$

$-\mu b(a - b) = \frac{\tau^2}{2}$

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Control of LTI Systems with Delays: Padé Approximation

Process

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Phase Lag due to the delay term

Function to approximate: $e^{-\tau s}$

McLaurin approximation: $e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$

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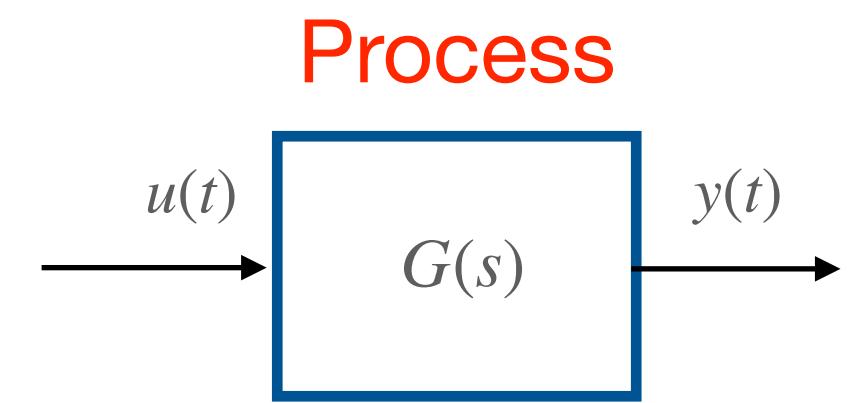
$\mu(a - b) = -\tau$

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Control of LTI Systems with Delays

Summary:

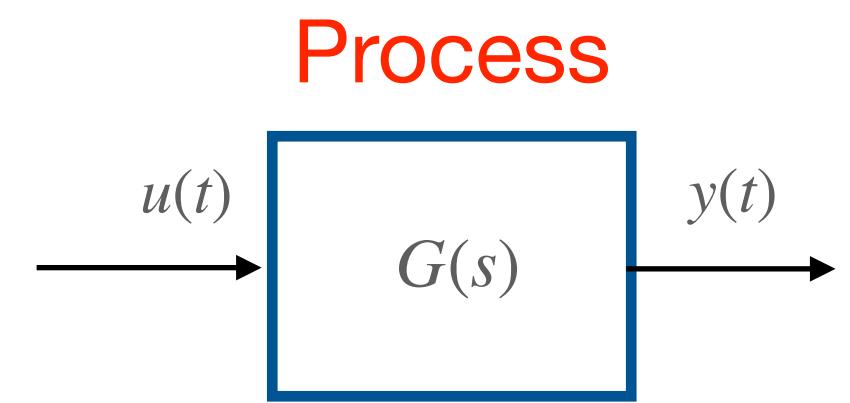


$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$



Control of LTI Systems with Delays

Summary:



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

$e^{-\tau s}$

Modelling alternatives:

$$G_I(s) = \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s}$$

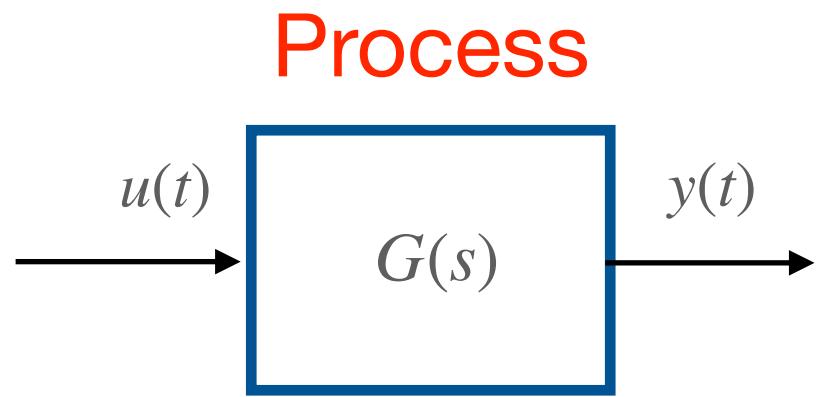
$$G_{II}(s) = \dots$$

$$G_{III}(s) = \dots$$



Control of LTI Systems with Delays

Summary:



Modelling alternatives:

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

$e^{-\tau s}$ $G_I(s) = \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s}$
 $G_{II}(s) = \dots$
 $G_{III}(s) = \dots$

Design alternatives for $R(s)$:

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

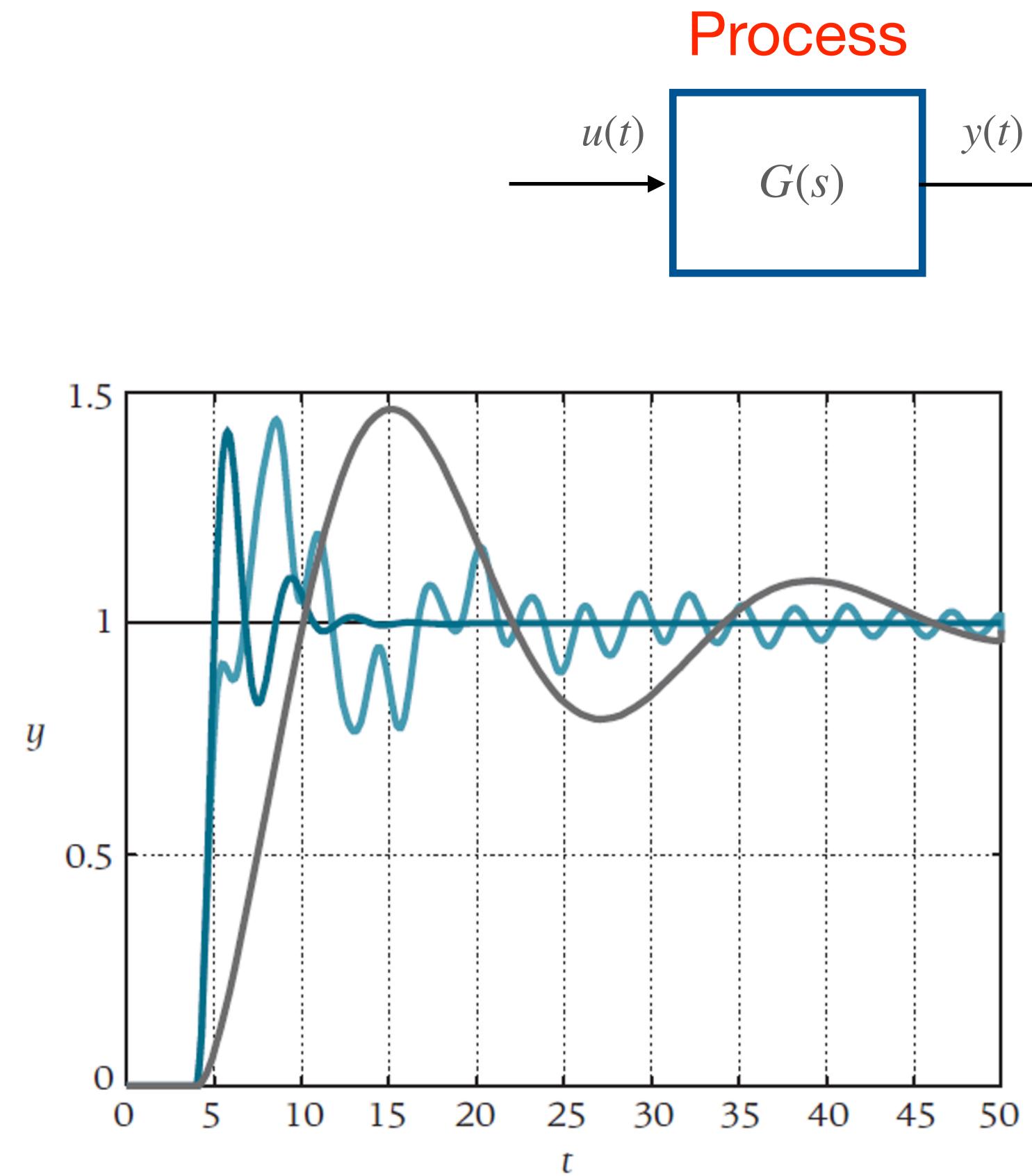
$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} \approx \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s} G'(s)$$

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Control of LTI Systems with Delays

Summary:



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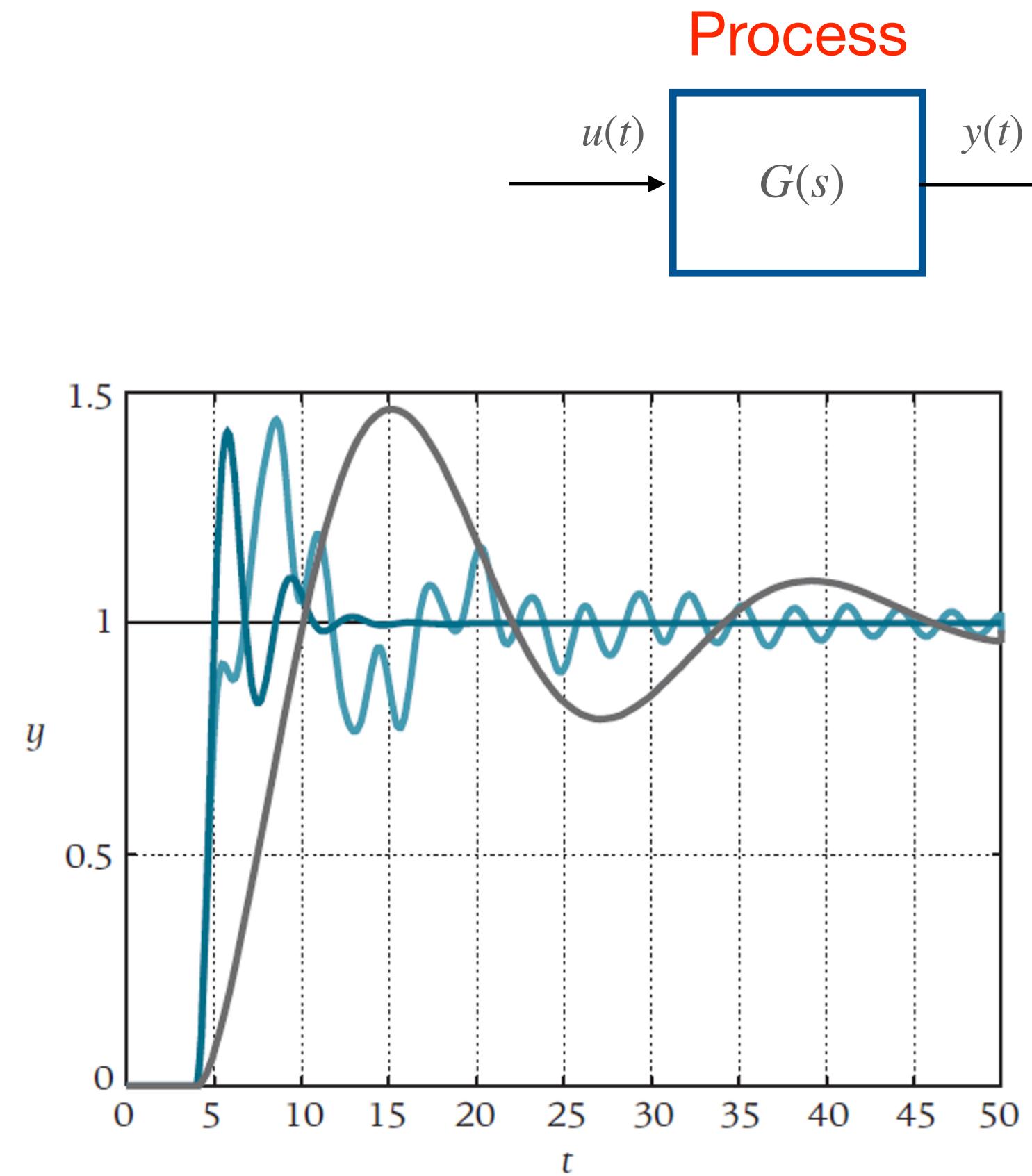
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Control of LTI Systems with Delays

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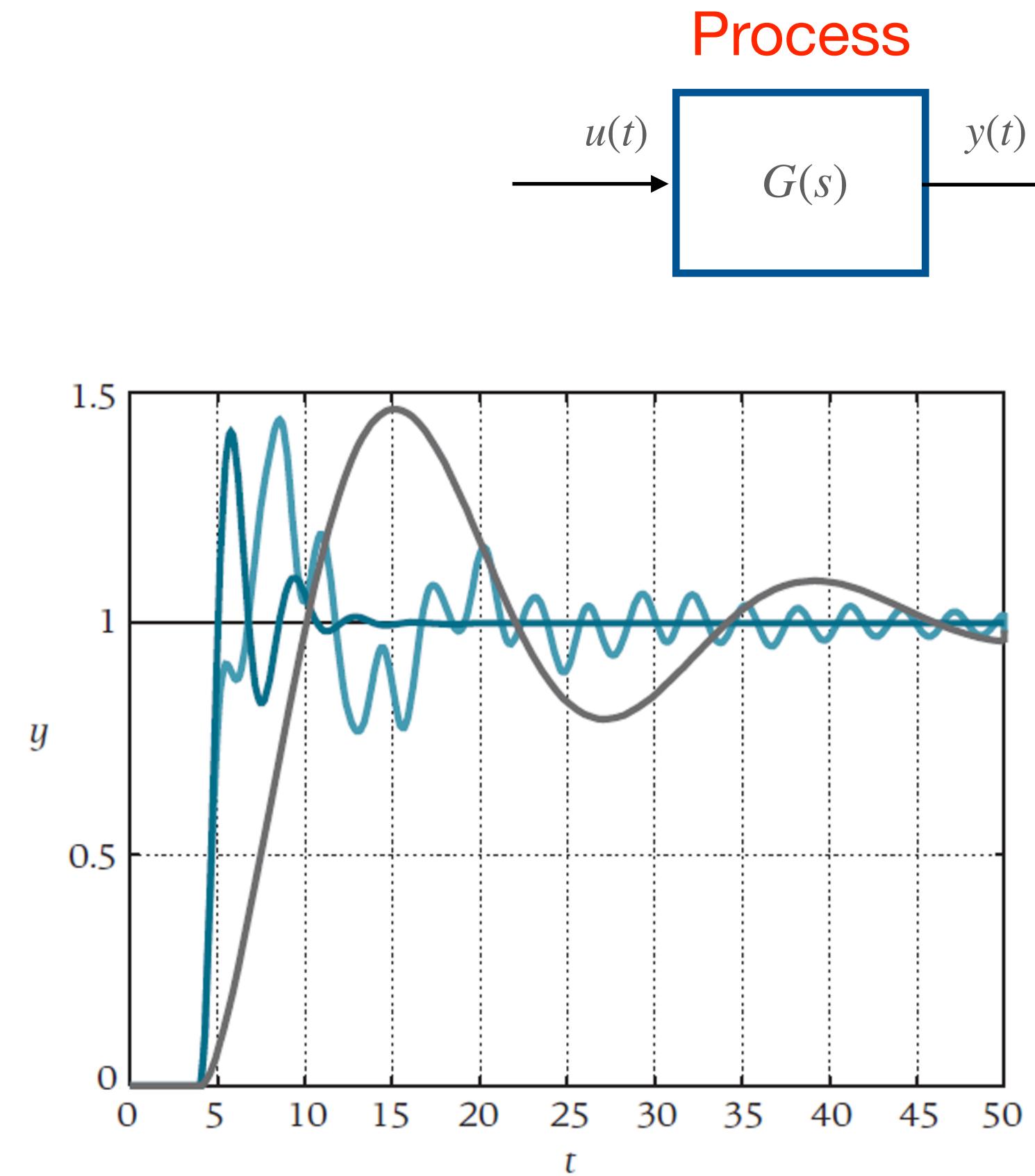
$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} \approx \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s} G'(s)$$

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Control of LTI Systems with Delays

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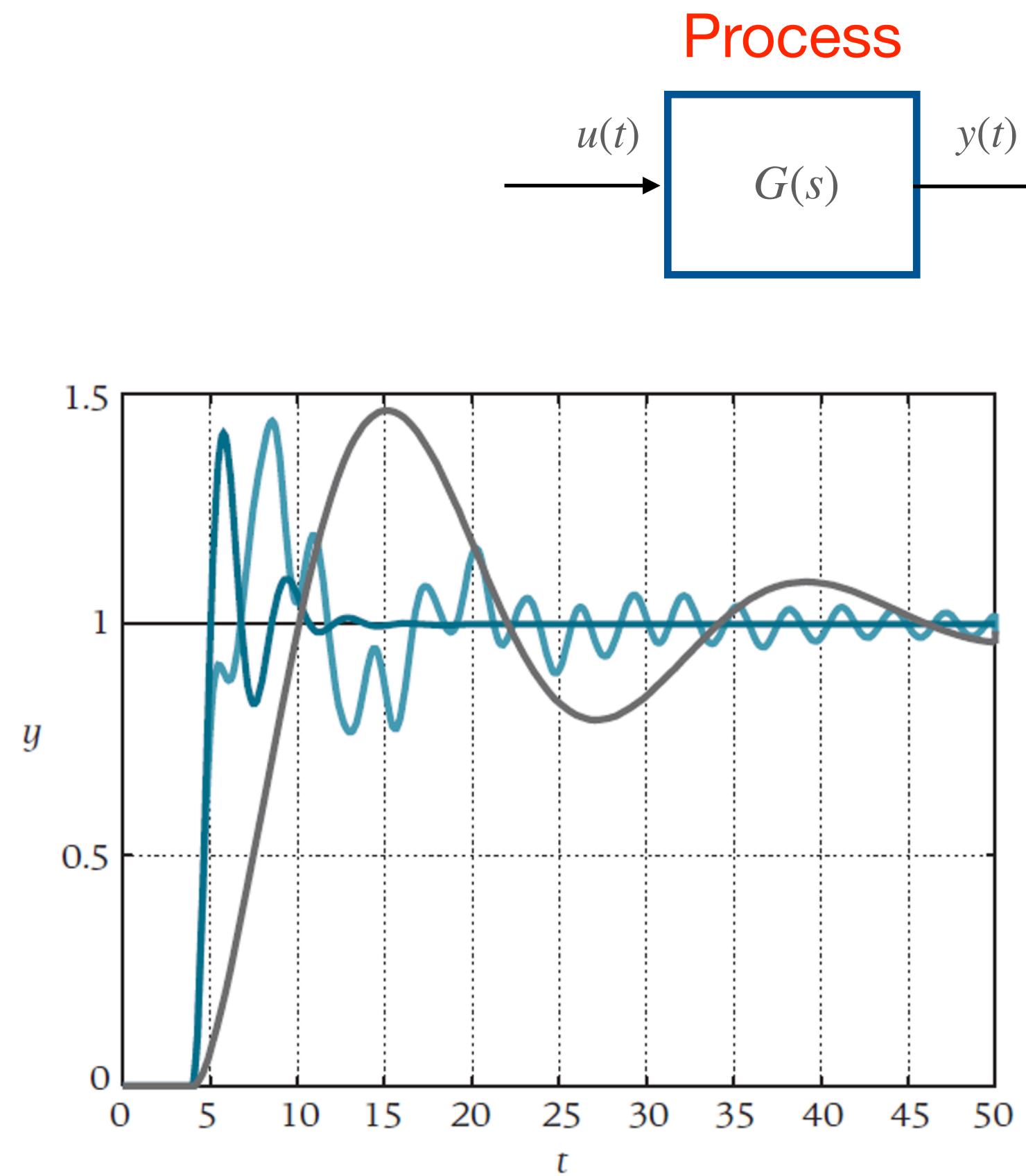
$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} \approx \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s} G'(s)$$

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Control of LTI Systems with Delays

Summary:



Exercise: Use the
UFCS and design
 $R(s)$ relying on

Design alternatives for $R(s)$:

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

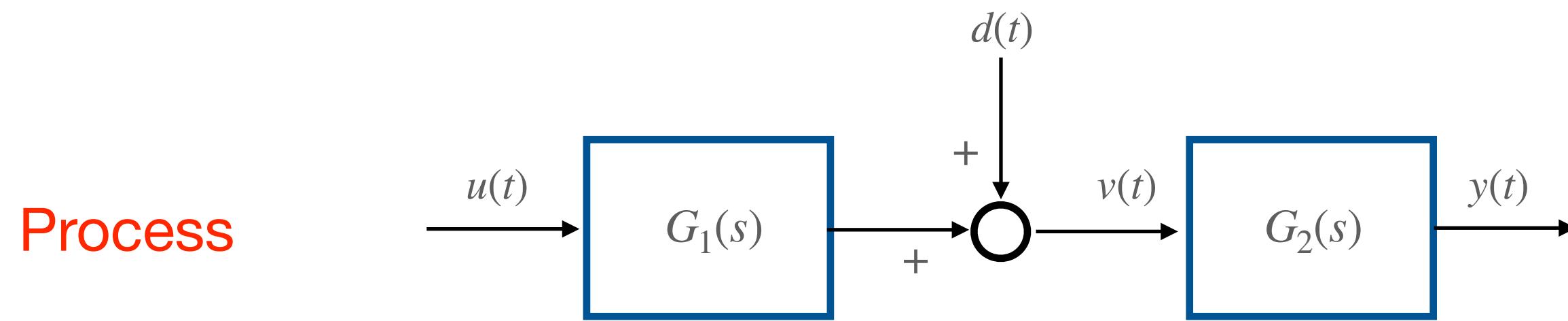
$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

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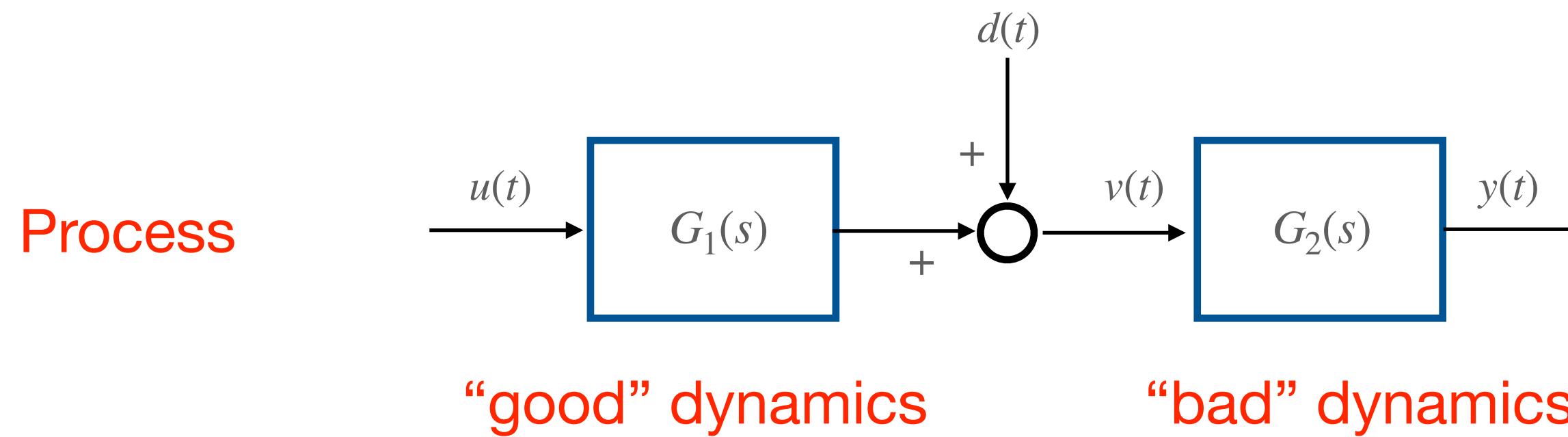
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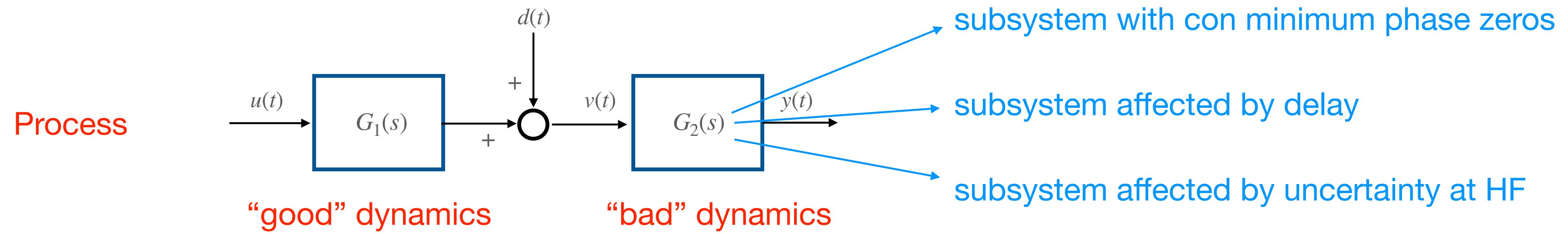
Control Scheme with Decoupling in the Frequency Domain



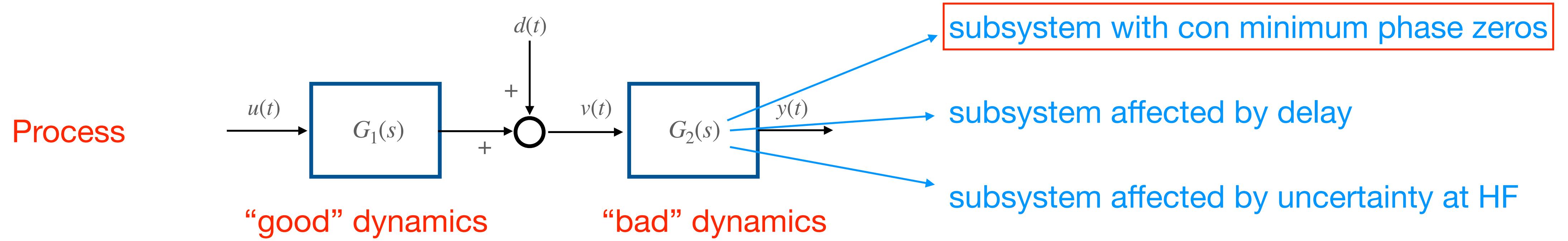
Control Scheme with Decoupling in the Frequency Domain



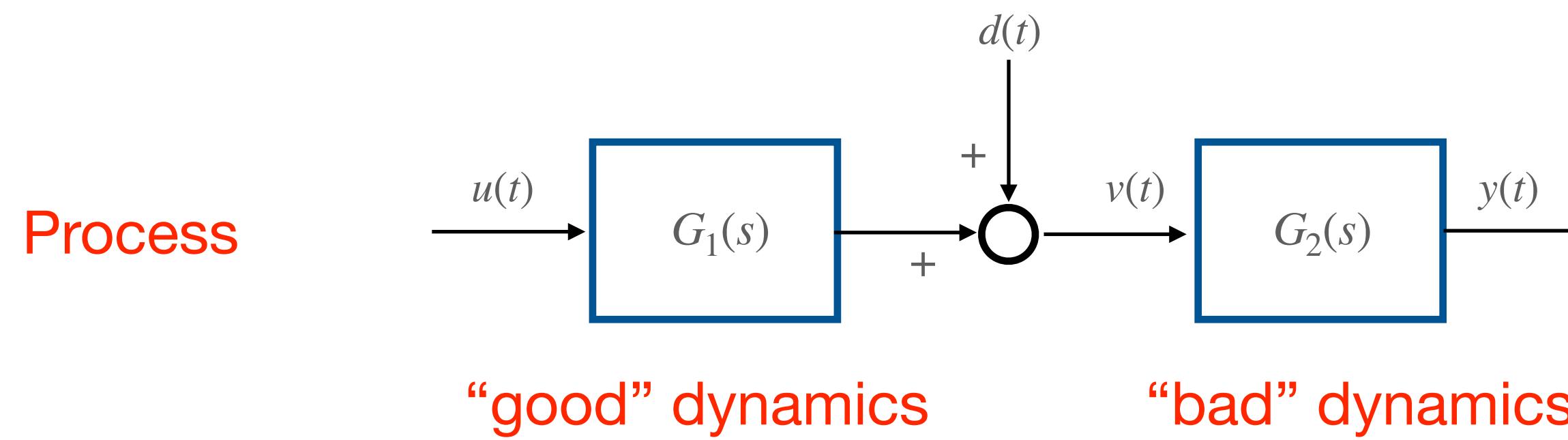
Control Scheme with Decoupling in the Frequency Domain



Control Scheme with Decoupling in the Frequency Domain



Control Scheme with Decoupling in the Frequency Domain

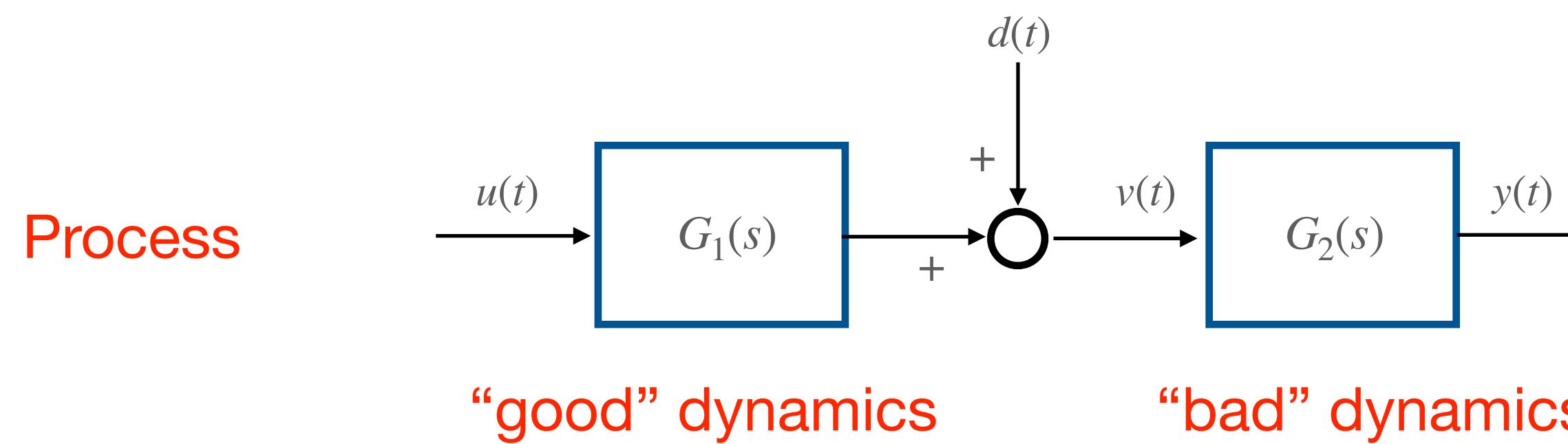


Assumptions:

- The performances that can be obtained by designing a control system for $G_1(s)$ alone are better than those that can be obtained for the entire system $G_1(s)G_2(s)$
- v is accessible (measurable)



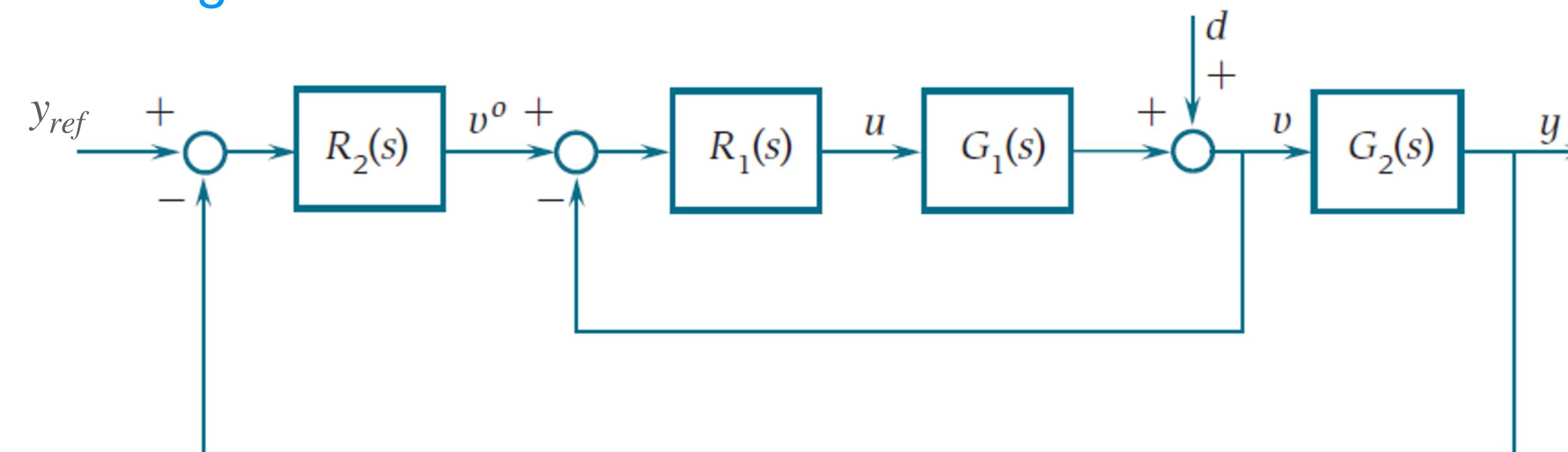
Control Scheme with Decoupling in the Frequency Domain



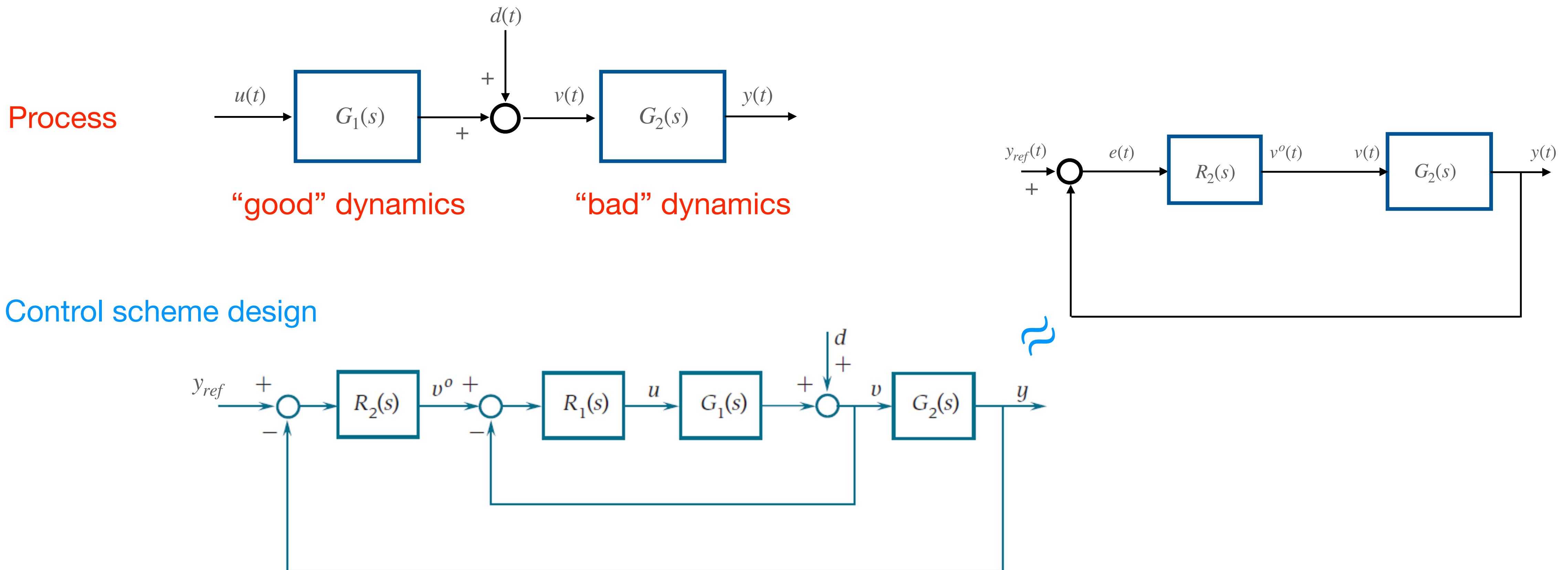
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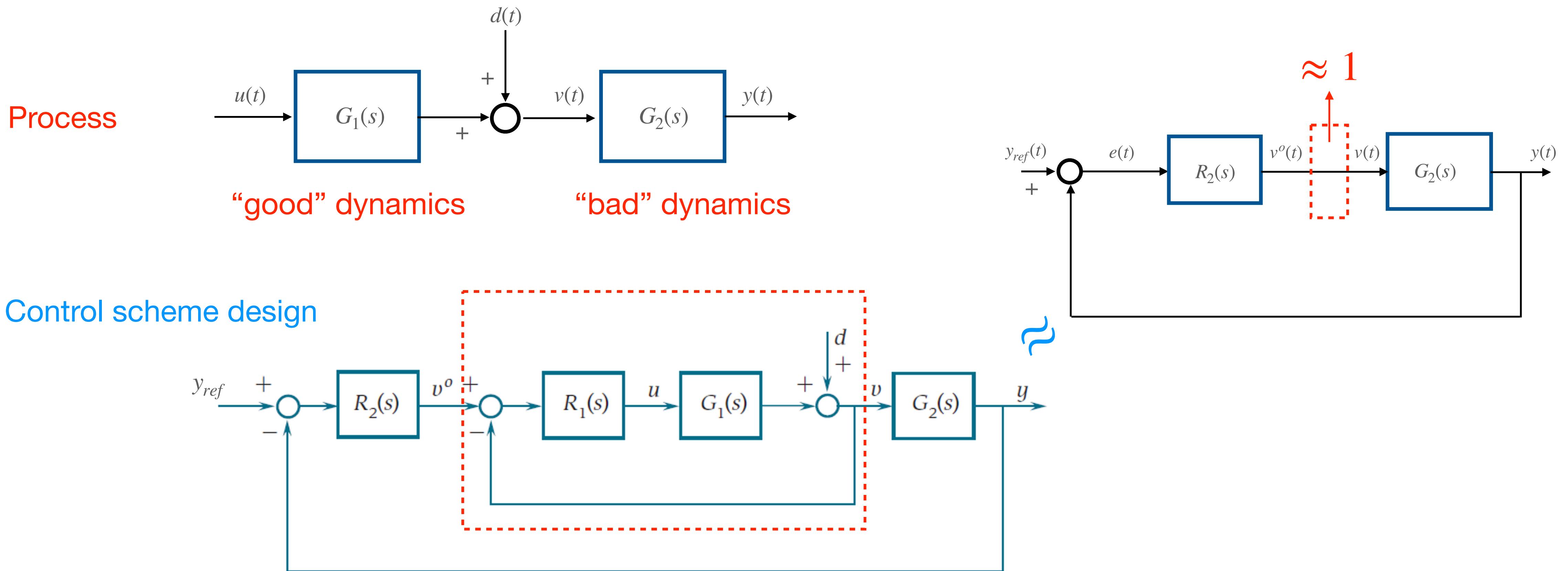
Control scheme design



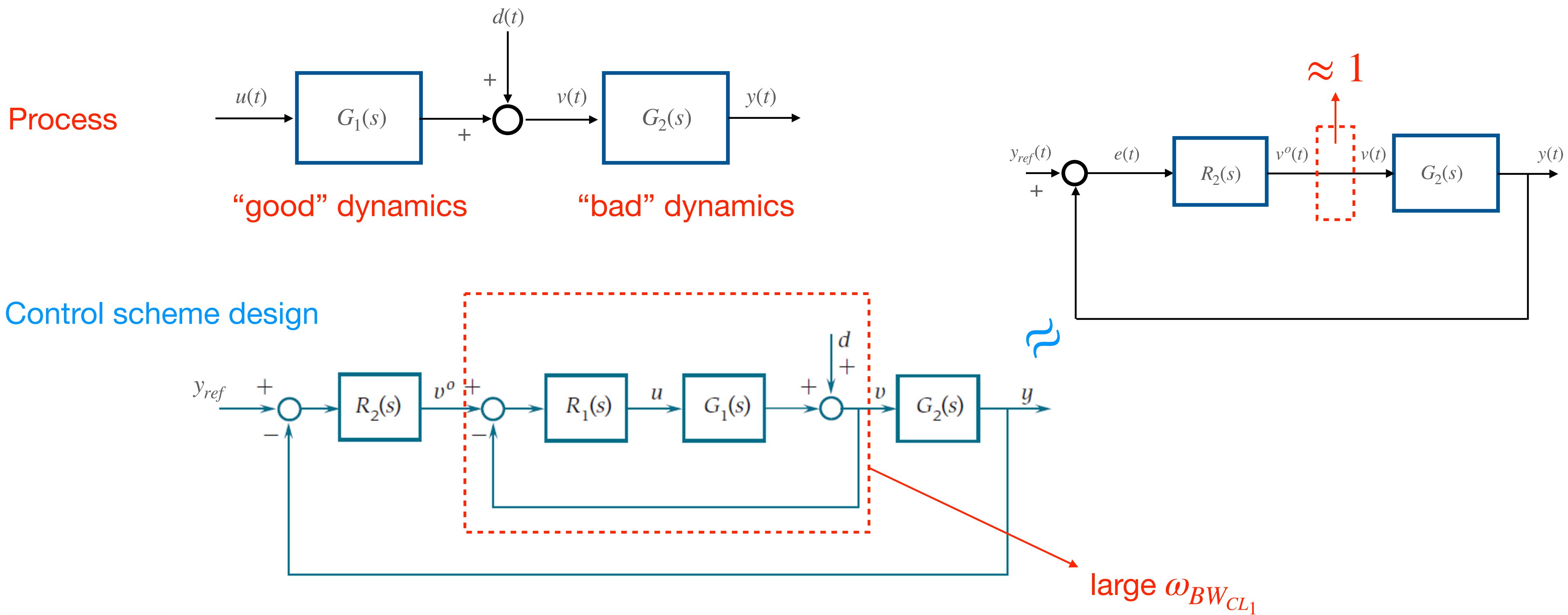
Control Scheme with Decoupling in the Frequency Domain



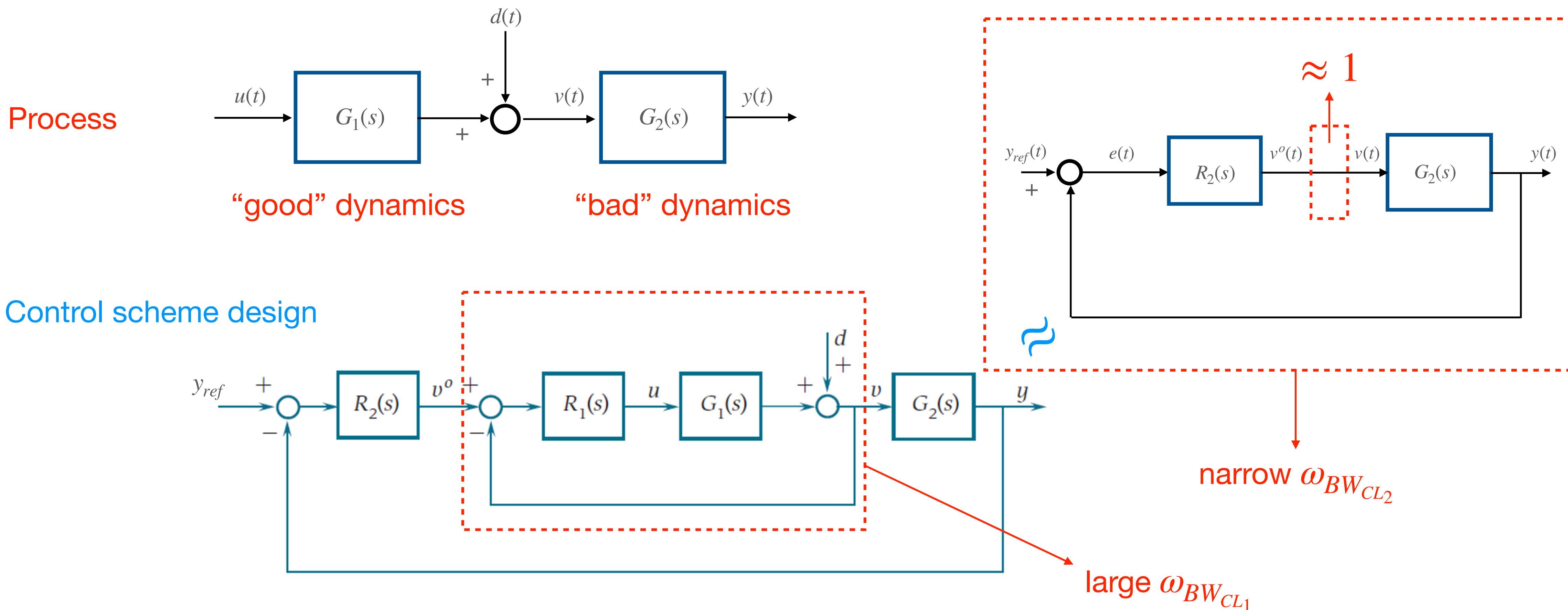
Control Scheme with Decoupling in the Frequency Domain



Control Scheme with Decoupling in the Frequency Domain

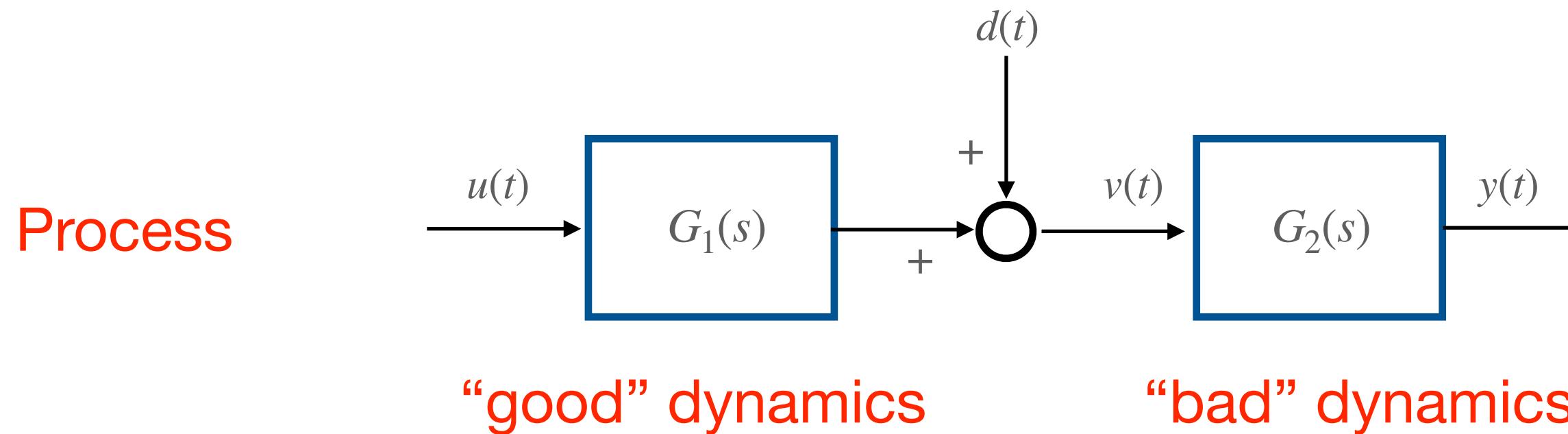


Control Scheme with Decoupling in the Frequency Domain



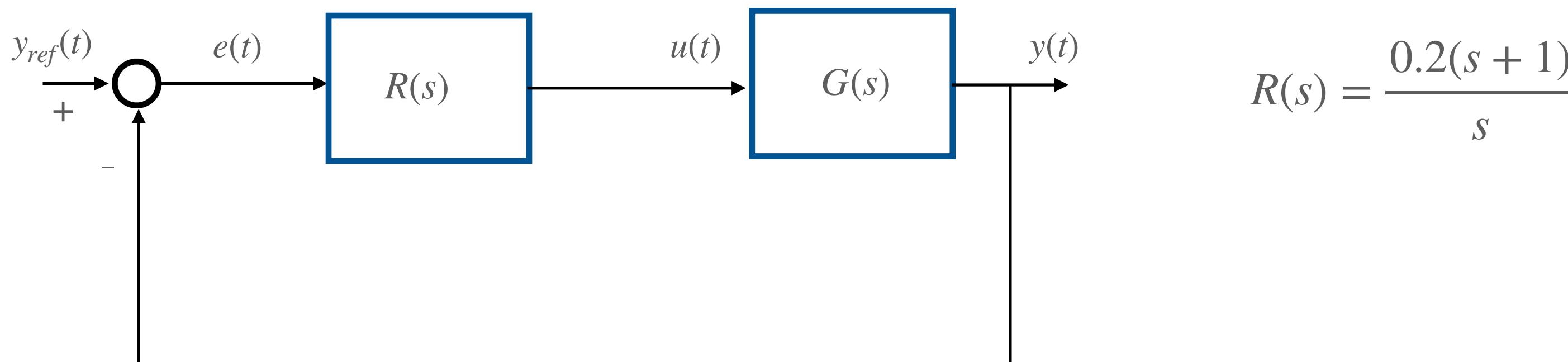
Control Scheme with Decoupling in the Frequency Domain

Example (from the textbook):



$$G_1(s) = \frac{1}{1 + 0.005s}$$
$$G_2(s) = \frac{e^{-4s}}{(1 + s)^2}$$

Control scheme design: Solution 1

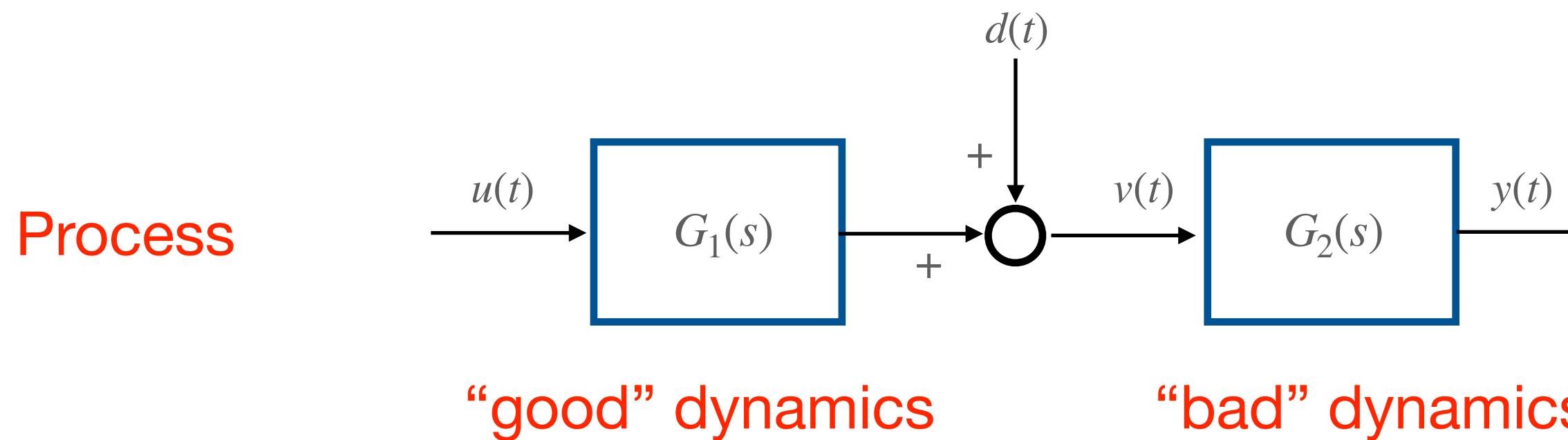


$$R(s) = \frac{0.2(s + 1)}{s}$$



Control Scheme with Decoupling in the Frequency Domain

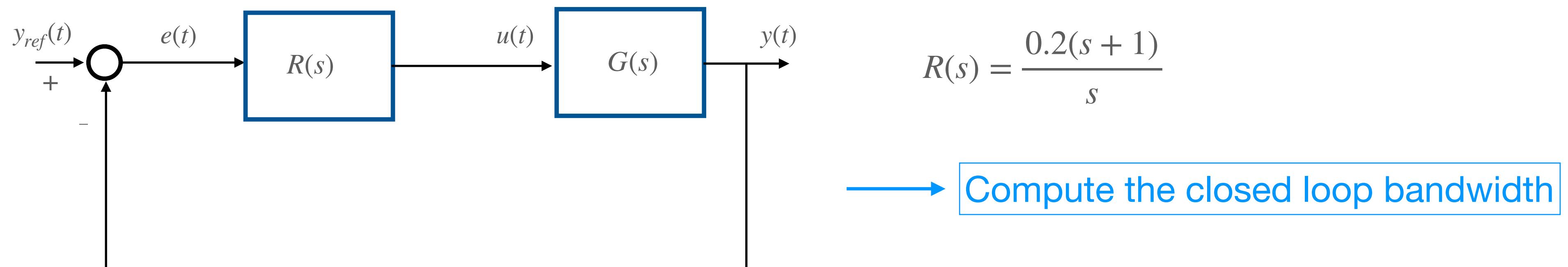
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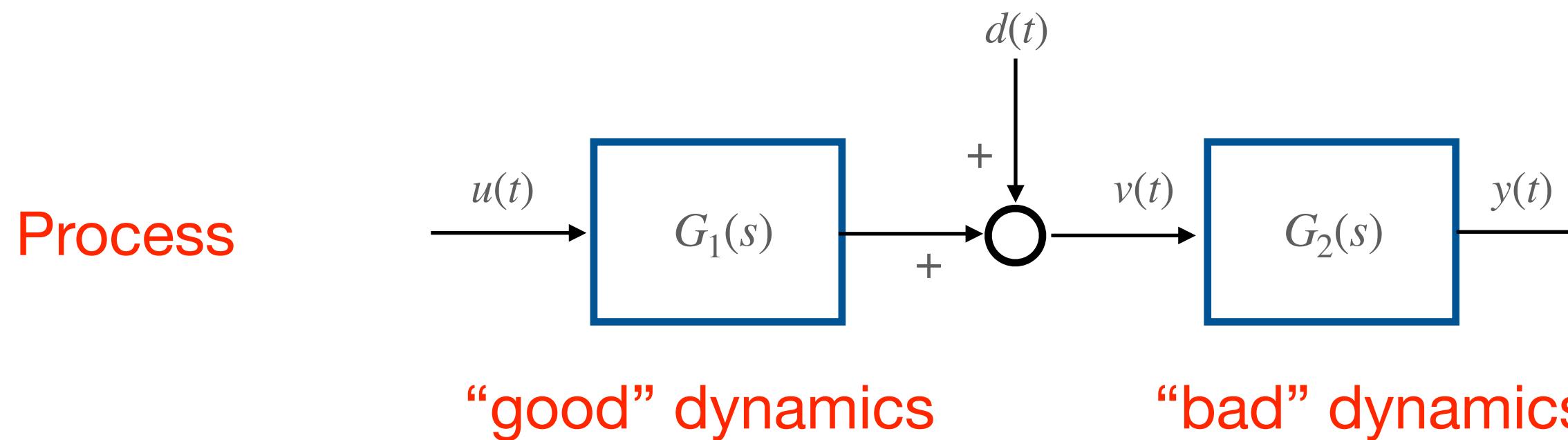
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Control Scheme with Decoupling in the Frequency Domain

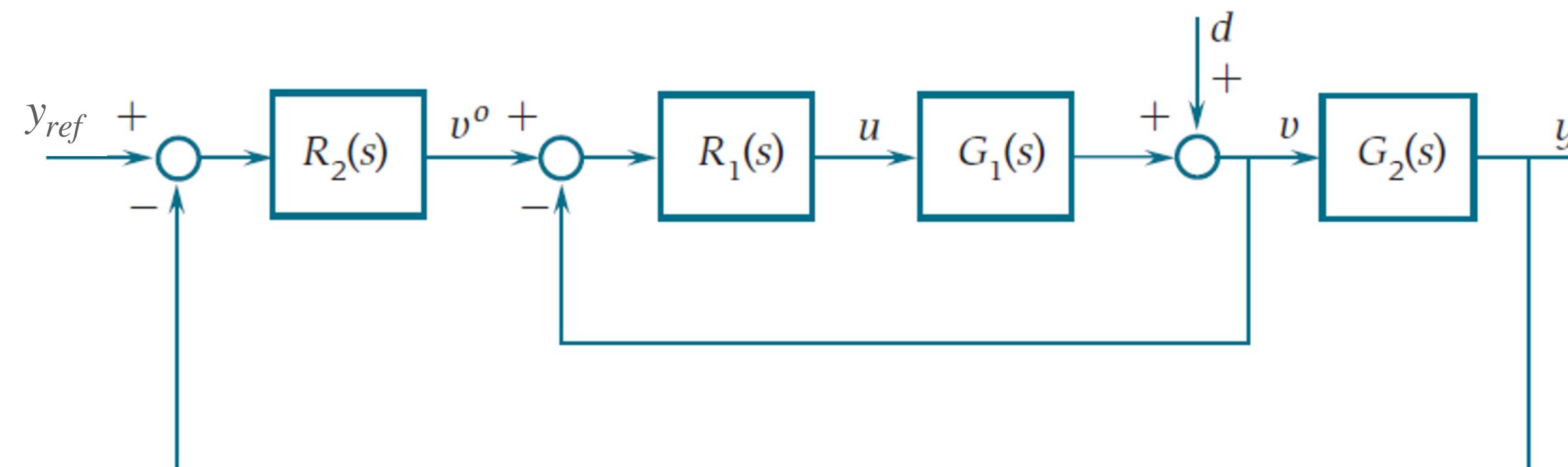
Example (from the textbook):



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$$G_2(s) = \frac{e^{-4s}}{(1 + s)^2}$$

Control scheme design: Solution 2



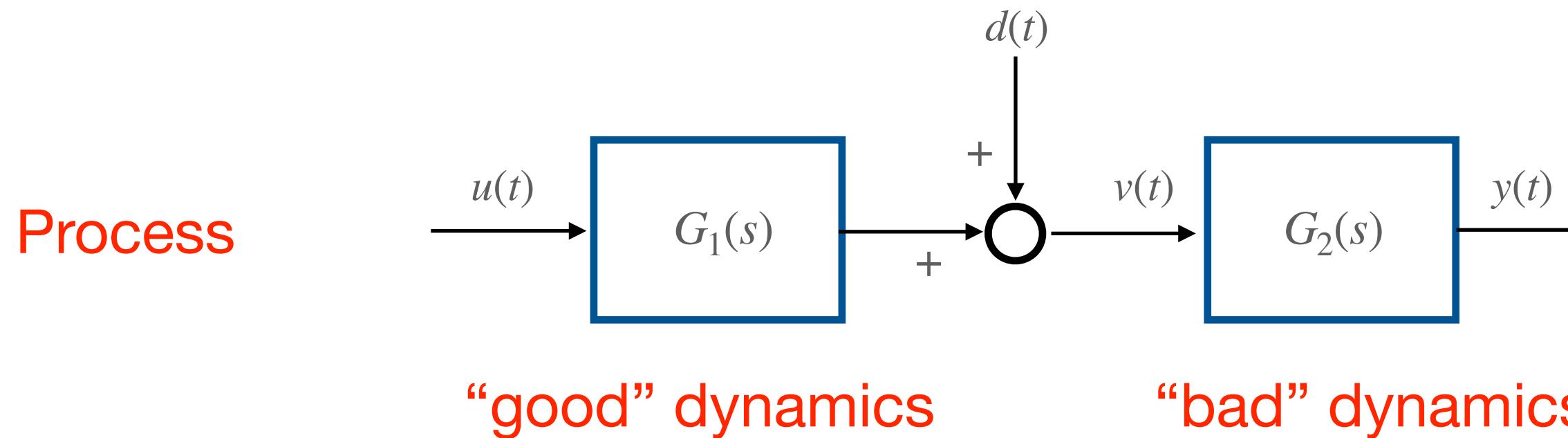
$$R_1(s) = 10$$

$$R_2(s) = \frac{0.2(s + 1)}{s}$$



Control Scheme with Decoupling in the Frequency Domain

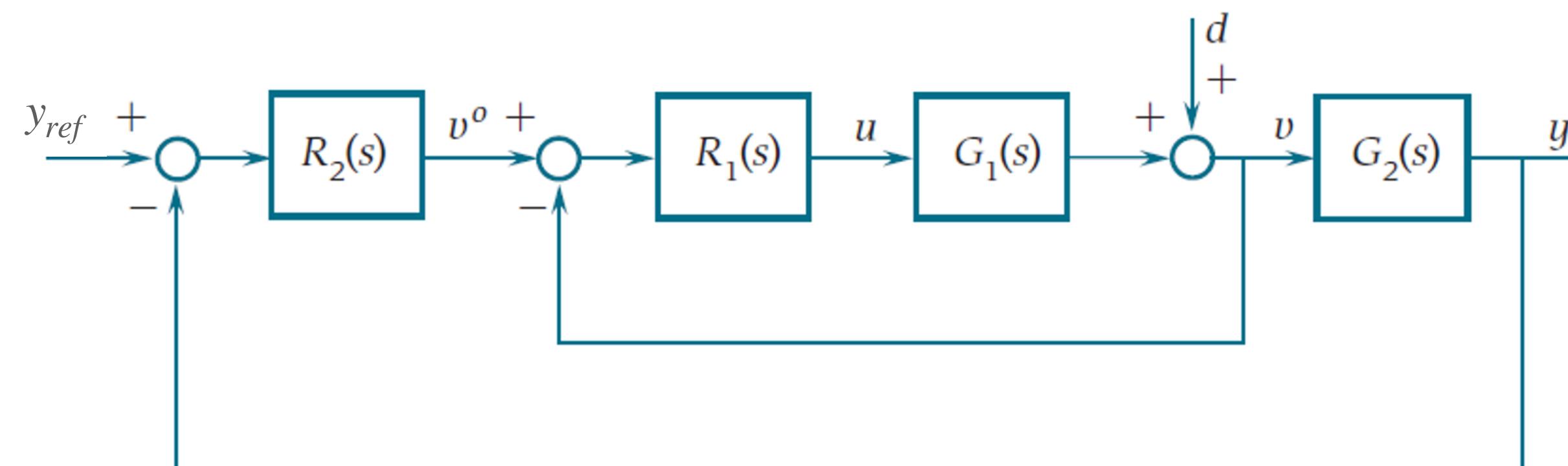
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Control scheme design: Solution 2



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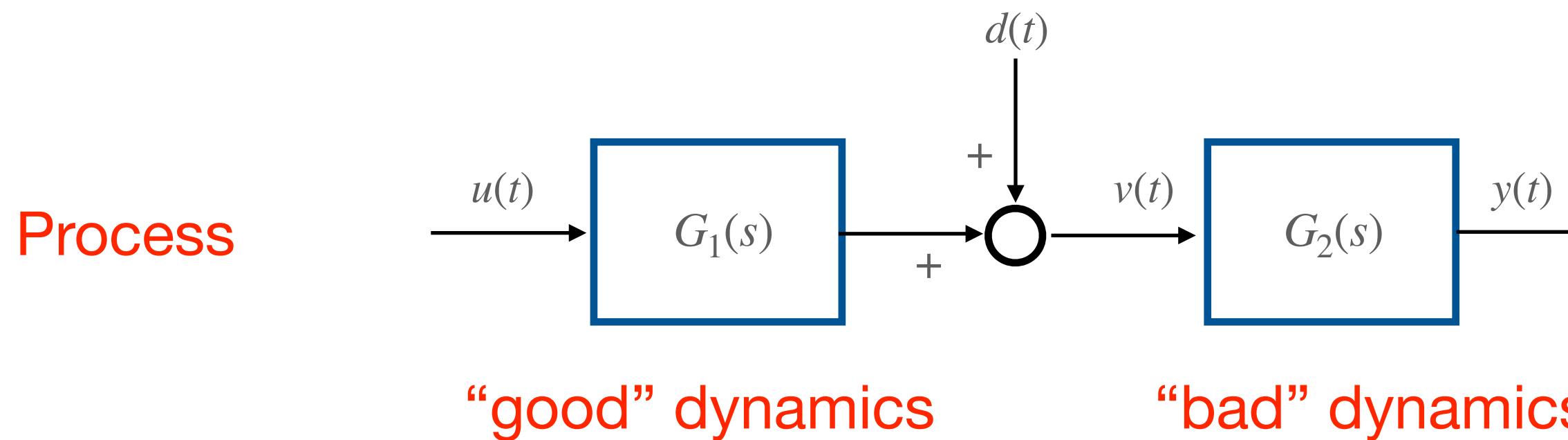
$$R_2(s) = \frac{0.2(s + 1)}{s}$$

Compute $\omega_{BW_{CL1}}$ and $\omega_{BW_{CL2}}$

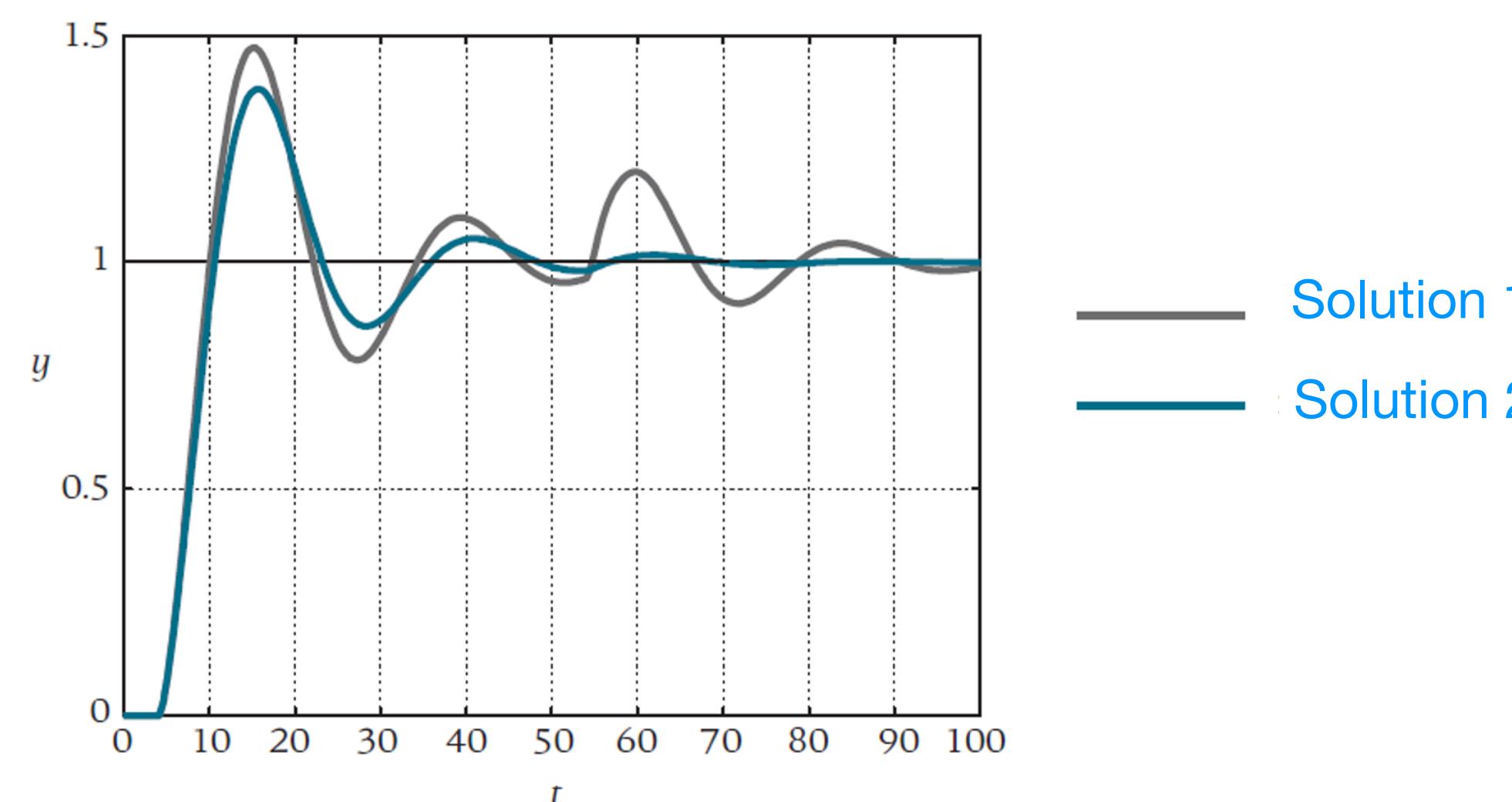


Control Scheme with Decoupling in the Frequency Domain

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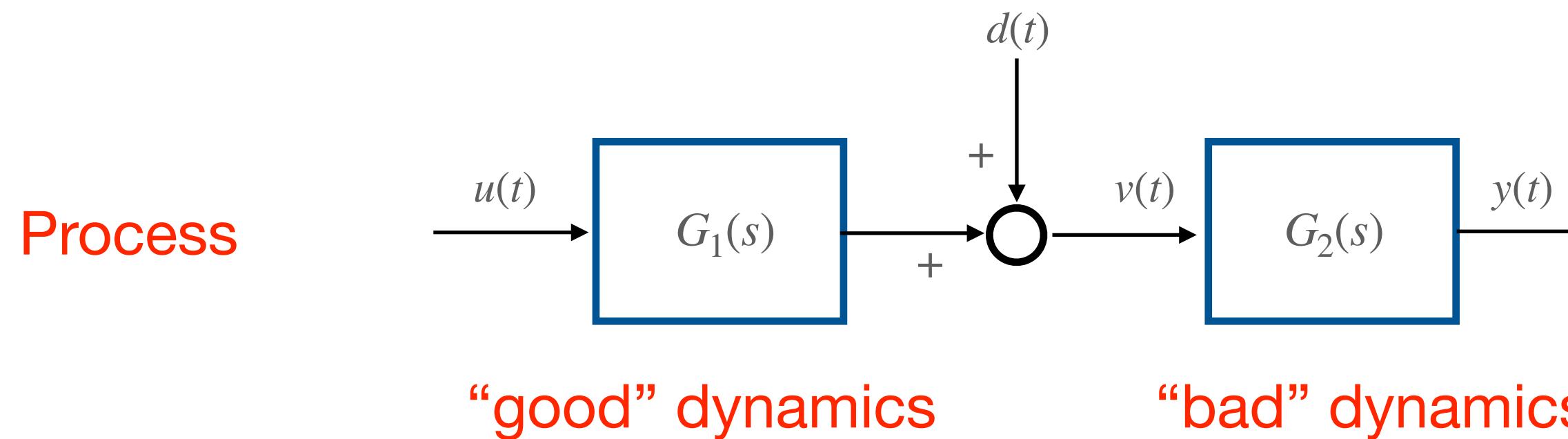


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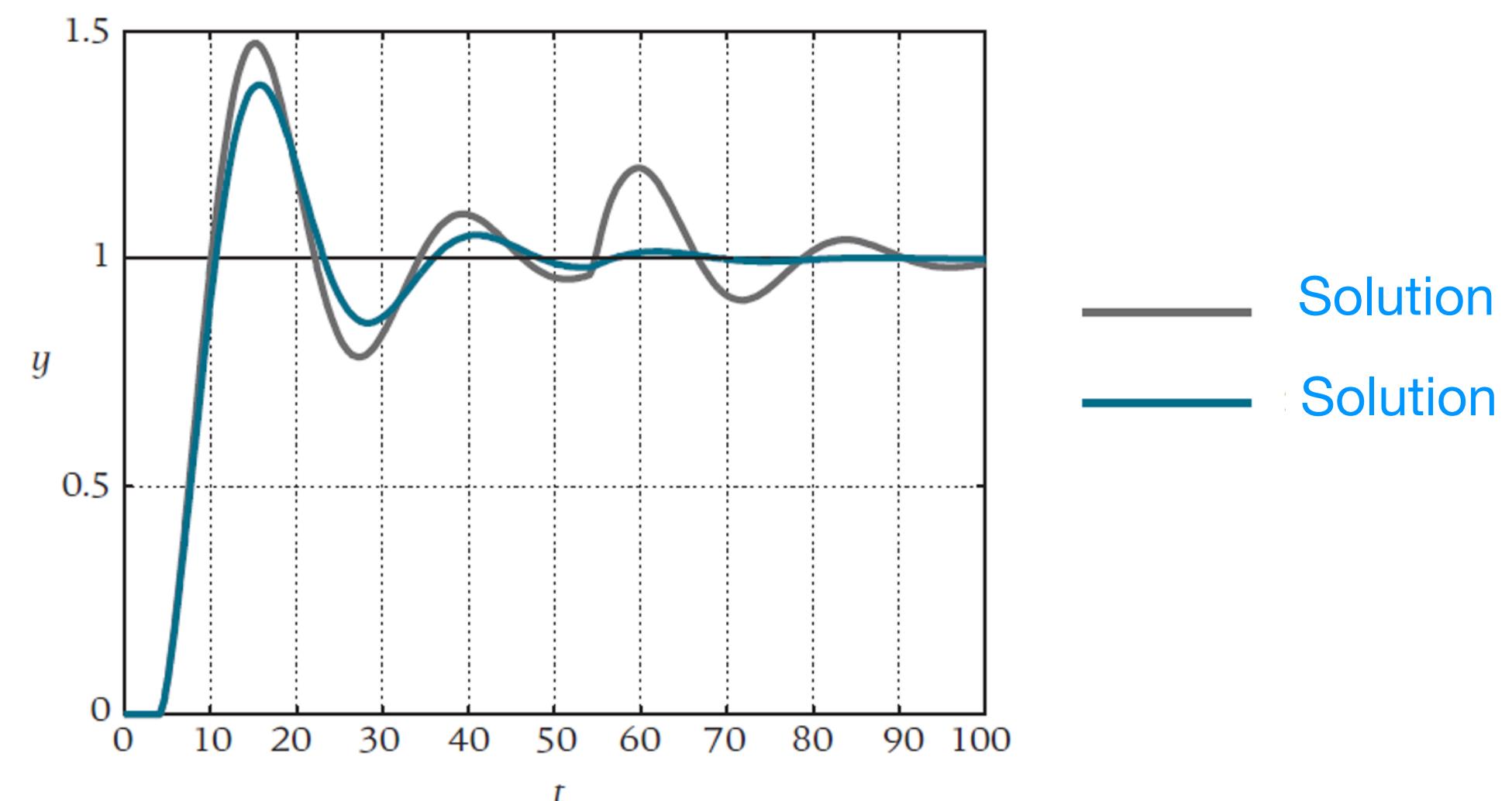


Control Scheme with Decoupling in the Frequency Domain

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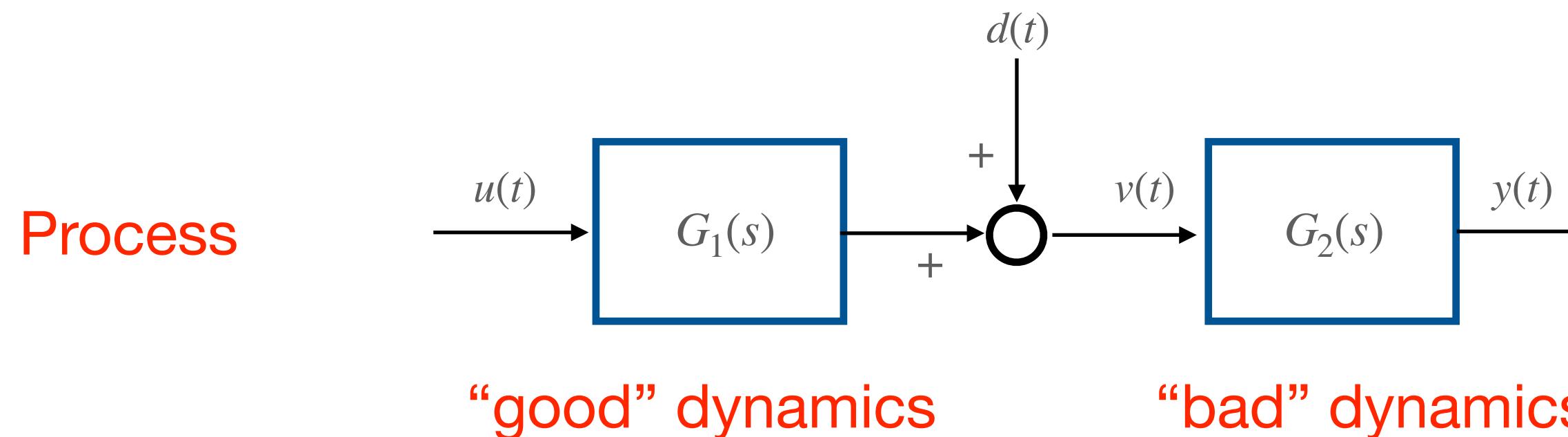


Exercise: formulate a similar example when $G_2(s)$ is rational with a real zero in +1



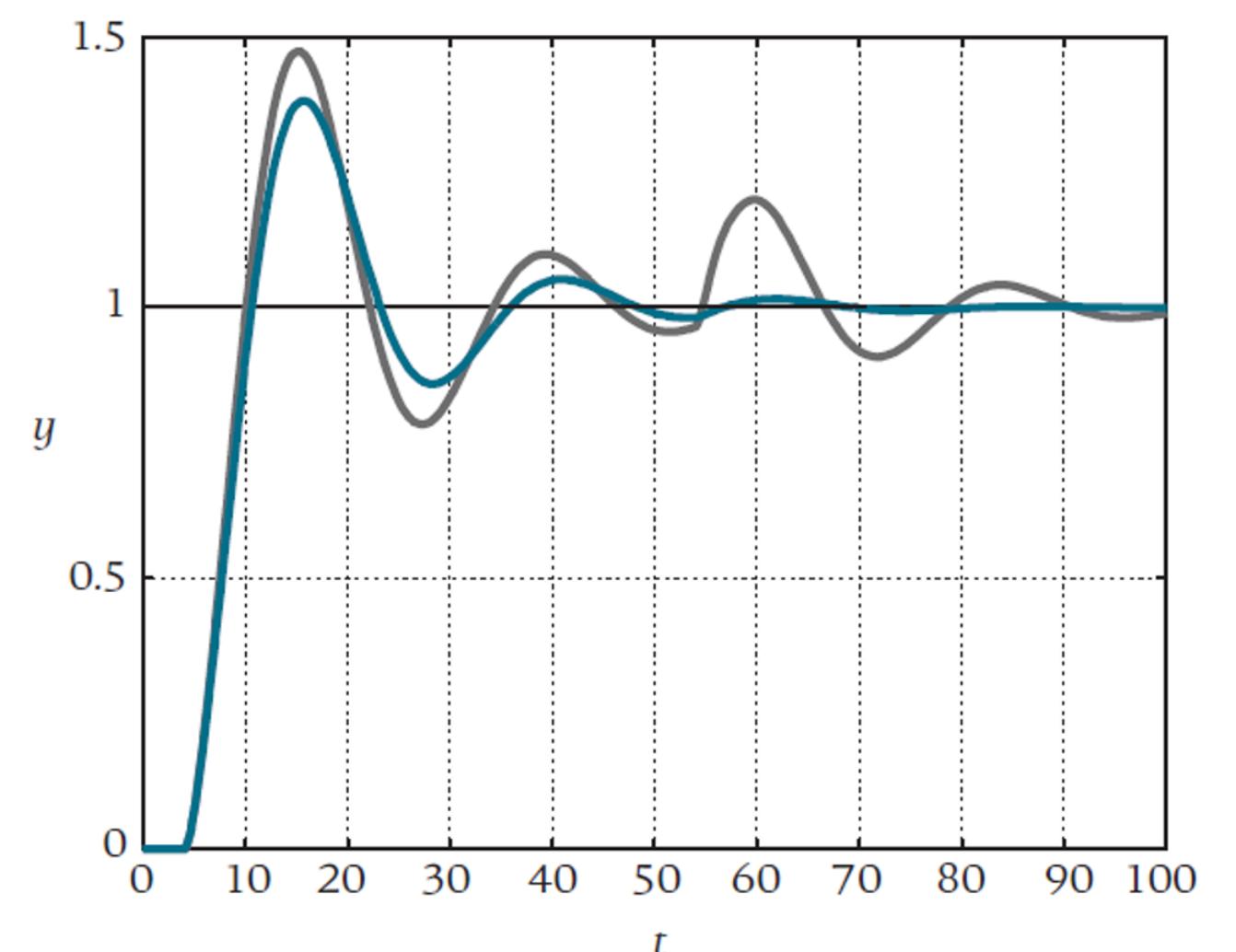
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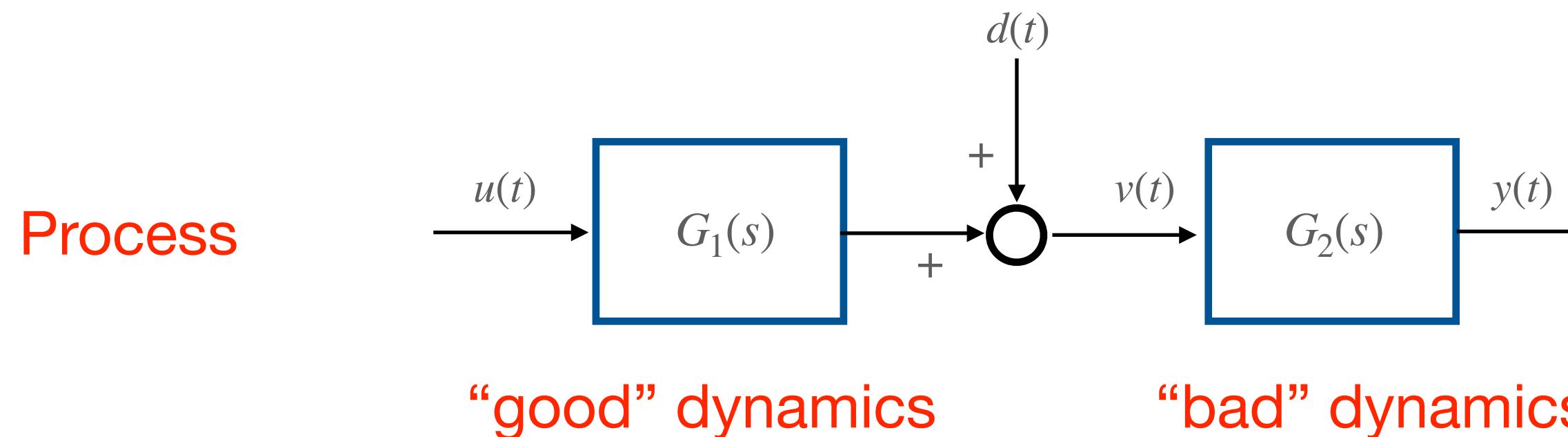
Exercise: formulate a similar example when $G_2(s)$ is rational with a real zero in +1

Can you use the unitary feedback control scheme and design a single controller $R(s)$? Try for different $d(t)$



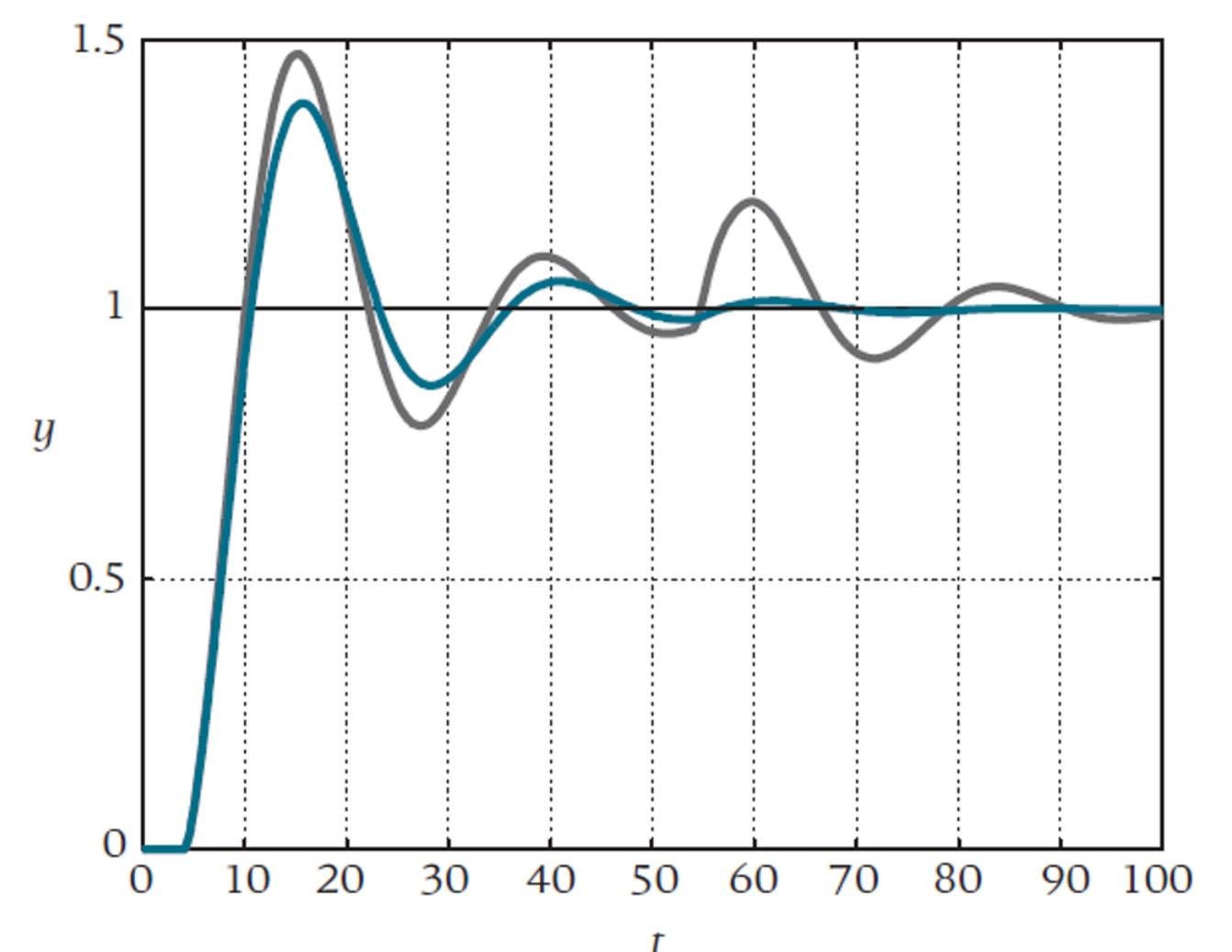
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Example (from the textbook):



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Exercise: formulate a similar example when $G_2(s)$ is rational with a real zero in +1

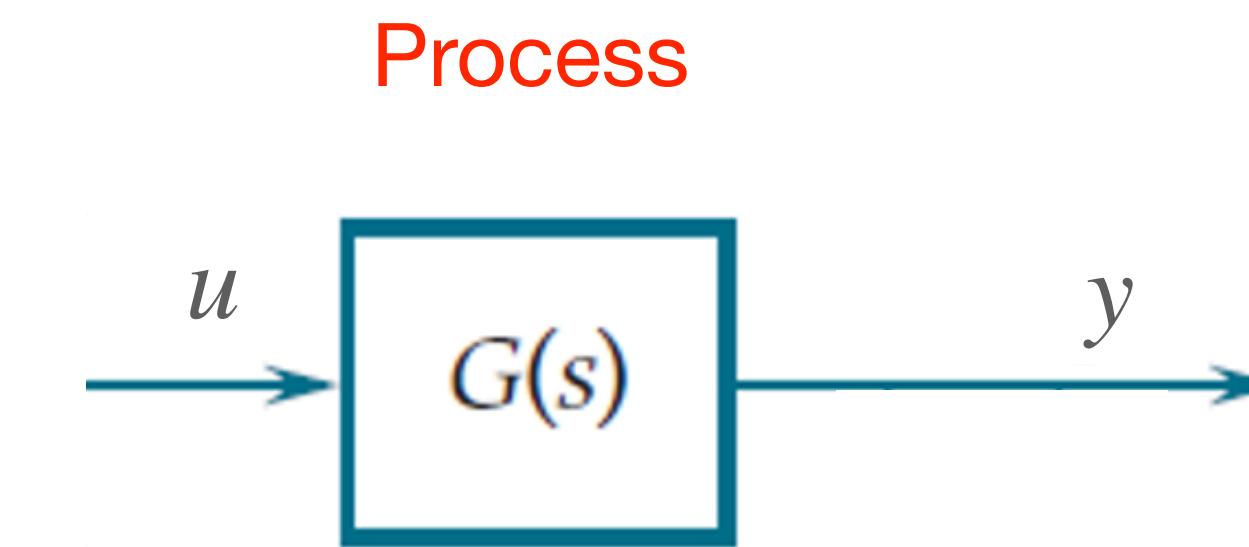
Use decoupling in the freq. dom.
Find $R_1(s)$ and $R_2(s)$ and simulate the step response in MATLAB



Control of Open-loop Unstable Systems

Assumption:

- $G(s)$ open-loop unstable



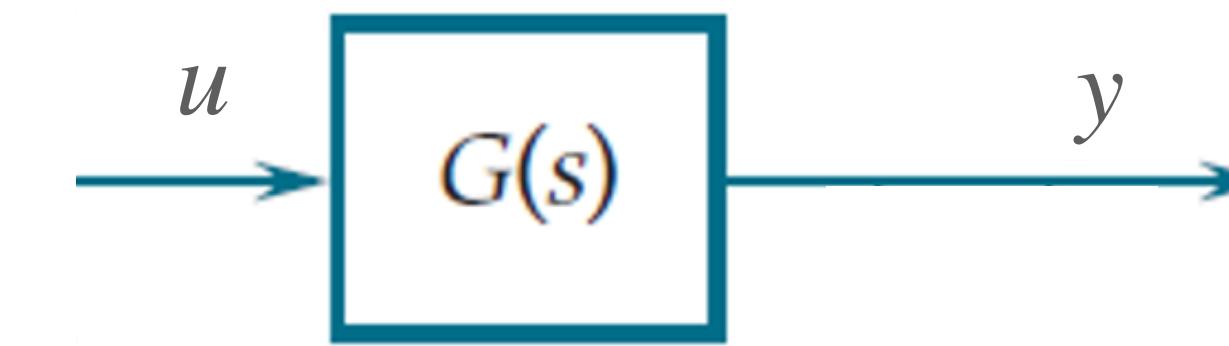
Control of Open-loop Unstable Systems

Assumption:

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Process



Bode Criterion: It requires that

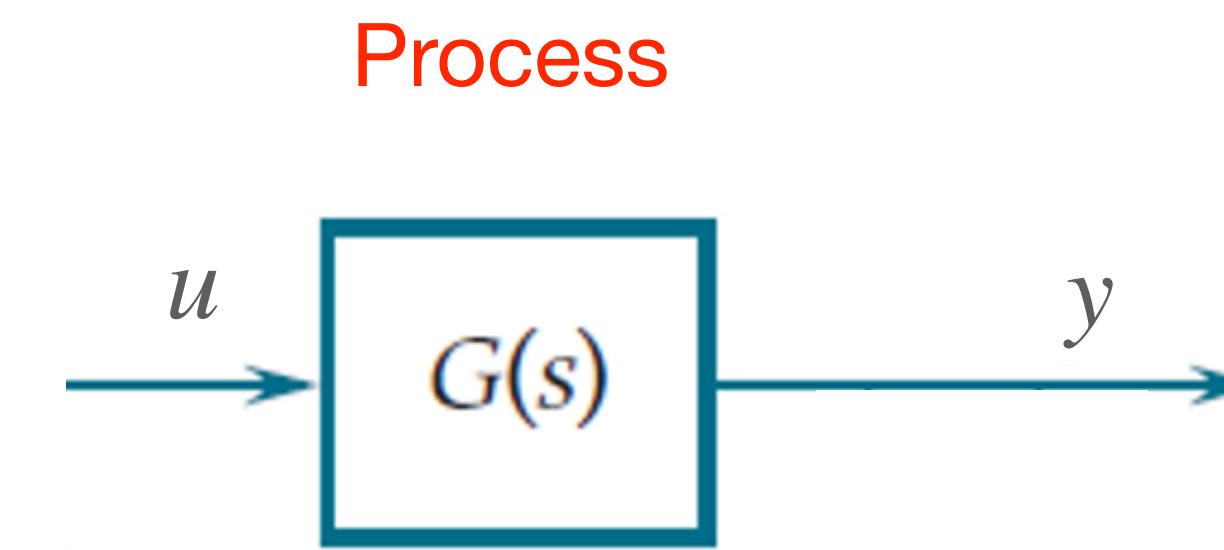
- $L(s)$ has no poles with positive real part
- the Bode plot of the magnitude of $L(j\omega)$ crosses the 0 dB axis only once



Control of Open-loop Unstable Systems

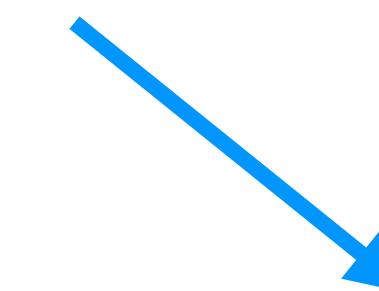
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It cannot be applied in this case!



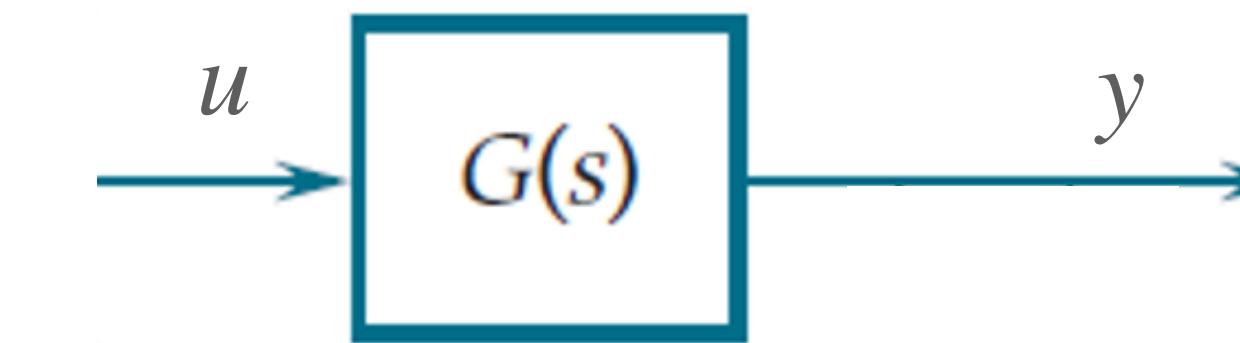
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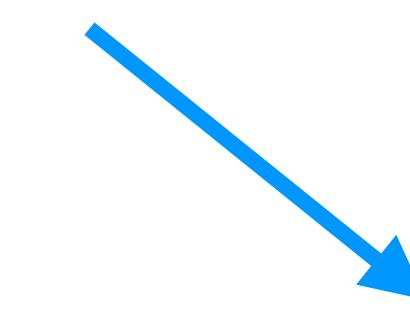


Process



Bode Criterion: It requires that

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It cannot be applied in this case!



Alternatives:

- Nyquist Criterion
- Poles Assignment
- Root Locus



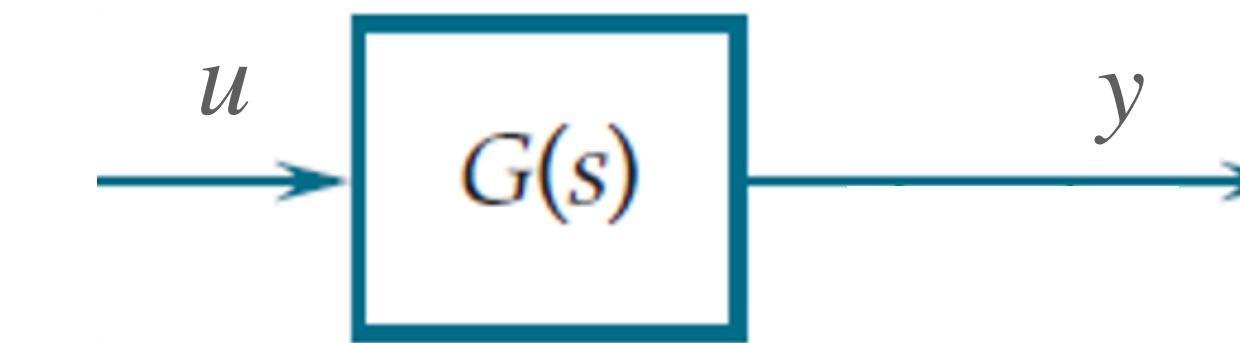
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Assumption:

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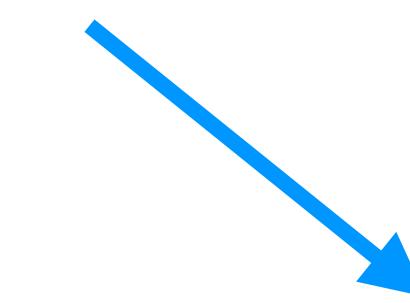


Process

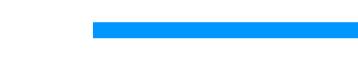


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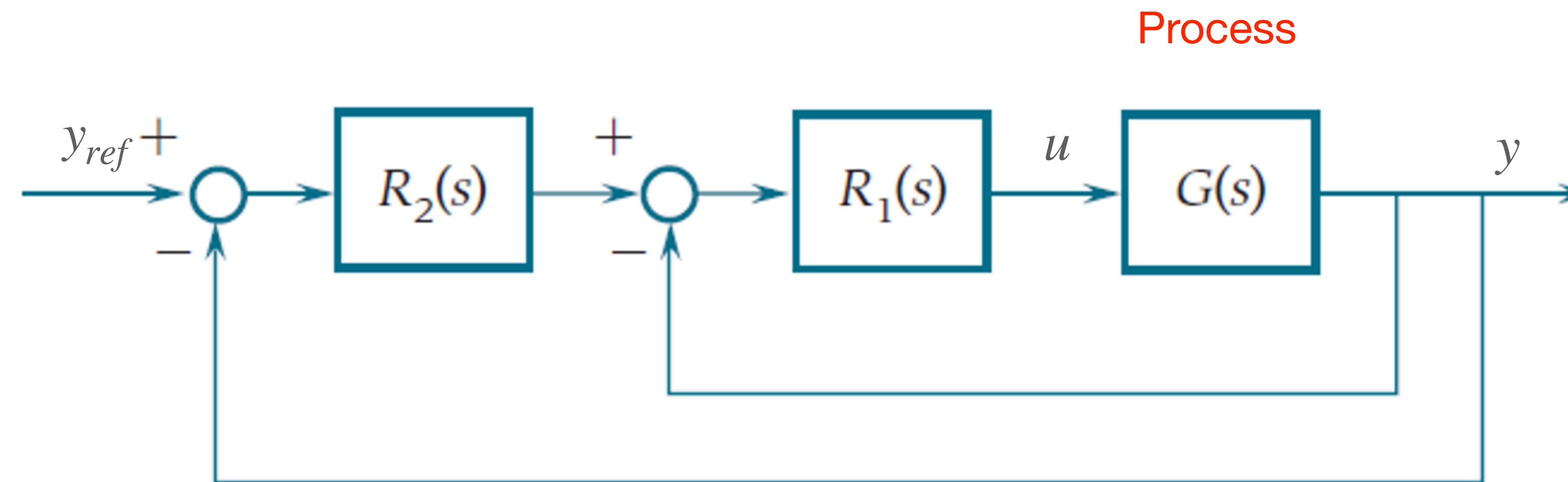


Alternatives:

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- Poles Assignment
- Root Locus



Control of Open-loop Unstable Systems



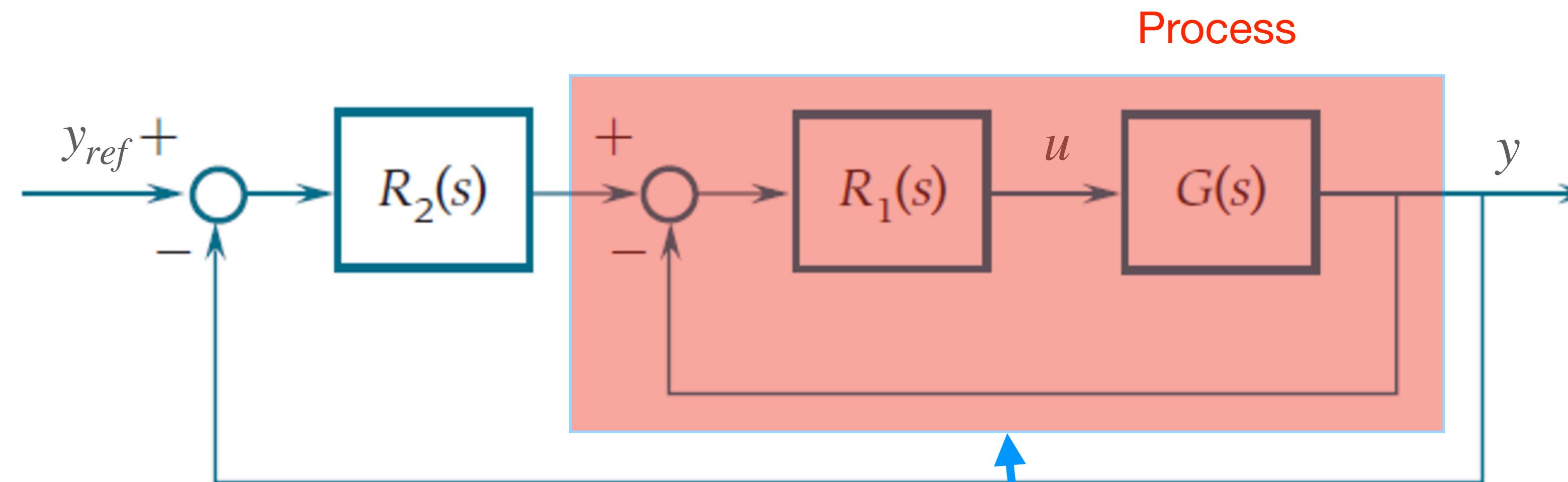
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Control of Open-loop Unstable Systems



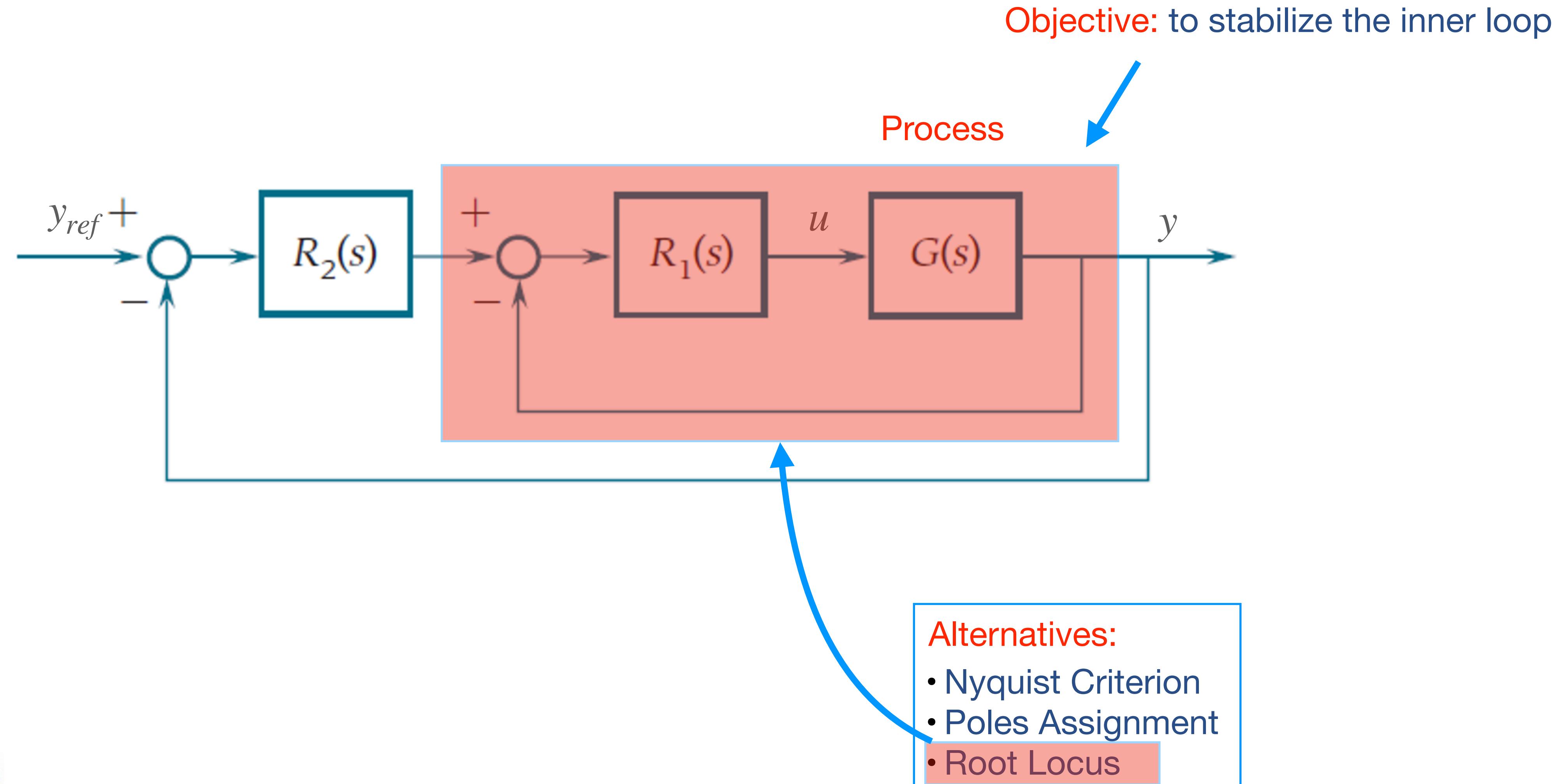
Process

Alternatives:

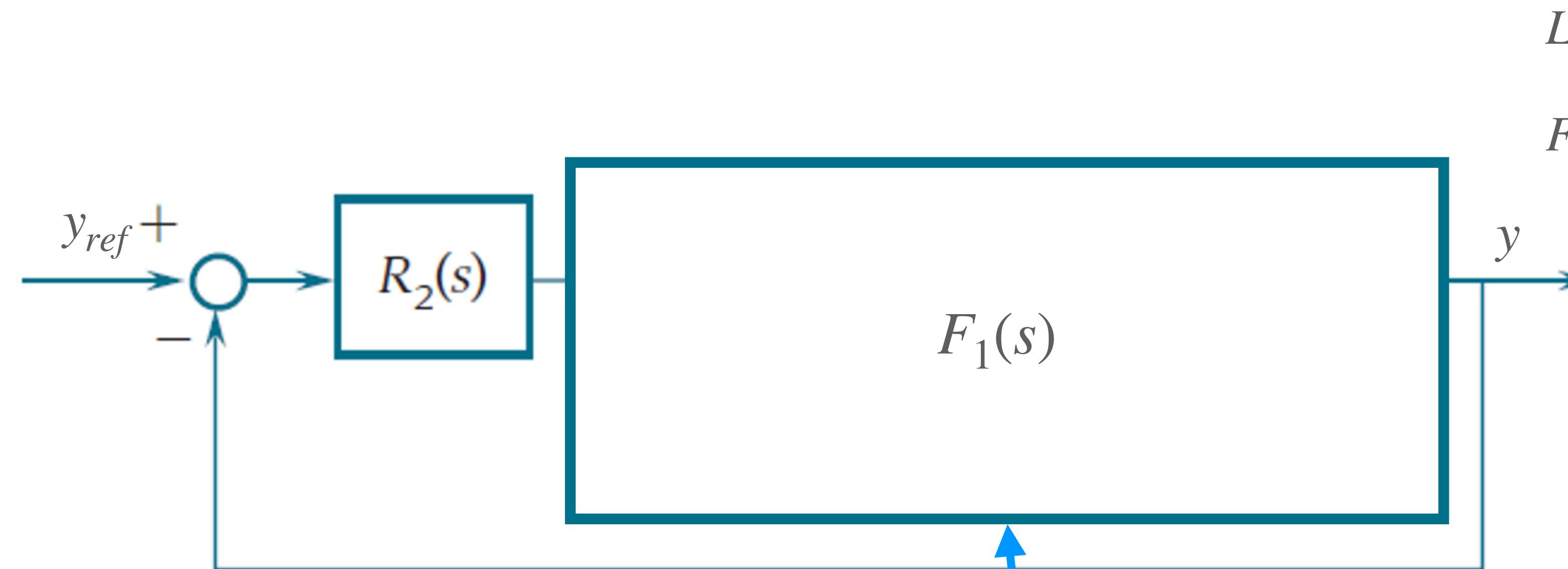
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Control of Open-loop Unstable Systems



Control of Open-loop Unstable Systems



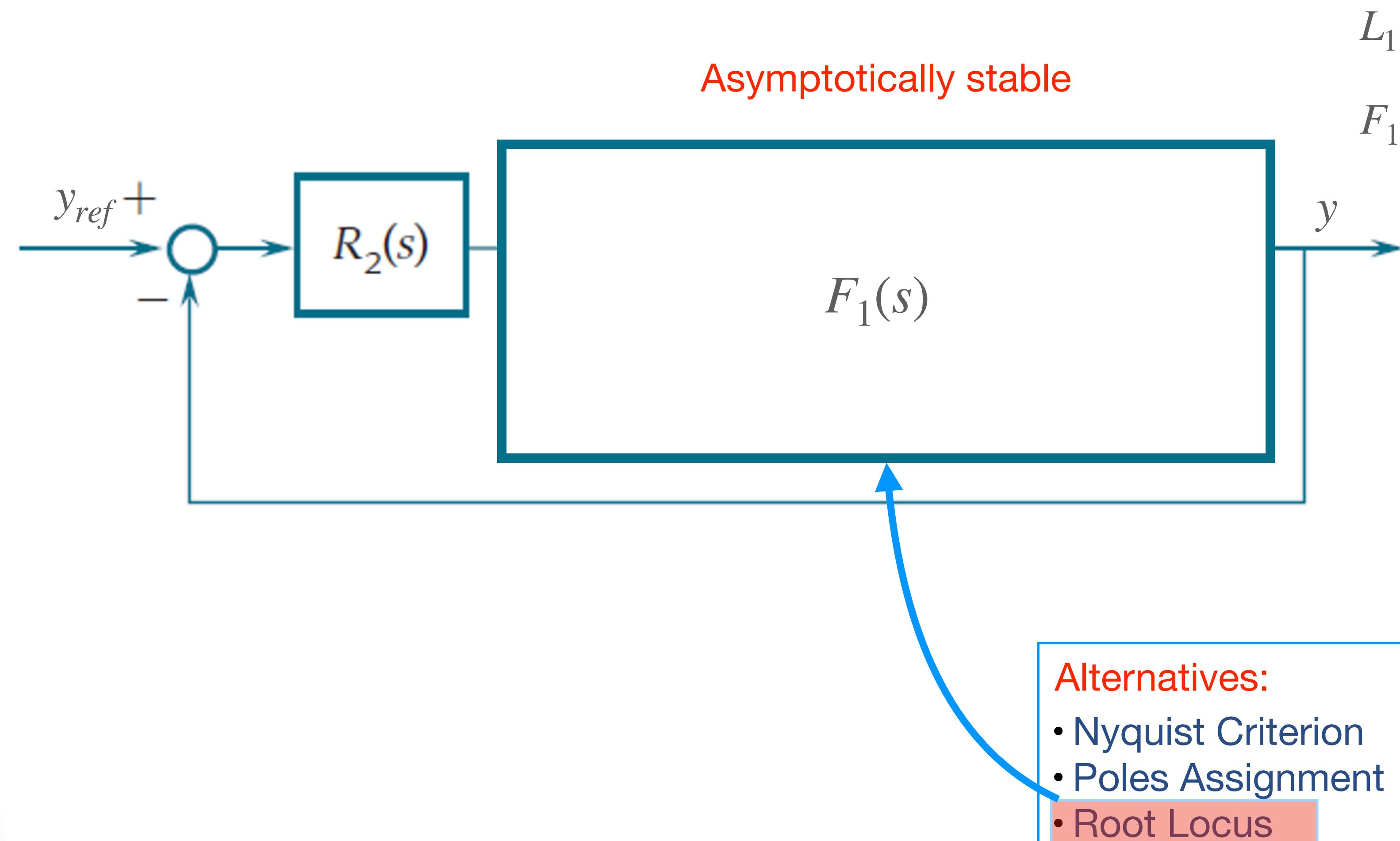
$$L_1(s) = R_1(s) G(s)$$

$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$

- Alternatives:
- Nyquist Criterion
 - Poles Assignment
 - Root Locus



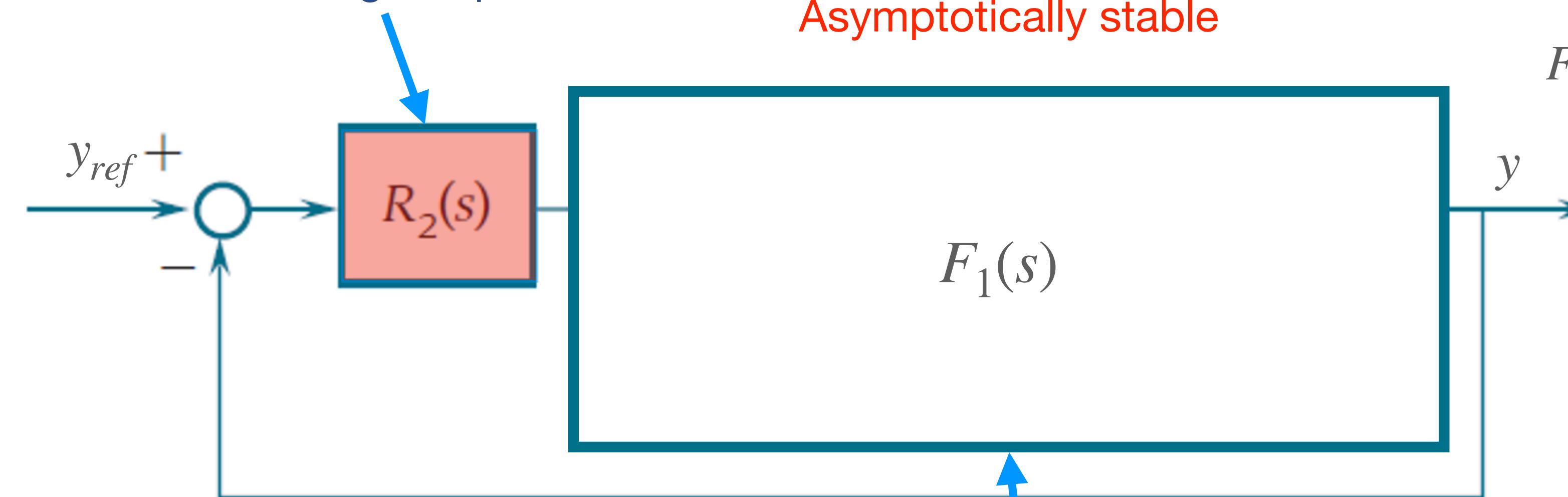
Control of Open-loop Unstable Systems



Control of Open-loop Unstable Systems

Objective: to stabilize the outer loop & satisfy the design requirements

Asymptotically stable



$$L_1(s) = R_1(s) G(s)$$

$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$

Alternatives:

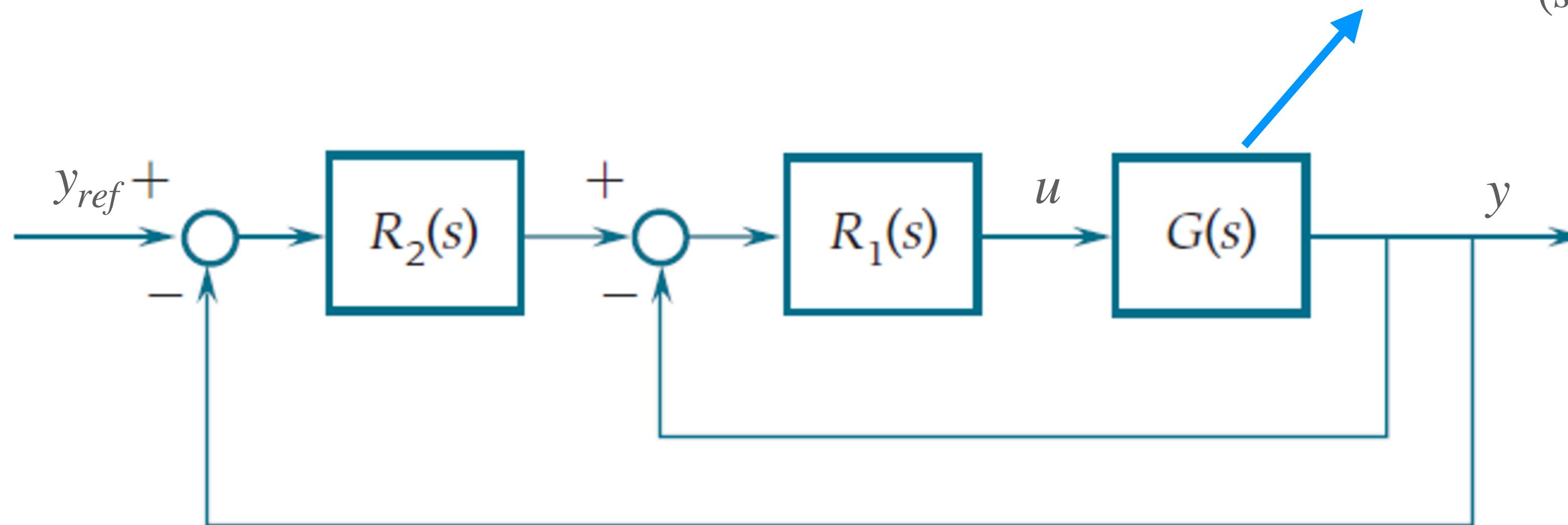
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Control of Open-loop Unstable Systems

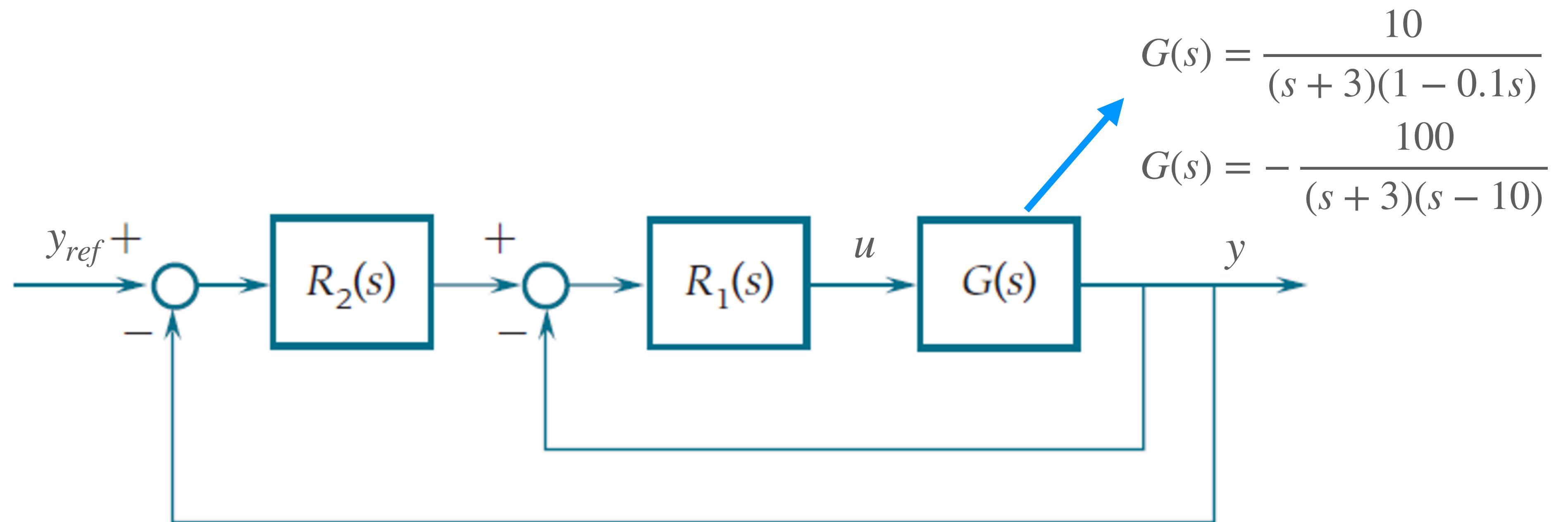
Example:

$$G(s) = \frac{10}{(s + 3)(1 - 0.1s)}$$



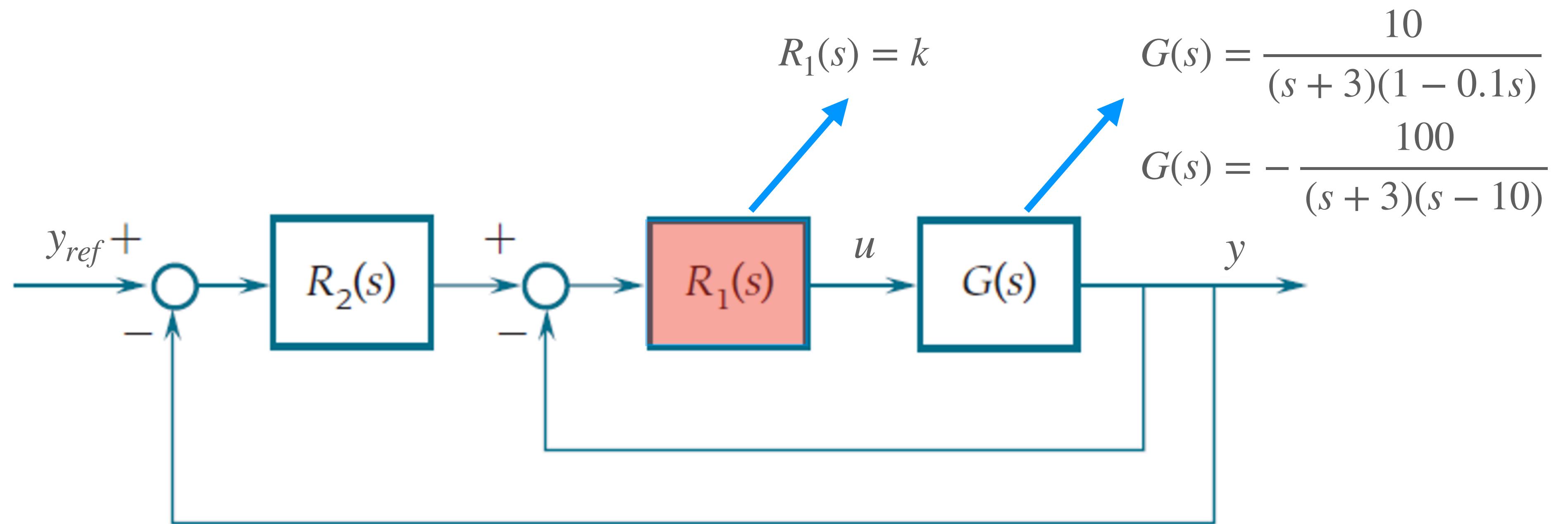
Control of Open-loop Unstable Systems

Example:



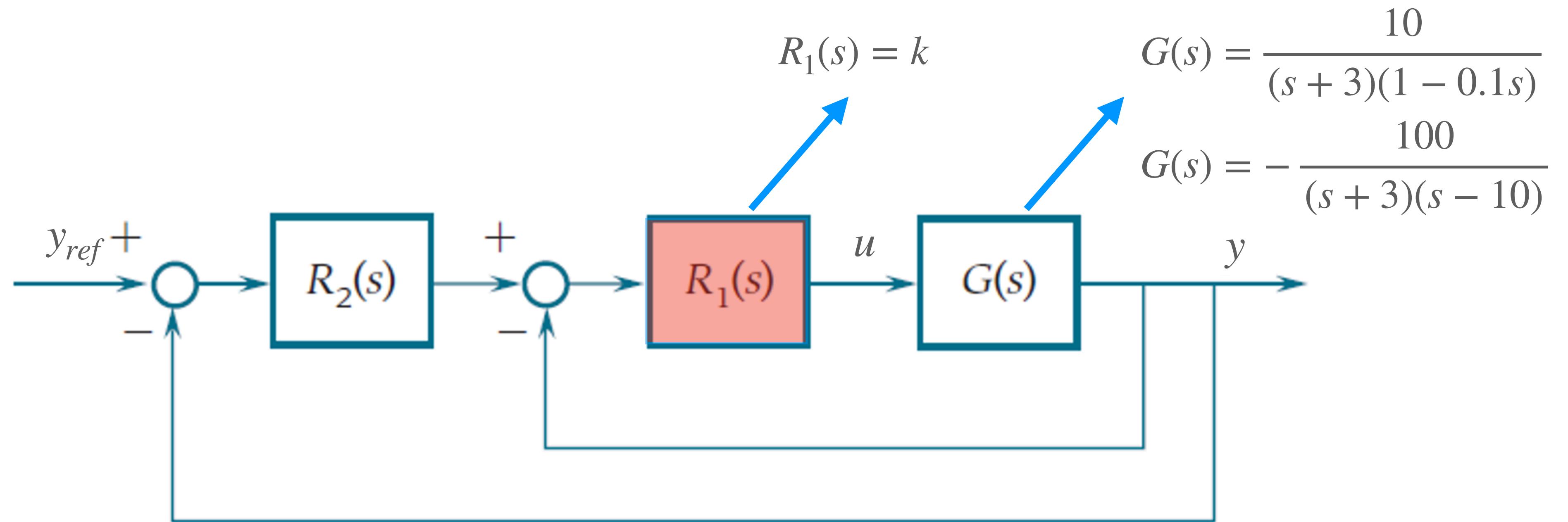
Control of Open-loop Unstable Systems

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Control of Open-loop Unstable Systems

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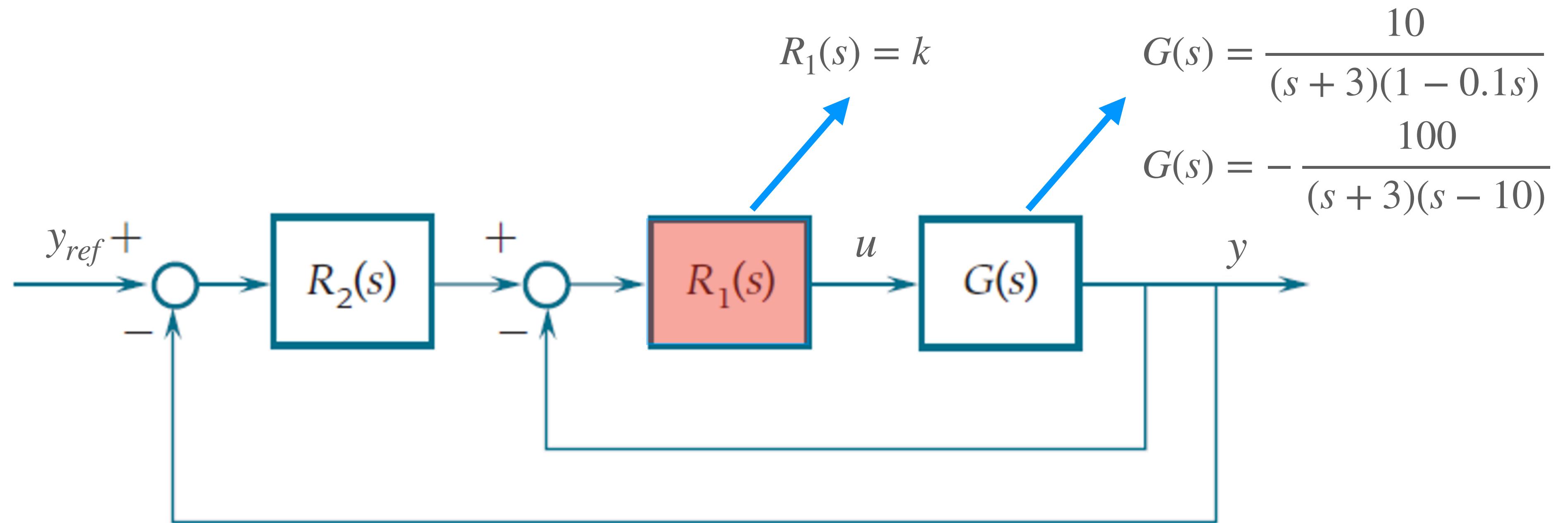
$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s + 3)(s - 10)}$$

$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$



Control of Open-loop Unstable Systems

Example:



$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s + 3)(s - 10)} = \frac{K}{(s + 3)(s - 10)}$$

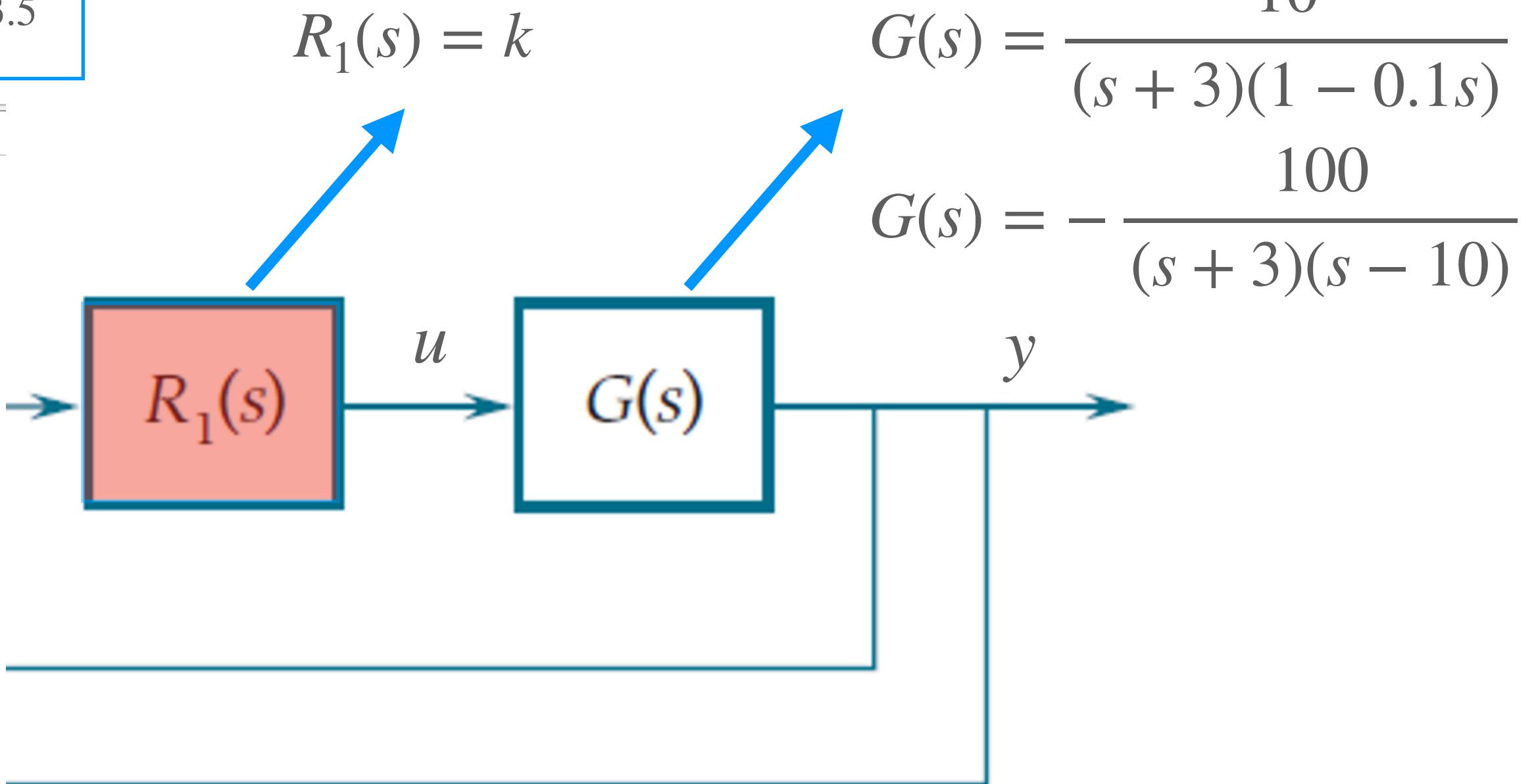
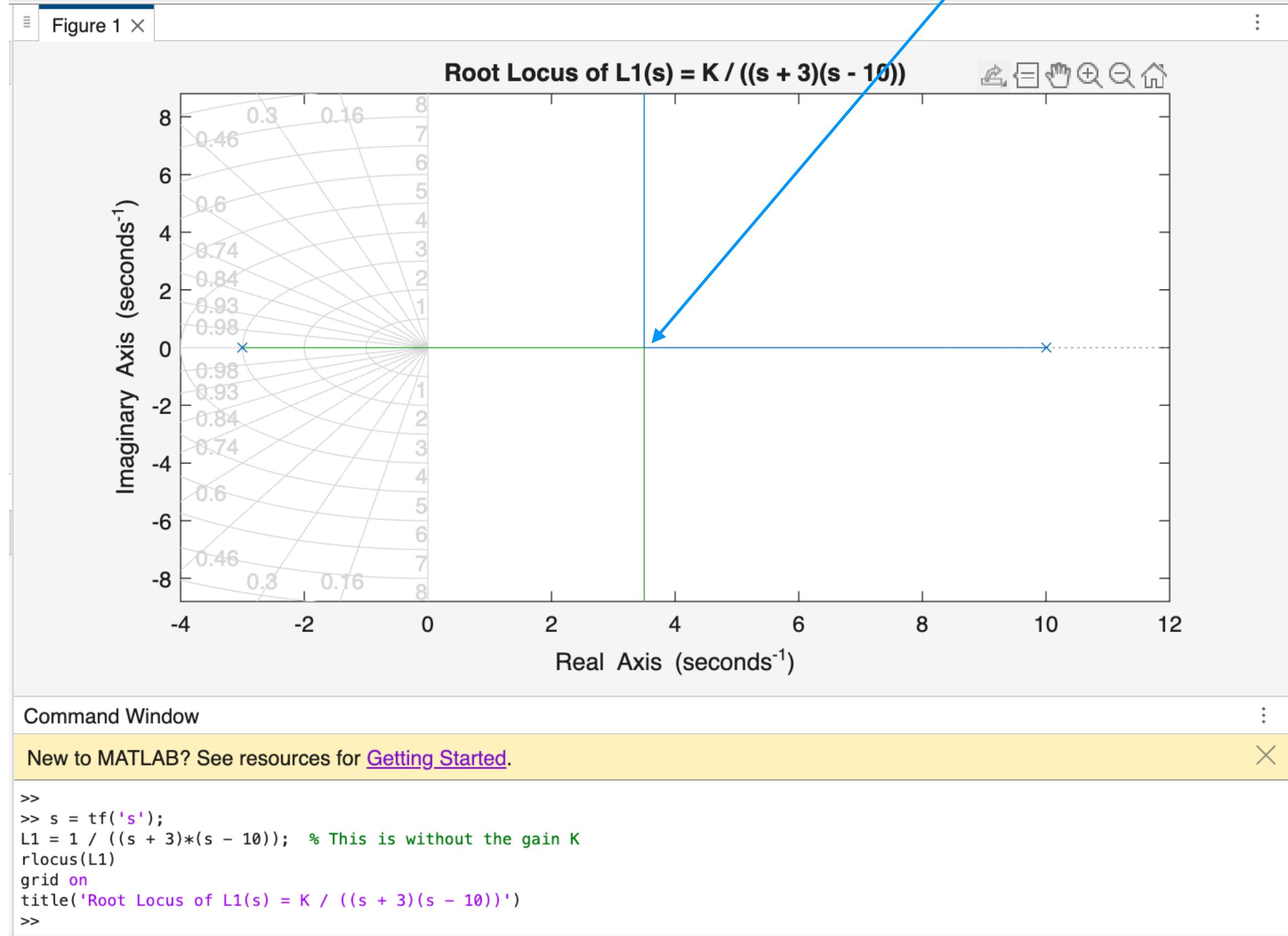
$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$



Control of Open-loop Unstable Systems

Example:

$$\sigma_a = \frac{\sum \text{poli} - \sum \text{zeri}}{n - m} \quad \sigma_a = \frac{-3 + 10}{2 - 0} = \frac{7}{2} = 3.5$$



$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s + 3)(s - 10)} = \frac{K}{(s + 3)(s - 10)}$$

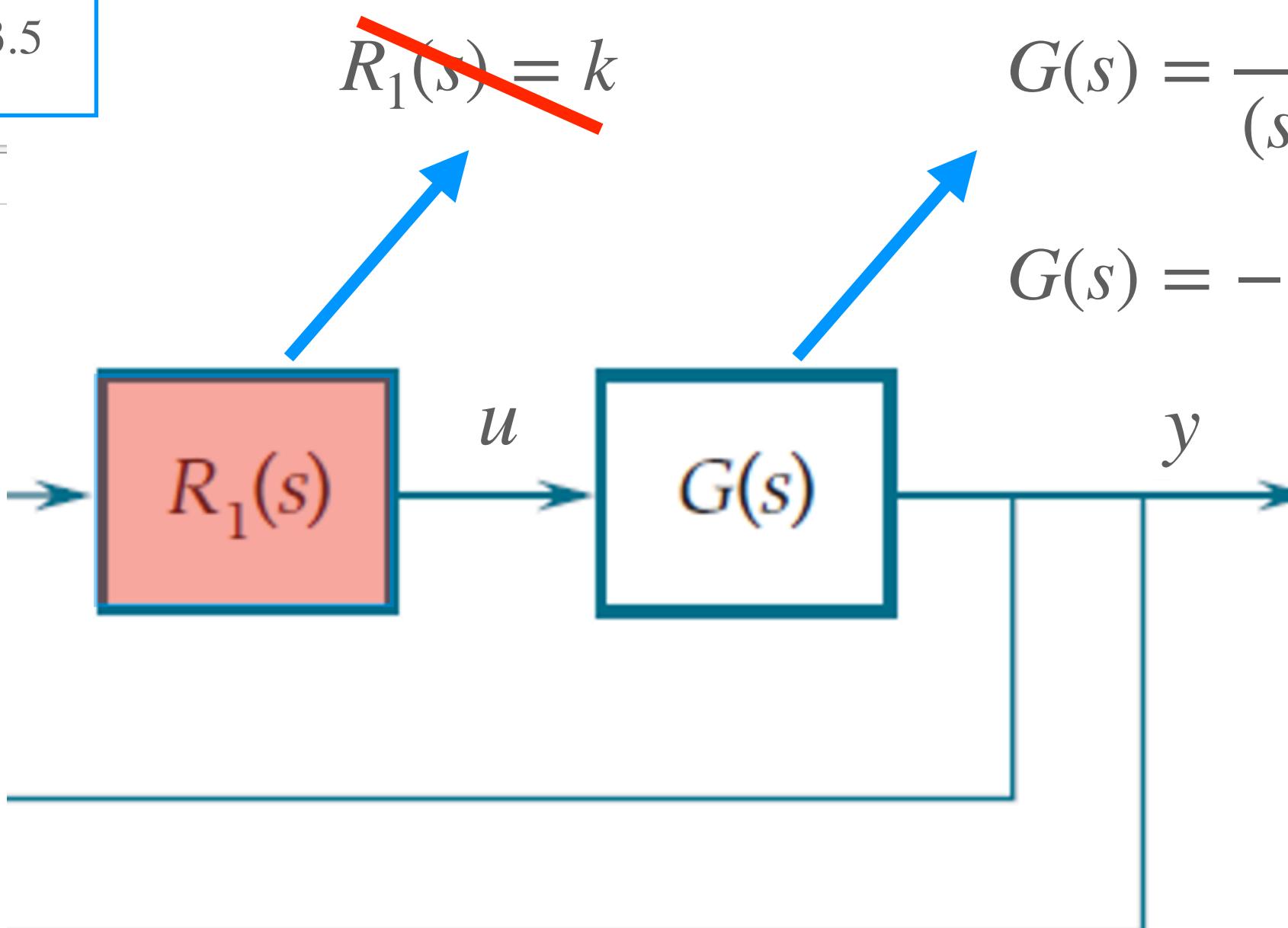
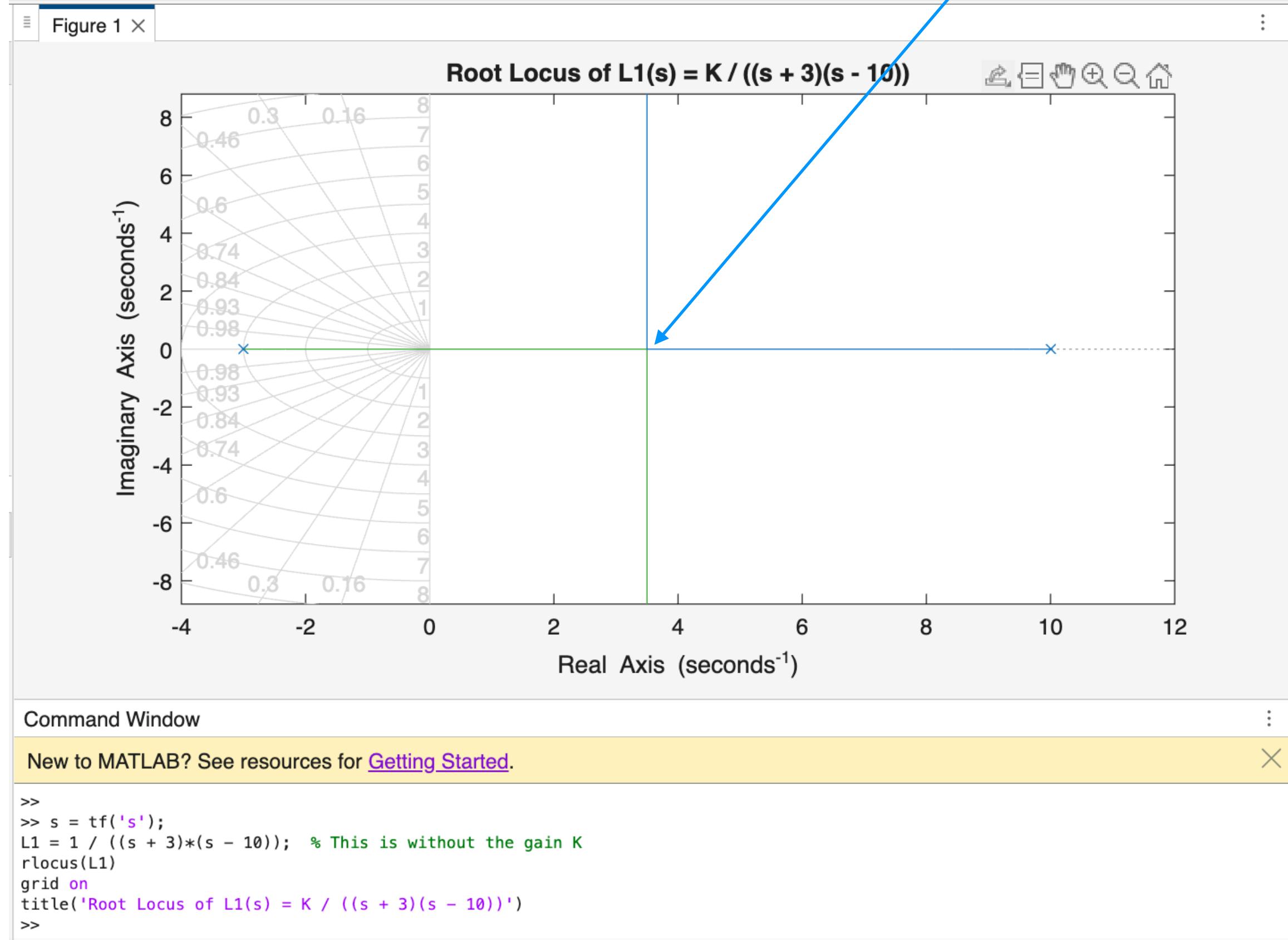
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Control of Open-loop Unstable Systems

Example:

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$$G(s) = \frac{10}{(s + 3)(1 - 0.1s)}$$

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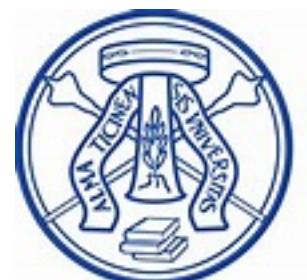
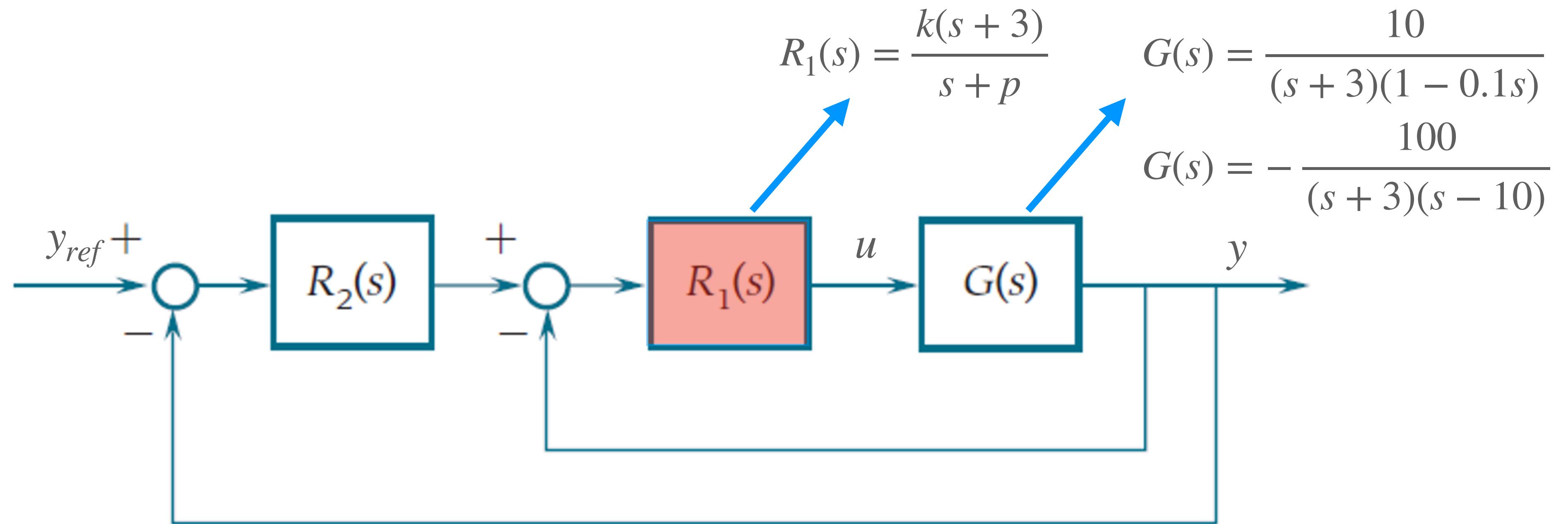
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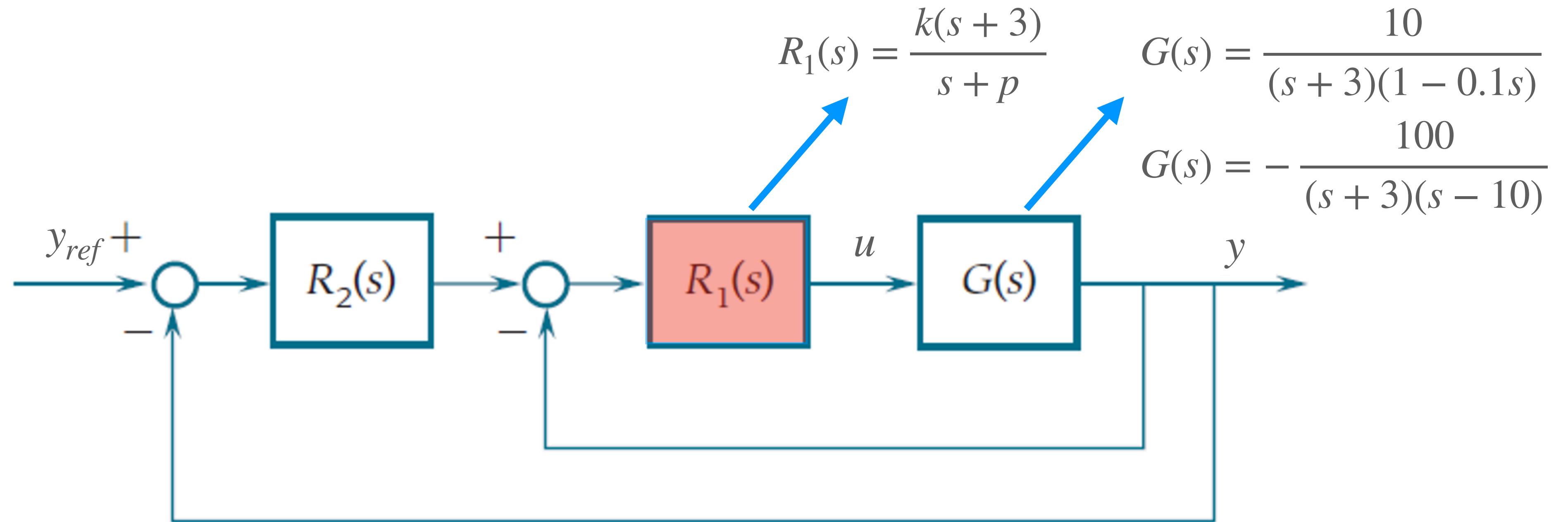
Control of Open-loop Unstable Systems

Example:



Control of Open-loop Unstable Systems

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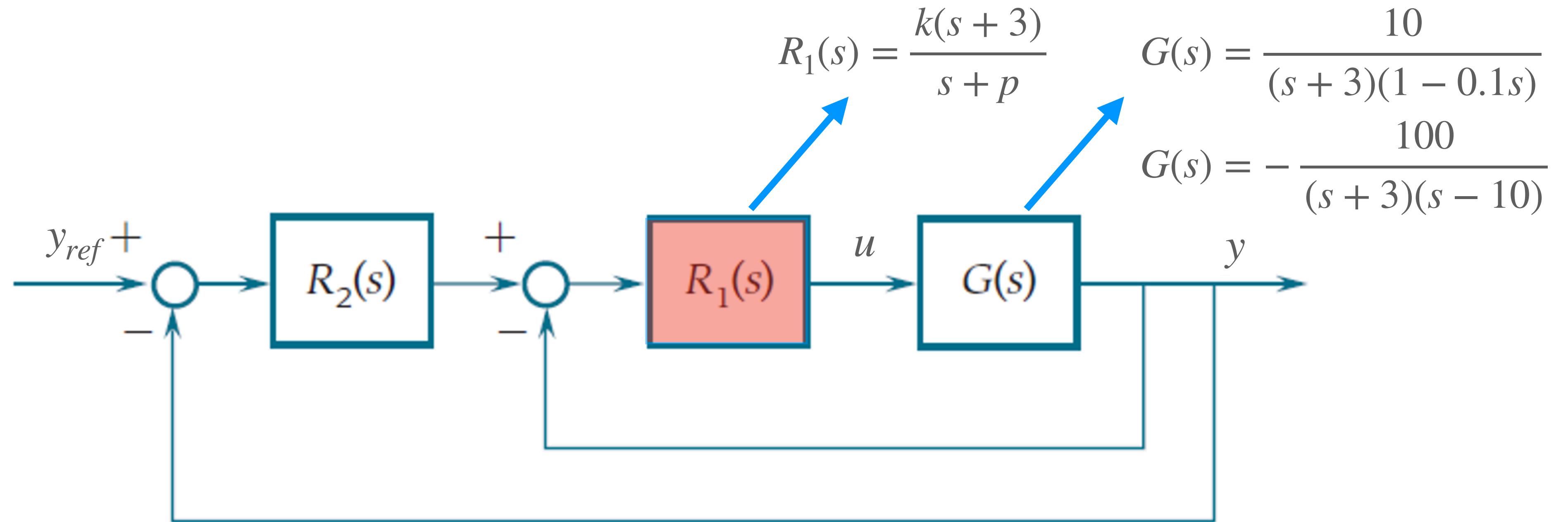
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Control of Open-loop Unstable Systems

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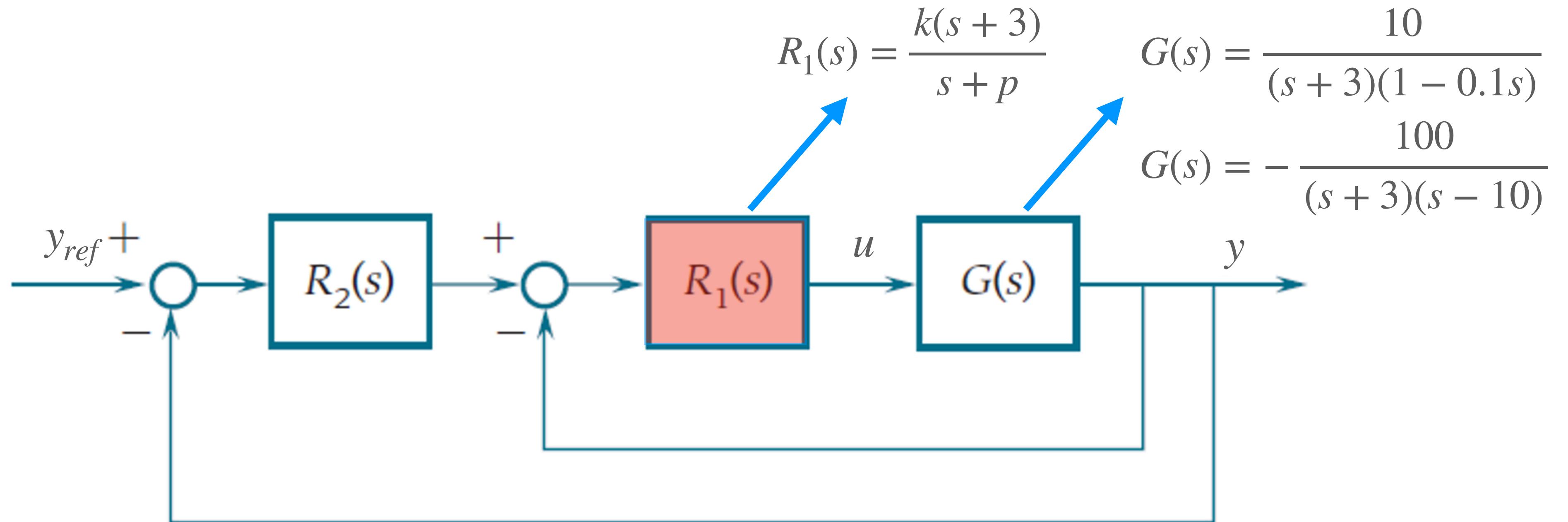
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Control of Open-loop Unstable Systems

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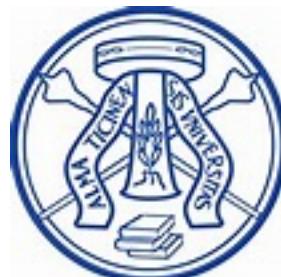
$$\sigma_a = \frac{\sum \text{poli} - \sum \text{zeri}}{n-m}$$

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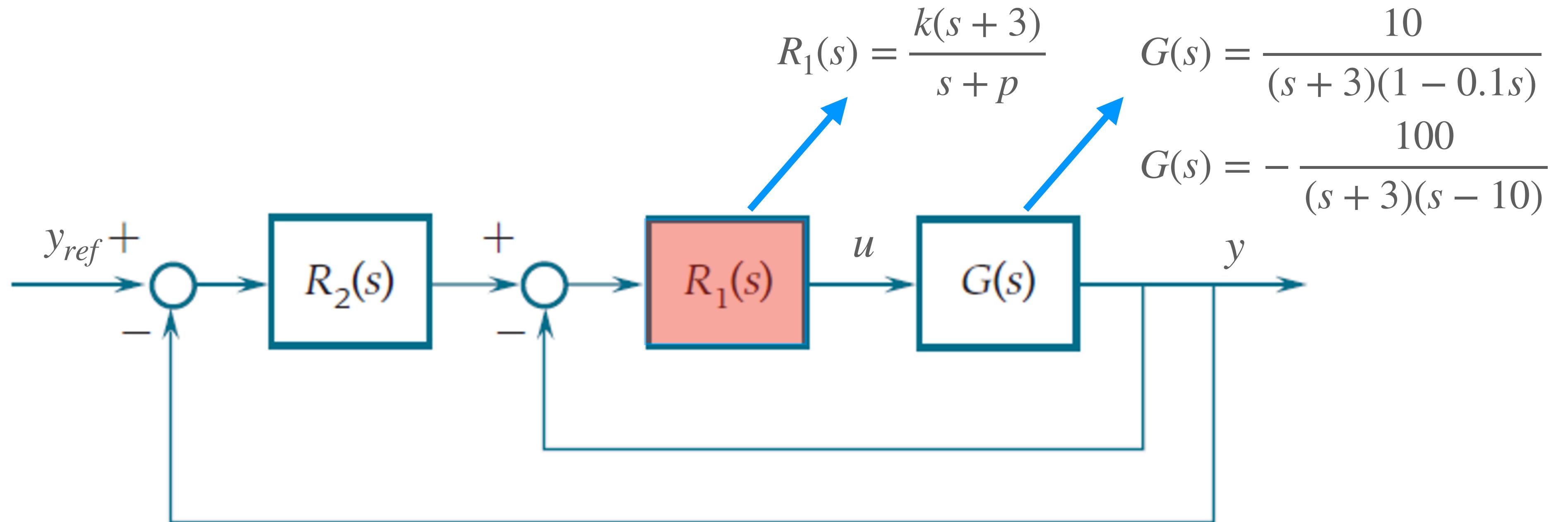
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Control of Open-loop Unstable Systems

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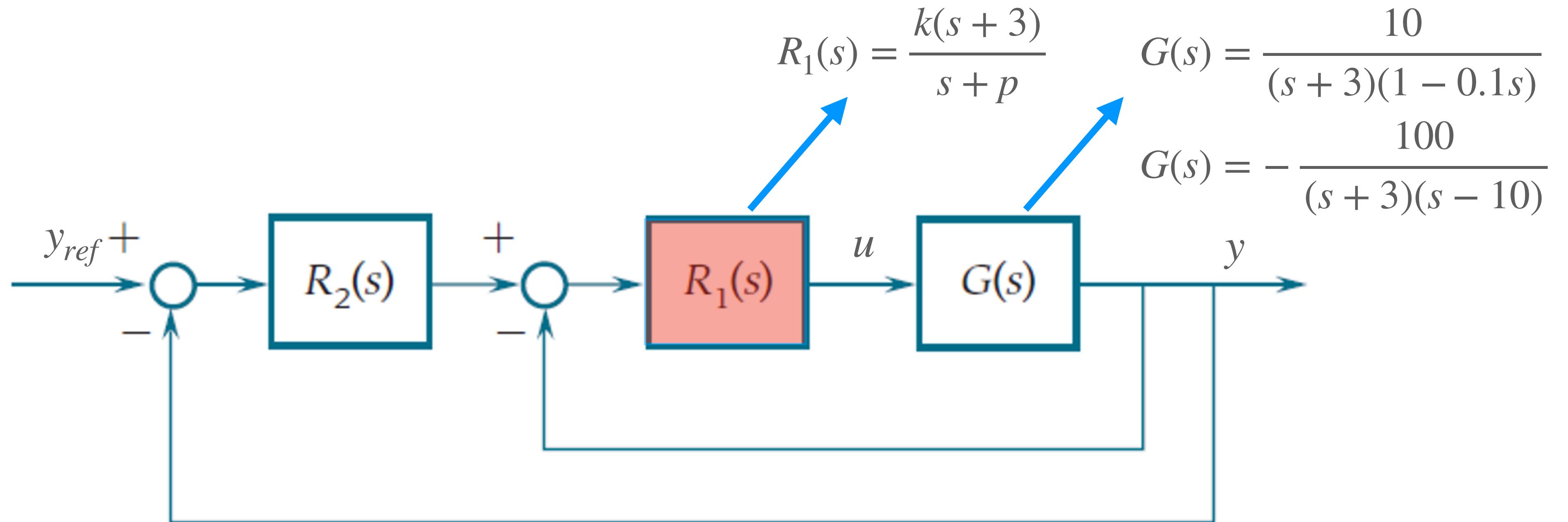
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Control of Open-loop Unstable Systems

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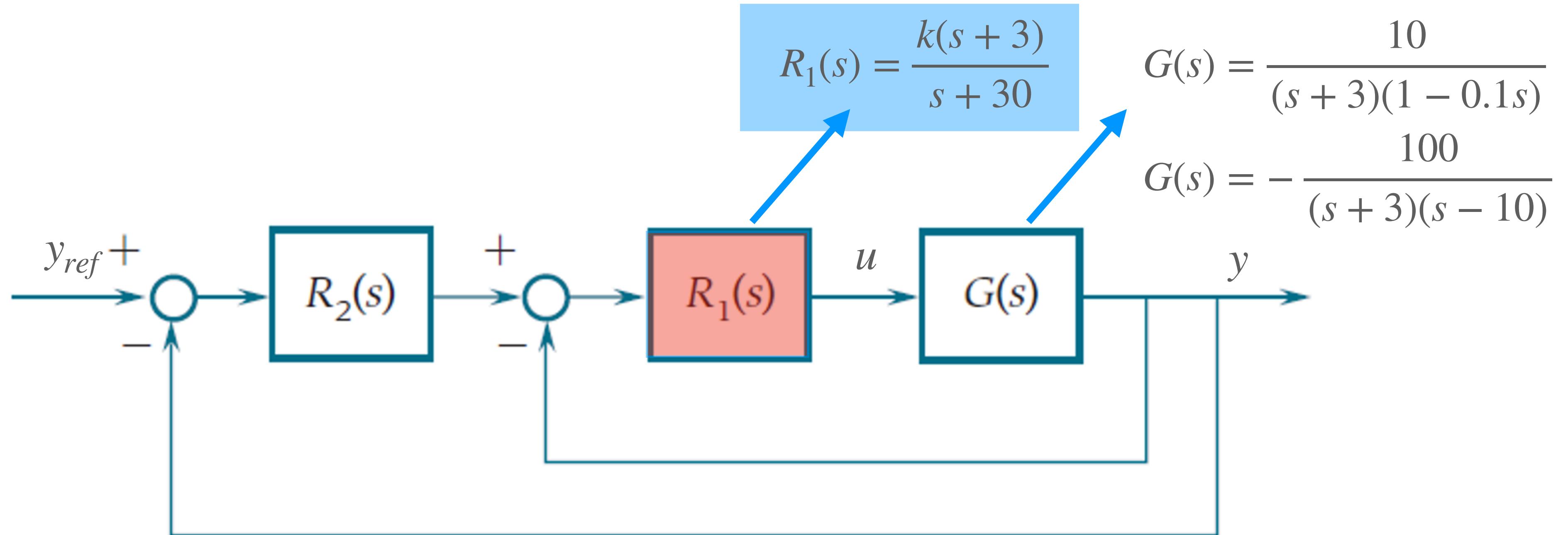
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Control of Open-loop Unstable Systems

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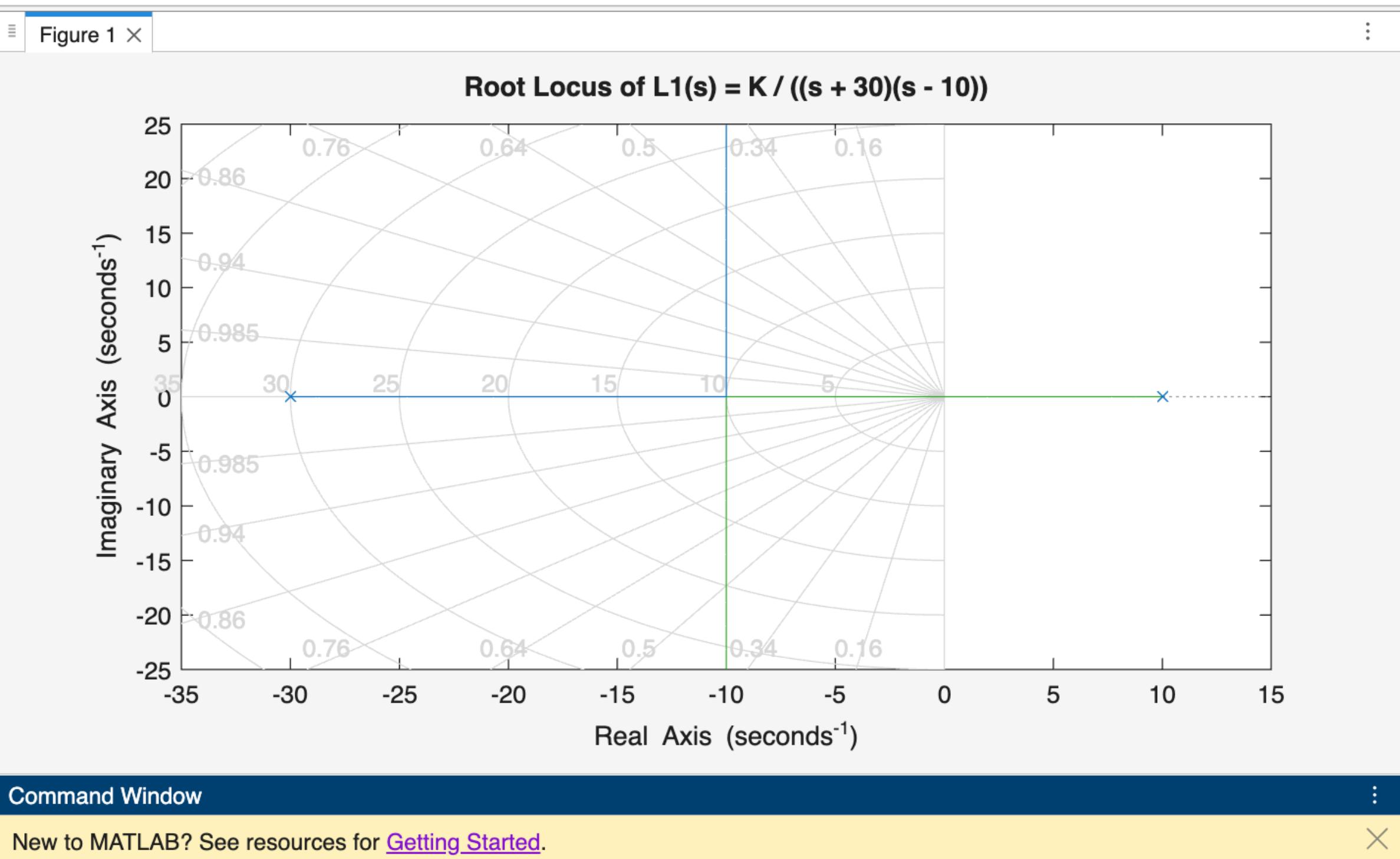
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Control of Open-loop Unstable Systems

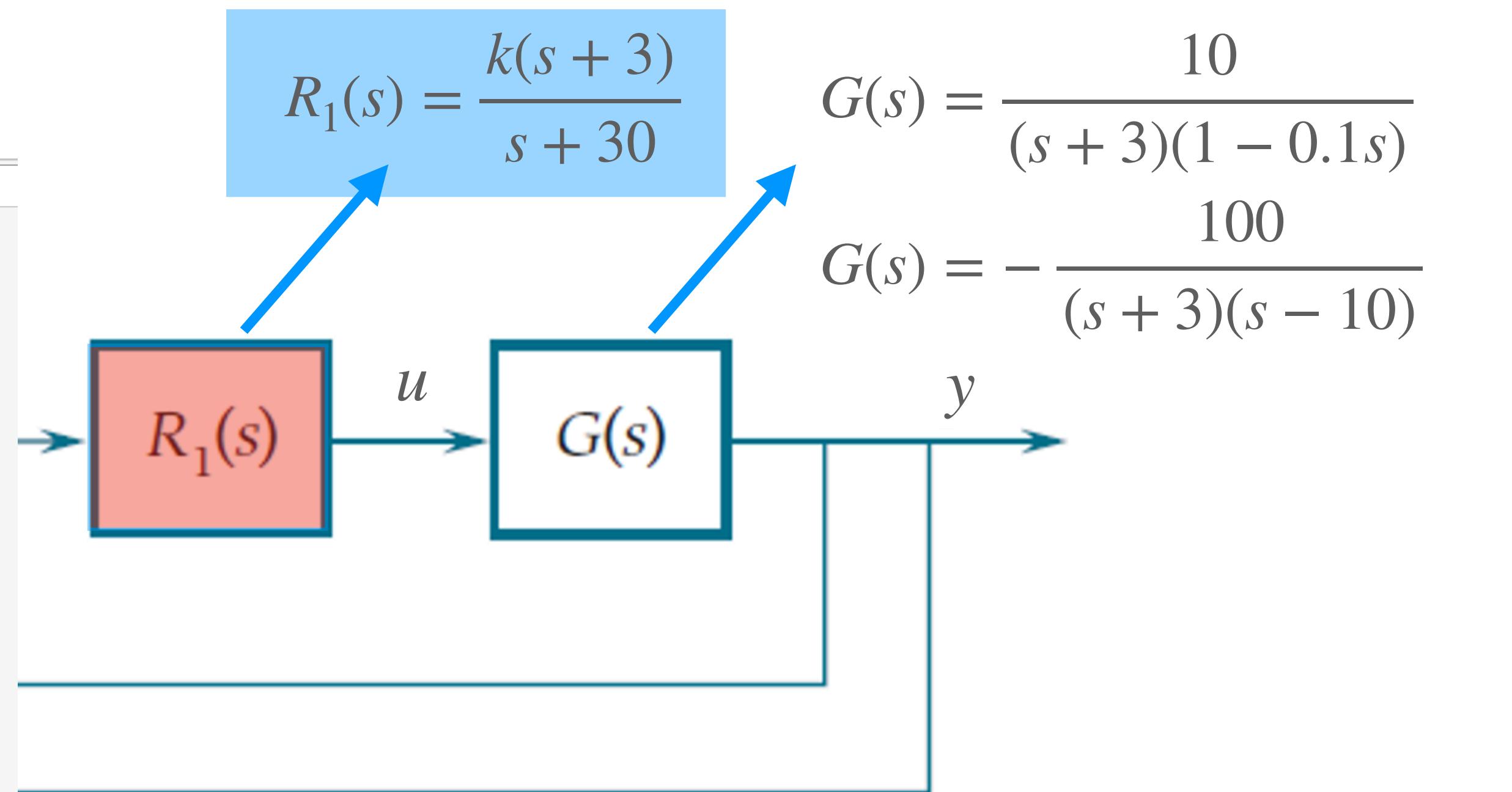
Example:



Command Window

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```
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L1 = 1 / ((s + 30)*(s - 10)); % This is without the gain K
rlocus(L1)
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title('Root Locus of L1(s) = K / ((s + 30)(s - 10))')
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```



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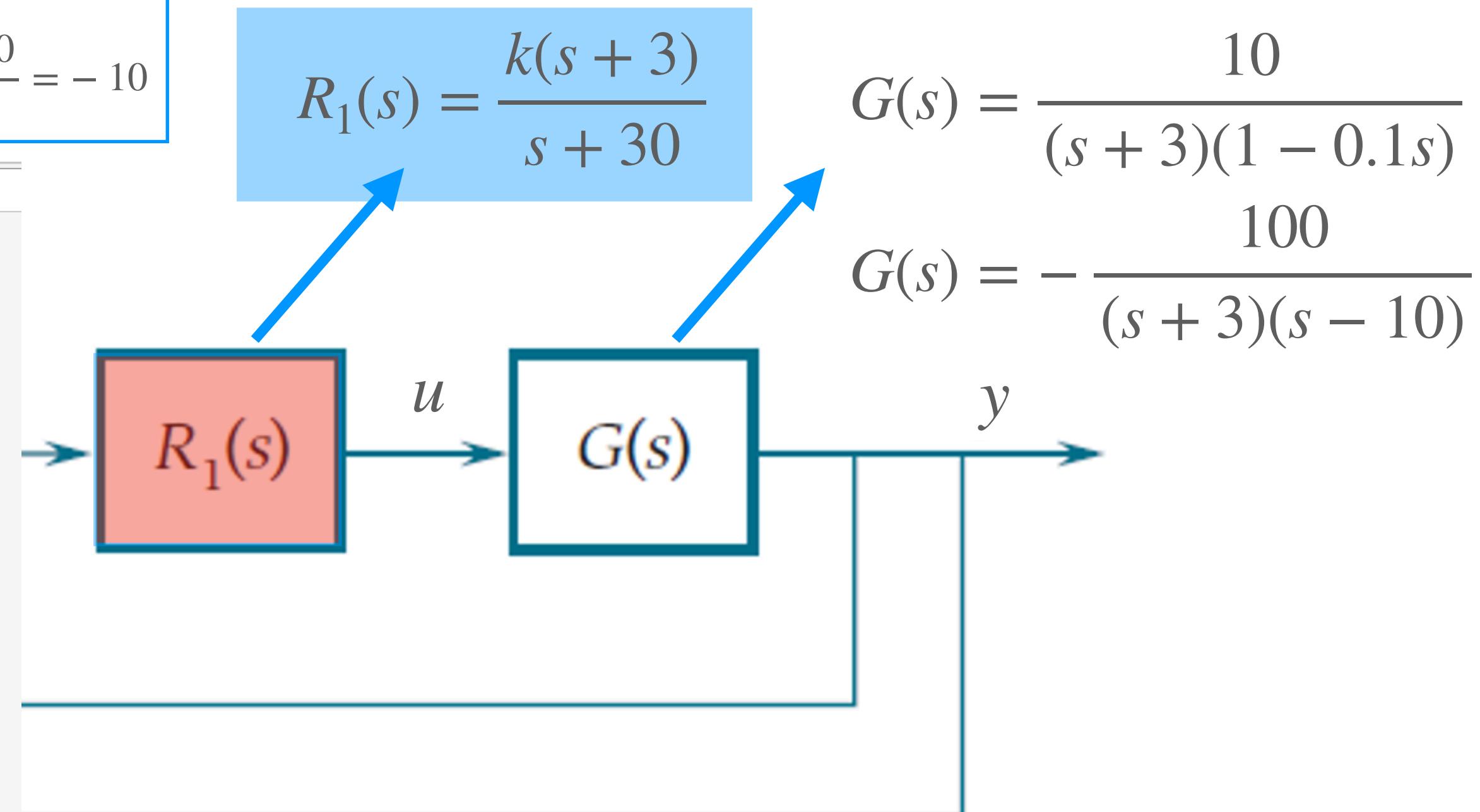
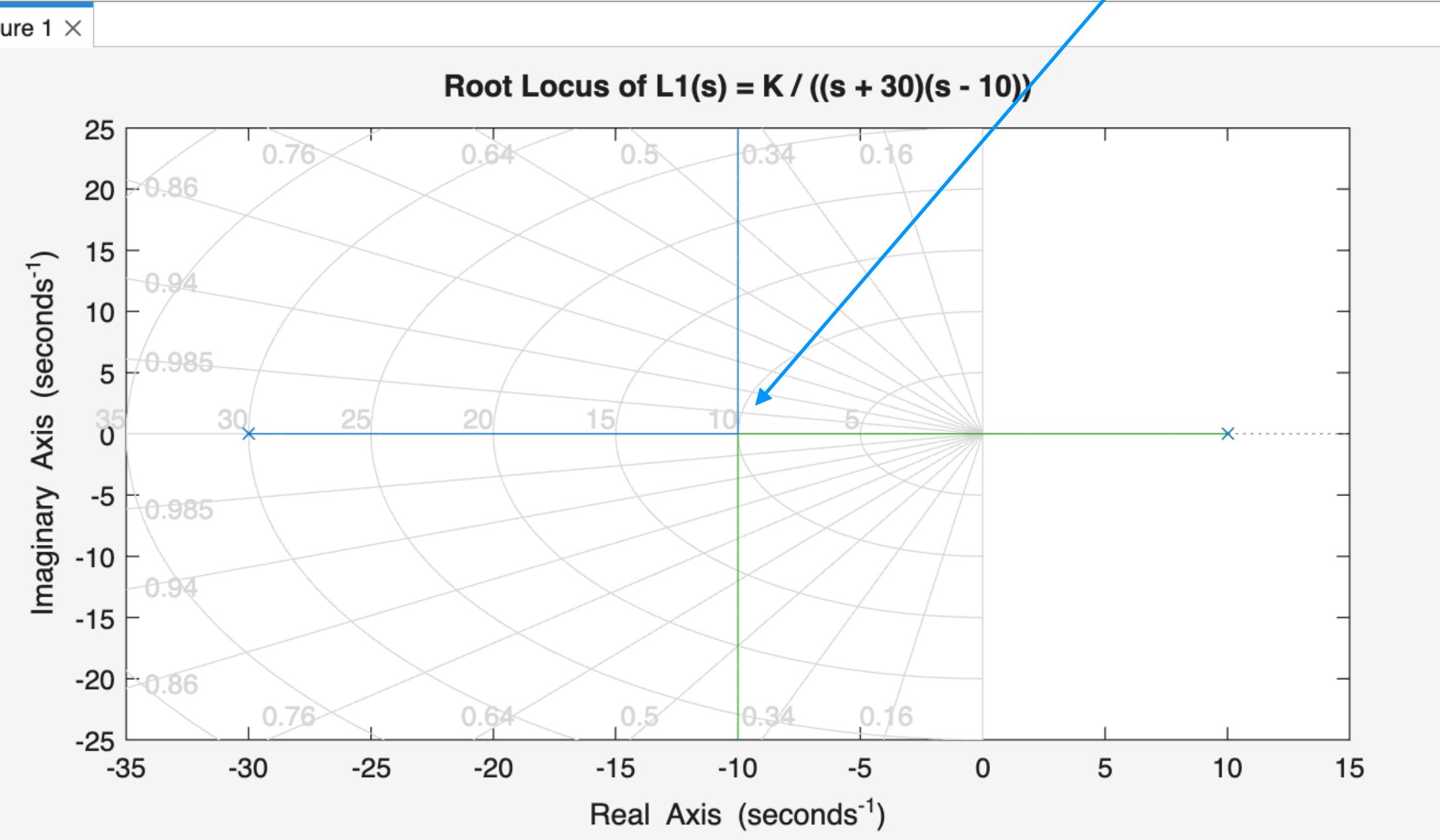


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Control of Open-loop Unstable Systems

Example:

$$\sigma_a = \frac{\sum \text{poli} - \sum \text{zeri}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



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Command Window
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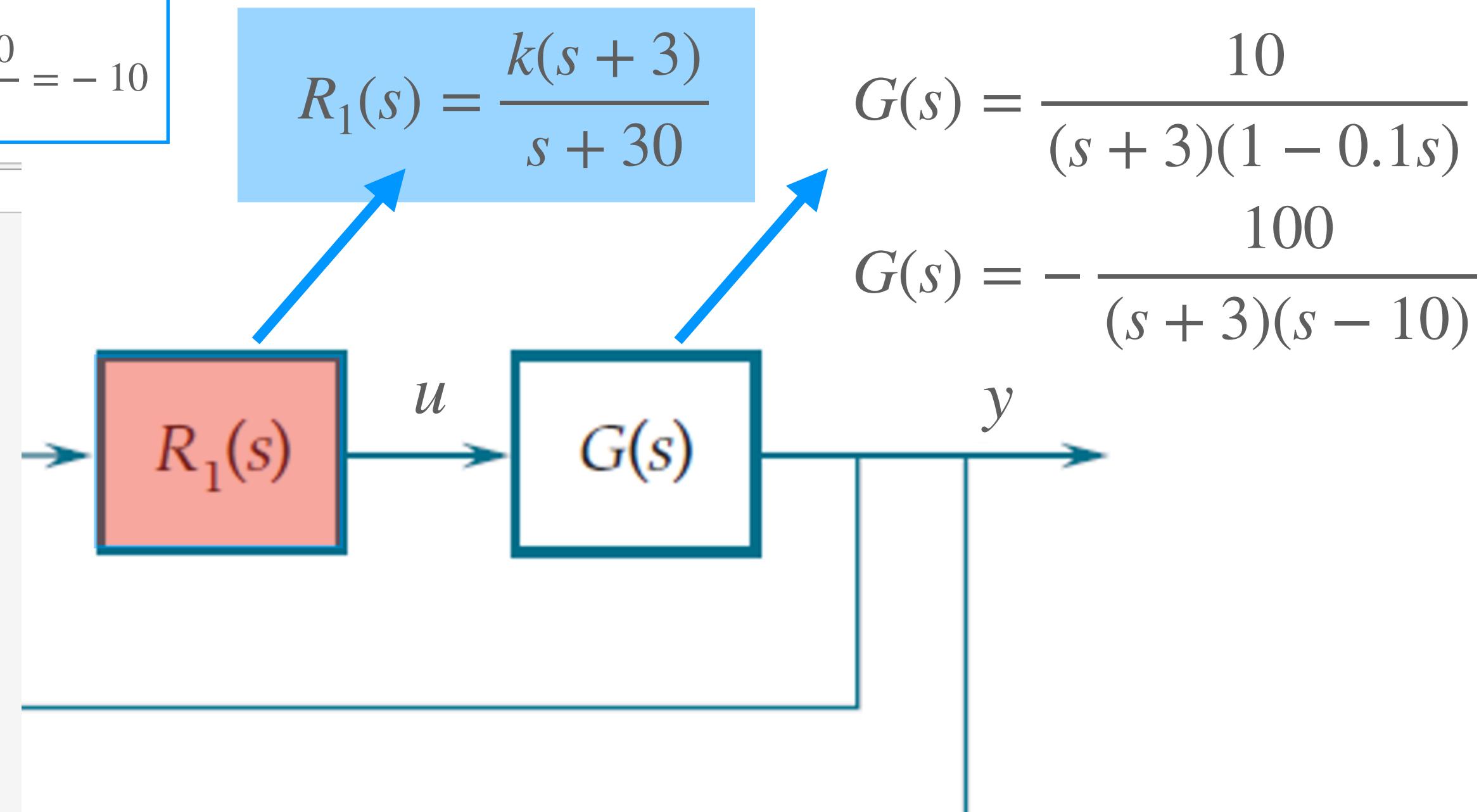
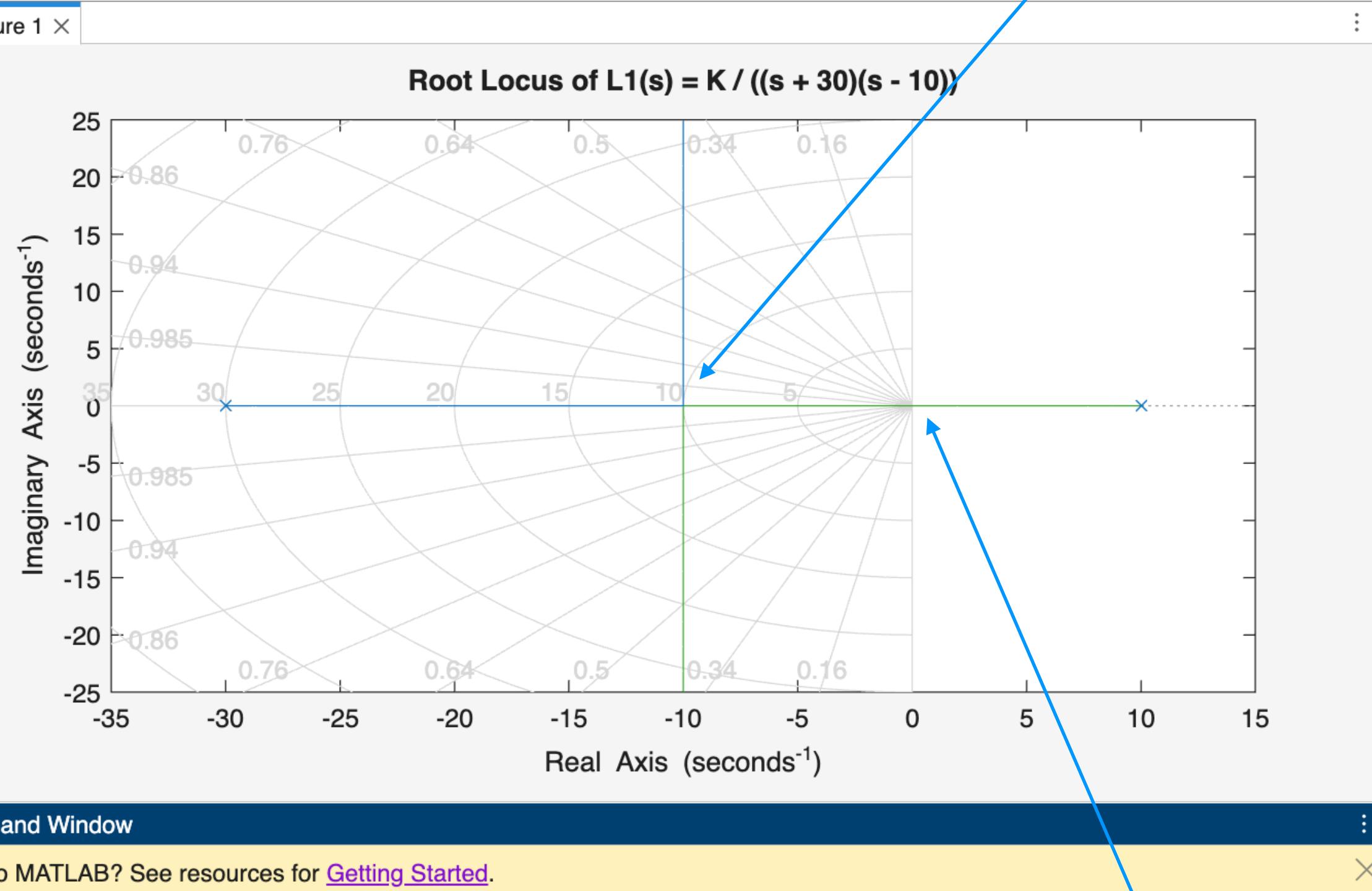
```
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L1 = 1 / ((s + 30)*(s - 10)); % This is without the gain K
rlocus(L1)
grid on
title('Root Locus of L1(s) = K / ((s + 30)(s - 10))')
>> |
```



Control of Open-loop Unstable Systems

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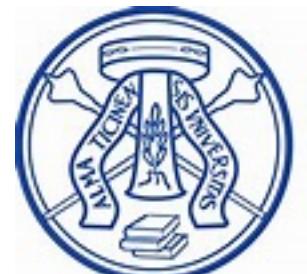


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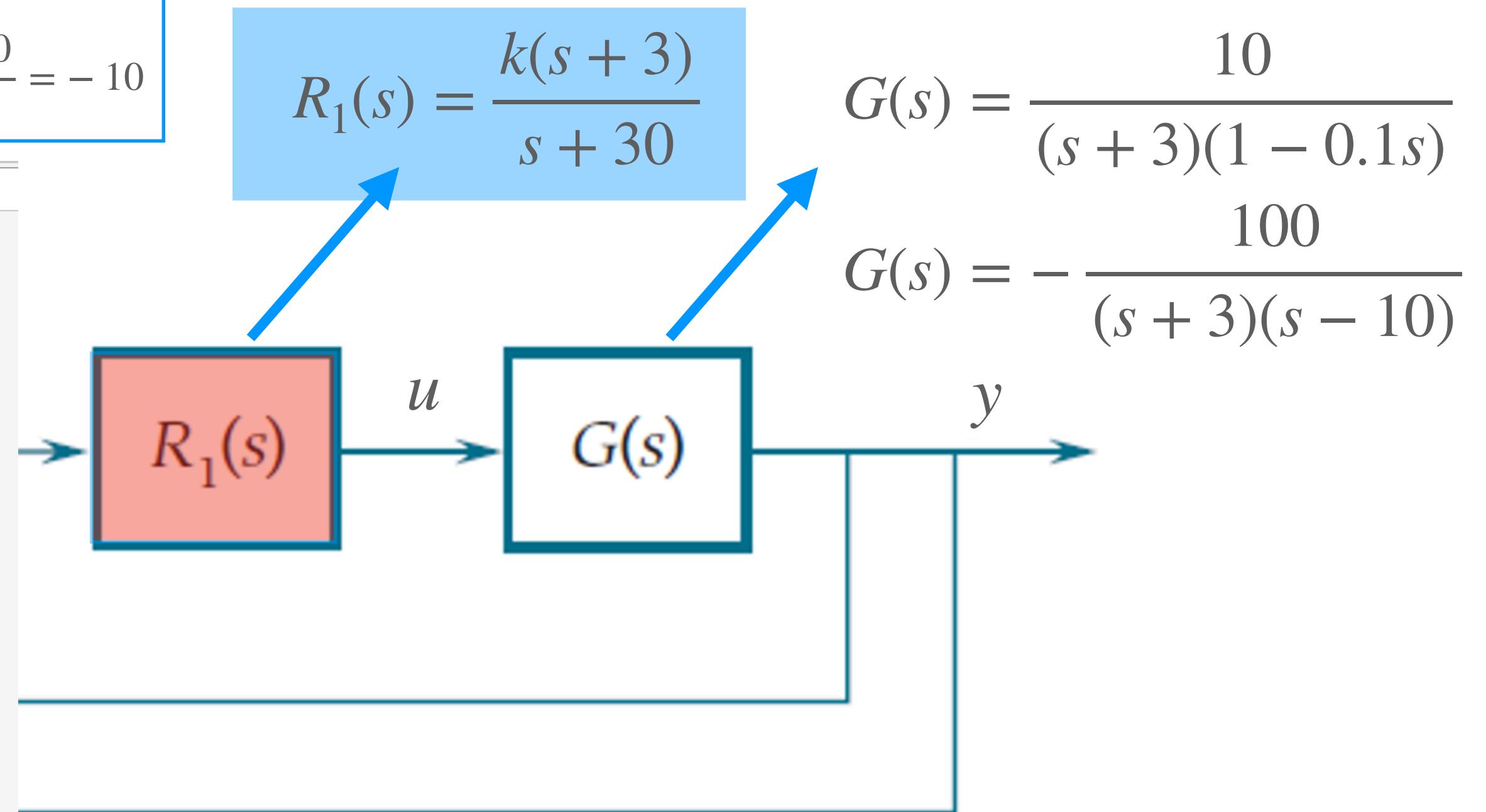
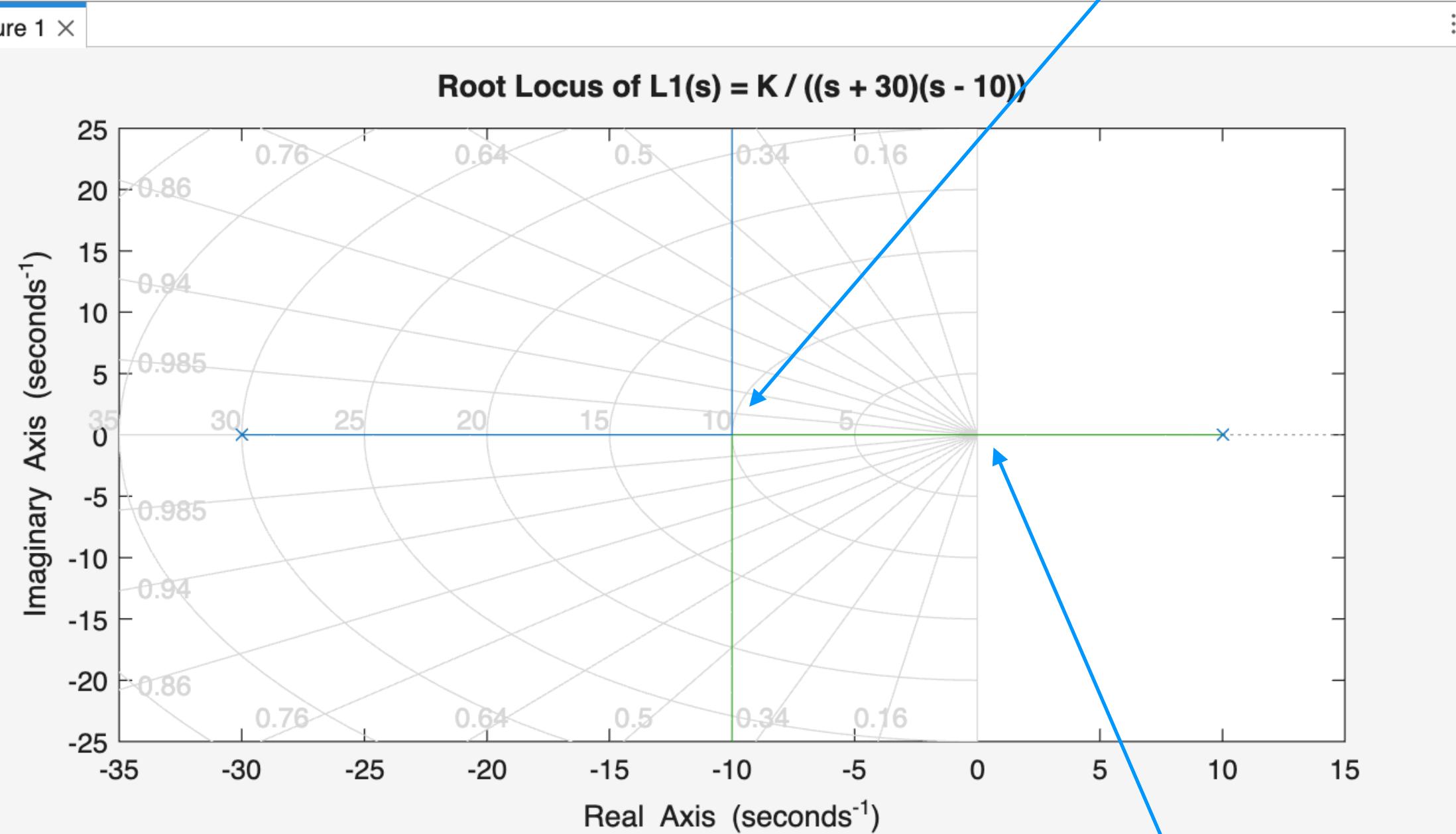
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Control of Open-loop Unstable Systems

Example:

$$\sigma_a = \frac{\sum \text{poli} - \sum \text{zeri}}{n - m}$$

$$\sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



Command Window

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```
desired_pole = 0;
K_at_origin = rlocfind(L1, desired_pole);
K_at_origin
|
K_at_origin =
300
```

Command Window

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```
>> s = tf('s');
L1 = 1 / ((s + 30)*(s - 10)); % This is without the gain K
rlocus(L1)
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$K^* ?$

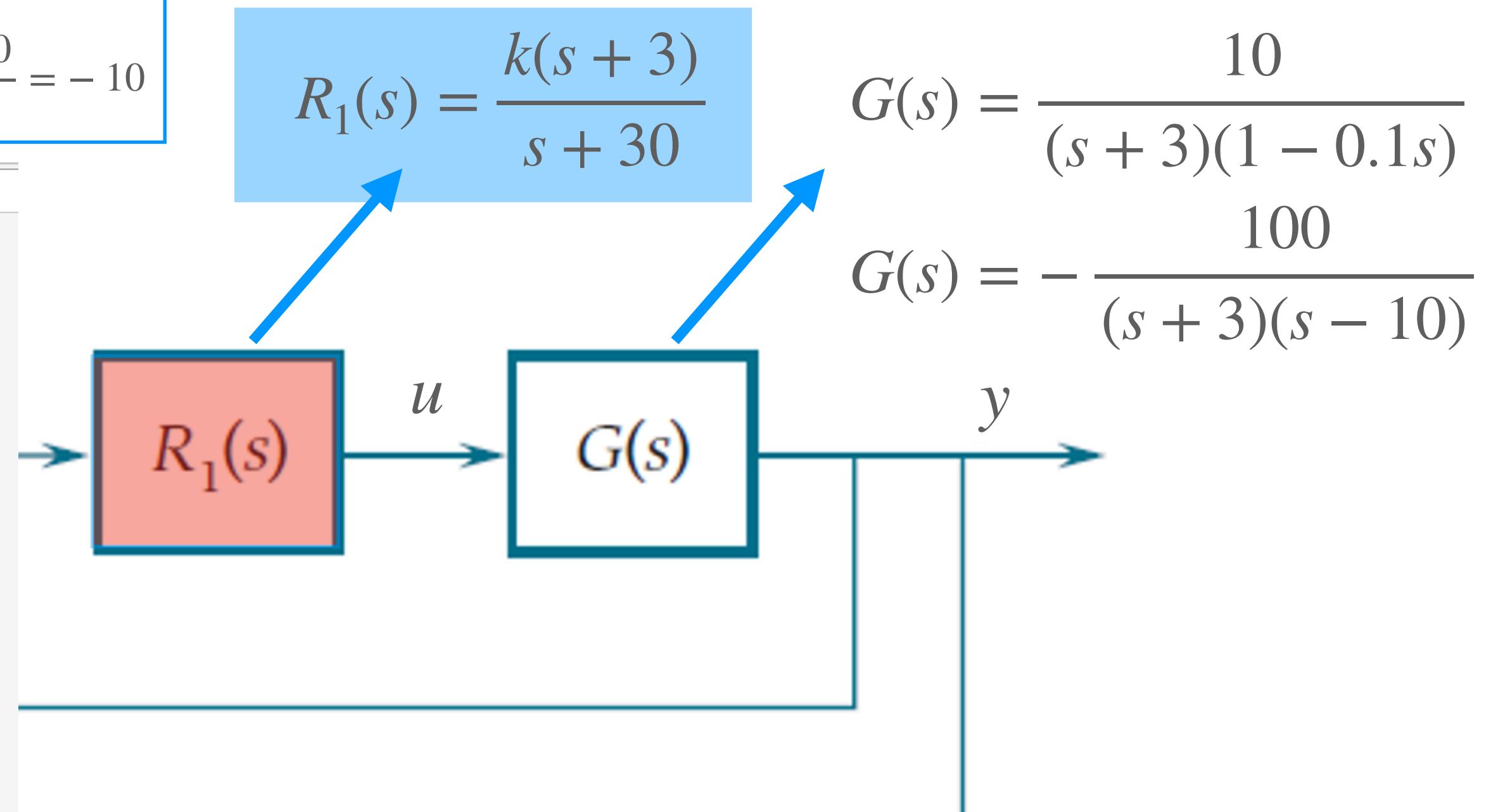
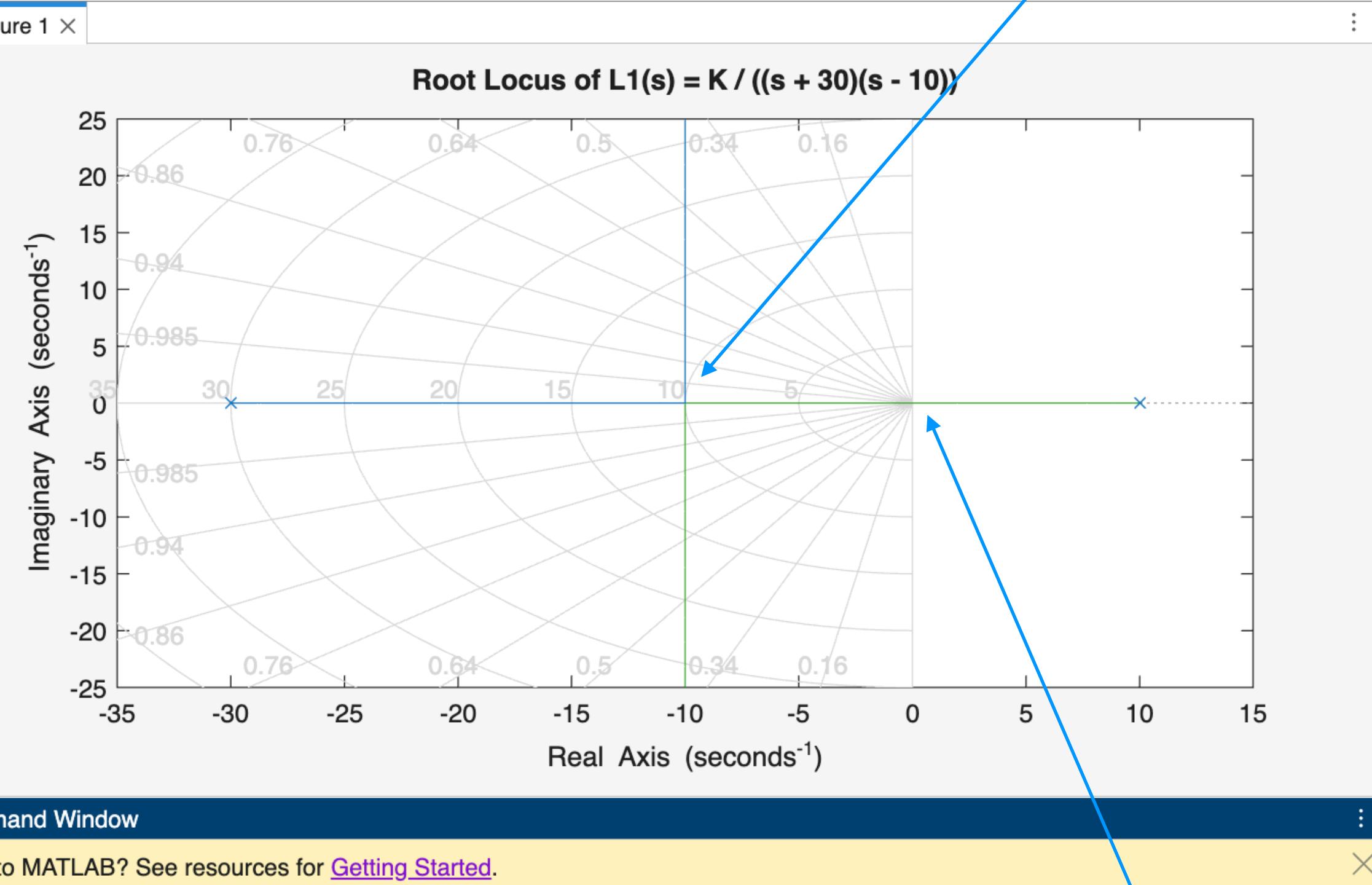


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Control of Open-loop Unstable Systems

Example:

$$\sigma_a = \frac{\sum \text{poli} - \sum \text{zeri}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right|$$

K^* ?

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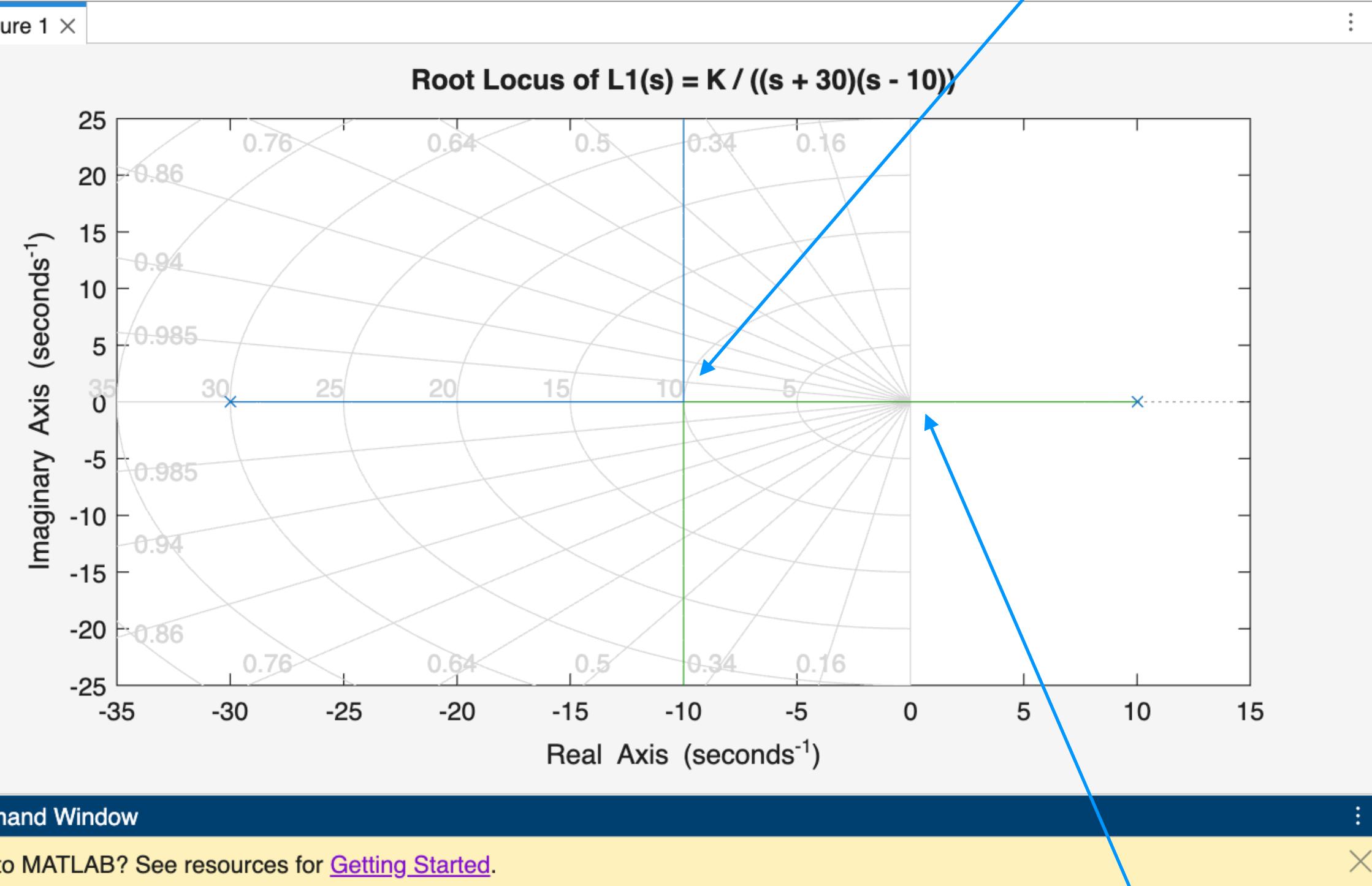


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Control of Open-loop Unstable Systems

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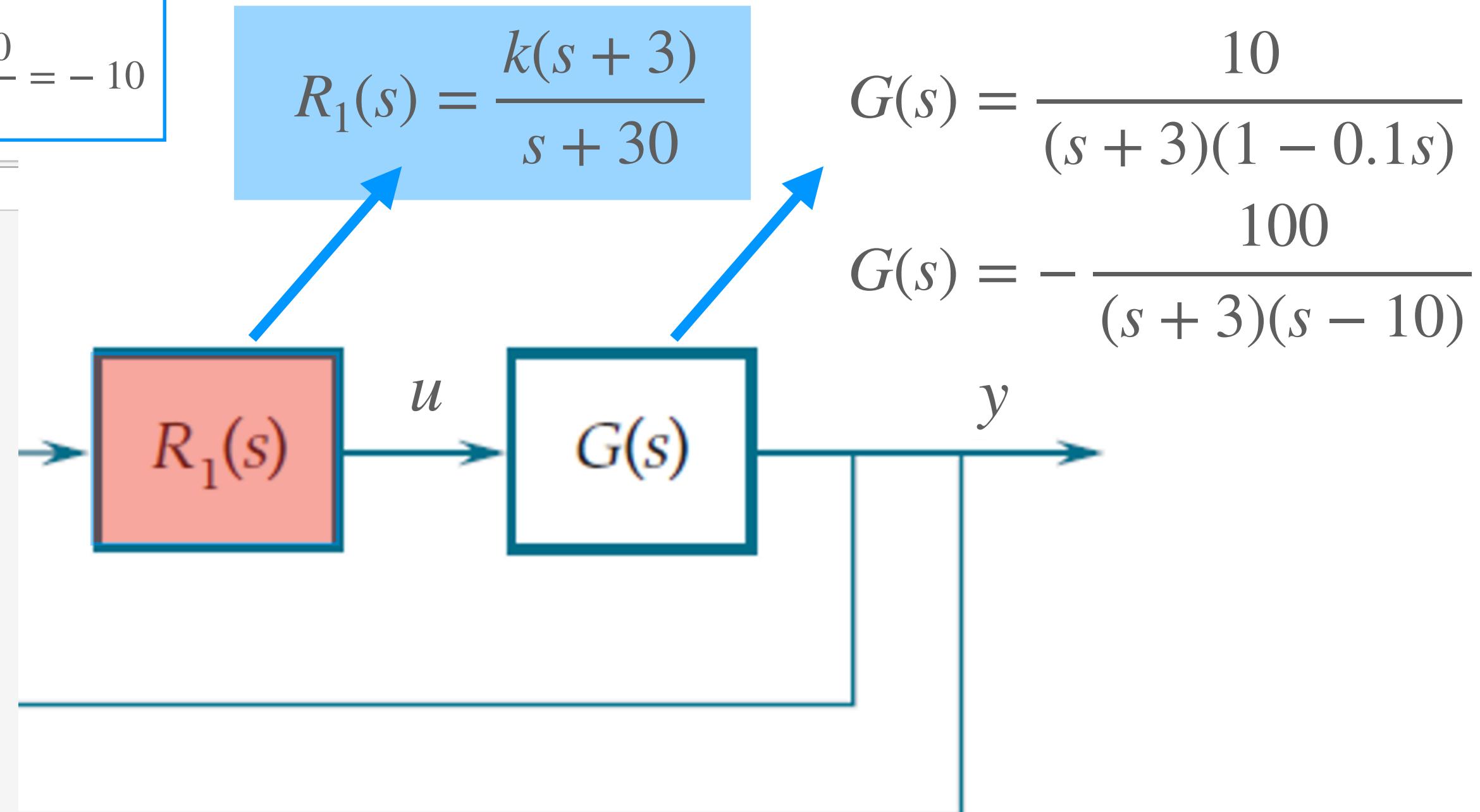


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```



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$$R_1(s) = \frac{k(s + 3)}{s + 30}$$



$$G(s) = \frac{10}{(s + 3)(1 - 0.1s)}$$

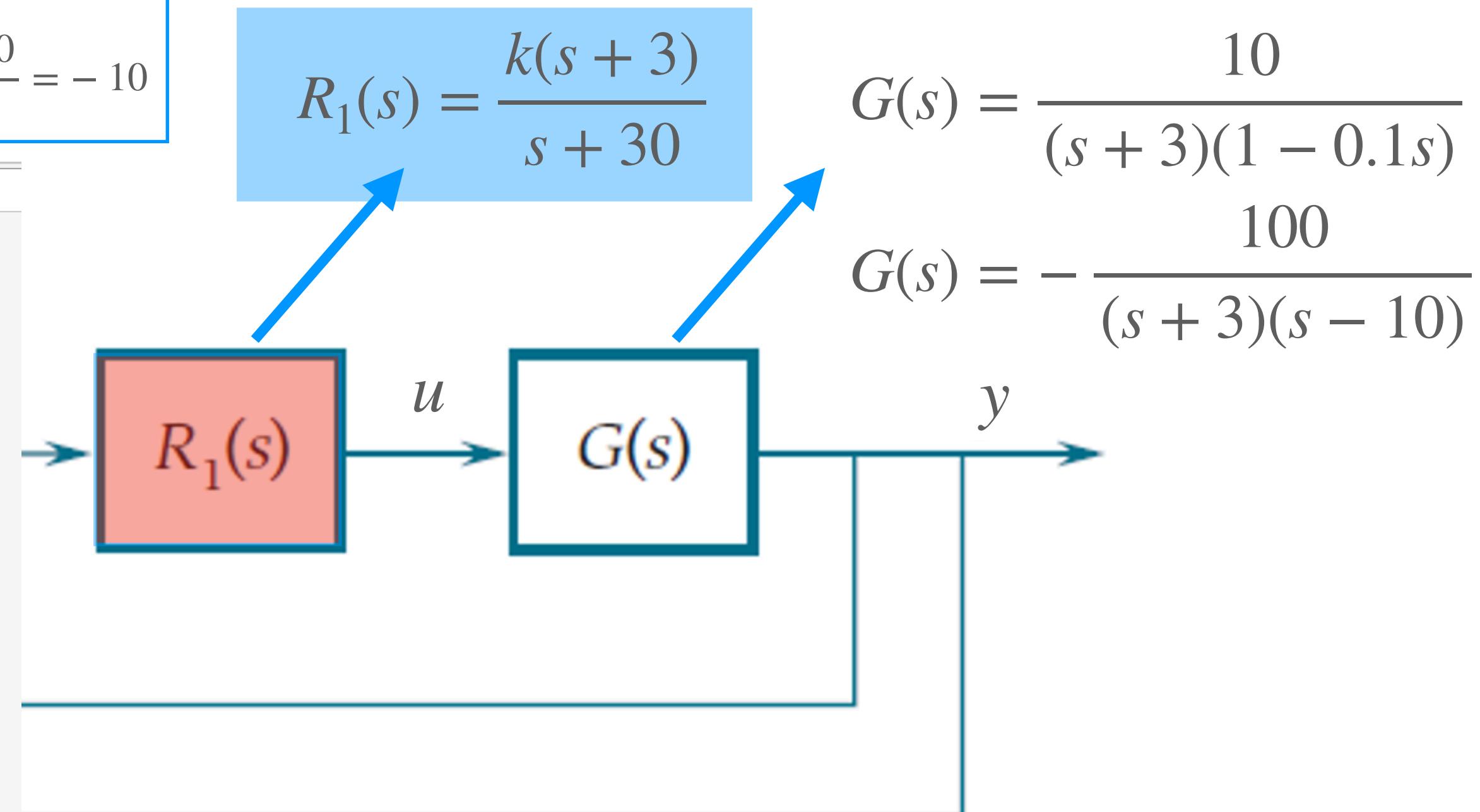
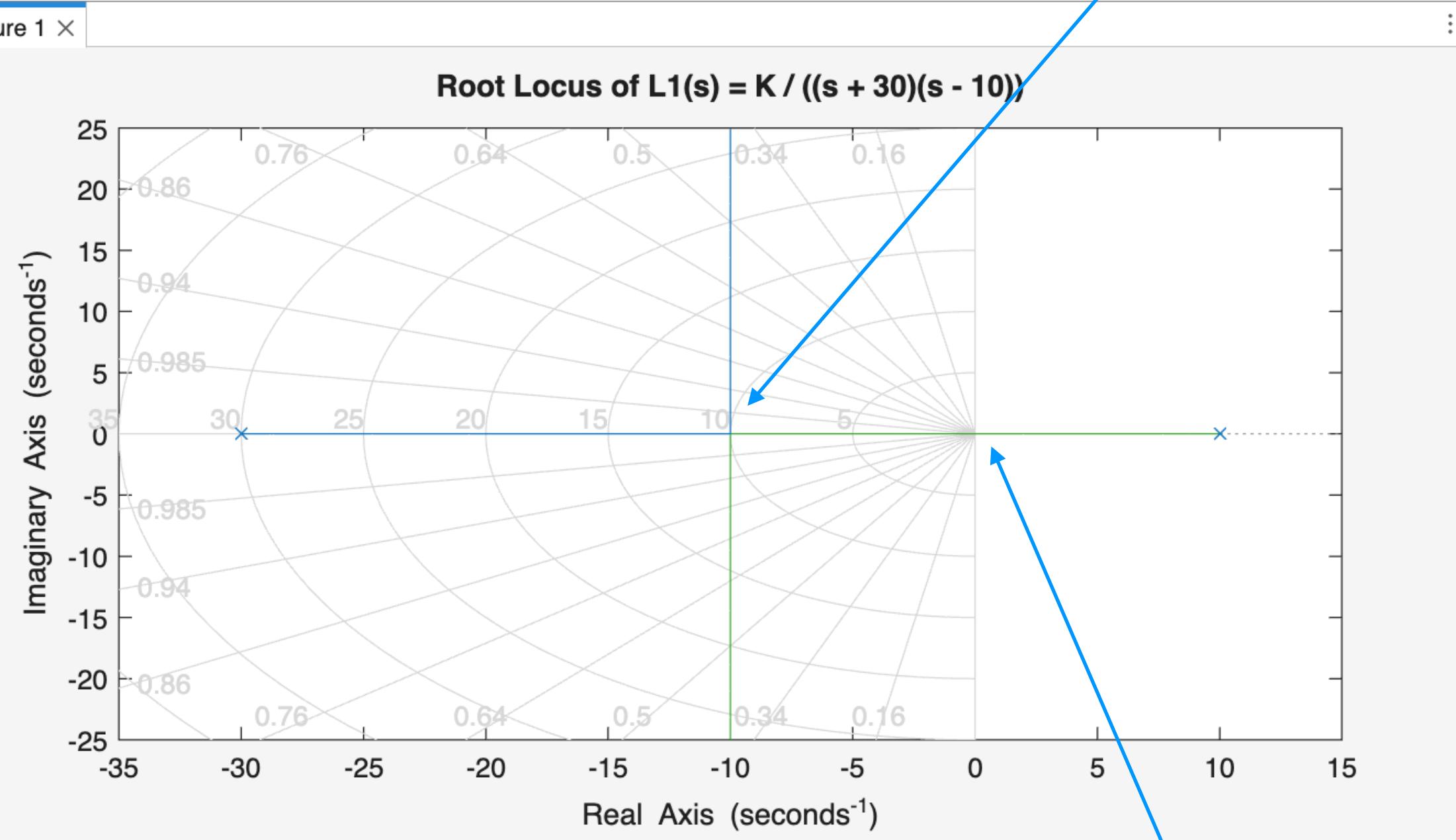
$$G(s) = -\frac{100}{(s + 3)(s - 10)}$$

$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^* + 30)(s^* - 10)}{1} \right|$$

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K^* ? →

$$|K|_{s=0} = \left| \frac{D(0)}{N(0)} \right| = \left| \frac{(0 + 30)(0 - 10)}{1} \right|$$

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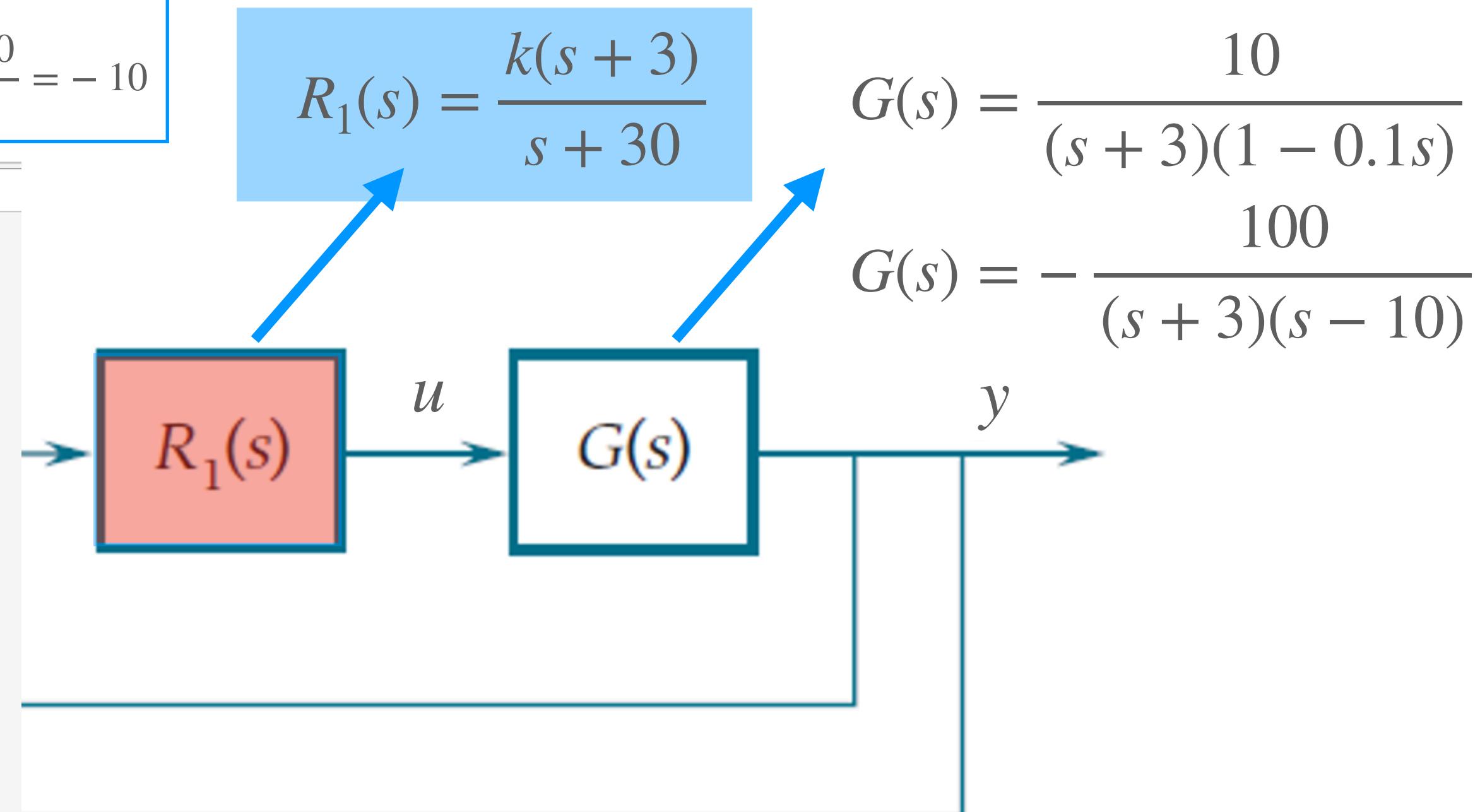
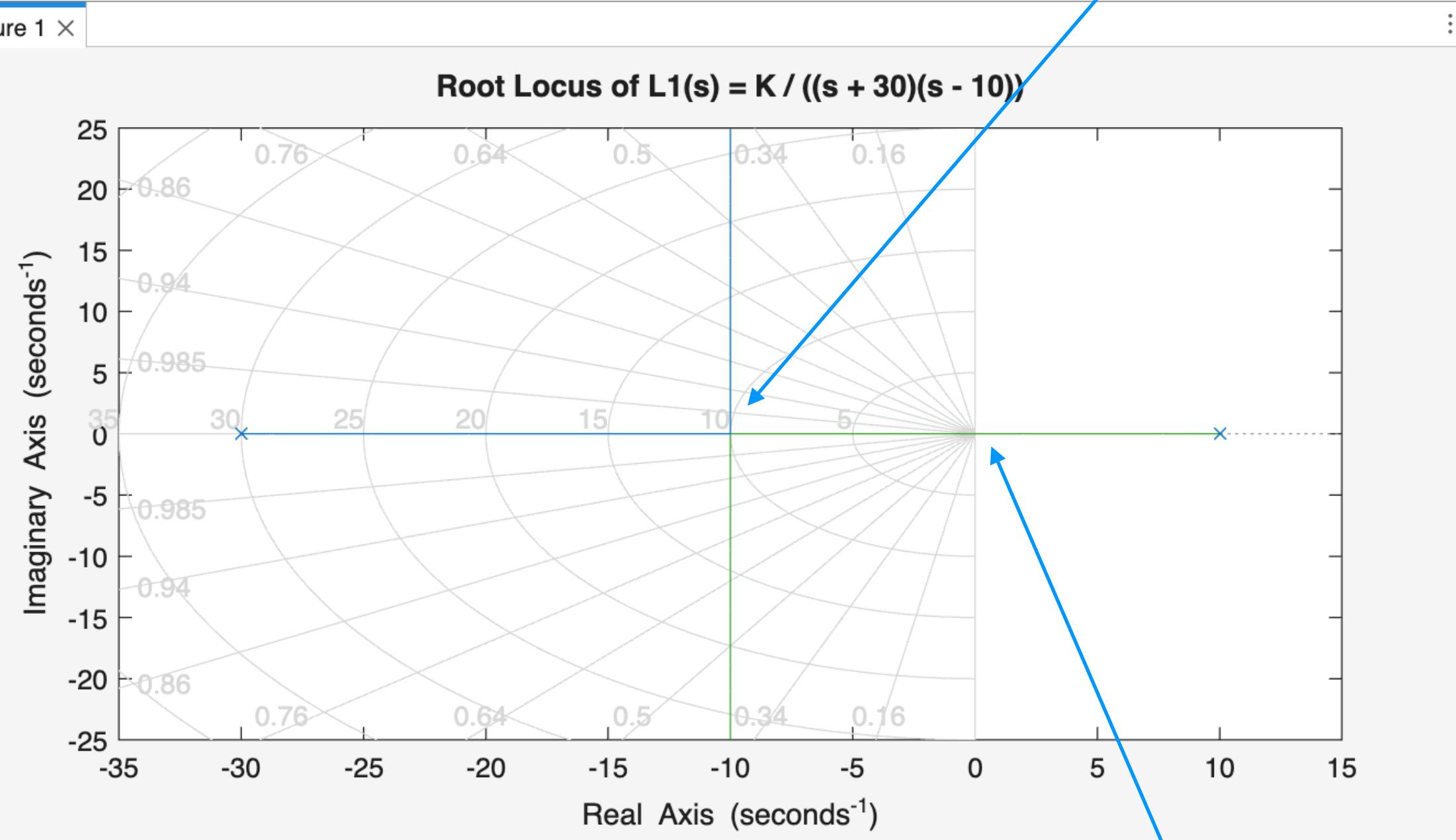
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Control of Open-loop Unstable Systems

Example:

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$$\sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^* + 30)(s^* - 10)}{1} \right|$$

K^* ? →

$$|K|_{s=0} = \left| \frac{D(0)}{N(0)} \right| = \left| \frac{(0 + 30)(0 - 10)}{1} \right| = 300$$

Command Window

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grid on
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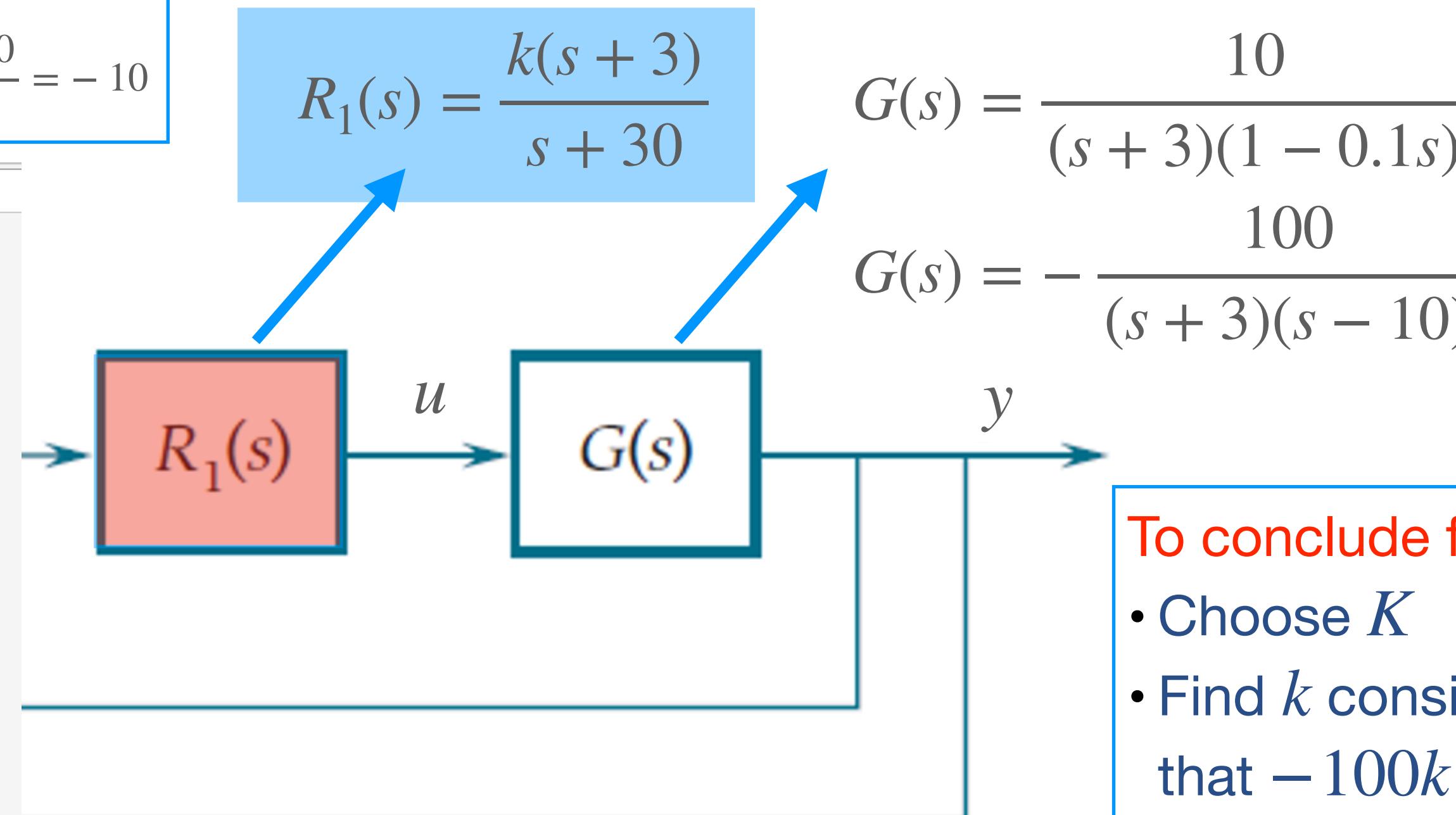
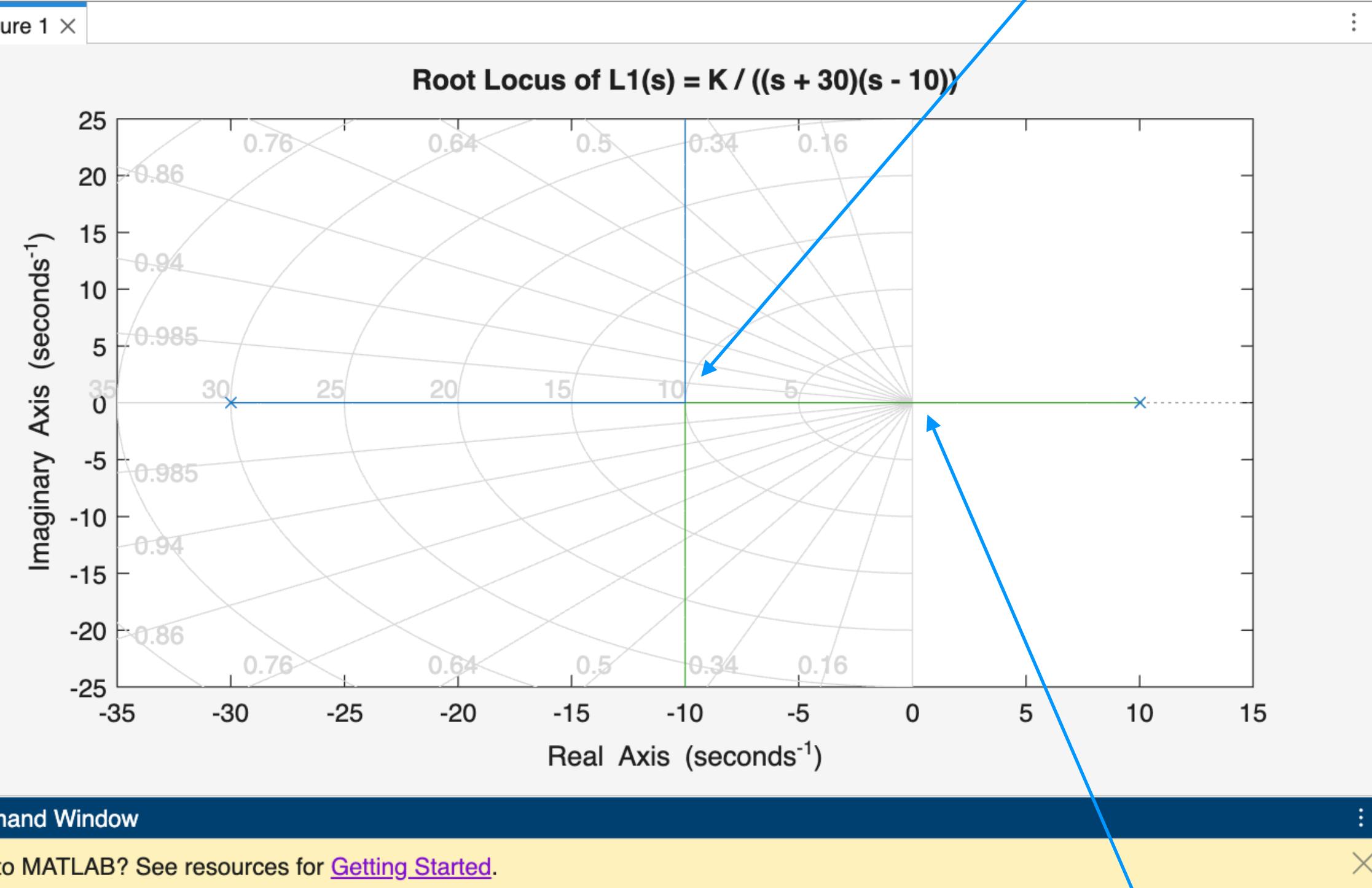


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Control of Open-loop Unstable Systems

Example:

$$\sigma_a = \frac{\sum \text{poli} - \sum \text{zeri}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



To conclude for $R_1(s)$:

- Choose K
- Find k considering that $-100k = K$

$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^* + 30)(s^* - 10)}{1} \right|$$

K^* ? →

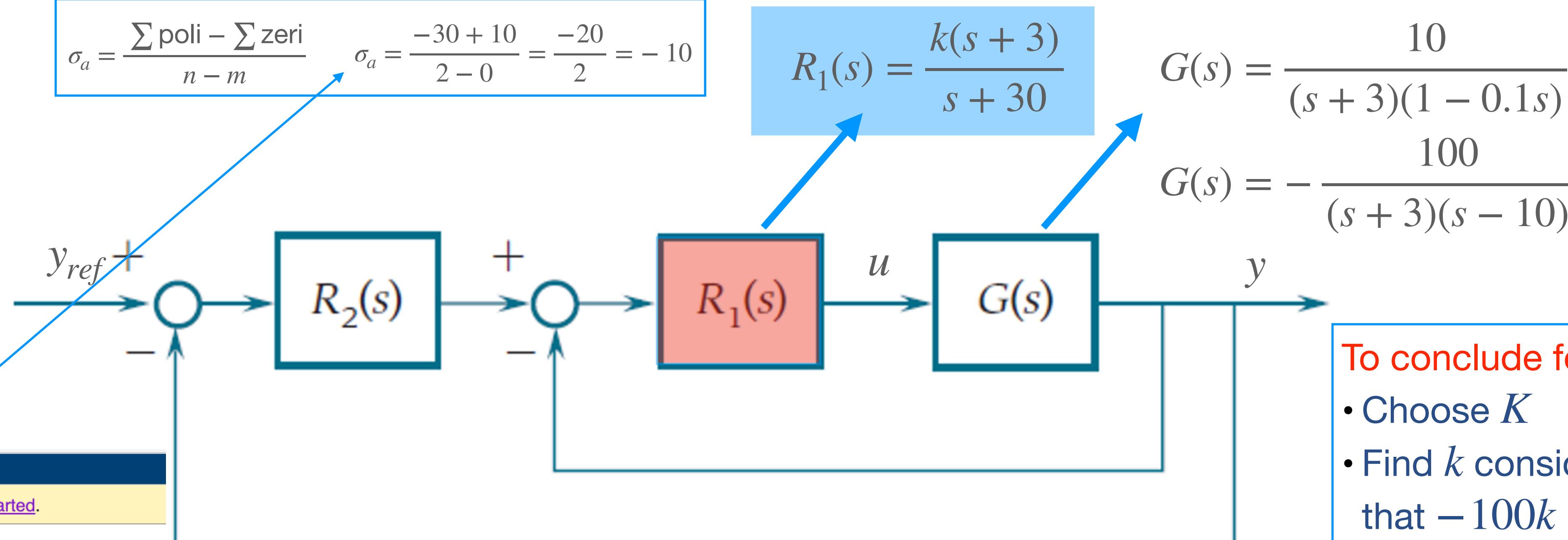
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Control of Open-loop Unstable Systems

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$$\sigma_a = \frac{\sum \text{poli} - \sum \text{zeri}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



My choice:

Command Window

```
New to MATLAB? See resources for Getting Started.
>> desired_pole = -10;
K_at_sigma_a = rlocfind(L1, desired_pole);
K_at_sigma_a

K_at_sigma_a =
    400
```

$$R_1(s) = \frac{k(s + 3)}{s + 30}$$

$$G(s) = \frac{10}{(s + 3)(1 - 0.1s)}$$

$$G(s) = -\frac{100}{(s + 3)(s - 10)}$$

To conclude for $R_1(s)$:

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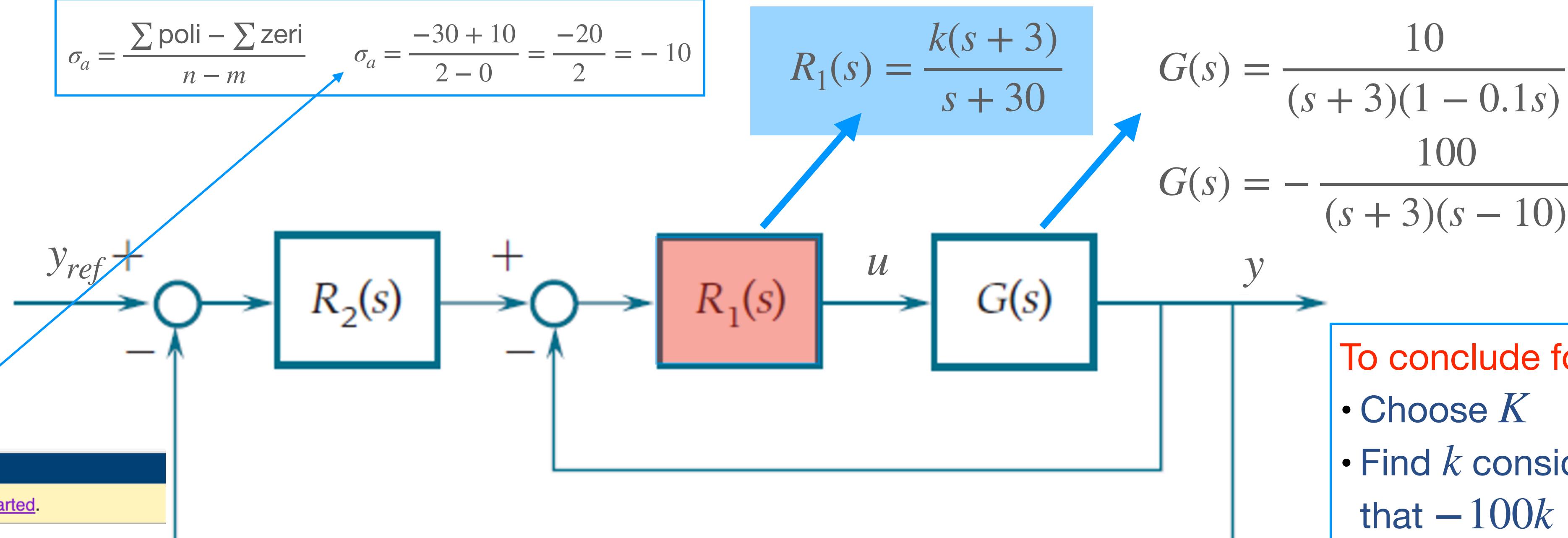
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$$\sigma_a = \frac{\sum \text{poli} - \sum \text{zeri}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



My choice:

Command Window

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>> desired_pole = -10;
K_at_sigma_a = rlocfind(L1, desired_pole);
K_at_sigma_a
```

K_at_sigma_a =

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$$-100k = K$$

$$k = -\frac{K}{100}$$

$$R_1(s) = \frac{k(s+3)}{s+30}$$

$$G(s) = \frac{10}{(s+3)(1-0.1s)}$$

$$G(s) = -\frac{100}{(s+3)(s-10)}$$

To conclude for $R_1(s)$:

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K* ? →

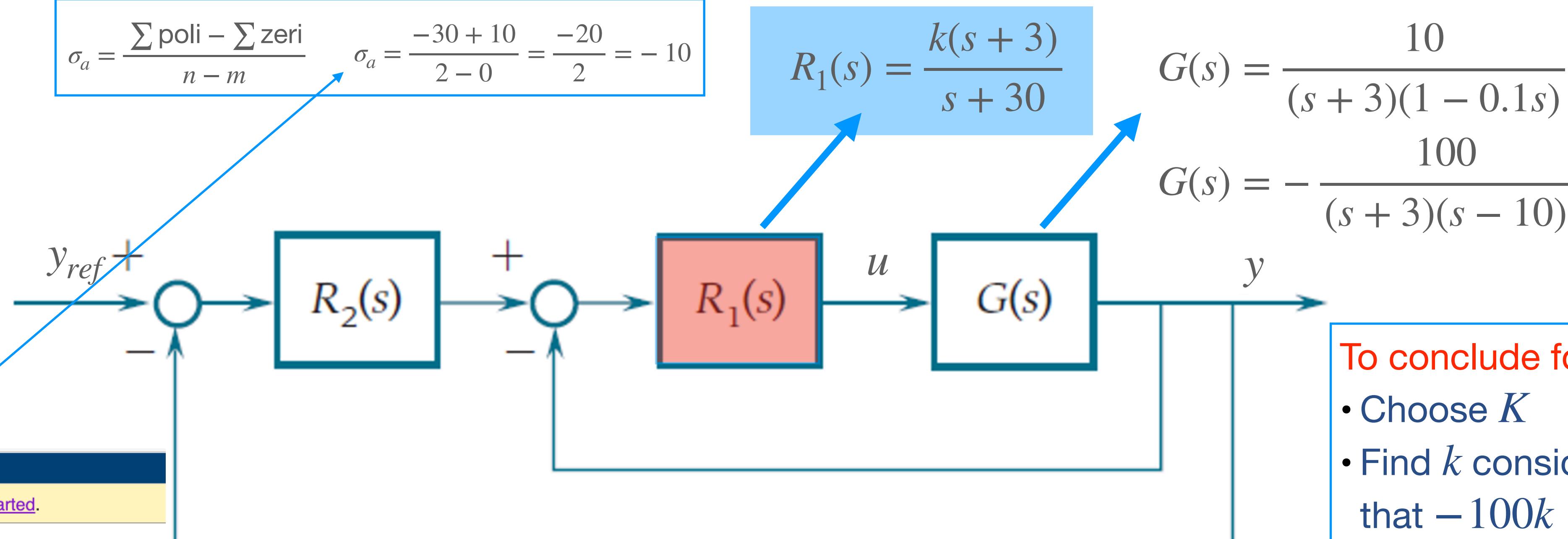
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>> desired_pole = -10;
K_at_sigma_a = rlocfind(L1, desired_pole);
K_at_sigma_a
```

K_at_sigma_a =

400

$$-100k = K$$

$$k = -\frac{K}{100}$$

$$R_1(s) = -\frac{4(s+3)}{s+30}$$

$$R_1(s) = \frac{k(s+3)}{s+30}$$

$$G(s) = \frac{10}{(s+3)(1-0.1s)}$$

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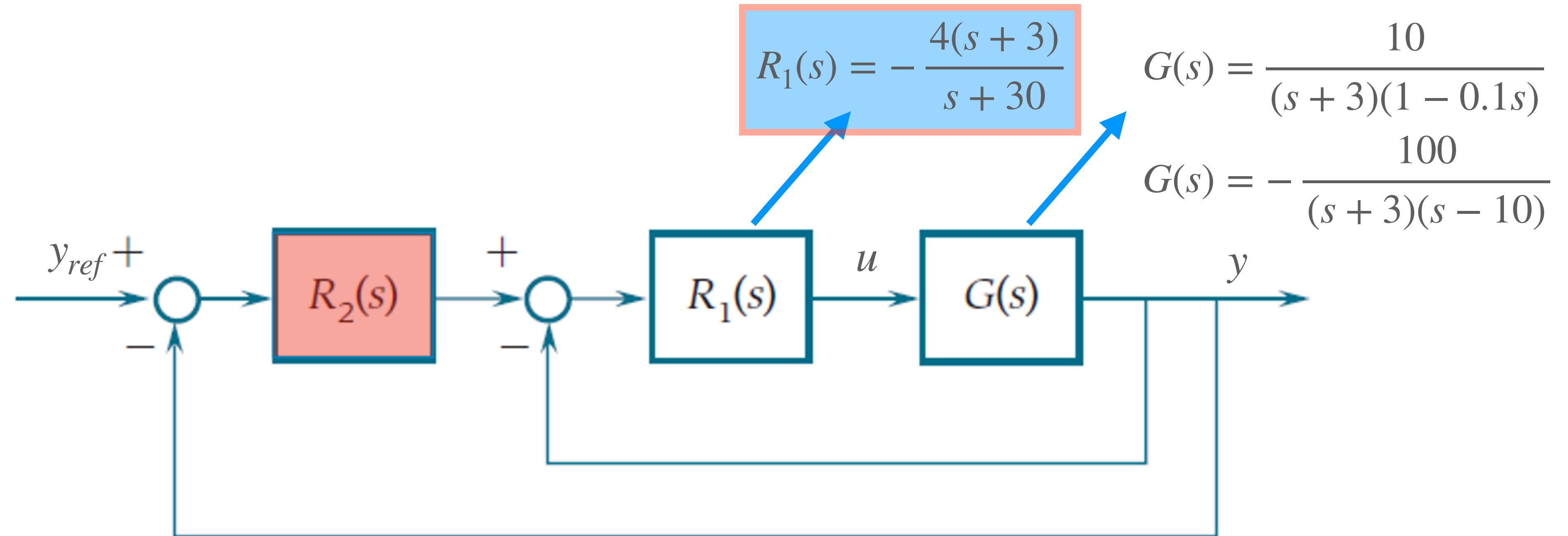
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$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^*+30)(s^*-10)}{1} \right|$$

$K^* ? \rightarrow |K|_{s=0} = \left| \frac{D(0)}{N(0)} \right| = \left| \frac{(0+30)(0-10)}{1} \right| = 300$

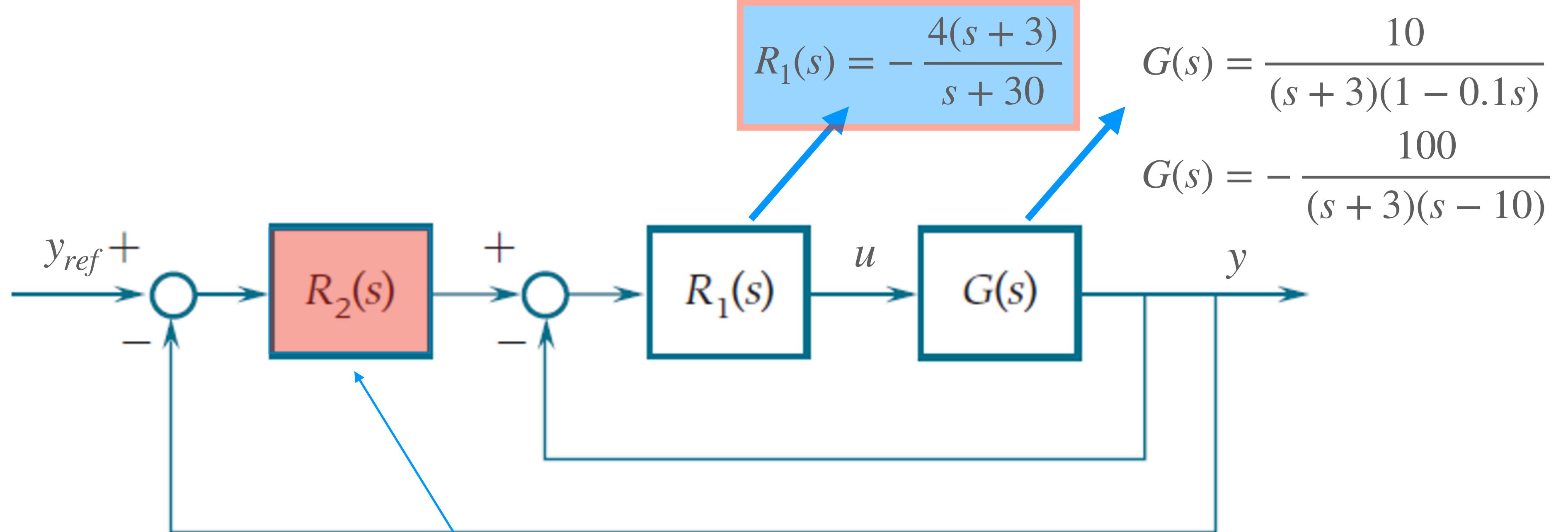
Control of Open-loop Unstable Systems

Example:



Control of Open-loop Unstable Systems

Example:



To conclude for $R_2(s)$:

- Compute $F_1(s)$
- Design $R_2(s)$ based on $F_1(s)$ using standard loop-shaping design guided by the Bode criterion

