

# General Information

**Prof. Antonella Ferrara**

<https://scholar.google.com/citations?user=r5JuMskAAAAJ&hl=en>

**Course Teaching Material:**

KIRO UNIPV <https://elearning.unipv.it/>

- [504462 - PROCESS CONTROL 2025-26 - PROF.SSA FERRARA ANTONELLA](#)

**Lecture Time-table:**

<http://www-3.unipv.it/ingserv/orario2526/1sem/insegnamenti/ProCont.html>

**Exams:**

<https://studentionline.unipv.it/esse3/Home.do>

<https://kirotesting.unipv.it/>



UNIVERSITÀ  
DI PAVIA

# Introduction

- The course and its objectives
- Kiro UNIPV
- Teaching material
- The preparatory course
- The exam
- Office hours
- The Intelligent Robotics Lab
- Theses and research
- The role of Ph.D. Students and PostDocs



# Introduction

- Program of the course:

## **Advanced SISO control schemes:**

Pre-filters and parallel compensators, two degrees of freedom control schemes, compensation of measurable disturbances, systems with delays and Smith Predictor, Padé approximation, decoupling in the frequency domain, control of open loop unstable systems.

## **Advanced MIMO control schemes:**

Decoupling based control schemes, decentralized control, relative gain array.

## **PID controllers:**

Features and properties. Rules for the empirical calibration. Wind-up and anti wind-up schemes.

## **Digital control:**

Discrete-time systems. The concept of equilibrium for discrete-time systems. Stability of linear time-invariant discrete-time systems. Jury test. Digital control schemes. Zeta transform and its properties. Transfer functions in the z domain. Sampling and aliasing. Choice of the sampling time. Zero-order-Hold. Discretization of continuous-time controllers. Bilinear transformation, Euler and Tustin methods.

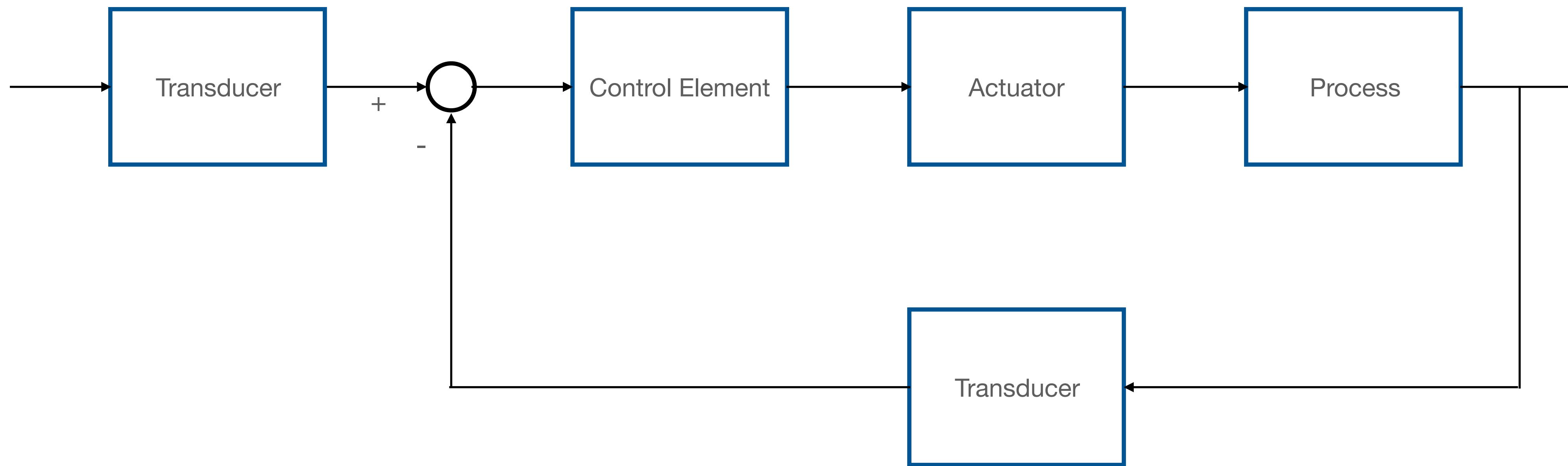


# Introduction

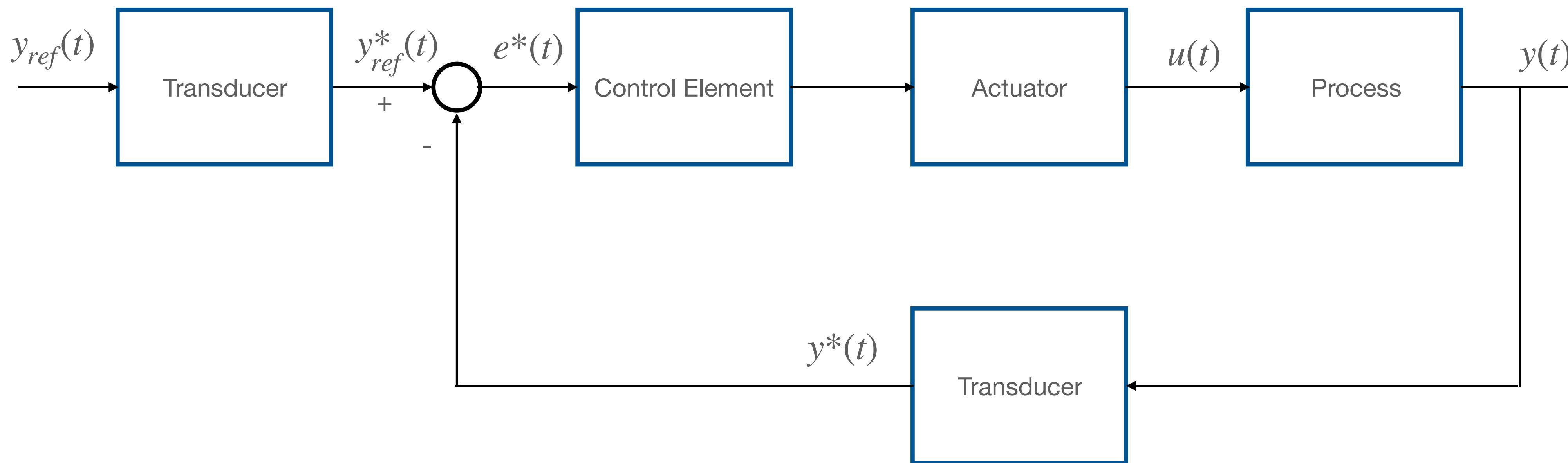
- Some of the figures in these slides, kindly provided by McGraw-Hill, are those of the Textbook:



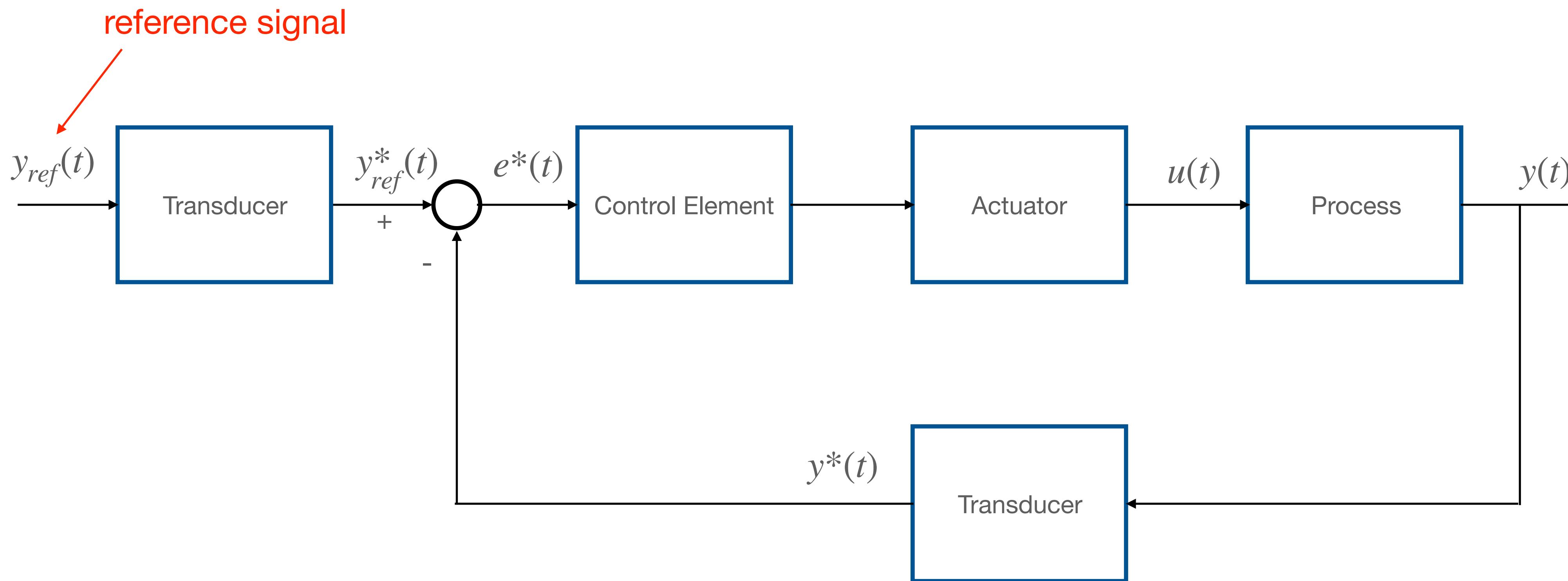
## Basic Control Scheme



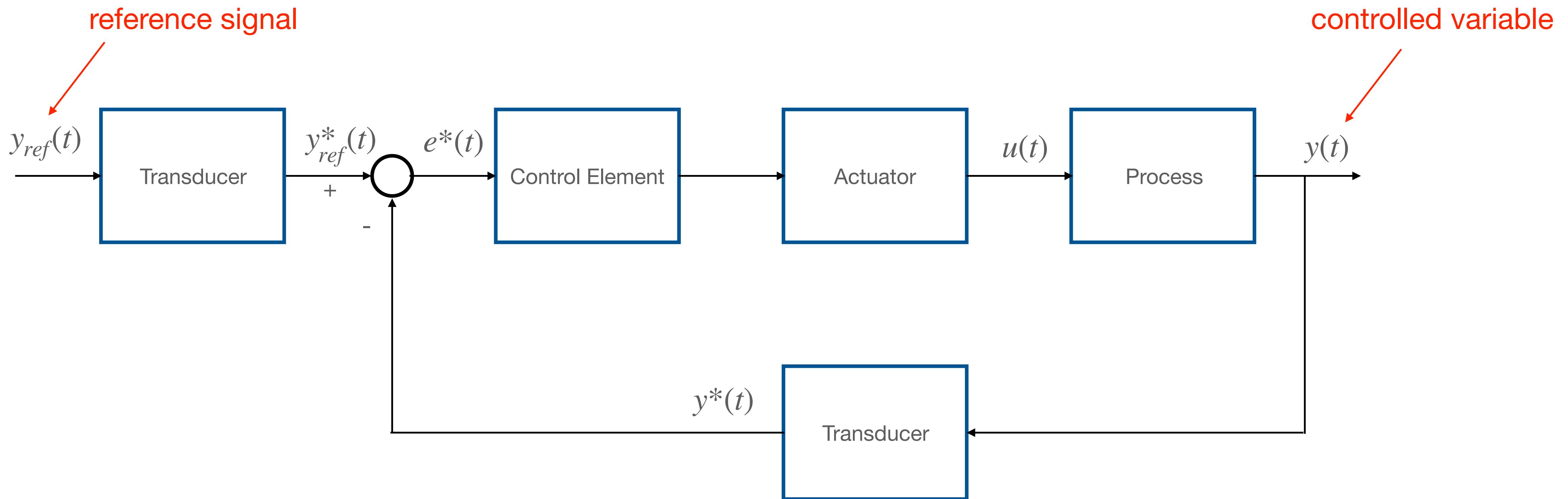
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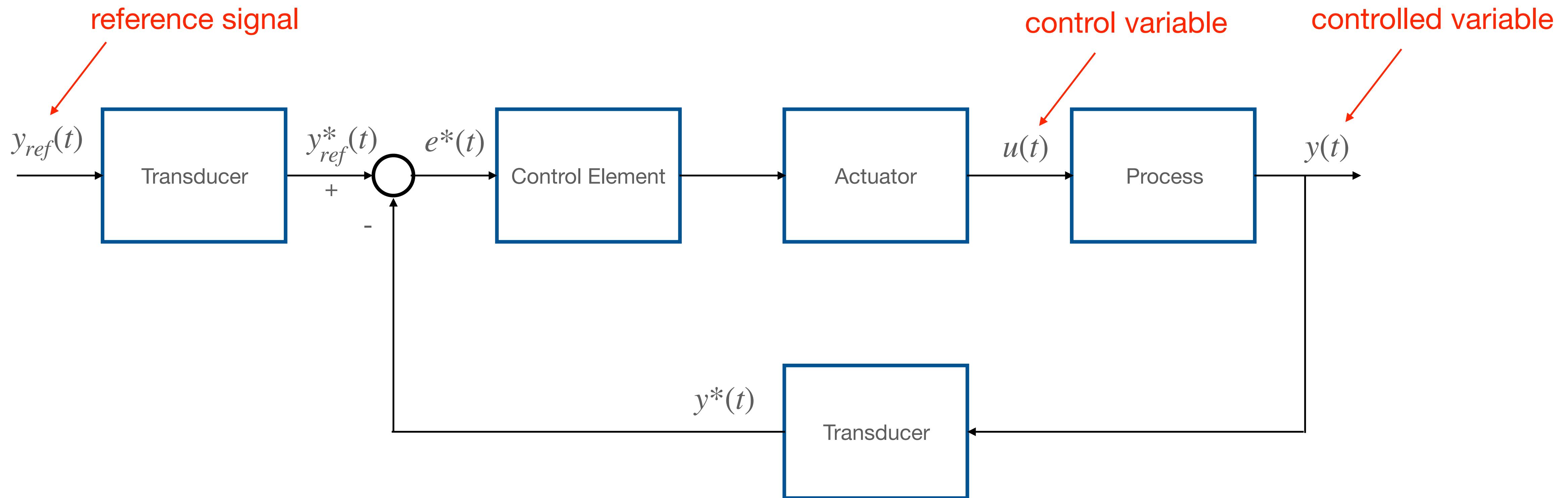
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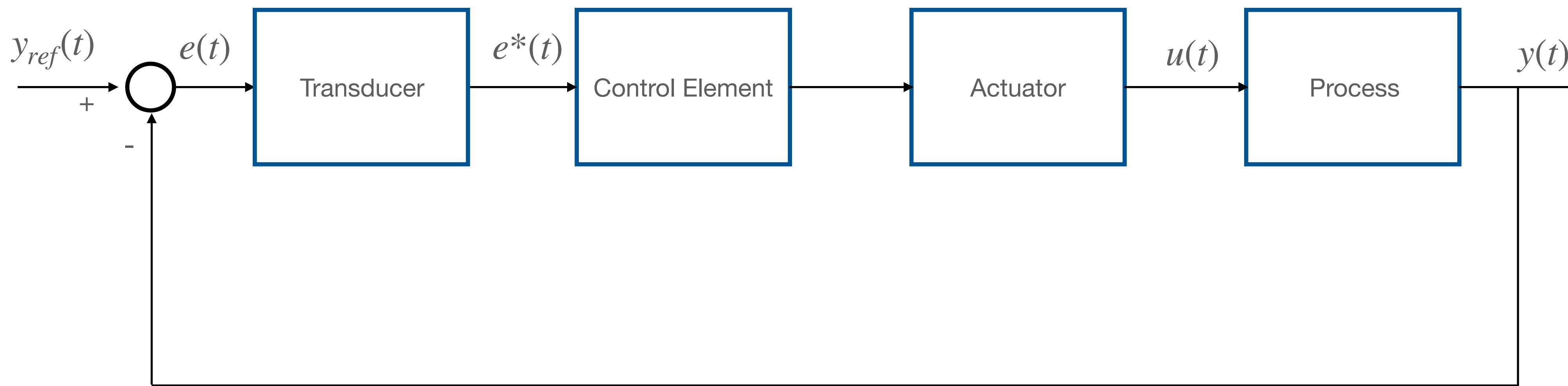
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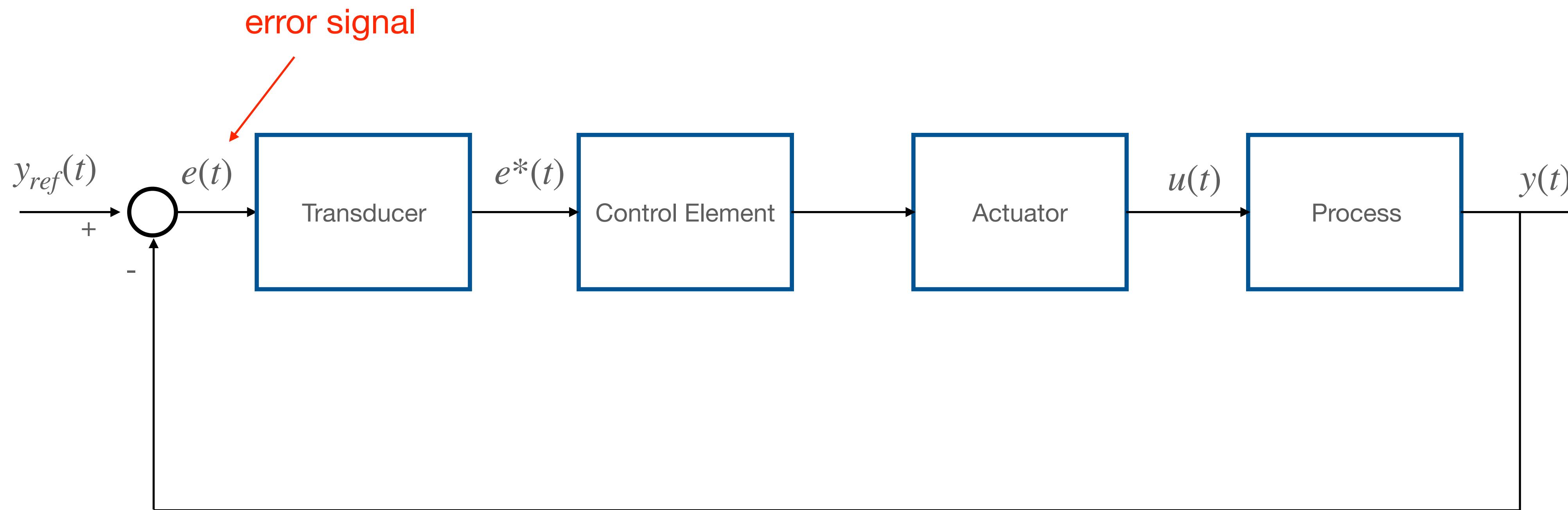
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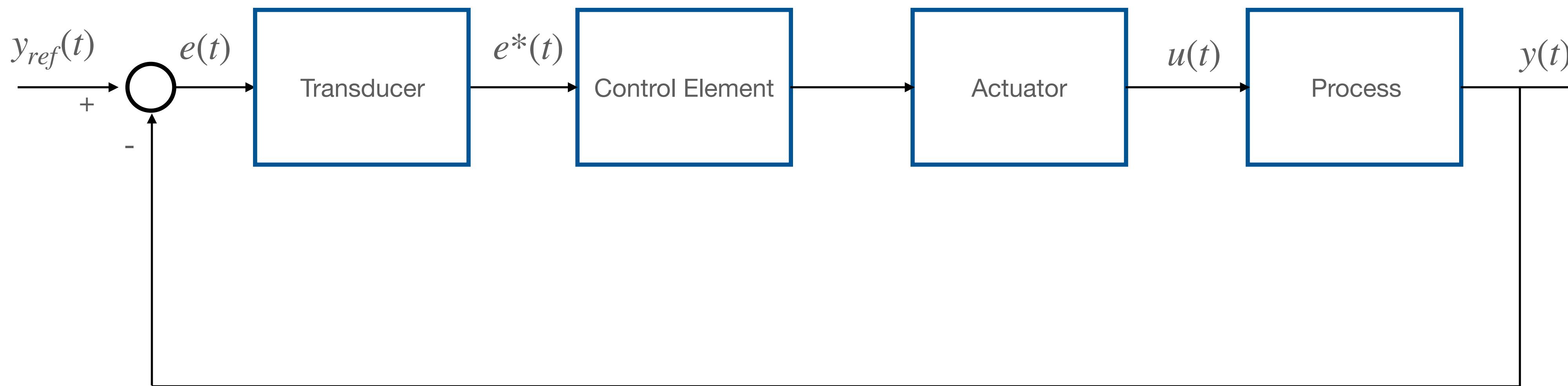
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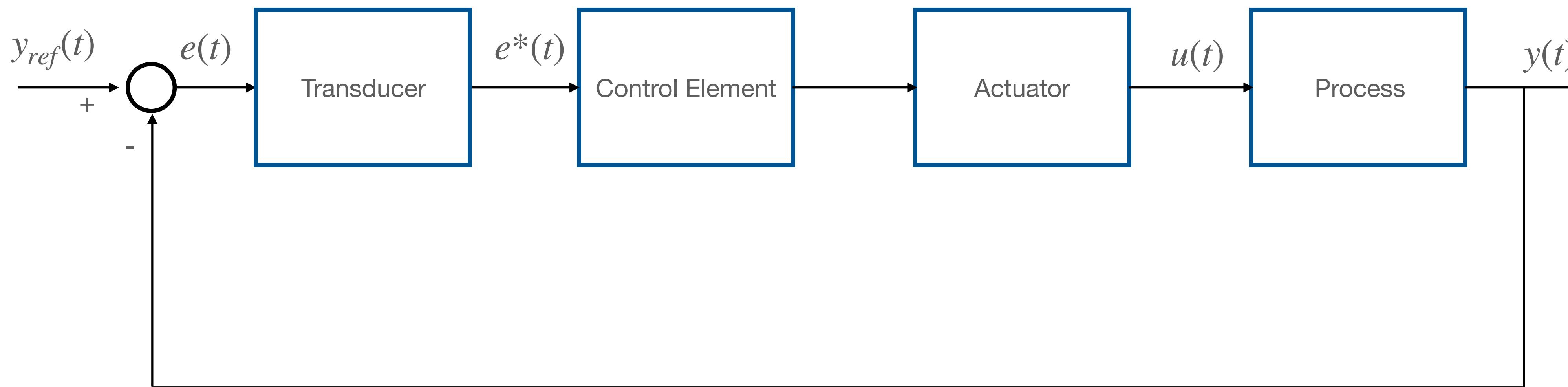
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Assumption: Linear Time-Invariant (LTI) Dynamic Systems



## Basic Control Scheme

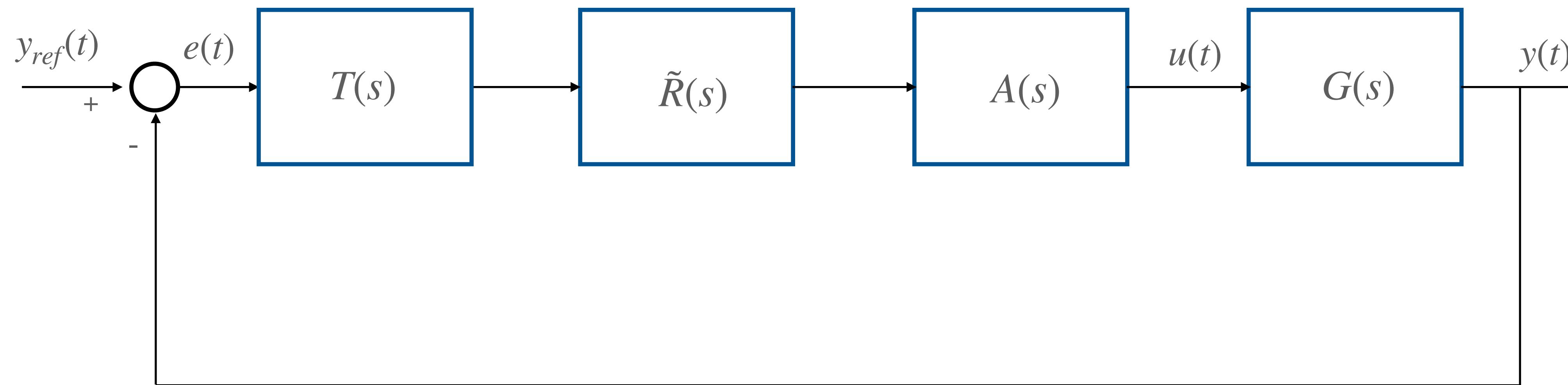


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They can be modelled via Transfer Functions



## Basic Control Scheme

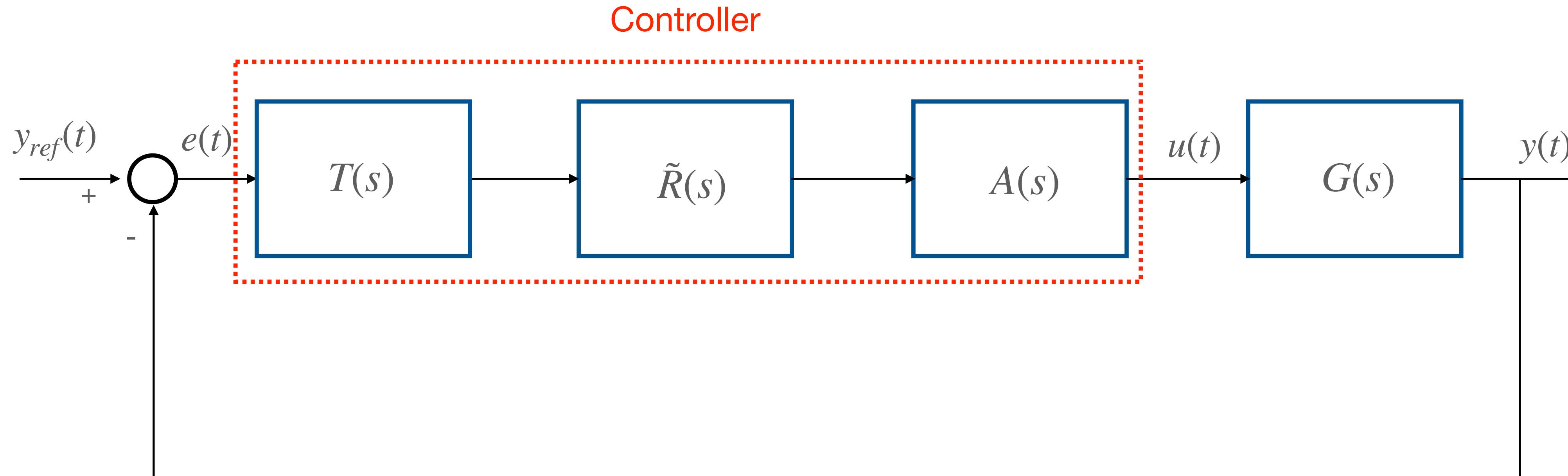


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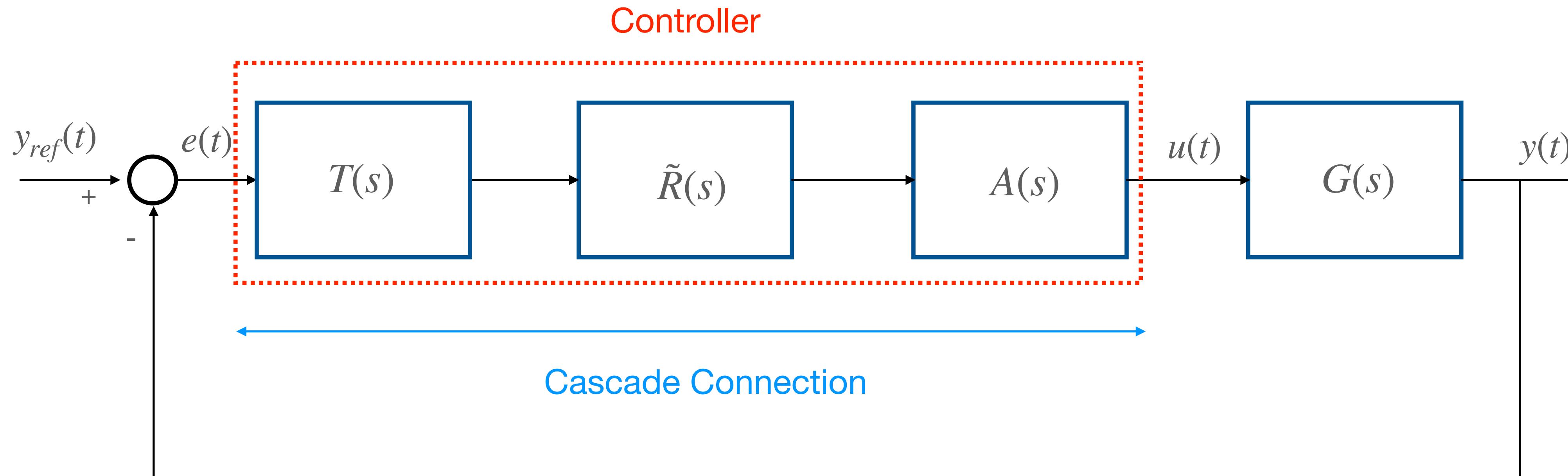
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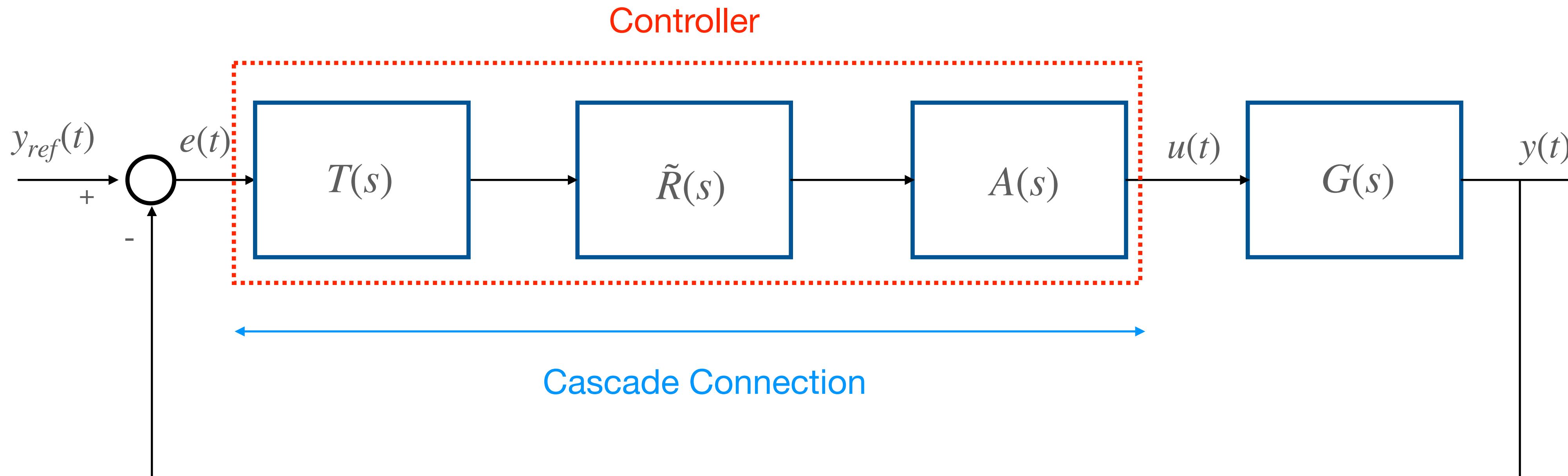
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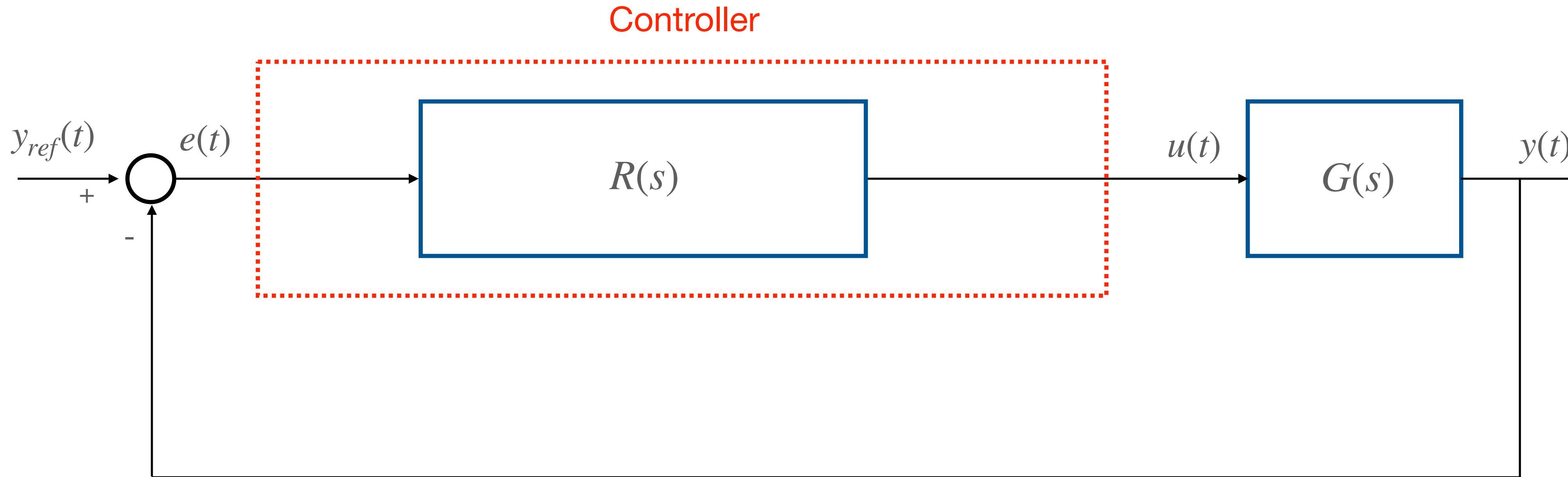


## Basic Control Scheme



Block Algebra Rule:  $T(s) \cdot \tilde{R}(s) \cdot A(s) = R(s)$

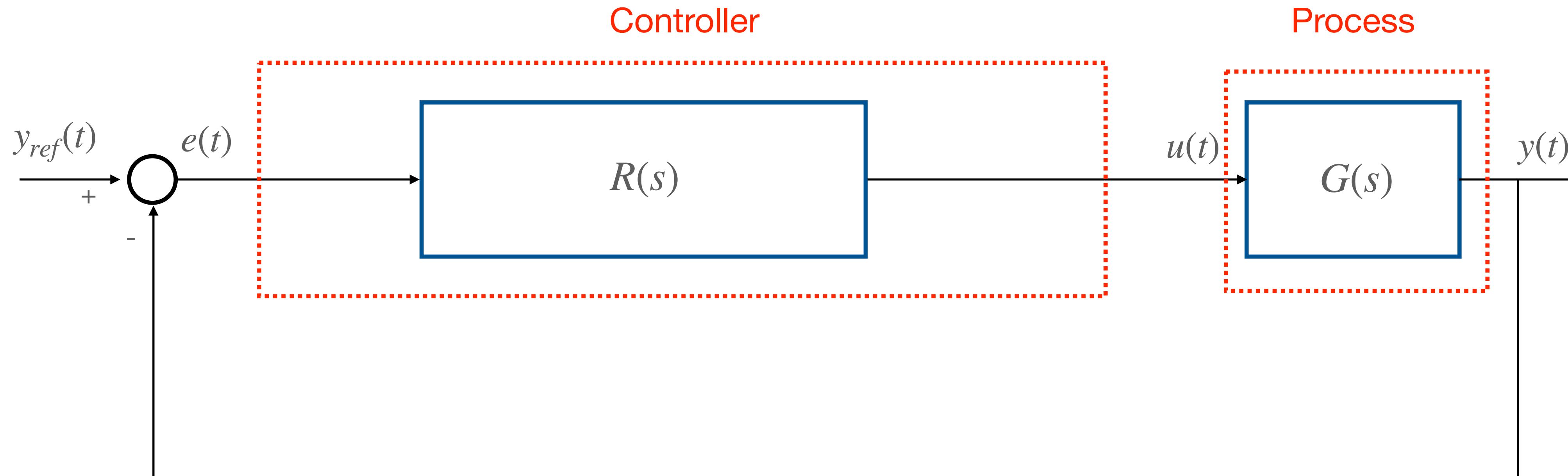
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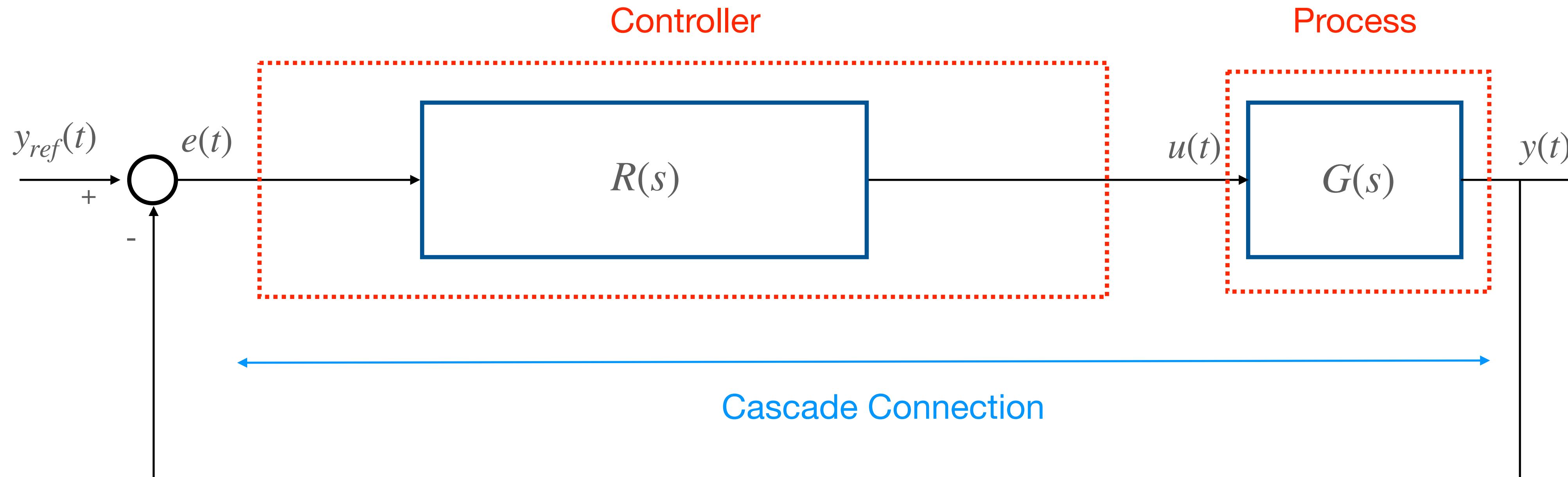
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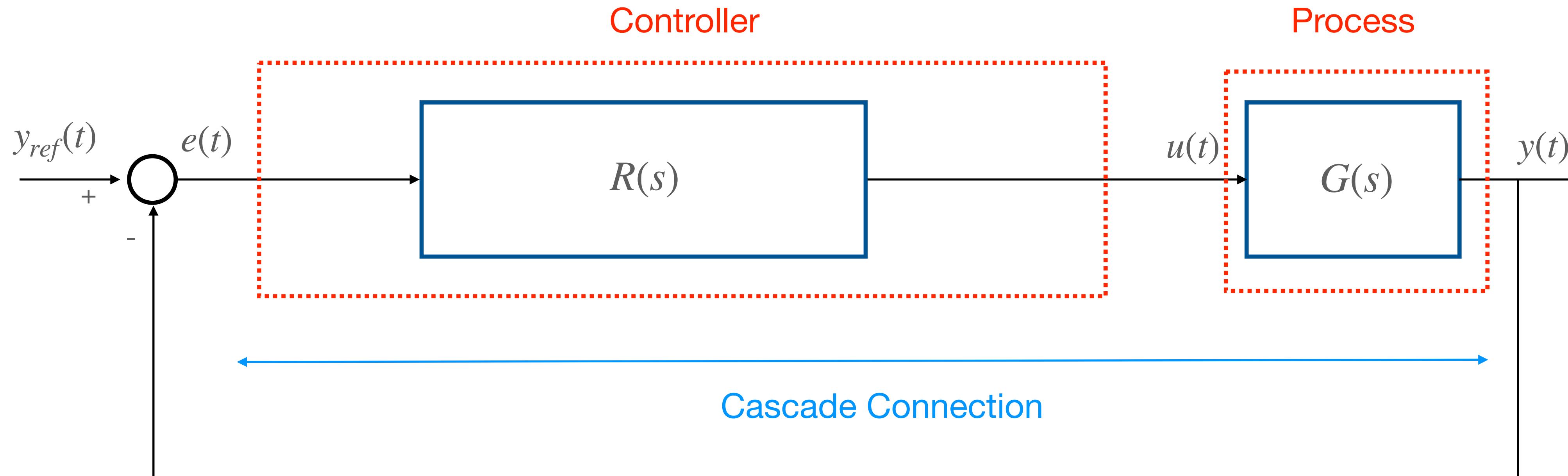
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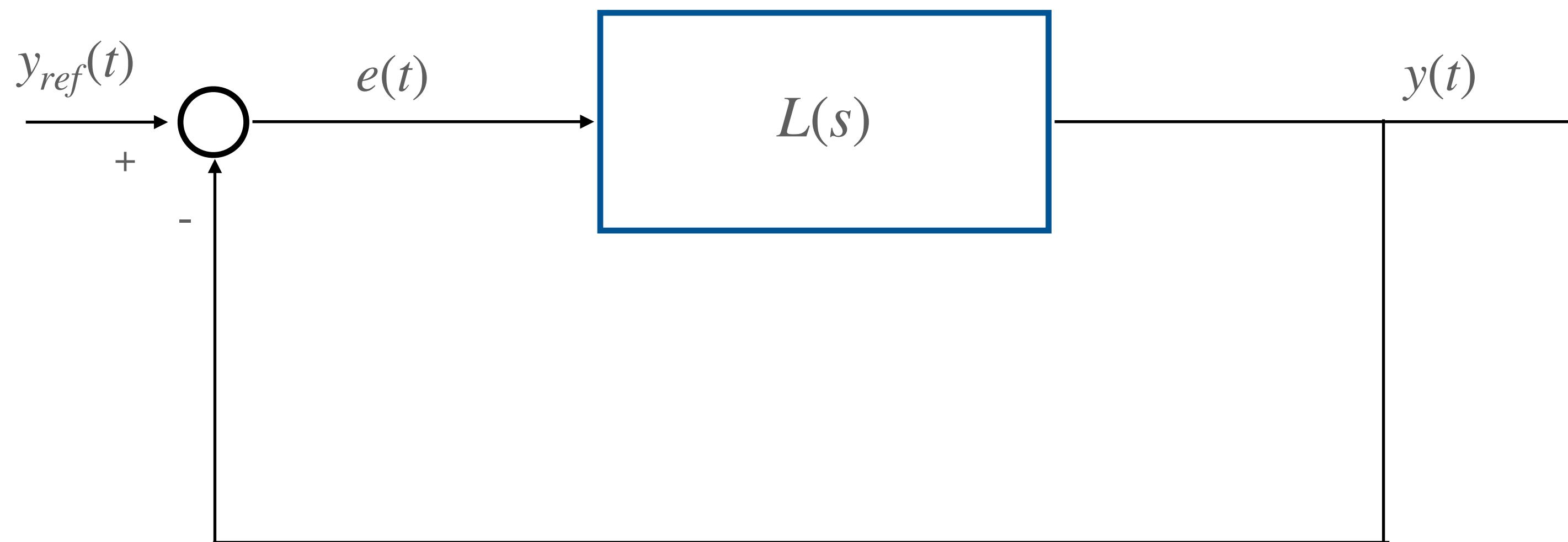


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Block Algebra Rule:  $R(s) \cdot G(s) = L(s)$

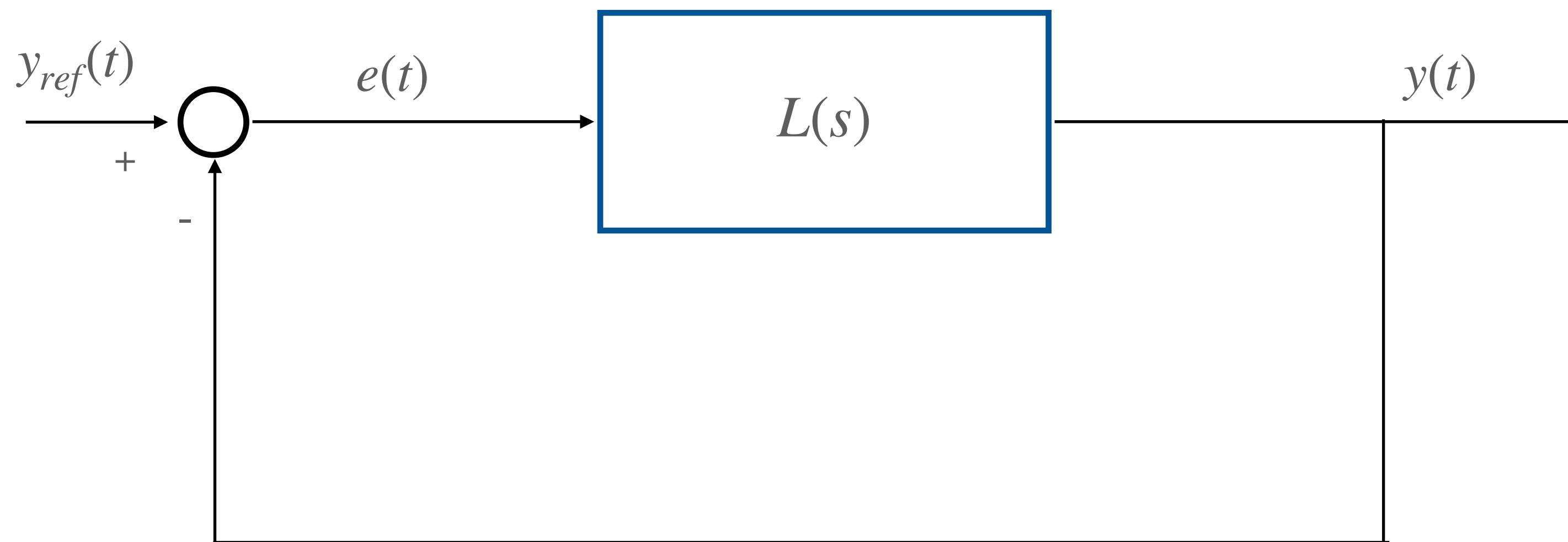
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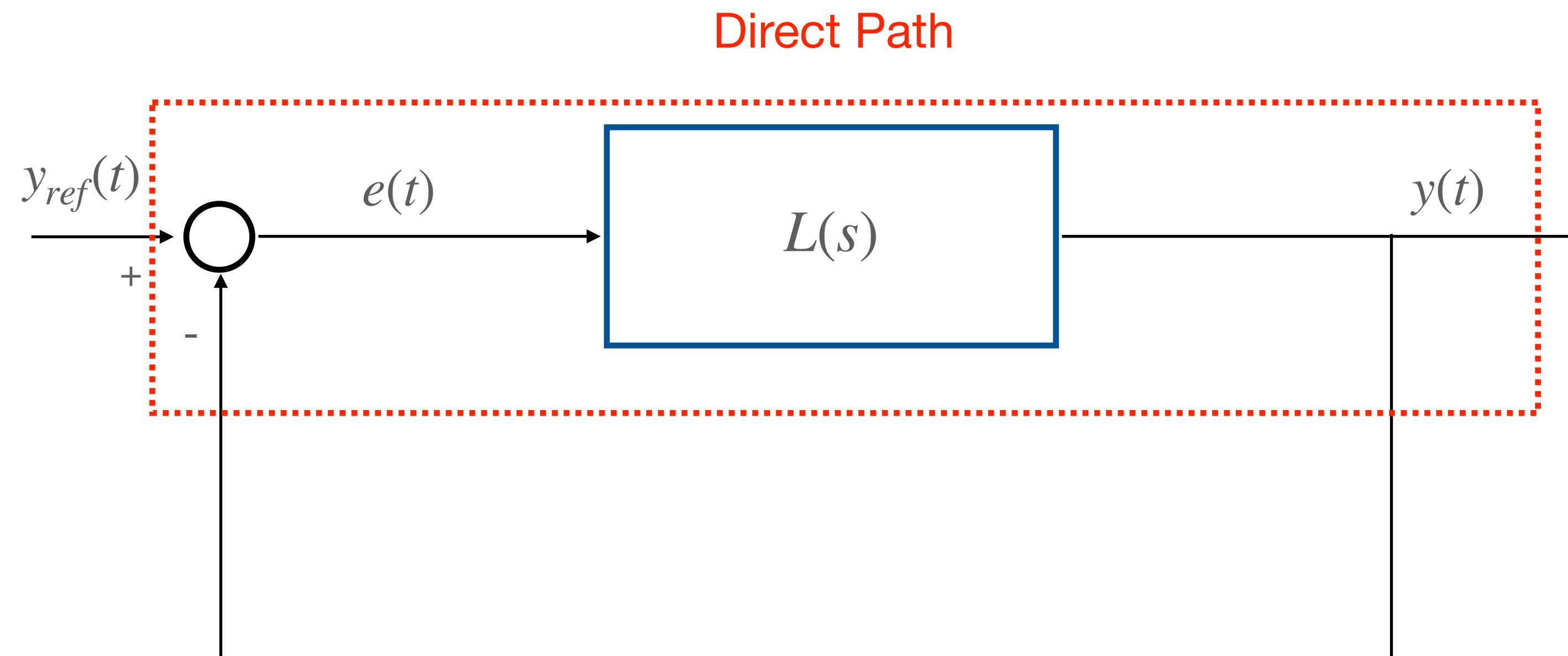


## Unitary Feedback Control Scheme



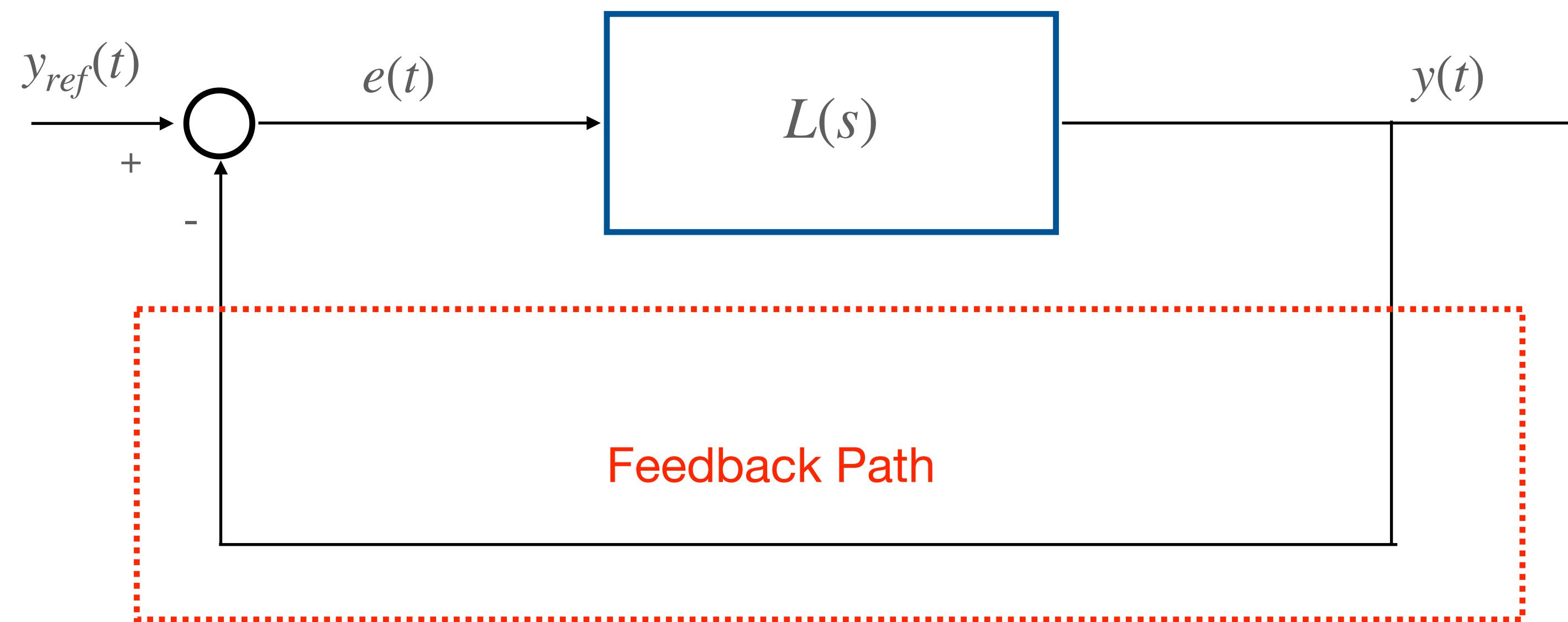
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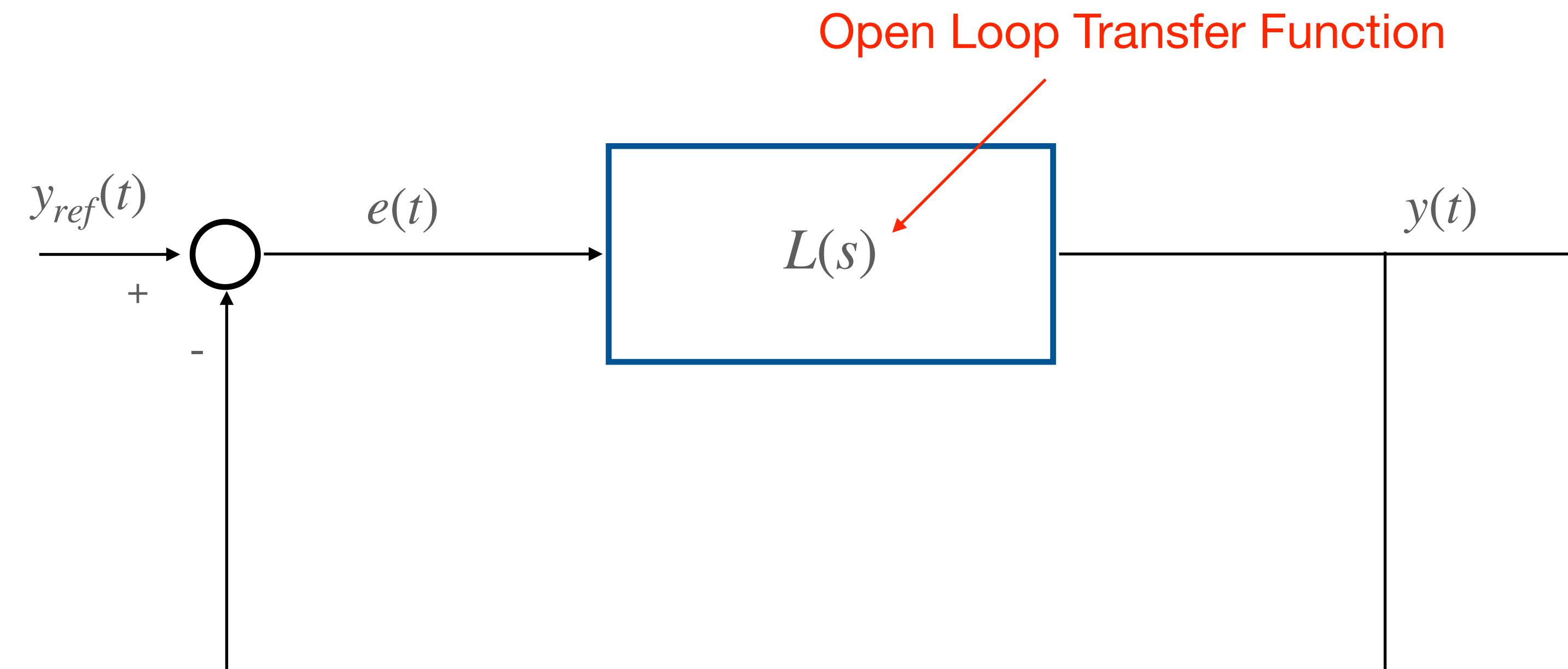
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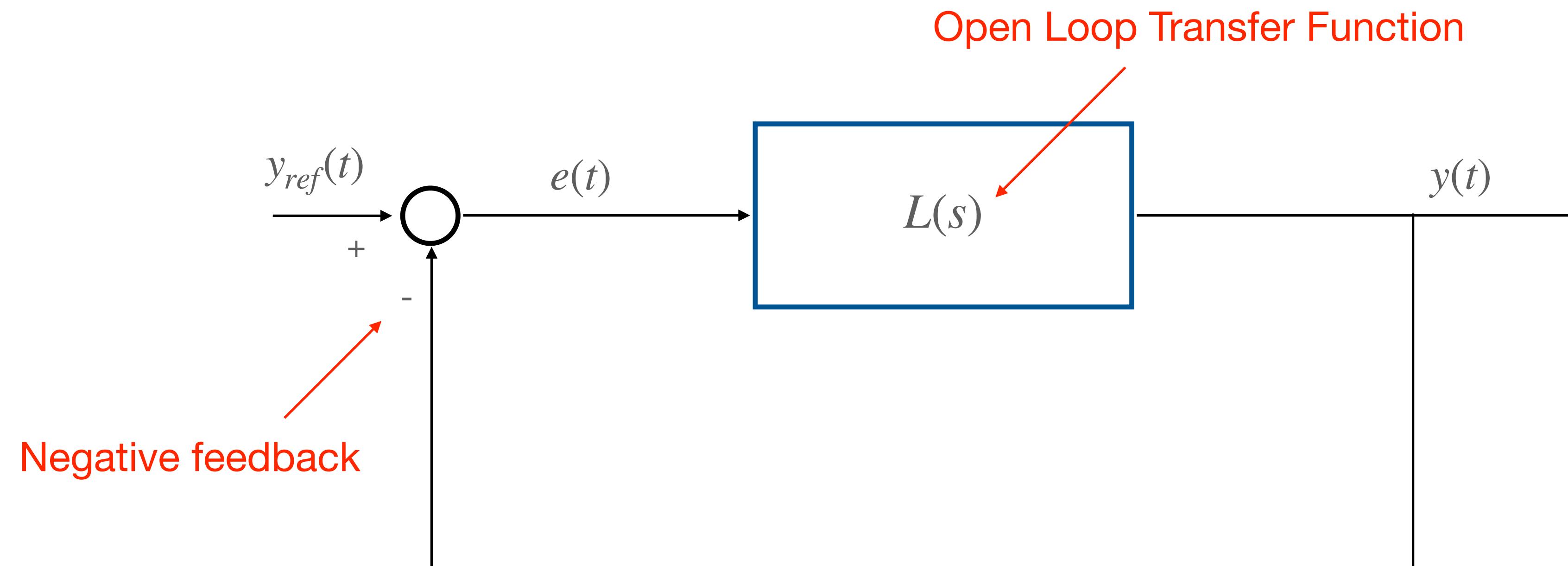
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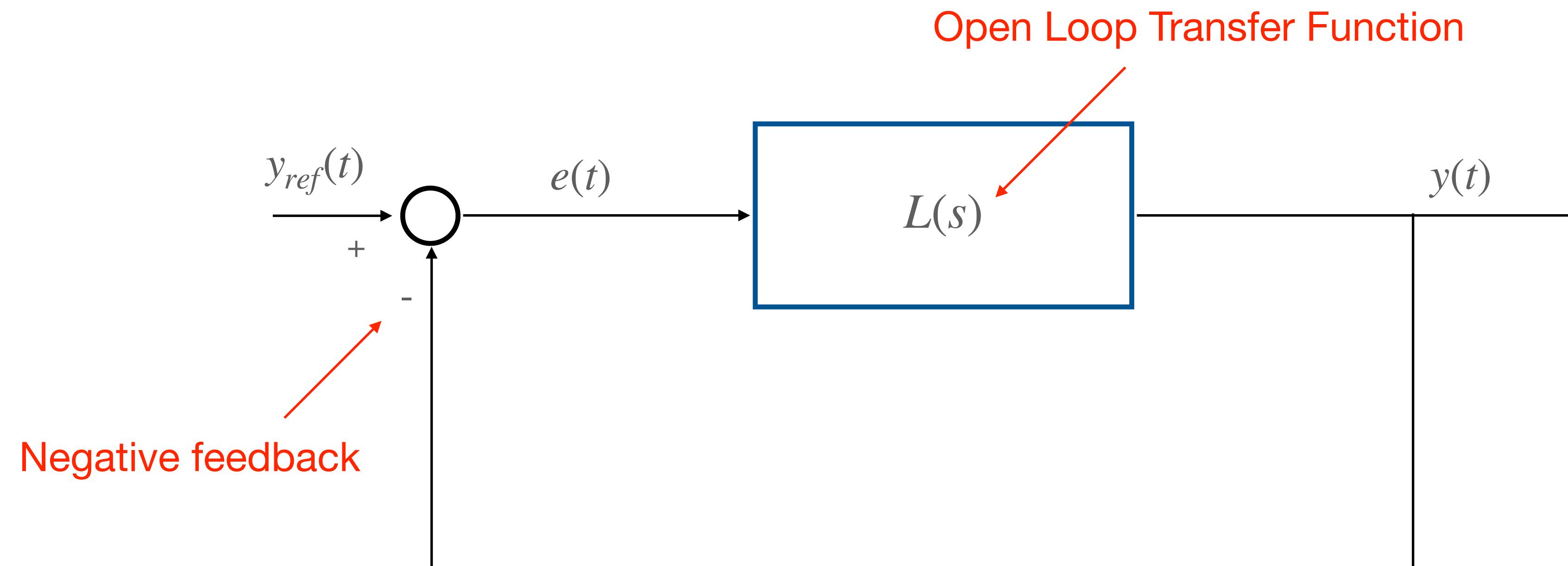
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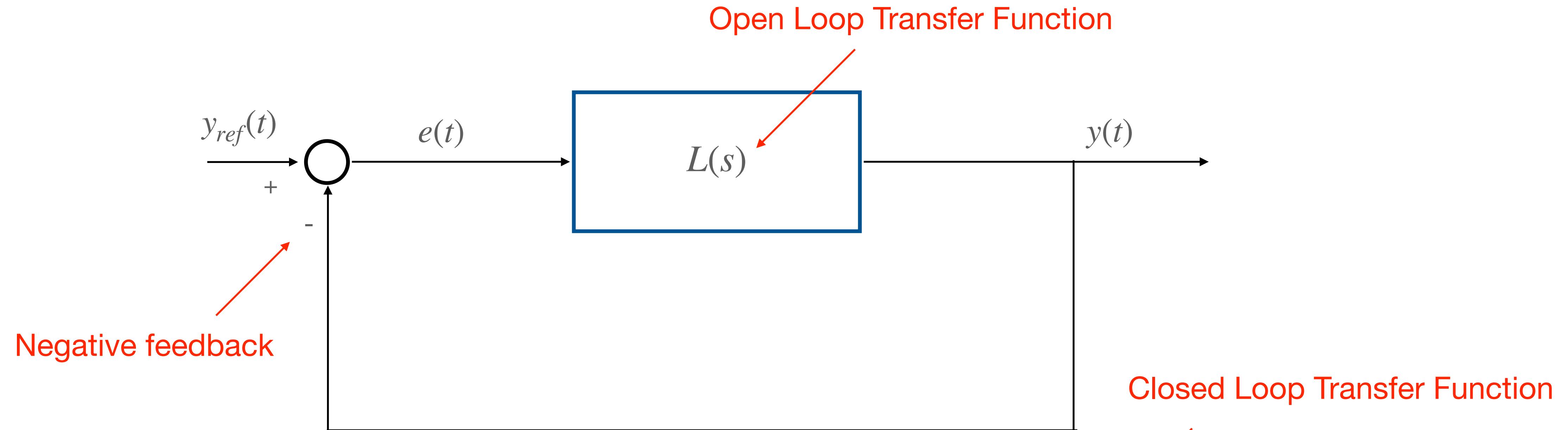


Block Algebra Rule:

$$F(s) = \frac{R(s) \cdot G(s)}{1 + R(s) \cdot G(s)} = \frac{L(s)}{1 + L(s)}$$



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## Limitations of the Basic Unitary Feedback Control Scheme

- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays
- No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
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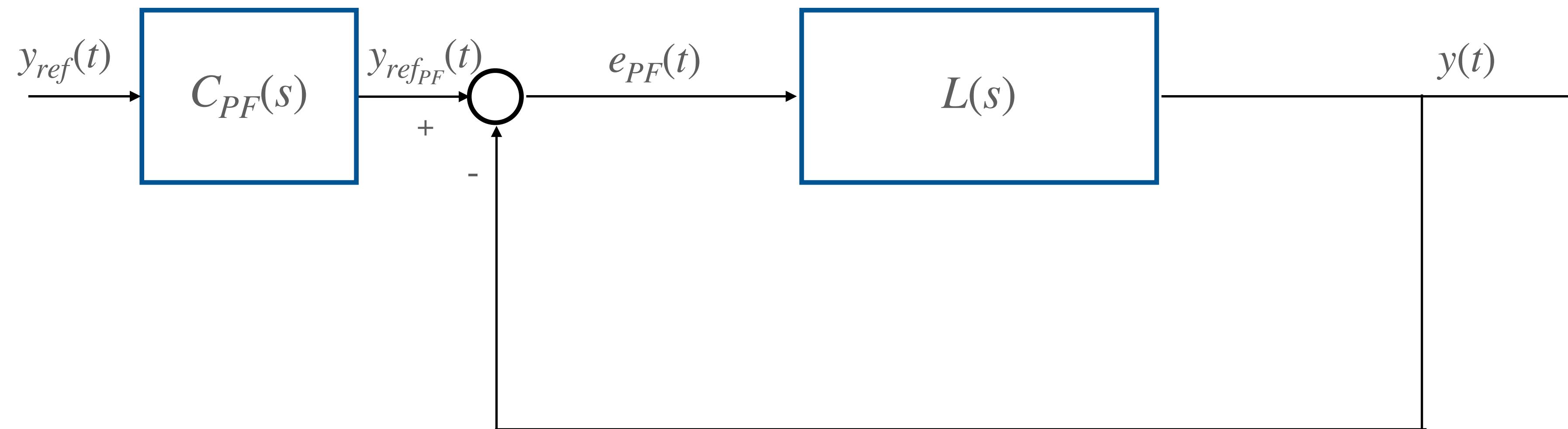
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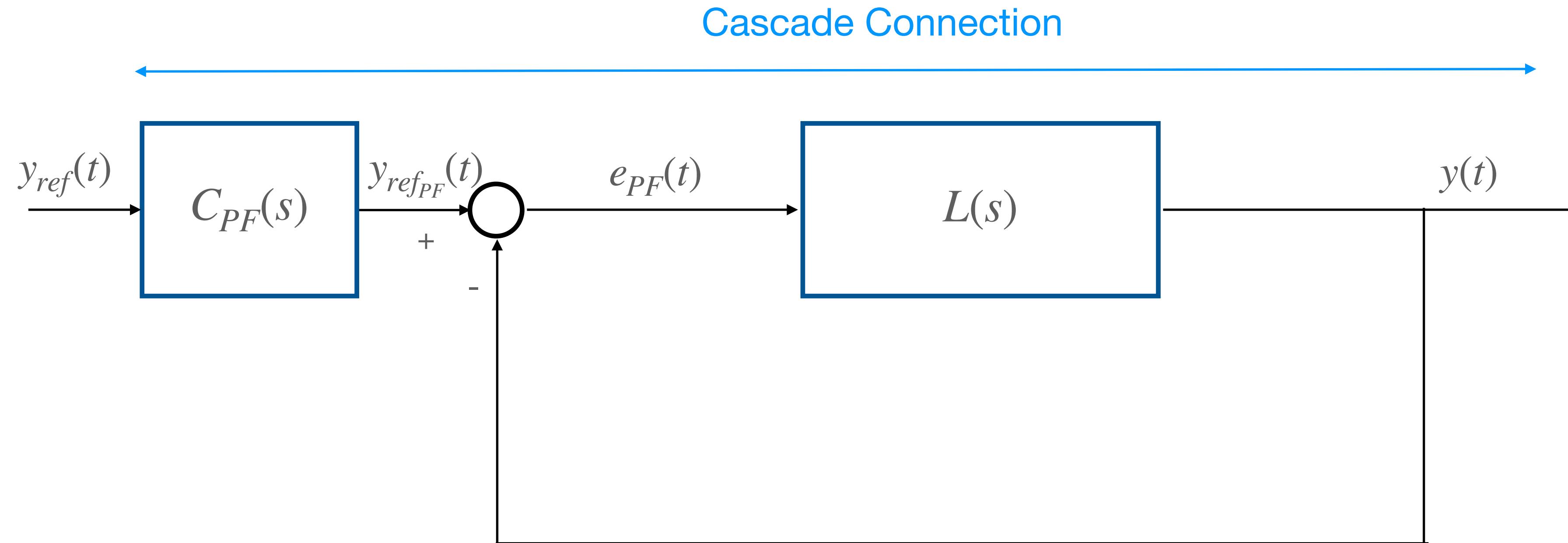
Possible solution: add a Pre-Filter to the basic unitary feedback control scheme



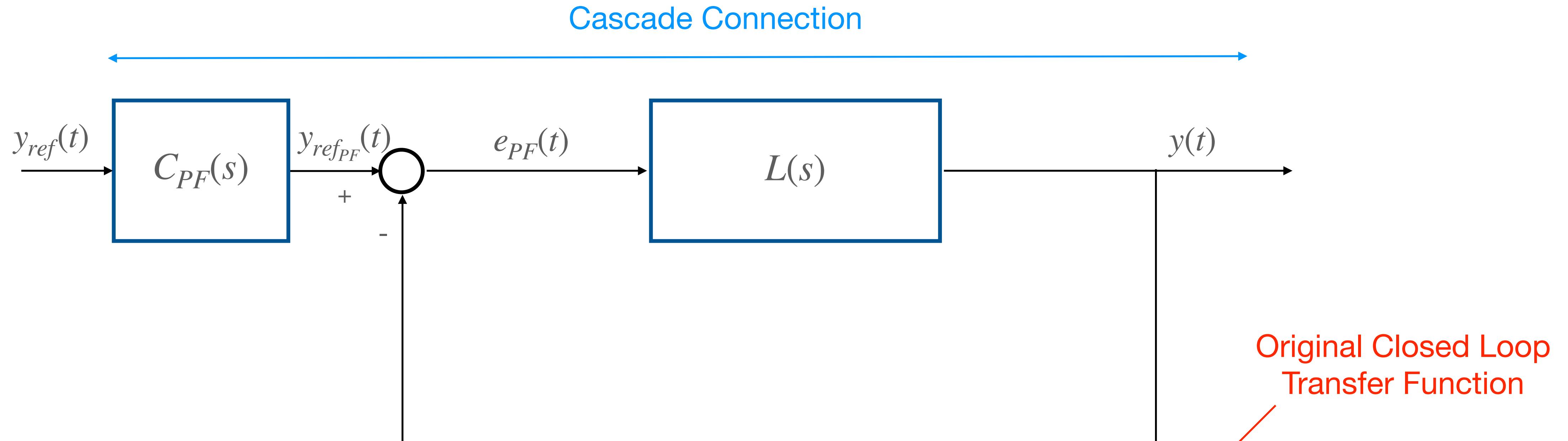
## Pre-filter Based Control Scheme



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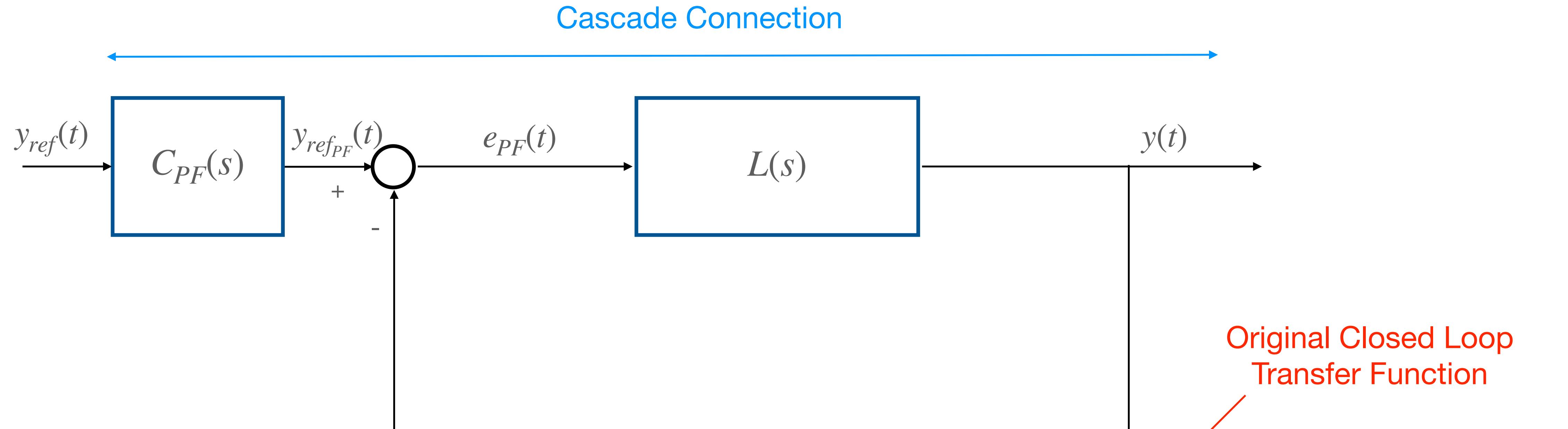


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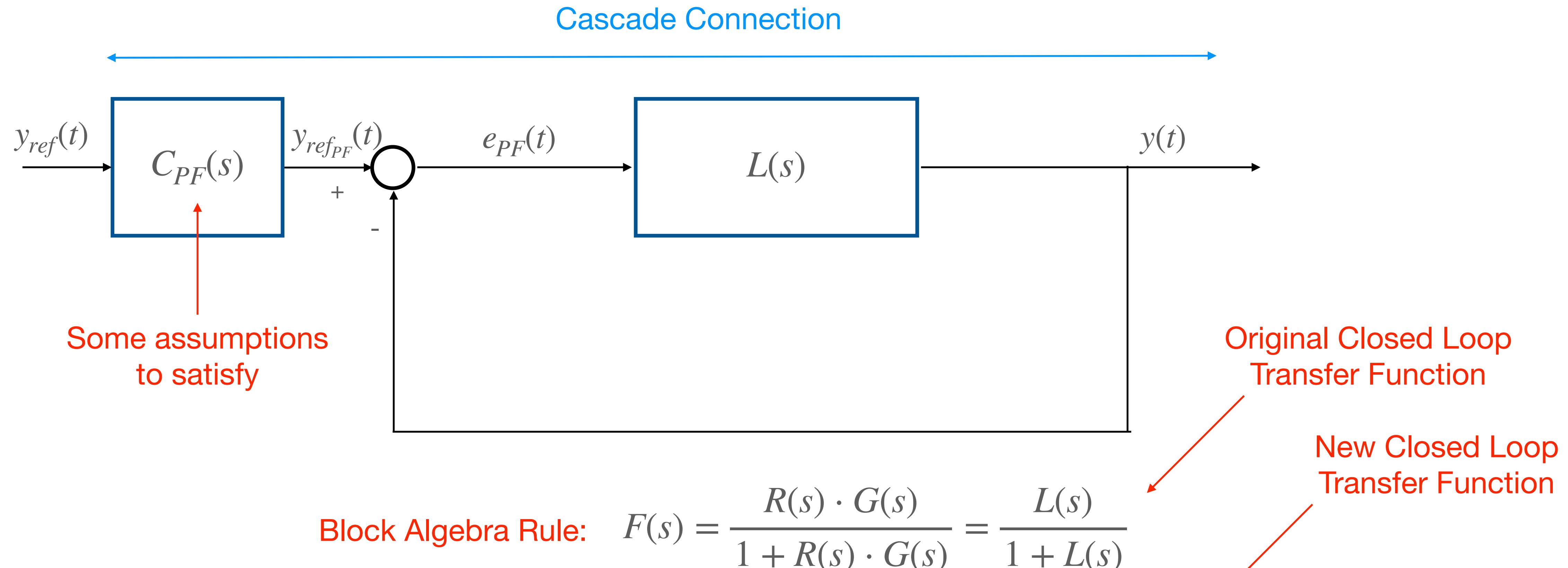
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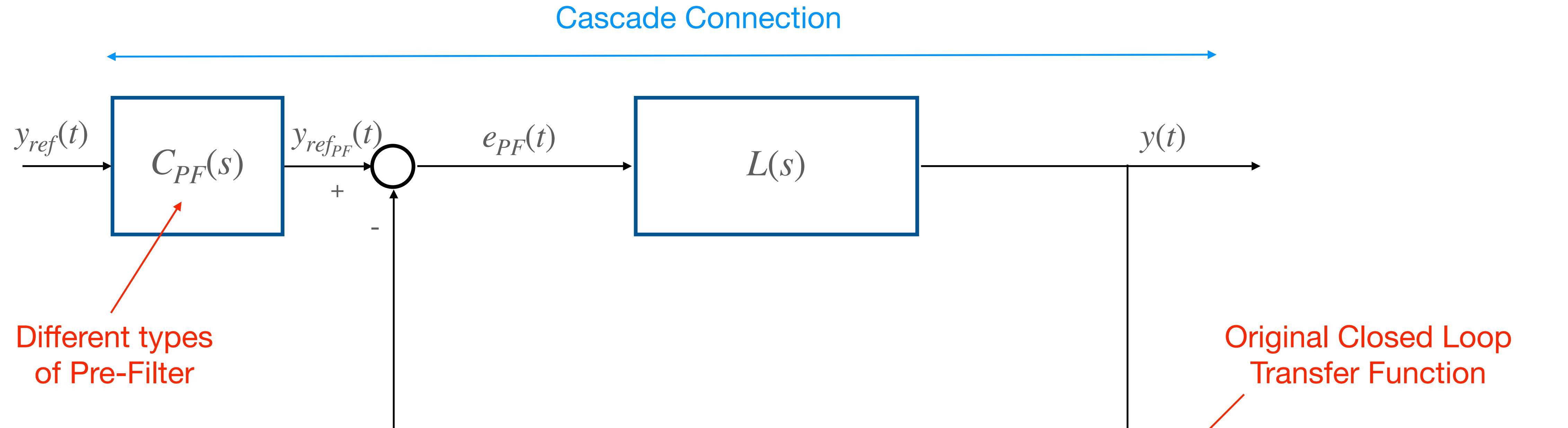
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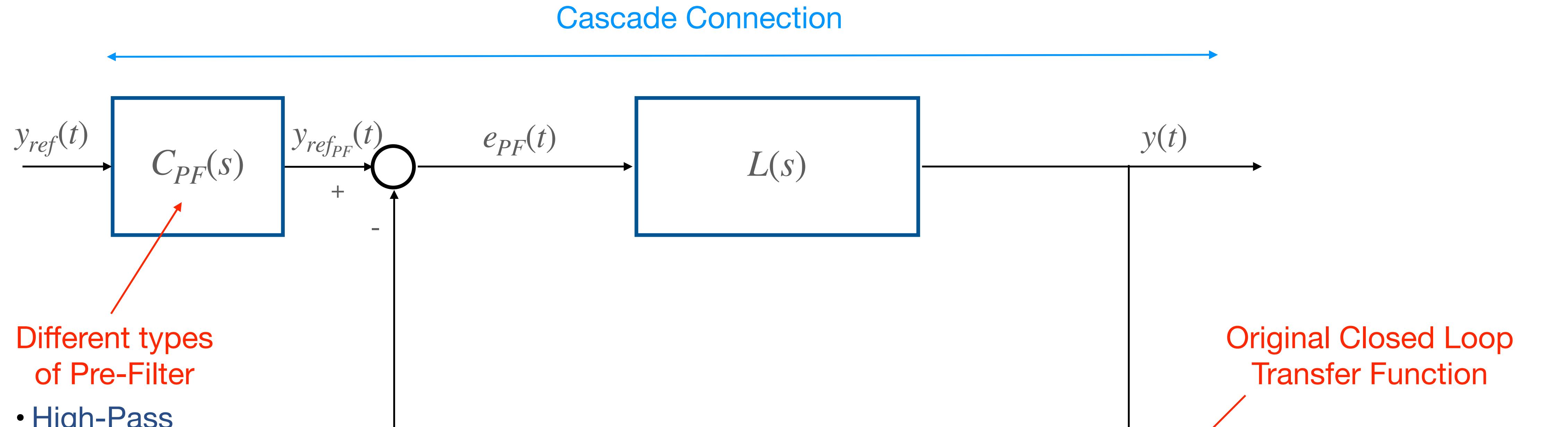
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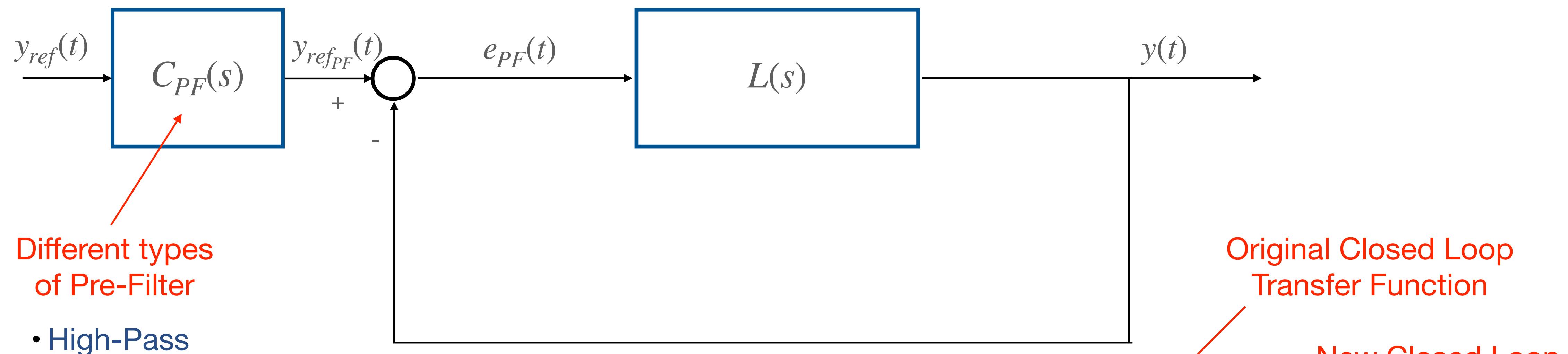
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## Pre-filter Based Control Scheme



## Pre-filter Based Control Scheme: Case 1



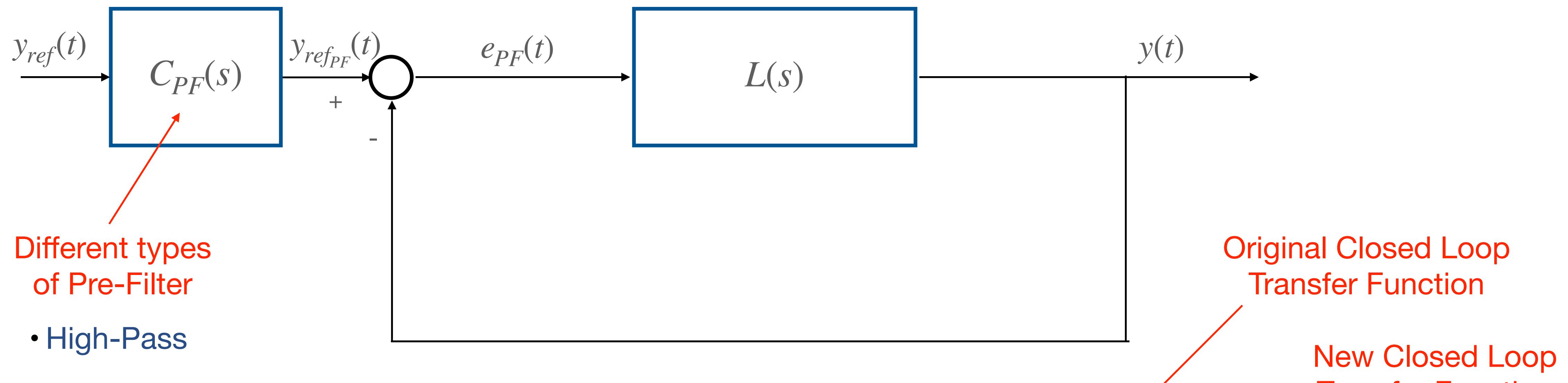
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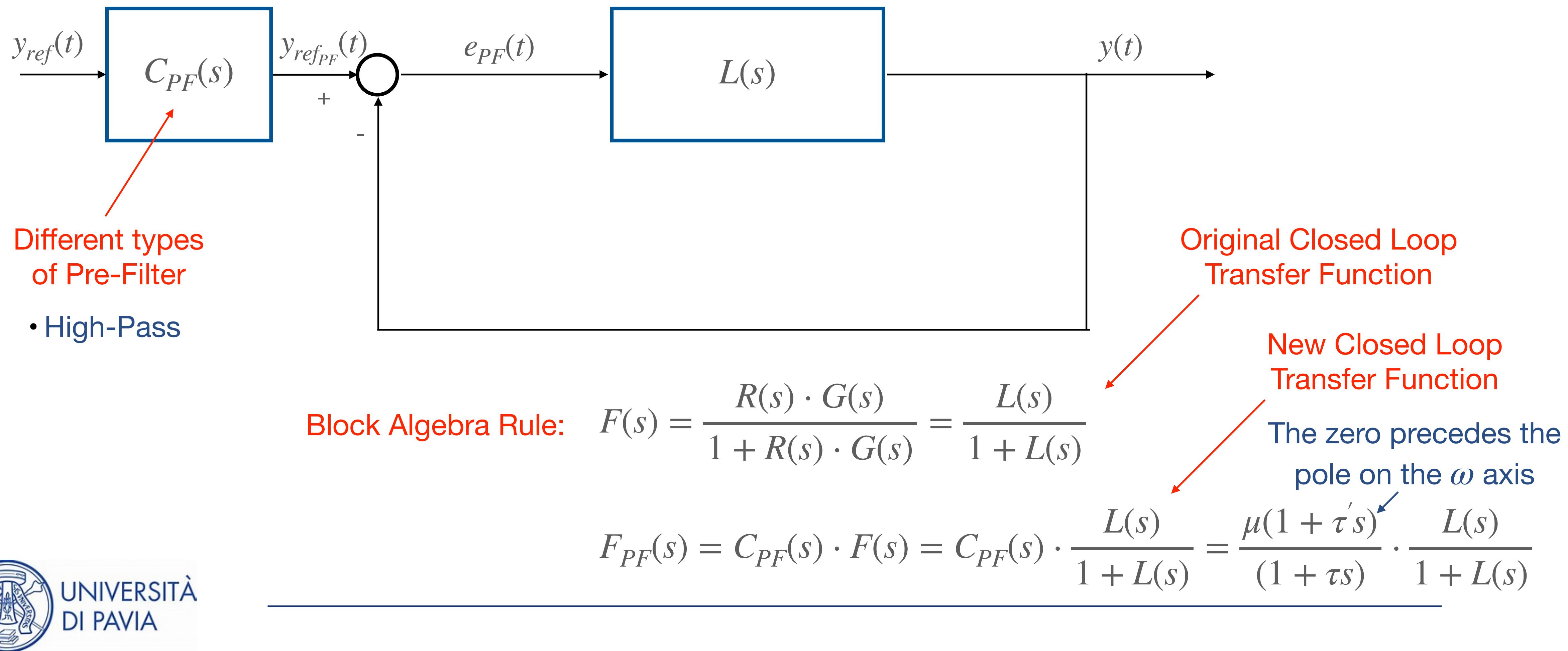
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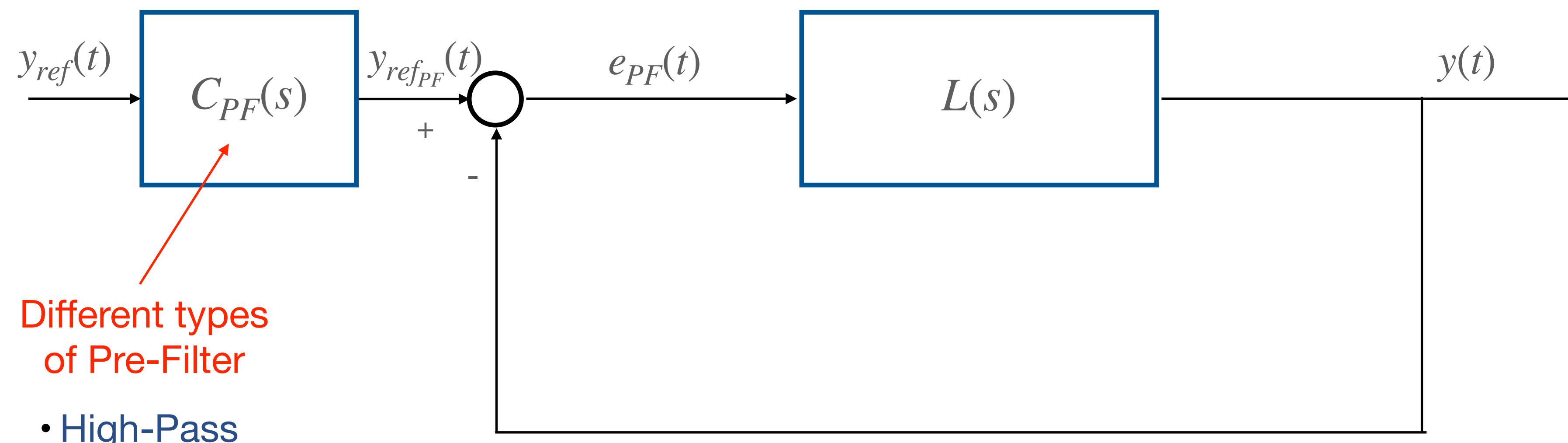
$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



## Pre-filter Based Control Scheme: Case 1



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Assumptions:

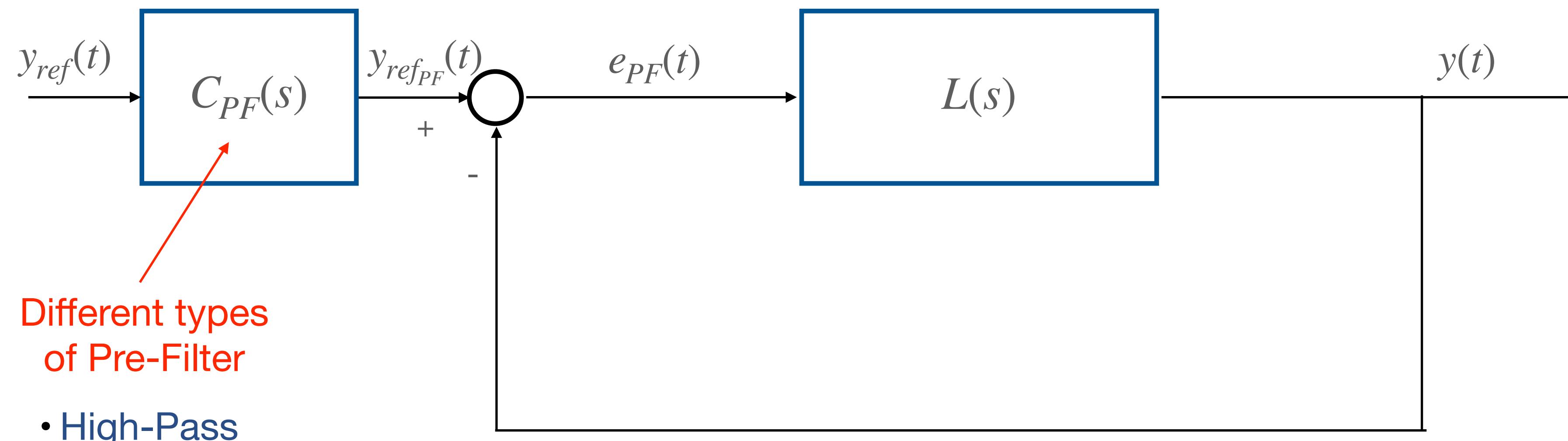
- $C_{PF}(s)$  As. Stable
- Proper
- Unitary gain

The zero precedes the pole on the  $\omega$  axis

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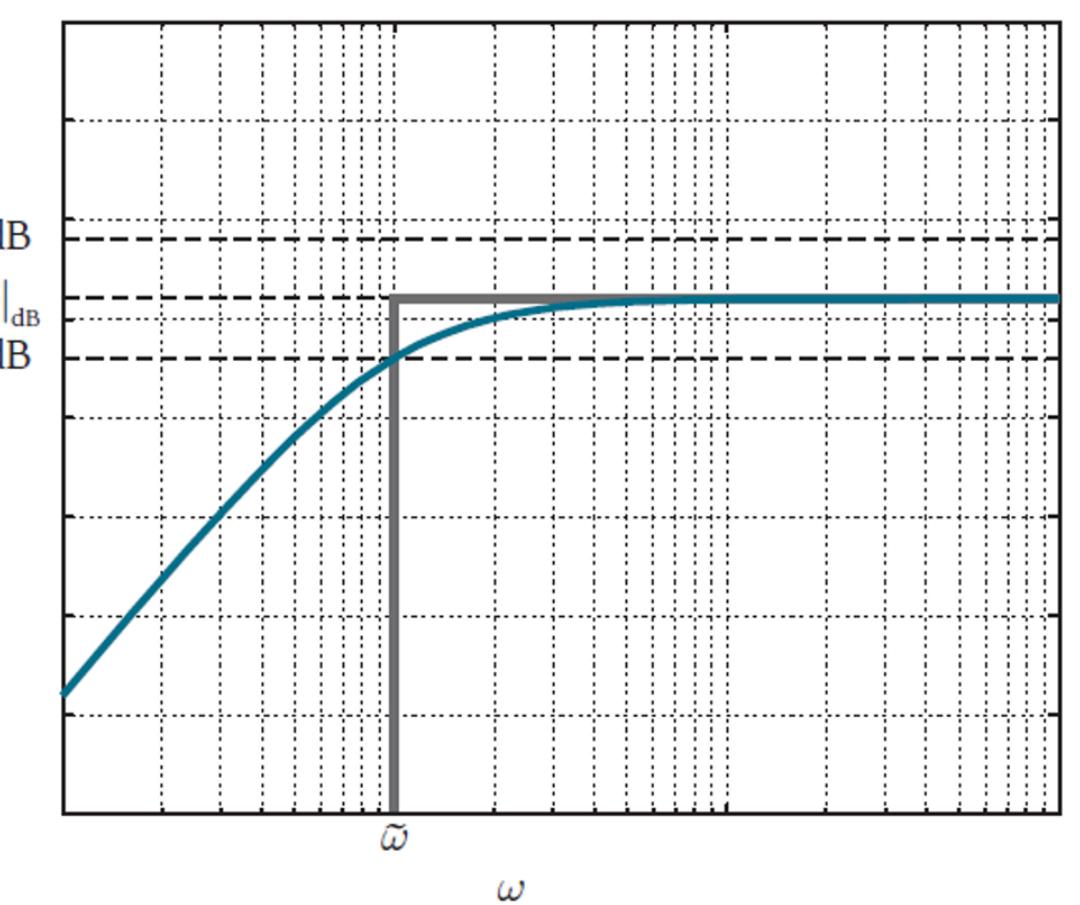
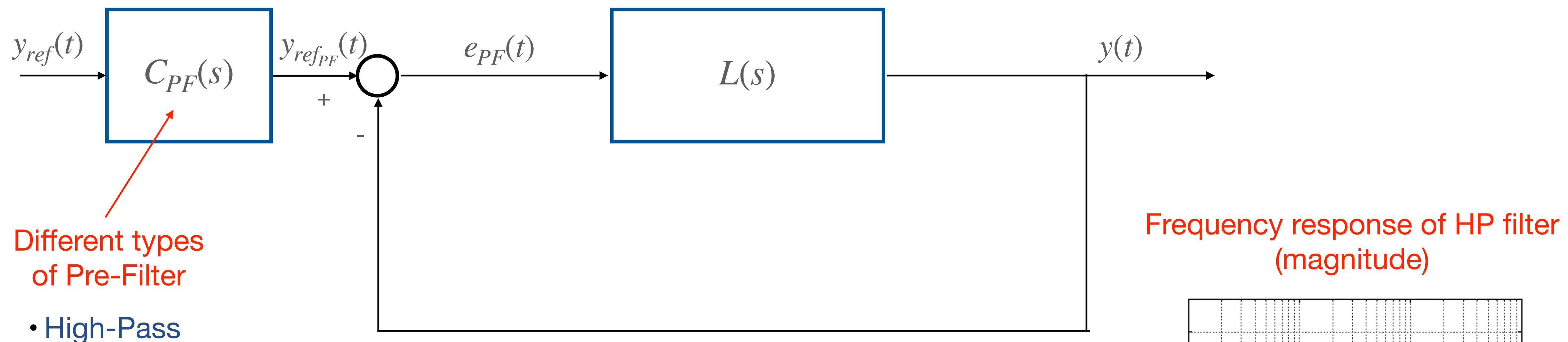
$$\lim_{s \rightarrow 0} \frac{\mu(1 + \tau' s)}{1 + \tau s} = \mu = 1$$

The zero precedes the pole on the  $\omega$  axis

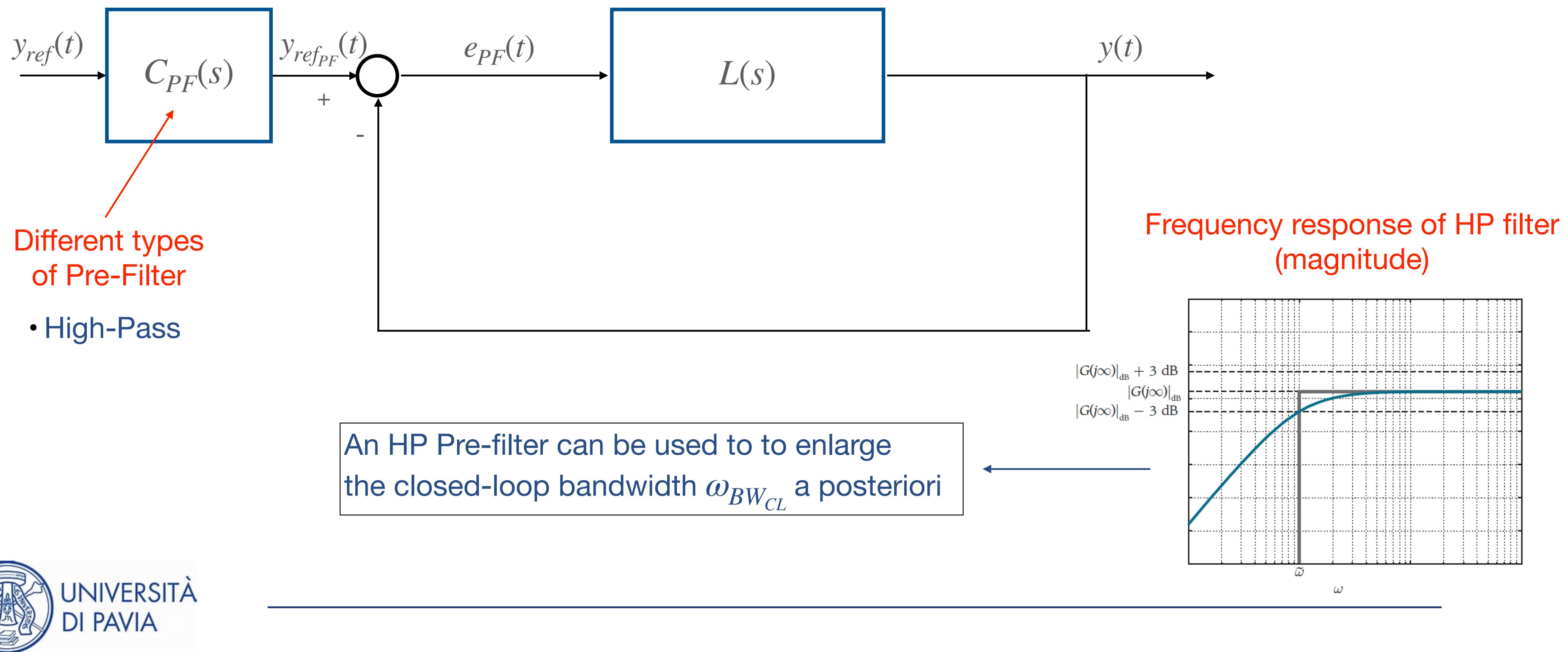
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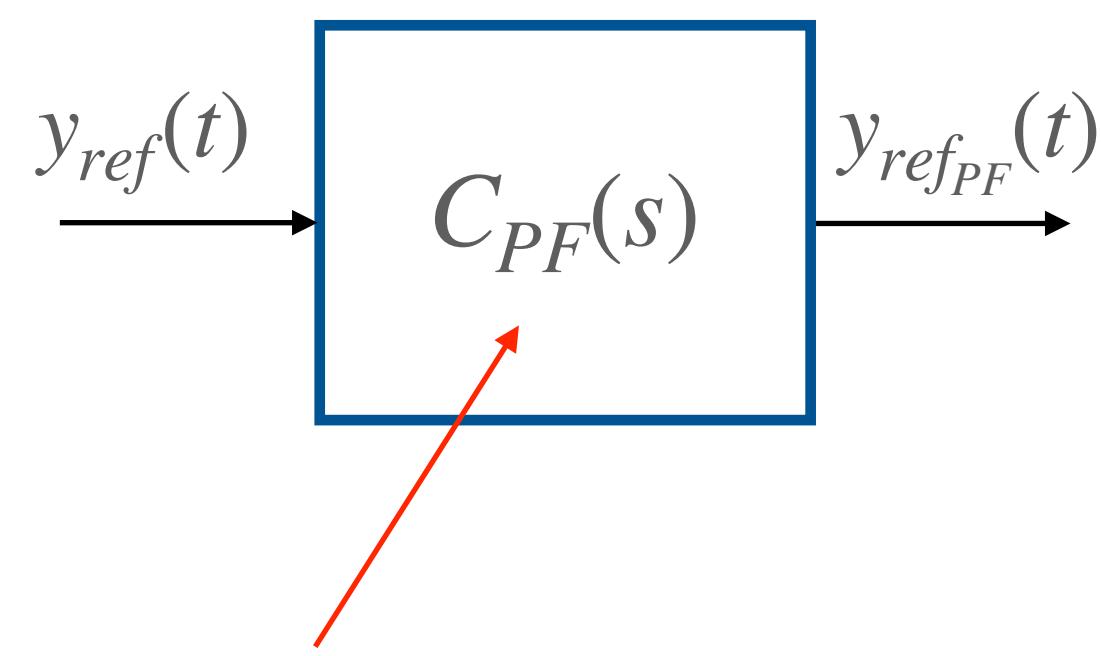
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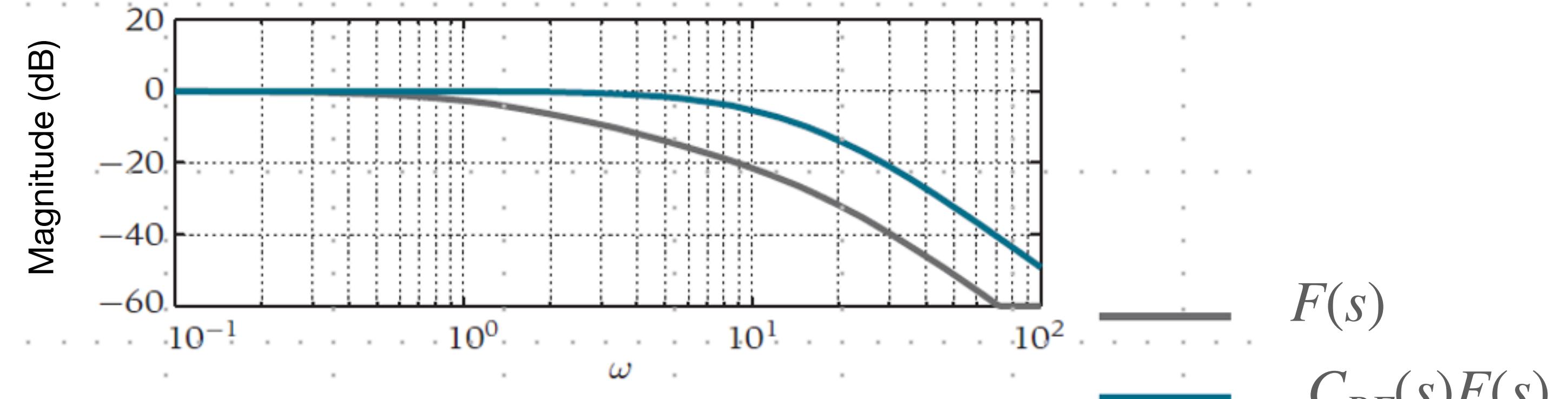
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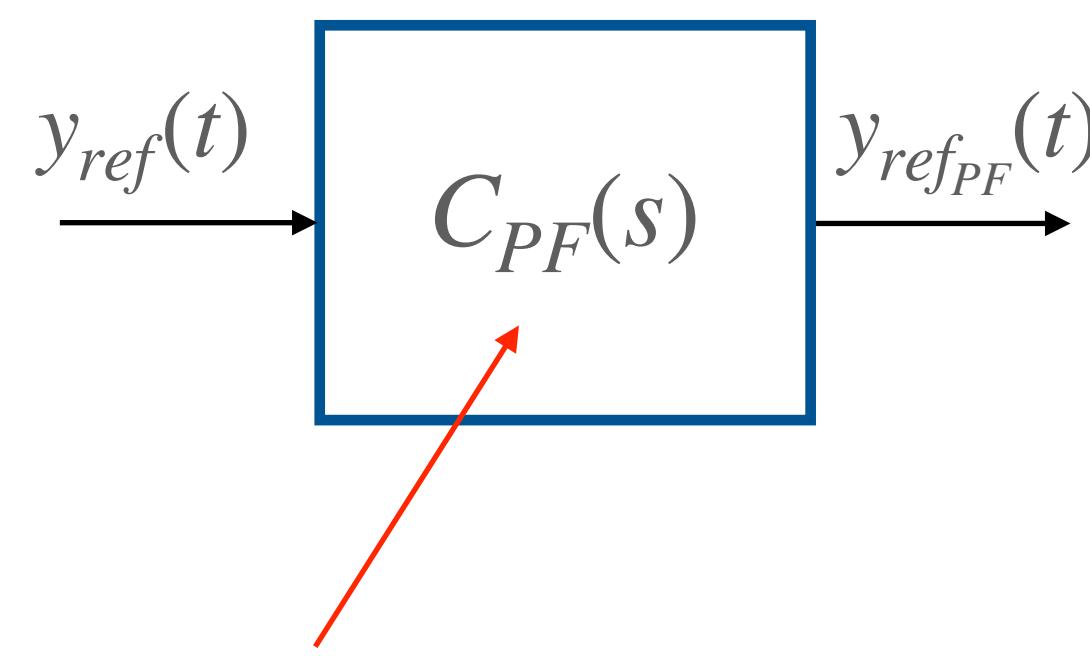
Different types  
of Pre-Filter

- High-Pass

Frequency domain



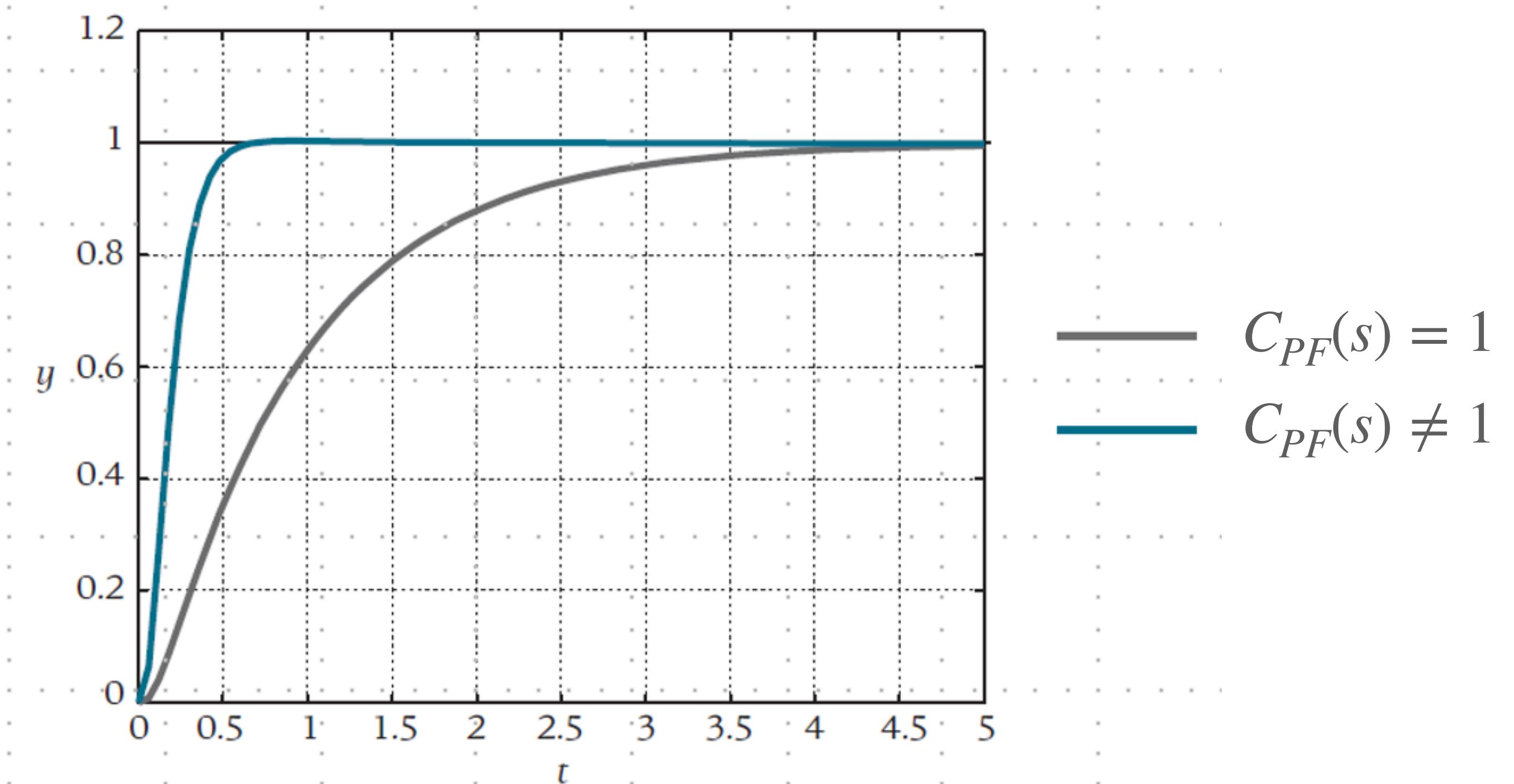
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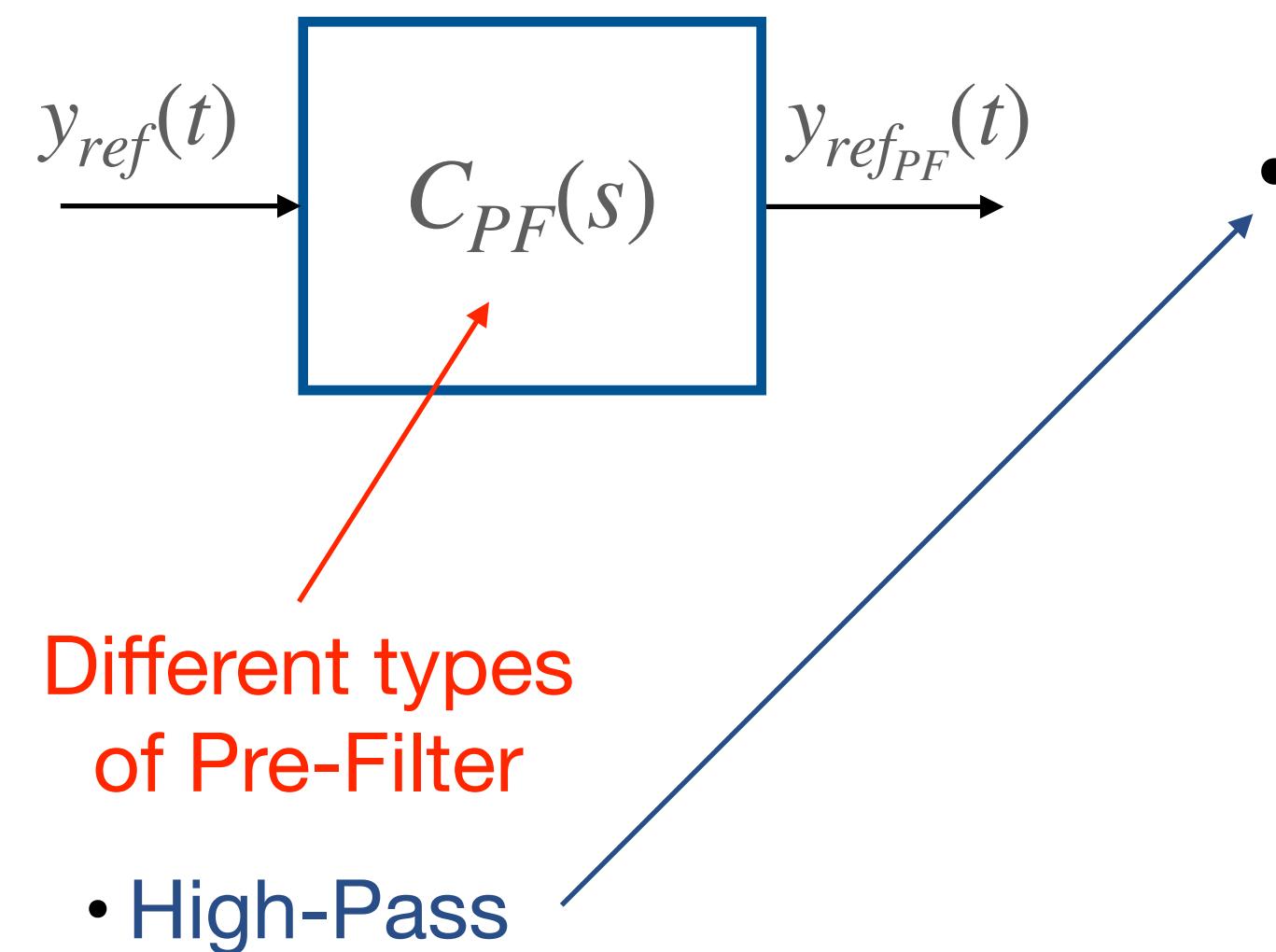
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Time domain



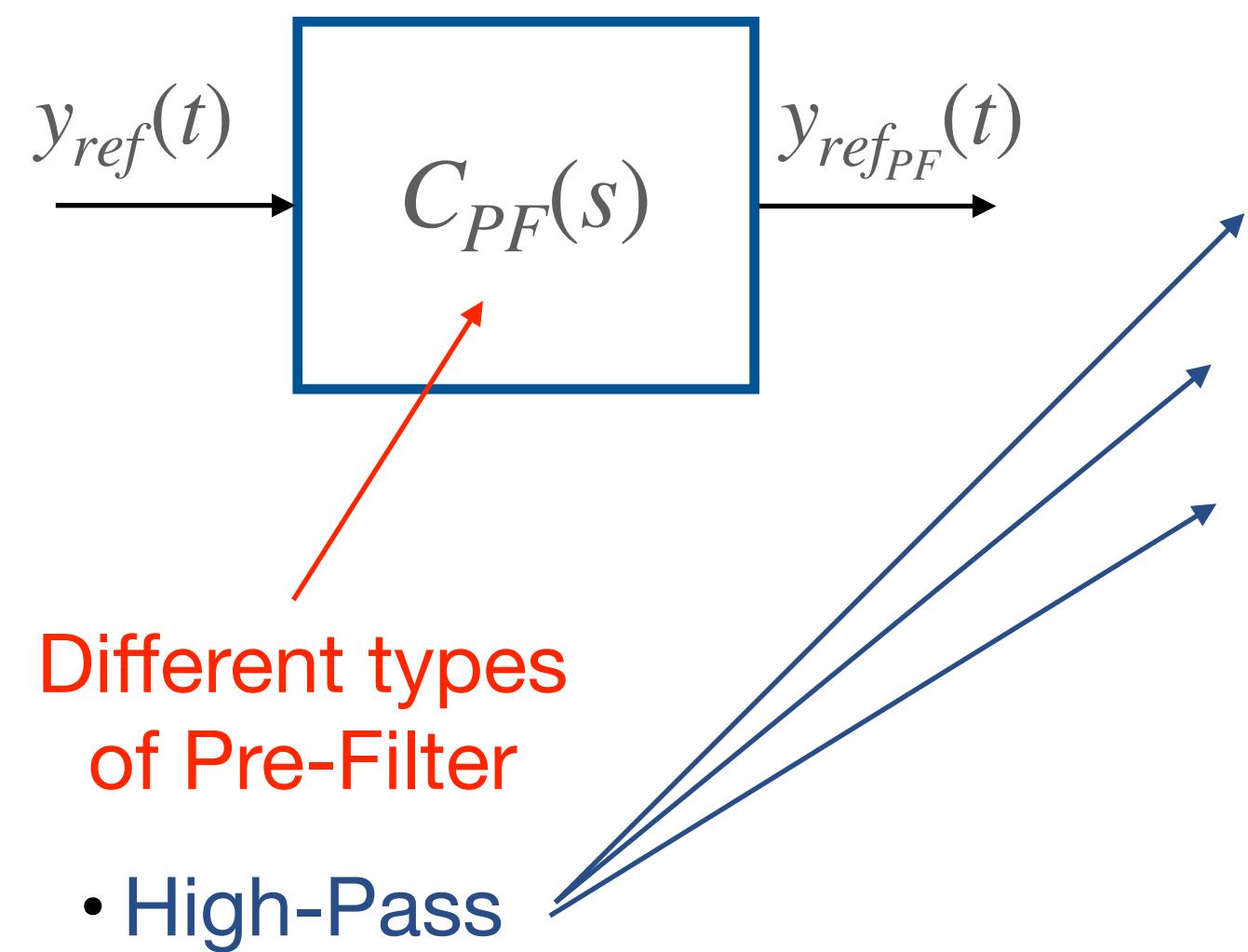
## Pre-filter Based Control Scheme: Case 1



- Limited bandwidth in case of measurement disturbances

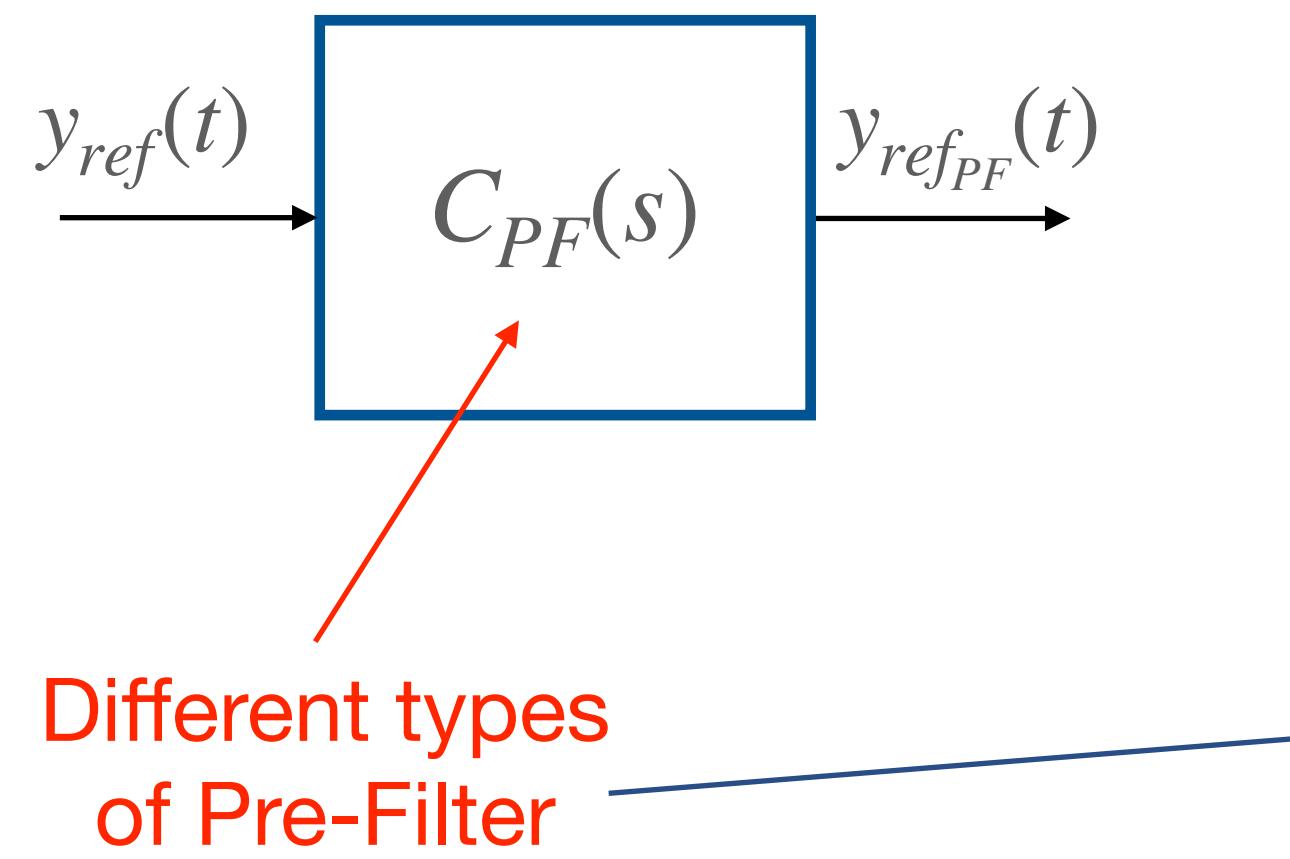


## Pre-filter Based Control Scheme: Case 1



- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays

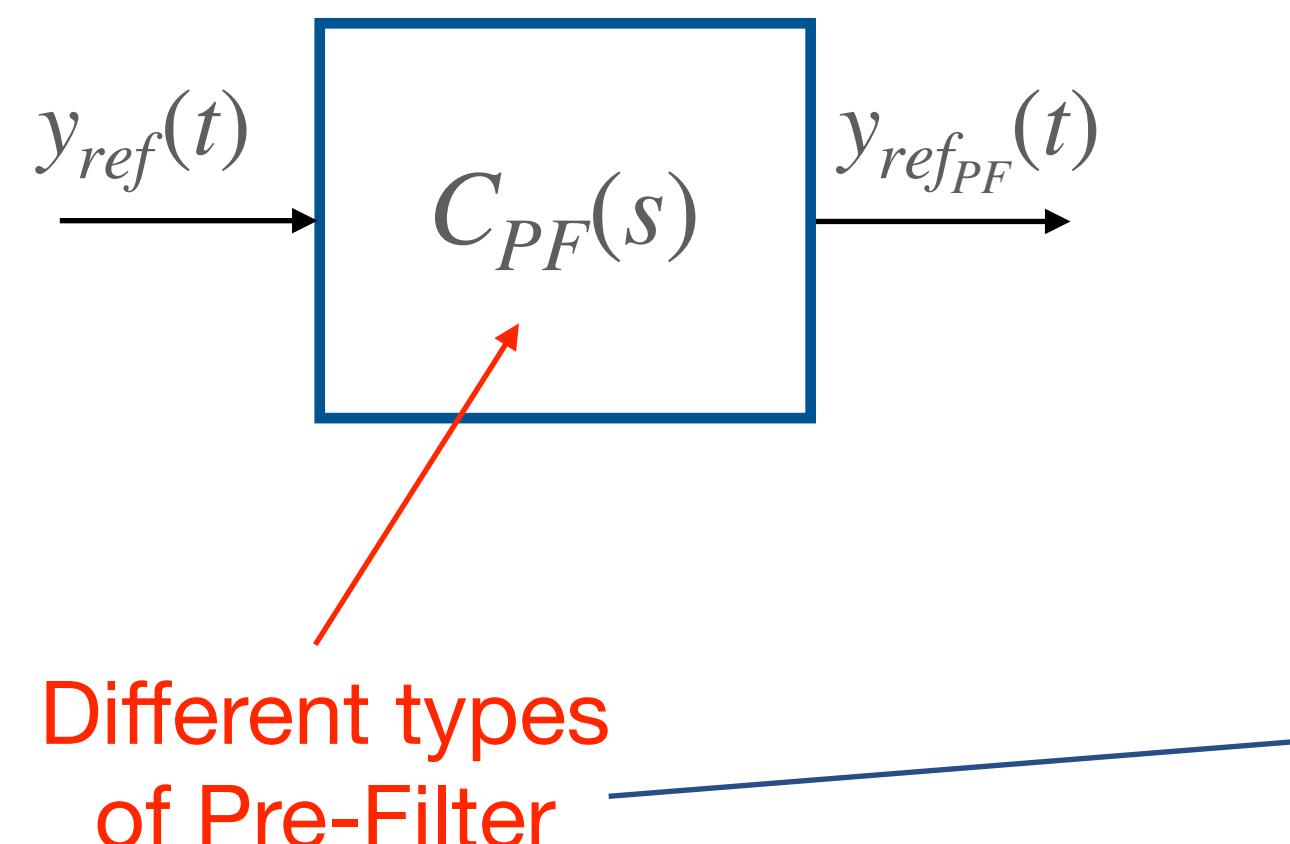
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- Limited bandwidth in case of measurement disturbances
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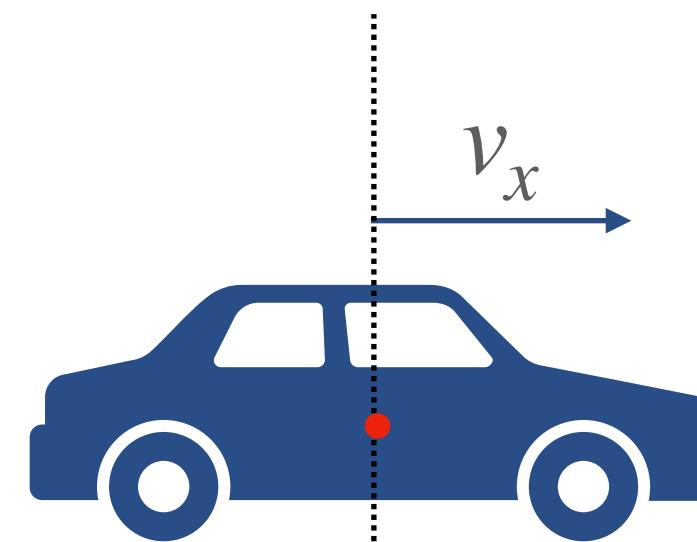


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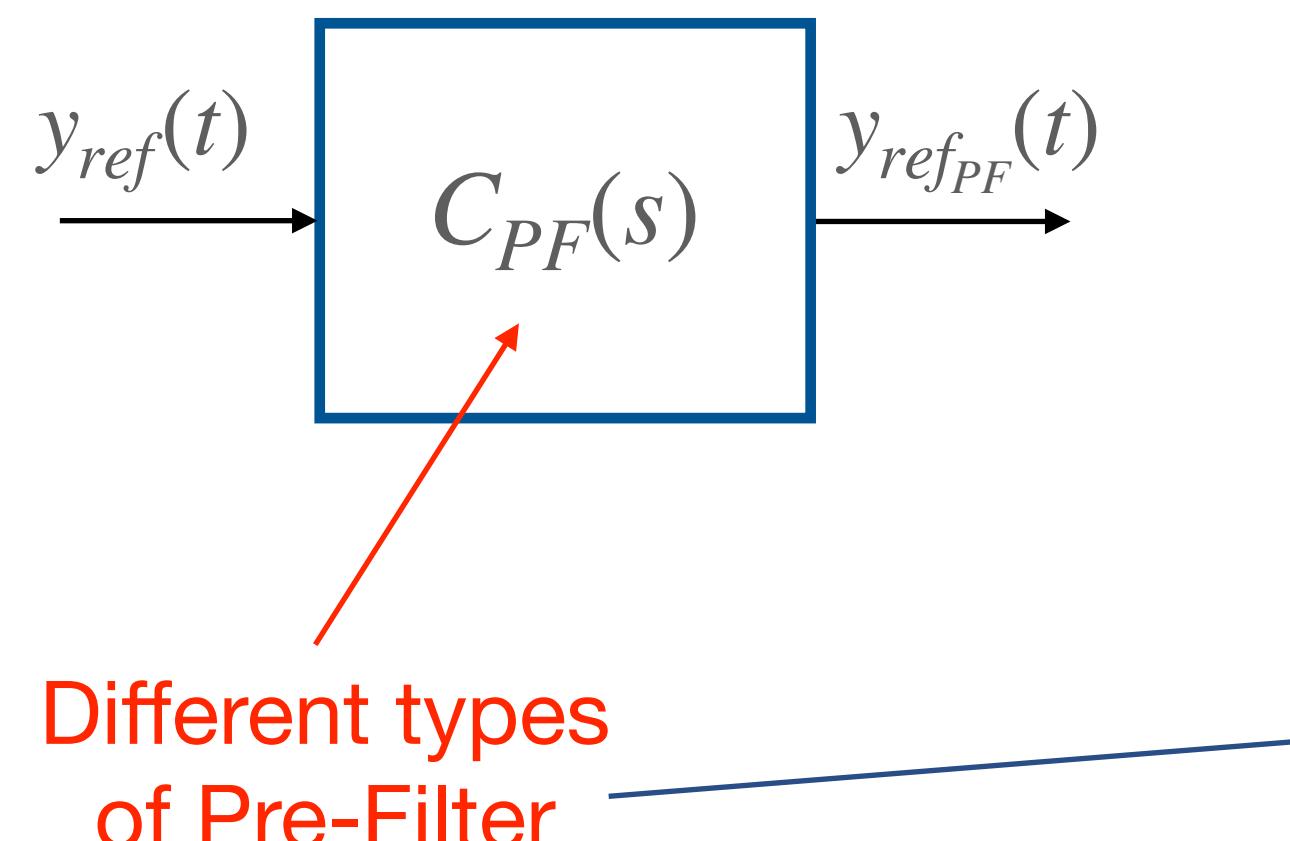


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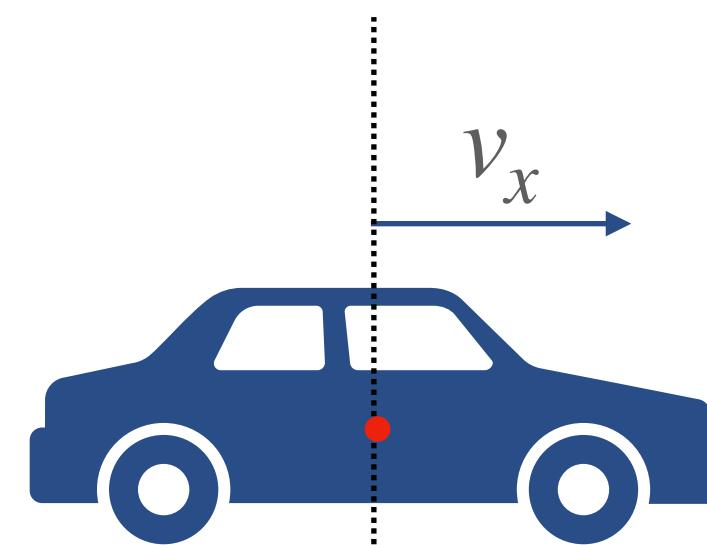
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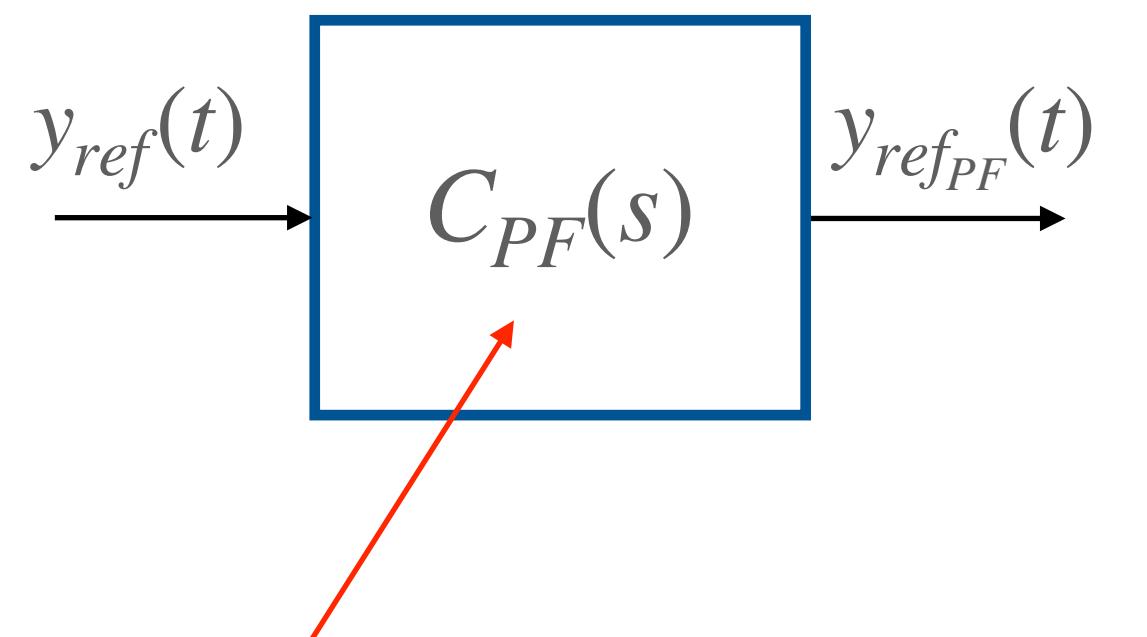
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$$\lambda := \frac{\omega r - v_x}{\max(\omega r, v_x)}$$



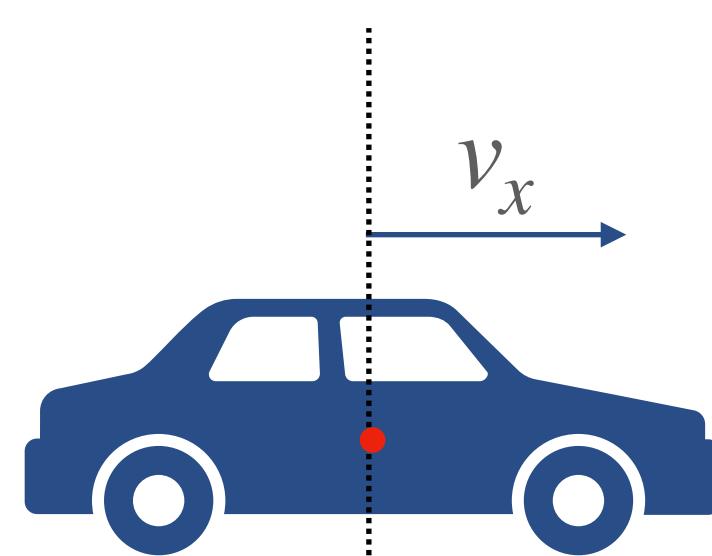
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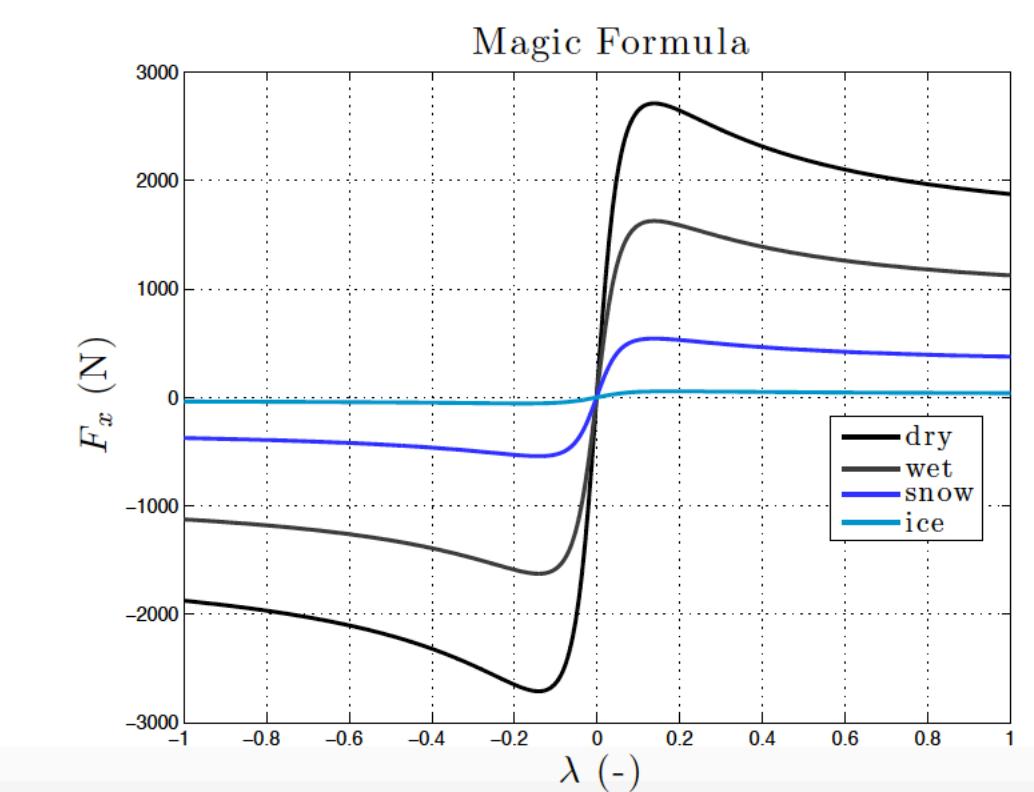
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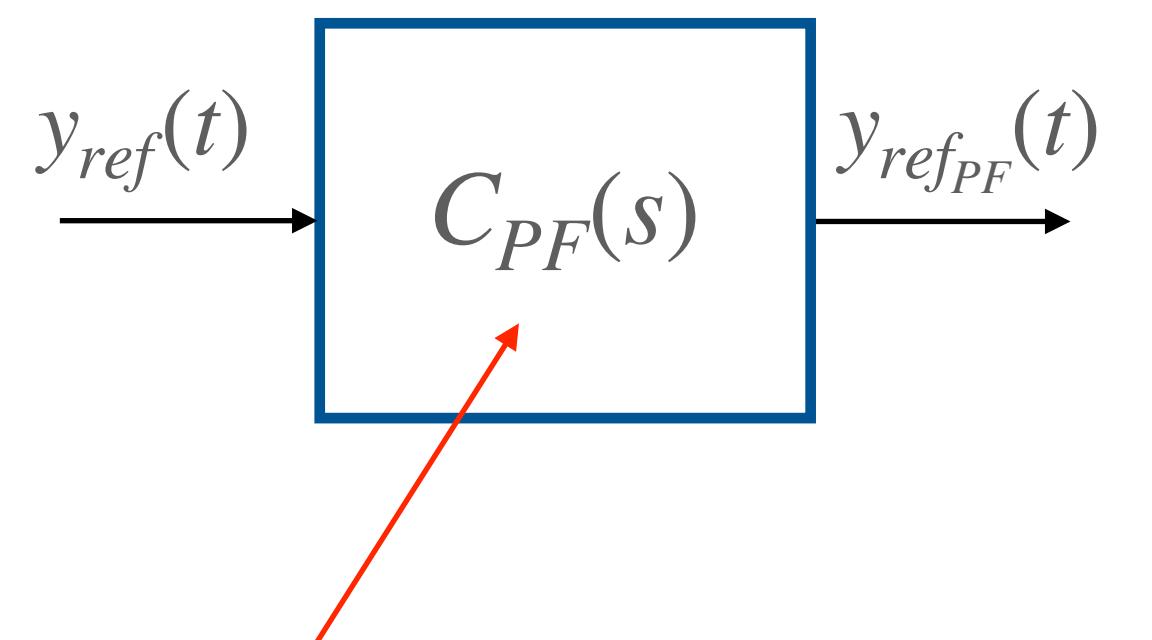
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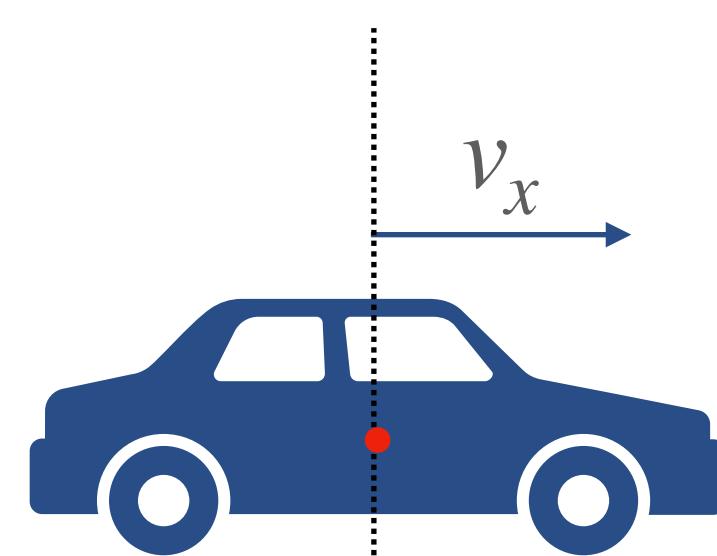
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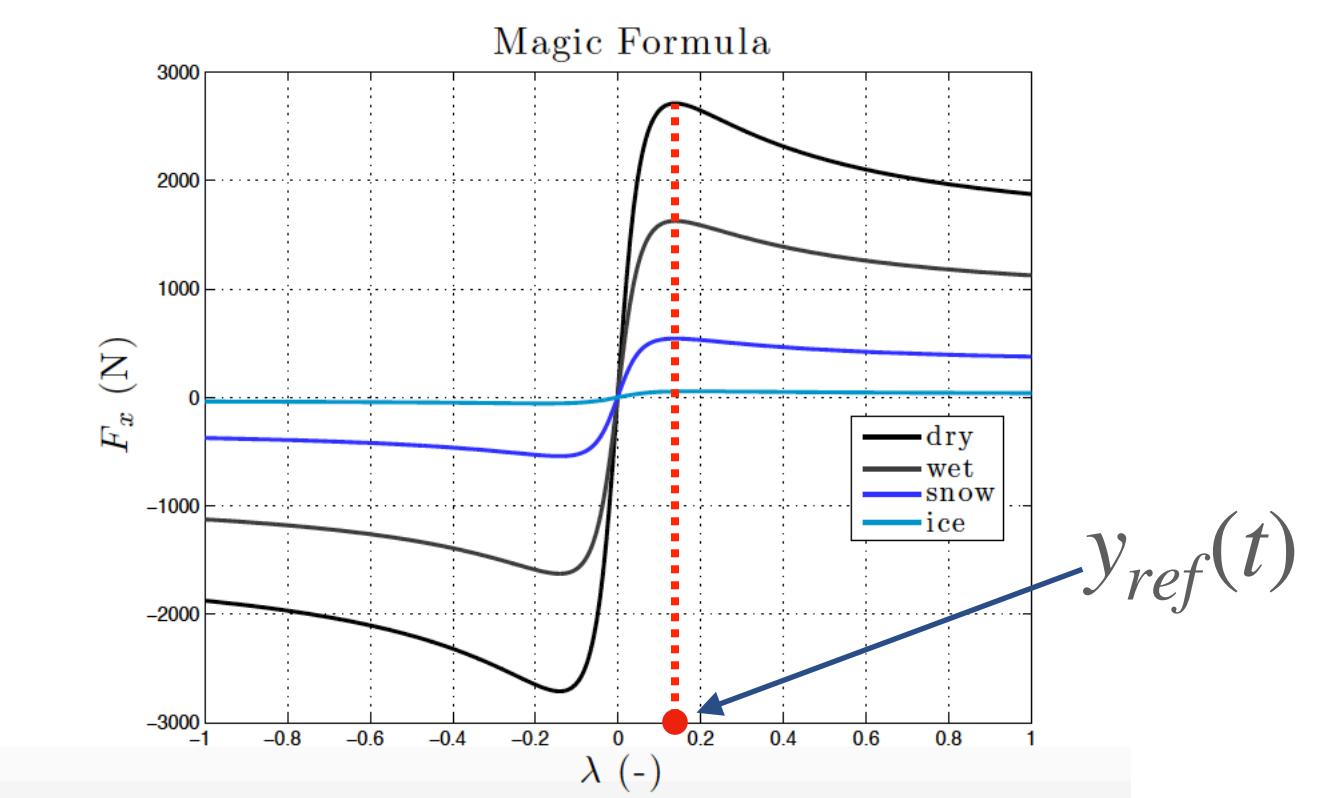
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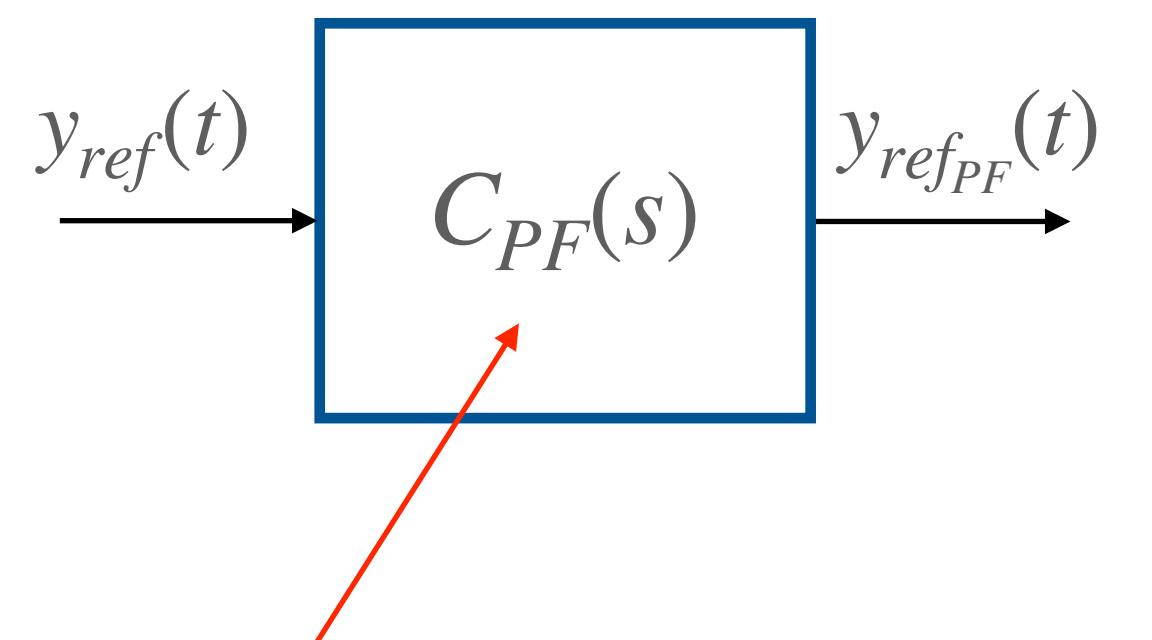
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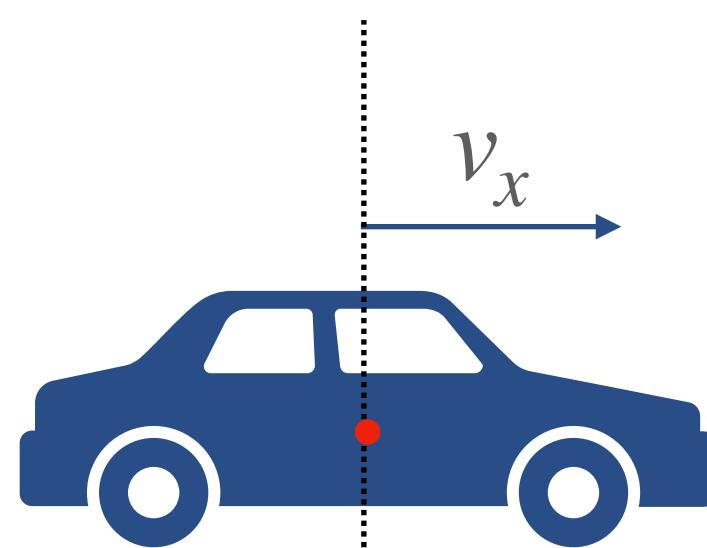
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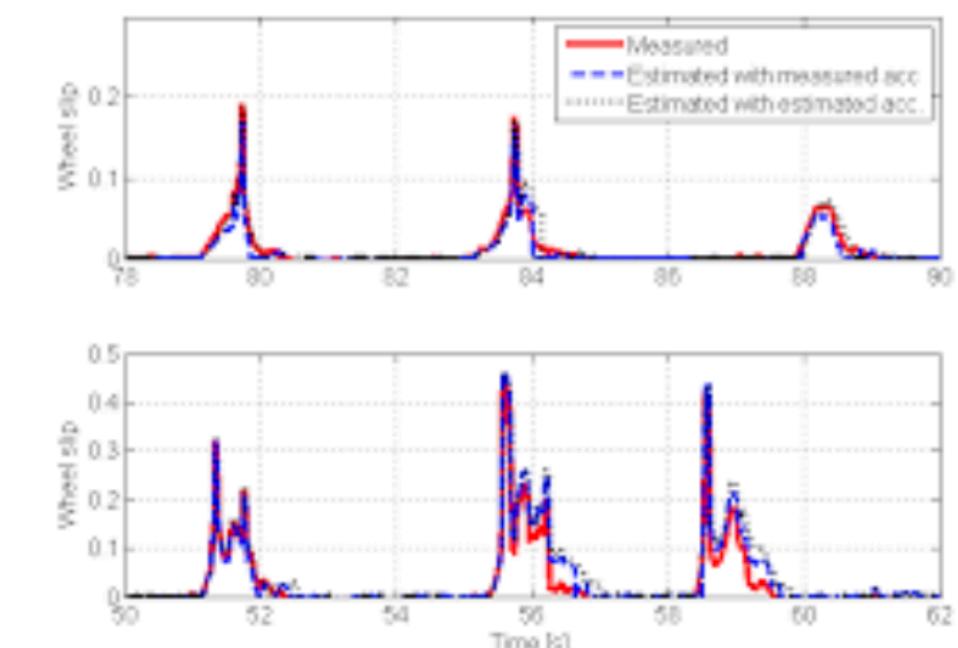
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Example:



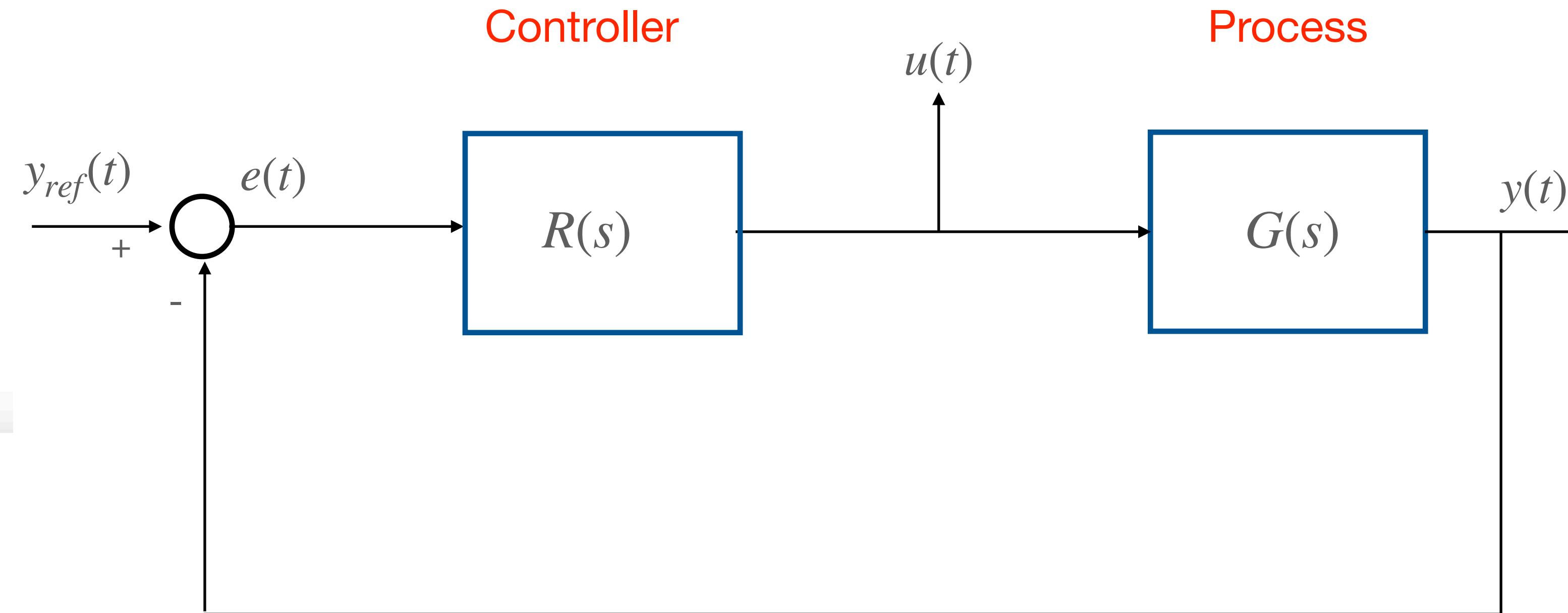
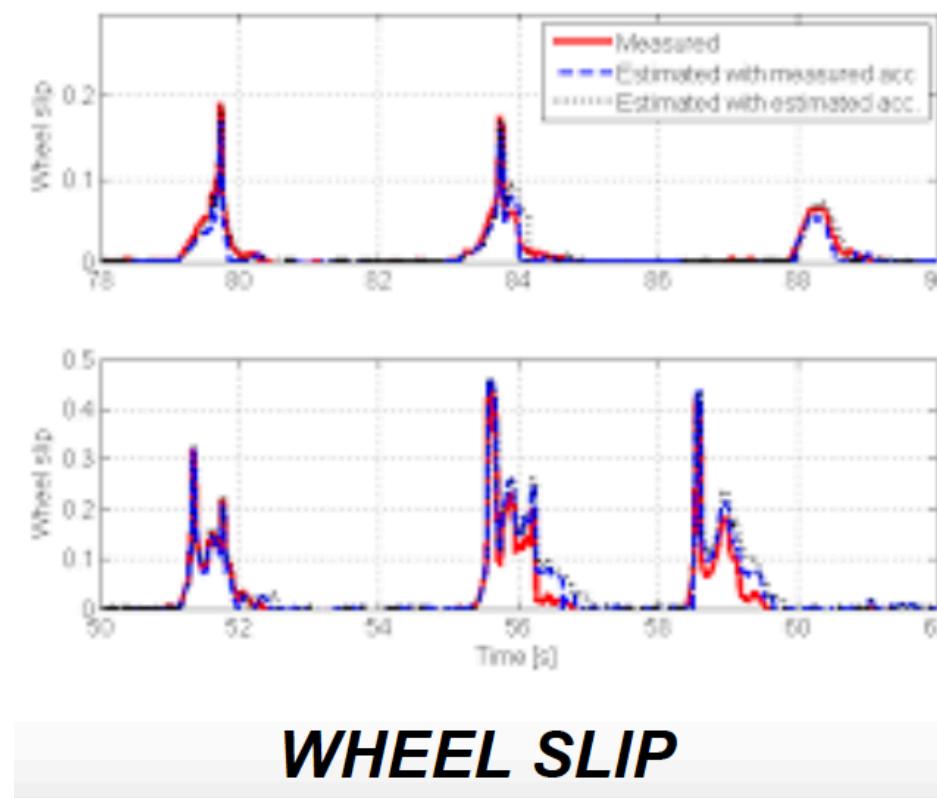
$$\lambda := \frac{\omega r - v_x}{\max(\omega r, v_x)}$$



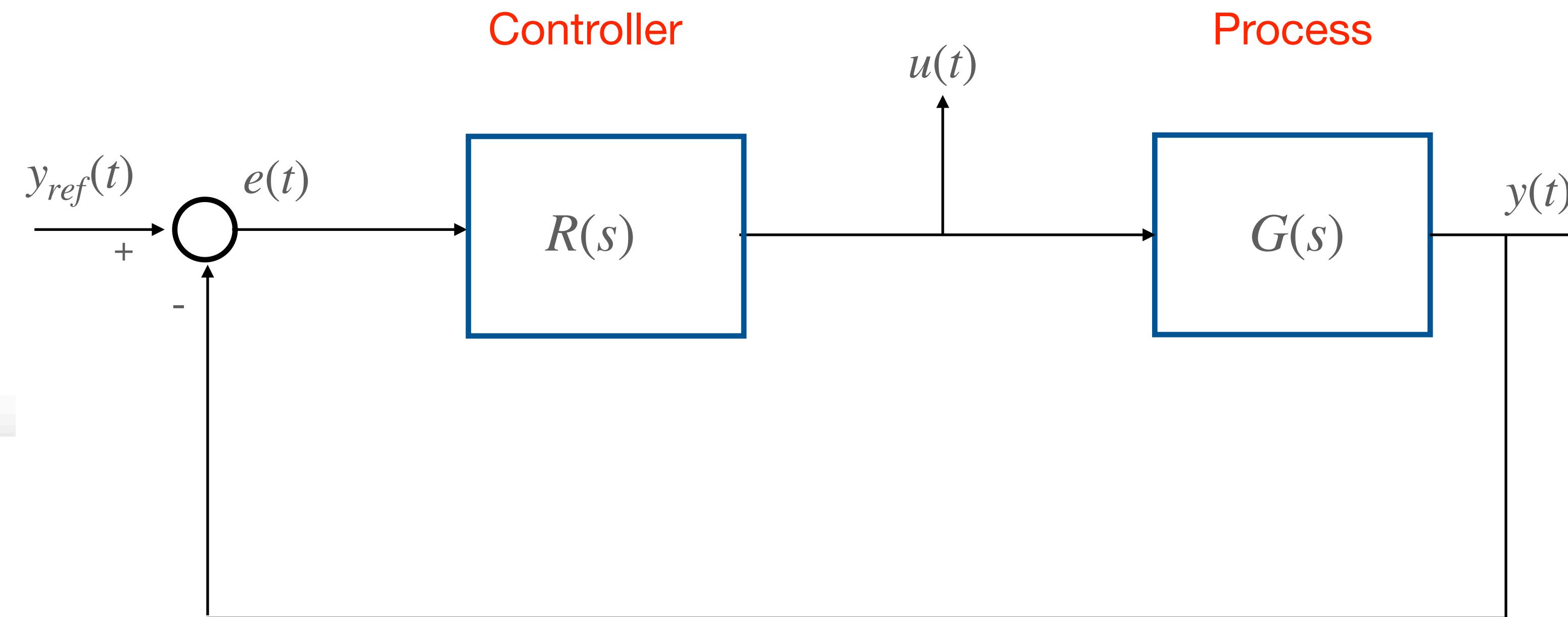
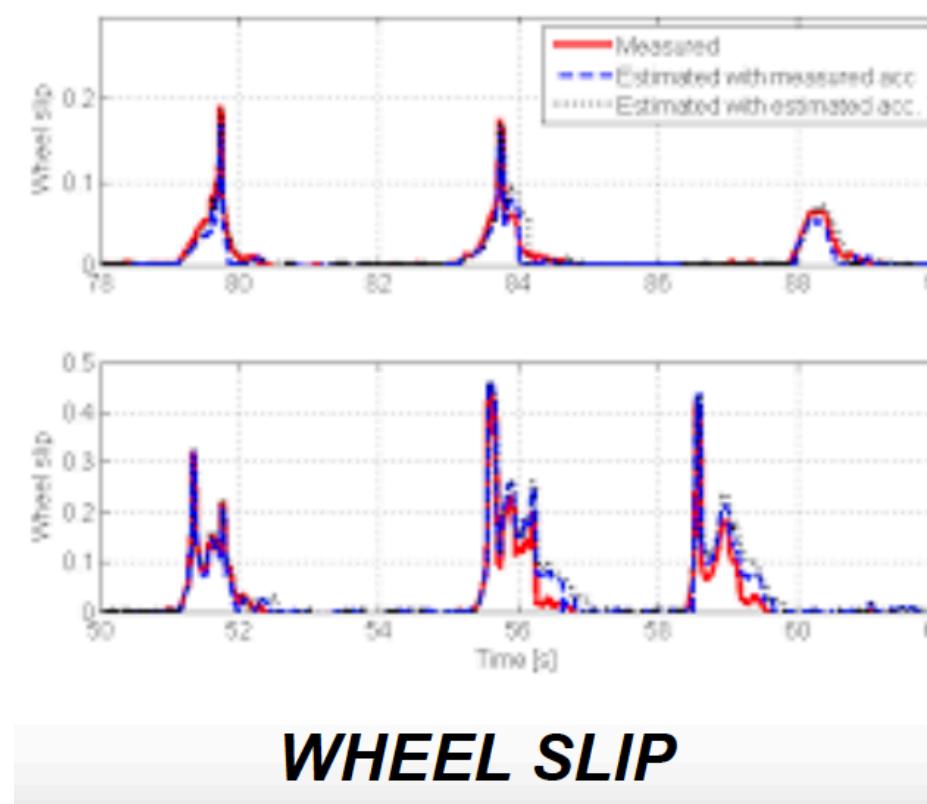
**WHEEL SLIP**



## Basic Control Scheme



## Basic Control Scheme

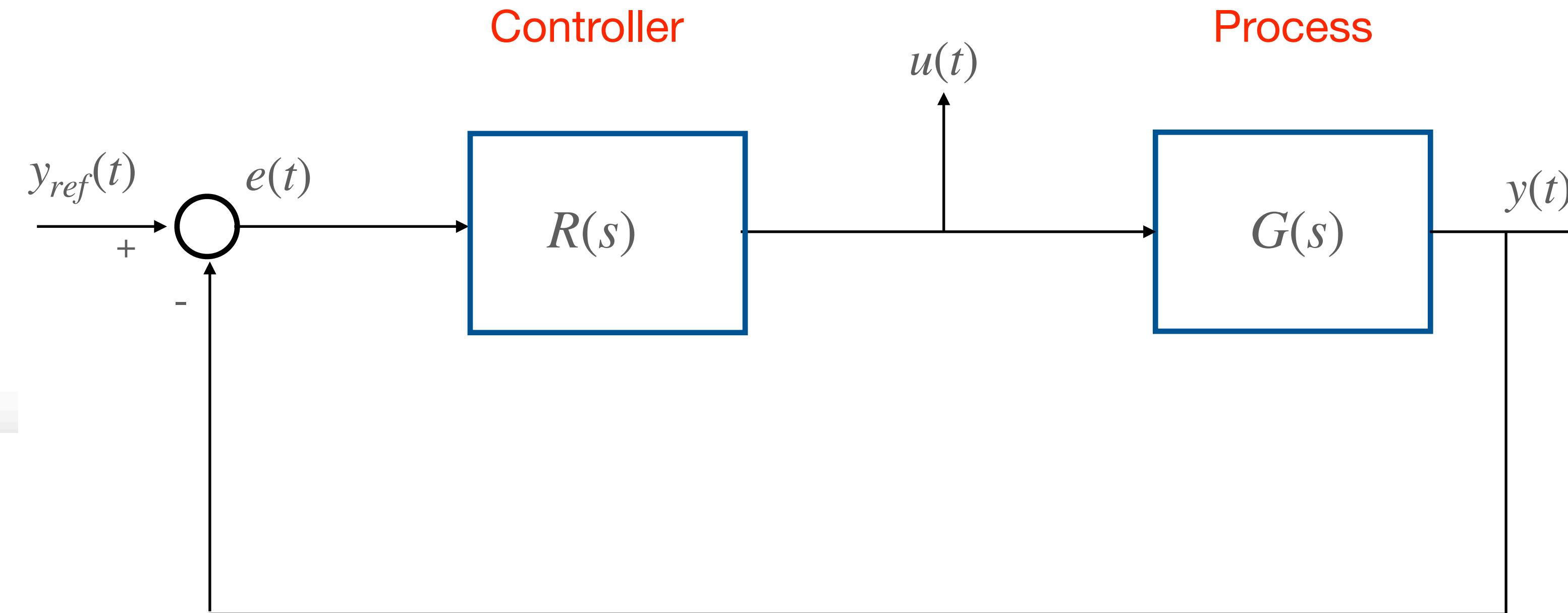
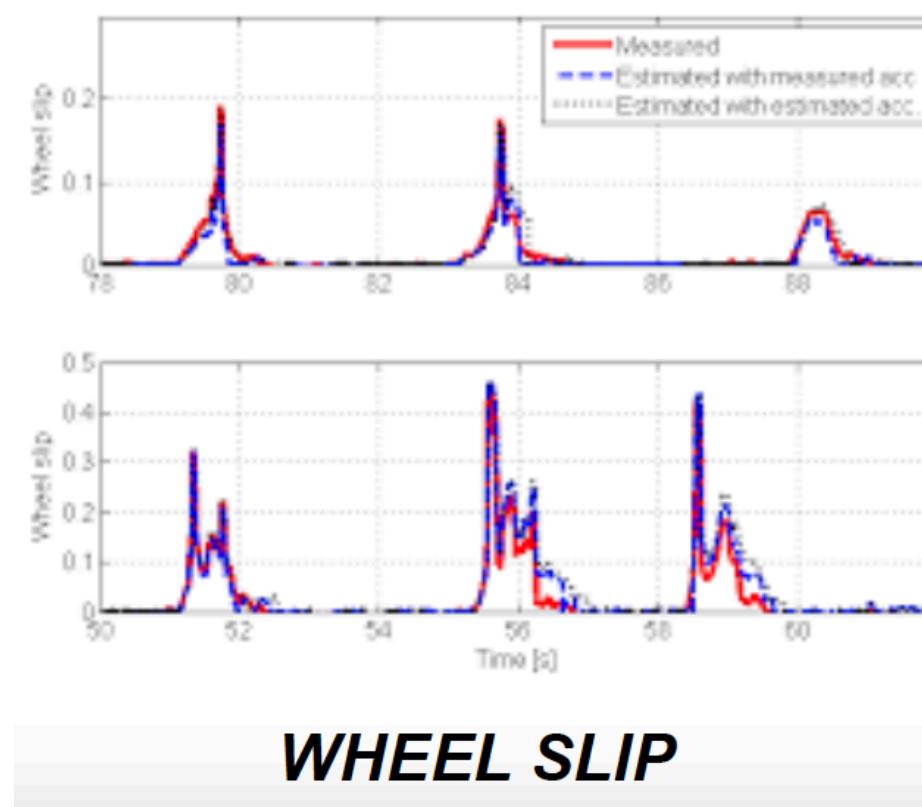


Block Algebra Rule:

$$y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$$



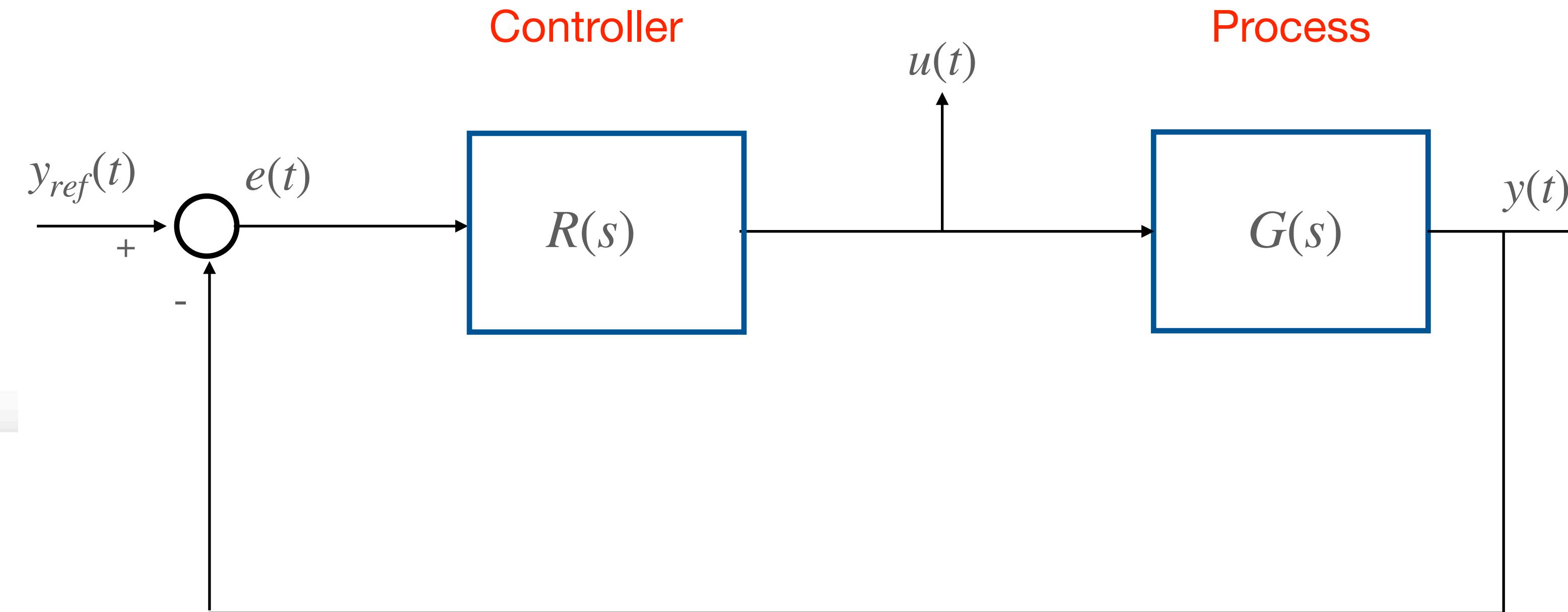
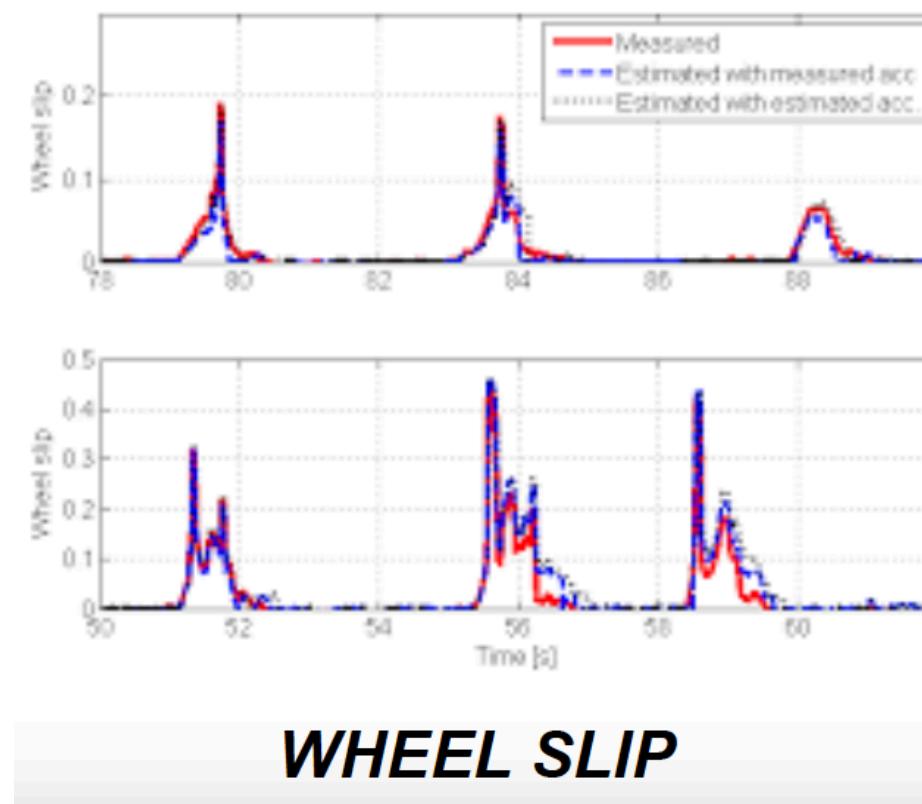
## Basic Control Scheme



Block Algebra Rule:  $y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$  Control Sensitivity Function



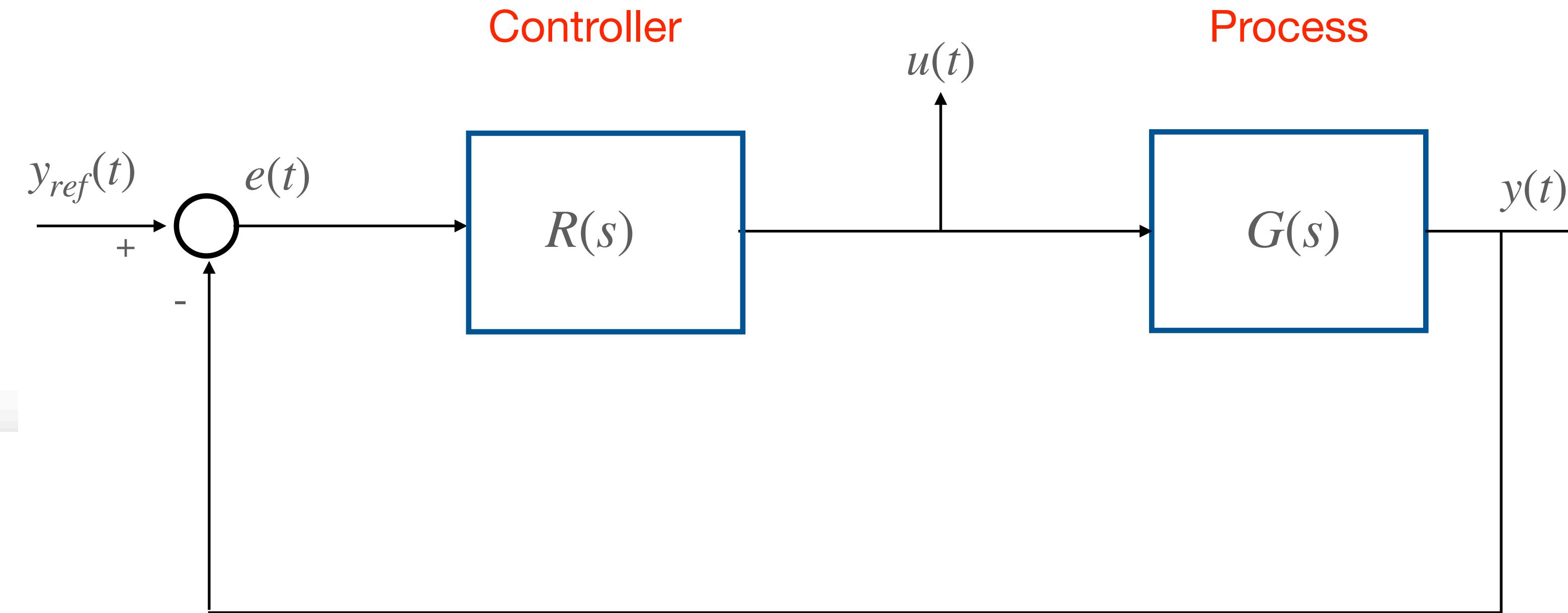
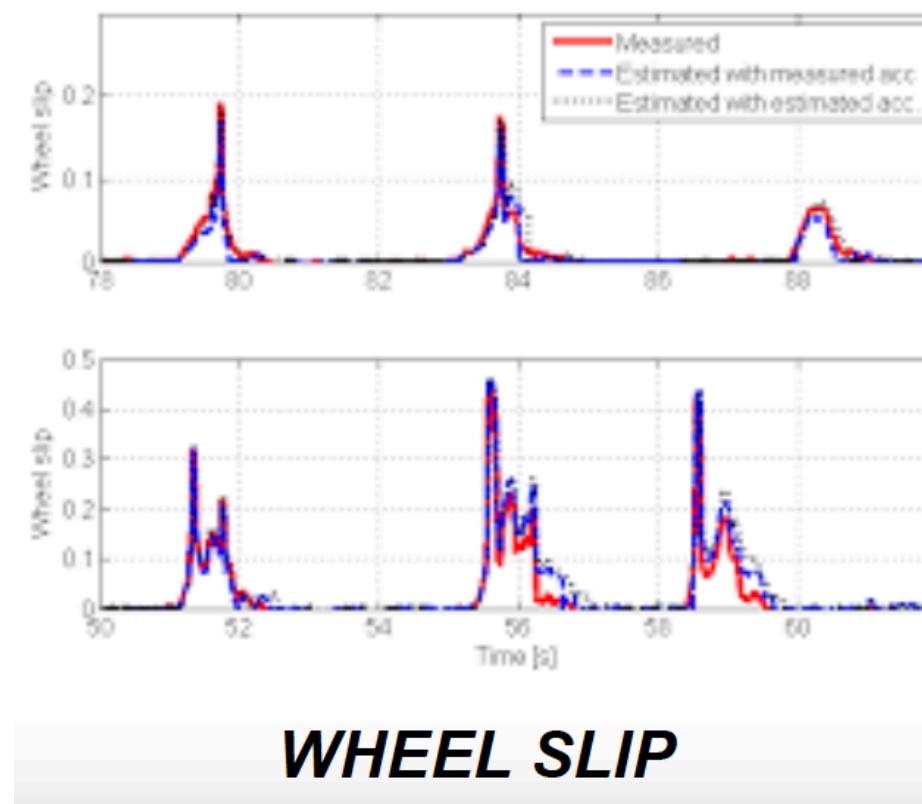
## Basic Control Scheme



Block Algebra Rule:  $y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$  Control Sensitivity Function

Recommendation:  $|Q(j\omega)|$  sufficiently low  $\forall \omega \in \mathbb{R}_+ \cup \{0\}$

## Basic Control Scheme



Block Algebra Rule:

$$y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$$

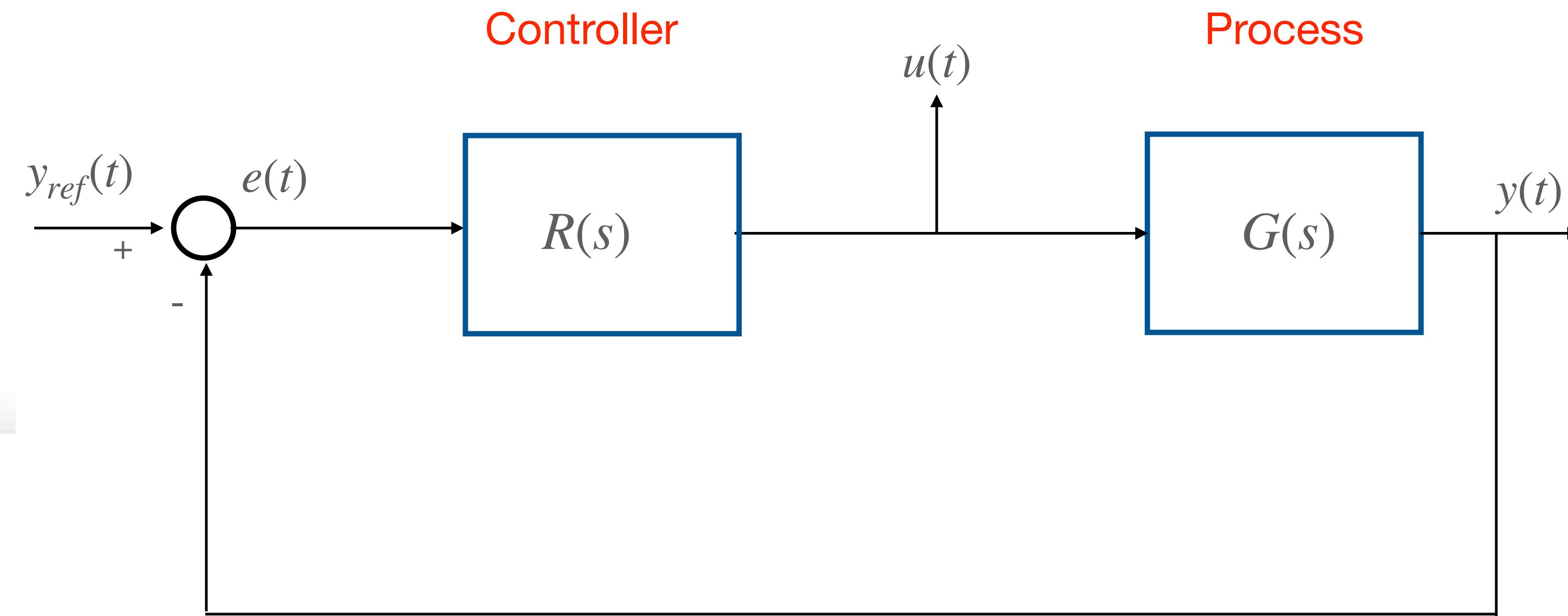
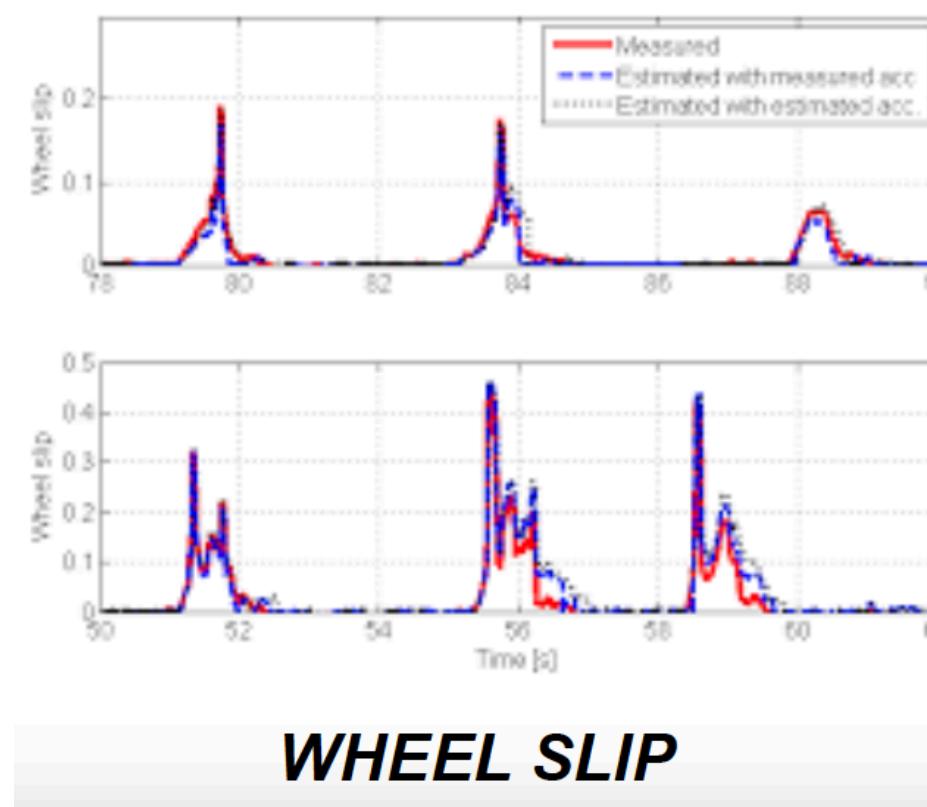
Control Sensitivity Function

Recommendation:  $|Q(j\omega)|$  sufficiently low  $\forall \omega \in \mathbb{R}_{+} \cup \{0\}$

LF:  $|Q(j\omega)| \rightarrow \left| \frac{1}{G(j\omega)} \right|$



## Basic Control Scheme



**Block Algebra Rule:**  $y_{ref} \rightarrow u : Q(s) = \frac{R(s)}{1 + R(s)G(s)}$  **Control Sensitivity Function**

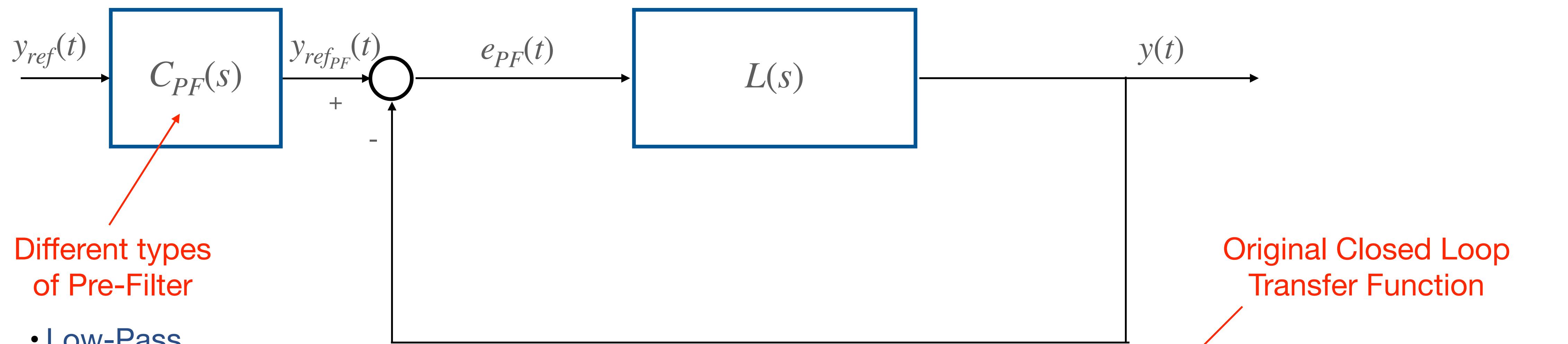
Recommendation:  $|Q(j\omega)|$  sufficiently low  $\forall \omega \in \mathbb{R}_{+} \cup \{0\}$

LF:  $|Q(j\omega)| \rightarrow \left| \frac{1}{G(j\omega)} \right|$

HF:  $|Q(j\omega)| \rightarrow |R(j\omega)|$



## Pre-filter Based Control Scheme: Case 2



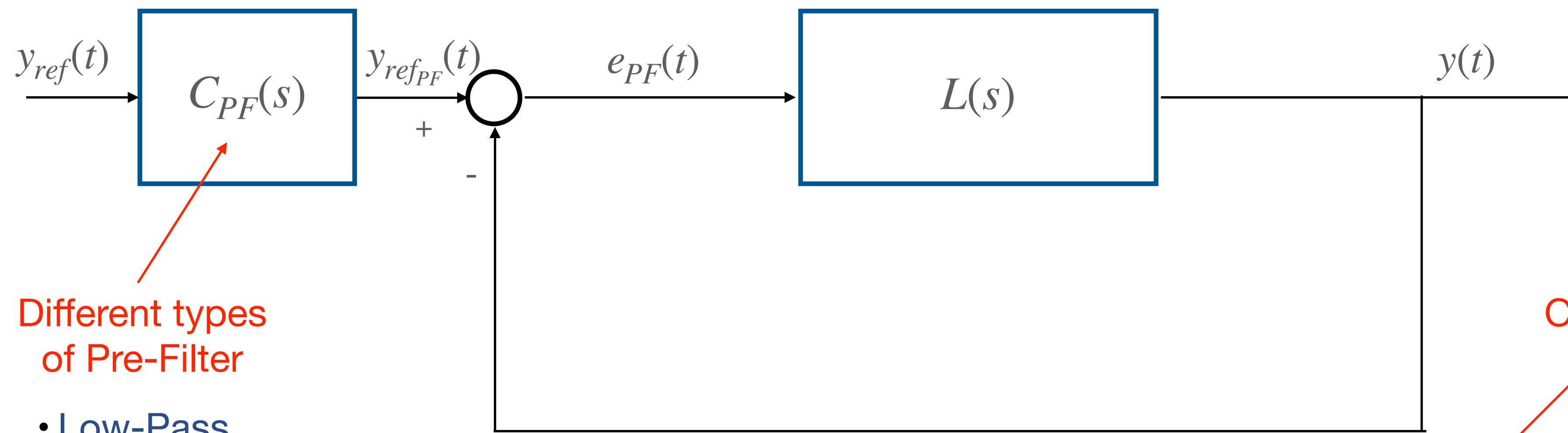
Block Algebra Rule:

$$F(s) = \frac{R(s) \cdot G(s)}{1 + R(s) \cdot G(s)} = \frac{L(s)}{1 + L(s)}$$

$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



## Pre-filter Based Control Scheme: Case 2



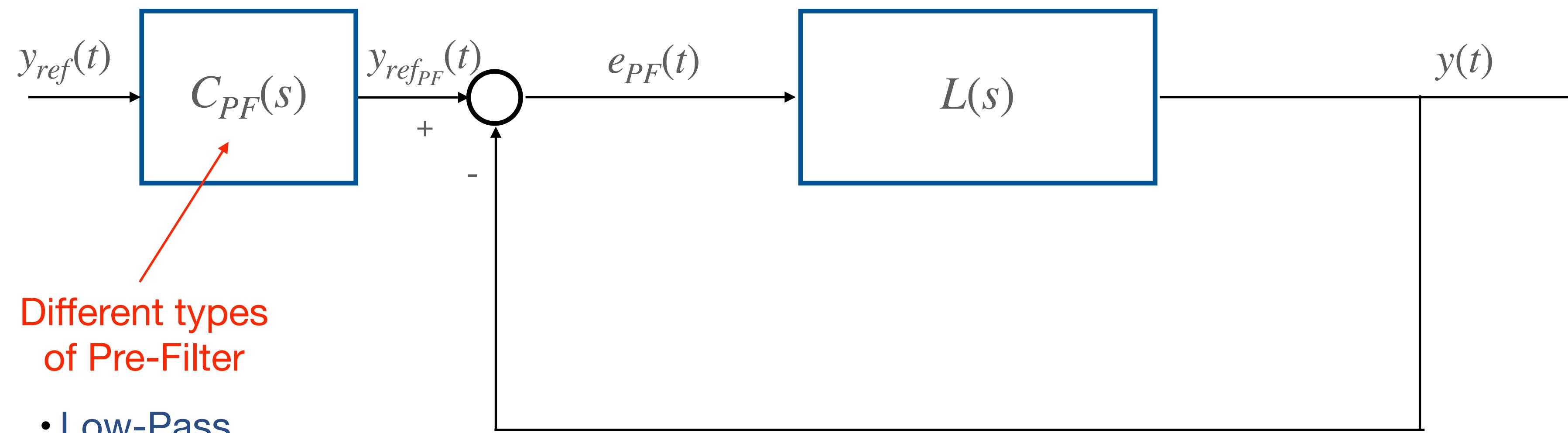
Block Algebra Rule:

$$F(s) = \frac{R(s) \cdot G(s)}{1 + R(s) \cdot G(s)} = \frac{L(s)}{1 + L(s)}$$

$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



## Pre-filter Based Control Scheme: Case 2



Assumptions:

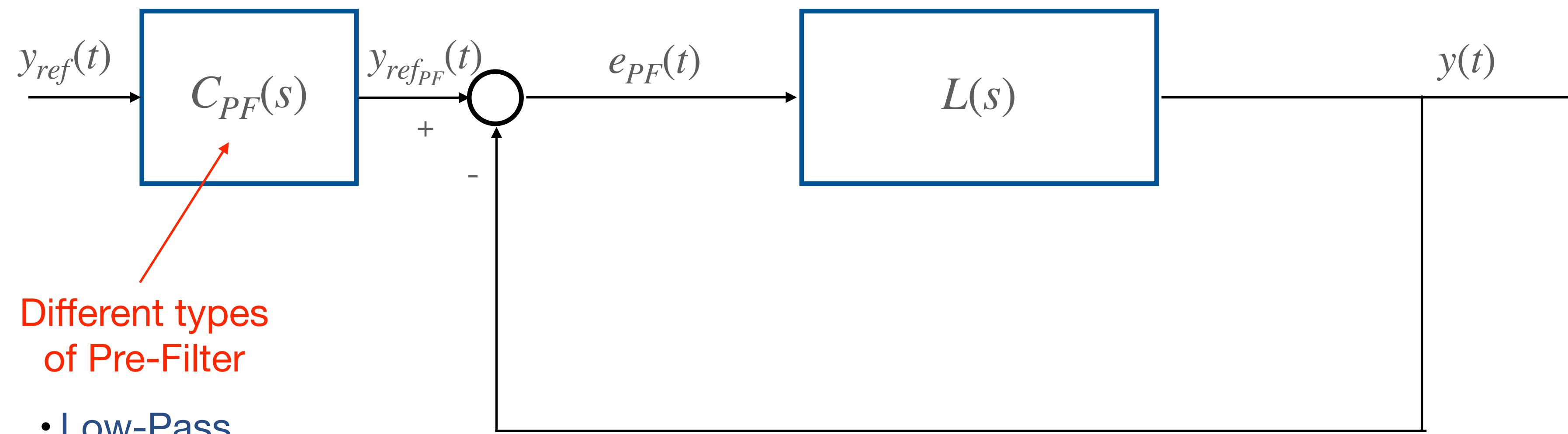
- $C_{PF}(s)$  As. Stable
- Proper
- Unitary gain

The pole precedes the zero on the  $\omega$  axis

$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



## Pre-filter Based Control Scheme: Case 2

**Assumptions:**

- $C_{PF}(s)$  As. Stable
- Proper
- Unitary gain

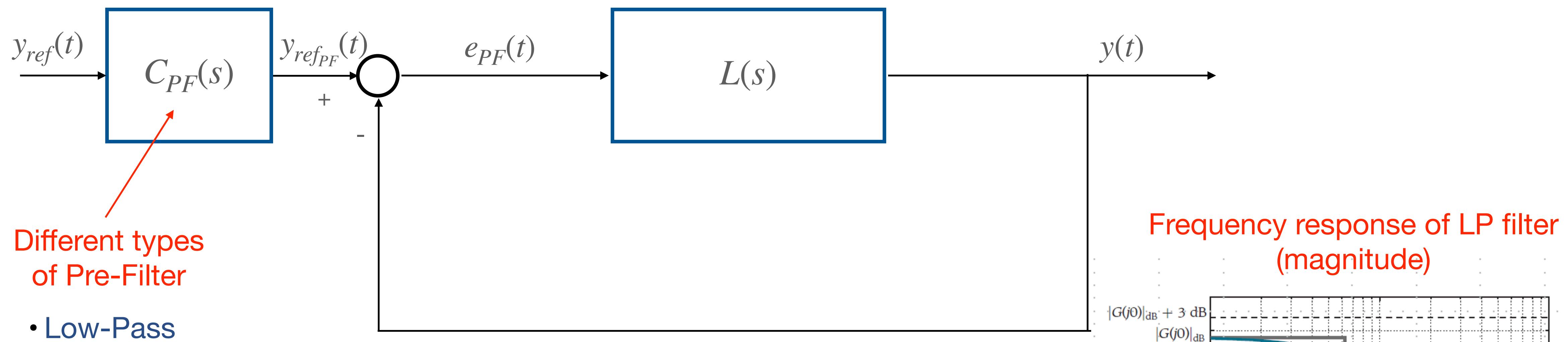
$$\lim_{s \rightarrow 0} \frac{\mu(1 + \tau' s)}{1 + \tau s} = \mu = 1$$

The pole precedes the zero on the  $\omega$  axis

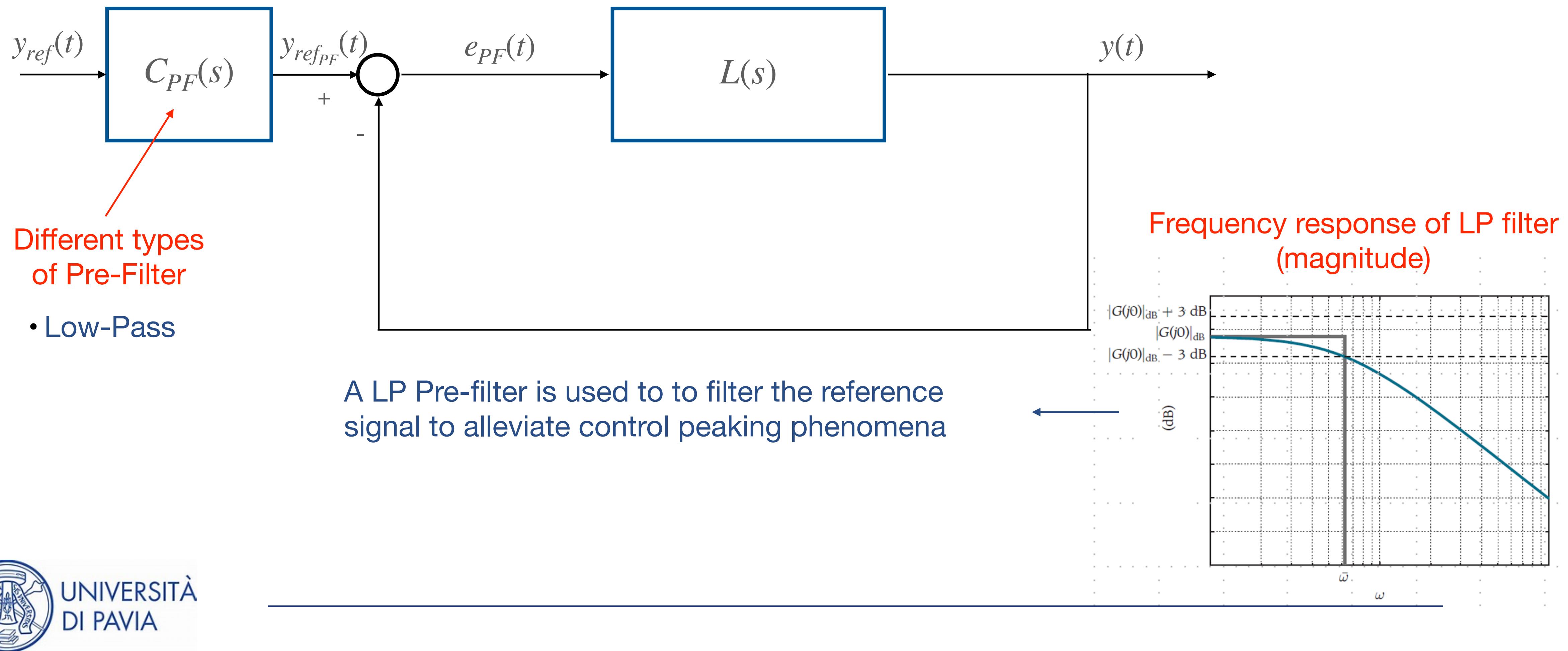
$$F_{PF}(s) = C_{PF}(s) \cdot F(s) = C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = \frac{\mu(1 + \tau' s)}{(1 + \tau s)} \cdot \frac{L(s)}{1 + L(s)}$$



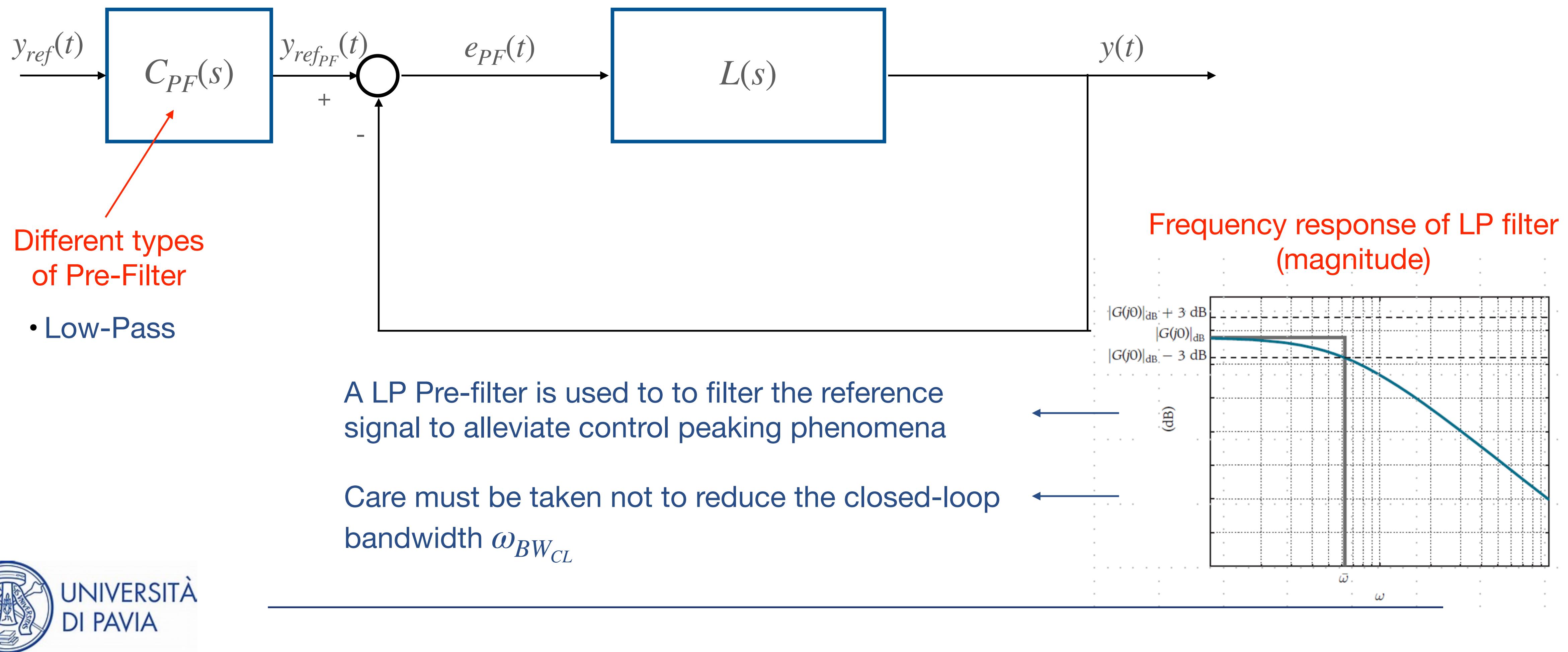
## Pre-filter Based Control Scheme: Case 2



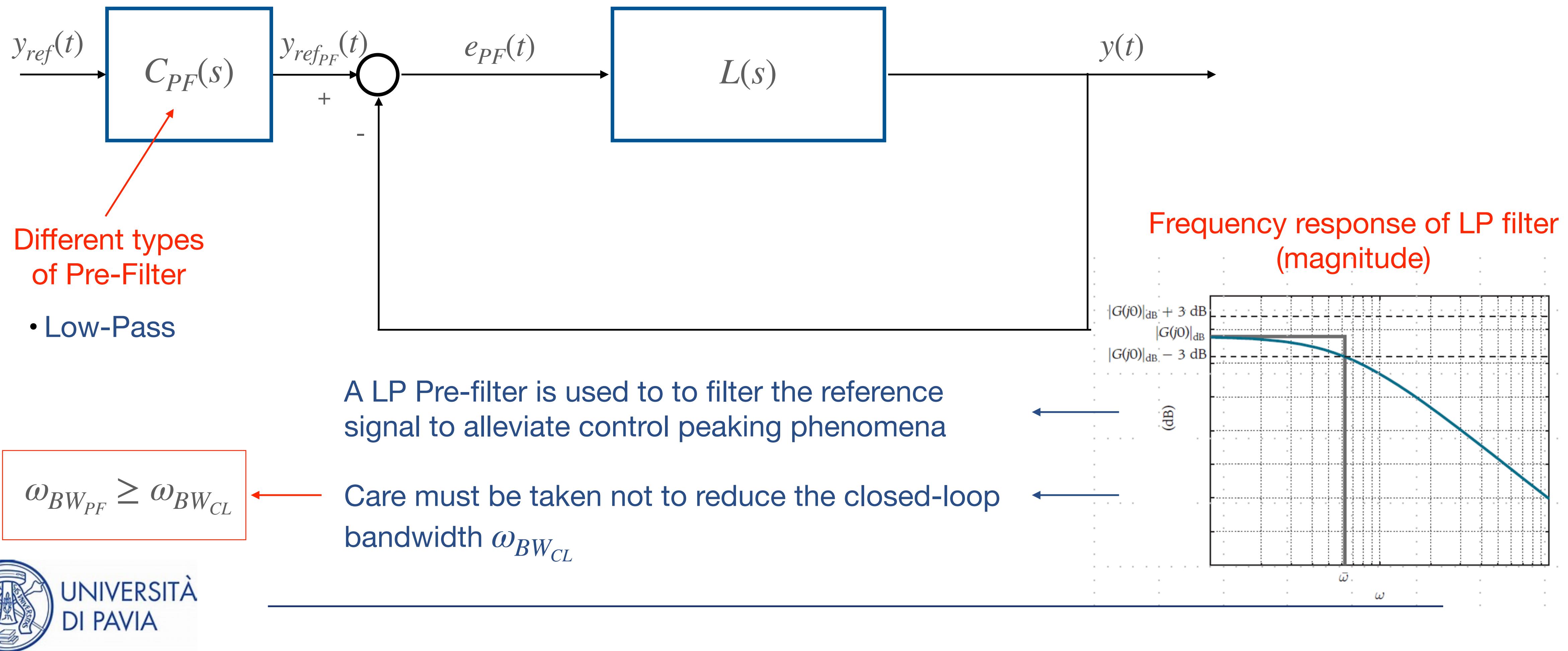
## Pre-filter Based Control Scheme: Case 2



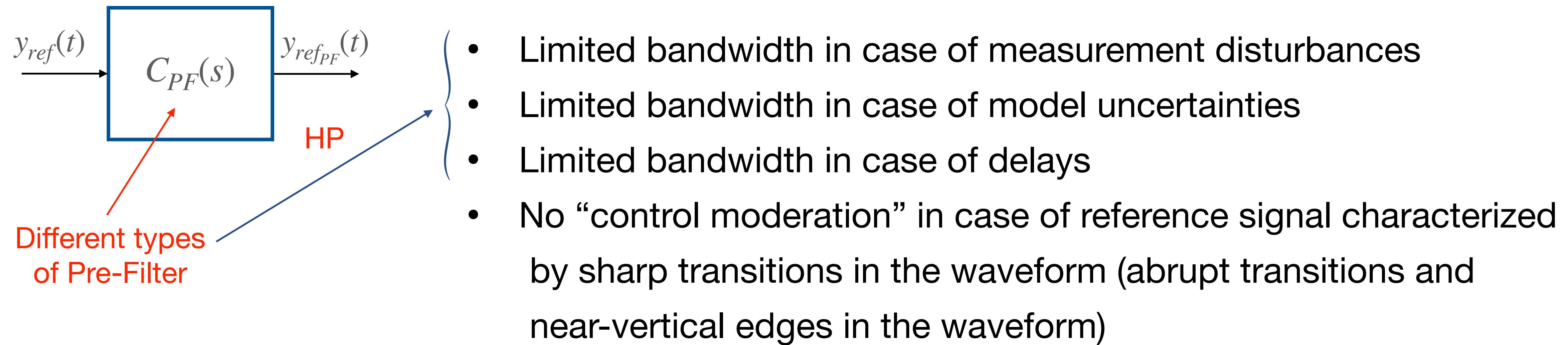
## Pre-filter Based Control Scheme: Case 2



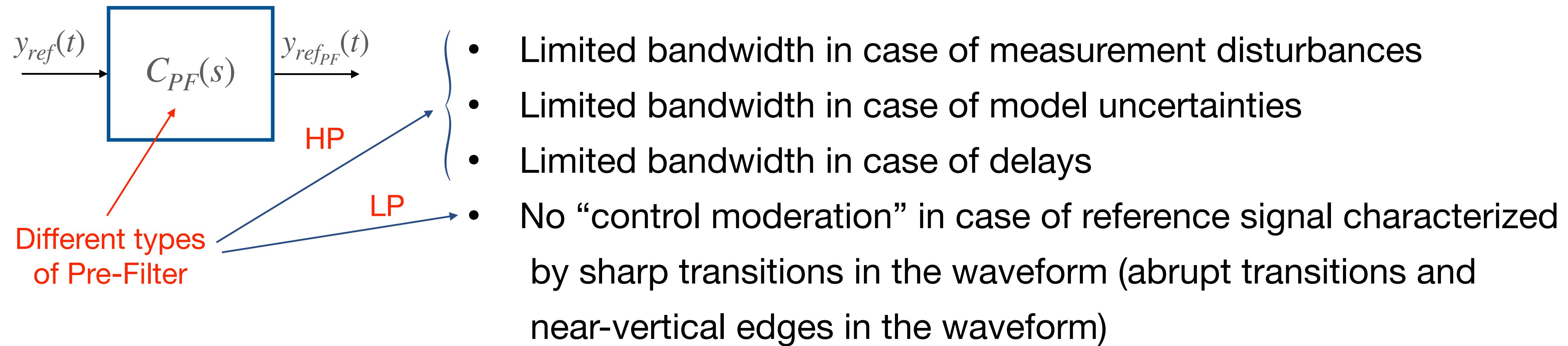
## Pre-filter Based Control Scheme: Case 2



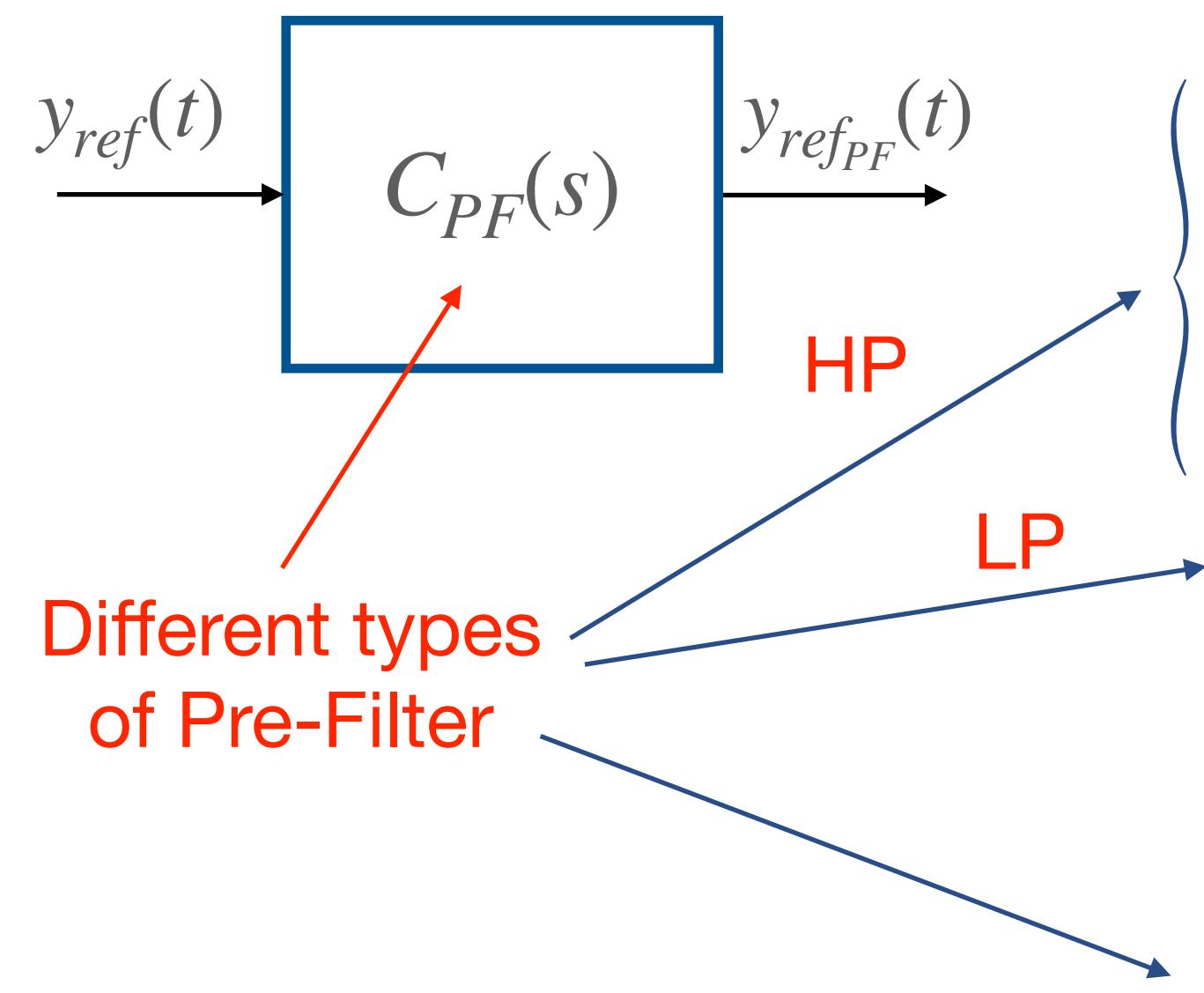
## Pre-filter Based Control Scheme: Case 3



## Pre-filter Based Control Scheme: Case 3



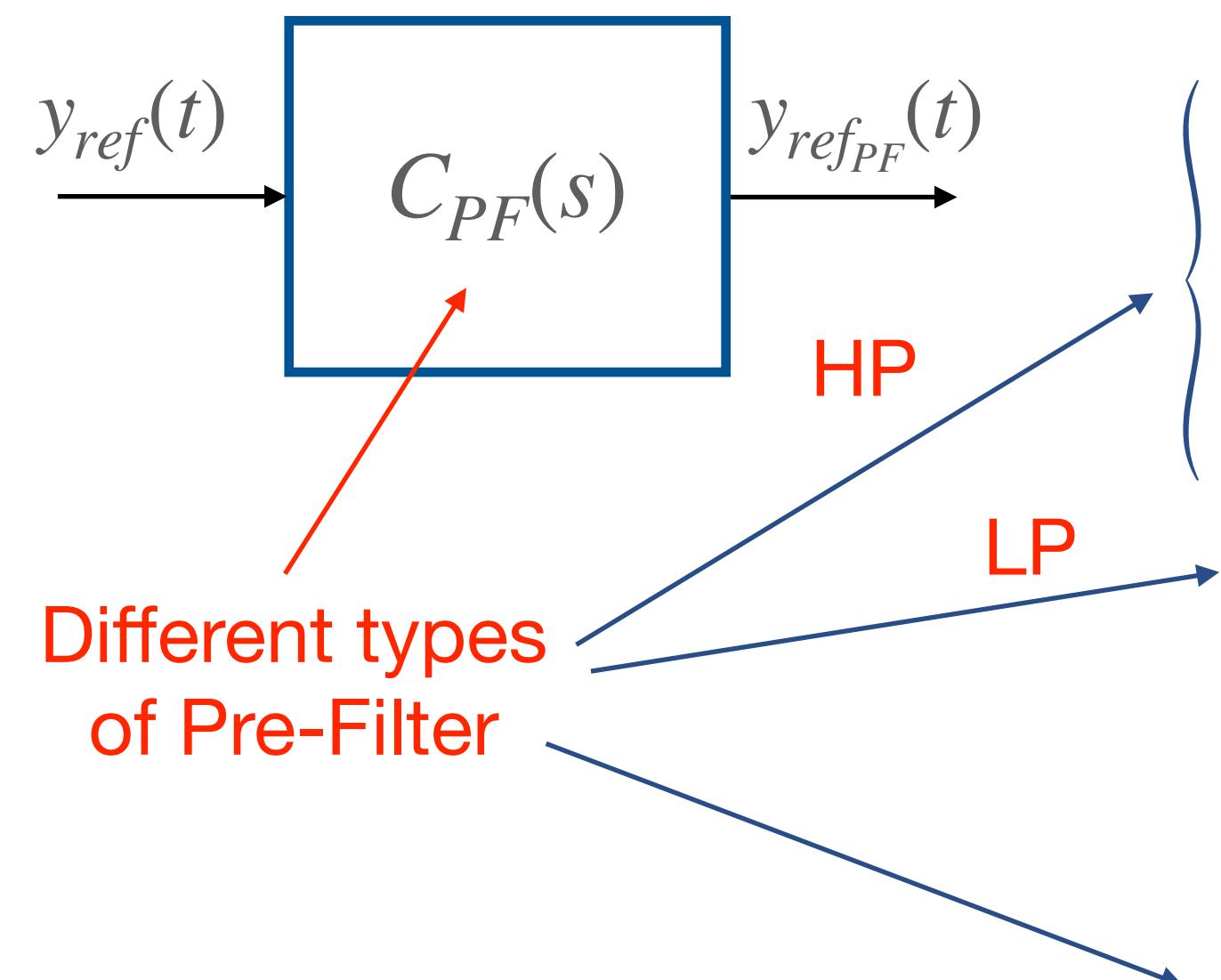
## Pre-filter Based Control Scheme: Case 3



- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays
- No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
- Possible difficulties in enforcing steady-state precision with step reference signal via the classical approach



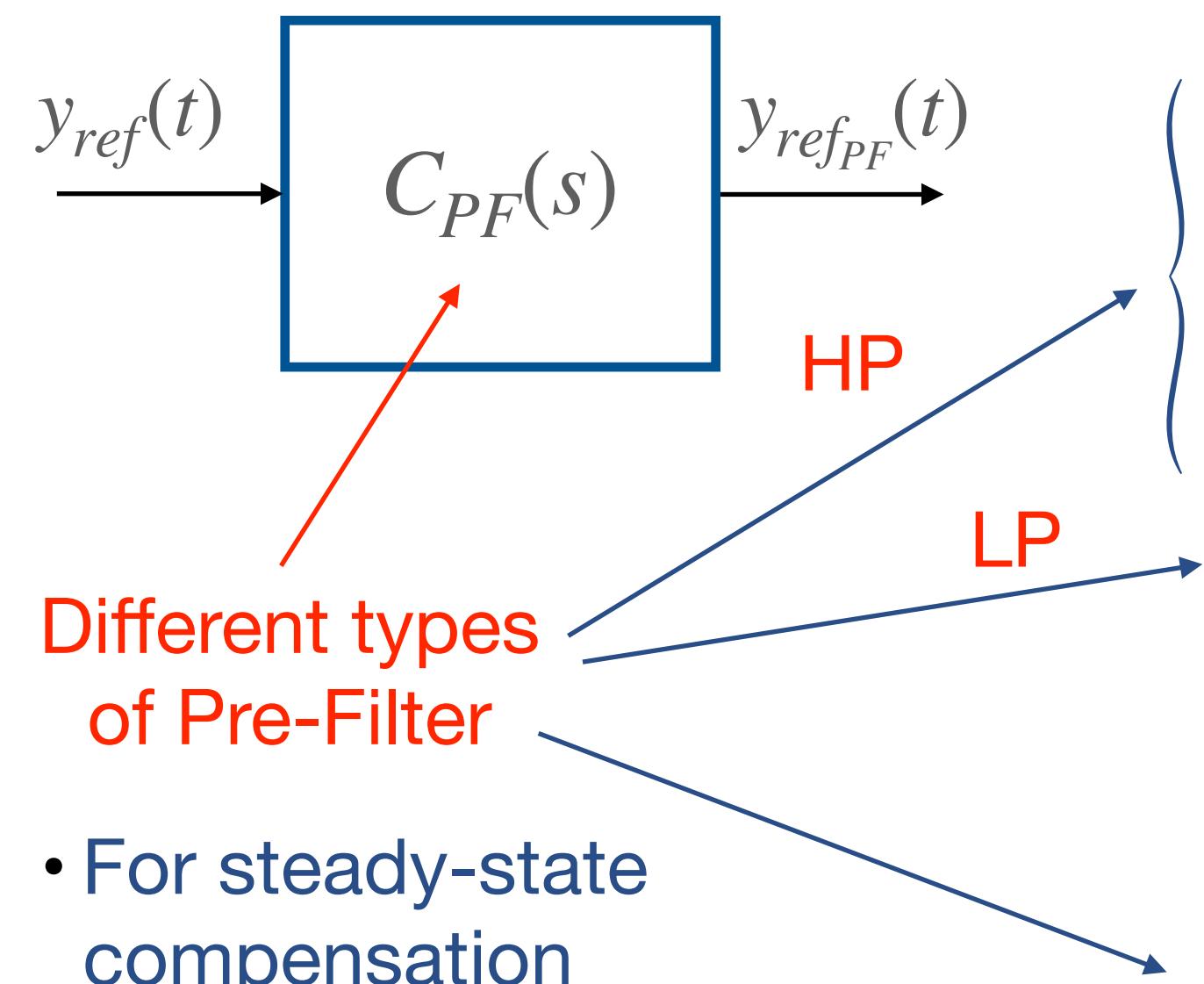
## Pre-filter Based Control Scheme: Case 3



- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays
- No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
- Possible difficulties in enforcing steady-state precision with step reference signal via the classical approach  
(by introducing integrators in the direct path)



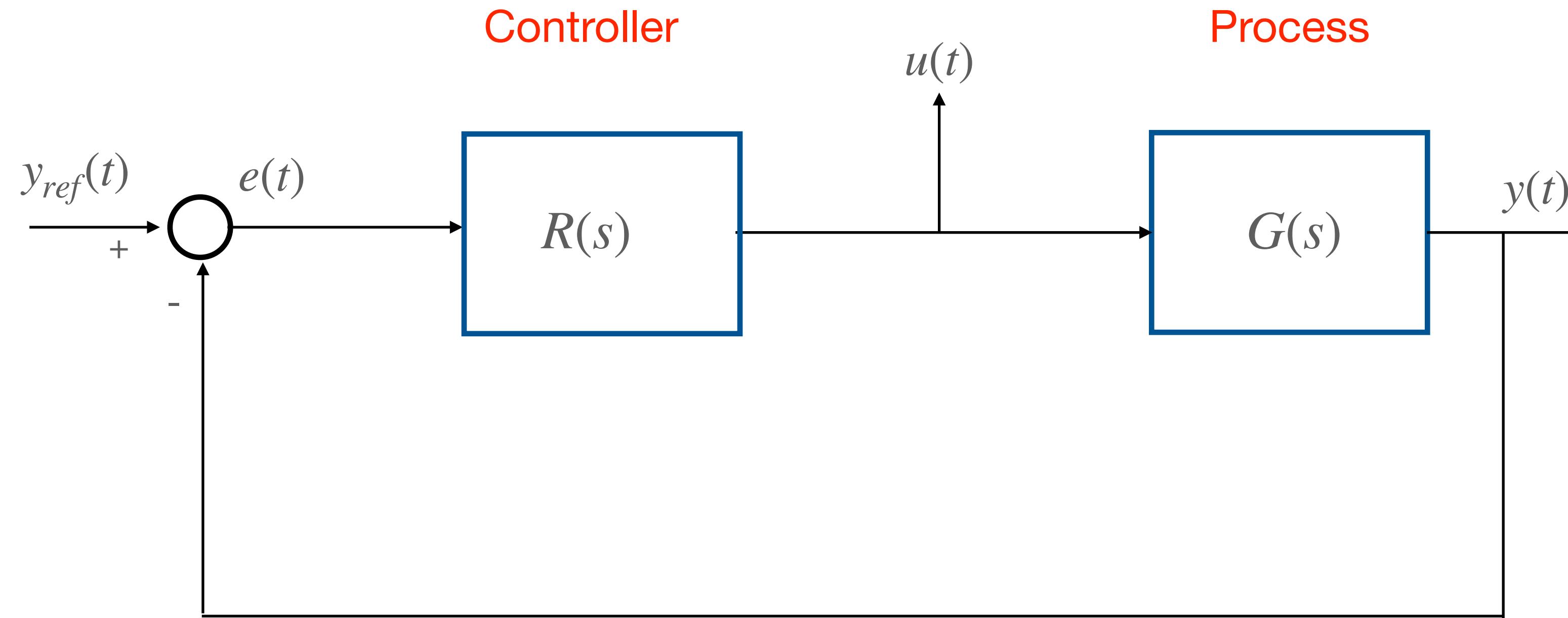
## Pre-filter Based Control Scheme: Case 3



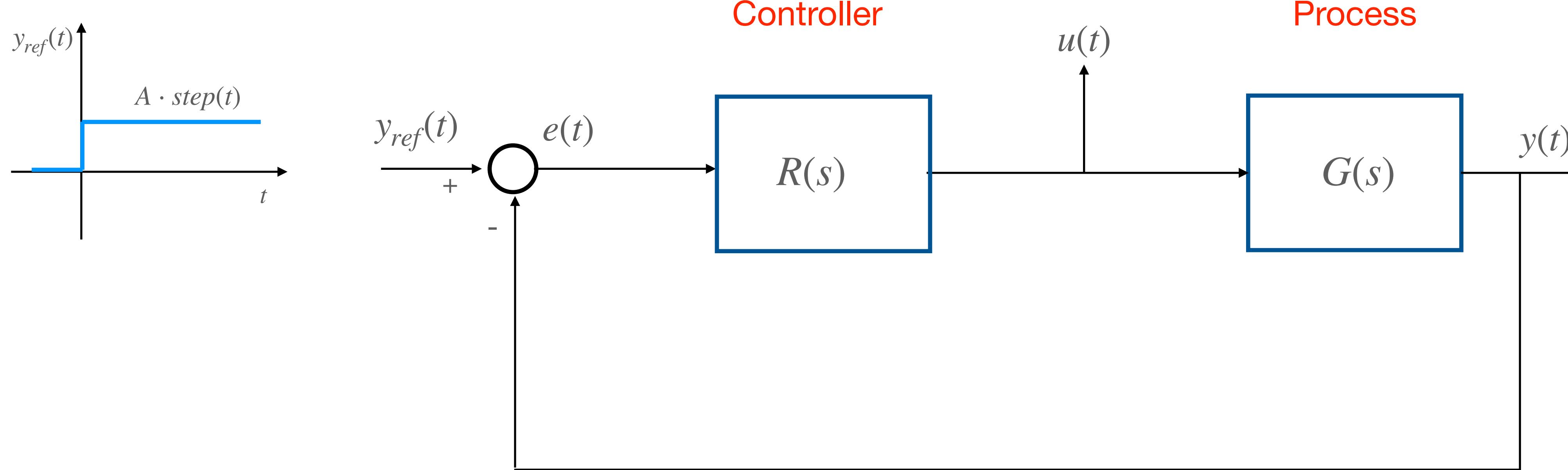
- Limited bandwidth in case of measurement disturbances
  - Limited bandwidth in case of model uncertainties
  - Limited bandwidth in case of delays
  - No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
  - Possible difficulties in enforcing steady-state precision with step reference signal via the classical approach (by introducing integrators in the direct path)
- For steady-state compensation



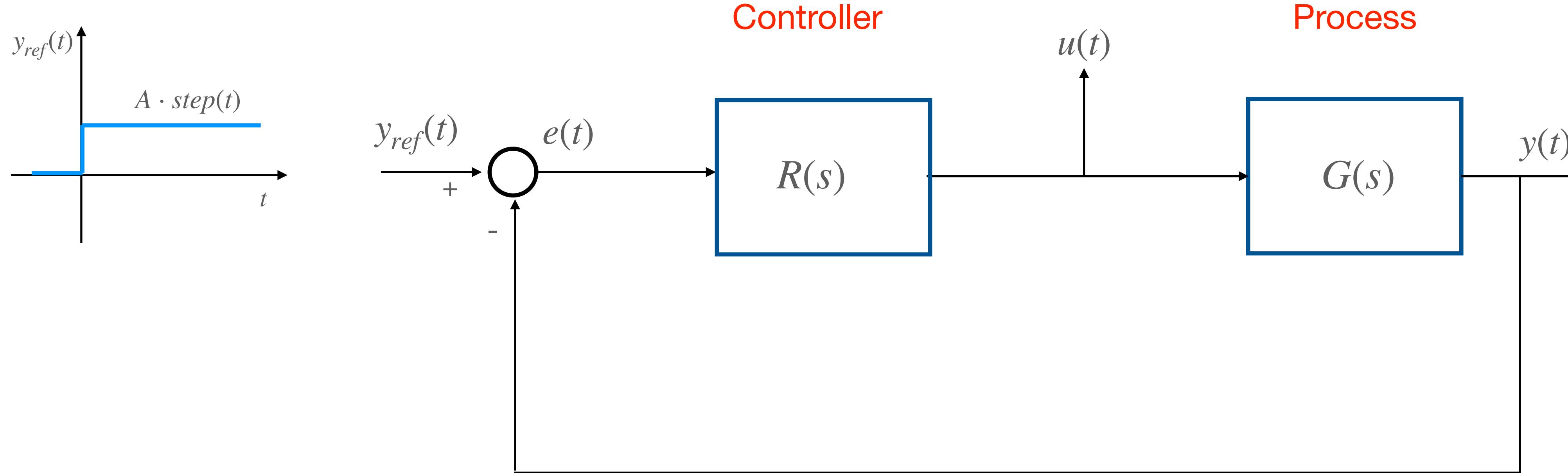
## Basic Control Scheme



## Basic Control Scheme



## Basic Control Scheme

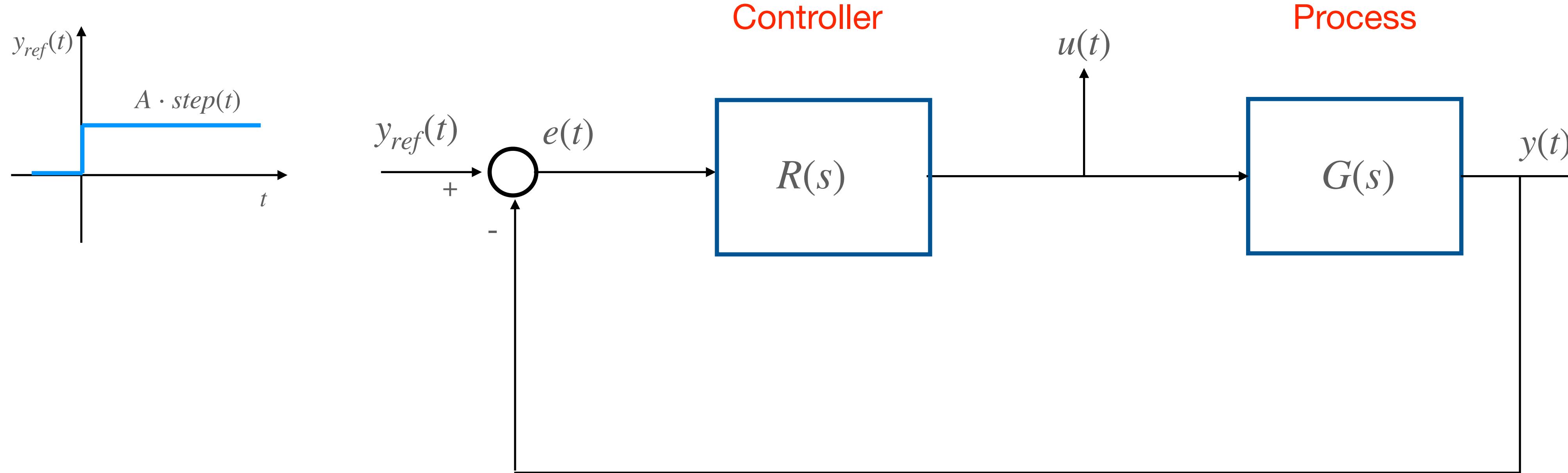


Block Algebra Rule:  $y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$

$$y \approx y_{ref} : F(0) = 1$$



## Basic Control Scheme

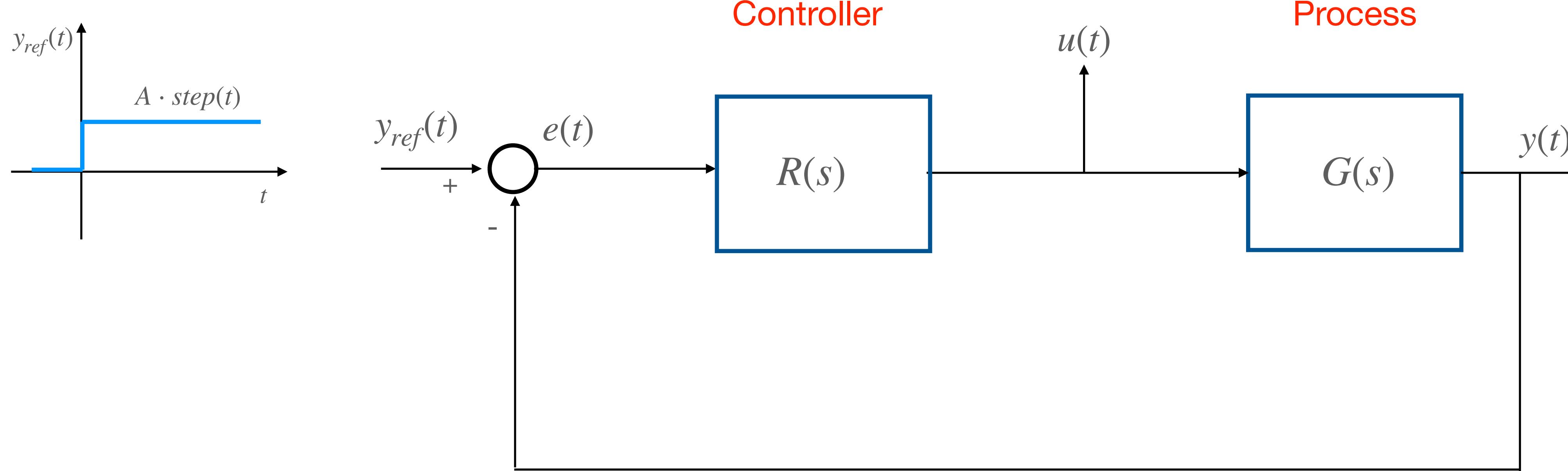


Block Algebra Rule:  $y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$

$y \approx y_{ref} : F(0) = 1 \longrightarrow \text{Impossible if } L(s) \text{ is not at least a TYPE 1 transfer function}$



## Basic Control Scheme: Example



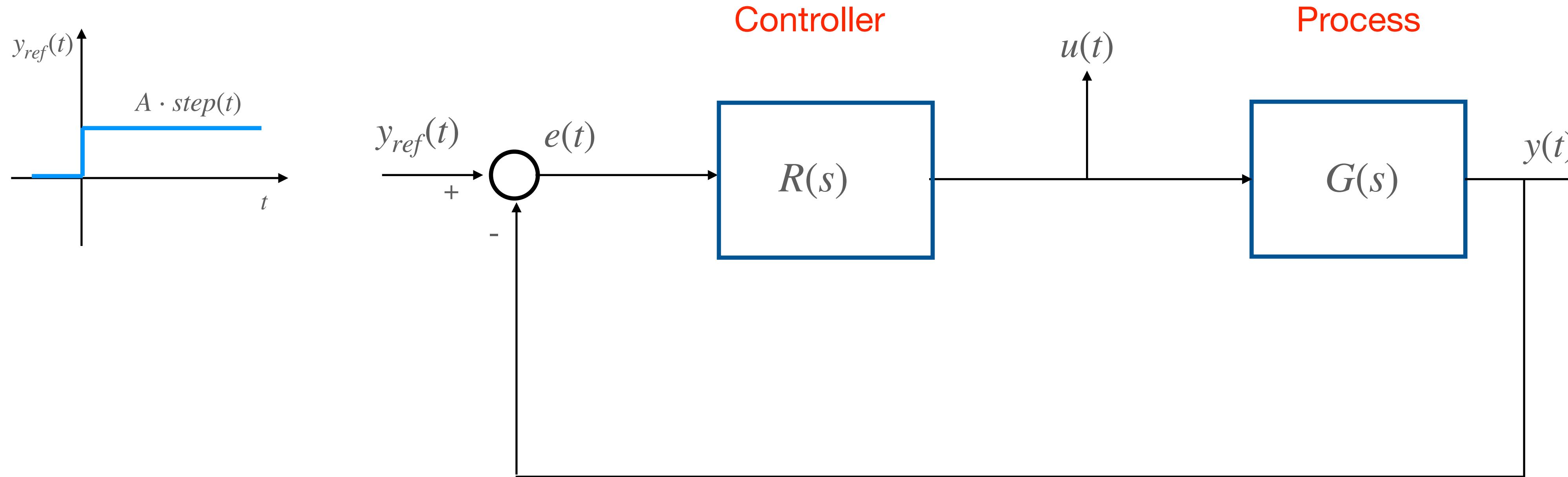
$$L(s) = \frac{100}{s + 10} \rightarrow F(s) = \frac{\frac{100}{s + 10}}{1 + \frac{100}{s + 10}} = \frac{100}{s + 110}$$

$$y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$$

$y \approx y_{ref} : F(0) = 1 \quad \longrightarrow \text{Impossible if } L(s) \text{ is not at least a TYPE 1 transfer function}$



## Basic Control Scheme: Example



$$L(s) = \frac{100}{s + 10} \rightarrow F(s) = \frac{\frac{100}{s+10}}{1 + \frac{100}{s+10}} = \frac{100}{s + 110}$$

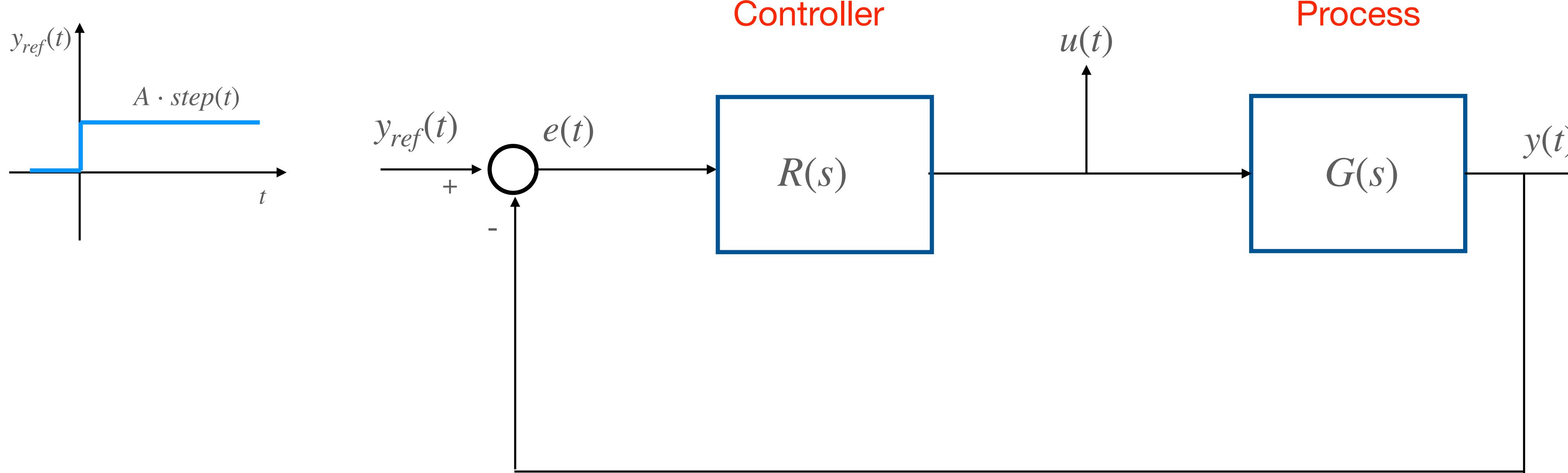
$$\lim_{s \rightarrow 0} F(s) = \frac{100}{110} = \frac{10}{11} \neq 1$$

$$y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$$

$y \approx y_{ref} : F(0) = 1 \quad \longrightarrow \quad \text{Impossible if } L(s) \text{ is not at least a TYPE 1 transfer function}$



## Basic Control Scheme: Example



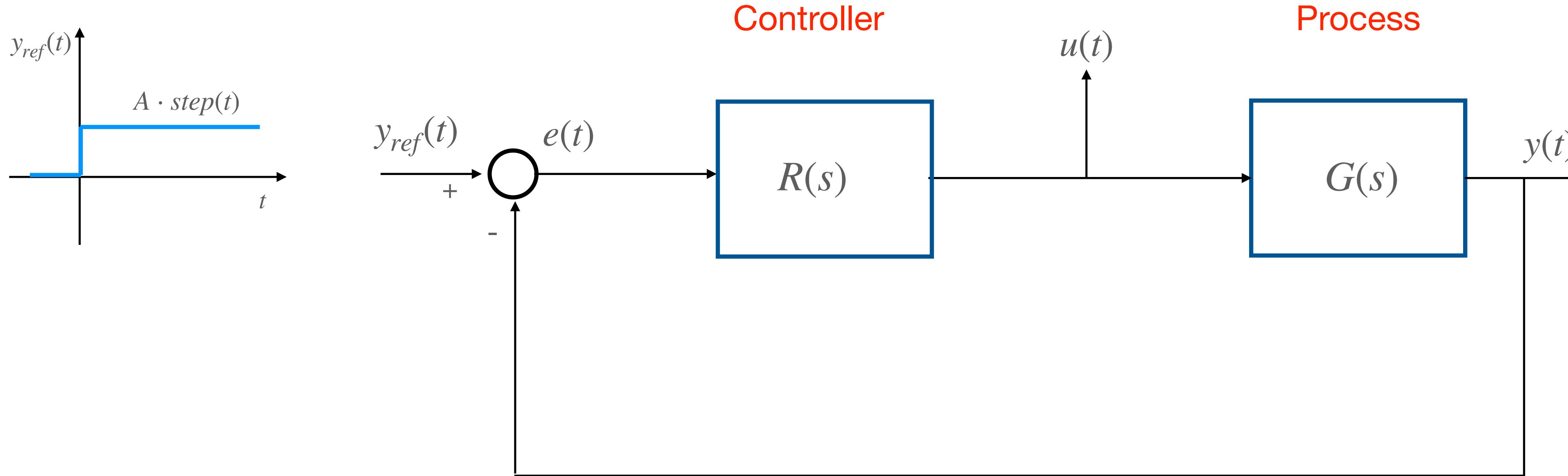
$$L(s) = \frac{100}{s(s+10)} \rightarrow F(s) = \frac{\frac{100}{s(s+10)}}{1 + \frac{100}{s(s+10)}} = \frac{100}{s(s+10)+100}$$

$$y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$$

$y \approx y_{ref} : F(0) = 1 \longrightarrow$  Impossible if  $L(s)$  is not at least a TYPE 1 transfer function



## Basic Control Scheme: Example



$$L(s) = \frac{100}{s(s+10)} \rightarrow F(s) = \frac{\frac{100}{s(s+10)}}{1 + \frac{100}{s(s+10)}} = \frac{100}{s(s+10)+100}$$

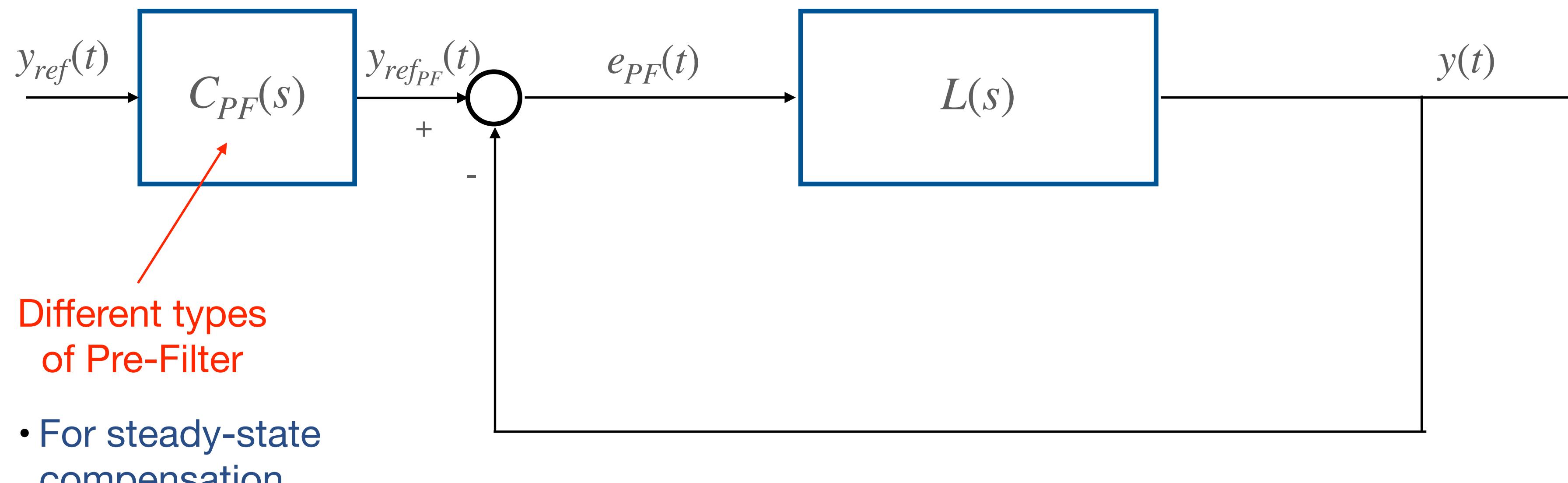
$$\lim_{s \rightarrow 0} F(s) = \frac{100}{100} = 1$$

$$y_{ref} \rightarrow y : F(s) = \frac{R(s)G(s)}{1 + R(s)G(s)} = \frac{L(s)}{1 + L(s)}$$

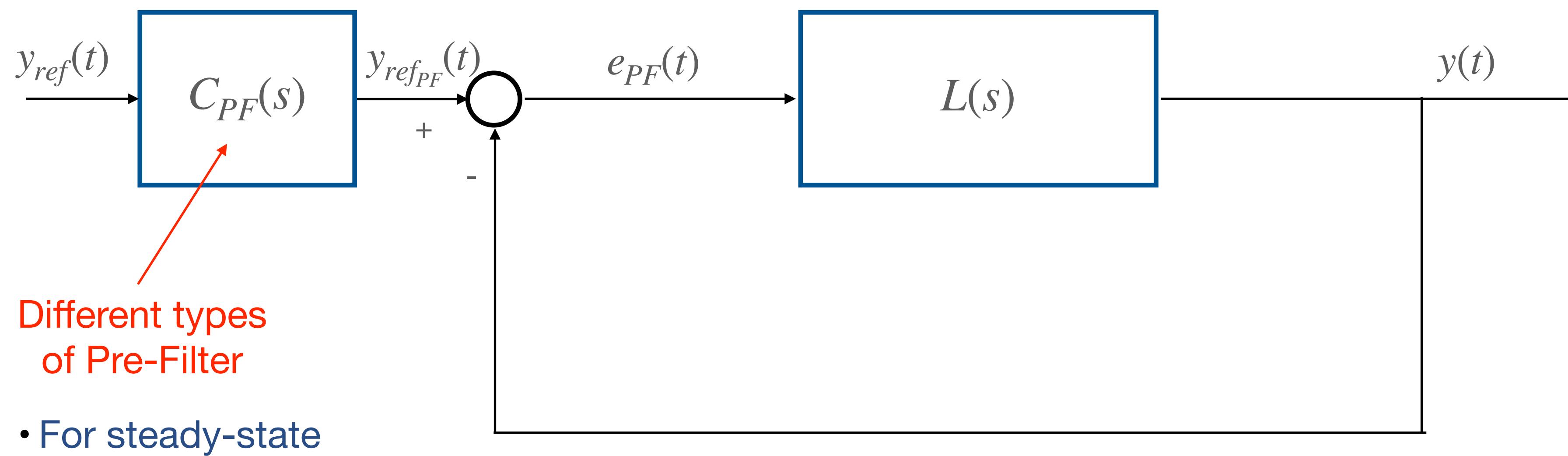
$y \approx y_{ref} : F(0) = 1 \longrightarrow$  Impossible if  $L(s)$  is not at least a TYPE 1 transfer function



## Pre-filter Based Control Scheme: Case 3



## Pre-filter Based Control Scheme: Case 3

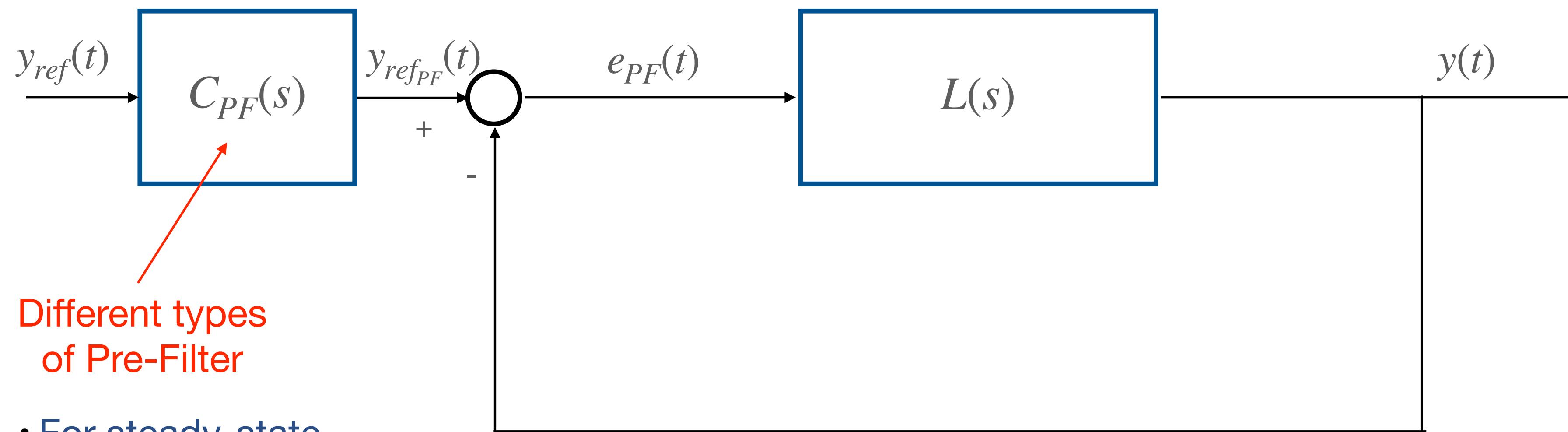


Assumptions:

- $C_{PF}(s)$  As. Stable
- Proper
- ~~Unitary gain~~



## Pre-filter Based Control Scheme: Case 3



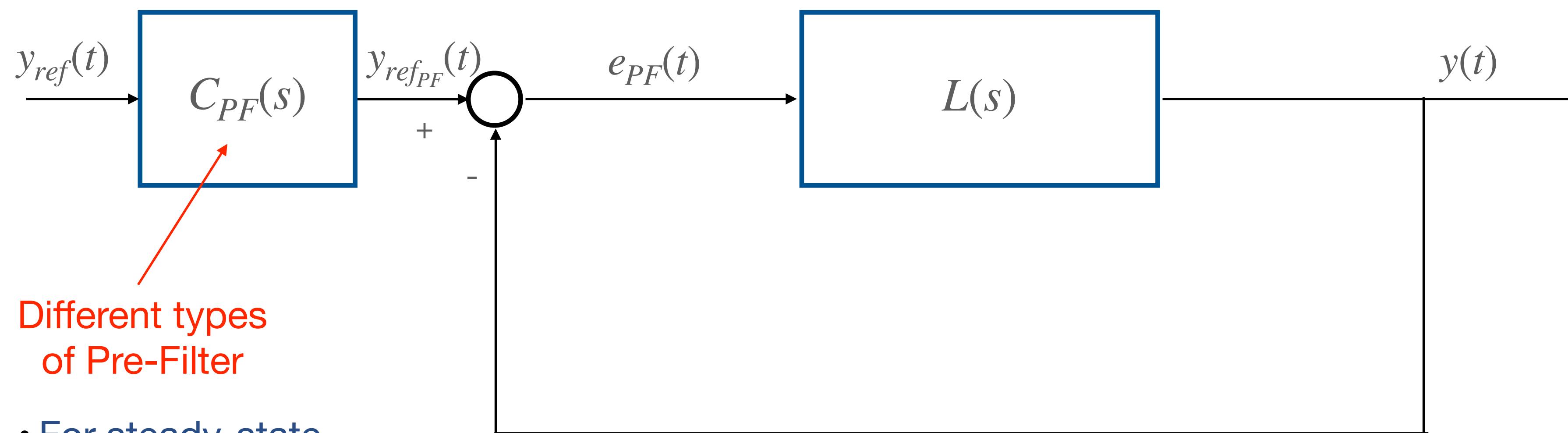
Assumptions:

- $C_{PF}(s)$  As. Stable
- Proper
- ~~Unitary gain~~

Block Algebra Rule:  $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$



## Pre-filter Based Control Scheme: Case 3



Assumptions:

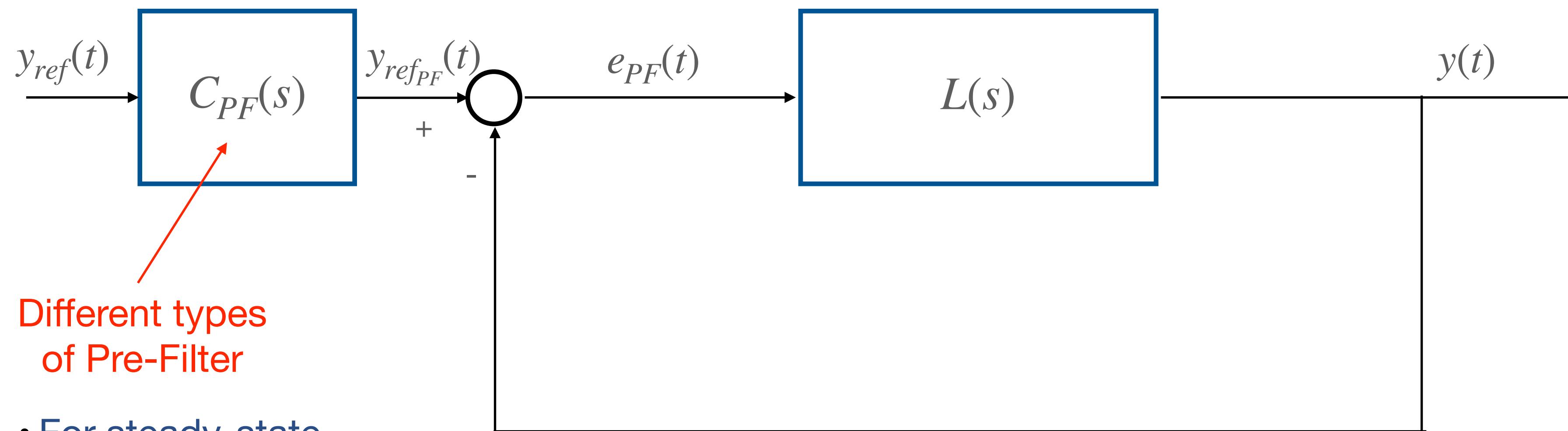
- $C_{PF}(s)$  As. Stable
- Proper
- ~~Unitary gain~~

Block Algebra Rule:  $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$

$$y \approx y_{ref} : C_{PF}(0) \cdot F(0) = 1$$



## Pre-filter Based Control Scheme: Case 3



Assumptions:

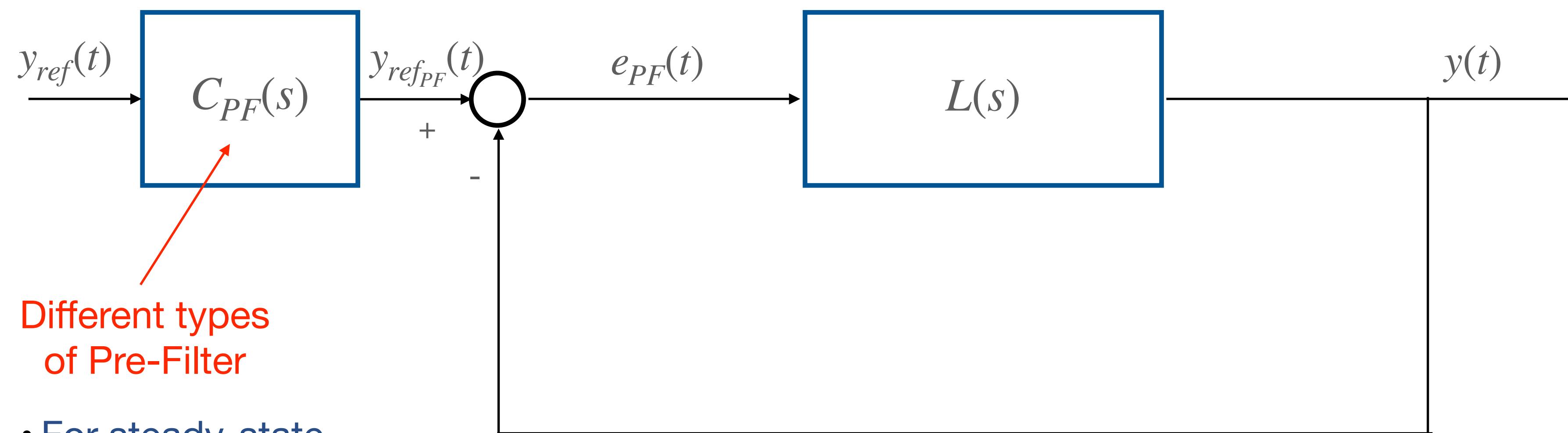
- $C_{PF}(s)$  As. Stable
- Proper
- ~~Unitary gain~~

Block Algebra Rule:  $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$

$$y \approx y_{ref} : C_{PF}(0) \cdot F(0) = 1 \longrightarrow C_{PF}(0) = \frac{1}{F(0)}$$



## Pre-filter Based Control Scheme: Case 3



**Assumptions:**

- $C_{PF}(s)$  As. Stable
- Proper
- ~~Unitary gain~~

The filter could be just a constant gain

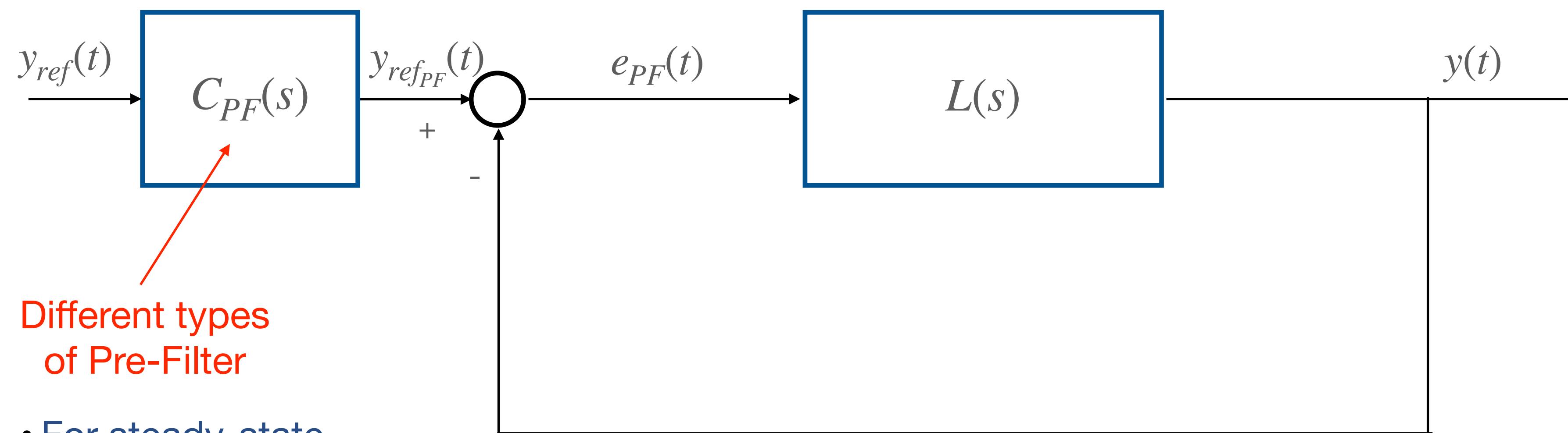
Block Algebra Rule:  $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$

$$y \approx y_{ref} : C_{PF}(0) \cdot F(0) = 1$$

$$C_{PF}(0) = \frac{1}{F(0)}$$



## Pre-filter Based Control Scheme: Case 3



**Assumptions:**

- $C_{PF}(s)$  As. Stable
- Proper
- ~~Unitary gain~~

**Warning:**  
Use only when strictly necessary!

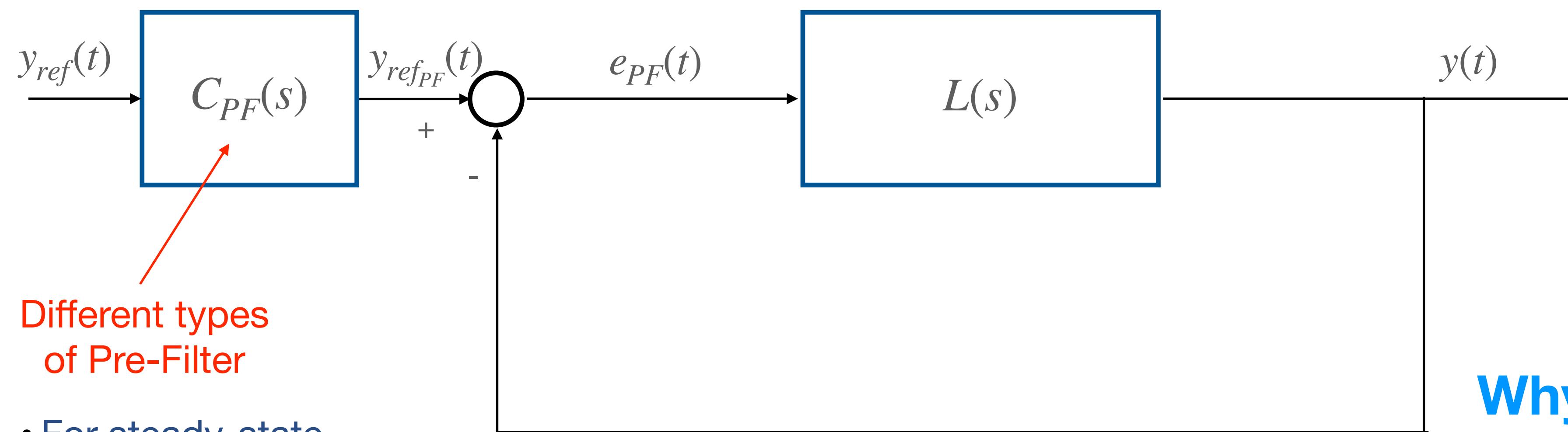
**Block Algebra Rule:**  $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$

$$y \approx y_{ref} : C_{PF}(0) \cdot F(0) = 1$$

$$C_{PF}(0) = \frac{1}{F(0)}$$



## Pre-filter Based Control Scheme: Case 3



**Assumptions:**

- $C_{PF}(s)$  As. Stable
- Proper
- ~~Unitary gain~~

**Why?**

**Warning:**  
Use only when strictly necessary!

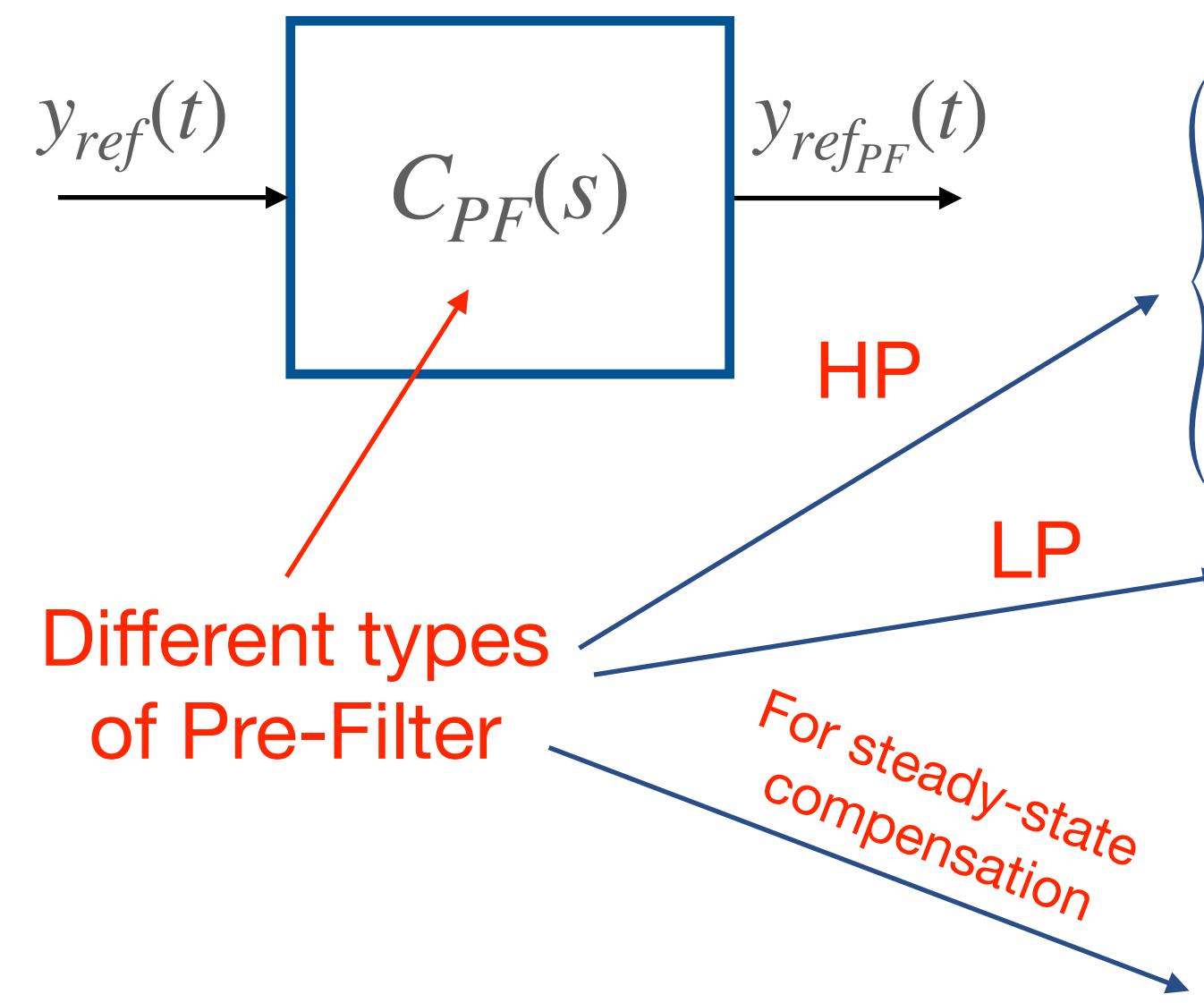
Block Algebra Rule:  $y_{ref} \rightarrow y : C_{PF}(s) \cdot \frac{L(s)}{1 + L(s)} = C_{PF}(s) \cdot F(s)$

$$y \approx y_{ref} : C_{PF}(0) \cdot F(0) = 1$$

$$C_{PF}(0) = \frac{1}{F(0)}$$



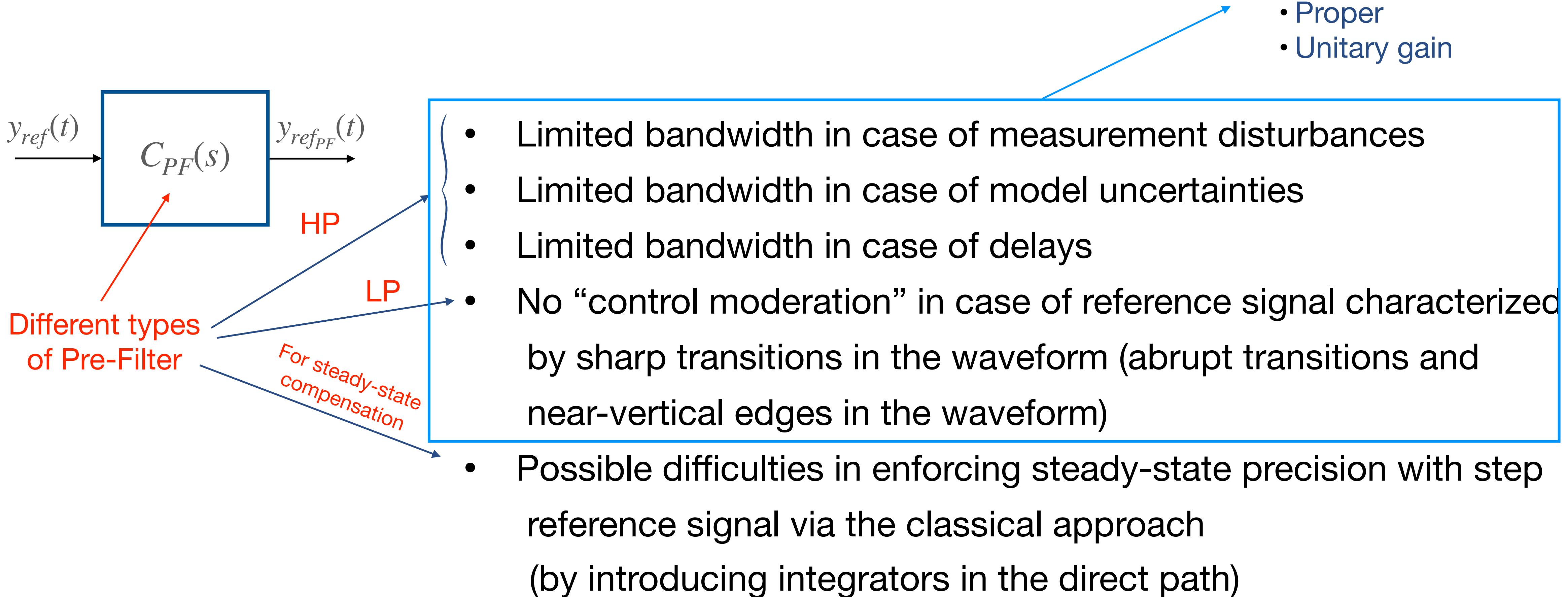
## Pre-filter Based Control Scheme: Summary



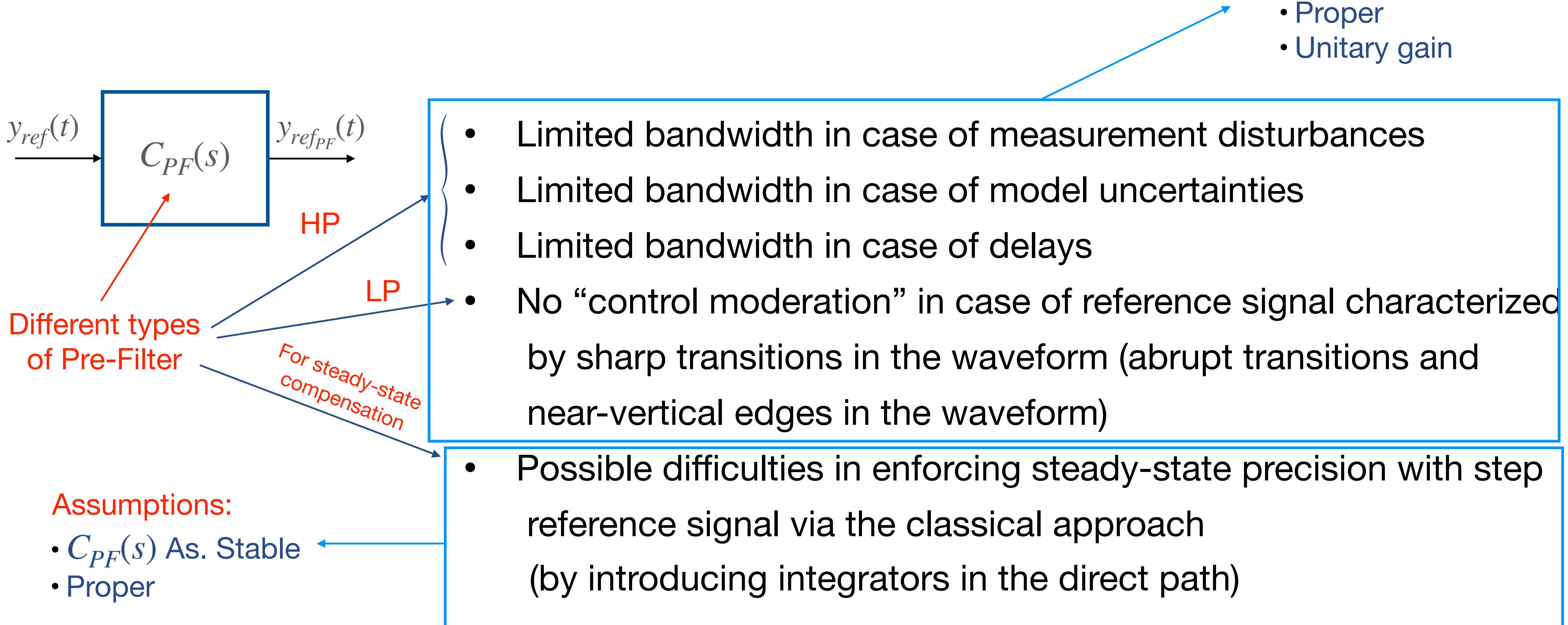
- Limited bandwidth in case of measurement disturbances
- Limited bandwidth in case of model uncertainties
- Limited bandwidth in case of delays
- No “control moderation” in case of reference signal characterized by sharp transitions in the waveform (abrupt transitions and near-vertical edges in the waveform)
- Possible difficulties in enforcing steady-state precision with step reference signal via the classical approach  
(by introducing integrators in the direct path)



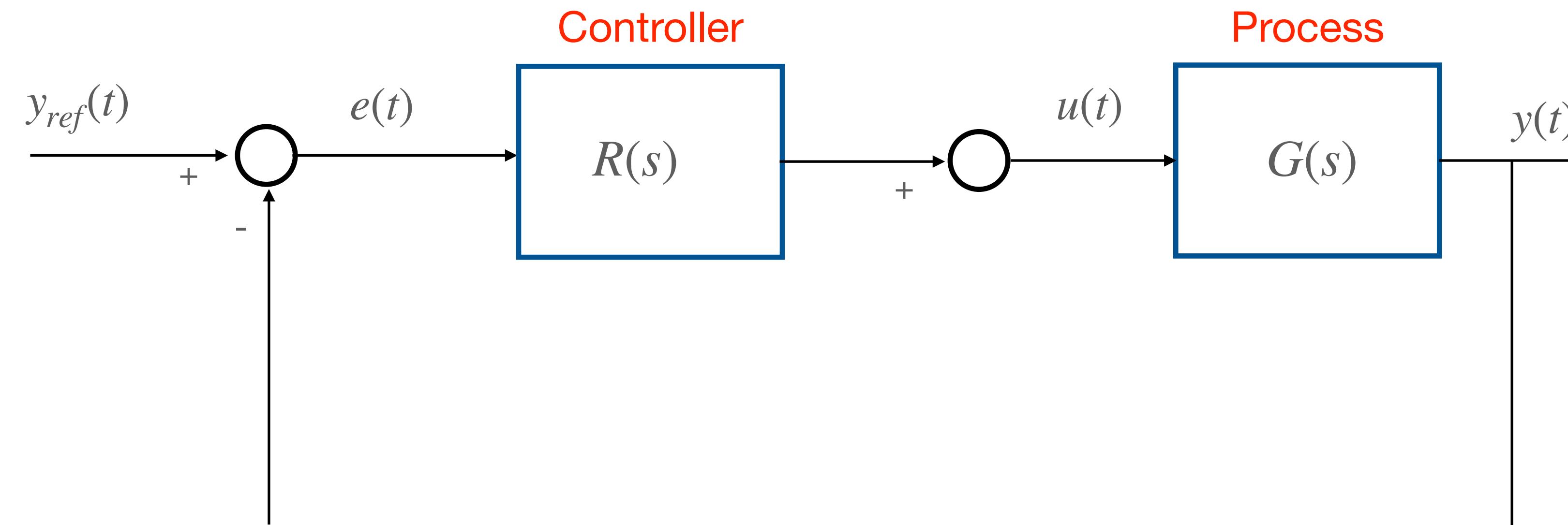
## Pre-filter Based Control Scheme: Summary



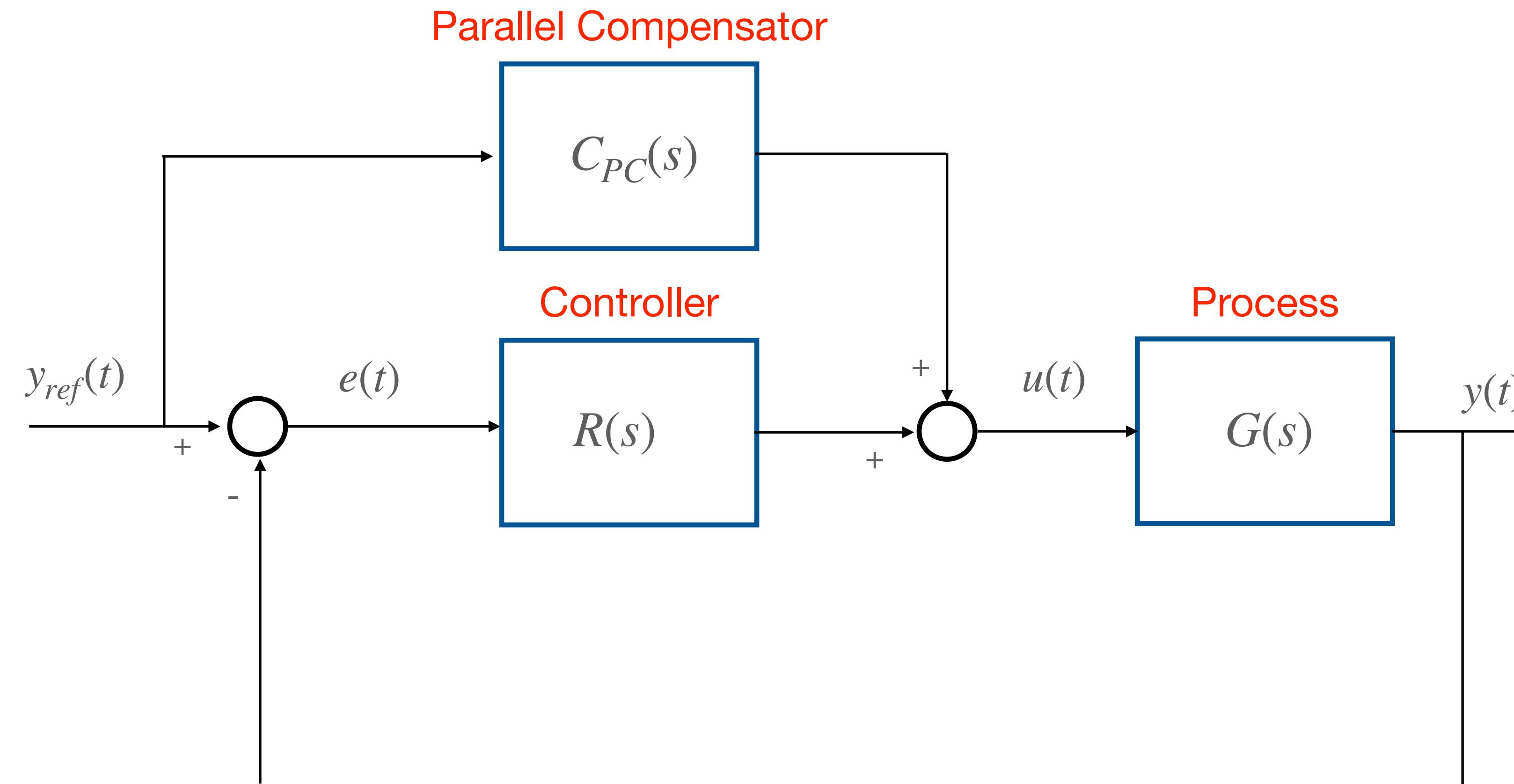
## Pre-filter Based Control Scheme: Summary



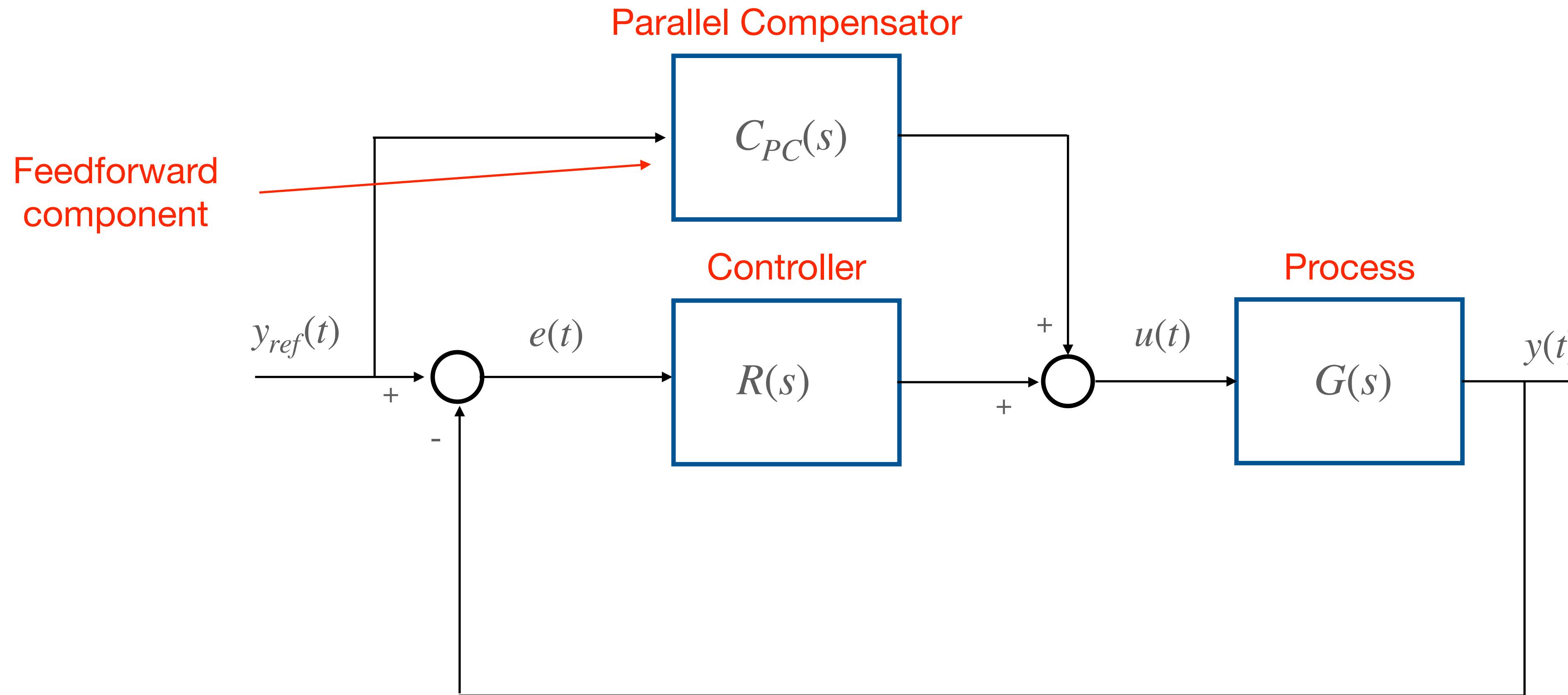
## Parallel-compensator Based Control Scheme



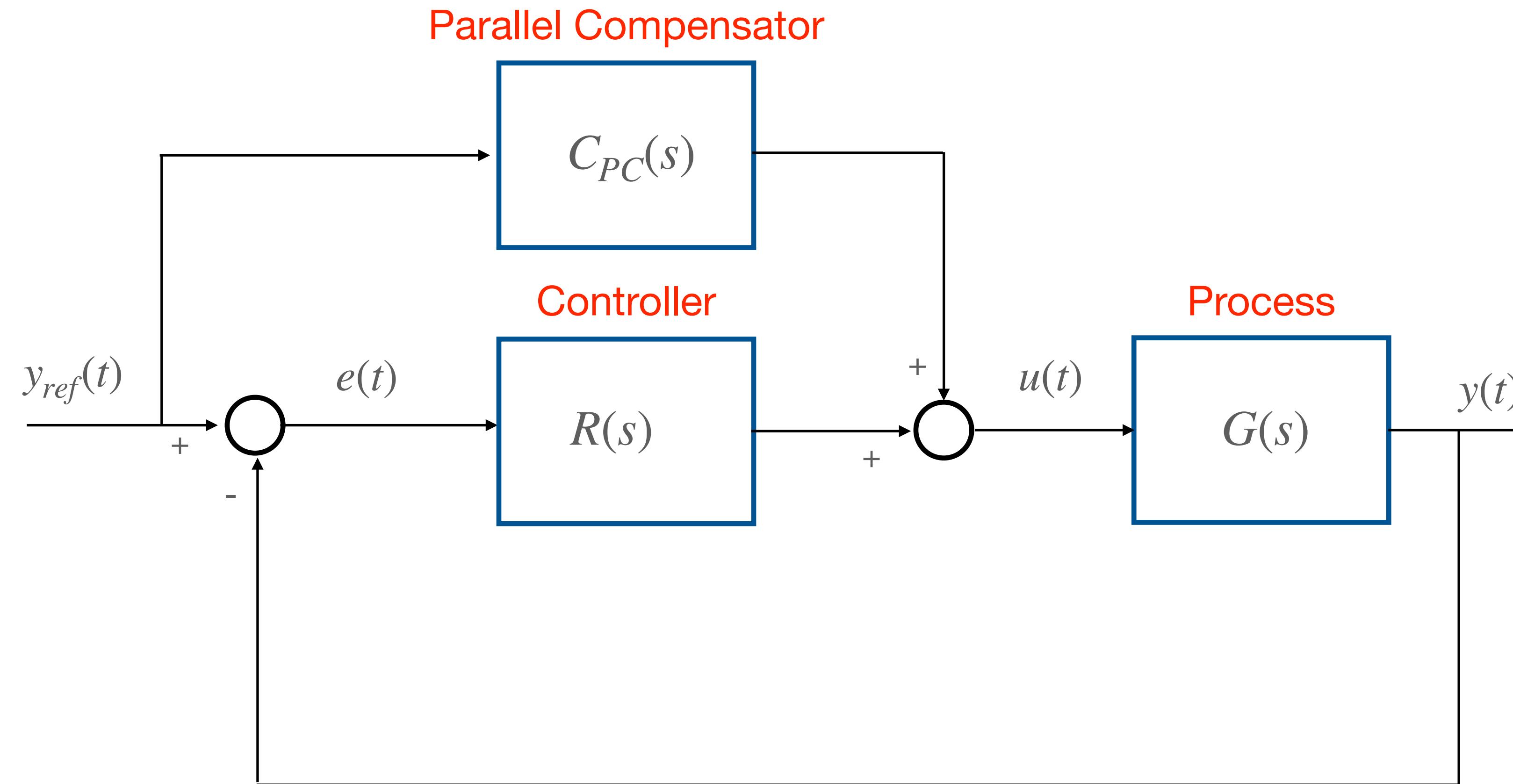
## Parallel-compensator Based Control Scheme



## Parallel-compensator Based Control Scheme



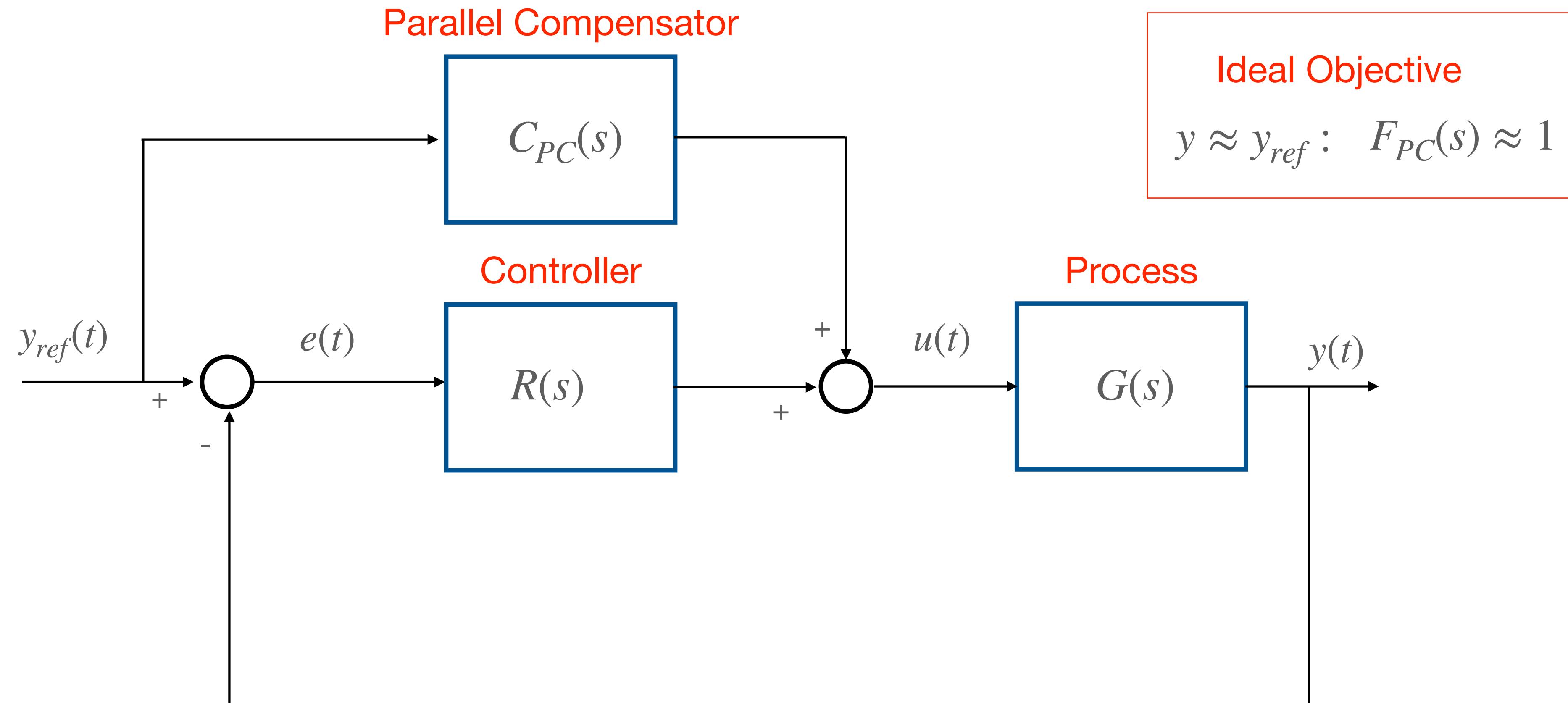
## Parallel-compensator Based Control Scheme



$$y_{ref} \rightarrow y : F_{PC}(s) = \frac{C_{PC}(s)G(s) + R(s)G(s)}{1 + R(s)G(s)}$$



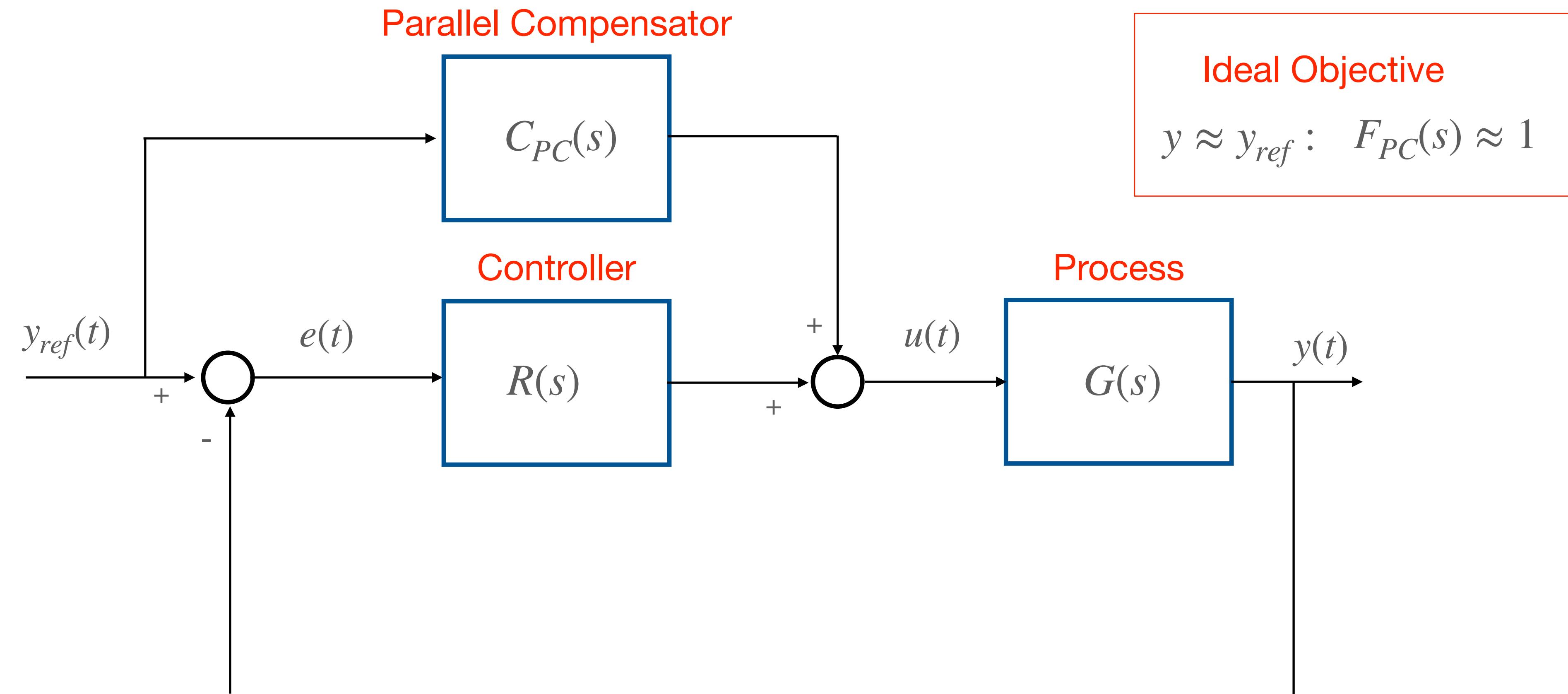
## Parallel-compensator Based Control Scheme



$$y_{ref} \rightarrow y : F_{PC}(s) = \frac{C_{PC}(s)G(s) + R(s)G(s)}{1 + R(s)G(s)}$$



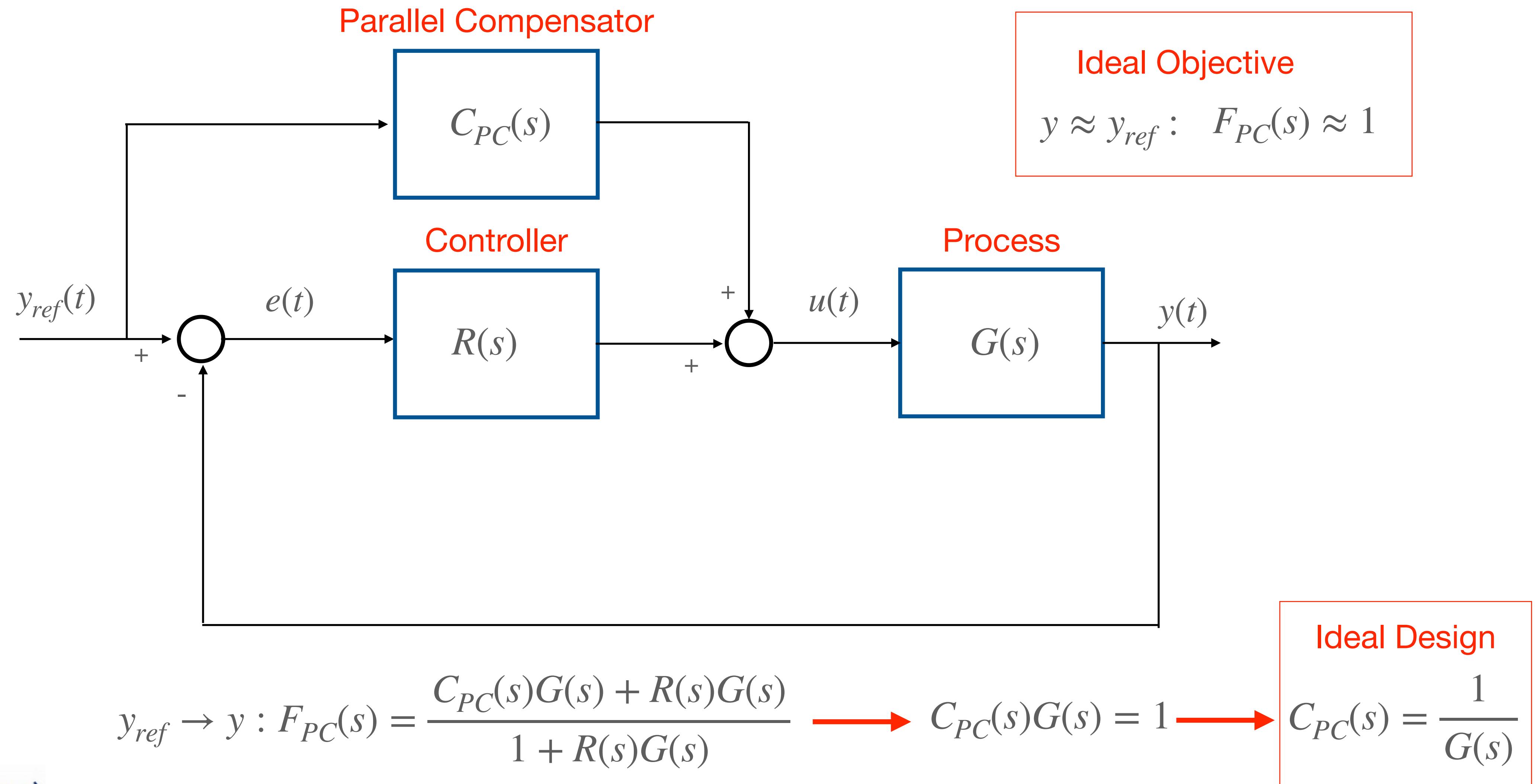
## Parallel-compensator Based Control Scheme



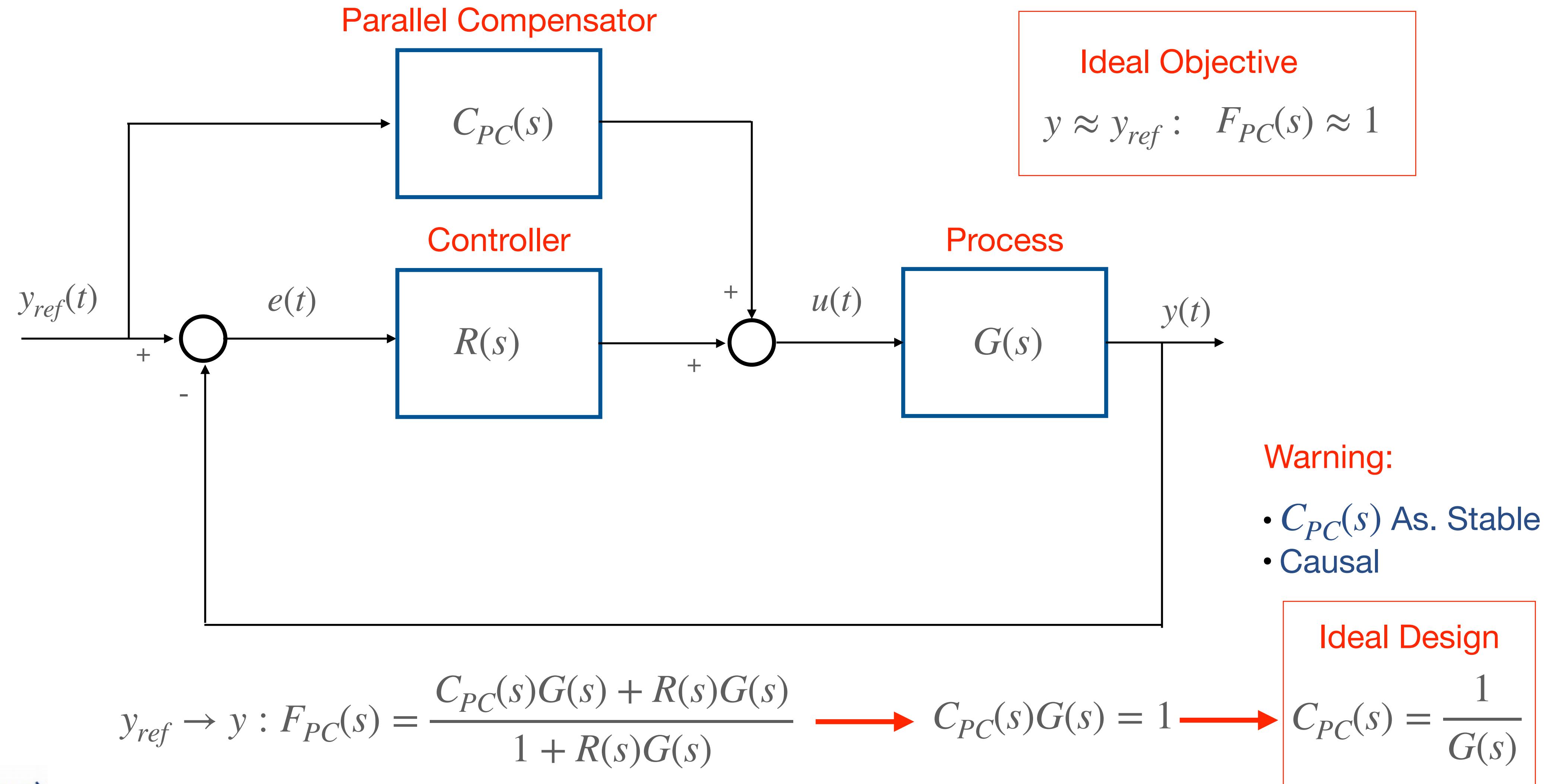
$$y_{ref} \rightarrow y : F_{PC}(s) = \frac{C_{PC}(s)G(s) + R(s)G(s)}{1 + R(s)G(s)} \longrightarrow C_{PC}(s)G(s) = 1$$



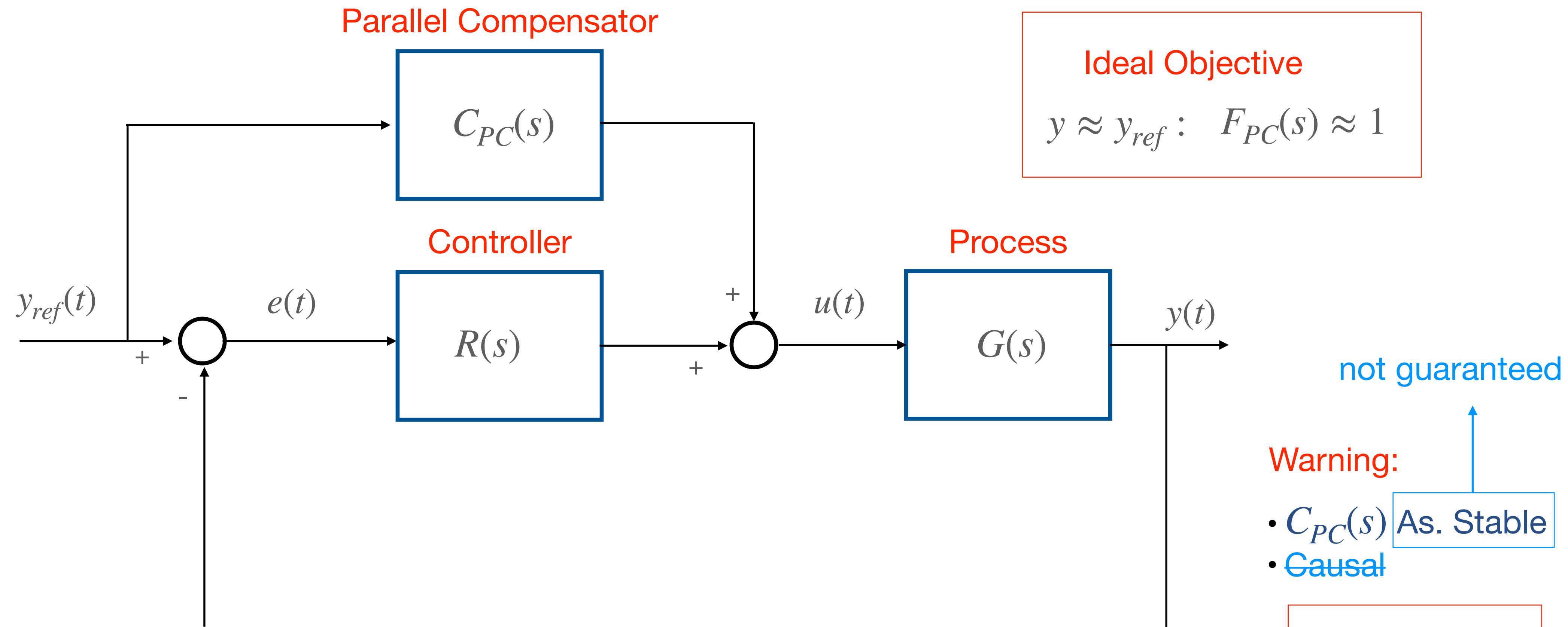
## Parallel-compensator Based Control Scheme



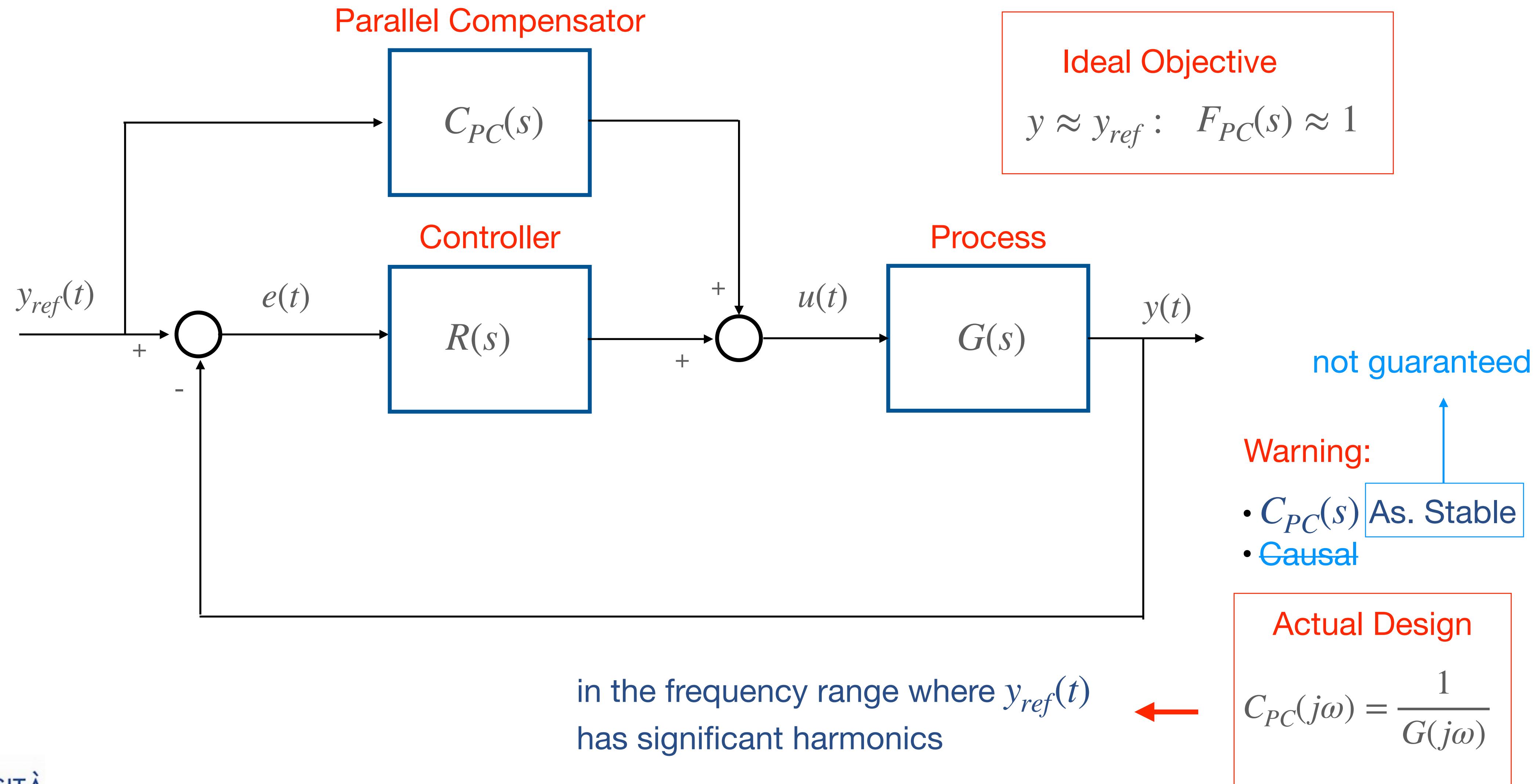
## Parallel-compensator Based Control Scheme



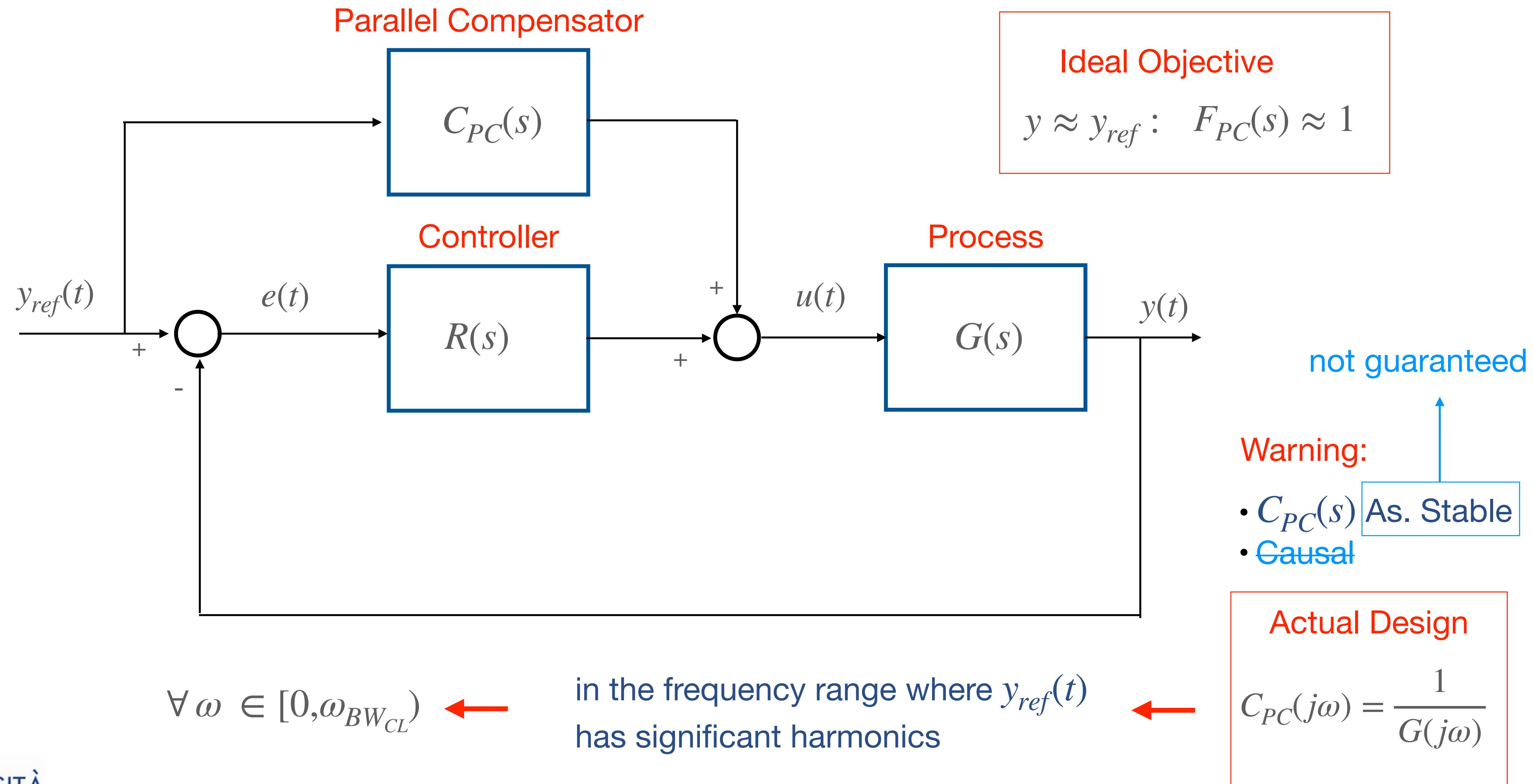
## Parallel-compensator Based Control Scheme



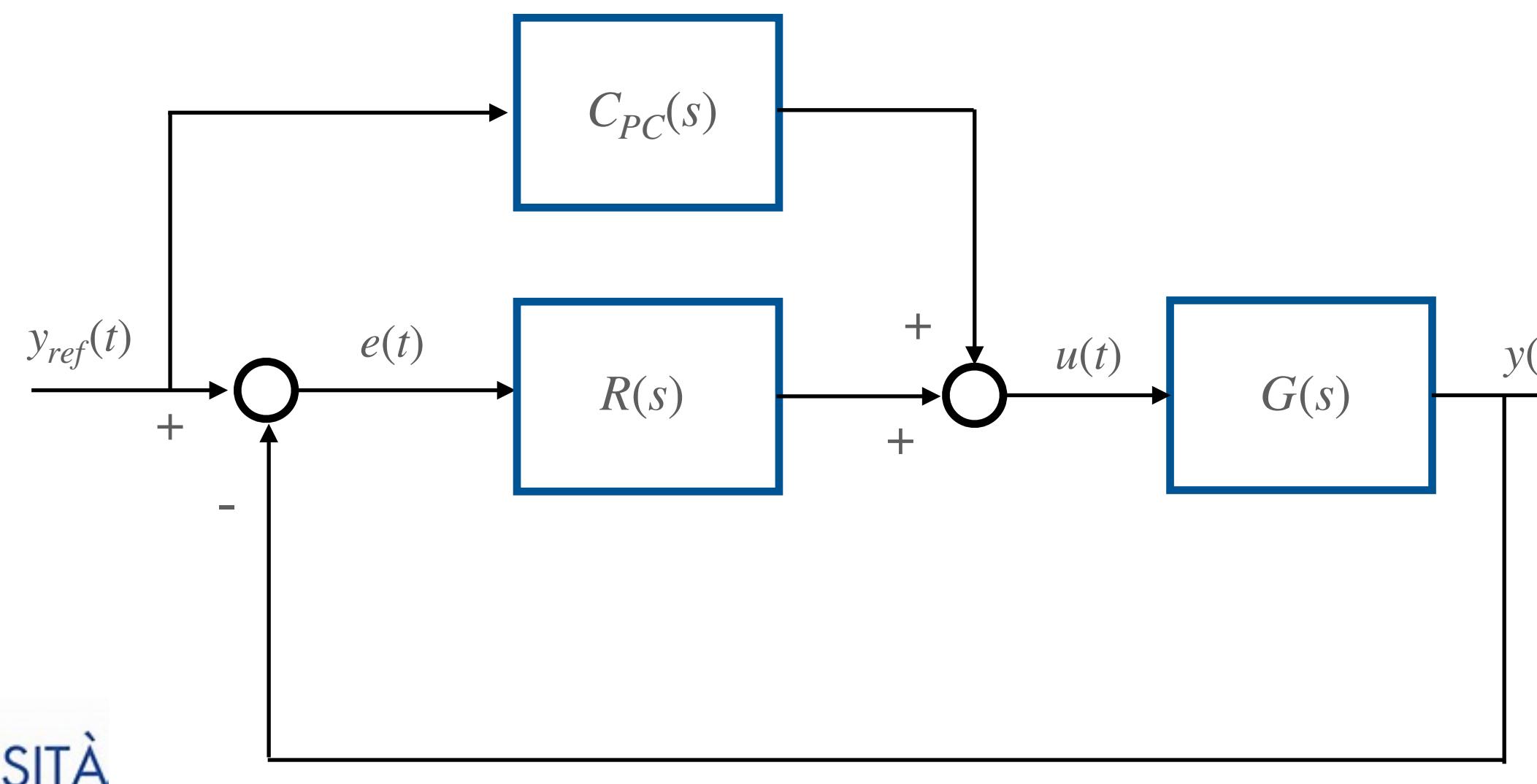
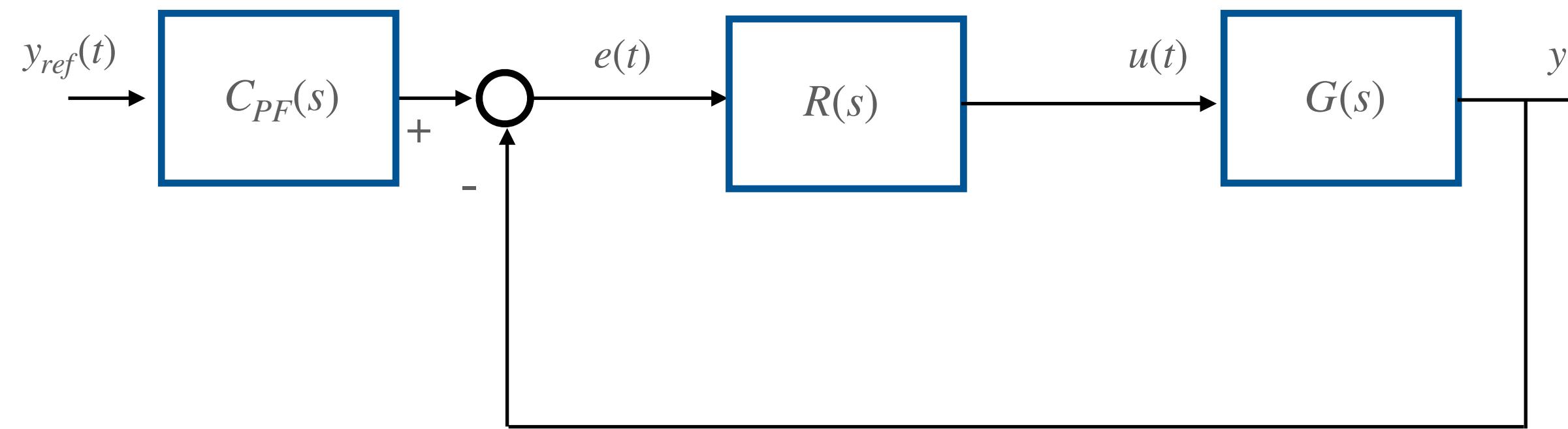
## Parallel-compensator Based Control Scheme



## Parallel-compensator Based Control Scheme

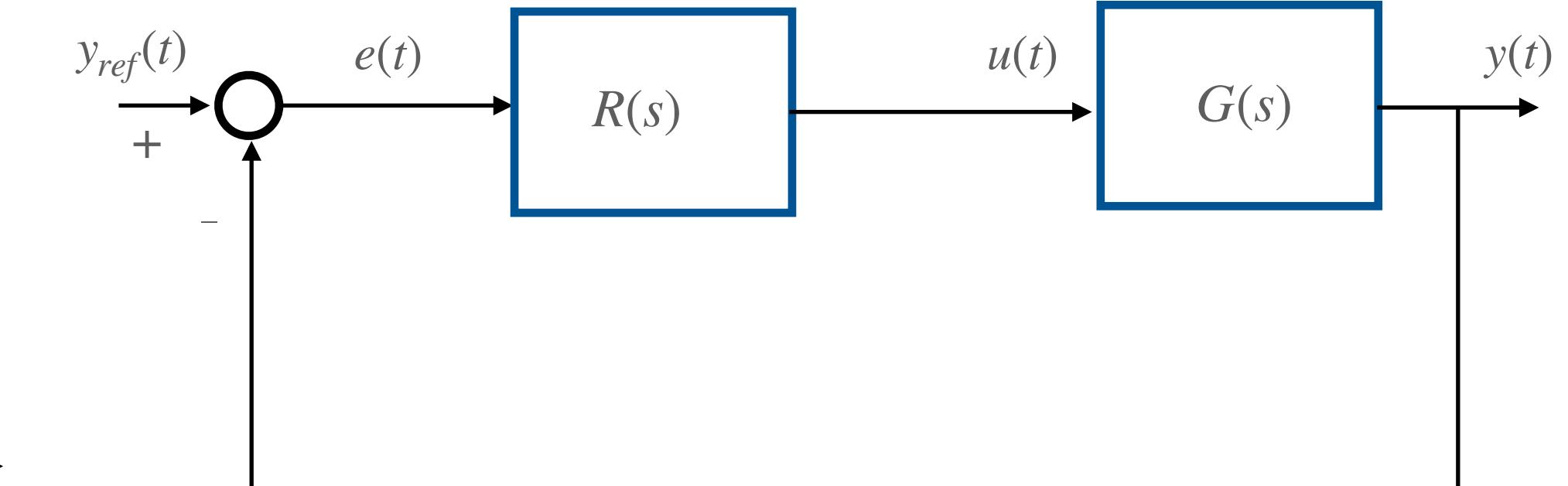
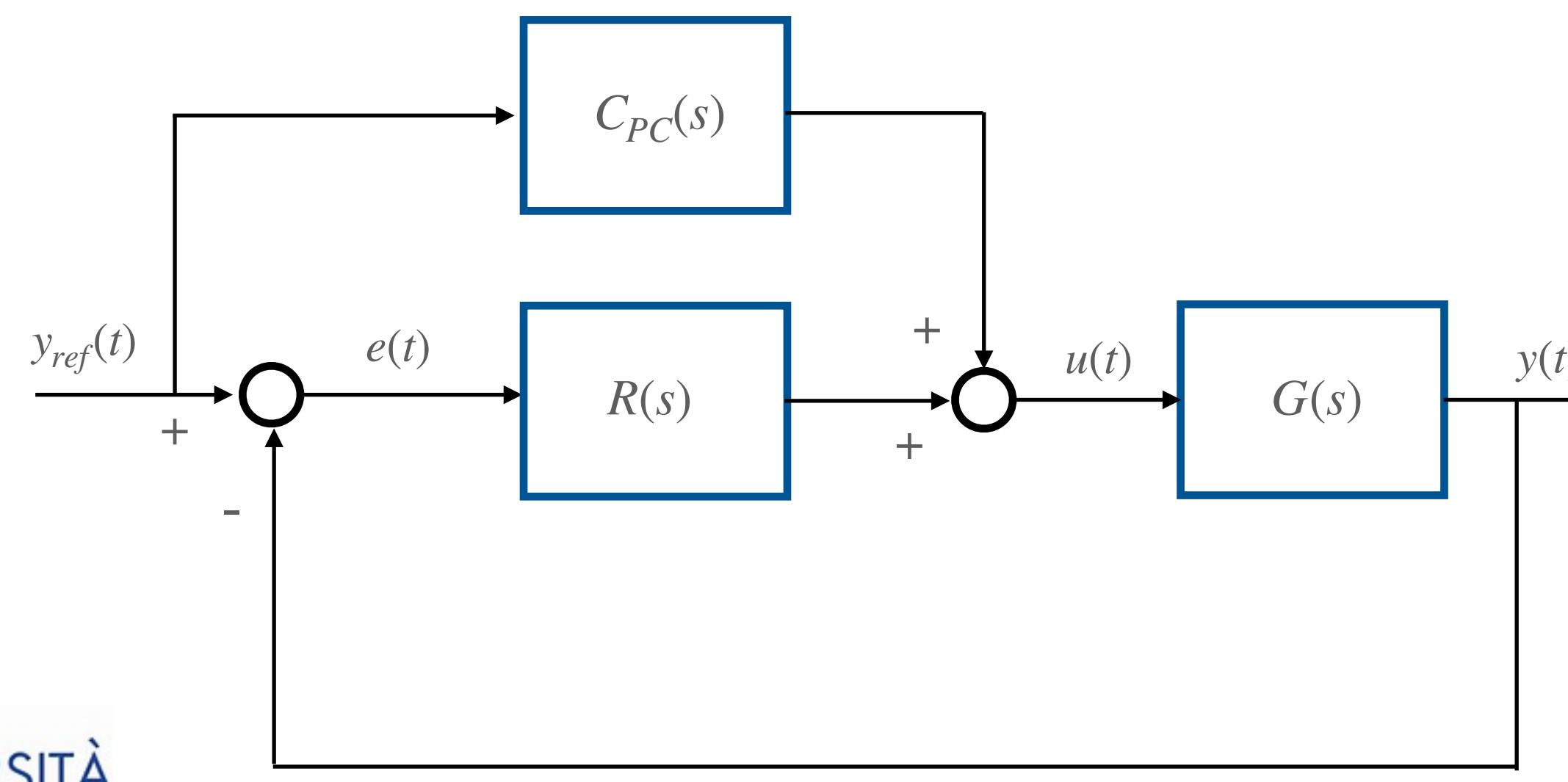
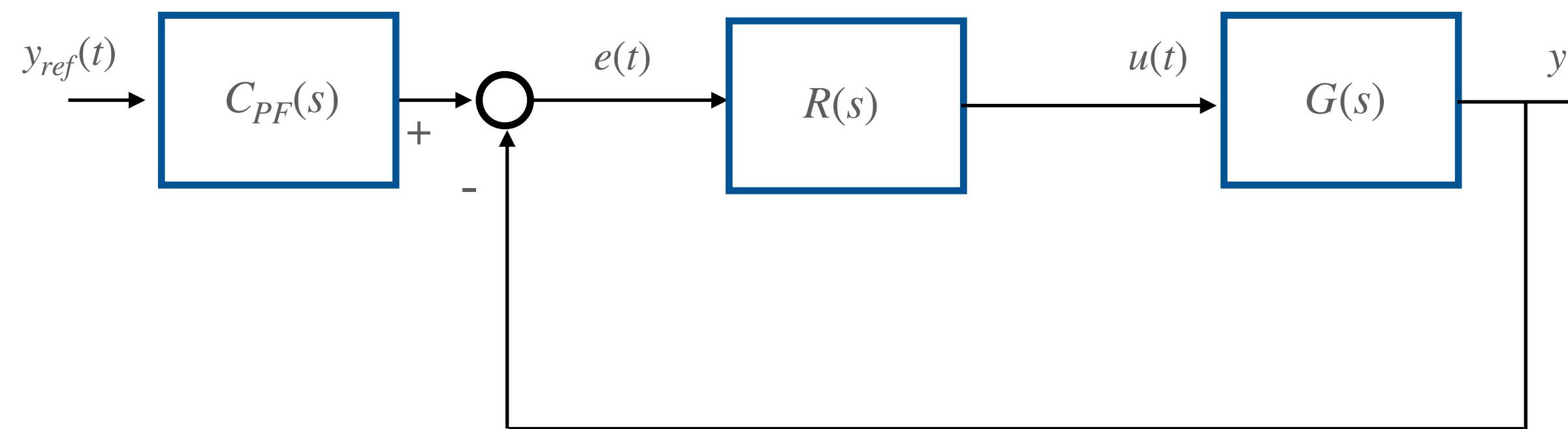


## Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes

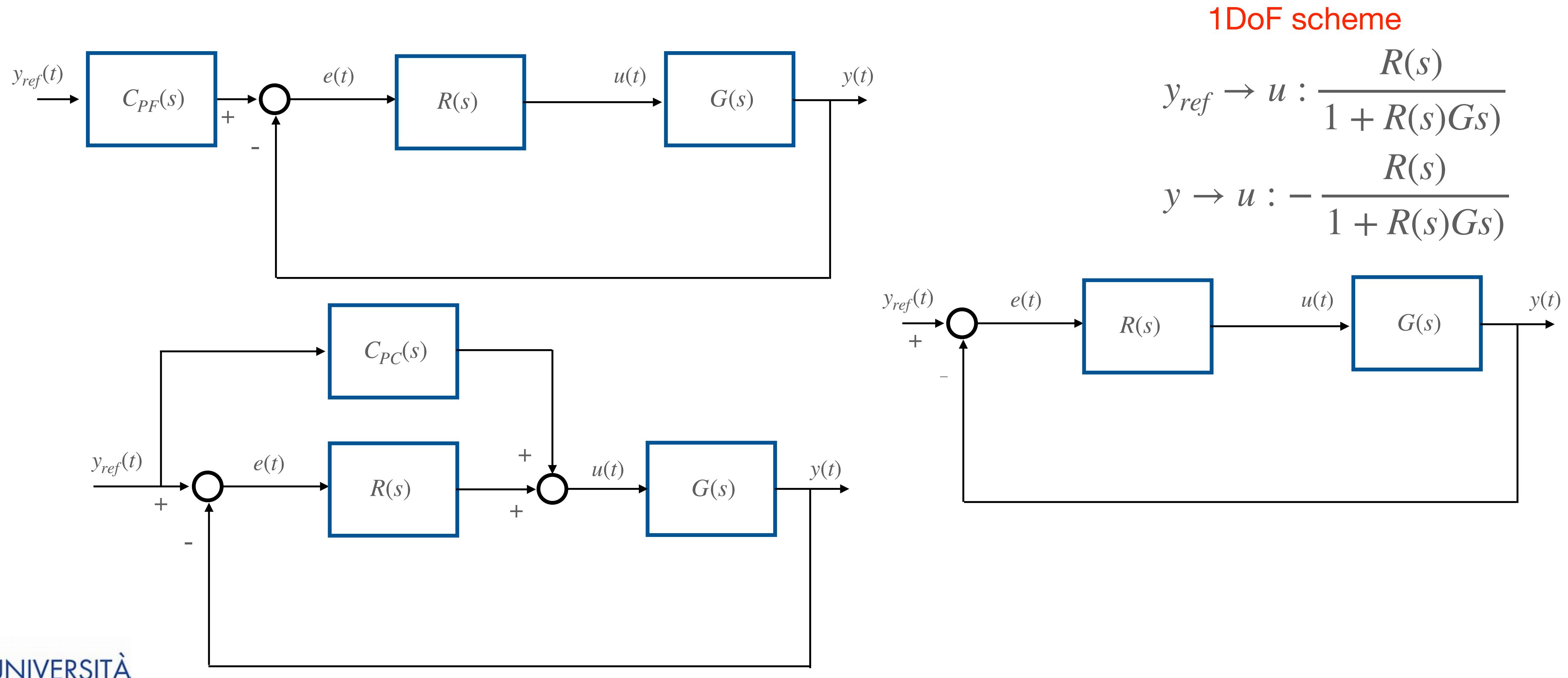


## Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes

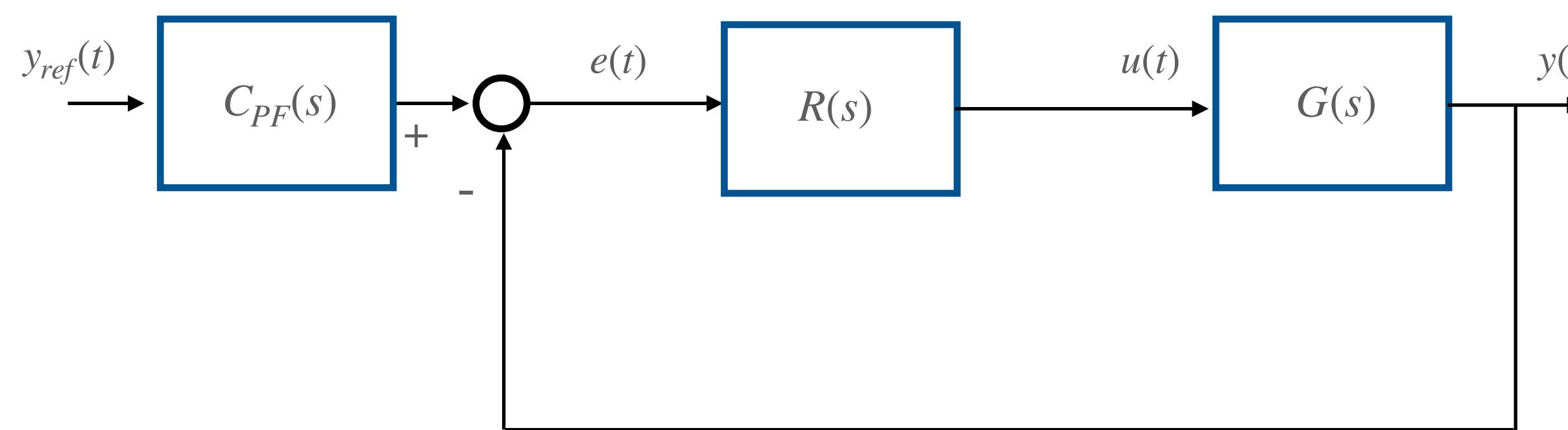
1DoF scheme



## Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



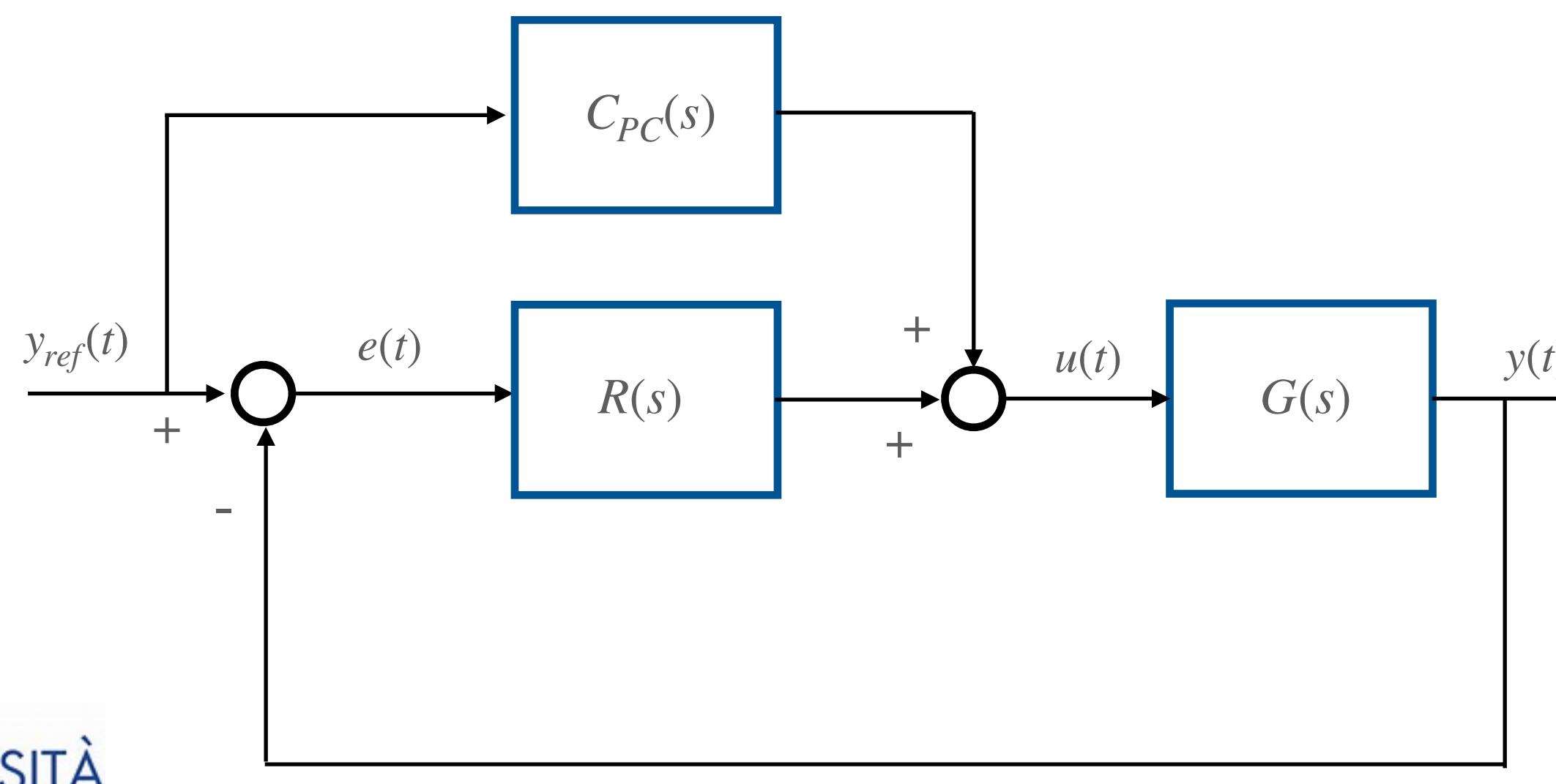
## Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



2DoF scheme

$$y_{ref} \rightarrow u : \frac{C_{PF}(s)R(s)}{1 + R(s)Gs}$$

$$y \rightarrow u : -\frac{R(s)}{1 + R(s)Gs}$$



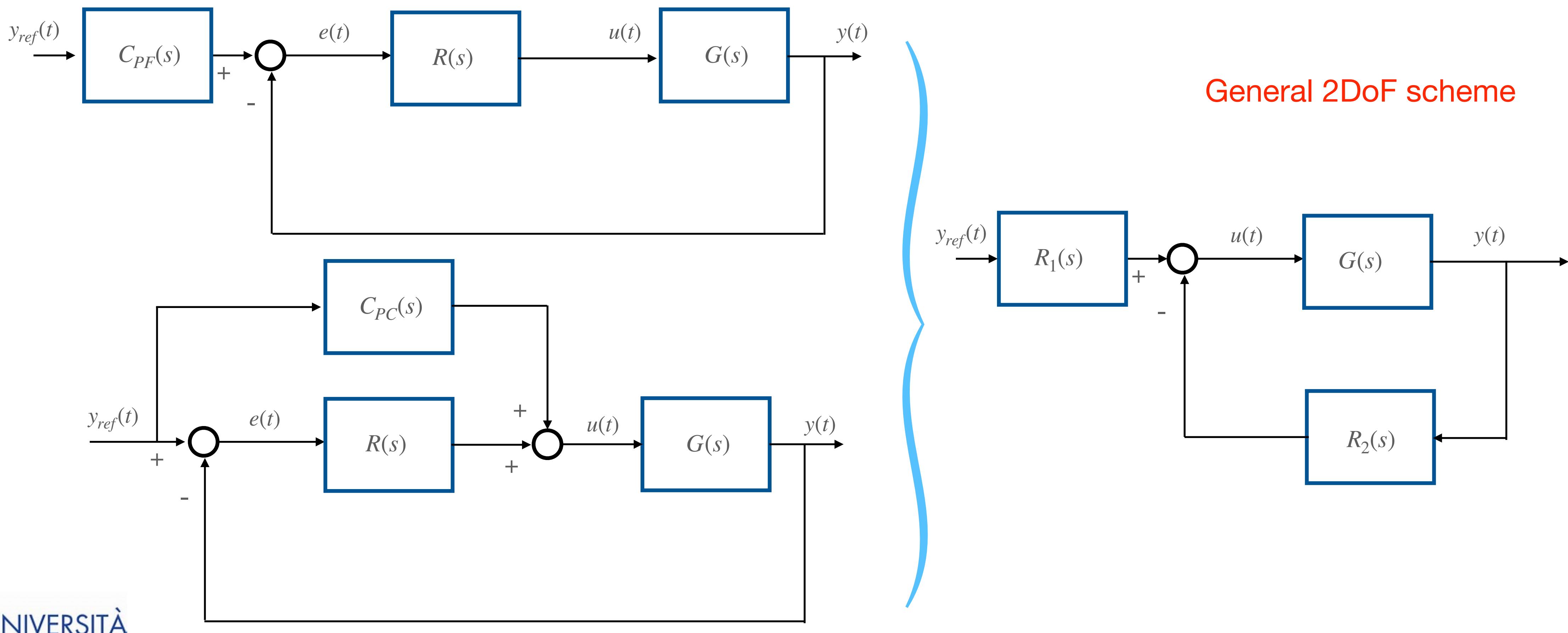
2DoF scheme

$$y_{ref} \rightarrow u : \frac{C_{PC}(s) + R(s)}{1 + R(s)Gs}$$

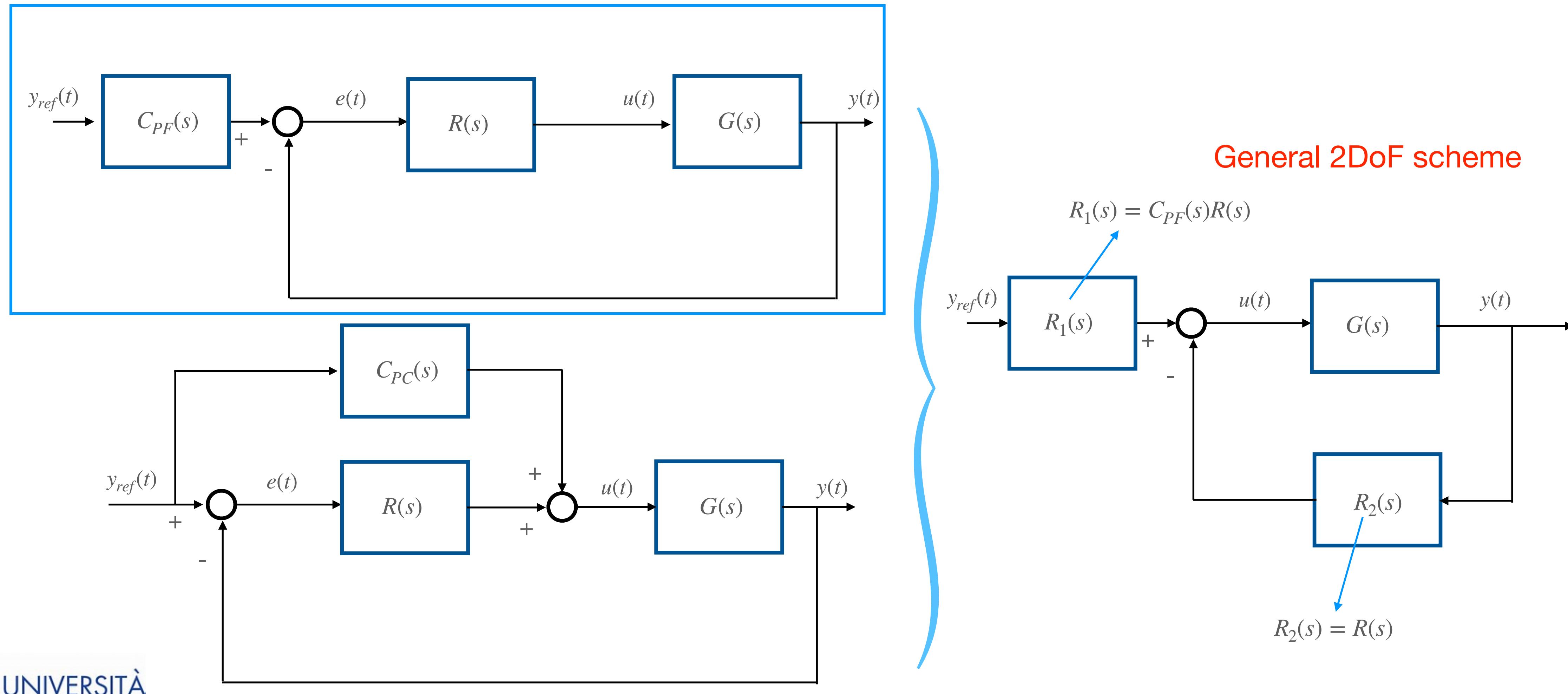
$$y \rightarrow u : -\frac{R(s)}{1 + R(s)Gs}$$



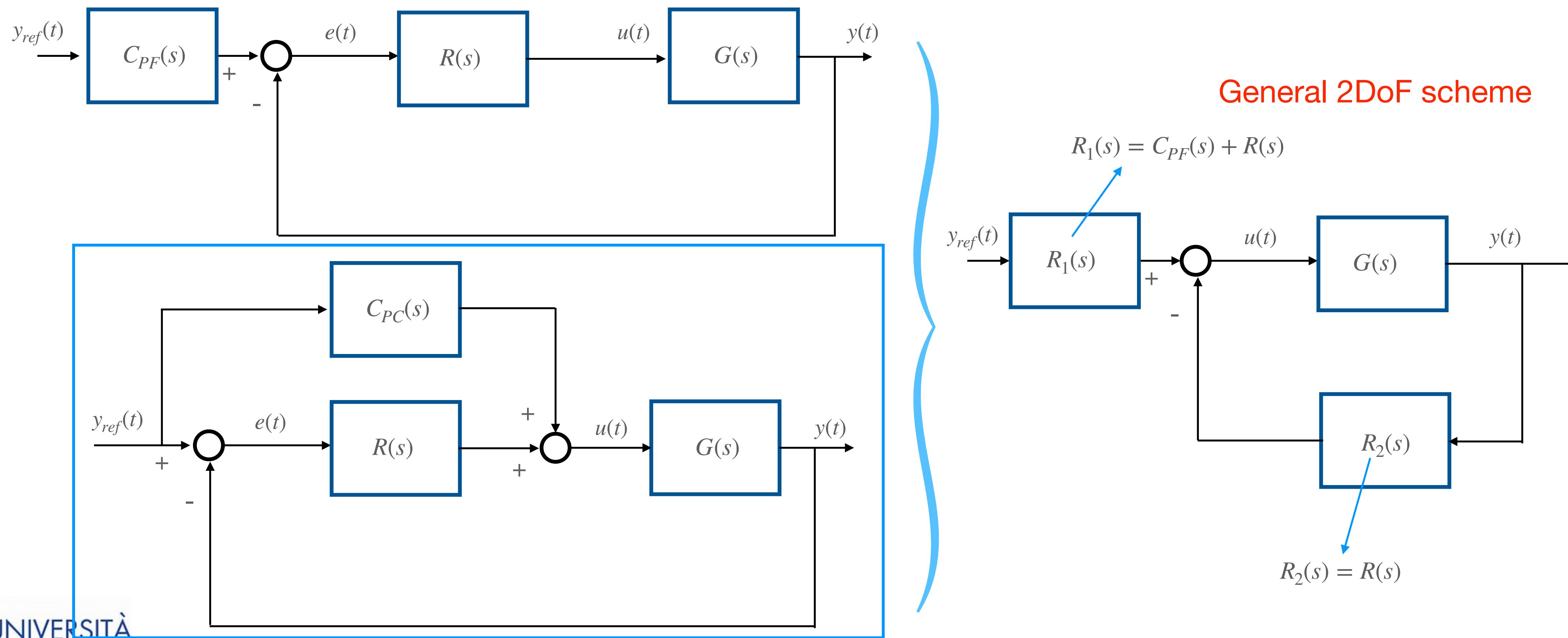
## Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



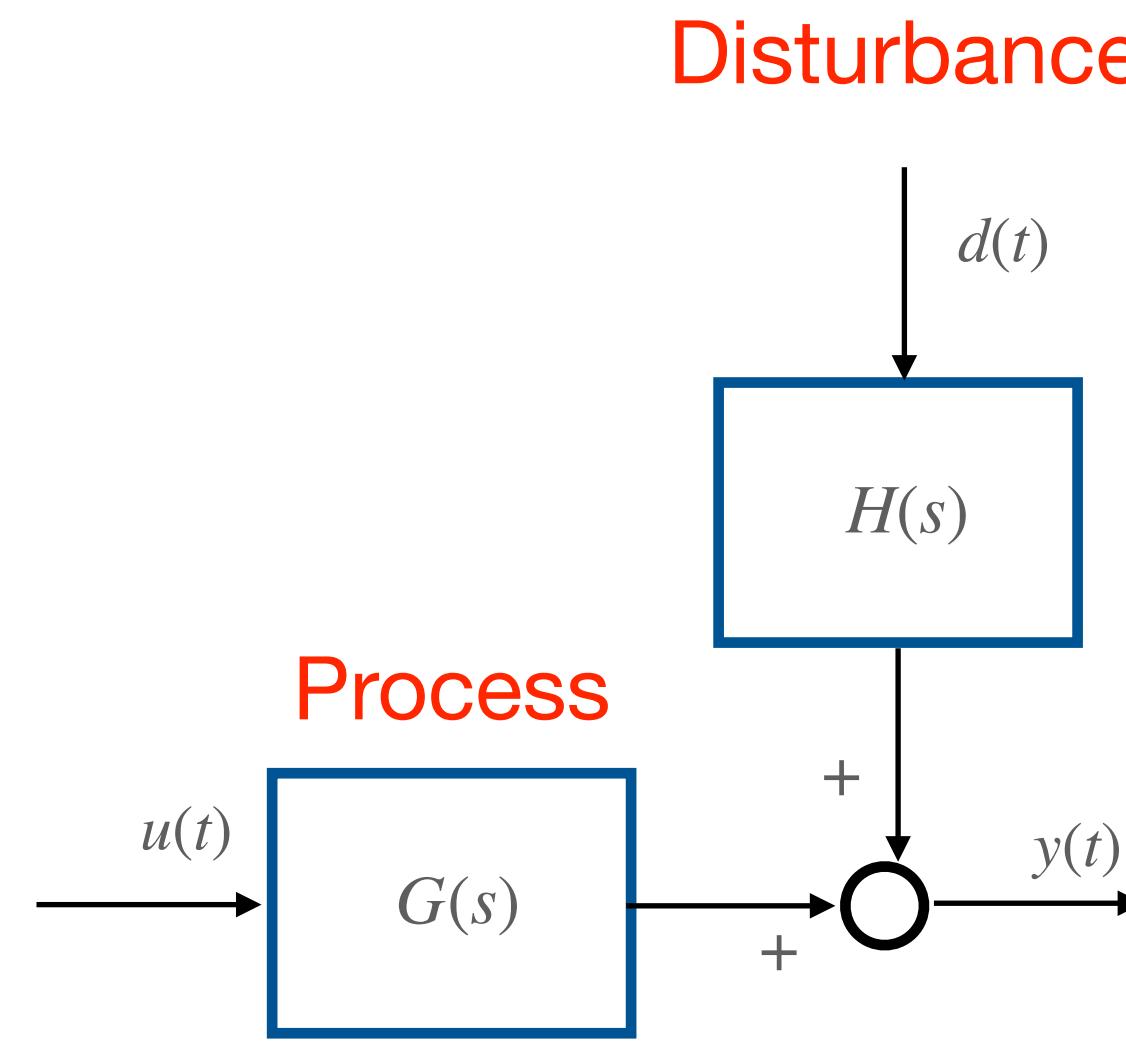
## Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



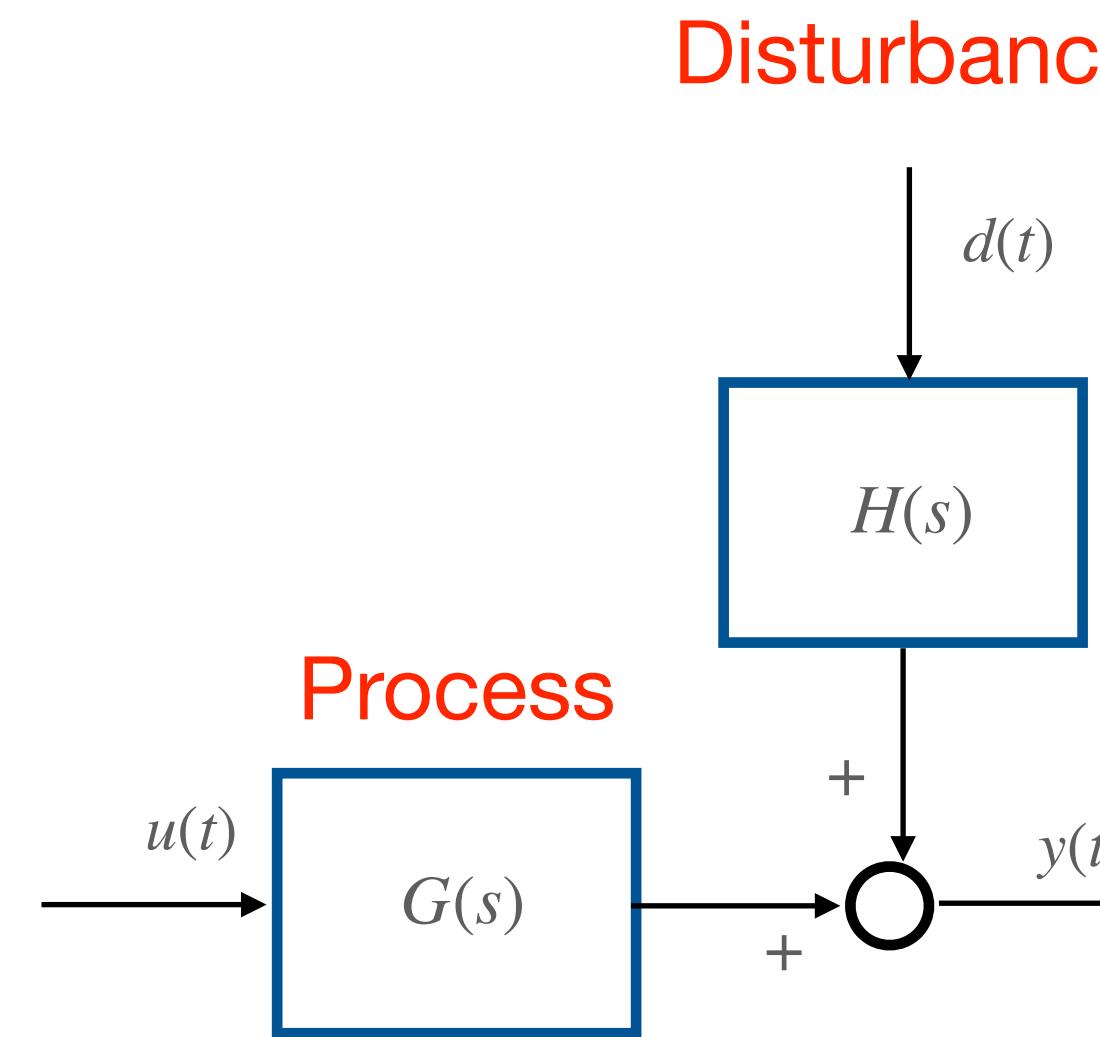
## Prefilter-based and Parallel-compensator Based Control Schemes: 2DoF Control Schemes



## Control Scheme with Measurable Disturbance Compensation



## Control Scheme with Measurable Disturbance Compensation



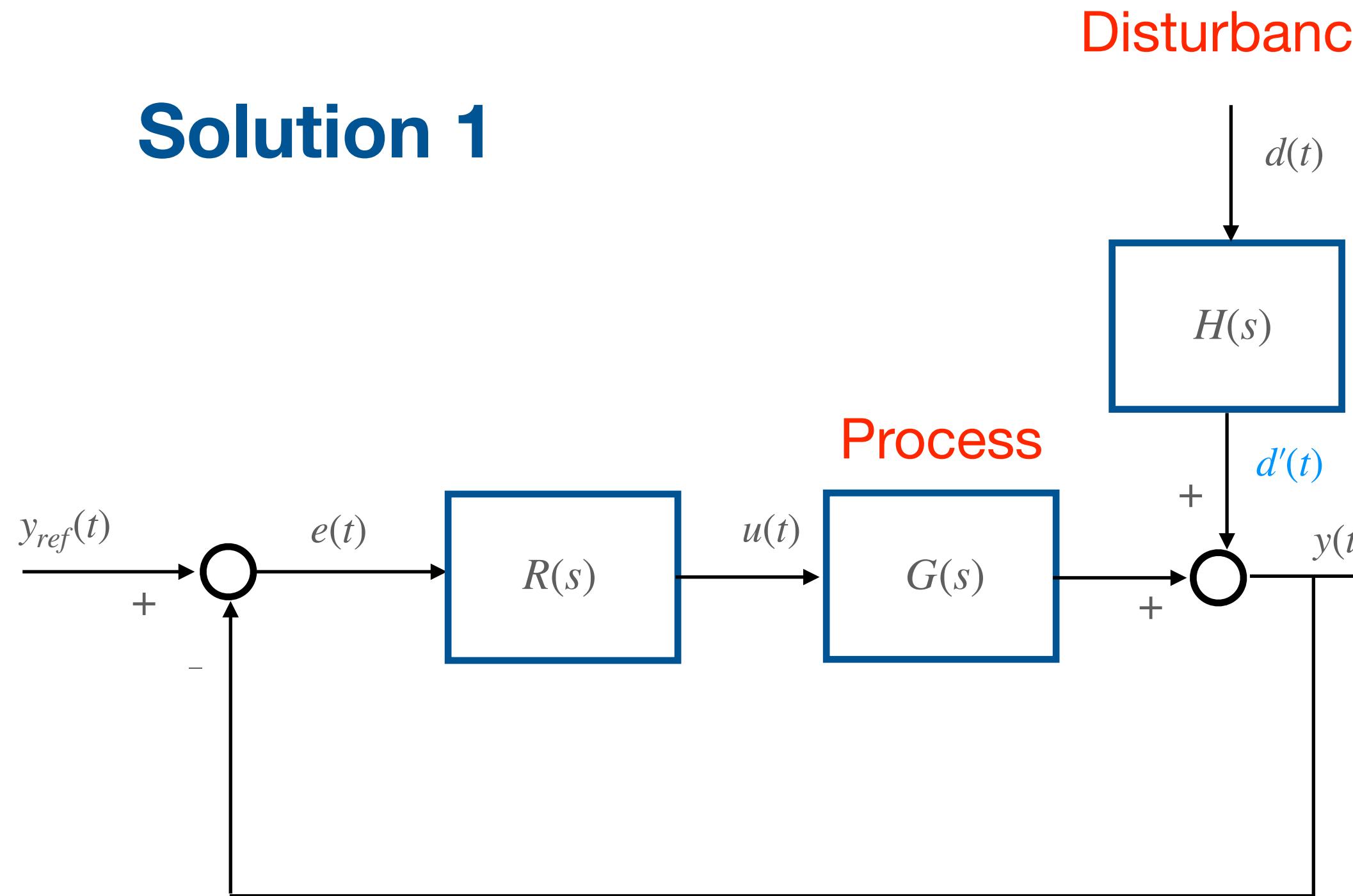
Assumptions:

- $d(t)$  measurable
- $H(s)$  known (possibly 1)



## Control Scheme with Measurable Disturbance Compensation

### Solution 1

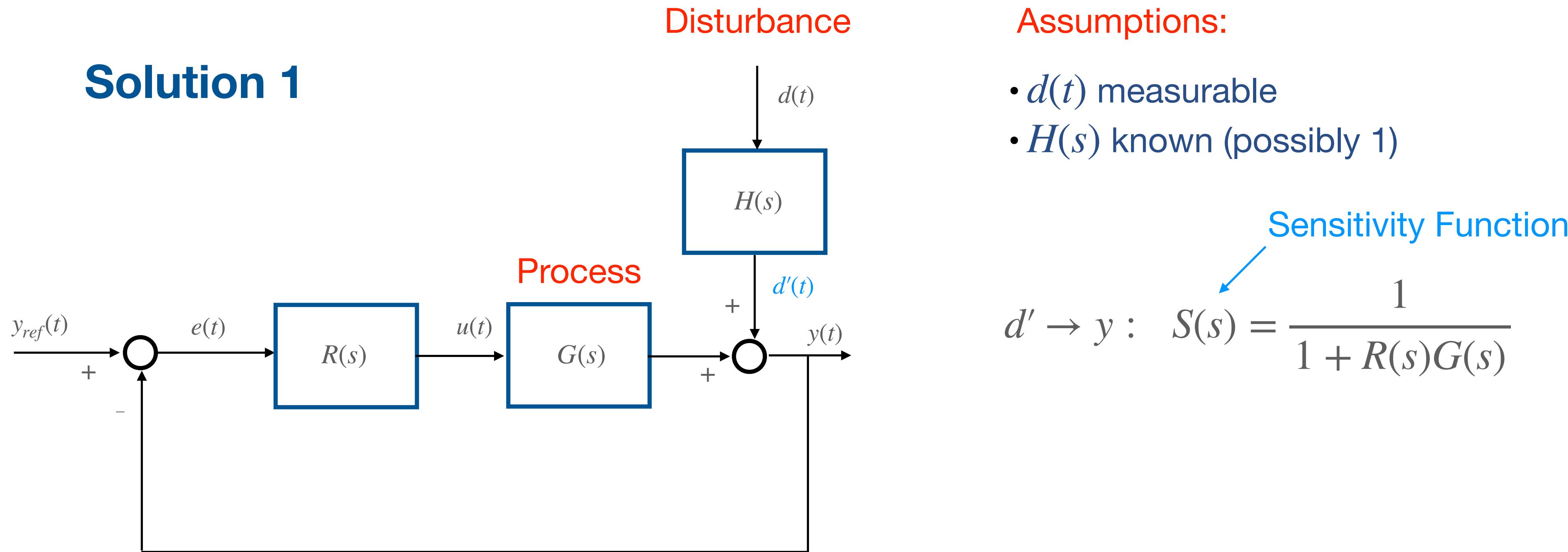


Assumptions:

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## Control Scheme with Measurable Disturbance Compensation

### Solution 1



Assumptions:

- $d(t)$  measurable
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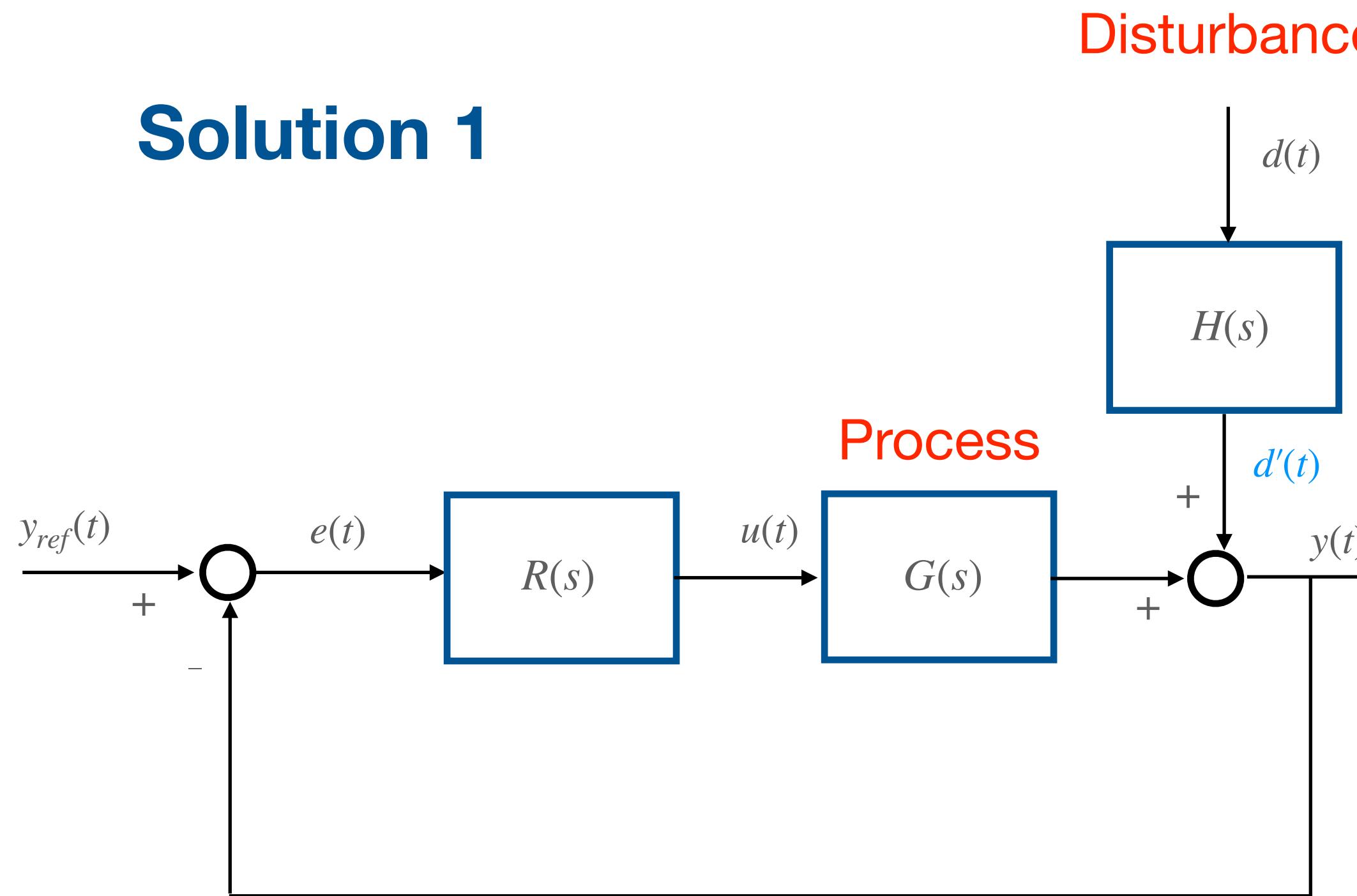
Sensitivity Function

$$d' \rightarrow y : S(s) = \frac{1}{1 + R(s)G(s)}$$



## Control Scheme with Measurable Disturbance Compensation

### Solution 1



Disturbance

Assumptions:

- $d(t)$  measurable
- $H(s)$  known (possibly 1)

**Sensitivity Function**

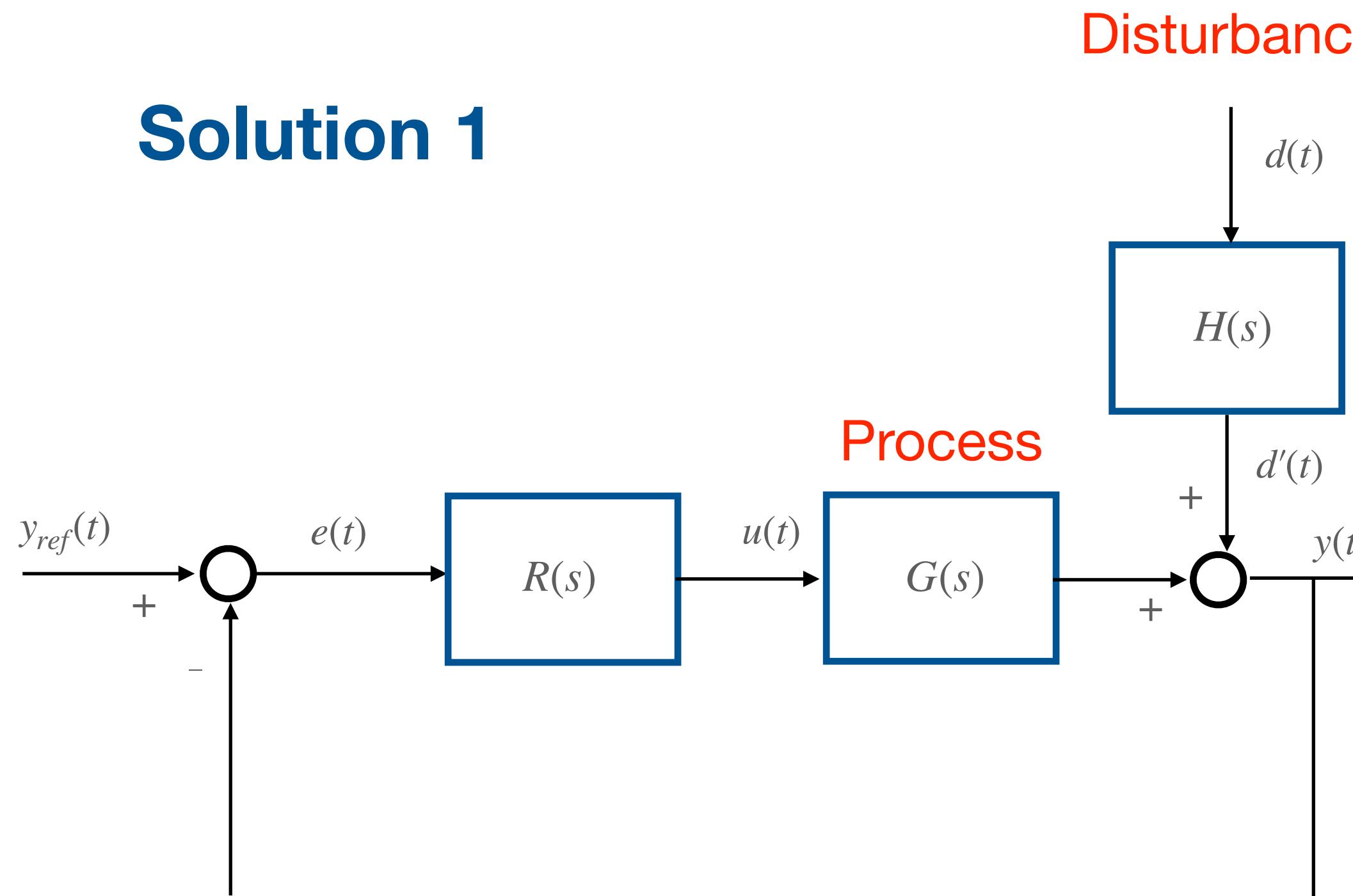
$$d' \rightarrow y : S(s) = \frac{1}{1 + R(s)G(s)}$$

Recommendation:  $|S(j\omega)|$  sufficiently low  $\forall \omega$   
in the frequency range where  $d'(t)$   
has significant harmonics



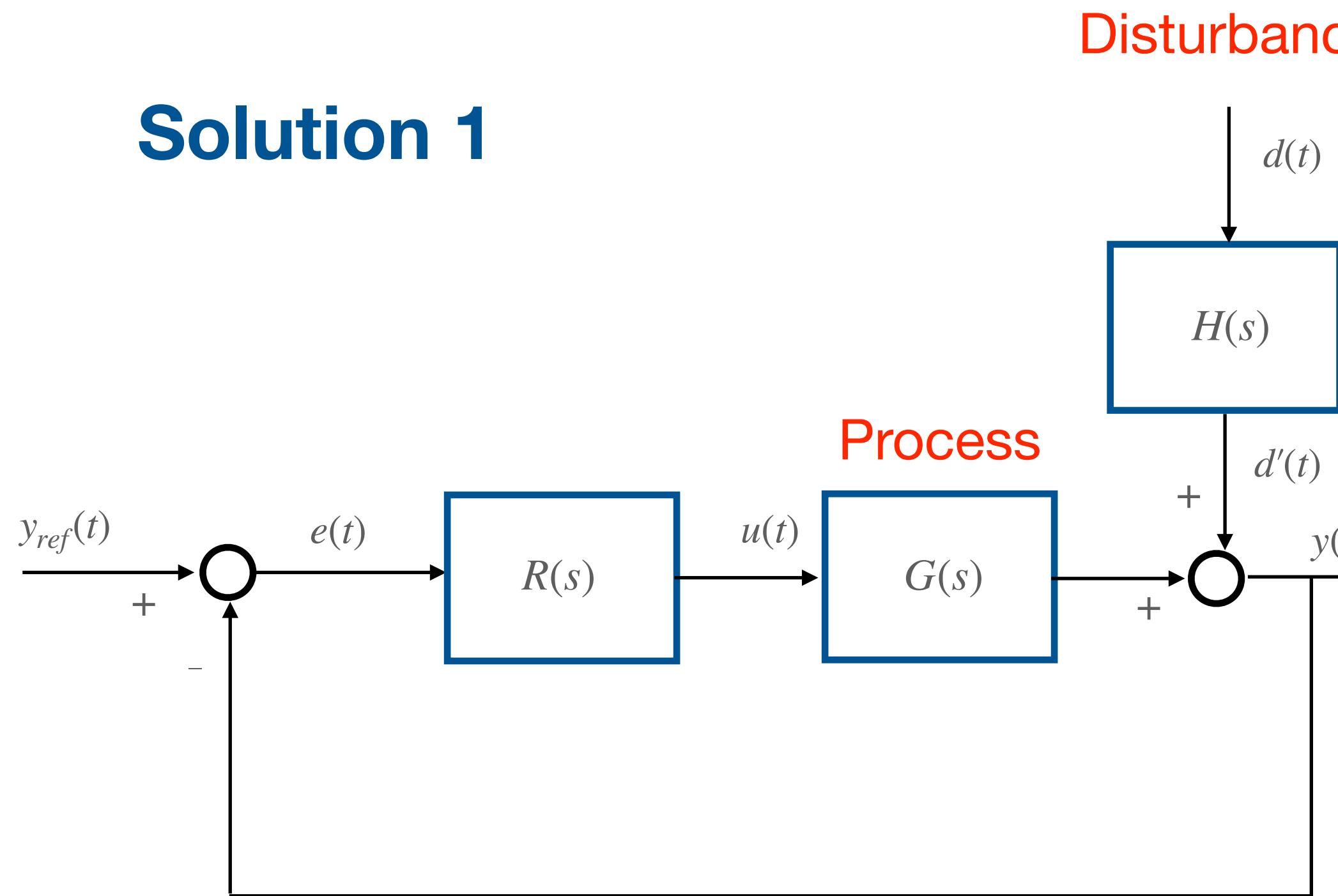
## Control Scheme with Measurable Disturbance Compensation

### Solution 1



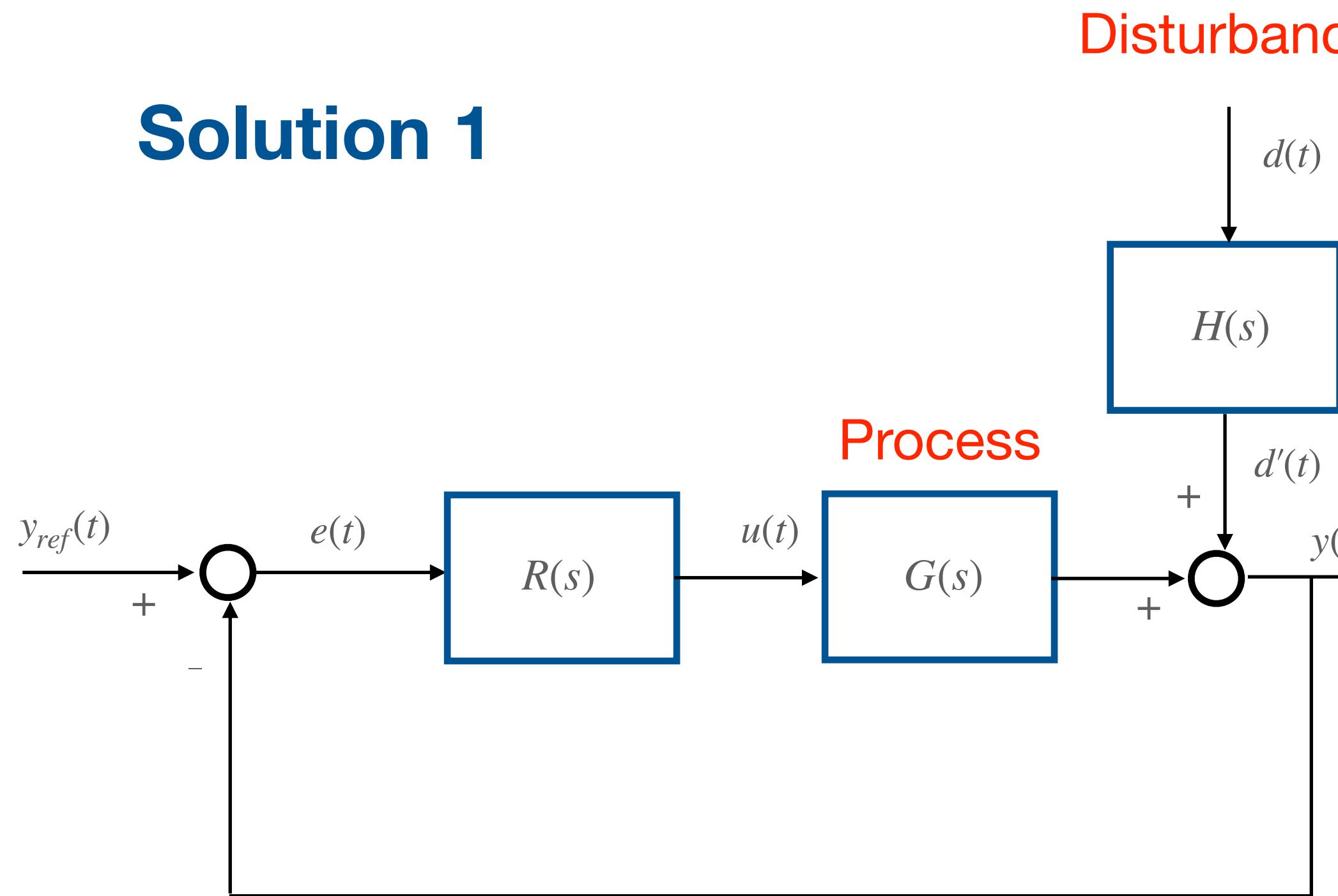
## Control Scheme with Measurable Disturbance Compensation

### Solution 1



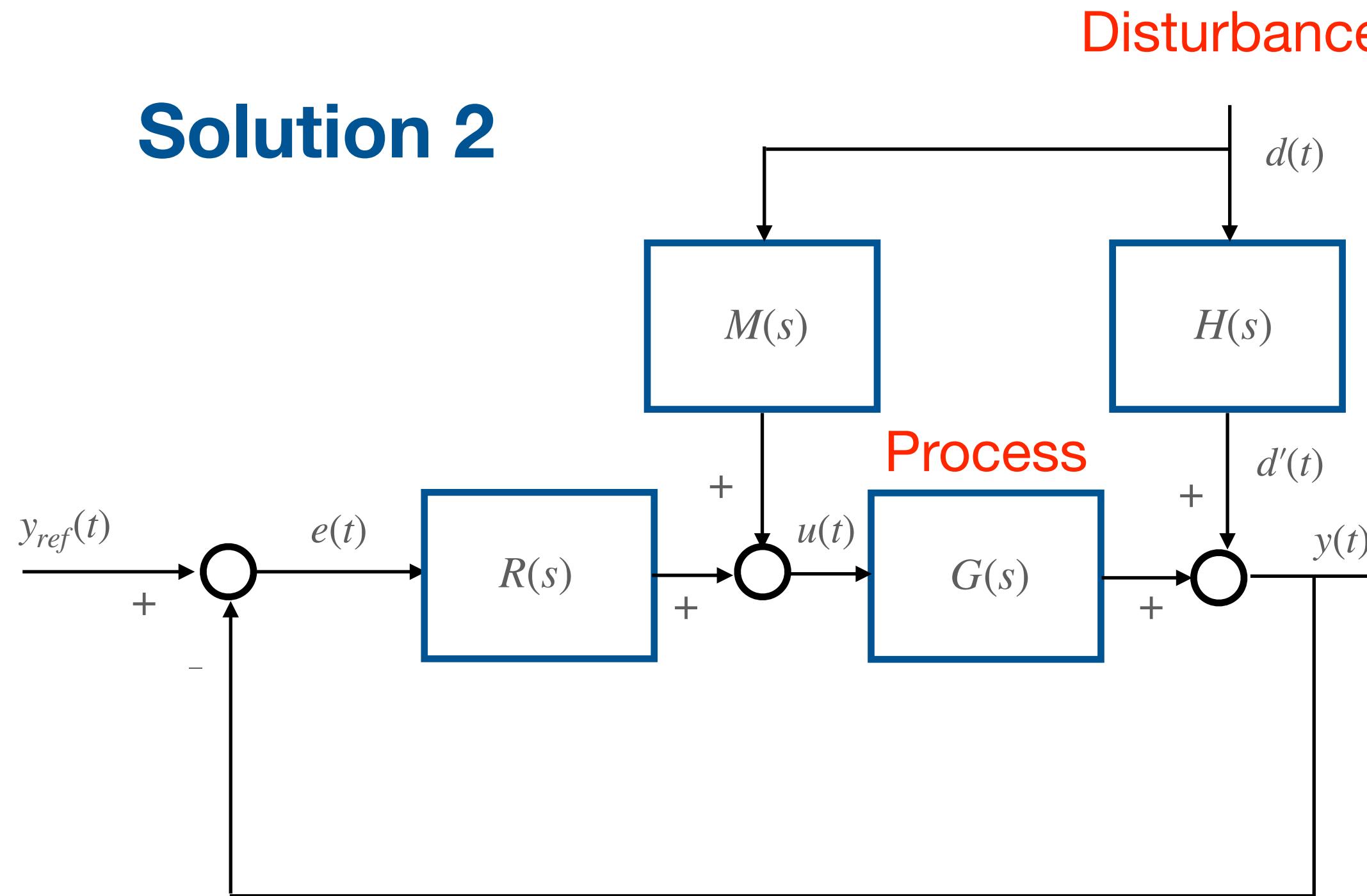
## Control Scheme with Measurable Disturbance Compensation

### Solution 1



## Control Scheme with Measurable Disturbance Compensation

### Solution 2



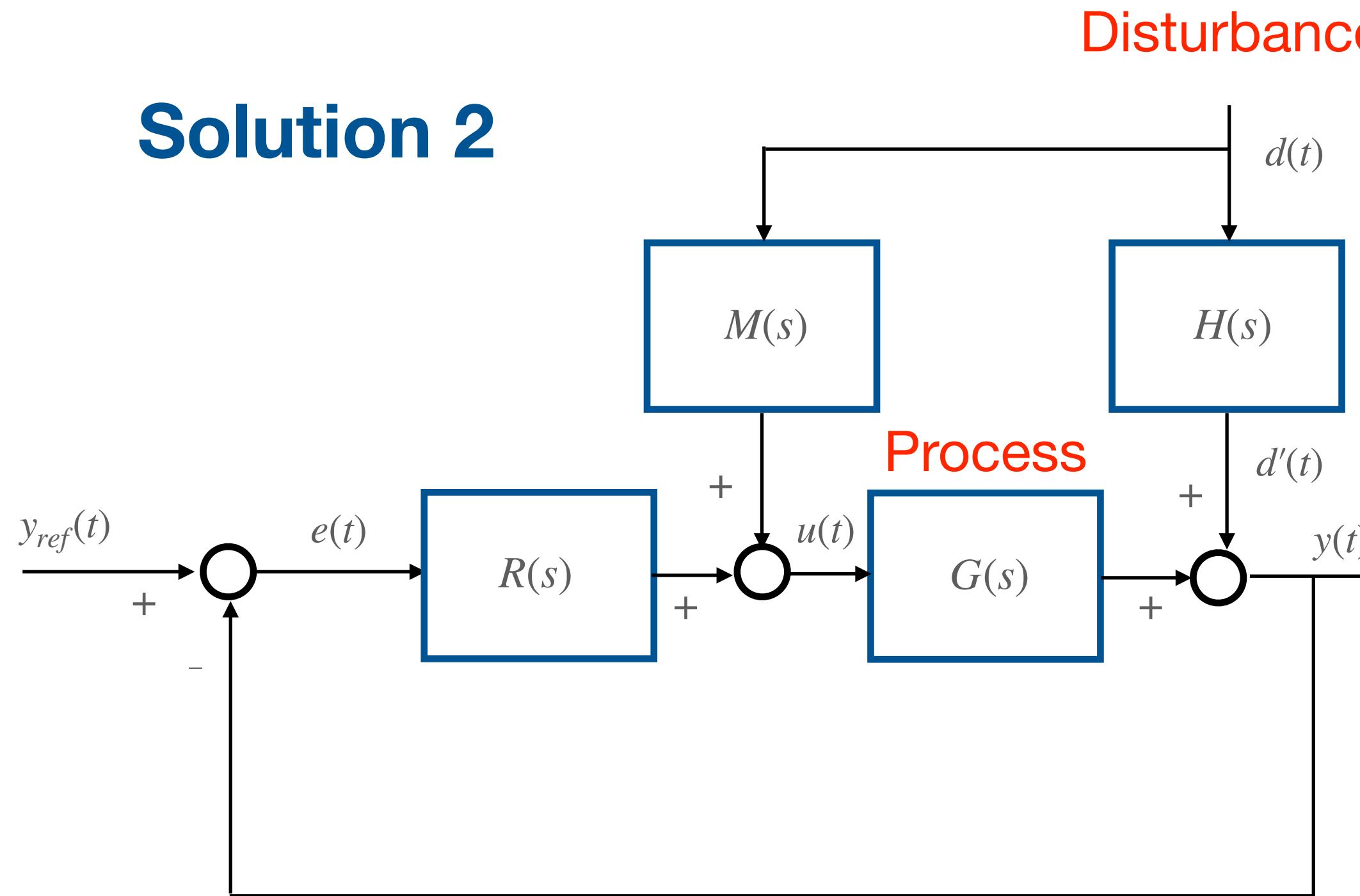
Assumptions:

- $d(t)$  measurable
- $H(s)$  known (possibly 1)



## Control Scheme with Measurable Disturbance Compensation

### Solution 2



Assumptions:

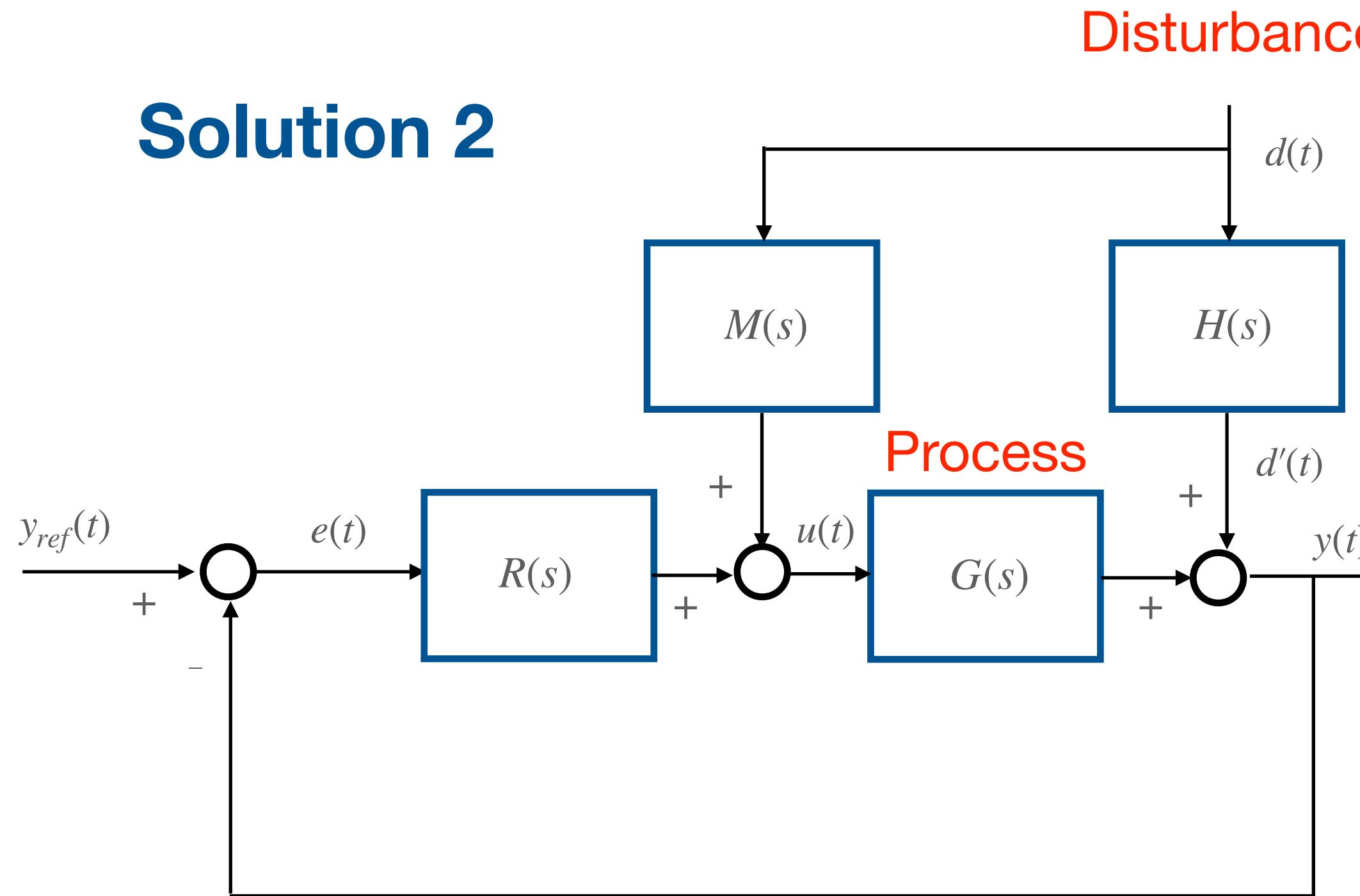
- $d(t)$  measurable
- $H(s)$  known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)}$$



## Control Scheme with Measurable Disturbance Compensation

### Solution 2



Assumptions:

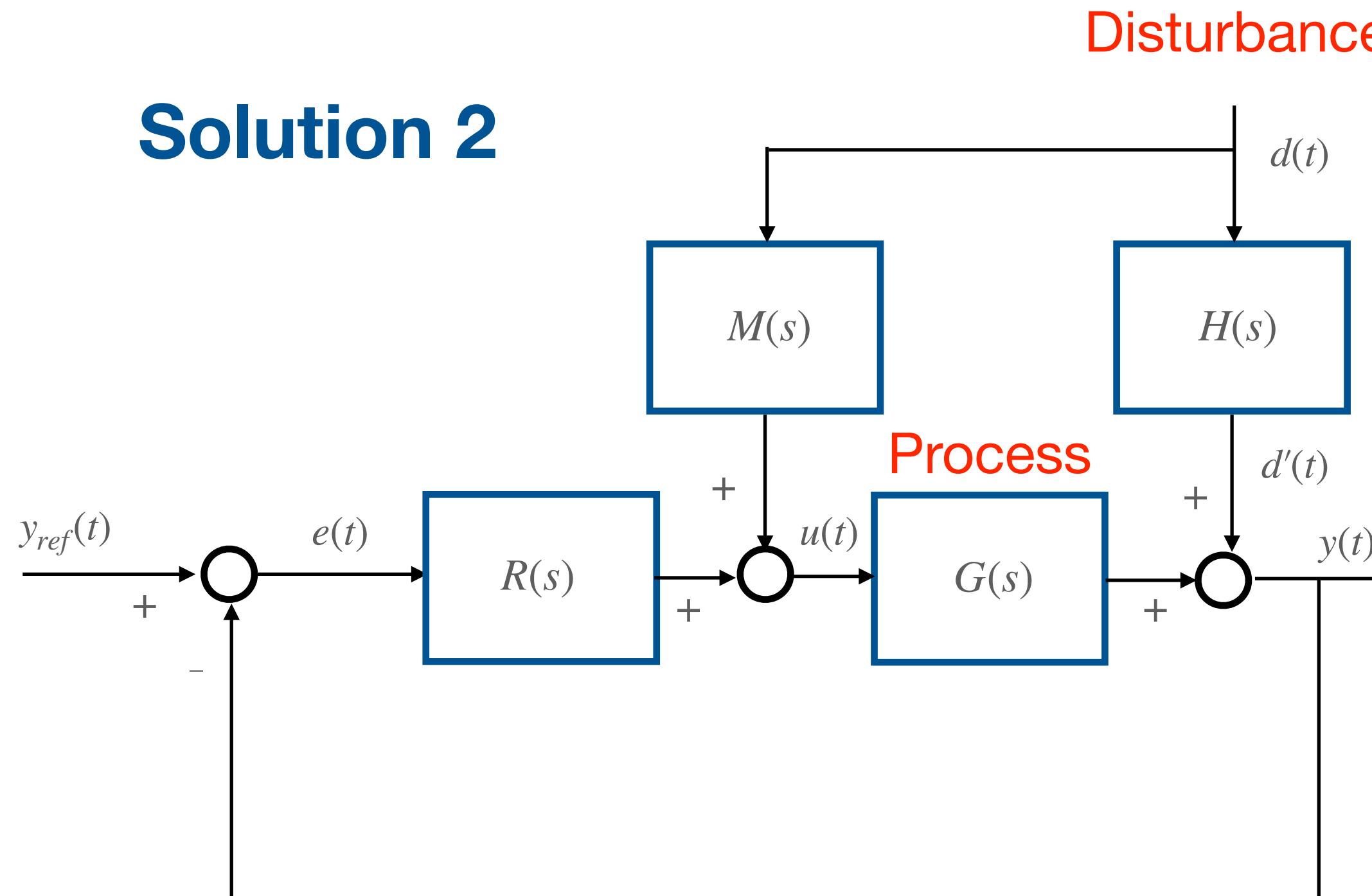
- $d(t)$  measurable
- $H(s)$  known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$



## Control Scheme with Measurable Disturbance Compensation

### Solution 2



Assumptions:

- $d(t)$  measurable
- $H(s)$  known (possibly 1)

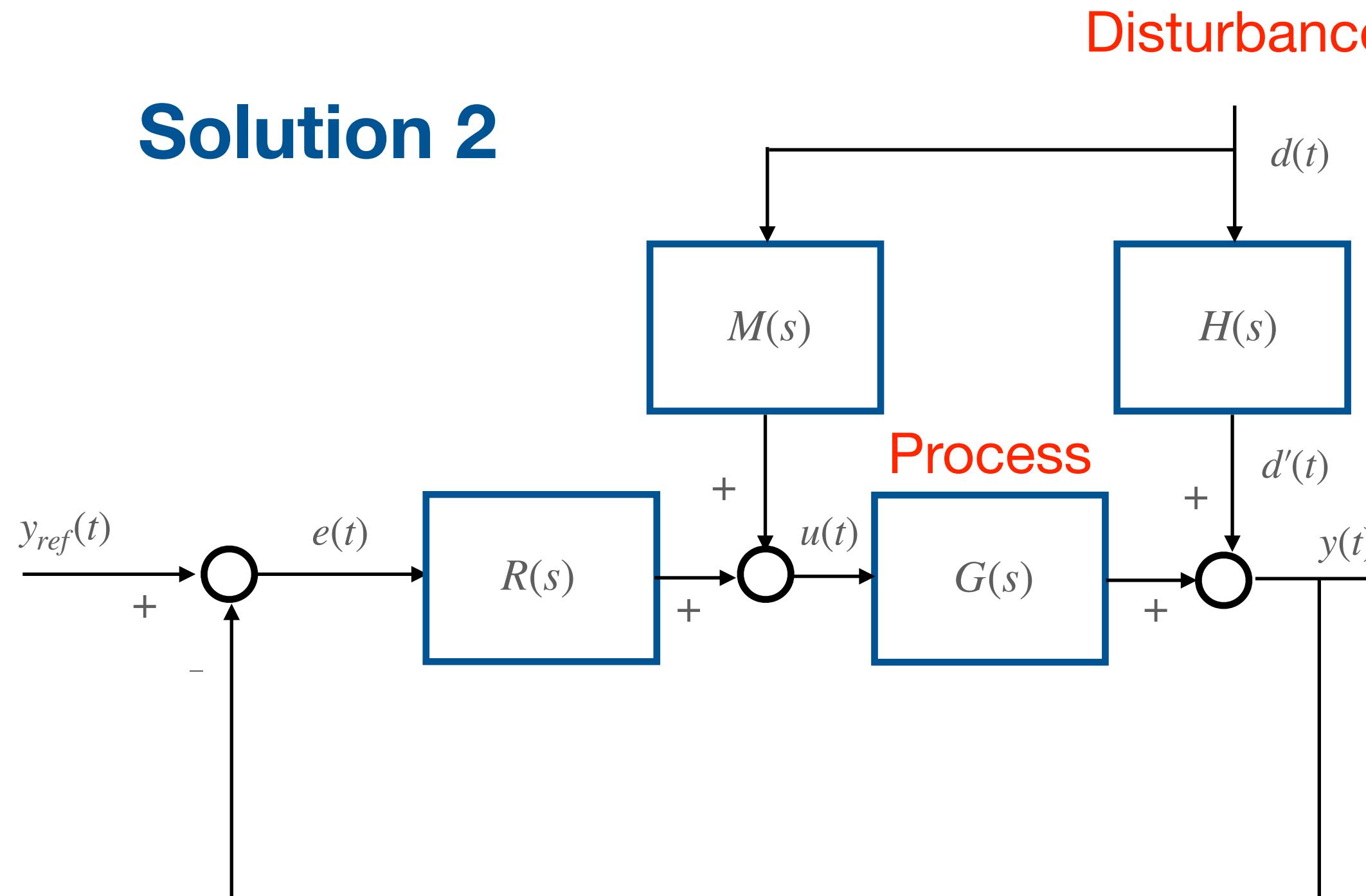
$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

$$M(s) = -\frac{H(s)}{G(s)}$$



## Control Scheme with Measurable Disturbance Compensation

### Solution 2



Disturbance

Assumptions:

- $d(t)$  measurable
- $H(s)$  known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

Warning:

- $M(s)$  As. Stable
- Causal

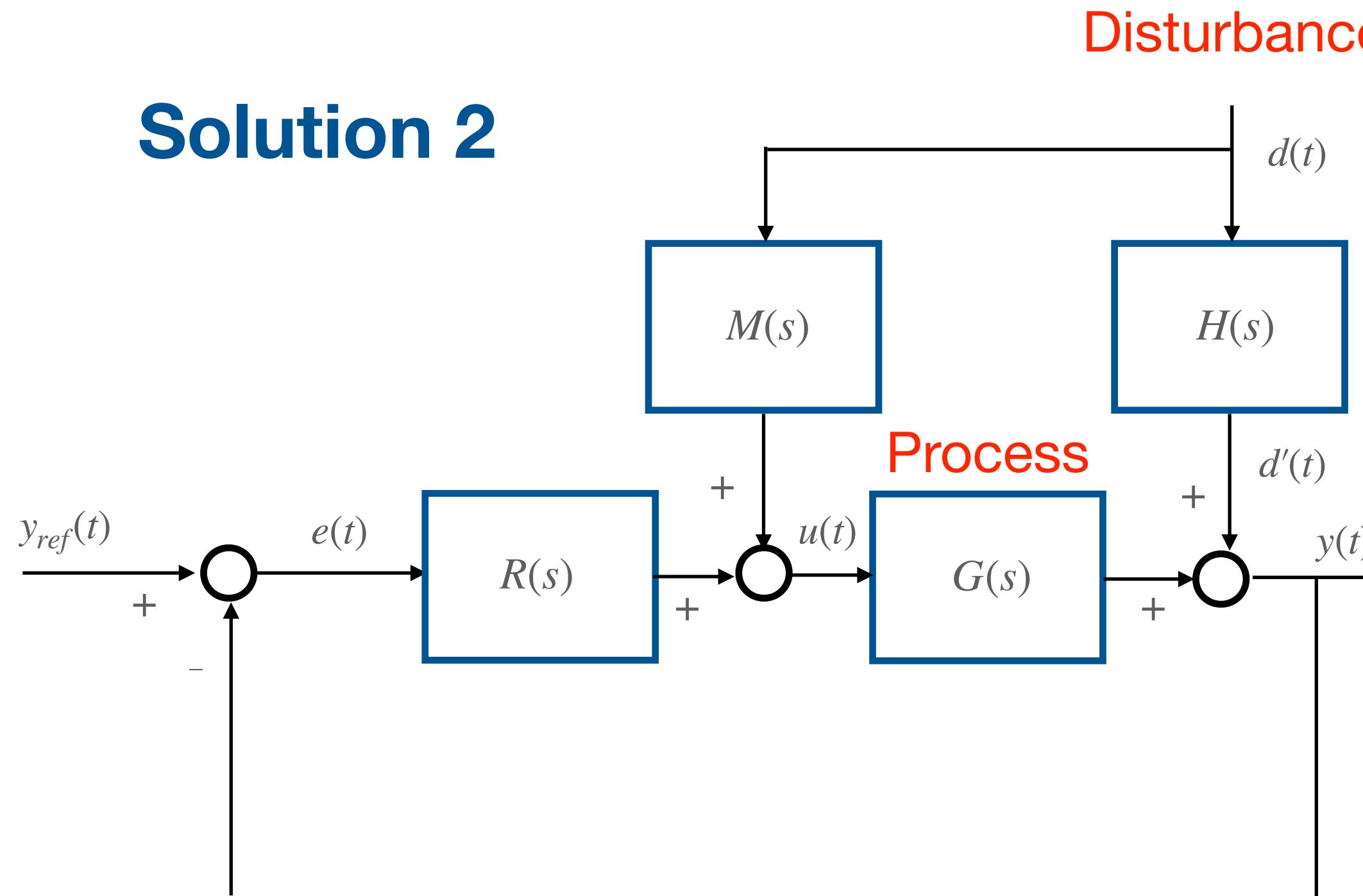
Ideal Design

$$M(s) = -\frac{H(s)}{G(s)}$$



## Control Scheme with Measurable Disturbance Compensation

### Solution 2



Assumptions:

- $d(t)$  measurable
- $H(s)$  known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

Ideal Design

$$M(s) = -\frac{H(s)}{G(s)}$$

not guaranteed

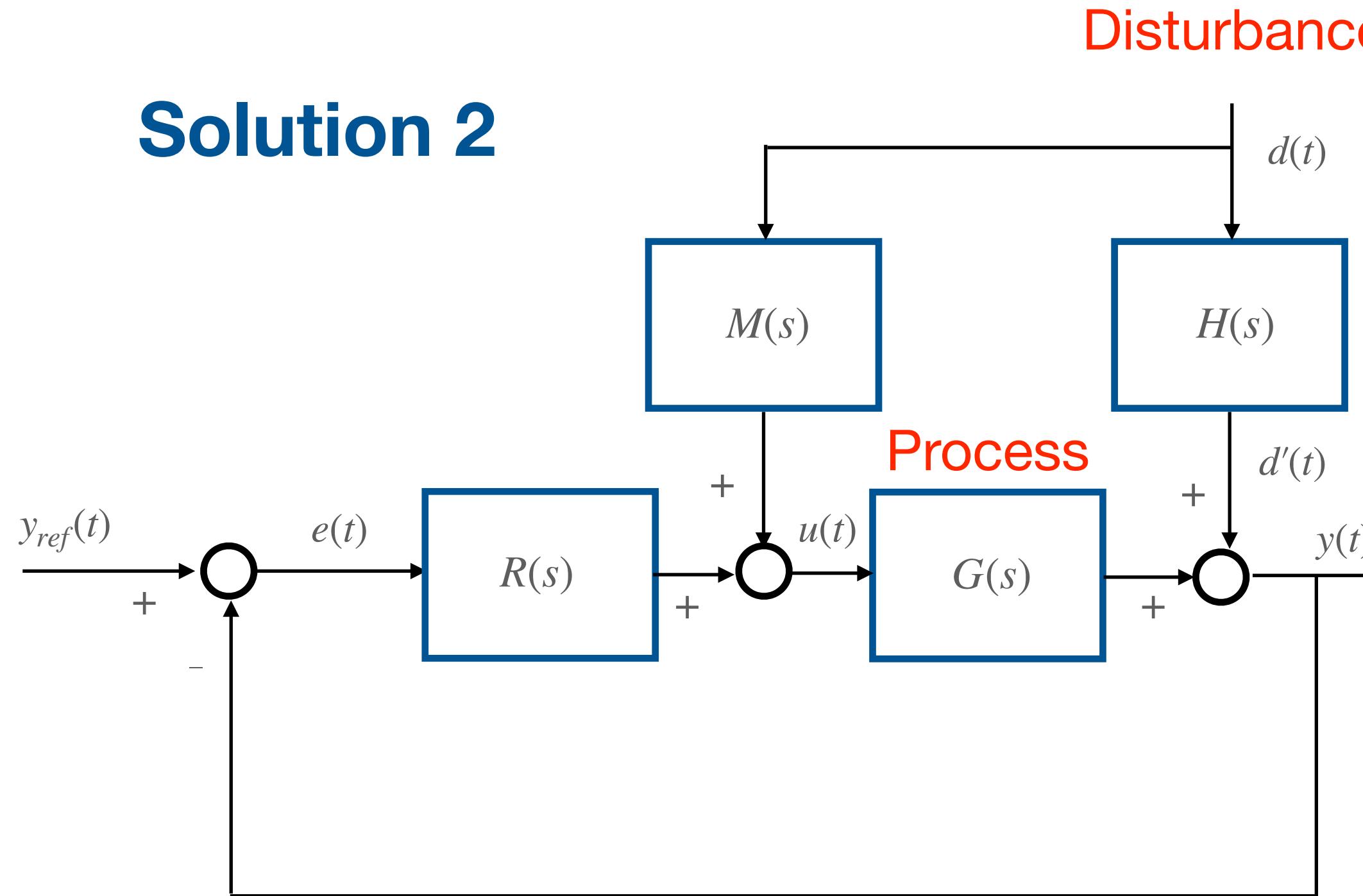
Warning:

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## Control Scheme with Measurable Disturbance Compensation

### Solution 2



Assumptions:

- $d(t)$  measurable
- $H(s)$  known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

not guaranteed

Warning:

- $M(s)$  As. Stable
- Causal

in the frequency range where  $d(t)$   
has significant harmonics

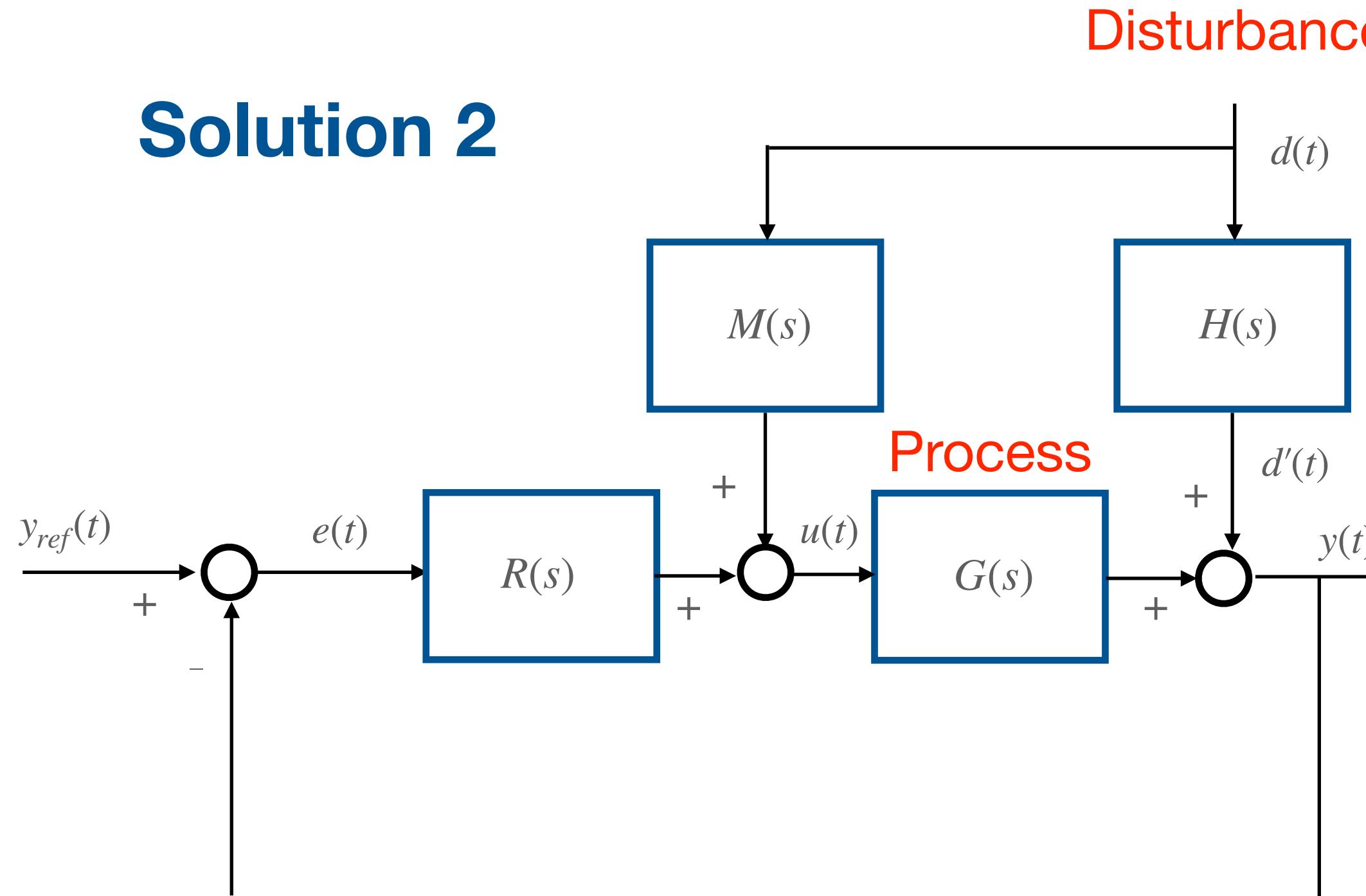
Actual Design

$$M(j\omega) = -\frac{H(j\omega)}{G(j\omega)}$$



## Control Scheme with Measurable Disturbance Compensation

### Solution 2



### Assumptions:

- $d(t)$  measurable
- $H(s)$  known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

not guaranteed

### Warning:

- $M(s)$  As. Stable
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### Actual Design

$$M(j\omega) = -\frac{H(j\omega)}{G(j\omega)}$$

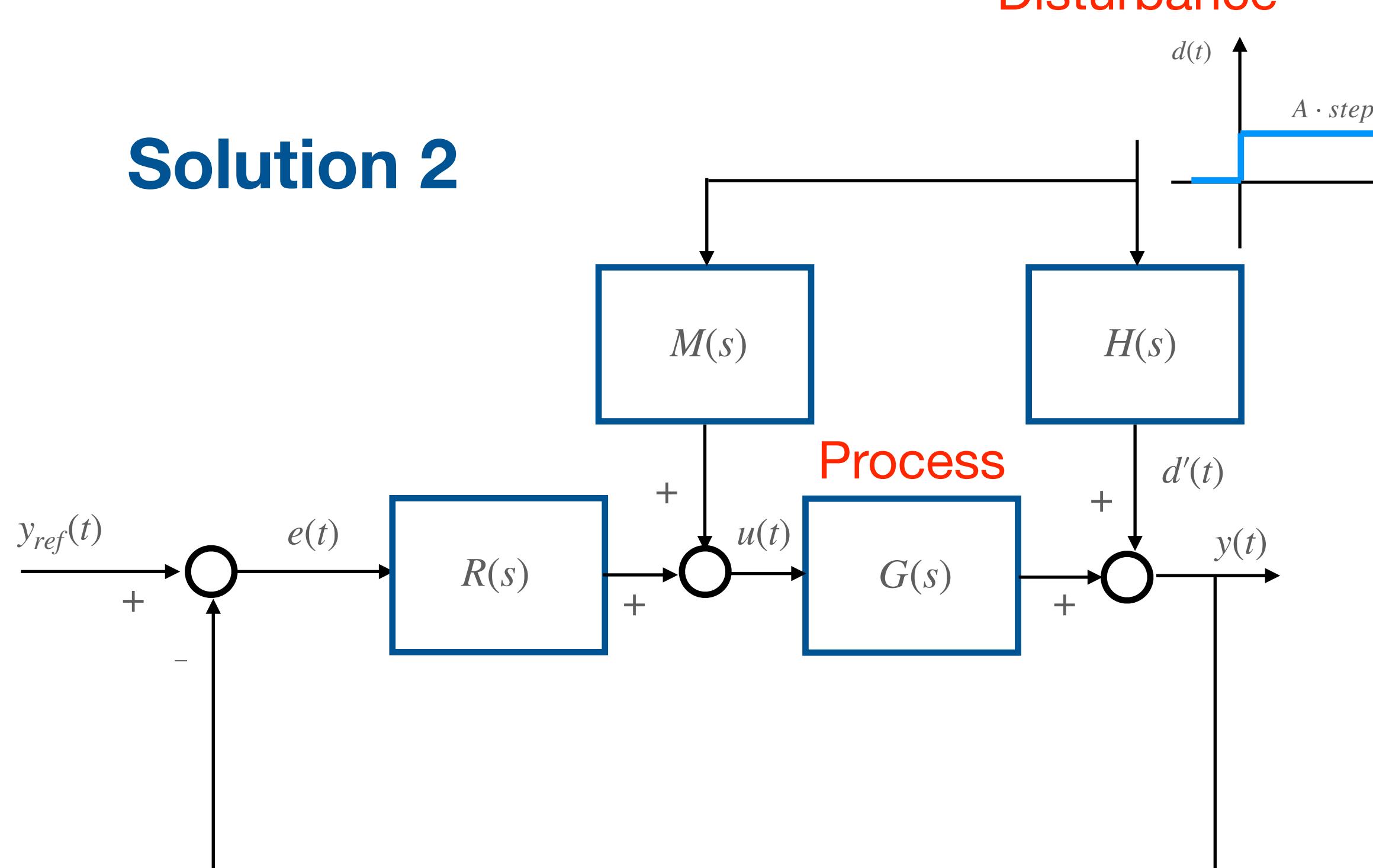
### Example 1:

$$d(t) = \sin(\hat{\omega} t) \longrightarrow M(s) \text{ s.t. } M(j\omega) \Big|_{\omega=\hat{\omega}} = -\frac{H(j\hat{\omega})}{G(j\hat{\omega})}$$



## Control Scheme with Measurable Disturbance Compensation

### Solution 2



### Assumptions:

- $d(t)$  measurable
- $H(s)$  known (possibly 1)

$$d \rightarrow y : \frac{H(s) + M(s)G(s)}{1 + R(s)G(s)} \approx 0$$

not guaranteed

### Warning:

- $M(s)$  As. Stable
- Causal

### Actual Design

$$M(j\omega) = -\frac{H(j\omega)}{G(j\omega)}$$

Example 2:  
static case

$$\hat{\omega} = 0$$

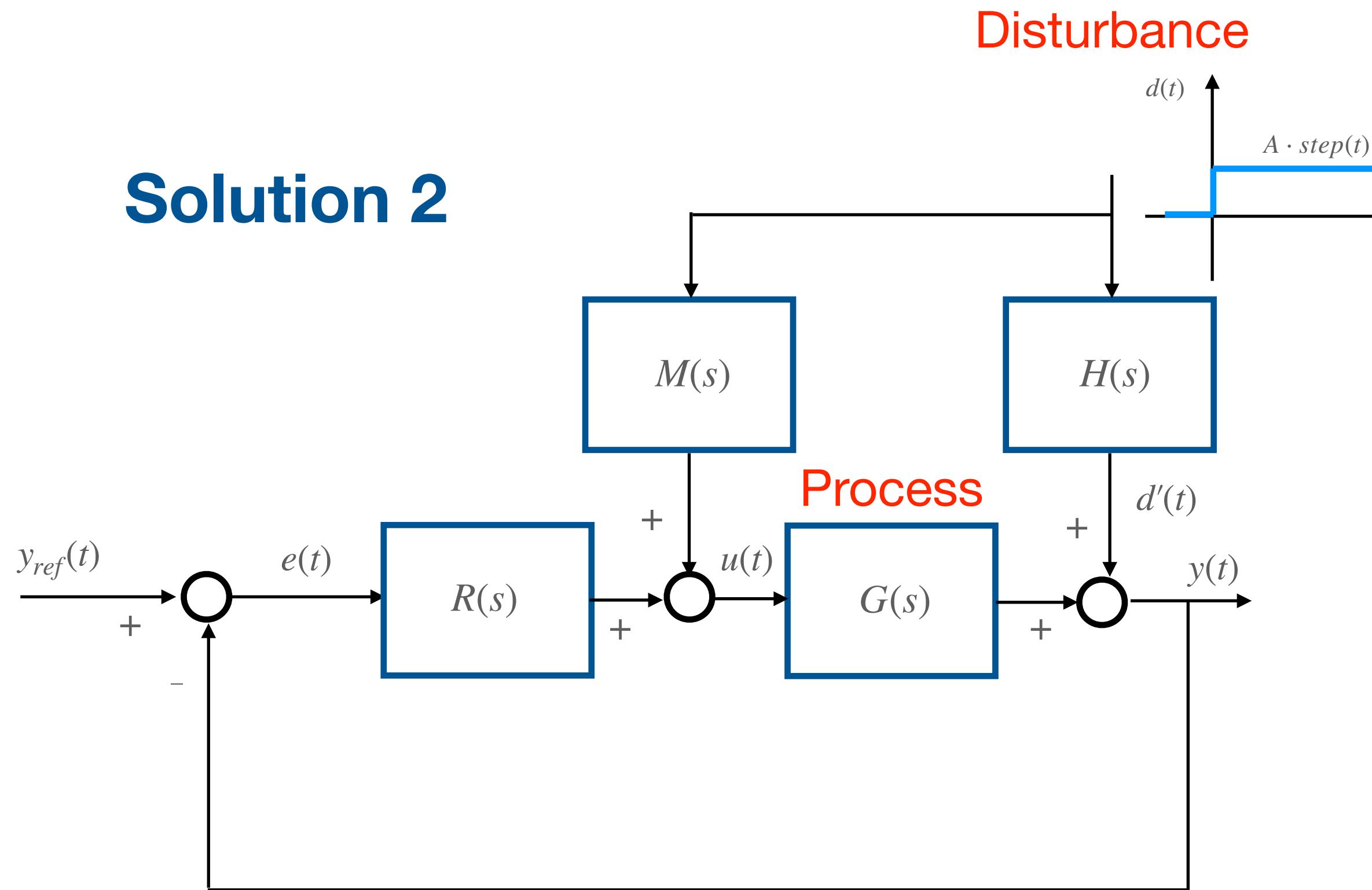
$M(s)$  s.t.

$$M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

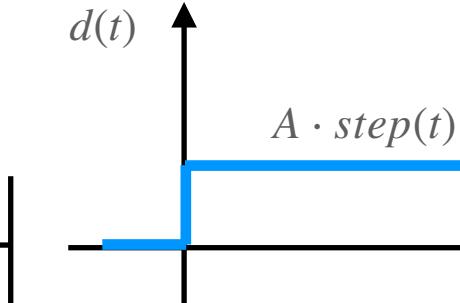


## Control Scheme with Measurable Disturbance Compensation

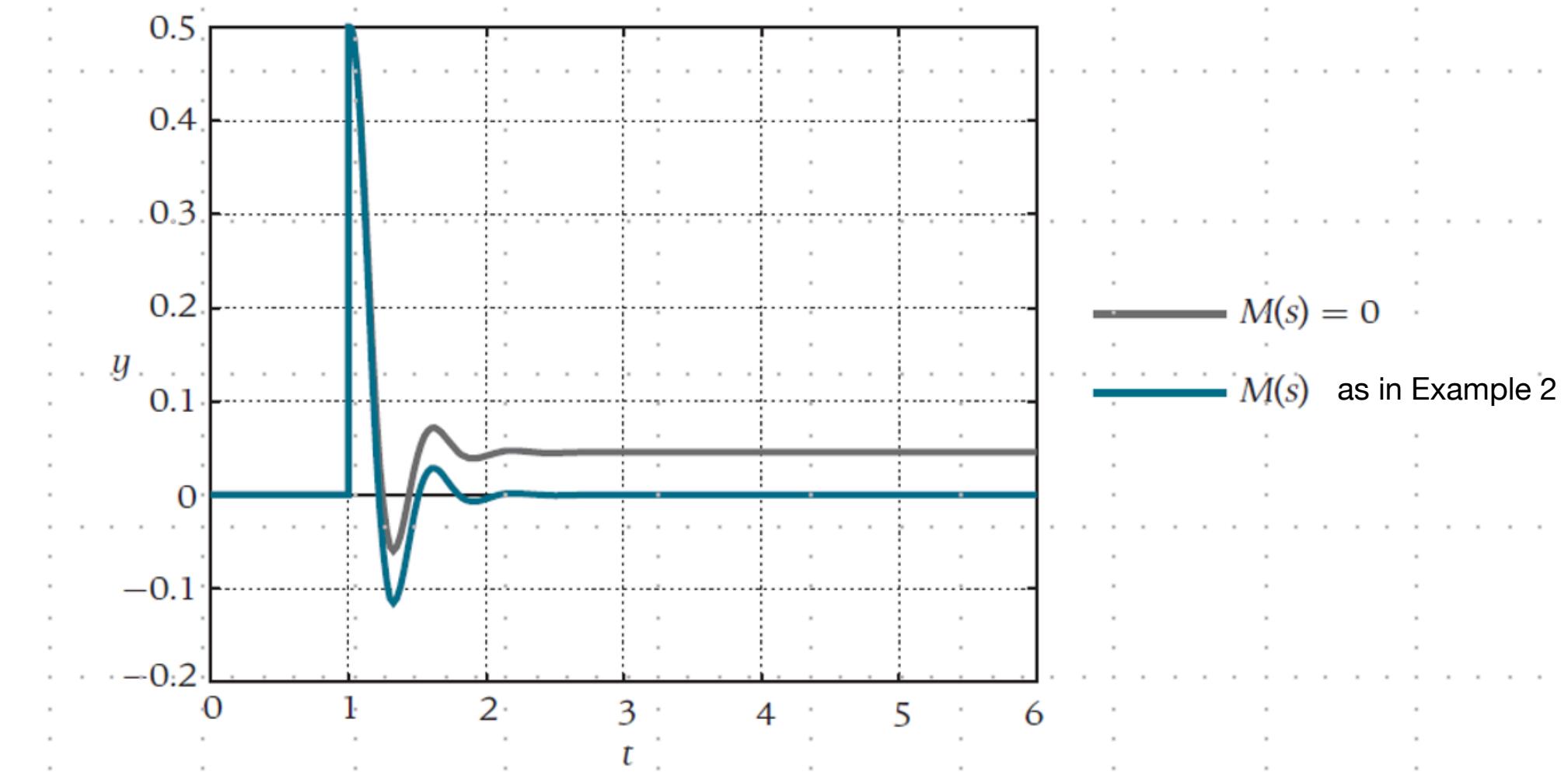
### Solution 2



Disturbance



Response to a step disturbance



Example 2:  
static case

$$\hat{\omega} = 0$$

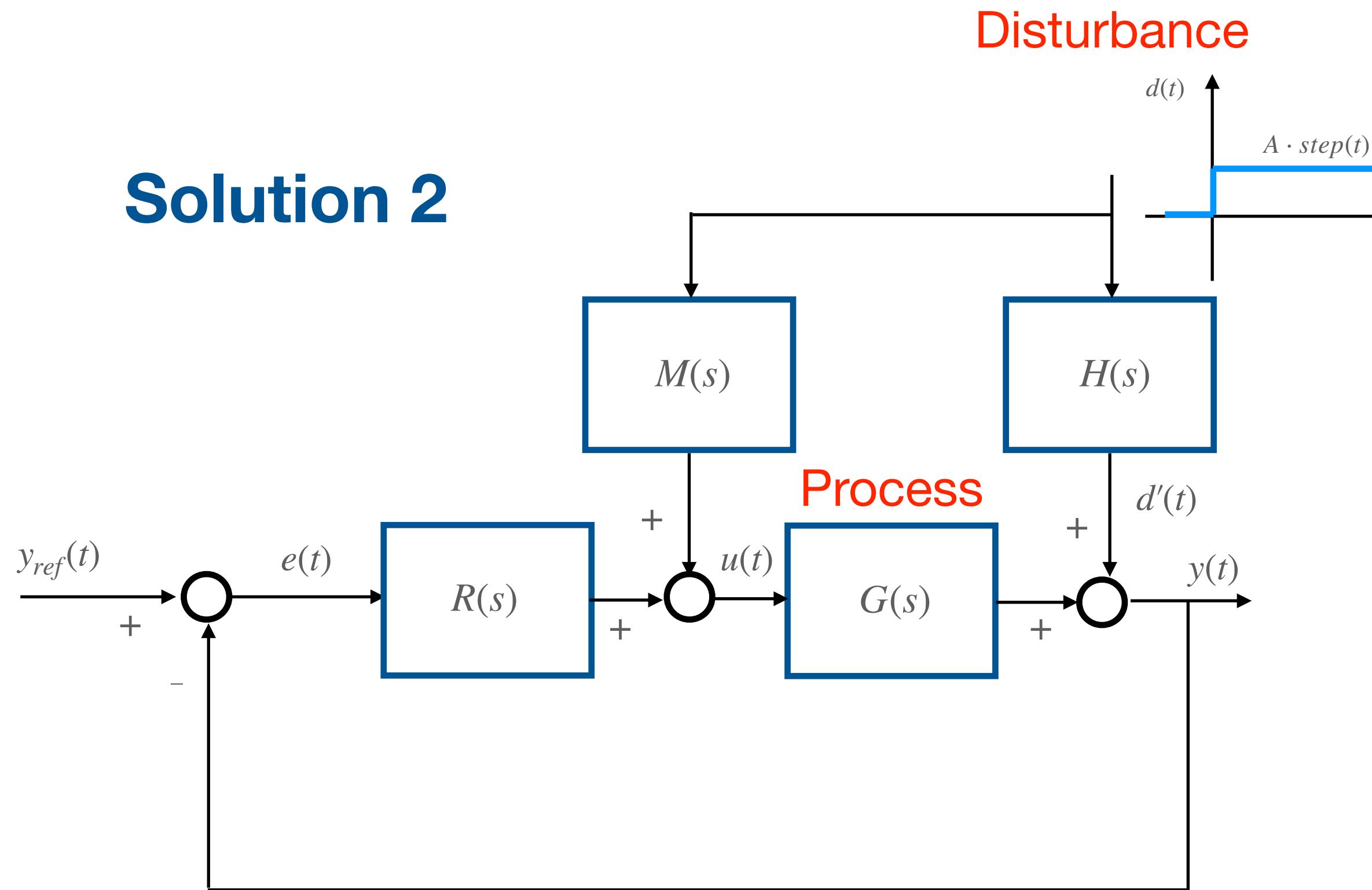
$\rightarrow M(s)$  s.t.

$$M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

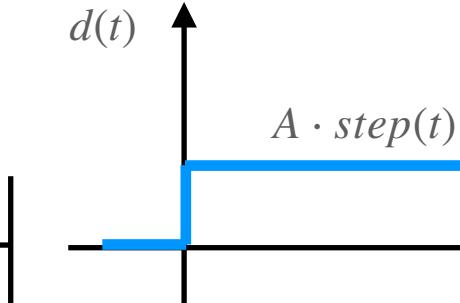


## Control Scheme with Measurable Disturbance Compensation

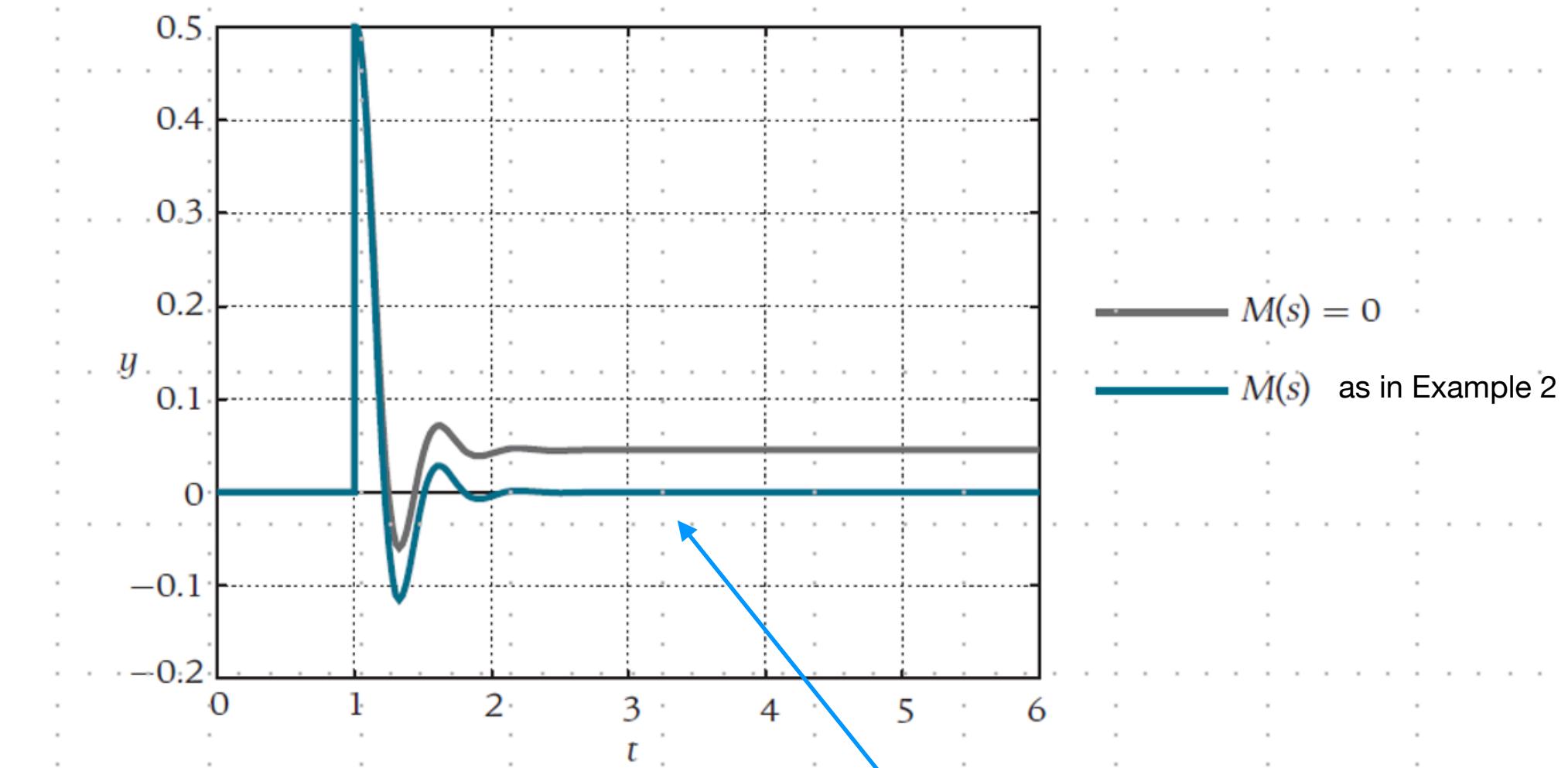
### Solution 2



Disturbance



Response to a step disturbance



Example 2:  
static case

$$\hat{\omega} = 0$$

$\longrightarrow M(s)$  s.t.

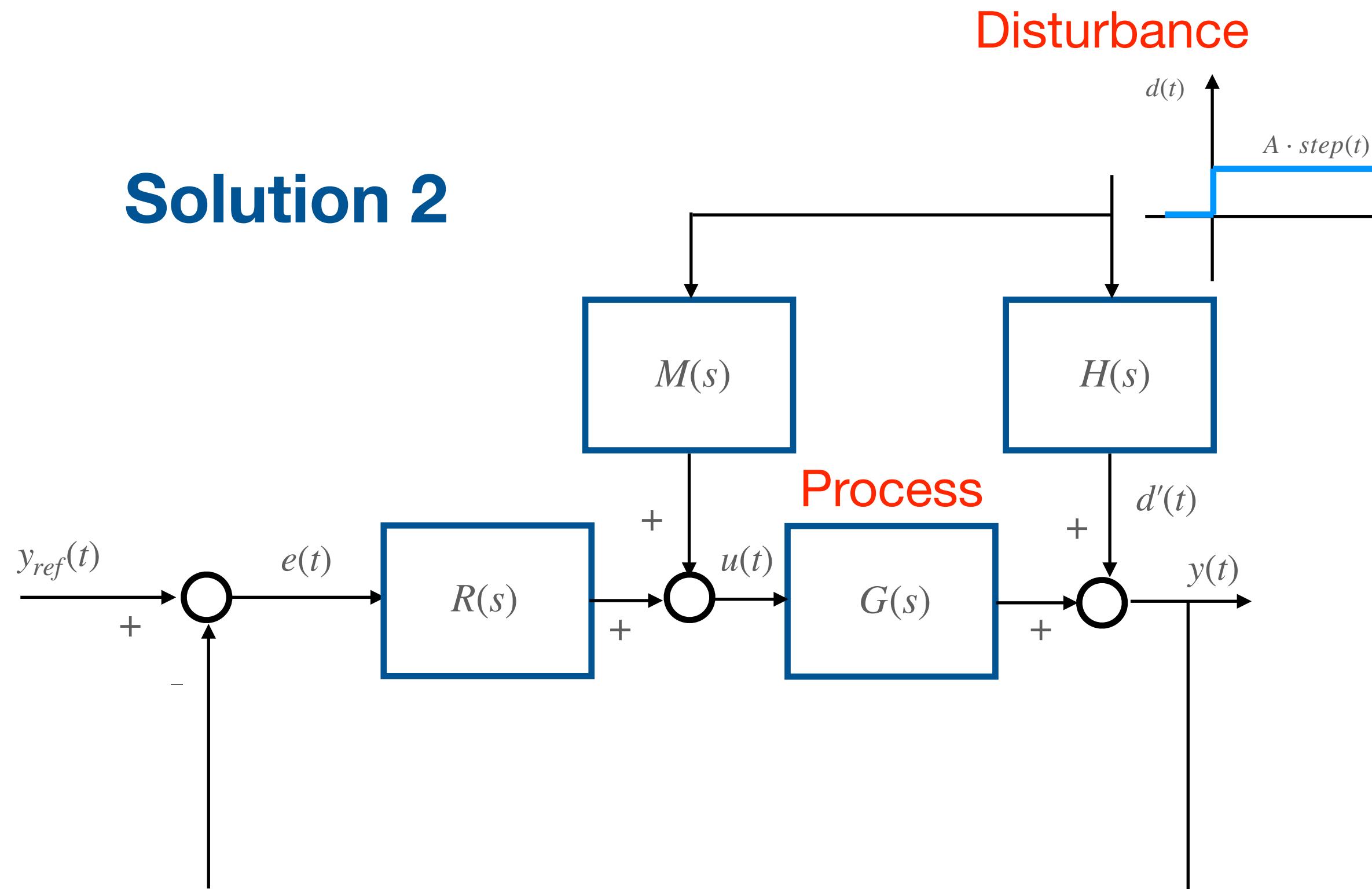
$$M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

the disturbance is rejected  
in steady-state

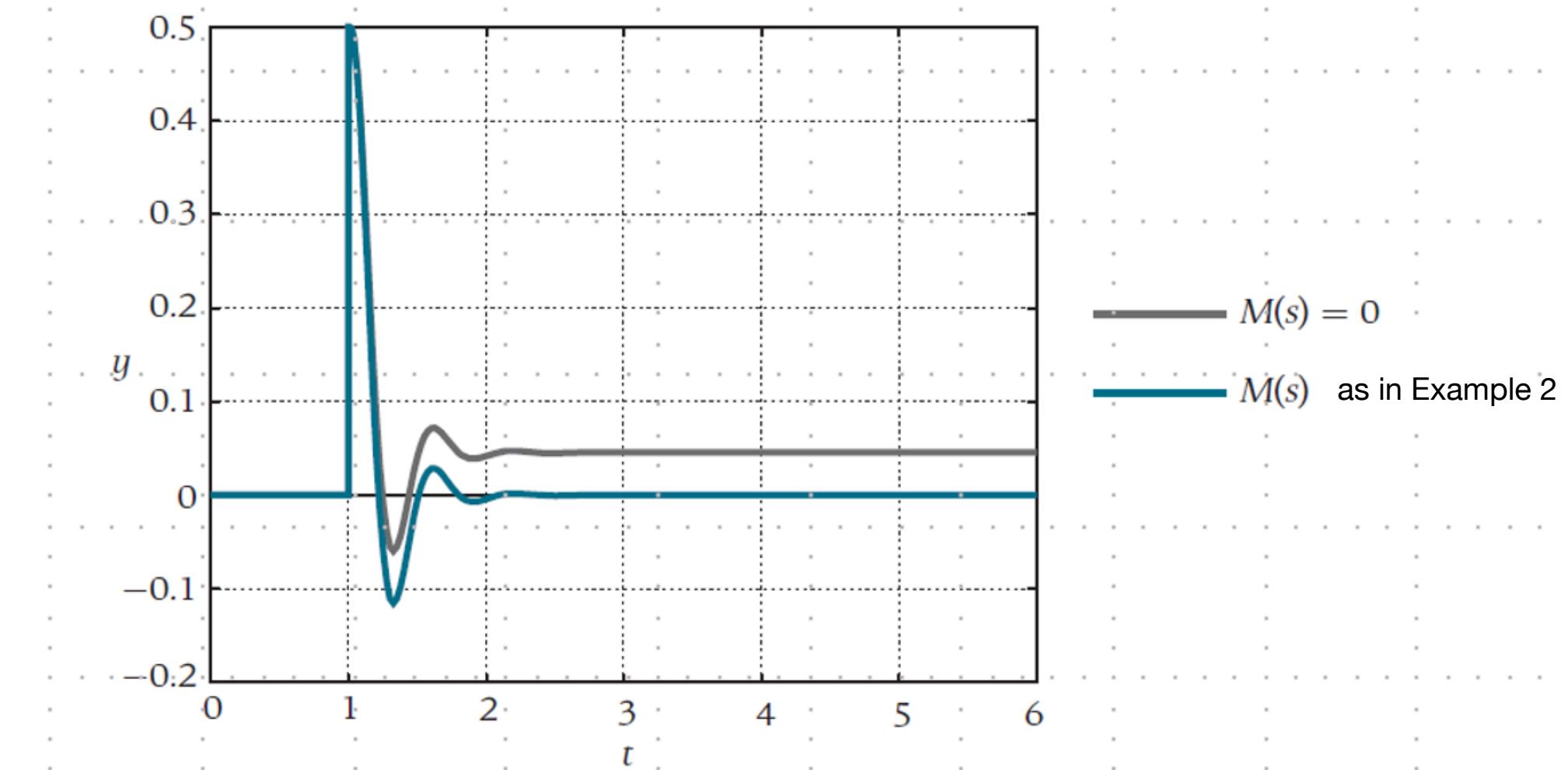


## Control Scheme with Measurable Disturbance Compensation

### Solution 2



Response to a step disturbance



Example 2:  
static case

$$\hat{\omega} = 0$$

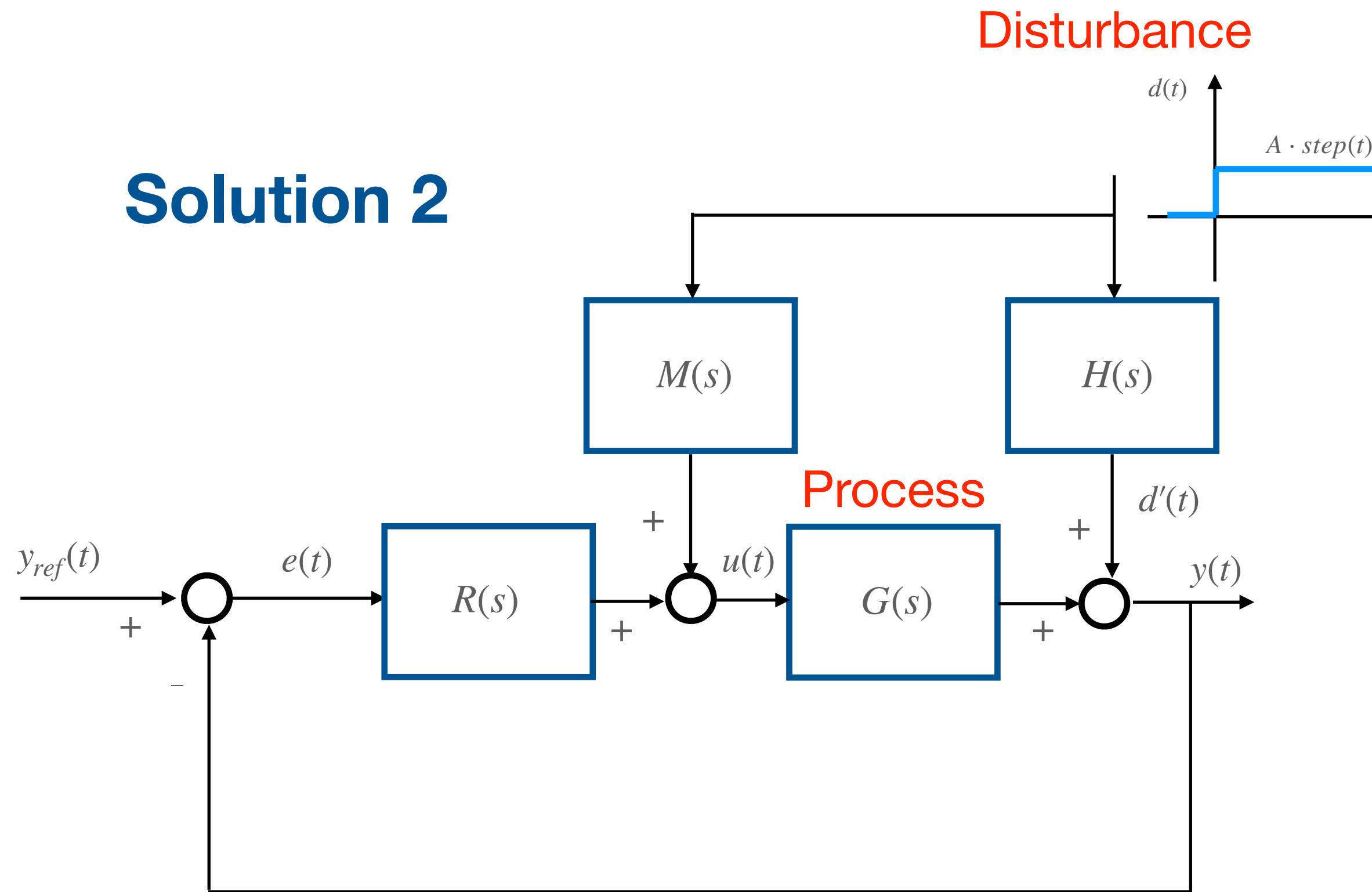
$$\xrightarrow{\quad M(s) \text{ s.t.} \quad} M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

Is this static  
compensation  
robust?

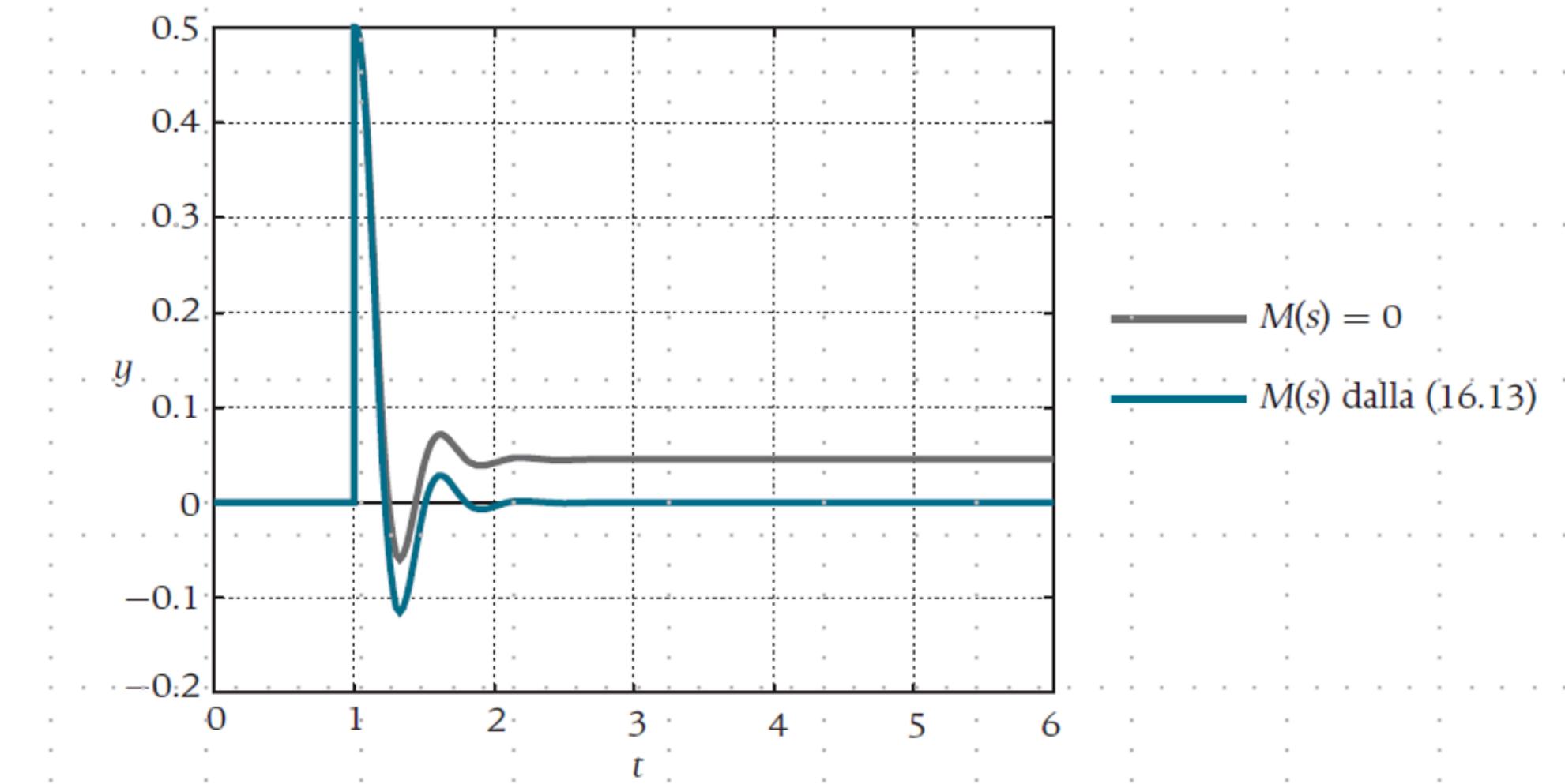


## Control Scheme with Measurable Disturbance Compensation

### Solution 2



Response to a step disturbance



Example 2:  
static case

$$\hat{\omega} = 0$$

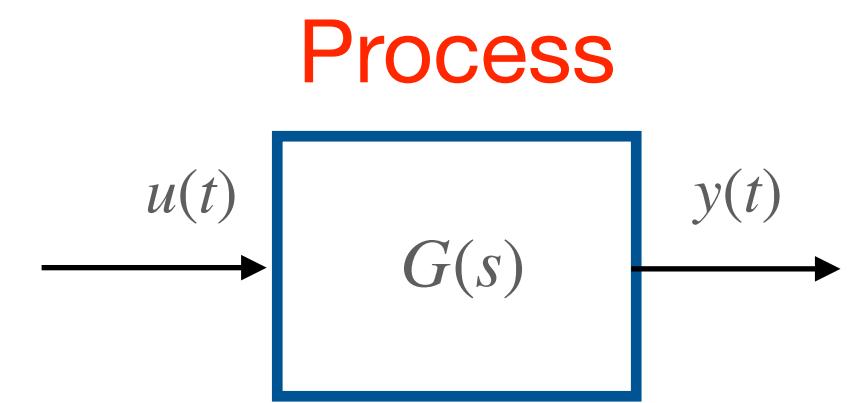
$$\xrightarrow{\quad M(s) \text{ s.t. } \quad} M(j\omega) \Big|_{\omega=0} = -\frac{H(0)}{G(0)}$$

Is this static  
compensation  
robust?

Is there a  
better  
alternative?



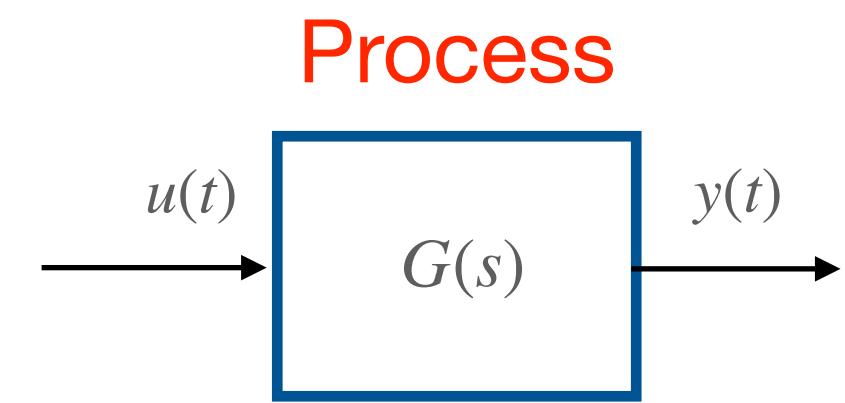
## Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$



## Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

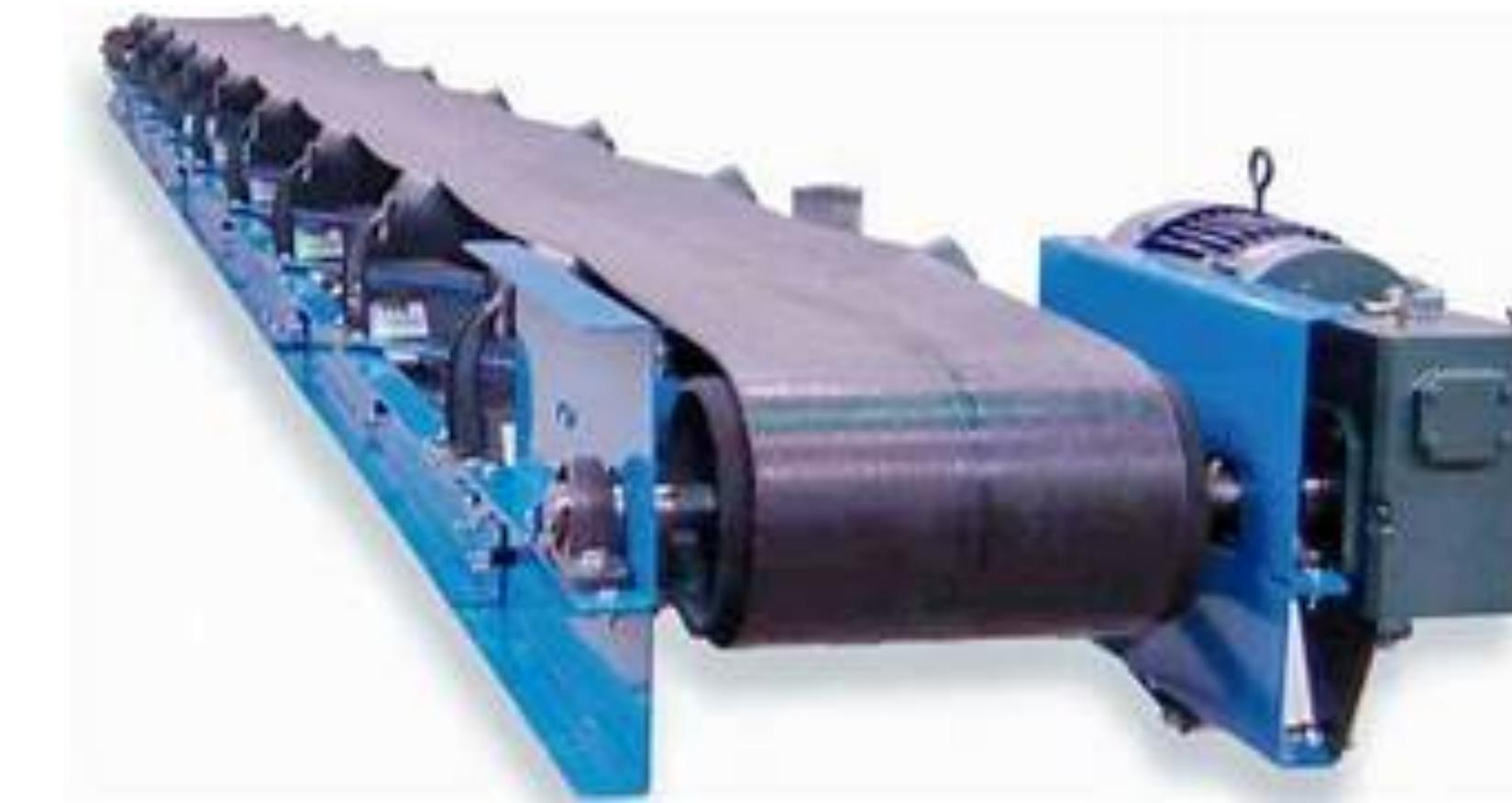
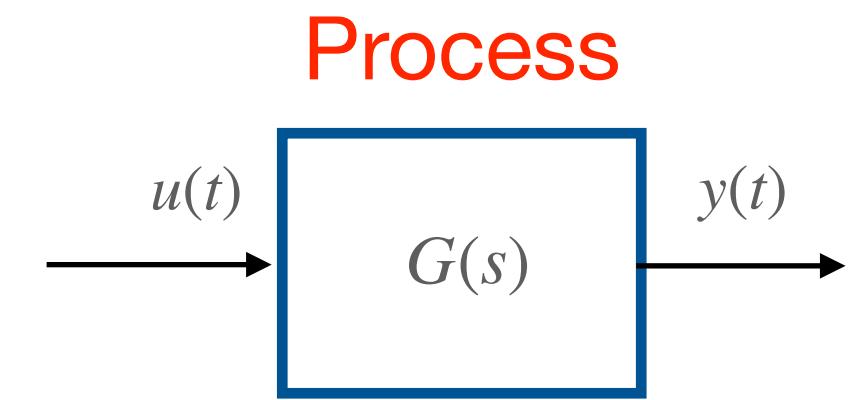


Image credit to: <https://www.gramconveyor.com/how-to-install-conveyor-belt/>

## Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

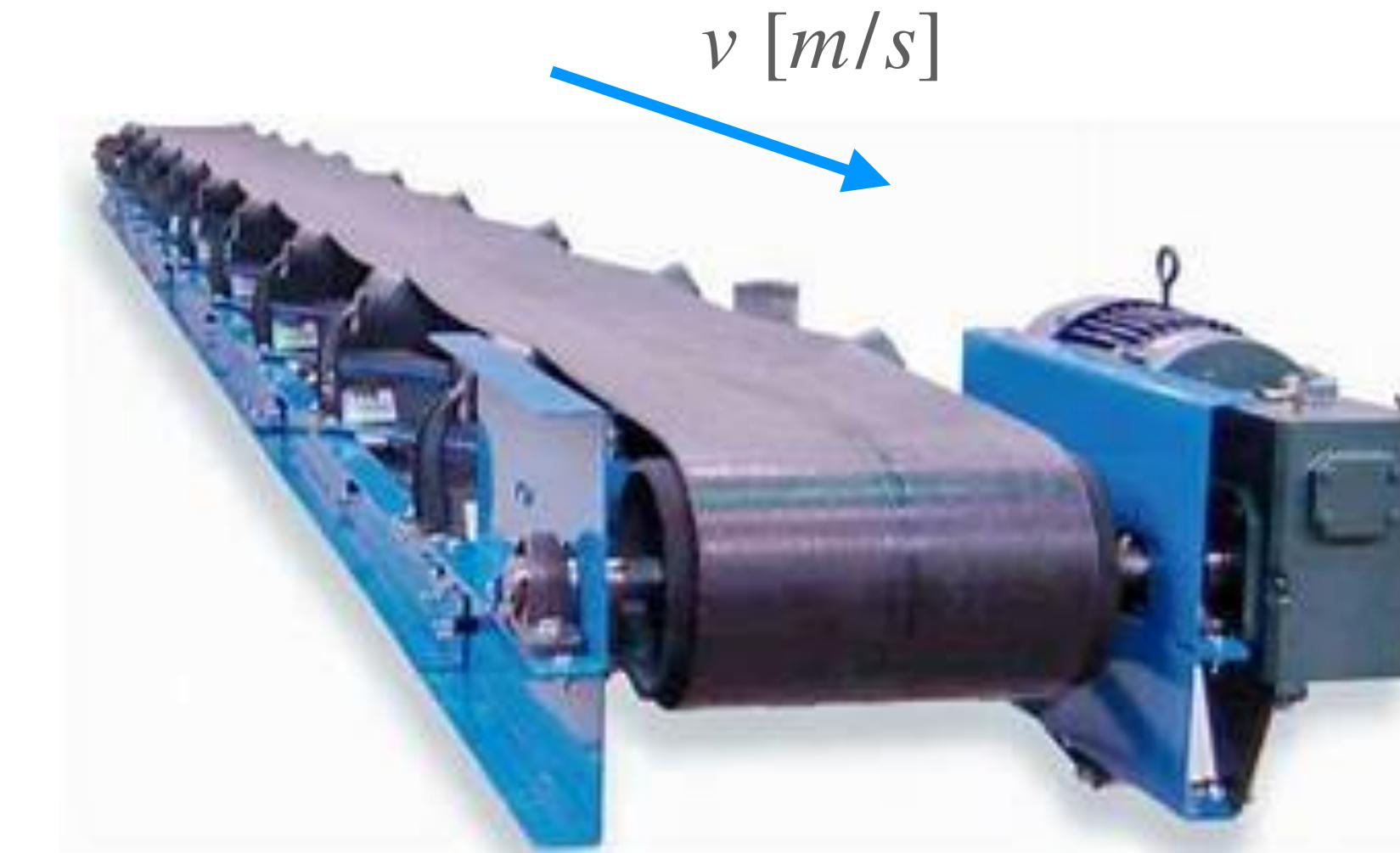
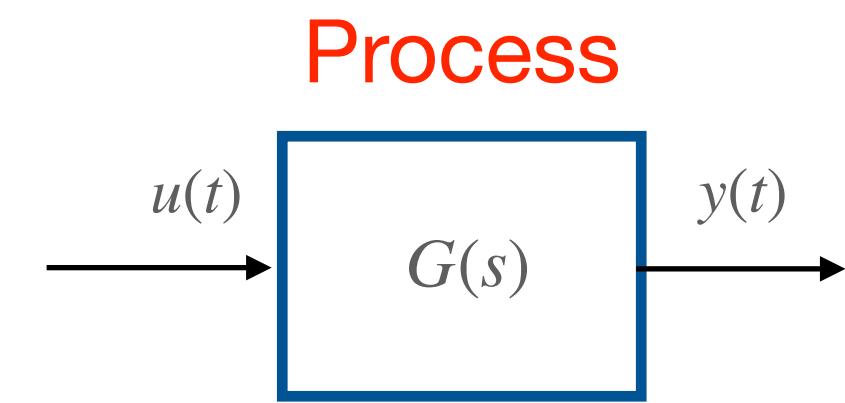


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## Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

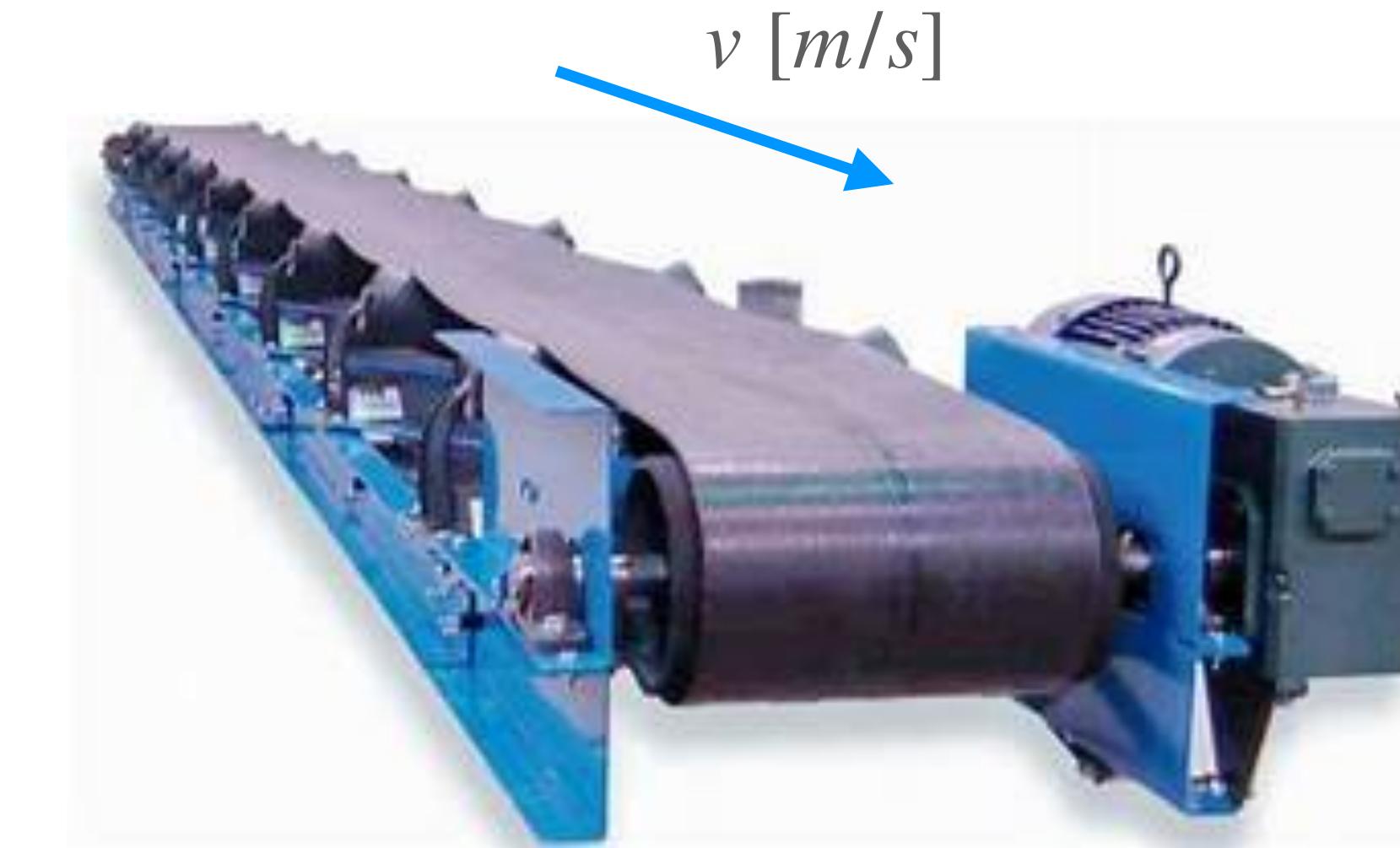
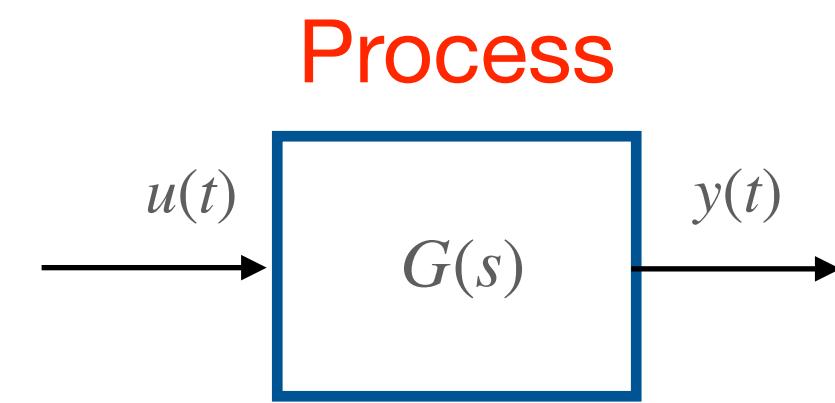


Image credit to: <https://www.gramconveyor.com/how-to-install-conveyor-belt/>

$$q_{OUT}(t) = q_{IN}(t - \tau) \quad \tau = \frac{L}{v}$$



## Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

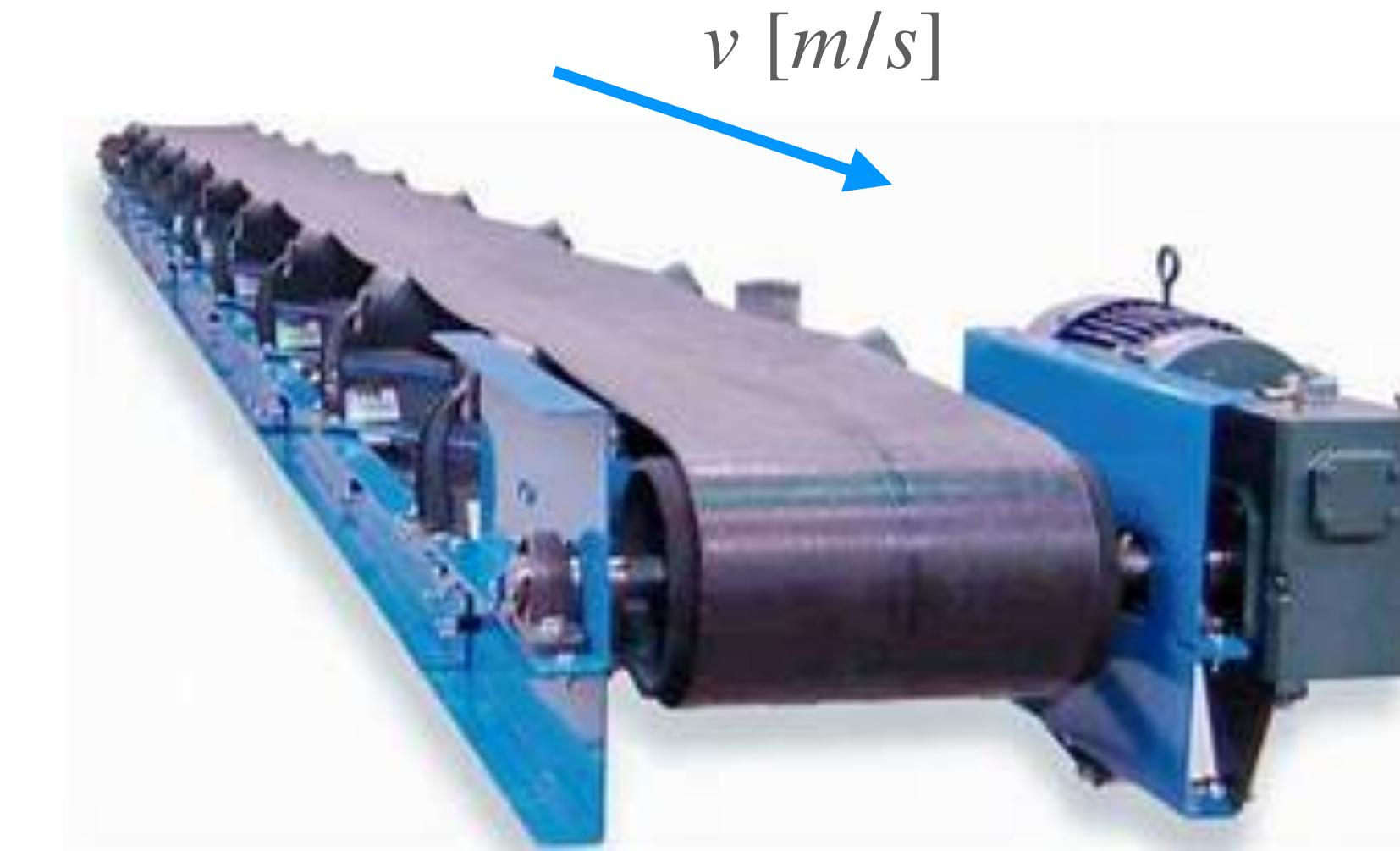
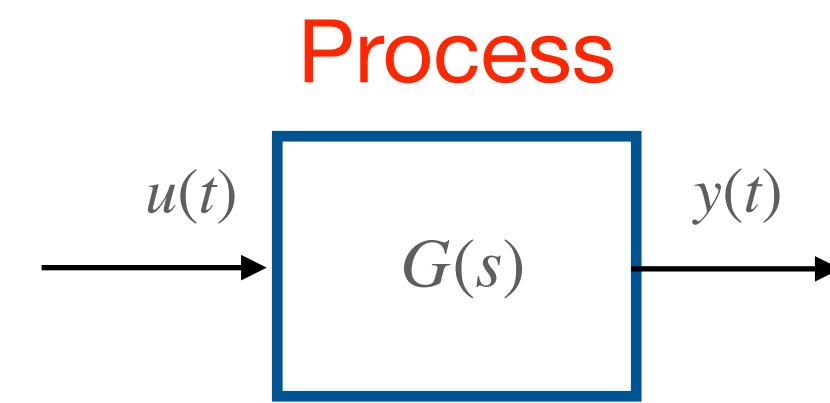


Image credit to: <https://www.gramconveyor.com/how-to-install-conveyor-belt/>

$$\mathcal{L} \leftarrow q_{OUT}(t) = q_{IN}(t - \tau) \quad \tau = \frac{L}{v}$$
$$Q_{OUT}(s) = e^{-\tau s} Q_{IN}(s)$$



## Control of LTI Systems with Delays



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

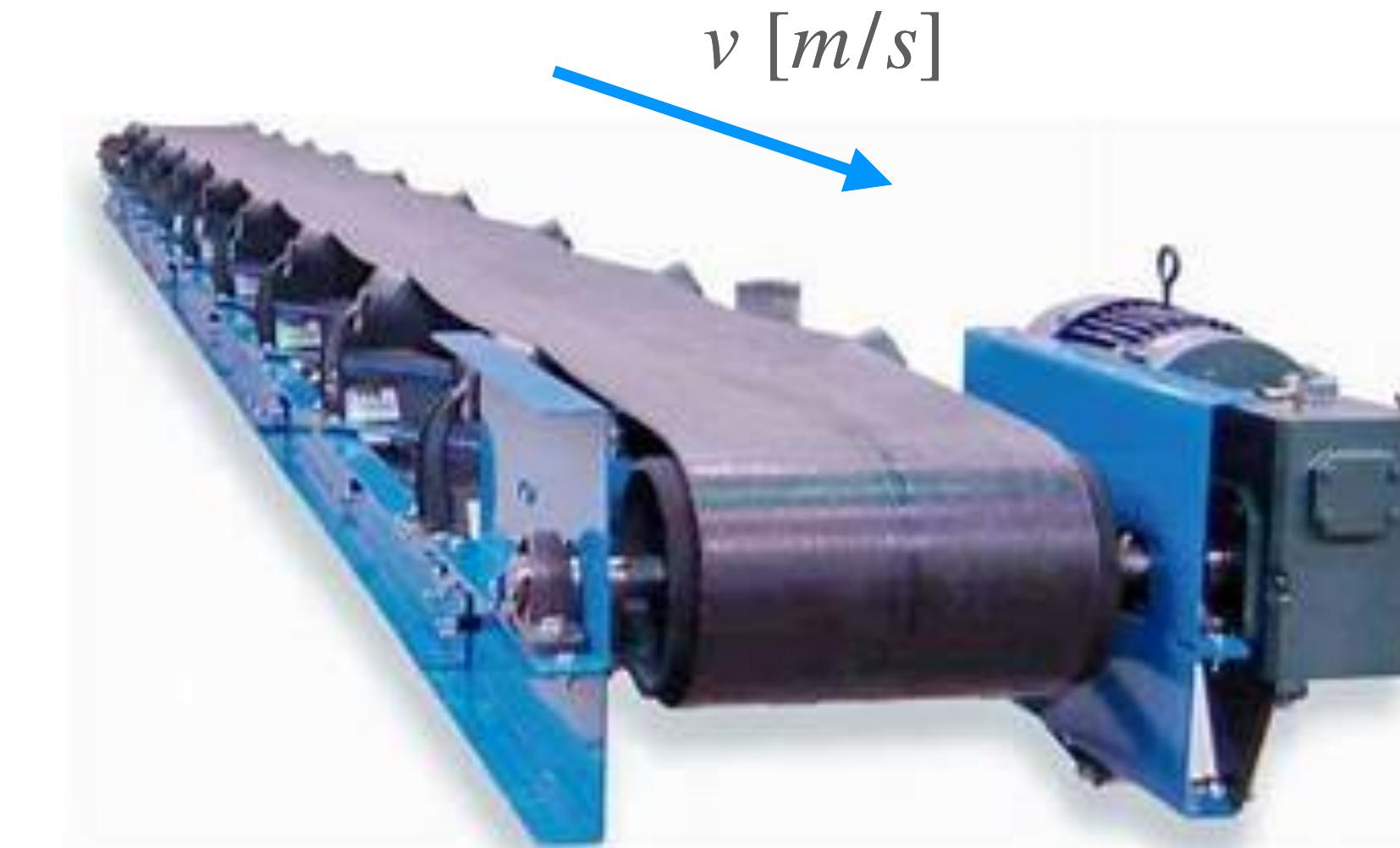
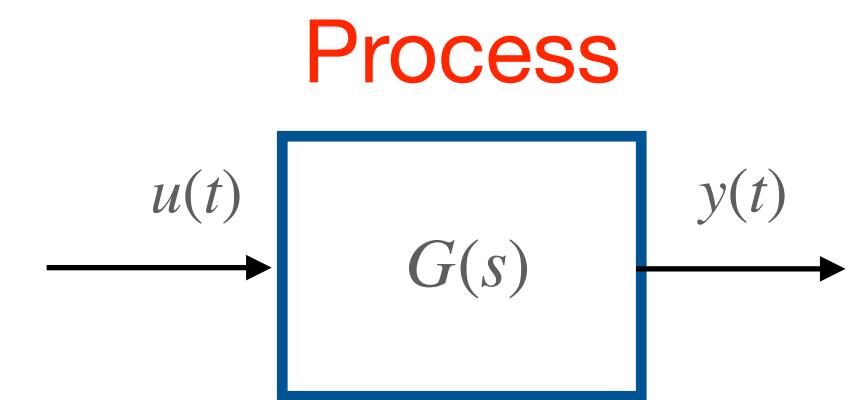


Image credit to: <https://www.gramconveyor.com/how-to-install-conveyor-belt/>

$$G(s) = e^{-\tau s} \quad \leftarrow \quad \mathcal{L} \quad q_{OUT}(t) = q_{IN}(t - \tau) \quad \tau = \frac{L}{v}$$
$$Q_{OUT}(s) = e^{-\tau s} Q_{IN}(s)$$

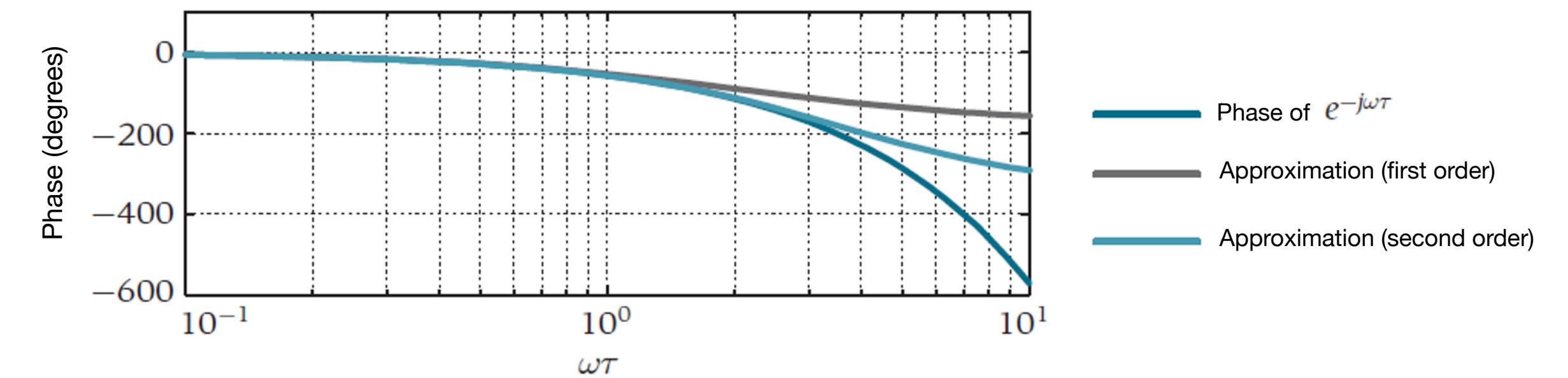


## Control of LTI Systems with Delays

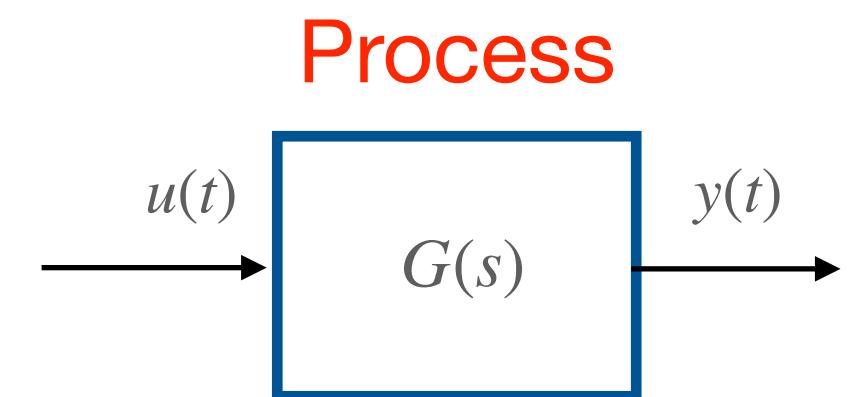


$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Phase Lag due to the delay term

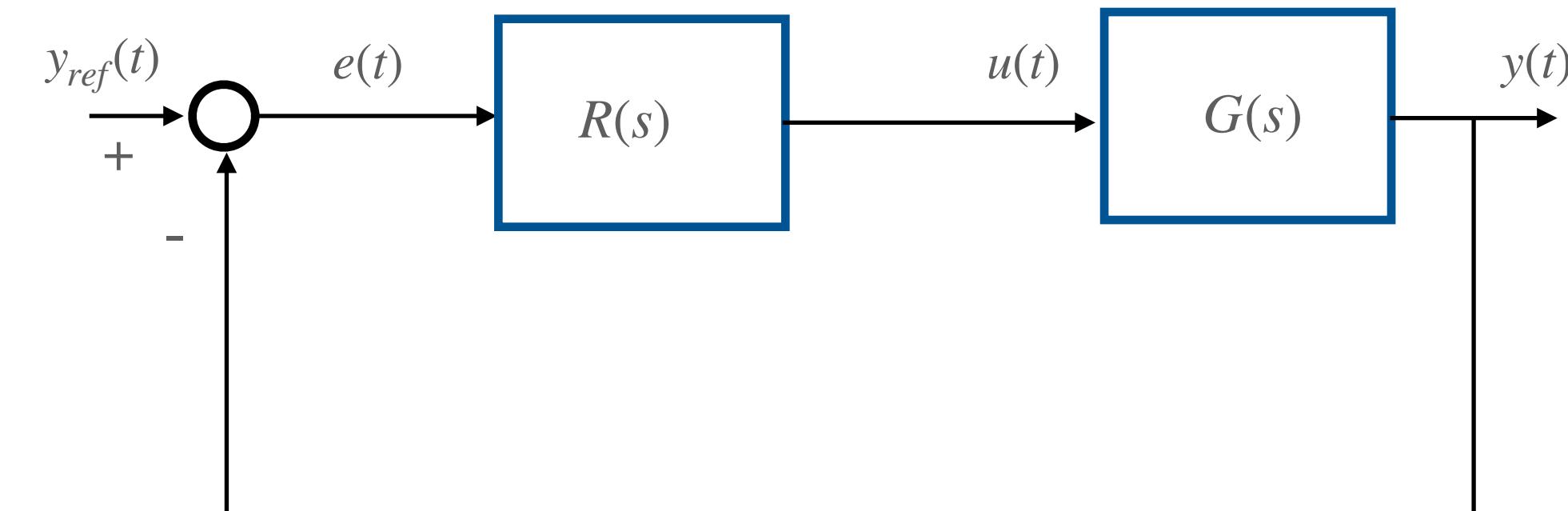
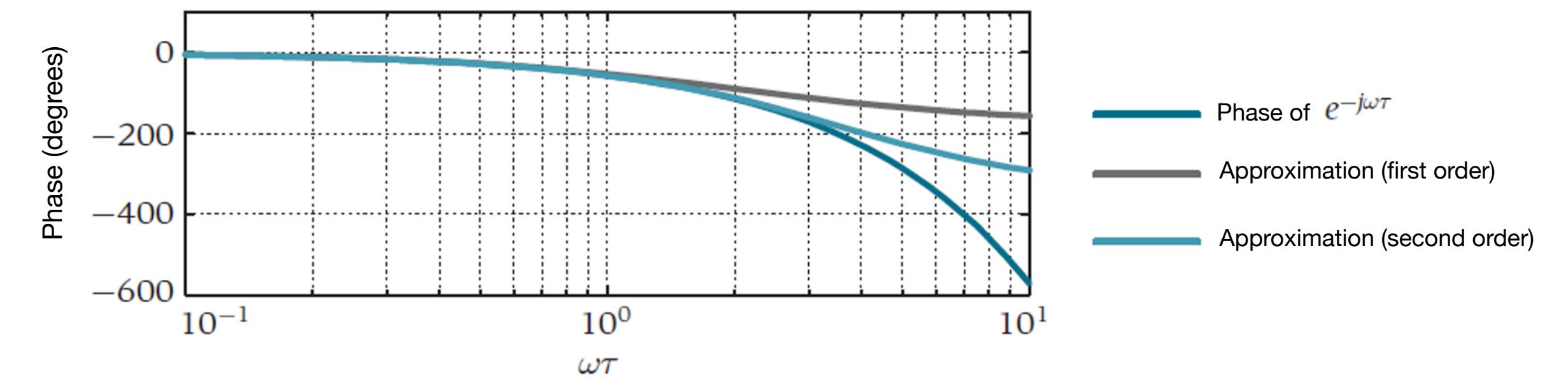


## Control of LTI Systems with Delays

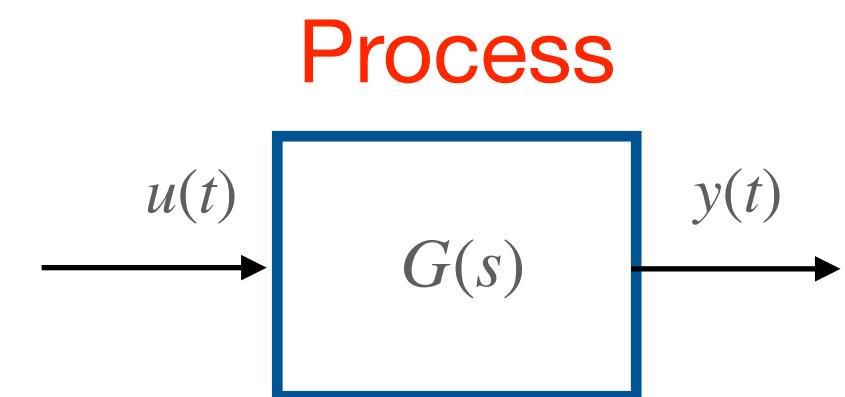


$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

**Phase Lag due to the delay term**



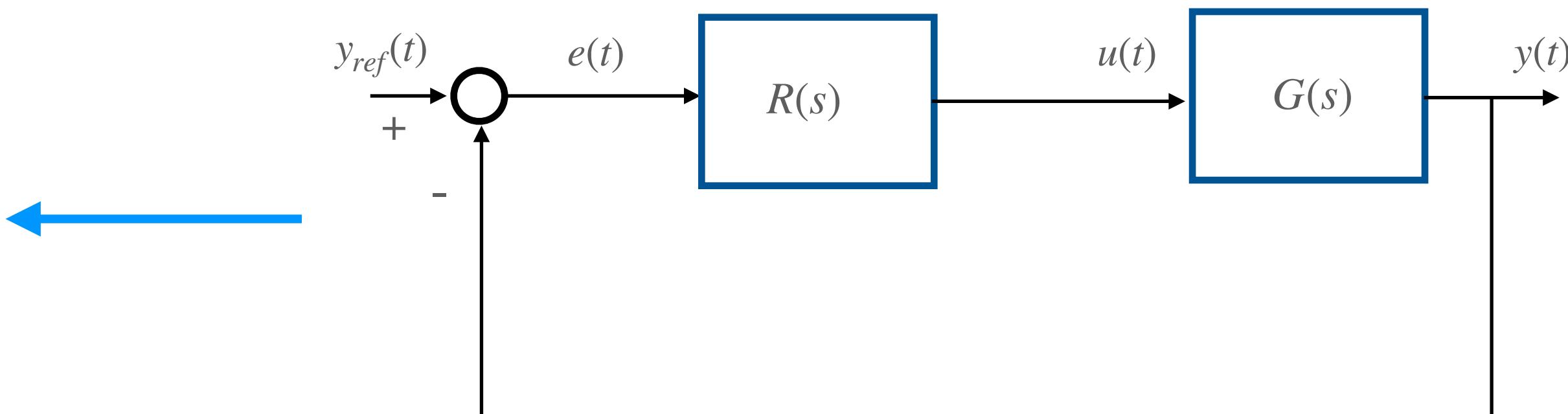
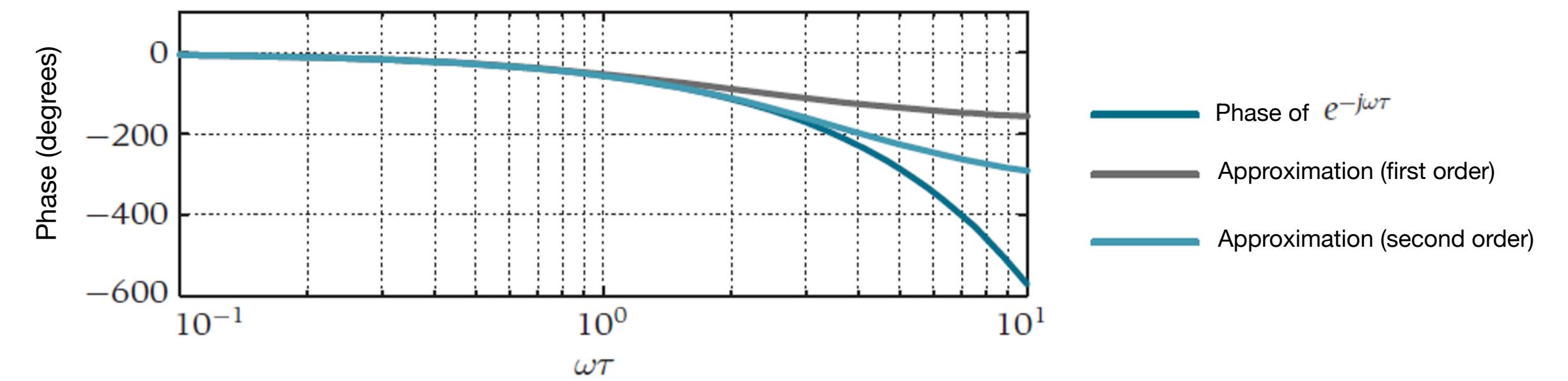
## Control of LTI Systems with Delays



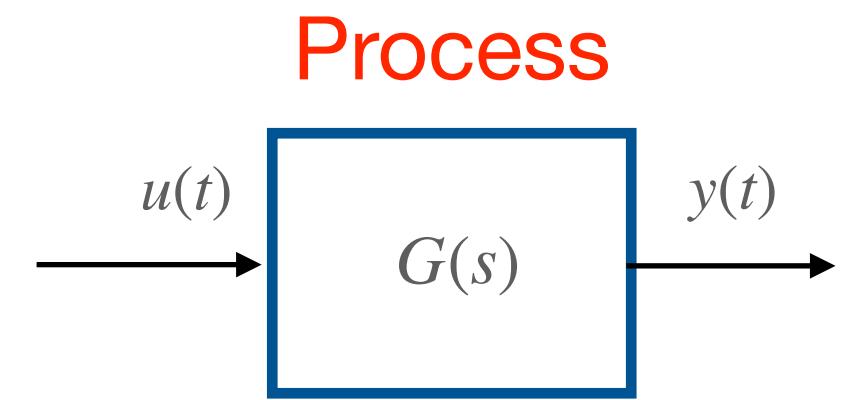
$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Recommendation:  $\omega_{BW_{CL}}$  sufficiently narrow

**Phase Lag due to the delay term**



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme



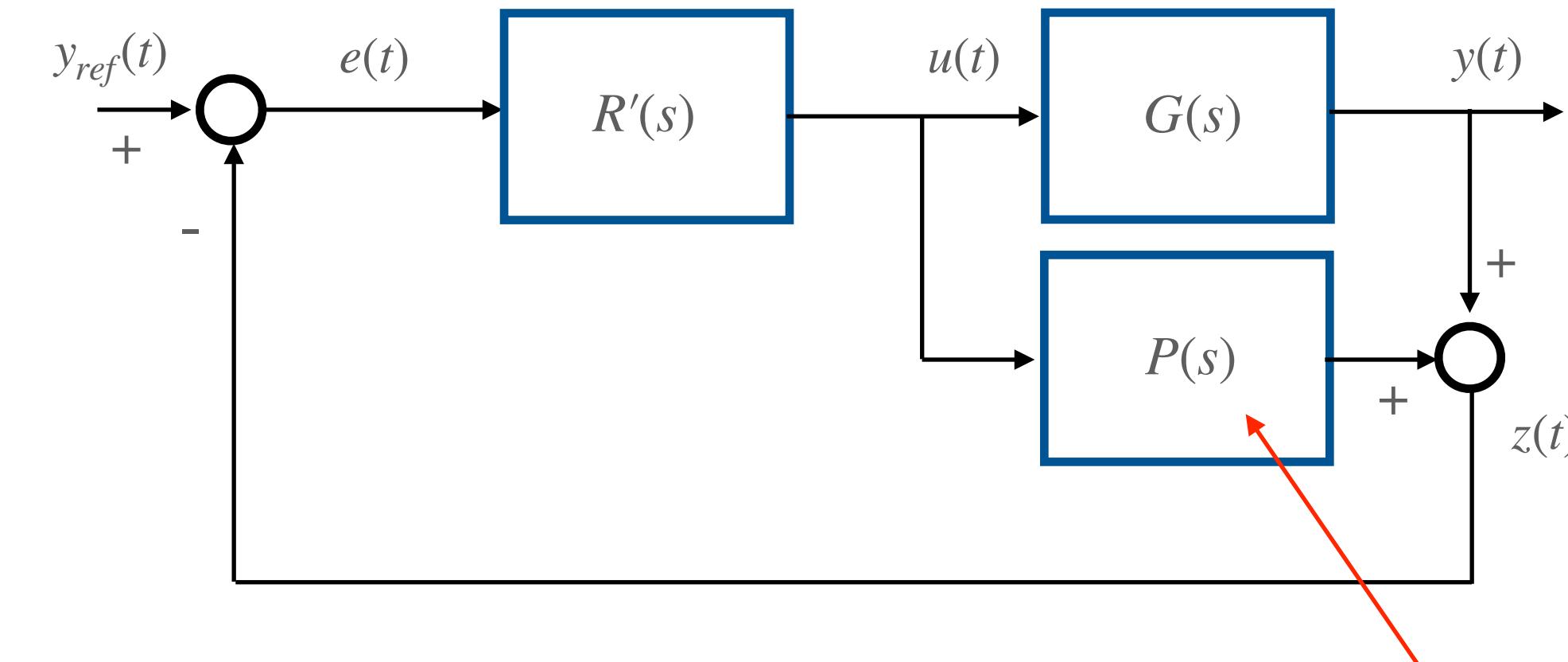
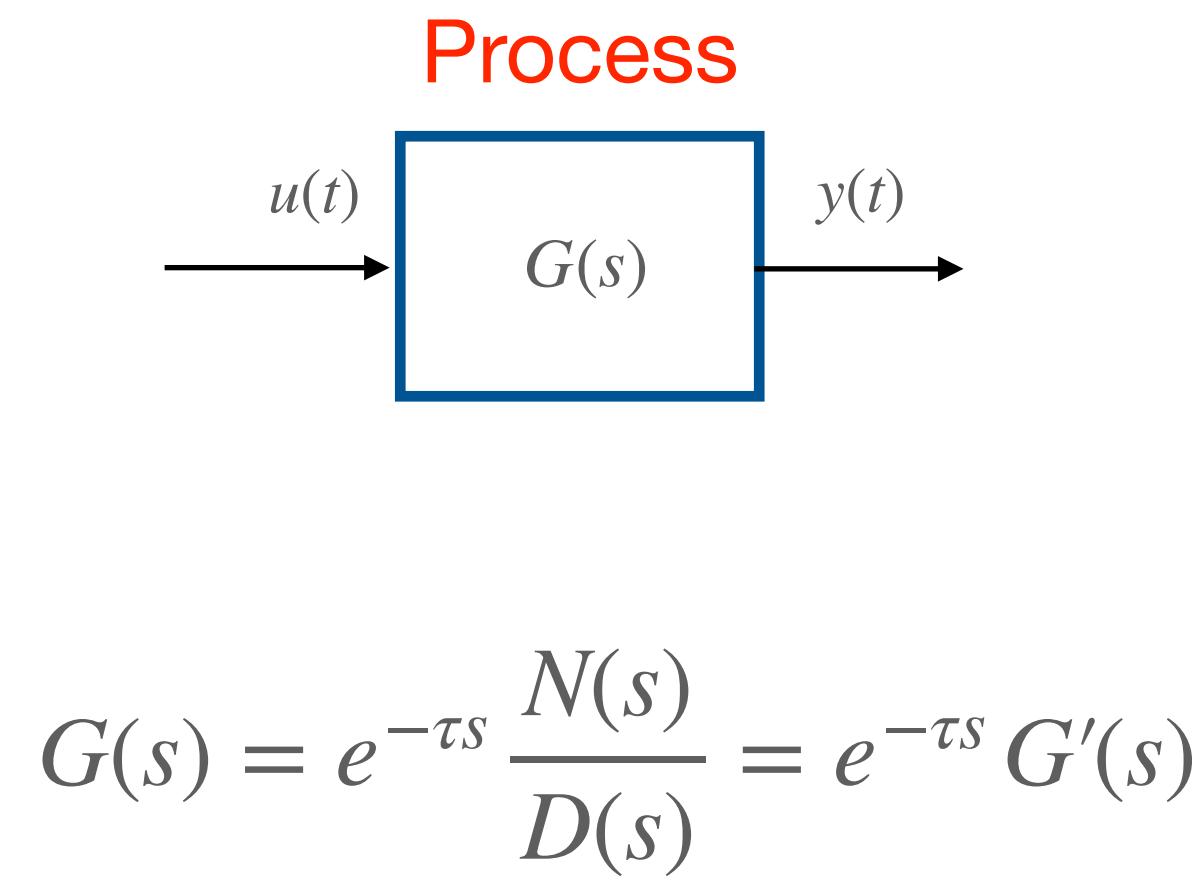
$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Assumptions:

- $G'(s)$  As. Stable
- $\tau$  known



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme



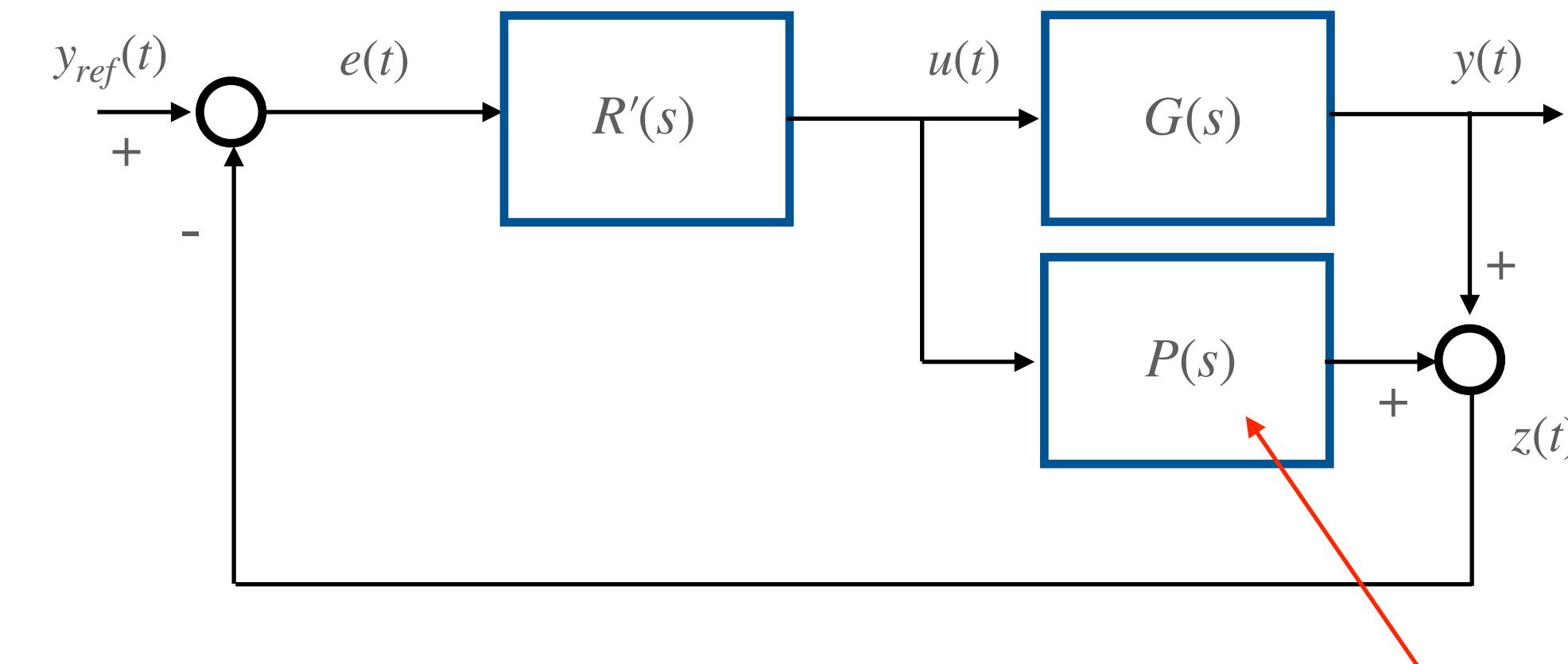
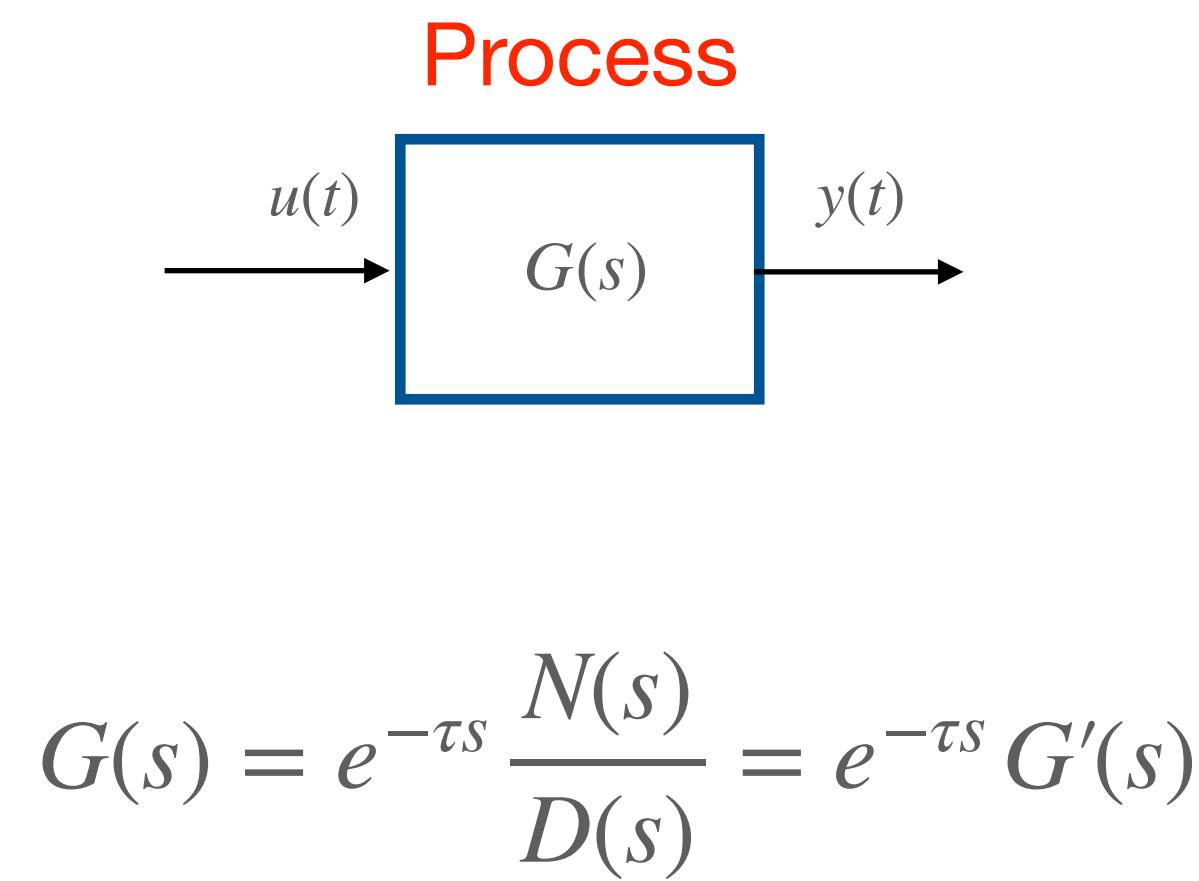
Smith Predictor

Assumptions:

- $G'(s)$  As. Stable
- $\tau$  known



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme



Smith Predictor

Assumptions:

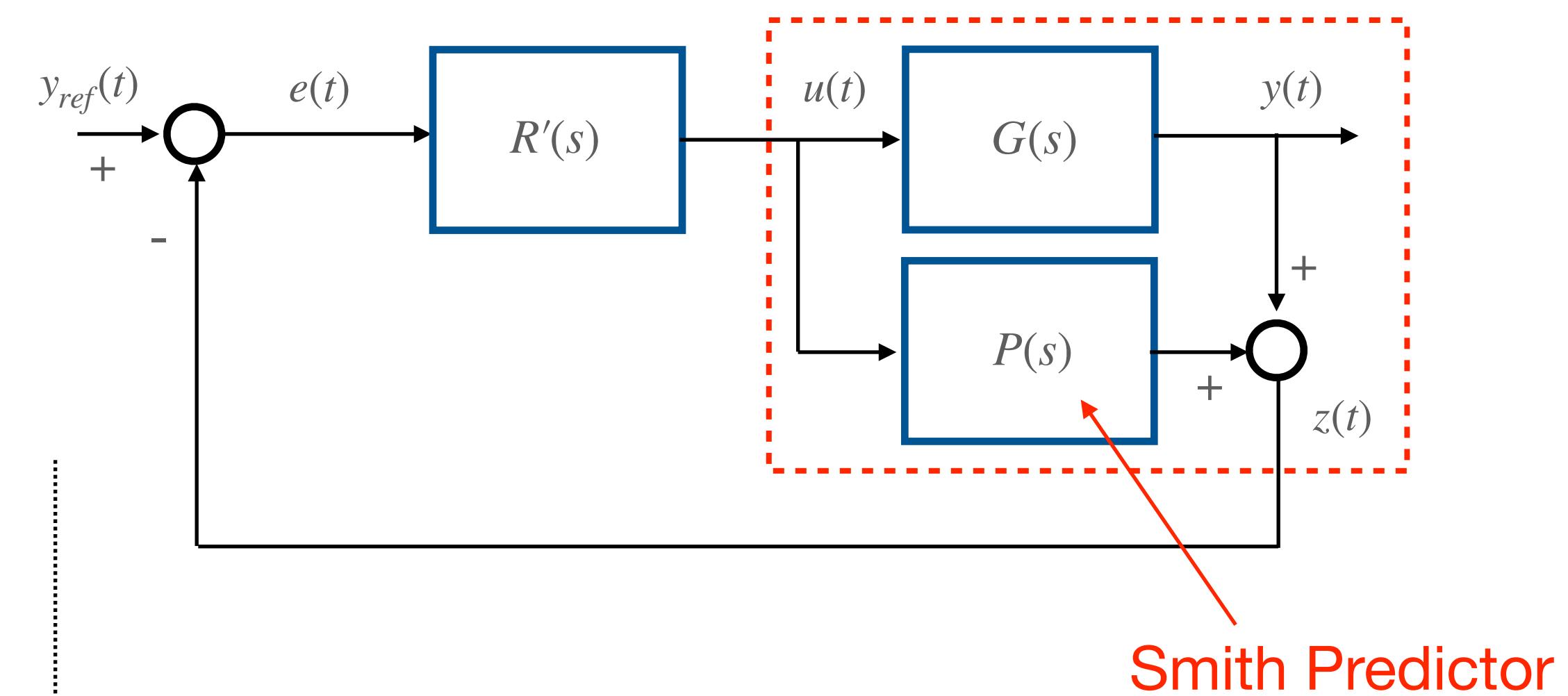
- $G'(s)$  As. Stable
- $\tau$  known

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$$\begin{aligned}
 u \rightarrow z : \quad G(s) + P(s) &= e^{-\tau s} G'(s) + (1 - e^{-\tau s}) G'(s) \\
 &= e^{-\tau s} G'(s) + G'(s) - e^{-\tau s} G'(s) \\
 &= G'(s)
 \end{aligned}$$

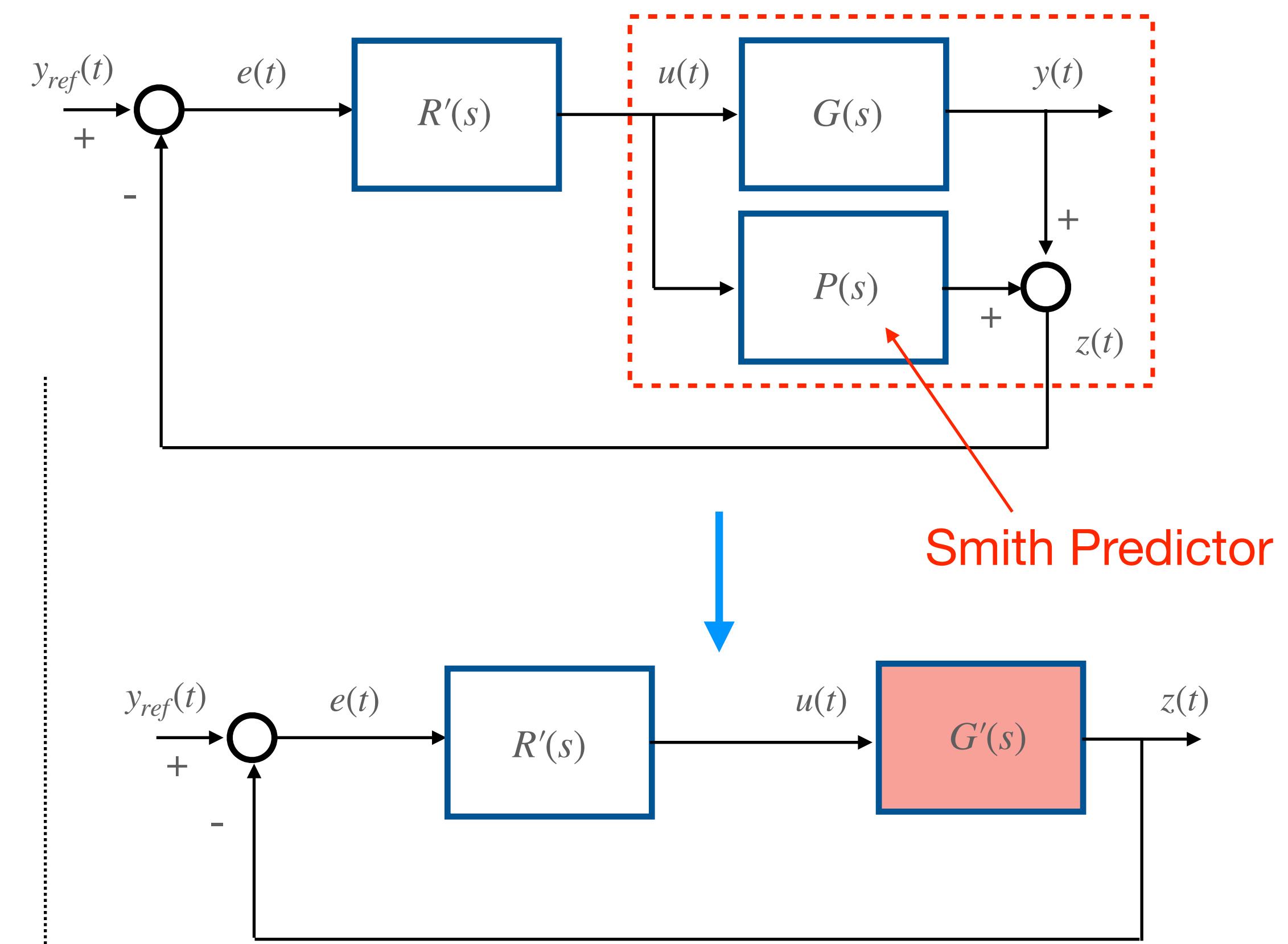


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

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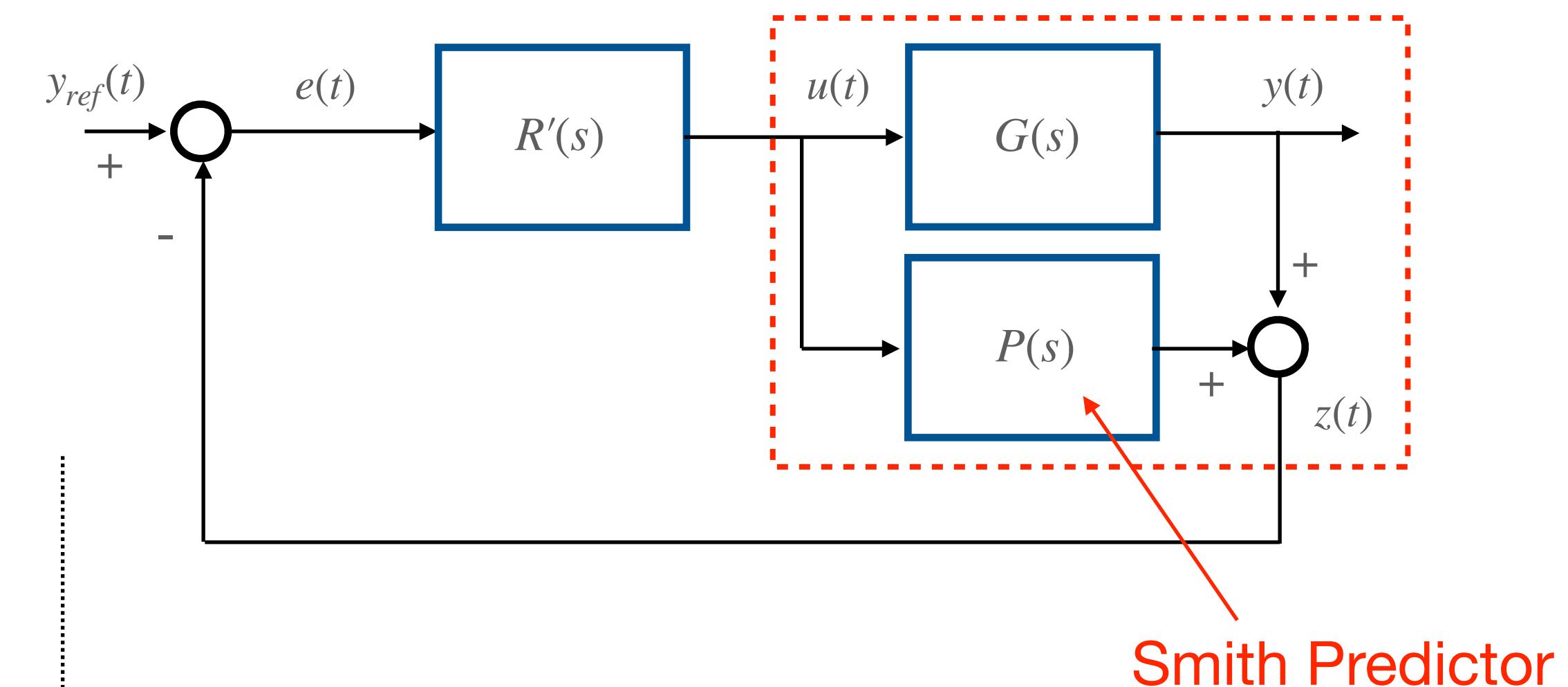
## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$$\begin{aligned}
 u \rightarrow z : \quad G(s) + P(s) &= e^{-\tau s} G'(s) + (1 - e^{-\tau s}) G'(s) \\
 &= e^{-\tau s} G'(s) + G'(s) - e^{-\tau s} G'(s) \\
 &= G'(s)
 \end{aligned}$$



**Warning:**

$$z(t) \neq y(t)$$

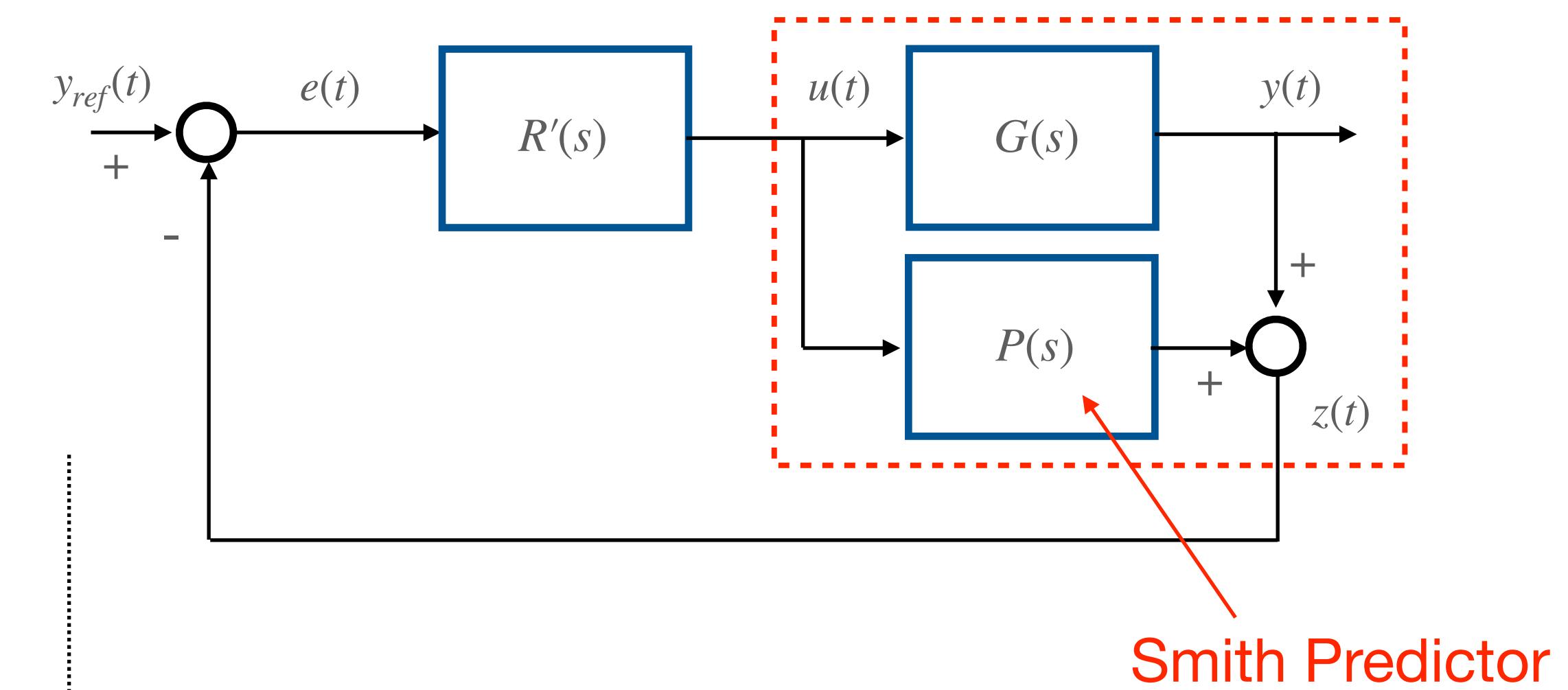


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$z(t) \neq y(t)$  →  $z(t) ?$



$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$z(t) \neq y(t)$

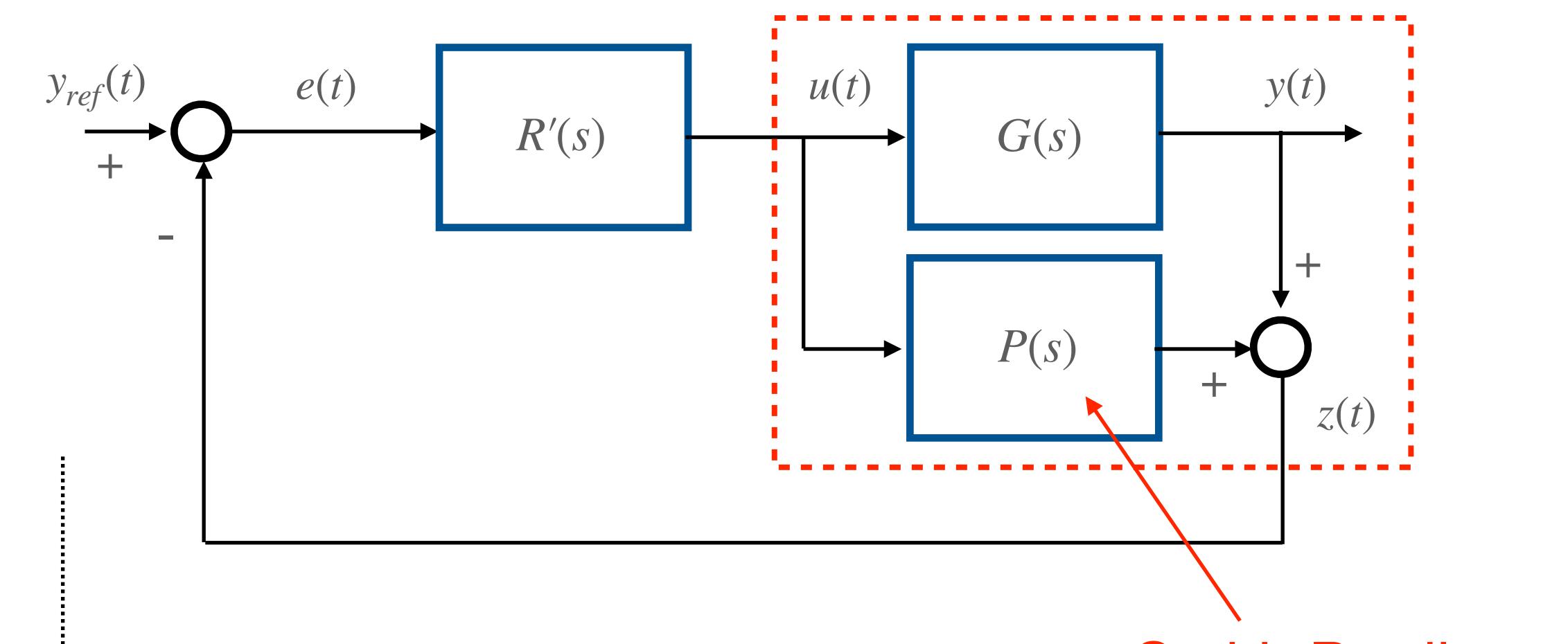


$z(t) ?$

In open loop:

$$u \rightarrow z : G'(s)$$

$$u \rightarrow y : e^{-\tau s} G'(s)$$



$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$z(t) \neq y(t)$



$z(t) ?$

In open loop:

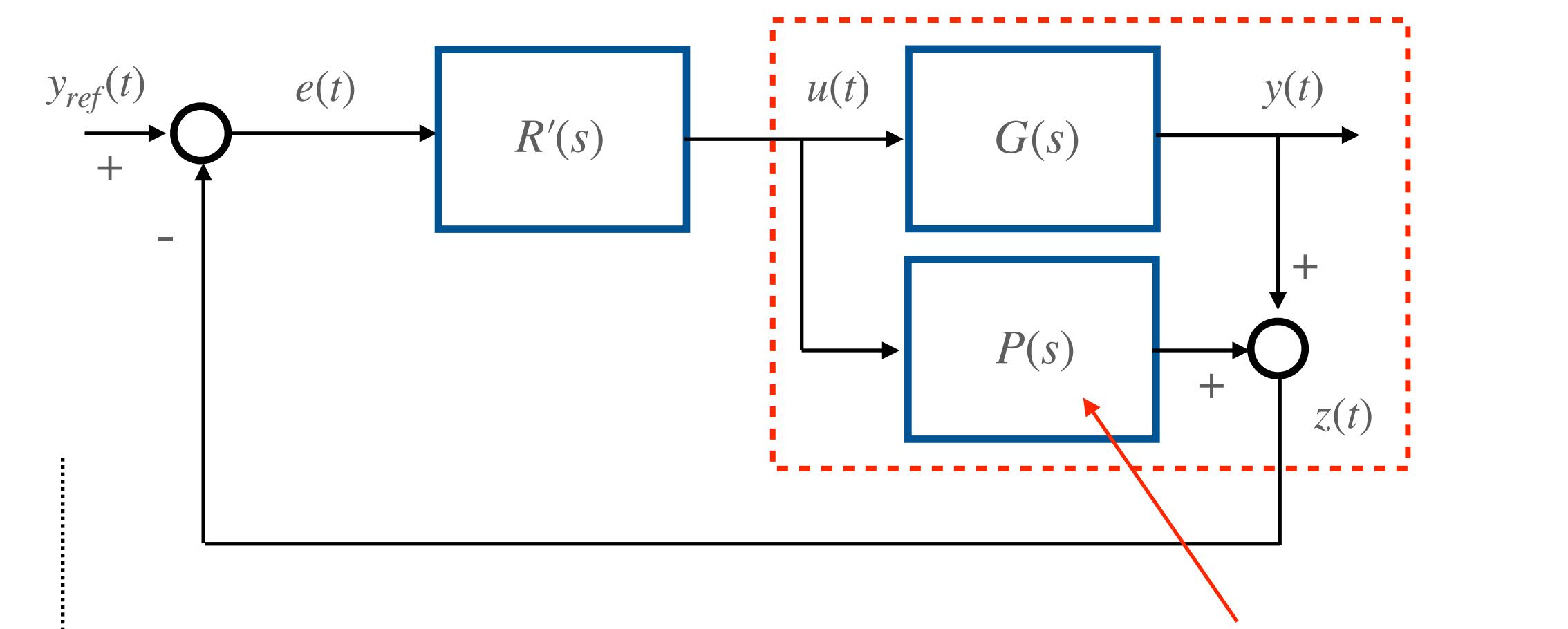
$$u \rightarrow z : G'(s)$$

$$u \rightarrow y : e^{-\tau s} G'(s)$$

$\mathcal{L}$

$$Z(s) = G'(s) U(s)$$

$$Y(s) = e^{-\tau s} G'(s) U(s)$$



$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

$$z(t) \neq y(t)$$



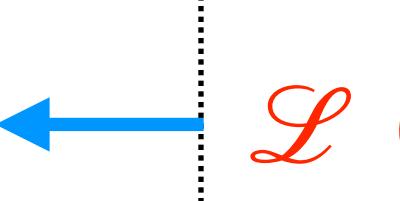
$z(t)$  ?

In open loop:

$$u \rightarrow z : G'(s)$$

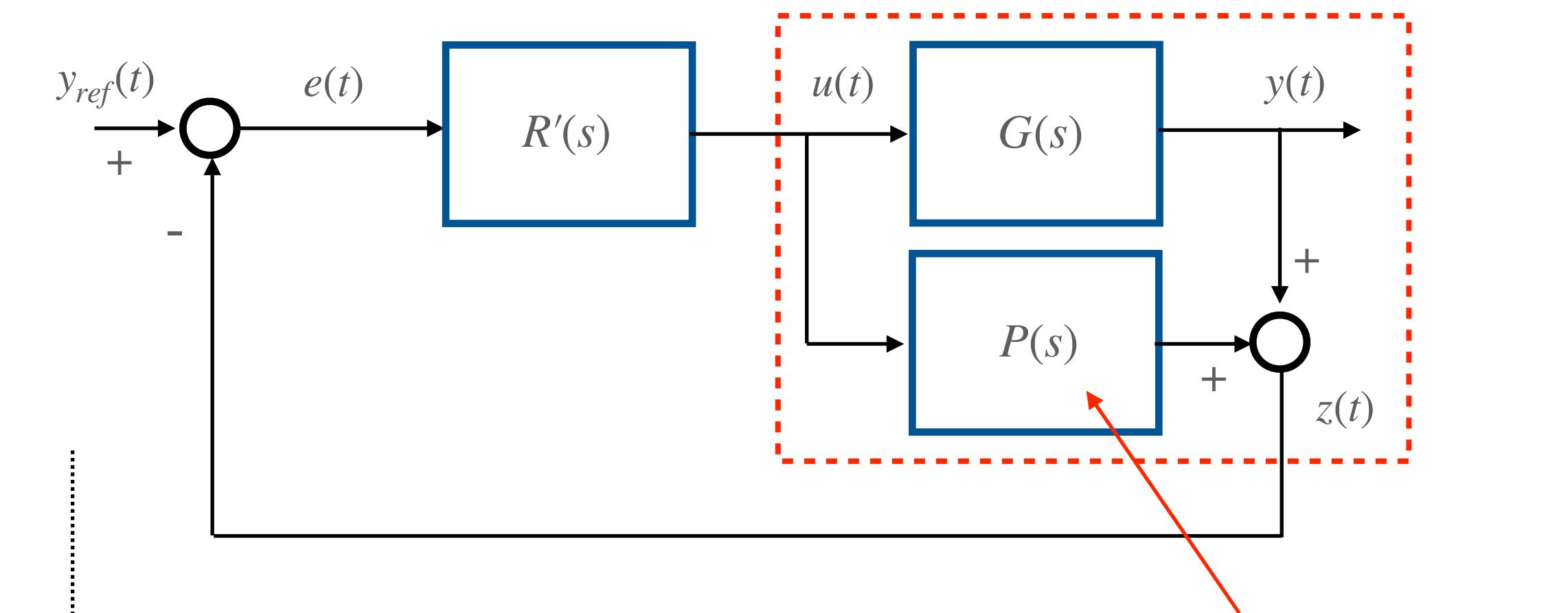
$$u \rightarrow y : e^{-\tau s} G'(s)$$

$$Z(s) = e^{\tau s} Y(s)$$



$$Z(s) = G'(s) U(s)$$

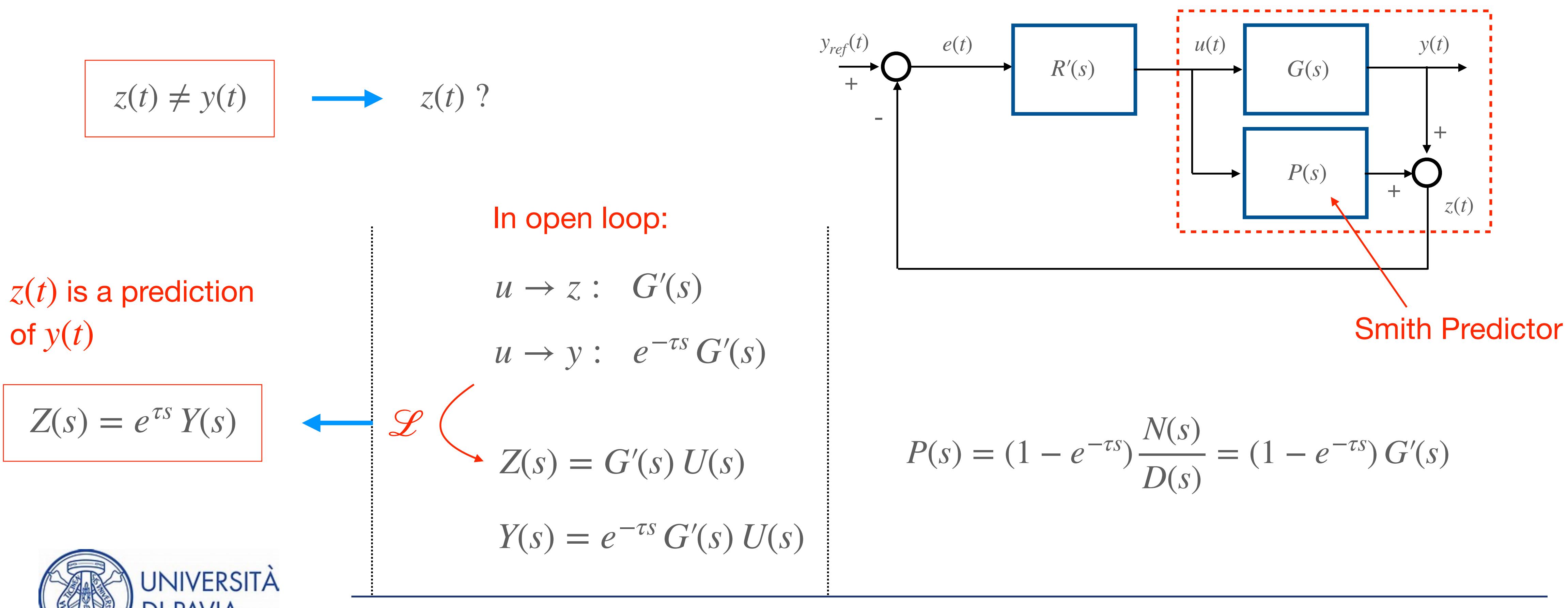
$$Y(s) = e^{-\tau s} G'(s) U(s)$$



$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

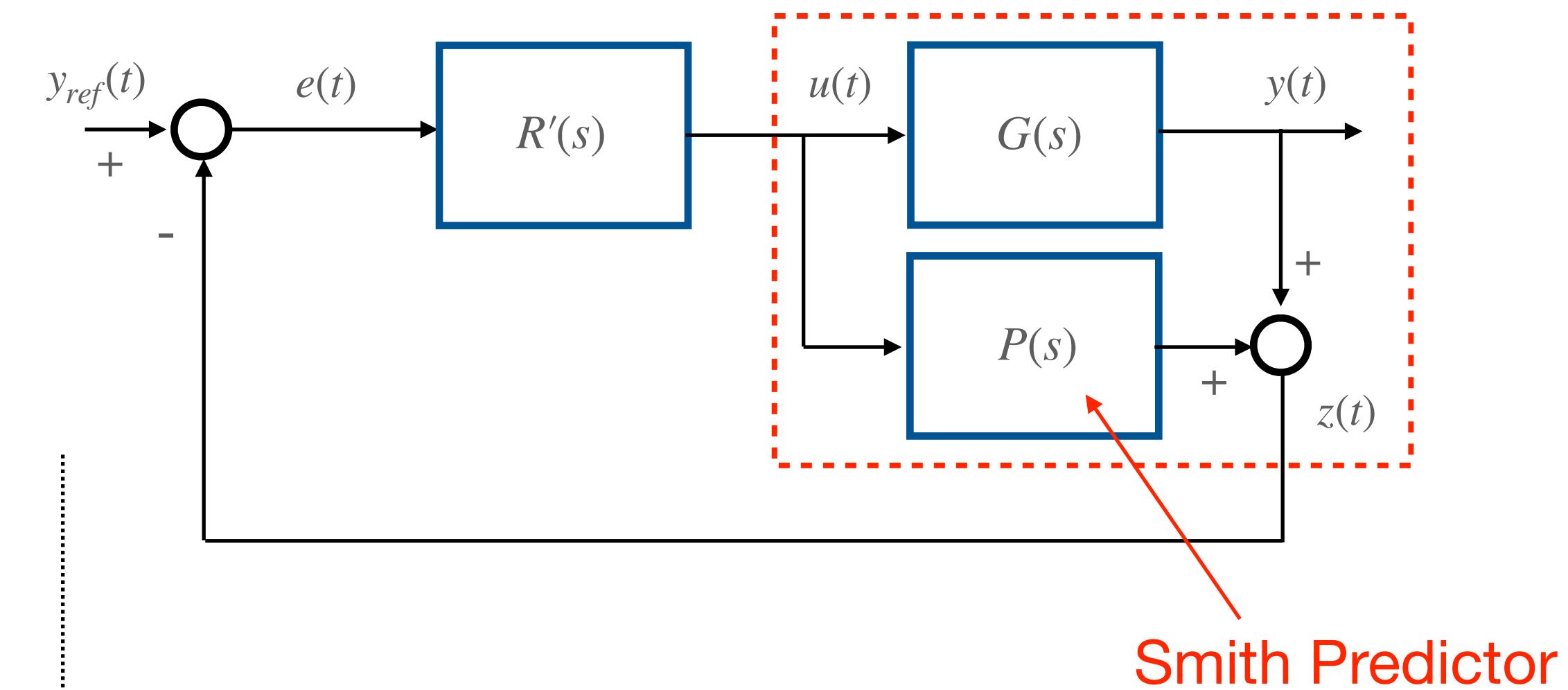


## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 1:

In the static case

$$P(0) = (1 - e^{-\tau_0}) G'(0) = 0$$



$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

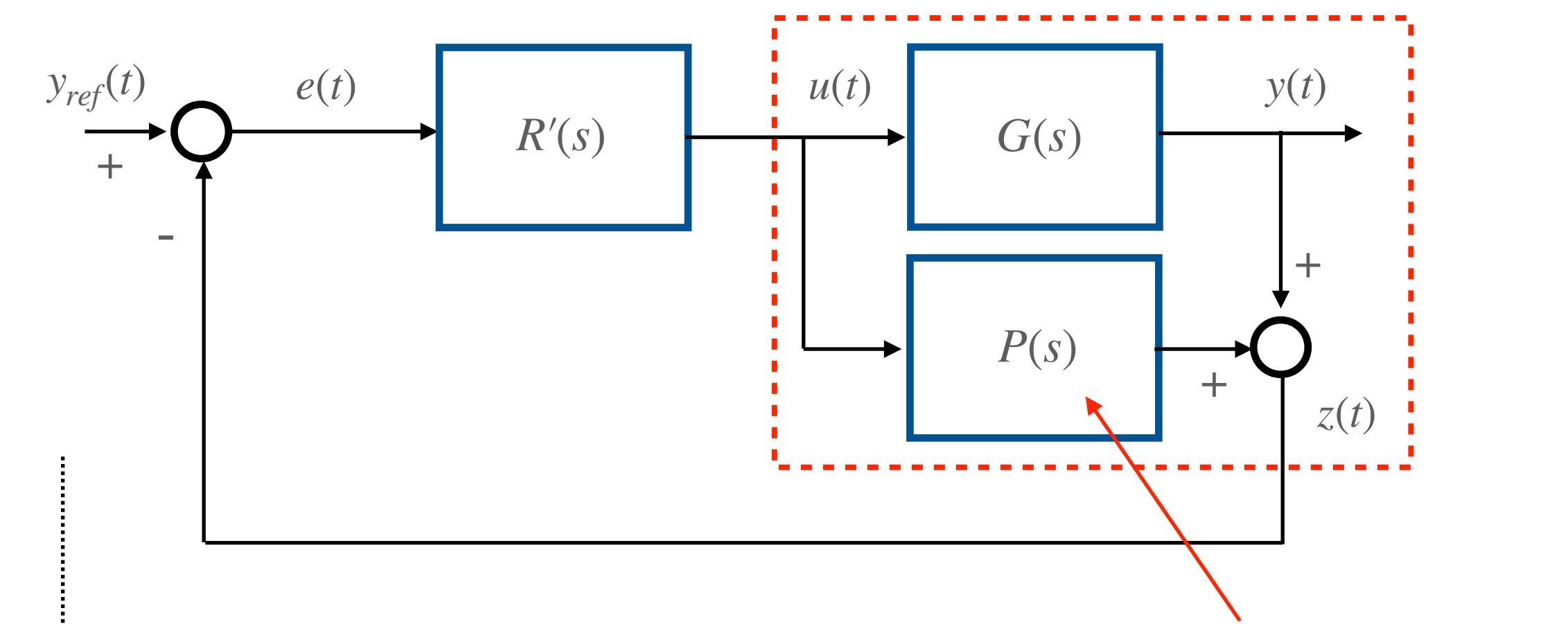
Observation 1:

In the static case

$$P(0) = (1 - e^{-\tau_0}) G'(0) = 0$$

Then, if the closed loop system is As. Stable  
and  $y_{ref}(t) = A \text{ step}(t)$ :

$$z(t) \rightarrow y(t) \quad \text{for} \quad t \rightarrow \infty$$

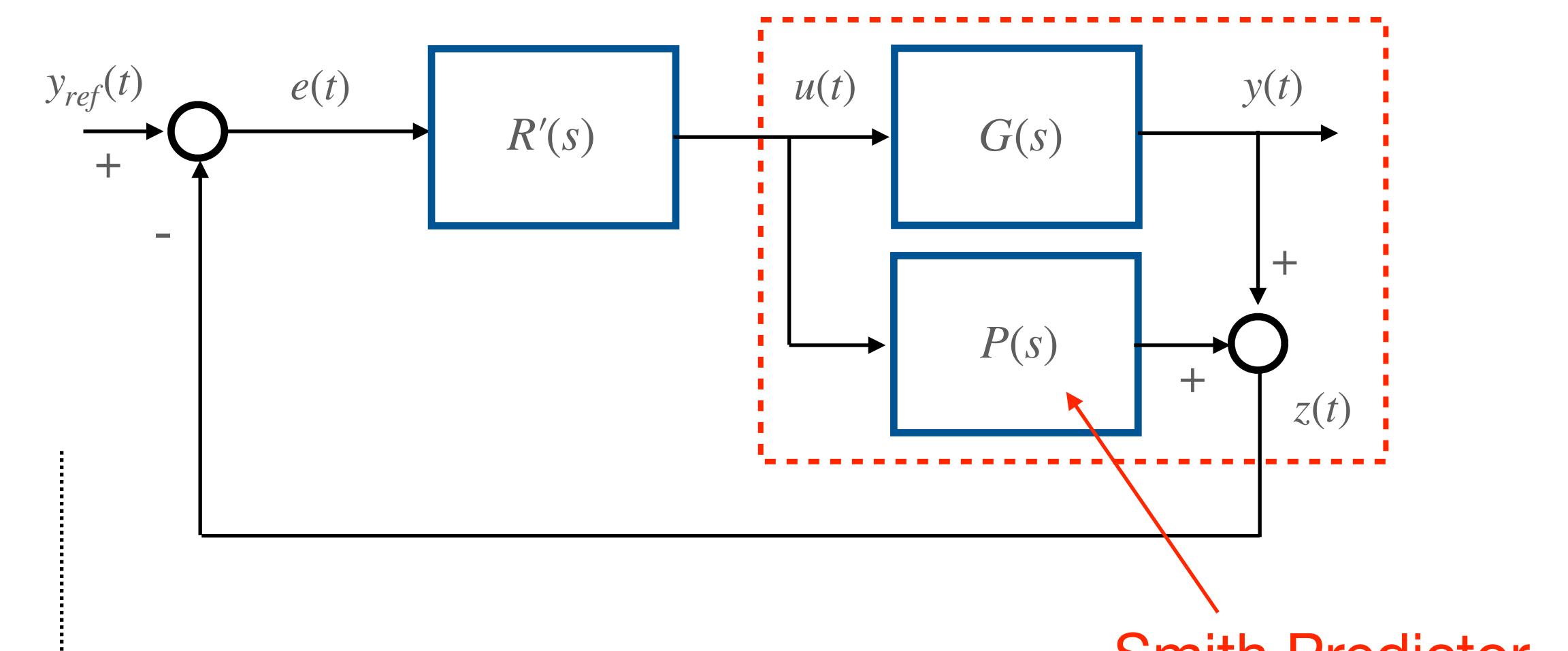
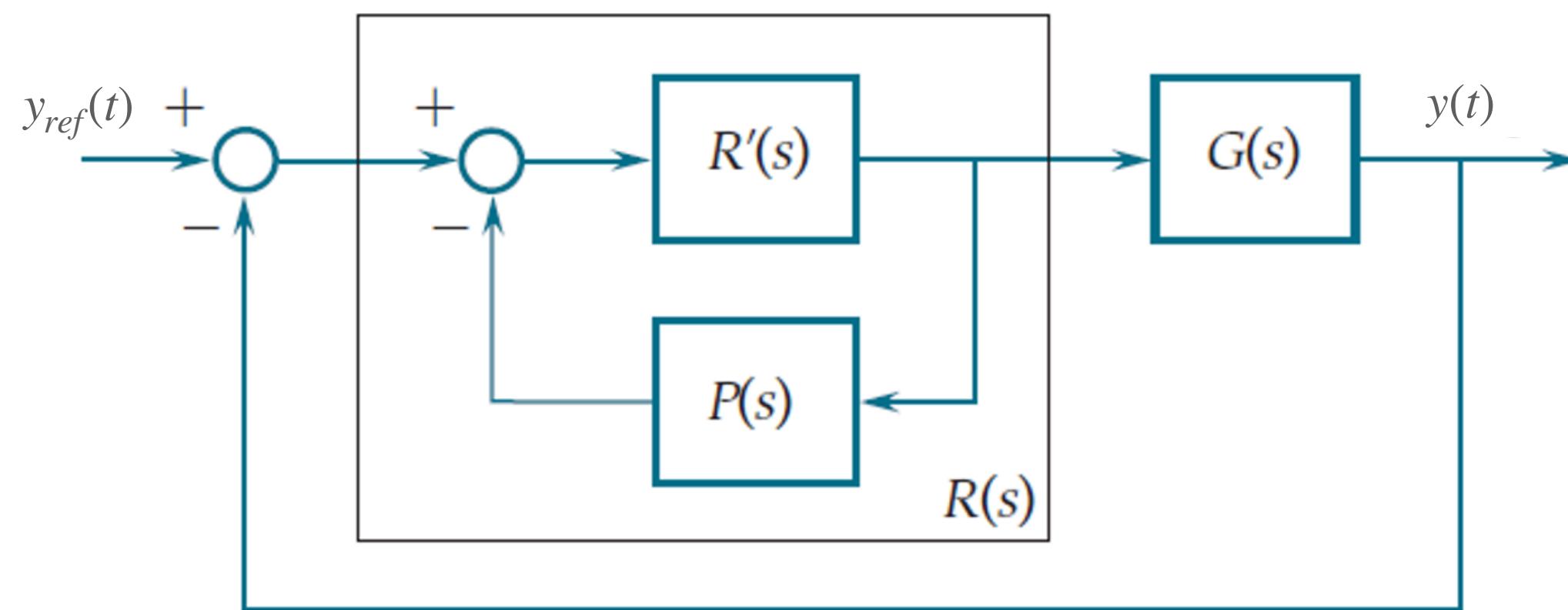


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation

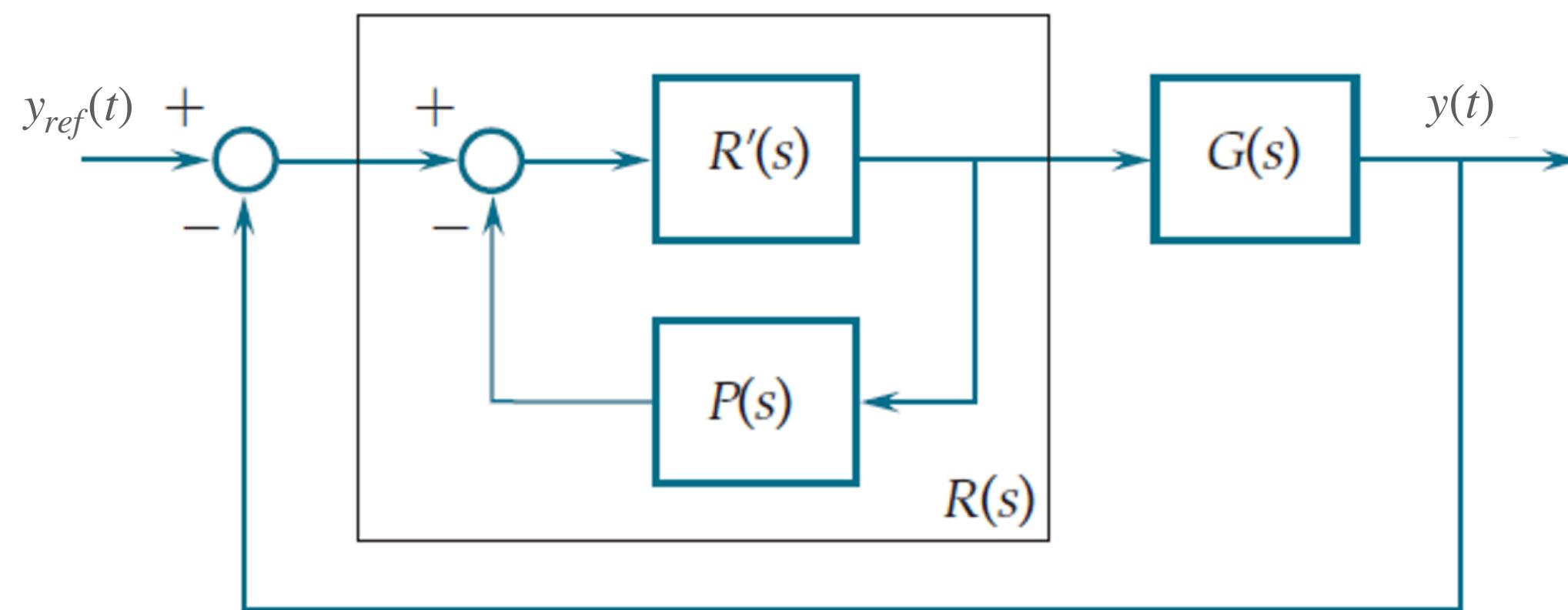


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

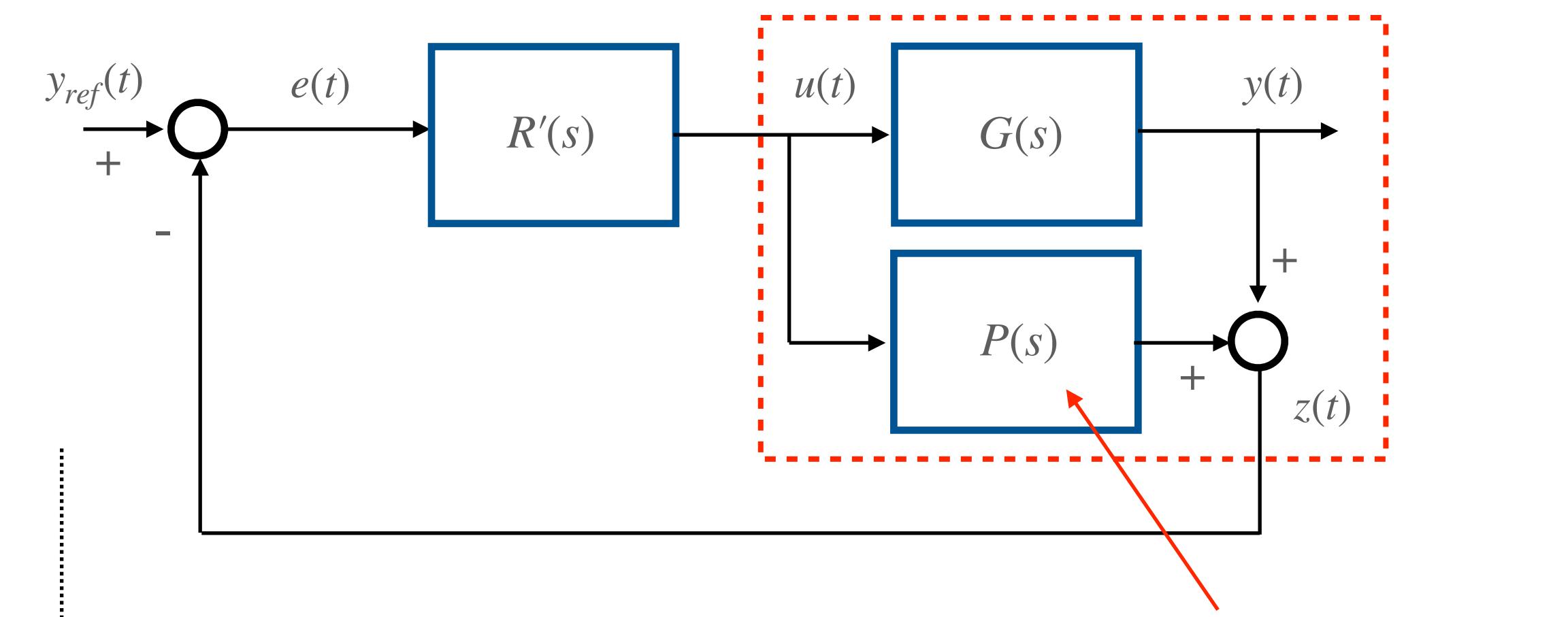


## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation



$$R(s) = \frac{R'(s)}{1 + R'(s)P(s)}$$

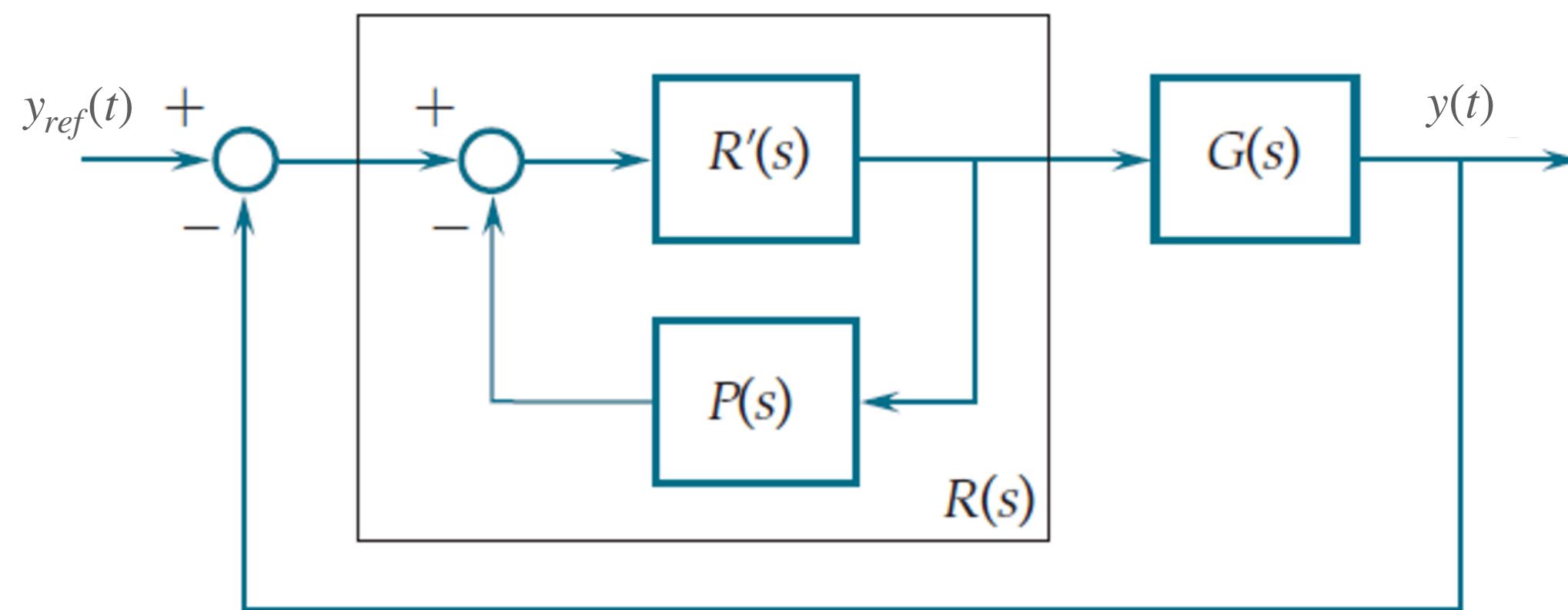


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

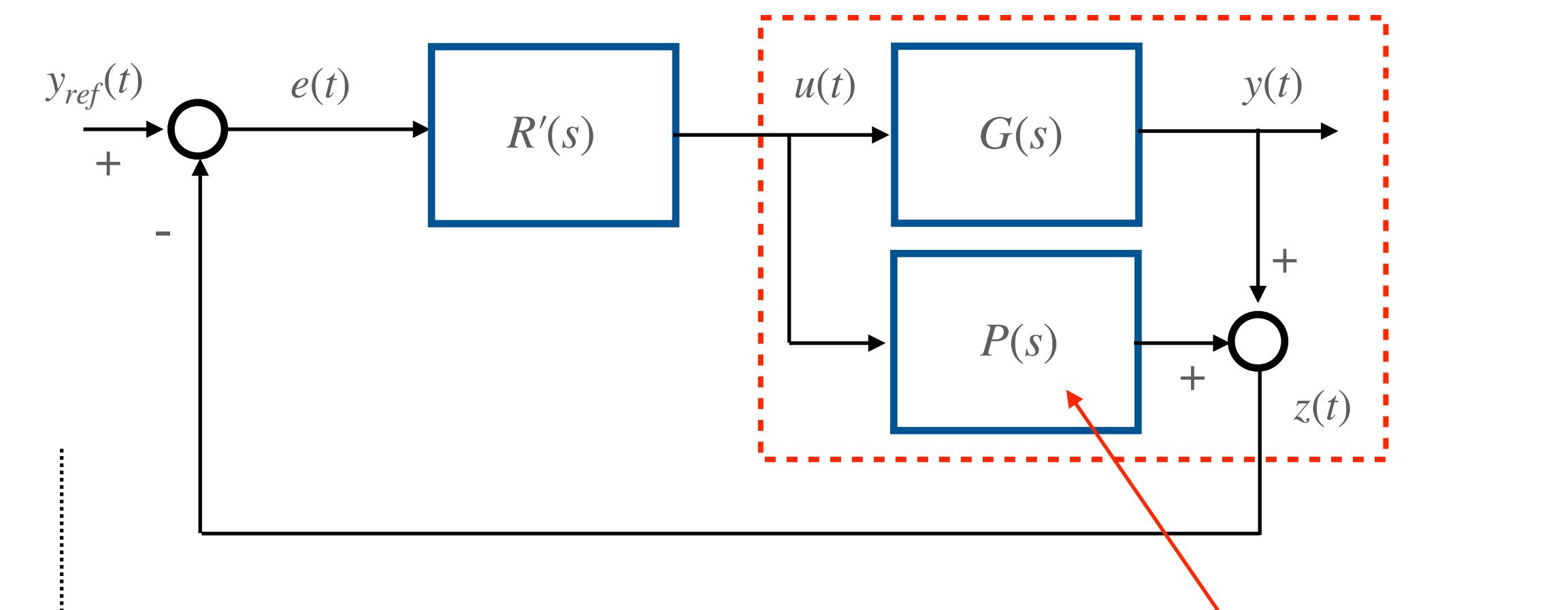


## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation



$$\begin{aligned} R(s) &= \frac{R'(s)}{1 + R'(s)P(s)} \\ &= \frac{N_{R'}(s)/D_{R'}(s)}{1 + (N_{R'}(s)/D_{R'}(s))(1 - e^{-\tau s})(N(s)/D(s))} \end{aligned}$$

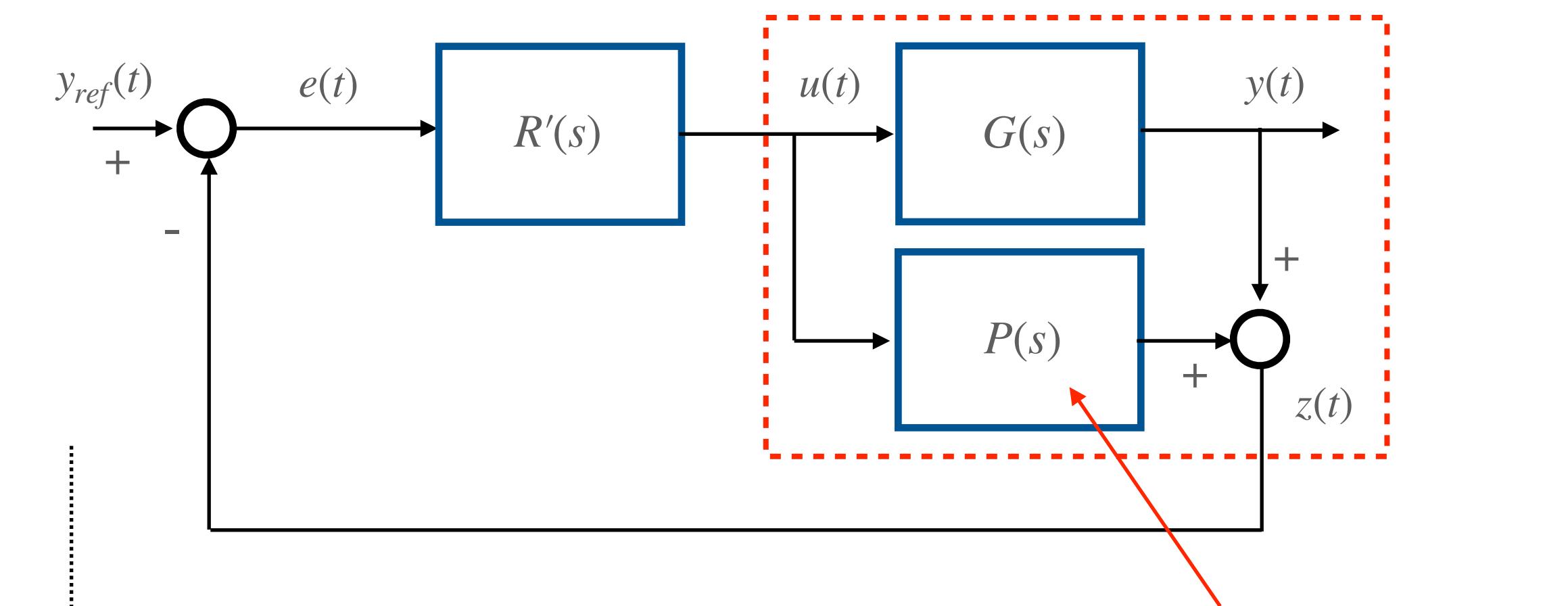
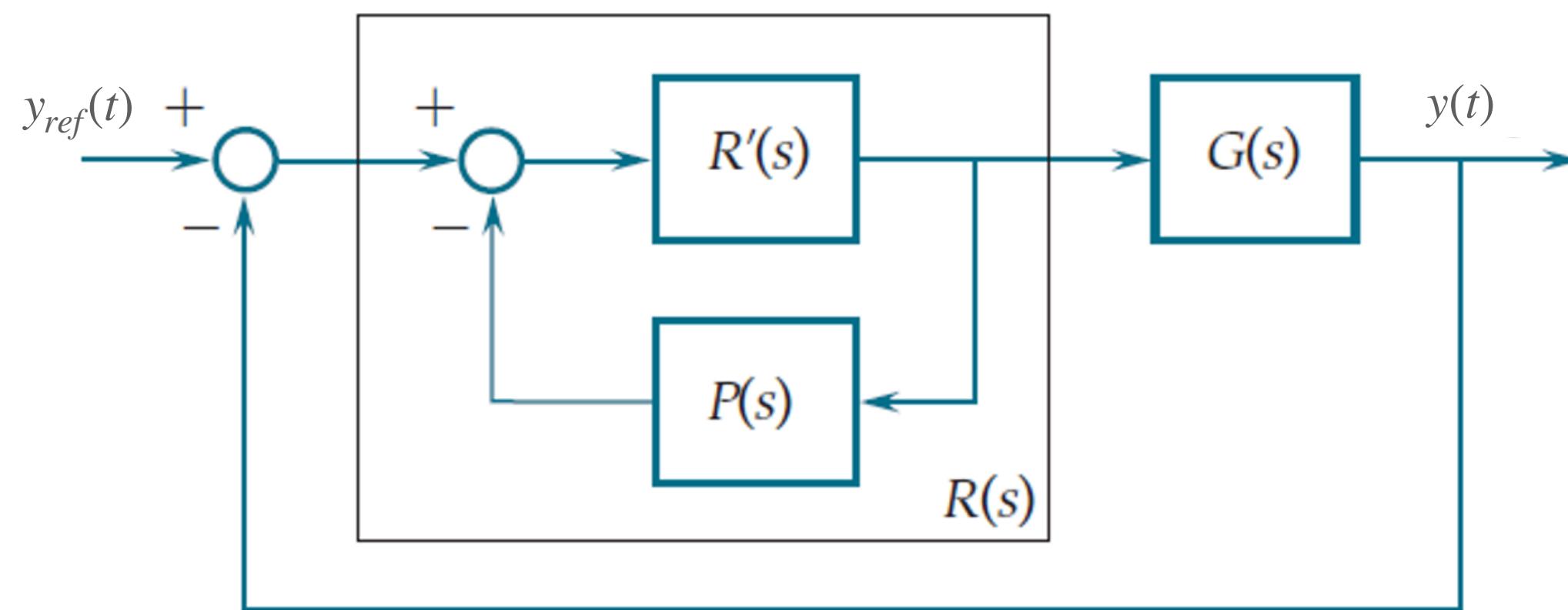


$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation



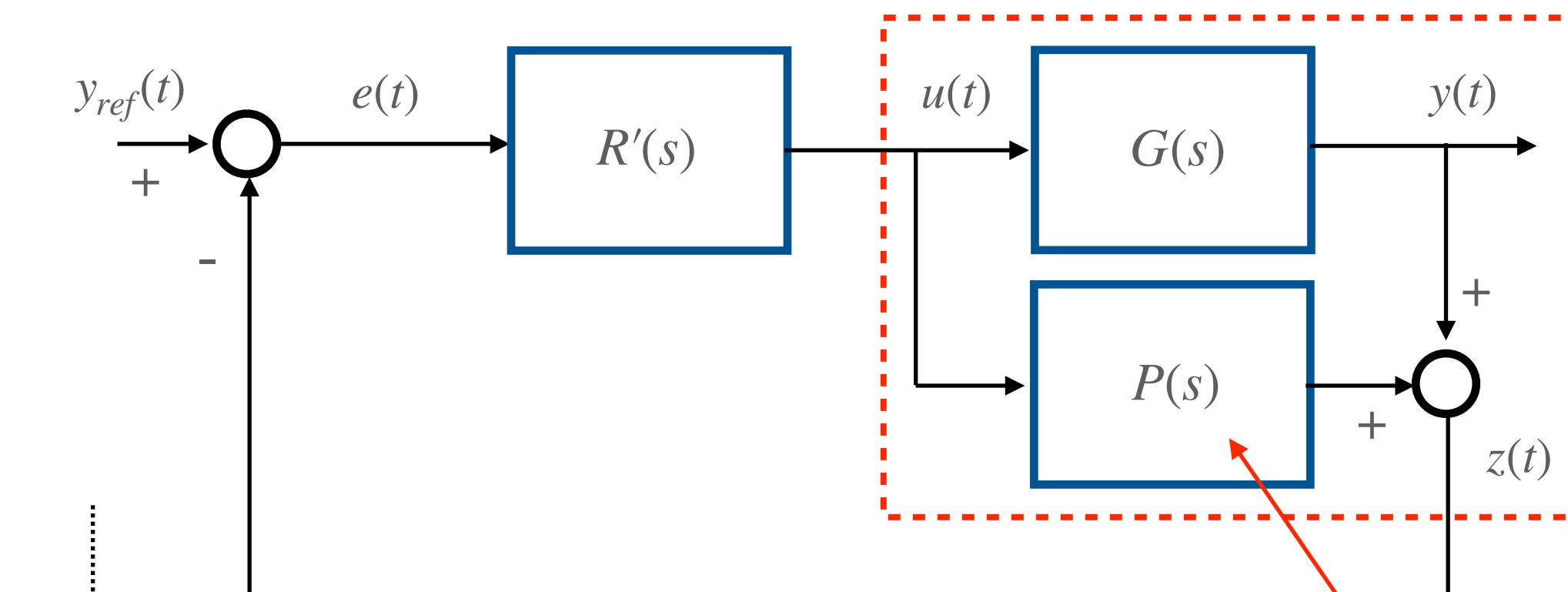
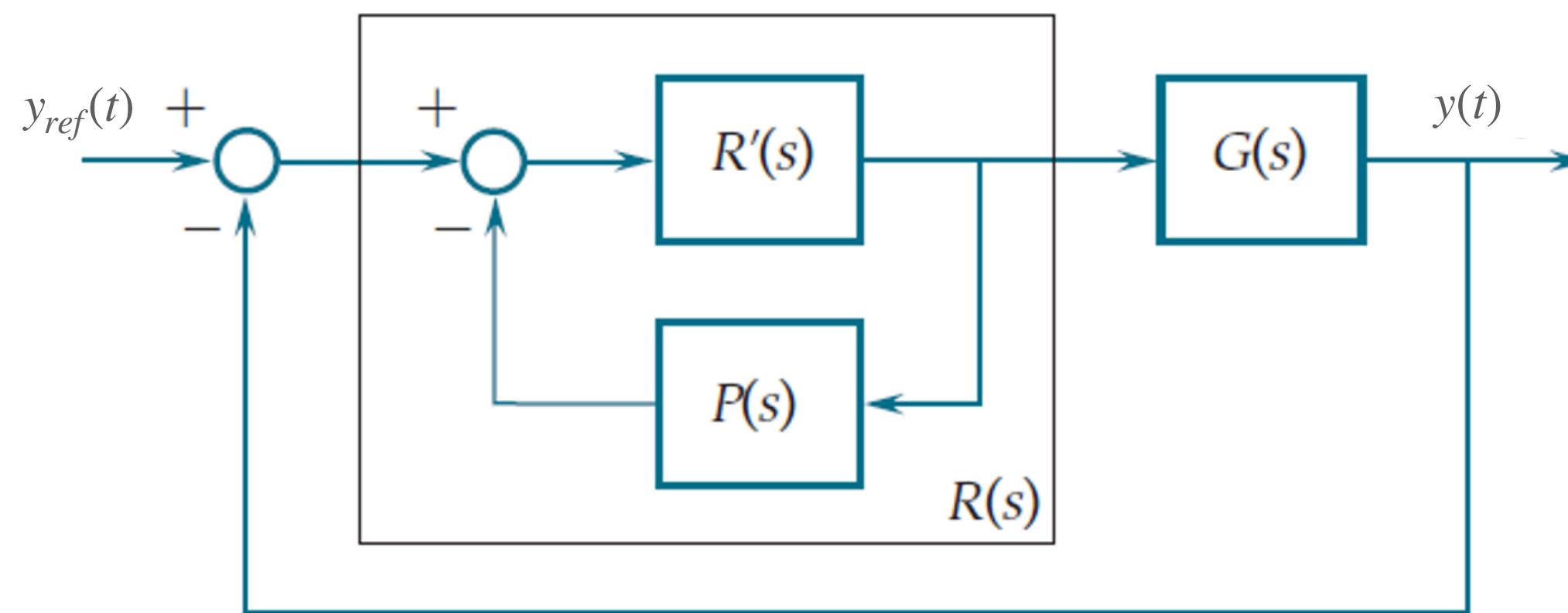
$$\begin{aligned}
 R(s) &= \frac{R'(s)}{1 + R'(s)P(s)} \\
 &= \frac{N_{R'}(s)/D_{R'}(s)}{1 + (N_{R'}(s)/D_{R'}(s))(1 - e^{-\tau s})(N(s)/D(s))} \\
 &= \frac{N_{R'}(s)D(s)}{D_{R'}(s)D(s) + N_{R'}(s)N(s)(1 - e^{-\tau s})}
 \end{aligned}$$

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 2: alternative formulation



$$\begin{aligned}
 R(s) &= \frac{R'(s)}{1 + R'(s)P(s)} \\
 &= \frac{N_{R'}(s)/D_{R'}(s)}{1 + (N_{R'}(s)/D_{R'}(s))(1 - e^{-\tau s})(N(s)/D(s))} \\
 &= \frac{N_{R'}(s)D(s)}{D_{R'}(s)D(s) + N_{R'}(s)N(s)(1 - e^{-\tau s})}
 \end{aligned}$$

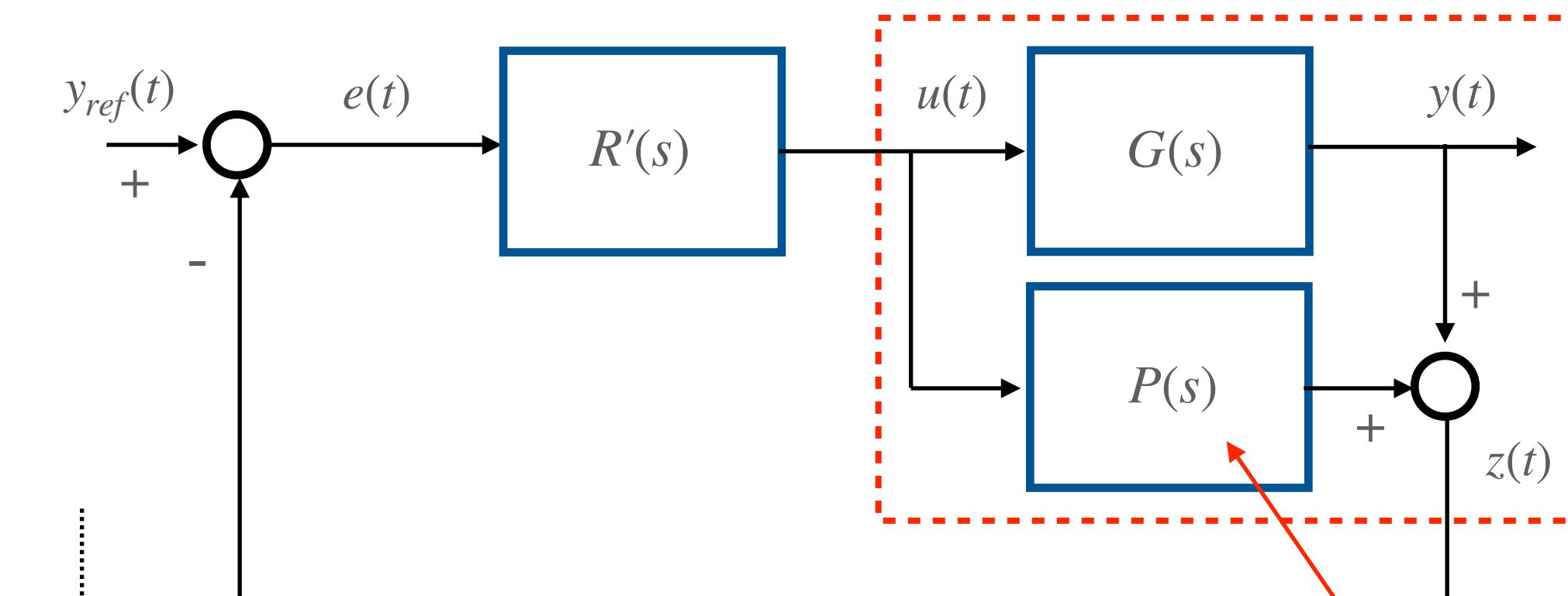
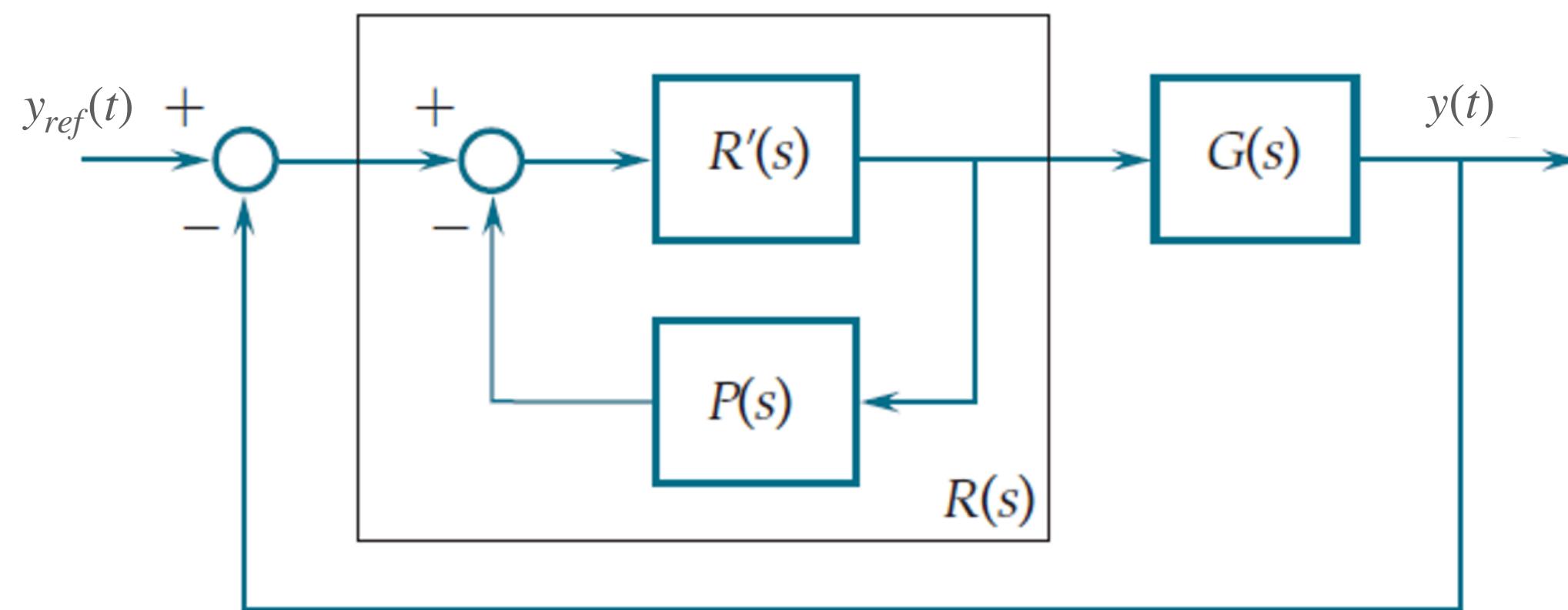
$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

synthesis by cancellation



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Observation 3:



$$\begin{aligned}
 R(s) &= \frac{R'(s)}{1 + R'(s)P(s)} \\
 &= \frac{N_{R'}(s)/D_{R'}(s)}{1 + (N_{R'}(s)/D_{R'}(s))(1 - e^{-\tau s})(N(s)/D(s))} \\
 &= \frac{N_{R'}(s)D(s)}{D_{R'}(s)D(s) + N_{R'}(s)N(s)(1 - e^{-\tau s})}
 \end{aligned}$$

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

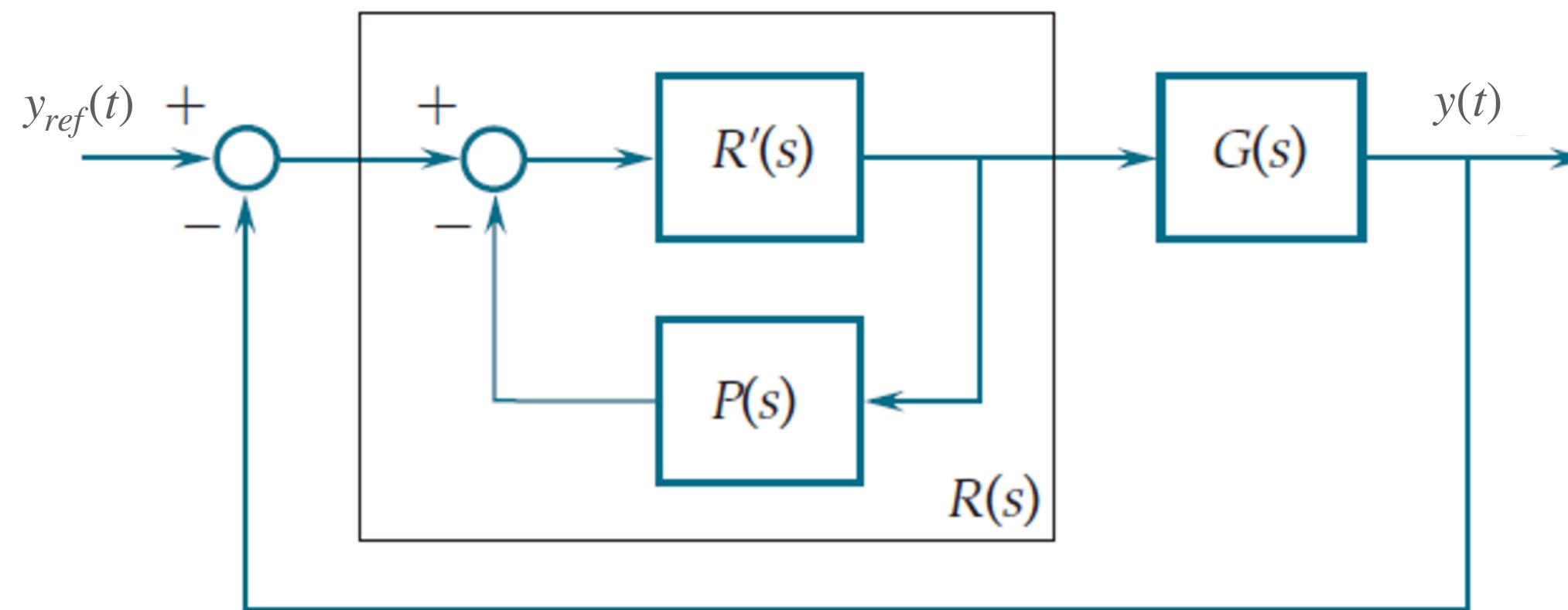
$R(s)$  is a controller with a non-rational transfer function



## Control of LTI Systems with Delays: Smith Predictor Based Control Scheme

Summary:

Alternative scheme



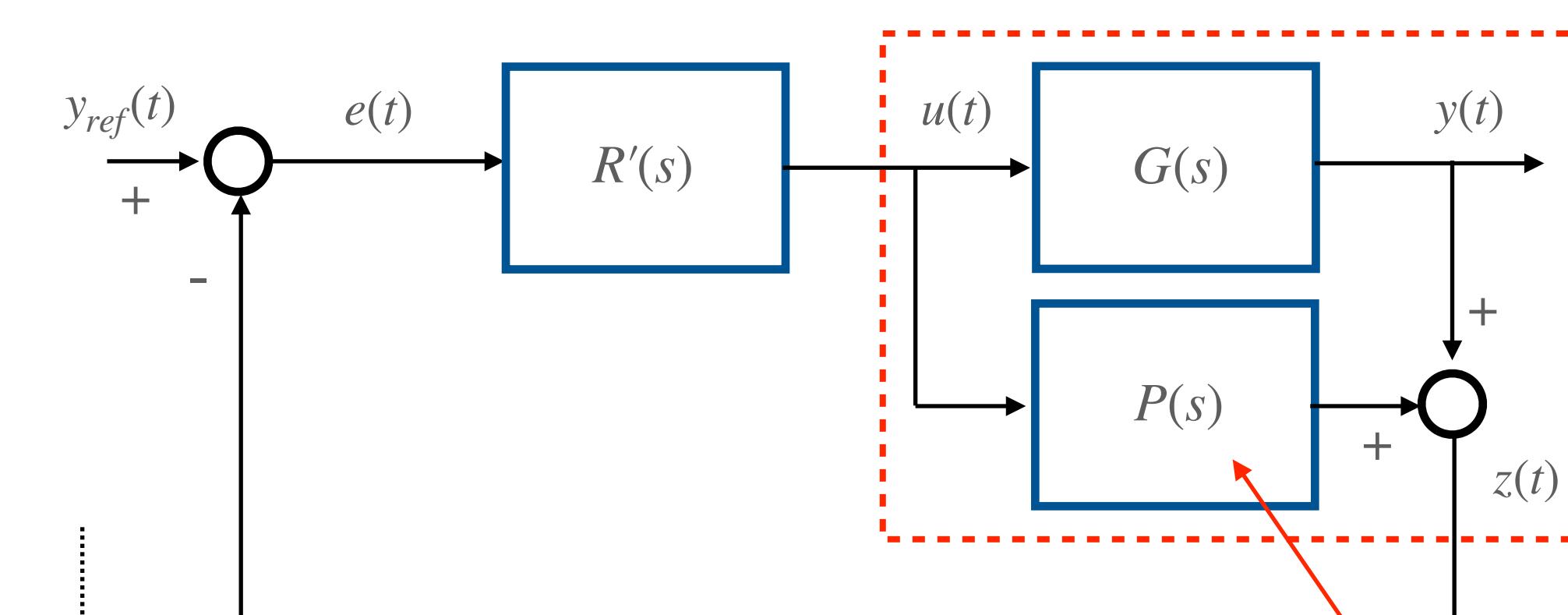
$z(t)$  is a prediction  
of  $y(t)$

$$Z(s) = e^{\tau s} Y(s)$$

$y(t)$  is  $z(t)$  delayed of  $\tau$  seconds

$$Y(s) = e^{-\tau s} Z(s)$$

Classical scheme



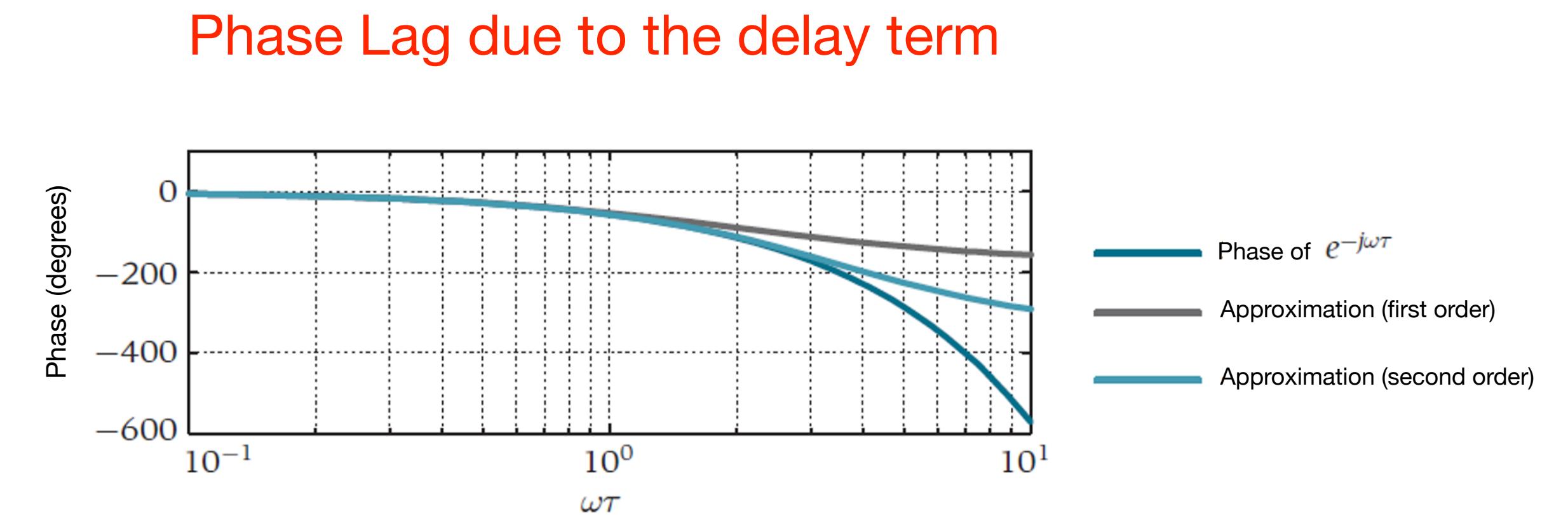
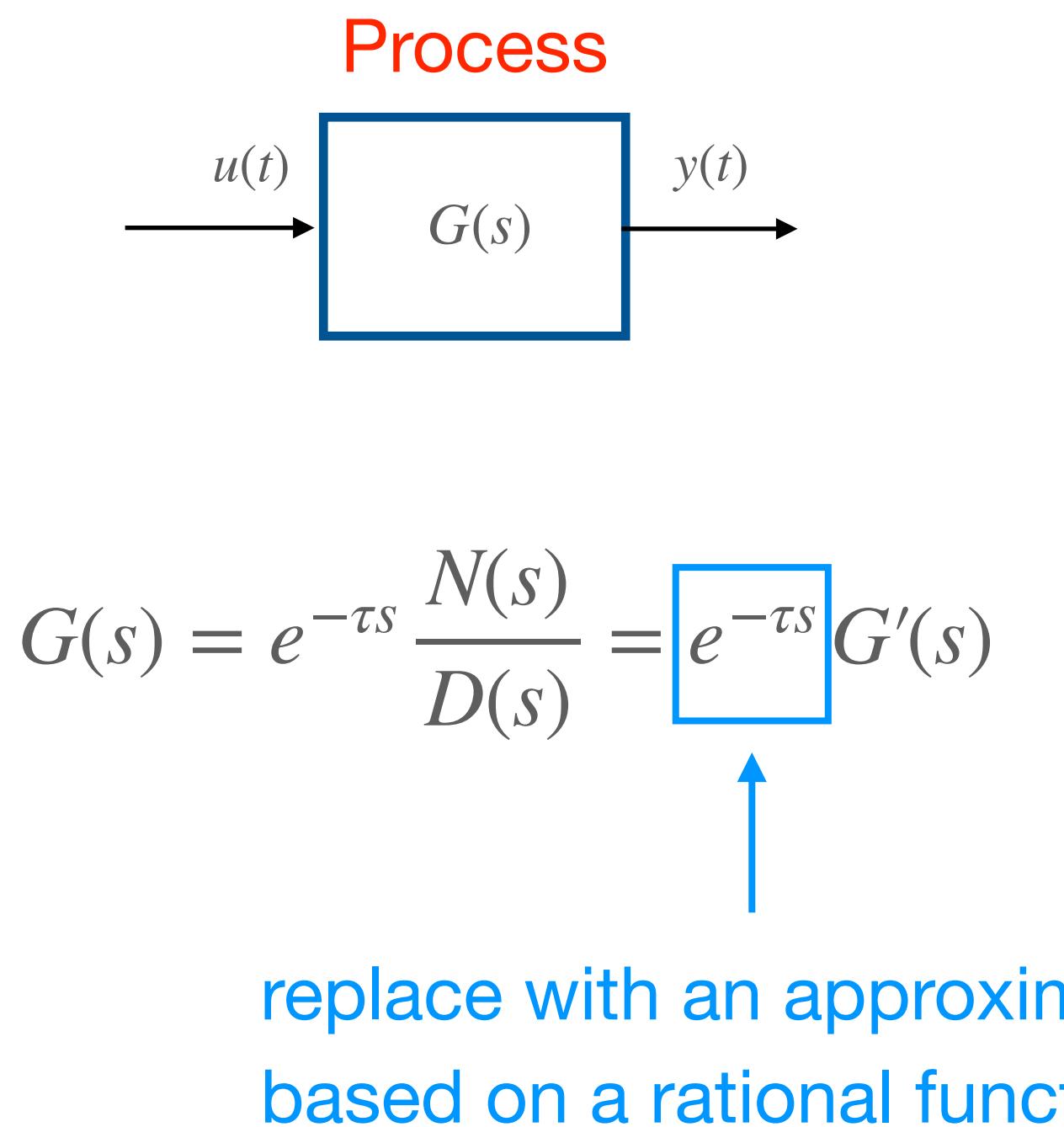
Smith Predictor

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

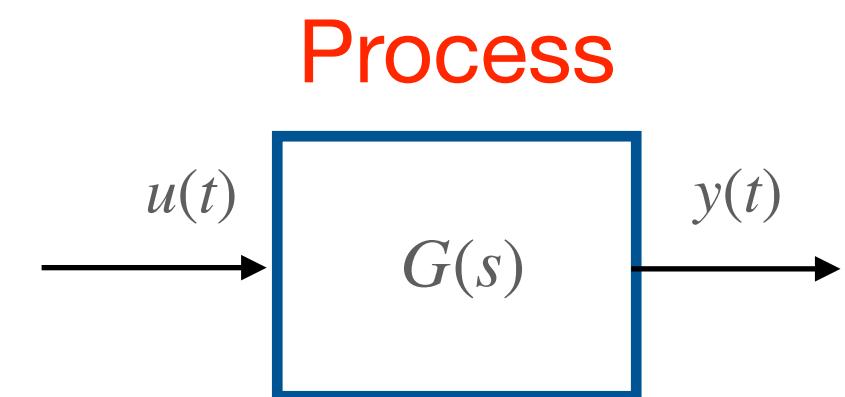
the delay has not disappeared!



## Control of LTI Systems with Delays: Padé Approximation



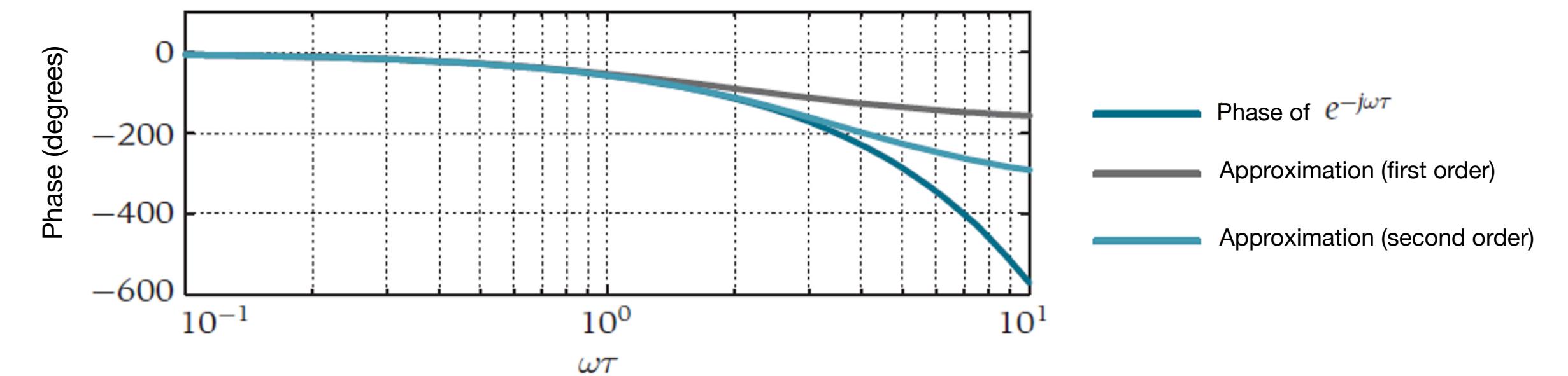
## Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Function to approximate:  $e^{-\tau s}$

Phase Lag due to the delay term

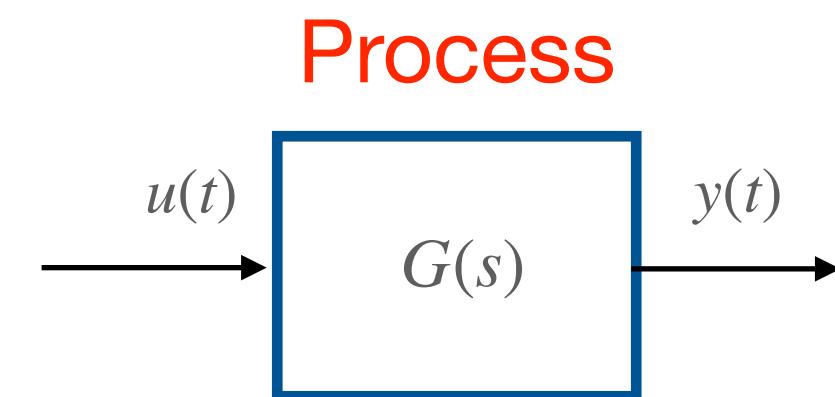


First order approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

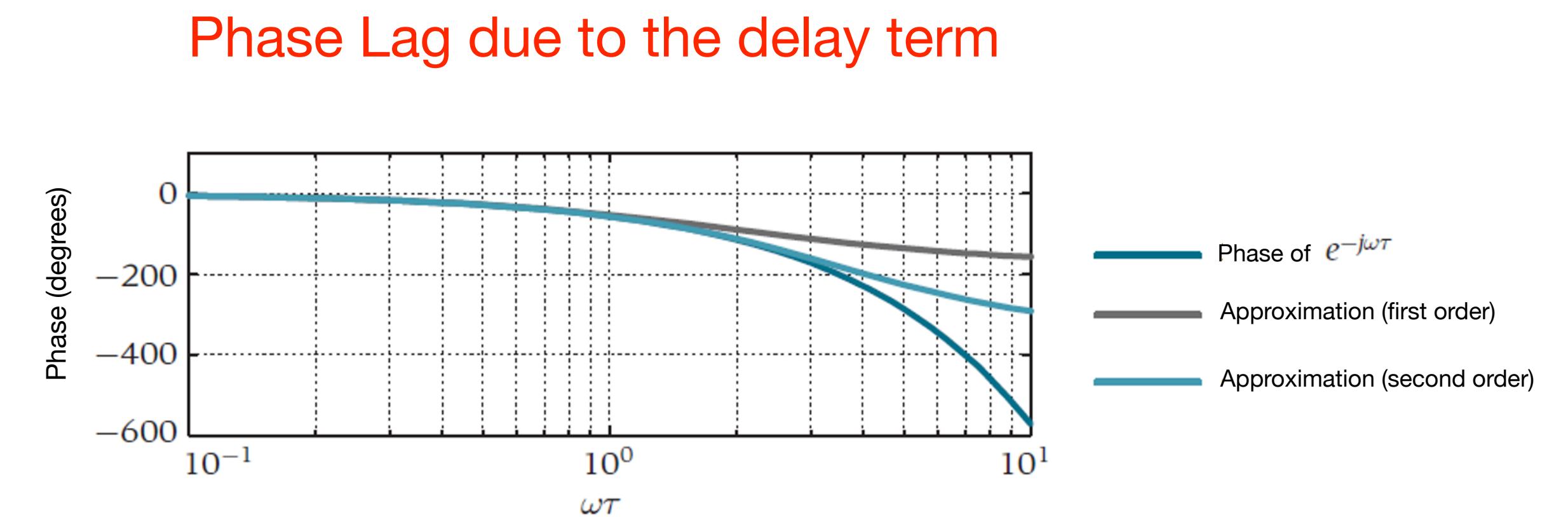


## Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

Function to approximate:  $e^{-\tau s}$



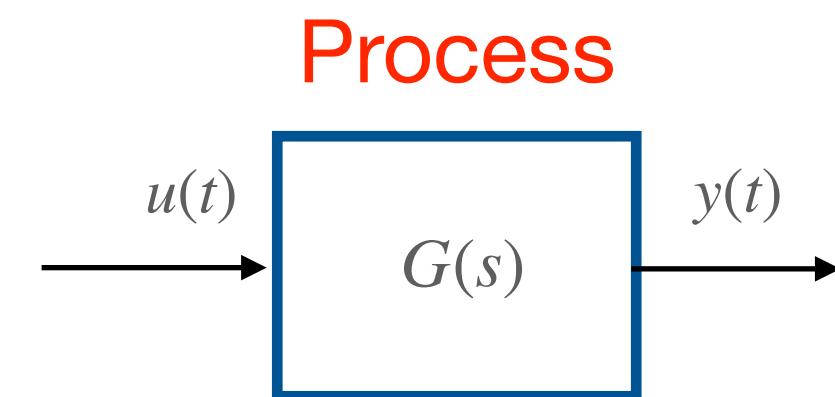
First order approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

How to select the parameters?



## Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

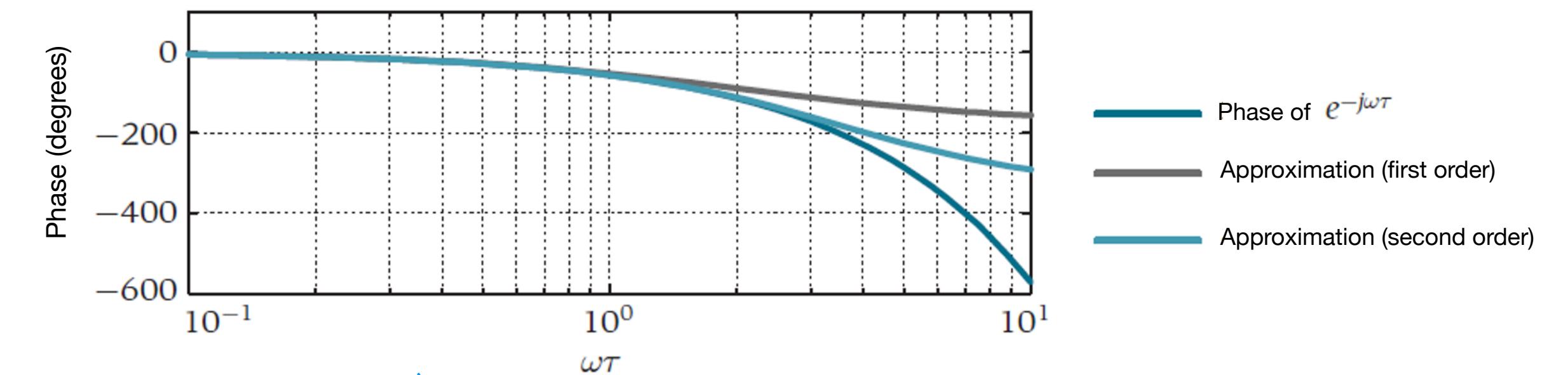
Function to approximate:

$$e^{-\tau s}$$

First order approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

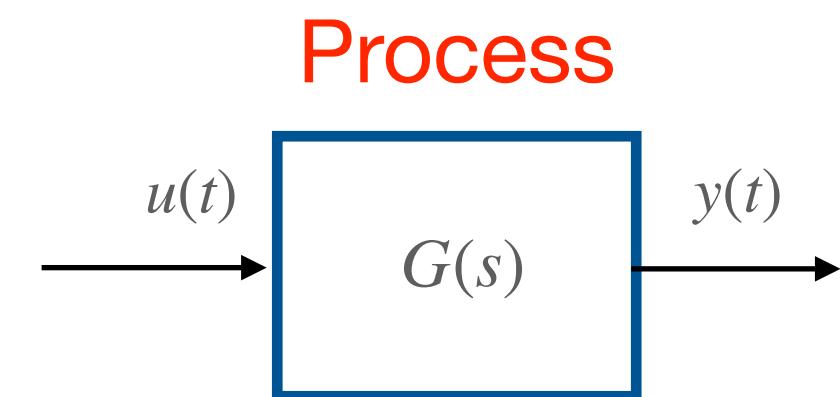
Phase Lag due to the delay term



Use MacLaurin series expansion



## Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

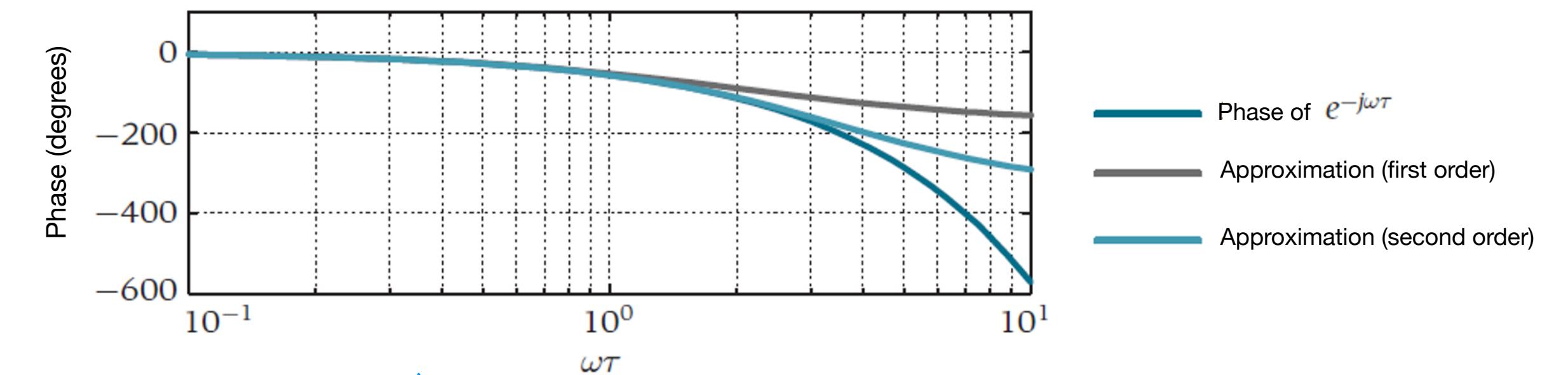
Function to approximate:

$$e^{-\tau s}$$

First order approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

Phase Lag due to the delay term

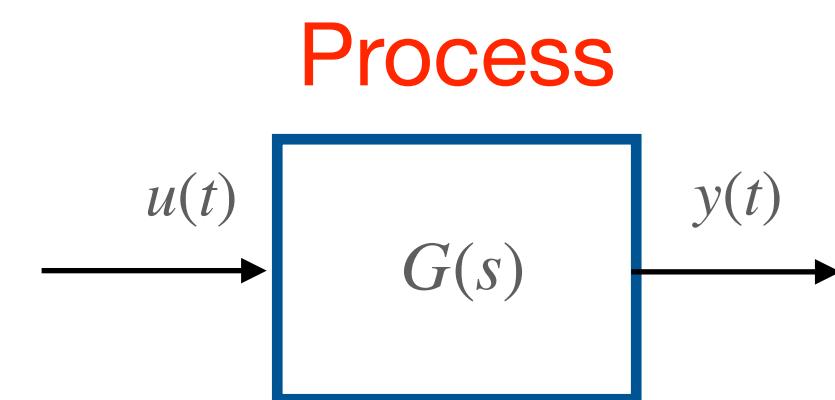


Use MacLaurin series expansion

$$f(s) = f(0) + \frac{df}{ds}(0)s + \frac{1}{2!} \frac{d^2f}{ds^2}(0)s^2 + \frac{1}{3!} \frac{d^3f}{ds^3}(0)s^3 + \mathcal{O}(s^4)$$



## Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

Function to approximate:

McLaurin approximation:

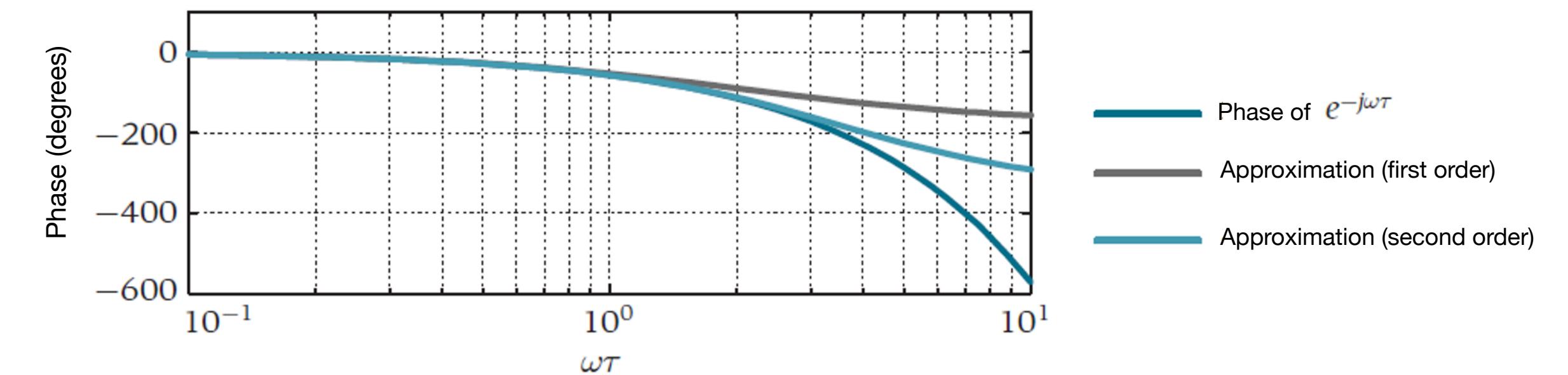
First order approximation:

$$e^{-\tau s}$$

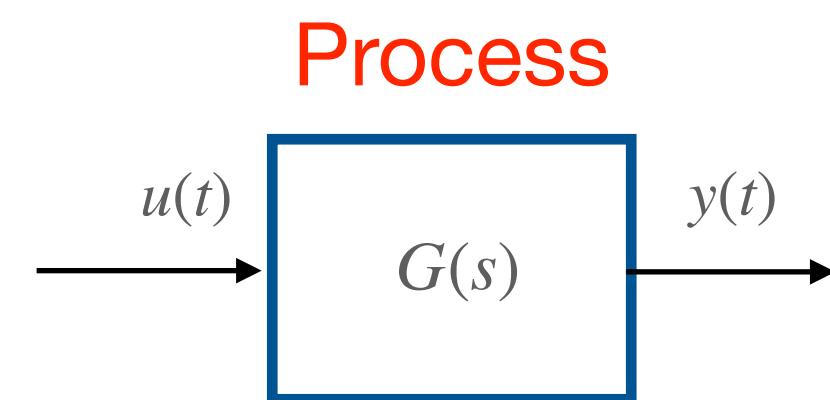
$$e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$$

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

Phase Lag due to the delay term



## Control of LTI Systems with Delays: Padé Approximation



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

Function to approximate:

$$e^{-\tau s}$$

McLaurin approximation:

$$e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$$

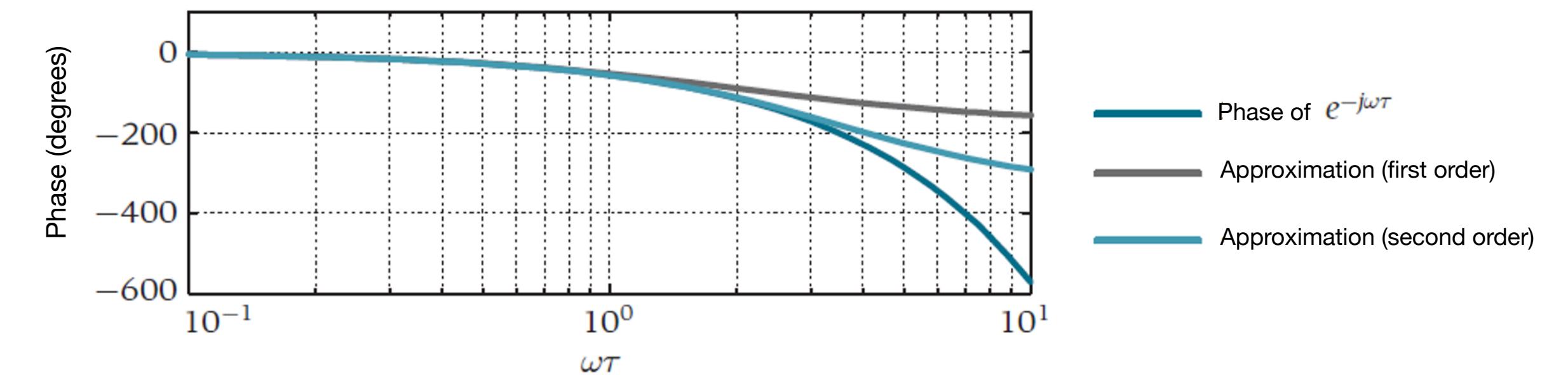
First order approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs}$$

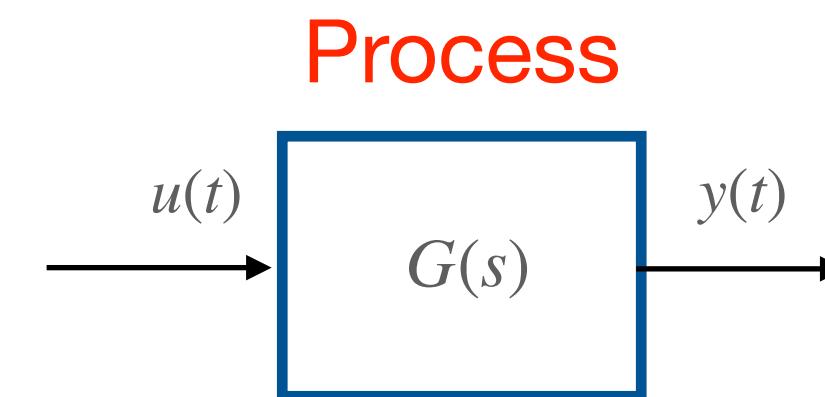
McLaurin approximation:

$$G_I(s) = \mu \frac{1 + as}{1 + bs} = \approx \mu + \mu(a - b)s - \mu b(a - b)s^2$$

Phase Lag due to the delay term



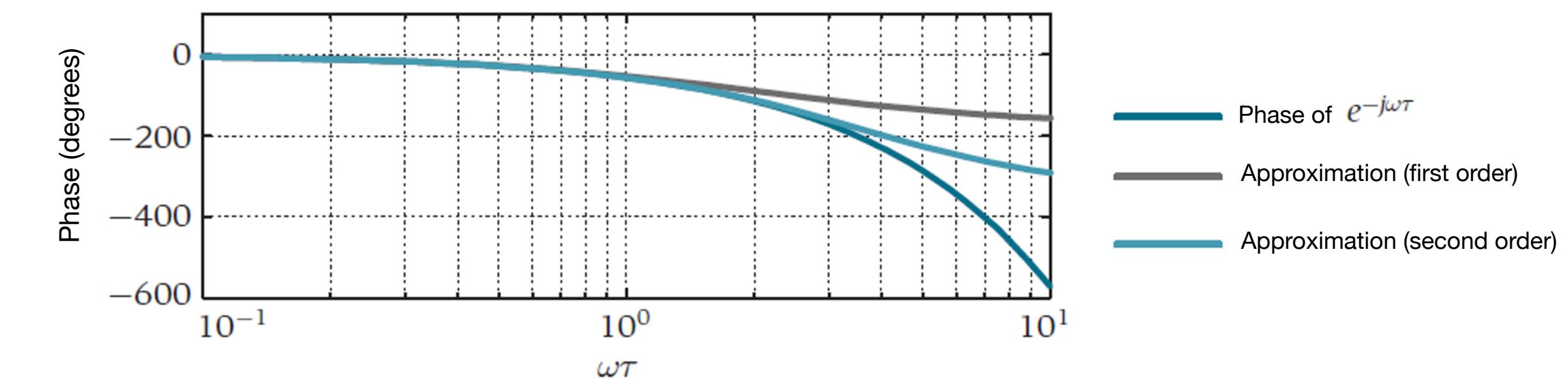
## Control of LTI Systems with Delays: Padé Approximation



$$f(s) = f(0) + \frac{df}{ds}(0)s + \frac{1}{2!} \frac{d^2f}{ds^2}(0)s^2 + \frac{1}{3!} \frac{d^3f}{ds^3}(0)s^3 + \mathcal{O}(s^4)$$

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

Phase Lag due to the delay term



$$\mu = 1$$

$$\mu(a-b) = -\tau$$

$$-\mu b(a-b) = \frac{\tau^2}{2}$$

Function to approximate:

McLaurin approximation:

First order approximation:

McLaurin approximation:

$$e^{-\tau s}$$

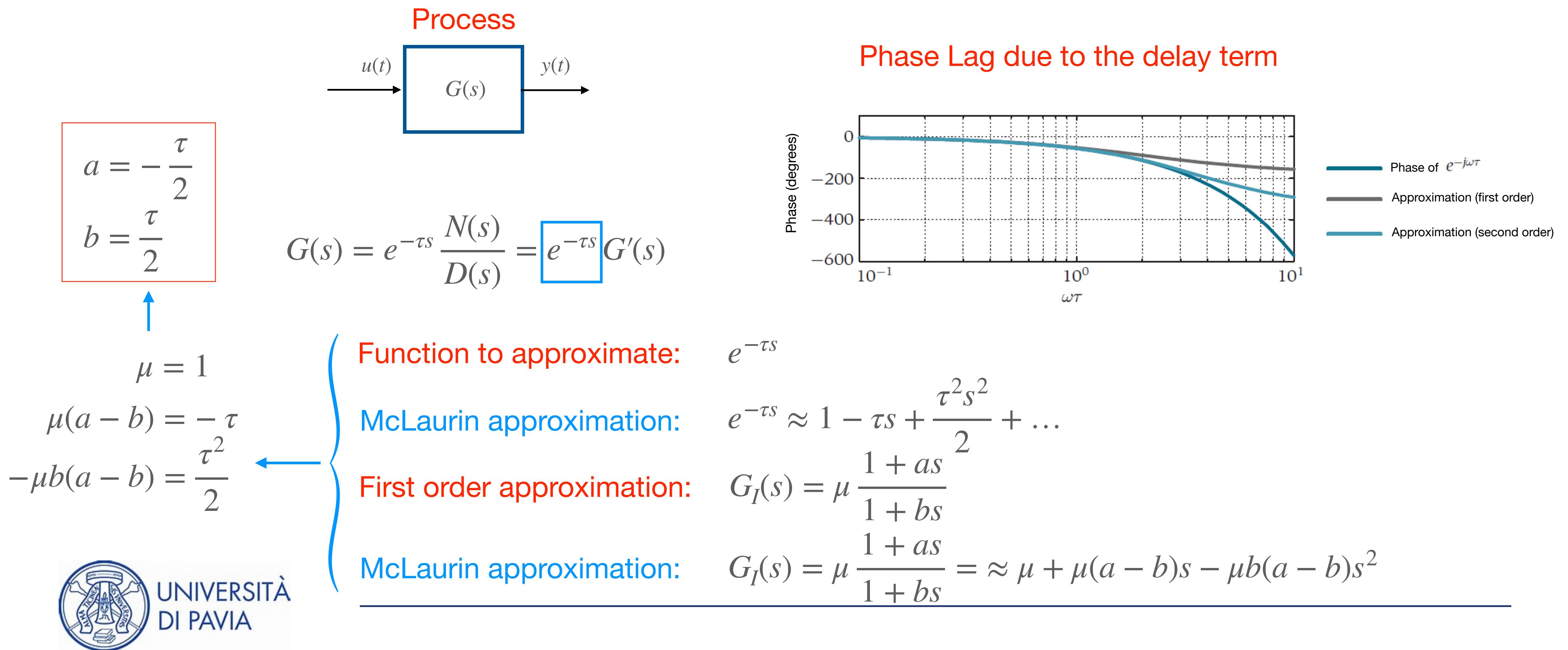
$$e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$$

$$G_I(s) = \mu \frac{1+as}{1+bs}$$

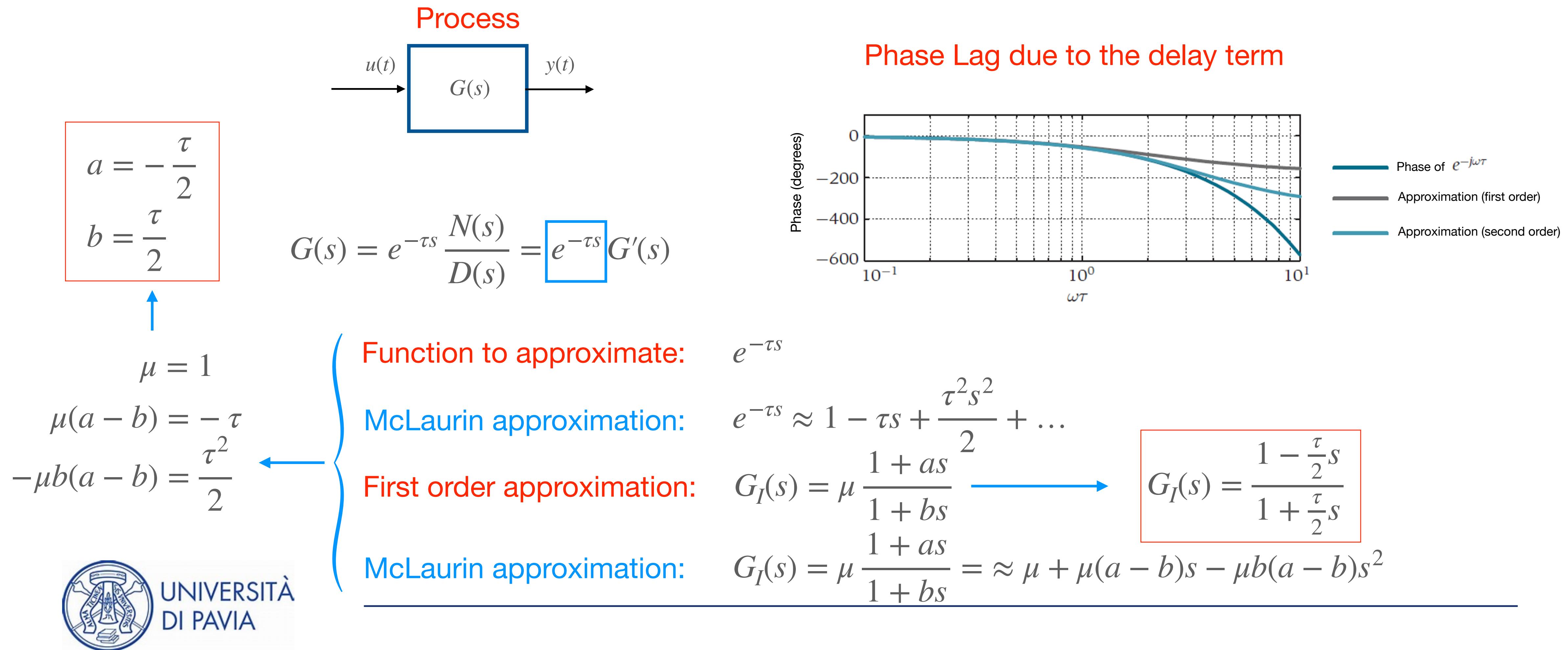
$$G_I(s) = \mu \frac{1+as}{1+bs} \approx \mu + \mu(a-b)s - \mu b(a-b)s^2$$



## Control of LTI Systems with Delays: Padé Approximation



## Control of LTI Systems with Delays: Padé Approximation



## Control of LTI Systems with Delays: Padé Approximation

**Process**

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

**Phase Lag due to the delay term**

**Function to approximate:**  $e^{-\tau s}$

**McLaurin approximation:**  $e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$

**First order approximation:**  $G_I(s) = \mu \frac{1 + as}{1 + bs}$

**McLaurin approximation:**  $G_I(s) = \mu \frac{1 + as}{1 + bs} = \approx \mu + \mu(a - b)s - \mu b(a - b)s^2$

**Magnitude and phase ?**

$a = -\frac{\tau}{2}$

$b = \frac{\tau}{2}$

$\mu = 1$

$\mu(a - b) = -\tau$

$-\mu b(a - b) = \frac{\tau^2}{2}$

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## Control of LTI Systems with Delays: Padé Approximation

**Process**

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

**Phase Lag due to the delay term**

**Function to approximate:**  $e^{-\tau s}$

**McLaurin approximation:**  $e^{-\tau s} \approx 1 - \tau s + \frac{\tau^2 s^2}{2} + \dots$

**First order approximation:**  $G_I(s) = \mu \frac{1 + as}{1 + bs}$

**McLaurin approximation:**  $G_I(s) = \mu \frac{1 + as}{1 + bs} = \approx \mu + \mu(a - b)s - \mu b(a - b)s^2$

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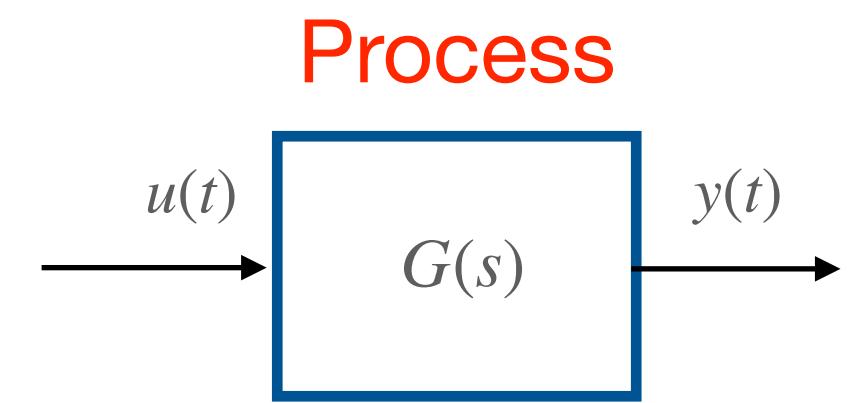
$\mu(a - b) = -\tau$

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## Control of LTI Systems with Delays

Summary:

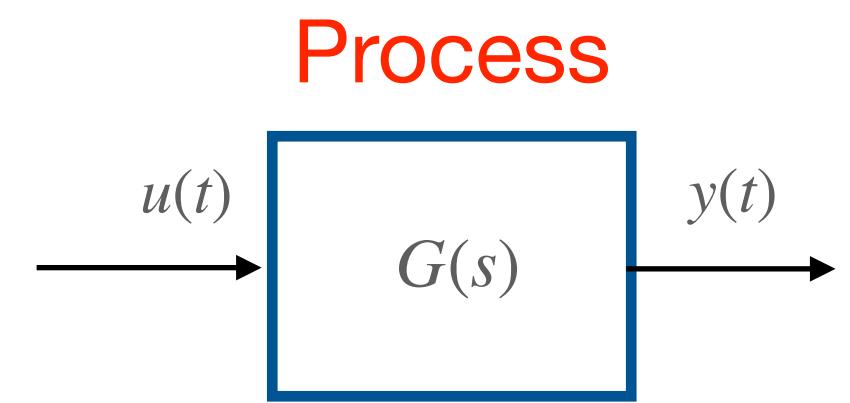


$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$



## Control of LTI Systems with Delays

Summary:



$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = \boxed{e^{-\tau s}} G'(s)$$

$e^{-\tau s}$

Modelling alternatives:

$$G_I(s) = \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s}$$

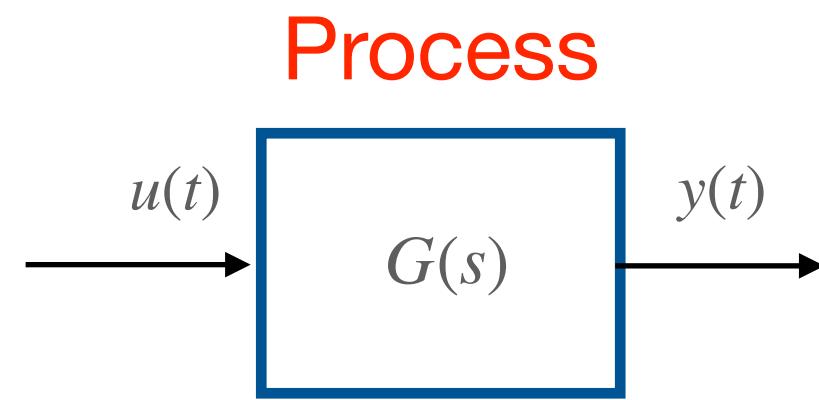
$$G_{II}(s) = \dots$$

$$G_{III}(s) = \dots$$



## Control of LTI Systems with Delays

Summary:



Modelling alternatives:

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

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$$G_I(s) = \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s}$$

$$G_{II}(s) = \dots$$

$$G_{III}(s) = \dots$$

Design alternatives for  $R(s)$ :

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

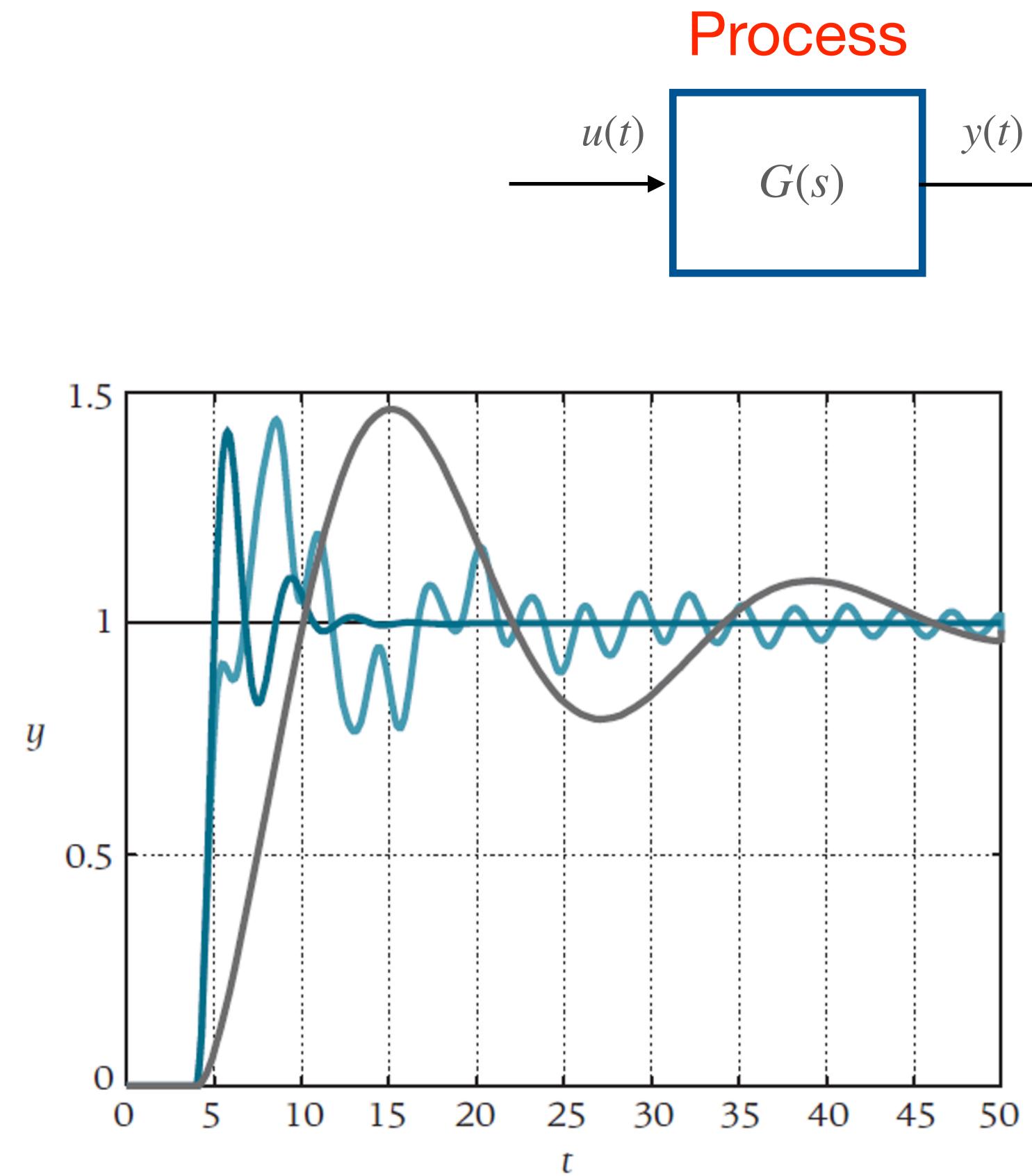
$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} \approx \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s} G'(s)$$

$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} \approx \left( 1 - \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s} \right) G'(s)$$



## Control of LTI Systems with Delays

Summary:



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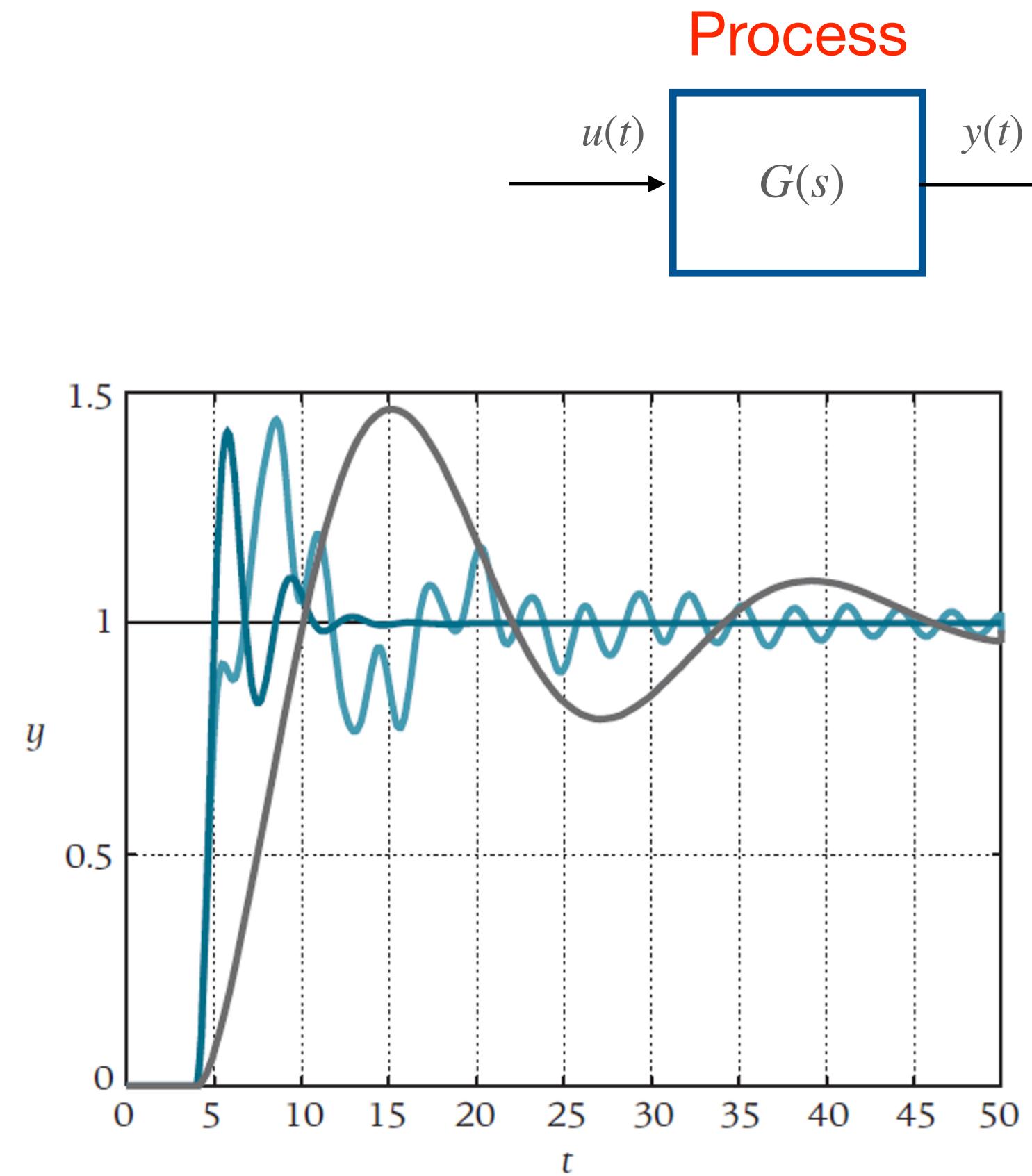
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## Control of LTI Systems with Delays

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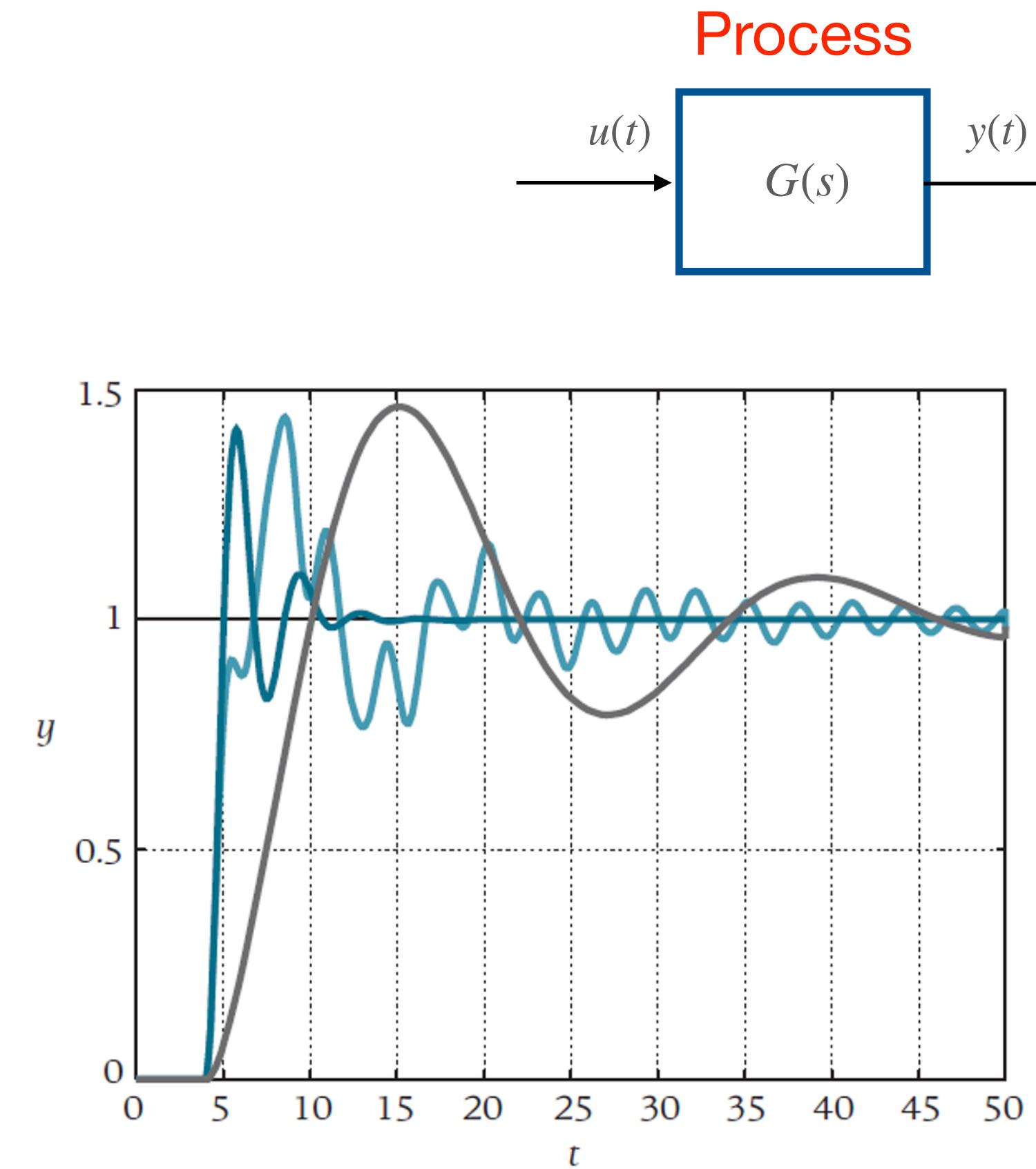
$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} \approx \frac{1 - \frac{\tau}{2}s}{1 + \frac{\tau}{2}s} G'(s)$$

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## Control of LTI Systems with Delays

Summary:



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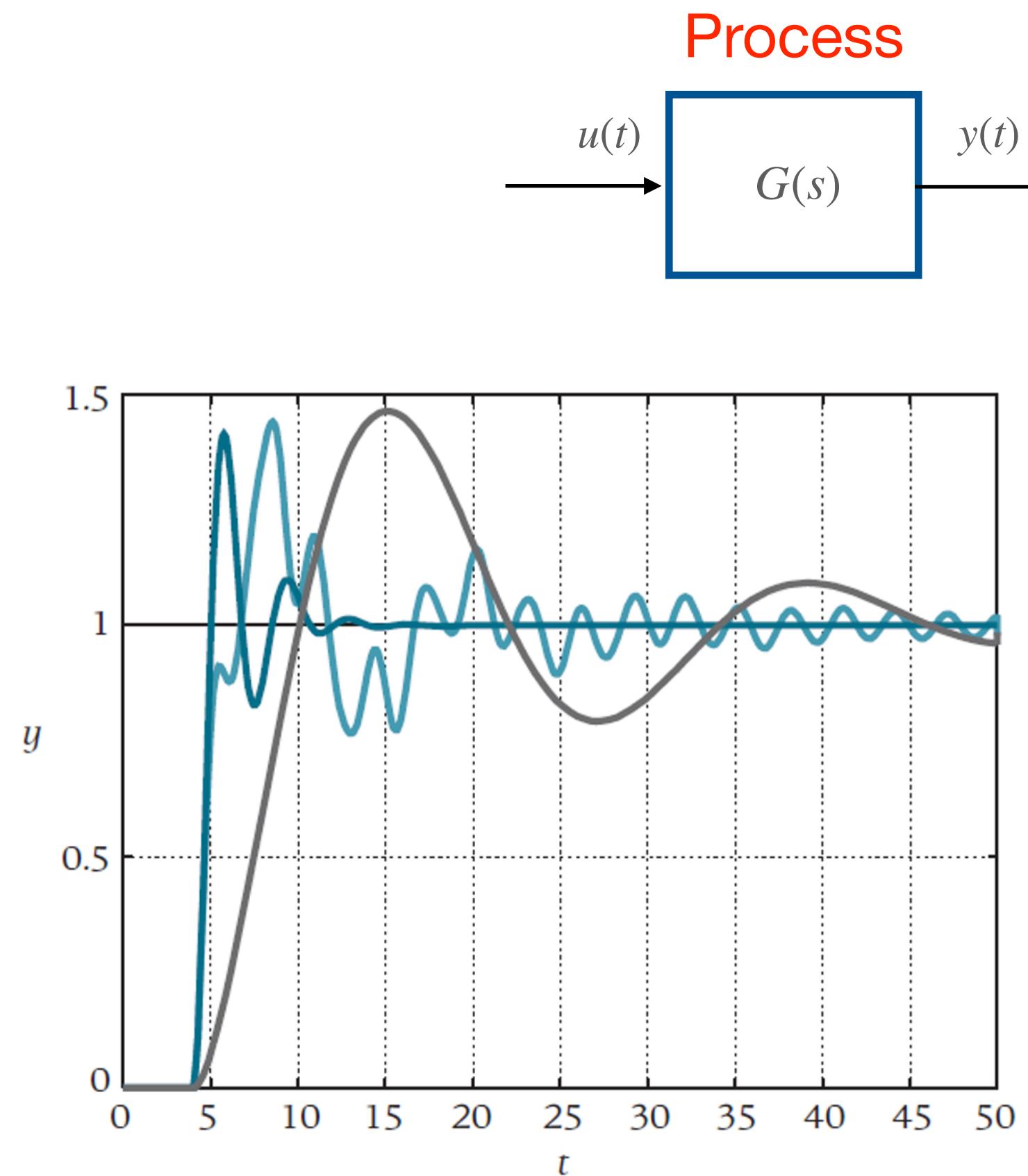
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## Control of LTI Systems with Delays

Summary:



Exercise: Use the  
UFCS and design  
 $R(s)$  relying on

Design alternatives for  $R(s)$ :

$$G(s) = e^{-\tau s} \frac{N(s)}{D(s)} = e^{-\tau s} G'(s)$$

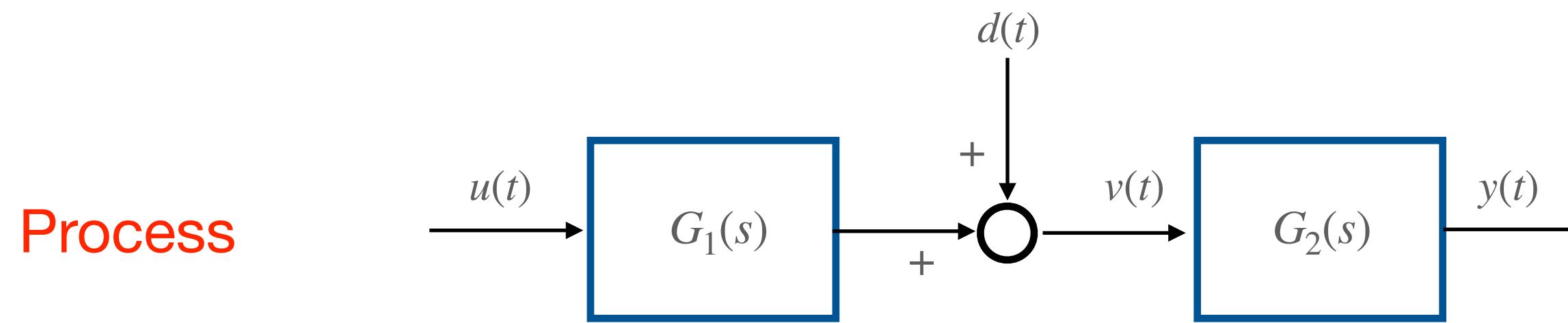
$$P(s) = (1 - e^{-\tau s}) \frac{N(s)}{D(s)} = (1 - e^{-\tau s}) G'(s)$$

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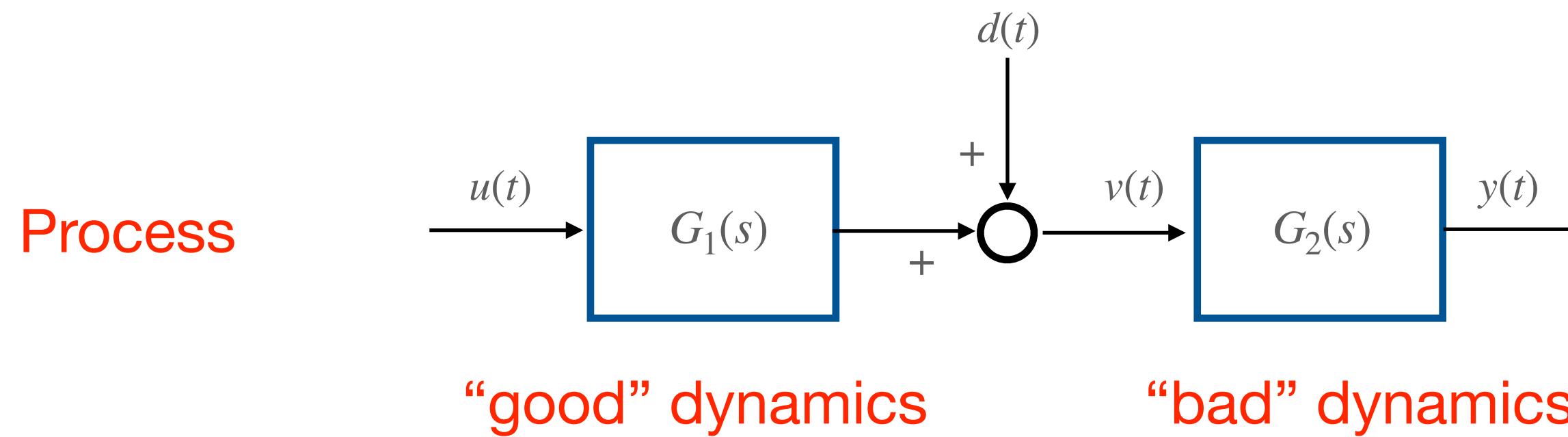
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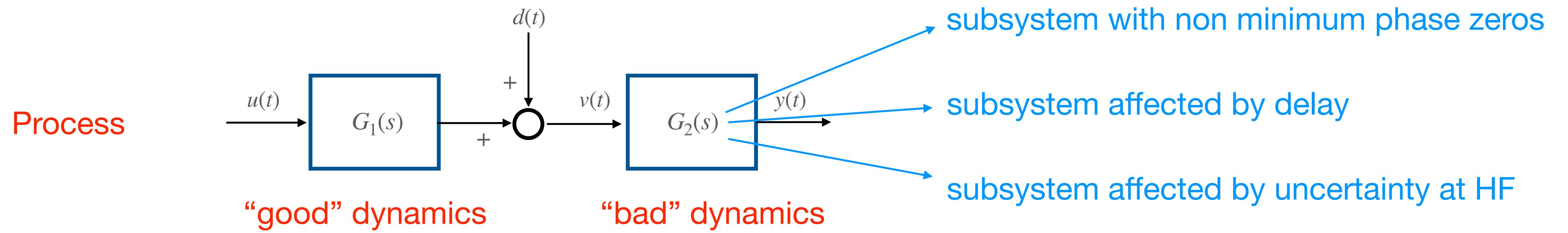
## Control Scheme with Decoupling in the Frequency Domain



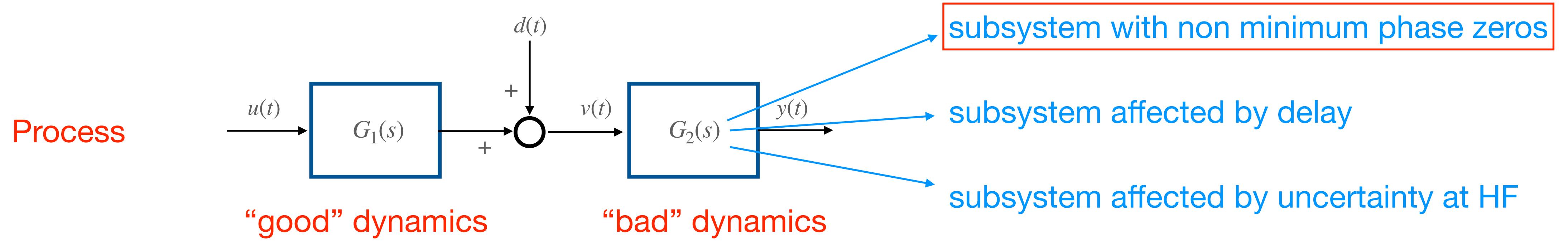
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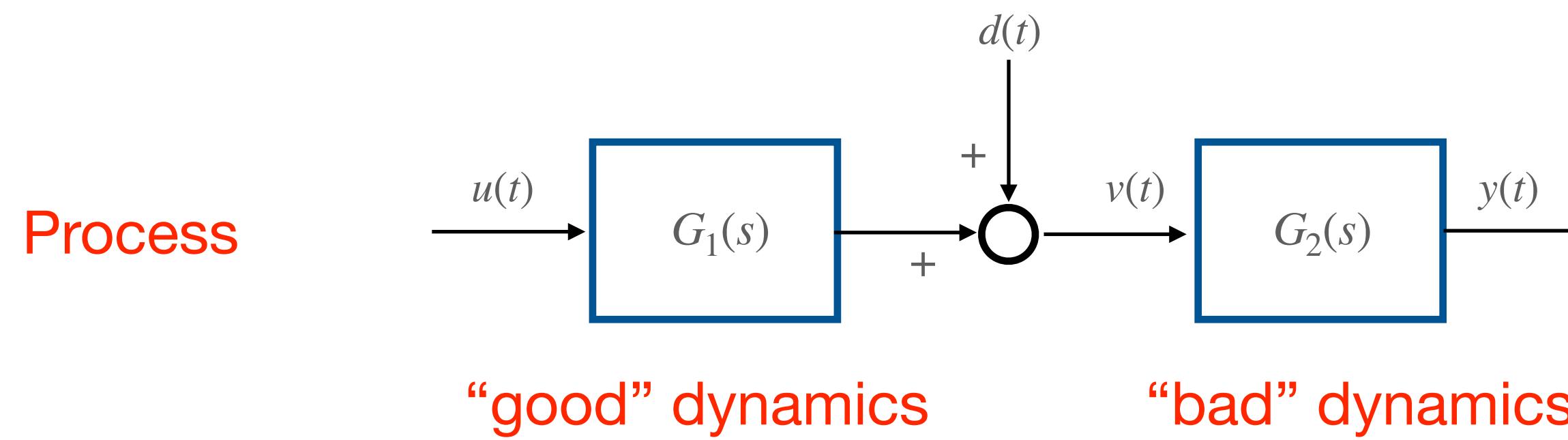
## Control Scheme with Decoupling in the Frequency Domain



## Control Scheme with Decoupling in the Frequency Domain



## Control Scheme with Decoupling in the Frequency Domain

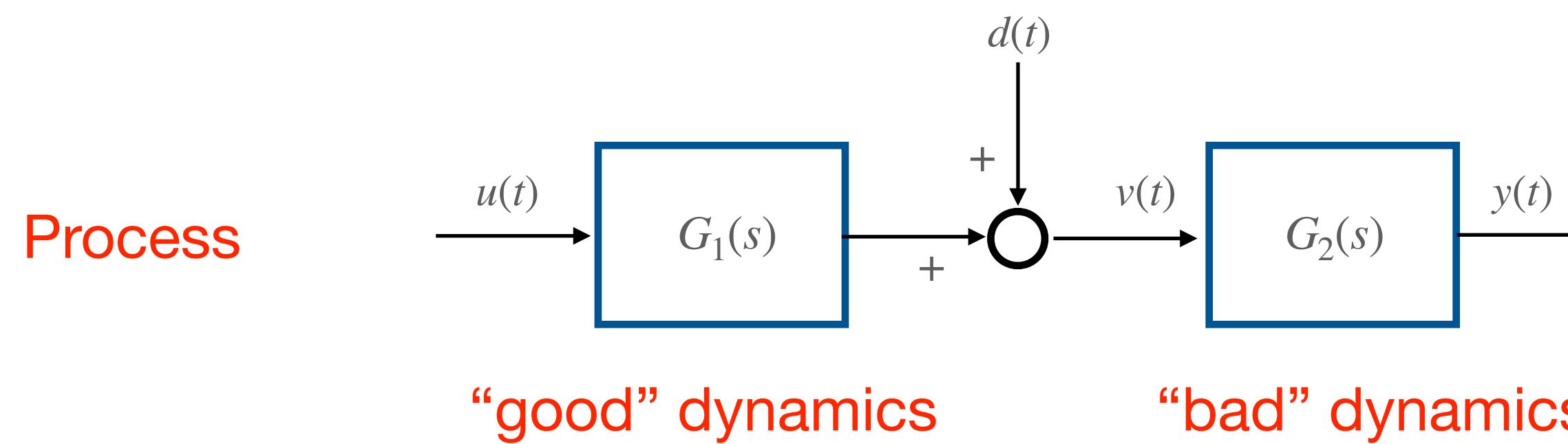


### Assumptions:

- The performances that can be obtained by designing a control system for  $G_1(s)$  alone are better than those that can be obtained for the entire system  $G_1(s)G_2(s)$
- $v$  is accessible (measurable)



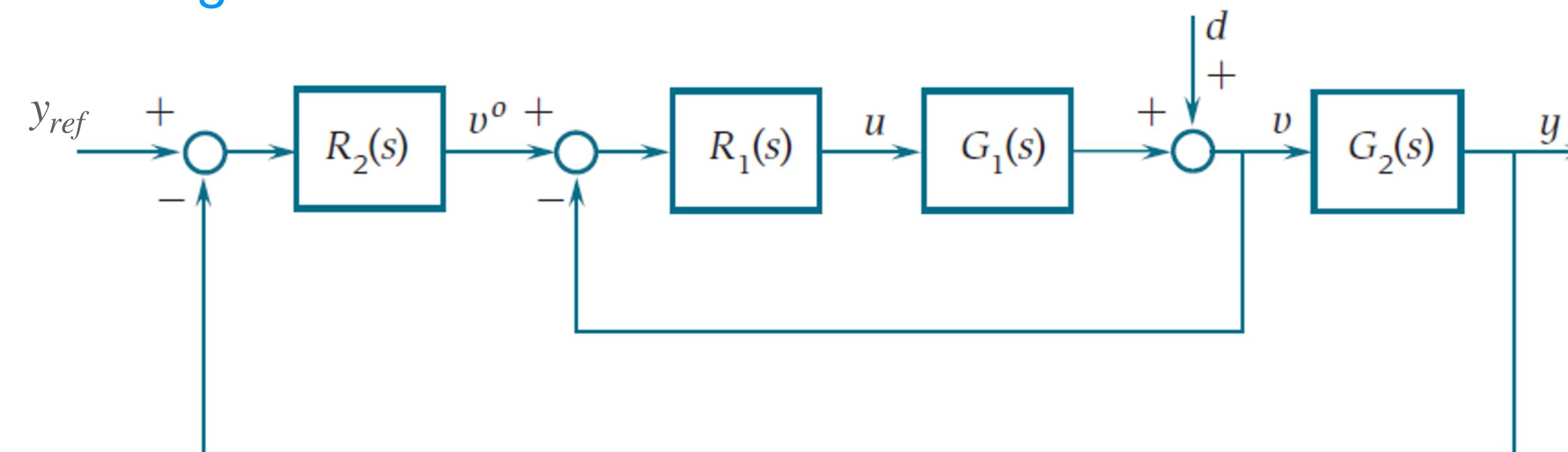
## Control Scheme with Decoupling in the Frequency Domain



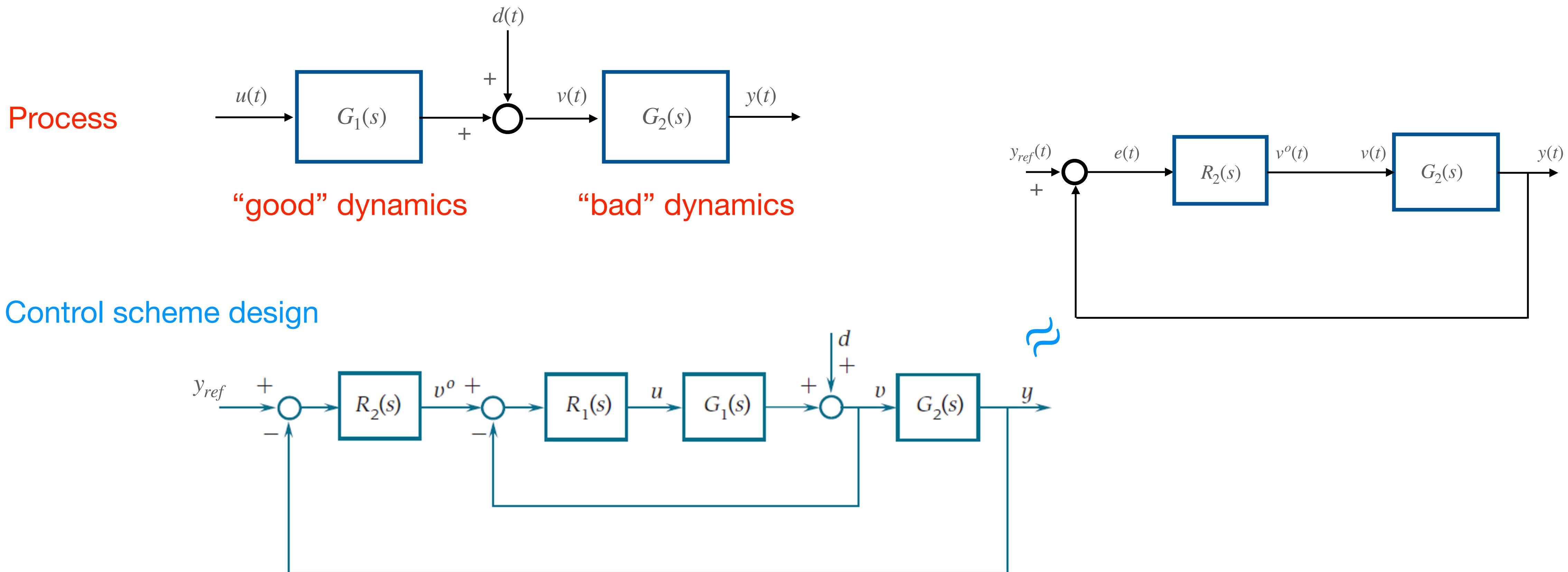
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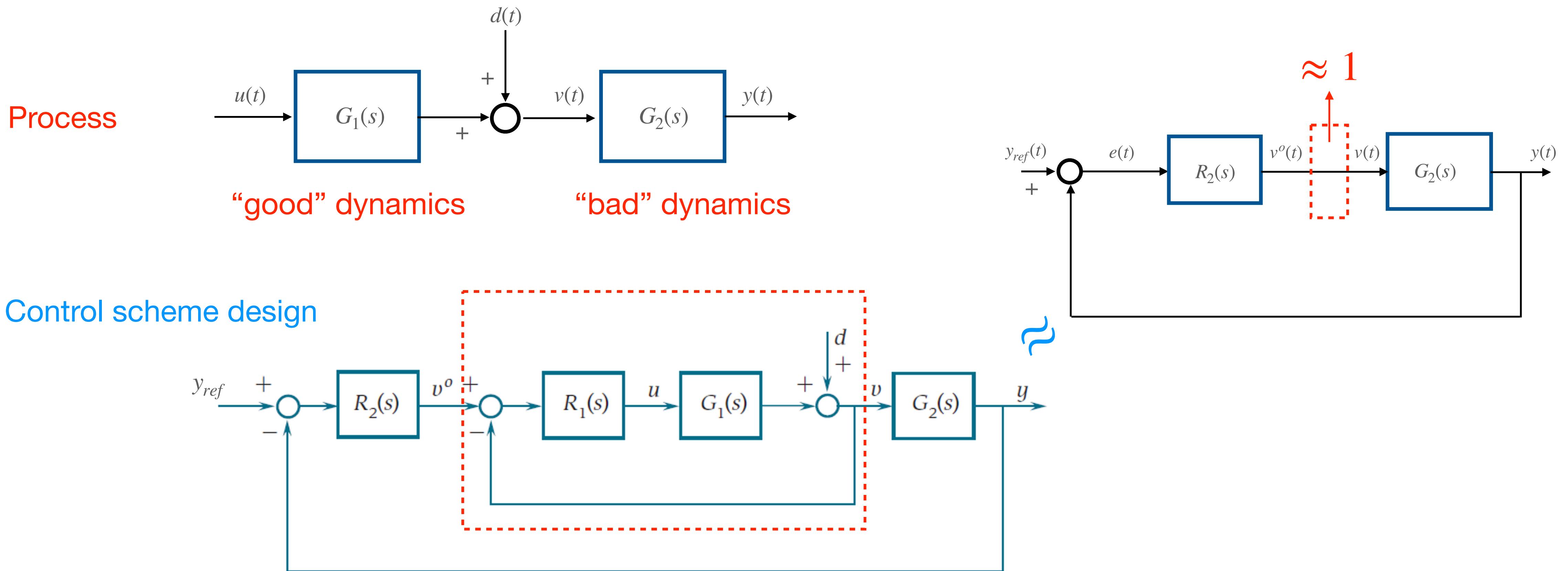
### Control scheme design



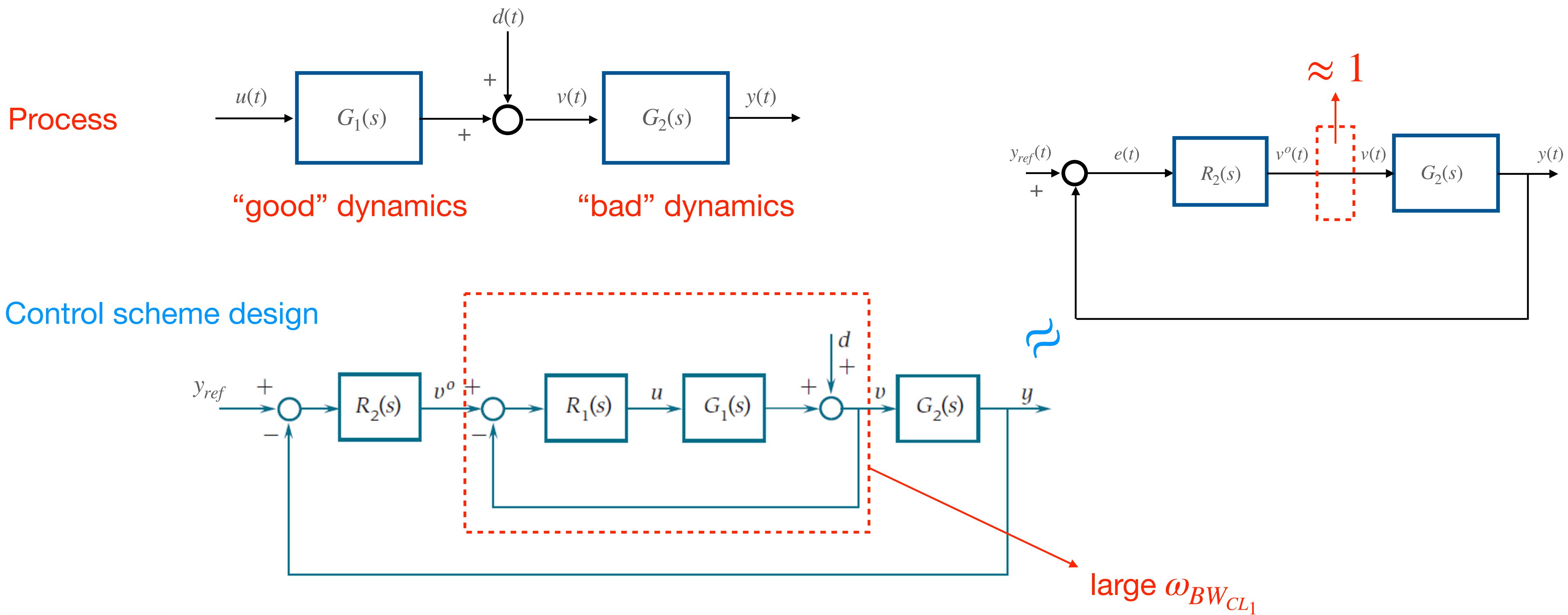
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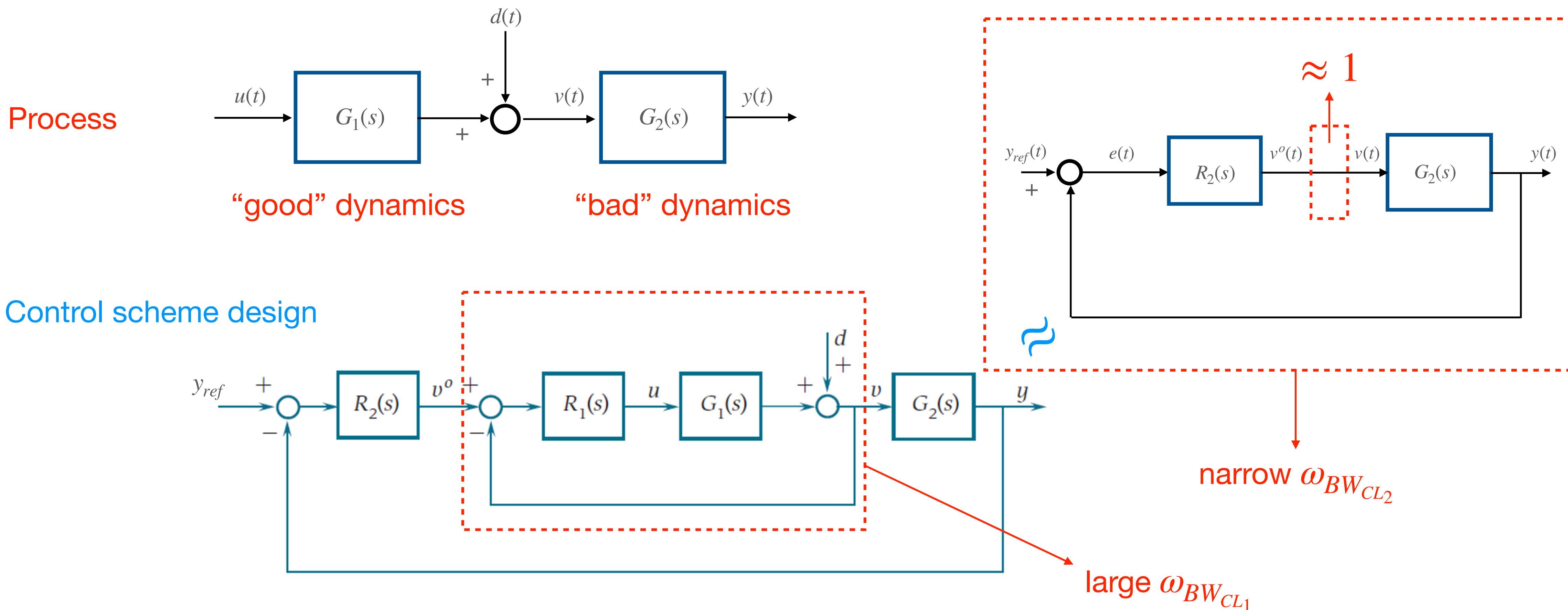
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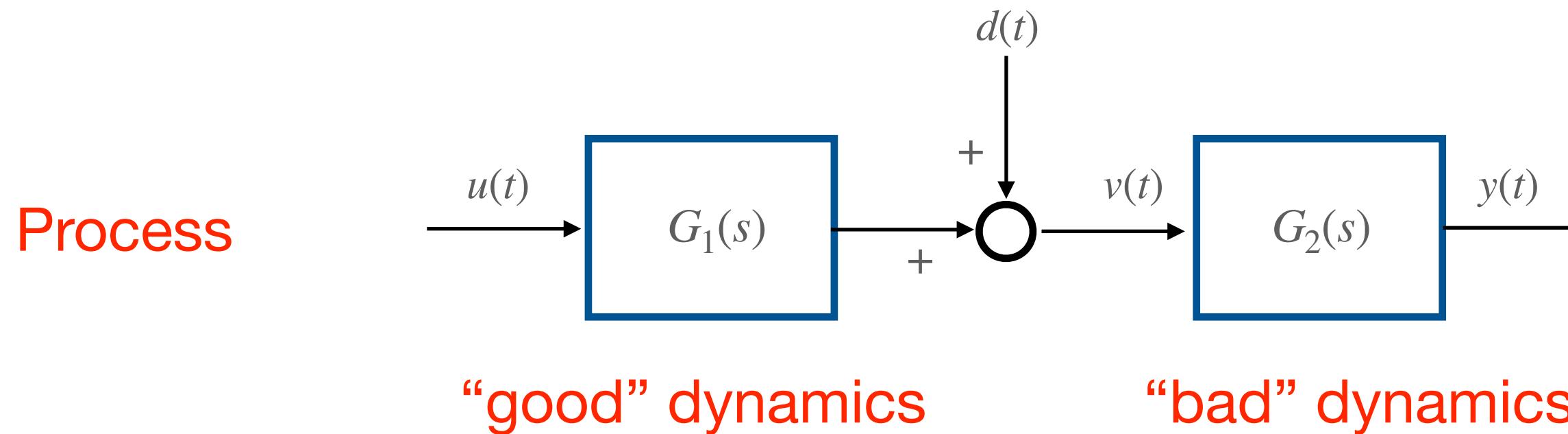


## Control Scheme with Decoupling in the Frequency Domain



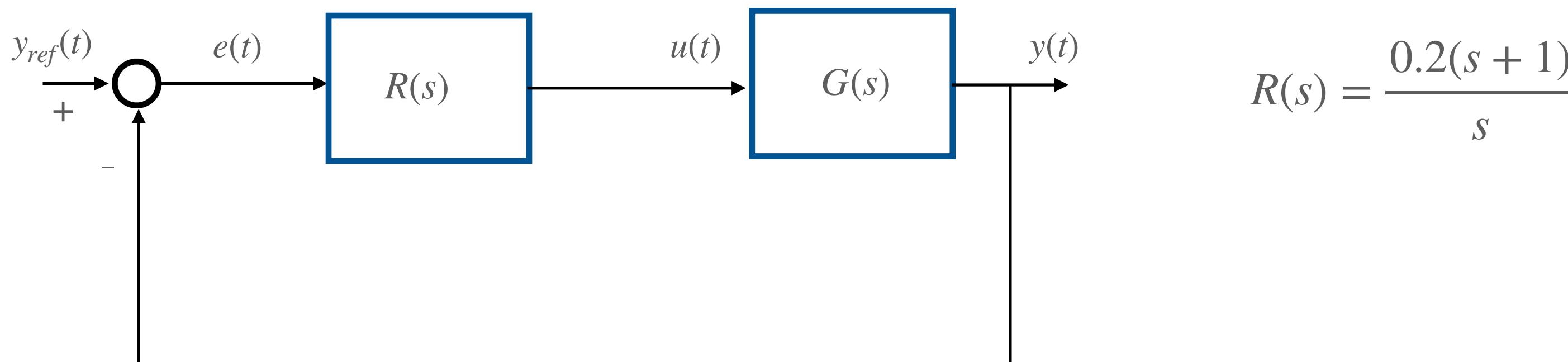
## Control Scheme with Decoupling in the Frequency Domain

Example (from the textbook):



$$G_1(s) = \frac{1}{1 + 0.005s}$$
$$G_2(s) = \frac{e^{-4s}}{(1 + s)^2}$$

Control scheme design: Solution 1

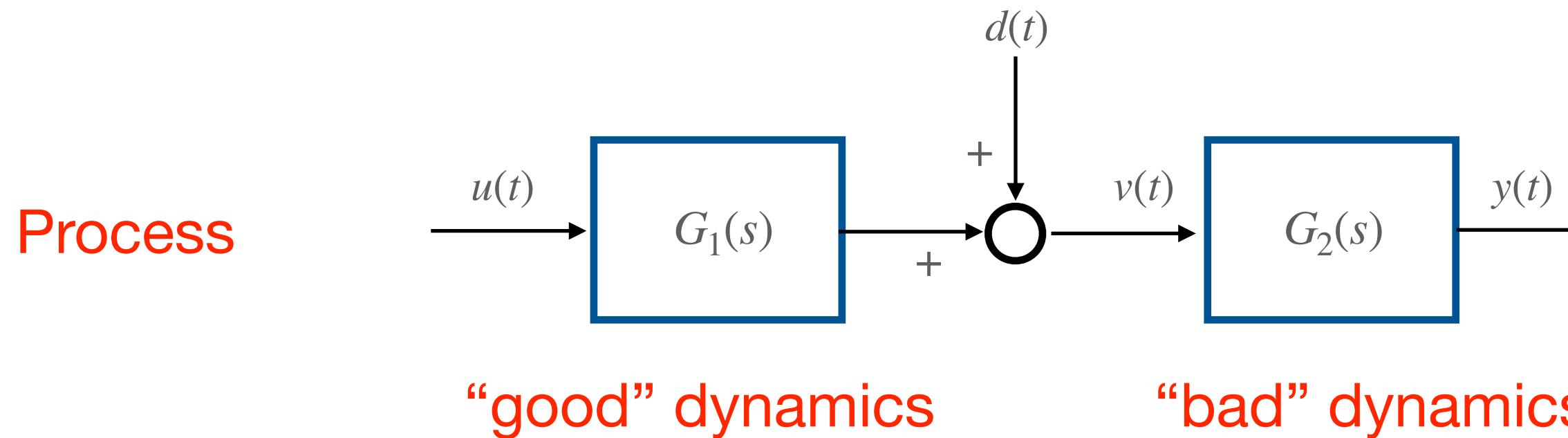


$$R(s) = \frac{0.2(s + 1)}{s}$$



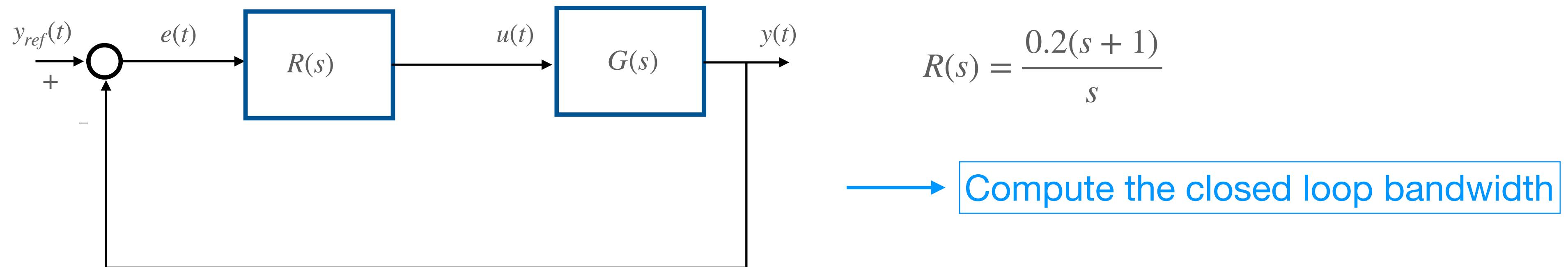
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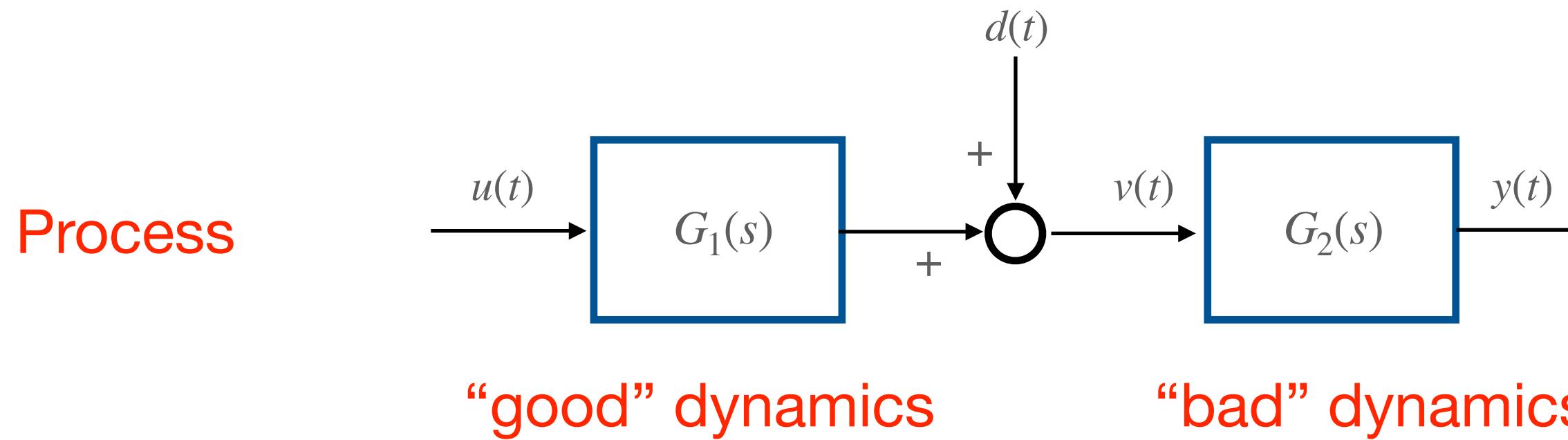
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Control scheme design: Solution 1



## Control Scheme with Decoupling in the Frequency Domain

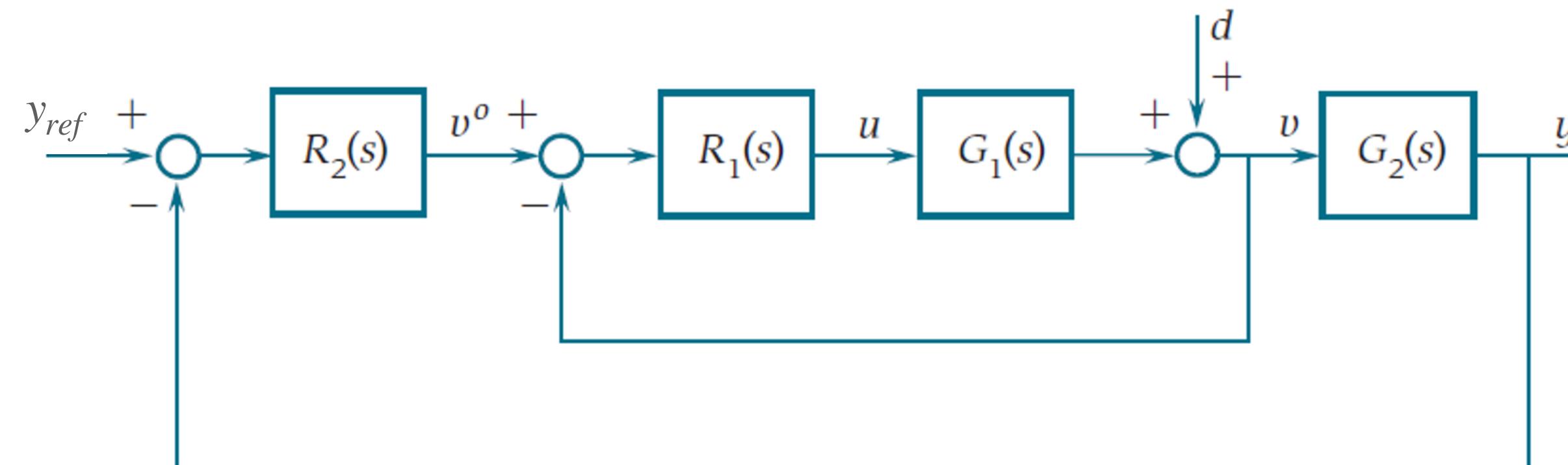
Example (from the textbook):



$$G_1(s) = \frac{1}{1 + 0.005s}$$

$$G_2(s) = \frac{e^{-4s}}{(1 + s)^2}$$

Control scheme design: Solution 2



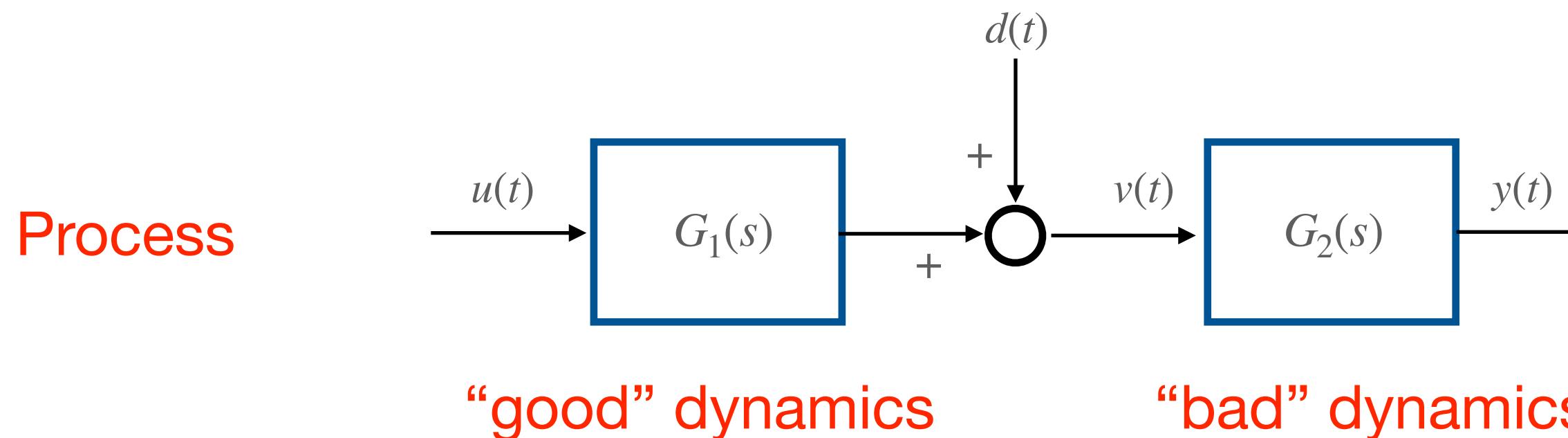
$$R_1(s) = 10$$

$$R_2(s) = \frac{0.2(s + 1)}{s}$$



## Control Scheme with Decoupling in the Frequency Domain

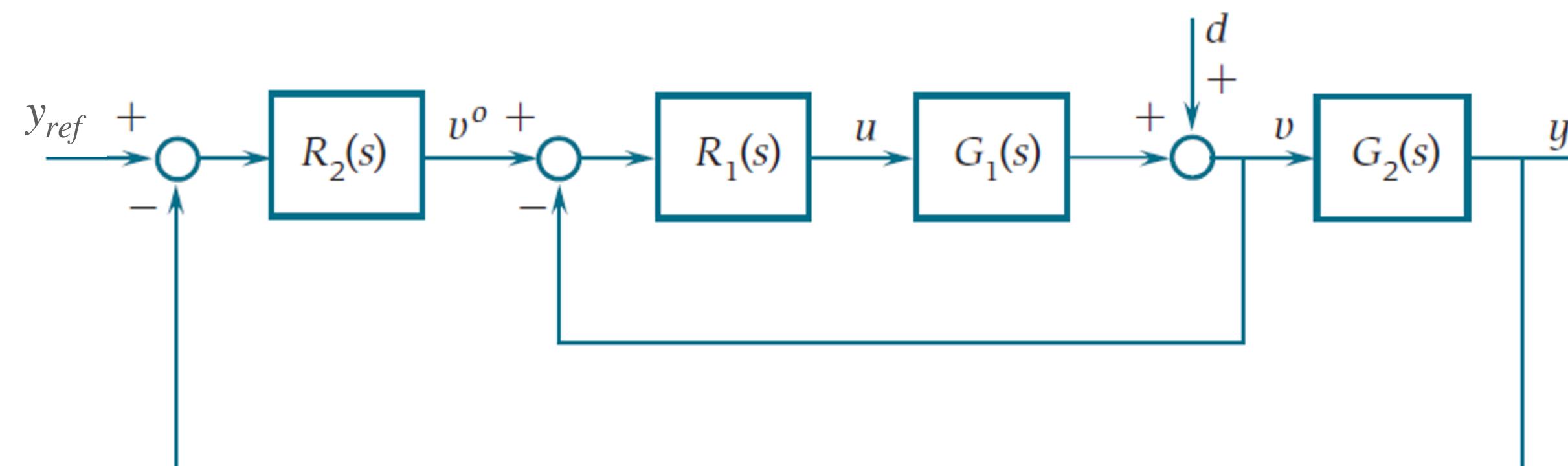
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Control scheme design: Solution 2



$$R_1(s) = 10$$

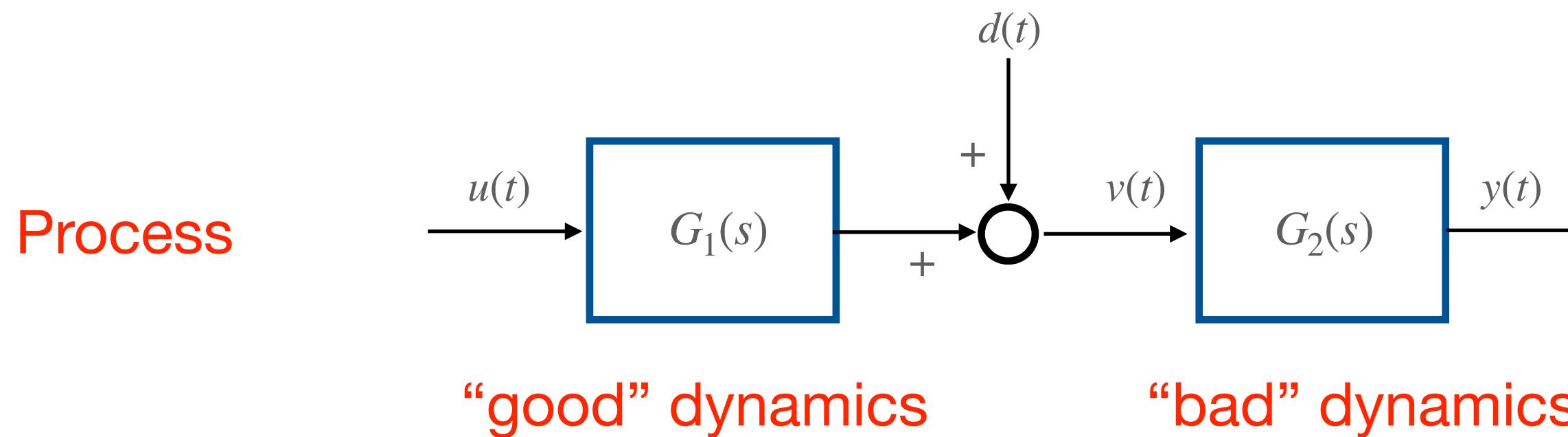
$$R_2(s) = \frac{0.2(s + 1)}{s}$$

Compute  $\omega_{BW_{CL1}}$  and  $\omega_{BW_{CL2}}$

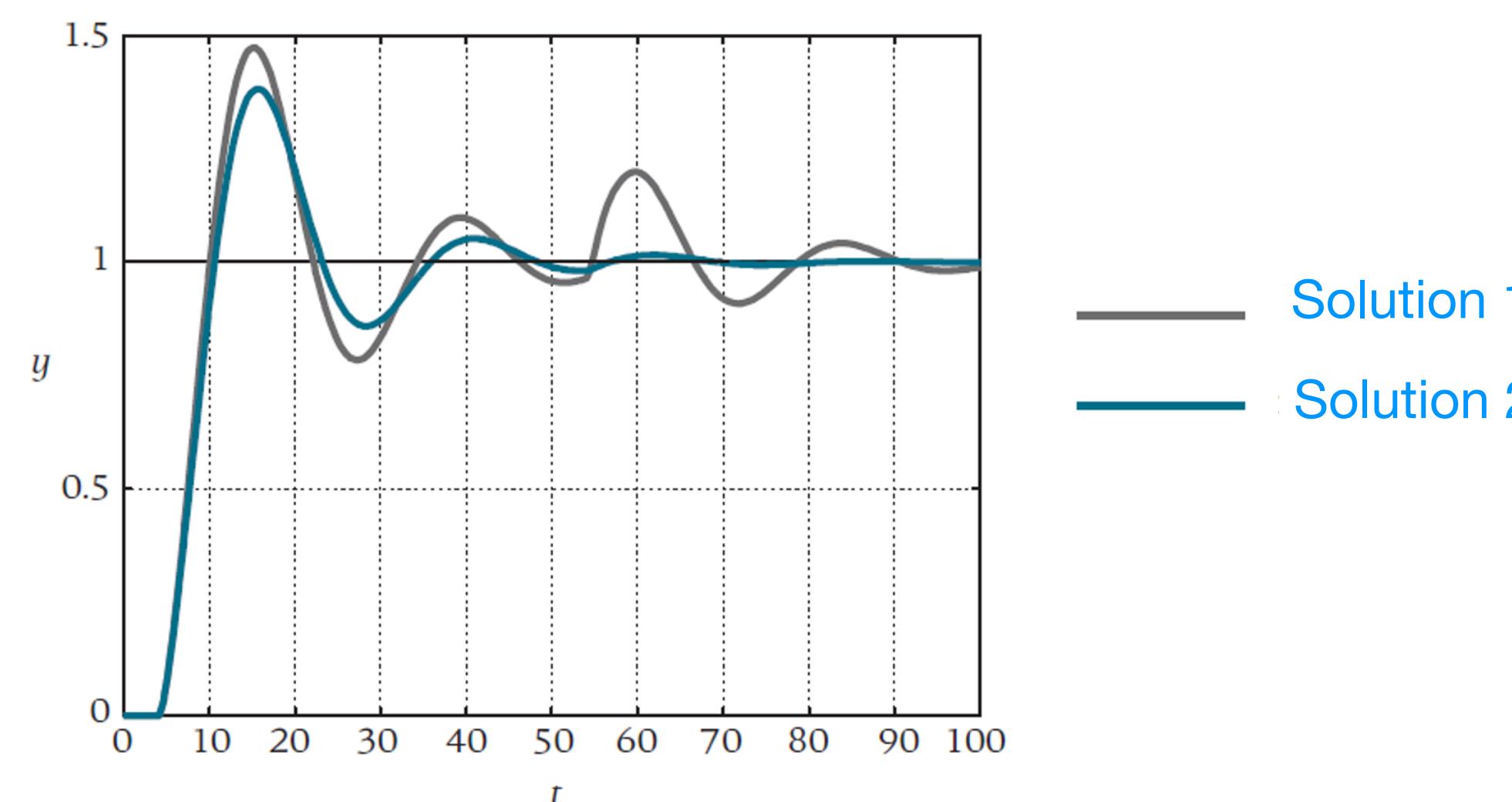


## Control Scheme with Decoupling in the Frequency Domain

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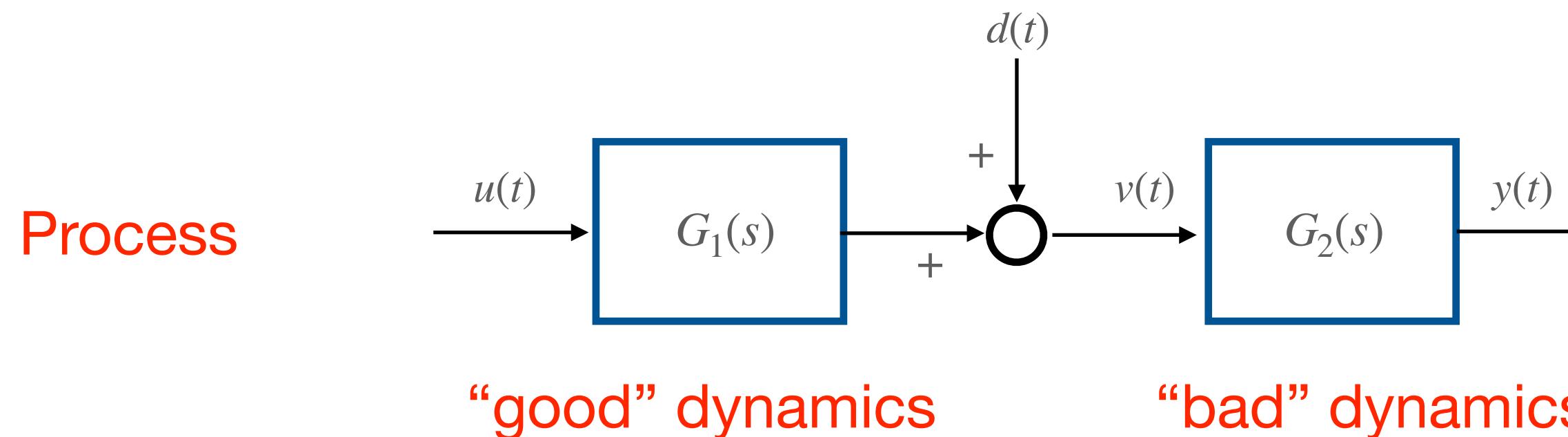


$$G_1(s) = \frac{1}{1 + 0.005s}$$
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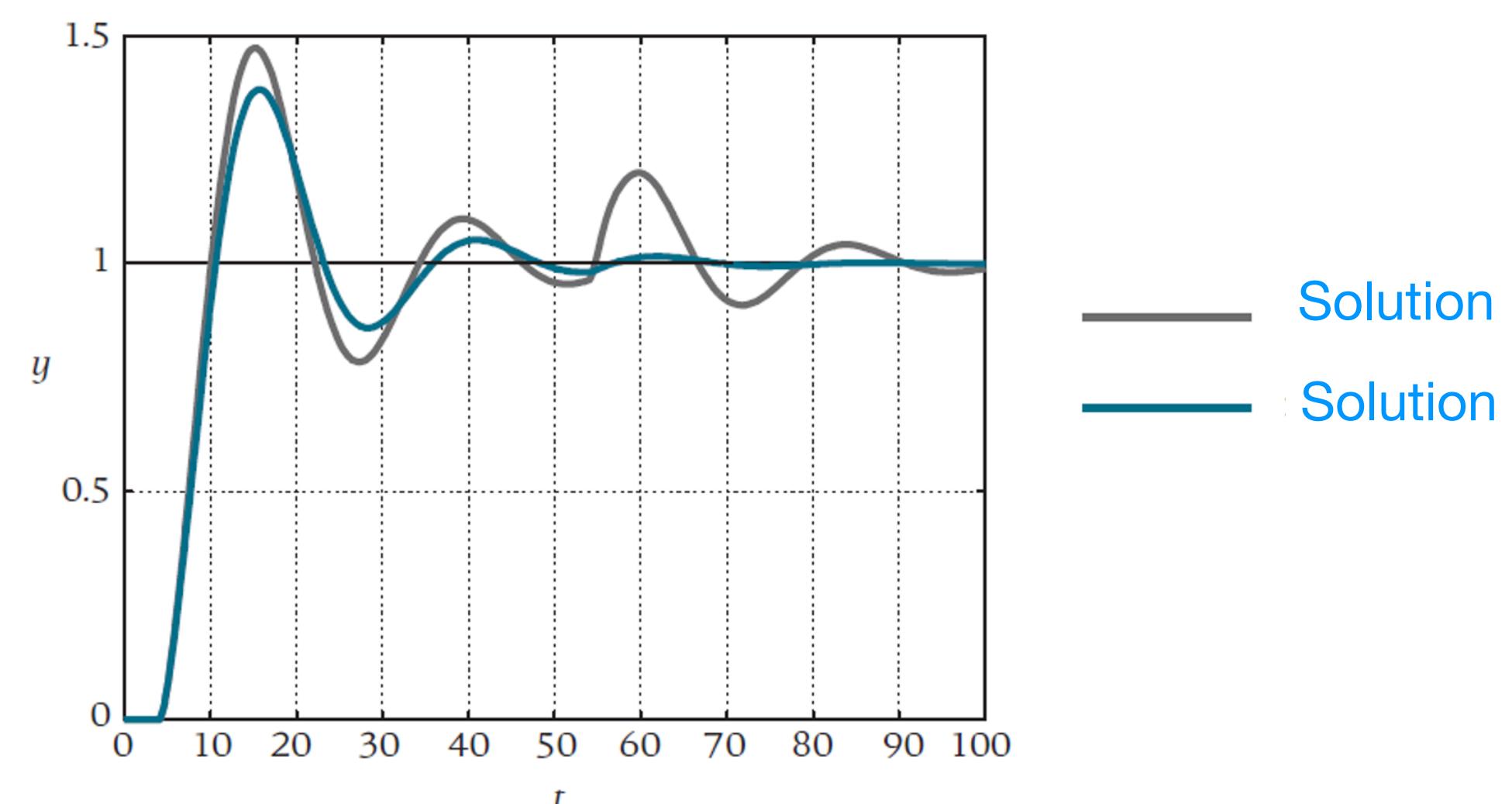


## Control Scheme with Decoupling in the Frequency Domain

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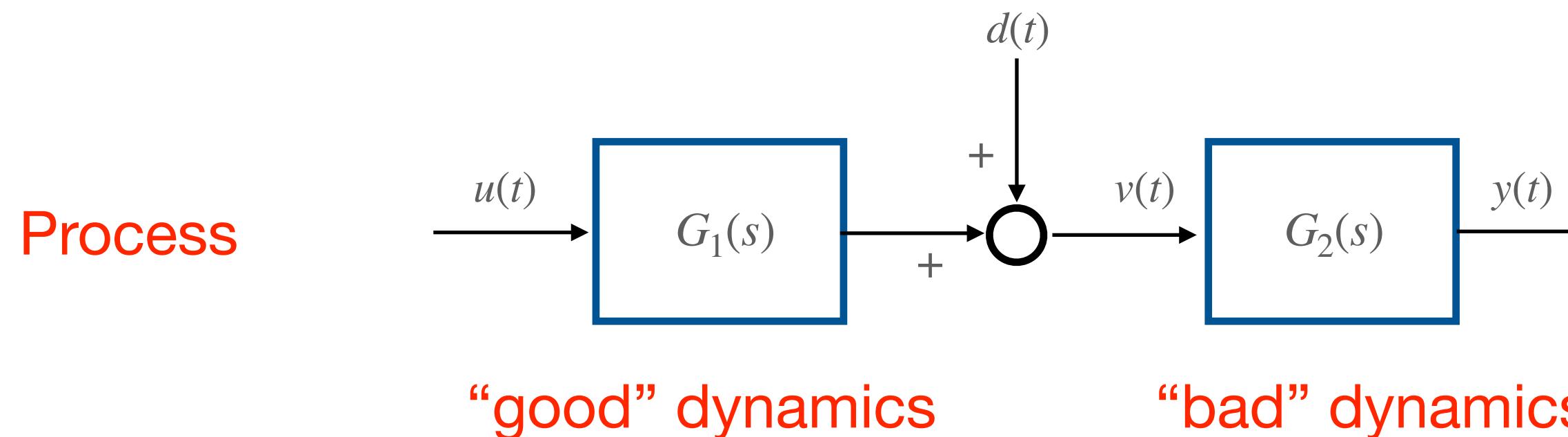


Exercise: formulate a similar example when  $G_2(s)$  is rational with a real zero in +1



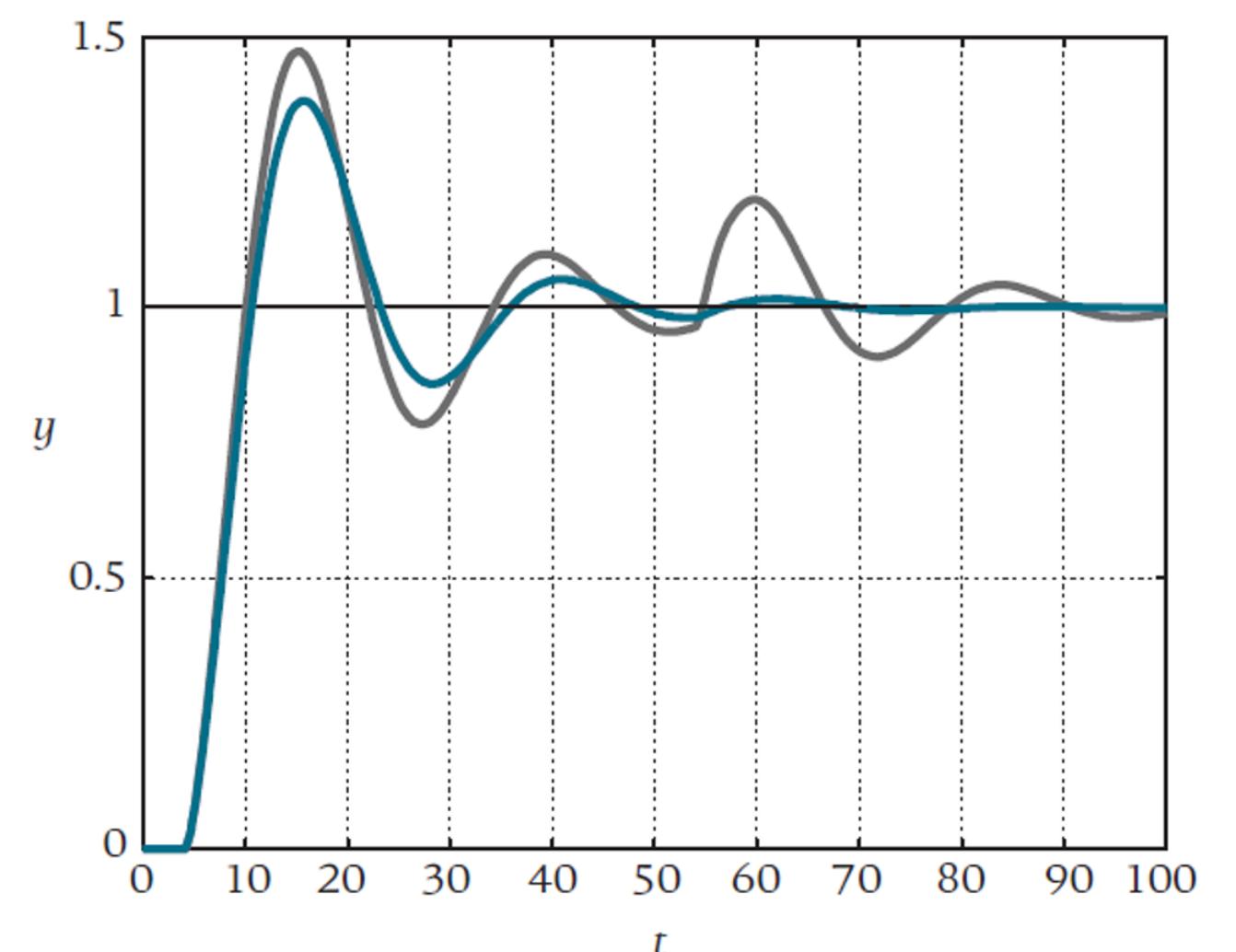
## Control Scheme with Decoupling in the Frequency Domain

**Example (from the textbook):**



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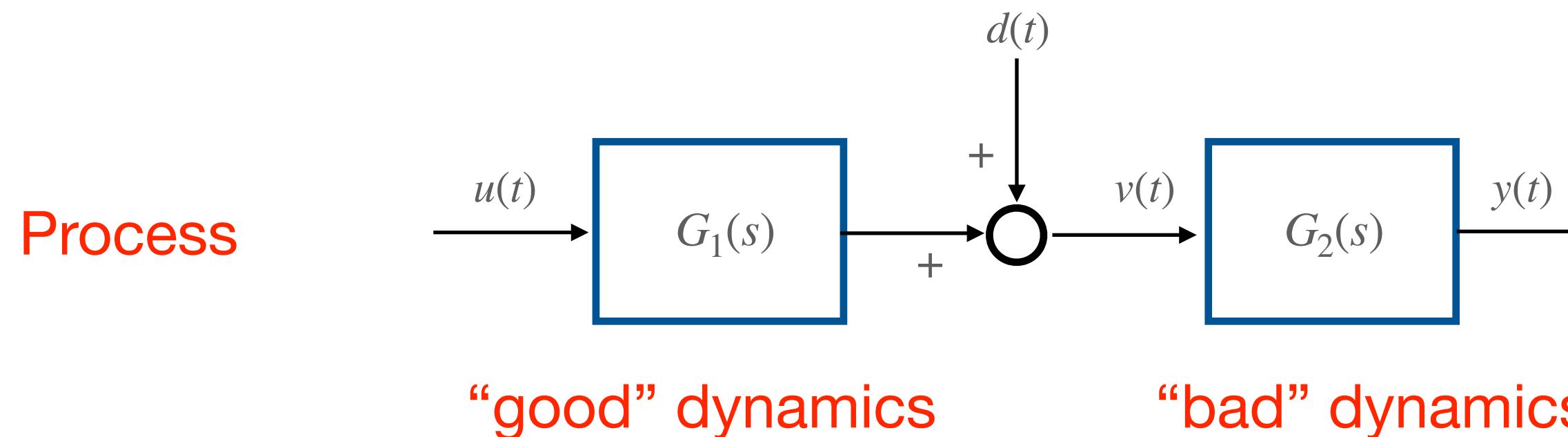
**Exercise:** formulate a similar example when  $G_2(s)$  is rational with a real zero in +1

Can you use the unitary feedback control scheme and design a single controller  $R(s)$ ? Try for different  $d(t)$



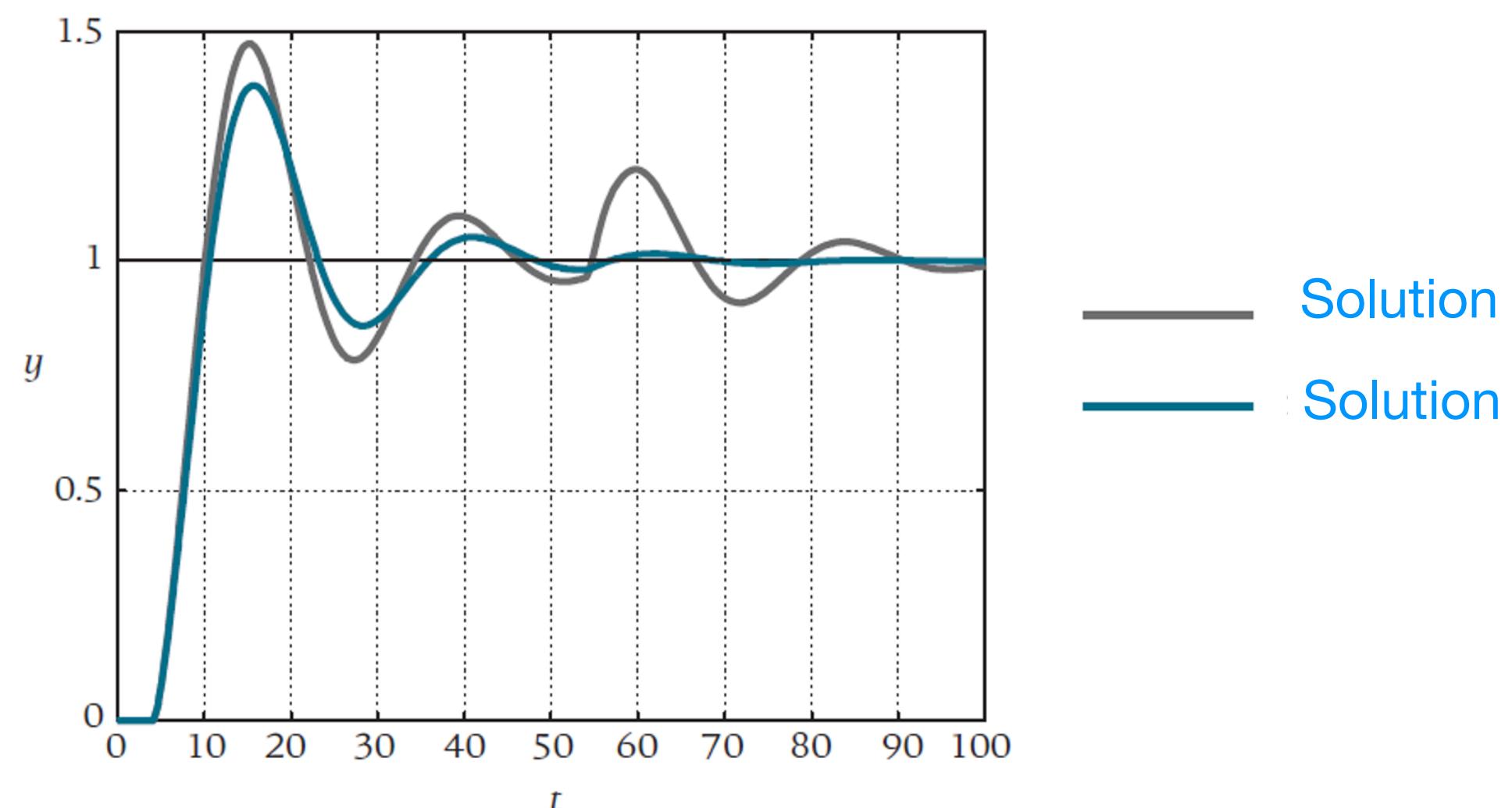
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Example (from the textbook):



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Exercise: formulate a similar example when  $G_2(s)$  is rational with a real zero in +1

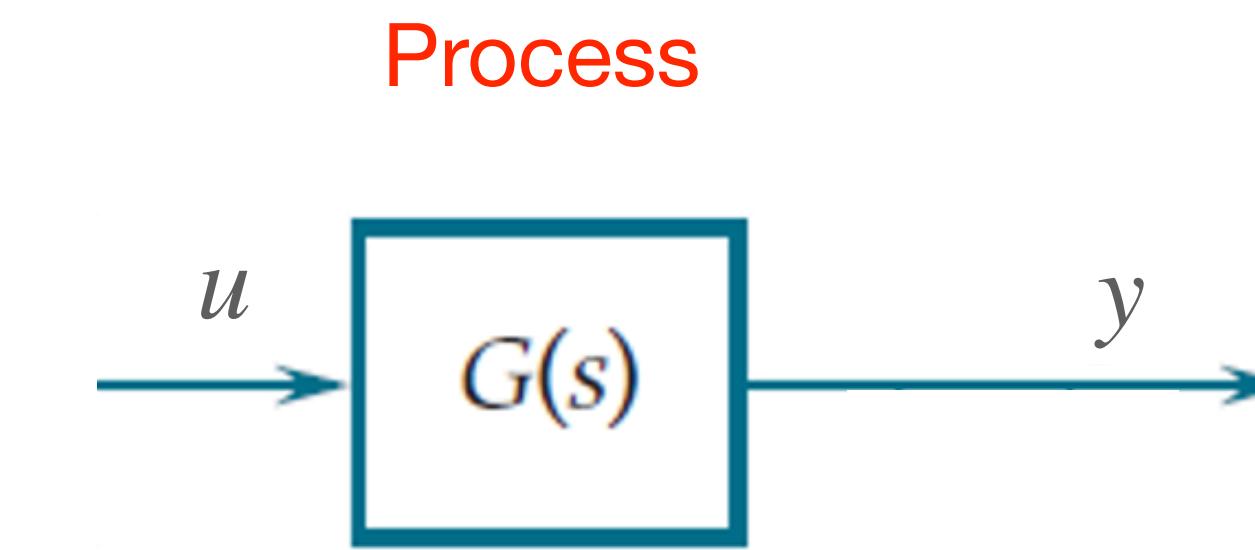
Use decoupling in the freq. dom.  
Find  $R_1(s)$  and  $R_2(s)$  and simulate the step response in MATLAB



## Control of Open-loop Unstable Systems

Assumption:

- $G(s)$  open-loop unstable



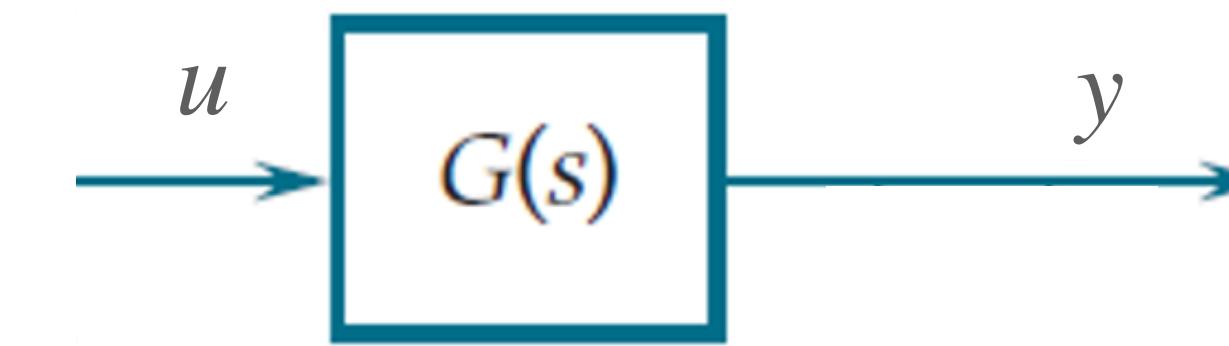
## Control of Open-loop Unstable Systems

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Process



Bode Criterion: It requires that

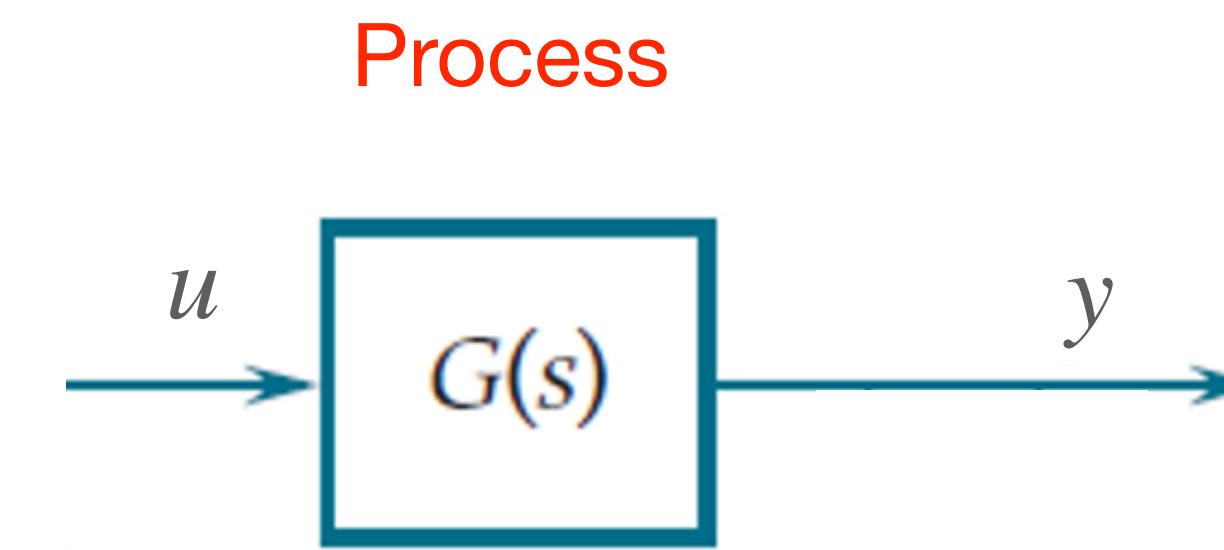
- $L(s)$  has no poles with positive real part
- the Bode plot of the magnitude of  $L(j\omega)$  crosses the 0 dB axis only once



## Control of Open-loop Unstable Systems

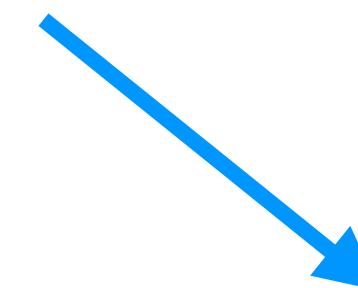
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It cannot be applied in this case!



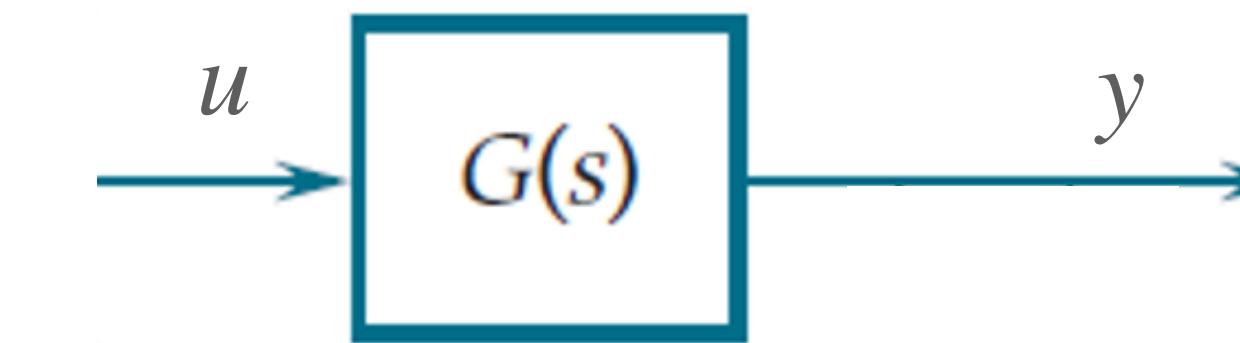
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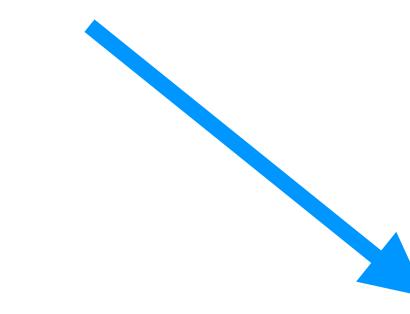


Process



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Alternatives:

- Nyquist Criterion
- Poles Assignment
- Root Locus



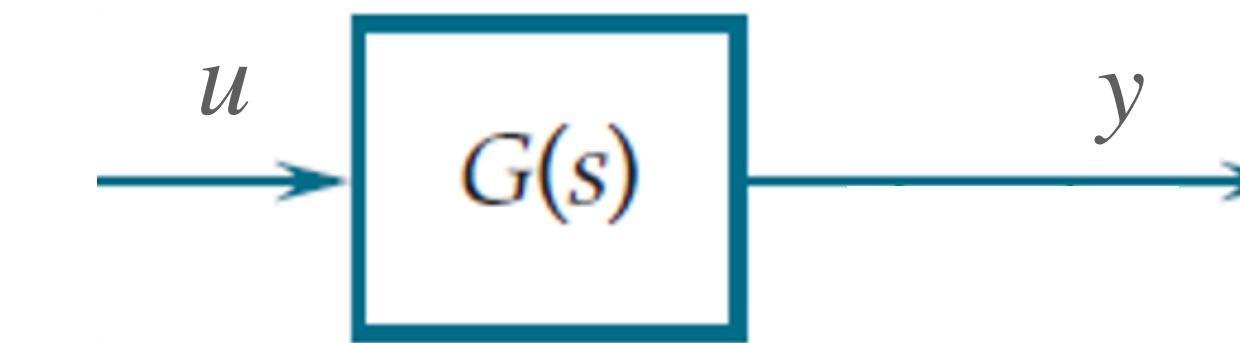
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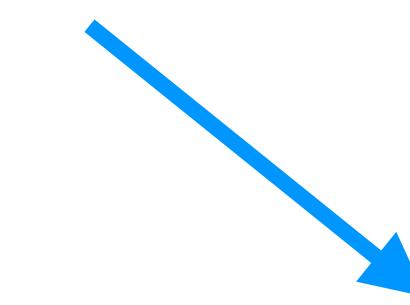


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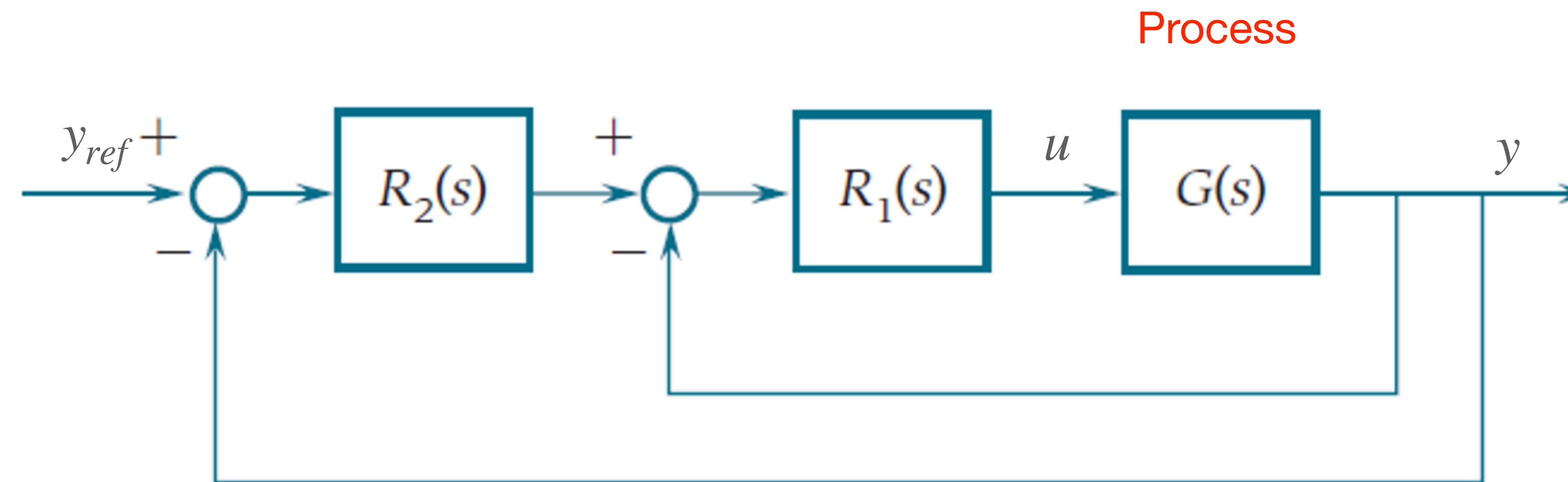


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## Control of Open-loop Unstable Systems



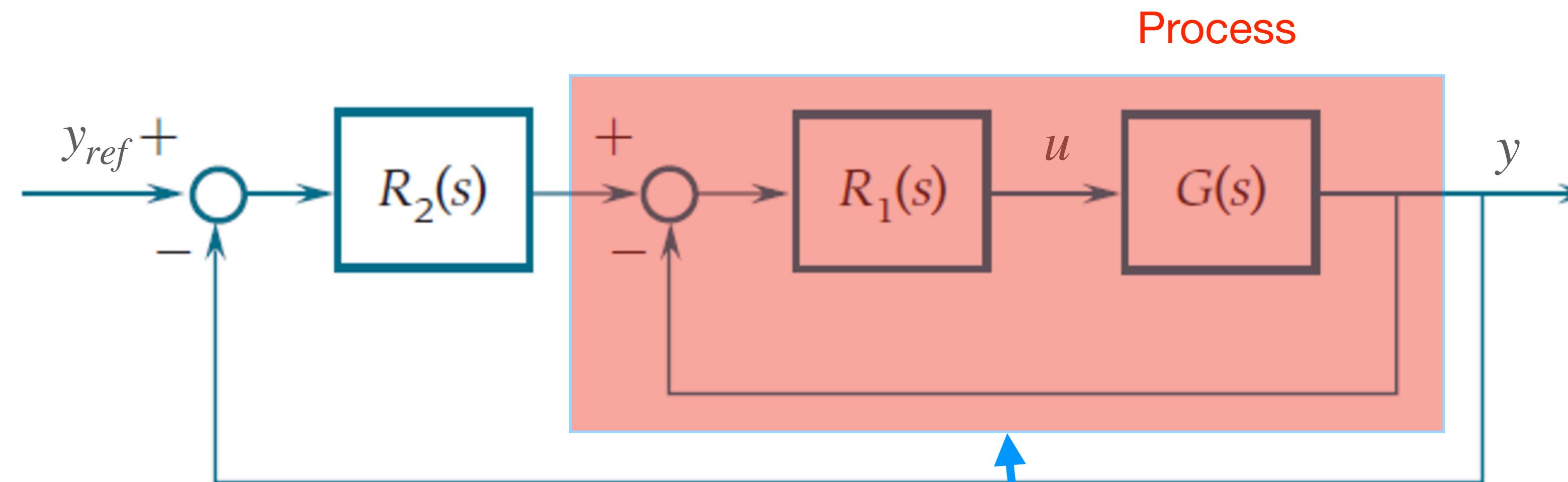
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## Control of Open-loop Unstable Systems

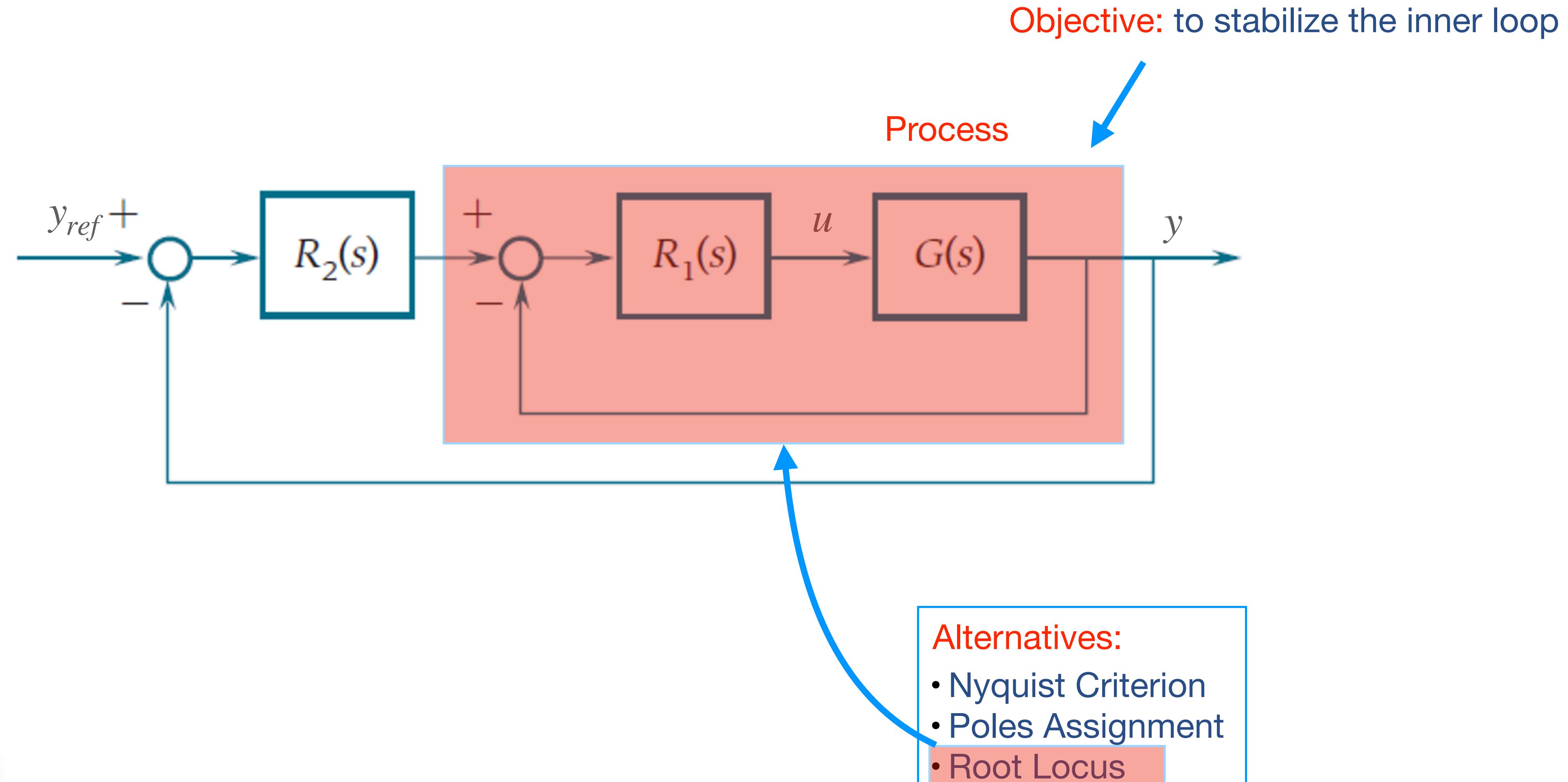


### Alternatives:

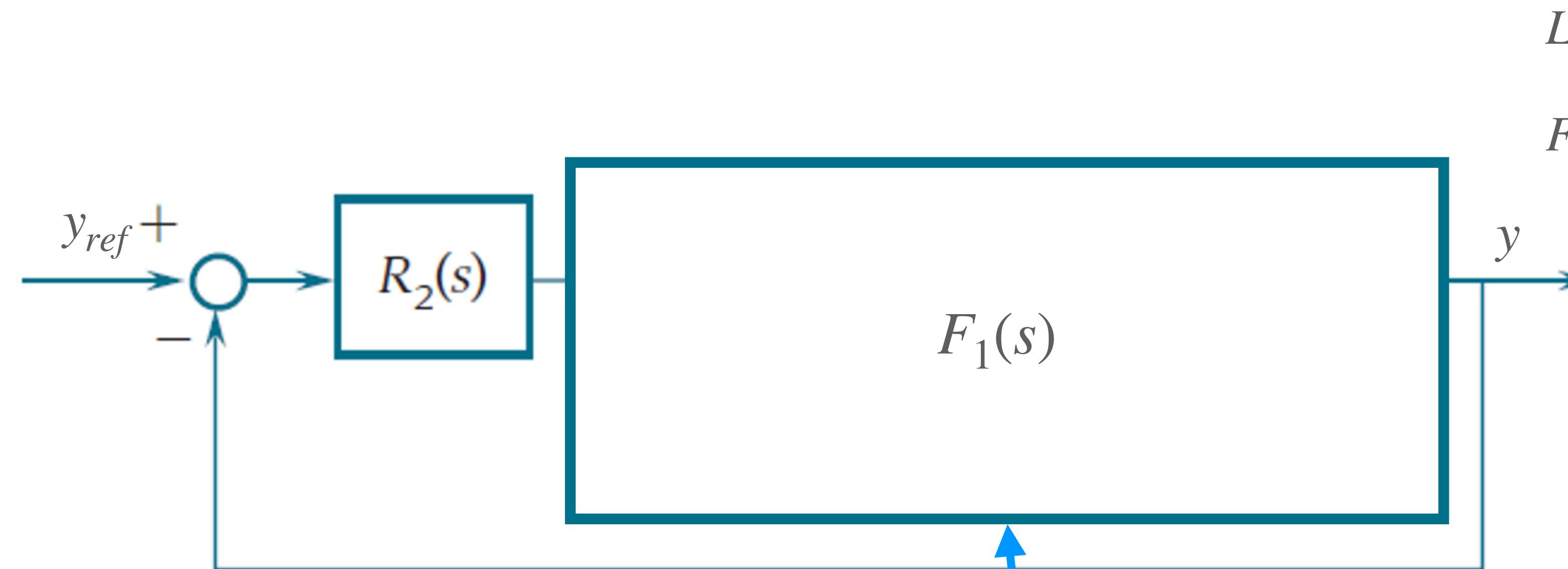
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## Control of Open-loop Unstable Systems



## Control of Open-loop Unstable Systems



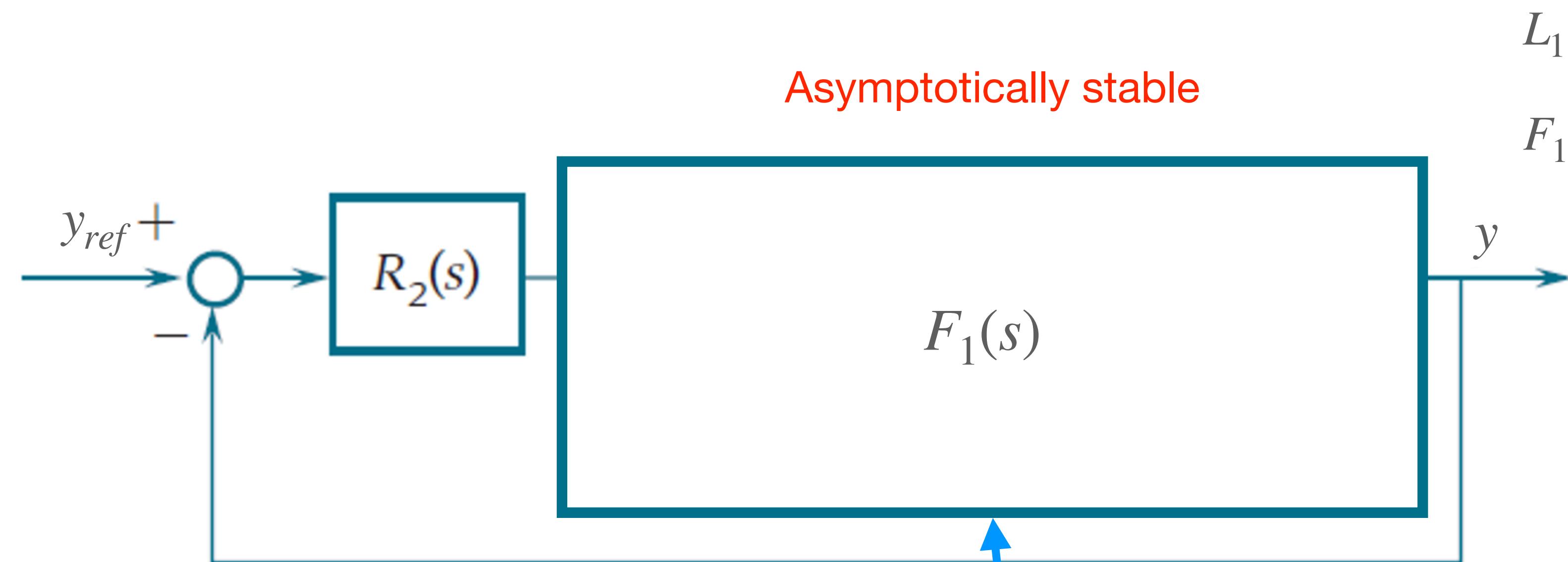
$$L_1(s) = R_1(s) G(s)$$

$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$

- Alternatives:
- Nyquist Criterion
  - Poles Assignment
  - Root Locus



## Control of Open-loop Unstable Systems



$$L_1(s) = R_1(s) G(s)$$

$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$

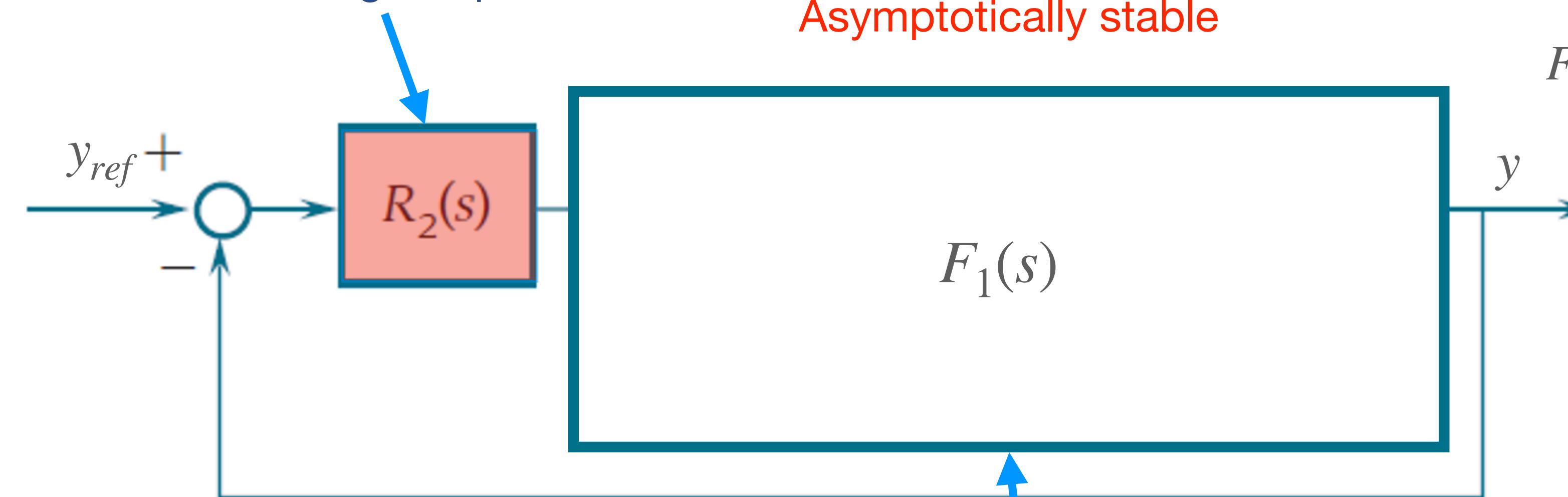
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## Control of Open-loop Unstable Systems

**Objective:** to stabilize the outer loop & satisfy the design requirements

Asymptotically stable



$$L_1(s) = R_1(s) G(s)$$

$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$

Alternatives:

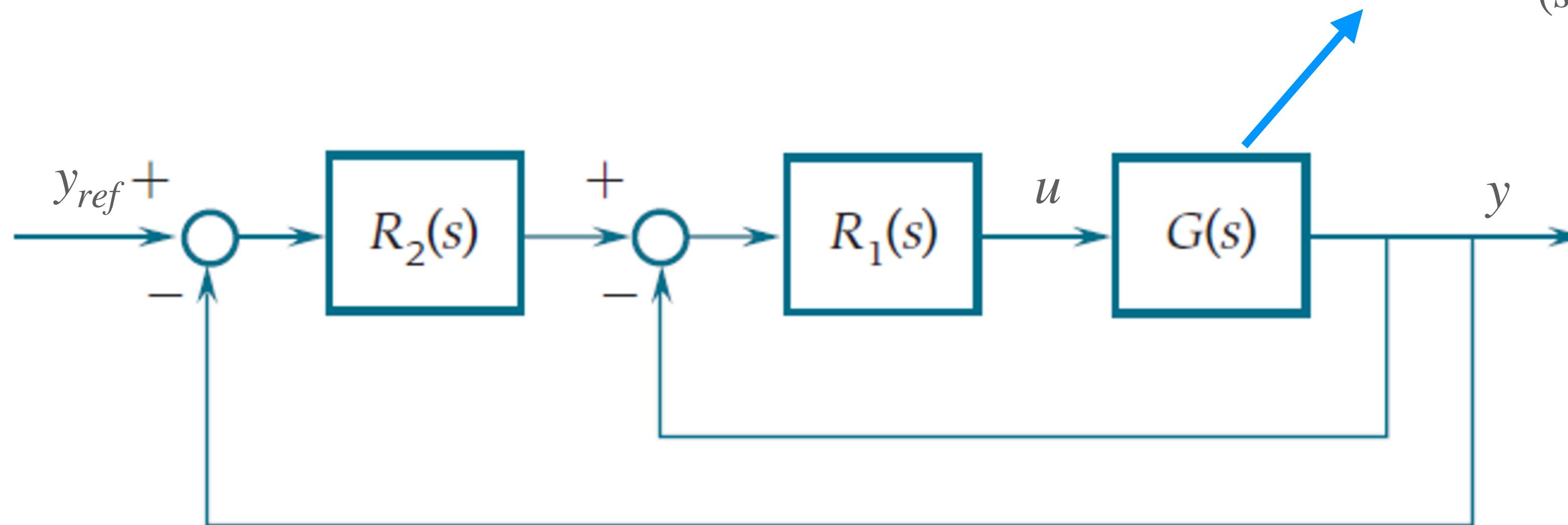
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## Control of Open-loop Unstable Systems

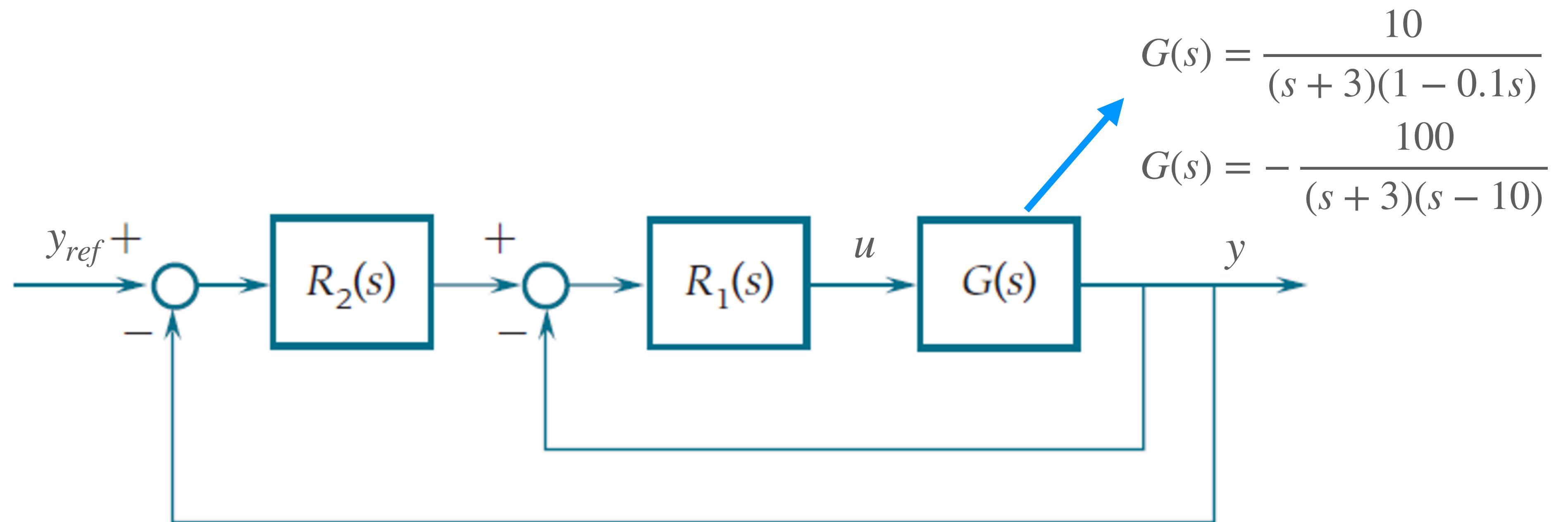
Example:

$$G(s) = \frac{10}{(s + 3)(1 - 0.1s)}$$



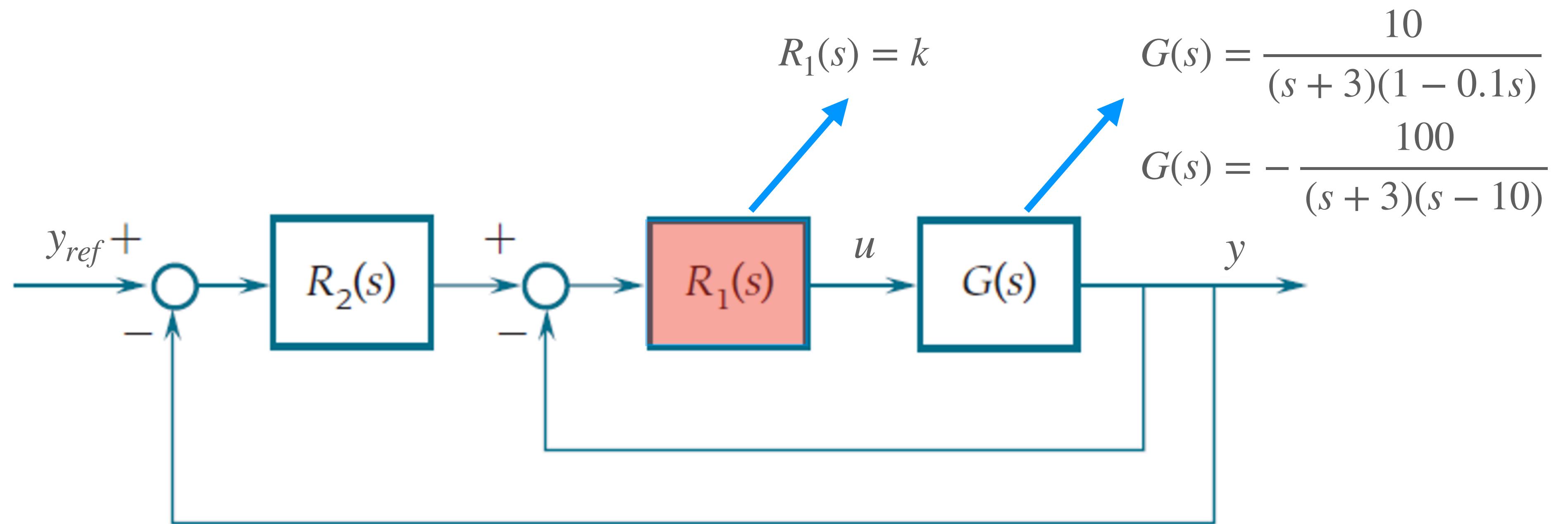
## Control of Open-loop Unstable Systems

Example:



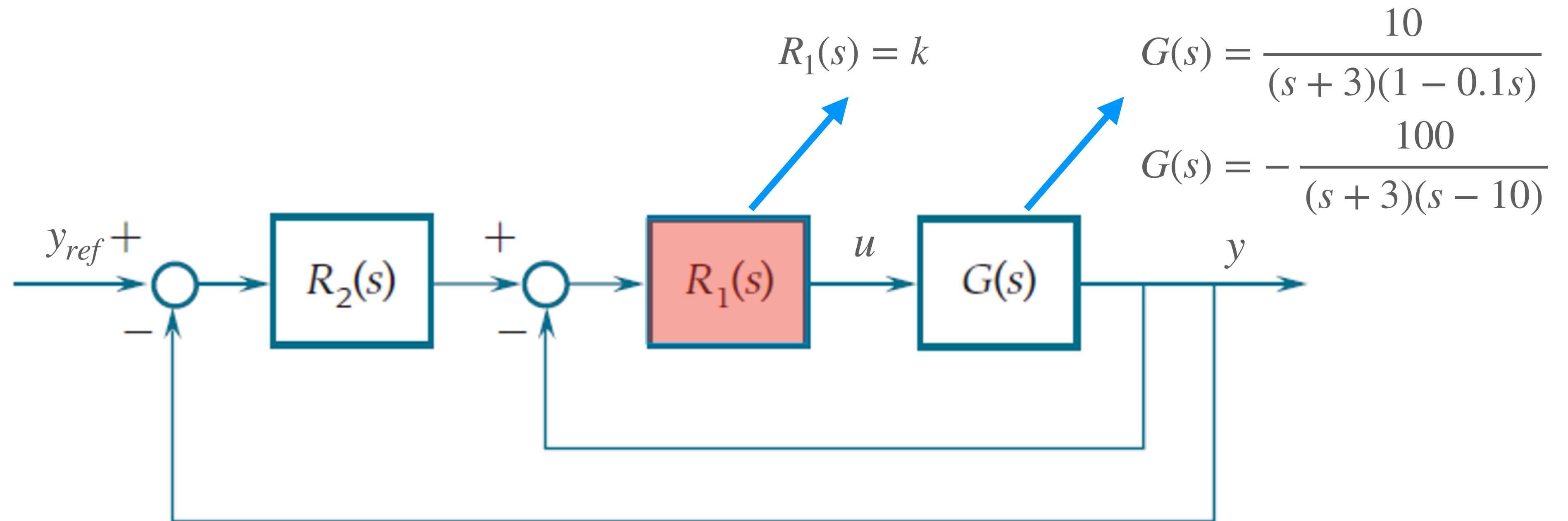
## Control of Open-loop Unstable Systems

Example:



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Example:



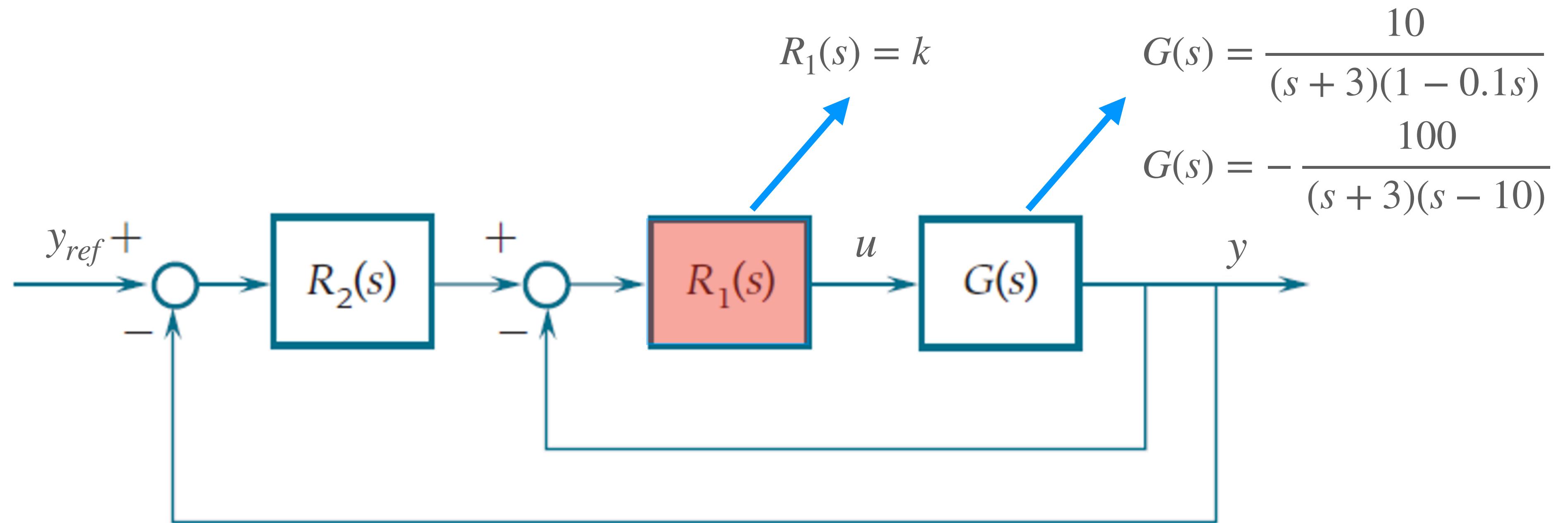
$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s + 3)(s - 10)}$$

$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$



## Control of Open-loop Unstable Systems

Example:



$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s+3)(s-10)} = \frac{K}{(s+3)(s-10)}$$

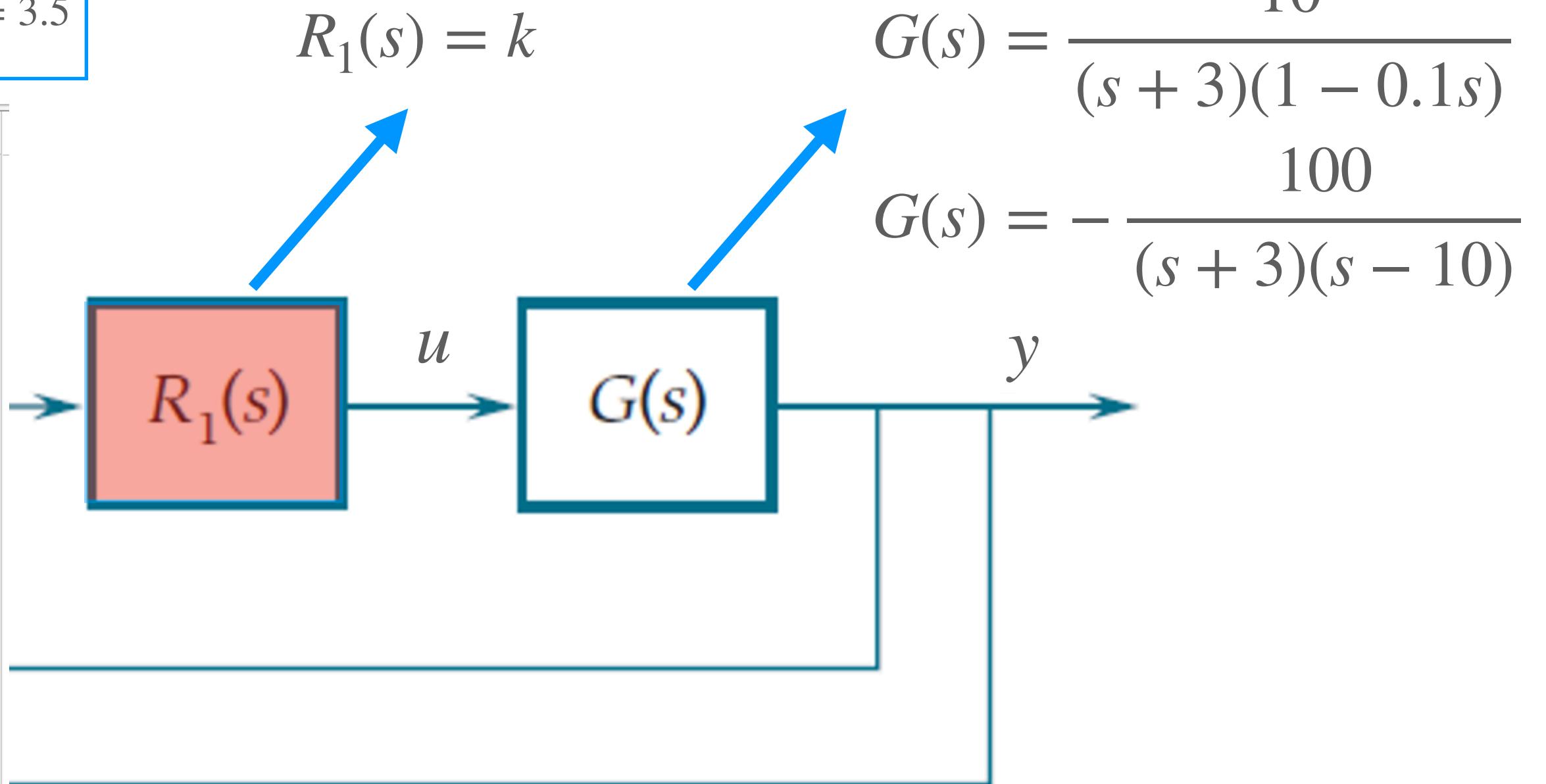
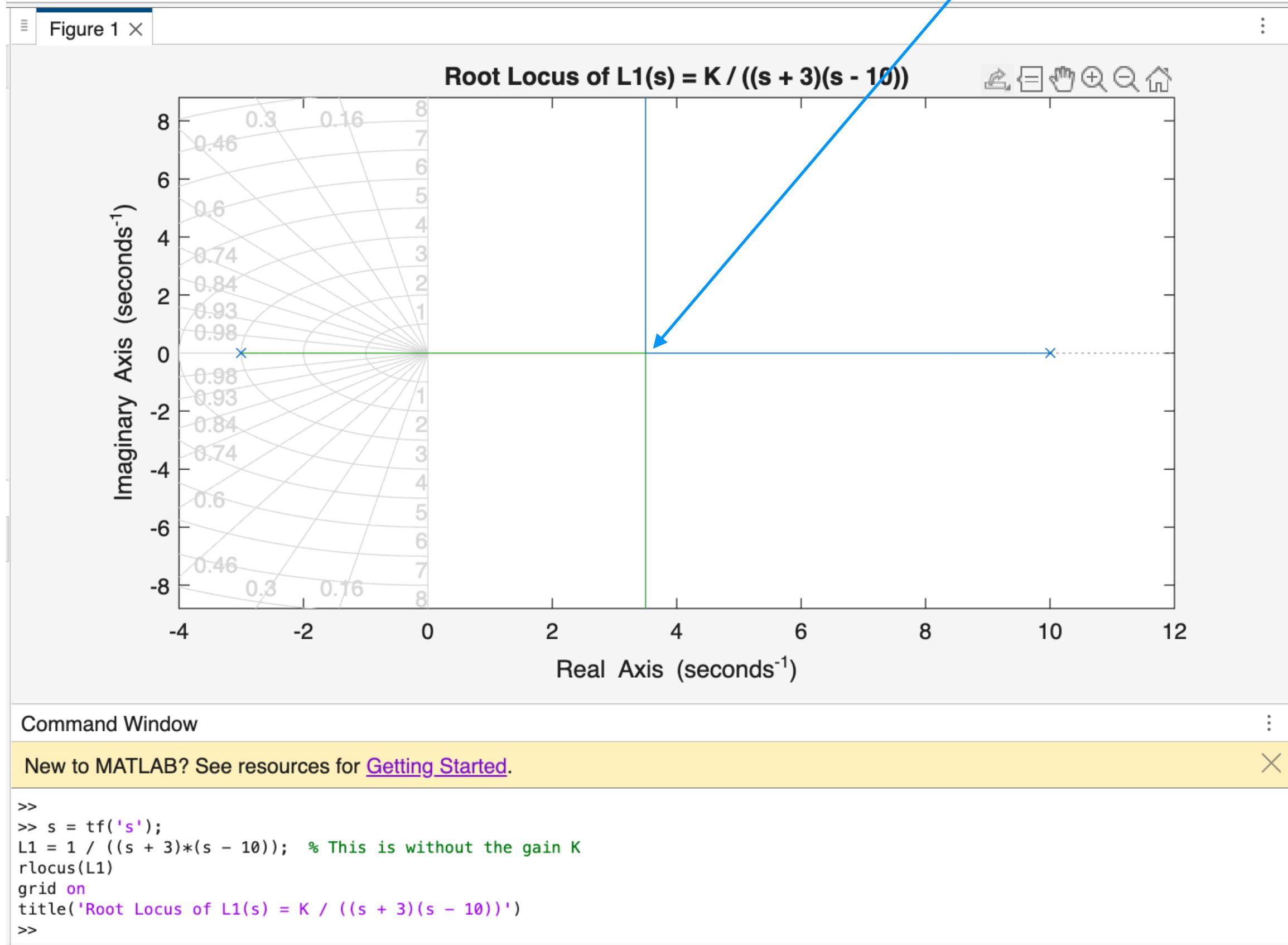
$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$



# Control of Open-loop Unstable Systems

**Example:**

$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-3 + 10}{2 - 0} = \frac{7}{2} = 3.5$$



$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s + 3)(s - 10)} = \frac{K}{(s + 3)(s - 10)}$$

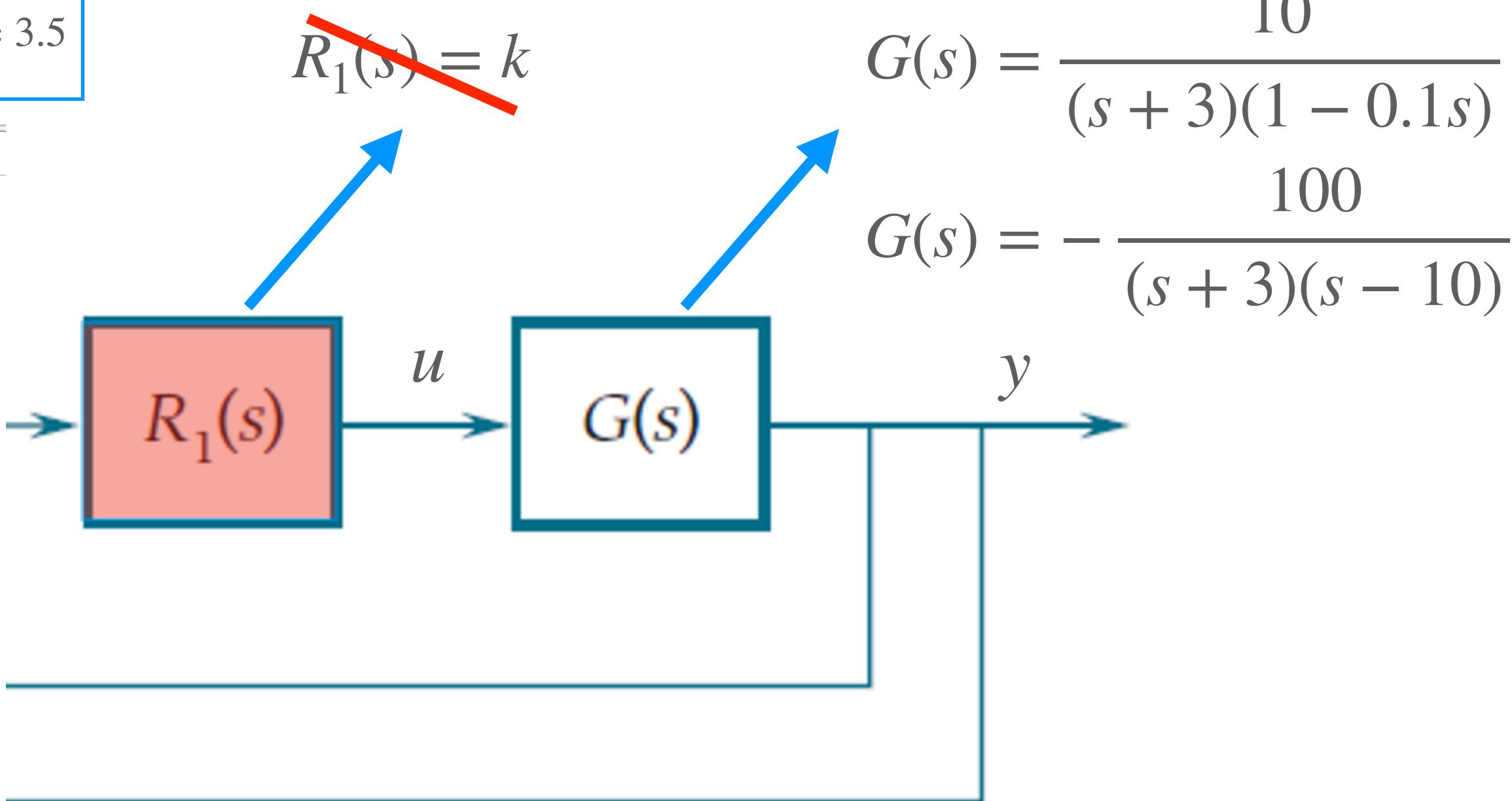
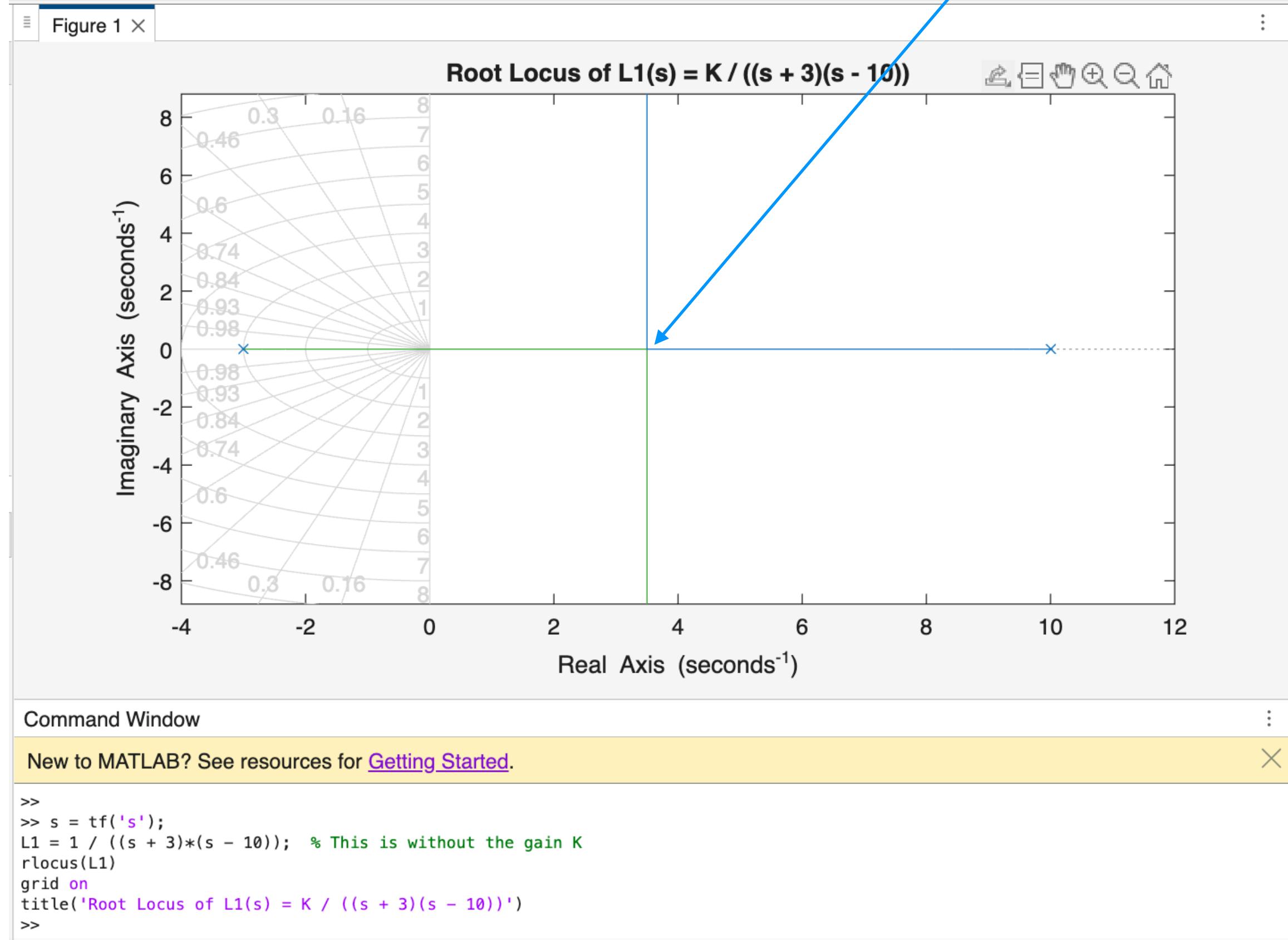
$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$



# Control of Open-loop Unstable Systems

**Example:**

$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-3 + 10}{2 - 0} = \frac{7}{2} = 3.5$$



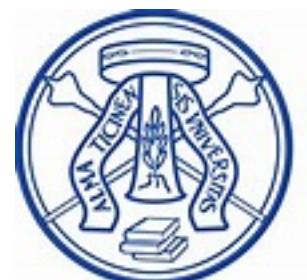
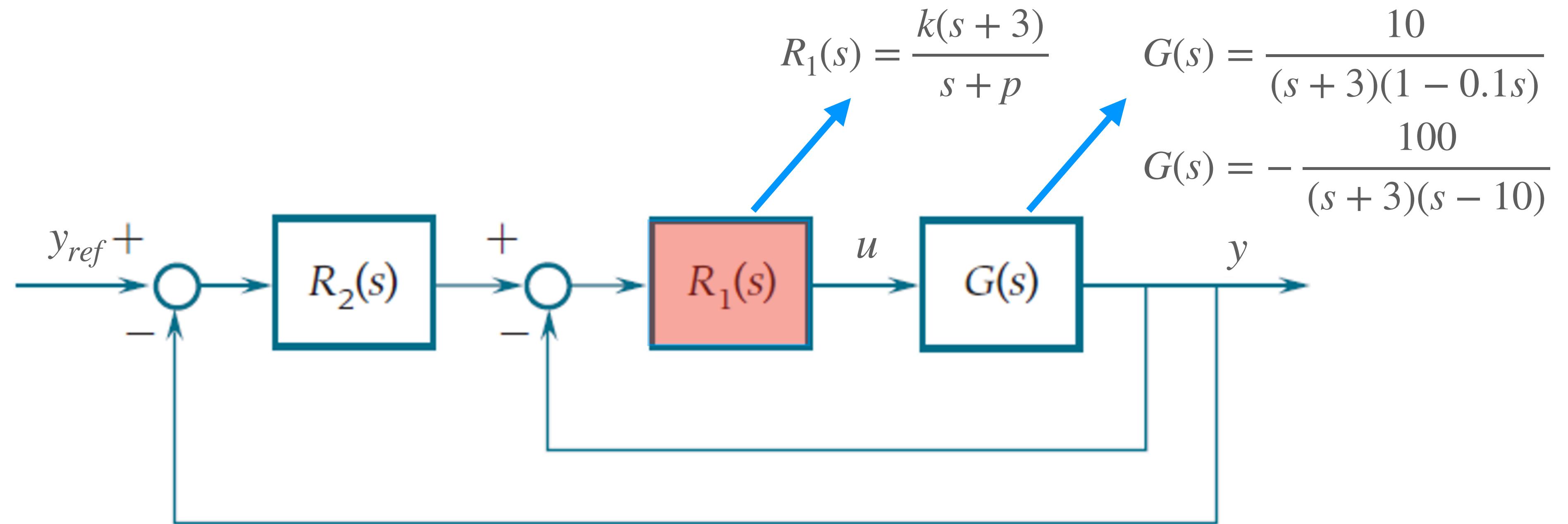
$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s+3)(s-10)} = \frac{K}{(s+3)(s-10)}$$

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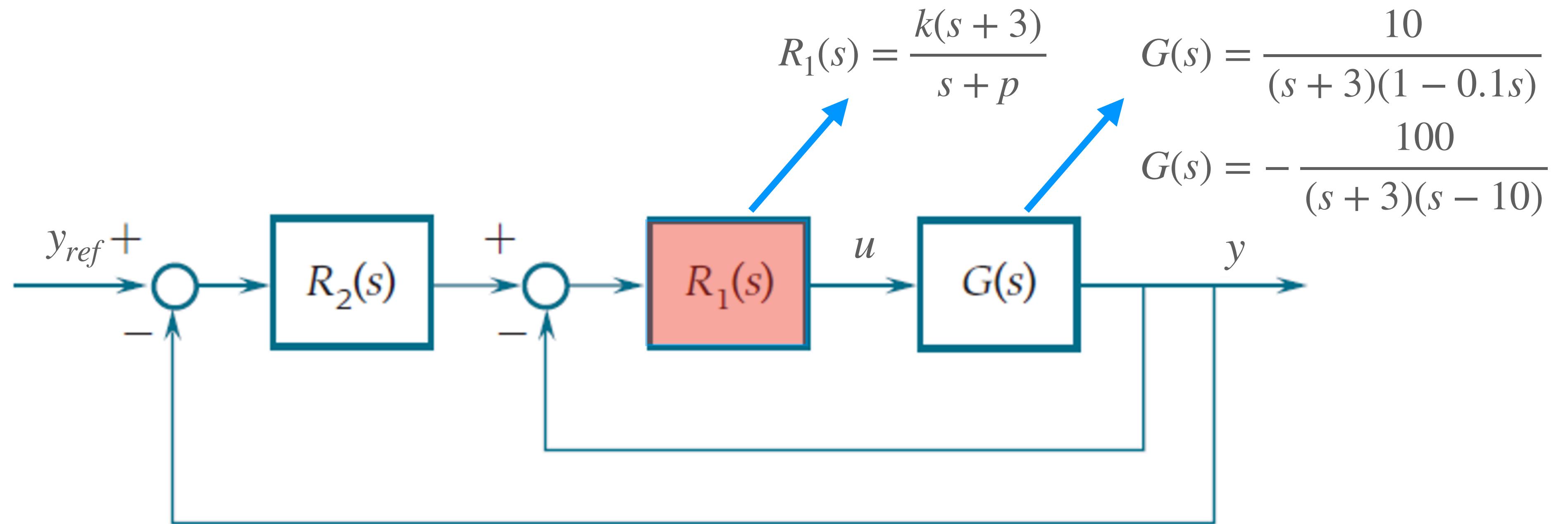
## Control of Open-loop Unstable Systems

Example:



## Control of Open-loop Unstable Systems

Example:



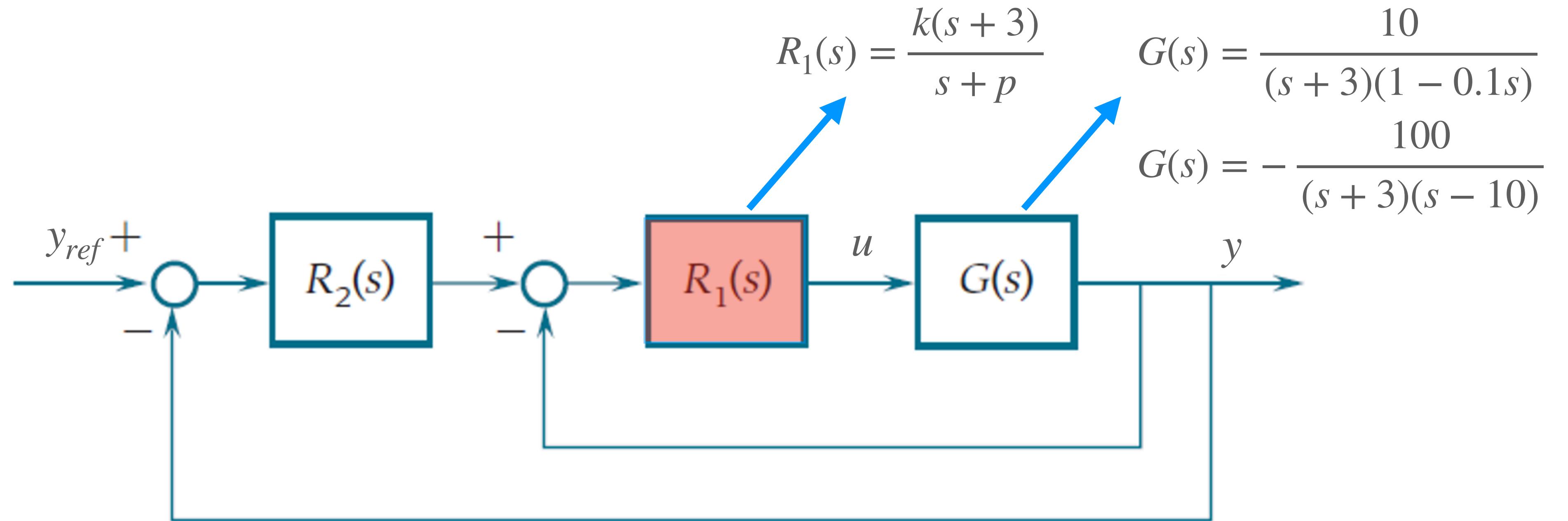
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## Control of Open-loop Unstable Systems

**Example:**



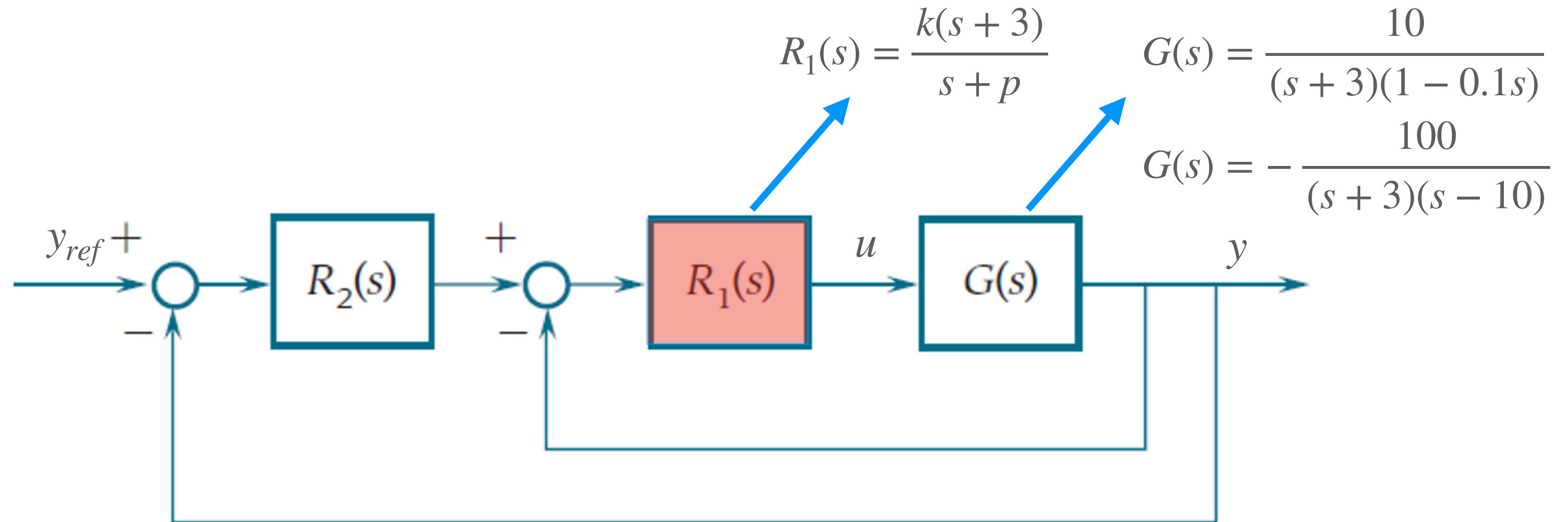
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?



## Control of Open-loop Unstable Systems

**Example:**



$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n-m} \quad \sigma_a = \frac{(-p) + 10}{2 - 0} = \frac{10 - p}{2} =$$

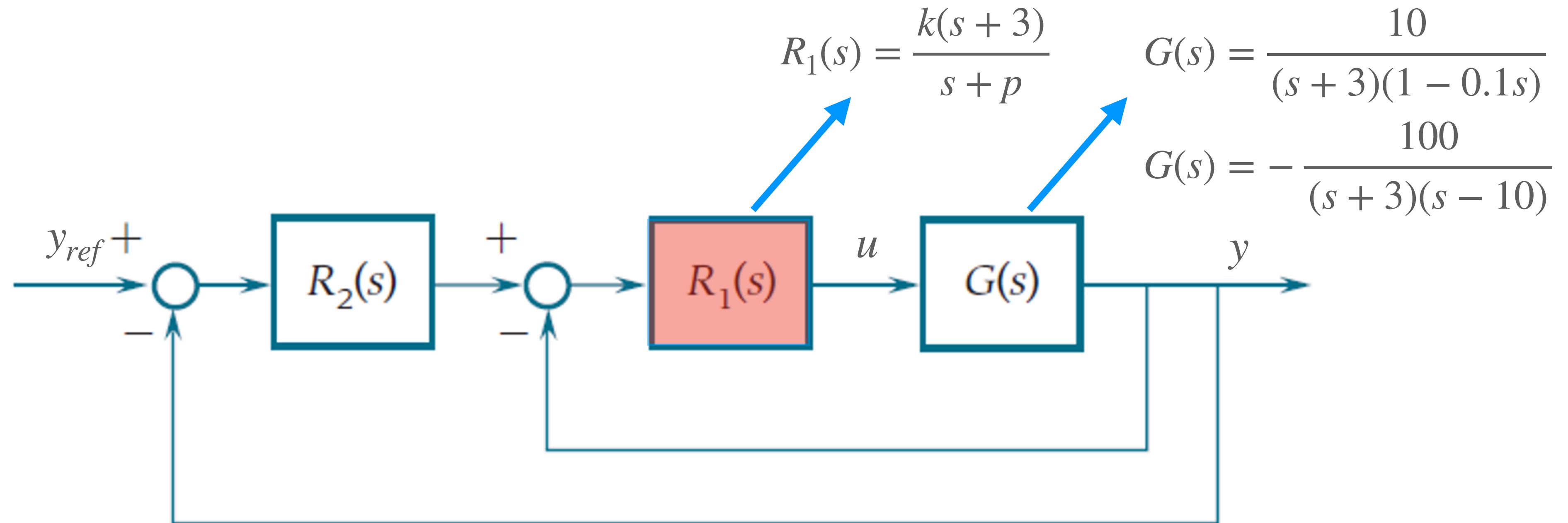
$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s+p)(s-10)} = \frac{K}{(s+p)(s-10)}$$

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## Control of Open-loop Unstable Systems

**Example:**



$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n-m} \quad \sigma_a = \frac{(-p) + 10}{2 - 0} = \frac{10 - p}{2} = -10$$

$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s+p)(s-10)} = \frac{K}{(s+p)(s-10)}$$

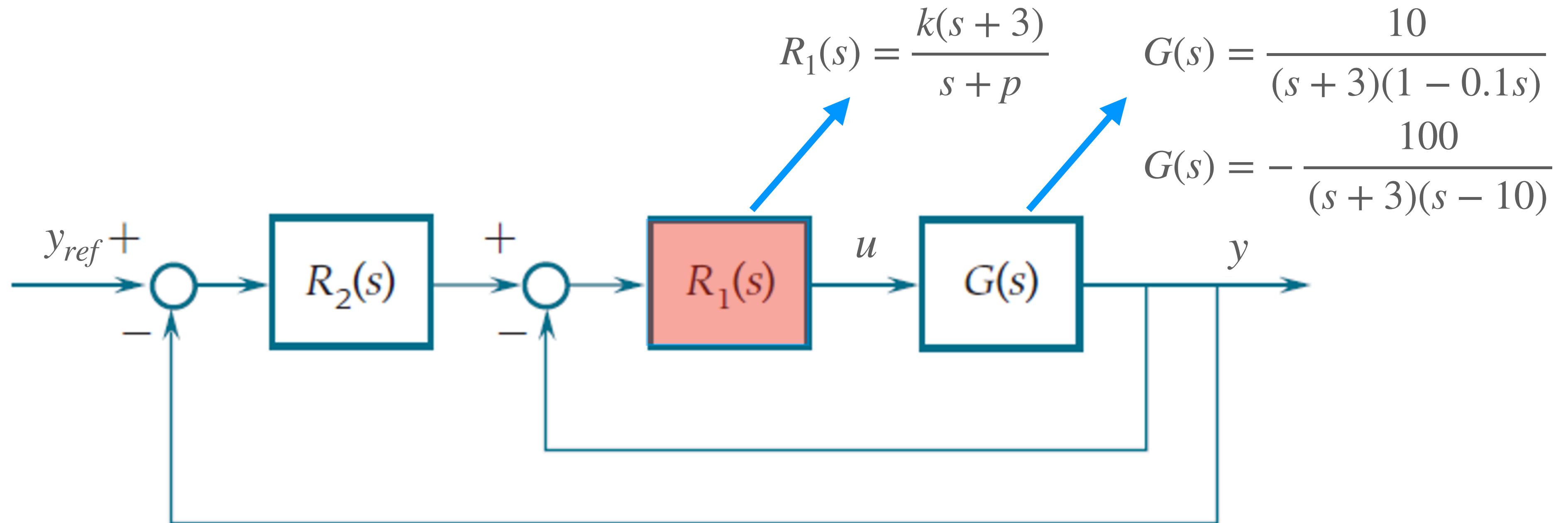
$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$

?



## Control of Open-loop Unstable Systems

**Example:**



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↓

p = 30

$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s+p)(s-10)} = \frac{K}{(s+p)(s-10)}$$

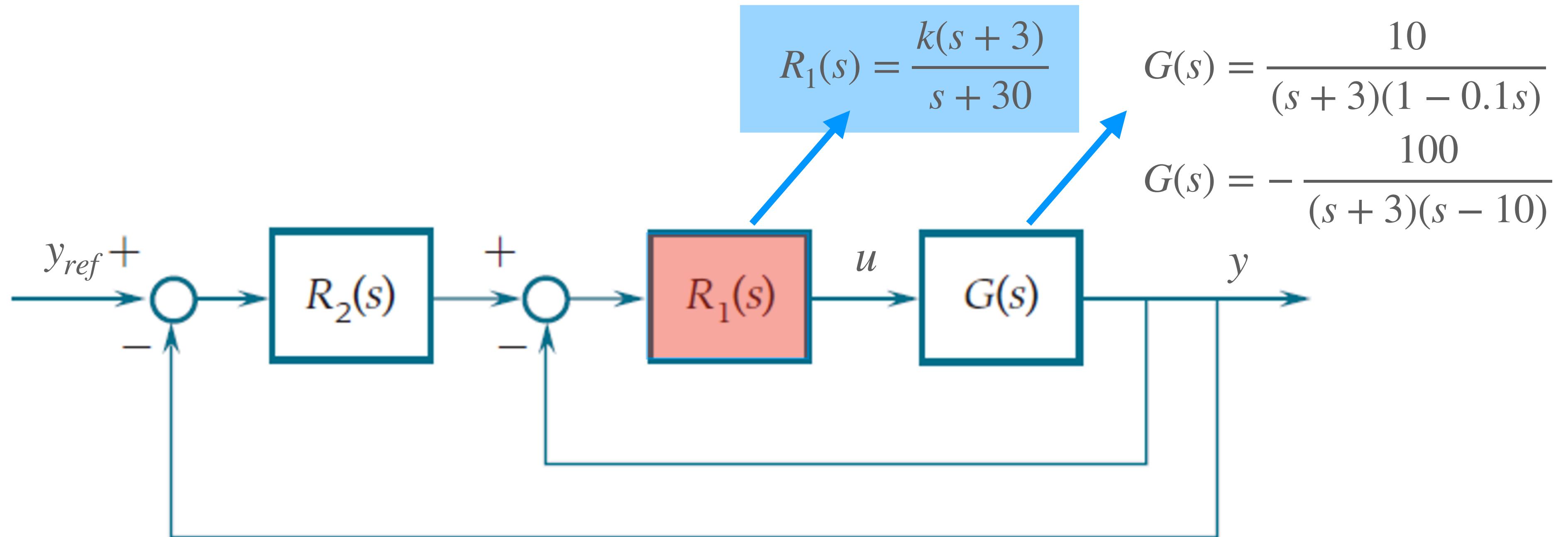
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## Control of Open-loop Unstable Systems

**Example:**



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$\downarrow$

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$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s+p)(s-10)} = \frac{K}{(s+p)(s-10)}$$

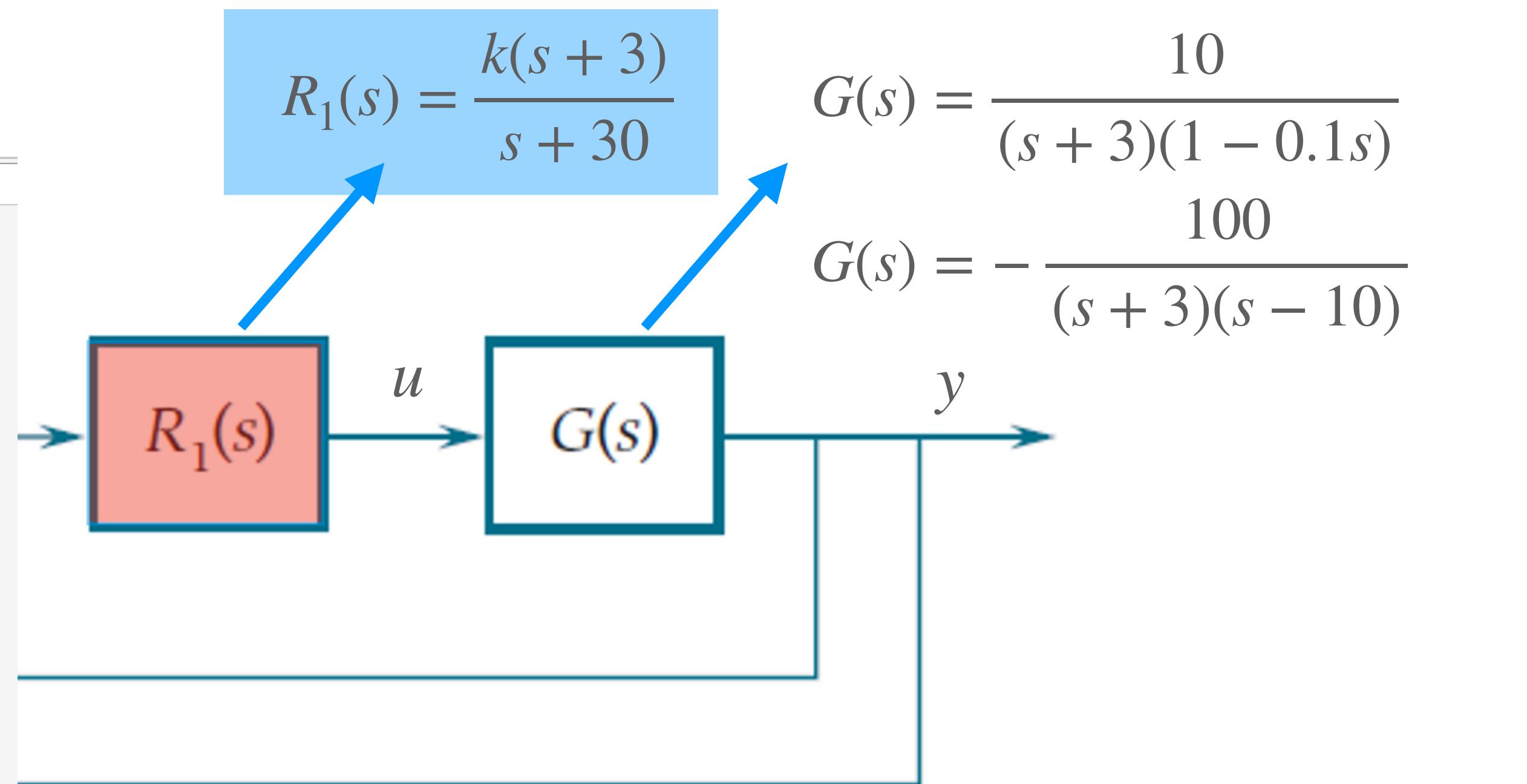
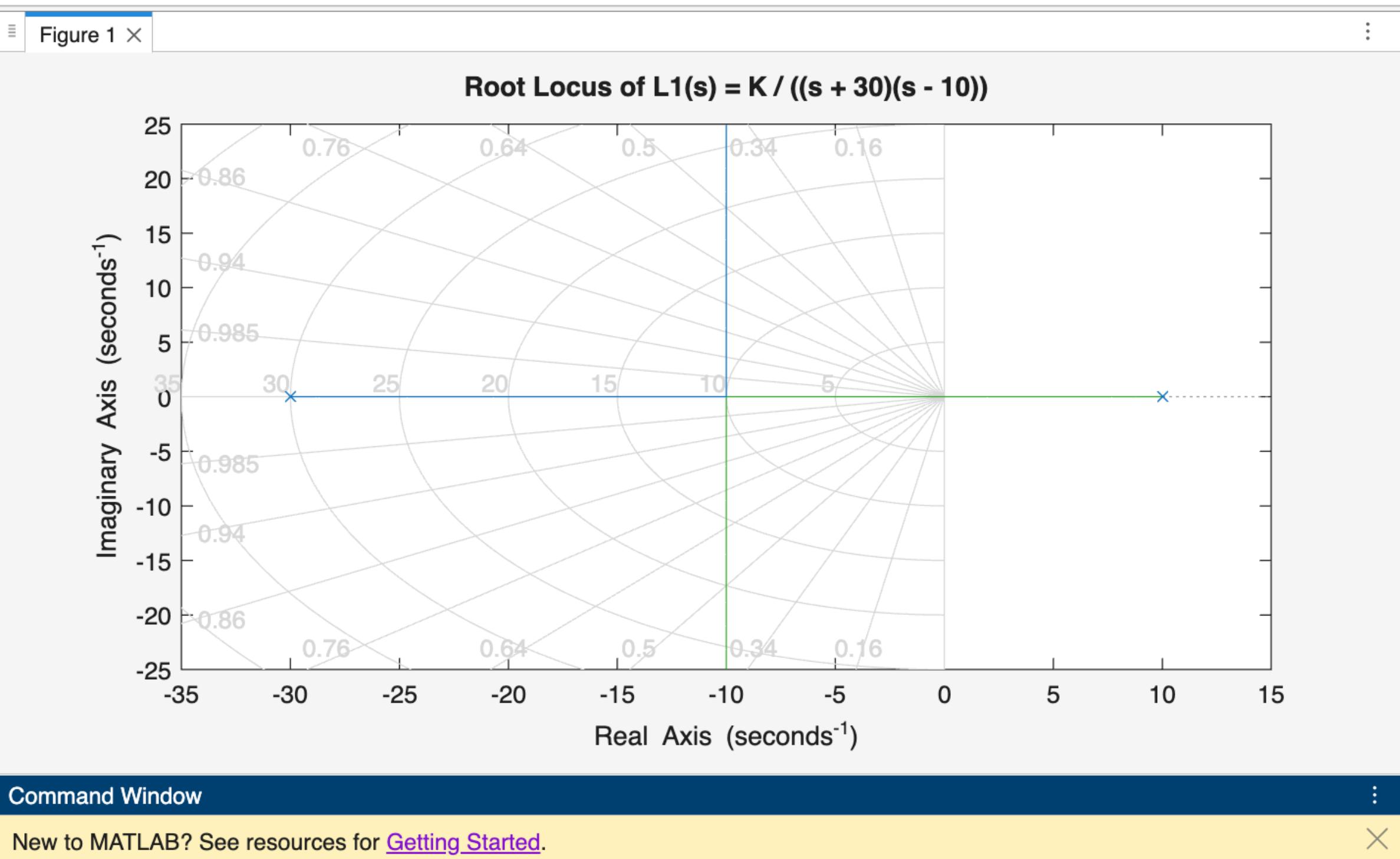
?

$$F_1(s) = \frac{L_1(s)}{1 + L_1(s)}$$



# Control of Open-loop Unstable Systems

**Example:**



$$L_1(s) = R_1(s) G(s) = -\frac{100k}{(s + p)(s - 10)} = \frac{K}{(s + p)(s - 10)}$$

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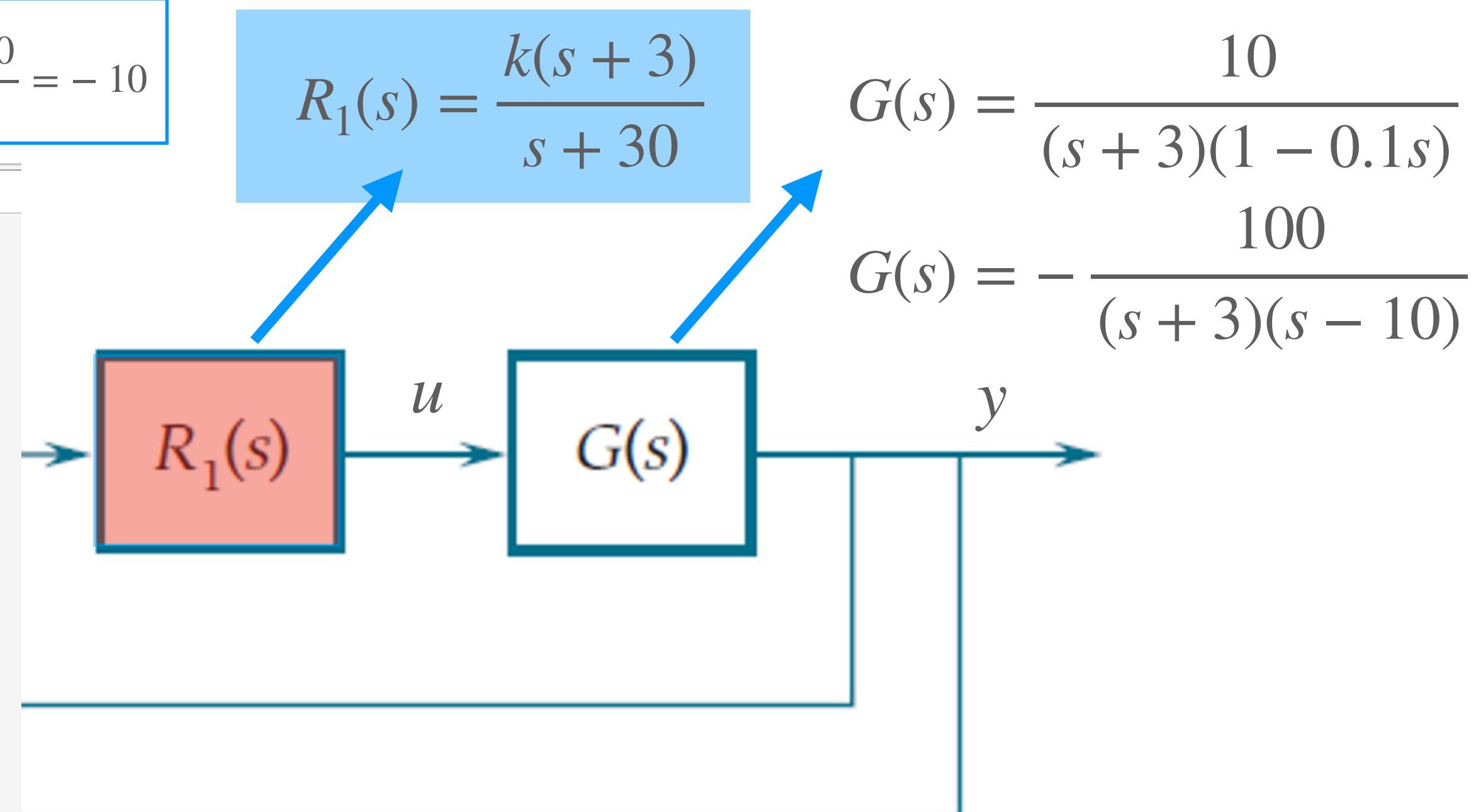
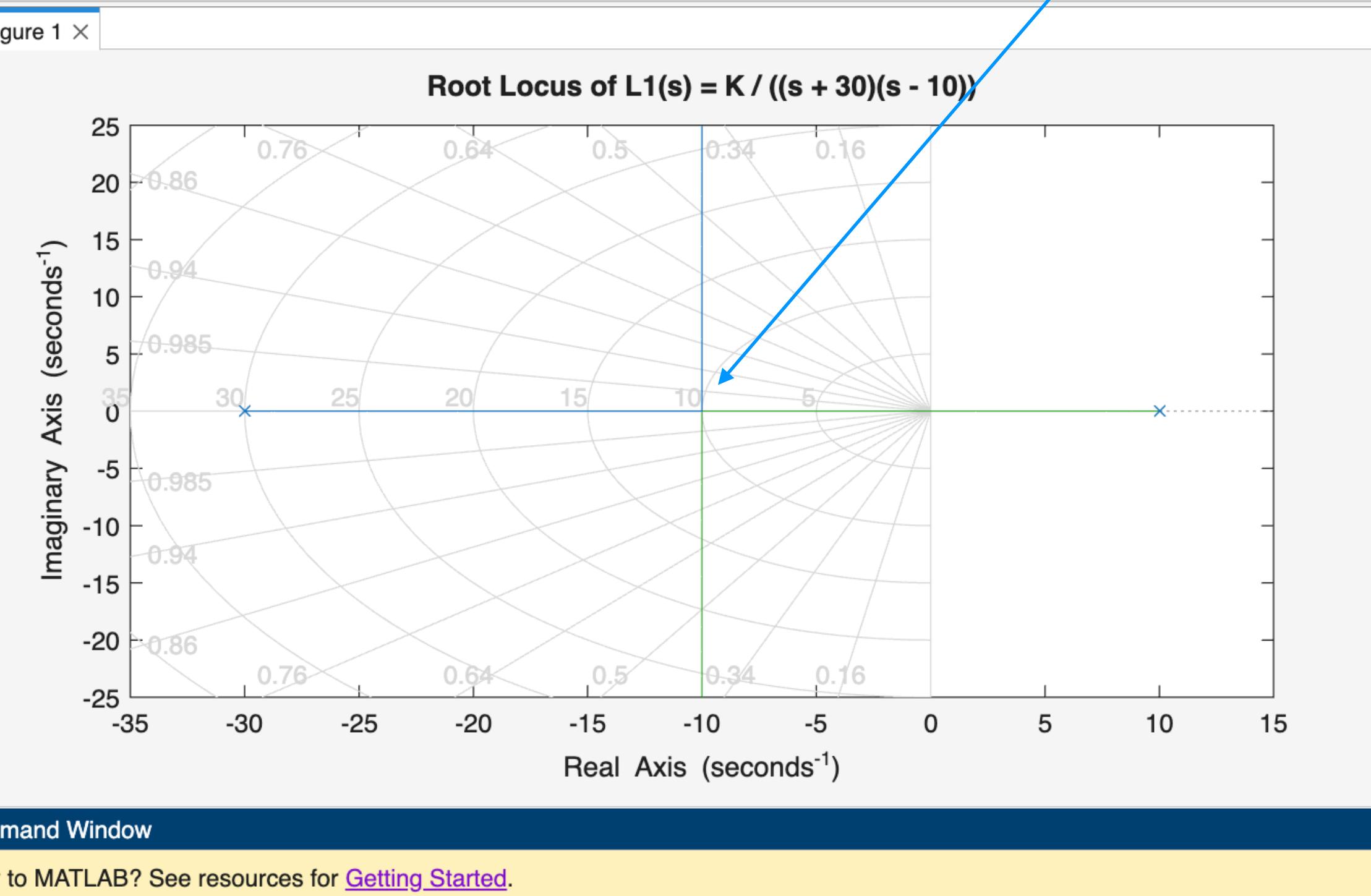


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# Control of Open-loop Unstable Systems

**Example:**

$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



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```
>> s = tf('s');
L1 = 1 / ((s + 30)*(s - 10)); % This is without the gain K
rlocus(L1)
grid on
title('Root Locus of L1(s) = K / ((s + 30)(s - 10))')
>> |
```

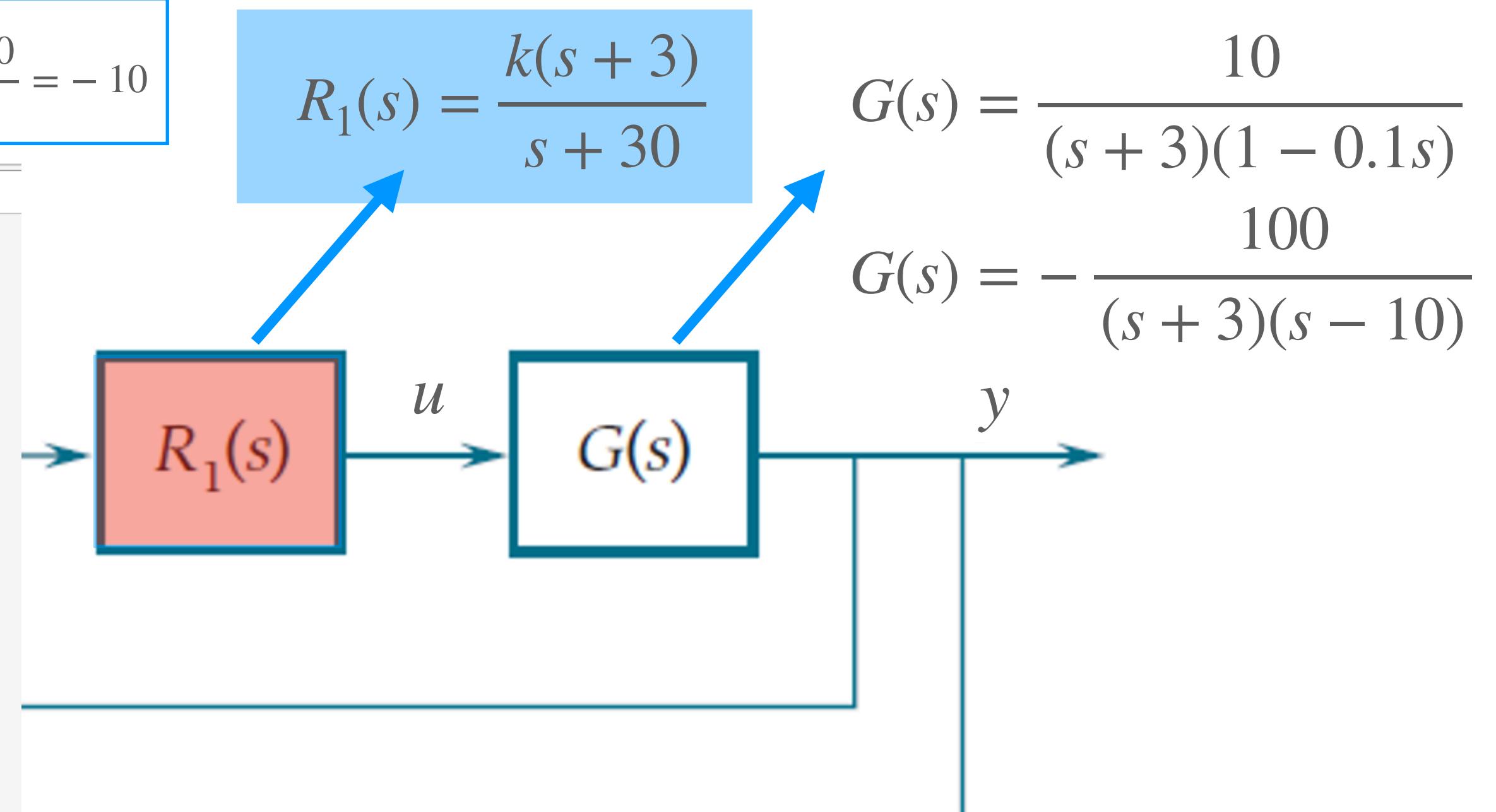
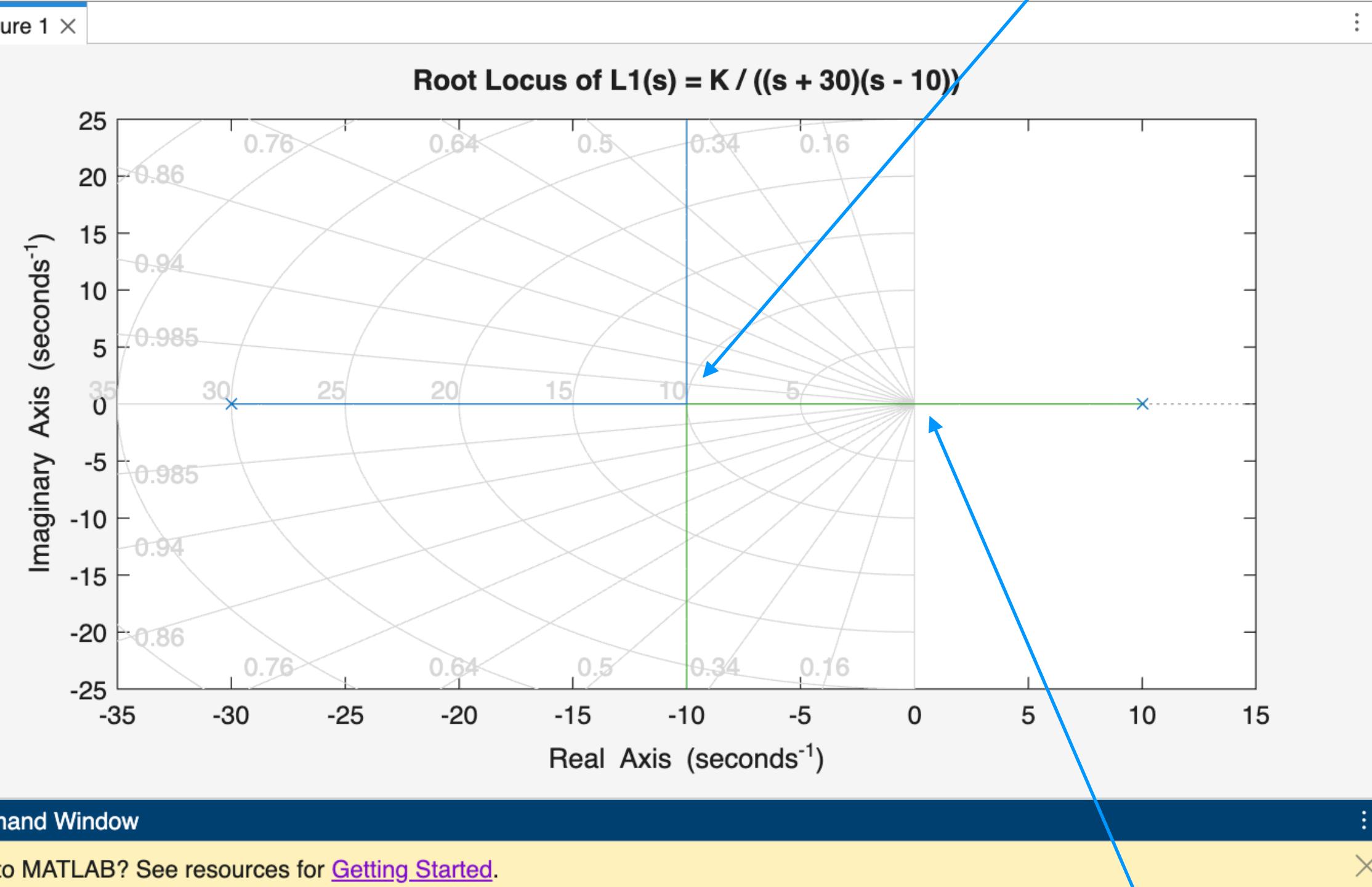


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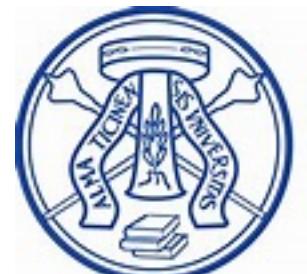


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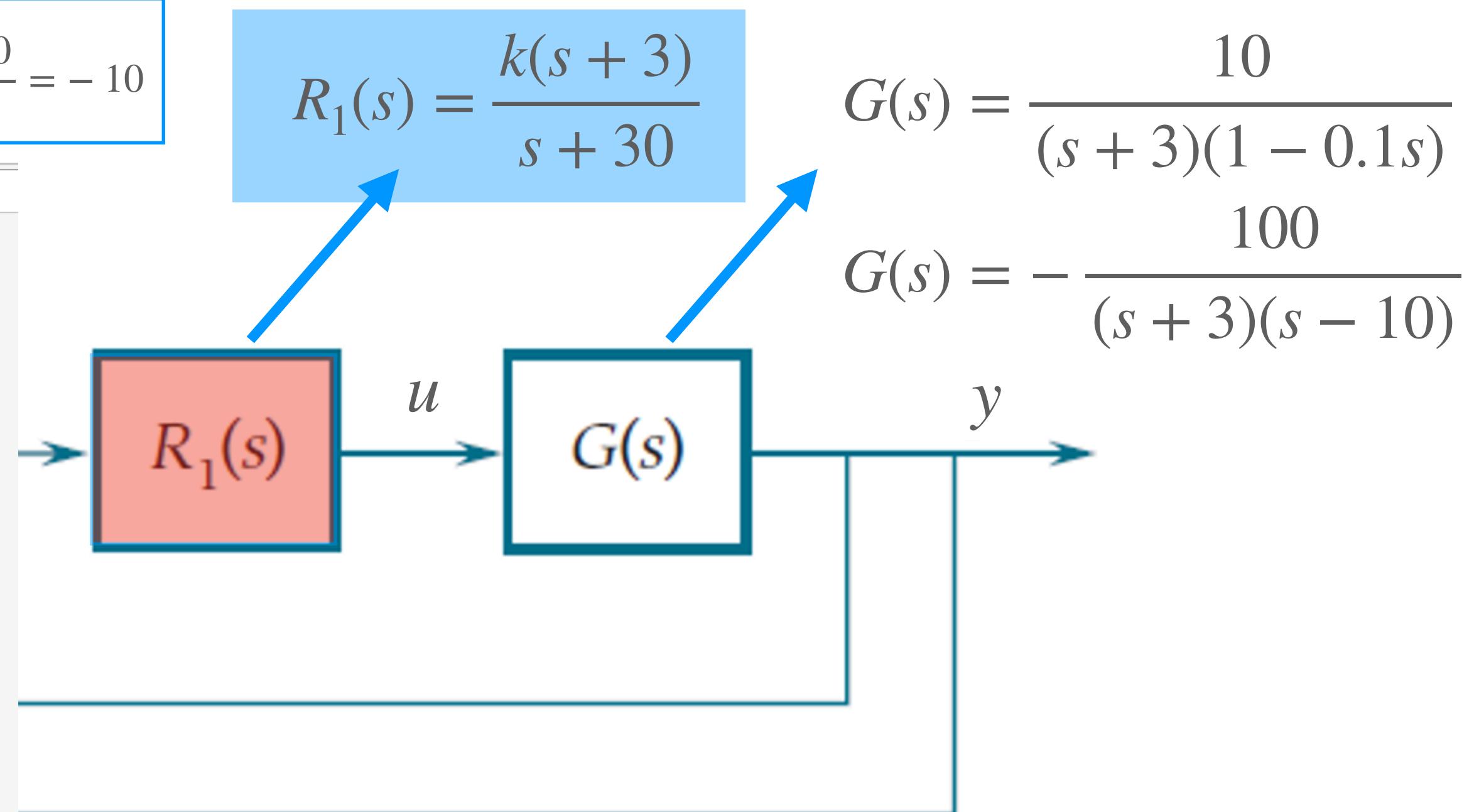
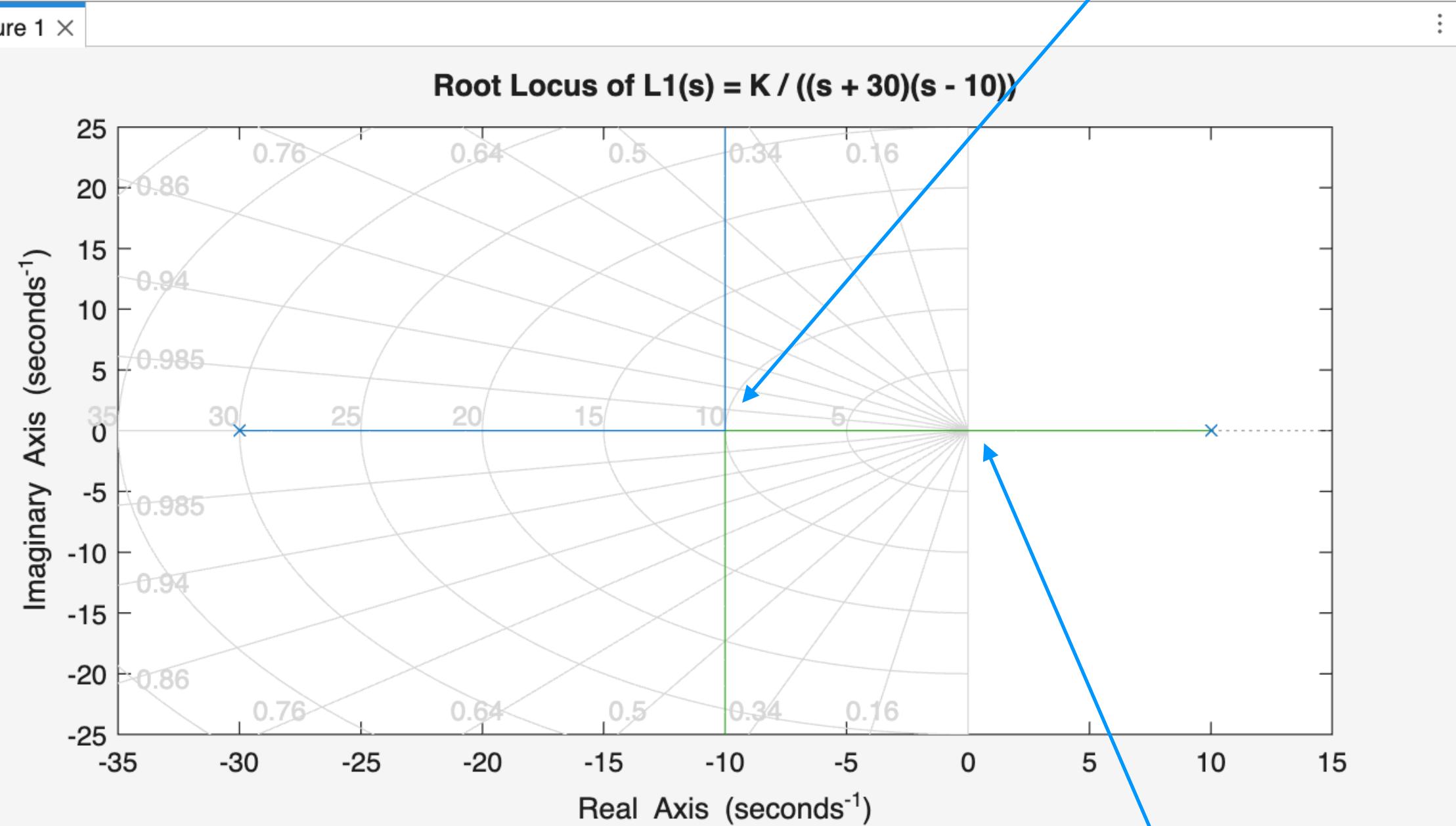


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$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



Command Window

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title('Root Locus of L1(s) = K / ((s + 30)(s - 10))')
>> |
```

*K\* ?*

Command Window

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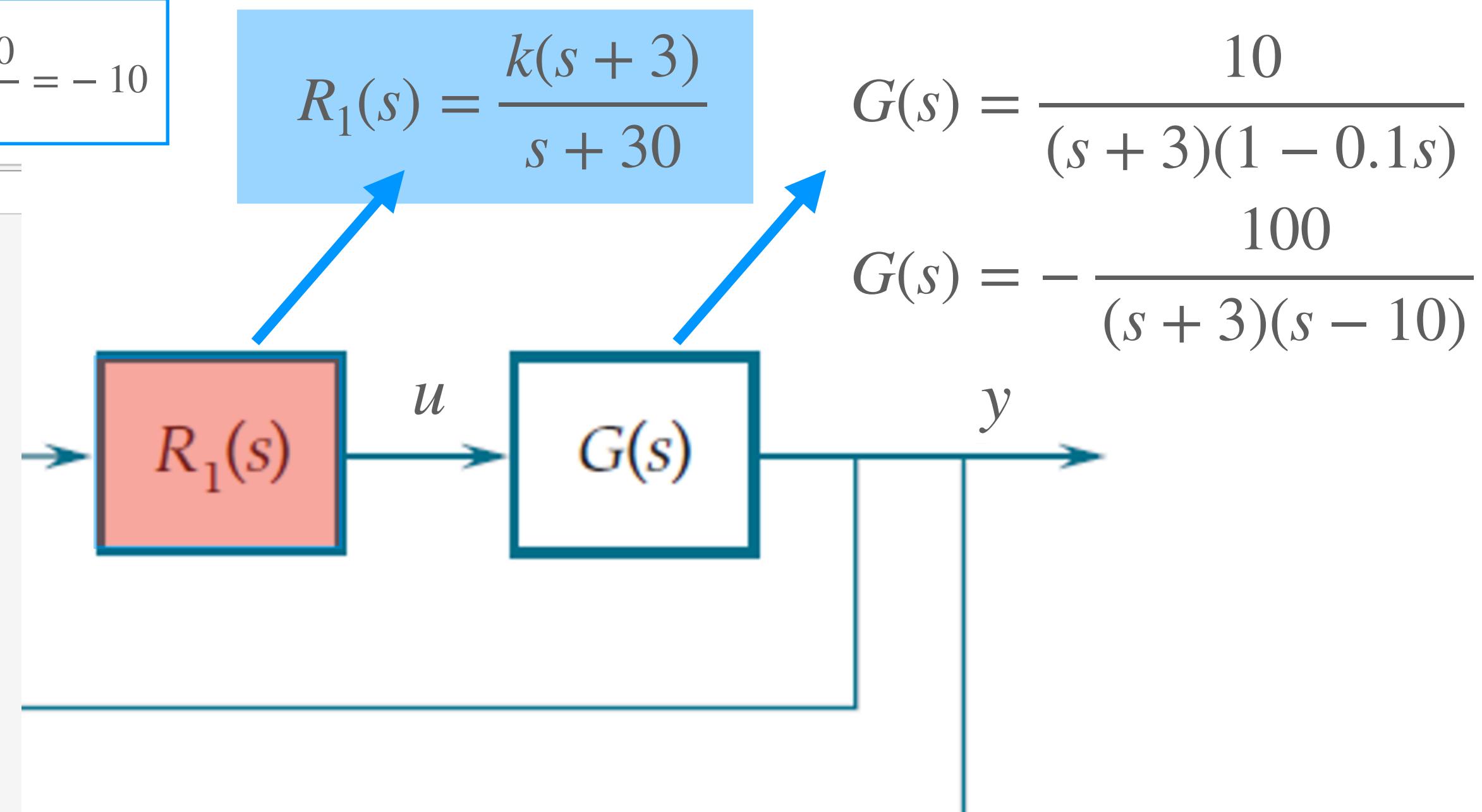
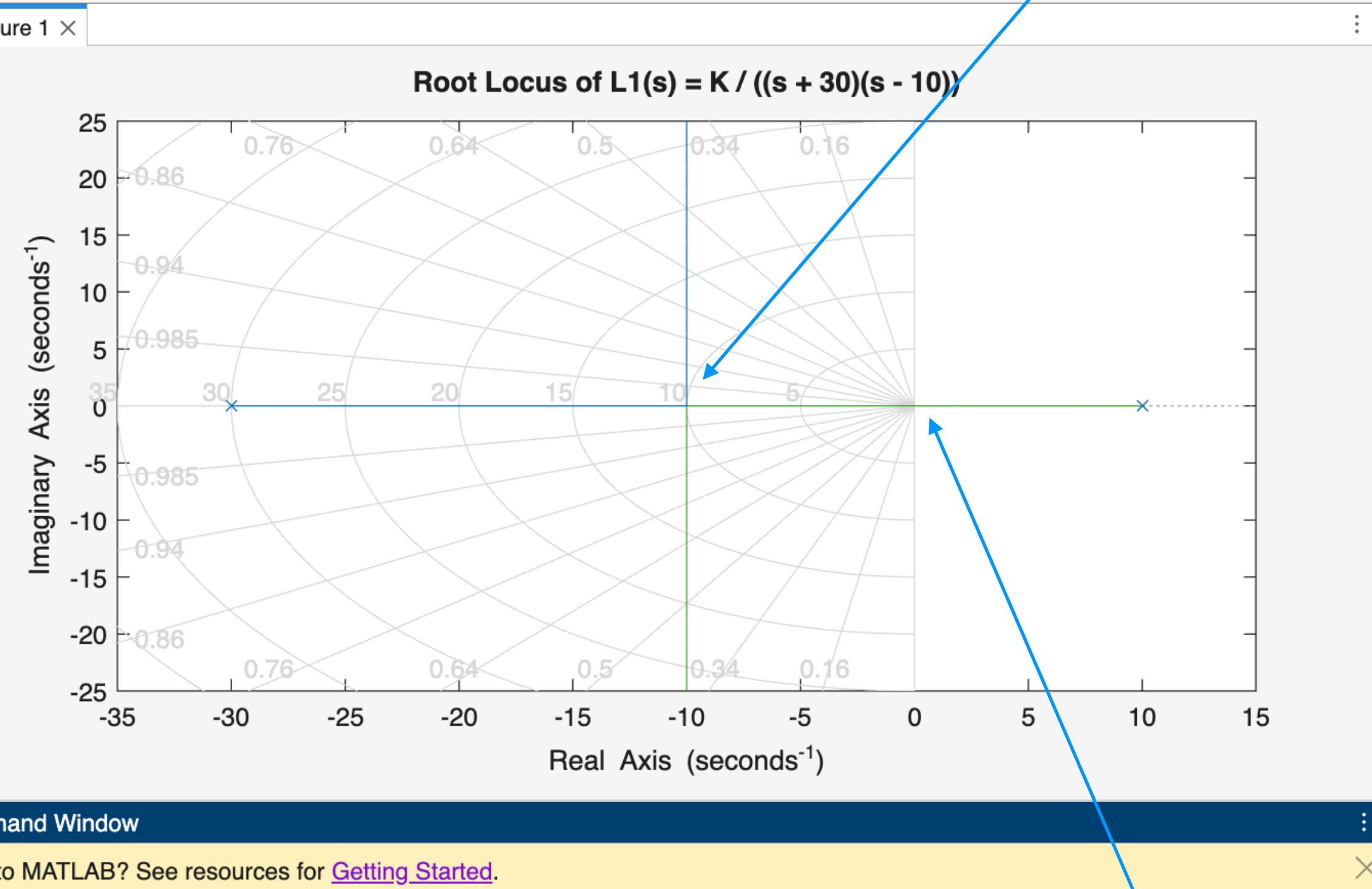
```
desired_pole = 0;
K_at_origin = rlocfind(L1, desired_pole);
K_at_origin
|
K_at_origin =
300
```



# Control of Open-loop Unstable Systems

**Example:**

$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right|$$

$K^*$  ?

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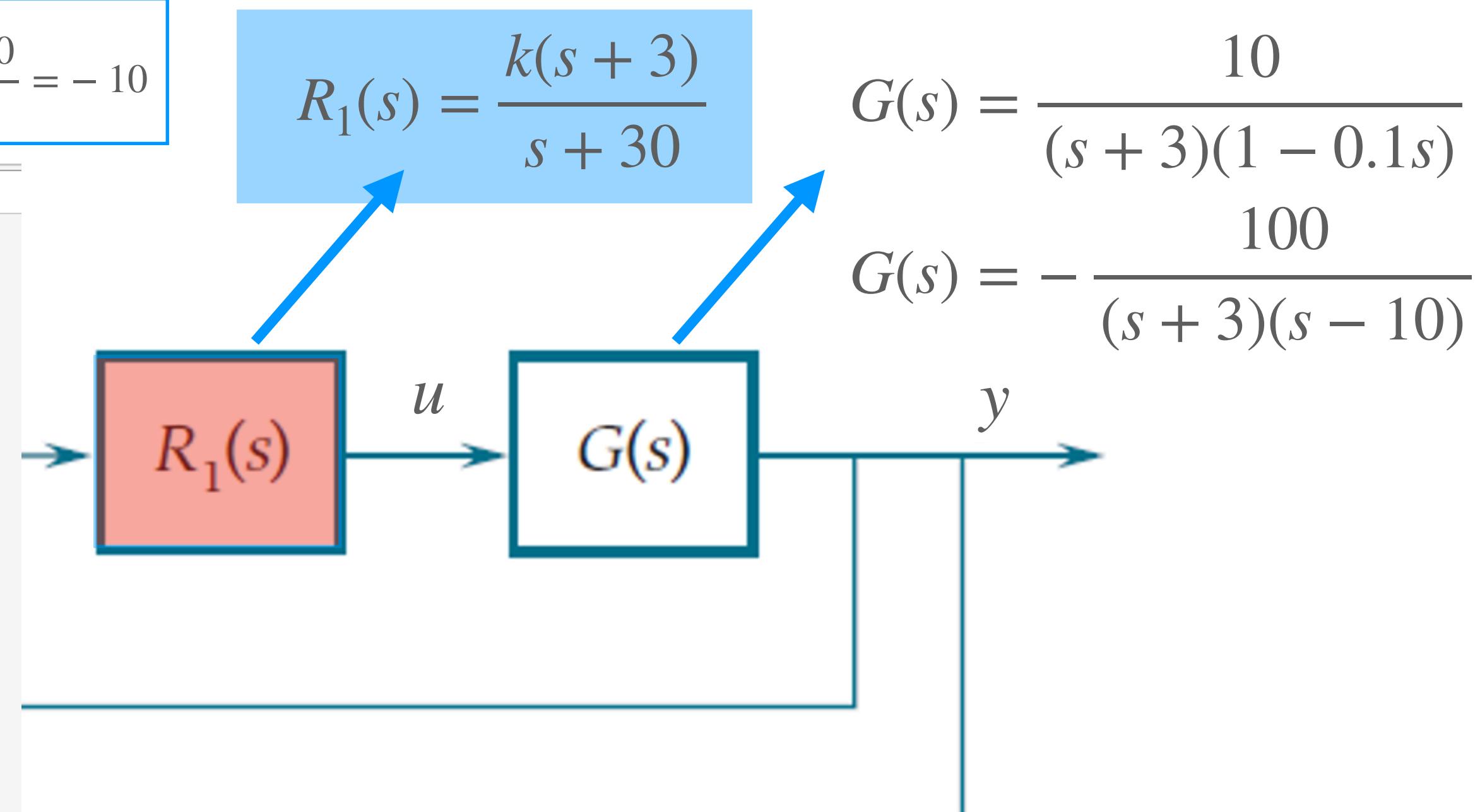
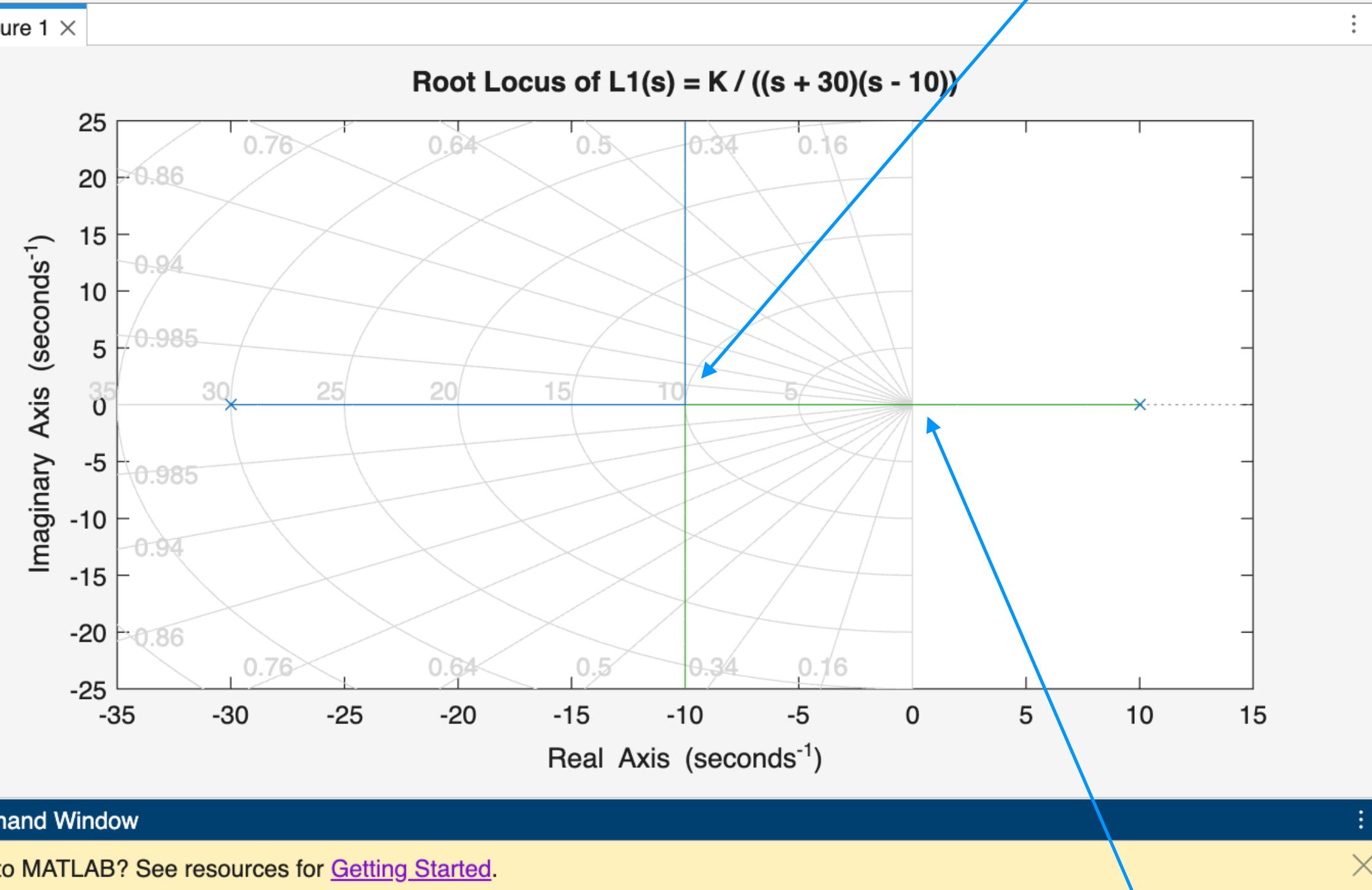


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# Control of Open-loop Unstable Systems

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$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^* + 30)(s^* - 10)}{1} \right|$$

$K^*$  ?

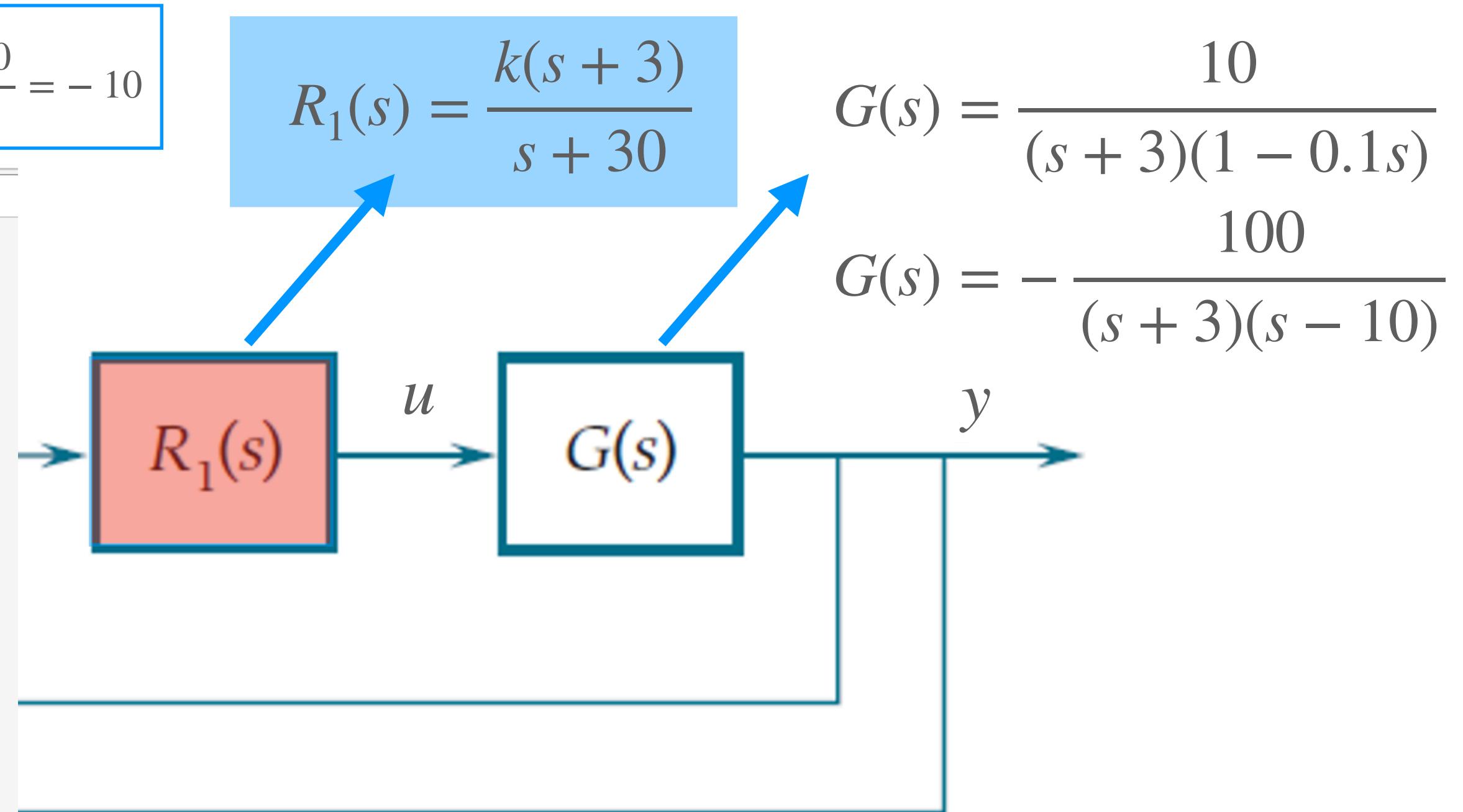
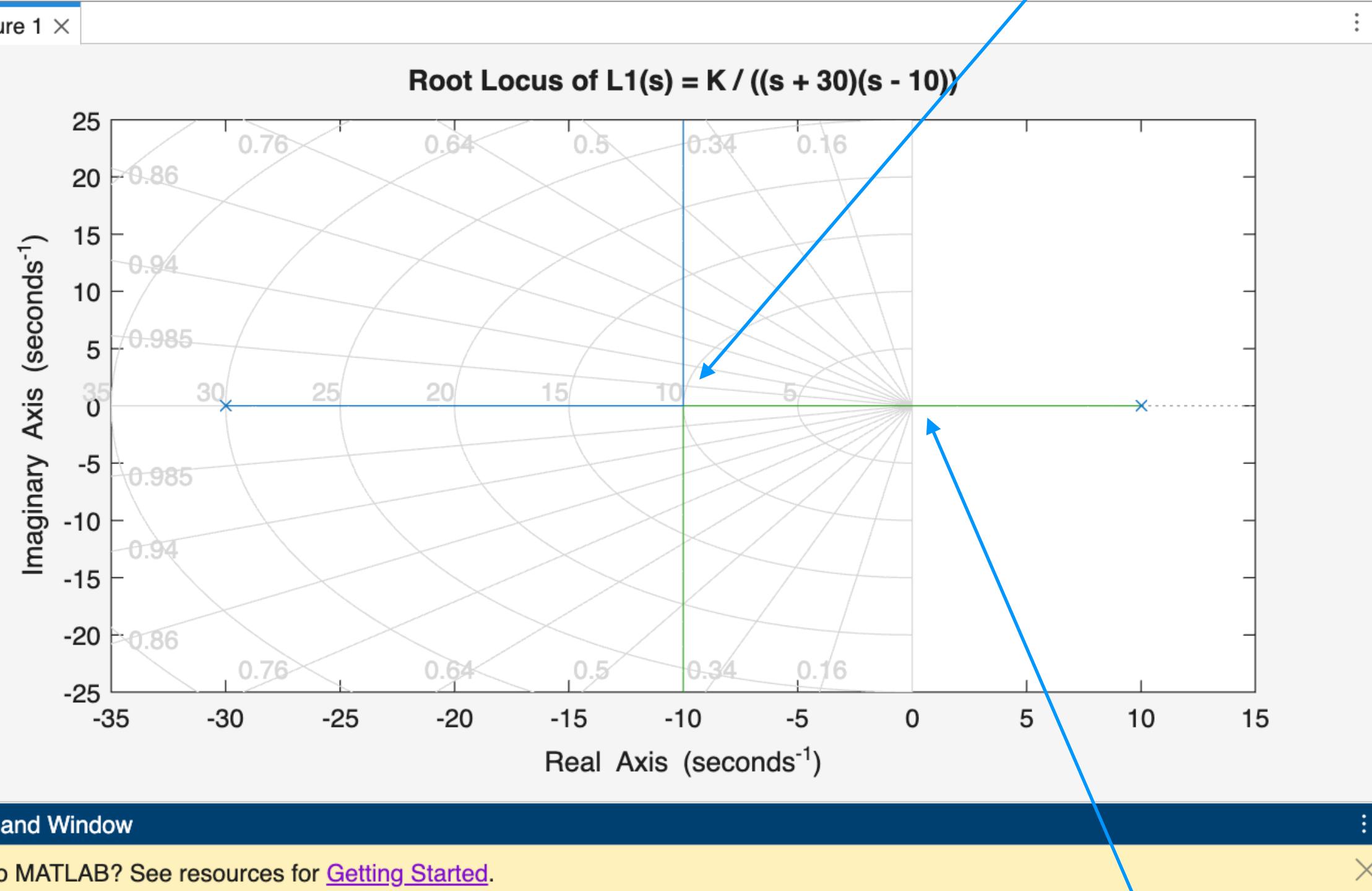


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# Control of Open-loop Unstable Systems

**Example:**

$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^* + 30)(s^* - 10)}{1} \right|$$

$K^*$  ? →

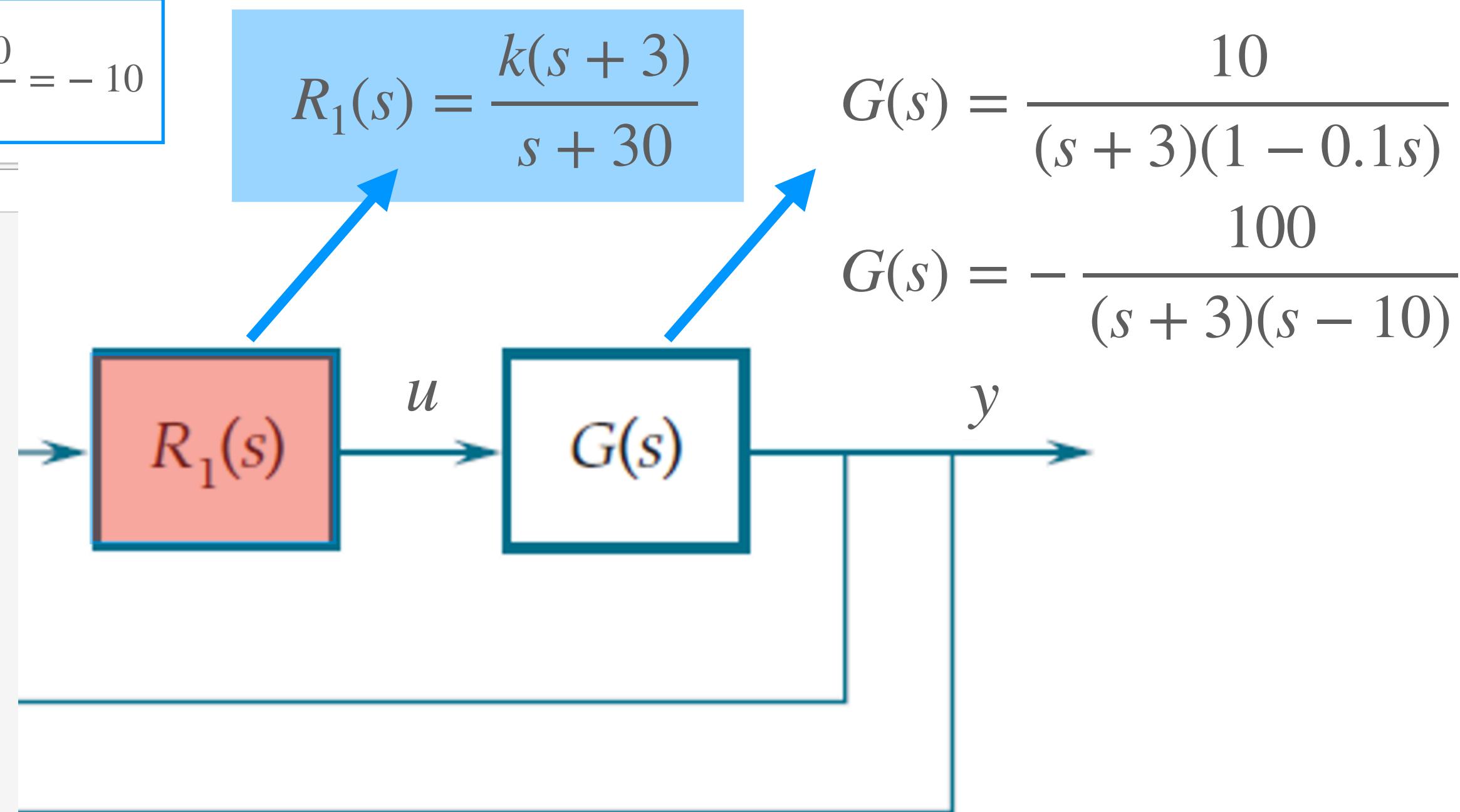
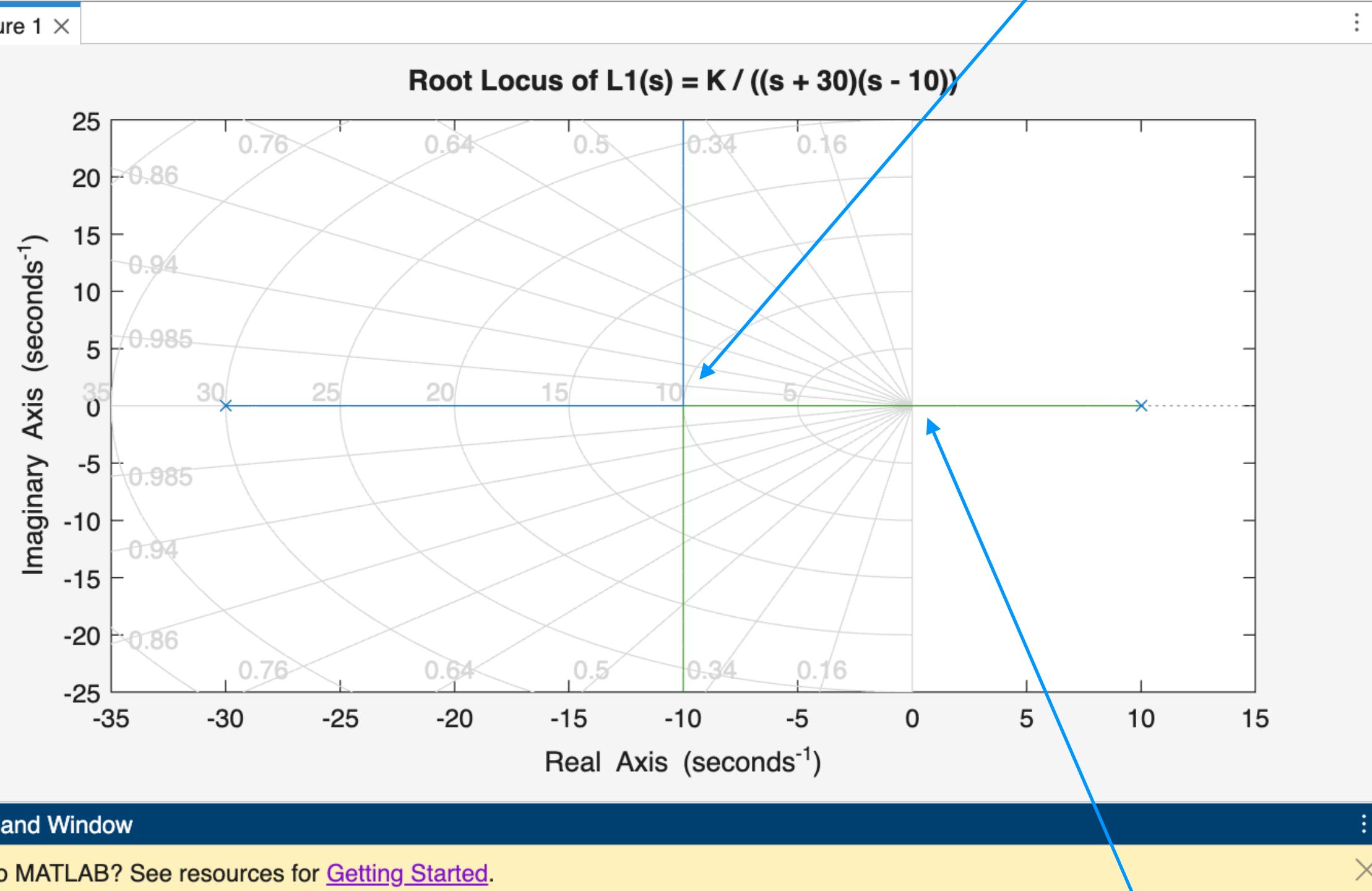
$$|K|_{s=0} = \left| \frac{D(0)}{N(0)} \right| = \left| \frac{(0 + 30)(0 - 10)}{1} \right|$$



# Control of Open-loop Unstable Systems

**Example:**

$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^* + 30)(s^* - 10)}{1} \right|$$

$K^*$  ? →

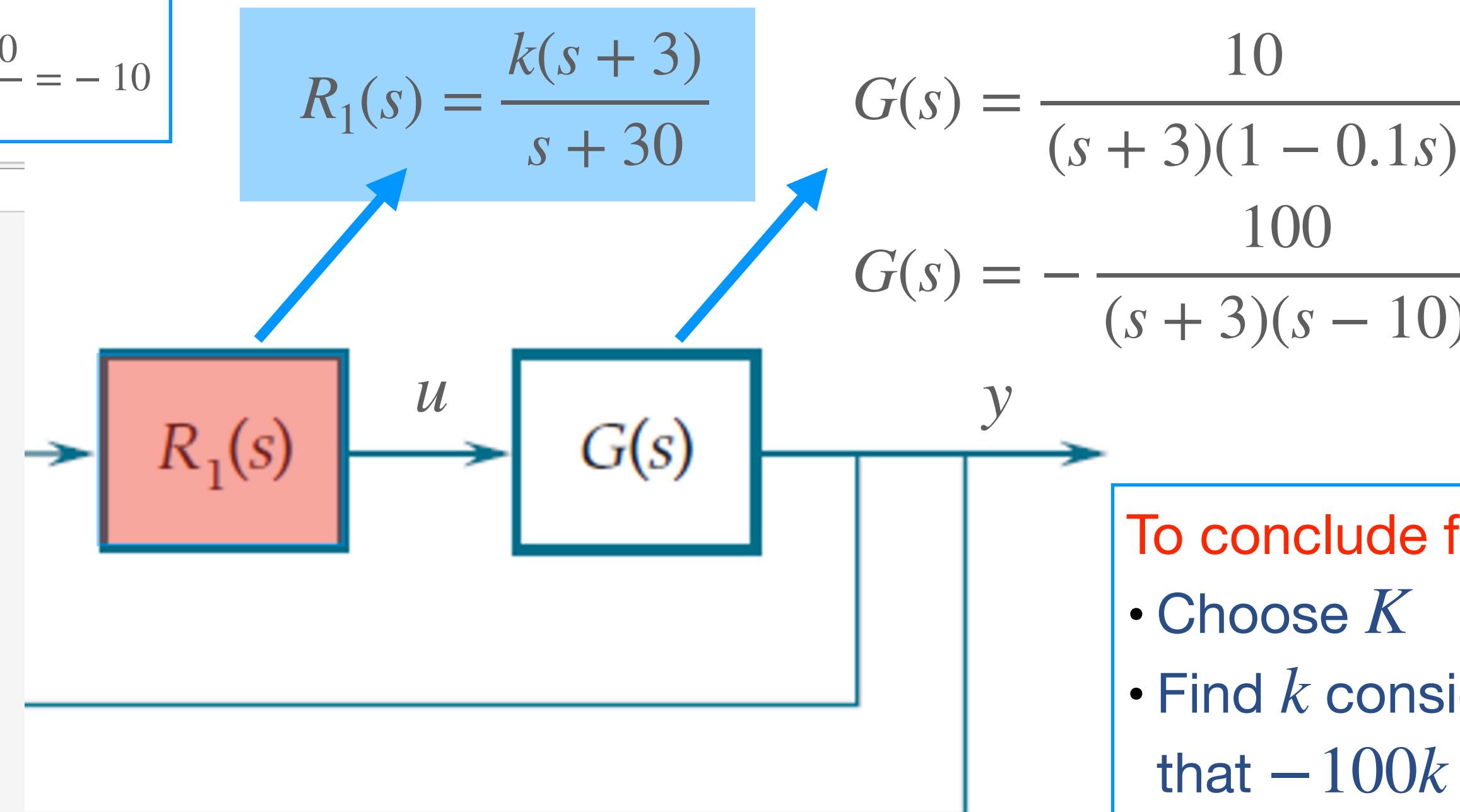
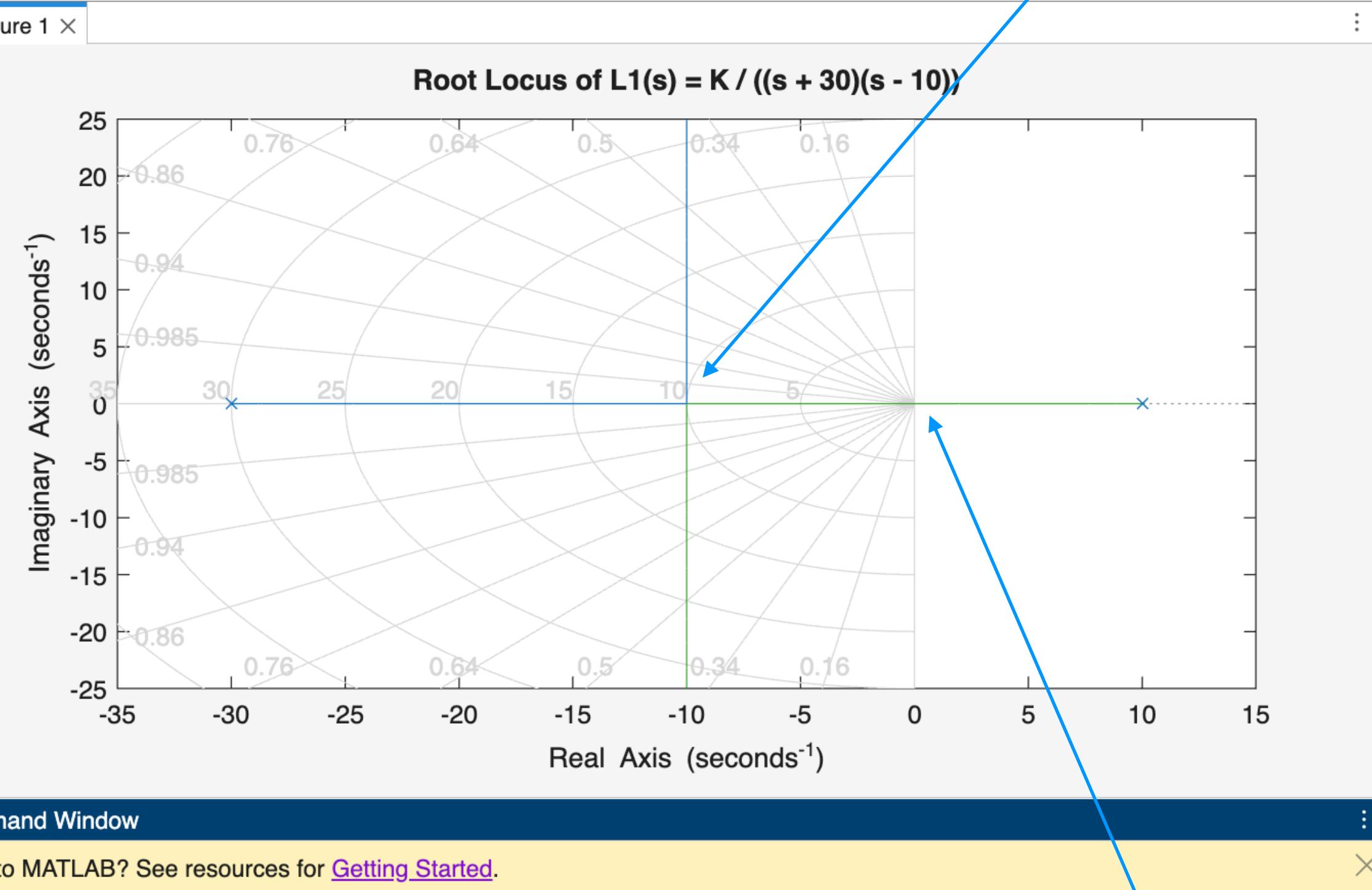
$$|K|_{s=0} = \left| \frac{D(0)}{N(0)} \right| = \left| \frac{(0 + 30)(0 - 10)}{1} \right| = 300$$



## Control of Open-loop Unstable Systems

**Example:**

$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^* + 30)(s^* - 10)}{1} \right|$$

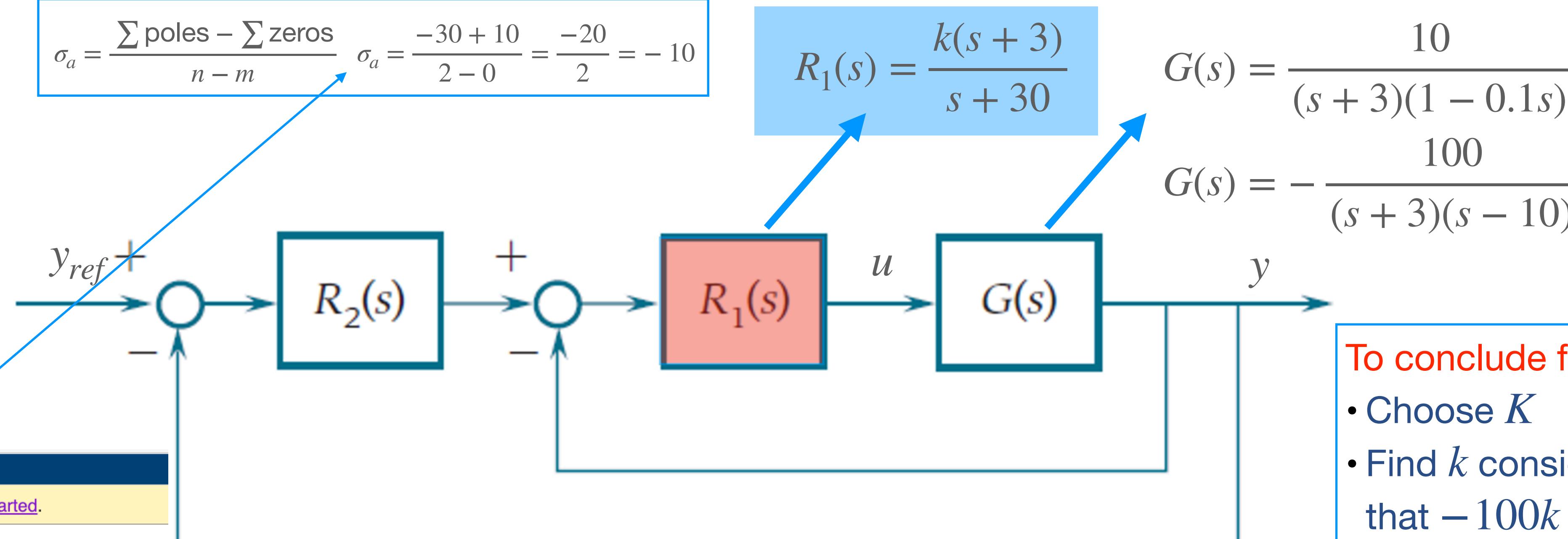
$$|K|_{s=0} = \left| \frac{D(0)}{N(0)} \right| = \left| \frac{(0 + 30)(0 - 10)}{1} \right| = 300$$



## Control of Open-loop Unstable Systems

**Example:**

$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



My choice:

Command Window

New to MATLAB? See resources for [Getting Started](#).

```
>> desired_pole = -10;
K_at_sigma_a = rlocfind(L1, desired_pole);
K_at_sigma_a
```

K\_at\_sigma\_a =

400

$$R_1(s) = \frac{k(s + 3)}{s + 30}$$

$$G(s) = \frac{10}{(s + 3)(1 - 0.1s)}$$

$$G(s) = -\frac{100}{(s + 3)(s - 10)}$$

To conclude for  $R_1(s)$ :

- Choose  $K$
- Find  $k$  considering that  $-100k = K$



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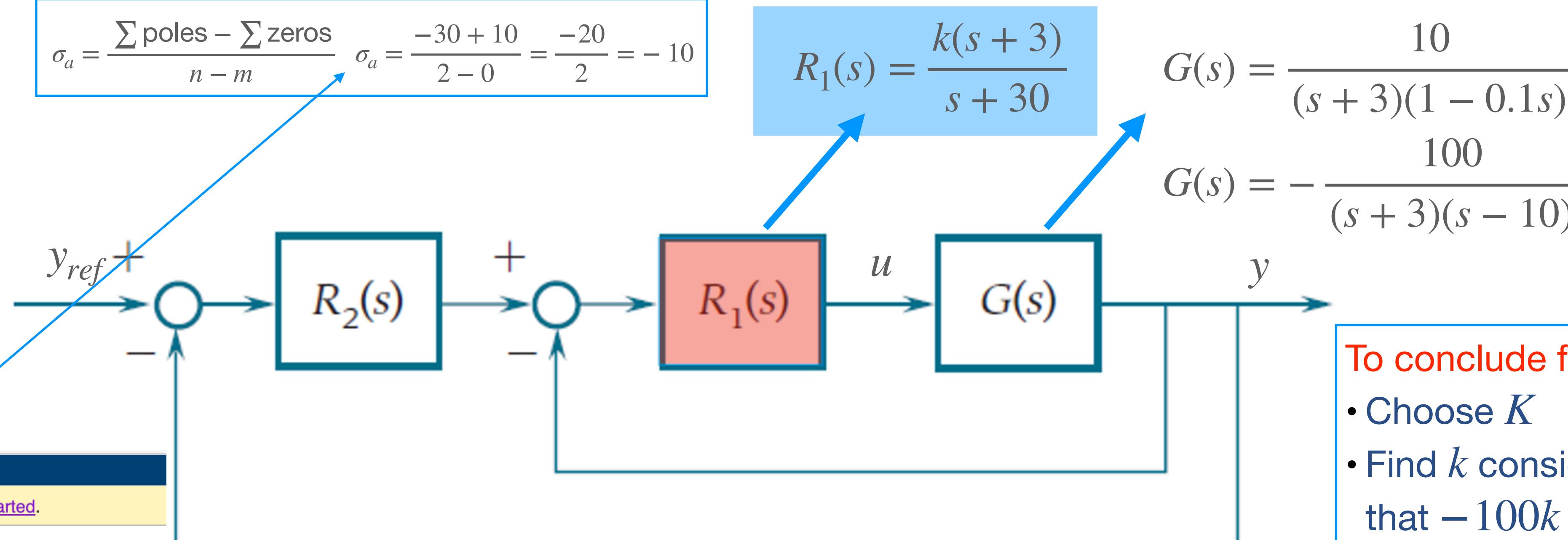
$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^* + 30)(s^* - 10)}{1} \right|$$

$K^* ? \rightarrow |K|_{s=0} = \left| \frac{D(0)}{N(0)} \right| = \left| \frac{(0 + 30)(0 - 10)}{1} \right| = 300$

## Control of Open-loop Unstable Systems

**Example:**

$$\sigma_a = \frac{\sum \text{poles} - \sum \text{zeros}}{n - m} \quad \sigma_a = \frac{-30 + 10}{2 - 0} = \frac{-20}{2} = -10$$



My choice:

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```
>> desired_pole = -10;
K_at_sigma_a = rlocfind(L1, desired_pole);
K_at_sigma_a
```

K\_at\_sigma\_a =

400

$$-100k = K$$

$$k = -\frac{K}{100}$$

$$R_1(s) = \frac{k(s+3)}{s+30}$$

$$G(s) = \frac{10}{(s+3)(1-0.1s)}$$

$$G(s) = -\frac{100}{(s+3)(s-10)}$$

To conclude for  $R_1(s)$ :

- Choose  $K$
- Find  $k$  considering that  $-100k = K$



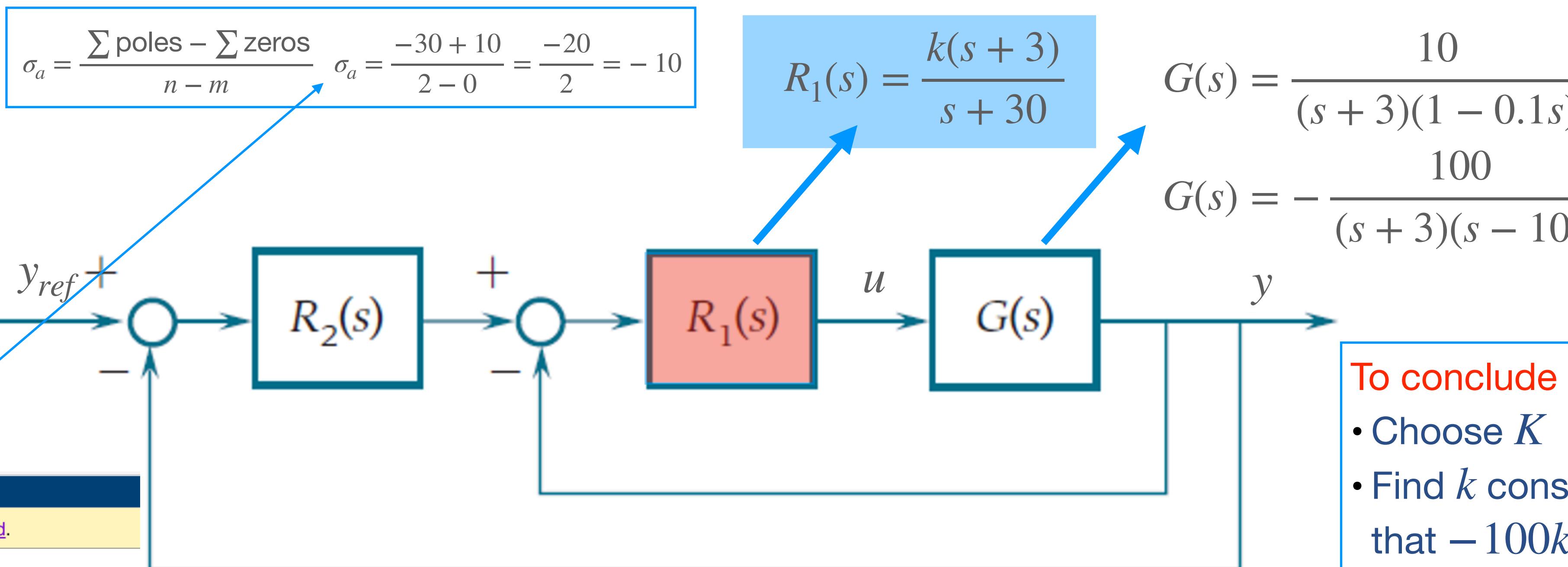
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$$|K|_{s=s^*} = \left| \frac{D(s^*)}{N(s^*)} \right| = \left| \frac{(s^* + 30)(s^* - 10)}{1} \right|$$

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## Control of Open-loop Unstable Systems

**Example:**



My choice:

Command Window

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```
>> desired_pole = -10;
K_at_sigma_a = rlocfind(L1, desired_pole);
K_at_sigma_a
```

K\_at\_sigma\_a =

400

$$-100k = K$$

$$k = -\frac{K}{100}$$

$$R_1(s) = -\frac{4(s + 3)}{s + 30}$$

To conclude for  $R_1(s)$ :

- Choose  $K$
- Find  $k$  considering that  $-100k = K$



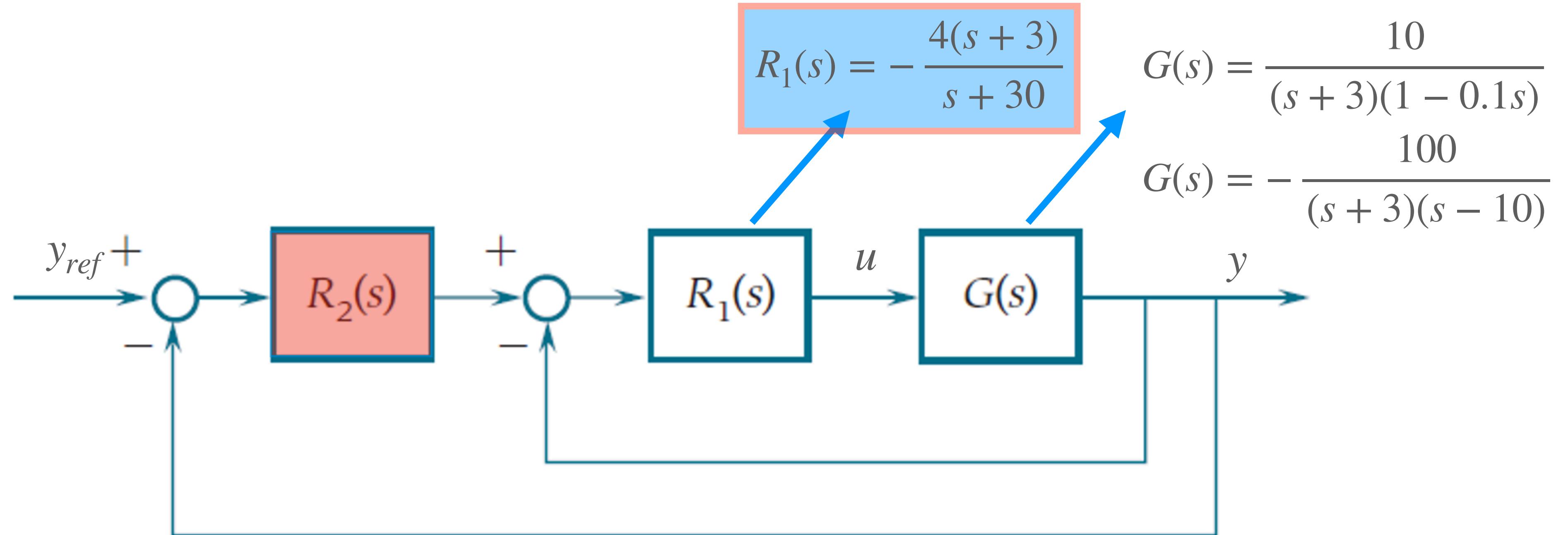
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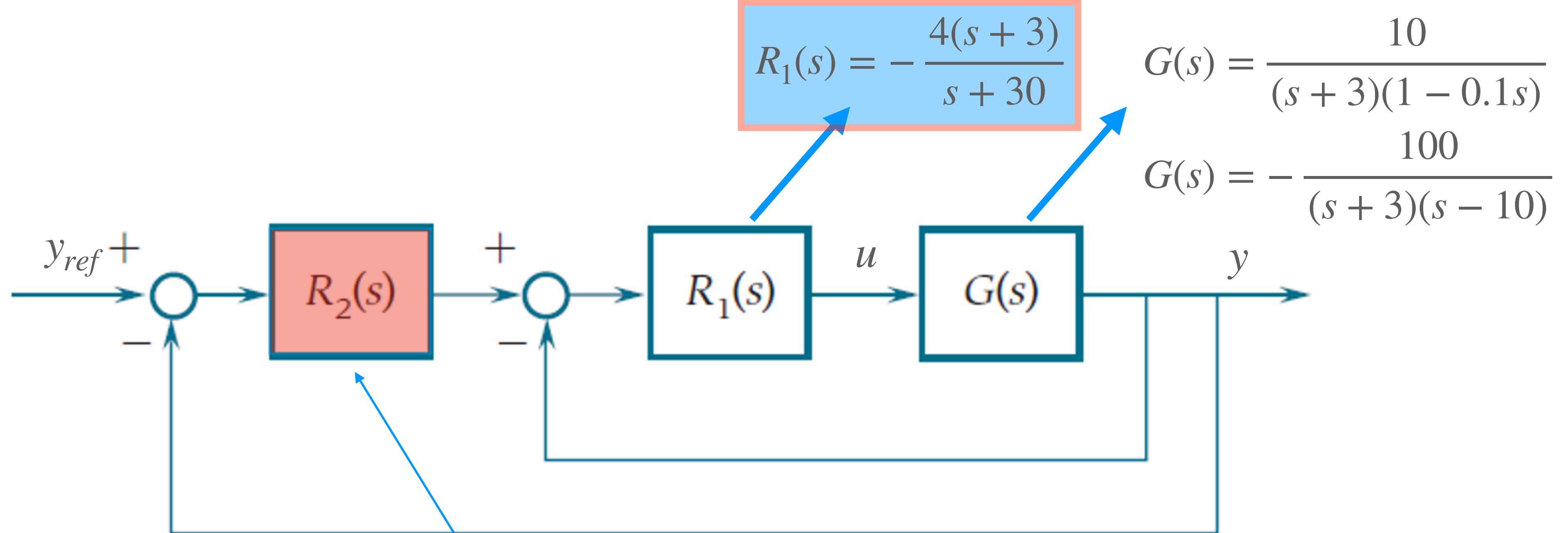
## Control of Open-loop Unstable Systems

Example:



## Control of Open-loop Unstable Systems

Example:



To conclude for  $R_2(s)$ :

- Compute  $F_1(s)$
- Design  $R_2(s)$  based on  $F_1(s)$  using standard loop-shaping design guided by the Bode criterion

