Indian Institute of Space Science and Technology AE 111 - Introduction to Aerospace Engineering (I Semester)

Test

Duration: 60 minutes Total Mark		Total Marks:20
	Name:	
	SC No: SC	
	Batch	
	For a turbojet aircraft, at the minimum thrust required condition	m, L/D is
	The tangent of glide angle is	
	The lift to weight ratio for steady climb angle of 45° is	
	For a piton-propeller aircraft, at the minimum power required conserved, zero-lift drag and lift-induced drag.	ondition, what is the relation
	What is absolute ceiling?	
	How does power required vary with altitude?	

7.	A typical glider has weight of 150N and has glide angle of 2.5°. Find Lift produced by the glider.	
3.	An aircraft is climbing with a 20° of climb angle and a 5° of angle of attack having a mass of 11,000 kg and aircraft drag is 9,000 N then determine engine thrust.	
•	To maximize the range for a reciprocating-engine, propeller-driven airplane, we want the following	
	1.	
	2.	
	3.	
	4.	
	For maximum endurance for a jet airplane, we want	
	1.	
	2.	
	3.	
	Define rate of climb.	
	During level turning operation an aircraft has load factor of 2. Find the required bank angle for this level turning operation.	