

goal: frequency-domain analysis of control system performance

(a) transfer matrix $\dot{x} = Ax + Bu \leadsto G_{yu}(s) = C(sI - A)^{-1}B + D$
 (i.e. matrix of transfer functions) $y = Cx + Du$ $u = e^{st} \cdot u_0 \Rightarrow y = e^{st} \cdot G_{yu}(s) \cdot u_0$

(b) Bode plots $u = e^{st} \cdot u_0 \leadsto y = e^{st} G_{yu}(s) u_0$, $\sin(\omega t) = \text{Im } e^{j\omega t}$
 so plotting $\log|G(j\omega)|$, $\angle G(j\omega)$ vs ω shows us how system responds

(c) effect of disturbances $y = \frac{1}{1+PC} w + \frac{P}{1+PC} v + \frac{PC}{1+PC} r$
 so we can use sensitivity $S = \frac{1}{1+PC}$ to assess performance

(d) fundamental limits $S = \frac{1}{1+PC} = 1 - \frac{PC}{1+PC} = 1 - T$; $\int_0^\infty \log|S(j\omega)| d\omega$ is conserved

(a) transfer matrix [AMv2 ch 6.3, 9.2] [NV7 ch 4.11]

\hookrightarrow i.e. a matrix of transfer functions relating a vector of inputs to a vector of outputs

• given LTI system in state-space form, $\dot{x} = Ax + Bu$, $x \in \mathbb{R}^n$, $u \in \mathbb{R}^p$
 $y = Cx + Du$ $y \in \mathbb{R}^q$

know that $x(t) = e^{At} x(0) + \int_0^t e^{A(t-\tau)} B u(\tau) d\tau$

so $y(t) = C e^{At} x(0) + C \int_0^t e^{A(t-\tau)} B u(\tau) d\tau + Du(t)$

• let's see what output is produced by $u(t) = e^{st} \cdot u_0$, $s \in \mathbb{C}$, $u_0 \in \mathbb{R}^p$

$$y(t) = C e^{At} x(0) + C \int_0^t e^{A(t-\tau)} B [e^{s\tau} \cdot u_0] d\tau + D e^{st} \cdot u_0$$

$$= C \left[\int_0^t e^{A(t-\tau)} e^{s\tau} d\tau \right] B u_0$$

$$= e^{At} \cdot e^{-A\tau} \cdot e^{sI\tau} \leftarrow e^{s\tau} \cdot I = e^{sI\tau}$$

$$y(t) = C e^{At} x(0) + C e^{At} \left[\int_0^t e^{(sI-A)\tau} d\tau \right] \cdot B \cdot u_0 + D e^{st} \cdot u_0$$

→ evaluate this integral hint: find expression whose derivative is the integrand $e^{(sI-A)\tau}$

– recall that $\frac{d}{d\tau} e^{(sI-A)\tau} = (sI-A) e^{(sI-A)\tau}$

– so $\frac{d}{d\tau} \left[\underbrace{(sI-A)^{-1}} \cdot e^{(sI-A)\tau} \right] = e^{(sI-A)\tau} \quad \checkmark$

when is this matrix invertible?

$\Leftrightarrow \det(sI-A) \neq 0 \Leftrightarrow s \notin \lambda(A)$, i.e. s is not an eigenvalue of A

• so $\int_0^t e^{(sI-A)\tau} d\tau = \left[(sI-A)^{-1} e^{(sI-A)\tau} \right]_{\tau=0}^{\tau=t}$

$$= (sI-A)^{-1} \left[\underbrace{e^{(sI-A)t}} - \cancel{e^{0 \cdot I}} \right]$$

$$= e^{-At} \cdot e^{sIt} = e^{-At} \cdot e^{st}$$

hence

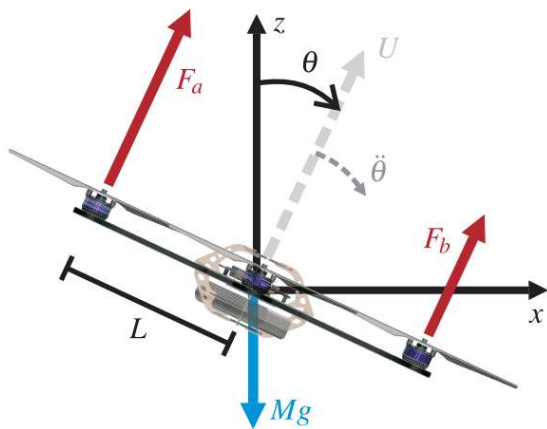
$$= u(t)$$

$$y(t) = \underbrace{C e^{At} (x(0) - (sI - A)^{-1} B u_0)}_{\text{transient response}} + \underbrace{[C(sI - A)^{-1} B + D] e^{st} \cdot u_0}_{\text{steady-state response}}$$

* assuming A stable: $\Rightarrow e^{At} \rightarrow 0$ as $t \rightarrow \infty$ $\rightarrow 0$ as $t \rightarrow \infty$ $G_{yu}(s) \in \mathbb{R}^{p \times p}$

takeaway: given $\dot{x} = Ax + Bu$ $y = Cx + Du$ get $G_{yu}(s) = C(sI - A)^{-1} B + D$
transfer matrix
(ie matrix of transfer functions)

ex: quadrotor



inputs $u = \begin{bmatrix} F_a \\ F_b \end{bmatrix} \in \mathbb{R}^2$ outputs: $\begin{bmatrix} x \\ z \\ \theta \end{bmatrix} = y \in \mathbb{R}^3$

$$G_{yu} = \begin{bmatrix} G_{x F_a} & G_{x F_b} \\ G_{z F_a} & G_{z F_b} \\ G_{\theta F_a} & G_{\theta F_b} \end{bmatrix} \in \mathbb{R}^{3 \times 2}$$

so if $u(t) = e^{st} \cdot u_0 \in \mathbb{R}^2$ then $y(t) = e^{st} \cdot G_{yu}(s) \cdot u_0 \in \mathbb{R}^3$

(b) Bode plots [AMv2 ch 9.6] [Nv7 ch 10.1]

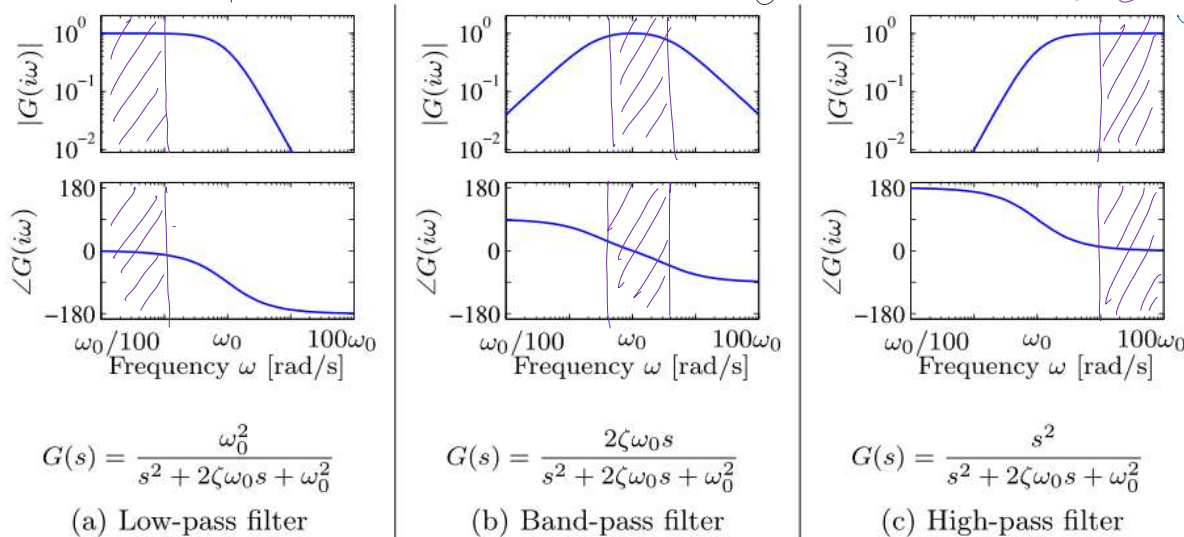
• since a complex exponential input $e^{st} \cdot u_0$ yields
complex exponential output $e^{st} \cdot G_{yu}(s) \cdot u_0$

and $\sin(\omega t) = \text{Im } e^{j\omega t}$, $\omega \in \mathbb{R}$ and $\cos(\omega t) = \text{Re } e^{j\omega t}$

then evaluating/visualizing $G_{yu}(j\omega)$ tells us how system responds to any input $\in \mathbb{C}$ physically-realizable (i.e. sin or cos)

• a Bode plot is a visualization of $|G_{yu}(j\omega)|$, $\angle G_{yu}(j\omega)$ vs ω

ex: Bode plots for second-order systems "pass" band



we usually use a logarithmic axis for gain but not phase

← also log axis for frequency

Figure 9.17: Bode plots for low-pass, band-pass, and high-pass filters. The upper plots are the gain curves and the lower plots are the phase curves. Each system passes frequencies in a different range and attenuates frequencies outside of that range.

* note: $Q(s) = \frac{b_1(s)b_2(s)}{a_1(s)a_2(s)}$

$$\Rightarrow \log |Q| = \log |b_1| + \log |b_2| - \log |a_1| - \log |a_2|$$

$$\Rightarrow \angle Q = \angle b_1 + \angle b_2 - \angle a_1 - \angle a_2$$

ex: (spring-mass-damper / RLC circuit)

$$m\ddot{g} + c\dot{g} + kg = u \text{ or } L\ddot{q} + R\dot{q} + \frac{1}{C}q = v$$

i.e. "passive" elements

$$m, c, k > 0$$

$$R, L, C > 0$$

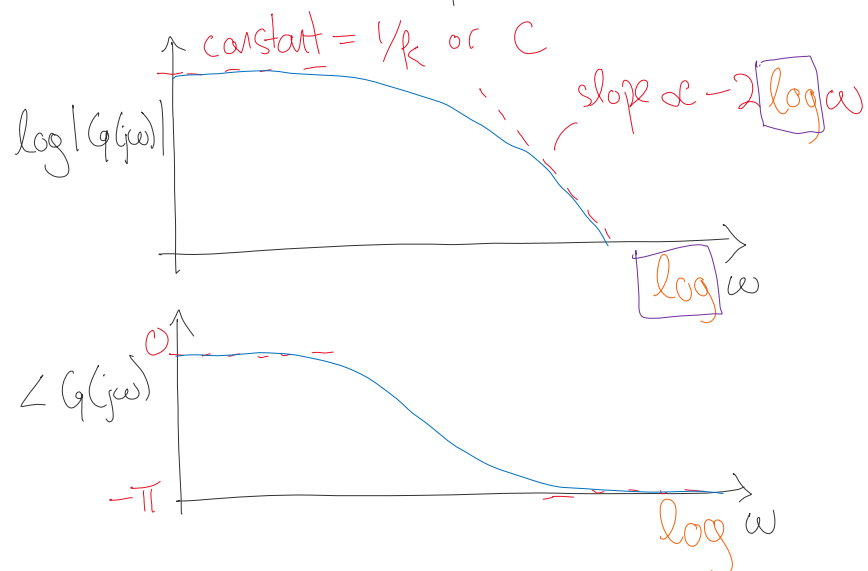
$$\leadsto Q(s) = \frac{1}{ms^2 + cs + k} \text{ or } = \frac{1}{Ls^2 + Rs + \frac{1}{C}}$$

$$\leadsto G(s) = \frac{1}{ms^2 + cs + k} \quad \text{or} \quad = \frac{1}{Ls^2 + Rs + 1/C}$$

note: as $s \rightarrow 0$, $G(s) \rightarrow 1/k$ or C , i.e. $|G(s)| \rightarrow \text{constant}$
 $\angle G(s) \rightarrow 0$

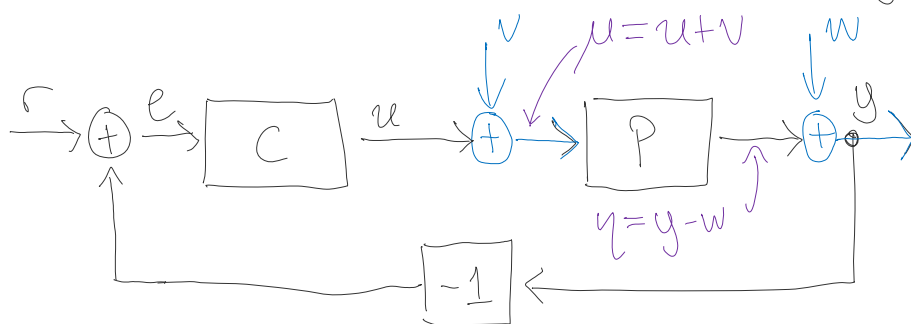
as $s \rightarrow \infty$, $G(s) \rightarrow 1/ms^2$ or $1/Ls^2$, i.e. $\log |G(j\omega)| \propto -2\omega$
 $\angle G(s) \rightarrow \pm \pi$

so we can sketch Bode plot:



(c) effect of disturbances [AMv2 Ch 12.1] [Nu7 Ch 7.5]

• consider negative feedback block diagram with disturbances:



r - reference; ideally $r \rightarrow y$
 v - input disturbance, eg actuator noise
 w - output disturbance, eg sensor noise

* assuming input disturbance v and output disturbance w

* assuming input disturbance v and output disturbance w are independent of other signals in the diagram, they affect y linearly:

→ derive transfer functions G_{yr} , G_{yv} , G_{yw}
so that $y = G_{yr} \cdot r + G_{yv} \cdot v + G_{yw} \cdot w$

$$\begin{aligned} - y &= w + \eta = w + P\mu = w + P(v + u) = w + Pv + PCe \\ &= w + Pv + PC(r - y) \end{aligned}$$

$$\Leftrightarrow (I + PC)y = w + Pv + PCr$$

* assuming invertible

$$\Leftrightarrow y = (I + PC)^{-1}w + (I + PC)^{-1}Pv + (I + PC)^{-1}PCr$$

in single-input/
single-output case:

$$= \underbrace{\frac{G_{yw}}{1+PC}}_1 w + \underbrace{\frac{G_{yv}}{1+PC}}_P v + \underbrace{\frac{G_{yr}}{1+PC}}_{PC} r$$

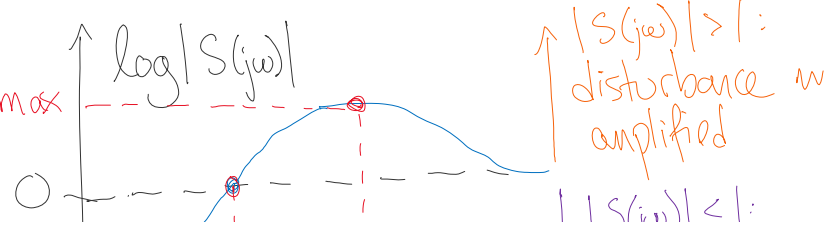
* the "ideal" transformations are different for each transformation:

- ideally $G_{yr} \simeq I$ so that $y = G_{yr} \cdot r \simeq r$

- ideally $G_{yv}, G_{yw} \simeq 0$ so that v & w don't affect y

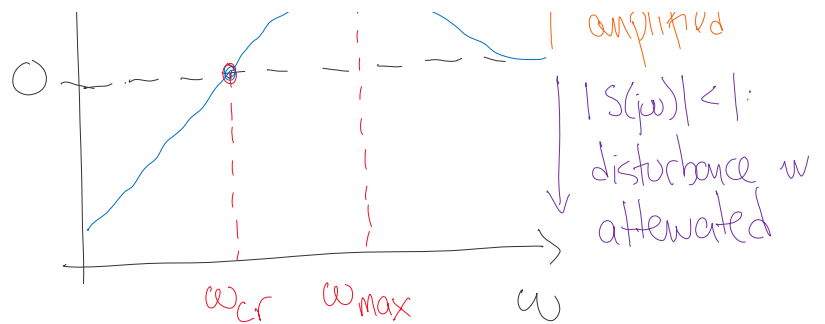
(we'll see that we can't achieve all "ideals" at the same time)

in terms of Bode plot: $|S|_{\max}$



let $S = G_{yw}$ be

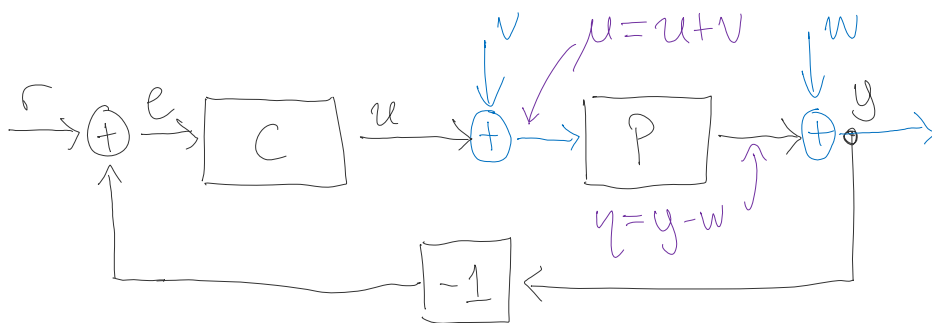
let $S = G_{yw}$ be
sensitivity of closed-loop system



- performance of closed-loop system can be assessed using
peak sensitivity $|S|_{\max}$ or crossover frequency ω_{cr}

(d) fundamental limits [AMv2 Ch 12.1, 14.2]

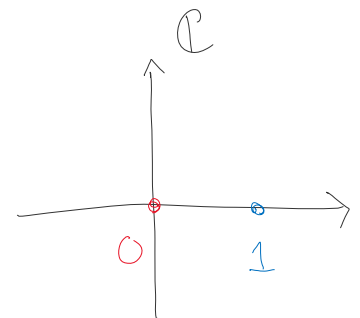
• consider the negative feedback block diagram with disturbances



recall that

$$y = G_{yr} \cdot r + G_{yv} \cdot v + G_{yw} \cdot w$$

$$= \frac{PC}{1+PC} \cdot r + \frac{P}{1+PC} \cdot v + \frac{1}{1+PC} \cdot w$$



and we want: 1°. disturbances rejected, i.e. $G_{yv}, G_{yw} \simeq 0$

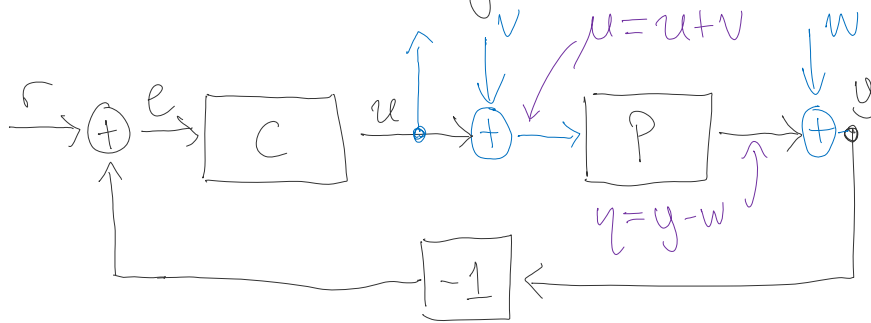
2°. references tracked, i.e. $G_{yr} \simeq 1$

- since $G_{yw} + G_{yr} = \frac{1}{1+PC} + \frac{PC}{1+PC} = \frac{1+PC}{1+PC} = 1$

so $G_{yr} \simeq 1 \iff G_{yw} \simeq 0$, which seems great!

→ but this happy coincidence is misleading...

- return to block diagram and consider effect of disturbances v, w on input u



→ determine G_{ur}, G_{uv}, G_{uw} s.t. $u = G_{ur} \cdot r + G_{uv} \cdot v + G_{uw} \cdot w$

$$- u = C(r - y) = Cr - C(w + Pu) = Cr - Cw - CP(u + u)$$

$$\Leftrightarrow (1 + PC)u = Cr - PCv - Cw$$

$$\Leftrightarrow u = \frac{C}{1+PC} r - \frac{PC}{1+PC} v - \frac{C}{1+PC} w$$

$$= G_{ur} \cdot r + G_{uv} \cdot v + G_{uw} \cdot w$$

recall:

$$y = G_{yr} \cdot r + G_{yv} \cdot v + G_{yw} \cdot w$$

$$= \frac{PC}{1+PC} \cdot r + \frac{P}{1+PC} \cdot v + \frac{1}{1+PC} \cdot w$$

observe: $G_{yw} - G_{uv} = \frac{1}{1+PC} + \frac{PC}{1+PC} = \frac{1+PC}{1+PC} = 1$

but we want to reject disturbances, i.e. $G_{yw}, G_{uv} \approx 0$

we can't have both! ✓

→ there's a tradeoff

• in many systems, frequency content of: ex: cruise control

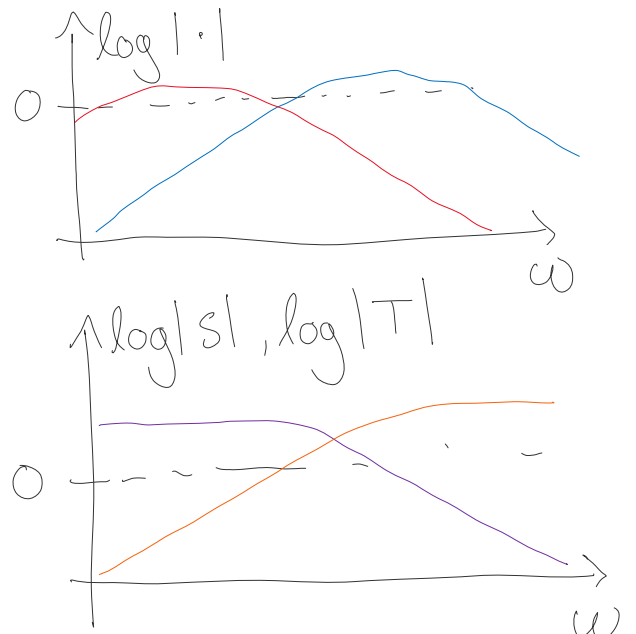
- in many systems, frequency content of: ex: cruise control
input disturbance v is low \leftarrow wind pushing on car
output disturbance w is high \leftarrow sensor noise

so if we look at Bode plots:

which means we can design

$$S = \frac{1}{1+PC} = G_{yw} \text{ (sensitivity)}$$

$$T = 1 - S = \frac{PC}{1+PC} = -G_{uv} \text{ (complementary sensitivity)}$$



to design frequency-dependent (complementary) sensitivity

- it turns out there are limits on how we can reshape S :

thm: (Bode integral formula / argument principle)

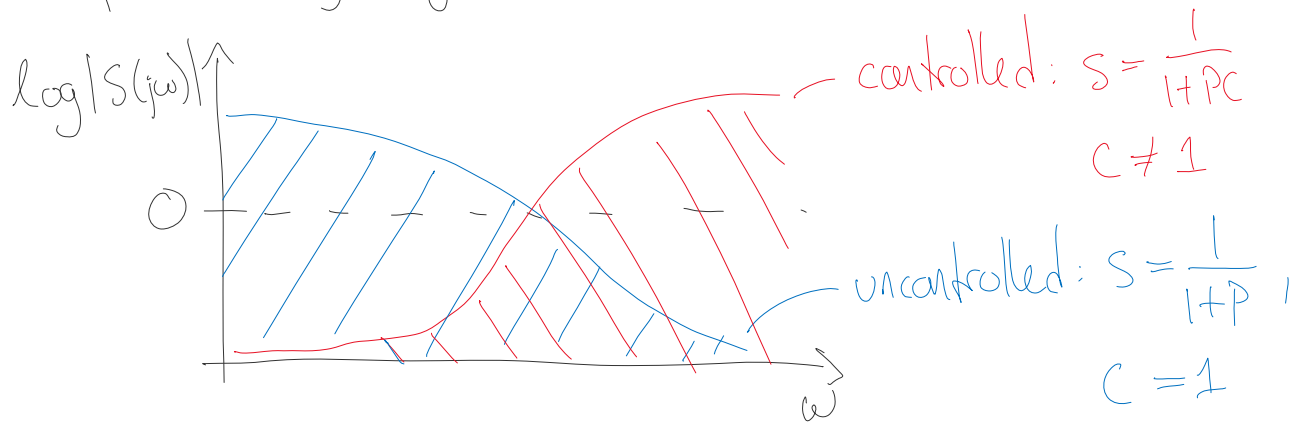
$$\int_0^{\infty} \log |S(j\omega)| d\omega \geq \text{constant, independent of } C$$

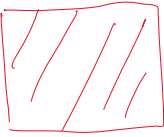
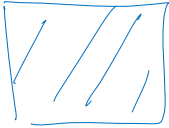
$$= \pi \cdot \sum \{ \operatorname{Re} p \mid p \text{ is a pole of } P \text{ in right-half plane} \}$$

$$= 0 \text{ if process is stable}$$

- * since $\log |S(j\omega)| < 0 \Leftrightarrow |S(j\omega)| < 1 \Leftrightarrow$ disturbance attenuated
any range of frequencies where controller attenuates disturbance
is a range of frequencies where controller amplifies disturbance

any range of frequencies where controller attenuates disturbance
is compensated by range where controller amplifies disturbance



thm:  =  ← we can't create or destroy "area" — only shift it