18 Euler Angles

How do we typically parameterize SO(3)—the space of rigid body rotations? Typically, particularly with mechanisms,⁵ we use Euler angles, defined by:

$$R_Z = e^{\theta \hat{\omega}_Z}$$
 $R_X = e^{\psi \hat{\omega}_X}$ $R_Y = e^{\phi \hat{\omega}_Y}$

where

$$\omega_Z = \left[egin{array}{c} 0 \\ 0 \\ 1 \end{array}
ight] \quad \omega_X = \left[egin{array}{c} 1 \\ 0 \\ 0 \end{array}
ight] \quad \omega_Y = \left[egin{array}{c} 0 \\ 1 \\ 0 \end{array}
ight]$$

Note that Euler angles do not commute:

$$R_X R_Y R_Z \neq R_Y R_X R_Z$$

but both of these are common choices for parameterizing SO(3).

Example: 3D mechanism

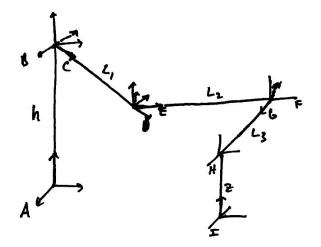


Figure 25: 3D mechanism

 $g_{AI} = g_{AB}g_{BC}g_{CD}g_{DE}g_{EF}g_{FG}g_{GH}g_{HI}$

$$=\begin{bmatrix} I & \begin{bmatrix} 0 \\ 0 \\ h \end{bmatrix} \\ 0 & 1 \end{bmatrix} \begin{bmatrix} R_Z(\theta_1) & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} I & \begin{bmatrix} L_1 \\ 0 \\ 0 \end{bmatrix} \\ 0 & 1 \end{bmatrix} \begin{bmatrix} R_Z(\theta_2) & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} I & \begin{bmatrix} L_2 \\ 0 \\ 0 \end{bmatrix} \\ 0 & 1 \end{bmatrix} \begin{bmatrix} R_Z(\theta_3) & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} I & \begin{bmatrix} L_3 \\ 0 \\ 0 \end{bmatrix} \\ 0 & 1 \end{bmatrix} \begin{bmatrix} I & \begin{bmatrix} 0 \\ 0 \\ z \end{bmatrix} \\ 0 & 1 \end{bmatrix}$$

Also,
$$q = (\theta_1, \theta_2, \theta_3, z)$$
 and $\frac{d}{dt} \begin{bmatrix} I & \begin{bmatrix} 0 \\ 0 \\ z \end{bmatrix} \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} 0 & \begin{bmatrix} 0 \\ 0 \\ \dot{z} \end{bmatrix} \\ 0 & 0 \end{bmatrix}$ and $\frac{d}{dz} \begin{bmatrix} I & \begin{bmatrix} 0 \\ 0 \\ z \end{bmatrix} \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} 0 & \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \\ 0 & 0 \end{bmatrix}$

⁵In other instances, like modeling satellites in orbit, Euler angles can be disasterous because of singularities in their parameterization of SO(3), but for our purposes they will be fine for almost everything we might come up with.