# AIRCRAFT FLIGHT MODEL VARIABLES

The tables below indicate the properties for the <u>Simulation Variables</u> that can be used to get and set properties related to the physical properties and flight model of an aircraft. For information on the units listed for each variable, please see here: <u>Simulation Variable Units</u>

**NOTE**: When Microsoft Flight Simulator is running in multiplayer mode, only a small number of variables are communicated between aircraft. Those variables that are available will say so in the description as being either for "All Aircraft" or for "Shared Cockpit".

You can find a complete index of all available SimVars here: SimVar Index

#### **General**

Simulation Variable	Description	Units	Settable
BETA DOT	Beta dot	Radians per second	
DECISION ALTITUDE MSL	Design decision altitude above mean sea level	Feet (ft)	
DECISION HEIGHT	Design decision height	Feet (ft)	
DESIGN CRUISE ALT	This design constant represents the optimal altitude the aircraft should maintain when in cruise. It is derived from the cruise_alt setting in the [REFERENCE SPEEDS] section of the flightmodel.cfg. Default is 1500ft.	Feet ( <i>ft</i> )	

DESIGN SPAWN ALTITUDE CRUISE	This design constant represents the spawn altitude for the aircraft when spawning in cruise. It is derived from the spawn_cruise_altitude setting in the [REFERENCE SPEEDS] section of the flightmodel.cfg. Default is 1500ft.	Feet ( <i>ft</i> )	
DESIGN SPAWN ALTITUDE DESCENT	This design constant represents the spawn altitude for the aircraft when spawning in descent. It is derived from the spawn_descent_altitude setting in the [REFERENCE SPEEDS] section of the flightmodel.cfg. Default is 500ft.	Feet (ft)	
DESIGN SPEED	This design constant represents the optimal climb speed for the aircraft. It is derived from the <code>climb_speed</code> setting in the <code>[REFERENCE SPEEDS]</code> section of the <code>flightmodel.cfg</code> . Default value is -1.	Feet ( <i>ft</i> ) per second	
DESIGN SPEED MIN ROTATION	This design constant represents the minimum speed required for aircraft rotation. It is derived from the rotation_speed_min setting in the [REFERENCE SPEEDS] section of the flightmodel.cfg. Default value is -1.	Feet ( <i>ft</i> ) per second	
DESIGN SPEED VC	This design constant represents the aircraft ideal cruising speed. It is derived from the cruise_speed setting in the [REFERENCE SPEEDS] section of the flightmodel.cfg. The default value is computed an internal function that uses the estimated cruise altitude and estimated cruise percent power,	Feet ( <i>ft</i> ) per second	

	according of the engine type, the number of engines, the density, the wing area and some drag parameters.  Normally this value is set in the CFG file and the default value is never used.		
DESIGN SPEED VS0	This design constant represents the the stall speed when flaps are fully extended. It is derived from the full_flaps_stall_speed setting in the [REFERENCE SPEEDS] section of the flightmodel.cfg. Default value is 0.8 x VS.	kias	
DESIGN SPEED VS1	This design constant represents the stall speed when flaps are fully retracted. It is derived from the flaps_up_stall_speed setting in the [REFERENCE SPEEDS] section of the flightmodel.cfg. Default value is 0.	kias	
DESIGN TAKEOFF SPEED	This design constant represents the aircraft ideal takoff speed. It is derived from the takeoff_speed setting in the [REFERENCE SPEEDS] section of the flightmodel.cfg.	Knots	
DYNAMIC PRESSURE	Dynamic pressure	Pounds per square foot (psf)	
ESTIMATED CRUISE SPEED	Estimated cruise speed	Feet ( <i>ft</i> ) per second	
G FORCE	Current g force	GForce	

G LIMITER SETTING	This returns the setting of the G-limiter, as set using the GLimiterSetting parameter.	Enum:  0 = Off  1 = On  2 =  Override
INCIDENCE	Angle of attack	Radians
INCIDENCE BETA	Sideslip angle	Radians
IS TAIL DRAGGER	True if the aircraft is a taildragger	Bool
LINEAR CL ALPHA	Linear CL alpha	Per radian
MACH MAX OPERATE	Maximum design mach	Mach
MAX G FORCE	Maximum G force attained	Gforce
MIN DRAG VELOCITY	Minimum drag velocity, in clean, with no input and no gears, when at 10000ft.	Feet ( <i>ft</i> ) per second
MIN G FORCE	Minimum G force attained	Gforce
SEMIBODY LOADFACTOR X Deprecated	Deprecated, do not use!	Number
SEMIBODY LOADFACTOR Y	Acceleration along the axis Y divided by the gravity constant g (usually around 9.81m.s²)	Number
SEMIBODY LOADFACTOR YDOT	Derivative of SEMIBODY LOADFACTOR Y in relation to time.	Per second
SEMIBODY LOADFACTOR Z Deprecated	Deprecated, do not use!	Number
SIGMA SQRT	Sigma sqrt	Number

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SIMULATED	Simulated radius	Feet (ft)	
STALL ALPHA	The angle of attack which produces the maximum lift coefficient before entering into stall conditions.	Radians	
STATIC PITCH	The angle at which static pitch stability is achieved.	Radians	
TYPICAL DESCENT RATE	the typical (normal) descent rate for the aircraft.	Feet ( <i>ft</i> ) per minute	
WING AREA	Total wing area	Square feet (ft)	
WING FLEX PCT:index	The current wing flex. Different values can be set for each wing (for example, during banking). Set an index of 1 for the left wing, and 2 for the right wing.	Percent Over 100	
WING SPAN	Total wing span	Feet (ft)	
YAW STRING ANGLE	The yaw string angle. Yaw strings are attached to gliders as visible indicators of the yaw angle. An animation of this is not implemented in ESP.	Radians	
YAW STRING PCT EXTENDED	Yaw string angle as a percentage	Percent Over 100	
ZERO LIFT	The angle of attack at which the wing has zero lift.	Radians	

## **Center Of Gravity**

Simulation Variable	Description	Units	Settable
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CG AFT LIMIT	Most backward authorized position of the <i>CG</i> according to the <i>POH</i> .  NOTE: This is only valid for airplanes.	Percent Over 100
CG FEET	The longitudinal <i>CG</i> position relative to the <u>Reference</u> <u>Datum Position</u> .  **NOTE: This is only valid for helicopters.	Feet (ft)
CG FEET AFT LIMIT	The aft <i>CG</i> limit position relative to the <u>Reference</u> <u>Datum Position</u> .  NOTE: This is only valid for helicopters.	Feet (ft)
CG FEET LATERAL	The lateral <i>CG</i> position relative to the <u>Reference</u> <u>Datum Position</u> .  NOTE: This is only valid for helicopters.	Feet (ft)
CG FEET LATERAL LEFT LIMIT	The left hand lateral <i>CG</i> position relative to the Reference Datum Position.  NOTE: This is only valid for helicopters.	Feet (ft)
CG FEET LATERAL RIGHT LIMIT	The right hand lateral <i>CG</i> position relative to the Reference Datum Position.	Feet (ft)

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	<b>NOTE:</b> This is only valid for <b>helicopters</b> .		
CG FEET FWD LIMIT	The forward <i>CG</i> limit position relative to the <u>Reference</u> <u>Datum Position</u> .  NOTE: This is only valid for helicopters.	Feet ( <i>ft</i> )	
CG FWD LIMIT	Most forward authorized position of the <i>CG</i> according to the <i>POH</i> .  NOTE: This is only valid for airplanes.	Percent Over 100	
CG MAX MACH Deprecated	Deprecated, do not use!	Mach	
CG MIN MACH Deprecated	Deprecated, do not use!	Mach	
CG PERCENT	Longitudinal <i>CG</i> position as a percent of reference <i>Chord</i> .  NOTE: This is only valid for airplanes.	Percent Over 100	
CG PERCENT LATERAL	Lateral <i>CG</i> position as a percent of reference <i>Chord</i> .  NOTE: This is only valid for airplanes.	Percent Over 100	

	Static <i>CG</i> position with reference to the ground.		
GROUND	<b>NOTE:</b> This is only valid for <b>airplanes</b> .	Feet ( <i>ft</i> )	

#### **Interactive Points**

Simulation Variable	Description	Units	Settable
INTERACTIVE POINT BANK	Interactive Point orientation: Bank	Degrees	
INTERACTIVE POINT HEADING	Interactive Point orientation: Heading	Degrees	
INTERACTIVE POINT JETWAY LEFT BEND	Interactive Point Jetway constant, determining the desired left bend ratio of jetway hood	Percent	
INTERACTIVE POINT JETWAY LEFT DEPLOYMENT	Interactive Point Jetway constant, determining the desired left deployment angle of jetway hood	Degrees	
INTERACTIVE POINT JETWAY RIGHT BEND	Interactive Point Jetway constant, determining the desired right bend ratio of jetway hood	Percent	
INTERACTIVE POINT JETWAY RIGHT DEPLOYMENT	Interactive Point Jetway constant, determining the desired right deployment angle of jetway hood	Degrees	

INTERACTIVE POINT JETWAY TOP HORIZONTAL	Interactive Point Jetway constant, determining the desired top horizontal ratio of displacement of jetway hood	Percent
INTERACTIVE POINT JETWAY TOP VERTICAL	Interactive Point Jetway constant, determining the desired top vertical ratio of displacement of jetway hood	Percent
INTERACTIVE POINT GOAL	The Interactive Point goal percentage of opening (if it's for a door) or percentage of deployment (if it's for a hose or cable).	Percent Over 100
INTERACTIVE POINT OPEN	Interactive Point current percentage of opening (if door) or deployment (if hose/cable)	Percent Over 100
INTERACTIVE POINT PITCH	Interactive Point orientation: Pitch	Degrees
INTERACTIVE POINT POSX	Interactive Point X position relative to datum reference point	Feet (ft)
INTERACTIVE POINT POSY	Interactive Point Y position relative to datum reference point	Feet ( <i>ft</i> )
INTERACTIVE POINT POSZ	Interactive Point Z position relative to datum reference point	Feet ( <i>ft</i> )
INTERACTIVE POINT TYPE	The type of interactive point	Enum:  0: Main exit  1: Cargo exit  2: Emergency exit

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		3: Fuel hose	
		4: Ground Power cable	
		99: Unknown (for error handling)	

## Weight

Simulation Variable	Description	Units	Settable
EMPTY WEIGHT	Empty weight of the aircraft	Pounds	
EMPTY WEIGHT CROSS	Empty weight cross coupled moment of inertia	Slugs per feet squared ( <i>Slug sqft</i> )	
EMPTY WEIGHT PITCH	Empty weight pitch moment of inertia	Slugs per feet squared ( <i>Slug sqft</i> )	
EMPTY WEIGHT ROLL	Empty weight roll moment of inertia	Slugs per feet squared ( <i>Slug sqft</i> )	
EMPTY WEIGHT YAW	Empty weight yaw moment of inertia	Slugs per feet squared ( <i>Slug sqft</i> )	
MAX GROSS WEIGHT	Maximum gross weight of the aircaft	Pounds	
TOTAL METCHT	Total weight of the aircraft	Dounde	