





FEniCS-Shells:

A modern open-source extensible finite element implementation of linear and nonlinear plate and shell models

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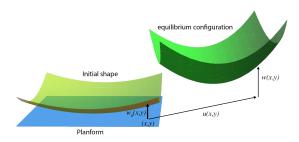
Plates and shells: deformable surfaces



In thin shells the deformations are small and the behaviour is usually linear elastic.

Non-linear complex phenomena are due to geometry and shape.

Shell modelling



Numerical modelling of thin shells is difficult:

- Singular perturbation problem (thickness $t \rightarrow 0$)
- \blacksquare H^2 regularity requirement
- Differential geometry: PDE on surfaces
- Competition between membrane and stretching, with possible change of the PDE character with the sign of the Gaussian curvature
- Complex nonlinearities phenomena: buckling, failure, stress-focusing, point and line localisations.

Objectives of FEniCS-shells

- Provide an extensible open-source collection of high-quality numerical methods for thin structures
- Translate complex multiphysical problems on surface to performing scalable numerical implementation through simple formulations in FEniCS/ UFL
- Use in research (singularities in shells, multistable shells, fracture in shells) and teaching (structural mechanics, advanced numerical methods)

Based on FEniCS

The FEniCS Project is a collection of free software with an extensive list of features for automated, efficient solution of differential equations.

https://fenicsproject.org

FEniCS-shells: Implemented models

√	Reissner-Mindlin plate	Shearable •	Bending	Linear •	Nonlinear
✓	von Kármán plate	•	•		•
\checkmark	Marguerre shallow shell	•	•		•
\checkmark	Naghdi shell	•			•

- Shearable models implemented through MITC/Duran-Liberman formulation [Dvorkin and Bathe 1986, Duran and Liberman 1992] in order to avoid numerical locking
- Bending models implemented through C/D Galerkin formulation [Engel et al., 2002] in order to avoid H^2 -finite elements

Other non considered modern alternatives: subdivision of surfaces, IGA, meshless

The "simple" linear plate theory

Non-trivial numerical problems arise also for flat shells (plates) and linear models

Reissner-Mindlin (RM) plate model

- $w : \Omega \to R$: transverse displacement (scalar)
- lacksquare $\theta:\Omega o R^2$ rotation of transverse segments (vector)

$$\mathcal{E}_{\mathsf{RM}}(\theta, w) = \frac{1}{2} \int_{\Omega} D \nabla^{\mathsf{s}} \theta : \underbrace{\nabla^{\mathsf{s}} \theta}_{\mathsf{curvature}} + \frac{t^{-2}}{2} \int_{\Omega} F \| \underbrace{\nabla w - \theta}_{\mathsf{shear}} \|^2 - \mathcal{L}_{\mathsf{e}}(w, \theta)$$

Kirchhoff-Love (KL) plate model ($t \rightarrow 0$)

$$\mathcal{E}_{\mathsf{KL}}(w) = \frac{1}{2} \int_{\Omega} D \nabla \nabla w : \nabla \nabla w - \mathcal{L}_{e}(w)$$

Remark: when $t \to 0$ the RM-model asymptotically converges to the KL-model with $\nabla w = \theta$.

Shear locking in the discrete problem in the RM model

Minimisation problem for the equilibrium of a plate of thickness t:

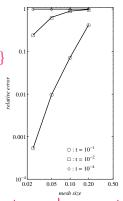
$$\min_{\theta,w \in V} \left\{ \frac{1}{2} a(\theta,\theta) + \frac{t^{-2}}{2} \int_{\Omega} \|\nabla w - \theta\|^2 - \mathcal{L}_e(w) \right\}, \ V = H^1(\Omega,R^2) \times H^1(\Omega,R)$$

For $t \to 0$ the limit problem is

$$\min_{\theta,w\in K} \left\{ \frac{1}{2} a(\theta,\theta) - \mathcal{L}_{e} \right\}, K = \left\{ (\theta,w) \in V : \nabla w = \theta \right\}$$

while the corresponding discrete problem has limit problem

$$\min_{\theta_h, w_h \in K_h} \left\{ \frac{1}{2} a(\theta_h, \theta_h) - \mathcal{L}_e \right\}, K_h = K \cap (\boldsymbol{\Theta}_h \times W_h)$$



If $(\Theta_h \times W_h)$ is not large enough the basis functions cannot properly represent the Kirchhoff's constraint $\nabla w = \theta$ and we can have $K_h \equiv \emptyset$.

A well-established locking cure: MITC formulation

The MITC formulation recast the mixed formulation into a displacement form

$$\min_{\theta_h, w_h \in U_h} \left\{ \frac{1}{2} a(\theta_h, \theta_h) + \frac{t^{-2}}{2} \int_{\Omega} \| \frac{R_h}{\gamma_h} (\nabla w_h - \theta_h) \|^2 - L_e \right\}$$

where the shear term is assembled using a reduction operator.

The element MITC3 (Bathe et al.) adopts an element-wise interpolation:

$$R_h \gamma_h = \left[\begin{array}{c} a - b y \\ a + c x \end{array} \right]$$

where the constants (a, b, c) are determined by imposing

$$R_h \gamma_h(x_i, y_i) = \gamma_h(x_i, y_i), \quad i = 1, 2, 3$$

at three tying points on the middle of the element edges (x_i, y_i)

Revisiting MITC as a mixed formulation

The locking issue can be solved also with the following mixed formulation

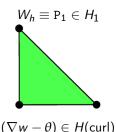
$$\operatorname{stat}\{\mathcal{E}, (\theta, u, \gamma, \lambda) \in \Theta_h \times W_h \times \Gamma_h \times \Gamma_h\}$$

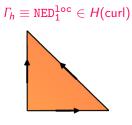
with

$$\mathcal{E}(\theta, u, \gamma, \lambda) = \left\{ \frac{1}{2} \mathsf{a}(\theta, \theta) + \frac{t^{-2}}{2} \int_{\Omega} \| \boldsymbol{\gamma} \|^2 - \mathcal{L}_{e}(w) + \int_{\Omega} \lambda \cdot ((\nabla w - \theta) - \boldsymbol{\gamma}) \right\}$$

Duran-Liberman [Duran, Liberman, 1992] propose the following stable and uniformly convergent choice

$$\Theta_h \equiv \mathtt{P}_2^2 \in H_1^2$$





Linear algebra and reduced formulation

Full space linear algebra formulation

$$\begin{bmatrix} A_{pp} & A_{pq} \\ A_{pq}^{\mathsf{T}} & A_{qq} \end{bmatrix} \begin{bmatrix} U_p \\ U_q \end{bmatrix} = \begin{bmatrix} F_p \\ F_q \end{bmatrix}, \qquad \begin{cases} U_p = (\theta_h, w_h) \in P_2^2 \times P_1 \\ U_q = (\gamma_h, \lambda_h) \in NED_1^{\mathsf{loc}} \times NED_1^{\mathsf{loc}} \end{cases}$$

Linear algebra formulation after elimination

$$A_R U_P = F_R, \quad \begin{cases} A_R = A_{pp} - A_{pq} A_{qq}^{-1} A_{pq}^T \\ F_R = F_p - A_{pq} A_{qq}^{-1} 1 F_q \end{cases}$$

- The reduction can be done as a static condensation at the element level.
- This kind of variational formulation is perfect for FEniCS. However static condensation is not possible because automatic code generation and assembling implementation.

FEniCS implementation

After the new/overloaded functions and classes of FEniCS-shells

■ Define a FunctionSpace with the partition in primal and local variable

■ Variational formulation including the constraint in the full space

```
kappa_ = grad(theta_); gamma_ = grad(w_) - theta_
bending_energy = (t**3*Et*((1.0 - nu)*tr(kappa_*kappa_) + nu*(tr(kappa_))**2))*dx
shear_energy = (t*Et*Constant(5.0/6.0)*inner(gamma_,gamma_)*dx
external_work = inner(Constant(1.0)*t**3, w_t)*dx
pontential_energy = bending_energy + shear_energy - external_work
constraint = inner_e(gamma_ - R_gamma_, p_)
lagrangian = pontential_energy + constraint
d1_lagrangian = derivative(lagrangian, u_, u_t)
L = - replace(d1_lagrangian, u_, u) # linear form
a = derivative(d1_lagrangian, u_, u) # quadratic form
```

■ Assemble with static condensation and solve

```
A, b = assemble(U_P, a, L)
solver.solve(A, u_p_.vector(), b)
reconstruct_full_space(u_, u_p_, a, L)
```

Implementing the static condensation in FEniCS

```
U = ProjectedFunctionSpace(mesh, element, num_projected_subspaces=2)
U_F = U.full_space
U_P = U.projected_space
A, b = assemble(U_P, a, L)
```

For e in elements:

- \blacksquare Assemble the element matrices A_{pp}^e , A_{pq}^e , A_{qq}^e , F_p^e , F_q^e in the full space
- Calculate through Eigen (dense linear algebra) the local static condensation

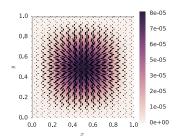
$$\begin{array}{lll} \textbf{A}_{R}^{e} & \leftarrow & \textbf{A}_{pp}^{e} - \textbf{A}_{pq}^{e} \textbf{A}_{qq}^{e-1} \textbf{A}_{pq}^{eT} \\ \textbf{F}_{R}^{e} & \leftarrow & \textbf{F}_{p}^{e} - \textbf{A}_{pq}^{e} \textbf{A}_{qq}^{e-1} \textbf{F}_{q}^{e} \end{array}$$

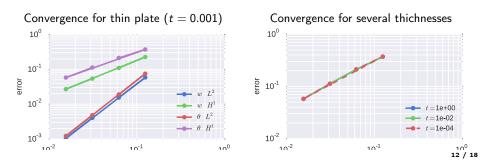
3 Add to global stiffness on the reduced space using the dofmap of U_P

Note: As an option, $A, b = assemble(U_F, a, L)$ assemble the matrix on the full space without static condensation.

RM plate: convergence

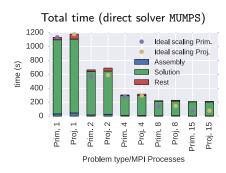
Clamped plate test (analytical solution by [Lovadina, 1995])

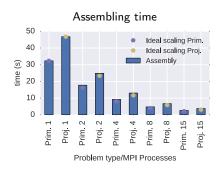




Assembling performance

Strong scaling with $\sim 2e^6$ elements, $\sim 10e^6$ dofs





Comments

- Assembling time is negligible
- Assembling scales almost perfectly
- Iterative solvers/preconditioners are needed (difficult issue)

Extension to nonlinear shells

Enjoy FEniCS/UFL syntax to write the variational formulation

```
# Naghdi strain measures
# Director vector and
d = lambda theta: as_vector([sin(theta[1])*cos(theta[0]), -sin(theta[0]), cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1])*cos(theta[1]
# Deformation gradient and strain measures
F = lambda u: as_tensor([[1.0, 0.0],[0.0, 1.0],[Constant(0), Constant(0)]]) + grad(u)
G = lambda FO: 0.5*(FO.T*FO - Identity(2)) # Stretching tensor (1st Naghdi strain measure)
K = lambda F0, d: 0.5*(F0.T*grad(d) + grad(d).T*F0) # Curvature tensor (2nd Naghdi strain meas
g = lambda FO, d: FO.T*d # Shear strain vector (3rd Naghdi strain measure)
# Dual stress tensor (constitutive law)
S = lambda X: 2.0*mu*X + ((2.0*mu*lb)/(2.0*mu + lb))*tr(X)*Identity(2)
# Energies
psi_G = (2.0 * eps) * inner(S(G(F(z_))), G(F(z_))) # Membrane energy density
psi_K = ((2.0*eps)**3/12.0)*inner(S(K(F(z_), d(theta_))), K(F(z_), d(theta_))) # Bending energy
psi_g = eps*2.0*mu*inner(Rgamma_, Rgamma_) # Shear energy density
psi = 0.5*(psi_G + psi_K + psi_g)*dx # Stored strain energy density
# constraint
L_R = inner_e(g(F(z_), d(theta_)) - Rgamma_, p_)
```

in the projected space a customised NonlinearProblem using the condensed assembler

```
problem = ProjectedNonlinearProblem(U_P, F, u_, u_p_, bcs=bcs, J=J)
solver = NewtonSolver()
solver.solve(problem, u_p_.vector())
```

Large transformations of nonlinear shells

Pulling of a slit annular plate Rolling of a strip 1.0 F Outer point 0.2

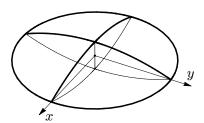
15

Vertical displacement

 w_C

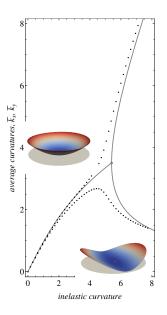
Bistable disc with spherical target curvature

Circular plate with lenticular cross section subjected to a temperature gradient through its thickness



Average curvatures at bifurcation [Mansfield, 1962]

$$\kappa_T = \frac{8}{(1+\nu)^{3/2}}$$



Adaptive remeshing

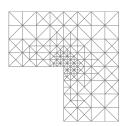
Thick plate

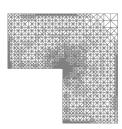






Thin plate





Conclusions an perspectives

- HPC open-source extensible tool for shell computations
- Projected and full formulations have similar performances
- Projected formulation is better for memory.
- Projected formulation is practical for stability analysis
- Full formulation is probably better for preconditioners (todo)
- Higher order MITC7 are possible (MITC7), the advantage of static condensation would be more evident
- Working on curvilinear coordinate mapping
- Aim at applications on growth, morphing structures, fracture with phase-field models, shape control, multiphysical coupling, analysis of singular structures in shells (d-cone, streching ridge)

Codes available at

https://bitbucket.org/unilucompmech/fenics-shells

We provide functional docker containers to run the application!

Mail to corrado.maurini@upmc.fr if interested in use/collaboration