

$$① \quad C_{L_{aw}} = 4.1 \text{ 1/rad} \cdot S_H = 45.2 \text{ ft}^2, \quad C_{L_{aH}} = 4.1 \text{ 1/rad}, \quad C_{m_a} = -1.59 \text{ 1/rad}$$

$$\bar{x}_{cg} = 0.5, \quad S = 169 \text{ ft}^2, \quad \bar{c} = 5.825 \text{ ft}, \quad \bar{x}_{acH} = 3.5, \quad \bar{x}_{acWB} = 0.6$$

$$\frac{d\epsilon}{d\alpha} = 0.5, \quad \eta_H = 0.95$$

$$② \quad SM = \frac{-C_{m_a}}{C_{L_a}} = \frac{-(-1.59)}{4.59} = 0.3464 \rightarrow 34.64\%$$

$$③ \quad SM' = 0.2, \quad C_{m_a}' = -SM' \cdot C_{L_a}'$$

$$C_{L_{\alpha \text{ tail}}} = C_{L_{aH}} \eta_H \frac{S_H}{S} = (4.1)(0.95) \left( \frac{45.2}{169} \right) = 1.03995 \text{ 1/rad}$$

$$C_{L_{\alpha}} = C_{L_{aWBH}} + C_{L_{aH}} \eta_H \frac{S_H}{S}, \quad C_{L_{aWBH}} = C_{L_a} - C_{L_{\alpha \text{ tail}}} = 4.59 - 1.03995$$

$$C_{m_{aH}} = -C_{L_{aH}} \eta_H \frac{S_H}{S} \cancel{4.1(0.95)} (\bar{x}_{acH} - \bar{x}_{cg}) \left( 1 - \frac{d\epsilon}{d\alpha} \right) = 3.55005$$

$$C_{m_{aH}} = -1.56 \text{ 1/rad}, \quad C_{m_{aWB}} = C_{m_a} - C_{m_{aH}} = -0.03 \text{ 1/rad}$$

$$\bar{x}_{acWB} = 0.6, \quad C_{L_{aw}} = 4.1, \quad C_{m_{aWB}} = -0.41$$

$$C_{L_a}' = C_{L_{aWBH}} + C_{L_{aH}} \eta_H \frac{S_H'}{S} = 3.55 + 0.023 S_H'$$

$$C_{m_{aH}}' = -C_{L_{aH}} \eta_H \frac{S_H'}{S} (\bar{x}_{acH} - \bar{x}_{cg}) \left( 1 - \frac{d\epsilon}{d\alpha} \right) = -0.03 S_H'$$

$$C_{m_a}' = C_{m_{aWB}} + C_{m_{aH}}' = -0.03 - 0.0345 S_H'$$

$$C_{m_a}' = -0.2 C_{L_a}' \rightarrow S_H' = 22.72 \text{ ft}^2, \quad S_H = 45.2 \text{ ft}^2$$

$$\frac{S_H' - S_H}{S_H} \times 100 = -49.72\%$$

$$(2) X_{cg} = 0.25\bar{c}, C_l = 0.03 + 0.08\alpha^\circ + 0.1\delta_E^\circ$$

$$-10^\circ \leq \delta_E^\circ \leq +10^\circ, \alpha > 15^\circ.$$

$$(2a) C_{m\delta_E} = \frac{\Delta C_m}{\Delta \delta_E}, \alpha = 0^\circ, \delta_E^\circ = 0^\circ C_m = 0.025, \delta_E^\circ = 5^\circ C_m = -0.11$$

$$C_{m\delta_E} = \frac{-0.11 - 0.025}{5^\circ - 0^\circ} = -0.027$$

$$(2b) C_m = C_{m0} + C_{m\alpha}\alpha^\circ + C_{m\delta_E}\delta_E^\circ, \alpha = 0^\circ, \delta_E^\circ = 0^\circ C_{m0} = 0.02$$

$$\delta_E^\circ = 0^\circ, \alpha = 0^\circ C_m = 0.025, \alpha = 10^\circ C_m = -0.08$$

$$C_{m\alpha} = \frac{-0.08 - 0.025}{10^\circ - 0^\circ} = -0.0105, C_m = 0.025 - 0.0105\alpha^\circ - 0.027\delta_E^\circ$$



$$\textcircled{2c} \quad C_m = 0 \rightarrow \alpha^\circ = \frac{0.025 - 0.027\delta_E^\circ}{0.0105}, \quad \delta_E^\circ \in [-10^\circ, 10^\circ]$$

$$\delta_E = -10^\circ : \alpha^\circ = 28^\circ$$

$$\delta_E = +10^\circ : \alpha^\circ = -23.33^\circ, \quad \alpha_{\text{stall}} = 15^\circ \rightarrow \delta_E^\circ = -4.907$$

$$[-23.33, 15^\circ]$$

$$\textcircled{2d} \quad C_m = 0, \quad \delta_E^\circ = \frac{0.025 - 0.0105\alpha^\circ}{0.027}, \quad C_L = 0.03 + 0.08\alpha^\circ + 0.1 \left( \frac{0.025 - 0.0105\alpha^\circ}{0.027} \right)$$

$$C_L = 0.12 + 0.041\alpha^\circ \rightarrow \alpha_{\min} = -23.33^\circ \rightarrow C_L = -0.8367$$

$$\alpha_{\max} = 15^\circ, \quad C_L = 0.7393 \rightarrow [-0.83, 0.74]$$

$$(2) (e) W = 93000 \text{ lb}, S_{\text{ref}} = 900 \text{ ft}^2, \rho_0 = 0.0023769$$

$$C_L = \frac{W}{\frac{1}{2} \rho V^2 S} \quad , \quad V = \sqrt{\frac{W}{\frac{1}{2} \rho S C_L}} = \sqrt{\frac{88817.8}{C_L}} \rightarrow C_L > 0$$

$$(0, 0.74) \quad , \quad V_{\min @ C_{L_{\max} - p.s}} = 0.74 \quad , \quad V_{\min} = \sqrt{\frac{88817.8}{0.74}} \sim 346$$

$$C_L = 0 \quad , \quad \alpha = -2.98 \quad \delta_E = 2.0855 \quad , \quad \alpha \in [0, 15]$$

$$\alpha = 0 \quad , \quad C_m = 0 \quad , \quad \delta_E \sim 0.926^\circ \quad , \quad C_{L_{\alpha=0}} = 0.03 + 0.08(0) + 0.1(0.926) \\ = 0.1226$$

$$V_{\max} = \sqrt{\frac{88817.6}{0.1226}} \sim 851.15 \text{ ft/s}$$



$$\textcircled{3} C_{Lr} = \frac{\partial C_L}{\partial (rb/2V)}, \quad \textcircled{a} C_{Lr} = -\frac{C_{La}}{12} \left( \frac{1+3b}{1+b} \right) - \mu_1 \cdot C_d \tan^2 \Lambda$$

$$C_{Lr} = -\frac{C_{La}}{12} \left( \frac{1+3b}{1+b} \right) - \frac{C_{La}}{4} \frac{AR}{AR+4} \tan^2 \Lambda$$

$$\textcircled{b} C_{Lr} = (C_{Lr})_{\Gamma=0} = \cos^2 \Gamma \approx 1$$

$$\textcircled{c} C_{Lr} = \left[ -\frac{C_{La}}{12} \left( \frac{1+3b}{1+b} \right) - \frac{C_{La}}{4} \frac{AR}{AR+4} \tan^2 \Lambda \right] \cos^2 \Gamma$$

$$\textcircled{d} C_{Lr} = -\frac{C_{La}}{12} \left( \frac{1+3(1)}{1+1} \right) = -\frac{C_{La}}{6}$$

$$\textcircled{4} \quad C_{n\beta} = 0.15' / \text{rad}, \quad C_{n\beta WB} = -0.3' / \text{rad}, \quad C_{L_{av}} = 4.0' / \text{rad}$$

$$\eta_V = 1, \quad 1 - \frac{d\sigma}{d\beta} = 2 + 0.75 \frac{S_V}{S}, \quad \kappa_{VS} = \frac{b}{2}, \quad l_V = \frac{b}{2}, \quad \frac{l_V}{b} = 0.5$$

$$C_{n\beta} = C_{n\beta WB} + C_{n\beta V}, \quad C_{n\beta V} = C_{L_{av}} \eta_V \frac{S_V}{S} \frac{l_V}{b} \left( 1 - \frac{d\sigma}{d\beta} \right)$$

$$C_{n\beta V} = C_{n\beta} - C_{n\beta WB} = 0.15 - (-0.3) = 0.45' / \text{rad}$$

$$\Rightarrow 0.45 = 4 \left( \frac{S_V}{S} \right) (0.5) \left( 2 + 0.75 \frac{S_V}{S} \right) \longrightarrow \frac{S_V}{S} = 0.1081$$

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All Questions Answered by AI

Gemini 2.5 pro