The LQR Controller

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In this easy-to-follow guide on the Linear Quadratic Regulator (LQR) controller, you'll learn about its practical and theoretical aspects. It equips you with the knowledge to not only run simulations but also to apply it in real-world scenarios. From understanding the underlying principles to discussing practical applications, and even a step-by-step demonstration via a computer simulation, this guide covers it all. Whether you're a beginner or an experienced practitioner, this guide aims to give you a complete picture of the LQR controller, and how it can be a game-changer in tackling complex control problems.

Linear Quadratic Regulator | LQR | Control Theory | System Control | Practical Application | Simulation | Performance Analysis | Case Study

1. Introduction

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A. Overview of LQR. Once upon a time, in the mid-20th century, the field of control systems was undergoing a revolution. Engineers and mathematicians were seeking new ways to manage complex, dynamic systems. Amidst this backdrop, the Linear Quadratic Regulator (LQR) was born, a brainchild of the need for optimal control strategies.

The LQR was first introduced in the 1960s, during a period of rapid advancement in aerospace technology. The space race was in full swing, and the need for precise, reliable control systems for spacecraft was paramount. The traditional methods of control system design were not sufficient for these high-stakes, complex tasks. The world needed a new approach, and the LQR provided just that.

The LQR was a game-changer. It provided a systematic, principled approach to control system design. By defining a cost function that quantified the trade-off between system performance and control effort, and then minimizing this cost function, the LQR could determine the optimal control law for a given system.

Over the years, the LQR has proven its worth in a wide range of applications, from aerospace and robotics to economics and beyond. It is particularly well-suited for systems that can be accurately modeled by linear equations, and where the cost to be minimized can be expressed as a quadratic function. However, even for nonlinear systems, LQR can often provide useful results when applied to a linear approximation of the system around a particular operating point.

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In the following sections, we will delve deeper into the story of LQR, its applications, and its implementation. We will start with a specific problem, then introduce the LQR as a solution, and finally explore the theory behind it. We will also discuss how to implement an LQR solution and analyze the results of a simulation. So, buckle up and get ready for an exciting journey into the world of LQR!

B. Importance and Applications of LQR. The Linear Quadratic Regulator (LQR) has been a cornerstone in the field of control systems engineering due to its systematic and optimal approach to control system design. Its ability to balance system performance against control effort has made it a valuable tool across a multitude of industries.

The parameters used in LQR control design are primarily the weights in the quadratic cost function. These weights are chosen based on the specific requirements of the system being controlled. They reflect the relative importance of achieving a particular system state versus the amount of control effort expended.

Let's delve into some specific applications and the parameters typically used:

Significance Statement

This document's significance lies in its comprehensive exploration of the Linear Quadratic Regulator (LQR) controller, a key tool in the realm of system control.

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1. **Aerospace**: In the design of control systems for aircraft and spacecraft, the cost function might be weighted more heavily towards achieving a specific altitude or speed (system state), with less weight given to control effort. This is because precise control is often critical in these applications.

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- 2. **Automotive**: In applications like cruise control systems and active suspension systems, the cost function might be weighted to balance ride comfort (a function of the vehicle's state) and fuel efficiency (a function of the control effort).
- 3. Robotics: In controlling robotic arms, unmanned aerial vehicles (drones), and autonomous vehicles, the weights in the cost function might be chosen to balance the precision of the task execution (system state) and energy efficiency (control effort).
- 4. **Energy Sector**: In optimizing the operation of power systems, such as power plants and electrical grids, the cost function might be weighted more heavily towards maintaining a stable power output (system state), with less weight given to the cost of operation (control effort).
- 5. **Economics**: In dynamic optimization problems, such as maximizing economic output while minimizing costs, the weights in the cost function would reflect the relative importance of these two factors.

In each of these applications, the specific weights used in the LQR design would depend on the specific requirements and constraints of the system and the task at hand. As we delve deeper into the theory and implementation of LQR in the following sections, you will gain a deeper understanding of how these parameters are chosen and how they influence the performance of the LQR controller.

C. Prerequisites and Assumptions. Before we dive deeper into the world of Linear Quadratic Regulator (LQR), it's important to understand the prerequisites and assumptions that underpin this control strategy. This will ensure that we have a solid foundation upon which to build our understanding of LQR.

C.1. Prerequisites:. Knowledge in the following areas is assumed for this course:

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- Linear Algebra: Understanding of matrices, 111 vectors, and operations on them is essential as the state-space representation of systems, which is central to LQR, is expressed in matrix form.
- **Differential Equations**: Familiarity with or- 116 dinary differential equations is crucial as the dynamics of the systems we'll be controlling are often expressed as such equations.
- Control Systems: Basic understanding of 120 control systems, particularly state-space representation of systems, is required. This includes knowledge of system dynamics, sta- 123 bility, and controllability.
- Laplace Transforms: Familiarity with Laplace transforms is helpful for understanding the analysis and design of control systems.
- **C.2.** Assumptions:. When working with LQR, we typically make the following assumptions:
 - Linearity: The system to be controlled is assumed to be linear, or at least wellapproximated by a linear model around the operating point of interest.
 - Quadratic Cost Function: We assume that 135 the cost to be minimized can be expressed as a quadratic function of the system states and control inputs.
 - Full State Accessibility: We assume that all states of the system are measurable and available for feedback. If this is not the case, techniques such as state estimation may be required.
 - **Deterministic System**: We assume that the system is deterministic, i.e., there are no random disturbances affecting the system. If such disturbances are present, a different approach, such as stochastic control, may be more appropriate.

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With these prerequisites and assumptions in mind, we are now ready to embark on our journey into the fascinating world of LQR.

2. Example Problem

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A. Problem Statement: The Inverted Pendu**lum.** Our challenge is the inverted pendulum problem, a topic that has intrigued researchers from the past to the present. With its canonical and nonlinear structure, the inverted pendulum problem has been a popular subject for control theories and applications.

The inverted pendulum problem comes in various forms: the cart-pole inverted pendulum, the double pendulum cart-pole, the rotary inverted pendulum, and the mobile inverted pendulum. The balancing problems of these systems are widely covered in the literature and are still relevant today.

In this study, realized with a different approach to the balance problem of the inverted pendulum, the nonlinear mathematical model of the pendulum is linearized, and control is performed with full state feedback. When designing the controller, the goal is to keep the pendulum balanced vertically and bring the cart back to the starting point. You can see the pendulum system at figure 1

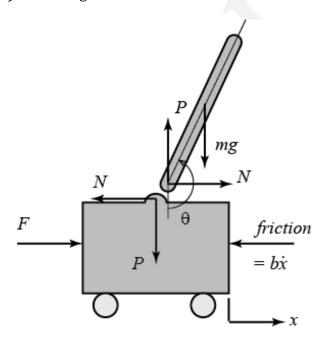


Fig. 1. Inverted Pendulum System

B. Problem Analysis. The inverted pendulum 177 problem is a classic in control theory due to its inherently unstable nature. It's a classic example of an unstable system that requires active control to maintain its equilibrium state.

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The system consists of a cart of mass M that can move horizontally and a pendulum of length l and mass m attached to the cart. The pendulum is free to rotate and its angle with the vertical, denoted by θ , is a critical parameter. The pendulum is in its unstable equilibrium position when θ is 180 degrees, i.e., when it's perfectly vertical.

The control input to the system is a force F applied horizontally to the cart. The goal is to adjust this force to keep the pendulum upright, 191 i.e., to keep θ as close to 180 degrees as possible. 192

However, the system is nonlinear and under- 193 actuated, meaning it has more degrees of freedom (two, in this case: the position of the cart and the angle of the pendulum) than it has independent control inputs (one, in this case: the force applied to the cart). This makes the control problem challenging.

Moreover, the system is highly sensitive to dis- 200 turbances and changes in the initial conditions. A small push to the pendulum or a small change in the initial angle can cause the pendulum to fall over.

In the next section, we will define the specific goals and objectives for the control problem.

C. Goals and Objectives. In this section, we will define the specific goals and objectives for the control of the inverted pendulum system. Our aim is to control both the pendulum's angle θ and the cart's position x.

To ensure the performance of our system, 212 we need to set some quantitative requirements. 213 These requirements will serve as our definition of a "good" system. In control theory, we often use specific performance measures to define these requirements. For our inverted pendulum system, we will focus on the following perfor- 218 mance measures:

• **Settling Time**: This is the time it takes for the system to converge to its desired final state. For our system, we want the cart to reach its commanded position and the pen- 223 dulum to reach its vertical position within 5

seconds.

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- **Rise Time**: This is the time it takes for the system to go from its initial state to its desired final state for the first time. For our system, we want the cart to reach its commanded position for the first time in less than 0.5 seconds.
- Overshoot: This is the extent to which the system exceeds its desired final state. For our system, we want the pendulum angle θ to never deviate more than 20 degrees (0.35 radians) from the vertical position.
- **Steady-State Error**: This is the difference between the desired final state and the actual final state that the system reaches after a long time. For our system, we want the steady-state error to be less than 2% for both x and θ .

In summary, our goal is to design a controller that can meet these performance requirements, effectively balancing the pendulum in its upright position while controlling the position of the cart.

3. Solution! LQR

A. Introduction to LQR as a Solution. Having defined our problem and set our performance requirements, we now turn to the task of designing a controller that can meet these requirements. For this, we will use the Linear Quadratic Regulator (LQR), a powerful method for designing optimal controllers for linear systems.

The LQR approach is based on the principle of minimizing a quadratic cost function, which represents a certain "cost" associated with the system's behavior. This cost function is typically a weighted sum of the system states and control inputs, reflecting the trade-off between system performance (which we want to maximize) and control effort (which we want to minimize).

By minimizing this cost function, the LQR can determine the optimal control law for a given system. This control law specifies how the control input should be adjusted based on the current system state to minimize the cost.

In the context of our inverted pendulum prob- 269 lem, the LQR will determine how the force F ap- 270 plied to the cart should be adjusted based on the current position of the cart and the angle of the pendulum, in order to keep the pendulum up- 273 right and control the position of the cart, while minimizing the use of force.

In the following sections, we will delve deeper into the theory behind LQR and how to implement it for our problem.

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- **B.** Benefits of Using LQR. The Linear Quadratic Regulator (LQR) offers several benefits that make it a powerful and popular method for control system design:
 - Optimality: LQR provides an optimal solu-283 tion in terms of minimizing a quadratic cost function. This means it finds the best possible control law under the given cost function and system dynamics.
 - Flexibility: The cost function in LQR can be tuned by adjusting the weights of the state and control variables. This allows us to balance the trade-off between system perfor- 291 mance and control effort according to the specific requirements of our system.
 - Robustness: LQR is robust to small pertur- 294 bations around the operating point. This means it can handle small disturbances and model uncertainties effectively.
 - Stability: Under certain conditions (like con- 298 trollability of the system), LQR guarantees stability of the closed-loop system. This is a crucial property for any control system.

These benefits make LQR a suitable choice for our inverted pendulum problem. By optimally balancing the pendulum and controlling the position of the cart, while minimizing the use of force, LQR can help us meet our performance requirements.

C. Limitations of LQR. While the Linear Quadratic Regulator (LQR) is a powerful method for control system design, it's important to be aware of its limitations:

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- Linearity Assumption: LQR is designed for linear systems. If the system is nonlinear, as is the case with our inverted pendulum problem, it needs to be linearized around an operating point. This means that the performance of the LQR controller may degrade if the system deviates significantly from this operating point.
- **Quadratic Cost Function**: LQR minimizes a quadratic cost function. If the actual cost associated with the system's behavior is not well-represented by a quadratic function, the LQR controller may not provide the best performance.
- Full State Accessibility: LQR assumes that all states of the system are measurable and available for feedback. If some states are not measurable, additional techniques such as state estimation may be needed.
- Deterministic Systems: LQR is designed for deterministic systems. If there are random disturbances or uncertainties in the system, a stochastic control approach may be more appropriate.

Despite these limitations, LQR can still provide a good solution for many control problems, including our inverted pendulum problem, especially when used in combination with other techniques to address these limitations.

4. Theory

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5. Theory

- A. Mathematical Foundations of LQR. The state-space representation of a system is a mathematical model of a physical system as a set of input, output, and state variables related by firstorder differential equations. It is written in matrix form and allows us to analyze and control systems. The matrices A, B, C, and D are part of the state-space representation of the system:
 - The A matrix, also known as the system matrix, relates the state vector to its derivative. It represents the dynamics of the system how the system evolves over time in the absence of input. The main logic is that we can

express the derivative of the state vector by its current values. For example, the element at the second row and third column of the A matrix represents how the angle of the pendulum affects the velocity of the cart.

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- The B matrix, also known as the input ma- 361 trix, relates the control input to the state vector. It represents how the input affects the evolution of the system. For example, the element at the second row of the B matrix represents how the force applied to the cart affects the velocity of the cart.
- The C matrix, also known as the output ma- 368 trix, relates the state vector to the output. It 369 represents how the states of the system af- 370 fect the output. In our case, the C matrix is a diagonal matrix, meaning that each output 372 is directly affected by only one state.
- The D matrix, also known as the feedfor- 374 ward matrix, directly relates the control input to the output, bypassing the state. In 376 many physical systems, including ours, this matrix is often the zero matrix, meaning that the input does not directly affect the output.

In the next sections, we will delve deeper into the theory behind LQR and how to implement it 382 for our problem.

But first lets understand what these matrices are and how to use them.

B. State-Space Representation. The state-space representation of our system is given by the following matrices:

$$A = \begin{bmatrix} A_{11} & A_{12} & A_{13} & A_{14} \\ A_{21} & A_{22} & A_{23} & A_{24} \\ A_{31} & A_{32} & A_{33} & A_{34} \\ A_{41} & A_{42} & A_{43} & A_{44} \end{bmatrix}$$
 [1] 389

$$B = \begin{bmatrix} B_1 \\ B_2 \\ B_3 \\ B_4 \end{bmatrix}$$
 [2] 390

The state-space representation of the system is given by the equation:

$$\dot{x} = Ax + Bu \tag{3}$$

This can be expanded as follows:

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$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \end{bmatrix} = \begin{bmatrix} A_{11} & A_{12} & A_{13} & A_{14} \\ A_{21} & A_{22} & A_{23} & A_{24} \\ A_{31} & A_{32} & A_{33} & A_{34} \\ A_{41} & A_{42} & A_{43} & A_{44} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} B_1 \\ B_2 \\ B_3 \\ B_4 \end{bmatrix} u$$
 [4]

So, for example, the derivative of the first state variable x_1 (the position of the cart) is given by:

$$\dot{x}_1 = A_{11}x_1 + A_{12}x_2 + A_{13}x_3 + A_{14}x_4 + B_1u$$
 [5]

This equation shows how the current state of the system and the control input u (the force applied to the cart) determine the rate of change of the position of the cart.

The elements of the A and B matrices in the state-space representation, such as A_{11} , A_{12} , B_{1} , and so on, are real numbers. These numbers are determined by the physical parameters of the system, such as the masses of the cart and the pendulum, the length of the pendulum, and the gravitational acceleration. The fact that we can directly find the derivative of the state vector by multiplying these numbers with the current state and input vectors is a characteristic of linear systems.

C. Cost Function. The Linear Quadratic Regulator (LQR) method is based on the principle of minimizing a quadratic cost function. This cost function is typically a weighted sum of the system states and control inputs, reflecting the trade-off between system performance (which we want to maximize) and control effort (which we want to minimize).

The cost function is usually defined as:

$$J = \int_0^\infty (x^T Q x + u^T R u) dt$$
 [6]

where: 424

- x is the state vector,
- u is the control input vector,
- Q is a positive semi-definite matrix that weights the states,

• *R* is a positive definite matrix that weights the control inputs,

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The matrices Q and R are chosen by the designer and reflect the relative importance of the states and control inputs in the cost function. 433 They can be thought of as tuning knobs or a GUI 434 for the LQR controller. For example, if a par- 435 ticular state is more important to control accu- 436 rately, the corresponding element in the Q ma- 437 trix would be set to a higher value. Similarly, if a particular control input needs to be used sparingly, the corresponding element in the R matrix would be set to a higher value.

The term x^TQx in the cost function represents the cost associated with the states of the system. For example, if we have a 2-dimensional state vector $x = [x_1, x_2]$, and a 2x2 matrix Q = $[Q_{11}, Q_{12}; Q_{21}, Q_{22}]$, then:

$$x^{T}Qx = x_1^2Q_{11} + x_1x_2(Q_{12} + Q_{21}) + x_2^2Q_{22}$$
 [7]

This shows that the cost is a weighted sum of the squares of the states and their cross product, where the weights are the elements of the Q matrix.

Similarly, the term $u^T R u$ in the cost function represents the cost associated with the control 453 inputs.

The goal of the LQR method is to find the control input *u* that minimizes this cost function. In the next section, we will discuss how this is achieved by solving the Riccati equation.

D. Riccati Equation. The solution to the LQR 459 problem is given by the solution to the Alge- 460 braic Riccati Equation (ARE). The ARE is a type of nonlinear equation that arises in the context 462 of infinite-horizon optimal control problems like the LQR problem.

For the LQR problem, the ARE takes the following form:

$$0 = A^{T}P + PA - PBR^{-1}B^{T}P + Q$$
 [8]

where:

• A, B, Q, and R are the matrices defined in the state-space representation and the cost function,

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• P is the solution to the ARE, a symmetric positive definite matrix that determines the optimal state feedback gain.

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The solution to the ARE can be found using various numerical methods. Once the matrix P is found, the optimal state feedback gain K can be calculated as:

$$K = R^{-1}B^TP [9]$$

This gain matrix K is then used in the state feedback controller to determine the control input *u* that minimizes the cost function:

$$u = -Kx ag{10}$$

This gain matrix K is then used in the state feedback controller to determine the control input u that minimizes the cost function. For example, if we have a 2-dimensional state vector $x = [x_1, x_2]$ and a 2-dimensional gain vector $K = [k_1, k_2]$, then the control input is given by:

$$u = -Kx = -k_1x_1 - k_2x_2$$
 [11]

This equation shows that the control input is a weighted sum of the states, where the weights are the elements of the gain vector K. So u is just a number.

E. What is K and its effect. The gain matrix Kin the LQR controller plays a crucial role in determining the behavior of the controlled system. It is calculated as $K = R^{-1}B^TP$, where P is the solution to the Riccati equation, and R and B are the matrices defined in the state-space representation and the cost function.

The control input u is given by u = -Kx, which means that the control input is a linear function of the state. The gain matrix K determines how much each state variable contributes to the control input.

When the LQR controller is applied to the system, the state-space equation $\dot{x} = Ax + Bu$ becomes $\dot{x} = (A - BK)x$. This means that the LQR controller effectively changes the system matrix A into a new matrix A - BK.

The stability of the controlled system depends on the eigenvalues of this new matrix. If all the eigenvalues have negative real parts, then the system is stable. The LQR method is designed to find the gain matrix K that makes the system 516 stable and minimizes the cost function.

Let's consider a simple example where the system matrix A and the gain matrix K are 2x2matrices:

$$A = \begin{bmatrix} A_{11} & A_{12} \\ A_{21} & A_{22} \end{bmatrix}, \quad K = \begin{bmatrix} k_1 & k_2 \end{bmatrix}$$
 [12] 521

And B is a 2x1 matrix (column vector):

$$B = \begin{bmatrix} B_1 \\ B_2 \end{bmatrix}$$
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The new system matrix after applying the LQR controller is:

$$A - BK = \begin{bmatrix} A_{11} - B_1 k_1 & A_{12} - B_1 k_2 \\ A_{21} - B_2 k_1 & A_{22} - B_2 k_2 \end{bmatrix}$$
 [14] 526

6. Solution Implementation

A. System Modeling. In this section, we will model our system using the state-space representation. The state-space representation of a system is a mathematical model of a physical system as a set of input, output and state variables related by first-order differential equations. To represent our system, we need to define the matrices A, B, C, and D, which represent the system dynamics, the control input, the output, and the direct transmission term, respectively.

This system is challenging to model because of the physical constraint (the pin joint) between the cart and pendulum which reduces the de- 540 grees of freedom in the system. Both the cart and the pendulum have one degree of freedom (x and θ , respectively). We will generate the differential equations for these degrees of freedom from first principles employing Newton's second law (F = ma).

The equations of motion for the system are:

$$\ddot{x} = \frac{1}{M}(F - N - b\dot{x})$$
 [15] 54

$$\ddot{\theta} = \frac{1}{I}(-Nl\cos\theta - Pl\sin\theta)$$
 [16]

It is necessary to include the interaction forces 550 N and P between the cart and the pendulum 551

in order to fully model the system's dynamics. The inclusion of these forces requires modeling the x- and y-components of the translation of the pendulum's center of mass in addition to its rotational dynamics.

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The additional x- and y-component equations for the pendulum are:

$$N = m\ddot{x}_p \tag{17}$$

$$P = m(\ddot{y}_p + g)$$
 [18]

However, the position coordinates x_p and y_p are exact functions of θ . Therefore, we can represent their derivatives in terms of the derivatives of θ . The equations for the derivatives of x_p and y_p are:

$$\ddot{x}_p = \ddot{x} - l\dot{\theta}^2 \sin\theta + l\ddot{\theta}\cos\theta$$
 [19]

$$\ddot{y}_p = l\dot{\theta}^2 \cos\theta + l\ddot{\theta} \sin\theta$$
 [20]

These expressions can then be substituted into the expressions for N and P to give:

$$N = m(\ddot{x} - l\dot{\theta}^2 \sin \theta + l\ddot{\theta} \cos \theta)$$
 [21]

$$P = m(l\dot{\theta}^2 \cos \theta + l\ddot{\theta} \sin \theta + g)$$
 [22]

We can now represent these equations within a simulation environment. The simulation environment can work directly with nonlinear equations, so it is unnecessary to linearize these equations.

B. Derivation of State-Space Equations.

C. State-Space Equations. From the main problem, the dynamic equations of the inverted pendulum system in state-space form are the follow-580 ing: 581

$$\begin{bmatrix}
\dot{x} \\
\dot{x} \\
\dot{\phi} \\
\dot{\phi}
\end{bmatrix} = \begin{bmatrix}
0 & 1 & 0 & 0 \\
0 & -0.1818 & 2.6727 & 0 \\
0 & 0 & 0 & 1 \\
0 & -0.4545 & 31.1818 & 0
\end{bmatrix} \begin{bmatrix}
x \\
\dot{x} \\
\phi \\
\dot{\phi}
\end{bmatrix} + \begin{bmatrix}
0 \\
1.8182 \\
0 \\
4.5455
\end{bmatrix} u$$
[1]

$$\mathbf{y} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} x \\ \dot{x} \\ \phi \\ \dot{\phi} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \end{bmatrix} u$$
 [2] 583

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