

SUBROUTINE ZERO (C,V)	L001
IMPLICIT REAL * 8 (A-H, O-Z)	L002
COMMON /B0/ AL,BL,HX2,HY2,HXY,HX,HY,HXT2,HYT2	L003
COMMON /B1/ NUM,NX4,NX3,NX2,NX1,NY3,NY2,NY1	L004
DIMENSION C(1),CXX(1),CXY(1),CYY(1),V(NY3,1)	L005
C	L006
C	L007
CCCCC DISTRIBUTE COLUMN ARRAY INTO PLANE ARRAY.	L008
C	L009
C	L010
II = 1	L011
DO 150 I = 2, NY2	L012
DO 140 J = 2 , NX2	L013
V(I,J) = C(II)	L014
II = II + 1	L015
140 CONTINUE	L016
150 CONTINUE	L017
DO 160 J = 2 , NX2	L018
V(1,J) = V(3,J)	L019
160 CONTINUE	L020
DO 170 I = 1 , NY2	L021
V(I,1) = V(I,3)	L022
170 CONTINUE	L023
GO TO 300	L024
C	L025
C	L026
C	L027
ENTRY FIRST (CXX,CYY,V)	L028
C	L029
C	L030
CCCCC FINITE DIFFERENCE APPROXIMATION OF	L031
C THE FIRST DERIVATIVE OF A FUNCTION.	L032
C	L033
C	L034
DO 210 I = 2 , NY2	L035
K = (I-2)*NX1	L036
DO 200 J = 2 , NX2	L037
II = K + J - 1	L038
CXX(II) = HXT2 * (V(I,J+1)-V(I,J-1))**2	L039
CYY(II) = HYT2 * (V(I+1,J)-V(I-1,J))**2	L040
200 CONTINUE	L041
210 CONTINUE	L042
GO TO 300	L043
C	L044
C	L045
C	L046
ENTRY SECOND (CXX,CXY,CYY,V)	L047
C	L048
C	L049
CCCCC FINITE DIFFERENCE APPROXIMATION OF	L050
C THE SECOND DERIVATIVE OF A FUNCTION.	L051
C	L052

C

DO 250 I = 2 , NY2	L053
K = (I-2)*NX1	L054
DO 240 J = 2 , NX2	L055
II = K + J - 1	L056
CXX(II) = (V(I,J+1) - 2.*V(I,J)	L057
+ V(I,J-1)) / HX2	L058
1 CYY(II) = (V(I+1,J) - 2.*V(I,J)	L059
+ V(I-1,J)) / HY2	L060
1 CXY(II) = (V(I+1,J+1) - V(I+1,J-1)	L061
- V(I-1,J+1) + V(I-1,J-1)) / HXY	L062
240 CONTINUE	L063
250 CONTINUE	L064
300 RETURN	L065
END	L066
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SUBROUTINE STRESS (W,WXX,WXY,WYY,FXX,FX,FYY,M001
1      ,SBX,SBY,SBXY,IB)M002
      IMPLICIT REAL * 8 (A-H,O-Z)M003
      COMMON /B1/ NUMM004
      COMMON /B2/ RF,V,VS,D,PRM005
      COMMON /B4/ T2M006
      DIMENSION W(1),WXX(1),WXY(1),WYY(1),FXX(1),FX(1),FYY(1)M007
1      ,SBX(1),SBY(1),SBXY(1),IB(1)M008
C      M009
C      M010
CCCCC      FIGURE OUT THE BENDING STRESSES AT THEM011
C      BOTTOM OF THE PLATE, THEN PRINTOUT THEM012
C      BENDING AND THE MEMBRANE STRESSES.M013
C      M014
C      M015
      LC = 0M016
      WRITE (6,900)M017
      DO 50 I = 1, NUMM018
          SBX(I) = -D * (WXX(I) + PR*WYY(I)) / T2M019
          SBY(I) = -D * (WYY(I) + PR*WXX(I)) / T2M020
          SBXY(I) = -D * (1.-PR) * WXY(I) / T2M021
          CPXY = - FXY(I)M022
          IF (LC .NE. 53) GO TO 20M023
          LC = 0M024
          WRITE (6,910)M025
          WRITE (6,900)M026
20      LC = LC + 1M027
          WRITE (6,920) I,IB(I),W(I),SBX(I),SBY(I)M028
1          ,SBXY(1),FYY(1),FXX(1),CPXYM029
50 CONTINUEM030
C      M031
C      M032
C      M033
      RETURNM034
C      M035
C      M036
900 FORMAT (1H1,1X,'(NOTE: *-BOUNDARY POINTS',M037
1      ', **--CORNER POINT)')//1X,M038
2      'NODE',6X,'DEFLECTION',9X,'BENDING STRESSES',M039
3      13X,'SHEARING',8X,M040
4      'MEMBRANE STRESSES IN THE MIDDLE SURFACE' /M041
5      26X, 'X-DIRECTION, Y-DIRECTION',M042
6      'STRESS X-DIRECTION',M043
7      ' Y-DIRECTION SHEARING' /)M044
910 FORMAT (1H0,30X,'(CONTINUED)')M045
920 FORMAT (I5,A4,7(E13.6,3X))M046
      ENDM047

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SUBROUTINE TITLE (W1,QT,NT,KK)
IMPLICIT REAL * 8 (A-H,O-Z)
COMMON /B0/ AL,BL
COMMON /B1/ NUM,NX4,NX3,NX2,NX1,NY3,NY2,NY1,NX,NY
COMMON /B2/ RF,V,VS,D,PR
COMMON /B3/ TH,RC,SOVERD,TOVERD,EHX4,Q(8),ELAS
1      ,LINC,LORNL
COMMON /B4/ T2,SN,CN,STIF,IDIS,IPRI,IREF,NBC
1      ,ITITLE(12)
DIMENSION NSC(4,4),LNL(4,2)
DATA NSC /'SIMP','LY S','UPPO','RTED'
1      , 'ELAS','TIC','SUPP','ORT'
2      , 'FREE','EDG','ES',' '
3      , 'BEAM','SUP','PORT','ED'
DATA LNL /'LINE','AR T','YPE',' '
1      , 'NONL','INEA','R TY','PE'

C
C
CCCCC      PRINT THE TITLE AND NECESSARY DATA AT
C            THE FIRST PAGE.
C
C
QT = QT * TH
CF = -2. * D * (1.-PR) * W1
WRITE (6,300) ITITLE,(NSC(I,NBC),I=1,4)
1      ,(LNL(I,LORNL),I=1,4),NX,NY,LINC
WRITE (6,315) NT,KK,IPRI,IDIS
WRITE (6,325) AL,BL,ELAS,PR,TH,D,STIF,QT,CF
GO TO 100

C
C
ENTRY REF (QT,W2,W3,W4,W5,W6)
C
C
CCCCC      PRINTOUT THE REFERENCE DATA.
C
C
WNON = W2 / TH
QNON = QT * CN
SBXN1 = -D * (W3 + PR*W4) / T2 * SN
SMXN1 = W5 * SN
SMYN4 = W6 * SN
BXY3 = -CF / 2. / T2 * SN
WRITE (6,350) WNON,QNON,SBXN1,SMXN1,SMYN4,BXY3

C
100 RETURN
C
300 FORMAT (1H1,64(1H*) / 1X,1H*,62X,1H*
1      / ' ***** ',12A4,' *****'
2      / 1X,1H*,62X,1H* / 1X,64(1H*) /
3      '0(NOTE:  PRESSURE IN PSI, FORCE IN ',
4      'POUNDS, LENGTH IN INCHES.)' /

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5  'OTYPE OF SUPPORT .....', N053
6  '.....',4A4 / N054
7  'OTYPE OF PROBLEM .....', N055
8  '.....',4A4 / N056
9  'ONUMBER OF SEGMENTS IN X-DIRECTION .....', N057
A  '.....',I10 / N058
B  'ONUMBER OF SEGMENTS IN Y-DIRECTION .....', N059
C  '.....',I10 / N060
D  'ONUMBER OF LOAD INCREMENTS .....', N061
E  '.....',I10 ) N062
315 FORMAT ( N063
1  'ONUMBER OF LOAD INCREMENT OF PRESENT CASE', N064
2  '.....',I10 / N065
3  'OITERATION TIMES OF PRESENT CASE .....', N066
4  '.....',I10 / N067
5  'OPRINCIPAL STRESSES OUTPUT INCREMENT ....', N068
6  '.....',I10 / N069
7  'OIN-PLANE DISPLACEMENT OUTPUT INCREMENT .', N070
8  '.....',I10) N071
325 FORMAT ( N072
1  'OHALF LENGTH OF PLATE (A) IN X-DIRECTION ', N073
2  '.....',E12.5 / N074
3  'OHALF LENGTH OF PLATE (B) IN Y-DIRECTION ', N075
4  '.....',E12.5 / N076
5  'OMODULUS OF ELASTICITY OF PLATE .....', N077
6  '.....',E12.5 / N078
7  'OPOISSON'S RATIO OF PLATE .....', N079
8  '.....',E12.5 / N080
9  'OTHICKNESS OF PLATE .....', N081
A  '.....',E12.5 / N082
B  'OFLEXURAL RIGIDITY OF THE PLATE .....', N083
C  '.....',E12.5 / N084
D  'OMODULUS K USED ON THE BOUNDARY .....', N085
E  '.....',E12.5 / N086
F  'OMAGNITUDE OF LOAD OF PRESENT CASE .....', N087
G  '.....',E12.5 / N088
H  'OREACTION FORCE AT CORNER POINT .....', N089
I  '.....',E12.5 / ) N090
350 FORMAT ( 1H0,15X,'***** REFERENCE DATA *****'// N091
1  'OWIDTH / THICKNESS .....', N092
2  '.....',E12.5 / N093
3  'O (Q/E) * (A/T) ** 4 .....', N094
4  '.....',E12.5 / N095
5  'O (SBX 1/E) * (A/T) ** 2 .....', N096
6  '.....',E12.5 / N097
7  'O (SMX 1/E) * (A/T) ** 2 .....', N098
8  '.....',E12.5 / N099
9  'O (SMY 4/E) * (A/T) ** 2 .....', N100
A  '.....',E12.5 / N101
B  'O (BXY 3/E) * (A/T) ** 2 .....', N102
C  '.....',E12.5 ) N103
END N104

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SUBROUTINE PRINST (FXX,FXY,FYY,SBX,SBY,SBXY,IB) P001
IMPLICIT REAL * 8 (A-H,O-Z) P002
COMMON /B1/ NUM P003
DIMENSION FXX(1),FXY(1),FYY(1),SBX(1),SBY(1) P004
1          ,SBXY(1),IB(1) P005
C P006
C P007
CCCCCC FIGUREOUT THE TOTAL STRESSES AT THE BOTTOM P008
C OF THE PLATE, THE PRINCIPAL STRESSES AND P009
C THEIR CORRESPONDING PRINCIPAL DIRECTIONS. P010
C P011
LC = 0 P012
WRITE (6,900) P013
DO 50 I = 1, NUM P014
    BSX = SBX(I) + FYY(I) P015
    BSY = SBY(I) + FXX(I) P016
    BSXY = SBXY(I) - FXY(I) P017
    CEN = .5*(BSX+BSY) P018
    RAD = DSQRT((0.5*(BSX-BSY)**2 + BSXY*BSXY) P019
    BS1 = CEN + RAD P020
    BS2 = CEN - RAD P021
    BS = BS1 - BSY P022
    IF (DABS(BS1-BSX) .LT. 0.00001) GO TO 20 P023
    IF (DABS(BS) .LT. 0.00001) GO TO 10 P024
    X1 = DATAN(BSXY/BS) * 180. / 3.141592 P025
    GO TO 30 P026
10    X1 = 90.0 P027
    IF ((SBXY(I+1)-FXY(I+1)) .LT. 0.0) X1 = -90. P028
    GO TO 30 P029
20    X1 = .0 P030
30    X2 = X1 + 90. P031
    IF (LC .NE. 53) GO TO 40 P032
    LC = 0 P033
    WRITE (6,910) P034
    WRITE (6,900) P035
40    LC = LC + 1 P036
    WRITE (6,920) I,IB(I),BSX,BSY,BSXY P037
1          ,BS1,BS2,X1,X2 P038
50 CONTINUE P039
RETURN P040
C P041
900 FORMAT (1H1,1X,'(NOTE: *-BOUNDARY POINTS, *,', P042
1 ' *-CORNER POINT) '///',10X,'TOTAL ', P043
2 'STRESSES',14X,'SHEARING',15X,'PRINCIPAL ', P044
3 'STRESSES',13X,'CORRESPONDING ANGLES' / 10X, P045
4 'X-DIRECTION Y-DIRECTION ', P046
5 'STRESS ', P047
6 'MAXIMUM MINIMUM ', P048
7 'MAXIMUM MINIMUM' /) P049
910 FORMAT (1H0,30X,'(CONTINUED) ') P050
920 FORMAT (15,A4,3(E13.6,3X),4(3X,E13.6)) P051
END P052

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