Package 'civilR'

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all_member_sizes

Extract a vector of all member sizes

Description

Extract a vector of all member sizes for specified steel grade (S355/S275) and member type (UC/UB)

Usage

```
all_member_sizes(steel_grade, member_type, list_reference_tables)
```

Arguments

```
steel_grade steel_grade [N/mm^2], categorical: 'S355' or 'S275' member_type member_type, categorical: 'UC' or 'UB' list_reference_tables List of reference tables
```

Value

Vector of all member sizes [height (mm) x width (mm) x mass (kg/m)]

```
axial\_compression\_force
```

Calculate the axial compression force

Description

Compute Axial Compression Force, N_{ed} [kN], for member without including Temperature effect. Used as trial for the top level strut where temperature changes could not be neglected. As well can be used to calculate final N_{ed} for struts from low levels of excavation, where temperature effect could be neglected.

Usage

```
axial_compression_force(isTopLevel = T, DL = 1, LL = 1, L = 12.5,
P = 247, theta = 90, spacing = 6, Lcry = 12.7, Lcrz = 1,
steel_grade = "S355", member_type = "UB", alpha_T = 1.2e-05,
delta_T = 10, k_T = 0.8, E = 210, AL = 50, gamma = 1.35,
list_reference_tables)
```

Arguments

isTopLevel	Is member located at top level? [boolean]
DL	Dead load / self-weight of member $[kN/m]$
LL	Live load / imposed load $[kN/m]$
L	Total length of member $[m]$
Р	Axial compression force of member per meter $[kN/m]$
theta	Angle to wall $[deg]$
spacing	spacing $[m]$
Lcry	critical length major axis $[m]$
Lcrz	critical length minor axis $[m]$
$steel_grade$	steel_grade $[N/mm^2]$, categorical: 'S355' or 'S275'
$member_type$	member_type, categorical: 'UC' or 'UB'
alpha_T	Thermal coef. of expansion $[degC]$
$delta_{-}T$	Change in temperature from the Installation temperature $[degC]$
k_T	Coefficient Of Temperature Effect [dimensionless]
Е	Young's Modulus of Elasticity $[GPa]$
AL	Accidental Impact Load $[kN/m]$
gamma	Partial factor for action [dimensionless], as per EN 1990:2002 standard
list_reference.	tables

Details

First of all function check which combination govern in ULS (Ultimate Limit State) without including Temperature load, TL [kN]. Then include TL calculations for Load Combinations applying partial factors based on the Table A1.2(B), EN1990-2002, p53 Compare maximum from ULS and ALS to define which mistake could govern.

List of reference tables

Value

- N_{ed} Axial compression force [kN]
- TL Temperature Load [kN]

axial_compression_force_given_member

Calculate the axial compression force for a given member size

Description

Compute Axial Compression Force, N_{ed} [kN], for member without including Temperature effect. Used as trial for the top level strut where temperature changes could not be neglected. As well can be used to calculate final N_{ed} for struts from low levels of excavation, where temperature effect could be neglected.

Usage

```
axial_compression_force_given_member(isTopLevel = T, Ned_no_TL = 6987,
  member_size, alpha_T = 1.2e-05, delta_T = 10, k_T = 0.8, E = 210,
  list_reference_tables)
```

Arguments

isTopLevel Is member located at top level? [boolean]

alpha_T Thermal coef. of expansion [degC]delta_T Change in temperature from the Installation temperature [degC]k_T Coefficient Of Temperature Effect [dimensionless]

E Young's Modulus of Elasticity [GPa]list_reference_tables

List of reference tables

Details

First of all function check which combination govern in ULS (Ultimate Limit State) without including Temperature load, TL [kN]. Then include TL calculations for Load Combinations applying partial factors based on the Table A1.2(B), EN1990-2002, p53 Compare maximum from ULS and ALS to define which mistake could govern.

- N_{ed} Axial compression force [kN]
- TL Temperature Load [kN]

calculated_NEd 5

	calculated_NEd	Generate calculated NEd
--	----------------	-------------------------

Description

```
Generate calculated N_{E_d}, N_{E_{dc}} [kN].
```

Usage

```
calculated_NEd(N_b_Rd, Ieff, MEd, h0, A)
```

Arguments

N_b_Rd Overall buckling resistance of the struts about the axis [kN]

Ieff Effective second moment of area $[mm^4]$

MEd Second order moment [kN.m]

h0 Distance between centroids of chords [m]

A Cross-section area of strut $[cm^2]$

Value

```
N_{E_{dc}} Calculated N_{E_d} [kN]
```

 $\begin{tabular}{ll} check-all_member_sizes & Generate\ a\ table\ with\ all\ member\ sizes\ and\ apply\ all\ checks\ on\ each\ of\ them \end{tabular}$

Description

Generate a table with all member sizes and apply all checks on each of them

Usage

```
check_all_member_sizes(steel_grade, member_type, k, L, E, h0, Lch, Ad, n,
  isTopLevel, alpha_T, delta_T, k_T, Ned_no_TL, LL, AL, Lcry, Lcrz, ml,
  strut_name, base_file_name, export_xlsx)
```

Arguments

$steel_grade$	steel_grade $[N/mm^2]$, categorical: 'S355' or 'S275'
$member_type$	member_type, categorical: 'UC' or 'UB'
k	Coefficient [dimensionless]
L	Total length of member $[m]$
E	Young's Modulus of Elasticity $[GPa]$
h0	Distance between centroids of chords $[mm]$
Lch	Length of chord $[mm]$
Ad	Section area of diagonal (lacing), $[cm^2]$

n Number of planes of lacing, default [n=2] is TopLevel Is member located at top level? [boolean]

alpha $_{-}$ T Thermal coef. of expansion [degC]

delta_T Change in temperature from the Installation temperature [degC]

k_T Coefficient Of Temperature Effect [dimensionless]

Ned_no_TL Axial Compressional Force without Temperature Load [kN]

LL Live load / imposed load [kN/m] AL Accidental Impact Load [kN/m] Lcry critical length about major axis [m]

Lcrz critical length minor axis [m] ml Lacing weight [kN/m]

ma Baoma wording [n

strut_name Strut name

export_xlsx Boolean to export Excel spreadsheet or not [T/F]

file_name Path and file name of the output table

Value

- df Dataframe containing relevant input and all computed data
- $optimal_member_size$ Optimal member size [height (mm) x width (mm) x mass (kg/m)]
- $optimal_TL$ Optimal Temperature Load [kN]
- \bullet $optimal_Ned$ Optimal @param Ned_no_TL Axial Compressional Force with Temperature Load [kN]

check_local_buckling_resistance_about_zz_axis

Perform check #3, calculating the local buckling resistance of struts about minor z - z axis

Description

Calculate the local buckling resistance of member about minor z-z axis, based on EC3 Approach.

$$L_e = kL_{ch}$$

[mm] where L is the critical length for buckling about minor axis z-z Steps of the check performed for laced struts:

1. Plastic resistance of the cross-section to compression [kN]

$$N_{pl,R_d,ch} = 2(A f y)$$

2. The Euler buckling load [kN]

$$N_{cr,ch} = \frac{\pi^2 E I_{zz}}{L_e^2}$$

3. Relative slenderness [dimensionless]

$$\bar{\lambda_{ch}} = \sqrt{\frac{N_{pl,R_d,ch}}{N_{cr,ch}}}$$

4. Calculate Φ_{ch} parameter for slenderness reduction factor

$$\Phi_{ch} = 0.5 \left[1 + \alpha \left(\bar{\lambda_{ch}} - 0.2 \right) + \bar{\lambda_{ch}}^{2} \right]$$

5. Slenderness reduction factor [dimensionless]

$$X_{ch} = \frac{1}{\Phi_{ch} + \sqrt{\Phi_{ch}^2 - \bar{\lambda_{ch}}^2}}$$

6. Output overall buckling resistance of the struts about z-z minor axis [kN]

$$N_{b,R_d,ch} = X_{ch} N_{pl,R_d,ch}$$

The partial factors γ_M that are applied to resistance of members to instability: $\gamma_{M_1}=1$

Usage

check_local_buckling_resistance_about_zz_axis(trial_member_size,
 member_type, steel_grade, k, Lch, E, list_reference_tables)

Arguments

trial_member_size

Trial member size

member_type, categorical: 'UC' or 'UB'

steel_grade steel_grade $[N/mm^2]$, categorical: 'S355' or 'S275'

 $k \hspace{1cm} {\rm Coefficient} \hspace{0.1cm} [{\rm dimensionless}]$

Lch Length of chord [mm]

E Young's Modulus of Elasticity [GPa]

 $list_reference_tables$

List of reference tables

- $N_{b,Rd,X}$ Local buckling resistance of struts about z-z axis [kN]
- f_y
- \bullet N_{pl,R_d}
- \bullet N_{cr}
- \bullet $\bar{\lambda}$
- α_{yy}
- X

check_overall_buckling_resistance_about_yy_axis

Perform check #1, calculating the overall buckling resistance of member about major y - y axis

Description

Calculate the overall buckling resistance of member about y-y axis, based on EC3 Approach.

$$L_e = kL$$

[mm] where L is the critical length for buckling about major axis y-y Steps of the check performed for laced struts:

1. Plastic resistance of the cross-section to compression [kN]

$$N_{pl,R_d} = 2(A f y)$$

2. The Euler buckling load [kN]

$$N_{cr,X} = \frac{\pi^2 E I_{yy}}{L_e^2}$$

3. Relative slenderness [dimensionless]

$$\bar{\lambda_X} = \sqrt{\frac{N_{pl,R_d}}{N_{cr,X}}}$$

4. Calculate Φ_X parameter for slenderness reduction factor

$$\Phi_X = 0.5 \left[1 + \alpha \left(\bar{\lambda_X} - 0.2 \right) + \bar{\lambda_X}^2 \right]$$

5. Slenderness reduction factor [dimensionless]

$$X_{x} = \frac{1}{\Phi_{X} + \sqrt{{\Phi_{X}}^{2} - {\bar{\lambda_{X}}^{2}}^{2}}}$$

6. Output overall buckling resistance of the struts about y - y axis [kN]

$$N_{b,R_d,X} = X_X N_{pl,R_d}$$

The partial factors γ_M that are applied to resistance of members to instability: $\gamma_{M_1} = 1$

Usage

check_overall_buckling_resistance_about_yy_axis(trial_member_size,
 member_type, steel_grade, k, L, E, list_reference_tables)

Arguments

trial_member_size

Trial member size

member_type, categorical: 'UC' or 'UB'

 $steel_grade = steel_grade = [N/mm^2], categorical: 'S355' or 'S275'$

 $\begin{tabular}{ll} k & Coefficient [dimensionless] \\ L & Total length of member $[m]$ \\ \end{tabular}$

E Young's Modulus of Elasticity [GPa]

list_reference_tables

List of reference tables

Value

• $N_{b,Rd,X}$ Overall buckling resistance of struts about major y-y axis [kN]

- \bullet $N_{b,R_d,X}$
- f_y
- N_{pl,R_d}
- $N_{cr,X}$
- $N_{cr,X}$
- \bullet $\bar{\lambda_X}$
- α_{yy}
- X

check_overall_buckling_resistance_about_zz_axis

Perform check #2, calculating the overall buckling resistance of struts about major z-z axis

Description

Calculate the overall buckling resistance of member about z-z axis, based on EC3 Approach.

$$L_e = k L$$

[mm] where L is the critical length for buckling about major axis z-z Steps of the check performed for laced struts:

1. Plastic resistance of the cross-section to compression [kN]

$$N_{pl,R_d} = 2(A f y)$$

2. The Euler buckling load [kN]

$$N_{cr,Y} = \frac{\pi^2 E I_{eff}}{L_e^2}$$

3. Relative slenderness [dimensionless]

$$\bar{\lambda_Y} = \sqrt{\frac{N_{pl,R_d}}{N_{cr,Y}}}$$

4. Calculate Φ_Y parameter for slenderness reduction factor

$$\Phi_Y = 0.5 \left[1 + \alpha \left(\bar{\lambda_Y} - 0.2 \right) + \bar{\lambda_Y}^2 \right]$$

5. Slenderness reduction factor [dimensionless]

$$X_Y = \frac{1}{\Phi_Y + \sqrt{\Phi_Y^2 - \bar{\lambda_Y}^2}}$$

6. Output overall buckling resistance of the struts about z-z axis [kN]

$$N_{b,R_d,Y} = X_Y N_{pl,R_d}$$

The partial factors γ_M that are applied to resistance of members to instability: $\gamma_{M_1} = 1$

Usage

```
check_overall_buckling_resistance_about_zz_axis(trial_member_size,
  member_type, steel_grade, k, L, E, h0, list_reference_tables)
```

Arguments

trial_member_size

Trial member size

member_type, categorical: 'UC' or 'UB'

steel_grade steel_grade $[N/mm^2]$, categorical: 'S355' or 'S275'

k Coefficient [dimensionless]

L Total length of member [m]

E Young's Modulus of Elasticity [GPa]

h0 Distance between centroids of chords [mm]

list_reference_tables

List of reference tables

- $N_{b,R_d,Y}$ Overall buckling resistance of struts about z-z axis [kN]
- f_y
- N_{pl,R_d}
- I_{eff}
- $N_{cr,Y}$
- $\bar{\lambda_Y}$
- α_{yy}
- X

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civilR Package civilR

Description

Civil Engineering package.

 ${\tt combined_vertical_load}$ ${\it USL}$ ${\it vs}$ ${\it ALS}$ ${\it verical}$ ${\it load}$ ${\it combinations}$

Description

USL vs ALS verical load combinations

Usage

```
combined_vertical_load(DL, LL, AL)
```

Arguments

DL Dead load / self-weight of member [kN/m]

LL Live load / imposed load [kN/m]

AL Accidental Impact Load [kN/m]

Details

Calculation steps are as follows:

```
1. ULS: F = (1.35 DL + 1.5 LL + 1.5 TL)
```

2.
$$ALS: F = (1.0 DL + 0.7 LL + 1.0 AL)$$

3.
$$ALS: F = (1.0 DL + 0.6 LL + 1.0 AL + 0.5 TL)$$

Value

combined_vertical_load [kN/m]

 ${\tt compute_output_table} \quad \textit{Export output table to Excel file}$

Description

Export output table to Excel file.

Usage

```
compute_output_table(file_name = "tables/input/output_processed_table.xlsx",
export_xlsx = T)
```

Arguments

file_name Path and file name of the output table

export_xlsx Boolean to export Excel spreadsheet or not [T/F]

Value

None

 ${\tt convert_member_dimensions_string_to_elements}$

Convert individual member dimensions to a string

Description

Convert individual member dimensions to a string.

Usage

```
convert_member_dimensions_string_to_elements(s)
```

Arguments

s String of the member dimensions

- h Member height [mm]
- b Member width [mm]
- m Member weight [kg/m]

convert_member_dimensions_to_string

Convert the member size individual dimensions to a standard string

Description

Generate a combined string from given three individual elements, separated by "x".

Usage

```
convert_member_dimensions_to_string(h, b, m)
```

Arguments

```
\begin{array}{lll} \mathbf{h} & & \mathbf{Member\ height}\ [mm] \\ \mathbf{b} & & \mathbf{Member\ width}\ [mm] \\ \mathbf{m} & & \mathbf{Member\ mass}\ [kg/m] \end{array}
```

Value

String of the member dimensions

```
effective_length_of_member
```

Calculate the effective length of member

Description

Calculate the effective length of member, L_e [m].

Usage

```
effective_length_of_member(k, L)
```

Arguments

k Effective length coefficient [dimensionless]
L Length of strut between restraints [m]

```
L_e Effective length of strut [m]
```

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effective_second_moment_of_area

Calculate the effective second moment of area

Description

Compute the effective second moment of area $[mm^4]$. I_{eff} is a function of the distance between the centroids of the chords and the section area of a chord, calculated as $I_{eff} = 0.5 h_0^2 A$.

Usage

```
effective_second_moment_of_area(h0, A)
```

Arguments

h0 Distance between centroids of chords [mm]

A Cross-section area of strut $[cm^2]$

Value

 I_{eff} Effective second moment of area $[mm^4]$

Euler_buckling_load Calculate the Euler buckling load

Description

Calculate the Euler buckling load [kN]

$$N_{cr,ch} = \frac{\pi^2 E I}{L_e^2}$$

Usage

Euler_buckling_load(Le, E, I)

Arguments

Le Effective length of strut [mm]

E Young modulus $[MPa \text{ or } MN/m^2]$

I - check 1: I_{yy} , second moment of area Axis y - y [cm^4]. Check 2: I_{eff} , Effective second moment of area [mm^4]. Check 3: I_{eff} or I_{zz} [mm^4]

Value

 N_{cr} Euler buckling load [kN]

extract_member_dimensions

Extract dimensions from reference table

Description

Function that looks into the Blue Book https://www.steelforlifebluebook.co.uk/ for dimensions and properties.

Usage

```
extract_member_dimensions(h, b, m, member_type, list_reference_tables)
```

Arguments

```
h Member height [mm]
b Member width [mm]
m Member mass [kg/m]
member_type Member type, 'UB' or 'UC'
list_reference_tables
List of reference tables
```

Value

- A Area of section $[cm^2]$
- tw Thickness of web [mm]
- tf Thickness of flange [mm]
- Iyy Second moment of area axis y y [cm^4]
- sh Depth of section [mm]
- sb Width of section [mm]
- Izz Second moment of area axis z z [cm^4]

```
first_order_bending_moment
```

Calculate first order bending moment about major axis y - y [kN.m]

Description

Calculate first order bending moment about major axis y - y [kN.m]

Usage

```
first_order_bending_moment(combined_vertical_load, L)
```

Arguments

combined_vertical_load

combined_vertical_load [kN/m]

Length of strut between restraints [m]

Details

Calculated as

$$M_{Ed}^{I} = 0.08 F L_{cr,y}^{2}$$

Value

 M_{Ed}^{I} First order bending moment [kN.m]

imperfection_factor_yy $Calculate\ the\ imperfection\ factor\ \alpha_yy\ for\ rolled\ section\ [dimensionless]$

Description

Calculate the imperfection factor α_{yy} for rolled section [dimensionless].

Usage

```
imperfection_factor_yy(h, b, tf)
```

Arguments

h Member height [mm] b Member width [mm]

tf thickness of the flange [mm]

Value

 α_{yy} Imperfection factor for y-y axis [dimensionless]

 ${\tt imperfection_factor_zz} \ \ {\it Calculate the imperfection factor} \ \alpha_{\it zz for rolled section}$

Description

Calculate the imperfection factor α_{zz} for rolled section.

Usage

```
imperfection_factor_zz(h, b, tf)
```

Arguments

h	Member	height	[mm]
b	${\bf Member}$	width [[mm]

tf thickness of the flange [mm]

Value

 α_{zz} Imperfection factor for z-z axis [dimensionless]

import_reference_BlueBook_tables

Import Reference BlueBook Tables from Excel files

Description

Import Reference BlueBook Tables from Excel files.

Usage

import_reference_BlueBook_tables()

Arguments

None

Value

List of 4 BlueBook reference tables

maximum_shear_force_in_the_lacing

Calculate the maximum shear force in the lacing

Description

Calculate the maximum shear force in the lacing, V_{E_d} [kN] (for a laced strut subject to a compressive axial force only)

 $V_{E_d} = \pi \, \frac{M_{E_d}}{L}$

Usage

maximum_shear_force_in_the_lacing(MEd, L)

Arguments

MEd Second order moment [kN.m]

Length of strut between restraints [m]

Value

 V_{E_d} Maximum shear force in the lacing [kN] (for a laced strut subject to a compressive axial force only)

max_compressive_axial_force_in_chords

Maximum compressive axial force in the chords

Description

Determine maximum compressive axial force in the chords at mid-length of the strut, N_{ch,E_d} [kN]

Usage

```
max_compressive_axial_force_in_chords(trial_member_size, member_type,
   steel_grade, k, L, n, Ad, Lch, E, h0, Ned, list_reference_tables,
   isTopLevel, DL, LL, AL, TL, Lcry)
```

Arguments

trial_member_size

Trial member size

member_type member_type, categorical: 'UC' or 'UB' steel_grade $[N/mm^2]$, categorical: 'S355' or 'S275'

k Coefficient of length as function of wall rigidity [dimensionless]

Length between two restraints [m]

n Number of lacing planes, default [n=2] Ad Section area of diagonal (lacing), $[cm^2]$

Length of chord of between restrains (lace points) [m]

E Young modulus $[GPa \text{ or } GN/m^2]$

h0 Distance between centroids of chords [m]

Ned Axial compression Force [kN]

list_reference_tables

List of reference tables

isTopLevel Is member located at top level? [boolean] DL Dead load / self-weight of member $\lceil kN/m \rceil$

LL Live load / imposed load [kN/m] AL Accidental Impact Load [kN/m]

TL Temperature load [kN/m]

Lcry critical length about major axis [m]

- N_{ch,E_d} Maximum compressive axial force in the chords [kN]
- S_v
- $N_{cr,ch}$
- M_{E_d}

overall_buckling_resistance_about_axis

 ${\it Calculate \ the \ overall \ buckling \ resistance \ of \ the \ memeber \ about \ the \ axis}$

Description

General case to compute the overall buckling resistance of the member, N_{b,R_d} [kN], about the axis, calculated as:

$$N_{b,R_d} = X N_{pl,R_d}$$

Usage

overall_buckling_resistance_about_axis(X, N_pl_Rd)

Arguments

X Slenderness reduction factor [dimentionless]

 N_pl_Rd Plastic resistance of the cross-section to compression [kN]

Value

 N_{b,R_d} Overall buckling resistance of the struts about the axis [kN]

plastic_resistance_of_cross_section_to_compression

 ${\it Calculate the plastic resistance of the cross-section \ to \ compression}$

Description

Calculate the plastic resistance of the cross-section to compression [N], based on cross-section area A and yield strength f_y .

Usage

plastic_resistance_of_cross_section_to_compression(A, fy)

Arguments

A Cross-section area of the strut $[cm^2]$

fy Yield strength $[kN/mm^2]$

Value

 N_{pl,R_d} Plastic resistance of the cross-section to compression [N]

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 $read_input_table$

Read input table from given Excel file

Description

Read input table from given Excel file.

Usage

```
read_input_table(file_name = "tables/input/trial1_kotik.xlsx")
```

Arguments

 ${\tt file_name}$

Path and file name of the input table

Value

Input table

relative_slenderness

Calculate the relative slenderness

Description

Calculate the relative slenderness [dimensionless]

$$\bar{\lambda} = \sqrt{\frac{N_{pl,R_d}}{N_{cr}}}$$

Usage

relative_slenderness(N_pl_Rd, Ncr)

Arguments

N_pl_Rd

Plastic resistance of the cross-section to compression [kN]

Ncr

Euler buckling load [kN]

Value

 $\bar{\lambda}$ Relative slenderness [dimentionless]

 ${\tt second_order_bending_moment}$

Calculate the second order bending moment

Description

Compute the second order bending moment, M_{E_d} [kN.m]. The maximum bending moment, including the bow imperfection and the second order effects, calculated as:

$$M_{E_d} = \frac{N_{E_d} e_0 + M_{E_d}^{I}}{1 - \frac{N_{E_d}}{N_{cr,Y}} - \frac{N_{E_d}}{S_v}}$$

Usage

second_order_bending_moment(L, Ned, Sv, Ncr, DL, LL, AL)

Arguments

L	Length of strut between restraints $[m]$
Ned	axial_compression_force $[kN]$
Sv	Shear stiffness for K-shape lacing $[kN]$
Ncr	Euler buckling load from check #2 global zz $[kN]$
DL	Dead load / self-weight of member $[kN/m]$
LL	Live load / imposed load $[kN/m]$
AL	Accidental Impact Load $[kN/m]$

Value

 M_{E_d} Second order moment [kN.m]

 $shear_force_at_support$ Shear force at support calculation

Description

Generate Shear force at support V_{Ed} [kN].

Usage

```
shear_force_at_support(DL, LL, L, AL)
```

Arguments

guments	
DL	Dead load / self-weight of member $[kN/m]$
LL	Live load / imposed load $[kN/m]$
L	Length of strut between restraints $[m]$
AL	Accidental Impact Load $[kN/m]$

Value

 V_{Ed} Shear force at support [kN]

shear_stiffness

Calculate the shear stiffness for K-shape lacing

Description

Calculate the shear stiffness for K-shape lacing [kN]. The expression of shear stiffness is:

$$S_v = \frac{n E A_d L_{ch} h_0^2}{d^3}$$

Usage

shear_stiffness(n = 2, Ad, Lch, E, h0)

Arguments

n	Number of planes of lacing, default $[n=2]$
Ad	Section area of diagonal (lacing), $[cm^2]$
Lch	Length of chord of betwee restrains (lace points) $[m]$
Е	Young modulus $[GPa \text{ or } GN/m^2]$
h0	Distance between centroids of chords $[m]$

Value

 S_v Shear stiffness for K-shape lacing [kN]

slenderness_reduction_factor

Calculate the slenderness reduction factor

Description

Calculate the slenderness reduction factor X [dimentionless] for the general case.

$$\Phi = 0.5 \left[1 + \alpha \left(\bar{\lambda} - 0.2 \right) + \bar{\lambda}^2 \right]$$

$$X = \frac{1}{\Phi + \sqrt{\Phi^2 - \bar{\lambda}^2}}$$

Usage

slenderness_reduction_factor(alpha, lambda_bar)

Arguments

alpha Check #1: imperfection factor α_{yy} for rolled section [dimentionless]. Check 2 & 3: imperfection factor α_{zz} for rolled section [dimentionless] lambda_bar, Relative slenderness $\bar{\lambda}$ [dimentionless]

Value

X Slenderness reduction factor [dimentionless]

temperature_load 23

temperature_load	Calculate	the	temperature	load
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Description

Calculate Temperature Load as a function of a surface changes of temperature, TL [kN]. Usually used for calculation of Axial Compression Force for the top level member.

$$TL = \alpha_T \, \delta_T \, k_T \, E \, A$$

Usage

```
temperature_load(alpha_T = 1.2e-05, delta_T = 10, k_T = 0.8, E = 210, A = 94.4)
```

Arguments

alpha $_{ extsf{T}}$	Thermal coefficient of expansion $[degC]$
delta_T	Change in temperature from the Installation temperature $[degC]$
k_T	Coefficient Of temperature effect [dimensionless]
Е	Young's Modulus of Elasticity $[GPa \text{ or } GN/m2]$

A Sectional area from table for given member size $[cm^2]$

Value

TL Temperature load [kN]

trial member size	Determine member size
triai_lilelliber_size	Determine memoer size

Description

Find optimized designation [height (mm) x width (mm) x mass (kg/m)] (also called member size) for given Axial Compression Force and critical length for major and minor axis. Searching into the tables based on the 'Compression' tables of the Blue Book https://www.steelforlifebluebook.co.uk/

Usage

```
trial_member_size(Lcry, Lcrz, Ned, steel_grade, member_type,
   list_reference_tables)
```

Arguments

 $\begin{array}{lll} \textbf{Lcry} & \text{critical length major axis } [m] \\ \textbf{Lcrz} & \text{critical length minor axis } [m] \\ \textbf{Ned} & \text{Axial compression force } [kN] \\ \end{array}$

steel_grade steel_grade $[N/mm^2]$, categorical: 'S355' or 'S275'

member_type member_type, categorical: 'UC' or 'UB'

 $list_reference_tables$

List of reference tables

24 yield_strength

Value

```
Member size [ height (mm) x width (mm) x mass (kg/m) ]
```

 $yield_strength$

Calculate the yield strength

Description

Calculate the yield strength, $f_y\ [N/mm^2]$

Usage

```
yield_strength(tw, tf, steel_grade)
```

Arguments

tw Thickness of the web [mm] tf Thickness of the flange [mm]

steel_grade steel_grade $[N/mm^2]$, categorical: 'S355' or 'S275'

Value

```
f_y Yield strength [N/mm^2]
```

Examples

```
yield_strength(tw=47.6, tf=77, steel_grade="S355")
```

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