# Package 'civilR'

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```

axial\_compression\_force

Calculate the axial compression force

#### Description

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Compute Axial Compression Force,  $N_{ed}$  [kN], for member without including Temperature effect. Used as trial for the top level strut where temperature changes could not be neglected. As well can be used to calculate final  $N_{ed}$  for struts from low levels of excavation, where temperature effect could be neglected.

## Usage

```
axial_compression_force(isTopLevel = T, DL = 1, LL = 1, L = 12.5,
P = 247, theta = 90, spacing = 6, Lcry = 12.5, Lcrz = 1.6,
steel_grade = "S355", member_type = "UB", alpha_T = 1.2e-05,
delta_T = 10, k_T = 0.8, E = 210, IL = 50, gamma = 1.35)
```

Is member leasted at ten level? [beelean]

## Arguments

icToployol

istopLevei	is member located at top level: [boolean]
DL	Dead load / self-weight of member $[kN/m]$
LL	Live load / imposed load $[kN/m]$
L	Total length of member $[m]$
P	Axial compression force of member per meter $\left[kN/m\right]$
theta	Angle to wall $[deg]$
spacing	spacing $[m]$
Lcry	critical length major axis $[m]$
Lcrz	critical length minor axis $[m]$
$steel\_grade$	steel_grade $[N/mm^2]$ , categorical: 'S355' or 'S275'

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$member_{-}type$	member_type, categorical: 'UC' or 'UB'
alpha_T	Thermal coef. of expansion $[degC]$
delta_T	Change in temperature from the Installation temperature $[degC]$
k_T	Coefficient Of Temperature Effect [dimensionless]

E Young's Modulus of Elasticity [GPa]

IL Accidental Impact Load [kN/m]

gamma Partial factor for action [dimensionless], as per EN 1990:2002 standard

## **Details**

First of all function check which combination govern in ULS (Ultimate Limit State) without including Temperature load, TL [kN]. Then include TL calculations for Load Combinations applying partial factors based on the Table A1.2(B), EN1990-2002, p53 Compare maximum from ULS and ALS to define which mistake could govern.

#### Value

 $N_{ed}$  Axial compression force [kN]

${\tt calculated\_NEd}$	$Generate\ calculated\ NEd$
-------------------------	-----------------------------

## Description

```
Generate calculated N_{E_d}, N_{E_{dc}} [kN].
```

#### Usage

```
calculated_NEd(N_b_Rd, Ieff, MEd, h0, A)
```

## Arguments

N_b_Rd	Overall buckling	resistance of	the struts a	about the	axis $[kN]$

Ieff Effective second moment of area  $[mm^4]$ 

MEd Second order moment [kN.m]

h0 Distance between centroids of chords [m]

A Cross-section area of strut  $[cm^2]$ 

## Value

```
N_{E_{dc}} Calculated N_{E_d} [kN]
```

check\_local\_buckling\_resistance\_about\_zz\_axis

Perform check #3, calculating the local buckling resistance of struts about minor z - z axis

## Description

Calculate the local buckling resistance of member about minor z-z axis, based on EC3 Approach.

$$L_e = kL_{ch}$$

[mm] where L is the critical length for buckling about minor axis z-z Steps of the check performed for laced struts:

1. Plastic resistance of the cross-section to compression [kN]

$$N_{pl,R_d,ch} = 2(A f y)$$

2. The Euler buckling load [kN]

$$N_{cr,ch} = \frac{\pi^2 E I_{zz}}{L_e^2}$$

3. Relative slenderness [dimensionless]

$$\bar{\lambda_{ch}} = \sqrt{\frac{N_{pl,R_d,ch}}{N_{cr,ch}}}$$

4. Calculate  $\Phi_{ch}$  parameter for slenderness reduction factor

$$\Phi_{ch} = 0.5 \left[ 1 + \alpha \left( \bar{\lambda_{ch}} - 0.2 \right) + \bar{\lambda_{ch}}^{2} \right]$$

5. Slenderness reduction factor [dimensionless]

$$X_{ch} = \frac{1}{\Phi_{ch} + \sqrt{{\Phi_{ch}}^2 - {\lambda_{ch}^-}^2}}$$

6. Output overall buckling resistance of the struts about z-z minor axis [kN]

$$N_{b,R_d,ch} = X_{ch} N_{pl,R_d,ch}$$

The partial factors  $\gamma_M$  that are applied to resistance of members to instability:  $\gamma_{M_1} = 1$ 

## Usage

check\_local\_buckling\_resistance\_about\_zz\_axis(trial\_member\_size,
 member\_type, steel\_grade, k, Lch, E)

#### Arguments

trial\_member\_size

Trial member size

member\_type, categorical: 'UC' or 'UB'

steel\_grade steel\_grade  $[N/mm^2]$ , categorical: 'S355' or 'S275'

k Coefficient [dimensionless] Lch Length of chord [mm]

E Young's Modulus of Elasticity [GPa]

#### Value

 $N_{b,Rd,X}$  Local buckling resistance of struts about z-z axis [kN]

check\_overall\_buckling\_resistance\_about\_yy\_axis

Perform check #1, calculating the overall buckling resistance of member about major y-y axis

#### Description

Calculate the overall buckling resistance of member about y-y axis, based on EC3 Approach.

$$L_e = kL$$

[mm] where L is the critical length for buckling about major axis y-y Steps of the check performed for laced struts:

1. Plastic resistance of the cross-section to compression [kN]

$$N_{pl,R_d} = 2(A f y)$$

2. The Euler buckling load [kN]

$$N_{cr,X} = \frac{\pi^2 E I_{yy}}{{L_c}^2}$$

3. Relative slenderness [dimensionless]

$$\bar{\lambda_X} = \sqrt{\frac{N_{pl,R_d}}{N_{cr,X}}}$$

4. Calculate  $\Phi_X$  parameter for slenderness reduction factor

$$\Phi_X = 0.5 \left[ 1 + \alpha \left( \bar{\lambda_X} - 0.2 \right) + \bar{\lambda_X}^2 \right]$$

5. Slenderness reduction factor [dimensionless]

$$X_x = \frac{1}{\Phi_X + \sqrt{\Phi_X^2 - \bar{\lambda_X}^2}}$$

6. Output overall buckling resistance of the struts about y-y axis [kN]

$$N_{b,R_d,X} = X_X N_{pl,R_d}$$

The partial factors  $\gamma_M$  that are applied to resistance of members to instability:  $\gamma_{M_1} = 1$ 

#### Usage

```
check_overall_buckling_resistance_about_yy_axis(trial_member_size,
  member_type, steel_grade, k, L, E)
```

## Arguments

trial\_member\_size

Trial member size

member\_type, categorical: 'UC' or 'UB'

 $steel\_grade$   $steel\_grade$   $[N/mm^2]$ , categorical: 'S355' or 'S275'

k Coefficient [dimensionless] L Total length of member [m]

E Young's Modulus of Elasticity [GPa]

#### Value

 $N_{b,Rd,X}$  Overall buckling resistance of struts about major y-y axis [kN]

 $check\_overall\_buckling\_resistance\_about\_zz\_axis$ 

Perform check #2, calculating the overall buckling resistance of struts about major z - z axis

## Description

Calculate the overall buckling resistance of member about z-z axis, based on EC3 Approach.

$$L_e = k L$$

[mm] where L is the critical length for buckling about major axis z-z Steps of the check performed for laced struts:

1. Plastic resistance of the cross-section to compression [kN]

$$N_{pl,R_d} = 2(A f y)$$

2. The Euler buckling load [kN]

$$N_{cr,Y} = \frac{\pi^2 E I_{eff}}{L_e^2}$$

3. Relative slenderness [dimensionless]

$$\bar{\lambda_Y} = \sqrt{\frac{N_{pl,R_d}}{N_{cr,Y}}}$$

4. Calculate  $\Phi_Y$  parameter for slenderness reduction factor

$$\Phi_Y = 0.5 \left[ 1 + \alpha \left( \bar{\lambda_Y} - 0.2 \right) + \bar{\lambda_Y}^2 \right]$$

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5. Slenderness reduction factor [dimensionless]

$$X_Y = \frac{1}{\Phi_Y + \sqrt{\Phi_Y^2 - \bar{\lambda_Y}^2}}$$

6. Output overall buckling resistance of the struts about z-z axis [kN]

$$N_{b,R_d,Y} = X_Y N_{pl,R_d}$$

The partial factors  $\gamma_M$  that are applied to resistance of members to instability:  $\gamma_{M_1}=1$ 

## Usage

```
check_overall_buckling_resistance_about_zz_axis(trial_member_size,
  member_type, steel_grade, k, L, E, h0)
```

## Arguments

trial\_member\_size

Trial member size

member\_type, categorical: 'UC' or 'UB'

steel\_grade steel\_grade  $[N/mm^2]$ , categorical: 'S355' or 'S275'

k Coefficient [dimensionless]

L Total length of member [m]

E Young's Modulus of Elasticity [GPa]

h0 Distance between centroids of chords [mm]

#### Value

 $N_{b,Rd,Y}$  Overall buckling resistance of struts about z-z axis [kN]

civilR Package civilR

## Description

Civil Engineering package.

 ${\tt compute\_output\_table} \quad \textit{Export output table to Excel file}$ 

## Description

Export output table to Excel file.

## Usage

```
compute_output_table(file_name = "tables/input/output_processed_table.xlsx")
```

## Arguments

file\_name

Path and file name of the output table

## Value

None

 ${\tt convert\_member\_dimensions\_string\_to\_elements}$ 

Convert individual member dimensions to a string

## Description

Convert individual member dimensions to a string.

## Usage

```
convert_member_dimensions_string_to_elements(s)
```

## Arguments

s String of the member dimensions

#### Value

- h Member height [mm]
- b Member width [mm]
- m Member weight [kg/m]

## convert\_member\_dimensions\_to\_string

 ${\it Convert \ the \ member \ size \ individual \ dimensions \ to \ a \ standard \ string}$ 

## Description

Generate a combined string from given three individual elements, separated by "x".

## Usage

```
convert_member_dimensions_to_string(h, b, m)
```

## Arguments

h	Member height $[mm]$
b	Member width $[mm]$
m	Member mass $[kg/m]$

## Value

String of the member dimensions

```
critical_length_major_axis_y
```

 $Calculate\ the\ critical\ length\ along\ major\ y\ axis$ 

## Description

Calculate the critical length along major y axis,  $L_{cry}$  [m]

#### Usage

```
critical_length_major_axis_y(L, Lkp, Lsp)
```

## Arguments

#### Value

 $L_{cry}$  Critical length along major y axis [m]

critical\_length\_minor\_axis\_z

Calculate the critical length along major z axis

## Description

Calculate the critical length along major z axis,  $L_{crz}$  [m]

## Usage

```
critical_length_minor_axis_z(L, Lkp, Lsp)
```

## Arguments

#### Value

 $L_{crz}$  Critical length along major z axis [m]

effective\_length\_of\_member

Calculate the effective length of member

## Description

Calculate the effective length of member,  $L_e$  [m].

## Usage

```
effective_length_of_member(k, L)
```

## Arguments

k Effective length coefficient [dimensionless] L Length of strut between restraints [m]

## Value

 $L_e$  Effective length of strut [m]

#### effective\_second\_moment\_of\_area

Calculate the effective second moment of area

## Description

Compute the effective second moment of area  $[mm^4]$ .  $I_{eff}$  is a function of the distance between the centroids of the chords and the section area of a chord, calculated as  $I_{eff} = 0.5 h_0^2 A$ .

## Usage

```
effective_second_moment_of_area(h0, A)
```

## Arguments

h0 Distance between centroids of chords [mm]

A Cross-section area of strut  $[cm^2]$ 

#### Value

 $I_{eff}$  Effective second moment of area  $[mm^4]$ 

Euler\_buckling\_load

Calculate the Euler buckling load

## Description

Calculate the Euler buckling load [kN]

$$N_{cr,ch} = \frac{\pi^2 E I}{L_e^2}$$

#### Usage

Euler\_buckling\_load(Le, E, I)

## Arguments

Le Effective length of strut [mm]

E Young modulus  $[MPa \text{ or } MN/m^2]$ 

I - check 1:  $I_{yy}$ , second moment of area Axis y - y [ $cm^4$ ]. Check 2:  $I_{eff}$ , Effective second moment of area [ $mm^4$ ]. Check 3:  $I_{eff}$  or  $I_{zz}$  [ $mm^4$ ]

## Value

 $N_{cr}$  Euler buckling load [kN]

extract\_member\_dimensions

Extract dimensions from reference table

## Description

Function that looks into the Blue Book https://www.steelforlifebluebook.co.uk/ for dimensions and properties.

## Usage

```
extract_member_dimensions(h, b, m, member_type)
```

## Arguments

 $\begin{array}{lll} & & & & & \\ & & & & \\ & & & & \\ & & & \\ & & & \\ & & & \\ & & & \\$ 

## Value

- A Area of section  $[cm^2]$
- tw Thickness of web [mm]
- tf Thickness of flange [mm]
- Iyy Second moment of area axis y y  $[cm^4]$
- sh Depth of section [mm]
- sb Width of section [mm]
- Izz Second moment of area axis z z  $[cm^4]$

imperfection\_factor\_yy  $Calculate\ the\ imperfection\ factor\ \alpha\_yy\ for\ rolled\ section\ [dimensionless]$ 

## Description

Calculate the imperfection factor  $\alpha_{yy}$  for rolled section [dimensionless].

## Usage

```
imperfection_factor_yy(h, b, tf)
```

## Arguments

h	Member height $[mm]$
b	Member width $[mm]$

tf thickness of the flange [mm]

## Value

 $\alpha_{yy}$  Imperfection factor for y-y axis [dimensionless]

imperfection\_factor\_zz Calculate the imperfection factor  $\alpha$ \_zz for rolled section

## Description

Calculate the imperfection factor  $\alpha_{zz}$  for rolled section.

## Usage

```
imperfection_factor_zz(h, b, tf)
```

## Arguments

h Member height [mm] b Member width [mm]

tf thickness of the flange [mm]

## Value

 $\alpha_{zz}$  Imperfection factor for z-z axis [dimensionless]

main

Run the high-level main flow

## Description

Run the high-level main flow.

## Usage

main()

## Value

Final results

maximum\_shear\_force\_in\_the\_lacing

Calculate the maximum shear force in the lacing

## Description

Calculate the maximum shear force in the lacing,  $V_{E_d}$  [kN] (for a laced strut subject to a compressive axial force only)

 $V_{E_d} = \pi \, \frac{M_{E_d}}{L}$ 

#### Usage

maximum\_shear\_force\_in\_the\_lacing(MEd, L)

#### Arguments

MEd Second order moment [kN.m]

Length of strut between restraints [m]

#### Value

 $V_{E_d}$  Maximum shear force in the lacing [kN] (for a laced strut subject to a compressive axial force only)

max\_compressive\_axial\_force\_in\_chords

Maximum compressive axial force in the chords

## Description

Determine maximum compressive axial force in the chords at mid-length of the strut,  $N_{ch,E_d}$  [kN]

## Usage

```
max_compressive_axial_force_in_chords(trial_member_size, member_type,
    steel_grade, k, L, n, Ad, Lch, E, h0, Ned)
```

## Arguments

trial\_member\_size

Trial member size

 ${\tt member\_type} \qquad {\tt member\_type}, \ {\tt categorical:} \ {\tt 'UC'} \ {\tt or} \ {\tt 'UB'}$ 

 ${\tt steel\_grade} \qquad {\tt steel\_grade} \ [N/mm^2], \ {\tt categorical:} \ {\tt 'S355'} \ {\tt or} \ {\tt 'S275'}$ 

k Coefficient of length as function of wall rigidity [dimensionless]

Length between two restraints [m]

n Number of lacing planes, default [n=2]

	e .		/= · ·	. 01
Ad	Section ar	ea of diagonal	(lacing)	$ cm^2 $
/ \U	SCOULDIT OIL	ou or aragonar	. (1001115/	10116

Length of chord of between restrains (lace points) [m]

E Young modulus  $[GPa \text{ or } GN/m^2]$ 

h0 Distance between centroids of chords [m]

Ned Axial compression Force [kN]

#### Value

 $N_{ch,E_d}$  Maximum compressive axial force in the chords [kN]

#### overall\_buckling\_resistance\_about\_axis

Calculate the overall buckling resistance of the memeber about the axis

## Description

General case to compute the overall buckling resistance of the member,  $N_{b,R_d}$  [kN], about the axis, calculated as:

$$N_{b,R_d} = X N_{pl,R_d}$$

## Usage

overall\_buckling\_resistance\_about\_axis(X, N\_pl\_Rd)

#### Arguments

X Slenderness reduction factor [dimentionless]

 $N_pl_Rd$  Plastic resistance of the cross-section to compression [kN]

#### Value

 $N_{b,R_d}$  Overall buckling resistance of the struts about the axis [kN]

## plastic\_resistance\_of\_cross\_section\_to\_compression

 ${\it Calculate}$  the plastic resistance of the cross-section to compression

## Description

Calculate the plastic resistance of the cross-section to compression [N], based on cross-section area A and yield strength  $f_y$ .

#### Usage

plastic\_resistance\_of\_cross\_section\_to\_compression(A, fy)

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## Arguments

A Cross-section area of the strut  $[cm^2]$ 

fy Yield strength  $[kN/mm^2]$ 

#### Value

 $N_{pl,R_d}$  Plastic resistance of the cross-section to compression [N]

process\_input\_table

 $Process\ input\ table$ 

## Description

Process input table.

## Usage

```
process_input_table()
```

#### Value

Processed table, adding computed outputs:

 $read\_input\_table$ 

Read input table from given Excel file

## Description

Read input table from given Excel file.

## Usage

```
read_input_table(file_name = "tables/input/trial1_kotik.xlsx")
```

## Arguments

 $file\_name$ 

Path and file name of the input table

## Value

Input table

relative\_slenderness 17

 $relative\_slenderness$   $Calculate\ the\ relative\ slenderness$ 

## Description

Calculate the relative slenderness [dimensionless]

$$\bar{\lambda} = \sqrt{\frac{N_{pl,R_d}}{N_{cr}}}$$

#### Usage

relative\_slenderness(N\_pl\_Rd, Ncr)

## Arguments

 $N_pl_Rd$  Plastic resistance of the cross-section to compression [kN]

Ncr Euler buckling load [kN]

## Value

 $\bar{\lambda}$  Relative slenderness [dimentionless]

 $second\_order\_bending\_moment$ 

Calculate the second order bending moment

## Description

Compute the second order bending moment,  $M_{E_d}$  [kN.m]. The maximum bending moment, including the bow imperfection and the second order effects, calculated as:

$$M_{E_d} = \frac{N_{E_d} e_0 + M_{E_d}^{\ I}}{1 - \frac{N_{E_d}}{N_{cr,Y}} - \frac{N_{E_d}}{S_v}}$$

## Usage

second\_order\_bending\_moment(L, Ned, Sv, Ncr)

## Arguments

Length of strut between restraints [mm]

Ned axial\_compression\_force [kN]

Sv Shear stiffness for K-shape lacing [kN]

Ncr Euler buckling load from check #2 global zz [kN]

#### Value

 $M_{E_d}$  Second order moment [kN.m]

shear\_stiffness

Calculate the shear stiffness for K-shape lacing

## Description

Calculate the shear stiffness for K-shape lacing [kN]. The expression of shear stiffness is:

$$S_v = \frac{n \, E \, A_d \, L_{ch} \, {h_0}^2}{d^3}$$

## Usage

shear\_stiffness(n = 2, Ad, Lch, E, h0)

#### Arguments

n	Number of planes of lacing, default $[n=2]$
Ad	Section area of diagonal (lacing), $[cm^2]$
Lch	Length of chord of betwee restrains (lace points) $[m]$
E	Young modulus $[GPa \text{ or } GN/m^2]$
h0	Distance between centroids of chords $[m]$

#### Value

 $S_v$  Shear stiffness for K-shape lacing [kN]

slenderness\_reduction\_factor

Calculate the slenderness reduction factor

## Description

Calculate the slenderness reduction factor X [dimentionless] for the general case.

$$\Phi = 0.5 \left[ 1 + \alpha \left( \bar{\lambda} - 0.2 \right) + \bar{\lambda}^2 \right]$$
 
$$X = \frac{1}{\Phi + \sqrt{\Phi^2 - \bar{\lambda}^2}}$$

## Usage

slenderness\_reduction\_factor(alpha, lambda\_bar)

## Arguments

alpha Check #1: imperfection factor  $\alpha_{yy}$  for rolled section [dimentionless]. Check 2 & 3: imperfection factor  $\alpha_{zz}$  for rolled section [dimentionless]

lambda\_bar, Relative slenderness  $\bar{\lambda}$  [dimentionless]

#### Value

X Slenderness reduction factor [dimentionless]

temperature load 19

temperature\_load

Calculate the temperature load

## Description

Calculate Temperature Load as a function of a surface changes of temperature, TL [kN]. Usually used for calculation of Axial Compression Force for the top level member.

$$TL = \alpha_T \, \delta_T \, k_T \, E \, A$$

#### Usage

```
temperature_load(alpha_T = 1.2e-05, delta_T = 10, k_T = 0.8, E = 210, A = 94.4)
```

#### Arguments

alpha\_T Thermal coefficient of expansion [degC]

delta\_T Change in temperature from the Installation temperature [degC]

k\_T Coefficient Of temperature effect [dimensionless] E Young's Modulus of Elasticity [GPa or GN/m2] A Sectional area from table for given member size [ $cm^2$ ]

#### Value

TL Temperature load [kN]

trial\_member\_size

Determine member size

## Description

Find optimized designation [ height (mm) x width (mm) x mass (kg/m) ] (also called member size) for given Axial Compression Force and critical length for major and minor axis. Searching into the tables based on the 'Compression' tables of the Blue Book https://www.steelforlifebluebook.co.uk/

#### Usage

```
trial_member_size(Lcry, Lcrz, Ned, steel_grade, member_type)
```

#### Arguments

 $\begin{array}{lll} \textbf{Lcry} & \text{critical length major axis } [m] \\ \textbf{Lcrz} & \text{critical length minor axis } [m] \\ \textbf{Ned} & \text{Axial compression force } [kN] \\ \end{array}$ 

steel\_grade steel\_grade  $[N/mm^2]$ , categorical: 'S355' or 'S275'

member\_type member\_type, categorical: 'UC' or 'UB'

## Value

```
Member size [ height (mm) x width (mm) x mass (kg/m) ]
```

20 yield\_strength

 $yield_strength$ 

Calculate the yield strength

## Description

```
Calculate the yield strength, f_y [N/mm^2]
```

## Usage

```
yield_strength(tw, tf, steel_grade)
```

## Arguments

tw Thickness of the web [mm] tf Thickness of the flange [mm]

 ${\tt steel\_grade} \qquad {\tt steel\_grade} \ [N/mm^2], \ {\tt categorical:} \ {\tt 'S355'} \ {\tt or} \ {\tt `S275'}$ 

## Value

```
f_y Yield strength [N/mm^2]
```

## Examples

```
yield_strength(tw=47.6, tf=77, steel_grade="S355")
```

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