

Chapter 7

Two-Body Systems

7.1 Two-Body Problem: Center-of-Mass Coordinates and Collisions

10/30:

- Announcements.
 - OH regular time but in KPTC 303.
- Today:
 - 2 body systems, i.e., 2 bodies in a uniform force field (usually gravity).
- Consider two particles with masses and positions m_1, \vec{r}_1 and m_2, \vec{r}_2 that exhibit forces on each other. We seek to describe their motion.
 - To do so, we'll first develop a coordinate system in which its easy to describe their motion.
 - Next, we'll write a Lagrangian for the system.
 - Then, we'll use it to find equations of motion.
- The first thing we'll do is develop a more convenient coordinate system than Cartesian coordinates in which to describe these two bodies.
 - We'll need the sum M of their masses, their center of mass \vec{R} , their relative position \vec{r} , and their reduced mass μ , given as follows.

$$M = m_1 + m_2 \quad \vec{R} = \frac{m_1 \vec{r}_1 + m_2 \vec{r}_2}{m_1 + m_2} \quad \vec{r} = \vec{r}_1 - \vec{r}_2 \quad \mu = \frac{m_1 m_2}{m_1 + m_2} = \frac{m_1 m_2}{M}$$

- In particular, (\vec{R}, \vec{r}) are our generalized coordinates.
 - Note: Switching to this new coordinate system is often colloquially referred to as a **diagonalization** of the system since the switch *uncouples* the equations of motion of the two particles.
 - Note: This is perhaps our first example of generalized coordinates (\vec{R}, \vec{r}) that aren't just shifted Cartesian coordinates.
- Next, we'll write the Lagrangian of the system, $L = T - V$.

- With respect to T , we can logically (albeit highly unintuitively) calculate that

$$\begin{aligned}
 T &= \frac{1}{2}m_1\dot{\vec{r}}_1^2 + \frac{1}{2}m_2\dot{\vec{r}}_2^2 \\
 &= \frac{1}{2} \left[\frac{(m_1^2 + m_1m_2)\dot{\vec{r}}_1^2 + (m_2^2 + m_1m_2)\dot{\vec{r}}_2^2}{m_1 + m_2} \right] \\
 &= \frac{1}{2} \frac{(m_1\dot{\vec{r}}_1 + m_2\dot{\vec{r}}_2)^2}{m_1 + m_2} + \frac{1}{2} \frac{m_1m_2}{m_1 + m_2} (\dot{\vec{r}}_1 - \dot{\vec{r}}_2)^2 \\
 &= \frac{1}{2}M\dot{\vec{R}}^2 + \frac{1}{2}\mu\dot{\vec{r}}^2
 \end{aligned}$$

- With respect to V , we have a uniform external force $m\vec{g}$ (e.g., $\vec{g} = -g\hat{i}$), so

$$\begin{aligned}
 V &= -m_1\vec{g} \cdot \vec{r}_1 - m_2\vec{g} \cdot \vec{r}_2 + V_{\text{int}}(\vec{r}_1 - \vec{r}_2) \\
 &= -M\vec{g} \cdot \vec{R} + V_{\text{int}}(\vec{r})
 \end{aligned}$$

- Thus, the final Lagrangian is

$$L = \frac{1}{2}M\dot{\vec{R}}^2 + M\vec{g} \cdot \vec{R} + \frac{1}{2}\mu\dot{\vec{r}}^2 - V_{\text{int}}(\vec{r})$$

- What is μ ?
 - The quantity that works. All of the above is “because it works” mathematics.
- We can now find equations of motion describing the two-body system.
 - Start with the E-L equations

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\vec{R}}_i} \right) = \frac{\partial L}{\partial \vec{R}_i} \qquad \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\vec{r}}_i} \right) = \frac{\partial L}{\partial \vec{r}_i}$$

- Substituting in the Lagrangian, we obtain

$$M\ddot{\vec{R}}_i = Mg_i \qquad \mu\ddot{\vec{r}}_i = -\frac{\partial V}{\partial \vec{r}_i} = F_i(\vec{r})$$

- The left equation tells us that the center of mass is uniformly accelerating.
 - The right equation is equivalent to a 1-particle problem.
- Summary of the above: The general method for solving two-body problems.
 1. Solve the 1-body EOM here.
 2. Transform back to \vec{r}_1, \vec{r}_2 coordinates, via

$$\vec{r}_1 = \vec{R} + \frac{m_2}{M}\vec{r} \qquad \vec{r}_2 = \vec{R} - \frac{m_1}{M}\vec{r}$$

- Descriptors of the system.
 - When L is separable, there are also 2 separately conserved energies.

$$\frac{1}{2}M\dot{\vec{R}}^2 - M\vec{g} \cdot \vec{R} = E_{\text{cm}} \qquad \frac{1}{2}\mu\dot{\vec{r}}^2 + V_{\text{int}}(\vec{r}) = E_{\text{int}}$$

- The total linear momentum of the system.

$$\vec{P} = m\dot{\vec{r}}_1 + m_2\dot{\vec{r}}_2 = M\dot{\vec{R}}$$

- The total angular momentum of the system.

$$\begin{aligned}
 \vec{J} &= m_1 \vec{r}_1 \times \dot{\vec{r}}_1 + m_2 \vec{r}_2 \times \dot{\vec{r}}_2 \\
 &= m_1 \left(\vec{R} + \frac{m_2}{M} \vec{r} \right) \times \left(\dot{\vec{R}} + \frac{m_2}{M} \dot{\vec{r}} \right) + m_2 \left(\vec{R} - \frac{m_1}{M} \vec{r} \right) \times \left(\dot{\vec{R}} - \frac{m_1}{M} \dot{\vec{r}} \right) \\
 &= M \vec{R} \times \dot{\vec{R}} + \mu \vec{r} \times \dot{\vec{r}}
 \end{aligned}$$

- The center of mass frame.

- Vectors in this frame are denoted with a superscript *.
- In the center of mass frame, we define $\vec{R}^* = 0$. That is, we let the origin of our coordinate system lie at the center of mass and move with it.
- We now explore some characteristics of this frame.
- It follows from this choice and the aforementioned coordinate transformations that

$$\vec{r}_1^* = \frac{m_2}{M} \vec{r} \quad \vec{r}_2^* = -\frac{m_1}{M} \vec{r}$$

- Additionally, the momenta of the two particle are equal and opposite:

$$m_1 \dot{\vec{r}}_1^* = -m_2 \dot{\vec{r}}_2^* = \mu \dot{\vec{r}} = \vec{p}^*$$

- It follows from the above that if the velocity of the center of mass is $\dot{\vec{R}}$, then we have

$$\vec{p}_1 = m_1 \dot{\vec{r}}_1 = m_1 \dot{\vec{R}} + \vec{p}^* \quad \vec{p}_2 = m_2 \dot{\vec{r}}_2 = m_2 \dot{\vec{R}} - \vec{p}^*$$

- The total momentum, angular momentum, and kinetic energy in the CM frame are

$$\vec{P}^* = 0 \quad \vec{J}^* = \mu \vec{r} \times \dot{\vec{r}} = \vec{r} \times \vec{p}^* \quad T^* = \frac{1}{2} \mu \dot{\vec{r}}^2 = \frac{(\vec{p}^*)^2}{2\mu}$$

- Once again, converting these values back to another frame in which the velocity of the center of mass is $\dot{\vec{R}}$, we obtain

$$\vec{P} = M \dot{\vec{R}} \quad \vec{J} = M \vec{R} \times \dot{\vec{R}} + \vec{J}^* \quad T = \frac{1}{2} M \dot{\vec{R}}^2 + T^*$$

- Example: Large satellite (e.g., moon around earth).

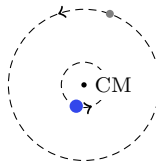


Figure 7.1: Moon and Earth in CM frame.

- Physically, the two tethered celestial bodies both orbit their center of mass.
- However, mathematically, this is equivalent to a particle of mass μ orbiting a fixed point mass M . Indeed, the EOM for \vec{r} is

$$\mu \ddot{\vec{r}} = -\hat{r} \frac{Gm_1m_2}{r^2} = -\hat{r} \frac{GM\mu}{r^2}$$

- Thus, the period of the (assumed) elliptical orbit can be calculated using the same methods as before. Indeed, we obtain

$$\left(\frac{\tau}{2\pi} \right) = \frac{a^3}{GM}$$

- However, note that a is the semimajor axis of the *relative* orbit (i.e., is the median distance between the bodies) and that M is the *sum* of the masses rather than the mass of the heavier body.
- Takeaway: Kepler's third law is only *approximately* correct.
- To conclude, let's discuss the motion of the Earth and moon in the CM frame.
 - Herein, the Earth orbits the CM with a small radius, and the moon orbits the CM directly across from the Earth in a much larger orbit.
 - Mathematically,

$$\vec{r}_1^* = \frac{m_2}{M} \vec{r} \qquad \vec{r}_2^* = -\frac{m_1}{M} \vec{r}$$

where we approximate

$$\frac{m_2}{M} \approx \frac{1}{82} \qquad \frac{m_1}{M} \approx \frac{81}{82}$$

- We now switch to an important application of this CM theory.
- Elastic collisions.

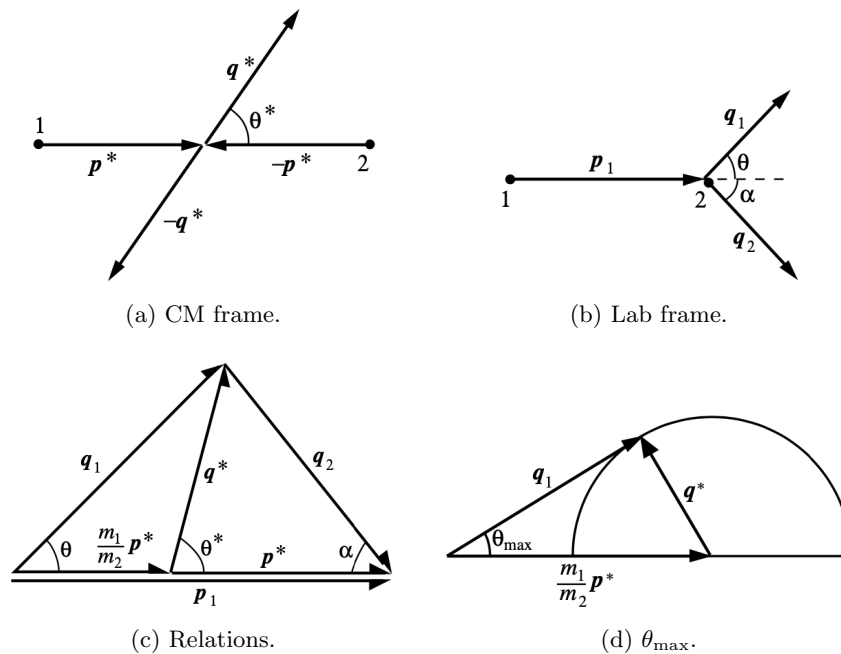


Figure 7.2: Elastic collisions.

- Background.
 - In an elastic collision, the kinetic energy is the same before and after.
 - Examples: Hard spheres, Coulomb force, gravity.
- Takeaways from Figure 7.2a.
 - Here's what an elastic collision looks like in the CM frame: We have two particles coming in, one with momentum \vec{p}^* and one with momentum $-\vec{p}^*$. After the collision, the particles separate with momenta \vec{q}^* and $-\vec{q}^*$.
 - Since energy is conserved,

$$T^* = \frac{(\vec{p}^*)^2}{2m} = \frac{(\vec{q}^*)^2}{2m}$$

- Thus, the magnitudes of the momenta before and after the collision are the same, i.e.,

$$p^* = q^*$$

– Takeaways from Figure 7.2b.

- In the lab, most elastic collision experiments begin with one incoming particle and one particle at rest.
- Denote by \vec{p}_1 the lab momentum of the incoming particle and by \vec{p}_2 the lab momentum of the resting particle. Note that

$$\vec{p}_1 = m_1 \dot{\vec{R}} + \vec{p}^* \qquad \vec{p}_2 = m_2 \dot{\vec{R}} - \vec{p}^*$$

- Now observe that $\vec{p}_2 = 0$. Then it follows from the right equation above that

$$\dot{\vec{R}} = \frac{1}{m_2} \vec{p}^*$$

- Substituting this into the left equation above yields

$$\vec{p}_1 = \frac{m_1}{m_2} \vec{p}^* + \vec{p}^* = \frac{M}{m_2} \vec{p}^*$$

- Therefore, employing the equations that shift you out of the CM frame and the above, we obtain

$$\begin{aligned} \vec{q}_1 &= m_1 \dot{\vec{R}} + \vec{q}^* & \vec{q}_2 &= m_2 \dot{\vec{R}} - \vec{q}^* \\ &= \frac{m_1}{m_2} \vec{p}^* + \vec{q}^* & &= \vec{p}^* - \vec{q}^* \end{aligned}$$

– Question to address: How much kinetic energy can be transferred during a collision?

- The lab kinetic energy transferred to the target particle is

$$T_2 = \frac{q_2^2}{2m_2}$$

- From Figure 7.2c, we have that

$$\alpha = \frac{1}{2}(\pi - \theta^*) \qquad q_2 = 2p^* \sin \frac{1}{2}\theta^*$$

- Combining these two results into the T_2 formula yields

$$\begin{aligned} T_2 &= \frac{2(p^*)^2}{m_2} \sin^2 \frac{1}{2}\theta^* \\ \frac{T_2}{T} &= \frac{\frac{2(p^*)^2}{m_2} \sin^2 \frac{1}{2}\theta^*}{\frac{p_1^2}{2m_1}} \\ &= \frac{\frac{2(p^*)^2}{m_2} \sin^2 \frac{1}{2}\theta^*}{\frac{M^2(p_1^*)^2}{2m_1 m_2^2}} \\ &= \frac{4m_1 m_2}{M^2} \sin^2 \frac{1}{2}\theta^* \end{aligned}$$

- The maximum occurs when $\theta^* = \pi$ and has value

$$\frac{T_2}{T} = \frac{4m_1 m_2}{M^2}$$

- Note that the expression on the right, above, equals unity when $m_1 = m_2$.

- Relating the lab and CM scattering angles.

$$\tan \theta = \frac{\sin \theta^*}{m_1/m_2 + \cos \theta^*}$$

- We read the above from Figure 7.2c by dropping a perpendicular from the upper vertex.
- If $m_1 = m_2$:

$$\theta = \frac{\theta^*}{2} \qquad \theta_{\max} = \frac{\pi}{2}$$

- If $m_1/m_2 > 1$:

$$\sin \theta_{\max} = \frac{m_2}{m_1}$$

- Example: An α particle can only be scattered by a proton by up to 14.5° , and a proton can only be scattered by an electron by up to 0.031° .
- Note that θ_{\max} can be visualized as in Figure 7.2d.