	<div>CONCESSION</div> <div>CONTINUATION SHEET</div>	Reference	TN-006536234
		Suffix	FW
		Revision	
		Page	2 of 2
		A/C Type	A350-900
		MSN	00790
<div>Description of Divergence</div> <div>V532-70357-202-00 MACHINED PANEL US V532-70166-200-00 BUTTSTRAP FR 20 US V532-70161-201-00 CFRP SHOVEL FWD US</div>		<div>Responsible Design Office Decision</div> <div>Reason for suffix FW: - Impacted drawing: V53124243 - Further work to be performed according to Sketches 3-4.</div>	



Validated

AIRBUS

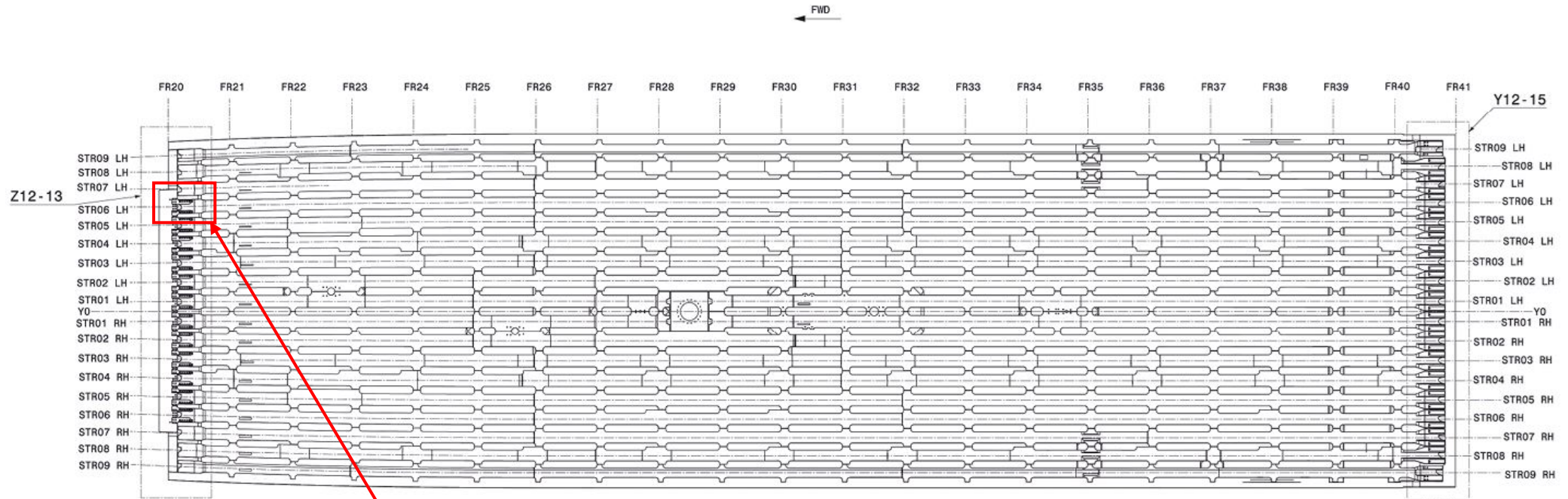
TN-006536234

Sketch

1

VIEW LOOKING OUTBOARD

20



AFFECTED AREA - FRAME 20, STRINGER 06 LH - 07 LH

NONCONFORMITY: OVERSIZE HOLE DRILLED

SEE MORE DETAILS IN SKETCHES.

SURFACE PROTECTION WILL BE COMPLETED DURING  
MANUFACTURING ACCORDING TO DESIGN SOLUTION.

TITLE	FIRST ANGLE PROJECTION	SIZE A0	E00
BUTTSTRAP INSTL C20	V53270011	SHEET 12	ISSUE



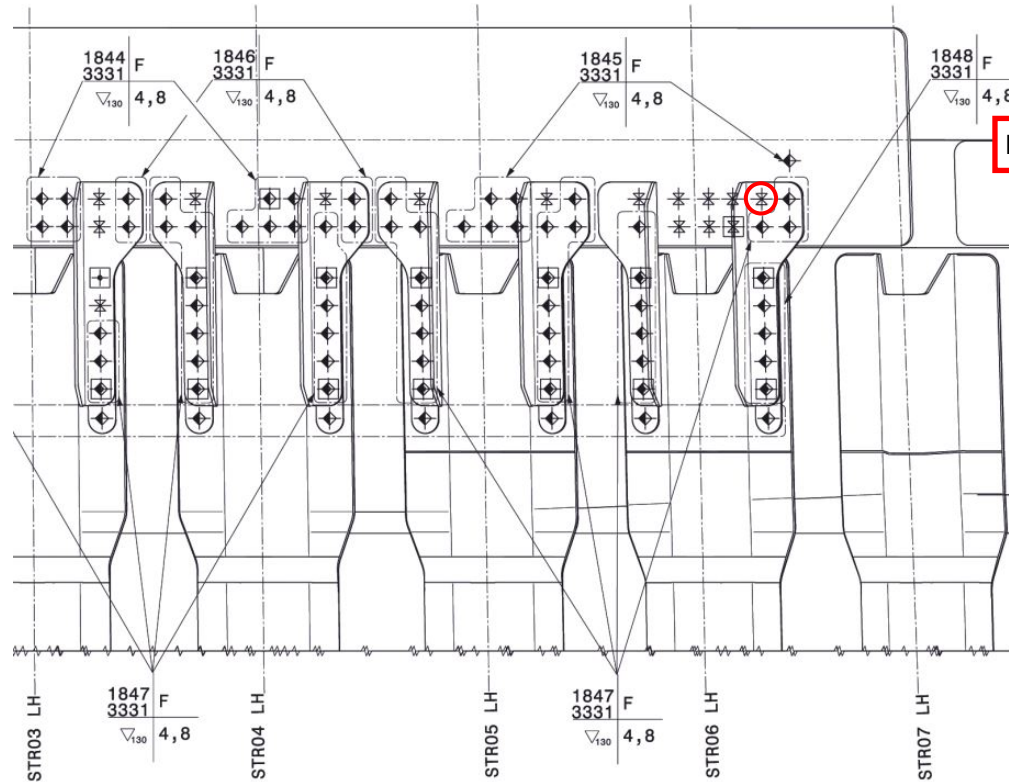
# AIRBUS

TN-006536234

Sketch

2

DETAIL V13-14



FR20

**NONCONFORMITY: 1x OVERSIZE HOLE DRILLED**

**SHOULD BE:**

**HOLE:**   $\rightarrow$  PILOT HOLE  $\varnothing 3,2\text{mm}$  H11

**IS:**

**HOLE:  $\varnothing 4,8$  (NO COUNTERSINK)**

**HOLE  $\varnothing 4,8$  DRILLED THROUGH "MACHINED PANEL US" AND "BUTTSTRAP FR 20 US"**

**HOLE DIAMETER IN "CFRP SHOVEL FWD US" IS  $\varnothing 3,2$**

**HOLE POSITION, EDGE DISTANCE, PITCH DISTANCE AND AFFECTED PARTS THICKNESS AS PER DRAWING.**

**AFFECTED PARTS:**


**V532-70357-202-00 MACHINED PANEL US**

**V532-70166-200-00 BUTTSTRAP FR 20 US**

**V532-70161-201-00 CFRP SHOVEL FWD US**

-   $\rightarrow$  ABS1707BC8V3A
-   $\rightarrow$  PILOT HOLE  $\varnothing 3,2\text{mm}$  H11
-   $\rightarrow$  DRILL OUT EXISTING TACK RIVET
-   $\rightarrow$  TACK HOLE

**All Dimensions in millimeters**

TITLE	FIRST ANGLE PROJECTION	SIZE A0	E00
BUTTSTRAP INSTL C20		V53270011	ISSUE
SHEET 14			



Validated

AIRBUS

TN-006536234

Sketch

3

**FURTHER WORK TO BE PERFORMED AT ASSEMBLY LEVEL FOR NON-CONFORMITY 0001:**

- Ream the non-conformity hole as per ABS1707BA0V3A geometry requirement through entire stack up.
- Complete the rest of installation as per nominal.

**CAUTION NOTE AT PRE-FAL LEVEL FOR NON-CONFORMITY 0001:**

- Take into account that, 1 hole was drilled to Ø4,8mm instead of Ø3,2mm on MACHINED PANEL US, BUTTSTRAP FR 20 US and CFRP SHOVEL FWD US at assembly level while performing the operations as per drawing V53124243 (See Figure 01)
- Special care to be taken while pinning or drilling template installation at Pre-FAL level as per DS: V53124243
- Complete the installation and protection as per definition.





AIRBUS

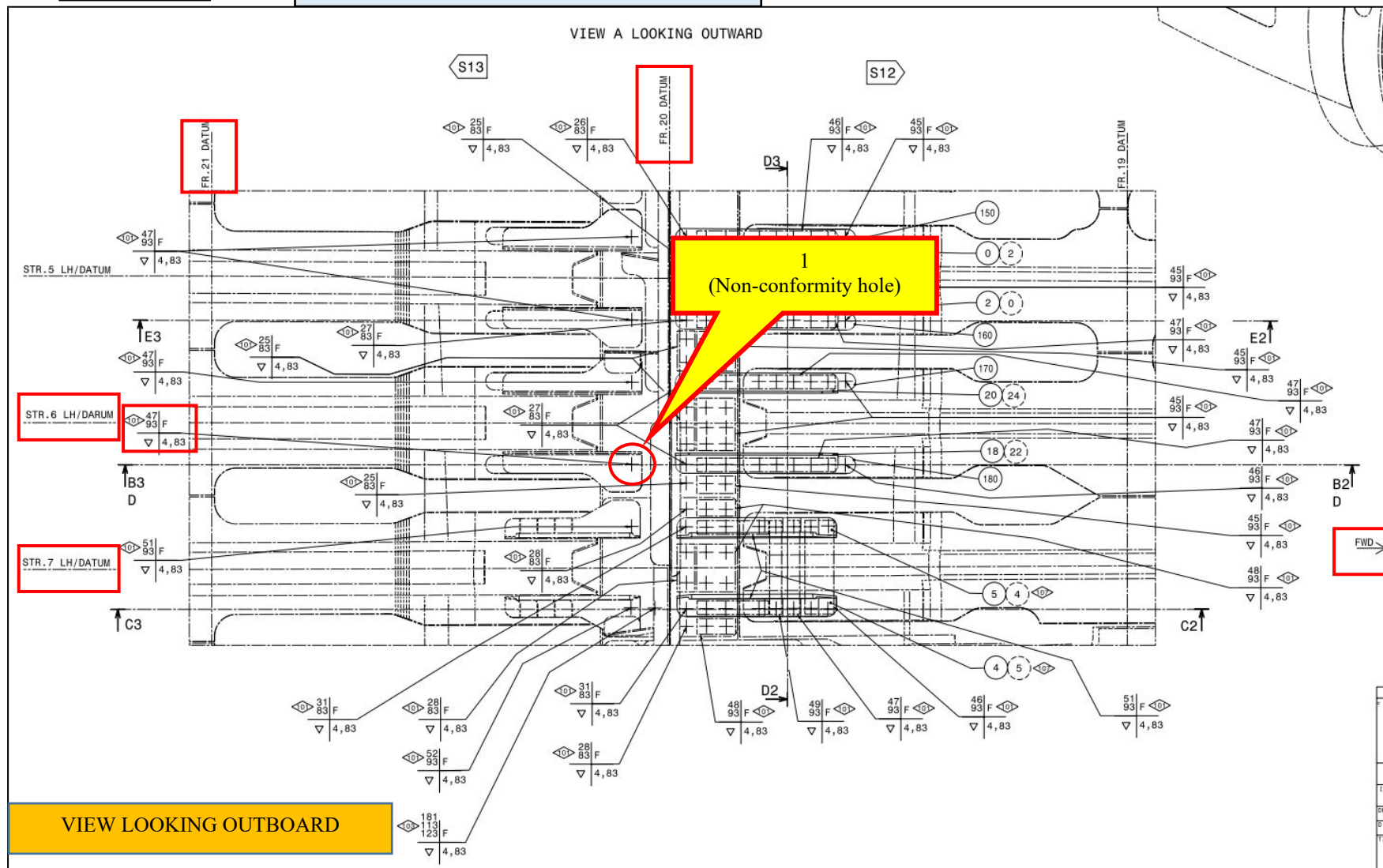
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Sketch

4

Figure 01

Extract from drawing V53124243



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