ME2055 Final Project (Solving Incompressible Navier-Stokes Equation in Parallel)

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This homework provides a overview of numerical solutions to the incompressible Navier-Stocks over a step profile. FTCS and point Gauss-Seidel are developed to solve unsteady vorticity equation and stream function equation respectively. To illustrate the effect of parallelization of the code, OpenMP (Shared Memory) is considered.

Categories and Subject Descriptors: [ME2055- Spring 2014]

1. INTRODUCTION

An important issue in fluid dynamics is the study of flow separation, which results in a global change of the flow field topology. One example is the flow separation and recirculation caused by a sudden contraction in a channel, in the form of a backward facing step. Incompressible Navier-Stokes equation is used and numerically applied to a 2D domain using uniform Cartesian mesh.

Fig. 1. Schematic of domain and Mesh grids.

The governing equation are given by the vonricity equation

$$\frac{\partial \omega}{\partial t} + u \frac{\partial \omega}{\partial x} + v \frac{\partial \omega}{\partial y} = \frac{1}{Re} \left(\frac{\partial^2 \omega}{\partial x^2} + \frac{\partial^2 \omega}{\partial y^2} \right) \tag{1}$$

and the stream function equation

$$\frac{\partial^2 \psi}{\partial x^2} + \frac{\partial^2 \psi}{\partial y^2} = -\omega \tag{2}$$

2. NUMERICAL SCHEME

Several numerical schemes for the solution of the vorticity transport equation have been reviewed in [1]. The FTCS scheme is used here to develop a numerical code. The formulation is rearranged as follows

$$\omega_{i,j}^{n+1} = \omega_{i,j}^{n} - \frac{\Delta t}{2\Delta x} u_{i,j}^{n} (\omega_{i+1,j}^{n} - \omega_{i-1,j}^{n}) - \frac{\Delta t}{2\Delta y} v_{i,j}^{n} (\omega_{i+1,j}^{n} - \omega_{i-1,j}^{n}) + \frac{\Delta t}{(\Delta x)^{2}} \frac{1}{Re} (\omega_{i+1,j}^{n} - 2\omega_{i,j}^{n} \omega_{i-1,j}^{n}) + \frac{\Delta t}{(\Delta y)^{2}} \frac{1}{Re} (\omega_{i,j+1}^{n} - 2\omega_{i,j}^{n} \omega_{i,j-1}^{n})$$
(3)

The stream function equation is an elliptic equation, and any one of the iterative schemes in [1] can be used to botain a solution. The point Gauss-Seidel scheme, given by Eq. (4) is used in numerical code.

$$\psi_{i,j}^{k+1} = \frac{1}{1+\beta^2} \left[(\Delta x)^2 \omega_{i,j}^{n+1} + \psi_{i+1,j}^k + \psi_{i-1,j}^{k+1} + \beta^2 (\psi_{i,j+1}^k + \omega_{i,j-1}^{k+1}) \right]$$
(4)

A step by step solution procedure us summarized as follows.

- 1- Initialize all the variables.
- 2- Update the vorticity within the domain by application of Eq. (3).
- 3- Solve Eq. (3) for the vorticity within the domain at the time level n+1.
- 4- Solve Eq. (4) for the stream function at the n+1 time level. Since an iterative scheme is used, a convergence criterion must be set.
- 5- Update the value of stream function where Neumann boundary condition is used.
- 6- Update the values of the vorticity at the boundaries.
- 7- Update the velocity field within the domain using finite difference equation for u and v (Eq. (5-6)).
- 8- Go to step 2 and repeat the computation up to specified final time level.

$$u_{i,j}^{n+1} = \frac{\psi_{i,j+1}^{n+1} - \psi_{i,j-1}^{n+1}}{2\Delta y}$$
 (5)

$$u_{i,j}^{n+1} = \frac{\psi_{i,j+1}^{n+1} - \psi_{i,j-1}^{n+1}}{2\Delta y}$$

$$v_{i,j}^{n+1} = -\frac{\psi_{i+1,j}^{n+1} - \psi_{i-1,j}^{n+1}}{2\Delta y}$$

$$\omega = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}$$

$$\omega = \frac{v_{i+1,j} - v_{i-1,j}}{2x} - \frac{u_{i+1,j} - u_{i-1,j}}{2y}$$
(7)

$$\omega = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}$$

$$\omega = \frac{v_{i+1,j} - v_{i-1,j}}{2x} - \frac{u_{i+1,j} - u_{i-1,j}}{2u} \tag{7}$$

During the computations as a measure of convergence to the steady state, I monitored residual for stream function and vorticity. The residual parameter, RES_{ψ} and RES_{ω} is normalized by the representative value at the previous time step. This then provides an indication of the maximum percent change in ψ in each iteration step.

$$RES_{\psi} = \sum_{i=2,j=2}^{i=n-1,j=m-1} \left| \frac{\psi_{i,j}^{k+1} - \psi_{i,j}^{k}}{\psi_{i,j}^{k}} \right|$$

$$RES_{\omega} = \sum_{i=2,j=2}^{i=n-1,j=m-1} \left| \frac{\omega_{i,j}^{k+1} - \omega_{i,j}^{k}}{\omega_{i,j}^{k}} \right|$$
(8)

In my calculations, I considered that convergence was achieved when RES_{ψ} and RES_{ω} are smaller than 10⁻⁴. Using the described numerical method, we obtained steady numerical solutions for up to Reynolds number of 400 and above this Reynolds number our numerical solution was not converging, but it was oscillating.

3. RESULTS

In Figs. (2-3), velocity contour patterns for two different Reynolds numbers (Re = 100, 400) are displayed, showing the increasing size of the recirculation region behind the step.

In addition to the primary recirculation zone, there exists a secondary recirculation zone near the upper wall for Re > 400. The adverse pressure gradient due to the sudden expansion at the edge of the step induces this separated flow [2].

3.1 Residual

One way to test the accuracy of CFD code is to validate physical laws or equations in the code. In order to do this for Incompressible Navier-Stokes equation, stream function could be tracked in every iteration. If

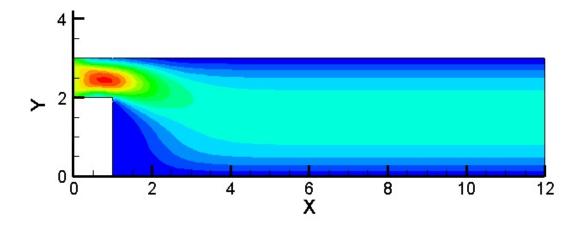


Fig. 2. Velocity contour for Re = 100

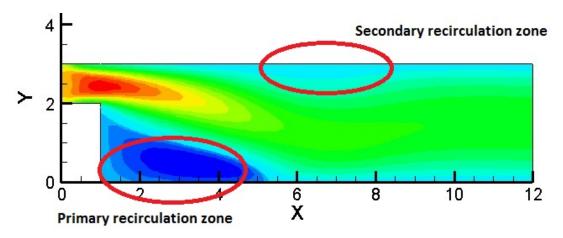


Fig. 3. Velocity contour for Re = 400

the difference of stream function values in n and n+1 iteration decreases, it could be concluded that, the code is going to converge. It leads us to conservation of mass.

In Fig. (4), the residual for both stream function and vorticity are shown. It can be concluded that, steady state solution is occurred after 600 number of iteration. The accuracy of solution is in order of 10^{-4} .

In Figs. (5-7), the velocity magnitude for different location in mesh domain is considered. They show that at first (X = 2) in the primary circulation zone, the velocity profile has slight change in magnitude but as y increases, the profile going to be parabolic. This pattern is also illustrated for (X = 6 and X = 10) with smaller effect of step on velocity profile. This behavior increases as Reynolds number grows.

3.2 Scalibility

As itwas mentioned before, to illustrate scalability, parallel programming on shared memory is used. Four different mesh sizes and three different computational procedure picked. In each mesh size, the running time

1:4 • S. Sammak

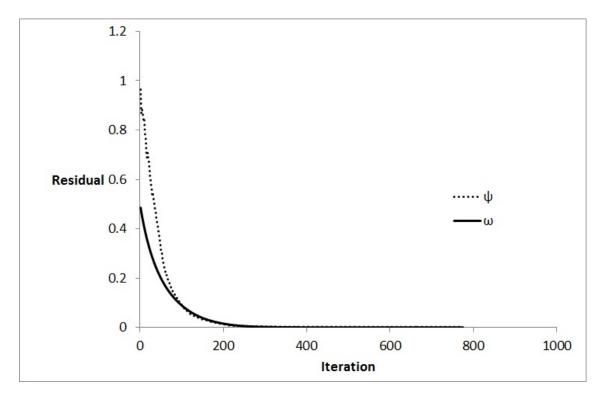


Fig. 4. Residual for stream function and vorticity

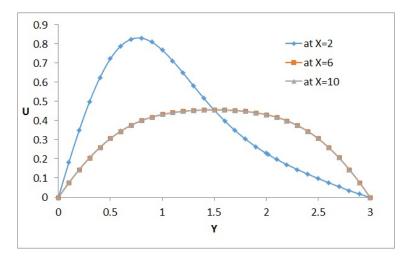


Fig. 5. Velocity magnitude for different location in y direction, for Re=100.

of the code on machine calculated. As the number of CPUs increases, the running time would decrease. In Table (I), the running time is shown serial and parallel and in the Fig. (8) they are compared to ideal scalability.

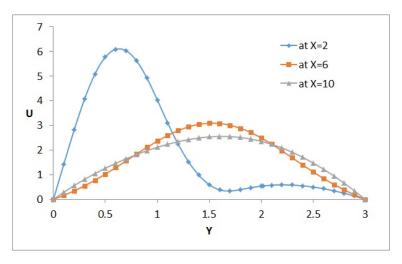


Fig. 6. Velocity magnitude for different location in y direction, from top to bottom Re = 200.

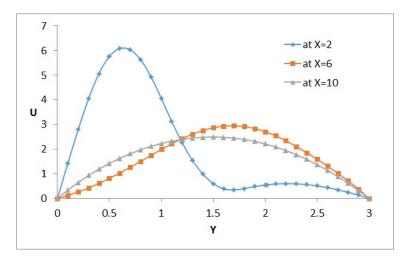


Fig. 7. Velocity magnitude for different location in y direction, for Re=400.

Table I. Run time for various grids and different number of CPUs in milliseconds

IIIIIIsecolids			
Mesh grids	One CPU	Two CPUs	Four CPUs
121*31	3622	2436	1332
241*61	30976	18948	8965
481*121	250352	141056	67213
961*241	2009519	1074875	519814

The number of mesh grids impact the run time of simulation. As it gets higher the speed up for two and four CPUs goes to two and four respectively. This scaling called weak speed up which is defined as how the solution time varies with the load per each CPU.

1:6 • S. Sammak

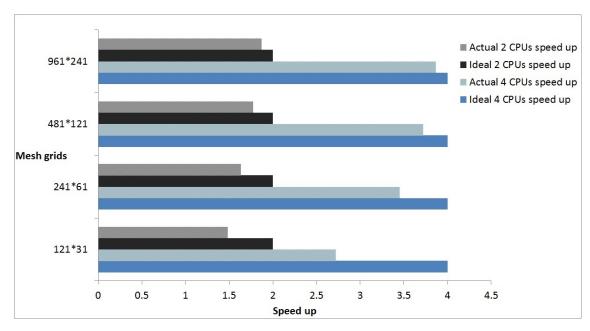


Fig. 8. Speed up for different mesh grids and number of CPUs.

4. CONCLUSION

The major numerical results are as follows. Contour plots of stream function, vorticity and velocities were presented at different Reynolds number.

I have presented accurate numerical solutions of the 2-D steady incompressible backward-facing step flow obtained using the upwind method. My results also indicate that the size of the recirculating regions grows as the Reynolds number increases.

References

- 1- Hoffmann, Klaus A., and Steve T. Chiang. "Computational fluid dynamics, Vol. 1." Wichita, KS: Engineering Education System (2000).
- 2- Erturk, Ercan. "Numerical solutions of 2-D steady incompressible flow over a backwardfacing step, Part I: High Reynolds number solutions." Computers and Fluids 37.6 (2008) 633-655.

5. APPENDIX

```
! main program
program incompns
implicit none
integer , parameter :: imax=100
integer , parameter :: jmax=300

integer im , jm , k , i , j
real*8 psi(imax , jmax)
real*8 w(imax , jmax)
, Vol. 2, No. 3, Article 1, Publication date: June 2010.
```

```
real *8 u(imax, jmax), v(imax, jmax)
real *8 check (imax, jmax)
real *8 dx, dy, duv
real*8 dt
real*8 nu
real*8 l,h
real *8 congs, conss
integer kmax
                                ! timing variables
integer :: t1, t2, cr
im=61
jm=61
l = 12.0
h = 3.0
dx=1/dfloat(im-1)
dy=h/dfloat(jm-1)
dt = 0.001
nu = 0.0025
congs = 0.001
conss = 0.002
! use only two threads
  !\$ call omp\_set\_num\_threads(4)
call system_clock( t1, cr )
do i = 1, im
         \mathbf{do} \hspace{0.1in} j \hspace{-0.1in} = \hspace{-0.1in} 1, jm
                  check(i,j)=0.0
         enddo
enddo
do i = 1,10
         do j = 1,10
                  check(i,j)=1.0
         enddo
enddo
kmax=100
k=0
! variable intialization
call initial (im, jm, psi, w, u, v)
```

```
1:8
       S. Sammak
5
       continue
k=k+1
call boundary_condition(im,jm,dx,dy,psi,w,u,v)
call update_omega(im,jm, dx, dy, w, u, v)
call ftcs(im, jm, dx, dy, dt, w, u, v, nu)
call pgs(im,jm, dx, dy, psi, w, congs)
call boundary_condition(im,jm,dx,dy,psi,w,u,v)
call velocity (im, jm, dx, dy, psi, u, v, duv)
call system_clock( t2, cr )
! display variation of velocity
write(*,15) k, duv
       format (2x, 'at_iteration', i6, ', duv=', f10.5)
! checking convergence
if(k.gt.kmax) then
       write(*,10) kmax
       format (2x, 'no_convegence_reached_in', i6, 'iteration')
else if (duv.gt.conss) then
       goto 5
endif
write (*,*) 'cpu_time_in_ms=', (t2-t1)*1000/cr
! write reasult in files
call write_in_file(im,jm, dx, dy, psi, w, u, v)
end program incompns
! initial subroutine
subroutine initial (im, jm, psi, w, u, v)
, Vol. 2, No. 3, Article 1, Publication date: June 2010.
```

```
implicit none
integer , parameter :: imax=100
integer , parameter :: jmax=300
\mathbf{integer} \ \mathrm{im} \, , \mathrm{jm} \, , \mathrm{k} \, , \ \mathrm{i} \, , \ \mathrm{j}
real *8 psi(imax, jmax)
real *8 w(imax, jmax)
real *8 u(imax, jmax), v(imax, jmax)
real*8 dx, dy
real*8 dt
real*8 nu
real*8 l,h
real *8 congs, conss
integer kmax
          !$omp parallel default(none) &
     !\$omp\ private(i,j)
     !$omp do
do i = 1, im
          do j=1, jm
                    psi(i, j) = 0.0
                   w(i, j) = 0.0
                   u(i, j) = 0.0
                    v(i, j) = 0.0
          enddo
enddo
          !$omp end do
     !$omp end parallel
return
end subroutine initial
! boundary condition subroutine
subroutine boundary_condition(im, jm, dx, dy, psi, w, u, v)
implicit none
integer,parameter :: imax=100
integer , parameter :: jmax=300
\mathbf{integer} \ \mathrm{im} \, , \mathrm{jm} \, , \mathrm{k} \, , \ \mathrm{i} \, , \ \mathrm{j}
real *8 psi(imax, jmax)
real *8 w(imax, jmax)
real *8 u(imax, jmax), v(imax, jmax)
```

```
real*8 dx, dy
real*8 dt
real*8 nu
real*8 l,h
\mathbf{real}*8 \text{ congs, conss}
integer kmax
real*8 :: psi_0, psi_20, psi_25, psi_30
integer i1, i2, i3, i4, j1, j2, j3
real*8 dx2, dy2
dx2=dx*dx
dy2=dy*dy
i1 = 20
j1 = 20
j2 = 6
j3 = 21
p s i_{-}0 = 0.0
psi_20 = 20.0
psi_25 = 25.0
psi_30 = 10.0
! boundary condition for psi
do i = 1, i1
        psi(i, j1) = psi_0
enddo
do j = 1, j1
        psi(i1,j) = psi_0
enddo
do i=i1+1, im
        psi(i,1) = psi_0
enddo
do i = 1, im
        psi(i, jm) = psi_30
enddo
do j=j1+1,jm
        psi(1,j) = psi(2,j)
enddo
```

[,] Vol. 2, No. 3, Article 1, Publication date: June 2010.

```
\mathbf{do} j=1,jm
        psi(im, j) = psi(im-1, j)
enddo
! boundary condition for vorticity w
do i = 1, i1
       w(i, j1) = 2.0*(psi(i, j1) - psi(i, j1+1))/dy2
enddo
do j = 1, j1
       w(1,j) = 2.0*(psi(i1,j)-psi(i1+1,j))/dx2
enddo
do i=i1+1,im
       w(i, j1) = 2.0*(psi(i, 1) - psi(i, 2))/dy2
enddo
do i = 1, im
       w(i, jm) = -2.0*(psi(i,jm)-psi(i,jm-1))/dy2
enddo
do j=j1, jm-1
       w(1,j) = 2.0*(psi(1,j)-psi(2,j))/dx2 &
                -(psi(1,j+1)-2.0*psi(1,j)+psi(1,j-1))*dy2
enddo
do j = 1, jm
       w(im, j) = 2.0*(psi(im, j)-psi(im-1, j))/dx2 &
                -(psi(im, j+1)-2.0*psi(im, j)+psi(im, j-1))*dy2
enddo
! boundary condition for velocity
do i=i1+1,im-1
                u(i,1)=0.0
                v(i, 1) = 0.0
enddo
do i = 1, i1
               u(i, j1+1)=0.0
                v(i, i1+1)=0.0
enddo
```

```
do j = 1, j1
                       u(i1, j) = 0.0
                       v(i1, j) = 0.0
enddo
do i = 1, im
                       u(i, jm) = 0.0
                       v(i, jm) = 0.0
enddo
do j=j1, jm-1
                       u(i,1) = -(psi(1,j1+1)-psi(1,j1))/dx
                       v(i, 1) = 0.0
enddo
do j = 1, jm-1
                       u(i,1) = -(psi(im, j+1)-psi(im, j))/dx
                       v(i,1) = 0.0
enddo
u(\mathrm{im}\,,\mathrm{jm})\!=\!0.0
v(im, jm) = 0.0
return
end subroutine boundary_condition
!\ subroutine\ update\_omega
subroutine update_omega(im,jm,dx,dy,w,u,v)
implicit none
integer , parameter :: imax=100
integer , parameter :: jmax=300
\mathbf{integer} \hspace{0.2cm} \mathrm{im} \hspace{0.1cm}, \mathrm{jm} \hspace{0.1cm}, \mathrm{k} \hspace{0.1cm}, \hspace{0.1cm} \mathrm{i} \hspace{0.1cm}, \hspace{0.1cm} \mathrm{j} \hspace{0.1cm}, \mathrm{i} \hspace{0.1cm} 1 \hspace{0.1cm}, \hspace{0.1cm} \mathrm{j} \hspace{0.1cm} 1
real *8 psi(imax, jmax)
real*8 w(imax,jmax)
real*8 u(imax,jmax),v(imax,jmax)
real*8 check(imax,jmax)
real*8 dx, dy
real*8 dt
```

```
real*8 nu
real*8 l,h
real *8 congs, conss
integer kmax
real *8 dvdx, dudy
i1 = 20
j1 = 20
        !$omp parallel default(none) &
    !\$omp\ private(i,j)
    !$omp do
do i = 2, im - 1
        i=0
        if(i < i1) j=j1
        do j = 2, jm - 1
                 dvdx = (v(i+1,j)-v(i-1,j))/2.0/dx
                 dudy = (u(i, j+1) - u(i, j-1))/2.0/dy
                 w(i, j) = dvdx - dudy
        enddo
enddo
         !$omp end do
        !\$omp\ end\ parallel
return
end subroutine update_omega
! subroutine ftcs for vorticity unsteady equation
subroutine ftcs (im, jm, dy, dt, w, u, v, nu)
implicit none
integer, parameter :: imax=100
integer,parameter :: jmax=300
integer im, jm, k, i, j, i1, j1
real *8 psi(imax, jmax)
real*8 w(imax,jmax)
real *8 u(imax, jmax), v(imax, jmax)
real *8 check (imax, jmax)
real*8 dx, dy
real*8 dt
real*8 nu
real*8 l,h
real *8 congs, conss
```

```
integer kmax
real*8 dw(imax,jmax)
real*8 dx2, dy2, dx22, dy22
real*8 udwdx, vdwdy, d2wdx2, d2wdy2
dx2=dx*dx
dy2=dy*dy
dx22 = 2.0 * dx
dy22 = 2.0 * dy
i1 = 20
j1 = 20
          !$omp parallel default(none) &
     !\$omp\ private(i,j)
     !$omp do
do i = 2, im - 1
          i=0
          if(i < i1) j=j1
          do j = 2, jm - 1
                    udwdx = u(i, j) *(w(i+1, j)-w(i-1, j))/dx22
                    vdwdy = v(i, j)*(w(i, j+1)-w(i, j-1))/dy22
                    d2wdx2 = (w(i+1,j)-2.0*w(i,j)+w(i-1,j))/dx2
                    d2wdy2 = (w(i, j+1)-2.0*w(i, j)+w(i, j-1))/dy2
                    dw(i,j) = dt * (-(udwdx + vdwdy) + nu*(d2wdx2 + d2wdy2))
          enddo
enddo
          !$omp end do
          !$omp end parallel
          !\$omp\ parallel\ default(none)\ \mathcal{E}
     !\$omp\ private(i,j)
     !$omp do
do i = 2, im - 1
          j = 0
          if(i < i1) j=j1
          do j = 2, jm - 1
                    w(\,i\,\,,\,j\,) = \,w(\,i\,\,,\,j\,) + dw(\,i\,\,,\,j\,\,)
          enddo
enddo
          !$omp end do
, Vol. 2, No. 3, Article 1, Publication date: June 2010.
```

```
!$omp end parallel
return
end subroutine ftcs
!\ subroutine\ point\ gauss-seidel
subroutine pgs (im, jm, dx, dy, psi, w, congs)
implicit none
integer , parameter :: imax=100
integer,parameter :: jmax=300
\mathbf{integer} \hspace{0.2cm} \mathrm{im} \hspace{0.1cm}, \mathrm{jm} \hspace{0.1cm}, \mathrm{k} \hspace{0.1cm}, \hspace{0.1cm} \mathrm{i} \hspace{0.1cm}, \hspace{0.1cm} \mathrm{j} \hspace{0.1cm}, \mathrm{i} \hspace{0.1cm} 1 \hspace{0.1cm}, \hspace{0.1cm} \mathrm{j} \hspace{0.1cm} 1
real *8 psi(imax, jmax)
real *8 w(imax, jmax)
real *8 u(imax, jmax), v(imax, jmax)
real*8 check(imax,jmax)
real*8 dx, dy
real*8 dt
real*8 nu
real*8 l,h
real *8 congs, conss
integer kmax
real*8 psi_old, error, errinit, err
real*8 :: dx2, dy2, beta2, cxy
dx2=dx*dx
dy2=dy*dy
beta2 = (dx2/dy2)
\cos 20.5/(1.0 + \cot 2)
k=0
i1 = 20
j1 = 20
5 continue
k=k+1
error = 0.0
            !$omp parallel default(none) &
      !\$omp\ private(i,j)
      !$omp do
do i = 2, im - 1
           j=0
```

```
1:16
         S. Sammak
        if(i < i1) j=j1
        do j = 2, jm-1
                 psi_old=psi(i,j)
                 psi(i,j)=cxy*(dx2*w(i,j)+psi(i+1,j)+psi(i-1,j) &
                         +beta2*(psi(i,j+1)+psi(i,j-1)))
                 error=error+abs(psi(i,j)-psi_old)
                 if(k.eq.1) then
                 errinit=error
                 endif
        enddo
enddo
        !$omp end do
        !$omp end parallel
err=error/errinit
if(k.gt.10000) then
        write(*,100)
else
        if(err.gt.congs) then
        goto 5
        endif
endif
100 format (2x, 'pgs_did_not_converge_in_10000_iterations')
```

```
return
end subroutine pgs
! subroutine velocity
subroutine velocity (im, jm, dx, dy, psi, u, v, duv)
implicit none
integer,parameter :: imax=100
integer,parameter :: jmax=300
\mathbf{integer} \ :: \ im\,, jm\,, k\,, \ i \;, \; j \;, i1 \;, j1
real *8 psi(imax, jmax)
real *8 w(imax, jmax)
real *8 u(imax, jmax), v(imax, jmax)
real *8 check (imax, jmax)
real*8 dx, dy, duv
real*8 dt
real*8 nu
real*8 l,h
```

```
real *8 congs, conss
integer kmax
real*8
         unew, vnew
real*8
         num, denum
num=0.0
denum=0.0
i1 = 20
j1 = 20
        !$omp parallel default(none) &
    !\$omp\ private(i,j)
    !$omp do
do i = 2, im - 1
        i=0
        if(i < i1) j=j1
        do j = 2, jm-1
                unew= (psi(i,j+1)-psi(i,j-1))/2.0/dy
                 vnew= (psi(i+1,j)-psi(i-1,j))/2.0/dx
                num = num + sqrt((unew - u(1,j)) **2.0 + (vnew - v(i,j)) **2.0)
                u(i,j)=unew
                 v(i, j) = vnew
        enddo
enddo
        !$omp end do
        !\$omp\ end\ parallel
        !$omp parallel default(none) &
    !\$omp\ private(i,j)
    !$omp do
do i = 2, im - 1
        i=0
        if(i < i1) j=j1
        do j = 2, jm-1
                denum=denum+ sqrt(u(i,j)**2.0+v(i,j)**2.0)
        enddo
enddo
        !$omp end do
        !\$omp\ end\ parallel
duv=num/denum
return
end subroutine velocity
! subroutine write_in_file
subroutine write_in_file(im,jm,dx,dy,psi,w,u,v)
```

```
implicit none
integer,parameter :: imax=100
integer,parameter :: jmax=300
integer im, jm, k, i, j
real *8 psi(imax, jmax)
real *8 w(imax, jmax)
real *8 u(imax, jmax), v(imax, jmax)
real *8 check (imax, jmax)
real *8 dx, dy
real*8 dt
real*8 nu
real*8 l,h
real *8 congs, conss
integer kmax
open(1, file='pbs8.dat')
write(1,111)im, jm, dx, dy
111 format ('im=', i4, '_; _jm=', i4, '; _dx=', f5.3, '; dy=', f5.3)
write(1,*) 'stream_function_distribution'
\mathbf{write}(1,*)
write (1,100) ((i-1)*dx, i=1,im,5)
do j=jm,1,-1
        write (1,200) (j-1)*dy, (psi(i,j), i=1,im,5)
enddo
write (1,*)
\mathbf{write}(1,*)
write(1,*)
write(1,*) 'vorticity_distribution'
\mathbf{write}(1,*)
write (1,100) ((i-1)*dx, i=1,im,5)
do j=jm,1,-1
        write (1,200) (j-1)*dy, (w(i,j), i=1,im,5)
enddo
write(1,*)
\mathbf{write}(1,*)
\mathbf{write}(1,*)
write(1,*) 'u_component_of_velocity'
write (1,*)
write(1,100) ((i-1)*dx, i=1,im,5)
, Vol. 2, No. 3, Article 1, Publication date: June 2010.
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```
do j = jm, 1, -1
                        write (1,200) (j-1)*dy, (u(i,j), i=1,im,5)
enddo
write(1,*)
 \mathbf{write}(1,*)
 \mathbf{write}(1,*)
 write(1,*) 'v_component_of_velocity'
write (1,*)
write (1,100) ((i-1)*dx, i=1,im,5)
do j=jm,1,-1
                        write (1,200) (j-1)*dy, (v(i,j), i=1,im,5)
enddo
write(1,*)
write(1,*)
 write (1,*)
 100 format (14x, 'y', 11('x=', f3.1, 6x))
200 format(f17.3, 2x, 11('x=', f3.1, 6x))
close(1)
 open(2, file='pbs8.dat')
write (2,221)
write(2,222)
 write(2,223) im, jm
221 format(1x, 'title="psi,w,u,v,velocity"')
222 format(1x, 'variables=_"x",_"y",_"psi",_"w",_"u",_"v","v","velocity"_')
223 format(1x, 'zone_t=_"1", _i=', i3, 2x, 'j=', i3,2x, 'f=point')
do j = 1, jm
                        do i = 1, im
                                                write(2,224) (i-1)*dx, (j-1)*dy, psi(i,j), w(i,j), u(i,j), v(i,j), sqrt(u(i,j), v(i,j), v(i
                        enddo
enddo
224 format (7(2x, d14.8))
close(2)
return
end subroutine write_in_file
```