

CHAPTER

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53-EFFECTIVE PAGES			53-050-00-01	STR (cont)		53-100-00-01	STR (cont)	
1 thru 14	JUN 15/2016		7	Feb 15/2016		7	Oct 15/2015	
53-010-00-01	STR		8	Oct 15/2015		8	Oct 15/2015	
1	Jun 15/2015		9	Oct 15/2015		9	Oct 15/2015	
2	Feb 15/2015		10	Oct 15/2015		53-110-00-01	STR	
3	Oct 15/2015		53-060-00-01	STR		1	Jun 15/2015	
4	Oct 15/2014		1	Jun 15/2015		2	Feb 15/2015	
5	Oct 15/2015		2	Feb 15/2015		3	Oct 15/2015	
53-010-00-02	STR		3	Oct 15/2015		4	Oct 15/2014	
1	Jun 15/2015		4	Oct 15/2014		5	Oct 15/2015	
2	Feb 15/2015		5	Oct 15/2015		6	Oct 15/2015	
3	Oct 15/2015		53-070-00-01	STR		7	Oct 15/2015	
4	Oct 15/2014		1	Jun 15/2015		8	Oct 15/2015	
5	Oct 15/2015		2	Feb 15/2015		9	Oct 15/2015	
53-020-00-01	STR		3	Oct 15/2015		53-120-00-01	STR	
1	Jun 15/2015		4	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		5	Oct 15/2015		2	Feb 15/2015	
3	Oct 15/2015		53-080-00-01	STR		3	Oct 15/2015	
4	Oct 15/2014		1	Jun 15/2015		4	Oct 15/2014	
5	Oct 15/2015		2	Feb 15/2015		5	Oct 15/2015	
53-030-00-01	STR		3	Oct 15/2015		6	Oct 15/2015	
1	Feb 15/2015		4	Oct 15/2014		7	Oct 15/2015	
2	Feb 15/2015		5	Oct 15/2015		8	Oct 15/2015	
3	Oct 15/2015		53-090-00-01	STR		9	Oct 15/2015	
4	Feb 15/2016		1	Jun 15/2015		10	Oct 15/2015	
53-030-00-02	STR		2	Feb 15/2015		53-130-00-01	STR	
1	Feb 15/2015		3	Oct 15/2015		1	Jun 15/2015	
2	Feb 15/2015		4	Oct 15/2014		2	Feb 15/2015	
3	Oct 15/2015		5	Oct 15/2015		3	Oct 15/2015	
4	Feb 15/2016		6	Oct 15/2015		4	Oct 15/2014	
53-050-00-01	STR		53-100-00-01	STR		5	Oct 15/2015	
1	Jun 15/2015		1	Jun 15/2015		6	Oct 15/2015	
2	Feb 15/2015		2	Feb 15/2015		7	Oct 15/2015	
3	Oct 15/2015		3	Oct 15/2015		8	Oct 15/2015	
4	Oct 15/2014		4	Oct 15/2014		53-140-00-01	STR	
5	Oct 15/2015		5	Oct 15/2015		1	Jun 15/2015	
6	Oct 15/2015		6	Oct 15/2015		2	Feb 15/2015	

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53-140-00-01	STR (cont)		53-180-00-01	STR (cont)		53-230-00-01	STR (cont)	
3	Oct 15/2015		4	Oct 15/2015		4	Oct 15/2014	
4	Oct 15/2014		5	Oct 15/2014		5	Oct 15/2015	
R 5	Jun 15/2016		6	Oct 15/2015		6	Oct 15/2015	
R 6	Jun 15/2016		7	Oct 15/2015		7	Oct 15/2015	
53-140-00-02	STR		53-190-00-01	STR		53-240-00-01	STR	
1	Jun 15/2015		1	Jun 15/2015		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
3	Oct 15/2015		3	Oct 15/2015		3	Oct 15/2015	
4	Oct 15/2014		4	Oct 15/2014		4	Oct 15/2014	
5	Oct 15/2014		5	Oct 15/2015		5	Oct 15/2014	
6	Oct 15/2014		6	Oct 15/2015		53-250-00-01	STR	
53-150-00-01	STR		53-200-00-01	STR		1	Jun 15/2015	
1	Jun 15/2015		1	Jun 15/2015		2	Feb 15/2015	
2	Feb 15/2015		2	Feb 15/2015		3	Oct 15/2015	
3	Oct 15/2015		3	Oct 15/2015		4	Oct 15/2014	
4	Oct 15/2014		4	Oct 15/2014		5	Oct 15/2014	
5	Oct 15/2014		5	Oct 15/2015		6	Oct 15/2014	
53-160-00-01	STR		6	Oct 15/2015		7	Oct 15/2014	
1	Jun 15/2015		53-210-00-01	STR		8	Oct 15/2014	
2	Feb 15/2015		1	Jun 15/2015		53-260-00-01	STR	
3	Oct 15/2015		2	Feb 15/2015		1	Jun 15/2015	
4	Oct 15/2014		3	Oct 15/2015		2	Feb 15/2015	
5	Oct 15/2014		4	Oct 15/2014		3	Oct 15/2015	
6	Oct 15/2014		5	Oct 15/2015		4	Oct 15/2014	
53-170-00-01	STR		6	Oct 15/2015		5	Oct 15/2015	
1	Jun 15/2015		7	Oct 15/2015		6	Oct 15/2015	
2	Feb 15/2015		53-220-00-01	STR		7	Oct 15/2015	
3	Oct 15/2015		1	Jun 15/2015		53-270-00-01	STR	
4	Oct 15/2014		2	Feb 15/2015		1	Jun 15/2015	
5	Oct 15/2015		3	Oct 15/2015		2	Feb 15/2015	
6	Oct 15/2015		4	Oct 15/2014		3	Feb 15/2015	
7	Oct 15/2015		5	Oct 15/2015				
53-180-00-01	STR		53-230-00-01	STR		4	Oct 15/2015	
1	Jun 15/2015		1	Jun 15/2015		5	Oct 15/2014	
R 2	Jun 15/2016		2	Feb 15/2015		6	Oct 15/2015	
3	Feb 15/2015		3	Oct 15/2015		7	Oct 15/2015	

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53-280-00-01	STR		53-330-00-01	STR		53-360-00-02	STR (cont)	
1	Jun 15/2015		1	Jun 15/2015		3	Oct 15/2015	
2	Feb 15/2015		2	Feb 15/2015		4	Oct 15/2014	
3	Oct 15/2015		3	Oct 15/2015		5	Oct 15/2014	
4	Oct 15/2014		4	Oct 15/2014		53-360-00-03	STR	
5	Oct 15/2015	R	5	Jun 15/2016		R	1	Jun 15/2016
6	Oct 15/2015	R	6	Jun 15/2016		2	Feb 15/2015	
53-290-00-01	STR		53-335-00-01	STR		3	Oct 15/2015	
1	Jun 15/2015		1	Jun 15/2015		4	Oct 15/2014	
2	Feb 15/2015		2	Oct 15/2015		5	Oct 15/2014	
3	Oct 15/2015		3	Oct 15/2014		53-360-00-04	STR	
4	Oct 15/2014	R	4	Jun 15/2016		R	1	Jun 15/2016
5	Oct 15/2014		5	Oct 15/2015		2	Feb 15/2015	
6	Oct 15/2014		6	Oct 15/2015		3	Oct 15/2015	
7	Oct 15/2014		53-340-00-01	STR		4	Oct 15/2014	
53-310-00-01	STR		1	Jun 15/2015		5	Oct 15/2014	
1	Feb 15/2015		2	Feb 15/2015		53-370-00-01	STR	
2	Feb 15/2015		3	Oct 15/2015		1	Jun 15/2015	
3	Oct 15/2015		4	Oct 15/2014		2	Feb 15/2015	
4	Oct 15/2015		5	Oct 15/2015		3	Oct 15/2015	
53-310-00-02	STR		6	Oct 15/2015		4	Oct 15/2014	
1	Feb 15/2015		53-350-00-01	STR		5	Oct 15/2015	
2	Feb 15/2015		1	Jun 15/2015		6	Oct 15/2015	
3	Oct 15/2015		2	Feb 15/2015		7	Oct 15/2015	
4	Oct 15/2015		3	Oct 15/2015		8	Oct 15/2015	
53-310-00-03	STR		4	Oct 15/2014		9	Oct 15/2015	
1	Feb 15/2015		5	Oct 15/2015		53-380-00-01	STR	
2	Feb 15/2015		6	Oct 15/2015		1	Jun 15/2015	
3	Oct 15/2015		53-360-00-01	STR		O	2	Jun 15/2016
4	Oct 15/2015	R	1	Jun 15/2016		3	Oct 15/2015	
53-310-00-04	STR		2	Feb 15/2015		4	Oct 15/2014	
1	Feb 15/2015		3	Oct 15/2015		5	Oct 15/2015	
2	Feb 15/2015		4	Oct 15/2014		6	Oct 15/2015	
3	Oct 15/2015		5	Oct 15/2014		7	Oct 15/2015	
4	Oct 15/2015		53-360-00-02	STR		8	Oct 15/2015	
		R	1	Jun 15/2016				
			2	Feb 15/2015				

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53-390-00-01	STR		53-430-00-01	STR (cont)		53-480-00-01	STR (cont)	
1	Jun 15/2015	R	3	Jun 15/2016		4	Oct 15/2014	
2	Feb 15/2015		4	Oct 15/2015		5	Oct 15/2015	
3	Oct 15/2015		5	Oct 15/2014		6	Oct 15/2015	
4	Oct 15/2014		6	Oct 15/2015		53-510-00-01	STR	
5	Oct 15/2015		53-440-00-01	STR		1	Feb 15/2016	
6	Oct 15/2015		1	Jun 15/2015		2	Feb 15/2016	
7	Oct 15/2015		2	Feb 15/2015		3	Feb 15/2016	
53-400-00-01	STR		3	Oct 15/2015		4	Feb 15/2016	
1	Jun 15/2015		4	Oct 15/2014		5	Feb 15/2016	
2	Oct 15/2015		5	Oct 15/2015		53-600-00-01	FAT	
3	Oct 15/2015		53-450-00-01	STR		1	Oct 15/2014	
4	Oct 15/2014		1	Jun 15/2015		2	Feb 15/2015	
5	Oct 15/2015		2	Feb 15/2015		53-600-00-02	FAT	
6	Oct 15/2015		3	Oct 15/2015		1	Oct 15/2014	
7	Oct 15/2015		4	Oct 15/2014		2	Feb 15/2015	
53-410-00-01	STR		5	Oct 15/2014		53-600-20-01	FAT	
1	Jun 15/2015		53-460-00-01	STR		1	Jun 15/2015	
2	Feb 15/2015		1	Jun 15/2015		2	Jun 15/2015	
3	Oct 15/2015		2	Feb 15/2015		53-600-20-02	FAT	
4	Oct 15/2014		3	Oct 15/2015		1	Jun 15/2015	
5	Oct 15/2015		4	Oct 15/2014		2	Jun 15/2015	
6	Oct 15/2015		5	Oct 15/2015		53-600-30-01	FAT	
7	Oct 15/2015		6	Oct 15/2015		1	Oct 15/2014	
8	Oct 15/2015		7	Oct 15/2015		2	Feb 15/2015	
53-420-00-01	STR		8	Oct 15/2015		53-600-30-02	FAT	
1	Jun 15/2015		53-470-00-01	STR		1	Oct 15/2014	
2	Feb 15/2015		1	Jun 15/2015		2	Feb 15/2015	
3	Oct 15/2015		2	Feb 15/2015		53-600-40-01	FAT	
4	Oct 15/2014		3	Oct 15/2015		1	Jun 15/2015	
5	Oct 15/2015		4	Oct 15/2014		2	Jun 15/2015	
6	Oct 15/2015		5	Oct 15/2015		53-600-40-02	FAT	
7	Oct 15/2015		6	Oct 15/2015		1	Jun 15/2015	
8	Oct 15/2015		53-480-00-01	STR		2	Jun 15/2015	
53-430-00-01	STR		1	Jun 15/2015		53-600-50-01	FAT	
1	Jun 15/2015		2	Feb 15/2015		1	Jun 15/2015	
R 2	Jun 15/2016		3	Oct 15/2015		2	Jun 15/2015	

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53-600-50-02	FAT		53-602-10-01	FAT		53-623-00-01	FAT	
1	Jun 15/2015		R 1	Jun 15/2016		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-601-01-01	FAT		53-602-10-02	FAT		53-624-00-01	FAT	
R 1	Jun 15/2016		R 1	Jun 15/2016		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		R 2	Jun 15/2016	
53-601-21-01	FAT		53-602-20-01	FAT		53-625-00-01	FAT	
R 1	Jun 15/2016		1	Oct 15/2014		R 1	Jun 15/2016	
2	Feb 15/2015		2	Feb 15/2015		R 2	Jun 15/2016	
53-601-21-02	FAT		53-605-10-01	FAT		53-626-00-01	FAT	
R 1	Jun 15/2016		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-601-30-01	FAT		53-606-40-01	FAT		53-626-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-601-30-02	FAT		53-606-50-01	FAT		53-627-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-601-41-01	FAT		53-606-70-01	FAT		53-627-00-02	FAT	
R 1	Jun 15/2016		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-601-41-02	FAT		53-618-00-01	FAT		53-628-00-01	FAT	
R 1	Jun 15/2016		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-601-50-01	FAT		53-619-00-01	FAT		53-628-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-601-50-02	FAT		53-620-00-01	FAT		53-629-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-601-60-01	FAT		53-621-00-01	FAT		53-629-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-601-60-02	FAT		53-622-00-01	FAT		53-630-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	

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53-631-00-01	FAT		53-638-00-02	FAT		53-644-00-02	FAT	
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2	Jun 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-633-00-01	FAT		53-639-00-01	FAT		53-645-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-634-00-01	FAT		53-639-00-02	FAT		53-645-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-635-00-01	FAT		53-640-00-01	FAT		53-645-01-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
R 2	Jun 15/2016		2	Feb 15/2015		2	Feb 15/2015	
53-635-00-02	FAT		53-640-00-02	FAT		53-645-01-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
R 2	Jun 15/2016		2	Feb 15/2015		2	Feb 15/2015	
53-636-00-01	FAT		53-641-00-01	FAT		53-646-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-636-00-02	FAT		53-641-00-02	FAT		53-646-10-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-636-10-01	FAT		53-642-00-01	FAT		53-646-20-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-636-10-02	FAT		53-642-00-02	FAT		53-646-30-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-637-00-01	FAT		53-643-00-01	FAT		53-646-40-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-637-00-02	FAT		53-643-00-02	FAT		53-646-50-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-638-00-01	FAT		53-644-00-01	FAT		53-646-60-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	

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53-646-61-01	FAT		53-649-01-01	FAT		53-656-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Oct 15/2015		2	Feb 15/2015	
53-646-62-01	FAT		53-649-01-02	FAT		53-657-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2015		1	Oct 15/2014	
R 2	Jun 15/2016		2	Oct 15/2015		2	Feb 15/2015	
53-646-63-01	FAT		53-650-00-01	FAT		53-657-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Oct 15/2015		2	Feb 15/2015	
53-646-70-01	FAT		53-650-00-02	FAT		53-658-00-01	FAT	
R 1	Jun 15/2016		1	Oct 15/2015		1	Oct 15/2014	
R 2	Jun 15/2016		2	Oct 15/2015		2	Feb 15/2015	
53-646-71-01	FAT		53-652-00-01	FAT		53-658-00-02	FAT	
R 1	Jun 15/2016		1	Oct 15/2014		1	Oct 15/2014	
R 2	Jun 15/2016		2	Feb 15/2015		2	Feb 15/2015	
53-646-75-01	FAT		53-653-00-01	FAT		53-659-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-646-80-01	FAT		53-653-00-02	FAT		53-659-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-647-00-01	FAT		53-654-00-01	FAT		53-660-00-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		R 1	Jun 15/2016	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-648-00-01	FAT		53-654-00-02	FAT		53-660-00-02	FAT	
1	Jun 15/2015		1	Jun 15/2015		R 1	Jun 15/2016	
2	Jun 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-648-00-02	FAT		53-655-00-01	FAT		53-661-00-01	FAT	
1	Jun 15/2015		1	Jun 15/2015		1	Jun 15/2015	
2	Jun 15/2015		2	Jun 15/2015		2	Jun 15/2015	
53-649-00-01	FAT		53-655-00-02	FAT		53-661-00-02	FAT	
1	Oct 15/2015		1	Jun 15/2015		1	Jun 15/2015	
2	Oct 15/2015		2	Jun 15/2015		2	Jun 15/2015	
53-649-00-02	FAT		53-656-00-01	FAT		53-662-00-01	FAT	
1	Oct 15/2015		1	Oct 15/2014		1	Jun 15/2015	
2	Oct 15/2015		2	Feb 15/2015		2	Jun 15/2015	

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53-662-00-02	FAT		53-668-10-01	FAT		53-674-00-02	FAT	
1	Jun 15/2015		1	Jun 15/2015		1	Jun 15/2015	
2	Jun 15/2015		2	Jun 15/2015		2	Jun 15/2015	
53-663-00-01	FAT		53-668-10-02	FAT		53-675-00-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Jun 15/2015	
2	Feb 15/2015		2	Jun 15/2015		2	Jun 15/2015	
53-663-00-02	FAT		53-669-00-01	FAT		53-675-00-02	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Jun 15/2015	
2	Feb 15/2015		2	Jun 15/2015		2	Jun 15/2015	
53-664-00-01	FAT		53-669-00-02	FAT		53-676-00-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Jun 15/2015	
2	Feb 15/2015		2	Jun 15/2015		2	Jun 15/2015	
53-665-00-01	FAT		53-670-00-01	FAT		53-676-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-665-00-02	FAT		53-671-00-01	FAT		53-676-10-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-666-00-01	FAT		53-671-00-02	FAT		53-676-10-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-666-00-02	FAT		53-672-00-01	FAT		53-677-00-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-667-00-01	FAT		53-672-00-02	FAT		53-677-00-02	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-667-00-02	FAT		53-673-00-01	FAT		53-678-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-667-10-01	FAT		53-673-00-02	FAT		53-678-00-02	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-667-10-02	FAT		53-674-00-01	FAT		53-679-00-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Jun 15/2015	
2	Feb 15/2015		2	Jun 15/2015		2	Jun 15/2015	

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53-679-00-02	FAT		53-686-00-02	FAT		53-697-00-02	FAT	
1	Jun 15/2015		1	Jun 15/2015		1	Jun 15/2015	
2	Jun 15/2015		2	Jun 15/2015		2	Jun 15/2015	
53-682-00-01	FAT		53-688-00-01	FAT		53-697-10-01	FAT	
1	Oct 15/2014		1	Jun 15/2015	R	1	Jun 15/2016	
2	Feb 15/2015		R 2	Jun 15/2016		2	Feb 15/2015	
53-682-00-02	FAT		53-688-00-02	FAT		53-698-00-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		R 2	Jun 15/2016		2	Feb 15/2015	
53-682-01-01	FAT		53-690-00-01	FAT		53-699-00-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Jun 15/2015	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-682-01-02	FAT		53-690-00-02	FAT		53-700-00-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-682-03-01	FAT		53-692-00-01	FAT		53-701-00-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-682-03-02	FAT		53-692-00-02	FAT		53-702-00-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-685-00-01	FAT		53-695-00-01	FAT		53-702-01-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-685-00-02	FAT		53-695-00-02	FAT		53-703-00-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-685-01-01	FAT		53-696-00-01	FAT		53-703-00-02	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-685-01-02	FAT		53-696-00-02	FAT		53-703-10-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-686-00-01	FAT		53-697-00-01	FAT		53-703-10-02	FAT	
1	Jun 15/2015		1	Jun 15/2015		1	Oct 15/2014	
2	Jun 15/2015		2	Jun 15/2015		2	Feb 15/2015	

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53-704-00-01	FAT		53-710-00-01	FAT		53-716-00-01	FAT	
1	Oct 15/2014		R 1	Jun 15/2016		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-704-00-02	FAT		53-710-00-02	FAT		53-717-00-01	FAT	
1	Oct 15/2014		R 1	Jun 15/2016		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-705-00-01	FAT		53-711-00-01	FAT		53-718-00-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Jun 15/2015	
2	Jun 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-705-00-02	FAT		53-711-00-02	FAT		53-719-00-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Jun 15/2015	
2	Jun 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-706-00-01	FAT		53-711-01-01	FAT		53-720-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-706-00-02	FAT		53-711-01-02	FAT		53-721-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-707-00-01	FAT		53-712-00-01	FAT		53-721-10-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-707-00-02	FAT		53-713-00-01	FAT		53-721-20-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-708-00-01	FAT		53-713-01-01	FAT		53-722-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-708-00-02	FAT		53-714-00-01	FAT		53-723-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-709-00-01	FAT		53-714-01-01	FAT		53-723-00-02	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-709-00-02	FAT		53-715-00-01	FAT		53-724-00-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Jun 15/2015	
2	Jun 15/2015		2	Feb 15/2015		2	Jun 15/2015	

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53-724-00-02	FAT		53-732-00-01	FAT		53-737-10-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Jun 15/2015	
2	Jun 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-725-00-01	FAT		53-732-01-01	FAT		53-738-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-725-00-02	FAT		53-732-10-01	FAT		53-738-00-02	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Jun 15/2015		2	Feb 15/2015	
53-726-00-01	FAT		53-733-00-01	FAT		53-738-01-01	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-726-00-02	FAT		53-733-01-01	FAT		53-738-01-02	FAT	
1	Jun 15/2015		1	Oct 15/2014		1	Oct 15/2014	
2	Jun 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-727-00-01	FAT		53-734-00-01	FAT		53-739-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-728-00-01	FAT		53-734-01-01	FAT		53-740-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-729-00-01	FAT		53-735-00-01	FAT		53-741-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Jun 15/2015	
2	Feb 15/2015		2	Feb 15/2015		2	Jun 15/2015	
53-729-01-01	FAT		53-735-01-01	FAT		53-742-00-01	FAT	
A 1	Jun 15/2016		A 1	Jun 15/2016		1	Jun 15/2015	
A 2	Jun 15/2016		A 2	Jun 15/2016		2	Jun 15/2015	
53-730-00-01	FAT		53-736-00-01	FAT		53-743-00-01	FAT	
1	Oct 15/2014		1	Oct 15/2014		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-731-00-01	FAT		53-737-00-01	FAT		53-744-00-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	
53-731-01-01	FAT		53-737-01-01	FAT		53-745-00-01	FAT	
1	Oct 15/2014		1	Jun 15/2015		1	Oct 15/2014	
2	Feb 15/2015		2	Feb 15/2015		2	Feb 15/2015	

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Subject/Page	Date	COC	Subject/Page	Date	COC	Subject/Page	Date	COC	
53-746-00-01	FAT		53-757-00-01	FAT		53-804-00-01	ZON (cont)		
1	Oct 15/2014		1	Jun 15/2015		4	Feb 15/2015		
2	Feb 15/2015		2	Jun 15/2015		53-806-00-01	ZON		
53-747-00-01	FAT		53-758-00-01	FAT		1	Feb 15/2015		
1	Jun 15/2015		1	Oct 15/2014		2	Feb 15/2015		
2	Jun 15/2015		2	Feb 15/2015		3	Feb 15/2015		
53-748-00-01	FAT		53-758-00-02	FAT		4	Feb 15/2015		
1	Jun 15/2015		1	Oct 15/2014		5	Feb 15/2015		
2	Jun 15/2015		2	Feb 15/2015		53-808-00-01	ZON		
53-749-00-01	FAT		53-759-00-01	FAT		1	Feb 15/2015		
1	Oct 15/2014		1	Oct 15/2014		2	Feb 15/2015		
2	Feb 15/2015		2	Feb 15/2015		3	Feb 15/2015		
53-749-10-01	FAT		53-759-00-02	FAT		4	Feb 15/2015		
1	Oct 15/2014		1	Oct 15/2014		53-810-00-01	ZON		
2	Feb 15/2015		2	Feb 15/2015		1	Feb 15/2015		
53-750-00-01	FAT		53-760-00-01	FAT		2	Feb 15/2015		
1	Oct 15/2014		1	Oct 15/2014		3	Feb 15/2015		
2	Feb 15/2015		2	Feb 15/2015		53-812-00-01	ZON		
53-751-00-01	FAT		53-760-00-02	FAT	R	1	Jun 15/2016		
1	Oct 15/2014		1	Oct 15/2014		2	Feb 15/2016		
2	Feb 15/2015		2	Feb 15/2015		3	Feb 15/2015		
53-752-00-01	FAT		53-800-00-01	ZON		4	Feb 15/2015		
1	Oct 15/2014	R	1	Jun 15/2016		53-814-00-01	ZON		
2	Feb 15/2015		2	Feb 15/2015		1	Feb 15/2015		
53-753-00-01	FAT		3	Feb 15/2015		2	Feb 15/2015		
1	Oct 15/2014		4	Feb 15/2015		3	Feb 15/2015		
2	Feb 15/2015		5	Feb 15/2015		4	Feb 15/2015		
53-754-00-01	FAT		6	Feb 15/2015		53-816-00-01	ZON		
1	Oct 15/2014		53-802-00-01	ZON		1	Oct 15/2014		
2	Feb 15/2015	R	1	Jun 15/2016		2	Feb 15/2015		
53-755-00-01	FAT		R	2	Jun 15/2016		3	Feb 15/2015	
1	Oct 15/2014		3	Feb 15/2015		4	Feb 15/2015		
2	Feb 15/2015		53-804-00-01	ZON		53-818-00-01	ZON		
53-756-00-01	FAT		1	Feb 15/2015		1	Feb 15/2015		
1	Oct 15/2014		2	Feb 15/2015		2	Feb 15/2015		
2	Feb 15/2015		3	Feb 15/2015		3	Feb 15/2015		

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53-820-00-01	ZON		53-836-00-01	ZON		53-850-00-01	ZON (cont)	
1	Feb 15/2015		1	Feb 15/2015		5	Feb 15/2015	
2	Feb 15/2015		2	Feb 15/2015		53-852-00-01	ZON	
3	Feb 15/2015		3	Feb 15/2015		1	Feb 15/2015	
4	Feb 15/2015		53-838-00-01	ZON		2	Feb 15/2015	
53-822-00-01	ZON		1	Feb 15/2015		3	Feb 15/2015	
1	Feb 15/2015		2	Feb 15/2015		53-854-02-01	ZON	
2	Feb 15/2015		3	Feb 15/2015		1	Jun 15/2015	
3	Feb 15/2015		53-840-00-01	ZON		2	Feb 15/2015	
53-824-01-01	ZON		1	Feb 15/2015		3	Feb 15/2015	
1	Oct 15/2014		2	Feb 15/2015		53-856-01-01	ZON	
2	Feb 15/2015		3	Feb 15/2015		1	Feb 15/2015	
53-826-00-01	ZON		4	Feb 15/2015		2	Feb 15/2015	
1	Feb 15/2015		53-842-00-01	ZON		3	Feb 15/2015	
2	Feb 15/2015		1	Feb 15/2015		53-858-00-01	ZON	
3	Feb 15/2015		2	Feb 15/2015		1	Feb 15/2015	
53-828-00-01	ZON		3	Feb 15/2015		2	Feb 15/2015	
1	Feb 15/2015		53-844-00-01	ZON		3	Feb 15/2015	
2	Feb 15/2015	R	1	Jun 15/2016		53-860-00-01	ZON	
3	Feb 15/2015		2	Feb 15/2015		1	Feb 15/2015	
4	Feb 15/2015		3	Feb 15/2015		2	Feb 15/2015	
5	Feb 15/2015		4	Feb 15/2015		3	Feb 15/2015	
53-830-00-01	ZON		53-846-00-01	ZON		4	Feb 15/2015	
R	1	Jun 15/2016	1	Feb 15/2015		5	Feb 15/2015	
2	Feb 15/2015		2	Feb 15/2015		53-862-00-01	ZON	
3	Feb 15/2015		3	Feb 15/2015		1	Feb 15/2015	
4	Feb 15/2015		4	Feb 15/2015		2	Feb 15/2015	
53-832-00-01	ZON		53-848-00-01	ZON		3	Feb 15/2015	
1	Jun 15/2015		1	Feb 15/2015		4	Feb 15/2015	
2	Feb 15/2015		2	Feb 15/2015		5	Feb 15/2015	
3	Feb 15/2015		3	Feb 15/2015		6	Feb 15/2015	
4	Feb 15/2015		4	Feb 15/2015		7	Feb 15/2015	
53-834-00-01	ZON		53-850-00-01	ZON		8	Feb 15/2015	
1	Feb 15/2015		1	Feb 15/2015		53-864-00-01	ZON	
2	Feb 15/2015		2	Feb 15/2015		1	Feb 15/2015	
3	Feb 15/2015		3	Feb 15/2015		2	Feb 15/2015	
4	Feb 15/2015		4	Feb 15/2015		3	Feb 15/2015	

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53-866-00-01	ZON		53-884-00-01	ZON		53-900-00-01	ZON (cont)	
1	Feb 15/2015		1	Oct 15/2014		3	Feb 15/2015	
2	Feb 15/2015		2	Feb 15/2015		53-902-00-01	ZON	
3	Feb 15/2015		3	Feb 15/2015		1	Feb 15/2015	
4	Feb 15/2015		53-886-00-01	ZON		2	Feb 15/2015	
53-868-00-01	ZON		1	Oct 15/2014		3	Feb 15/2015	
1	Feb 15/2015		2	Feb 15/2015		53-904-01-01	ZON	
2	Feb 15/2015		3	Feb 15/2015		R 1	Jun 15/2016	
3	Feb 15/2015		53-888-00-01	ZON		R 2	Jun 15/2016	
53-870-00-01	ZON		1	Oct 15/2014		3	Feb 15/2015	
1	Feb 15/2015		2	Feb 15/2015		53-906-02-01	ZON	
2	Feb 15/2015		3	Feb 15/2015		R 1	Jun 15/2016	
3	Feb 15/2015		53-890-00-01	ZON		R 2	Jun 15/2016	
53-872-00-01	ZON		1	Feb 15/2015		3	Feb 15/2015	
1	Feb 15/2015		2	Feb 15/2015				
2	Feb 15/2015		3	Feb 15/2015				
3	Feb 15/2015		53-892-00-01	ZON				
53-874-00-01	ZON		1	Feb 15/2015				
1	Feb 15/2015		2	Feb 15/2015				
2	Feb 15/2015		3	Feb 15/2015				
3	Feb 15/2015		53-894-00-01	ZON				
53-876-00-01	ZON		R 1	Jun 15/2016				
1	Feb 15/2015		2	Feb 15/2015				
2	Feb 15/2015		3	Feb 15/2015				
3	Feb 15/2015		53-896-00-01	ZON				
53-878-00-01	ZON		1	Feb 15/2015				
1	Oct 15/2014		2	Feb 15/2015				
2	Feb 15/2015		3	Feb 15/2015				
3	Feb 15/2015		4	Feb 15/2015				
53-880-00-01	ZON		5	Feb 15/2015				
1	Oct 15/2014		53-898-00-01	ZON				
2	Feb 15/2015		1	Feb 15/2015				
3	Feb 15/2015		2	Feb 15/2015				
53-882-00-01	ZON		3	Feb 15/2015				
1	Feb 15/2015		53-900-00-01	ZON				
2	Feb 15/2015		1	Oct 15/2014				
3	Feb 15/2015		2	Feb 15/2015				

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TASK CARDS

AIRLINE CARD NO		TITLE FUSELAGE LOWER LOBE, FORWARD ACCESS DOOR CUTOUT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-010-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 36 MO 4000 FC	REPEAT 36 MO 4000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 112A NOTE			ZONE 112

Inspect door cutout at forward access door.

INTERVAL NOTE: Whichever comes first.**ACCESS NOTE:** Open forward access door.**A. References**

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE LOWER LOBE, FORWARD ACCESS DOOR CUTOUT
		D633A109-AKS 53-010-00-01

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-010-00-01
TASK 53-05-03-210-801				MECH INSP

1. EXTERNAL - GENERAL VISUAL: FUSELAGE LOWER LOBE, FORWARD ACCESS DOOR CUTOUT
(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-001

(1) Open this access panel:

Number Name/Location
112A Forward Access Door

NOTE: Open forward access door.

SUBTASK 53-05-03-210-001

(2) Do a General Visual inspection of the door cutout at forward access door.

SUBTASK 53-05-03-910-001

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-801.

SUBTASK 53-05-03-410-001

(4) Close this access panel:

Number Name/Location
112A Forward Access Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE LOWER LOBE, FORWARD ACCESS DOOR CUTOUT D633A109-AKS 53-010-00-01	Page 2 of 5 Feb 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-010-00-01
				MECH INSP
TASK 51-05-01-210-801				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-022				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-023				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-024				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-025				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-026				
(5) CPCP Basic Task Item 5 is not applicable.				
SUBTASK 51-05-01-210-027				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				
1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.				

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE LOWER LOBE, FORWARD ACCESS DOOR CUTOUT	D633A109-AKS 53-010-00-01	Page 3 of 5 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-010-00-01		
					MECH	INSP
2) Not applicable	3) List of areas / items where water displacing/anti-corrosion compounds should not be applied: Water displacing / anti-corrosion compounds should not be applied in the following areas: <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.					

SUBTASK 51-05-01-210-028

(7) CPCP Basic Task Item 7 is not applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE LOWER LOBE, FORWARD ACCESS DOOR CUTOUT D633A109-AKS 53-010-00-01	Page 4 of 5 Oct 15/2014
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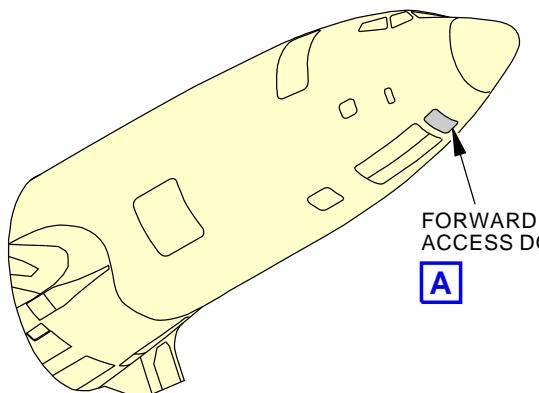
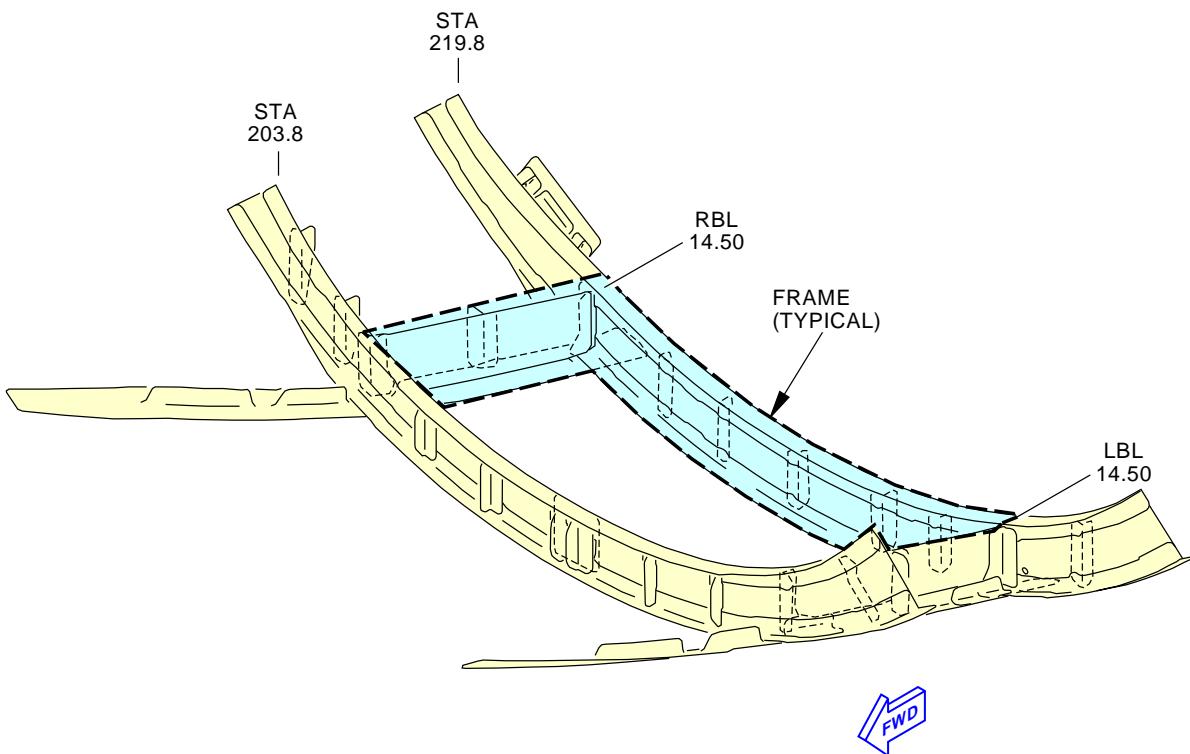
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-010-00-01FORWARD
ACCESS DOOR**A**

FORWARD ACCESS DOOR FRAME

A

H45940 S0006584592_V2

**External - Forward Access Door Cutout Frame
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**FUSELAGE LOWER LOBE, FORWARD ACCESS DOOR
CUTOUT****D633A109-AKS
53-010-00-01****Page 5 of 5
Oct 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FUSELAGE LOWER LOBE, EE COMPARTMENT DOOR CUTOUT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-010-00-02
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 36 MO 4000 FC	REPEAT 36 MO 4000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 117A NOTE			ZONE 112

Inspect door cutout at EE Compartment door.

INTERVAL NOTE: Whichever comes first.**ACCESS NOTE:** Open EE compartment door.**A. References**

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE LOWER LOBE, EE COMPARTMENT DOOR CUTOUT
		D633A109-AKS 53-010-00-02

Page 1 of 5
Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-010-00-02
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TASK 53-05-03-210-802

MECH

INSP

1. EXTERNAL - GENERAL VISUAL: FUSELAGE LOWER LOBE, EE COMPARTMENT DOOR**CUTOUT**

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-002

- (1) Open this access panel:

Number Name/Location

117A Electronic Equipment Access Door

NOTE: Open EE compartment door.

SUBTASK 53-05-03-210-002

- (2) Do a General Visual inspection of the door cutout at EE Compartment door.

SUBTASK 53-05-03-910-002

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-801.

SUBTASK 53-05-03-410-002

- (4) Close this access panel:

Number Name/Location

117A Electronic Equipment Access Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE LOWER LOBE, EE COMPARTMENT DOOR CUTOUT
		D633A109-AKS 53-010-00-02

**Page 2 of 5
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-010-00-02
				MECH INSP
TASK 51-05-01-210-801				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-022				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-023				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-024				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-025				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-026				
(5) CPCP Basic Task Item 5 is not applicable.				
SUBTASK 51-05-01-210-027				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				
1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.				

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE LOWER LOBE, EE COMPARTMENT DOOR CUTOUT D633A109-AKS 53-010-00-02	Page 3 of 5 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-010-00-02
	2) Not applicable 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied: Water displacing / anti-corrosion compounds should not be applied in the following areas: <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.		MECH	INSP

SUBTASK 51-05-01-210-028

(7) CPCP Basic Task Item 7 is not applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE LOWER LOBE, EE COMPARTMENT DOOR CUTOUT D633A109-AKS 53-010-00-02	Page 4 of 5 Oct 15/2014
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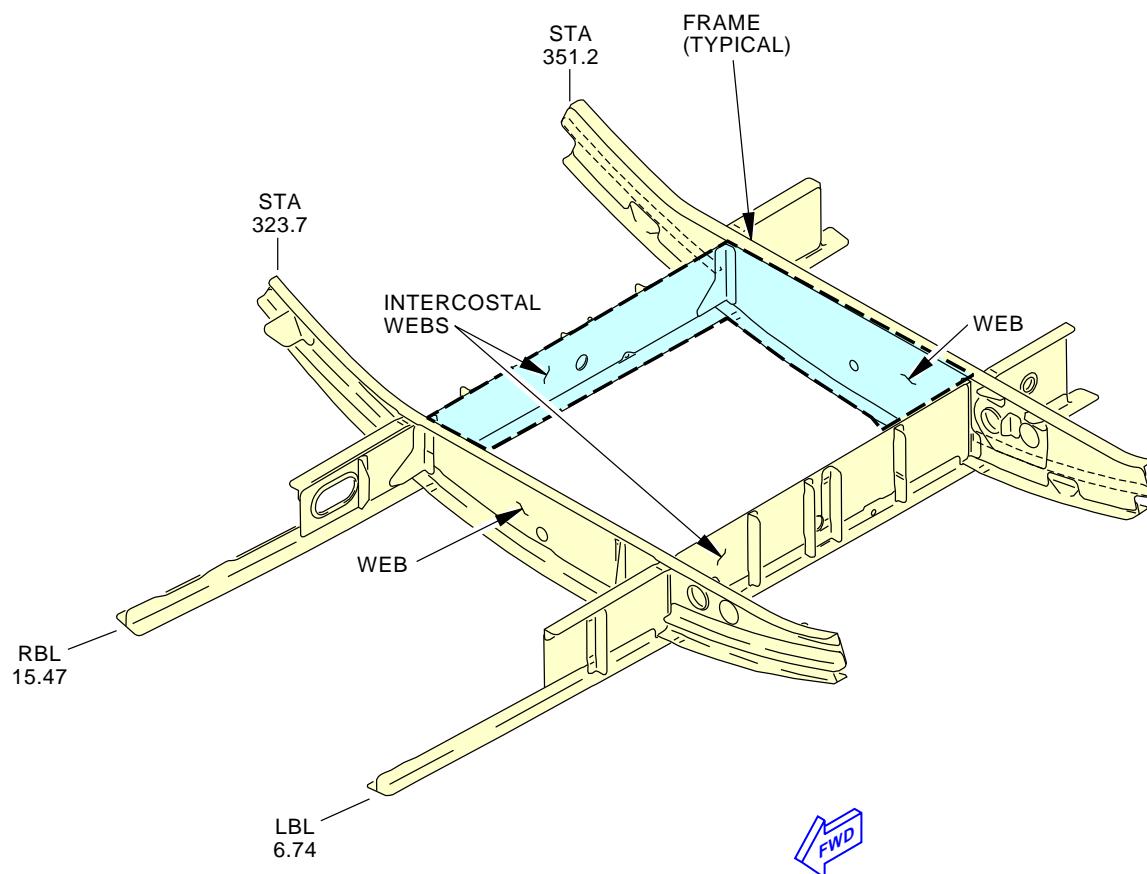
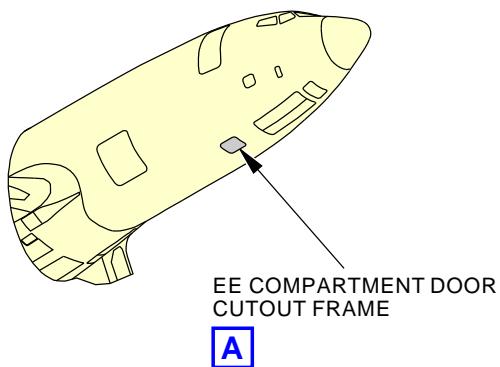
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-010-00-02**A**

H45755 S0006584595_V2

**External - EE Compartment Door Cutout Frame
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**FUSELAGE LOWER LOBE, EE COMPARTMENT DOOR
CUTOUT****D633A109-AKS
53-010-00-02****Page 5 of 5
Oct 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE NOSE WHEEL WELL			BOEING CARD NO. 53-020-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA NOSE W/W	VERSION 1.1	THRESHOLD 12 YR	REPEAT 6 YR	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 115 116

Inspect nose landing gear wheel well, including canted bulkhead (Sta 224.8 to 227.8), Sta 294.5 bulkhead, side and top panels, trunnion support fitting, actuator support fitting, and drag brace fitting.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	NOSE WHEEL WELL D633A109-AKS 53-020-00-01	Page 1 of 5 Jun 15/2015
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-020-00-01	MECH	INSP
TASK 53-05-03-210-804						

1. EXTERNAL - GENERAL VISUAL: NOSE WHEEL WELL

(Figure 1)

A. Inspection

SUBTASK 53-05-03-210-004

(1) Do a General Visual inspection of the nose landing gear wheel well, including canted bulkhead (Sta 224.8 to 227.8), Sta 294.5 bulkhead, side and top panels, trunnion support fitting, actuator support fitting, and drag brace fitting.

SUBTASK 53-05-03-910-004

(2) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	NOSE WHEEL WELL	
		D633A109-AKS 53-020-00-01	Page 2 of 5 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-020-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	NOSE WHEEL WELL D633A109-AKS 53-020-00-01	Page 3 of 5 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-020-00-01			
				<table border="1"><tr><td>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</td><td>MECH</td><td>INSP</td></tr></table>	1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP
1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP					

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

- 2) Not applicable

- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

Water displacing / anti-corrosion compounds should not be applied in the following areas:

- Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
- Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
- Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
- Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
- Areas with electrical arc potential.
- Interior materials, including cargo liners (change of flammability properties).
- Fiber-glass ducts where temperature exceeds 220 degrees F.
- Selected areas noted in baseline program.

SUBTASK 51-05-01-210-063

- (7) CPCP Basic Task Item 7 is not applicable.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	NOSE WHEEL WELL	
		D633A109-AKS 53-020-00-01	Page 4 of 5 Oct 15/2014

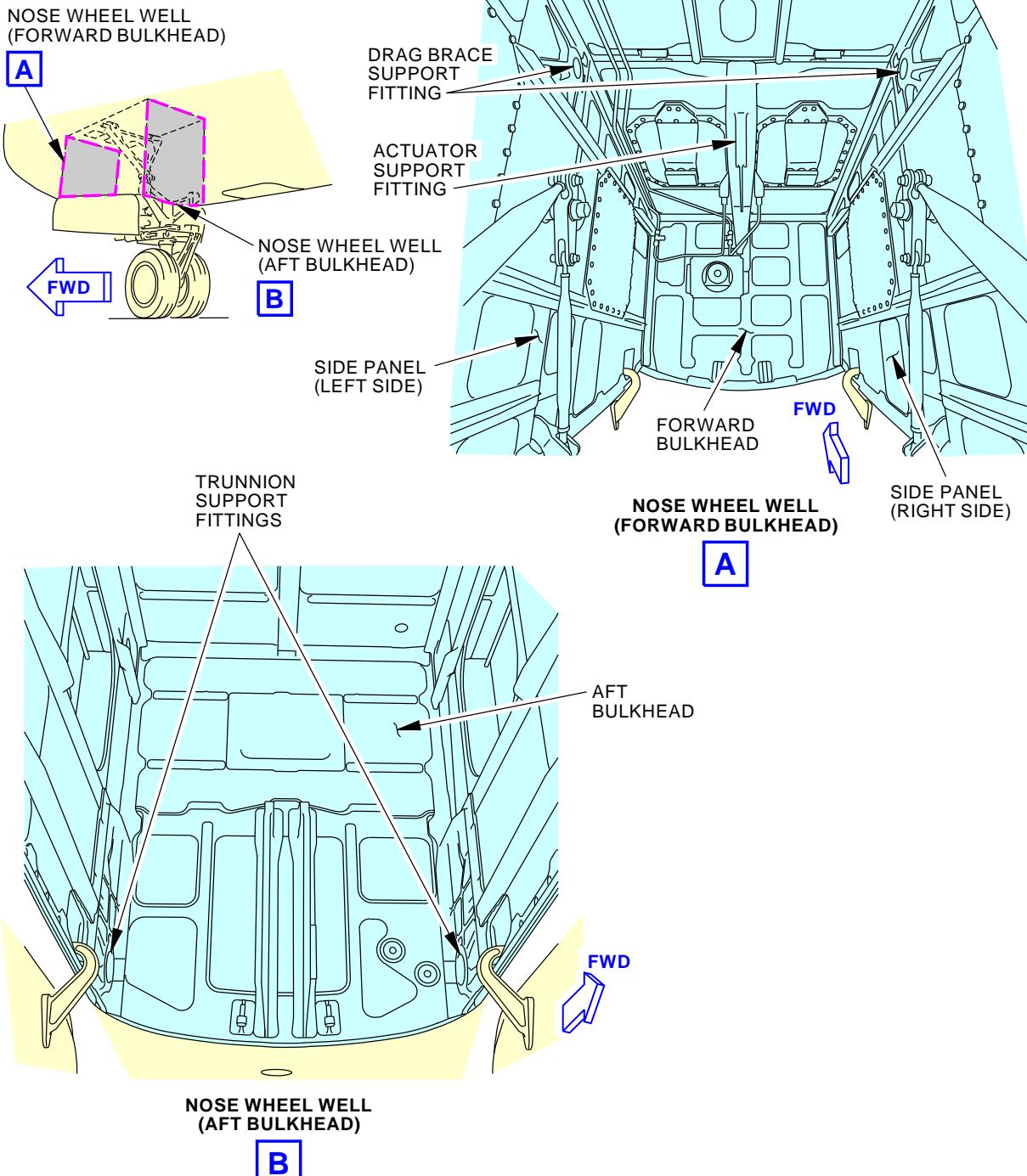
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-020-00-01

**Nose Landing Gear Wheel Well
Figure 1**

D63215 S0000162566_V2

EFFECTIVITY
AKS ALLSOURCE
MRB**NOSE WHEEL WELL****D633A109-AKS
53-020-00-01****Page 5 of 5
Oct 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS			BOEING CARD NO. 53-030-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1 1.2 NOTE	THRESHOLD 36 MO 6600 FC	REPEAT 36 MO 6600 FC	APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	ACCESS 821			ZONE 122

Inspect the forward cargo door surround structure, fittings and stops.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS
		D633A109-AKS 53-030-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-030-00-01
TASK 53-05-03-211-801				MECH INSP

1. EXTERNAL - DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-055

(1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
821	Forward Cargo Door

SUBTASK 53-05-03-211-001

(2) Do a Detailed inspection of the forward cargo door surround structure, fittings and stops.

SUBTASK 53-05-03-910-005

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.

SUBTASK 53-05-03-410-055

(4) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
821	Forward Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS
		D633A109-AKS 53-030-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-030-00-01
				MECH INSP
TASK 51-05-01-210-809				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-078				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-079				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-080				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-081				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-082				
(5) CPCP Basic Task Item 5 is not applicable.				
SUBTASK 51-05-01-210-100				
(6) Do the CPCP Basic Task Item 6 (Not Applicable)				
SUBTASK 51-05-01-210-084				
(7) CPCP Basic Task Item 7 is not applicable.				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS	
		D633A109-AKS 53-030-00-01	Page 3 of 4 Oct 15/2015

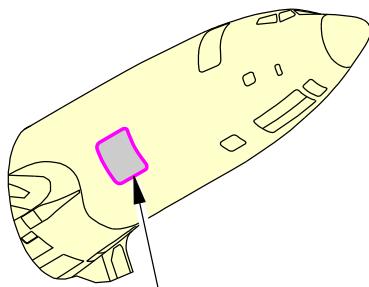
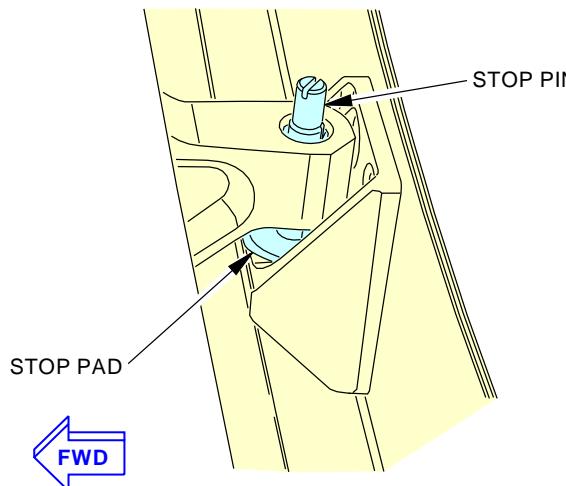
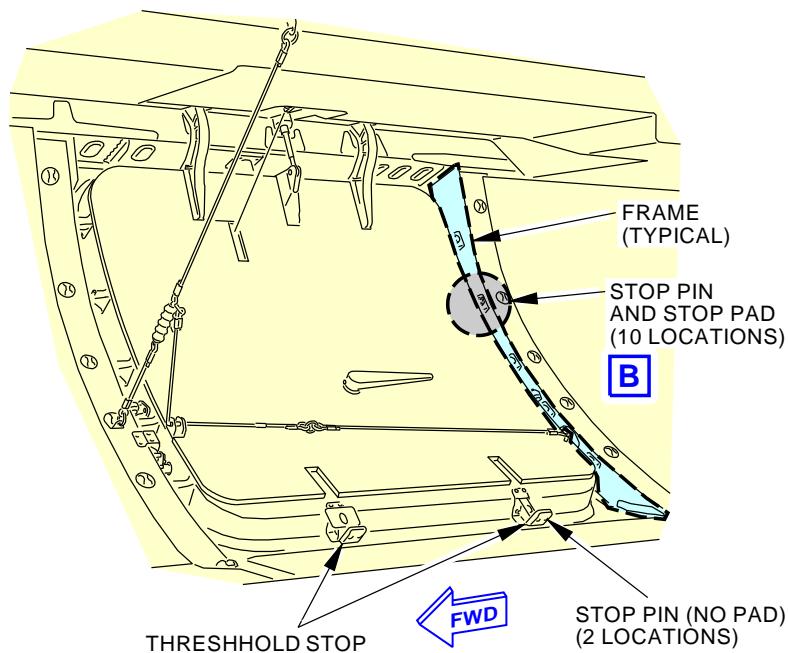
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-030-00-01**FORWARD CARGO COMPARTMENT DOOR****A****STOP PIN AND STOP PAD (EXAMPLE)****B**

H45965 S0006584602_V4

**External - Forward Cargo Door Surround Structure Fitting and Stops
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS
		D633A109-AKS 53-030-00-01

Page 4 of 4
Feb 15/2016

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS			BOEING CARD NO. 53-030-00-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA AFT CARGO DOOR	VERSION 1.1 1.2	THRESHOLD 36 MO 6600 FC	REPEAT 36 MO 6600 FC	APPLICABILITY	
STATION	SKILL AIRPL	NOTE			AIRPLANE ALL	ENGINE ALL
		ACCESS 822			ZONE 142	

Inspect the aft cargo door surround structure, fittings and stops.

INTERVAL NOTE: Whichever comes first.**A. References**

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS	D633A109-AKS 53-030-00-02	Page 1 of 4 Feb 15/2015
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-030-00-02
------	-------------	---------	------------------	--

TASK 53-05-03-211-802

1. EXTERNAL - DETAILED: AFT CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-056

- (1) Open this access panel:

Number **Name/Location**

822 Aft Cargo Door

SUBTASK 53-05-03-211-002

- (2) Do a Detailed inspection of the aft cargo door surround structure, fittings and stops.

SUBTASK 53-05-03-910-006

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.

SUBTASK 53-05-03-410-056

- (4) Close this access panel:

Number Name/Location

822 Aft Cargo Door

— END OF TASK —

EFFECTIVITY
AKS ALL

SOURCE
MRB

AFT CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS

D633A109-AKS
53-030-00-02

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Feb 15/2015

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-030-00-02
				MECH INSP
TASK 51-05-01-210-809				
2. 737-6789 Basic Task Description				
<p>A. CPCP Basic Task</p> <p>SUBTASK 51-05-01-210-078</p> <p>(1) CPCP Basic Task Item 1 is not applicable.</p> <p>SUBTASK 51-05-01-210-079</p> <p>(2) Do the CPCP Basic Task Item 2 as follows:</p> <p>(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.</p> <p>SUBTASK 51-05-01-210-080</p> <p>(3) Do the CPCP Basic Task Item 3 as follows:</p> <p>(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.</p> <p>SUBTASK 51-05-01-210-081</p> <p>(4) Do the CPCP Basic Task item 4 as follows:</p> <p>(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.</p> <p>SUBTASK 51-05-01-210-082</p> <p>(5) CPCP Basic Task Item 5 is not applicable.</p> <p>SUBTASK 51-05-01-210-100</p> <p>(6) Do the CPCP Basic Task Item 6 (Not Applicable)</p> <p>SUBTASK 51-05-01-210-084</p> <p>(7) CPCP Basic Task Item 7 is not applicable.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO DOOR SURROUND STRUCTURE, FITTINGS AND STOPS		
		D633A109-AKS 53-030-00-02		
				Page 3 of 4 Oct 15/2015

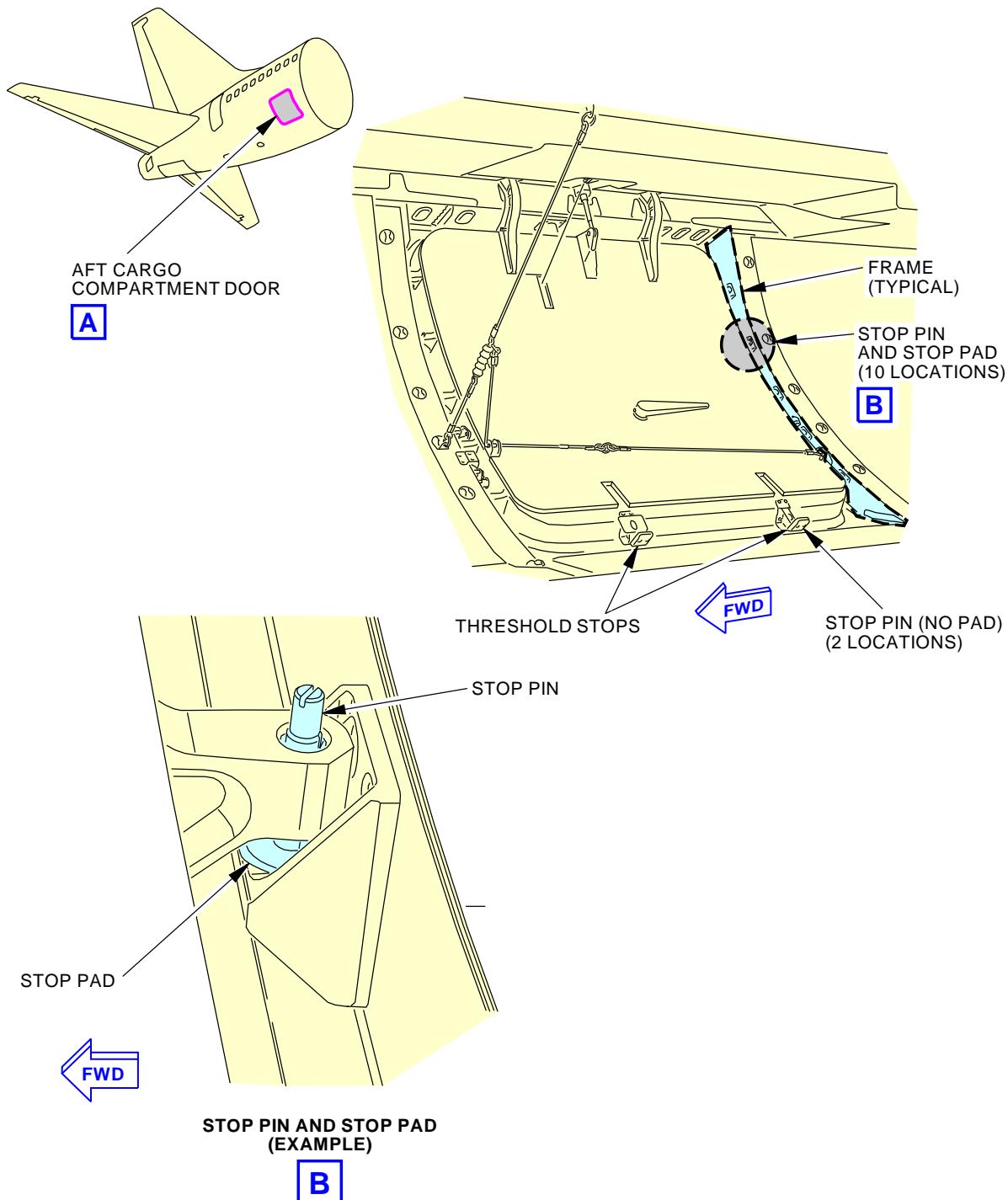
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-030-00-02

H45969 S0006584605_V4

External - Aft Cargo Door Surround Structure Fittings and Stops
Figure 1EFFECTIVITY
AKS ALLSOURCE
MRB**AFT CARGO DOOR SURROUND STRUCTURE, FITTINGS AND
STOPS****D633A109-AKS**
53-030-00-02**Page 4 of 4**
Feb 15/2016

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE MAIN LANDING GEAR WHEEL WELL			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-050-00-01
TAIL NUMBER	WORK AREA KEEL BEAM	VERSION 1.1	THRESHOLD 9 YR	REPEAT 3 YR	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS			ZONE 133 134 139 193

Inspect main landing gear wheel well, including:

1. Pressure deck web and stiffeners, including attachment to wing center section rear spar at Sta 663; 2. Bulkhead at STA 663; 3. Bulkhead and pressure web at STA 727; 4. Keel beam chords, webs, stiffeners and splice, keel beam/rear spar attachment angles; 5. Stringer 18A web, chord and links; 6. Side strut support frame at STA 706; 7. Main landing gear support frame at STA 695 and 716; 8. Wheel well frame at STA 685; 9. Flap track support fittings.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL	
		D633A109-AKS 53-050-00-01	Page 1 of 10 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-050-00-01
				MECH INSP
TASK 53-05-03-210-805				
1. EXTERNAL - GENERAL VISUAL: MAIN LANDING GEAR WHEEL WELL (Figure 1,Figure 2,Figure 3)				
A. Inspection				
SUBTASK 53-05-03-210-005				
(1) Do a General Visual inspection of the main landing gear wheel well, including: <ol style="list-style-type: none">1. Pressure deck web and stiffeners, including attachment to wing center section rear spar at Sta 663.2. Bulkhead at STA 663.3. Bulkhead and pressure web at STA 727.4. Keel beam chords, webs, stiffeners and splice, keel beam/rear spar attachment angles.5. Stringer 18A web, chord and links.6. Side strut support frame at STA 706.7. Main landing gear support frame at STA 695 and 716.8. Wheel well frame at STA 685.9. Flap track support fittings.				
SUBTASK 53-05-03-910-007				
(2) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.				
———— END OF TASK ———				

EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL	
		D633A109-AKS 53-050-00-01	Page 2 of 10 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-050-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-050-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-050-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-063

(7) CPCP Basic Task Item 7 is not applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-050-00-01

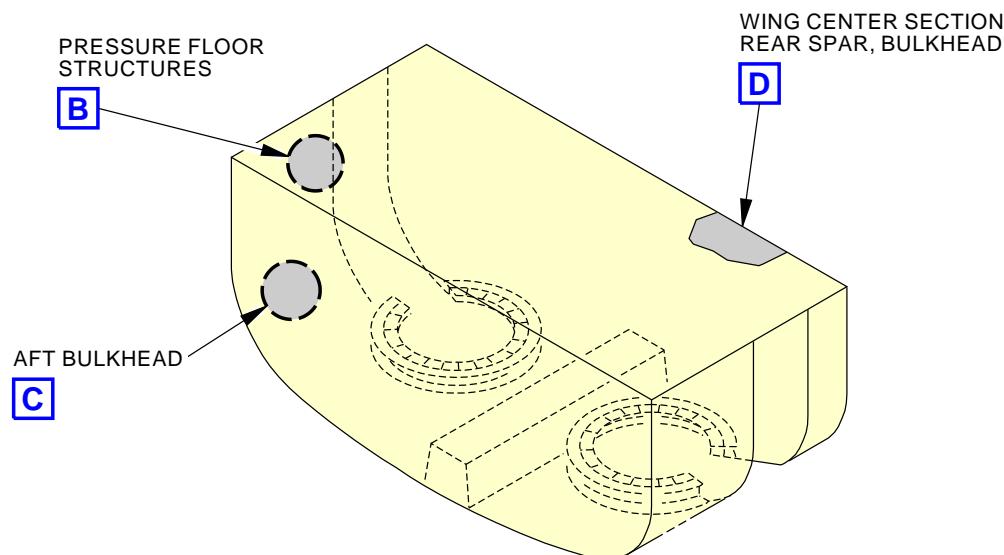
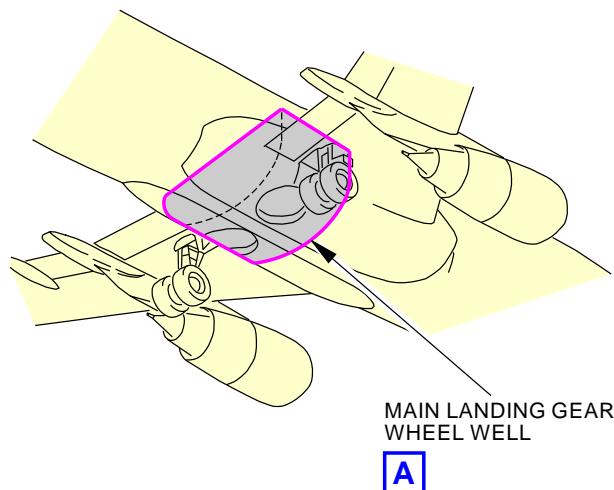
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-050-00-01

EXTERNAL-GENERAL VISUAL: MAIN LANDING GEAR WHEEL WELL
Figure 1 (Sheet 1 of 4)

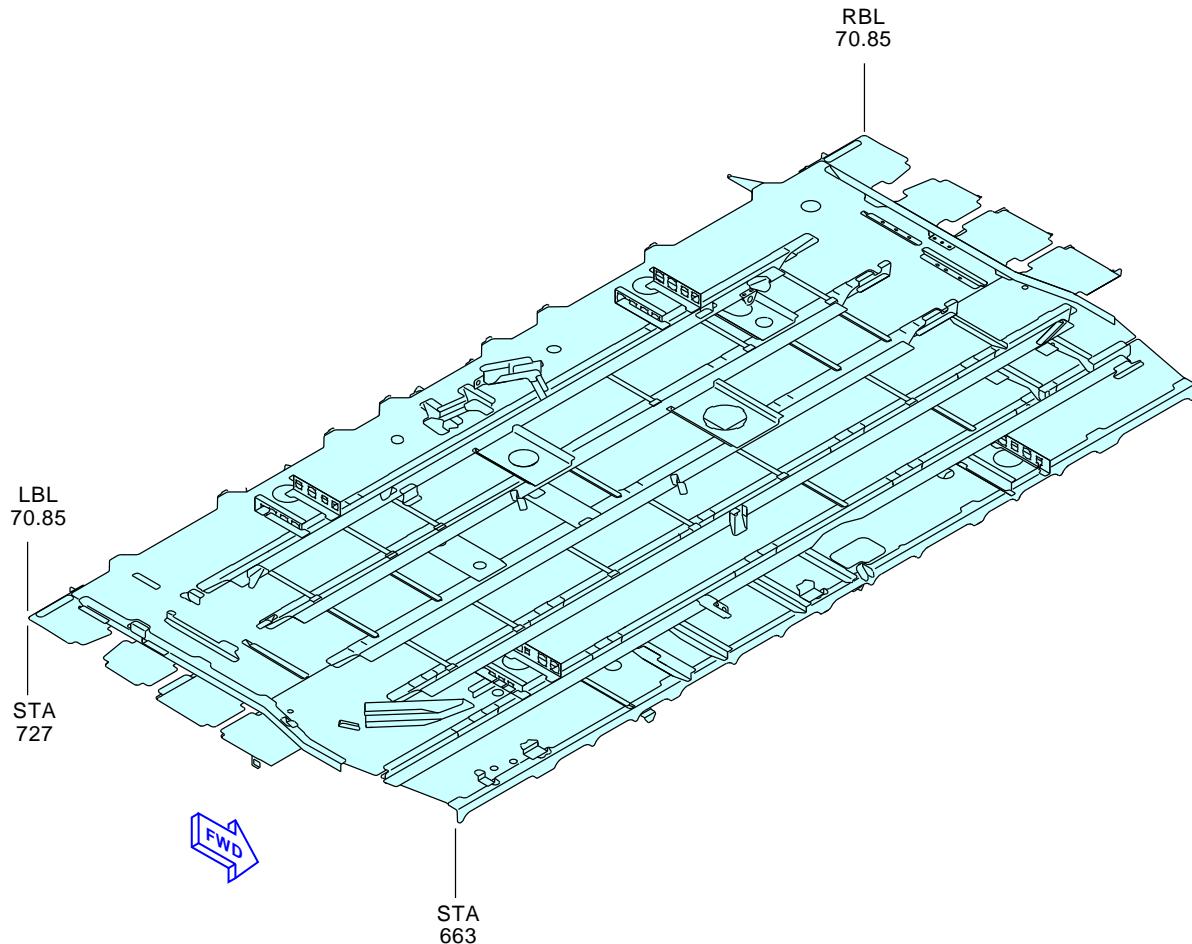
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EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-050-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-050-00-01

**PRESSURE FLOOR STRUCTURES
(BOTTOM VIEW LOOKING UP)**

D79097 S0000164679_V2

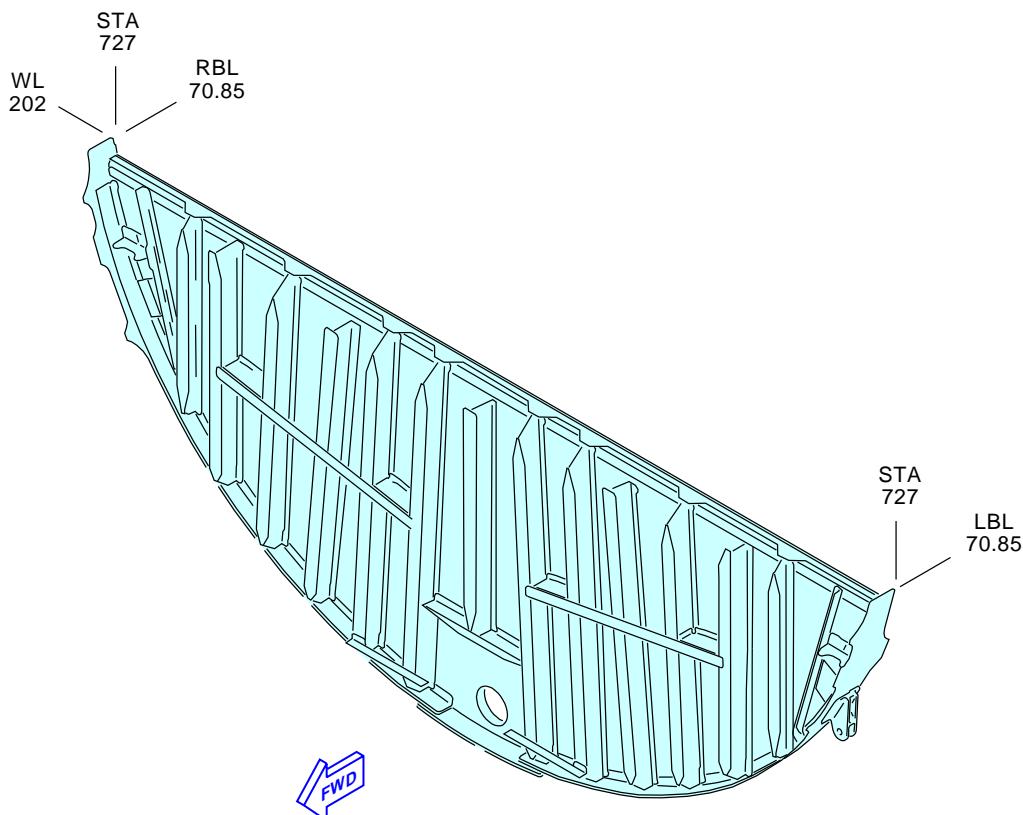
EXTERNAL-GENERAL VISUAL: MAIN LANDING GEAR WHEEL WELL
Figure 1 (Sheet 2 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-050-00-01

**Page 6 of 10
Oct 15/2015**

AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-050-00-01

**AFT BULKHEAD
(STA 727)****C**

D79170 S0000164681_V3

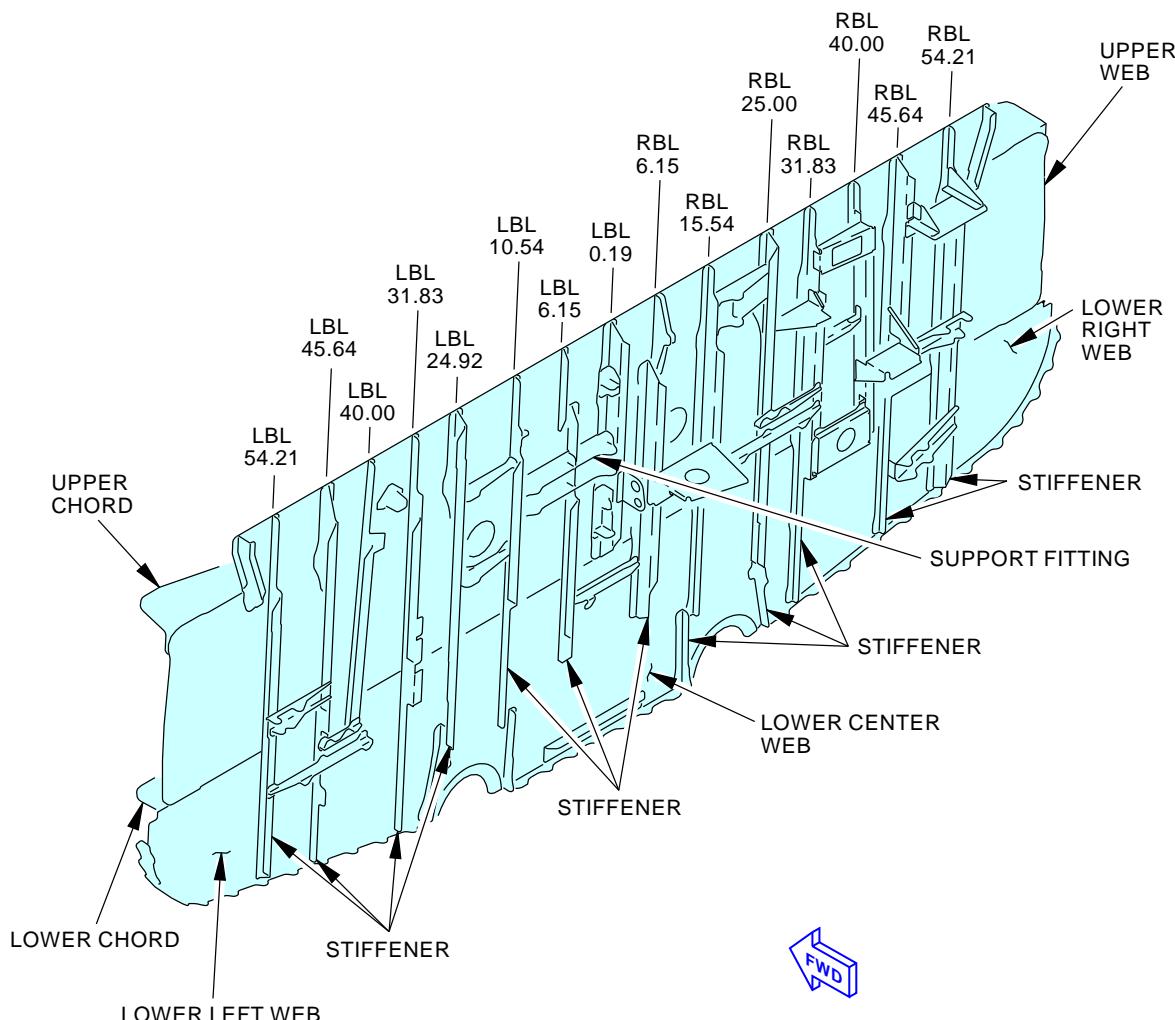
EXTERNAL-GENERAL VISUAL: MAIN LANDING GEAR WHEEL WELL
Figure 1 (Sheet 3 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-050-00-01

**Page 7 of 10
Feb 15/2016**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-050-00-01
------	-------------	---------	------------------	--



WING CENTER SECTION
REAR SPAR, BULKHEAD
(STA 663)

D

D79184 S0000164682_V2

EXTERNAL-GENERAL VISUAL: MAIN LANDING GEAR WHEEL WELL
Figure 1 (Sheet 4 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-050-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-050-00-01
				D79276 S0000164948_V2
EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL	D633A109-AKS 53-050-00-01	Page 9 of 10 Oct 15/2015

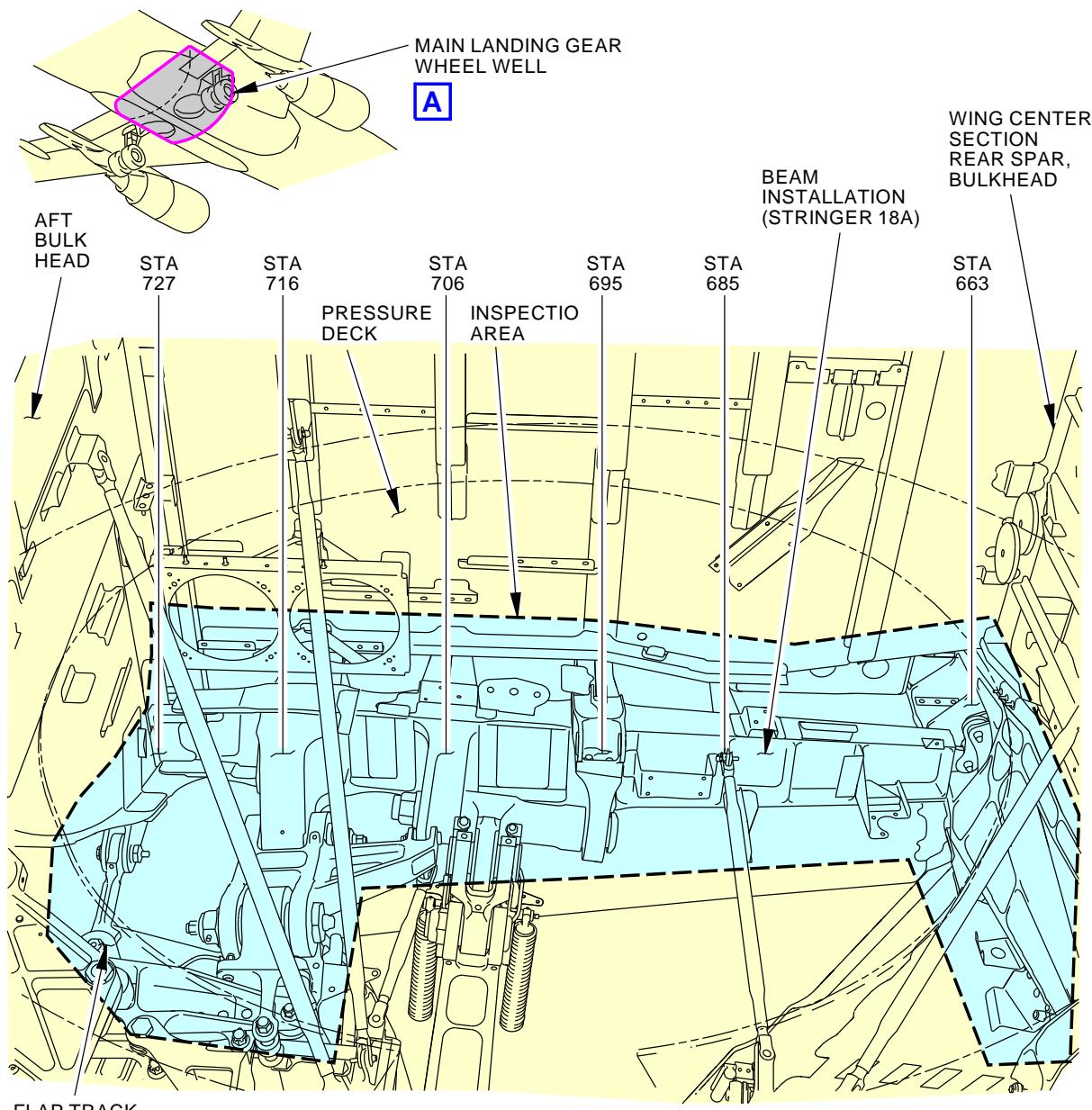
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-050-00-01**NOTE:**

LEFT SIDE SHOWN, RIGHT
SIDE OPPOSITE, LESS
FIREBOTTLE BRACKET

D80805 S0000164953_V2

EXTERNAL-GENERAL VISUAL: MAIN LANDING GEAR WHEEL WELL**Figure 3**

EFFECTIVITY AKS ALL	SOURCE MRB	MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-050-00-01

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD PRESSURE BULKHEAD			BOEING CARD NO. 53-060-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA RADOME	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC	AIRPLANE ALL ENGINE ALL
		ACCESS 111			ZONE 111
		NOTE			

Inspect the forward side of STA 178 bulkhead.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Open nose radome.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PRESSURE BULKHEAD	
		D633A109-AKS 53-060-00-01	Page 1 of 5 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-060-00-01
TASK 53-05-03-210-806				MECH INSP

1. INTERNAL - GENERAL VISUAL: FORWARD PRESSURE BULKHEAD

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-004

- (1) Open this access panel:

Number Name/Location

111 Radome

NOTE: Open nose radome.

SUBTASK 53-05-03-210-006

- (2) Do a General Visual inspection of the forward side of STA 178 bulkhead.

SUBTASK 53-05-03-910-008

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.

SUBTASK 53-05-03-410-004

- (4) Close this access panel:

Number Name/Location

111 Radome

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PRESSURE BULKHEAD
		D633A109-AKS 53-060-00-01

Page 2 of 5
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-060-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PRESSURE BULKHEAD
		D633A109-AKS 53-060-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-060-00-01			
				<table border="1"><tr><td>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</td><td>MECH</td><td>INSP</td></tr></table>	1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP
1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP					

SUBTASK 51-05-01-210-063

- (7) CPCP Basic Task Item 7 is not applicable.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PRESSURE BULKHEAD
		D633A109-AKS 53-060-00-01

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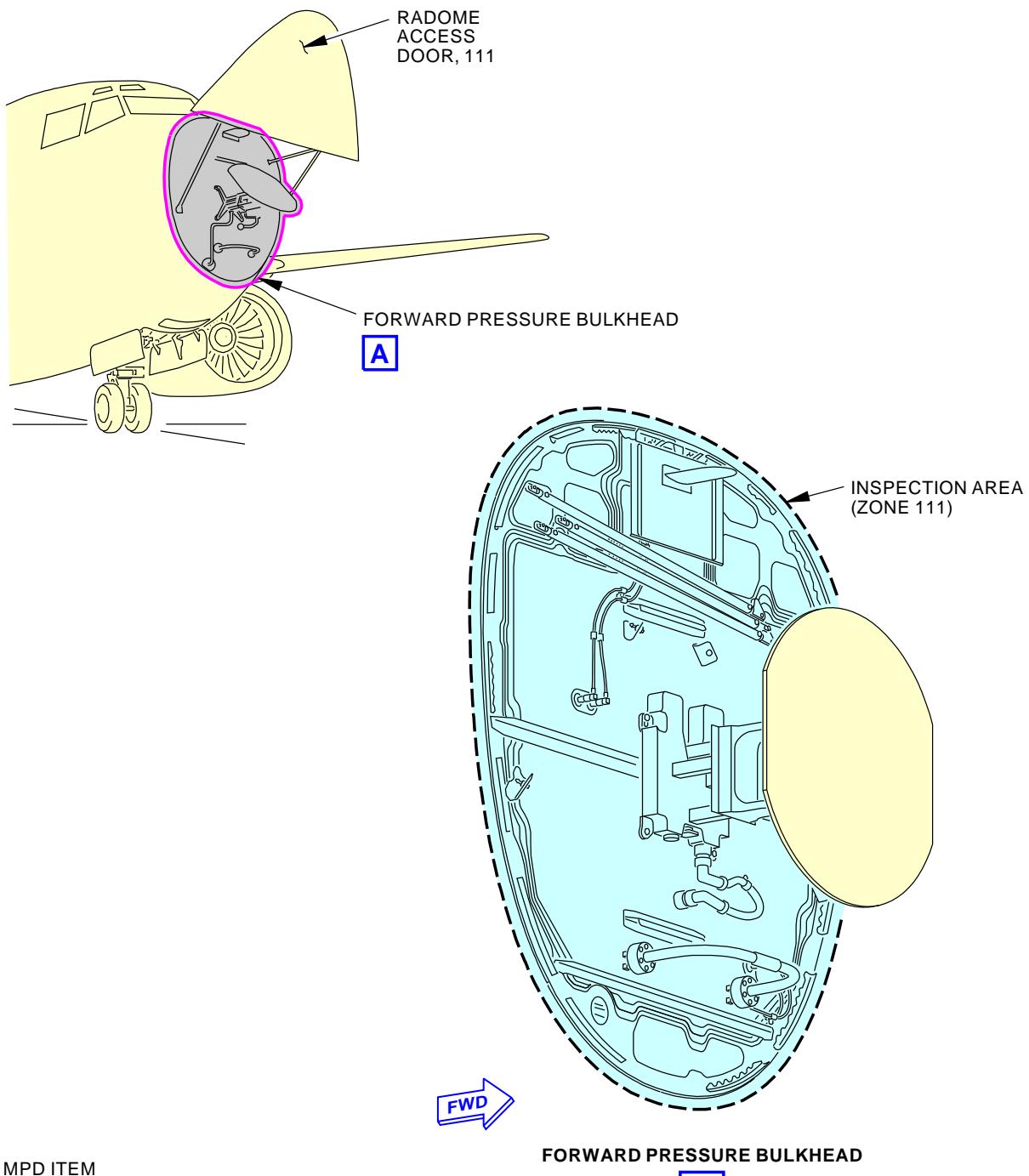
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-060-00-01MPD ITEM
53-060-00

2056123 S0000420182_V2

**Forward Pressure Bulkhead, Sta. 178
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PRESSURE BULKHEAD
		D633A109-AKS 53-060-00-01

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Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA FORWARD OF NOSE WHEEL WELL			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-070-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 112A S1122 NOTE			ZONE 112

Inspect fuselage lower lobe from STA 178 bulkhead to canted bulkhead (STA 224.8 to 227.8), including bulkheads, skin panels (skins, frames, stringers), longitudinal lap splices, forward access door cutout, and nose wheel well cutout.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove weather radar RT mount. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA FORWARD OF NOSE WHEEL WELL
		D633A109-AKS 53-070-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-070-00-01	MECH	INSP
TASK 53-05-03-210-807						
1.	INTERNAL - GENERAL VISUAL: AREA FORWARD OF NOSE WHEEL WELL					
	(Figure 1)					
	A. Inspection					
	SUBTASK 53-05-03-010-005					
	(1) Open these access panels:					
	Number Name/Location					
	112A Forward Access Door					
	S1122 Area Forward Of Nose Wheel Well Inspection					
	NOTE: Remove weather radar RT mount. Remove/displace insulation blankets as required.					
	SUBTASK 53-05-03-210-007					
	(2) Do a General Visual inspection of the fuselage lower lobe from STA 178 bulkhead to canted bulkhead (STA 224.8 to 227.8), including bulkheads, skin panels (skins, frames, stringers), longitudinal lap splices, forward access door cutout, and nose wheel well cutout.					
	SUBTASK 53-05-03-910-009					
	(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.					
	SUBTASK 53-05-03-410-005					
	(4) Close these access panels:					
	Number Name/Location					
	112A Forward Access Door					
	S1122 Area Forward Of Nose Wheel Well Inspection					
	— END OF TASK —					

EFFECTIVITY AKS ALL	SOURCE MRB	AREA FORWARD OF NOSE WHEEL WELL	
		D633A109-AKS 53-070-00-01	Page 2 of 5 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-070-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA FORWARD OF NOSE WHEEL WELL	
		D633A109-AKS 53-070-00-01	Page 3 of 5 Oct 15/2015

AKS**737-600/700/800/900****TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-070-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-077

(7) Do the CPCP Basic Task Item 7 as follows:

(a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AREA FORWARD OF NOSE WHEEL WELL
		D633A109-AKS 53-070-00-01

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Oct 15/2014

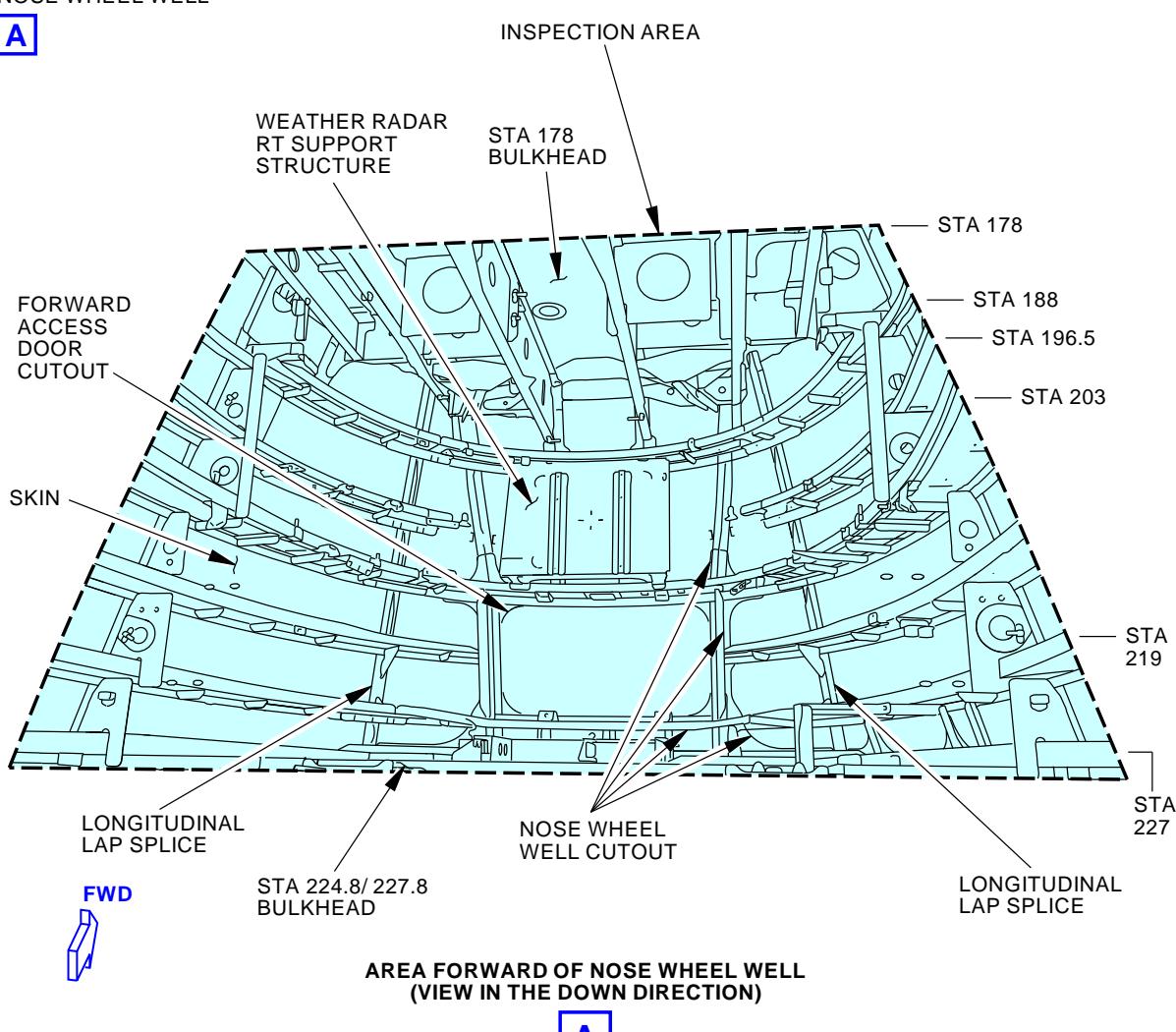
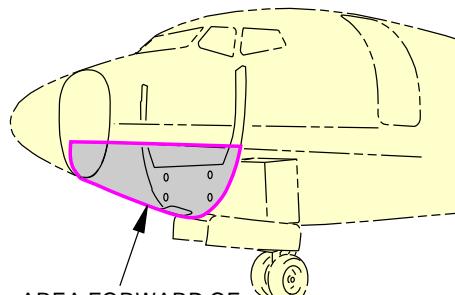
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-070-00-01MPD ITEM
53-070-00

2069529 S0000429375_V2

**Forward of Nose Landing Gear Wheel Well General Visual (Internal)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA FORWARD OF NOSE WHEEL WELL
		D633A109-AKS 53-070-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FLIGHT COMPARTMENT FLOOR STRUCTURE			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-080-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1	THRESHOLD 10 YR	REPEAT 10 YR	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS 112A			ZONE 112
		NOTE			

Inspect flight compartment floor structure from lower lobe.

ACCESS NOTE: Access through forward access door**A. References**

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-080-00-01

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Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-080-00-01
TASK 53-05-03-210-808				MECH INSP

1. INTERNAL - GENERAL VISUAL: FLIGHT COMPARTMENT FLOOR STRUCTURE
(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-006

(1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
112A	Forward Access Door

NOTE: Access through forward access door.

SUBTASK 53-05-03-210-008

(2) Do a General Visual inspection of the flight compartment floor structure from lower lobe.

SUBTASK 53-05-03-910-010

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.

SUBTASK 53-05-03-410-006

(4) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
112A	Forward Access Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-080-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-080-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-080-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-080-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-063

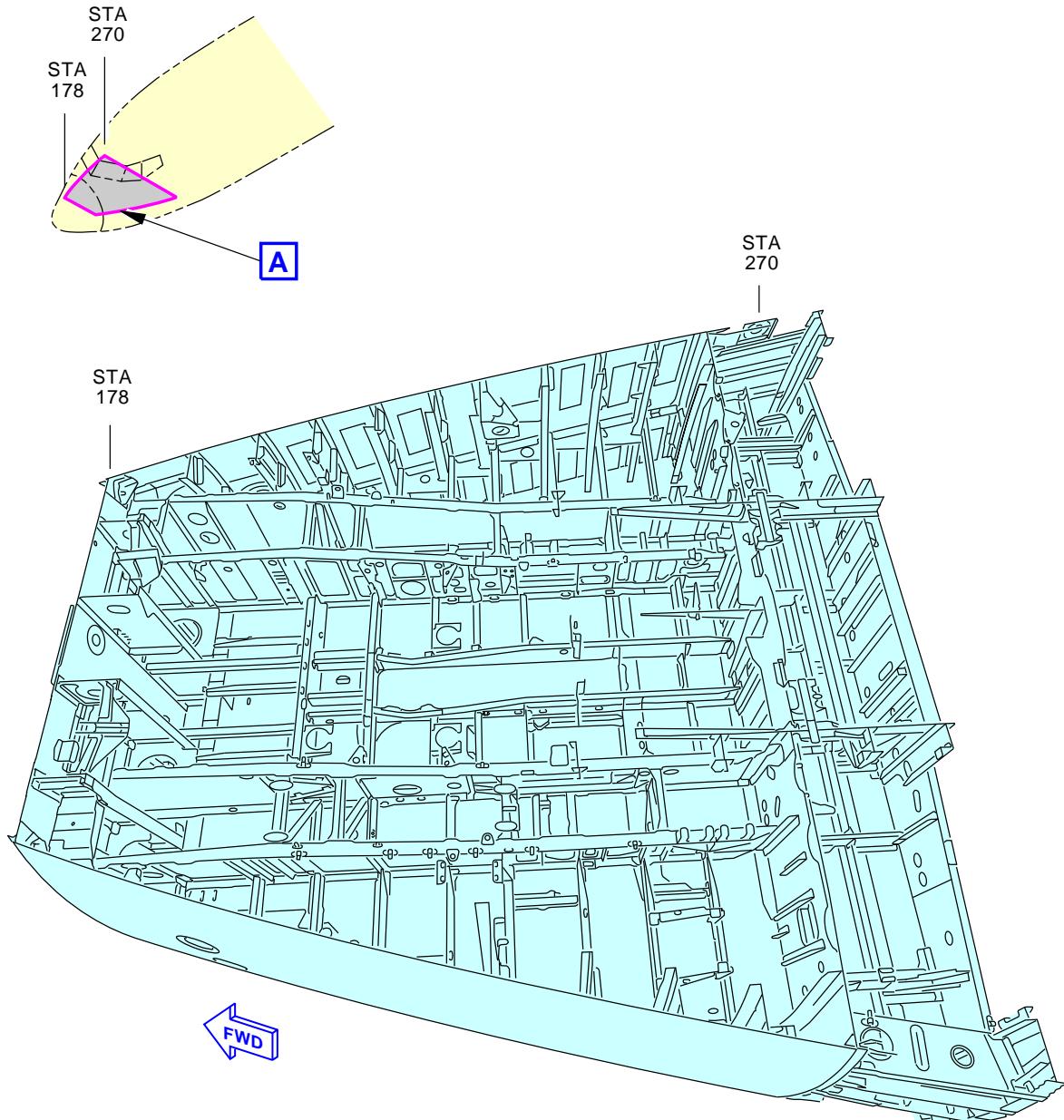
(7) CPCP Basic Task Item 7 is not applicable.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-080-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-080-00-01



(VIEW IN THE UP DIRECTION)

MPD ITEM
53-080-00

D63217 S0000162565_V3

**Flight Deck Floor Structure
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-080-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-090-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 113AW 113BW 114AW 114BW S1101			ZONE 113 114
		NOTE			

Inspect fuselage lower lobe from canted bulkhead (Sta 224.8 to 227.8) to Sta 294, including:

1. Skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices, bulkhead at Sta 259.5;
2. Nose wheel well cutout surround structure, nose wheel well side and top panels;
3. Trunnion support fitting, actuator support fitting and drag brace fitting.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Access through nose wheel well side and top access panels, and through access panel in crew floor. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL
		D633A109-AKS 53-090-00-01

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Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-090-00-01												
				MECH INSP												
TASK 53-05-03-210-809																
1. INTERNAL - GENERAL VISUAL: AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL (Figure 1)																
A. Inspection																
SUBTASK 53-05-03-010-007																
(1) Open these access panels:																
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>113AW</td><td>Forward Nose Wheel Well Panel</td></tr><tr><td>113BW</td><td>Forward Nose Wheel Well Panel</td></tr><tr><td>114AW</td><td>Forward Nose Wheel Well Panel</td></tr><tr><td>114BW</td><td>Forward Nose Wheel Well Panel</td></tr><tr><td>S1101</td><td>Area Above And Outboard of Nose Wheel Well Inspection</td></tr></tbody></table>					Number	Name/Location	113AW	Forward Nose Wheel Well Panel	113BW	Forward Nose Wheel Well Panel	114AW	Forward Nose Wheel Well Panel	114BW	Forward Nose Wheel Well Panel	S1101	Area Above And Outboard of Nose Wheel Well Inspection
Number	Name/Location															
113AW	Forward Nose Wheel Well Panel															
113BW	Forward Nose Wheel Well Panel															
114AW	Forward Nose Wheel Well Panel															
114BW	Forward Nose Wheel Well Panel															
S1101	Area Above And Outboard of Nose Wheel Well Inspection															
NOTE: Access through nose wheel well side and top access panels, and through access panel in crew floor. Remove/displace insulation blankets as required.																
SUBTASK 53-05-03-210-009																
(2) Do a General Visual inspection of the fuselage lower lobe from canted bulkhead (Sta 224.8 to 227.8) to Sta 294, including:																
1. Skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices, bulkhead at Sta 259.5.																
2. Nose wheel well cutout surround structure, nose wheel well side and top panels.																
3. Trunnion support fitting, actuator support fitting and drag brace fitting.																
SUBTASK 53-05-03-910-011																
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.																
SUBTASK 53-05-03-410-007																
(4) Close these access panels:																
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>113AW</td><td>Forward Nose Wheel Well Panel</td></tr><tr><td>113BW</td><td>Forward Nose Wheel Well Panel</td></tr><tr><td>114AW</td><td>Forward Nose Wheel Well Panel</td></tr><tr><td>114BW</td><td>Forward Nose Wheel Well Panel</td></tr><tr><td>S1101</td><td>Area Above And Outboard of Nose Wheel Well Inspection</td></tr></tbody></table>					Number	Name/Location	113AW	Forward Nose Wheel Well Panel	113BW	Forward Nose Wheel Well Panel	114AW	Forward Nose Wheel Well Panel	114BW	Forward Nose Wheel Well Panel	S1101	Area Above And Outboard of Nose Wheel Well Inspection
Number	Name/Location															
113AW	Forward Nose Wheel Well Panel															
113BW	Forward Nose Wheel Well Panel															
114AW	Forward Nose Wheel Well Panel															
114BW	Forward Nose Wheel Well Panel															
S1101	Area Above And Outboard of Nose Wheel Well Inspection															
— END OF TASK —																
EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL														
		D633A109-AKS 53-090-00-01	Page 2 of 6 Feb 15/2015													

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-090-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL	
		D633A109-AKS 53-090-00-01	Page 3 of 6 Oct 15/2015

AKS**737-600/700/800/900****TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-090-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-077

(7) Do the CPCP Basic Task Item 7 as follows:

(a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL
		D633A109-AKS 53-090-00-01

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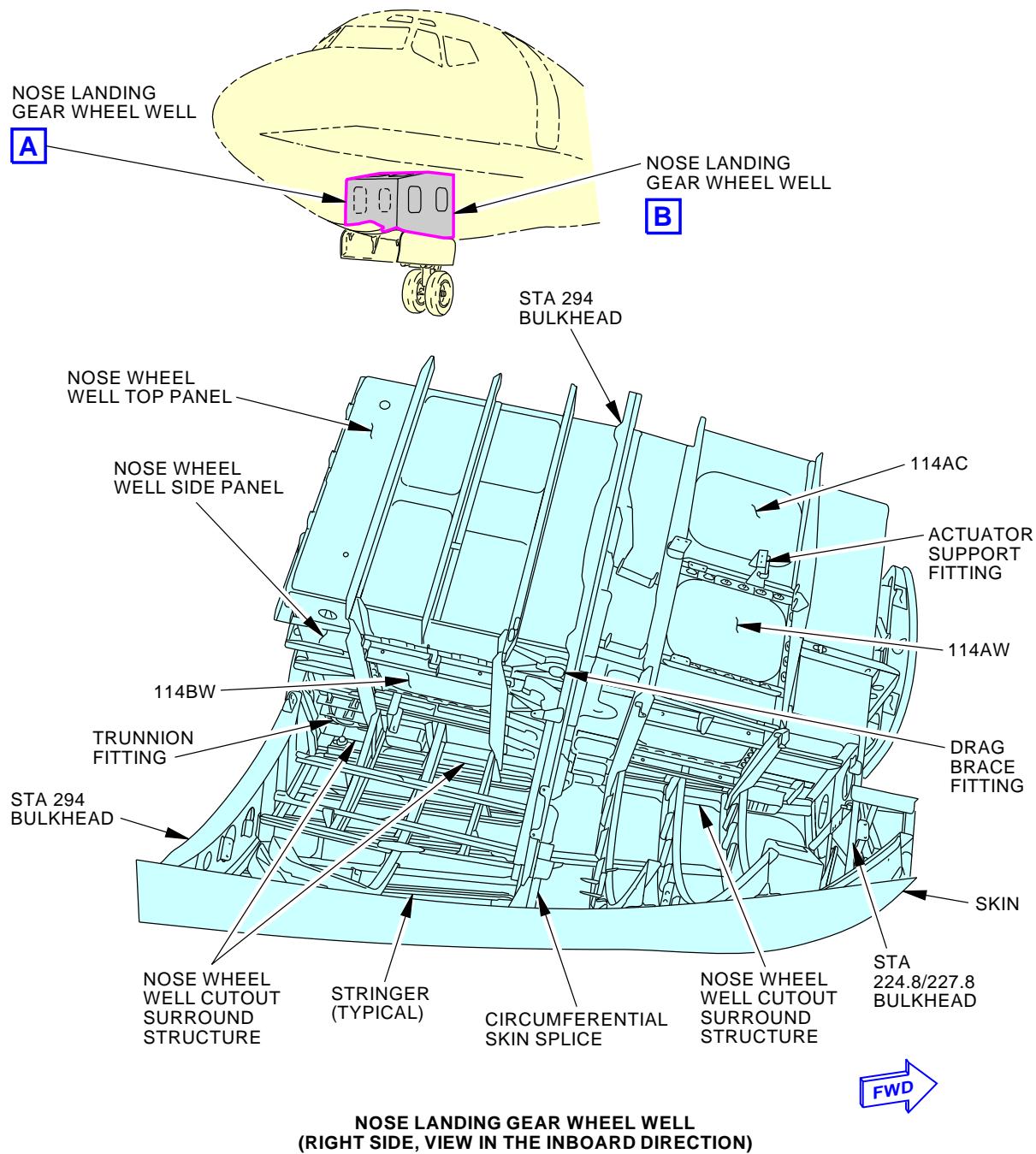
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-090-00-01MPD ITEM
53-090-00

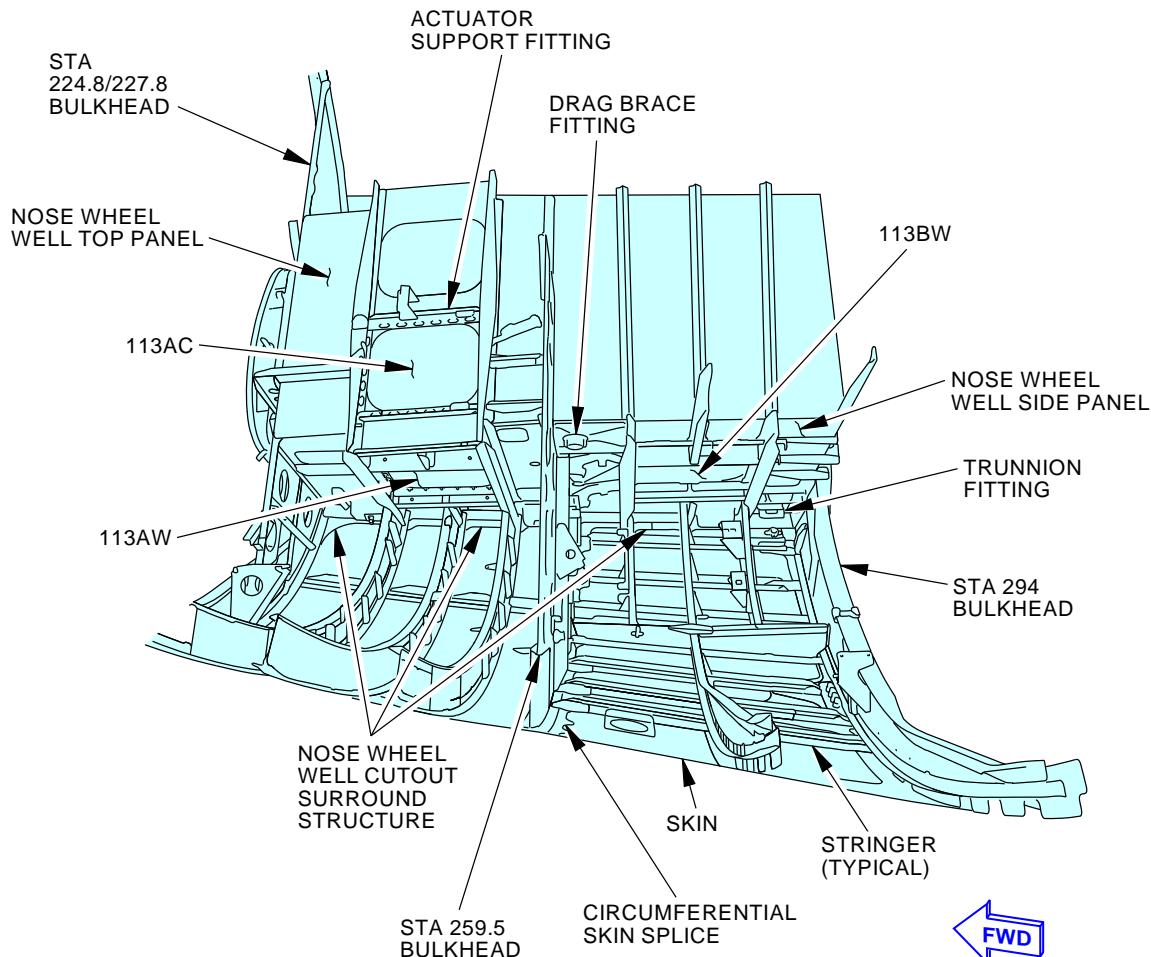
2069612 S0000429415_V2

**Above and Outboard of the Nose Landing Gear Wheel Well General Visual (Internal)
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL
		D633A109-AKS 53-090-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-090-00-01
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**NOSE LANDING GEAR WHEEL WELL
(LEFT SIDE, VIEW IN THE DOWN DIRECTION)****B**MPD ITEM
53-090-00

2070792 S0000429416_V2

**Above and Outboard of the Nose Landing Gear Wheel Well General Visual (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL
		D633A109-AKS 53-090-00-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA			BOEING CARD NO. 53-100-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 12 YR	REPEAT 12 YR	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS 117BL S1101			ZONE 113 114 117 118 121 122 125 126 141 142 145 146
		NOTE			

Inspect passenger compartment floor structure in dry areas (away from doors, galleys and lavs) from lower lobe.

ACCESS NOTE: Remove ceiling and sidewall panels as required. Remove/displace insulation blankets as required. Remove or displace auxiliary fuel tank as required (business jet only). Remove forward airstairs and airstairs compartment (if installed).

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-100-00-01

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737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-100-00-01						
				MECH INSP						
TASK 53-05-03-210-810										
1. INTERNAL - GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA										
(Figure 1)										
A. Inspection										
SUBTASK 53-05-03-010-008										
(1) Open these access panels:										
<table style="width: 100%; border-collapse: collapse;"> <thead> <tr> <th style="width: 15%;"><u>Number</u></th> <th style="width: 85%;"><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>117BL</td> <td>Forward Airstair Door</td> </tr> <tr> <td>S1101</td> <td>Area Above And Outboard of Nose Wheel Well Inspection</td> </tr> </tbody> </table>					<u>Number</u>	<u>Name/Location</u>	117BL	Forward Airstair Door	S1101	Area Above And Outboard of Nose Wheel Well Inspection
<u>Number</u>	<u>Name/Location</u>									
117BL	Forward Airstair Door									
S1101	Area Above And Outboard of Nose Wheel Well Inspection									
<u>NOTE:</u> Remove ceiling and sidewall panels as required. Remove/displace insulation blankets as required. Remove or displace auxiliary fuel tank as required (business jet only). Remove forward airstairs and airstairs compartment (if installed).										
SUBTASK 53-05-03-210-010										
(2) Do a General Visual inspection of the passenger compartment floor structure in dry areas (away from doors, galleys and lavs) from lower lobe.										
SUBTASK 53-05-03-910-012										
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.										
SUBTASK 53-05-03-410-008										
(4) Close these access panels:										
<table style="width: 100%; border-collapse: collapse;"> <thead> <tr> <th style="width: 15%;"><u>Number</u></th> <th style="width: 85%;"><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>117BL</td> <td>Forward Airstair Door</td> </tr> <tr> <td>S1101</td> <td>Area Above And Outboard of Nose Wheel Well Inspection</td> </tr> </tbody> </table>					<u>Number</u>	<u>Name/Location</u>	117BL	Forward Airstair Door	S1101	Area Above And Outboard of Nose Wheel Well Inspection
<u>Number</u>	<u>Name/Location</u>									
117BL	Forward Airstair Door									
S1101	Area Above And Outboard of Nose Wheel Well Inspection									
— END OF TASK —										

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-100-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA	
		D633A109-AKS 53-100-00-01	Page 3 of 9 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-100-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

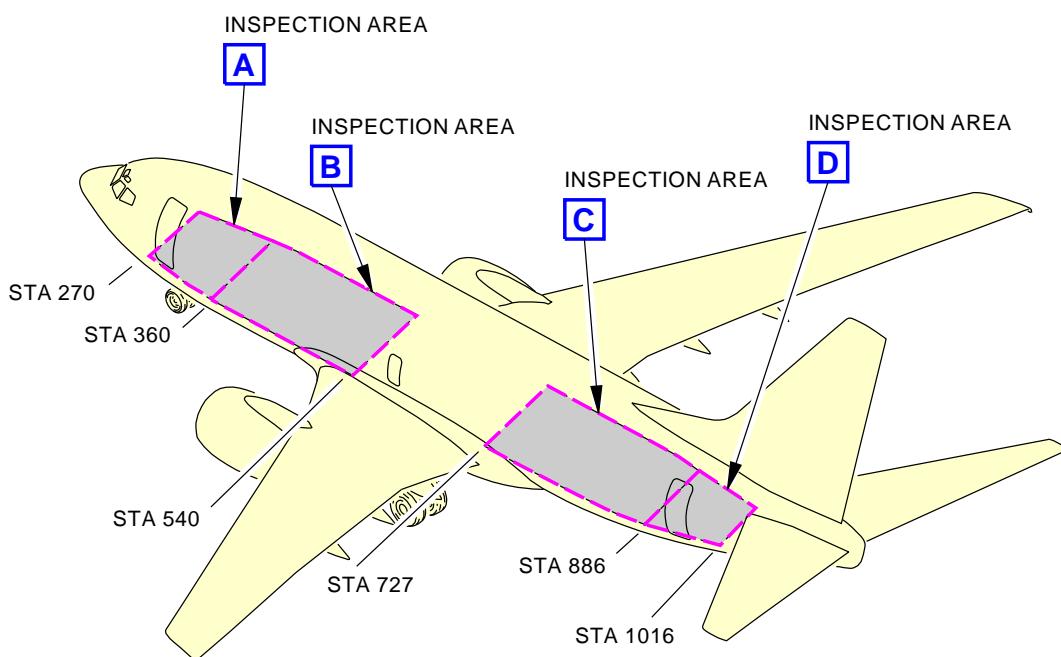
- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA	
		D633A109-AKS 53-100-00-01	Page 4 of 9 Oct 15/2014

AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-100-00-01

MPD ITEM
53-100-00

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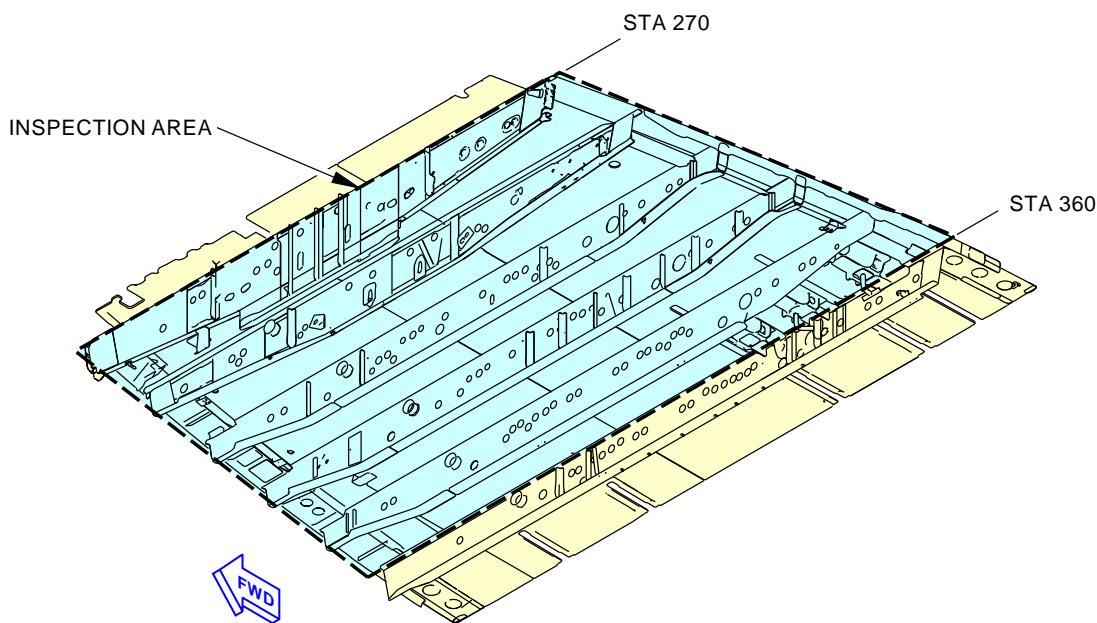
INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE
Figure 1 (Sheet 1 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-100-00-01

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Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-100-00-01

**PASSENGER COMPARTMENT FLOOR STRUCTURE
(BOTTOM VIEW)****A**MPD ITEM
53-100-00

2089843 S0000440308_V2

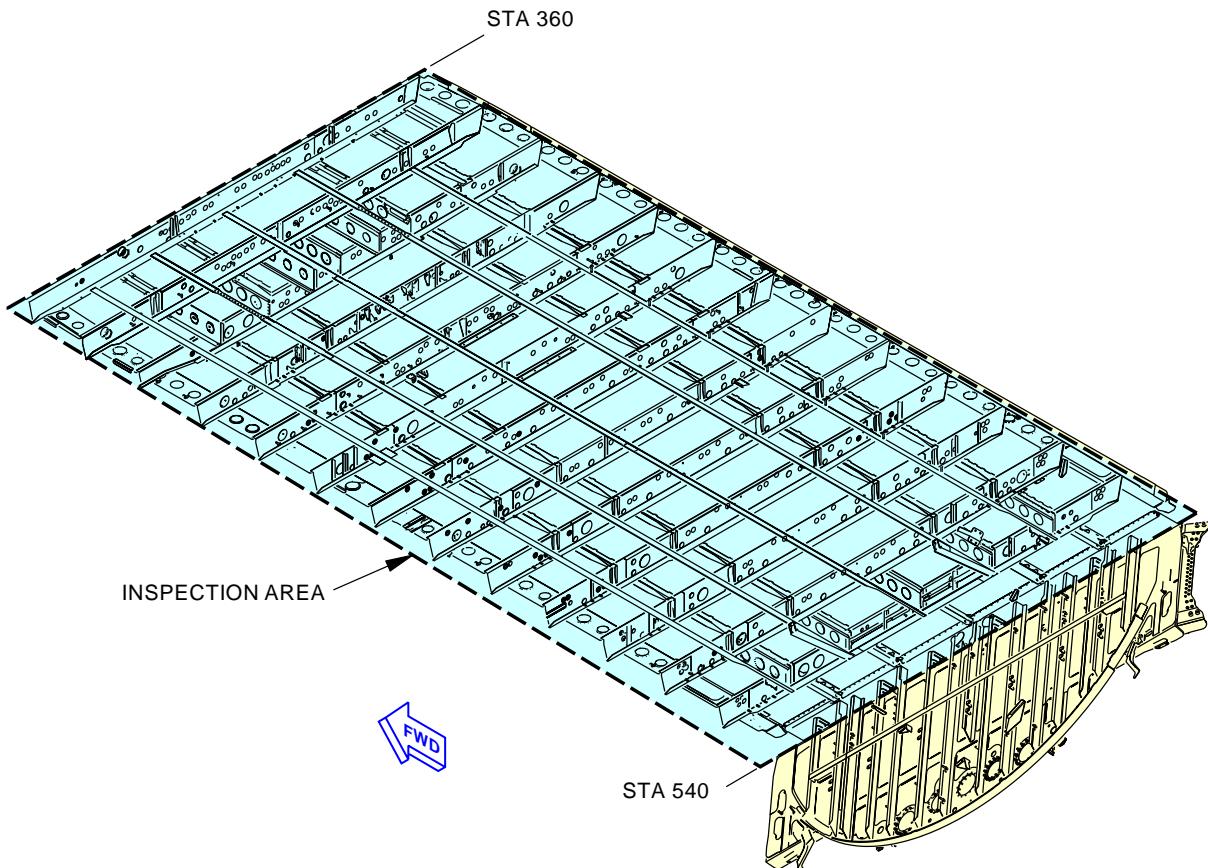
INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE
Figure 1 (Sheet 2 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-100-00-01

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Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-100-00-01

**PASSENGER COMPARTMENT FLOOR STRUCTURE
(BOTTOM VIEW)****B**MPD ITEM
53-100-00

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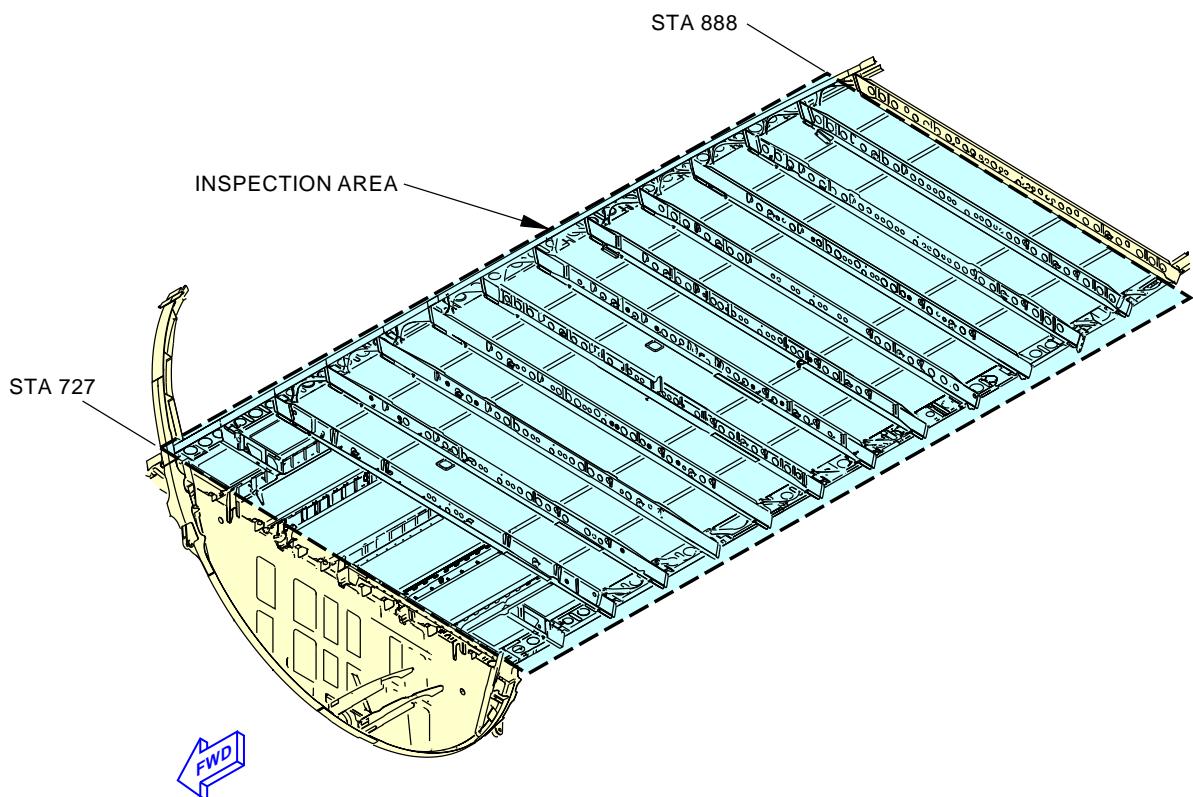
**INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE
Figure 1 (Sheet 3 of 5)**

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-100-00-01

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Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-100-00-01

**PASSENGER COMPARTMENT FLOOR STRUCTURE
(BOTTOM VIEW)****C**MPD ITEM
53-100-00

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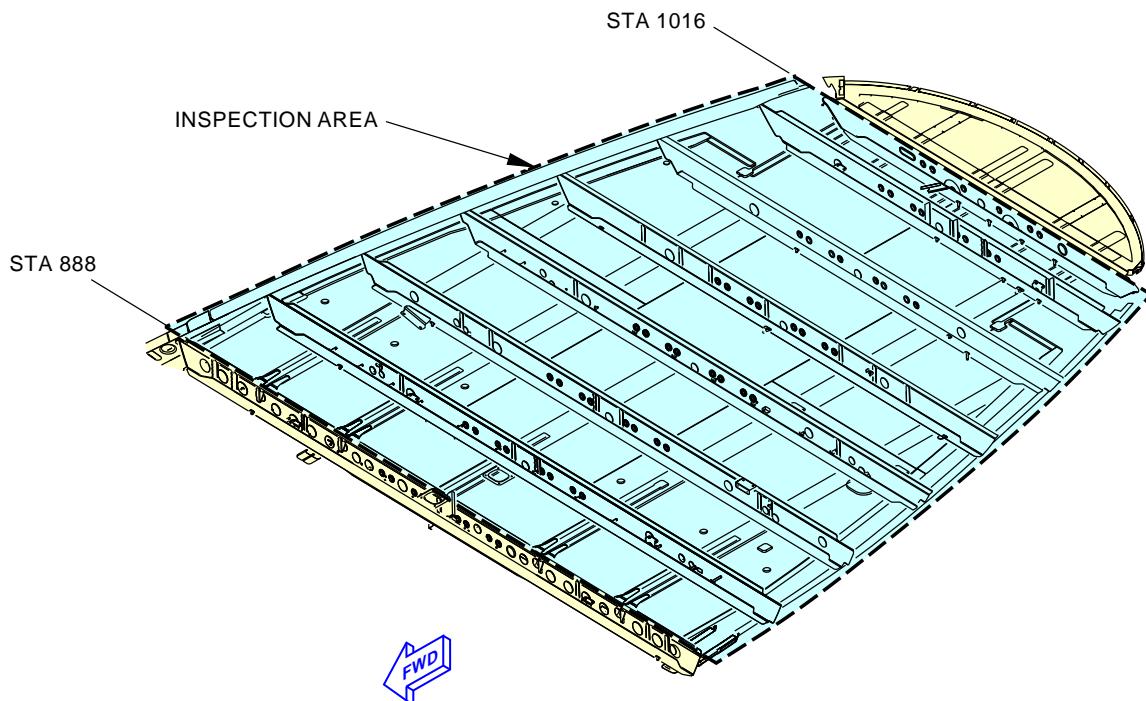
INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE
Figure 1 (Sheet 4 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-100-00-01

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Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-100-00-01

**PASSENGER COMPARTMENT FLOOR STRUCTURE
(BOTTOM VIEW)****D**MPD ITEM
53-100-00

2089863 S0000440311_V2

**INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE
Figure 1 (Sheet 5 of 5)**

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-100-00-01

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Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-110-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 NOTE	THRESHOLD NOTE	REPEAT	RELATED CARD
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 117BL S1002			ZONE 113 114 117 118 121 122 125 126 141 142 145 146
		NOTE			

Inspect passenger compartment floor structure in wet areas (within approximately 20 inches from doors, galleys and lavs) from lower lobe.

INTERVAL NOTE: Threshold Interval 8YR / Repeat interval 6YR, whichever occurs first, applicable to Airplanes L/N# 1-2412 and those that have not incorporated the HI-TAK Gel Tape. Threshold Interval 9YR / Repeat interval 6YR, whichever occurs first, applicable to Airplanes L/N# 2413 and on, or those that have incorporated the HI-TAK Gel Tape.

ACCESS NOTE: Remove ceiling and sidewall panels as required. Remove/displace insulation blankets as required. Remove or displace auxiliary fuel tank as required (business jet only). Remove forward airstairs and airstair compartment (if installed. 117BL).

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-110-00-01

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Jun 15/2015

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-110-00-01	MECH	INSP
TASK 53-05-03-210-811						
1. INTERNAL - GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA						
(Figure 1)						
A. Inspection						
SUBTASK 53-05-03-010-009						
(1) Open these access panels:						
Number Name/Location						
117BL Forward Airstair Door						
S1101 Area Above And Outboard of Nose Wheel Well Inspection						
NOTE: Remove ceiling and sidewall panels as required. Remove/displace insulation blankets as required. Remove or displace auxiliary fuel tank as required (business jet only). Remove forward airstairs and airstair compartment (if installed. 117BL).						
SUBTASK 53-05-03-210-011						
(2) Do a General Visual inspection of the passenger compartment floor structure in wet areas (within approximately 20 inches from doors, galleys and lavs) from lower lobe.						
SUBTASK 53-05-03-910-013						
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.						
SUBTASK 53-05-03-410-009						
(4) Close these access panels:						
Number Name/Location						
117BL Forward Airstair Door						
S1101 Area Above And Outboard of Nose Wheel Well Inspection						
———— END OF TASK ————						

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	D633A109-AKS 53-110-00-01	Page 2 of 9 Feb 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-110-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	D633A109-AKS 53-110-00-01	Page 3 of 9 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-110-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

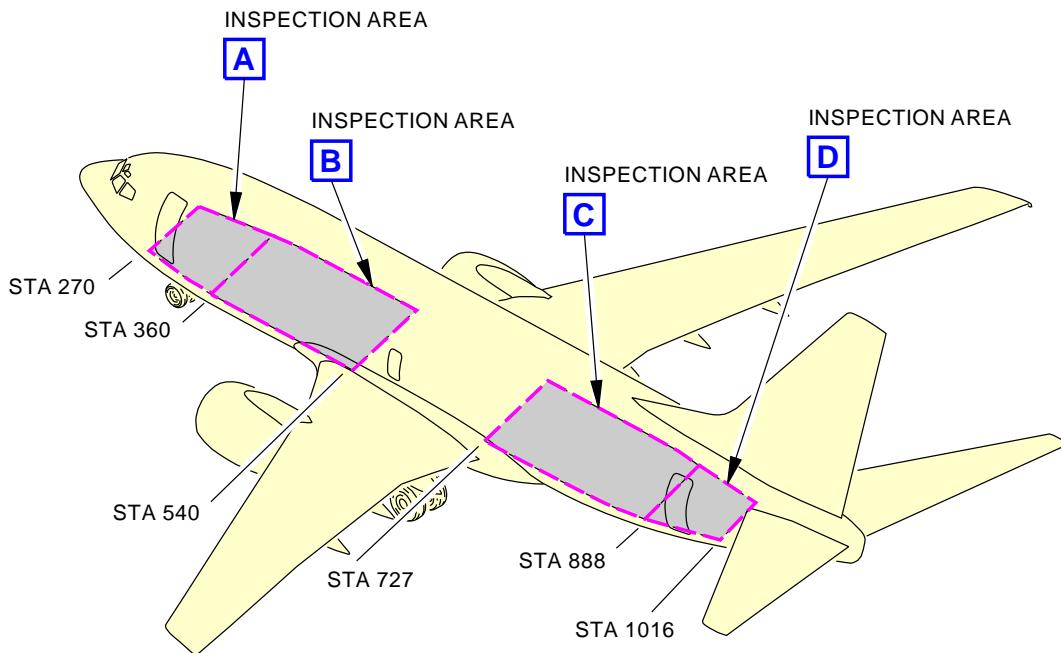
- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	
		D633A109-AKS 53-110-00-01	Page 4 of 9 Oct 15/2014

AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-110-00-01
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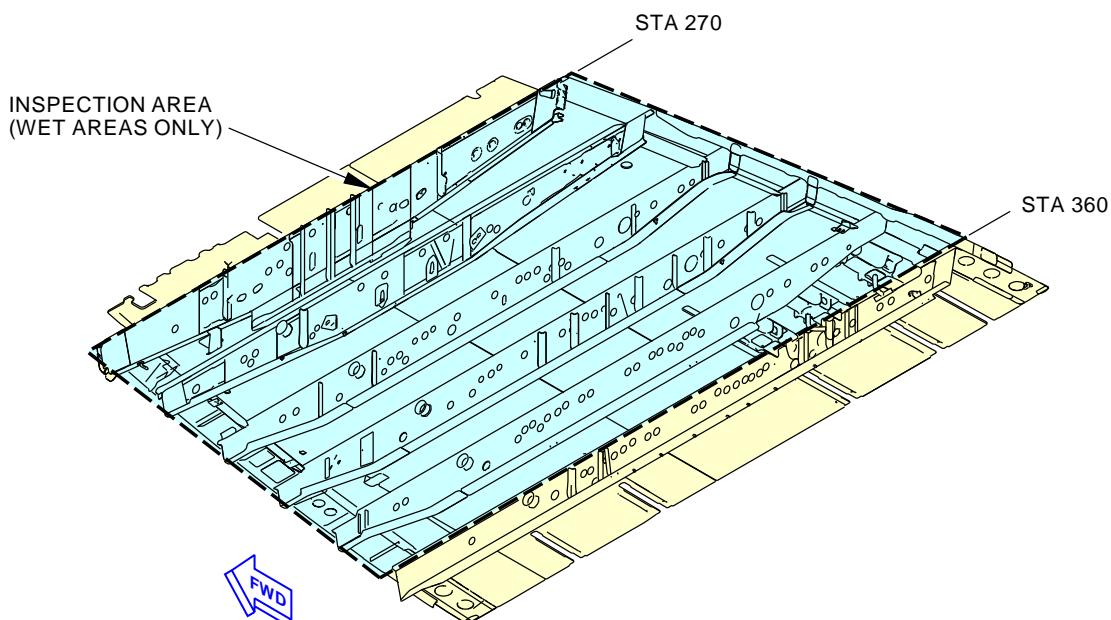
MPD ITEM
53-110-00

2123126 S0000457776_V2
Passenger Compartment Structure-Wet Area General Visual (Internal)
Figure 1 (Sheet 1 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-110-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-110-00-01

**PASSENGER COMPARTMENT FLOOR STRUCTURE
(BOTTOM VIEW)****A**MPD ITEM
53-110-00

2123135 S0000457777_V2

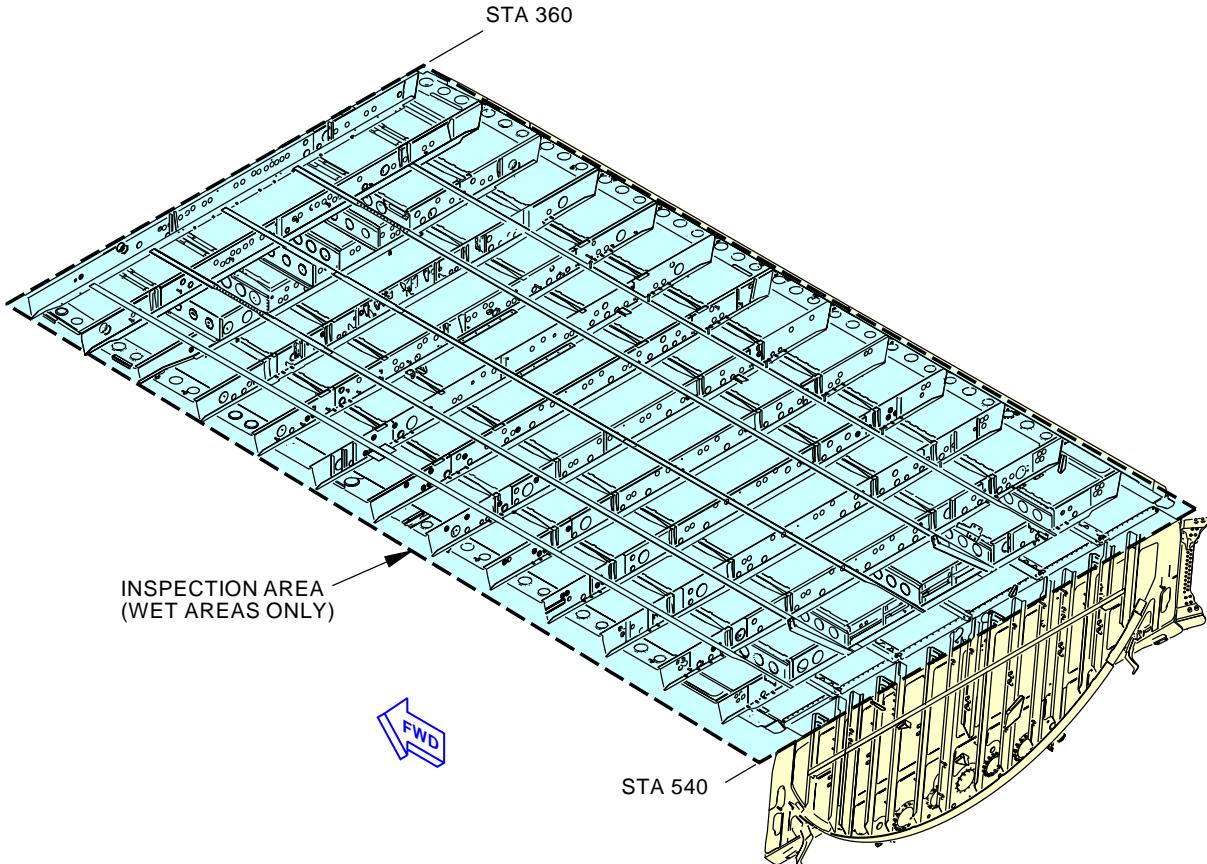
**Passenger Compartment Structure-Wet Area General Visual (Internal)
Figure 1 (Sheet 2 of 5)**

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-110-00-01

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Oct 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-110-00-01
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**PASSENGER COMPARTMENT FLOOR STRUCTURE
(BOTTOM VIEW)****B**MPD ITEM
53-110-00

2123138 S0000457778_V2

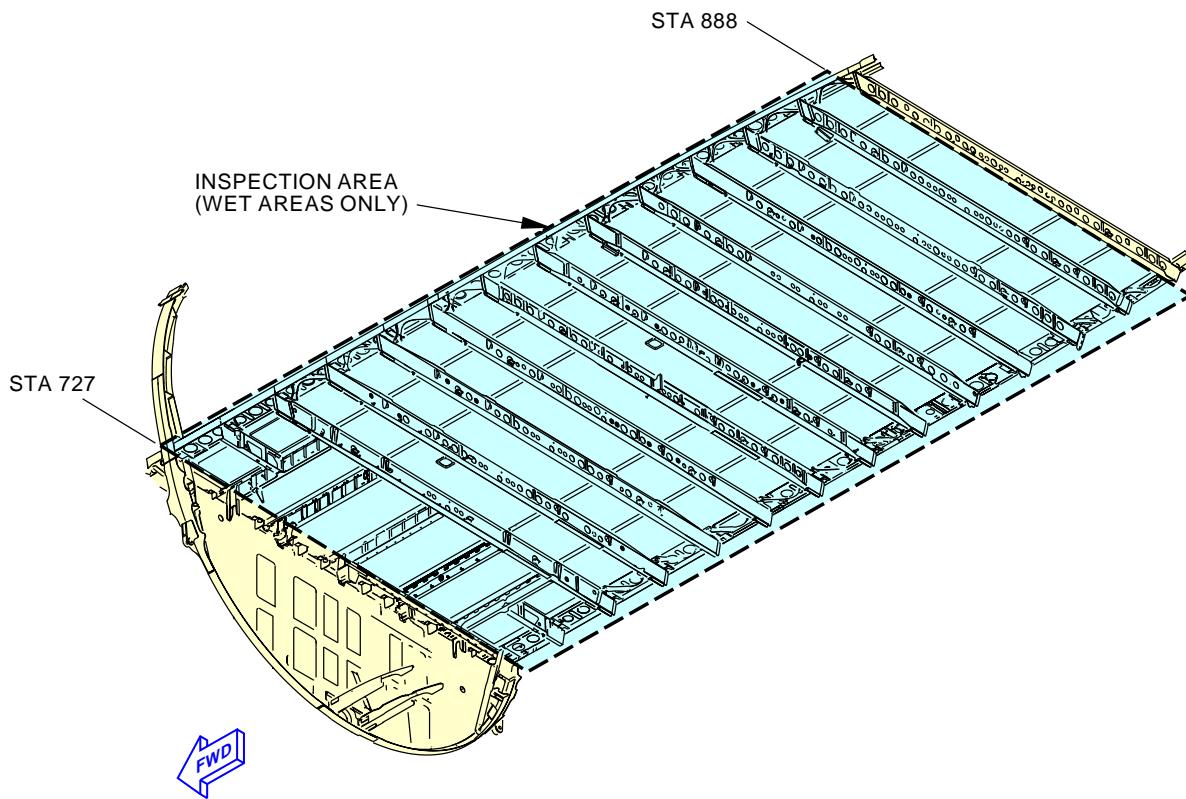
**Passenger Compartment Structure-Wet Area General Visual (Internal)
Figure 1 (Sheet 3 of 5)**

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-110-00-01

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AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-110-00-01



PASSENGER COMPARTMENT FLOOR STRUCTURE
(BOTTOM VIEW)

CMPD ITEM
53-110-00

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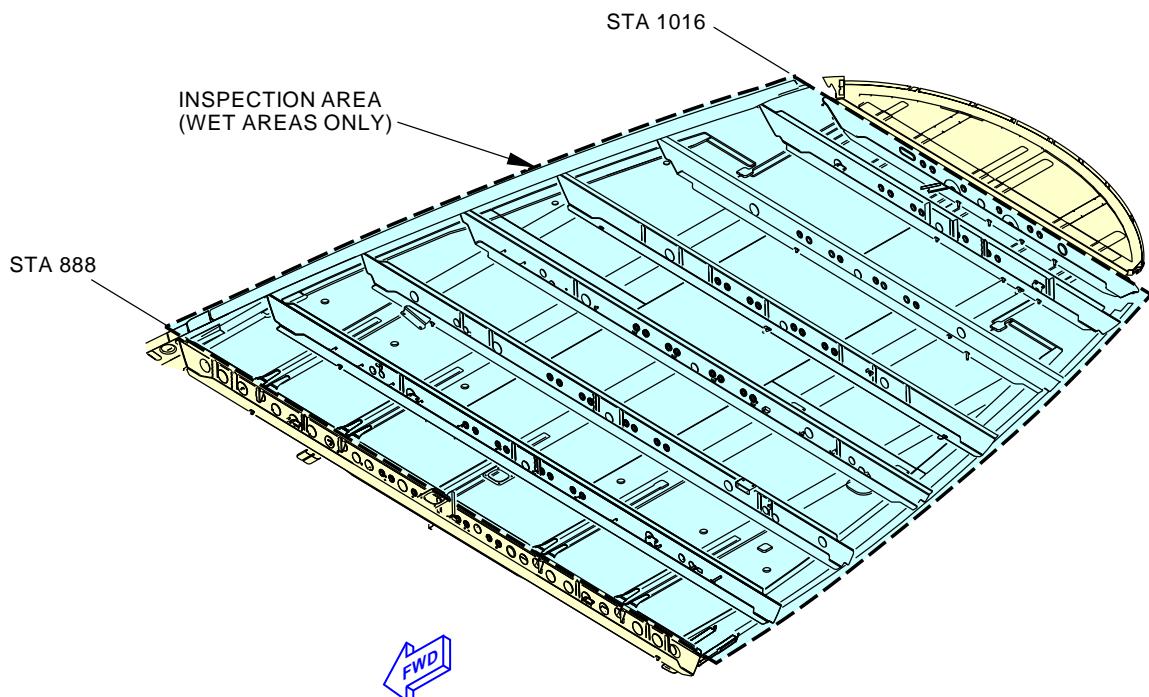
Passenger Compartment Structure-Wet Area General Visual (Internal)
Figure 1 (Sheet 4 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-110-00-01

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Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-110-00-01

**PASSENGER COMPARTMENT FLOOR STRUCTURE
(BOTTOM VIEW)****D**MPD ITEM
53-110-00

2123145 S0000457780_V2

**Passenger Compartment Structure-Wet Area General Visual (Internal)
Figure 1 (Sheet 5 of 5)**

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-110-00-01

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Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE ELECTRICAL AND ELECTRONICS COMPARTMENT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-120-00-01
TAIL NUMBER	WORK AREA E/E COMPARTMENT	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 117A S1102 NOTE			ZONE 117 118

Inspect EE compartment (STA 294.5 to 396 (STA 400 for 737-700C)), including: 1. Skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices, bulkhead at STA 294.5; 2. EE compartment door and airstairs door cutout surround structure; 3. Forward entry and galley door cutout surround structure (portion in lower lobe).

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove LRUs and racks, do not remove permanently installed structure. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT	
		D633A109-AKS 53-120-00-01	Page 1 of 10 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-120-00-01
TASK 53-05-03-210-812				MECH INSP

1. INTERNAL - GENERAL VISUAL: ELECTRICAL AND ELECTRONICS COMPARTMENT
(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-010

(1) Open these access panels:

<u>Number</u>	<u>Name/Location</u>
117A	Electronic Equipment Access Door
S1102	Electrical And Electronics Compartment Inspection

NOTE: Remove LRUs and racks, do not remove permanently installed structure.
Remove/displace insulation blankets as required.

SUBTASK 53-05-03-210-012

(2) Do a General Visual inspection of the EE compartment (STA 294.5 to 396 (STA 400 for 737-700C)), including:

1. Skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices, bulkhead at STA 294.5.
2. EE compartment door and airstairs door cutout surround structure.
3. Forward entry and galley door cutout surround structure (portion in lower lobe).

SUBTASK 53-05-03-910-014

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

SUBTASK 53-05-03-410-010

(4) Close these access panels:

<u>Number</u>	<u>Name/Location</u>
117A	Electronic Equipment Access Door
S1102	Electrical And Electronics Compartment Inspection

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT	
		D633A109-AKS 53-120-00-01	Page 2 of 10 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-120-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT	
		D633A109-AKS 53-120-00-01	Page 3 of 10 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-120-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

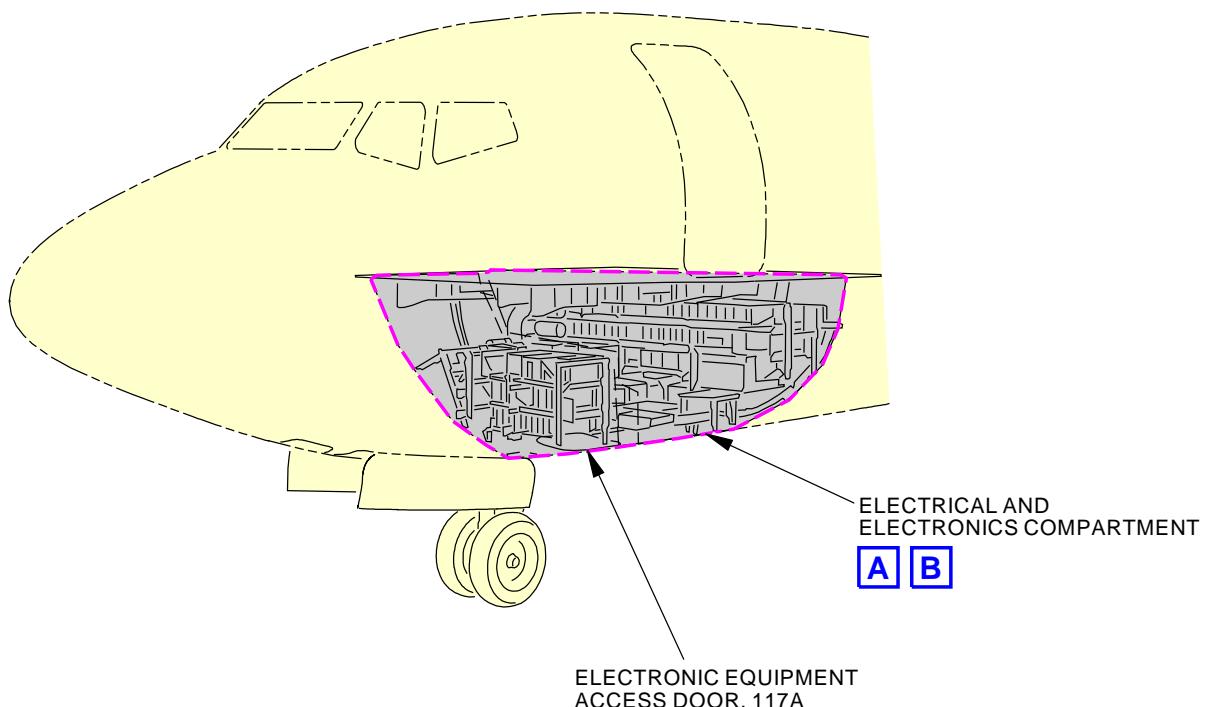
- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT	
		D633A109-AKS 53-120-00-01	Page 4 of 10 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-120-00-01

MPD ITEM
53-120-00

2084073 S0000437120_V2

**Electrical Equipment Access Door General Visual (Internal)
Figure 1 (Sheet 1 of 6)**

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT
		D633A109-AKS 53-120-00-01

**Page 5 of 10
Oct 15/2015**

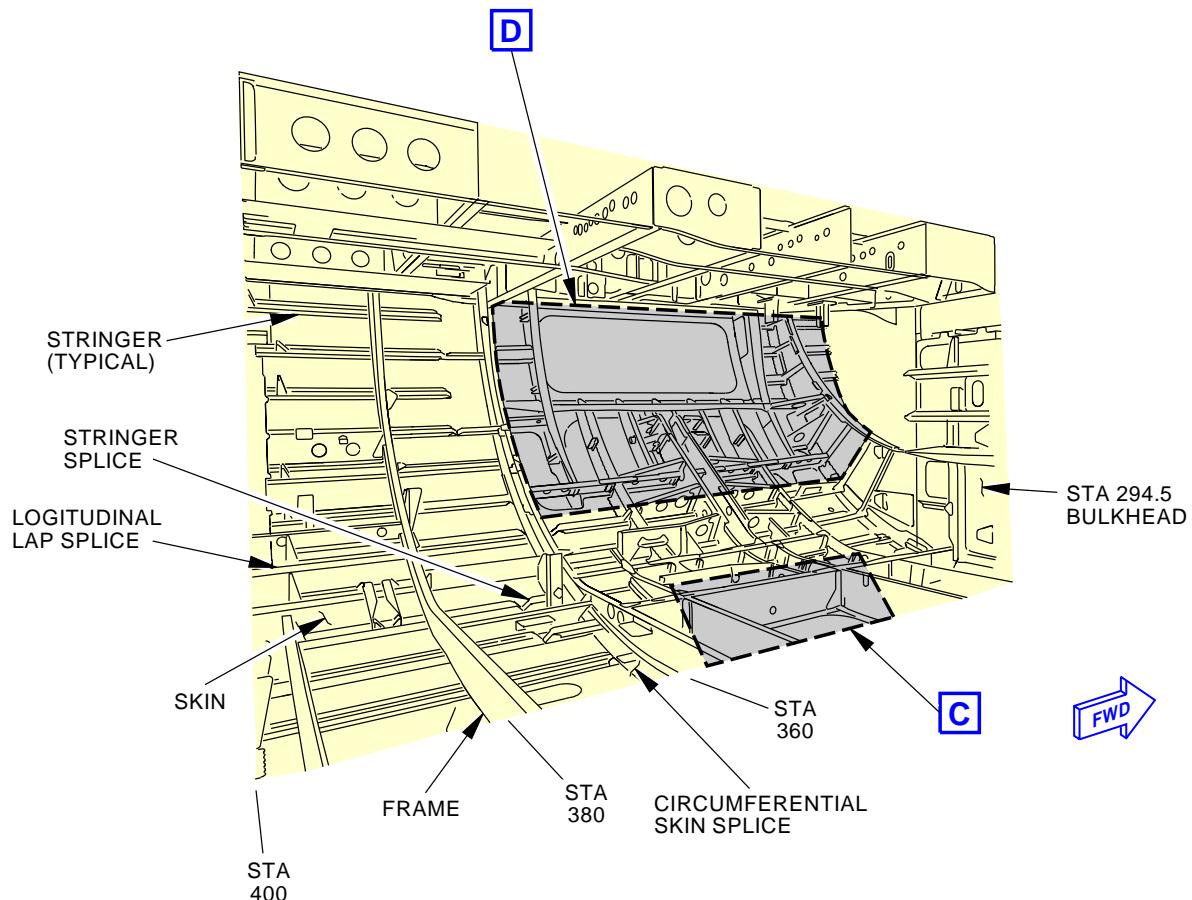
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-120-00-01**ELECTRICAL AND ELECTRONIC COMPARTMENT
(LEFT SIDE)**MPD ITEM
53-120-00

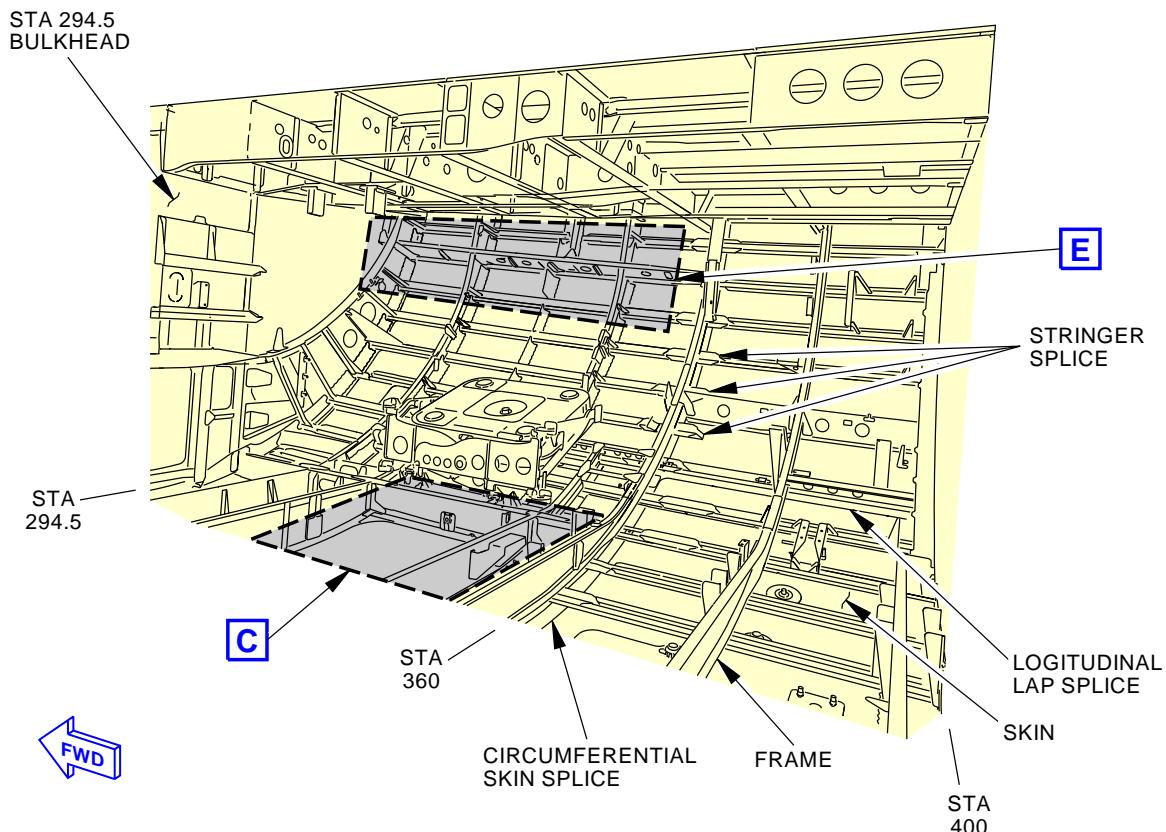
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**Electrical Equipment Access Door General Visual (Internal)
Figure 1 (Sheet 2 of 6)**

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT
		D633A109-AKS 53-120-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-120-00-01
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**ELECTRICAL AND ELECTRONIC COMPARTMENT
(RIGHT SIDE)**MPD ITEM
53-120-00

2081267 S0000437122_V2

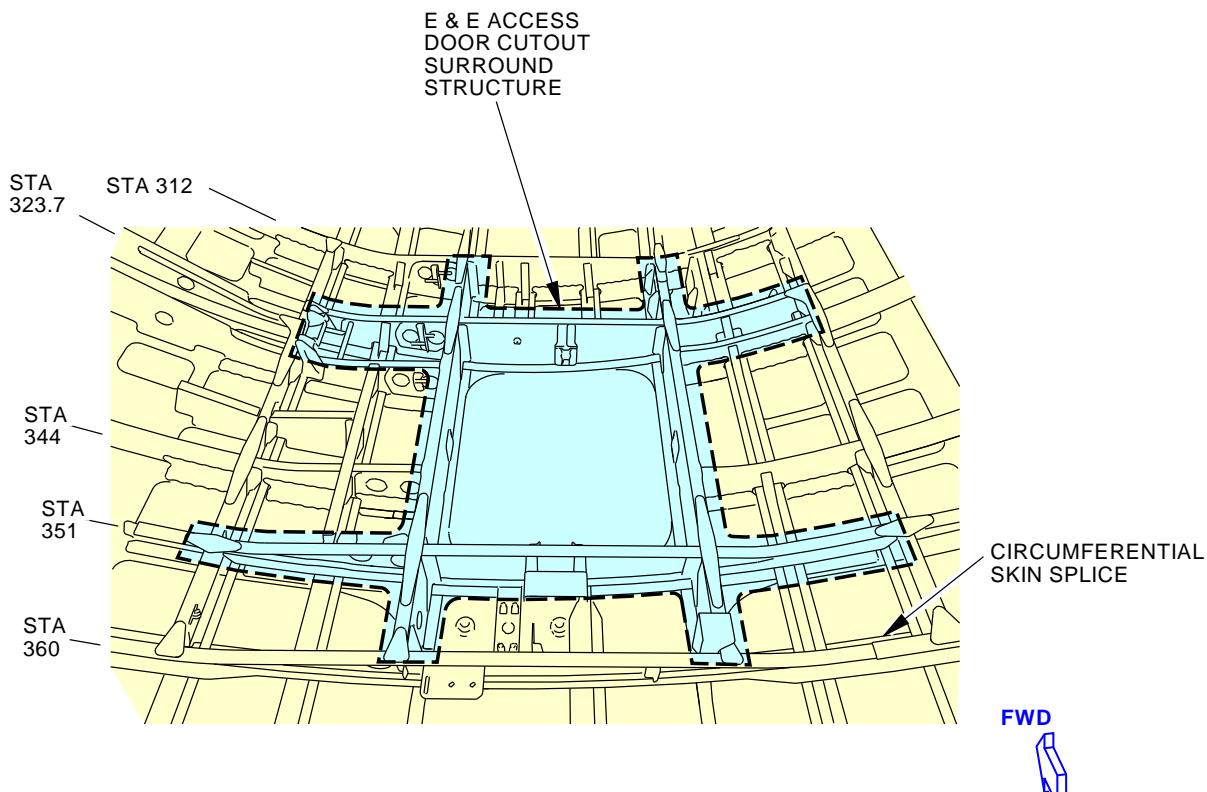
**Electrical Equipment Access Door General Visual (Internal)
Figure 1 (Sheet 3 of 6)**

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT
		D633A109-AKS 53-120-00-01

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Oct 15/2015

AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-120-00-01

MPD ITEM
53-120-00

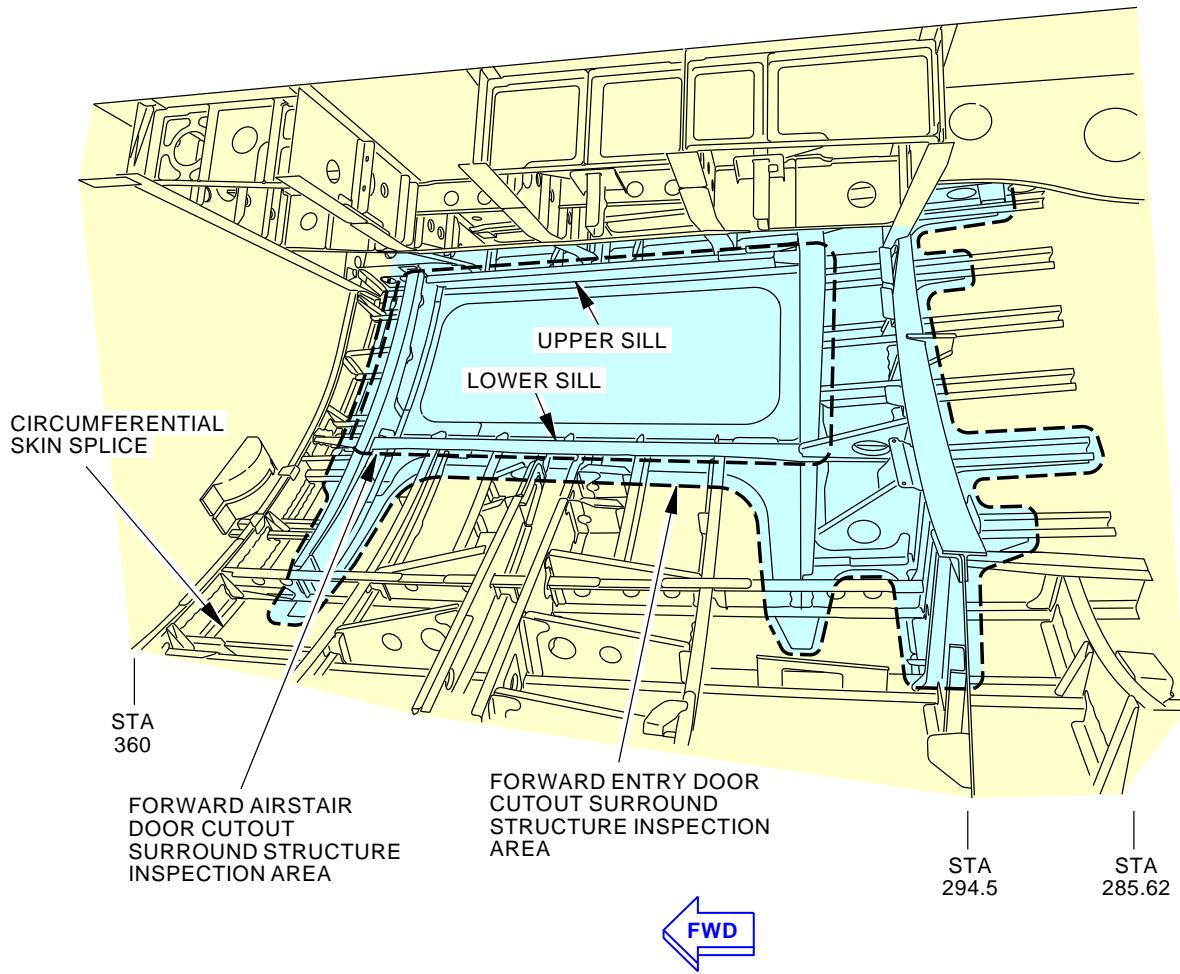
2084481 S0000437123_V2

**Electrical Equipment Access Door General Visual (Internal)
Figure 1 (Sheet 4 of 6)**

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT
		D633A109-AKS 53-120-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-120-00-01

**FORWARD AIRSTAIR DOOR AND FORWARD ENTRY DOOR CUTOUT
AND SURROUND STRUCTURE**MPD ITEM
53-120-00

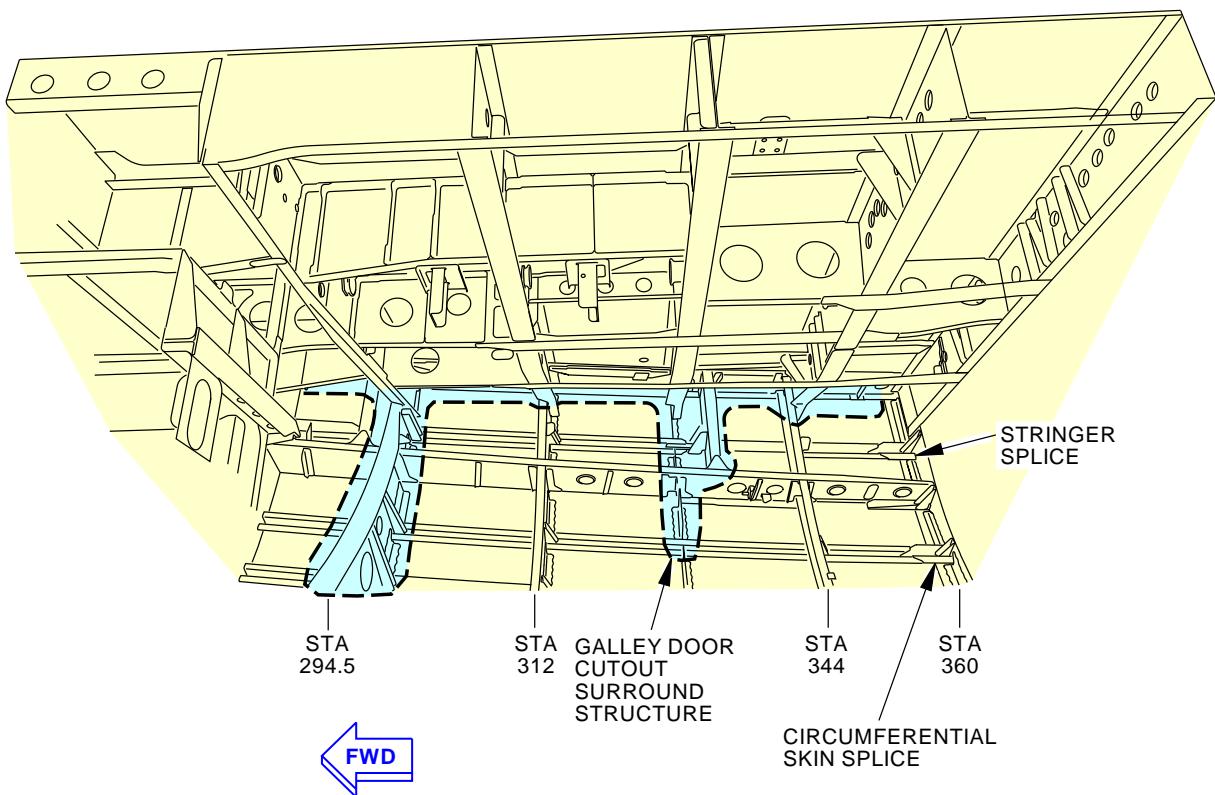
2088473 S0000437124_V3

**Electrical Equipment Access Door General Visual (Internal)
Figure 1 (Sheet 5 of 6)**

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT
		D633A109-AKS 53-120-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-120-00-01

**GALLEY DOOR CUTOUT SURROUND STRUCTURE****E**MPD ITEM
53-120-00

2086454 S0000437125_V2

**Electrical Equipment Access Door General Visual (Internal)
Figure 1 (Sheet 6 of 6)**

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT
		D633A109-AKS 53-120-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO COMPARTMENT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-130-00-01
TAIL NUMBER	WORK AREA FWD CARGO	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC	
		ACCESS S1201			ZONE 121 122
		NOTE			

Inspect forward cargo compartment skin panels including skins, frames, and stringers (note: inspection includes the circumferential skin and stringer splice at STA 500E for the -900 models).

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove sidewalls and ceiling panels. Remove/displace insulation blankets as required.
Remove/displace auxiliary fuel tank as required (business jet only).

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-130-00-01

Page 1 of 8
Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-130-00-01
				MECH INSP
TASK 53-05-03-210-813				
1. INTERNAL - GENERAL VISUAL: FORWARD CARGO COMPARTMENT				
(Figure 1)				
A. Inspection				
SUBTASK 53-05-03-010-011				
(1) Open this access panel:				
Number Name/Location				
S1201 Forward Cargo Compartment Inspection				
NOTE: Remove sidewalls and ceiling panels. Remove/displace insulation blankets as required. Remove/displace auxiliary fuel tank as required (business jet only).				
SUBTASK 53-05-03-210-013				
(2) Do a General Visual inspection of the forward cargo compartment skin panels including skins, frames, and stringers (note: inspection includes the circumferential skin and stringer splice at STA 500E for the -900 models).				
SUBTASK 53-05-03-910-015				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.				
SUBTASK 53-05-03-410-011				
(4) Close this access panel:				
Number Name/Location				
S1201 Forward Cargo Compartment Inspection				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-130-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-130-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT	
		D633A109-AKS 53-130-00-01	Page 3 of 8 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-130-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

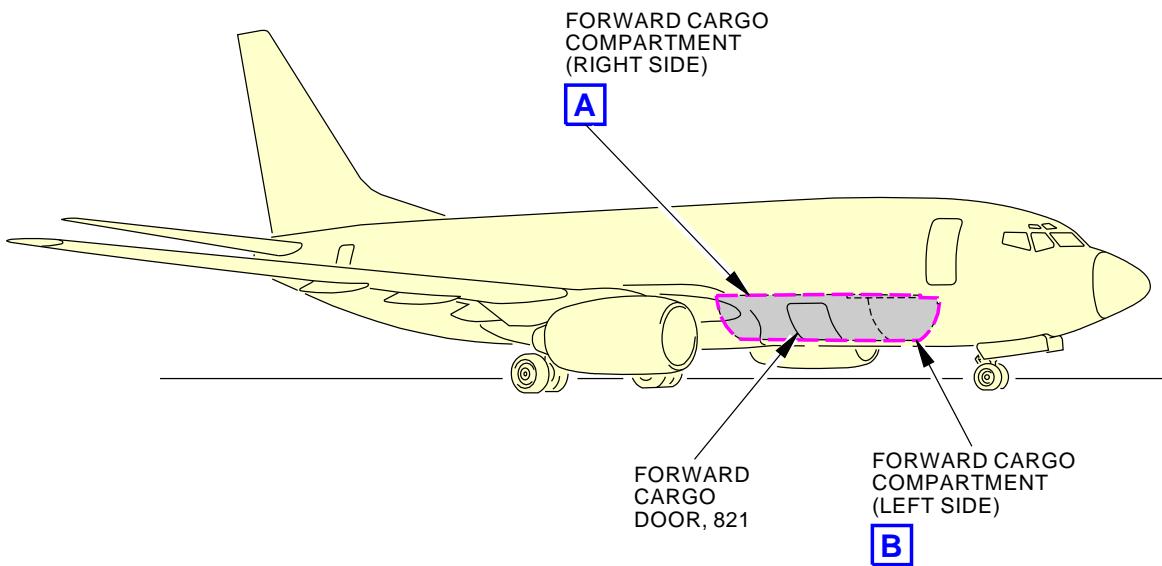
END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-130-00-01

Page 4 of 8
Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-130-00-01

MPD ITEM
53-130-00

2080203 S0000436833_V3

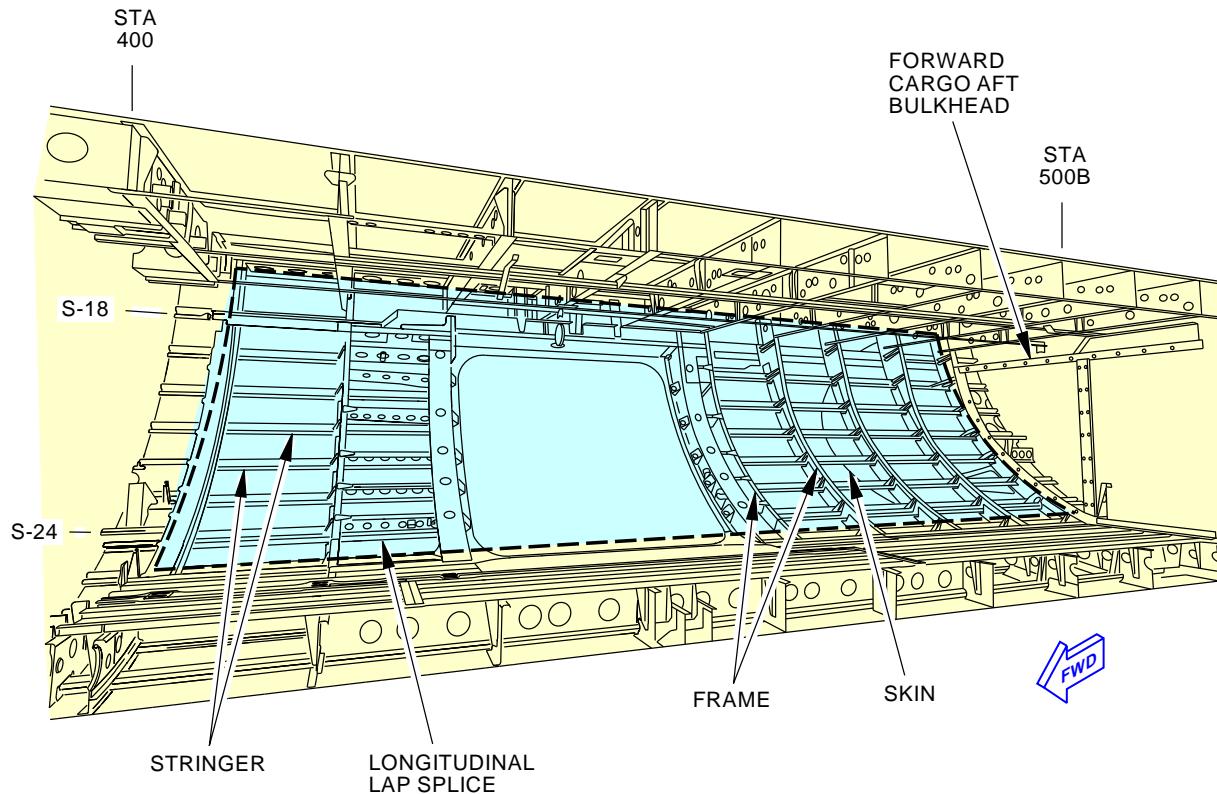
**Forward Cargo Compartment General Visual (Internal) (Sidewall Liners and Insulation Removed)
Figure 1 (Sheet 1 of 4)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-130-00-01

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Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-130-00-01
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**FORWARD CARGO COMPARTMENT (STA 400 TO AFT BULKHEAD)
(RIGHT SIDE VIEW)****A**MPD ITEM
53-130-00

2080217 S0000436834_V3

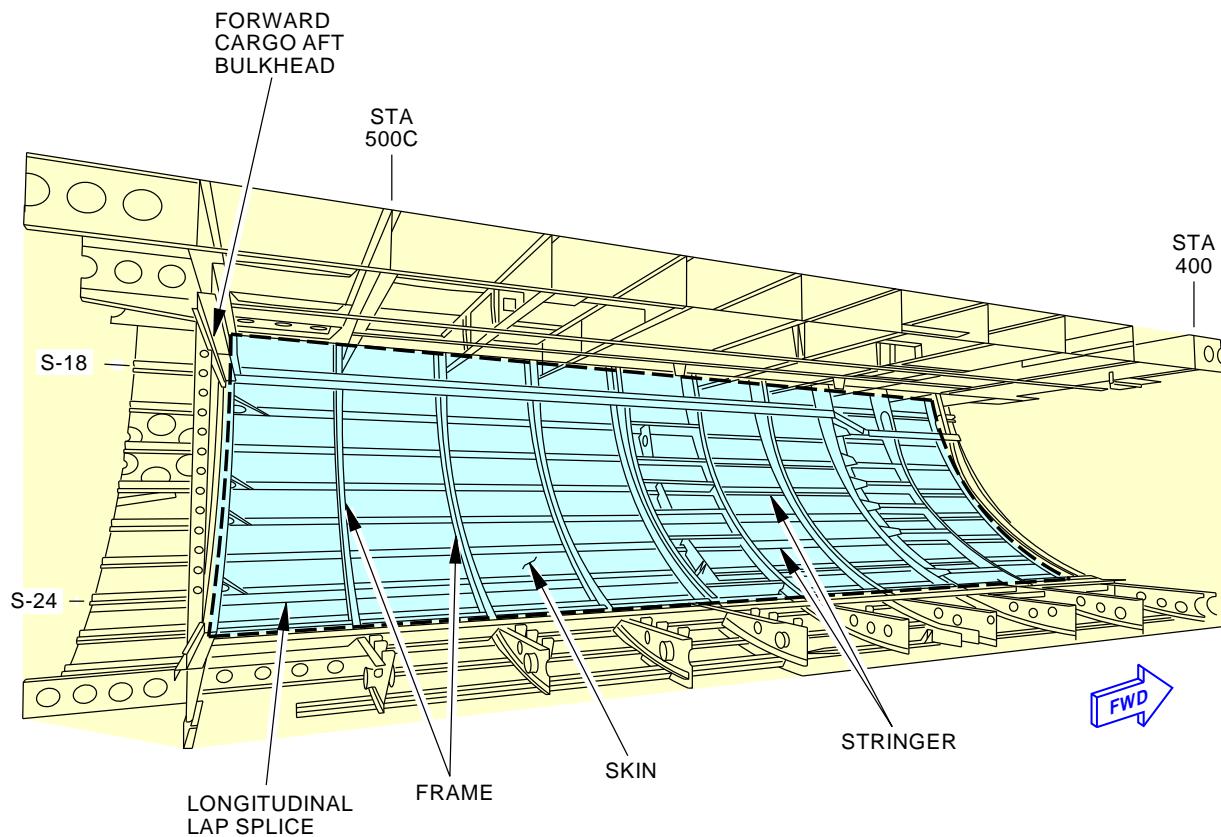
**Forward Cargo Compartment General Visual (Internal) (Sidewall Liners and Insulation Removed)
Figure 1 (Sheet 2 of 4)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-130-00-01

Page 6 of 8
Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-130-00-01
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**FORWARD CARGO COMPARTMENT (STA 400 TO AFT BULKHEAD)
(LEFT SIDE VIEW)****B**MPD ITEM
53-130-00

2082663 S0000436835_V3

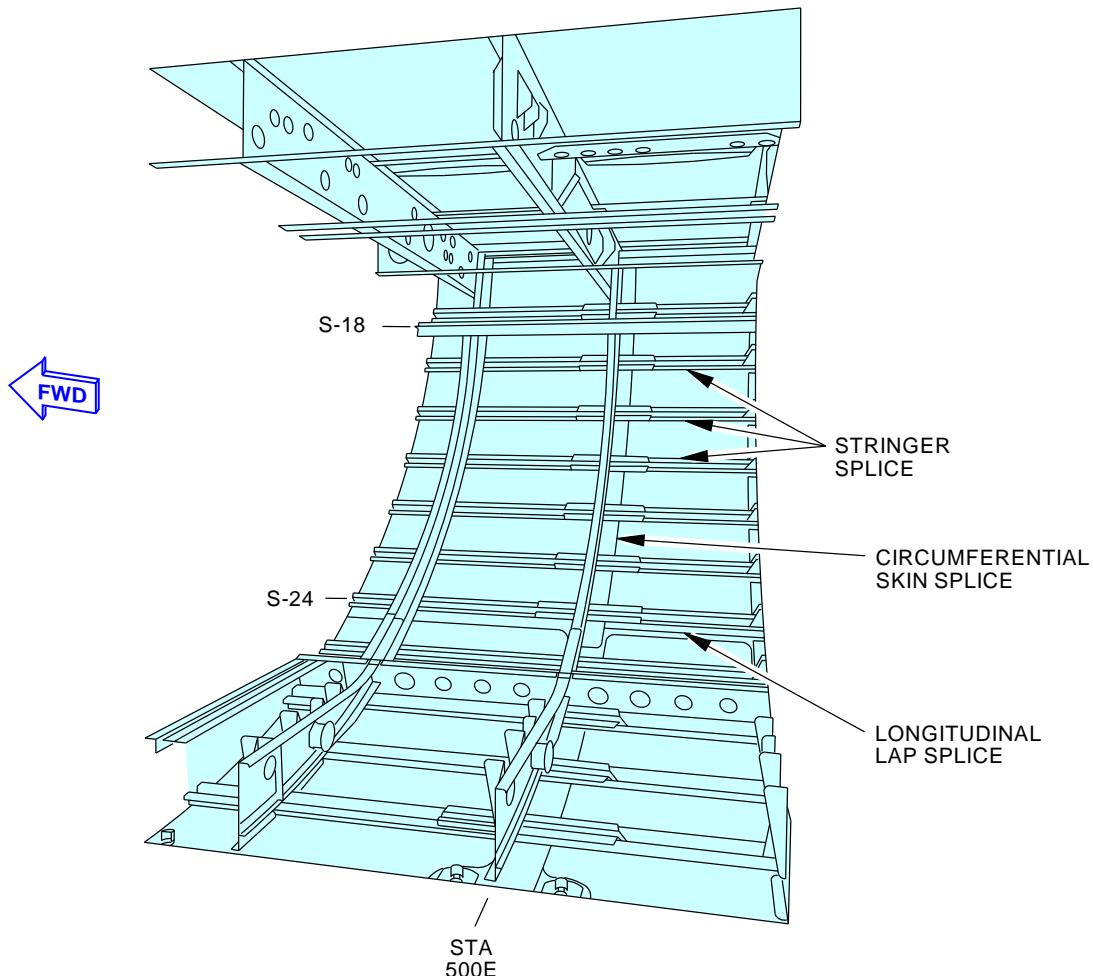
**Forward Cargo Compartment General Visual (Internal) (Sidewall Liners and Insulation Removed)
Figure 1 (Sheet 3 of 4)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-130-00-01

Page 7 of 8
Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-130-00-01



**CIRCUMFERENTIAL SKIN AND STRINGER SPLICE -900 MODEL
(RIGHT SIDE VIEW, LEFT SIDE OPPOSITE)**

MPD ITEM
53-130-00

2082409 S0000436836_V3

**Forward Cargo Compartment General Visual (Internal) (Sidewall Liners and Insulation Removed)
Figure 1 (Sheet 4 of 4)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-130-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO COMPARTMENT - FLOOR STRUCTURE			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-140-00-01
TAIL NUMBER	WORK AREA FWD CARGO	VERSION 1.1	THRESHOLD 8 YR	REPEAT 6 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	24000 FC	18000 FC	AIRPLANE ALL ENGINE ALL
		ACCESS S1003			ZONE 121 122
		NOTE			

Inspect forward cargo compartment floor structure.

INTERVAL NOTE: Whichever comes first.**ACCESS NOTE:** Remove cargo floor panels.**A. References**

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT - FLOOR STRUCTURE	
		D633A109-AKS 53-140-00-01	Page 1 of 6 Jun 15/2015

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-140-00-01
				MECH INSP
TASK 53-05-03-210-814				
1. INTERNAL - GENERAL VISUAL: FORWARD CARGO COMPARTMENT FLOOR STRUCTURE				
(Figure 1)				
A. Inspection				
SUBTASK 53-05-03-010-012				
(1) Open this access panel:				
Number Name/Location				
S1003 Forward and Aft Cargo Compartment Floor Structure Inspection				
NOTE: Remove cargo floor panels.				
SUBTASK 53-05-03-210-014				
(2) Do a General Visual inspection of the forward cargo compartment floor structure.				
SUBTASK 53-05-03-910-016				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.				
SUBTASK 53-05-03-410-012				
(4) Close this access panel:				
Number Name/Location				
S1003 Forward and Aft Cargo Compartment Floor Structure Inspection				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT - FLOOR STRUCTURE		
D633A109-AKS 53-140-00-01				

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-140-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT - FLOOR STRUCTURE	
		D633A109-AKS 53-140-00-01	Page 3 of 6 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-140-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT - FLOOR STRUCTURE
		D633A109-AKS 53-140-00-01

Page 4 of 6
Oct 15/2014

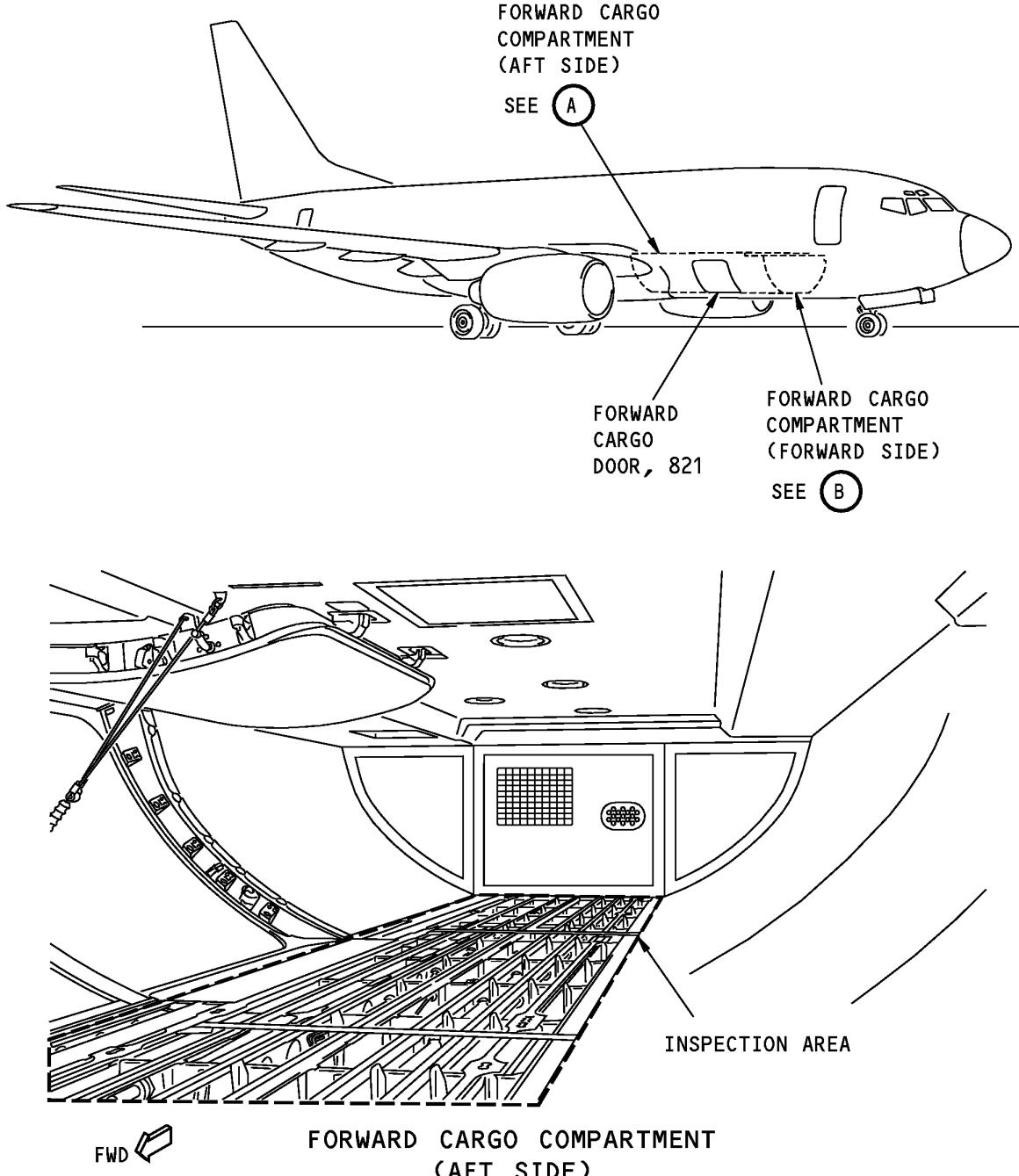
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

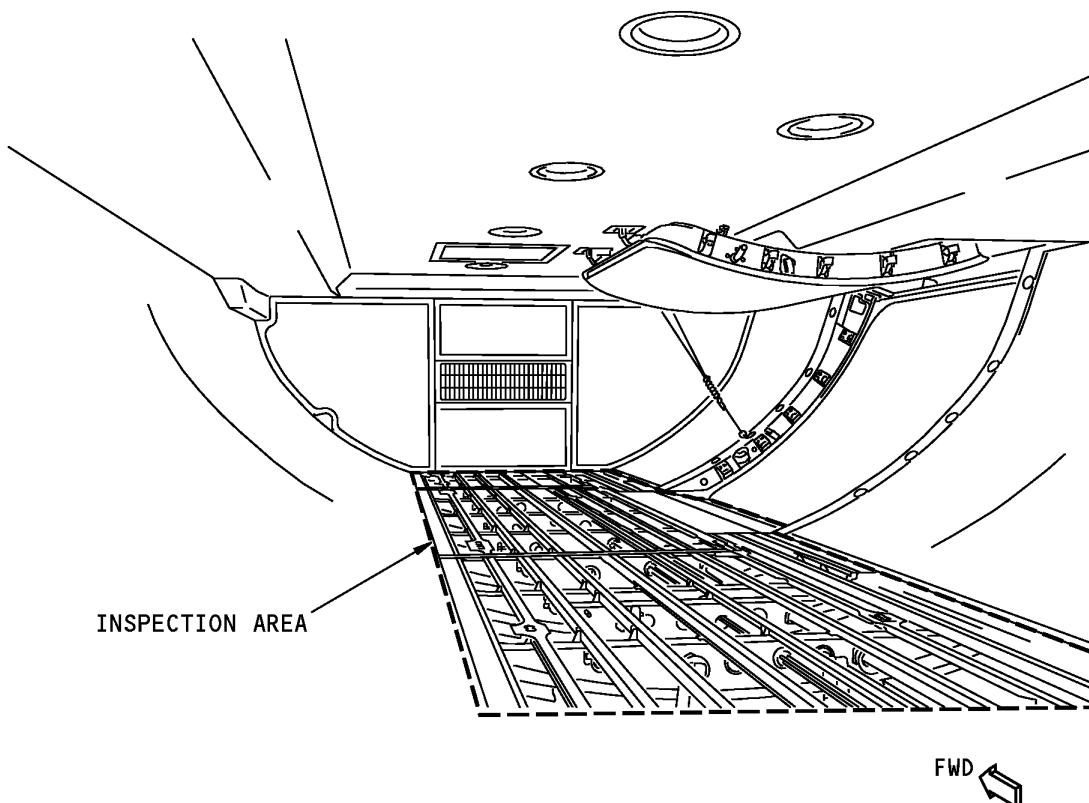
BOEING CARD NO.
53-140-00-01

Forward Cargo Compartment Floor Structure General Visual (Internal)
Figure 1 (Sheet 1 of 2)

EFFECTIVITY
AKS ALLSOURCE
MRB**FORWARD CARGO COMPARTMENT - FLOOR STRUCTURE****D633A109-AKS
53-140-00-01****Page 5 of 6
Jun 15/2016**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-140-00-01

**FORWARD CARGO COMPARTMENT
(FORWARD SIDE)****B****Forward Cargo Compartment Floor Structure General Visual (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT - FLOOR STRUCTURE
		D633A109-AKS 53-140-00-01

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO COMPARTMENT - FLOOR STRUCTURE			BOEING CARD NO. 53-140-00-02
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 8 YR	REPEAT 6 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	24000 FC	18000 FC	AIRPLANE ALL ENGINE ALL
		ACCESS S1003			ZONE 141 142
		NOTE			

Inspect Aft cargo compartment floor structure.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove cargo floor panels.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT - FLOOR STRUCTURE
		D633A109-AKS 53-140-00-02

**Page 1 of 6
Jun 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-140-00-02
------	-------------	---------	------------------	--

TASK 53-05-03-210-815

MECH

INSP

1. INTERNAL - GENERAL VISUAL: AFT CARGO COMPARTMENT FLOOR STRUCTURE

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-013

- (1) Open this access panel:

Number Name/Location

S1003 Forward and Aft Cargo Compartment Floor Structure Inspection

NOTE: Remove cargo floor panels.

SUBTASK 53-05-03-210-015

- (2) Do a General Visual inspection of the aft cargo compartment floor structure.

SUBTASK 53-05-03-910-017

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

SUBTASK 53-05-03-410-013

- (4) Close this access panel:

Number Name/Location

S1003 Forward and Aft Cargo Compartment Floor Structure Inspection

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT - FLOOR STRUCTURE
		D633A109-AKS 53-140-00-02

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Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-140-00-02
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT - FLOOR STRUCTURE	
		D633A109-AKS 53-140-00-02	Page 3 of 6 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-140-00-02				
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MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT - FLOOR STRUCTURE
		D633A109-AKS 53-140-00-02

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Oct 15/2014

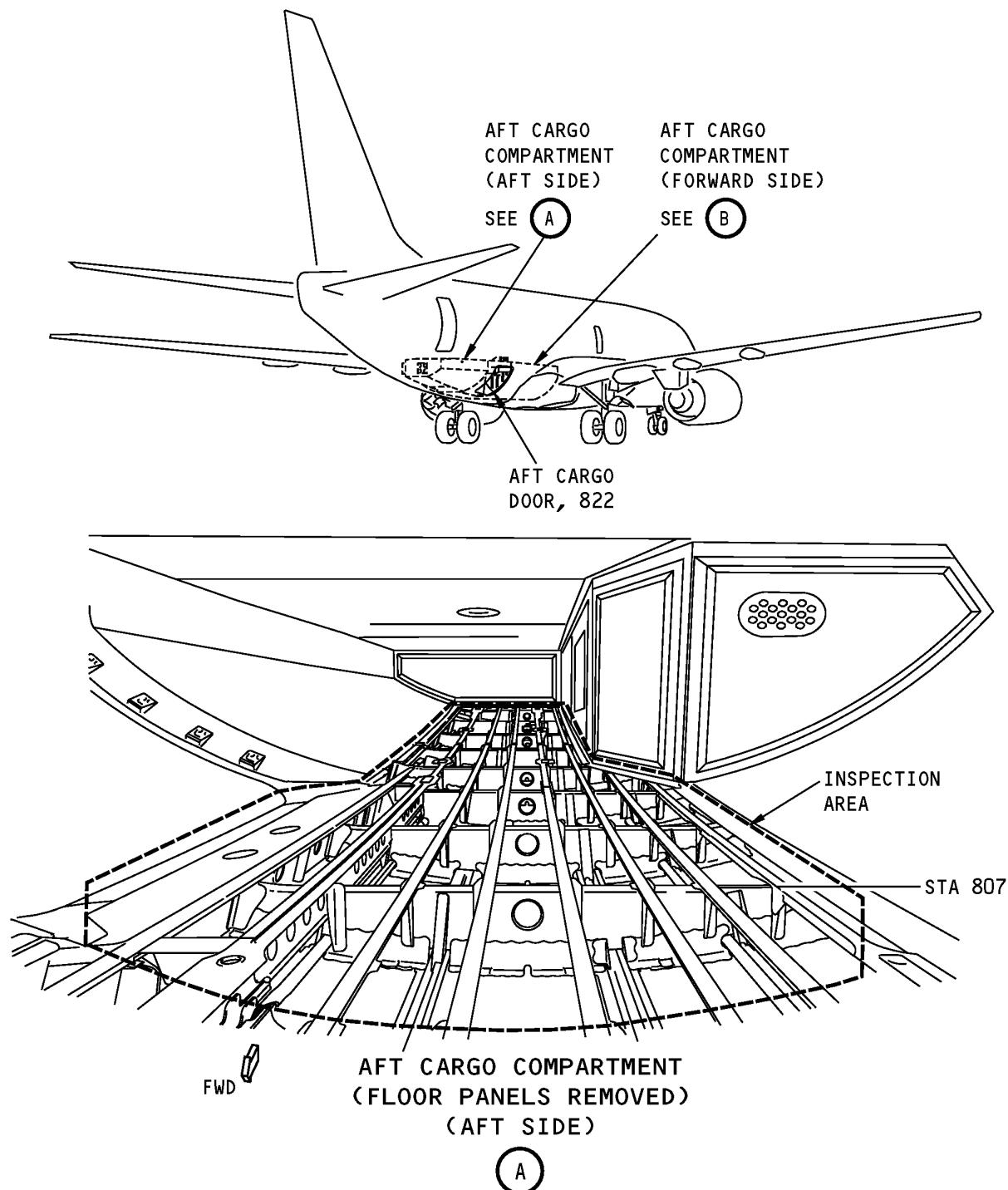
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

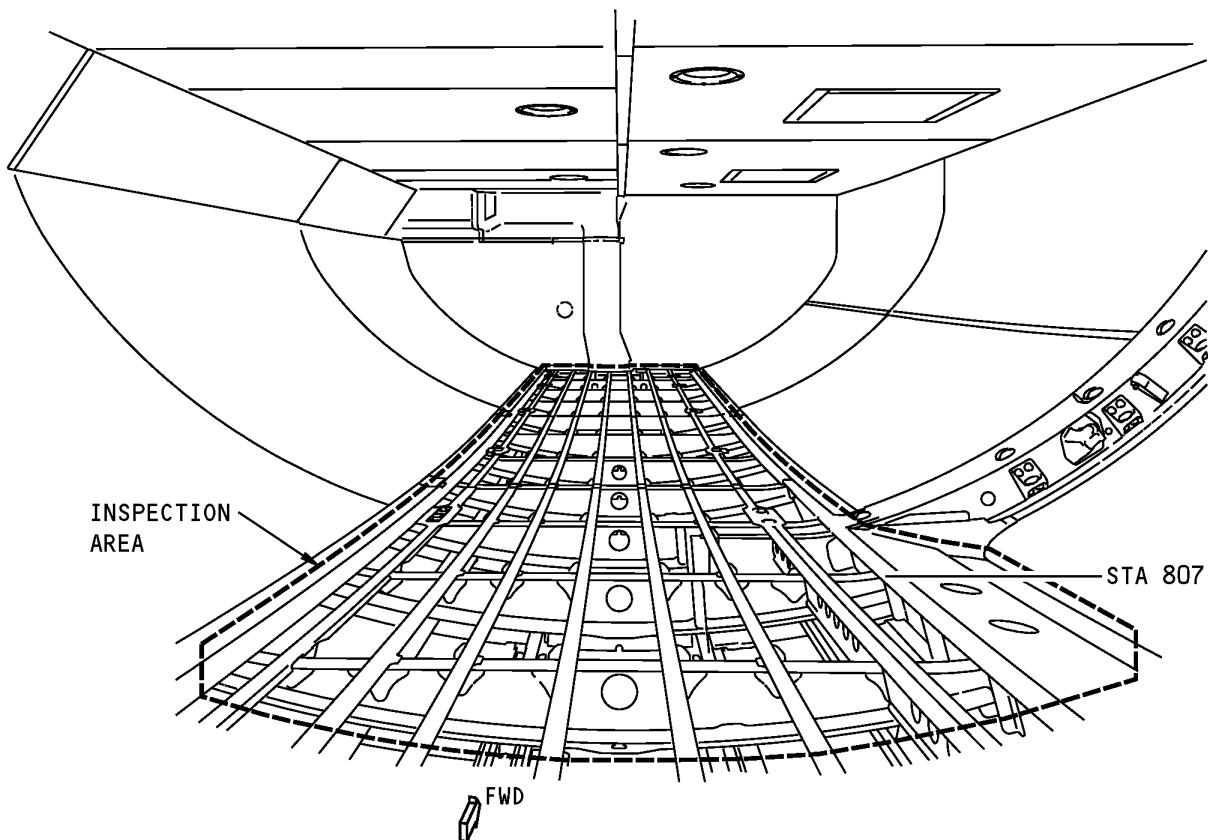
STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-140-00-02**INTERNAL - GENERAL VISUAL: AFT CARGO COMPARTMENT FLOOR STRUCTURE**
Figure 1 (Sheet 1 of 2)EFFECTIVITY
AKS ALLSOURCE
MRB**AFT CARGO COMPARTMENT - FLOOR STRUCTURE****D633A109-AKS
53-140-00-02****Page 5 of 6
Oct 15/2014**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-140-00-02



**AFT CARGO COMPARTMENT
(FLOOR PANELS REMOVED)
(FORWARD SIDE)**

B

INTERNAL - GENERAL VISUAL: AFT CARGO COMPARTMENT FLOOR STRUCTURE
Figure 1 (Sheet 2 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT - FLOOR STRUCTURE
		D633A109-AKS 53-140-00-02

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR CUTOUT			BOEING CARD NO. 53-150-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1 1.2	THRESHOLD 9 YR 24000 FC	REPEAT 6 YR 18000 FC	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL	NOTE			ENGINE ALL
		ACCESS S1221			ZONE 122
		NOTE			

Inspect forward cargo door cutout surround structure.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove door reveals. Remove sidewalls as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR CUTOUT	
		D633A109-AKS 53-150-00-01	Page 1 of 5 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-150-00-01
TASK 53-05-03-211-803				MECH INSP

1. INTERNAL - DETAILED: FORWARD CARGO DOOR CUTOUT

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-057

- (1) Open this access panel:

Number Name/Location

S1221 Forward Cargo Door Cutout Inspection

NOTE: Remove door reveals. Remove sidewalls as required. Remove/displace insulation blankets as required.

SUBTASK 53-05-03-211-003

- (2) Do a Detailed inspection of the forward cargo door cutout surround structure.

SUBTASK 53-05-03-910-018

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

SUBTASK 53-05-03-410-057

- (4) Close this access panel:

Number Name/Location

S1221 Forward Cargo Door Cutout Inspection

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR CUTOUT
		D633A109-AKS 53-150-00-01

Page 2 of 5
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-150-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR CUTOUT	
		D633A109-AKS 53-150-00-01	Page 3 of 5 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-150-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

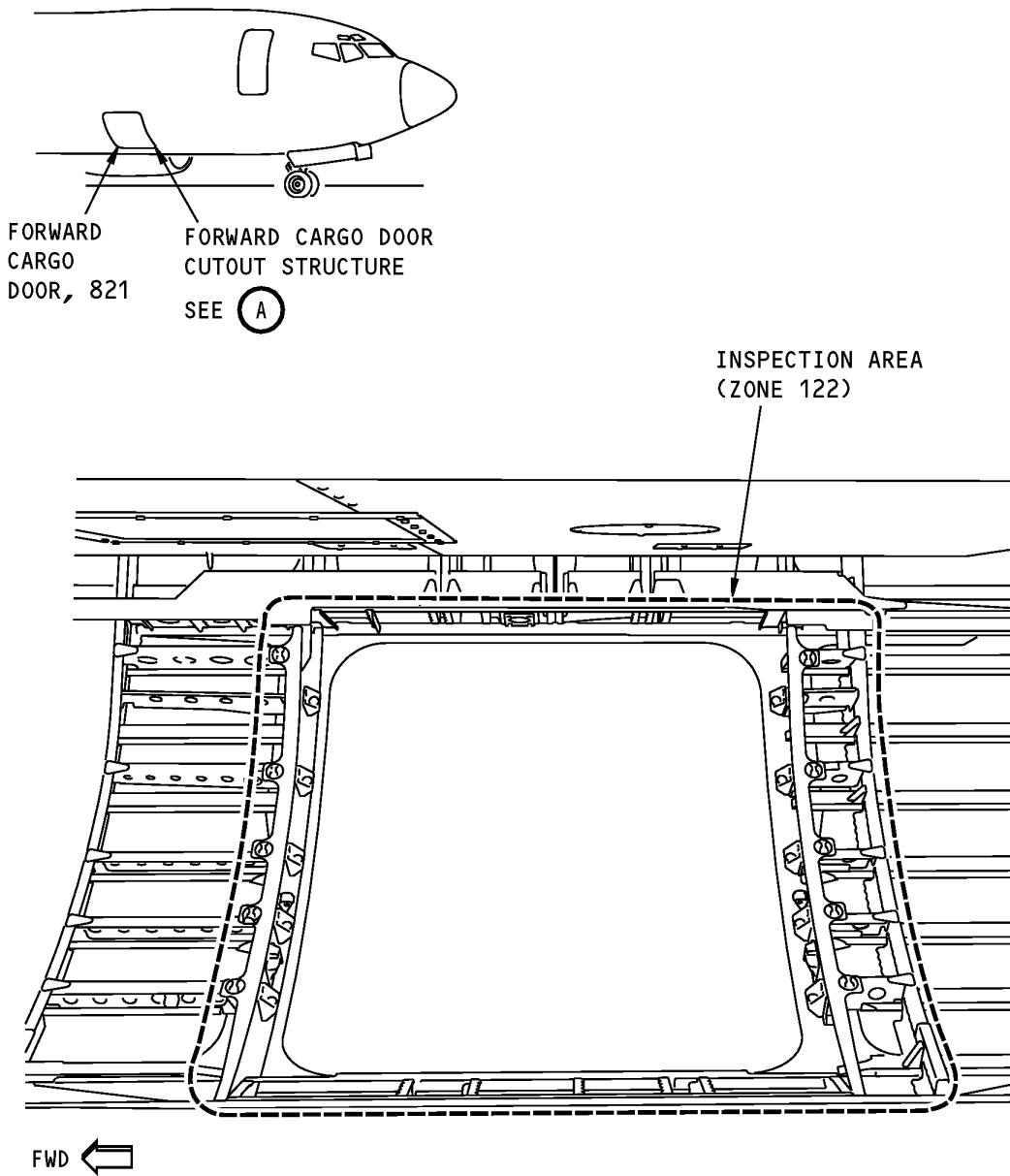
END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR CUTOUT
		D633A109-AKS 53-150-00-01

Page 4 of 5
Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-150-00-01



**FORWARD CARGO DOOR CUTOUT STRUCTURE
(DOOR REVEALS AND SIDEWALL PANELS REMOVED)**



**Forward Cargo Door Cutout Detailed (Internal)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR CUTOUT
		D633A109-AKS 53-150-00-01

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD BILGE			BOEING CARD NO. 53-160-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 8 YR 24000 FC	REPEAT 6 YR 18000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS S1202			ZONE 123 124
		NOTE			

Inspect forward bilge skin panels including skins, frames, stringers, longitudinal lap splices, and cargo door cutout surround structure in bilge (note: inspection includes the circumferential skin and stringer splice at STA 500E for the -900 models).

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove cargo floor panels and scuff plates. Remove/Displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD BILGE	
		D633A109-AKS 53-160-00-01	Page 1 of 6 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-160-00-01
				MECH INSP
TASK 53-05-03-210-816				
1. INTERNAL - GENERAL VISUAL: FORWARD BILGE				
(Figure 1)				
A. Inspection				
SUBTASK 53-05-03-010-014				
(1) Open this access panel:				
Number Name/Location				
S1202 Forward Bilge Inspection				
NOTE: Remove cargo floor panels and scuff plates. Remove/Displace insulation blankets as required.				
SUBTASK 53-05-03-210-016				
(2) Do a General Visual inspection of the forward bilge skin panels including skins, frames, stringers, longitudinal lap splices, and cargo door cutout surround structure in bilge (note: inspection includes the circumferential skin and stringer splice at STA 500E for the -900 models).				
SUBTASK 53-05-03-910-019				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.				
SUBTASK 53-05-03-410-014				
(4) Close this access panel:				
Number Name/Location				
S1202 Forward Bilge Inspection				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD BILGE	
		D633A109-AKS 53-160-00-01	Page 2 of 6 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-160-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD BILGE D633A109-AKS 53-160-00-01	Page 3 of 6 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-160-00-01				
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MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

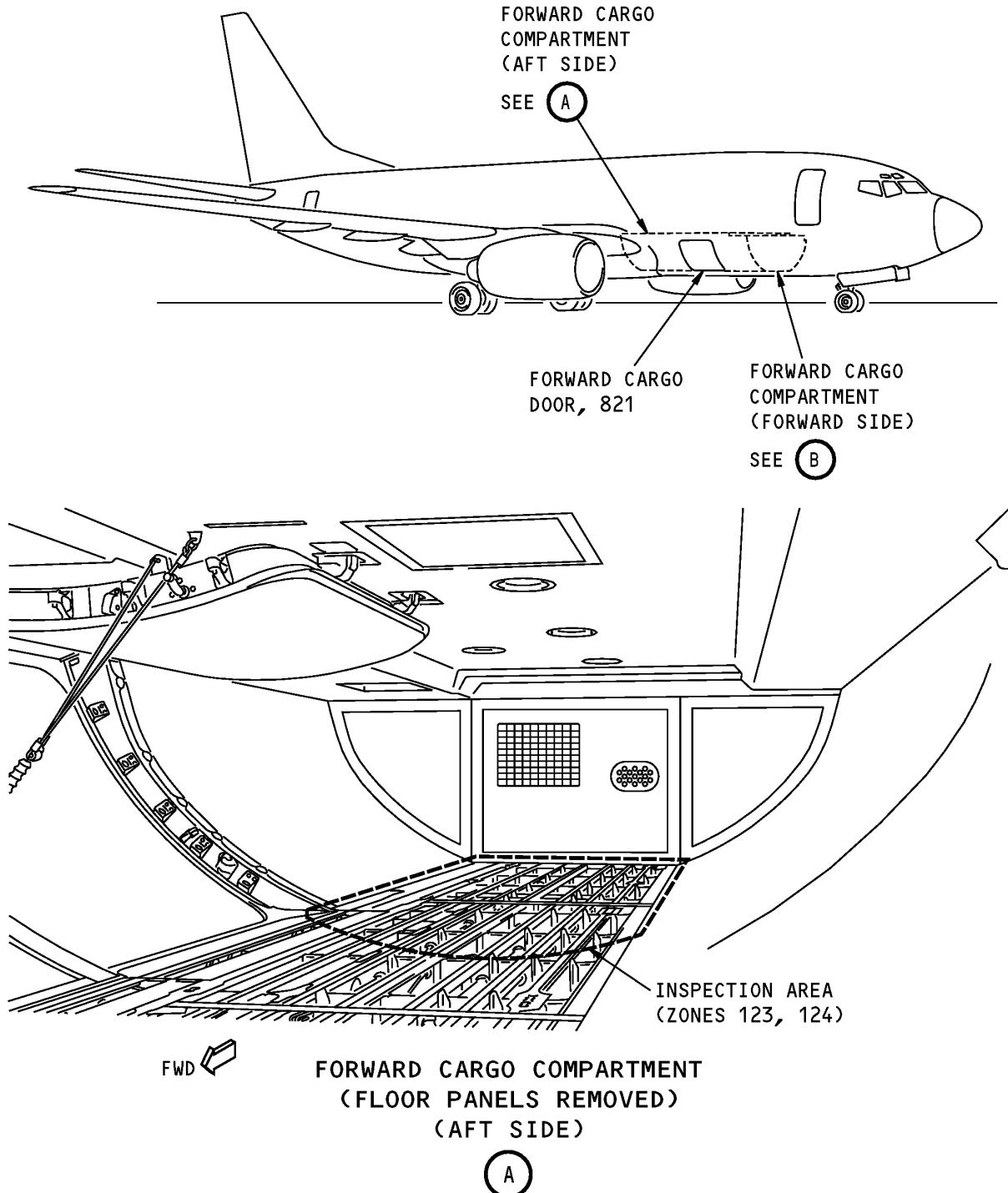
- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD BILGE D633A109-AKS 53-160-00-01	Page 4 of 6 Oct 15/2014
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-160-00-01

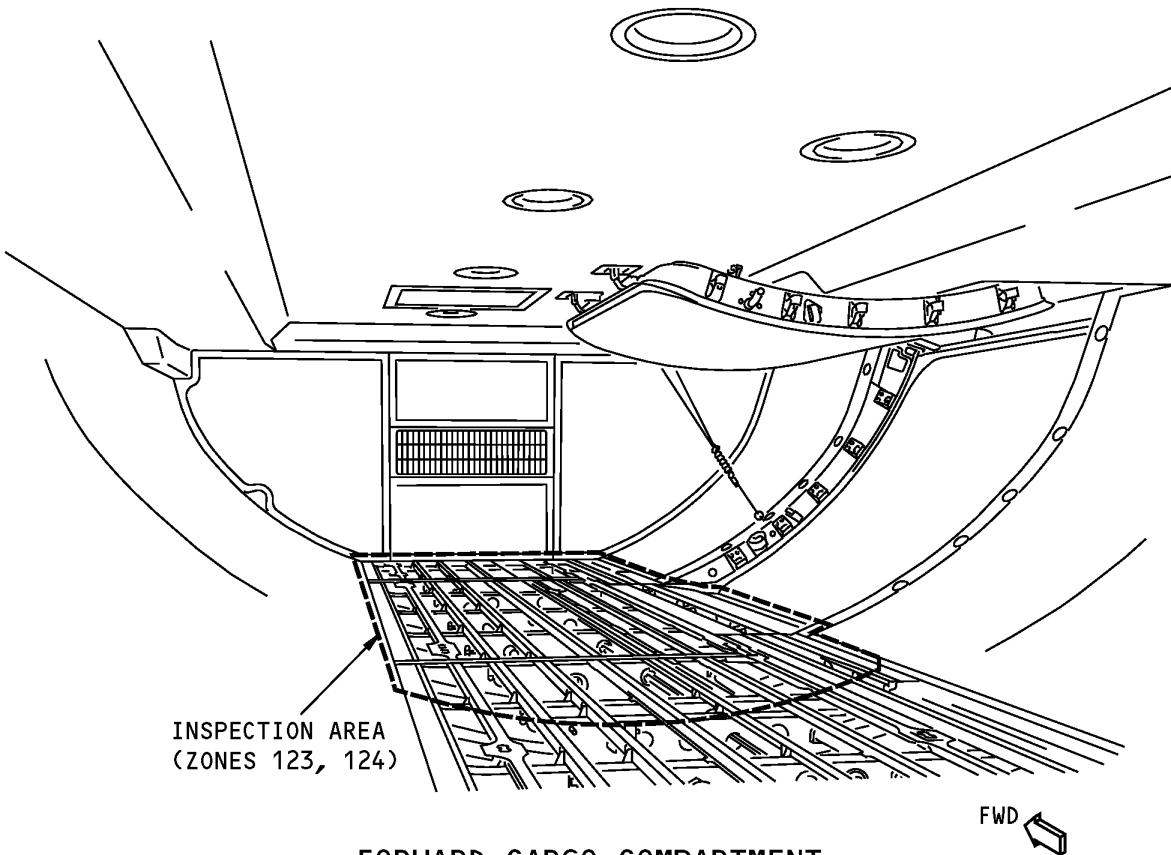


Below the Forward Cargo Comp - Forward Bilge General Visual (Int)
Figure 1 (Sheet 1 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD BILGE
		D633A109-AKS 53-160-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-160-00-01



**FORWARD CARGO COMPARTMENT
(FLOOR PANELS REMOVED)
(FORWARD SIDE)**

B

**Below the Forward Cargo Comp - Forward Bilge General Visual (Int)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD BILGE
		D633A109-AKS 53-160-00-01

**Page 6 of 6
Oct 15/2014**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA AFT OF FORWARD CARGO COMPARTMENT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-170-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 10 YR 36000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS S1203 NOTE			ZONE 125 126

Inspect area aft of forward cargo compartment, including skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices, and forward side of Sta 540 bulkhead and bulkhead splices.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove forward cargo compartment aft bulkhead panels. Remove/displace insulation blankets as required. Remove ducting as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-170-00-01

Page 1 of 7
Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-170-00-01
TASK 53-05-03-210-817				MECH INSP

1. INTERNAL - GENERAL VISUAL: AREA AFT OF FORWARD CARGO COMPARTMENT
(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-015

(1) Open this access panel:

Number Name/Location
S1203 Area Aft Of Forward Cargo Compartment Inspection

NOTE: Remove forward cargo compartment aft bulkhead panels. Remove/displace insulation blankets as required. Remove ducting as required.

SUBTASK 53-05-03-210-017

(2) Do a General Visual inspection of the area aft of forward cargo compartment, including skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices, and forward side of Sta 540 bulkhead and bulkhead splices.

SUBTASK 53-05-03-910-020

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

SUBTASK 53-05-03-410-015

(4) Close this access panel:

Number Name/Location
S1203 Area Aft Of Forward Cargo Compartment Inspection

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF FORWARD CARGO COMPARTMENT	
		D633A109-AKS 53-170-00-01	Page 2 of 7 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-170-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-170-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-170-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF FORWARD CARGO COMPARTMENT	
		D633A109-AKS 53-170-00-01	Page 4 of 7 Oct 15/2014

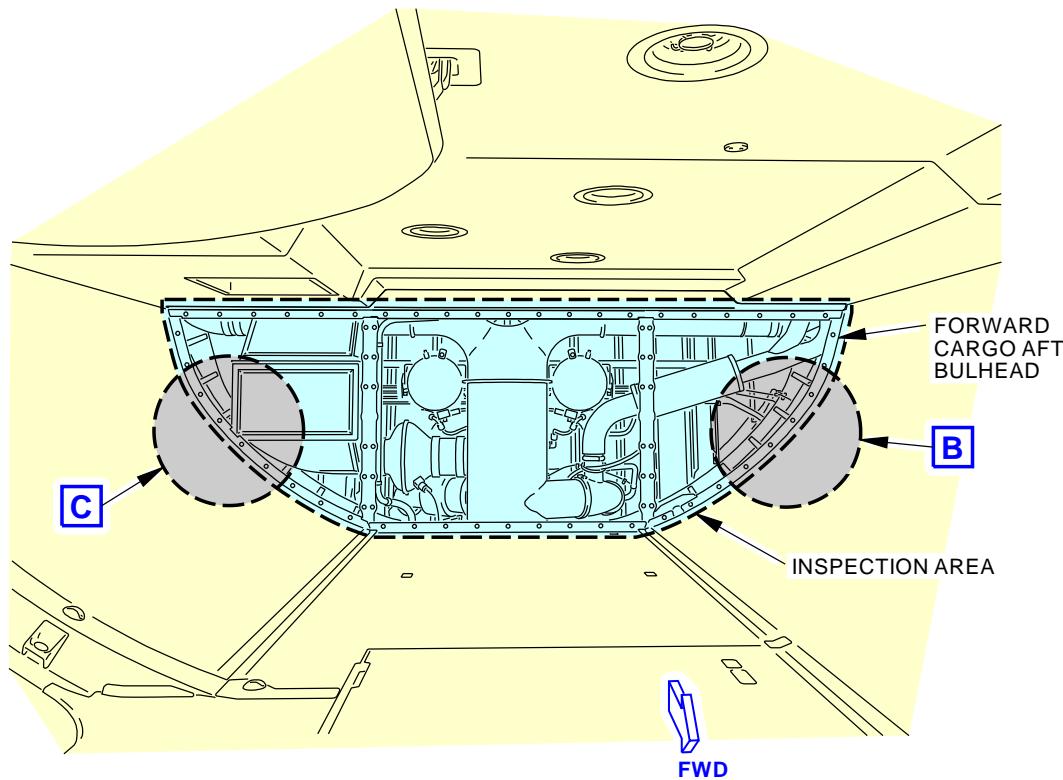
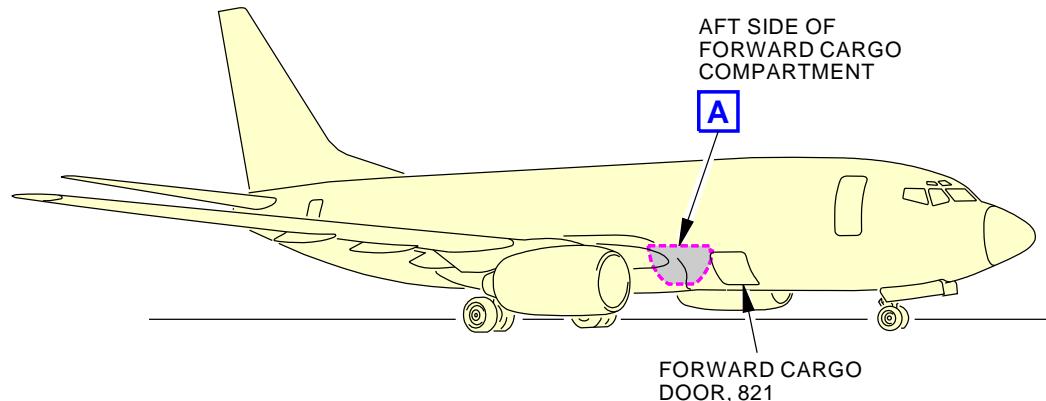
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-170-00-01MPD ITEM
53-170-00

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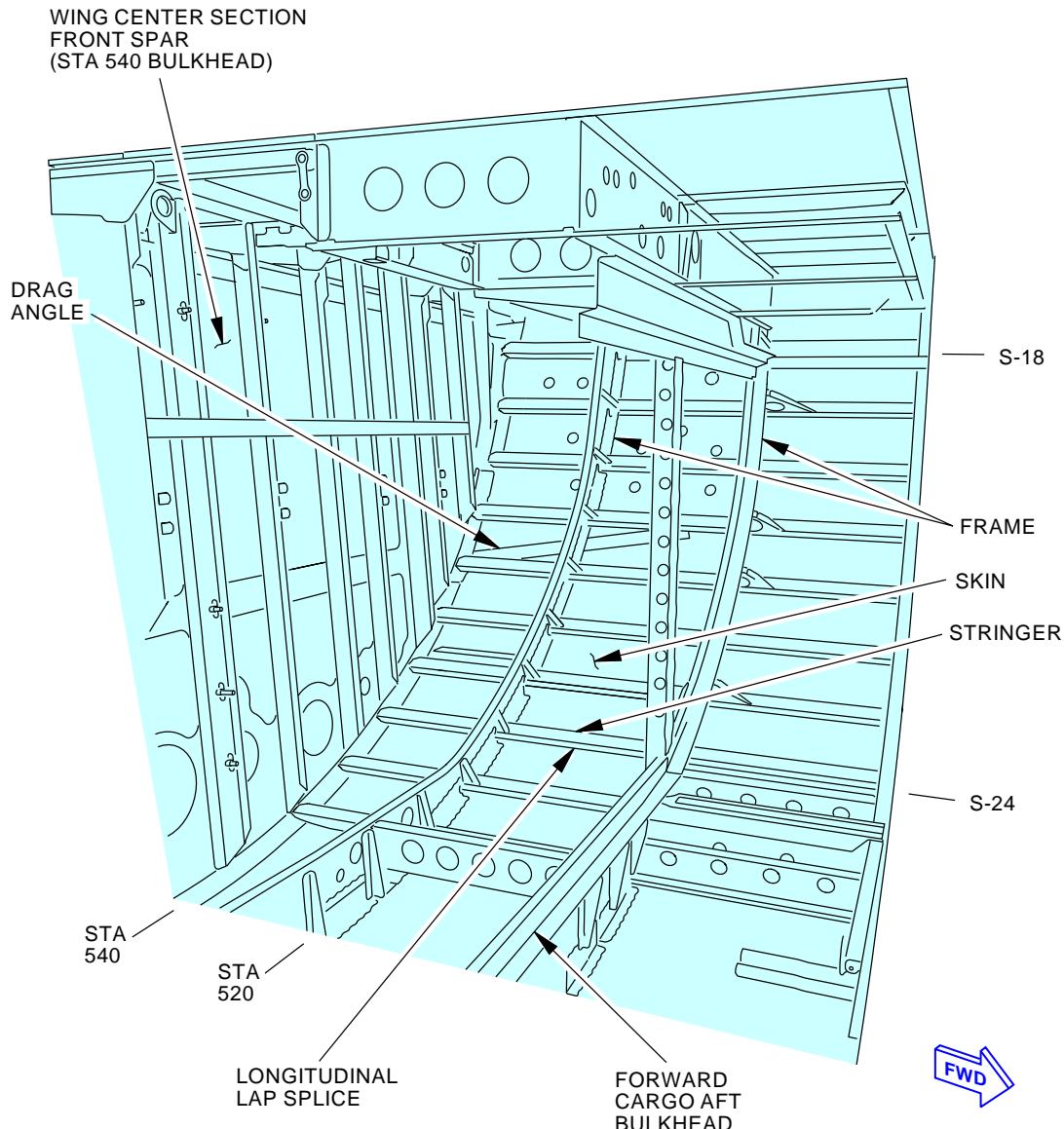
INTERNAL-GENERAL VISUAL: AREA AFT OF FORWARD CARGO COMPARTMENT
Figure 1 (Sheet 1 of 3)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-170-00-01

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Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-170-00-01
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**AREA AFT OF FORWARD CARGO
(LEFT SIDE VIEW)**MPD ITEM
53-170-00

2079756 S0000435862_V2

INTERNAL-GENERAL VISUAL: AREA AFT OF FORWARD CARGO COMPARTMENT
Figure 1 (Sheet 2 of 3)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-170-00-01

Page 6 of 7
Oct 15/2015

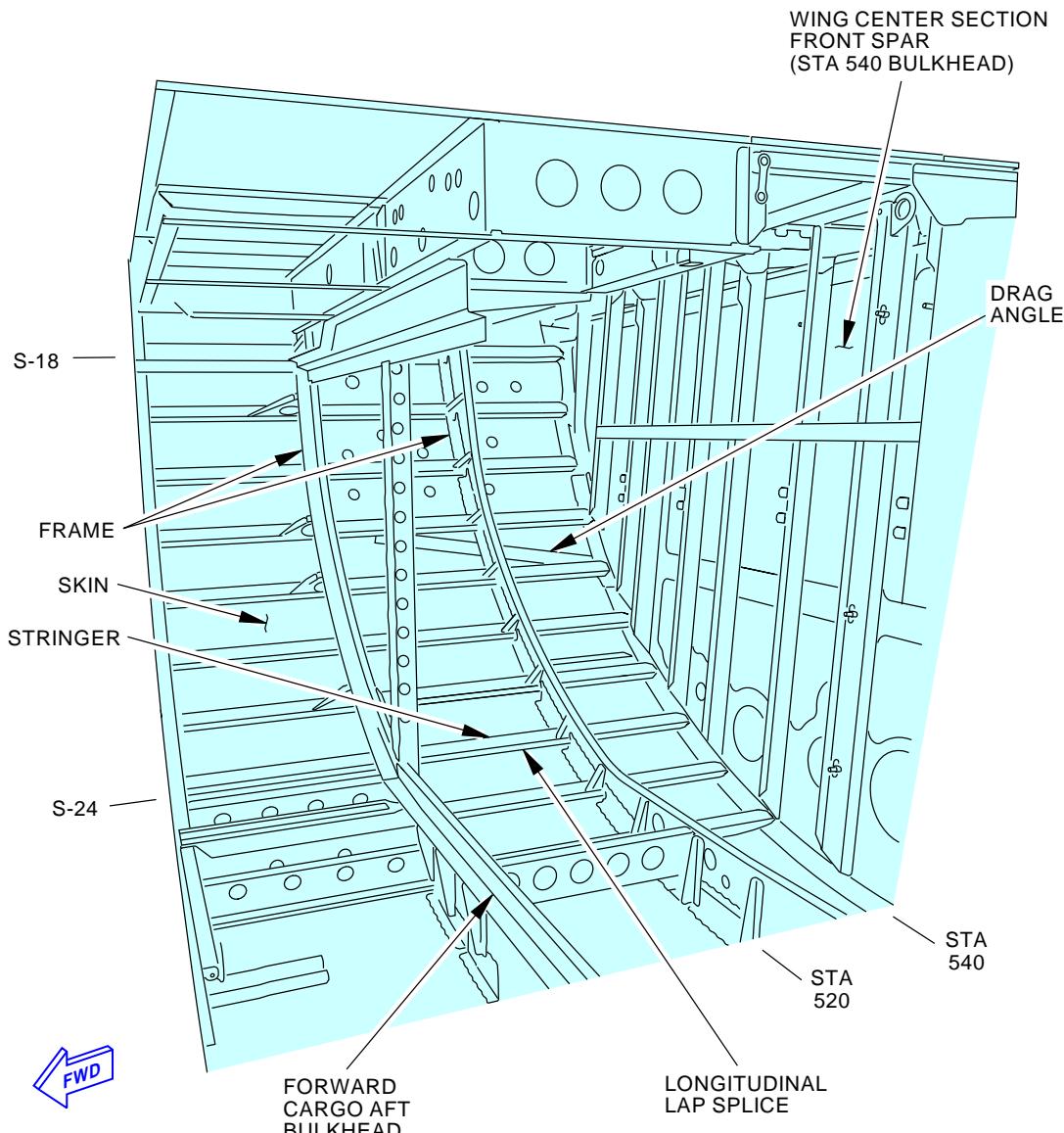
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-170-00-01MPD ITEM
53-170-00

2080303 S0000435861_V2

INTERNAL-GENERAL VISUAL: AREA AFT OF FORWARD CARGO COMPARTMENT
Figure 1 (Sheet 3 of 3)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-170-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA UNDER LOWER WING-TO-BODY FAIRING (FORWARD OF WING BOX)			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-180-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 191AL 191AR 191BL 191BR 191CL 191CR 191D 191FL 191FR 191GL 191GR 191HL 191HR			ZONE 129 191

Inspect area under lower wing-to-body fairing (forward of wing box), including skin panels, longitudinal lap splices, keel beam extension, wing-to-body drag angles, and Sta 540 bulkhead.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (FORWARD OF WING BOX)
		D633A109-AKS 53-180-00-01

Page 1 of 7
Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-180-00-01																												
				MECH INSP																												
TASK 53-05-03-210-818																																
1. INTERNAL - GENERAL VISUAL: AREA UNDER LOWER WING-TO-BODY FAIRING (Forward Of Wing Box) (Figure 1)																																
A. Inspection																																
SUBTASK 53-05-03-010-016																																
(1) Open these access panels:																																
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>191AL</td><td>Forward Wing To Body Fairing Panel - Upper</td></tr><tr><td>191AR</td><td>Forward Wing To Body Fairing Panel - Upper</td></tr><tr><td>191BL</td><td>Forward Wing To Body Fairing Panel, Ram Air Inlet</td></tr><tr><td>191BR</td><td>Forward Wing To Body Fairing Panel, Ram Air Inlet</td></tr><tr><td>191CL</td><td>Forward Wing To Body Fairing Panel - Middle</td></tr><tr><td>191CR</td><td>Forward Wing To Body Fairing Panel - Middle</td></tr><tr><td>191D</td><td>Forward Wing To Body Fairing Panel - Lower</td></tr><tr><td>191FL</td><td>Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet</td></tr><tr><td>191FR</td><td>Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet</td></tr><tr><td>191GL</td><td>Ram Air Actuator Panel - Forward</td></tr><tr><td>191GR</td><td>Ram Air Actuator Panel - Forward</td></tr><tr><td>191HL</td><td>Ram Air Inlet Lip Panel - Forward</td></tr><tr><td>191HR</td><td>Ram Air Inlet Lip Panel - Forward</td></tr></tbody></table>				Number	Name/Location	191AL	Forward Wing To Body Fairing Panel - Upper	191AR	Forward Wing To Body Fairing Panel - Upper	191BL	Forward Wing To Body Fairing Panel, Ram Air Inlet	191BR	Forward Wing To Body Fairing Panel, Ram Air Inlet	191CL	Forward Wing To Body Fairing Panel - Middle	191CR	Forward Wing To Body Fairing Panel - Middle	191D	Forward Wing To Body Fairing Panel - Lower	191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet	191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet	191GL	Ram Air Actuator Panel - Forward	191GR	Ram Air Actuator Panel - Forward	191HL	Ram Air Inlet Lip Panel - Forward	191HR	Ram Air Inlet Lip Panel - Forward	
Number	Name/Location																															
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191GR	Ram Air Actuator Panel - Forward																															
191HL	Ram Air Inlet Lip Panel - Forward																															
191HR	Ram Air Inlet Lip Panel - Forward																															
SUBTASK 53-05-03-210-018																																
(2) Do a General Visual inspection of the area under lower wing-to-body fairing (forward of wing box), including skin panels, longitudinal lap splices, keel beam extension, wing-to-body drag angles, and Sta 540 bulkhead.																																
SUBTASK 53-05-03-910-021																																
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.																																
SUBTASK 53-05-03-410-016																																
(4) Close these access panels:																																
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191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet																															
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet																															
EFFECTIVITY AKS ALL		SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (FORWARD OF WING BOX)																													
			D633A109-AKS 53-180-00-01																													
				Page 2 of 7 Jun 15/2016																												

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-180-00-01
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(Continued)

Number Name/Location

- | | |
|-------|-----------------------------------|
| 191GL | Ram Air Actuator Panel - Forward |
| 191GR | Ram Air Actuator Panel - Forward |
| 191HL | Ram Air Inlet Lip Panel - Forward |
| 191HR | Ram Air Inlet Lip Panel - Forward |

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (FORWARD OF WING BOX)
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**D633A109-AKS
53-180-00-01****Page 3 of 7
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-180-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (FORWARD OF WING BOX) D633A109-AKS 53-180-00-01	Page 4 of 7 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-180-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.					

SUBTASK 51-05-01-210-063

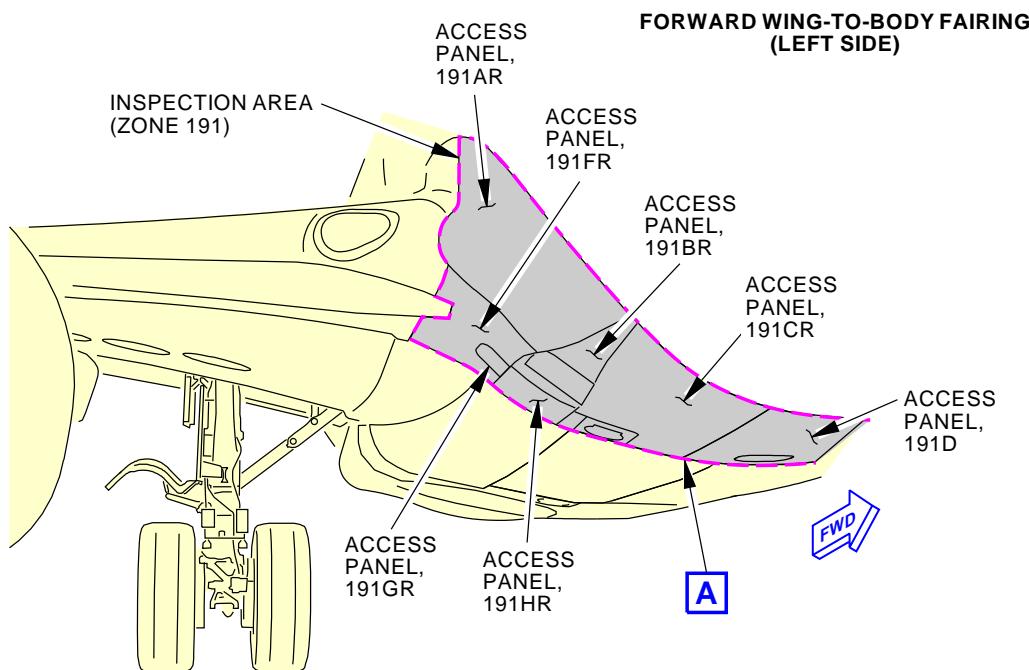
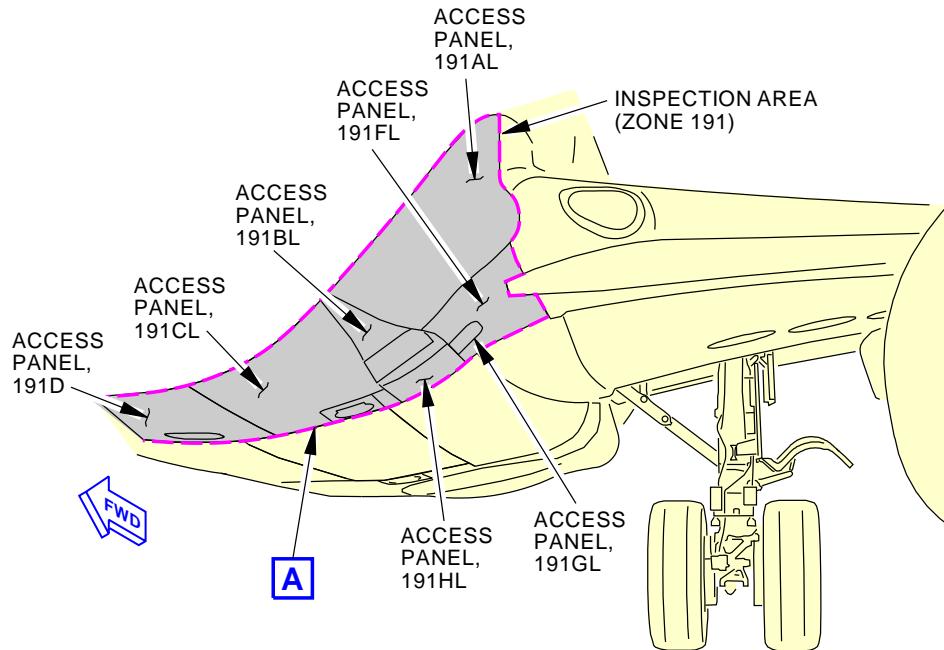
(7) CPCP Basic Task Item 7 is not applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (FORWARD OF WING BOX)	
		D633A109-AKS 53-180-00-01	Page 5 of 7 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-180-00-01

**FORWARD WING-TO-BODY FAIRING
(RIGHT SIDE)**MPD ITEM
53-180-00

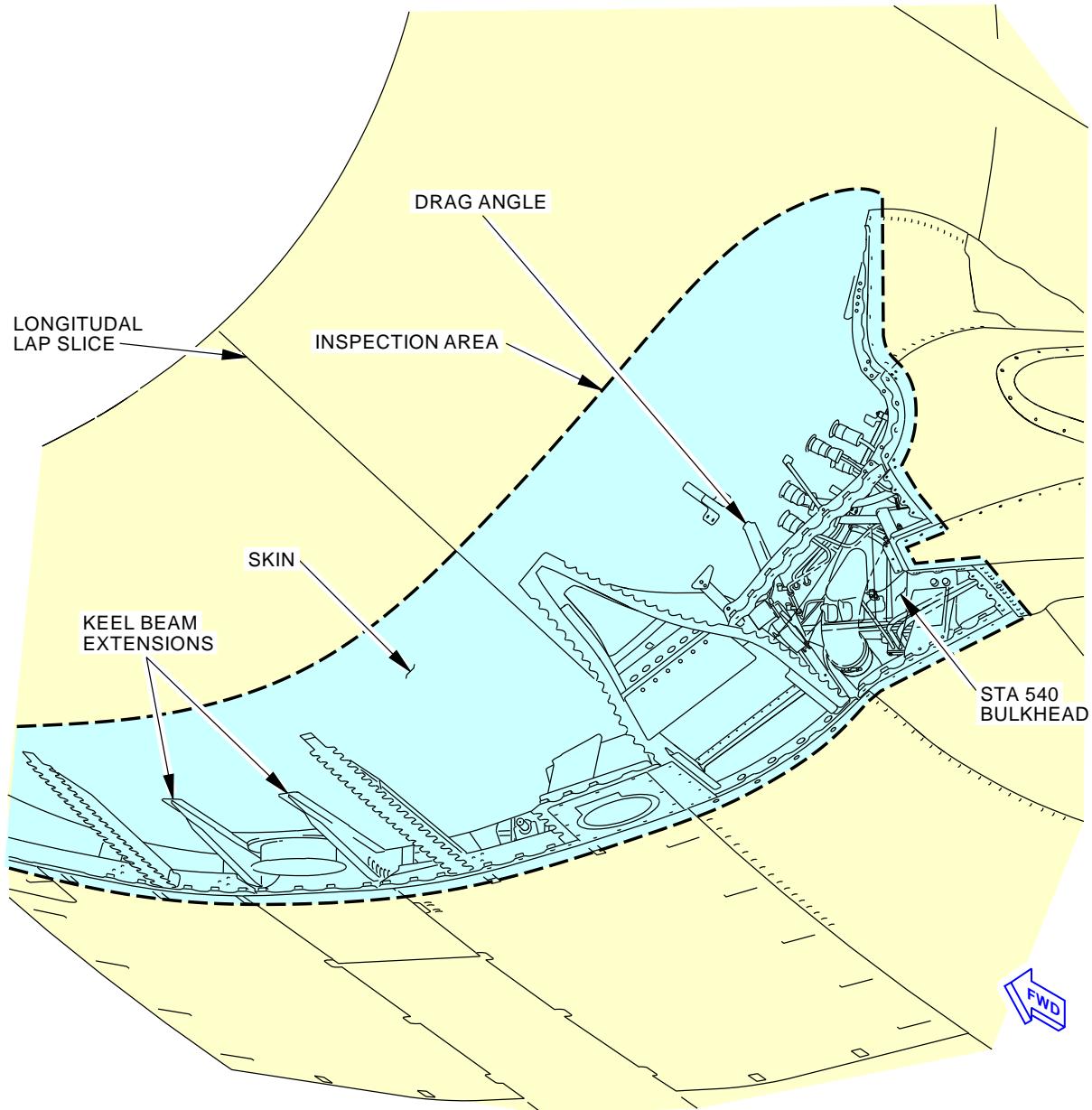
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INTERNAL-GENERAL VISUAL: AREA UNDER LOWER WING-TO-BODY FAIRING (Forward Of Wing Box)
Figure 1 (Sheet 1 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (FORWARD OF WING BOX)
		D633A109-AKS 53-180-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-180-00-01
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**A**MPD ITEM
53-180-00

2078266 S0000435773_V2

INTERNAL-GENERAL VISUAL: AREA UNDER LOWER WING-TO-BODY FAIRING (Forward Of Wing Box)
Figure 1 (Sheet 2 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (FORWARD OF WING BOX)
		D633A109-AKS 53-180-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA ABOVE WING BOX CENTER SECTION			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-190-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 10 YR 36000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS S1301			ZONE 135 136
		NOTE			

Inspect fuselage lower lobe above wing box center section upper panel, including side skin panels (skins, frames and stringers), Sta 540 bulkhead, and overwing frames and stub beams.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove floor panels. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE WING BOX CENTER SECTION	
		D633A109-AKS 53-190-00-01	Page 1 of 6 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-190-00-01
				MECH INSP
TASK 53-05-03-210-819				
1. INTERNAL - GENERAL VISUAL: AREA ABOVE WING BOX CENTER SECTION				
(Figure 1)				
A. Inspection				
SUBTASK 53-05-03-010-076				
(1) Special Access:				
Number Name/Location				
S1301 Area Above Wing Box Center Section Inspection				
NOTE: Remove floor panels. Remove/displace insulation blankets as required.				
SUBTASK 53-05-03-210-019				
(2) Do a General Visual inspection of the fuselage lower lobe above wing box center section upper panel, including side skin panels (skins, frames and stringers), Sta 540 bulkhead, and overwing frames and stub beams.				
SUBTASK 53-05-03-910-022				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.				
———— END OF TASK ——				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE WING BOX CENTER SECTION	
		D633A109-AKS 53-190-00-01	Page 2 of 6 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-190-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE WING BOX CENTER SECTION	
		D633A109-AKS 53-190-00-01	Page 3 of 6 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-190-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-077

(7) Do the CPCP Basic Task Item 7 as follows:

(a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE WING BOX CENTER SECTION
		D633A109-AKS 53-190-00-01

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Oct 15/2014

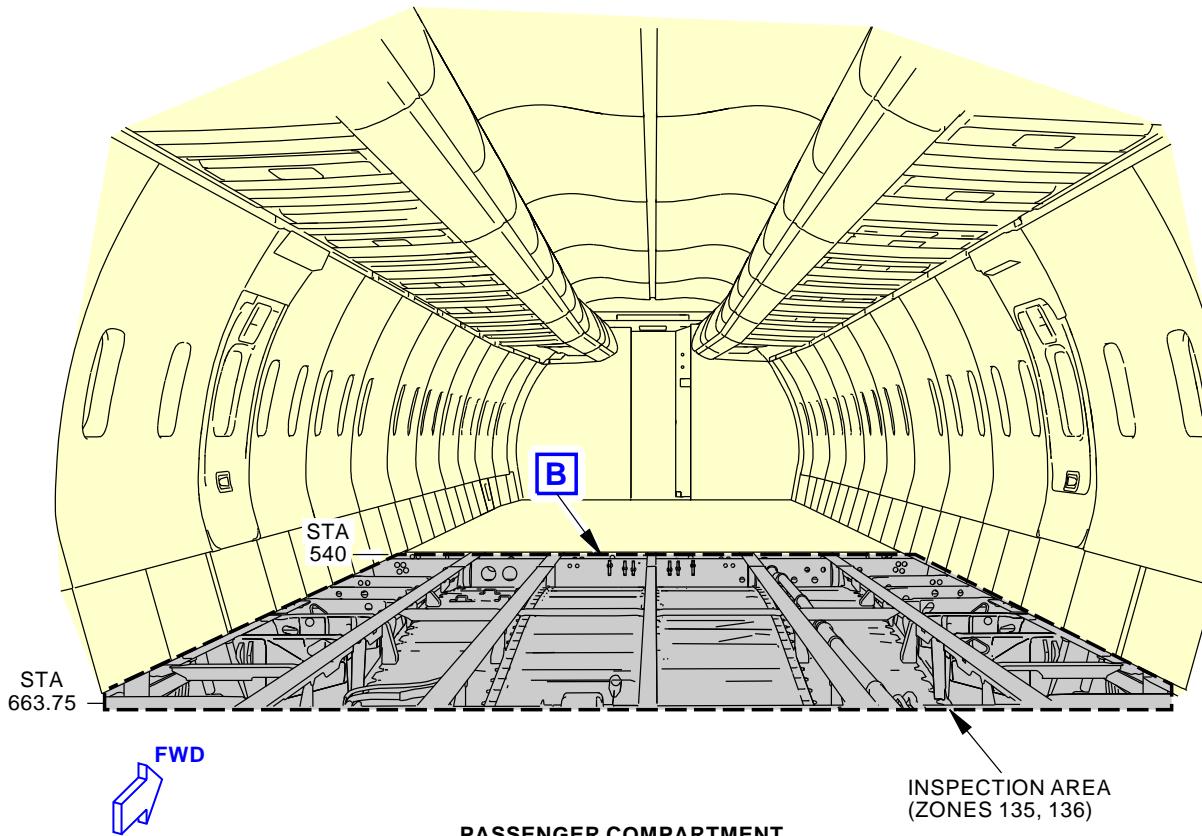
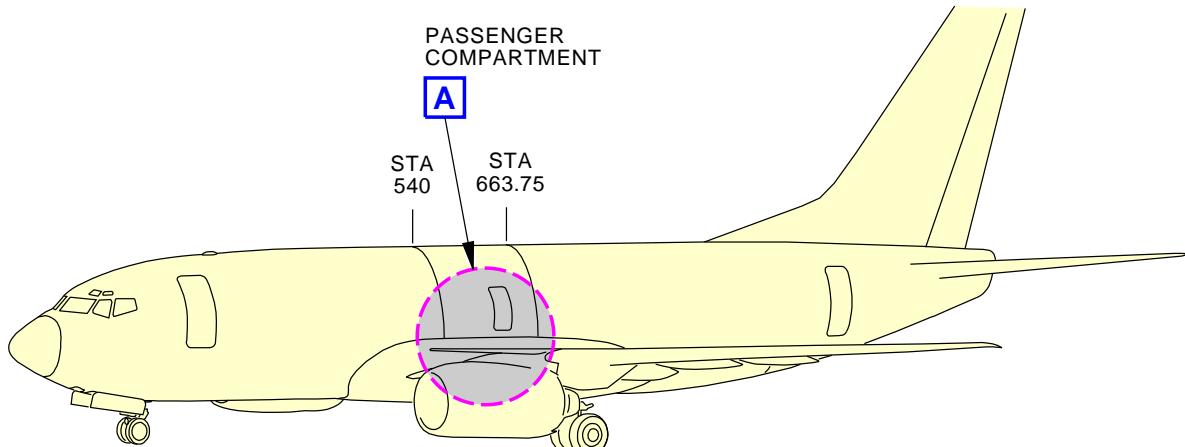
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-190-00-01MPD ITEM
53-190-00

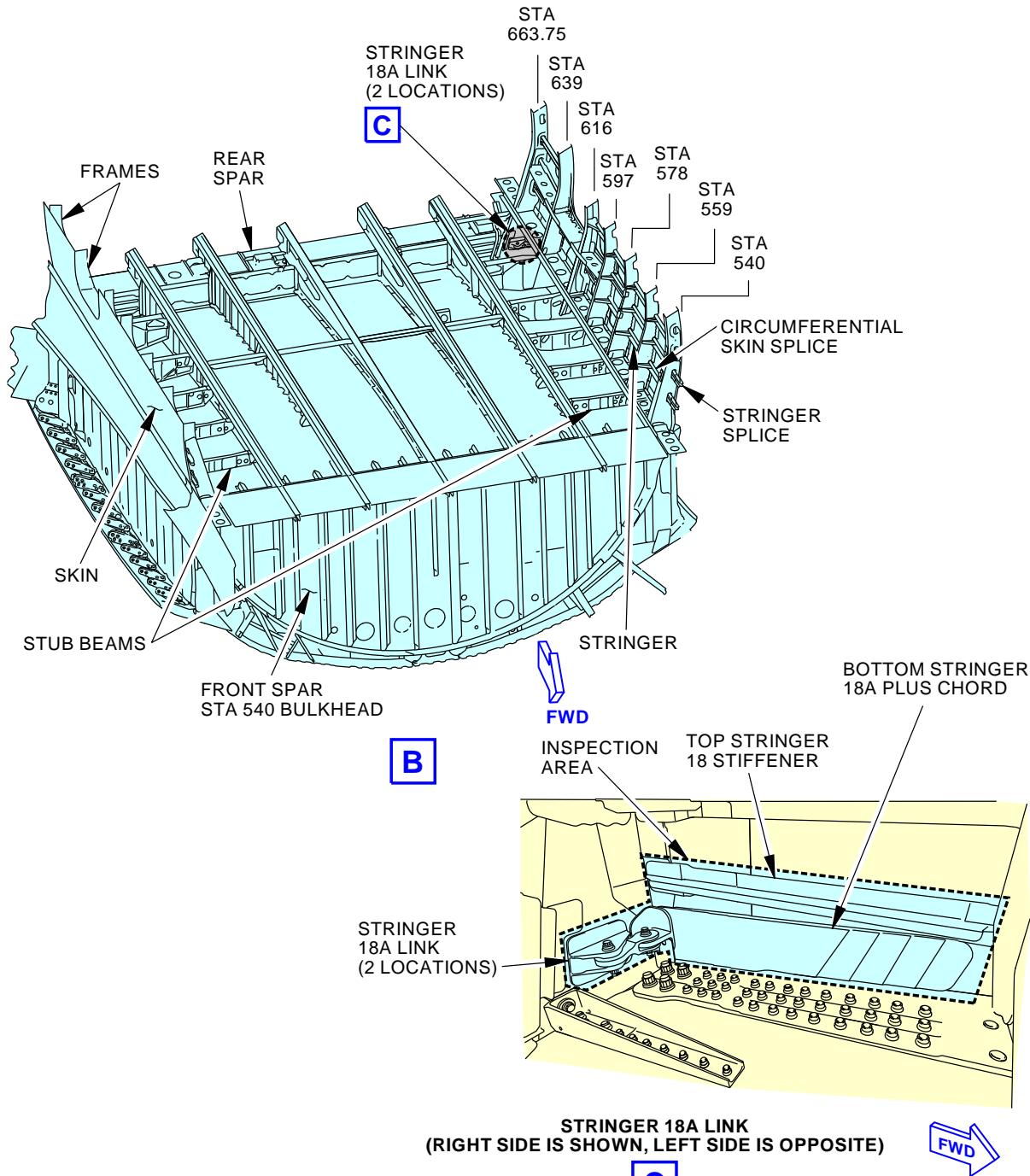
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INTERNAL-GENERAL VISUAL: AREA ABOVE WING BOX CENTER SECTION
Figure 1 (Sheet 1 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE WING BOX CENTER SECTION
		D633A109-AKS 53-190-00-01

AKS737-600/700/800/900
TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-190-00-01

MPD ITEM
53-190-00

2077198 S0000435834_V3

INTERNAL-GENERAL VISUAL: AREA ABOVE WING BOX CENTER SECTION
Figure 1 (Sheet 2 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE WING BOX CENTER SECTION
		D633A109-AKS 53-190-00-01

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Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA ABOVE MAIN LANDING GEAR WHEEL WELL			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-200-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 10 YR 36000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS S1302 NOTE			ZONE 137 138

Inspect fuselage lower lobe above main landing gear wheel well, including:

1. Pressure deck web to stiffeners, stiffener attachment to floor beam at STA 727; 2. Side skin panels, circumferential skin and stringer splice; 3. Bulkheads at STA 663 and 727; 4. Side strut support frame at STA 706; 5. Main landing gear support frame at STA 695 and 716; 6. Wheel well frame at STA 685.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove floor panels. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-200-00-01

Page 1 of 6
Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-200-00-01
TASK 53-05-03-210-820				MECH INSP

1. INTERNAL - GENERAL VISUAL: AREA ABOVE MAIN LANDING GEAR WHEEL WELL
(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-018

(1) Open this access panel:

Number Name/Location
S1302 Area Above Main Landing Gear Wheel Well Inspection

NOTE: Remove floor panels. Remove/displace insulation blankets as required.

SUBTASK 53-05-03-210-020

(2) Do a General Visual inspection of the fuselage lower lobe above main landing gear wheel well, including:

1. Pressure deck web to stiffeners, stiffener attachment to floor beam at STA 727.
2. Side skin panels, circumferential skin and stringer splice.
3. Bulkheads at STA 663 and 727.
4. Side strut support frame at STA 706.
5. Main landing gear support frame at STA 695 and 716.
6. Wheel well frame at STA 685.

SUBTASK 53-05-03-910-023

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

SUBTASK 53-05-03-410-018

(4) Close this access panel:

Number Name/Location
S1302 Area Above Main Landing Gear Wheel Well Inspection

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE MAIN LANDING GEAR WHEEL WELL	
		D633A109-AKS 53-200-00-01	Page 2 of 6 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-200-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE MAIN LANDING GEAR WHEEL WELL	
		D633A109-AKS 53-200-00-01	Page 3 of 6 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-200-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

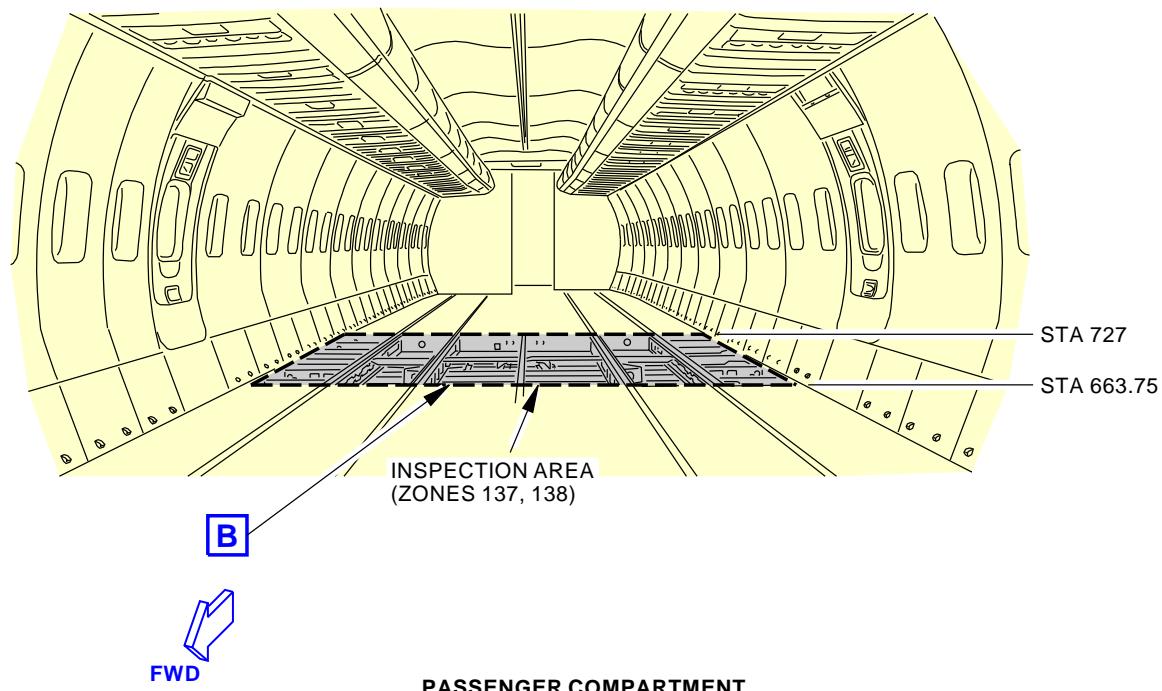
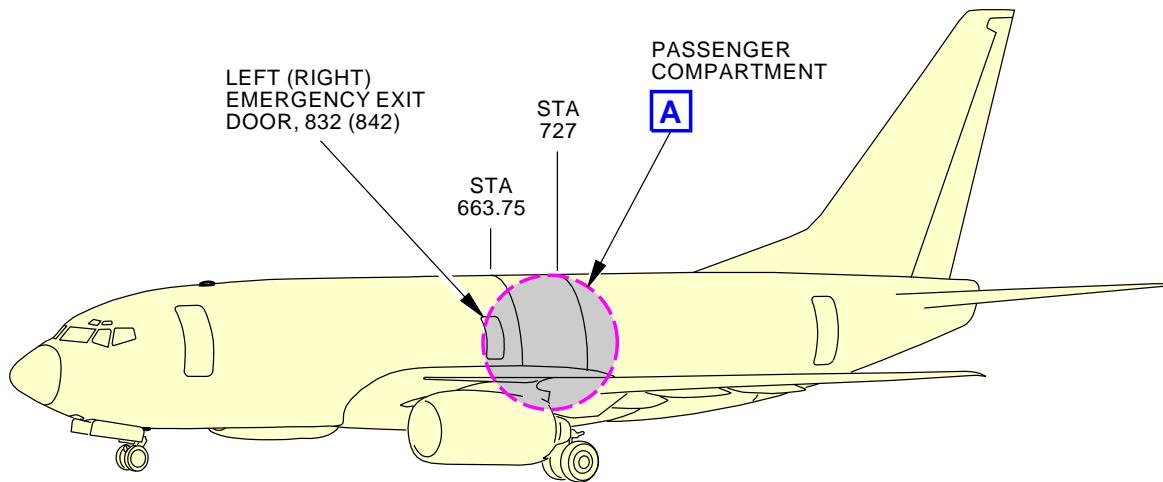
- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE MAIN LANDING GEAR WHEEL WELL	
		D633A109-AKS 53-200-00-01	Page 4 of 6 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-200-00-01

MPD ITEM
53-200-00

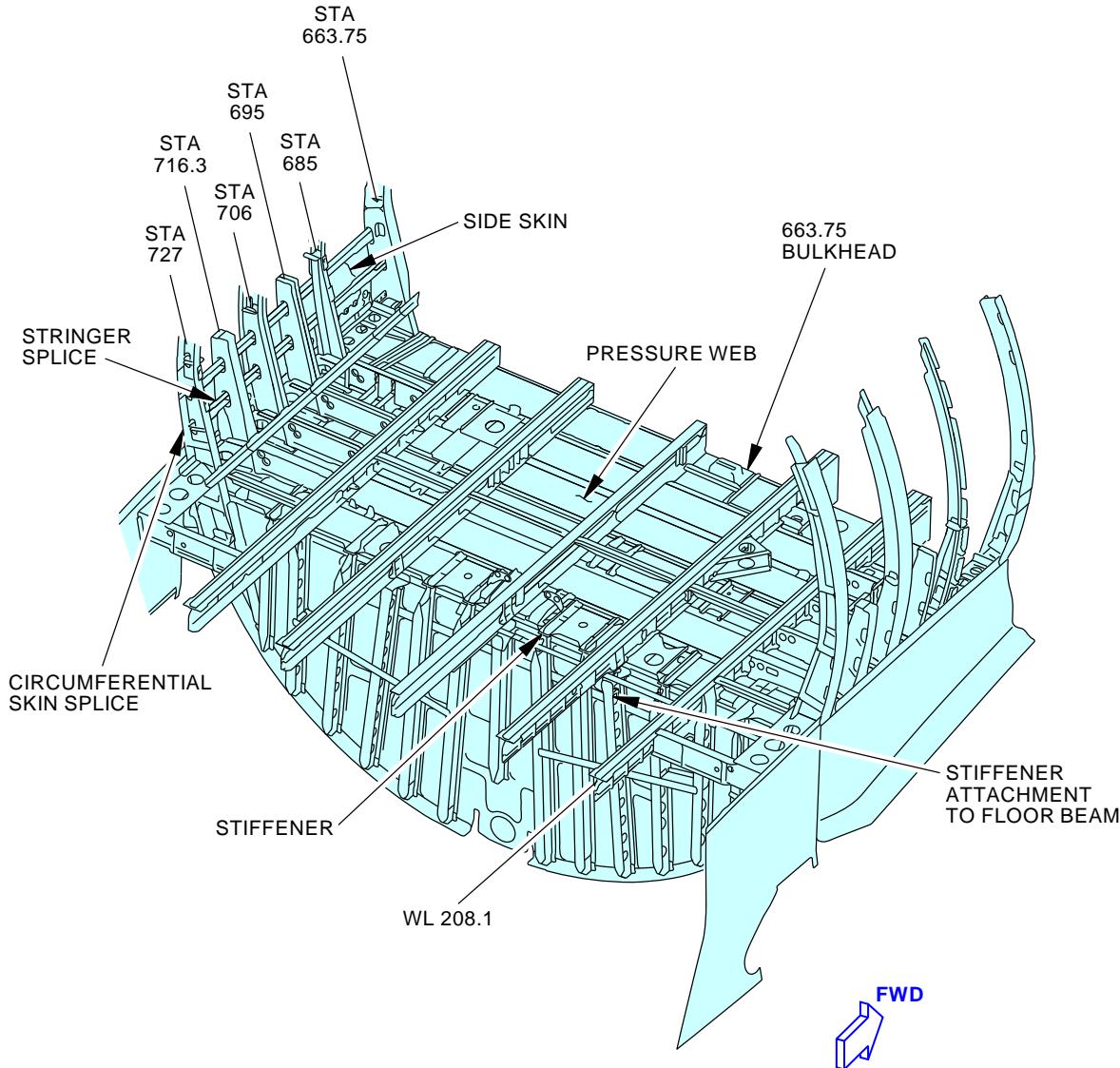
2068994 S0000429080_V2

**Above and Outboard of the Main Landing Gear Wheel Well General Visual (Internal)
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-200-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-200-00-01
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(TOP OF PRESSURE DECK)

BMPD ITEM
53-200-00

2070775 S0000429084_V2

**Above and Outboard of the Main Landing Gear Wheel Well General Visual (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-200-00-01

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Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE KEEL BEAM UNDER WING-TO-BODY FAIRING (UNDER WING BOX)			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-210-00-01
TAIL NUMBER	WORK AREA KEEL BEAM	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC	AIRPLANE ALL ENGINE ALL
		ACCESS 192CL 192CR 192E 192F S1004			ZONE 139 192
		NOTE			

Inspect keel beam under wing-to-body fairing (under wing box, Sta 540 to 663.75), including keel beam chords, webs, stiffeners, splice, keel beam/rear spar attachment angles.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Option 1: Remove center wing-to-body fairing (192CL, 192CR, 192E, 192F) and open AC bay access door. Option 2: Open AC bay door and remove AC pack to gain access to access holes.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM UNDER WING-TO-BODY FAIRING (UNDER WING BOX)	D633A109-AKS 53-210-00-01	Page 1 of 7 Jun 15/2015
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-210-00-01
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TASK 53-05-03-210-821

- ## 1. INTERNAL - GENERAL VISUAL: KEEL BEAM UNDER WING-TO-BODY FAIRING (under wing box)

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-071

- (1) Open these access panels:

<u>Number</u>	<u>Name/Location</u>
192CL	ECS Access Door
192CR	ECS Access Door
192E	ECS Under Keel Panel - Forward
192F	ECS Under Keel Panel - Middle
S1004	Keel Beam Under Wing-to-Body Fairing (Under Wing Box) Inspection

NOTE: Option 1: Remove center wing-to-body fairing (192CL, 192CR, 192E, 192F) and open AC bay access door.

Option 2: Open AC bay door and remove AC pack to gain access to access holes.

SUBTASK 53-05-03-210-021

- (2) Do a General Visual inspection of the keel beam under wing-to-body fairing (under wing box, Sta 540 to 663.75), including keel beam chords, webs, stiffeners, splice, keel beam/rear spar attachment angles.

SUBTASK 53-05-03-910-026

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-804.

SUBTASK 53-05-03-410-071

- (4) Close these access panels:

<u>Number</u>	<u>Name/Location</u>
192CL	ECS Access Door
192CR	ECS Access Door
192E	ECS Under Keel Panel - Forward
192F	ECS Under Keel Panel - Middle
S1004	Keel Beam Under Wing-to-Body Fairing (Under Wing Box) Inspection

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM UNDER WING-TO-BODY FAIRING (UNDER WING BOX)
		D633A109-AKS 53-210-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-210-00-01
				MECH INSP
TASK 51-05-01-210-804				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-043				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-044				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-045				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-046				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-111				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-048				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM UNDER WING-TO-BODY FAIRING (UNDER WING BOX)	D633A109-AKS 53-210-00-01	Page 3 of 7 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-210-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM UNDER WING-TO-BODY FAIRING (UNDER WING BOX)
		D633A109-AKS 53-210-00-01

Page 4 of 7
Oct 15/2014

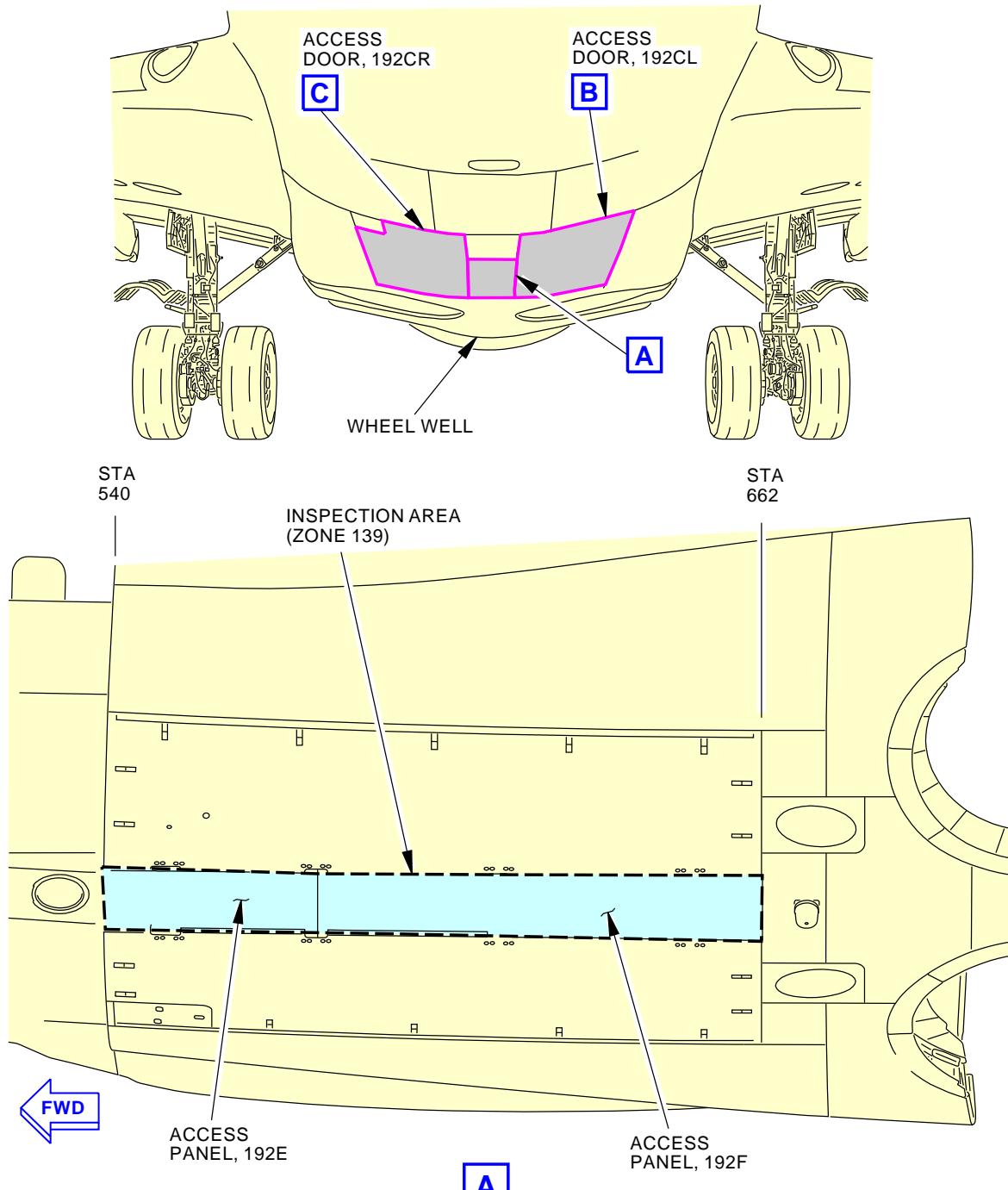
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-210-00-01MPD ITEM
53-210-00

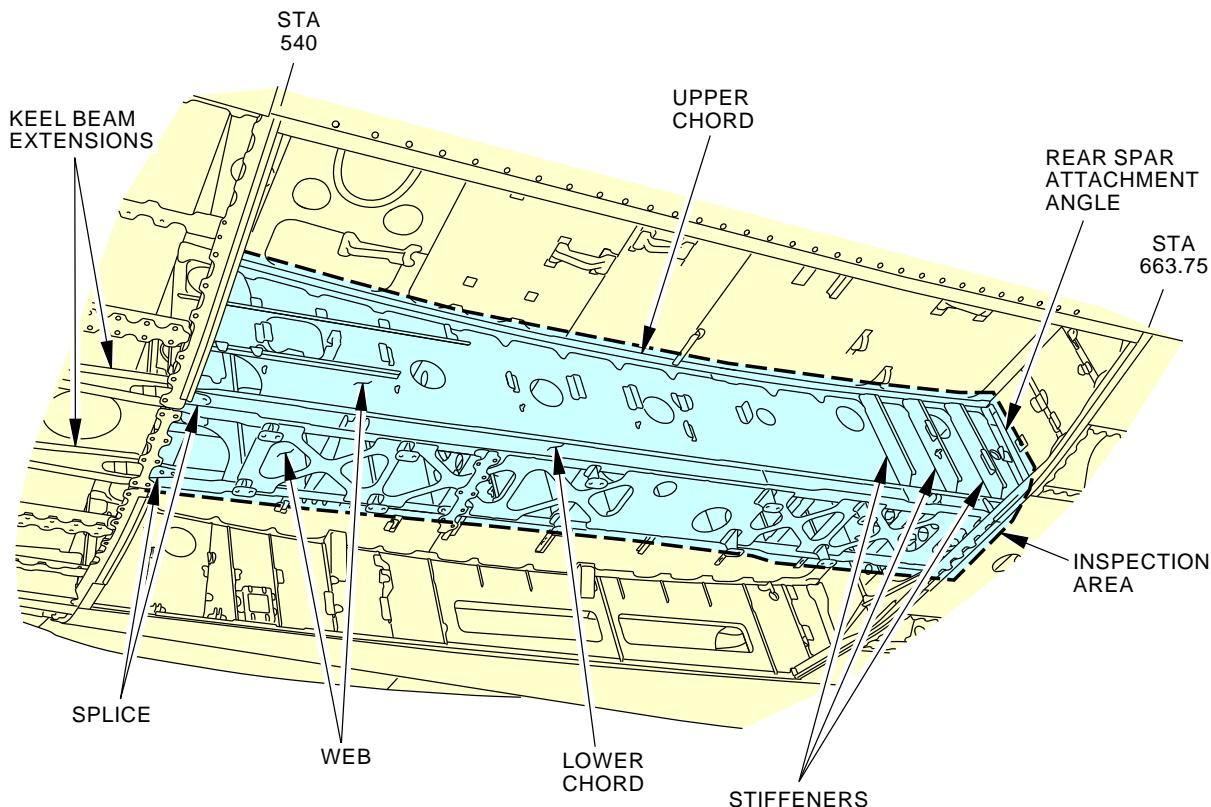
2084307 S0000436344_V2

**Keel Beam
Figure 1 (Sheet 1 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM UNDER WING-TO-BODY FAIRING (UNDER WING BOX)
		D633A109-AKS 53-210-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-210-00-01
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**KEEL BEAM
(STA 540 TO STA 663.75)
(ECS DOORS REMOVED)**

BMPD ITEM
53-210-00

2085210 S0000436345_V3

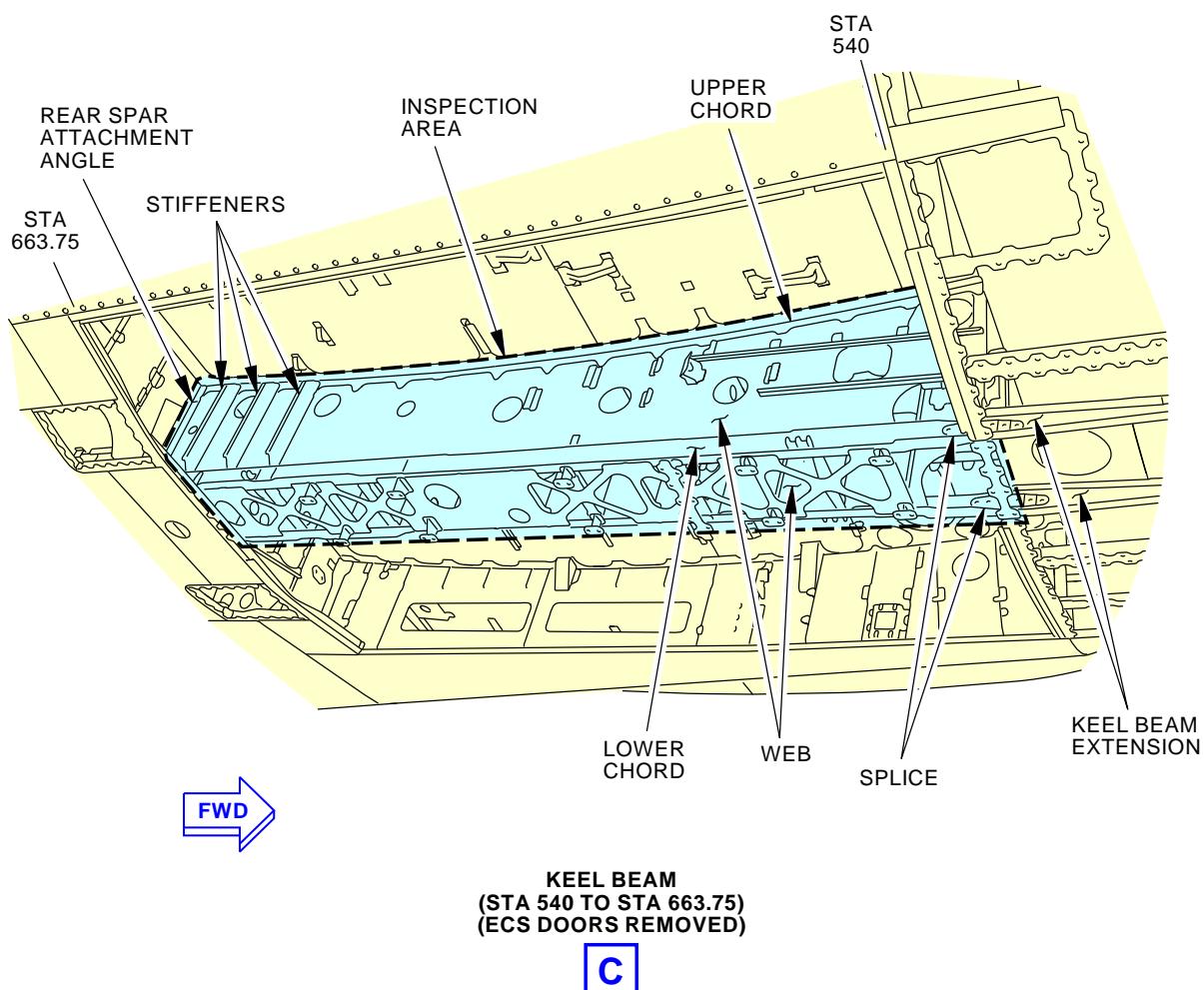
**Keel Beam
Figure 1 (Sheet 2 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM UNDER WING-TO-BODY FAIRING (UNDER WING BOX)
		D633A109-AKS 53-210-00-01

Page 6 of 7
Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-210-00-01
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MPD ITEM
53-210-00

2084810 S0000436346_V4

**Keel Beam
Figure 1 (Sheet 3 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM UNDER WING-TO-BODY FAIRING (UNDER WING BOX)
		D633A109-AKS 53-210-00-01

Page 7 of 7
Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE KEEL BEAM IN WHEEL WELL			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-220-00-01
TAIL NUMBER	WORK AREA KEEL BEAM	VERSION 1.1	THRESHOLD 6 YR	REPEAT 6 YR	APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	
		ACCESS 193B 193D			ZONE 139 193

Inspect keel beam in wheel well (Sta 663.75 to 727), including keel beam chords, webs, stiffeners, splice, keel beam/rear spar attachment angles.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM IN WHEEL WELL	
		D633A109-AKS 53-220-00-01	Page 1 of 5 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-220-00-01
				MECH INSP
TASK 53-05-03-210-823				
1. INTERNAL - GENERAL VISUAL: KEEL BEAM IN WHEEL WELL				
(Figure 1)				
A. Inspection				
SUBTASK 53-05-03-010-020				
(1) Open these access panels:				
Number Name/Location				
193B Wheel Well Panel - Forward Inboard				
193D Wheel Well Panel - Aft Inboard				
SUBTASK 53-05-03-210-023				
(2) Do a General Visual inspection of the keel beam in wheel well (Sta 663.75 to 727), including keel beam chords, webs, stiffeners, splice, keel beam/rear spar attachment angles.				
SUBTASK 53-05-03-910-024				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.				
SUBTASK 53-05-03-410-020				
(4) Close these access panels:				
Number Name/Location				
193B Wheel Well Panel - Forward Inboard				
193D Wheel Well Panel - Aft Inboard				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM IN WHEEL WELL	
		D633A109-AKS 53-220-00-01	Page 2 of 5 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-220-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM IN WHEEL WELL	
		D633A109-AKS 53-220-00-01	Page 3 of 5 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-220-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-063

- (7) CPCP Basic Task Item 7 is not applicable.

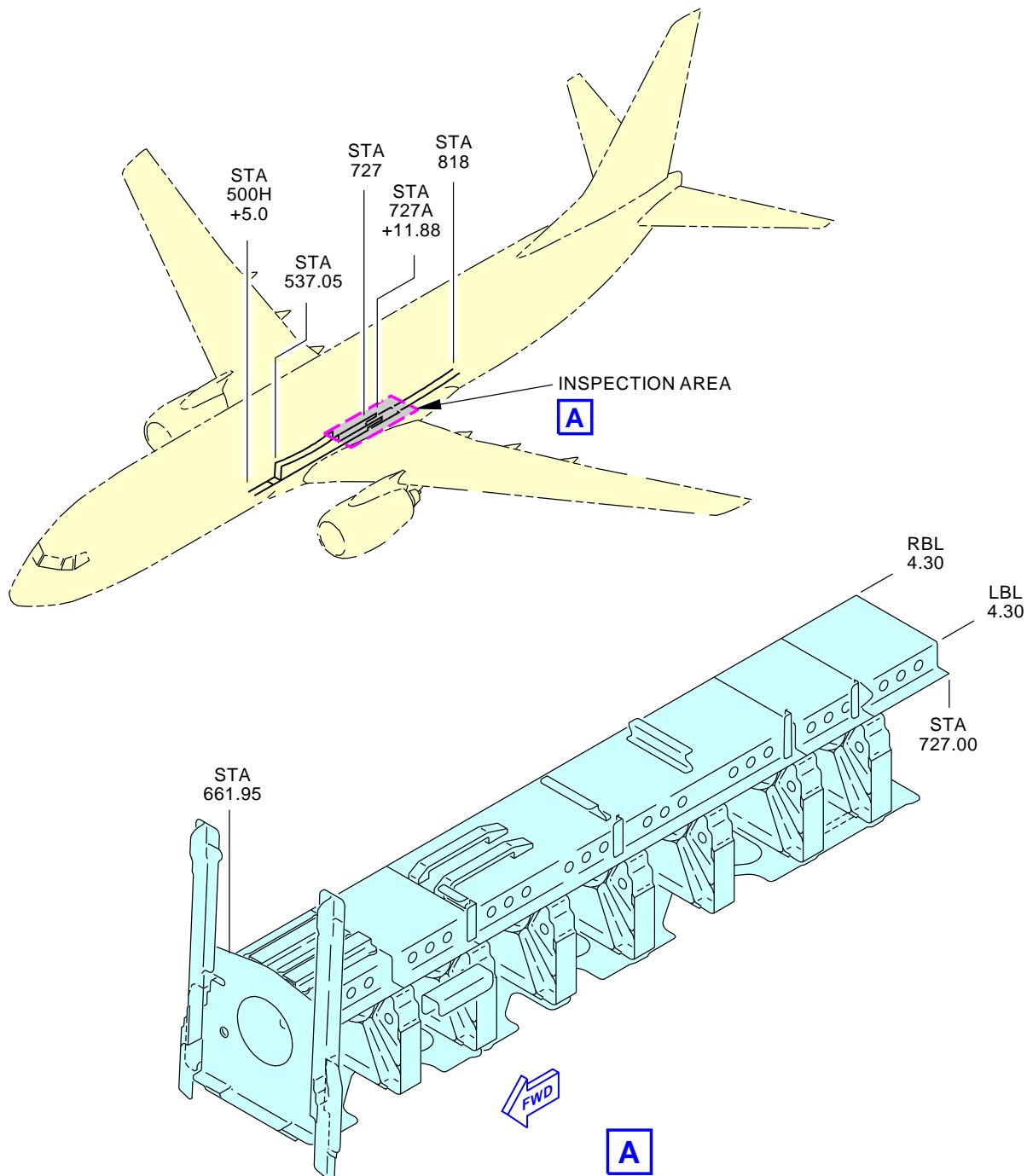
———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM IN WHEEL WELL
		D633A109-AKS 53-220-00-01

Page 4 of 5
Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-220-00-01



D65252 S0000161683_V2

INTERNAL - GENERAL VISUAL: KEEL BEAM IN WHEEL WELL
Figure 1

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM IN WHEEL WELL
		D633A109-AKS 53-220-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO COMPARTMENT			BOEING CARD NO.	
DATE	TASK GENERAL VISUAL				53-230-00-01	
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY AIRPLANE ALL ENGINE ALL	
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC		
		NOTE	ACCESS S1401			
		NOTE	ZONE 141 142			

Inspect aft cargo compartment, including: 1. Side skin panels (skin, frames, stringers), circumferential skin and stringer splices, (note: located at Sta 727I for -900 and 727L for -900ER models); 2. Stringer 18 strap at side of body; 3. Stringer 18A web, chord and links; 4. Aft side of STA 727 bulkhead and pressure web.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove sidewall and ceiling panels, E6 LRU, access panels around vacuum lav tank. Remove/displace insulation blankets as required. Remove/displace vacuum lav components as required. Remove/displace auxiliary fuel tank as required (business jet only).

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-230-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-230-00-01
				MECH INSP
TASK 53-05-03-210-824				
1. INTERNAL - GENERAL VISUAL: AFT CARGO COMPARTMENT				
(Figure 1)				
A. Inspection				
SUBTASK 53-05-03-010-021				
(1) Open this access panel:				
Number Name/Location				
S1401 AFT Cargo Compartment Inspection				
NOTE: Remove sidewall and ceiling panels, E6 LRU, access panels around vacuum lav tank. Remove/displace insulation blankets as required. Remove/displace vacuum lav components as required. Remove/displace auxiliary fuel tank as required (business jet only).				
SUBTASK 53-05-03-210-024				
(2) Do a General Visual inspection of the aft cargo compartment, including:				
1. Side skin panels (skin, frames, stringers), circumferential skin and stringer splices, (note: located at Sta 727I for -900 and 727L for -900ER models).				
2. Stringer 18 strap at side of body.				
3. Stringer 18A web, chord and links.				
4. Aft side of STA 727 bulkhead and pressure web.				
SUBTASK 53-05-03-910-025				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.				
SUBTASK 53-05-03-410-021				
(4) Close this access panel:				
Number Name/Location				
S1401 AFT Cargo Compartment Inspection				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-230-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-230-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT	
		D633A109-AKS 53-230-00-01	Page 3 of 7 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-230-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-230-00-01

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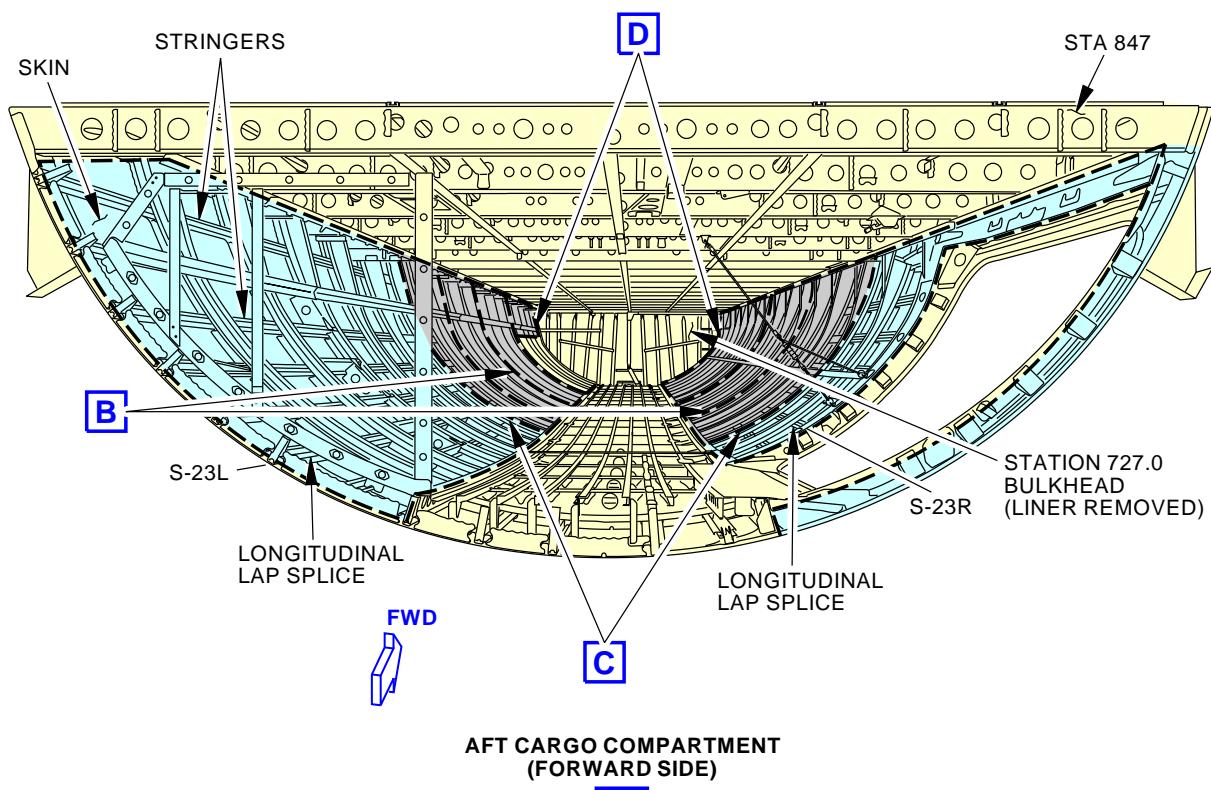
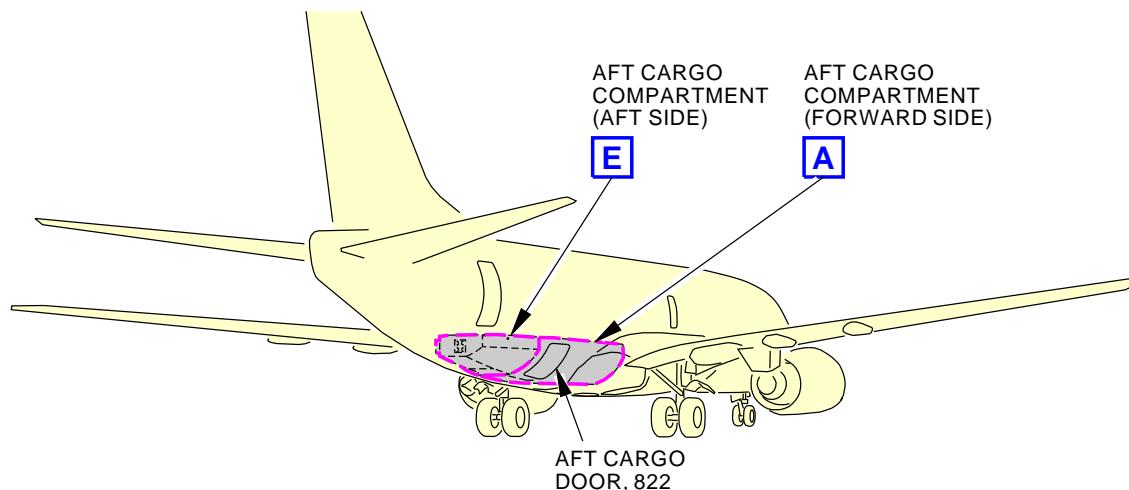
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-230-00-01MPD ITEM
53-230-00

2094308 S0000439209_V2

INTERNAL-GENERAL VISUAL: AFT CARGO COMPARTMENT
Figure 1 (Sheet 1 of 3)

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-230-00-01

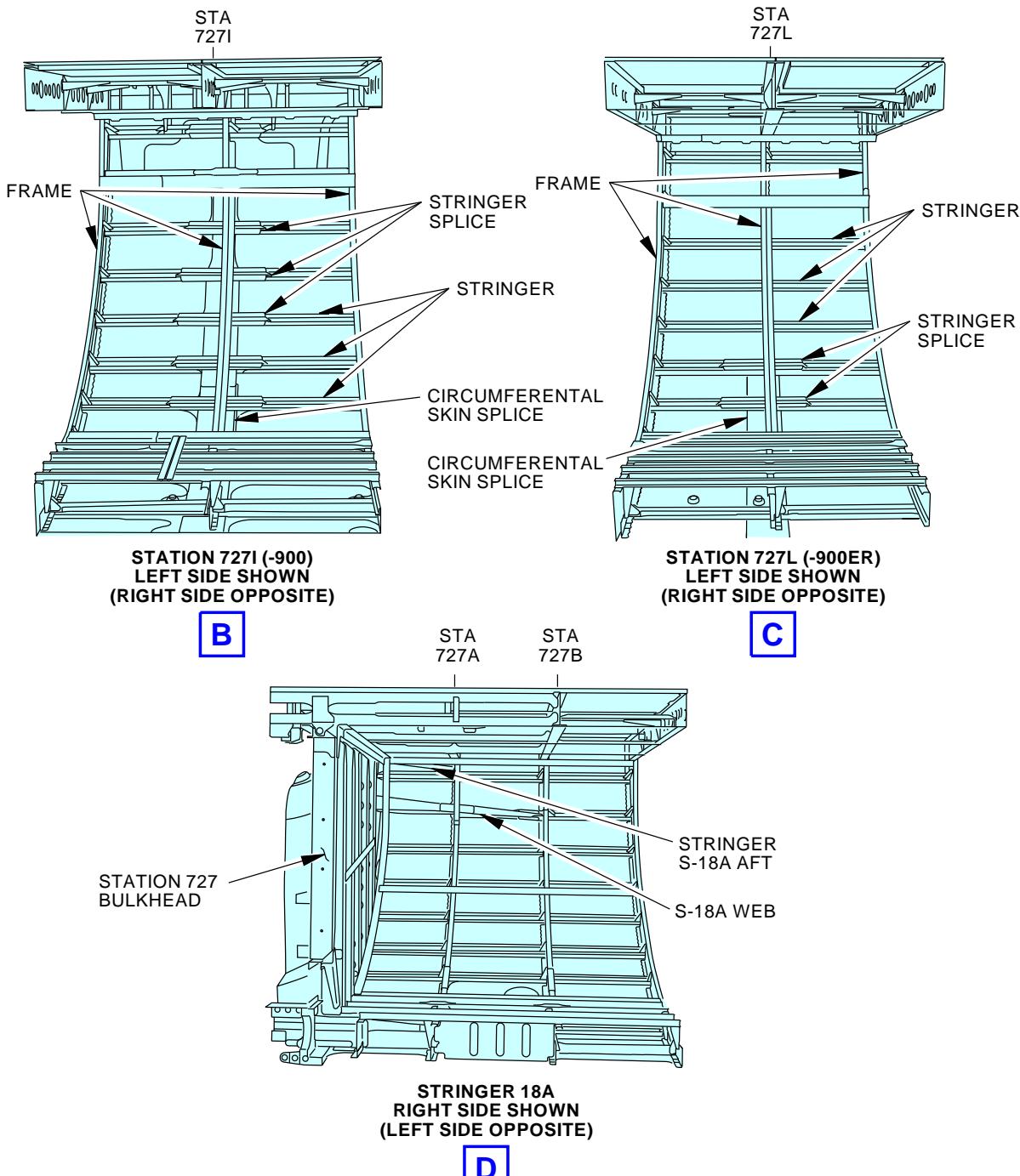
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-230-00-01MPD ITEM
53-230-00

2094379 S0000439211_V2

INTERNAL-GENERAL VISUAL: AFT CARGO COMPARTMENT
Figure 1 (Sheet 2 of 3)

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-230-00-01

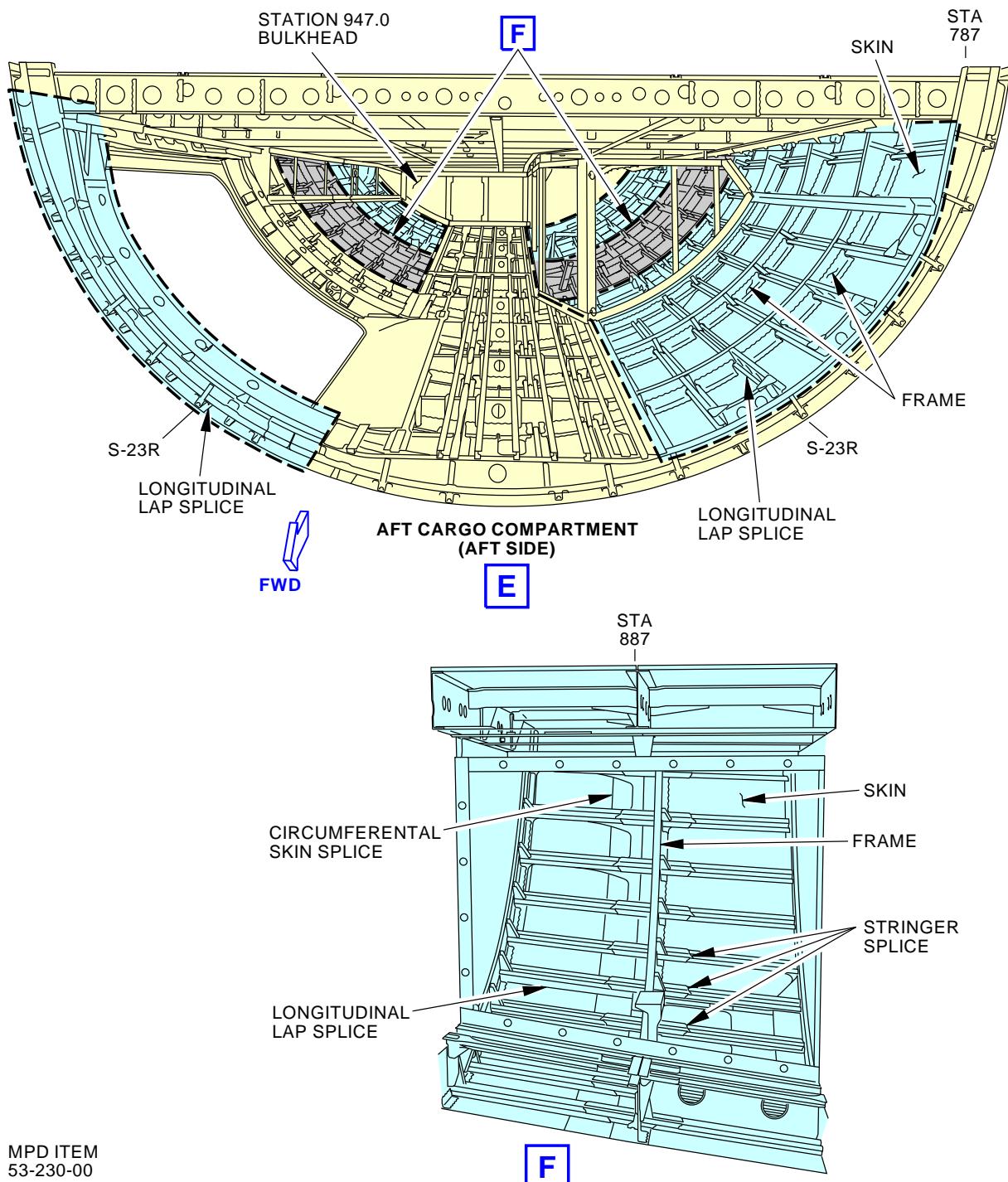
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-230-00-01MPD ITEM
53-230-00

2094603 S0000439210_V2

INTERNAL-GENERAL VISUAL: AFT CARGO COMPARTMENT
Figure 1 (Sheet 3 of 3)

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-230-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR CUTOUT			BOEING CARD NO.
DATE	TASK DETAILED				53-240-00-01
TAIL NUMBER	WORK AREA CARGO DOOR	VERSION 1.1 1.2 NOTE	THRESHOLD 8 YR 24000 FC	REPEAT 6 YR 18000 FC	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS S1421 NOTE			ENGINE ALL
					ZONE 142

Inspect aft cargo door cutout surround structure.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove door reveals. Remove sidewalls as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO DOOR CUTOUT
		D633A109-AKS 53-240-00-01

Page 1 of 5
Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-240-00-01
TASK 53-05-03-211-804				MECH INSP

1. INTERNAL - DETAILED: AFT CARGO DOOR CUTOUT
(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-058

(1) Special Access:

Number Name/Location
S1421 Aft Cargo Door Cutout Inspection

NOTE: Remove door reveals. Remove sidewalls as required. Remove/displace insulation blankets as required.

SUBTASK 53-05-03-211-004

(2) Do a Detailed inspection of the aft cargo door cutout surround structure.

SUBTASK 53-05-03-910-029

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

SUBTASK 53-05-03-410-058

(4) Close this access panel:

Number Name/Location
S1421 Aft Cargo Door Cutout Inspection

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO DOOR CUTOUT
		D633A109-AKS 53-240-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-240-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO DOOR CUTOUT	
		D633A109-AKS 53-240-00-01	Page 3 of 5 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-240-00-01
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH INSP

SUBTASK 51-05-01-210-077

(7) Do the CPCP Basic Task Item 7 as follows:

(a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

 END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO DOOR CUTOUT
		D633A109-AKS 53-240-00-01

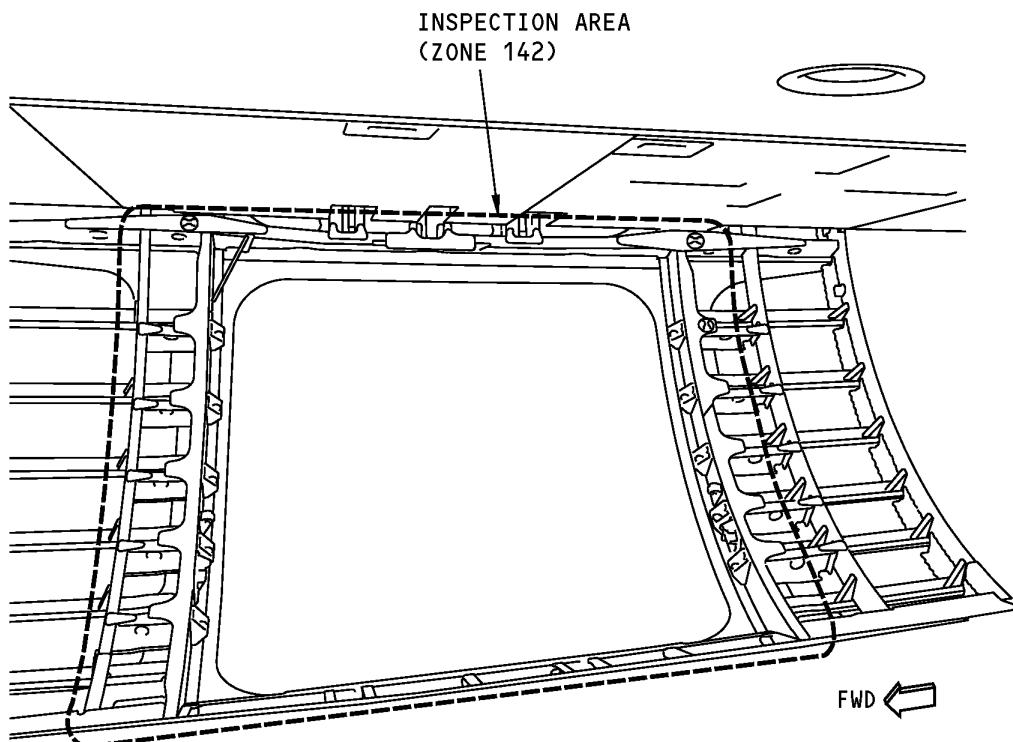
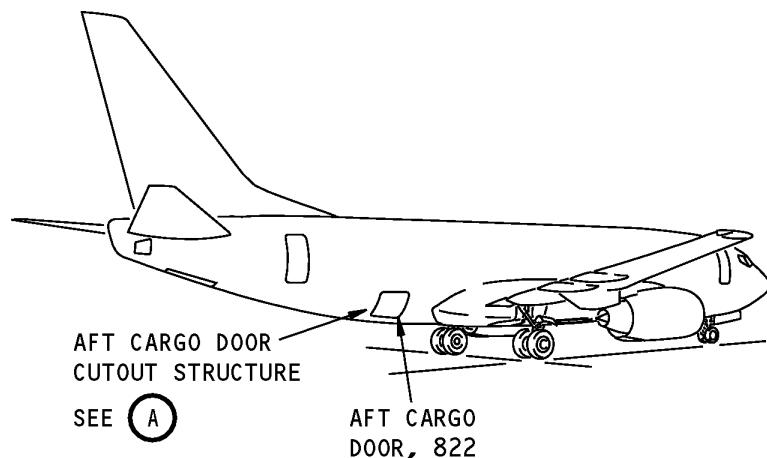
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-240-00-01

**AFT CARGO DOOR CUTOUT STRUCTURE
(DOOR REVEALS AND SIDEWALL PANELS REMOVED)**



**Aft Cargo Door Cutout Detailed (Internal)
Figure 1**

EFFECTIVITY
AKS ALLSOURCE
MRB**AFT CARGO DOOR CUTOUT****D633A109-AKS
53-240-00-01****Page 5 of 5
Oct 15/2014**

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT BILGE			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-250-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 8 YR 24000 FC	REPEAT 6 YR 18000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS S1402 NOTE			ZONE 143 144

Inspect aft bilge skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices, (note: located at Sta 727I for -900 and 727L for -900ER models); Sta 727 bulkhead and pressure web, and cargo door cutout surround structure in bilge.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove cargo floor panels and scuff plates. Remove/Displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AFT BILGE D633A109-AKS 53-250-00-01	Page 1 of 8 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-250-00-01
				MECH INSP
TASK 53-05-03-210-825				
1. INTERNAL - GENERAL VISUAL: AFT BILGE				
(Figure 1,Figure 2)				
A. Inspection				
SUBTASK 53-05-03-010-022				
(1) Special Access:				
Number Name/Location				
S1402 Aft Bilge Inspection				
NOTE: Remove cargo floor panels and scuff plates. Remove/Displace insulation blankets as required.				
SUBTASK 53-05-03-210-025				
(2) Do a General Visual inspection of the aft bilge skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices, (note: located at Sta 727I for -900 and 727L for -900ER models); Sta 727 bulkhead and pressure web, and cargo door cutout surround structure in bilge.				
SUBTASK 53-05-03-910-030				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	AFT BILGE D633A109-AKS 53-250-00-01	Page 2 of 8 Feb 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-250-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AFT BILGE D633A109-AKS 53-250-00-01	Page 3 of 8 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-250-00-01			
				<table border="1"><tr><td>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</td><td>MECH</td><td>INSP</td></tr></table>	1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP
1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP					

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

- 2) Not applicable

- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

Water displacing / anti-corrosion compounds should not be applied in the following areas:

- Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
- Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
- Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
- Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
- Areas with electrical arc potential.
- Interior materials, including cargo liners (change of flammability properties).
- Fiber-glass ducts where temperature exceeds 220 degrees F.
- Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:

- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AFT BILGE D633A109-AKS 53-250-00-01	Page 4 of 8 Oct 15/2014
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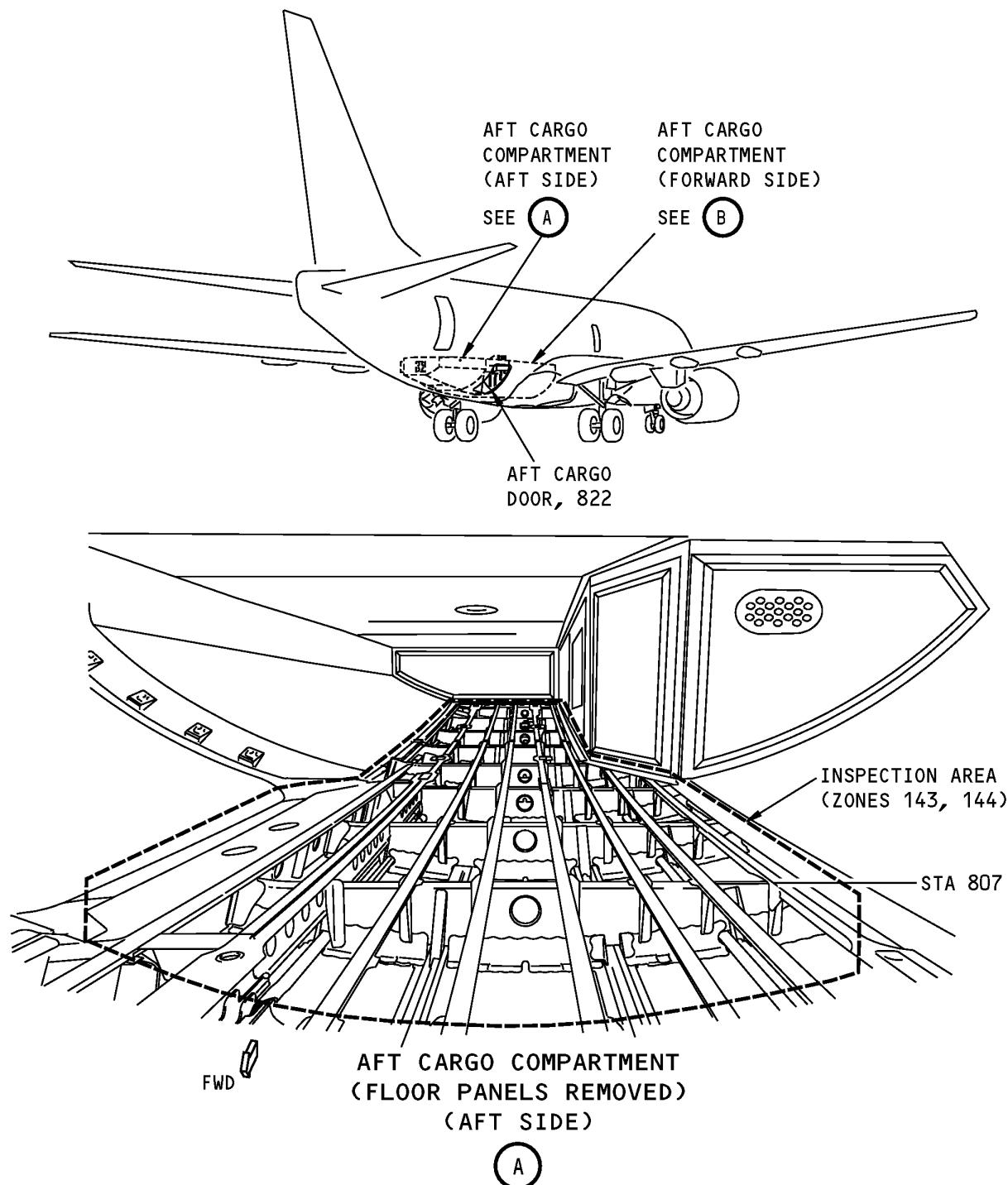
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

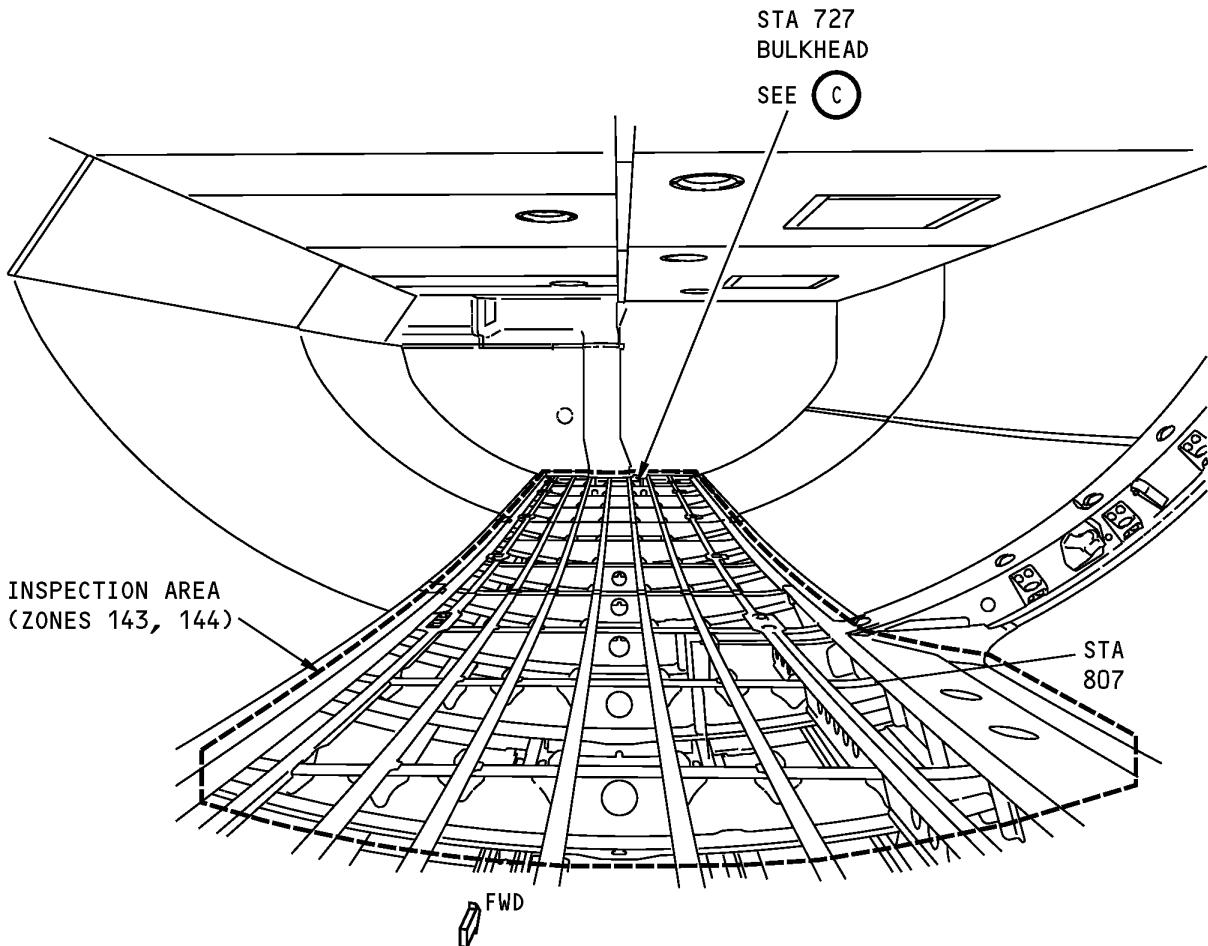
BOEING CARD NO.
53-250-00-01

Below the Aft Cargo Comp. - Aft Bilge General Visual (Int)
Figure 1 (Sheet 1 of 3)

EFFECTIVITY
AKS ALLSOURCE
MRB**AFT BILGE****D633A109-AKS
53-250-00-01****Page 5 of 8
Oct 15/2014**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-250-00-01



**AFT CARGO COMPARTMENT
(FLOOR PANELS REMOVED)
(FORWARD SIDE)**



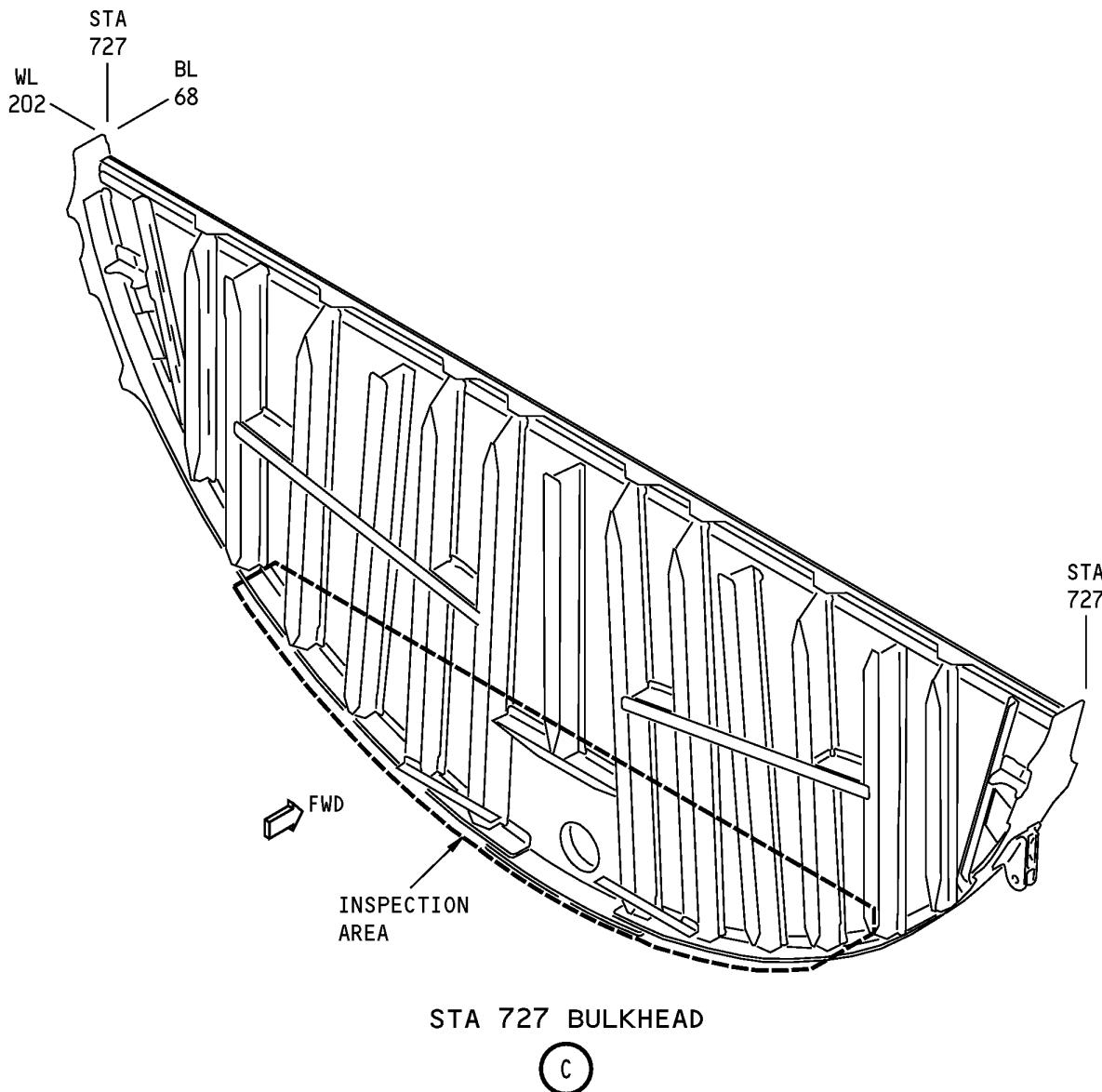
**Below the Aft Cargo Comp. - Aft Bilge General Visual (Int)
Figure 1 (Sheet 2 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	AFT BILGE
		D633A109-AKS 53-250-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-250-00-01



**Below the Aft Cargo Comp. - Aft Bilge General Visual (Int)
Figure 1 (Sheet 3 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	AFT BILGE D633A109-AKS 53-250-00-01	Page 7 of 8 Oct 15/2014
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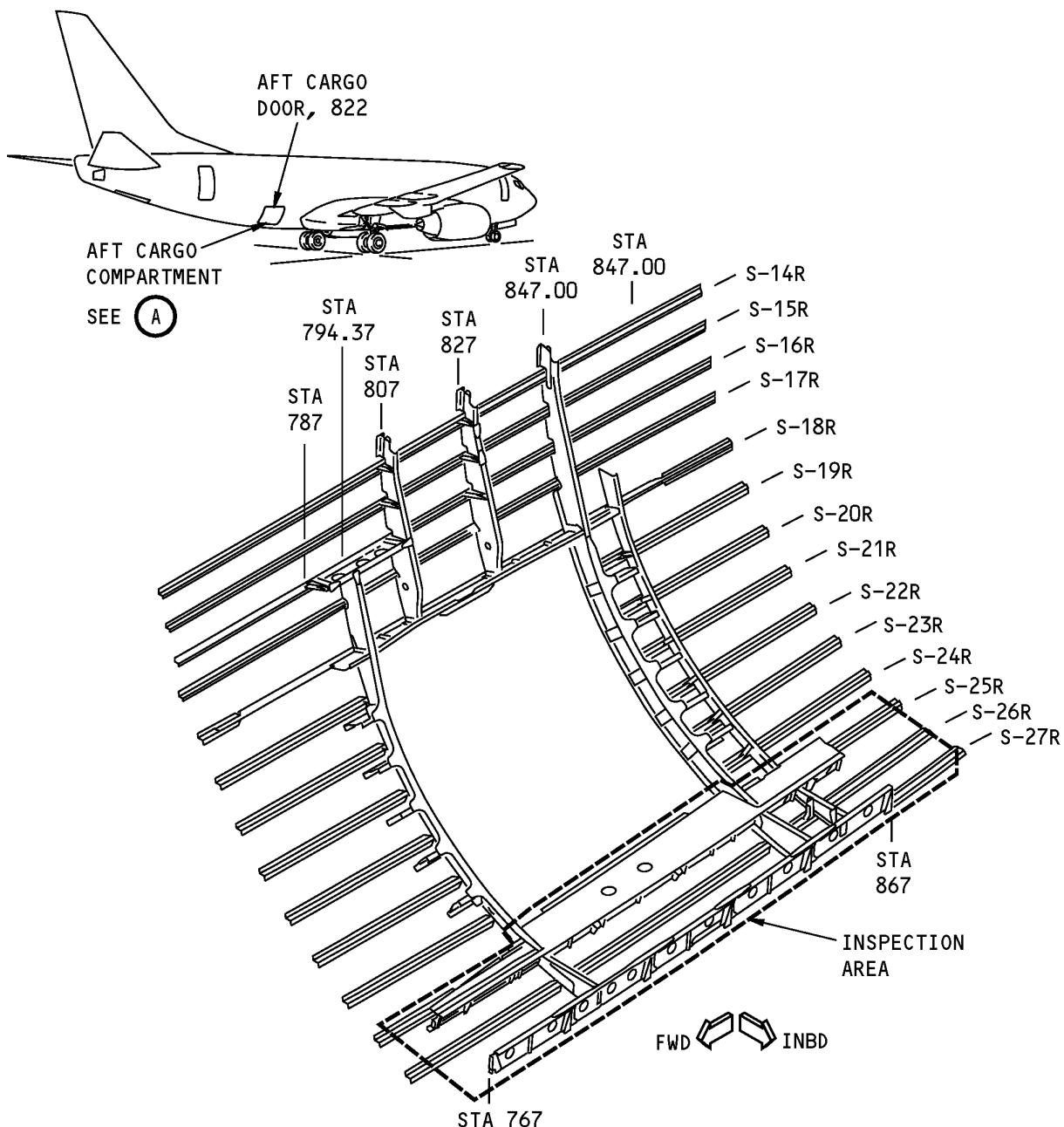
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-250-00-01

**AFT CARGO COMPARTMENT
(DOOR REVEALS AND SIDEWALL PANELS REMOVED)**

(A)

**Section 46 Aft Cargo Door Surround Structure Locations - General Visual (Internal)
Figure 2**

EFFECTIVITY
AKS ALLSOURCE
MRB**AFT BILGE****D633A109-AKS
53-250-00-01****Page 8 of 8
Oct 15/2014**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA AFT OF CARGO COMPARTMENT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-260-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS S1403			ZONE 145 146
		NOTE			

Inspect area aft of cargo compartment, including: 1. Skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices; 2. Aft entry and galley door cutout surround structure in lower lobe; 3. STA 1016 bulkhead, including chords, pressure web, stiffeners, chord/web attachments; 4. Stringer splice fittings and tension bolts at STA 1016.

INTERVAL NOTE: Whichever comes first.

AIRPLANE NOTE: Task not applicable to -900ER and -800 with Flat Pressure Bulkhead installed.

ACCESS NOTE: Remove aft cargo compartment aft bulkhead panel and potable water tank.

Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF CARGO COMPARTMENT	
		D633A109-AKS 53-260-00-01	Page 1 of 7 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-260-00-01
TASK 53-05-03-210-826				MECH INSP

1. INTERNAL - GENERAL VISUAL: AREA AFT OF CARGO COMPARTMENT

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-023

(1) Special Access:

Number Name/Location

S1403 Area Aft of Cargo Compartment Inspection

NOTE: Remove aft cargo compartment aft bulkhead panel and potable water tank.
Remove/displace insulation blankets as required.

SUBTASK 53-05-03-210-026

(2) Do a General Visual inspection of the area aft of cargo compartment, including:

1. Skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices.
2. Aft entry and galley door cutout surround structure in lower lobe.
3. STA 1016 bulkhead, including chords, pressure web, stiffeners, chord/web attachments.
4. Stringer splice fittings and tension bolts at STA 1016.

SUBTASK 53-05-03-910-031

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF CARGO COMPARTMENT
		D633A109-AKS 53-260-00-01

Page 2 of 7
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-260-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF CARGO COMPARTMENT	
		D633A109-AKS 53-260-00-01	Page 3 of 7 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-260-00-01			
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	MECH	INSP					

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF CARGO COMPARTMENT	
		D633A109-AKS 53-260-00-01	Page 4 of 7 Oct 15/2014

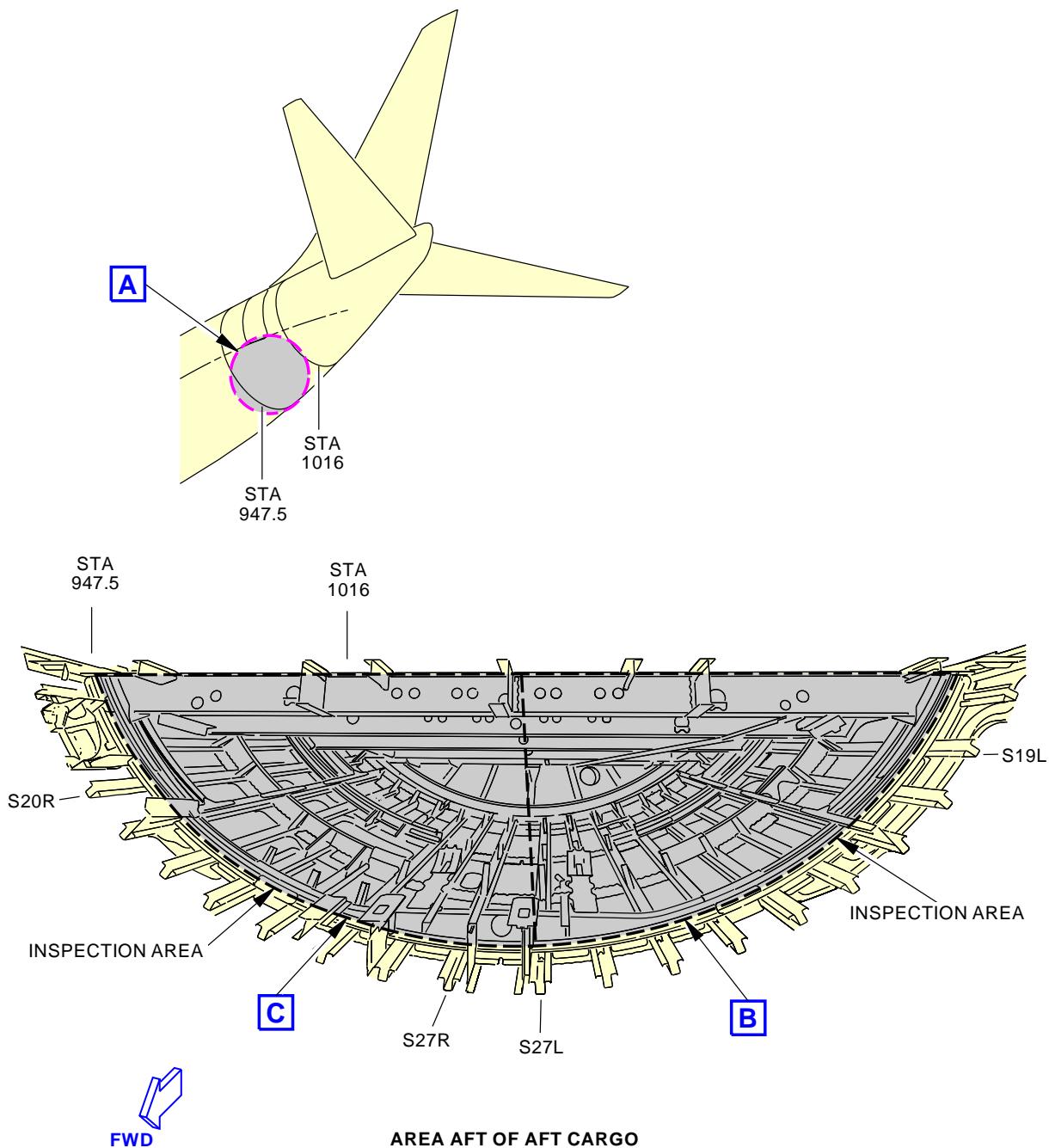
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-260-00-01MPD ITEM
53-260-00

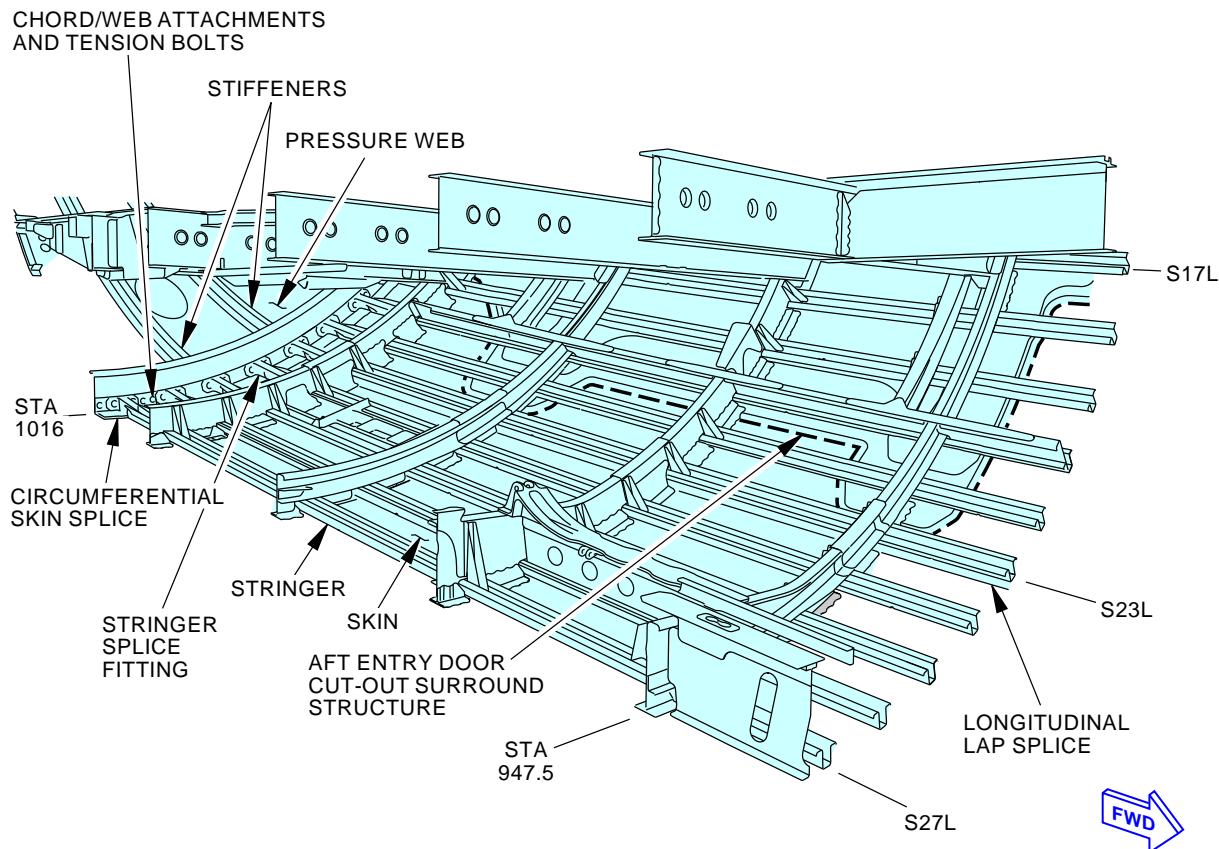
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**Internal-General Visual: Internal-Area Aft Of Cargo Compartment
Figure 1 (Sheet 1 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF CARGO COMPARTMENT
		D633A109-AKS 53-260-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-260-00-01
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MPD ITEM
53-260-00

2094973 S0000442106_V2

**Internal-General Visual: Internal-Area Aft Of Cargo Compartment
Figure 1 (Sheet 2 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF CARGO COMPARTMENT
		D633A109-AKS 53-260-00-01

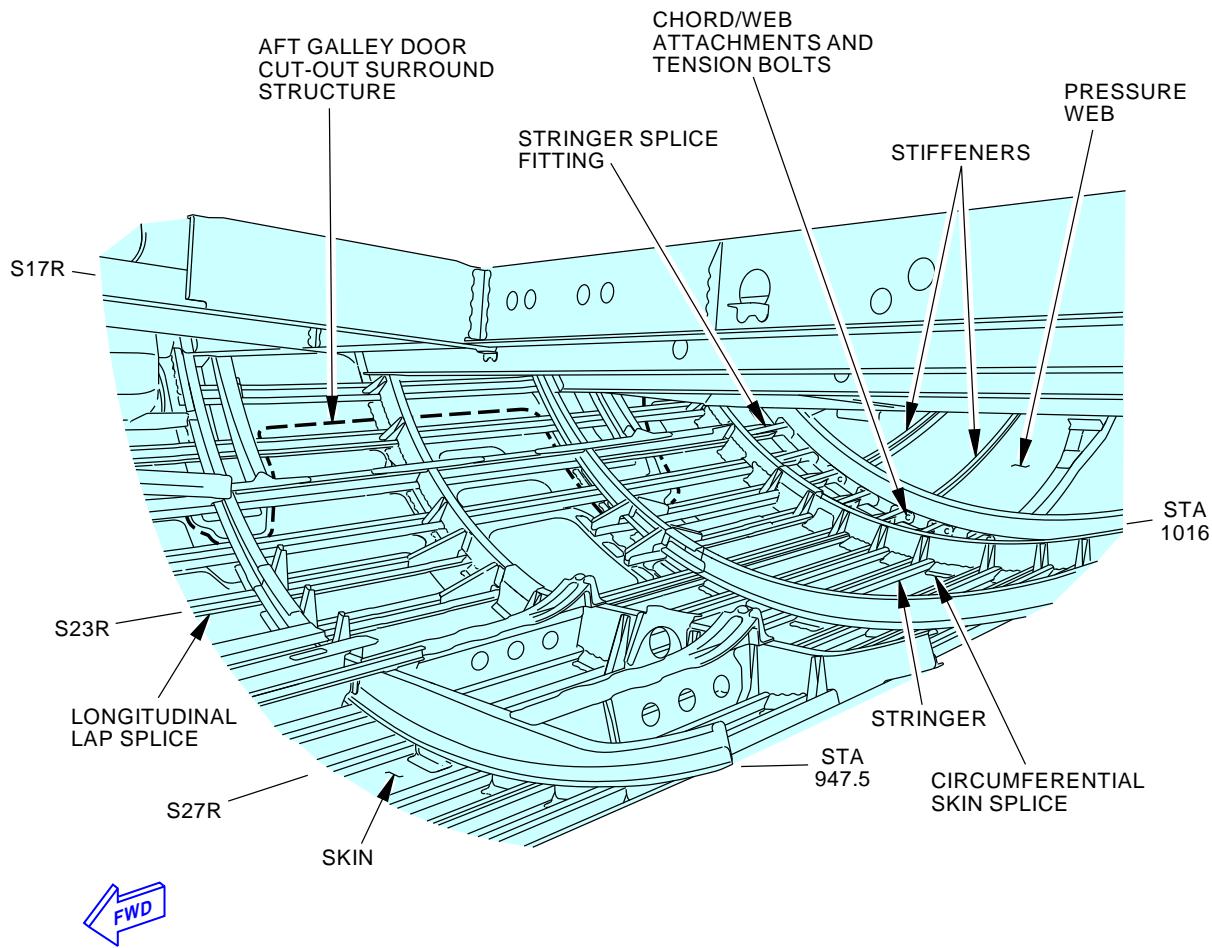
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-260-00-01MPD ITEM
53-260-00

2095515 S0000442107_V2

**Internal-General Visual: Internal-Area Aft Of Cargo Compartment
Figure 1 (Sheet 3 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF CARGO COMPARTMENT
		D633A109-AKS 53-260-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA UNDER LOWER WING-TO-BODY FAIRING (AFT OF WHEEL WELL)			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-270-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 194AL 194AR 194BL 194BR 194CL 194CR 194DL 194DR 194E 194FL 194FR 194GL 194GR 194HL 194HR			ZONE 149 194

Inspect area under lower wing-to-body fairing (aft of wheel well), including skin panels, longitudinal lap splices, circumferential skin splice, stringer 18 strap at side of body, stringer 18A (web, chords and links), and keel beam extension.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (AFT OF WHEEL WELL)	
		D633A109-AKS 53-270-00-01	Page 1 of 7 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-270-00-01																																
				MECH INSP																																
TASK 53-05-03-210-827																																				
1. INTERNAL - GENERAL VISUAL: AREA UNDER LOWER WING-TO-BODY FAIRING (aft of wheel well)																																				
(Figure 1)																																				
A. Inspection																																				
SUBTASK 53-05-03-010-024																																				
(1) Open these access panels:																																				
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>194AL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194AR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194BL</td><td>Flap Track Lubrication Panel - Aft</td></tr><tr><td>194BR</td><td>Flap Track Lubrication Panel - Aft</td></tr><tr><td>194CL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194CR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194DL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194DR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194E</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194FL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194FR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194HL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194HR</td><td>Aft Wing To Body Fairing Panel</td></tr></tbody></table>				Number	Name/Location	194AL	Aft Wing To Body Fairing Panel	194AR	Aft Wing To Body Fairing Panel	194BL	Flap Track Lubrication Panel - Aft	194BR	Flap Track Lubrication Panel - Aft	194CL	Aft Wing To Body Fairing Panel	194CR	Aft Wing To Body Fairing Panel	194DL	Aft Wing To Body Fairing Panel	194DR	Aft Wing To Body Fairing Panel	194E	Aft Wing To Body Fairing Panel	194FL	Aft Wing To Body Fairing Panel	194FR	Aft Wing To Body Fairing Panel	194GL	Aft Wing To Body Fairing Panel	194GR	Aft Wing To Body Fairing Panel	194HL	Aft Wing To Body Fairing Panel	194HR	Aft Wing To Body Fairing Panel	
Number	Name/Location																																			
194AL	Aft Wing To Body Fairing Panel																																			
194AR	Aft Wing To Body Fairing Panel																																			
194BL	Flap Track Lubrication Panel - Aft																																			
194BR	Flap Track Lubrication Panel - Aft																																			
194CL	Aft Wing To Body Fairing Panel																																			
194CR	Aft Wing To Body Fairing Panel																																			
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194GR	Aft Wing To Body Fairing Panel																																			
194HL	Aft Wing To Body Fairing Panel																																			
194HR	Aft Wing To Body Fairing Panel																																			
SUBTASK 53-05-03-210-027																																				
(2) Do a General Visual inspection of the area under lower wing-to-body fairing (aft of wheel well), including skin panels, longitudinal lap splices, circumferential skin splice, stringer 18 strap at side of body, stringer 18A (web, chords and links), and keel beam extension.																																				
SUBTASK 53-05-03-910-032																																				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.																																				
SUBTASK 53-05-03-410-024																																				
(4) Close these access panels:																																				
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>194AL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194AR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194BL</td><td>Flap Track Lubrication Panel - Aft</td></tr><tr><td>194BR</td><td>Flap Track Lubrication Panel - Aft</td></tr><tr><td>194CL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194CR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194DL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194DR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194E</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194FL</td><td>Aft Wing To Body Fairing Panel</td></tr></tbody></table>				Number	Name/Location	194AL	Aft Wing To Body Fairing Panel	194AR	Aft Wing To Body Fairing Panel	194BL	Flap Track Lubrication Panel - Aft	194BR	Flap Track Lubrication Panel - Aft	194CL	Aft Wing To Body Fairing Panel	194CR	Aft Wing To Body Fairing Panel	194DL	Aft Wing To Body Fairing Panel	194DR	Aft Wing To Body Fairing Panel	194E	Aft Wing To Body Fairing Panel	194FL	Aft Wing To Body Fairing Panel											
Number	Name/Location																																			
194AL	Aft Wing To Body Fairing Panel																																			
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194BL	Flap Track Lubrication Panel - Aft																																			
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194CR	Aft Wing To Body Fairing Panel																																			
194DL	Aft Wing To Body Fairing Panel																																			
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194E	Aft Wing To Body Fairing Panel																																			
194FL	Aft Wing To Body Fairing Panel																																			
EFFECTIVITY AKS ALL				SOURCE MRB																																
AREA UNDER LOWER WING-TO-BODY FAIRING (AFT OF WHEEL WELL)				D633A109-AKS 53-270-00-01																																
				Page 2 of 7 Feb 15/2015																																

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-270-00-01
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(Continued)

Number Name/Location

- | | |
|-------|--------------------------------|
| 194FR | Aft Wing To Body Fairing Panel |
| 194GL | Aft Wing To Body Fairing Panel |
| 194GR | Aft Wing To Body Fairing Panel |
| 194HL | Aft Wing To Body Fairing Panel |
| 194HR | Aft Wing To Body Fairing Panel |

MECH

INSP

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (AFT OF WHEEL WELL)
		D633A109-AKS 53-270-00-01

Page 3 of 7
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-270-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (AFT OF WHEEL WELL)	D633A109-AKS 53-270-00-01	Page 4 of 7 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-270-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

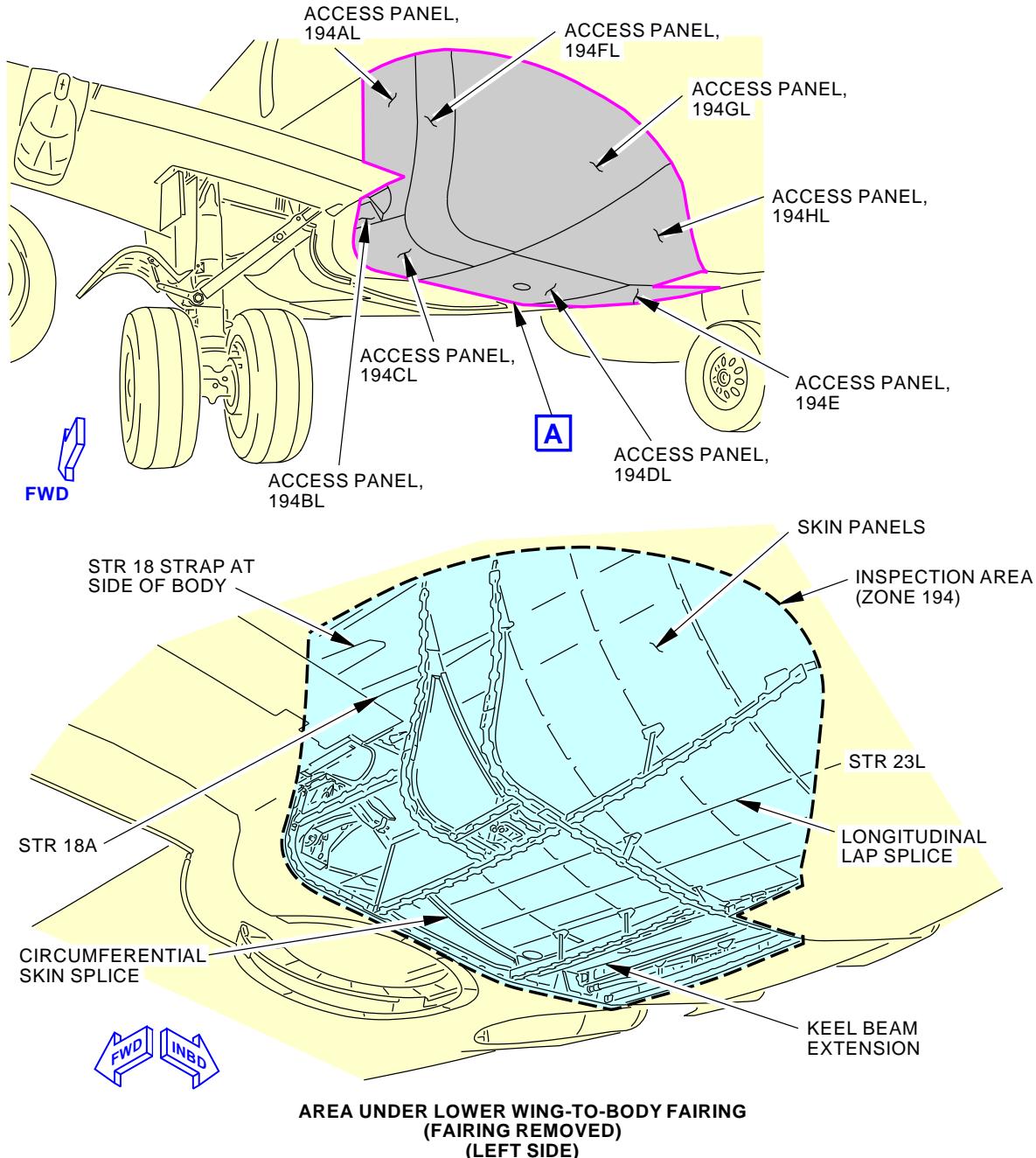
- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (AFT OF WHEEL WELL)	
		D633A109-AKS 53-270-00-01	Page 5 of 7 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-270-00-01
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MPD ITEM
53-270-00

2089485 S0000441011_V2

INTERNAL-GENERAL VISUAL: AREA UNDER LOWER WING-TO-BODY FAIRING (aft of wheel well)
Figure 1 (Sheet 1 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (AFT OF WHEEL WELL)
		D633A109-AKS 53-270-00-01

Page 6 of 7
Oct 15/2015

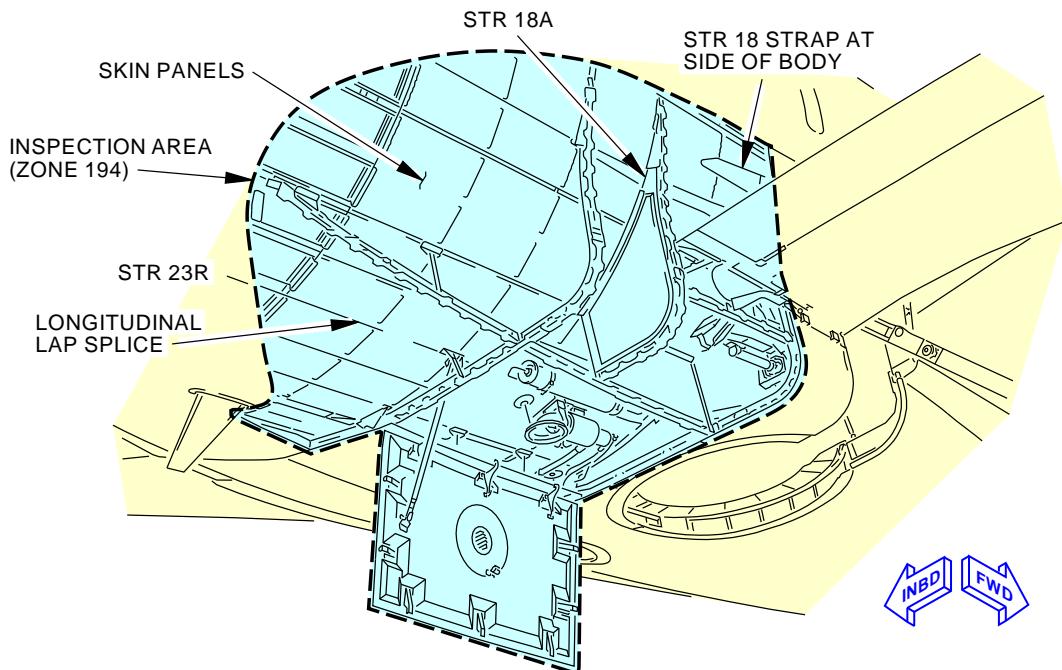
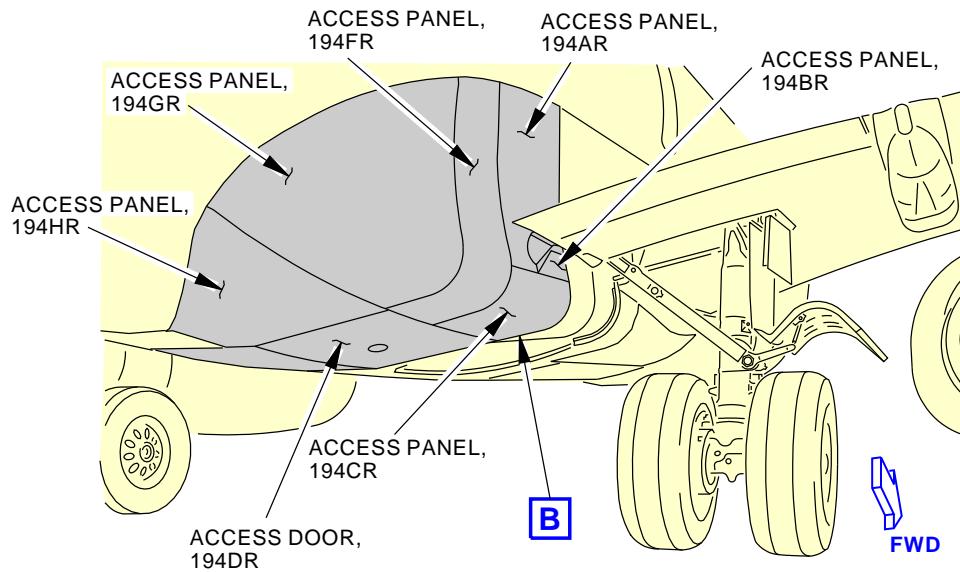
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-270-00-01MPD ITEM
57-270-00

2089514 S0000441012_V2

INTERNAL-GENERAL VISUAL: AREA UNDER LOWER WING-TO-BODY FAIRING (aft of wheel well)
Figure 1 (Sheet 2 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER LOWER WING-TO-BODY FAIRING (AFT OF WHEEL WELL)
		D633A109-AKS 53-270-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA UNDER WING-TO-BODY FAIRING (ABOVE WING)			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-280-00-01
TAIL NUMBER	WORK AREA LWR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 195AL 195AR 195BL 195BR 195CL 195CR			ZONE 195 196

Inspect area under above-wing wing-to-body fairing, including skin panels, circumferential skin splices, and stringer 18 strap at side of body.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER WING-TO-BODY FAIRING (ABOVE WING)
		D633A109-AKS 53-280-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-280-00-01
TASK 53-05-03-210-828				MECH INSP

1. INTERNAL - GENERAL VISUAL: AREA UNDER WING-TO-BODY FAIRING (above wing)
(Figure 1,Figure 2)

A. Inspection

SUBTASK 53-05-03-010-025

(1) Open these access panels:

<u>Number</u>	<u>Name/Location</u>
195AL	Wing To Body Fairing - Left Side
195AR	Wing To Body Fairing - Right Side
195BL	Wing To Body Fairing - Left Side
195BR	Wing To Body Fairing - Right Side
195CL	Wing To Body Fairing - Left Side
195CR	Wing To Body Fairings - Right Side

SUBTASK 53-05-03-210-028

(2) Do a General Visual inspection of the area under above-wing wing-to-body fairing, including skin panels, circumferential skin splices, and stringer 18 strap at side of body.

SUBTASK 53-05-03-910-035

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.

SUBTASK 53-05-03-410-025

(4) Close these access panels:

<u>Number</u>	<u>Name/Location</u>
195AL	Wing To Body Fairing - Left Side
195AR	Wing To Body Fairing - Right Side
195BL	Wing To Body Fairing - Left Side
195BR	Wing To Body Fairing - Right Side
195CL	Wing To Body Fairing - Left Side
195CR	Wing To Body Fairings - Right Side

———— END OF TASK ——

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER WING-TO-BODY FAIRING (ABOVE WING)	
		D633A109-AKS 53-280-00-01	Page 2 of 6 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-280-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER WING-TO-BODY FAIRING (ABOVE WING)	
		D633A109-AKS 53-280-00-01	Page 3 of 6 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-280-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-063

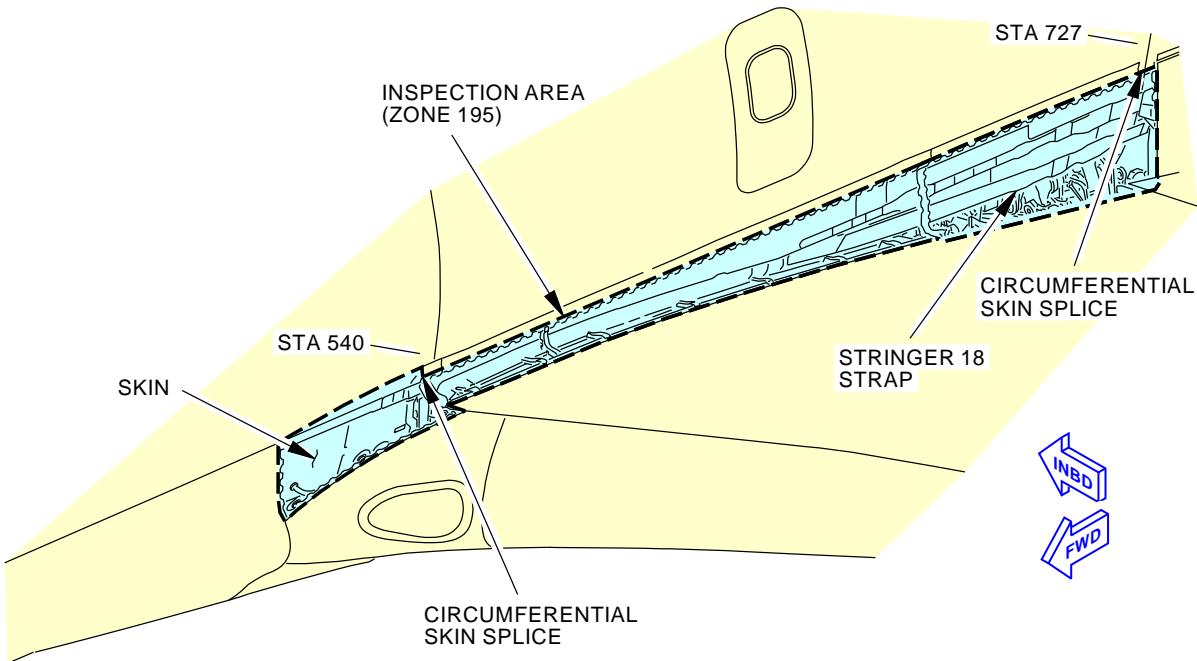
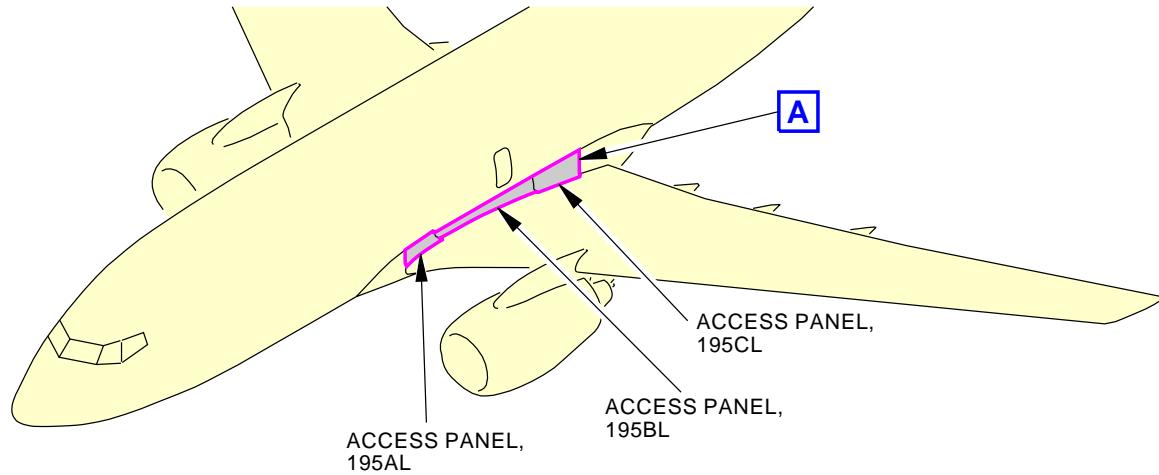
- (7) CPCP Basic Task Item 7 is not applicable.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER WING-TO-BODY FAIRING (ABOVE WING)	
		D633A109-AKS 53-280-00-01	Page 4 of 6 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-280-00-01
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MPD ITEM
53-280-00**A**

2083283 S0000437150_V2

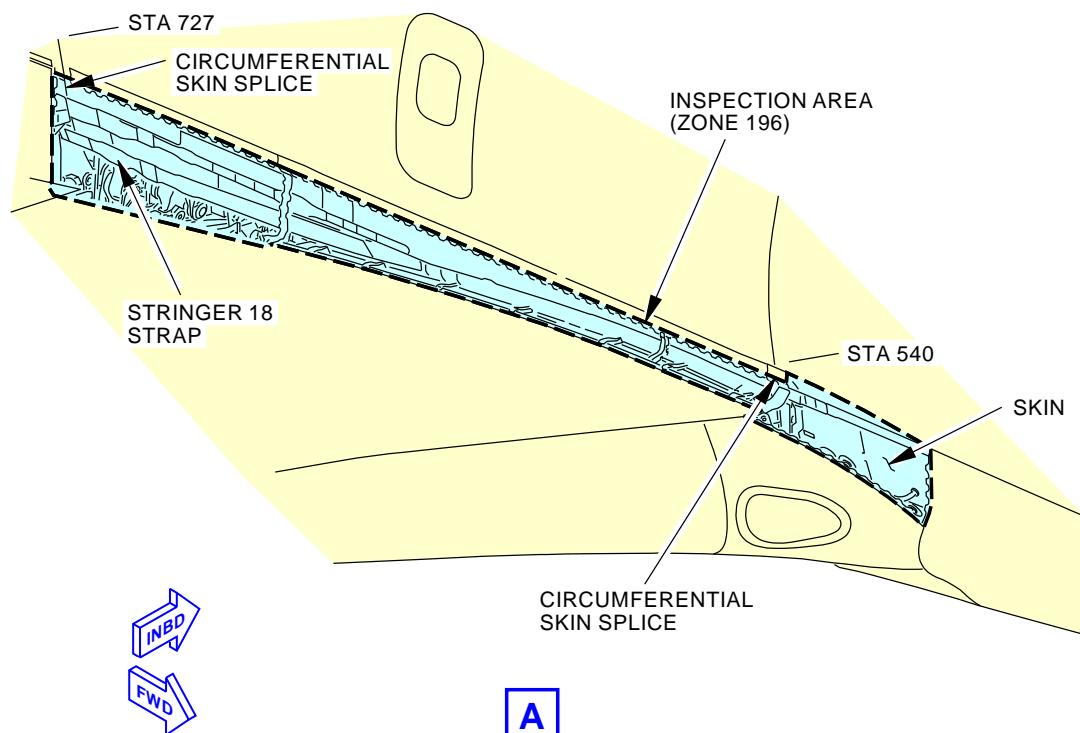
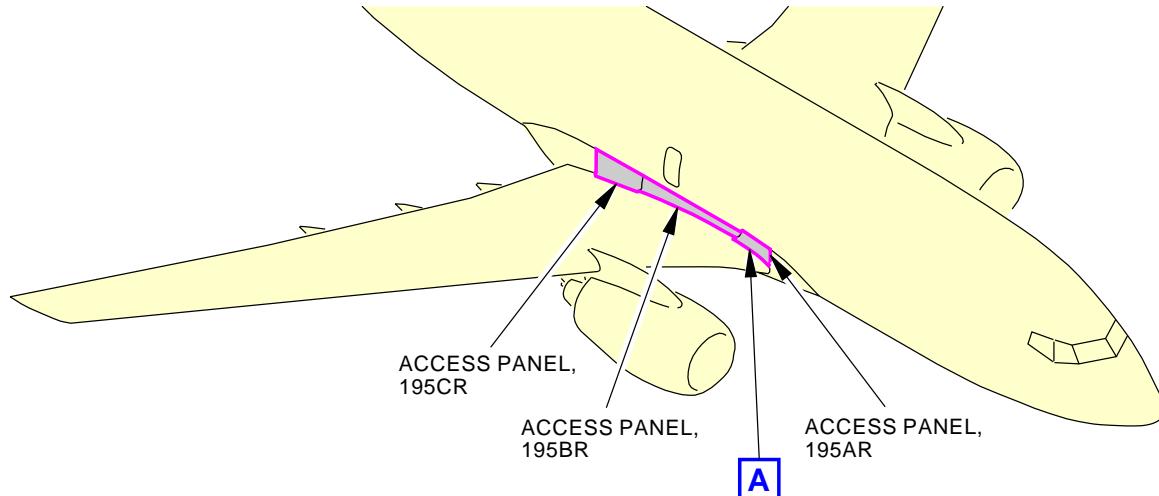
**Internal - General Visual: Area Under Left Wing-to-Body Fairing
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER WING-TO-BODY FAIRING (ABOVE WING)
		D633A109-AKS 53-280-00-01

Page 5 of 6
Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-280-00-01

MPD ITEM
53-280-00

2083297 S0000437151_V3

**Internal - General Visual: Area Under Right Wing-to-Body Fairing
Figure 2**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA UNDER WING-TO-BODY FAIRING (ABOVE WING)
		D633A109-AKS 53-280-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE OVERWING EMERGENCY EXIT CUTOUT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-290-00-01 RELATED CARD
TAIL NUMBER	WORK AREA UPR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 9 YR 24000 FC	REPEAT 8 YR 24000 FC	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL	NOTE			ENGINE ALL
		ACCESS 832 833 842 843			ZONE 231 232
		NOTE			

Inspect the automatic overwing exit cutout structure, fittings and stops;

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Open automatic overwing exits.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	OVERWING EMERGENCY EXIT CUTOUT
		D633A109-AKS 53-290-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-290-00-01										
				MECH INSP										
TASK 53-05-03-210-829														
1. EXTERNAL - GENERAL VISUAL: OVERWING EMERGENCY EXIT CUTOUT														
(Figure 1,Figure 2)														
A. Inspection														
SUBTASK 53-05-03-010-026														
(1) Open these access panels:														
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>832</td><td>Emergency Exit</td></tr><tr><td>833</td><td>Emergency Exit</td></tr><tr><td>842</td><td>Emergency Exit</td></tr><tr><td>843</td><td>Emergency Exit</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	832	Emergency Exit	833	Emergency Exit	842	Emergency Exit	843	Emergency Exit
<u>Number</u>	<u>Name/Location</u>													
832	Emergency Exit													
833	Emergency Exit													
842	Emergency Exit													
843	Emergency Exit													
NOTE: Open automatic overwing exits.														
SUBTASK 53-05-03-210-029														
(2) Do a General Visual inspection of the automatic overwing exit cutout structure, fittings and stops.														
SUBTASK 53-05-03-910-034														
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.														
SUBTASK 53-05-03-410-026														
(4) Close these access panels:														
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>832</td><td>Emergency Exit</td></tr><tr><td>833</td><td>Emergency Exit</td></tr><tr><td>842</td><td>Emergency Exit</td></tr><tr><td>843</td><td>Emergency Exit</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	832	Emergency Exit	833	Emergency Exit	842	Emergency Exit	843	Emergency Exit
<u>Number</u>	<u>Name/Location</u>													
832	Emergency Exit													
833	Emergency Exit													
842	Emergency Exit													
843	Emergency Exit													
———— END OF TASK ————														

EFFECTIVITY AKS ALL	SOURCE MRB	OVERWING EMERGENCY EXIT CUTOUT	
		D633A109-AKS 53-290-00-01	Page 2 of 7 Feb 15/2015

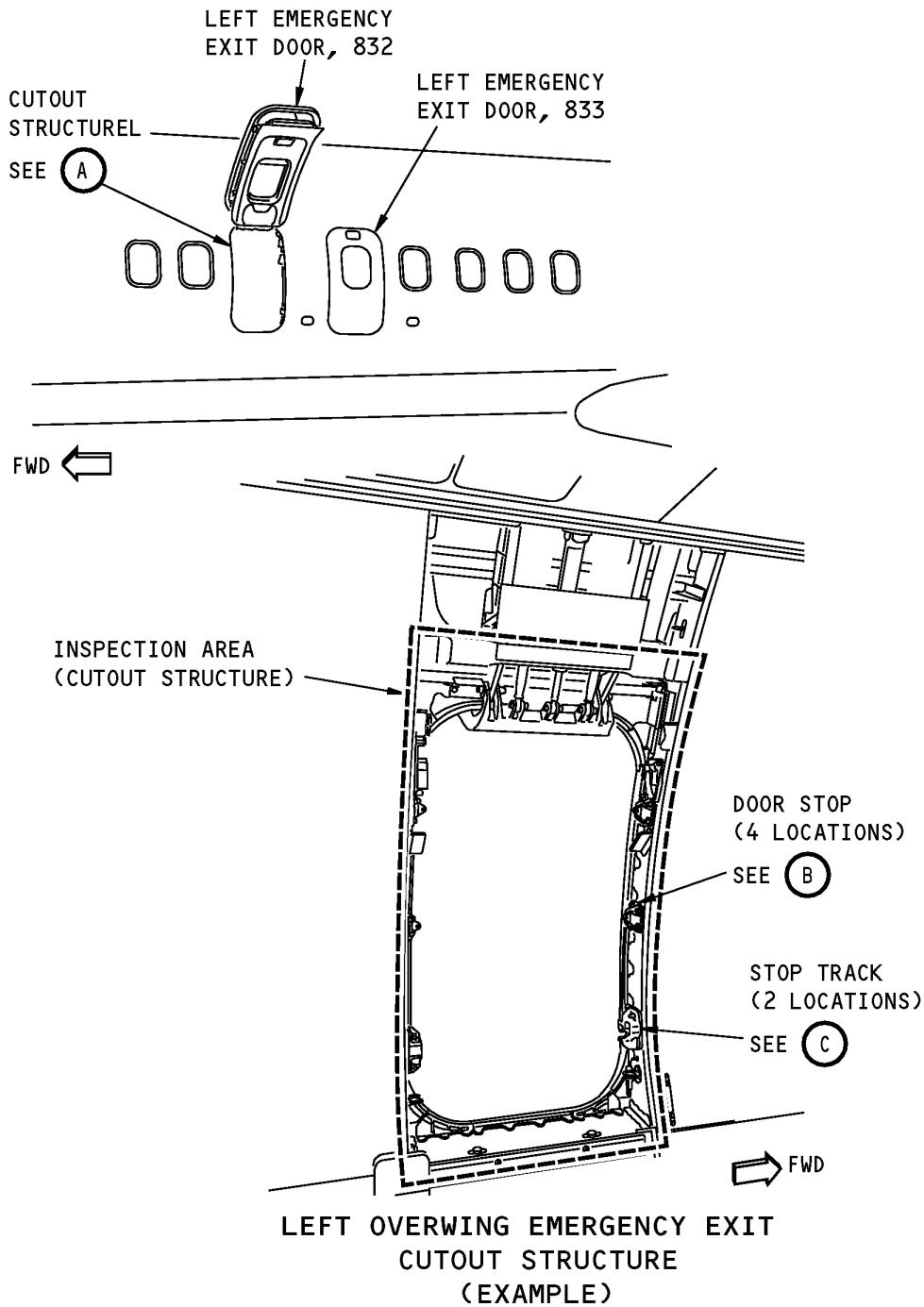
AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-290-00-01
				MECH INSP
TASK 51-05-01-210-809				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-078				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-079				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-080				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-081				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-082				
(5) CPCP Basic Task Item 5 is not applicable.				
SUBTASK 51-05-01-210-100				
(6) Do the CPCP Basic Task Item 6 (Not Applicable)				
SUBTASK 51-05-01-210-084				
(7) CPCP Basic Task Item 7 is not applicable.				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	OVERWING EMERGENCY EXIT CUTOUT	
		D633A109-AKS 53-290-00-01	Page 3 of 7 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-290-00-01



A
Left Overwing Emergency Exit Cutout
Figure 1 (Sheet 1 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	OVERWING EMERGENCY EXIT CUTOUT
		D633A109-AKS 53-290-00-01

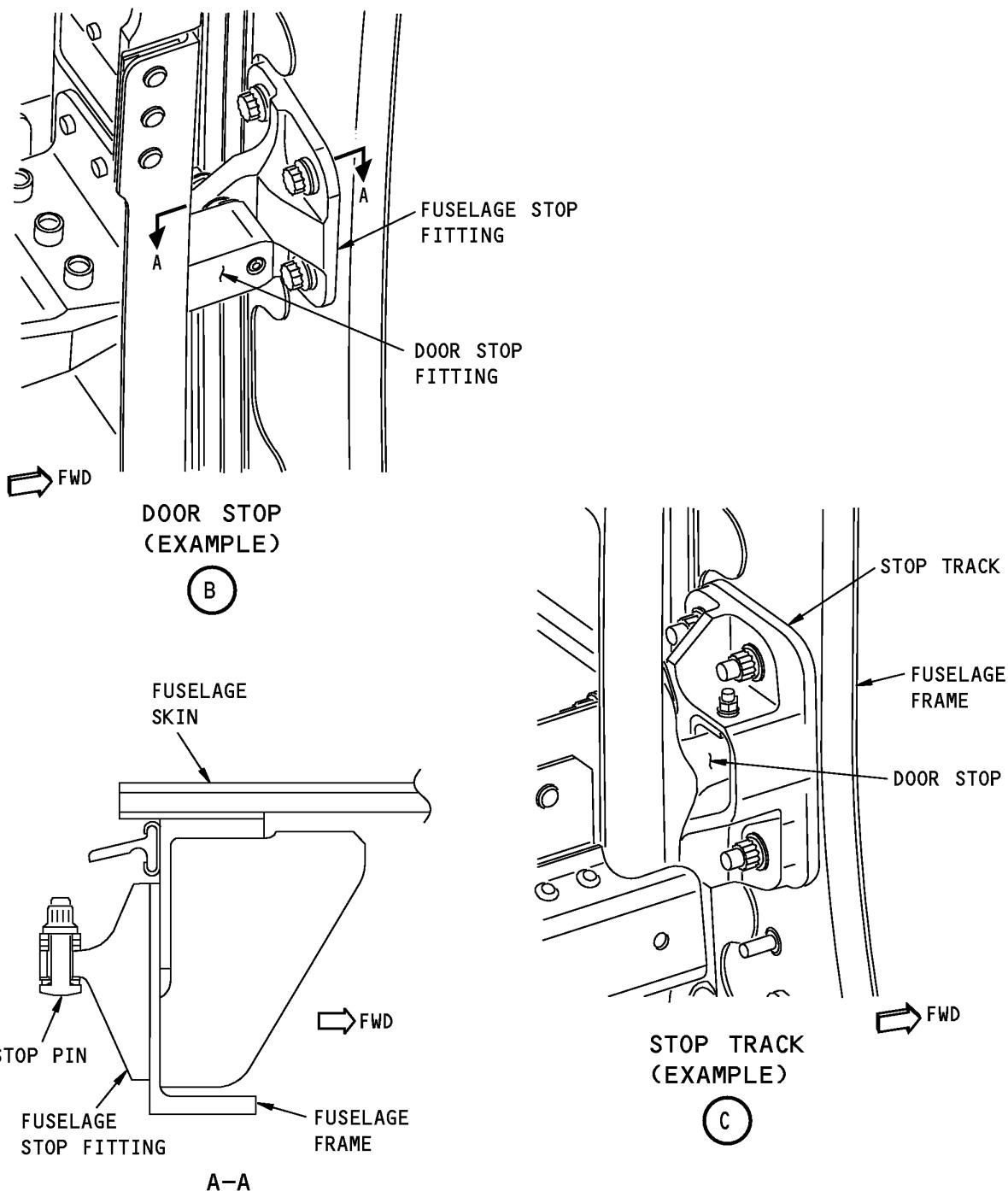
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-290-00-01

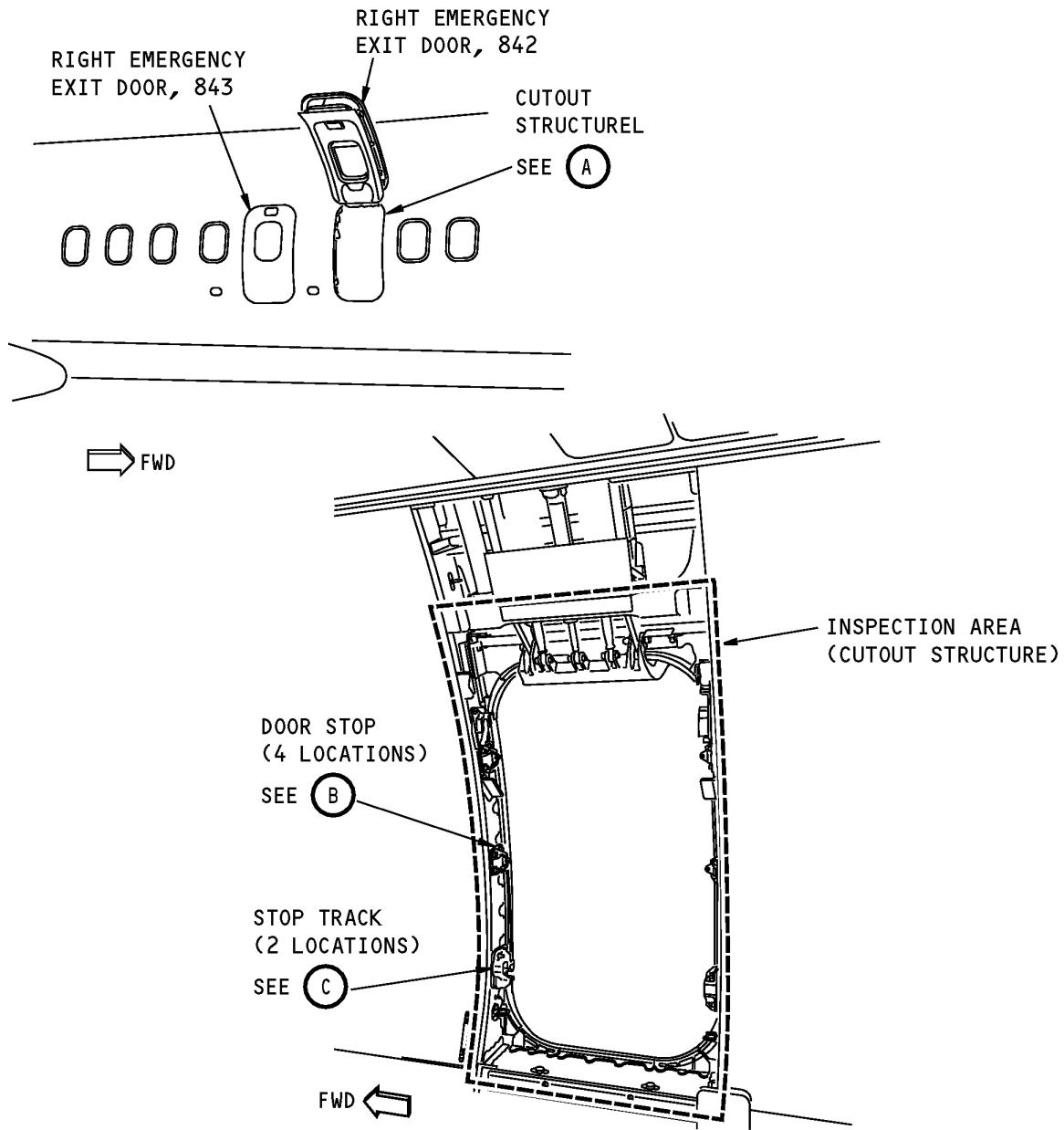
Left Overwing Emergency Exit Cutout
Figure 1 (Sheet 2 of 2)

484672 S0000143827_V2

EFFECTIVITY AKS ALL	SOURCE MRB	OVERWING EMERGENCY EXIT CUTOUT
		D633A109-AKS 53-290-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-290-00-01
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**RIGHT OVERWING EMERGENCY EXIT
CUTOUT STRUCTURE
(EXAMPLE)**

A

**Right Overwing Emergency Exit Cutout
Figure 2 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	OVERWING EMERGENCY EXIT CUTOUT
		D633A109-AKS 53-290-00-01

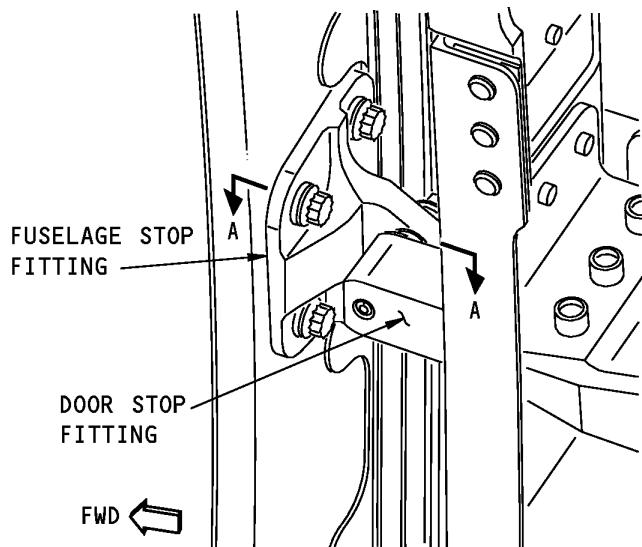
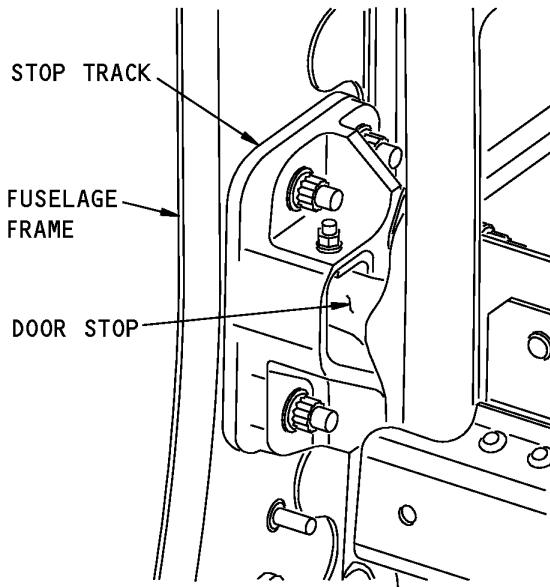
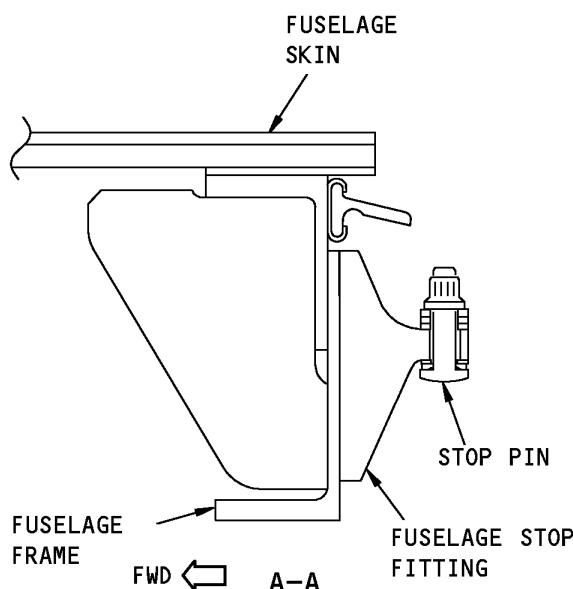
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-290-00-01**DOOR STOP
(EXAMPLE)****B****STOP TRACK
(EXAMPLE)****C****Right Overwing Emergency Exit Cutout
Figure 2 (Sheet 2 of 2)**

484701 S0000143840_V2

EFFECTIVITY
AKS ALLSOURCE
MRB**OVERWING EMERGENCY EXIT CUTOUT****D633A109-AKS
53-290-00-01****Page 7 of 7
Oct 15/2014**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES			BOEING CARD NO.
DATE	TASK DETAILED				53-310-00-01 RELATED CARD
TAIL NUMBER	WORK AREA UPR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 36 MO 6600 FC	REPEAT 36 MO 6600 FC	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL	NOTE			ENGINE ALL
		ACCESS 831			ZONE 221
		NOTE			

Inspect door frames, stops, latches and hinges on the forward door cutout surround structure.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Open forward entry door.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES D633A109-AKS 53-310-00-01
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-310-00-01			
				MECH INSP			
TASK 53-05-03-211-805							
1. EXTERNAL - DETAILED: FORWARD ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES (Figure 1)							
A. Inspection							
SUBTASK 53-05-03-010-059							
(1) Open this access panel:							
Number Name/Location							
831 Forward Entry Door							
NOTE: Open forward entry door.							
SUBTASK 53-05-03-211-005							
(2) Do a Detailed inspection of the door frames, stops, latches and hinges on the forward door cutout surround structure.							
SUBTASK 53-05-03-910-036							
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.							
SUBTASK 53-05-03-410-059							
(4) Close this access panel:							
Number Name/Location							
831 Forward Entry Door							
———— END OF TASK ————							
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES					
		D633A109-AKS					
		53-310-00-01					
Page 2 of 4							
Feb 15/2015							

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-310-00-01
				MECH INSP
TASK 51-05-01-210-809				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-078				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-079				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-080				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-081				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-082				
(5) CPCP Basic Task Item 5 is not applicable.				
SUBTASK 51-05-01-210-100				
(6) Do the CPCP Basic Task Item 6 (Not Applicable)				
SUBTASK 51-05-01-210-084				
(7) CPCP Basic Task Item 7 is not applicable.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES		
		D633A109-AKS		
		53-310-00-01		
		Page 3 of 4		
		Oct 15/2015		

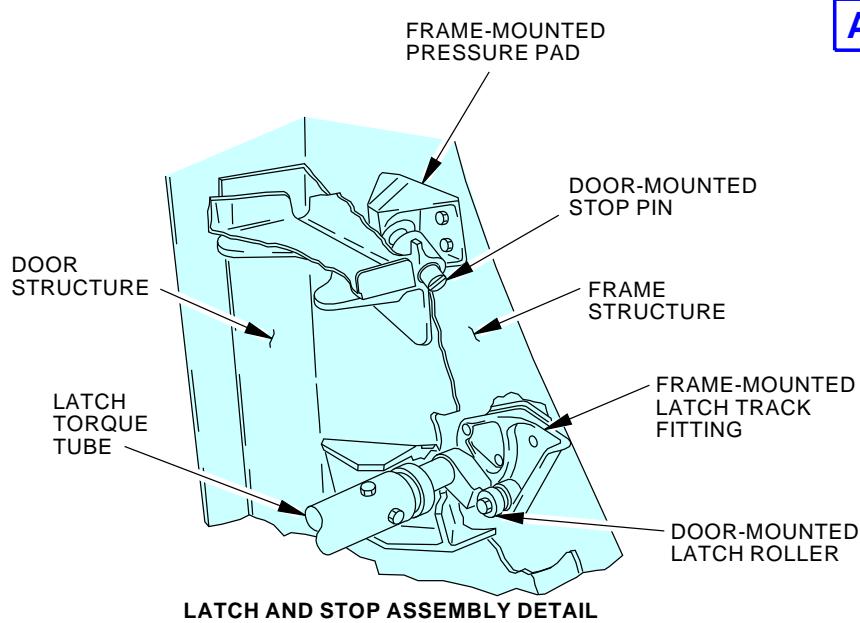
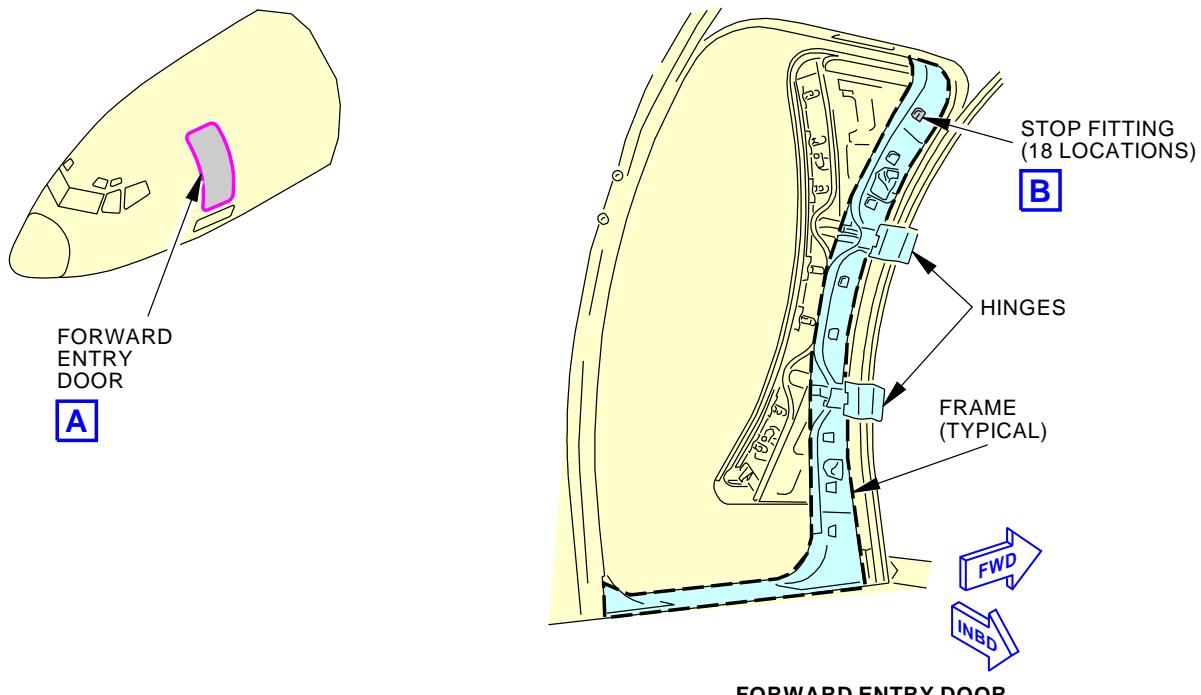
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-310-00-01

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**External - Forward Entry Door Frame, Stop, Latches and Hinges
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**FORWARD ENTRY DOOR FRAME, STOPS, LATCHES AND
HINGES****D633A109-AKS
53-310-00-01****Page 4 of 4
Oct 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES			BOEING CARD NO.
DATE	TASK DETAILED				53-310-00-02 RELATED CARD
TAIL NUMBER	WORK AREA UPR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 36 MO 6600 FC	REPEAT 36 MO 6600 FC	APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 841			ZONE 222
		NOTE			

Inspect door frames, stops, latches and hinges on forward galley door cutout surround structure.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Open forward galley service door.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES
		D633A109-AKS 53-310-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-310-00-02
TASK 53-05-03-211-806				MECH INSP

1. EXTERNAL - DETAILED: FORWARD GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-060

(1) Open this access panel:

Number Name/Location

841 Forward Galley Service Door

NOTE: Open forward galley service door.

SUBTASK 53-05-03-211-006

(2) Do a Detailed inspection of the door frames, stops, latches and hinges on forward galley door cutout surround structure.

SUBTASK 53-05-03-910-037

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.

SUBTASK 53-05-03-410-060

(4) Close this access panel:

Number Name/Location

841 Forward Galley Service Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES
		D633A109-AKS 53-310-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-310-00-02	MECH	INSP
TASK 51-05-01-210-809						
2. 737-6789 Basic Task Description						
A. CPCP Basic Task						
SUBTASK 51-05-01-210-078						
(1) CPCP Basic Task Item 1 is not applicable.						
SUBTASK 51-05-01-210-079						
(2) Do the CPCP Basic Task Item 2 as follows:						
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.						
SUBTASK 51-05-01-210-080						
(3) Do the CPCP Basic Task Item 3 as follows:						
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.						
SUBTASK 51-05-01-210-081						
(4) Do the CPCP Basic Task item 4 as follows:						
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.						
SUBTASK 51-05-01-210-082						
(5) CPCP Basic Task Item 5 is not applicable.						
SUBTASK 51-05-01-210-100						
(6) Do the CPCP Basic Task Item 6 (Not Applicable)						
SUBTASK 51-05-01-210-084						
(7) CPCP Basic Task Item 7 is not applicable.						
— END OF TASK —						

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES	
		D633A109-AKS 53-310-00-02	Page 3 of 4 Oct 15/2015

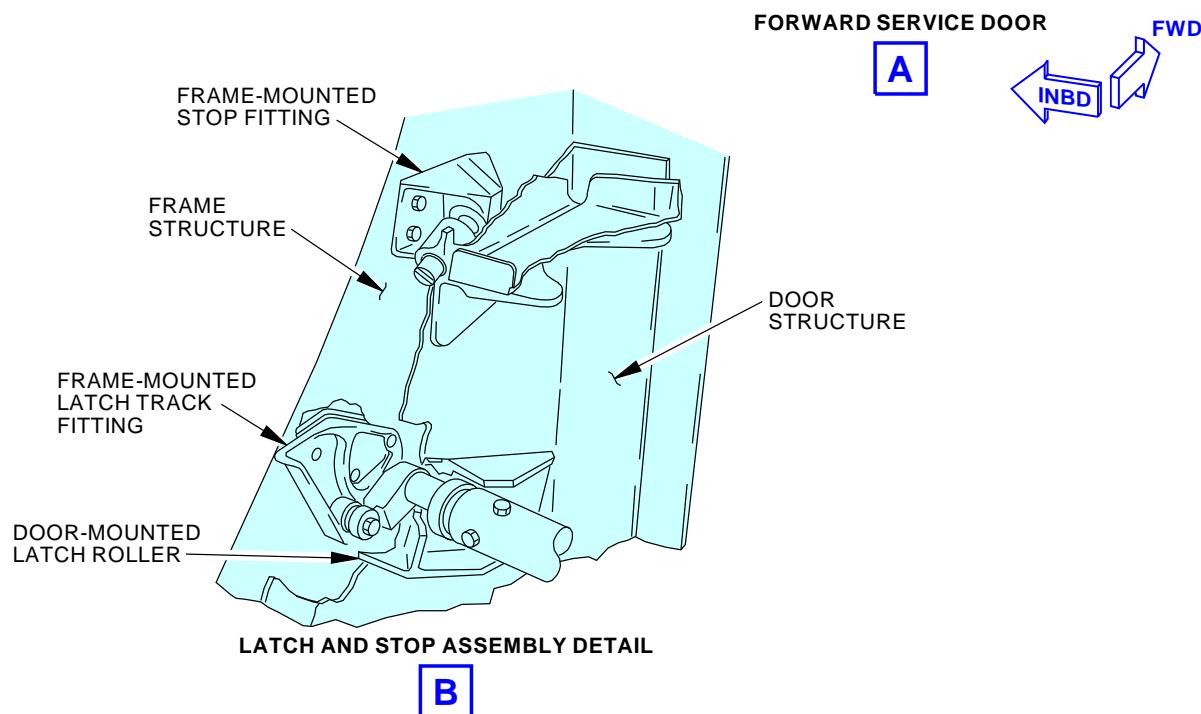
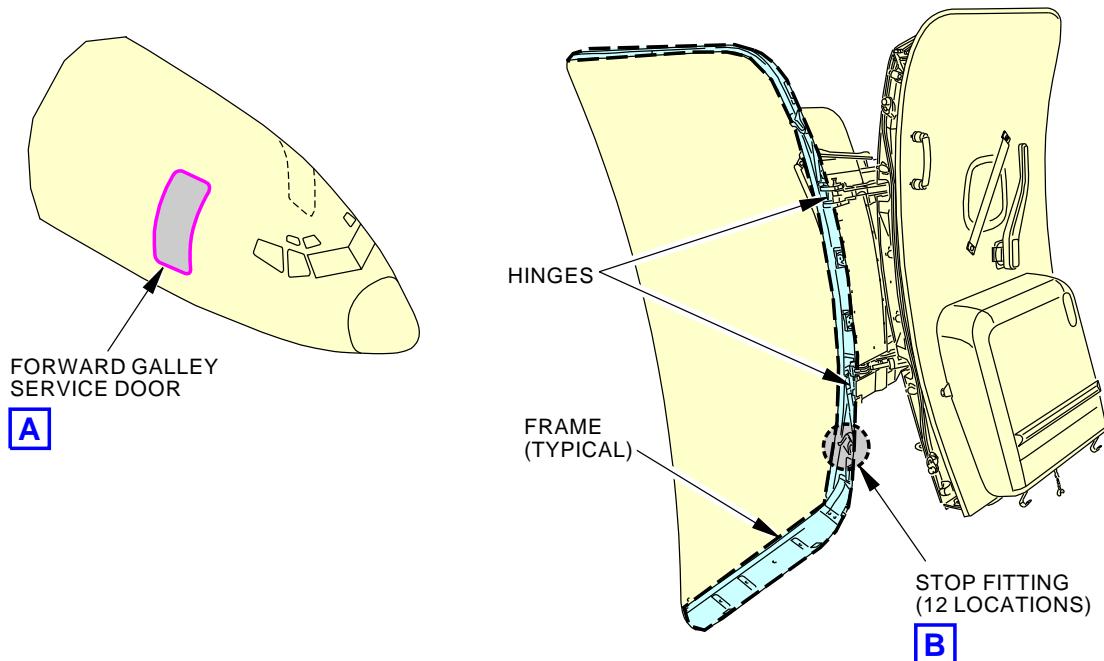
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-310-00-02

External - Forward Galley Service Door Frame, Stops, Latches and Hinges
Figure 1

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EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES
		D633A109-AKS 53-310-00-02

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES			BOEING CARD NO. 53-310-00-03	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA UPR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 36 MO 6600 FC	REPEAT 36 MO 6600 FC	APPLICABILITY	
STATION	SKILL AIRPL	NOTE			AIRPLANE ALL	ENGINE ALL
		ACCESS 834			ZONE 241	
		NOTE				

Inspect door frames, stops, latches and hinges on aft entry cutout surround structure.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Open aft entry door.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	AFT ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES
		D633A109-AKS 53-310-00-03

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-310-00-03
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TASK 53-05-03-211-807

1. EXTERNAL - DETAILED: AFT ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-061

- (1) Open this access panel:

Number **Name/Location**

834 Aft Entry Door

SUBTASK 53-05-03-211-007

- (2) Do a Detailed inspection of the door frames, stops, latches and hinges on aft entry cutout surround structure.

SUBTASK 53-05-03-910-038

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.

SUBTASK 53-05-03-410-061

- (4) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
---------------	----------------------

834 Aft Entry Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AFT ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES
		D633A109-AKS 53-310-00-03

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-310-00-03
				MECH INSP
TASK 51-05-01-210-809				
2. 737-6789 Basic Task Description				
<p>A. CPCP Basic Task</p> <p>SUBTASK 51-05-01-210-078</p> <p>(1) CPCP Basic Task Item 1 is not applicable.</p> <p>SUBTASK 51-05-01-210-079</p> <p>(2) Do the CPCP Basic Task Item 2 as follows:</p> <p class="list-item-l1">(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.</p> <p>SUBTASK 51-05-01-210-080</p> <p>(3) Do the CPCP Basic Task Item 3 as follows:</p> <p class="list-item-l1">(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.</p> <p>SUBTASK 51-05-01-210-081</p> <p>(4) Do the CPCP Basic Task item 4 as follows:</p> <p class="list-item-l1">(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.</p> <p>SUBTASK 51-05-01-210-082</p> <p>(5) CPCP Basic Task Item 5 is not applicable.</p> <p>SUBTASK 51-05-01-210-100</p> <p>(6) Do the CPCP Basic Task Item 6 (Not Applicable)</p> <p>SUBTASK 51-05-01-210-084</p> <p>(7) CPCP Basic Task Item 7 is not applicable.</p>				
— END OF TASK —				

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AFT ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES
		D633A109-AKS 53-310-00-03

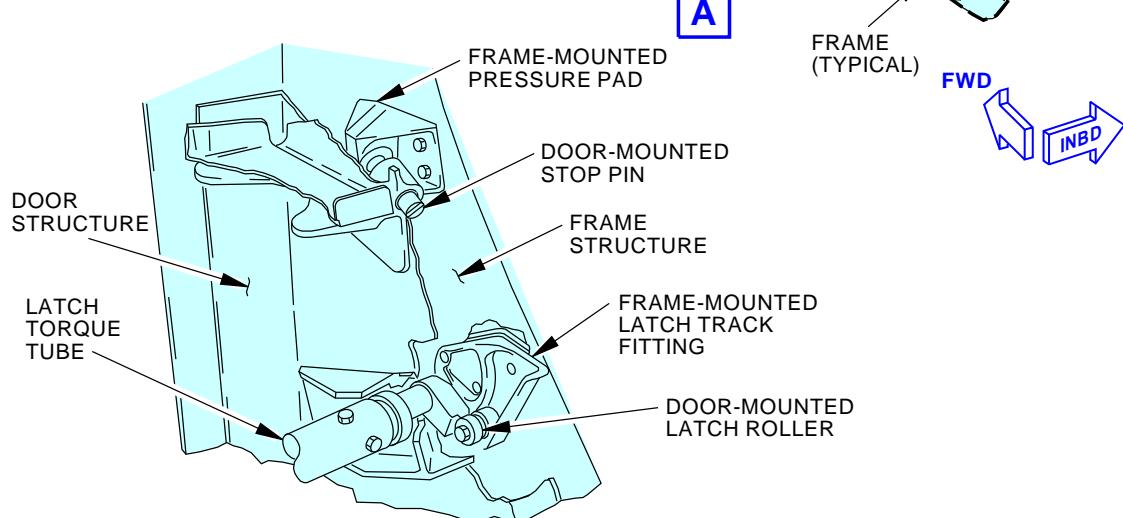
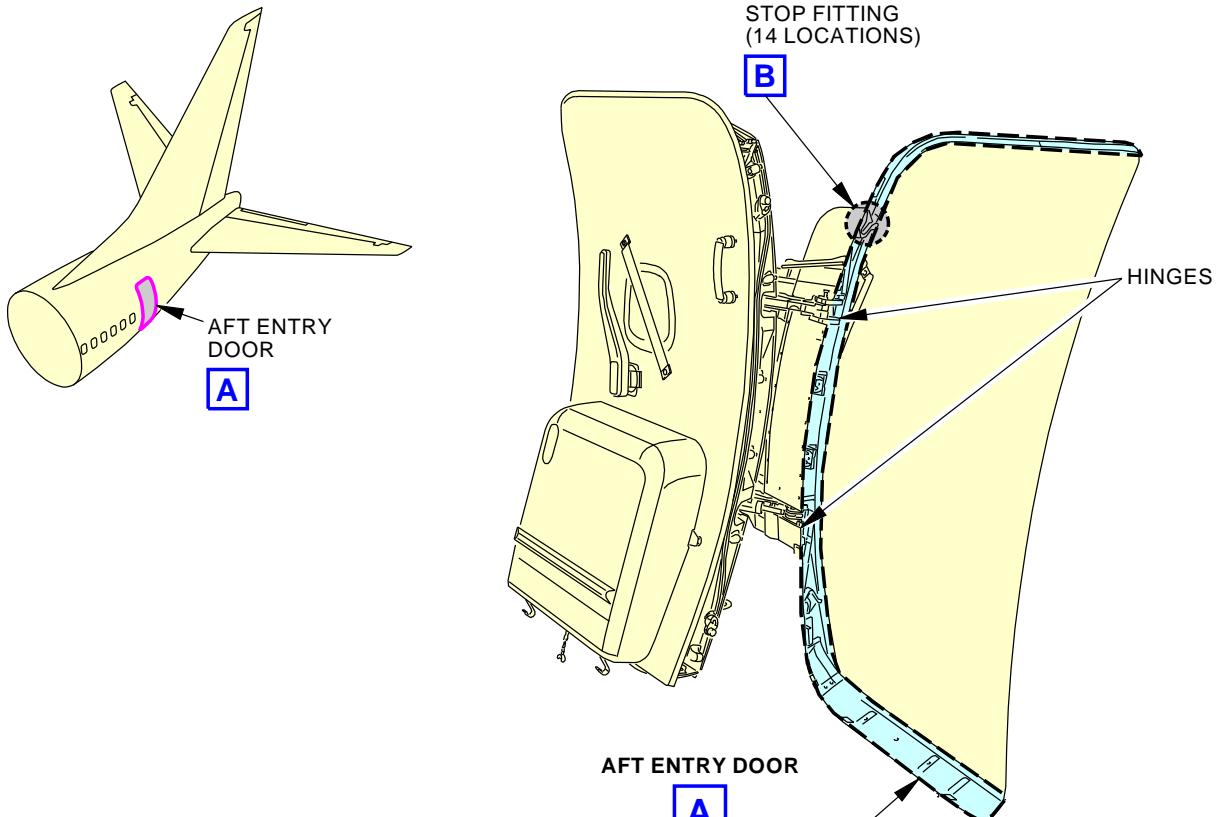
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-310-00-03**LATCH AND STOP ASSEMBLY DETAIL****AFT Entry Door Frame, Stops, Latches and Hinges
Figure 1**

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EFFECTIVITY
AKS ALLSOURCE
MRB**AFT ENTRY DOOR FRAME, STOPS, LATCHES AND HINGES****D633A109-AKS
53-310-00-03****Page 4 of 4
Oct 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES			BOEING CARD NO. 53-310-00-04	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA UPR FUSELAGE	VERSION 1.1 1.2	THRESHOLD 36 MO 6600 FC	REPEAT 36 MO 6600 FC	APPLICABILITY	
STATION	SKILL AIRPL	NOTE			AIRPLANE ALL	ENGINE ALL
		ACCESS 844			ZONE 242	
		NOTE				

Inspect door frames, stops, latches and hinges on aft galley door cutout surround structure.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Open aft galley door.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

EFFECTIVITY AKS ALL	SOURCE MRB	AFT GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES
		D633A109-AKS 53-310-00-04

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-310-00-04
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TASK 53-05-03-211-808

1. EXTERNAL - DETAILED: AFT GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-062

- (1) Open this access panel:

Number **Name/Location**

844 Aft Galley Service Door

NOTE: Open aft galley door.

SUBTASK 53-05-03-211-008

- (2) Do a Detailed inspection of the door frames, stops, latches and hinges on aft galley door cutout surround structure.

SUBTASK 53-05-03-910-039

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-809.

SUBTASK 53-05-03-410-062

- (4) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
---------------	----------------------

844 Aft Galley Service Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AFT GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES
		D633A109-AKS 53-310-00-04

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-310-00-04
				MECH INSP
TASK 51-05-01-210-809				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-078				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-079				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-080				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-081				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-082				
(5) CPCP Basic Task Item 5 is not applicable.				
SUBTASK 51-05-01-210-100				
(6) Do the CPCP Basic Task Item 6 (Not Applicable)				
SUBTASK 51-05-01-210-084				
(7) CPCP Basic Task Item 7 is not applicable.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE MRB	AFT GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES		
		D633A109-AKS 53-310-00-04		

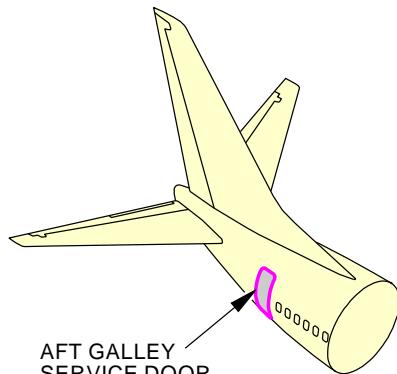
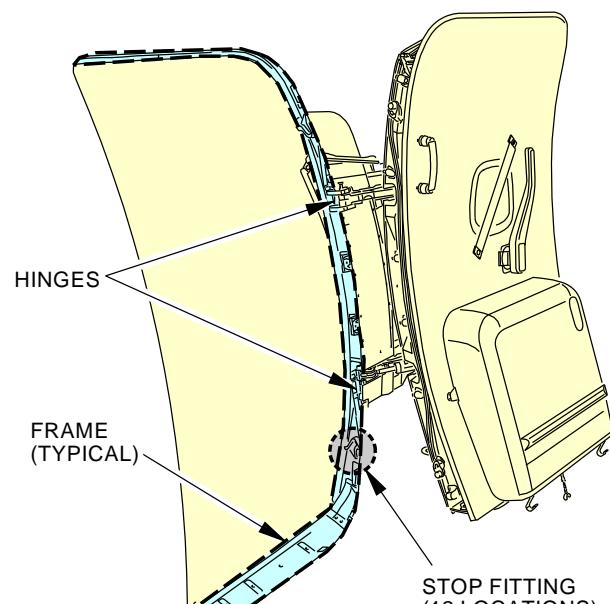
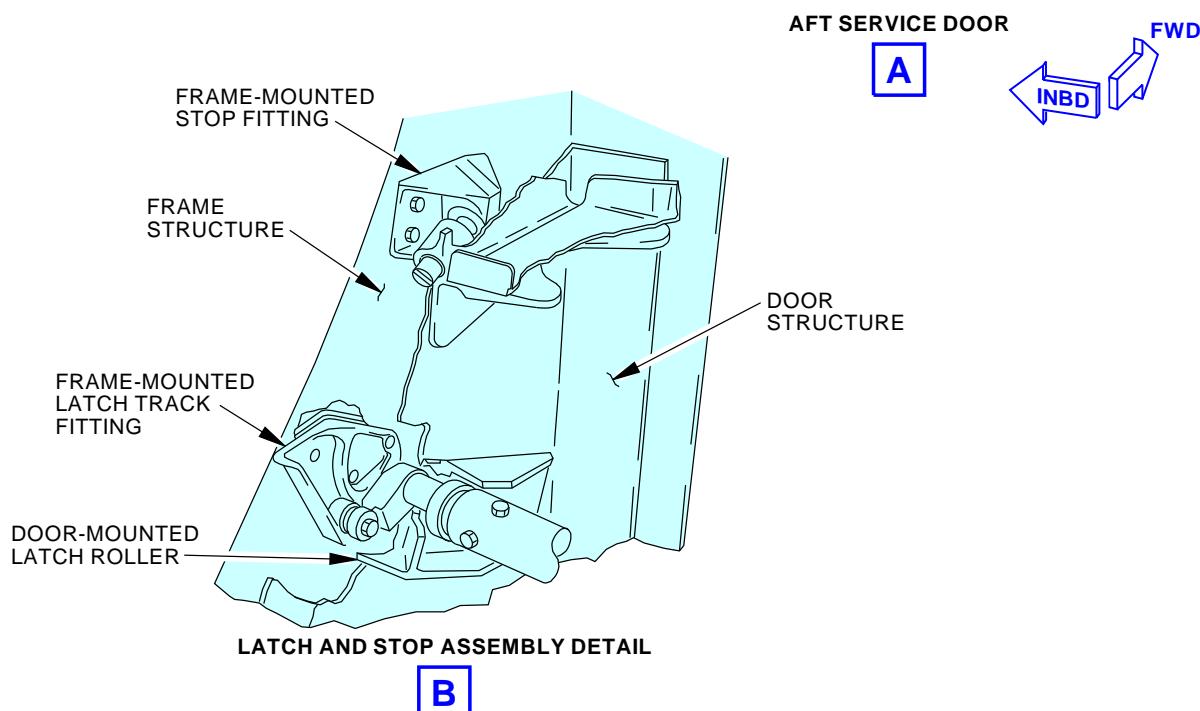
AKS737-600/700/800/900
TASK CARDS

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-310-00-04**A****B**Aft Galley Door Frame, Stop, Latches and Hinges
Figure 1

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EFFECTIVITY
AKS ALLSOURCE
MRB**AFT GALLEY SERVICE DOOR FRAME, STOPS, LATCHES AND HINGES**D633A109-AKS
53-310-00-04Page 4 of 4
Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FLIGHT COMPARTMENT FROM STA 178 TO 270			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-330-00-01
TAIL NUMBER	WORK AREA CREW CABIN	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC	AIRPLANE ALL ENGINE ALL
		ACCESS S2101			ZONE 211 212 221 222
		NOTE			

Inspect flight compartment, including skin panels (skins, frames, stringers), circumferential skin and stringer splice, crew cabin window cutout structure, and structure adjacent to ground block behind rudder pedal.

Inspection area does not include: Forward side of frame at STA 259.5 and structure 3 inches forward of STA 259.5; BL 0 + 4 inches (Left and Right); Forward and aft side of Frame 259.5 and structure from STA 249 to STA 263 between floor and S-5L (excluding window and window frame structure); Forward side of Frame 259.5 and structure 3 inches forward of STA 259.5 between floor and S-3R; Structure forward of STA 203.8 to STA 178 and from floor up to window frame (except structure adjacent to ground block behind rudder pedal); skin, frames and stringers above P5 panel.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove glare shield, liners, overhead units and panels as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270
		D633A109-AKS 53-330-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-330-00-01				
				MECH INSP				
TASK 53-05-03-210-830								
1. INTERNAL - GENERAL VISUAL: FLIGHT COMPARTMENT FROM STA 178 TO 270								
A. Inspection								
SUBTASK 53-05-03-010-078								
(1) Special Access:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>S2101</td><td>Flight Compartment Inspection</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	S2101	Flight Compartment Inspection
<u>Number</u>	<u>Name/Location</u>							
S2101	Flight Compartment Inspection							
<p><u>NOTE:</u> Remove glare shield, liners, overhead units and panels as required. Remove/displace insulation blankets as required.</p>								
SUBTASK 53-05-03-210-030								
(2) Do a General Visual inspection of the flight compartment from Sta 178 to 270, including skin panels (skins, frames, stringers), circumferential skin and stringer splice, crew cabin window cutout structure, and structure adjacent to ground block behind rudder pedal. Inspection area does not include: Forward side of frame at STA 259.5 and structure 3 inches forward of STA 259.5; BL 0 + 4 inches (Left and Right); Forward and aft side of Frame 259.5 and structure from STA 249 to STA 263 between floor and S-5L (excluding window and window frame structure); Forward side of Frame 259.5 and structure 3 inches forward of STA 259.5 between floor and S-3R; Structure forward of STA 203.8 to STA 178 and from floor up to window frame (except structure adjacent to ground block behind rudder pedal); skin, frames and stringers above P5 panel.								
SUBTASK 53-05-03-910-040								
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.								
— END OF TASK —								

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270
		D633A109-AKS 53-330-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-330-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270	
		D633A109-AKS 53-330-00-01	Page 3 of 6 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-330-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-077

(7) Do the CPCP Basic Task Item 7 as follows:

(a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

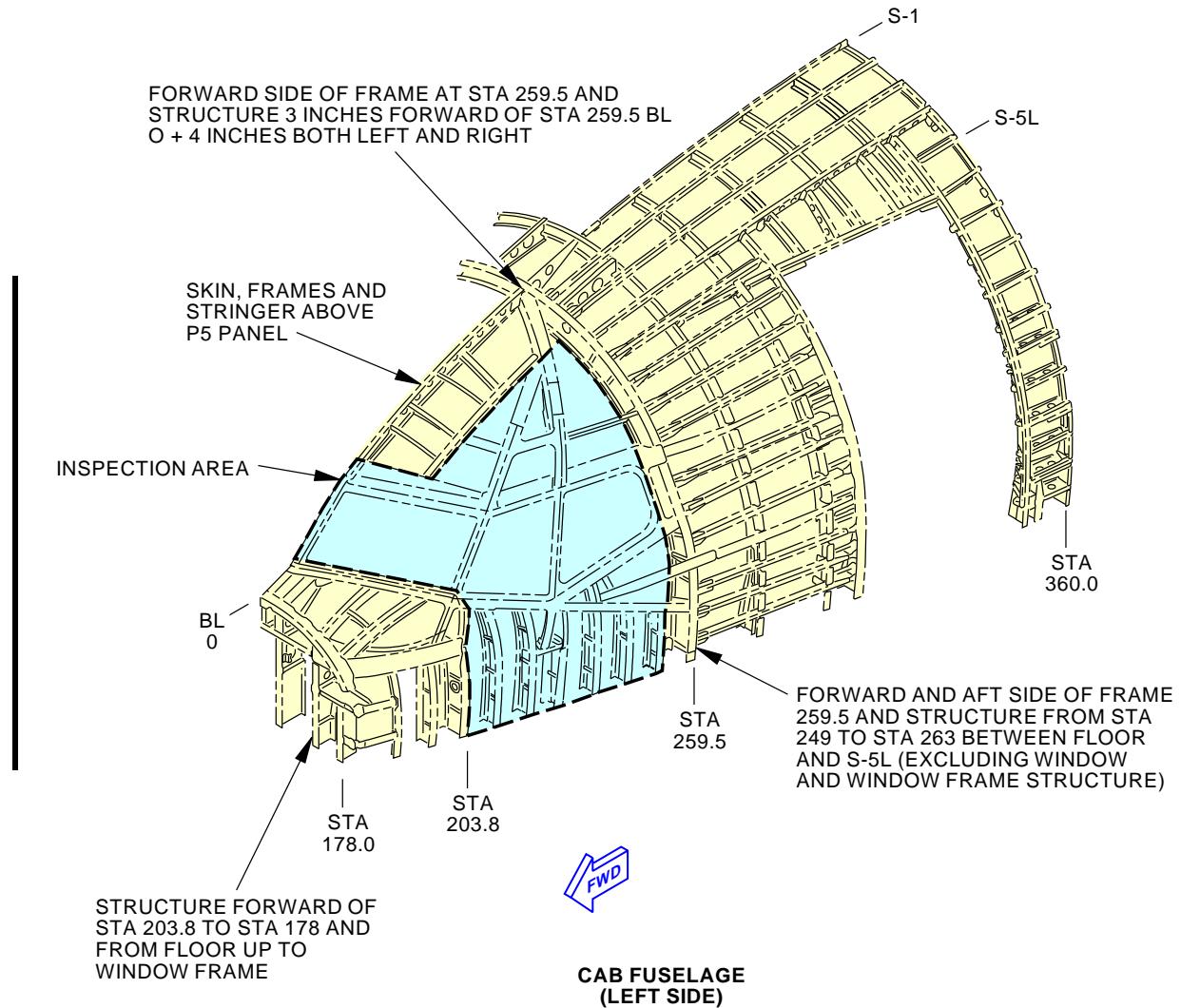
— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270
		D633A109-AKS 53-330-00-01

Page 4 of 6
Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-330-00-01
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MPD ITEM 53-330-00

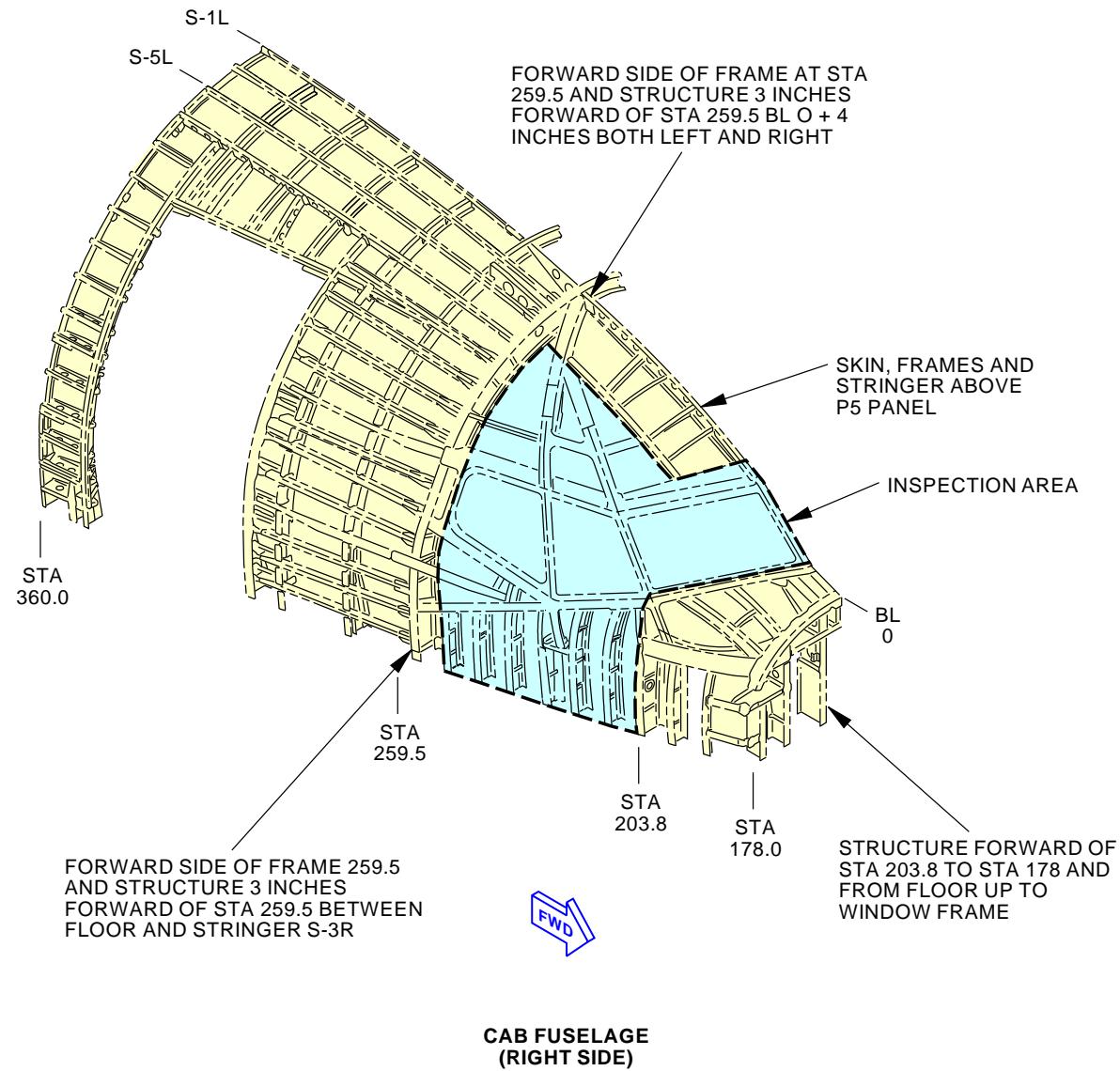
2293027 S0000519158_V3

FLIGHT COMPARTMENT FROM STA 178 TO 270
Figure 1 (Sheet 1 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270
		D633A109-AKS 53-330-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-330-00-01
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MPD ITEM 53-330-00

2293019 S0000519160_V3

**FLIGHT COMPARTMENT FROM STA 178 TO 270
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270
		D633A109-AKS 53-330-00-01

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Jun 15/2016

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FLIGHT COMPARTMENT FROM STA 178 TO 270			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-335-00-01
TAIL NUMBER	WORK AREA CREW CABIN	VERSION 1.1	THRESHOLD 20 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	50000 FC	24000 FC	AIRPLANE ALL ENGINE ALL
		ACCESS S2101			ZONE 211 212 221 222
		NOTE			

Inspect flight compartment, including skin panels (skins, frames, stringers), circumferential skin and stringer splice, crew cabin window cutout structure, and forward pressure bulkhead within the following areas: Forward side of frame at STA 259.5 and structure 3 inches forward of STA 259.5; BL 0 + 4 inches (Left and Right); Forward and aft side of Frame 259.5 and structure from STA 249 to STA 263 between floor and S-5L (excluding window and window frame structure); Forward side of Frame 259.5 and structure 3 inches forward of STA 259.5 between floor and S-3R; Structure forward of STA 203.8 to STA 178 and from floor up to window frame; and skin, frames and stringers above P5 panel.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove glare shield, liners, overhead units and panels as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270
		D633A109-AKS 53-335-00-01

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Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-335-00-01
				MECH INSP
TASK 51-05-01-210-808				
1. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270
		D633A109-AKS 53-335-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-335-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270	
		D633A109-AKS 53-335-00-01	Page 3 of 6 Oct 15/2014

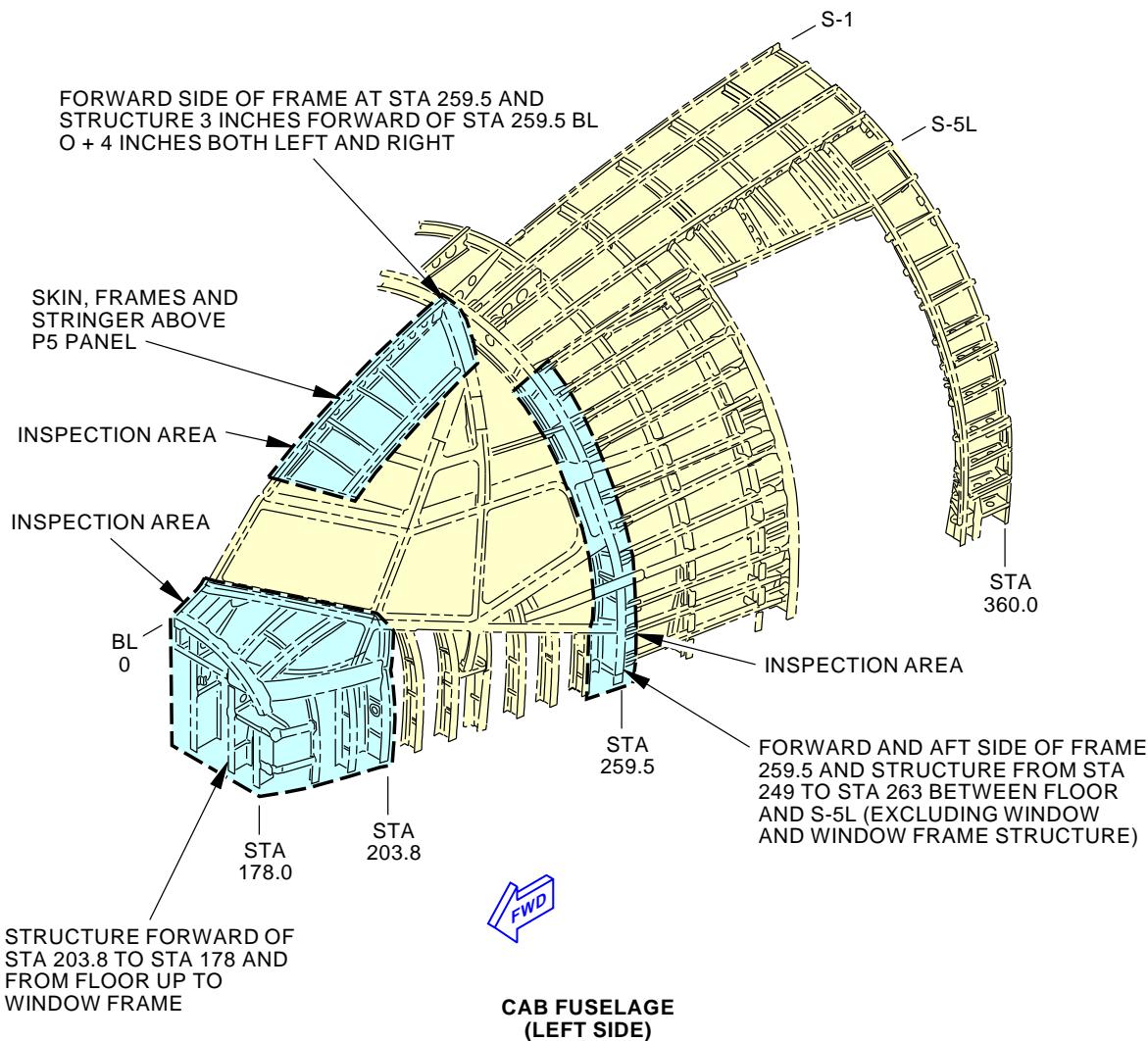
AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-335-00-01				
				MECH INSP				
TASK 53-05-03-210-861								
2. INTERNAL - GENERAL VISUAL: FLIGHT COMPARTMENT FROM STA 178 TO 270								
A. Inspection								
SUBTASK 53-05-03-010-079								
(1) Special Access:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>S2101</td><td>Flight Compartment Inspection</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	S2101	Flight Compartment Inspection
<u>Number</u>	<u>Name/Location</u>							
S2101	Flight Compartment Inspection							
<u>NOTE:</u> Remove glare shield, liners, overhead units and panels. Remove/displace insulation blankets as required.								
SUBTASK 53-05-03-210-049								
(2) Do a General Visual inspection of the flight compartment, including skin panels (skins, frames, stringers), circumferential skin and stringer splice, crew cabin window cutout structure, and forward pressure bulkhead within the following areas: Forward side of frame at STA 259.5 and structure 3 inches forward of STA 259.5; BL 0 + 4 inches (Left and Right); Forward and aft side of Frame 259.5 and structure from STA 249 to STA 263 between floor and S-5L (excluding window and window frame structure); Forward side of Frame 259.5 and structure 3 inches forward of STA 259.5 between floor and S-3R; Structure forward of STA 203.8 to STA 178 and from floor up to window frame; and skin, frames and stringers above P5 panel.								
SUBTASK 53-05-03-910-077								
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.								
— END OF TASK —								

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270	
		D633A109-AKS 53-335-00-01	Page 4 of 6 Jun 15/2016

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-335-00-01
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MPD ITEM 53-335-00

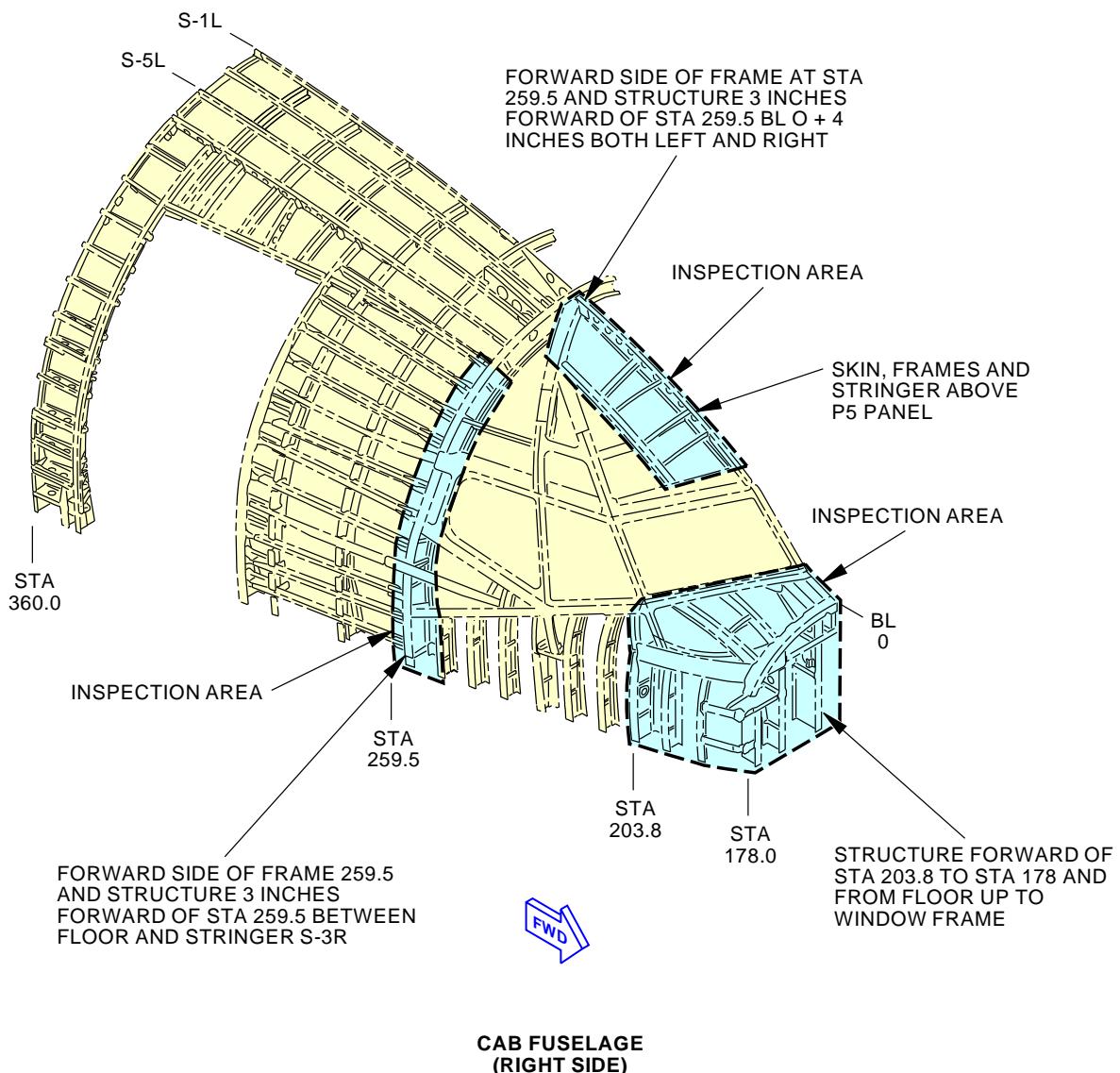
2293154 S0000519282_V2

**FLIGHT COMPARTMENT FROM STA 178 TO 270
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270
		D633A109-AKS 53-335-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-335-00-01
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MPD ITEM 53-335-00

2293157 S0000519280_V2

FLIGHT COMPARTMENT FROM STA 178 TO 270
Figure 1 (Sheet 2 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FROM STA 178 TO 270
		D633A109-AKS 53-335-00-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FLIGHT COMPARTMENT FLOOR STRUCTURE			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-340-00-01
TAIL NUMBER	WORK AREA CREW CABIN	VERSION 1.1	THRESHOLD 10 YR	REPEAT 10 YR	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS S2102			ZONE 211 212
		NOTE			

Inspect flight compartment floor structure.

ACCESS NOTE: Remove sidewalls and floor panels as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-340-00-01

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Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-340-00-01
TASK 53-05-03-210-831				MECH INSP

1. INTERNAL - GENERAL VISUAL: FLIGHT COMPARTMENT FLOOR STRUCTURE
(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-028

(1) Special Access:

<u>Number</u>	<u>Name/Location</u>
S1202	Forward Bilge Inspection

NOTE: Remove sidewalls and floor panels as required. Remove/displace insulation blankets as required.

SUBTASK 53-05-03-210-031

(2) Do a General Visual inspection of the flight compartment floor structure.

SUBTASK 53-05-03-910-041

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-340-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-340-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE	
		D633A109-AKS 53-340-00-01	Page 3 of 6 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-340-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-077

(7) Do the CPCP Basic Task Item 7 as follows:

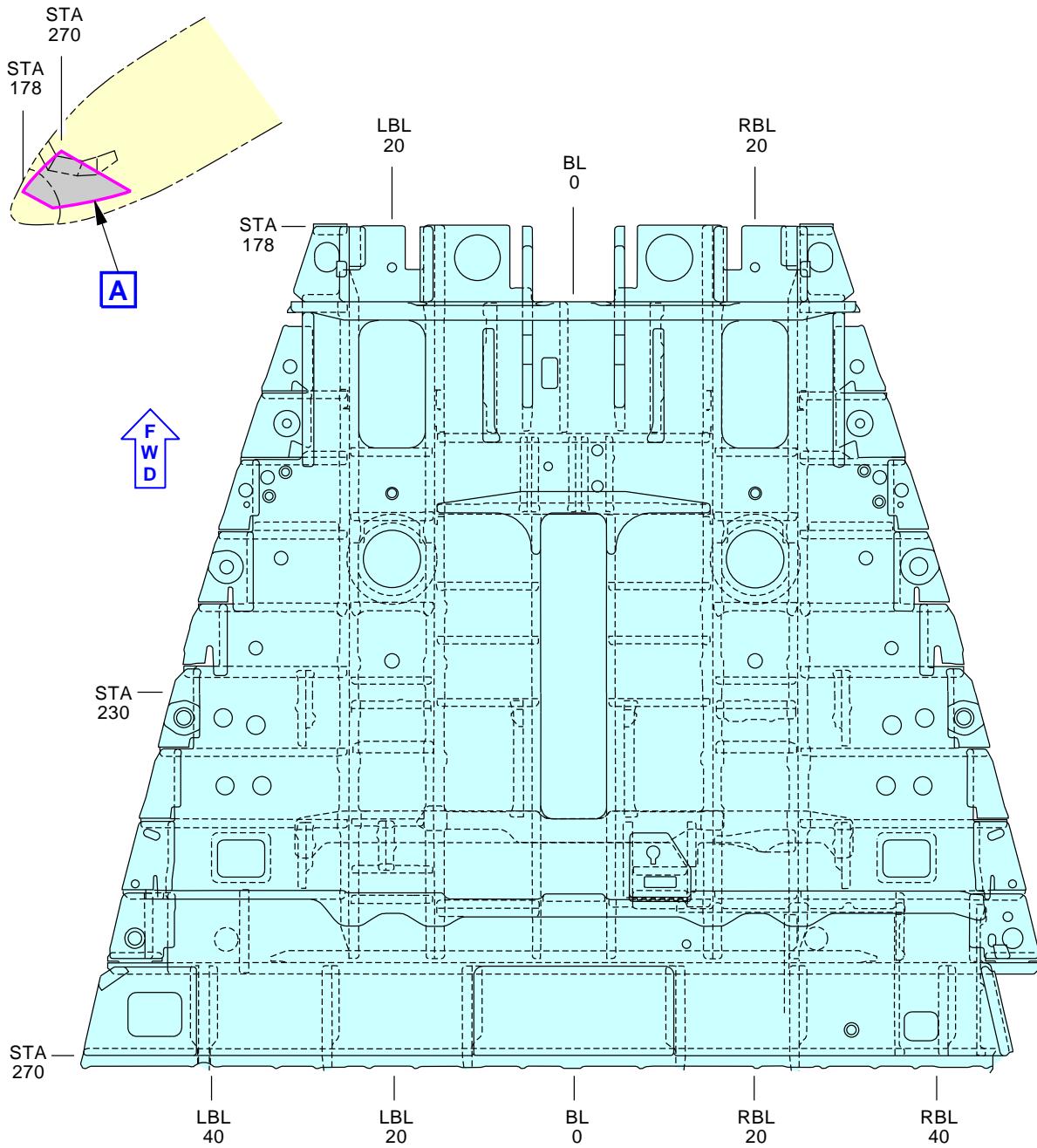
(a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-340-00-01

AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-340-00-01



**Flight Deck Floor Structure
Figure 1 (Sheet 1 of 2)**

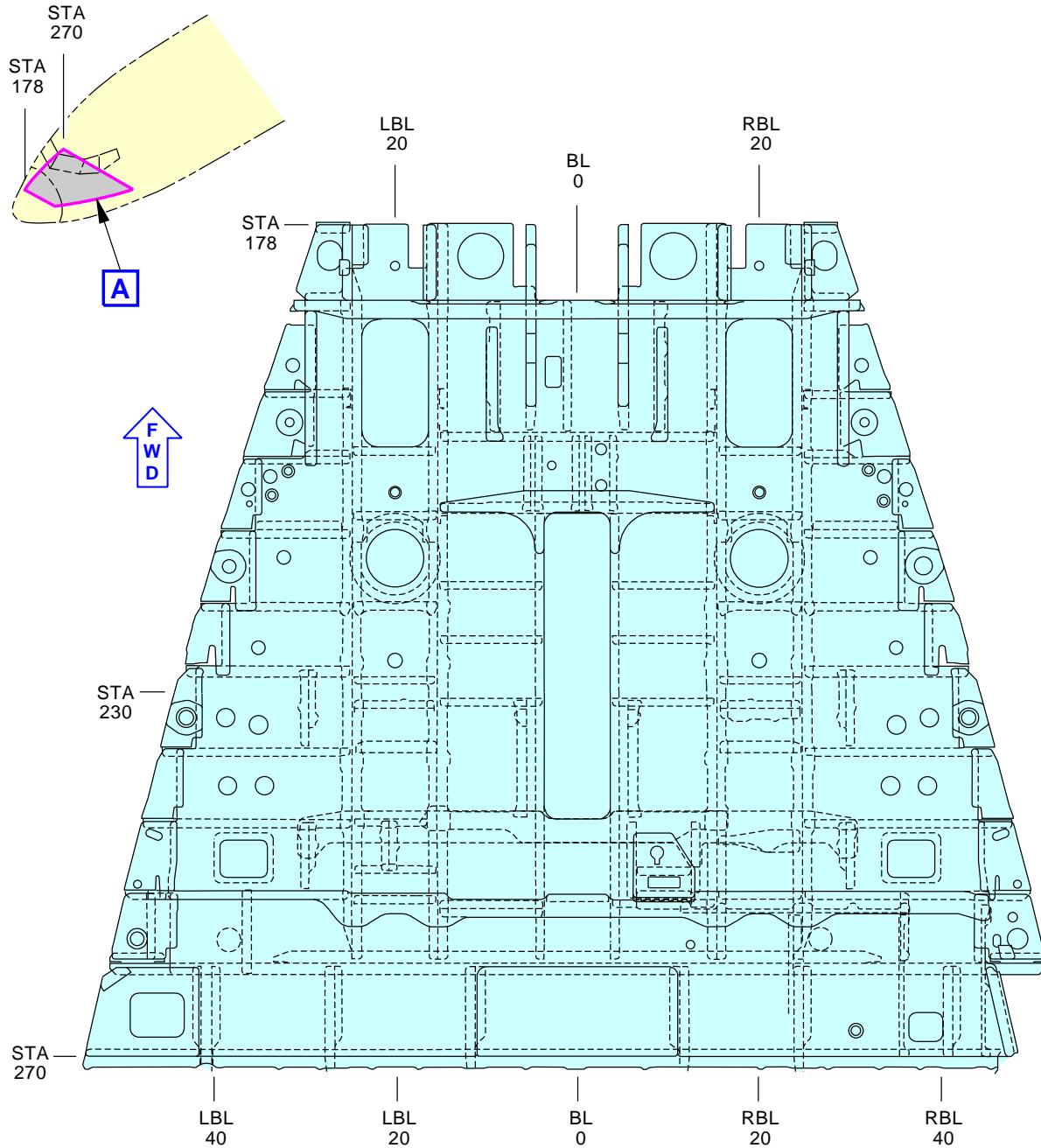
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EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-340-00-01

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Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-340-00-01

MPD ITEM
53-340-00**A**

2102637 S0000447408_V2

**Flight Deck Floor Structure
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT COMPARTMENT FLOOR STRUCTURE
		D633A109-AKS 53-340-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT FROM STA 270 TO 360			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-350-00-01
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC	AIRPLANE ALL ENGINE ALL
		ACCESS S2201		NOTE	ZONE 221 222
		NOTE			

Inspect passenger compartment from Sta 270 to 360 (except areas around door cutouts), including skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FROM STA 270 TO 360
		D633A109-AKS 53-350-00-01

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Jun 15/2015

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-350-00-01				
				MECH INSP				
TASK 53-05-03-210-832								
1. INTERNAL - GENERAL VISUAL: PASSENGER COMPARTMENT FROM STA 270 to 360								
(Figure 1)								
A. Inspection								
SUBTASK 53-05-03-010-029								
(1) Special Access:								
<table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>S2201</td> <td>Passenger Compartment From STA 259.5 to 360</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	S2201	Passenger Compartment From STA 259.5 to 360	
<u>Number</u>	<u>Name/Location</u>							
S2201	Passenger Compartment From STA 259.5 to 360							
NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.								
SUBTASK 53-05-03-210-032								
(2) Do a General Visual inspection of the passenger compartment from Sta 270 to 360 (except areas around door cutouts), including skin panels (skins, frames, stringers), longitudinal lap splices, circumferential skin and stringer splices.								
SUBTASK 53-05-03-910-042								
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.								
———— END OF TASK ————								

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FROM STA 270 TO 360
		D633A109-AKS 53-350-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-350-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FROM STA 270 TO 360
		D633A109-AKS 53-350-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-350-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-077

(7) Do the CPCP Basic Task Item 7 as follows:

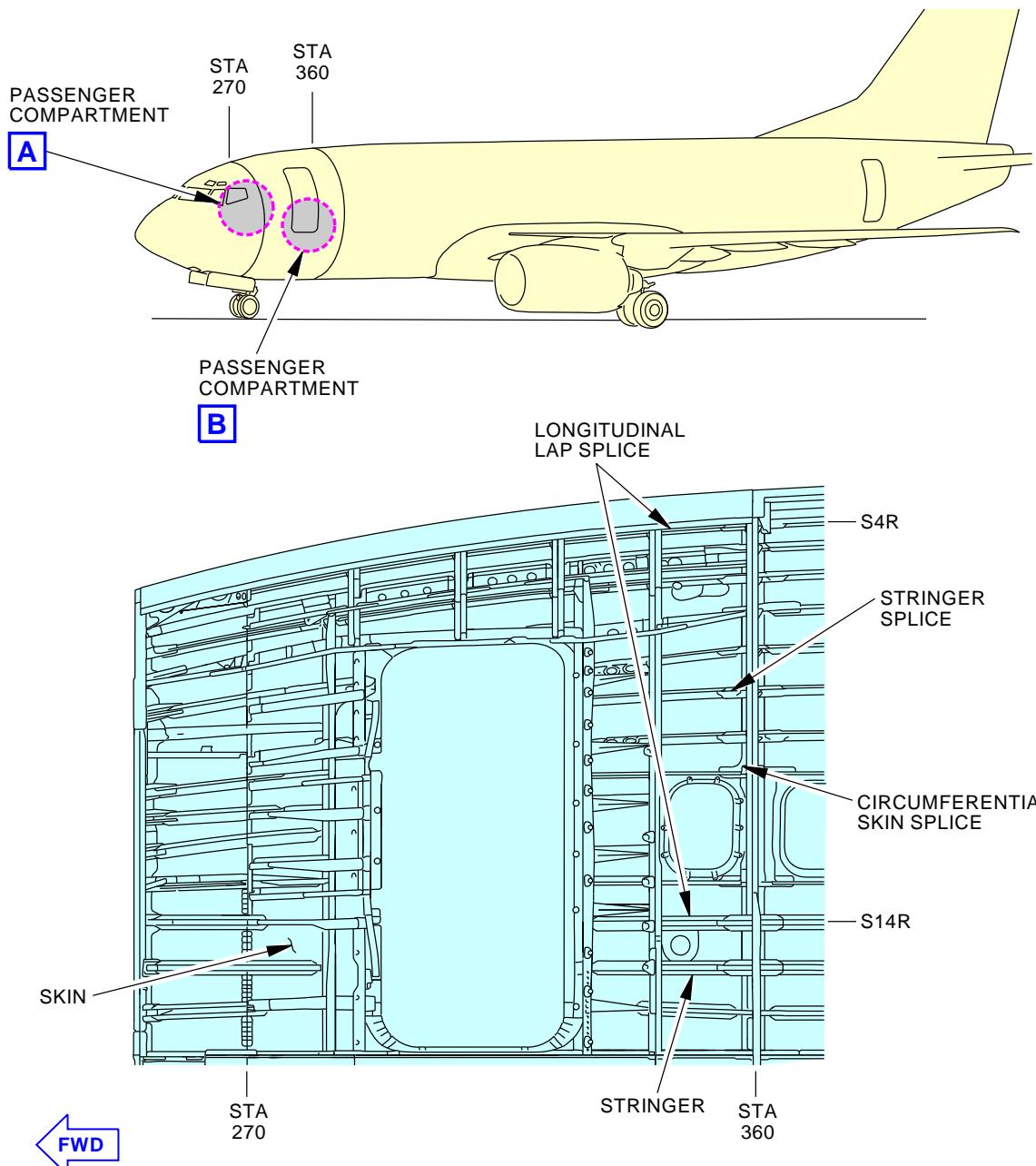
(a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FROM STA 270 TO 360
		D633A109-AKS 53-350-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-350-00-01
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**PASSENGER COMPARTMENT (STA 270 TO STA 360)
(RIGHT SIDE, VIEW IN THE OUTBOARD DIRECTION)****A**MPD ITEM
53-350-00

2071151 S0000430778_V2

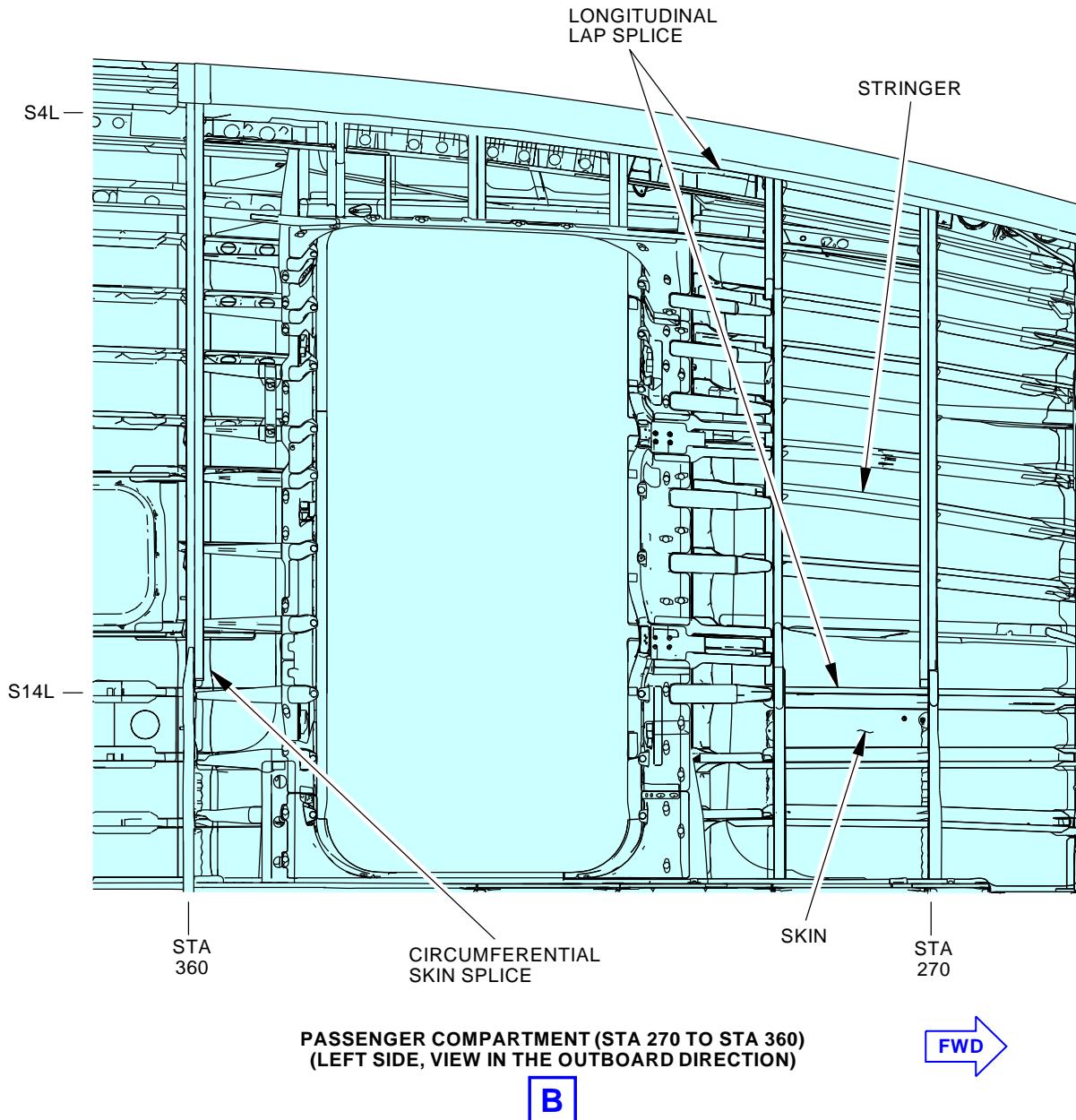
INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FROM STA 270 TO STA 360
Figure 1 (Sheet 1 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FROM STA 270 TO 360
		D633A109-AKS 53-350-00-01

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AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-350-00-01

MPD ITEM
53-350-00

2071155 S0000430781_V2

INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FROM STA 270 TO STA 360
Figure 1 (Sheet 2 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FROM STA 270 TO 360
		D633A109-AKS 53-350-00-01

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Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT DOOR CUTOUTS			BOEING CARD NO.
DATE	TASK DETAILED				53-360-00-01
TAIL NUMBER	WORK AREA UPR FUSELAGE	VERSION 1.1	THRESHOLD NOTE	REPEAT	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS S2001			ZONE 221
		NOTE			

Inspect forward entry door cutout surround structure (the door cutout to, and including, the door side of the first frame from the door in both the forward and aft directions).

INTERVAL NOTE: Threshold interval 8YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes first), applicable to Airplanes L/N# 1-5645 and L/N# 5653-5659 and those that have not incorporated the doorsill corrosion protection/enhanced moisture barrier. Threshold interval 9YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes first), applicable to Airplanes L/N# 5646-5652 and L/N# 5660 and on, or those that have incorporated the doorsill corrosion protection/enhanced moisture barrier.

ACCESS NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS
		D633A109-AKS 53-360-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-01
------	-------------	---------	------------------	--

TASK 53-05-03-211-809

MECH

INSP

1. INTERNAL - DETAILED: PASSENGER COMPARTMENT DOOR CUTOUTS

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-063

- (1) Special Access:

Number Name/Location

S2001 Passenger Compartment Door Cutouts Inspection

NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.

SUBTASK 53-05-03-211-009

- (2) Do a Detailed inspection of the forward entry door cutout surround structure (the door cutout to, and including, the door side of the first frame from the door in both the forward and aft directions).

SUBTASK 53-05-03-910-043

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

———— END OF TASK ———

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-01	Page 2 of 5 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-01	Page 3 of 5 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-01
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH INSP

SUBTASK 51-05-01-210-077

(7) Do the CPCP Basic Task Item 7 as follows:

(a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

 END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-01	Page 4 of 5 Oct 15/2014

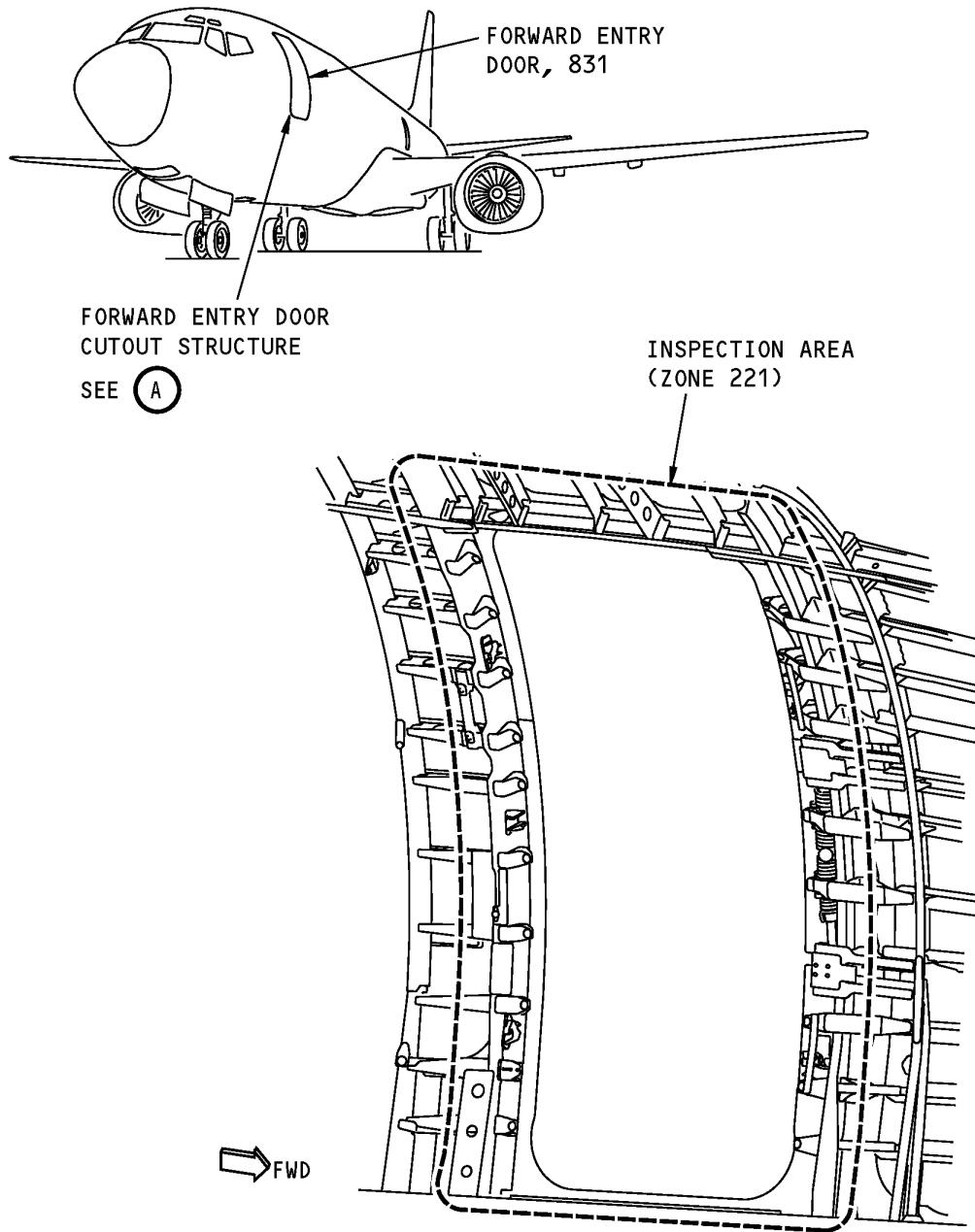
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-360-00-01

**FORWARD ENTRY DOOR CUTOUT STRUCTURE
(LAVATORY, SIDEWALL PANELS AND INSULATION REMOVED)**

A

**Forward Entry Door Cutout Surround Structure (Lavatory, Sidewall Panels And Insulation Removed)
Figure 1**

**EFFECTIVITY
AKS ALL****SOURCE
MRB****PASSENGER COMPARTMENT DOOR CUTOUTS****D633A109-AKS
53-360-00-01****Page 5 of 5
Oct 15/2014**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT DOOR CUTOUTS			BOEING CARD NO.
DATE	TASK DETAILED				53-360-00-02
TAIL NUMBER	WORK AREA UPR FUSELAGE	VERSION 1.1	THRESHOLD NOTE	REPEAT	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS 222AR S2001			ZONE 222
		NOTE			

Inspect forward galley door cutout surround structure (the door cutout to, and including, the door side of the first frame from the door in both the forward and aft directions).

INTERVAL NOTE: Threshold interval 8YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes first), applicable to Airplanes L/N# 1-5645 and L/N# 5653-5659 and those that have not incorporated the doorsill corrosion protection/enhanced moisture barrier. Threshold interval 9YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes first), applicable to Airplanes L/N# 5646-5652 and L/N# 5660 and on, or those that have incorporated the doorsill corrosion protection/enhanced moisture barrier.

ACCESS NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS
		D633A109-AKS 53-360-00-02

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Jun 15/2016

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-02
TASK 53-05-03-211-810				MECH INSP

1. INTERNAL - DETAILED: PASSENGER COMPARTMENT DOOR CUTOUTS

Figure 1

A. Inspection

SUBTASK 53-05-03-010-064

(1) Open this access panel:

Number Name/Location

222AR Forward Galley Service Door Hinge and Torque Tube Access Panel

Special Access:

Number Name/Location

S2001 Passenger Compartment Door Cutouts Inspection

NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.

SUBTASK 53-05-03-211-010

(2) Do a Detailed inspection of the forward galley door cutout surround structure (the door cutout to, and including, the door side of the first frame from the door in both the forward and aft directions).

SUBTASK 53-05-03-910-044

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

SUBTASK 53-05-03-410-064

(4) Close this access panel:

Number Name/Location

222AR Forward Galley Service Door Hinge and Torque Tube Access Panel

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-02	Page 2 of 5 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-02
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-02	Page 3 of 5 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-02				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

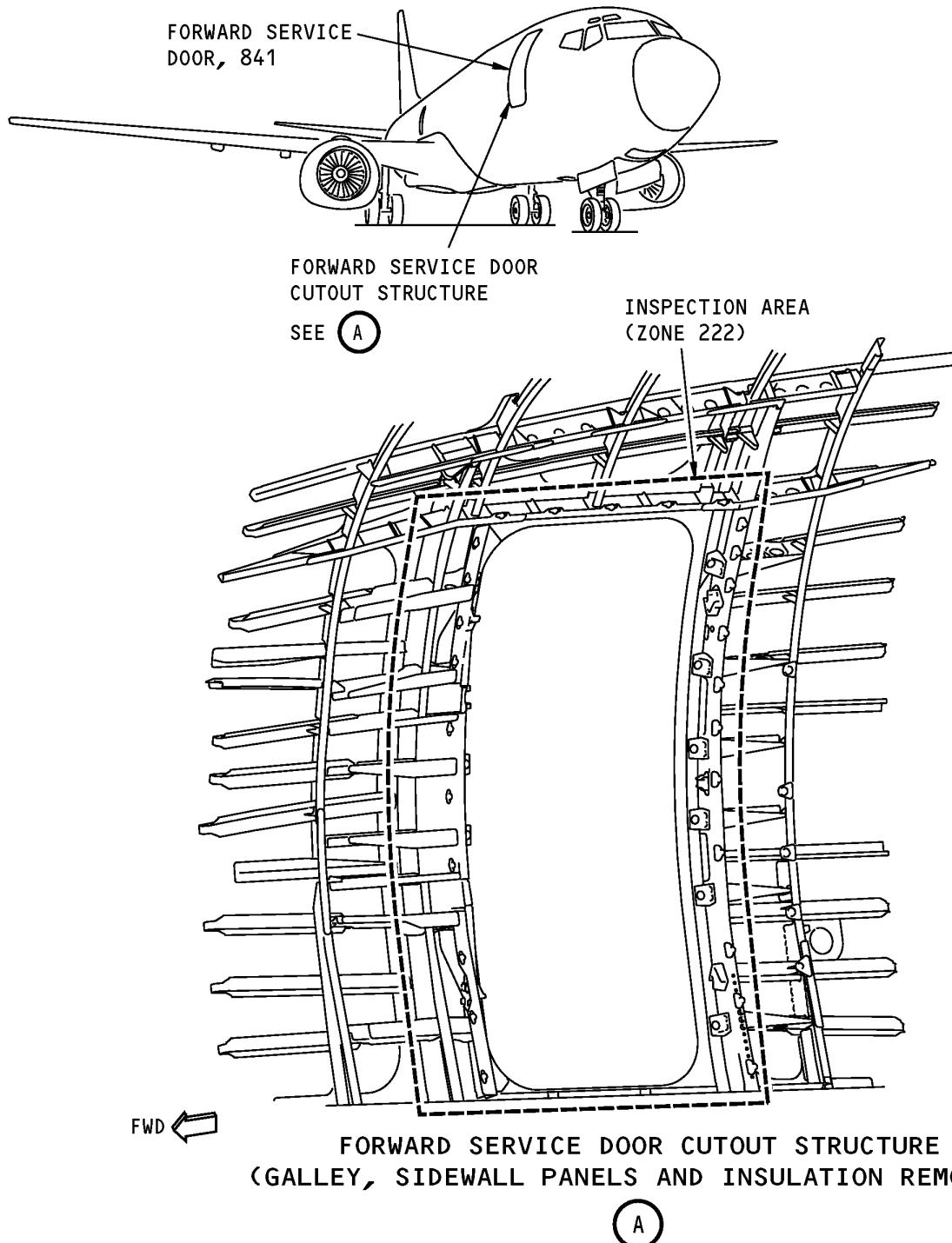
- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-02	Page 4 of 5 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-02
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**Forward Galley Service Door Cutout Surround Structure (Galley, Sidewall Panels and Insulation Removed)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS
		D633A109-AKS 53-360-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT DOOR CUTOUTS			BOEING CARD NO.
DATE	TASK DETAILED				53-360-00-03
TAIL NUMBER	WORK AREA UPR FUSELAGE	VERSION 1.1	THRESHOLD NOTE	REPEAT	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ENGINE ALL ALL
		ACCESS S2001			ZONE 241
		NOTE			

Inspect aft entry door cutout surround structure (the door cutout to, and including, the door side of the first frame from the door in both the forward and aft directions).

INTERVAL NOTE: Threshold interval 8YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes first), applicable to Airplanes L/N# 1-5645 and L/N# 5653-5659 and those that have not incorporated the doorsill corrosion protection/enhanced moisture barrier. Threshold interval 9YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes first), applicable to Airplanes L/N# 5646-5652 and L/N# 5660 and on, or those that have incorporated the doorsill corrosion protection/enhanced moisture barrier.

ACCESS NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS
		D633A109-AKS 53-360-00-03

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Jun 15/2016

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-03
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TASK 53-05-03-211-811

MECH

INSP

1. INTERNAL - DETAILED: PASSENGER COMPARTMENT DOOR CUTOUTS

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-065

(1) Special Access:

Number Name/Location

S2001 Passenger Compartment Door Cutouts Inspection

NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.

SUBTASK 53-05-03-211-011

(2) Do a Detailed inspection of the aft entry door cutout surround structure (the door cutout to, and including, the door side of the first frame from the door in both the forward and aft directions).

SUBTASK 53-05-03-910-045

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-03	Page 2 of 5 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-03
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-03	Page 3 of 5 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-03				
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MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

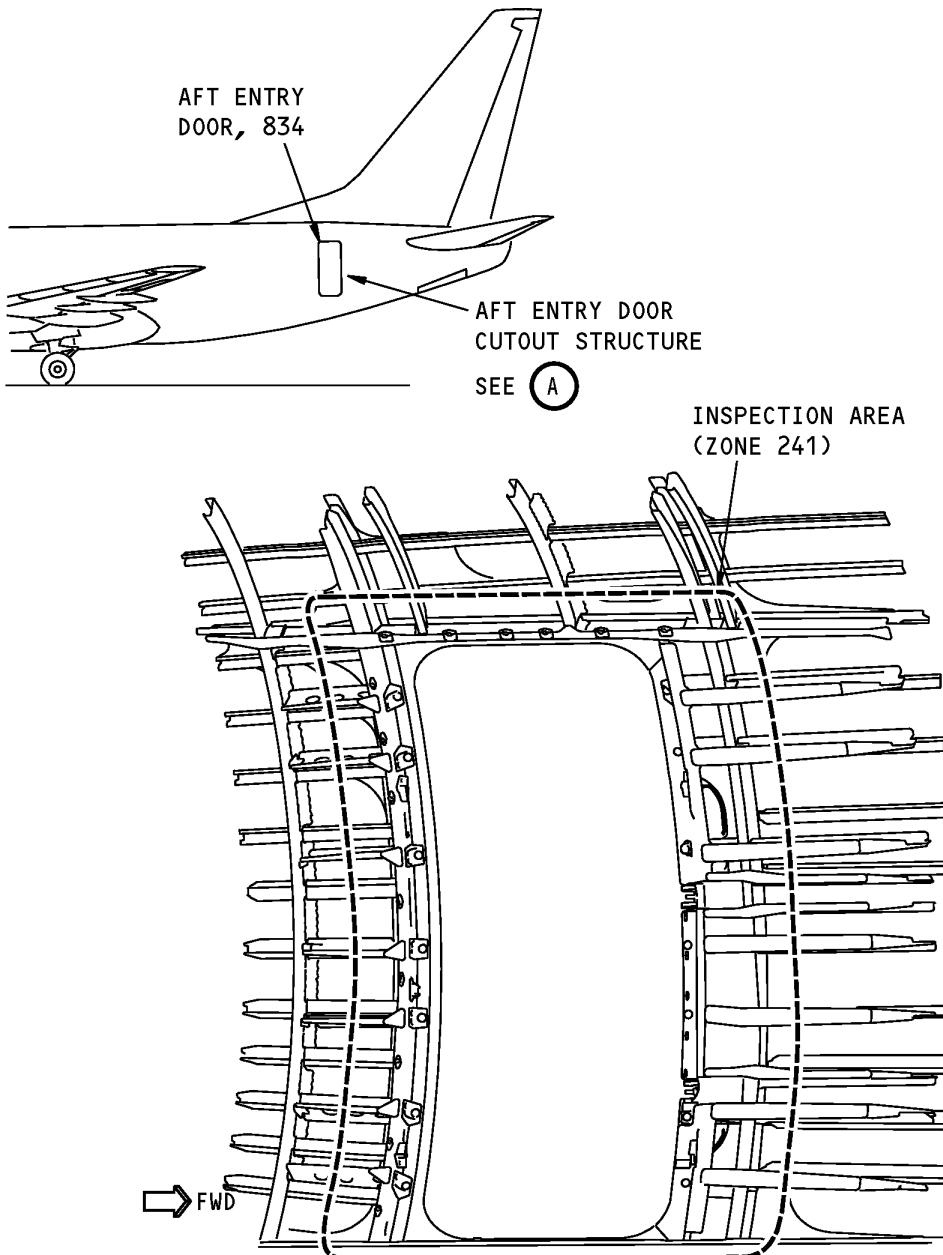
- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-03	Page 4 of 5 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-03
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**AFT ENTRY DOOR CUTOUT STRUCTURE
(LAVATORY, SIDEWALL PANELS AND INSULATION REMOVED)**

(A)

**Passenger Compartment Door Cutouts
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS
		D633A109-AKS 53-360-00-03

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT DOOR CUTOUTS			BOEING CARD NO.
DATE	TASK DETAILED				53-360-00-04
TAIL NUMBER	WORK AREA UPR FUSELAGE	VERSION 1.1	THRESHOLD NOTE	REPEAT	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS S2001			ZONE 242
		NOTE			

Inspect aft galley door cutout surround structure (the door cutout to, and including, the door side of the first frame from the door in both the forward and aft directions).

INTERVAL NOTE: Threshold interval 8YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes first), applicable to Airplanes L/N# 1-5645 and L/N# 5653-5659 and those that have not incorporated the doorsill corrosion protection/enhanced moisture barrier. Threshold interval 9YR / 24000 FC and Repeat interval 6YR / 18000 FC (Whichever comes first), applicable to Airplanes L/N# 5646-5652 and L/N# 5660 and on, or those that have incorporated the doorsill corrosion protection/enhanced moisture barrier.

ACCESS NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS
		D633A109-AKS 53-360-00-04

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Jun 15/2016

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-04
------	-------------	---------	------------------	--

TASK 53-05-03-211-812

MECH

INSP

1. INTERNAL - DETAILED: PASSENGER COMPARTMENT DOOR CUTOUTS

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-066

- (1) Special Access:

Number Name/Location

S2001 Passenger Compartment Door Cutouts Inspection

NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/displace insulation blankets as required.

SUBTASK 53-05-03-211-012

- (2) Do a Detailed inspection of the aft galley door cutout surround structure (the door cutout to, and including, the door side of the first frame from the door in both the forward and aft directions).

SUBTASK 53-05-03-910-046

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

———— END OF TASK ———

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-04	Page 2 of 5 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-04
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-04	Page 3 of 5 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-360-00-04			
				<table border="1"><tr><td style="width: 80%;"></td><td style="width: 10%; text-align: center;">MECH</td><td style="width: 10%; text-align: center;">INSP</td></tr></table>		MECH	INSP
	MECH	INSP					

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT DOOR CUTOUTS	
		D633A109-AKS 53-360-00-04	Page 4 of 5 Oct 15/2014

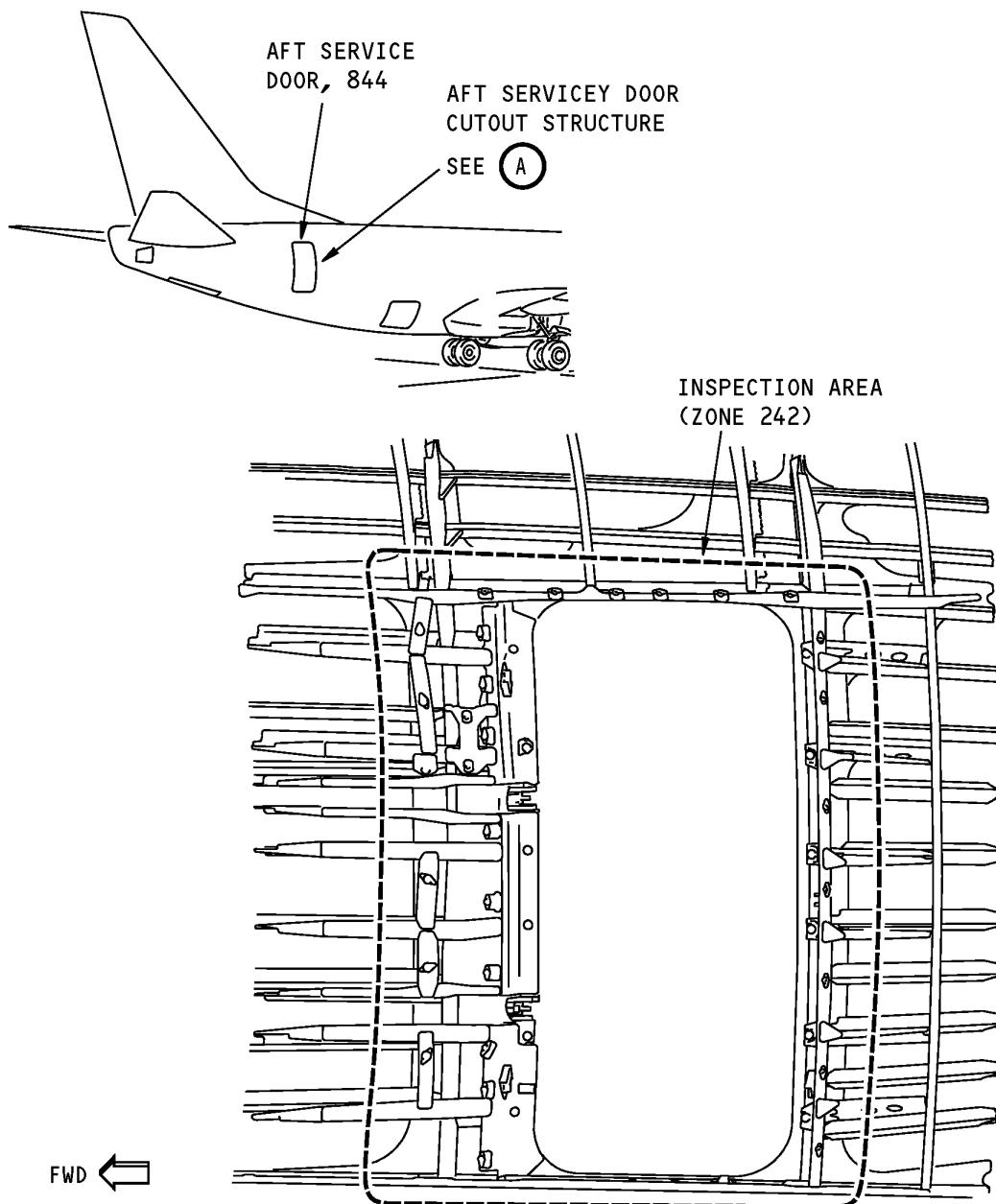
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-360-00-04

**AFT SERVICE DOOR CUTOUT STRUCTURE
GALLEY, SIDEWALL PANELS AND INSULATION REMOVED)**

(A)

**Passenger Compartment Door Cutouts
Figure 1**

**EFFECTIVITY
AKS ALL****SOURCE
MRB****PASSENGER COMPARTMENT DOOR CUTOUTS****D633A109-AKS
53-360-00-04****Page 5 of 5
Oct 15/2014**

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA			BOEING CARD NO. 53-370-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 12 YR	REPEAT 12 YR	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS S2002			
		NOTE			
			ZONE 221 222 231 232 241 242		

Inspect passenger compartment floor structure in dry area (away from doors, galleys and lavs). Exclude floor structure from Sta 540 to 727.

NOTE: Not applicable to airplanes with flat aft pressure bulkhead.

ACCESS NOTE: Remove floor panels and sidewalls as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-370-00-01

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Jun 15/2015

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-370-00-01
				MECH INSP
TASK 53-05-03-210-833				
1. INTERNAL - GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA				
(Figure 1)				
A. Inspection				
SUBTASK 53-05-03-010-030				
(1) Special Access: <u>Number</u> <u>Name/Location</u> S2002 Passenger Compartment Floor Structure - Dry Area Inspection <u>NOTE:</u> Remove floor panels and sidewalls as required. Remove/displace insulation blankets as required.				
SUBTASK 53-05-03-210-033				
(2) Do a General Visual inspection of the passenger compartment floor structure in dry area (away from doors, galleys and lavs). Exclude floor structure from Sta 540 to 727.				
SUBTASK 53-05-03-910-047				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.				
— END OF TASK —				
EFFECTIVITY AKS ALL		SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA D633A109-AKS 53-370-00-01	

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-370-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				
EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA		
		D633A109-AKS 53-370-00-01	Page 3 of 9 Oct 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-370-00-01				
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MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

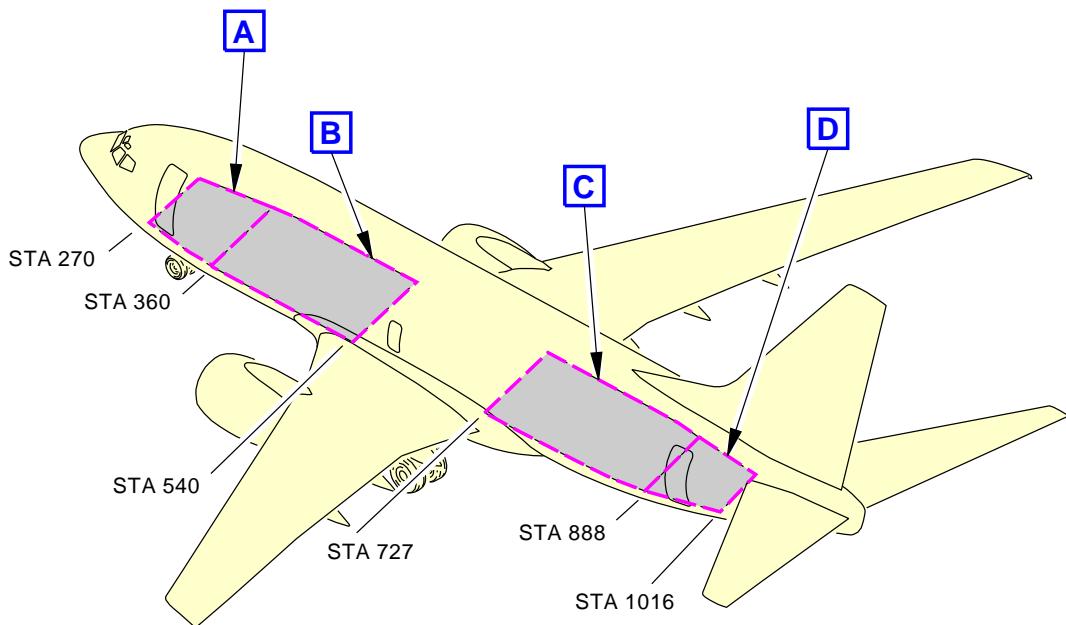
- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA	
		D633A109-AKS 53-370-00-01	Page 4 of 9 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-370-00-01

MPD ITEM
53-370-00

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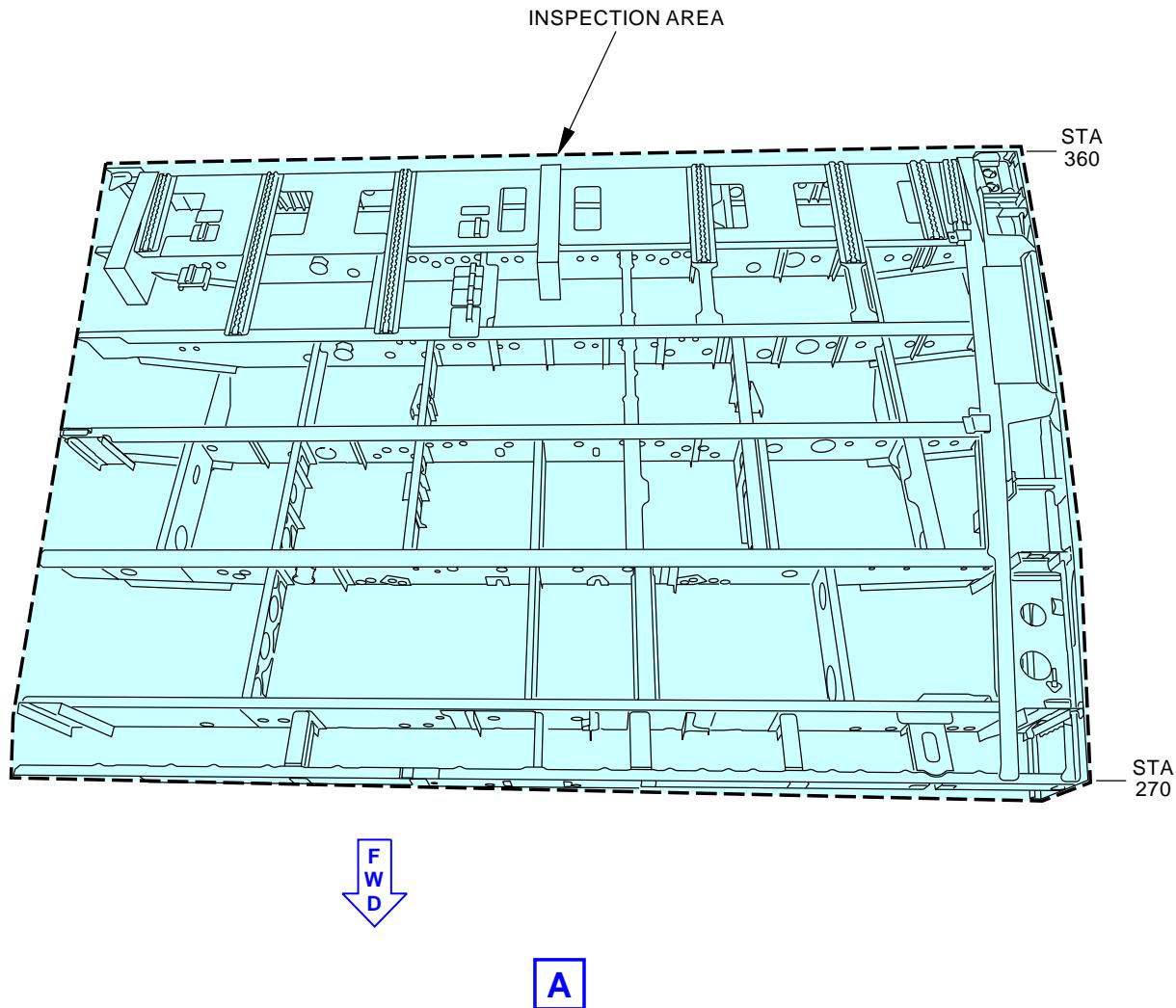
INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE-DRY AREA
Figure 1 (Sheet 1 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-370-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-370-00-01

MPD ITEM
53-370-00

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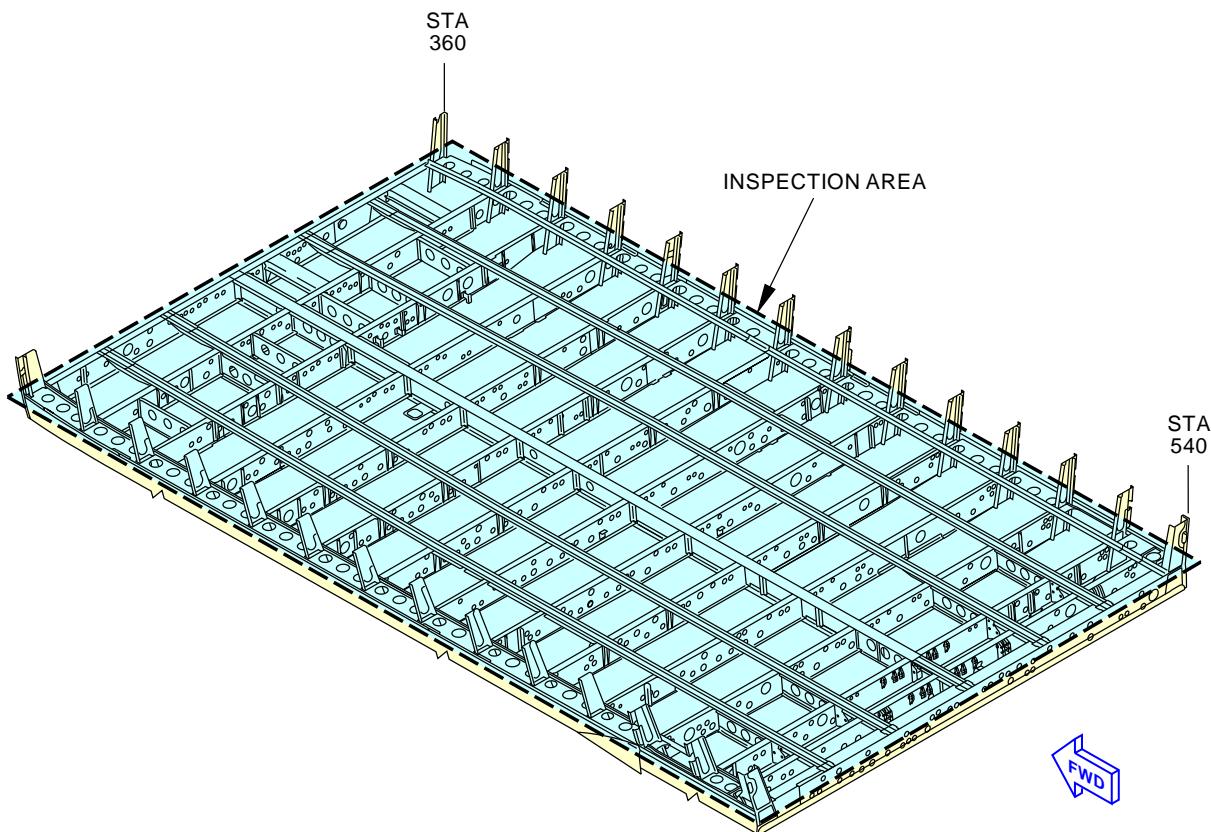
INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE-DRY AREA
Figure 1 (Sheet 2 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-370-00-01

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Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-370-00-01
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**B**MPD ITEM
53-370-00

2102696 S0000444462_V2

INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE-DRY AREA
Figure 1 (Sheet 3 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-370-00-01

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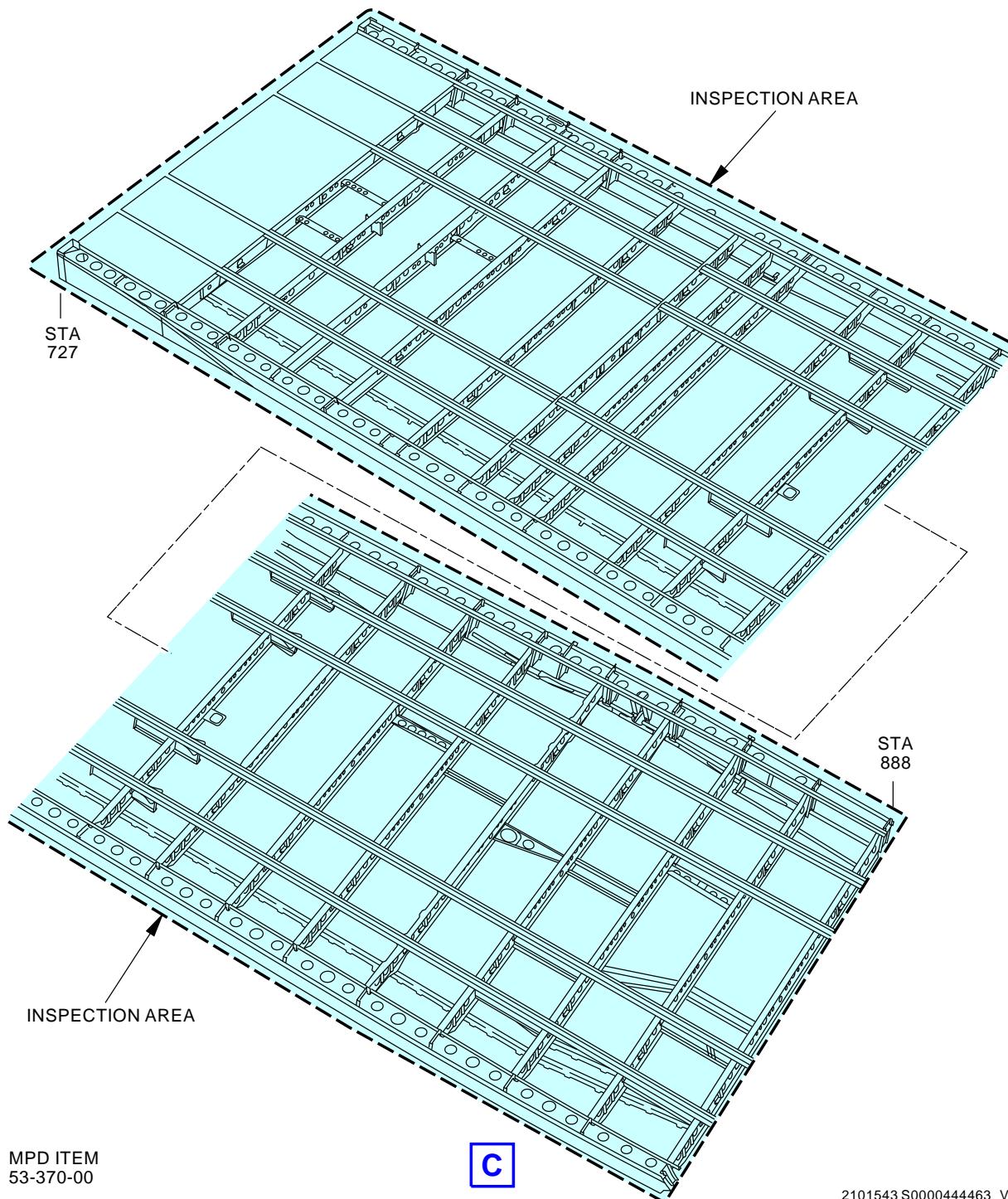
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-370-00-01MPD ITEM
53-370-00

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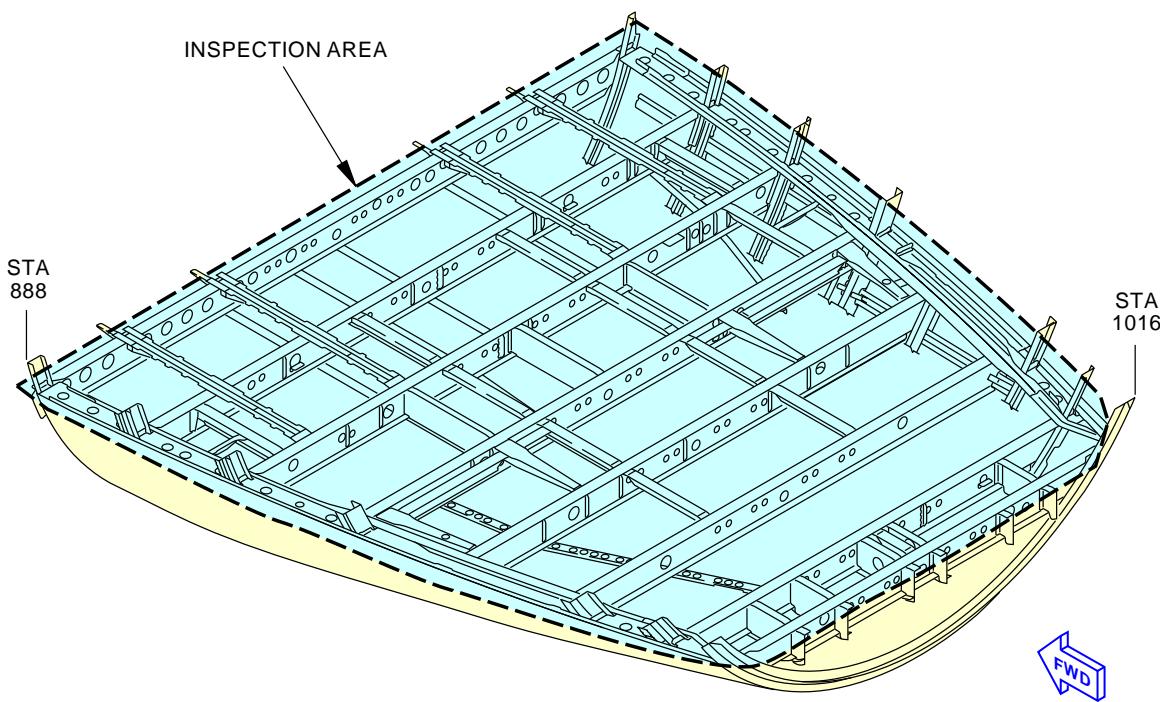
INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE-DRY AREA
Figure 1 (Sheet 4 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-370-00-01

Page 8 of 9
Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-370-00-01

MPD ITEM
53-370-00

2102072 S0000444464_V2

INTERNAL-GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE-DRY AREA
Figure 1 (Sheet 5 of 5)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - DRY AREA
		D633A109-AKS 53-370-00-01

Page 9 of 9
Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-380-00-01
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD NOTE	REPEAT	RELATED CARD
STATION	SKILL AIRPL	NOTE			AIRPLANE ALL ENGINE ALL
		ACCESS S2003			ZONE 221 222 231 232 241 242
		NOTE			

Inspect passenger compartment floor structure in wet area (within approximately 20 inches from doors, galleys and lavs, and the floor structure below the door to, and including, the door side of the first frame from the door in both the forward and aft directions).

NOTE: Not applicable to airplanes with flat aft pressure bulkhead.

INTERVAL NOTE: Threshold interval 8YR / Repeat interval 6YR, whichever comes first, applicable to Airplanes L/N# 1-2412 and those that have not incorporated the HI-TAK Gel Tape. Threshold interval 9YR / Repeat interval 6YR, whichever comes first, applicable to Airplanes L/N# 2413 and on, or those that have incorporated the HI-TAK Gel Tape.

ACCESS NOTE: Remove galleys and lavs. Remove floor panels and sidewalls as required.

Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	D633A109-AKS 53-380-00-01	Page 1 of 8 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-380-00-01
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AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD

MECH

INSP

TASK 53-05-03-210-834**1. INTERNAL - GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA**

(Figure 1,Figure 2)

A. Inspection

SUBTASK 53-05-03-010-031

- (1) Special Access:

Number Name/Location

S2003 Passenger Compartment Floor Structure - Wet Area Inspection

NOTE: Remove galleys and lavs. Remove floor panels and sidewalls as required.
Remove/displace insulation blankets as required.

SUBTASK 53-05-03-210-034

- (2) Do a General Visual inspection of the passenger compartment floor structure in wet area (within approximately 20 inches from doors, galleys and lavs, and the floor structure below the door to, and including, the door side of the first frame from the door in both the forward and aft directions).

SUBTASK 53-05-03-910-048

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.

AKS ALL**———— END OF TASK ————**

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-380-00-01

**Page 2 of 8
Jun 15/2016**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-380-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	D633A109-AKS 53-380-00-01	Page 3 of 8 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-380-00-01				
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MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	
		D633A109-AKS 53-380-00-01	Page 4 of 8 Oct 15/2014

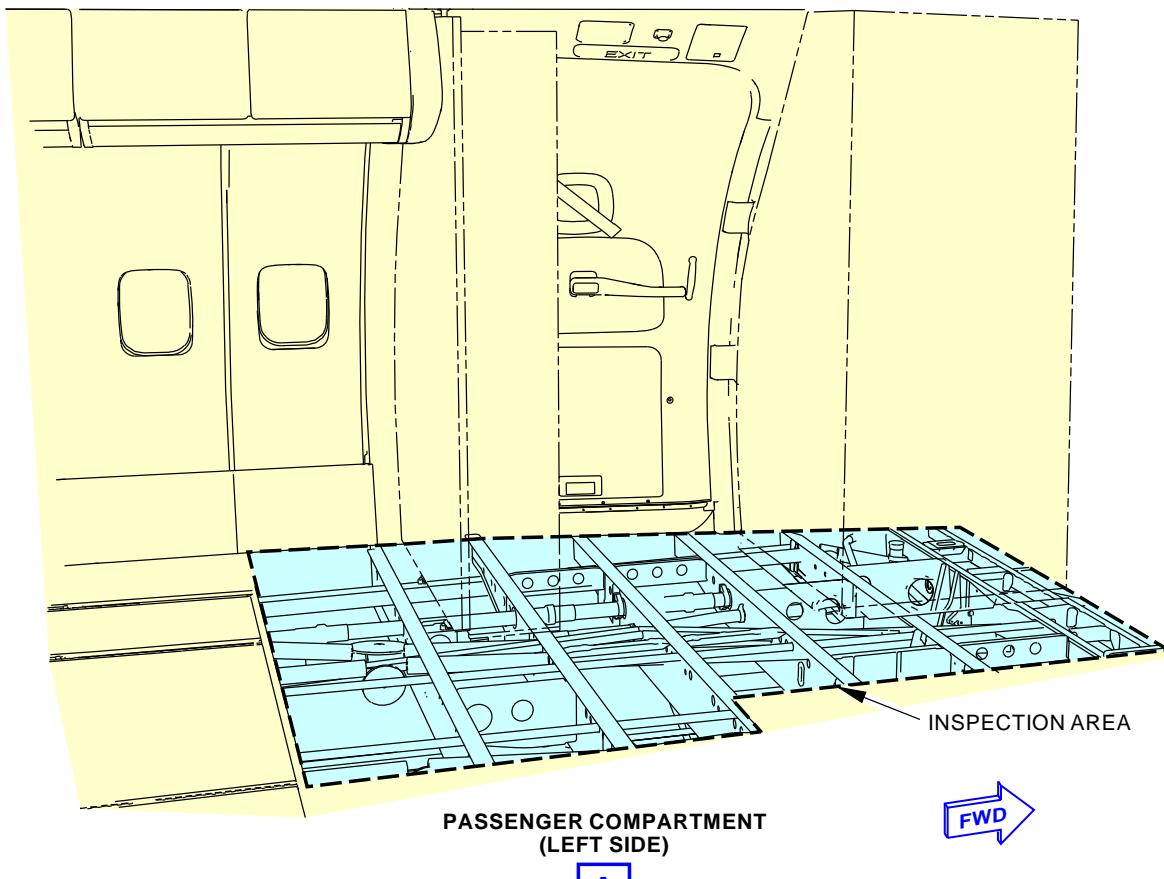
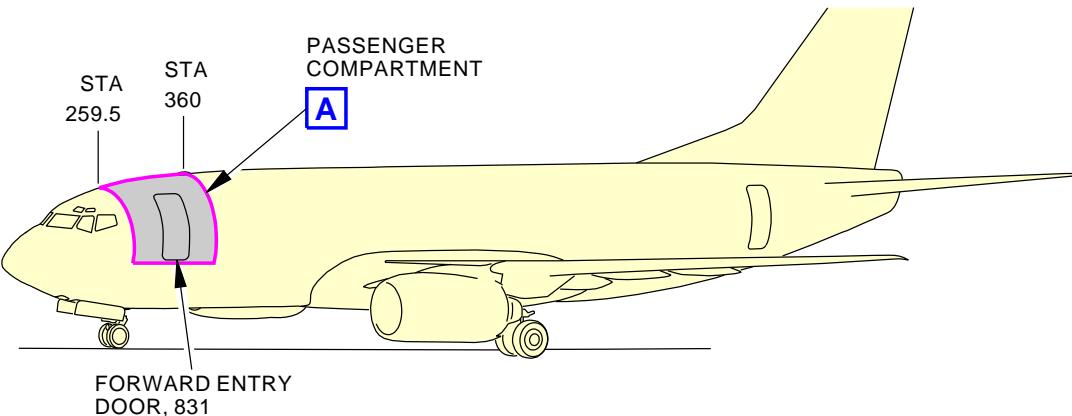
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-380-00-01MPD ITEM
53-380-00

1363524 S0000246348_V4

**Passenger Compartment Structure - Wet Area General Visual (Internal)
Figure 1 (Sheet 1 of 2)**EFFECTIVITY
**AKS ALL; AIRPLANES WITH A CURVED AFT
PRESSURE BULKHEAD**SOURCE
MRB**PASSENGER COMPARTMENT FLOOR STRUCTURE - WET
AREA****D633A109-AKS
53-380-00-01****Page 5 of 8
Oct 15/2015**

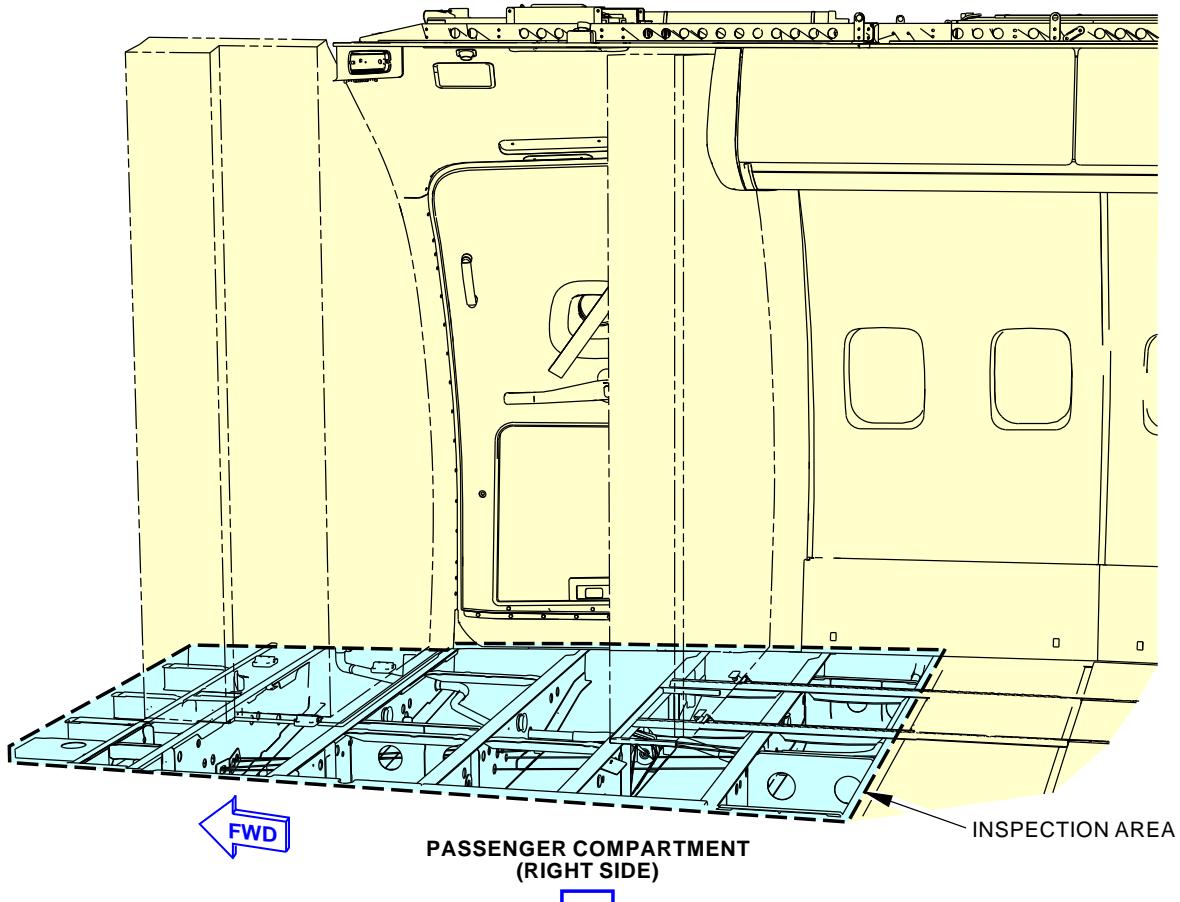
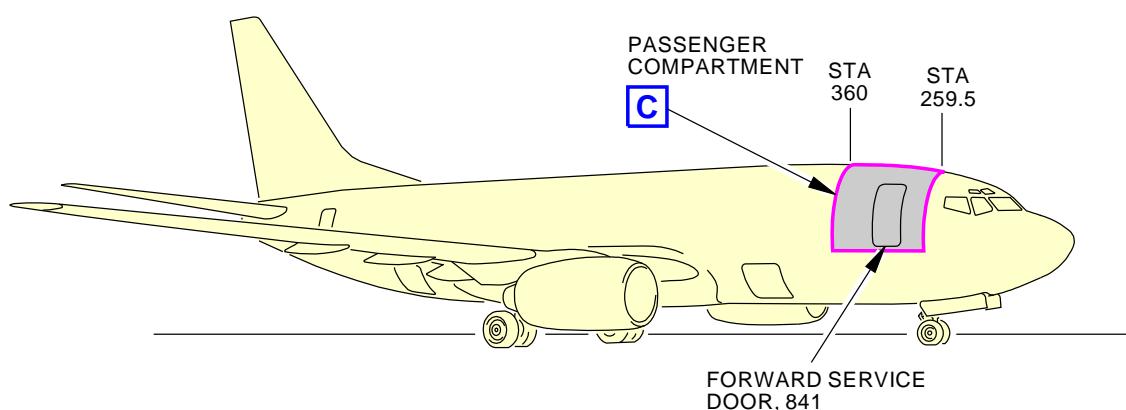
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-380-00-01MPD ITEM
53-380-00

1363522 S0000246349_V4

**Passenger Compartment Structure - Wet Area General Visual (Internal)
Figure 1 (Sheet 2 of 2)****EFFECTIVITY
AKS ALL; AIRPLANES WITH A CURVED AFT
PRESSURE BULKHEAD****SOURCE
MRB****PASSENGER COMPARTMENT FLOOR STRUCTURE - WET
AREA****D633A109-AKS
53-380-00-01****Page 6 of 8
Oct 15/2015**

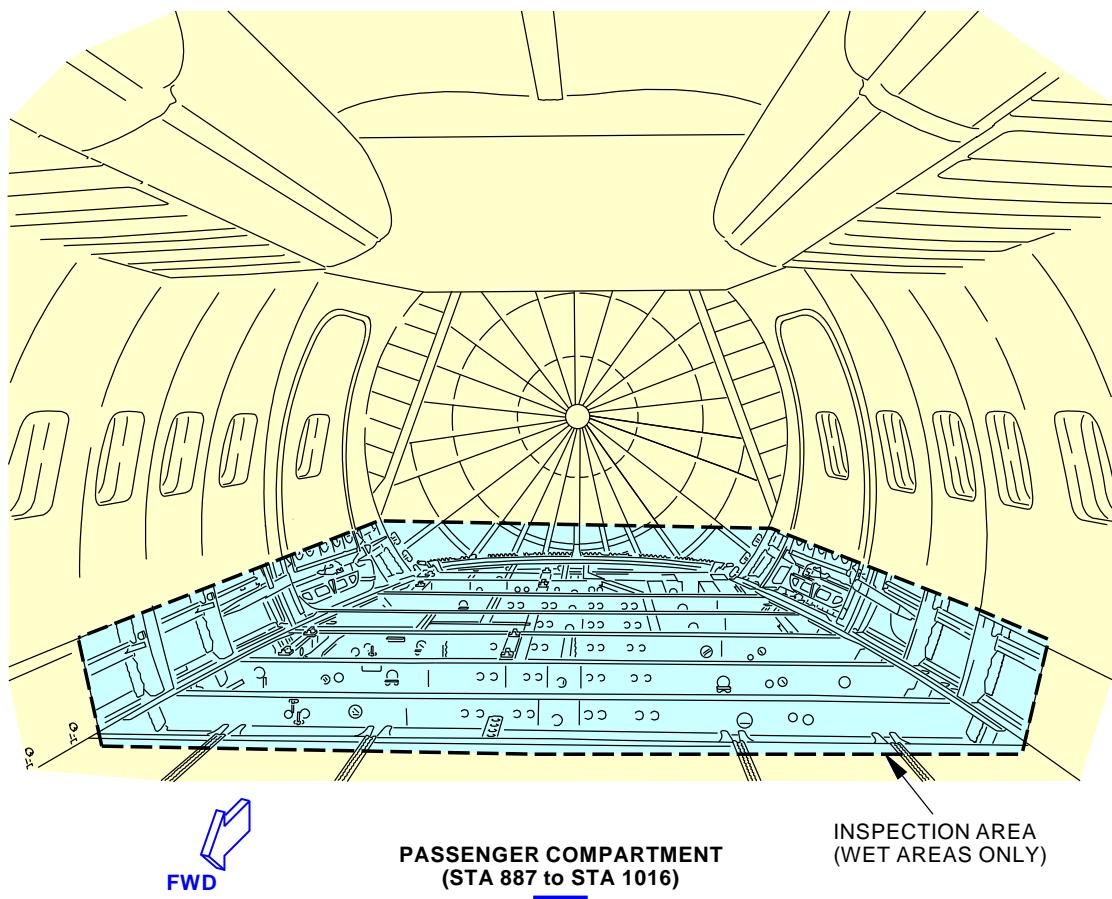
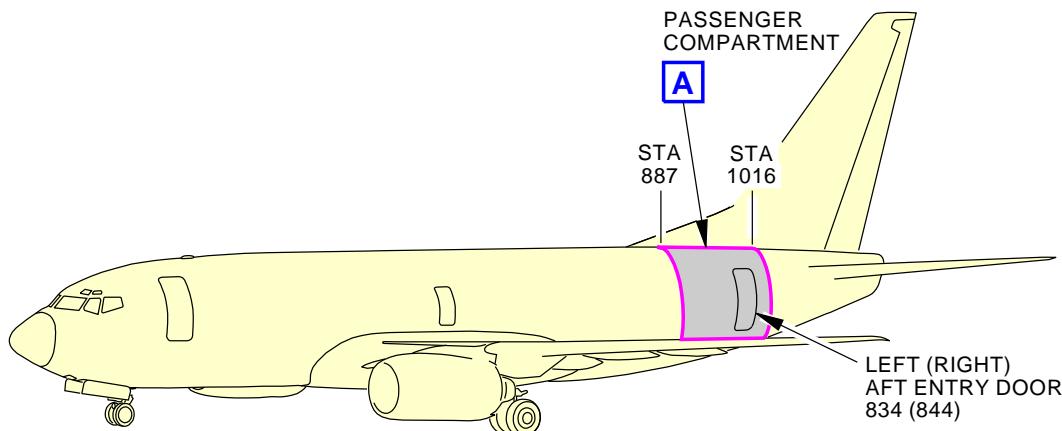
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

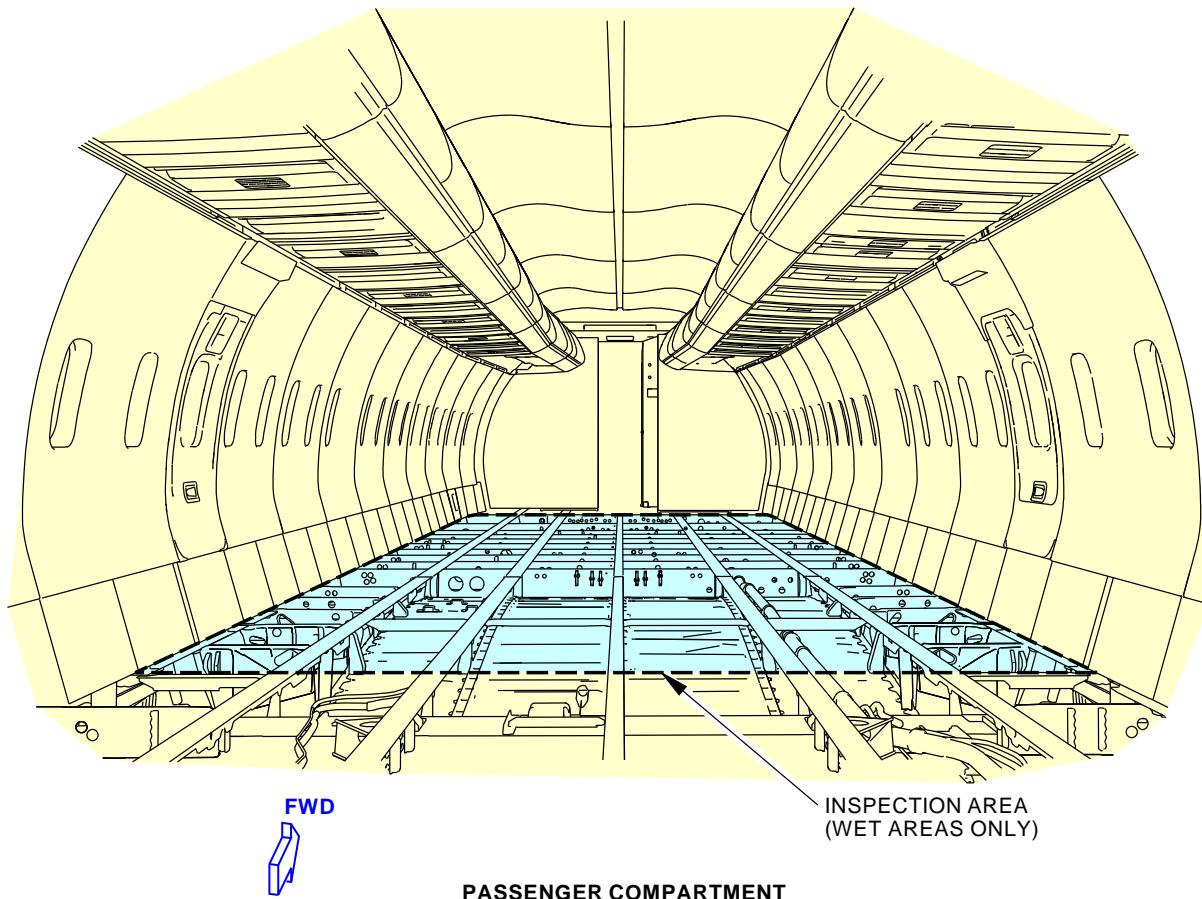
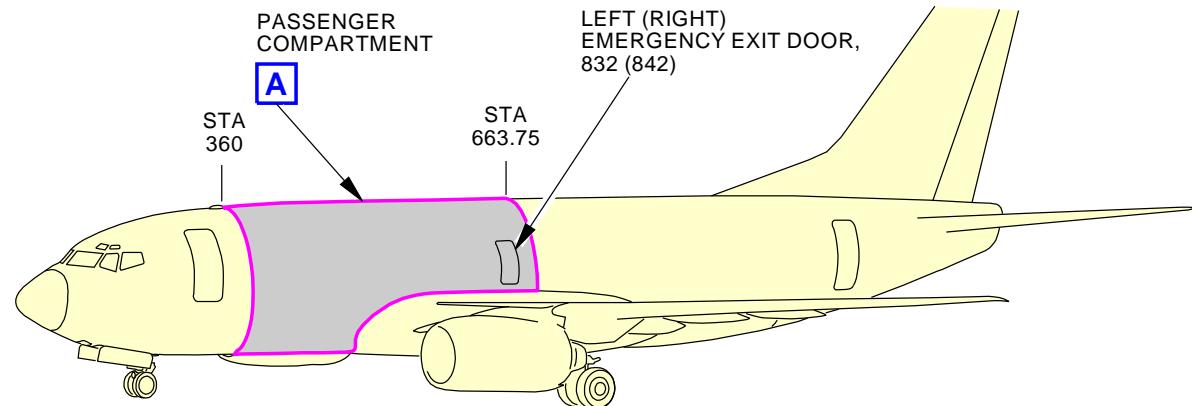
BOEING CARD NO.
53-380-00-01MPD ITEM
53-380-00

J76715 S0000179157_V4

**Passenger Compartment Structure - Wet Area General Visual (Internal)
Figure 2 (Sheet 1 of 2)****EFFECTIVITY
AKS ALL; AIRPLANES WITH A CURVED AFT
PRESSURE BULKHEAD**SOURCE
MRB**PASSENGER COMPARTMENT FLOOR STRUCTURE - WET
AREA****D633A109-AKS
53-380-00-01****Page 7 of 8
Oct 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-380-00-01

MPD ITEM
53-380-00

D37065 S0000154098_V3

**Passenger Compartment Structure - Wet Area General Visual (Internal)
Figure 2 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-380-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-390-00-01
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 9 YR	REPEAT 6 YR	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS S2004			ZONE 221 222 231 232 241 242
		NOTE			

Inspect galley and lav attach fittings and any other easily visible portions of the floor structure in wet area (within approximately 20 inches from galleys and lavs, and the floor structure below the door to, and including, the door side of the first frame from the door in both the forward and aft directions).

ACCESS NOTE: Galleys and lavs removal is not required. Remove galley kick-plates and any other easily removable panels that may help inspect areas under galleys and lavs.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-390-00-01

Page 1 of 7
Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-390-00-01
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TASK 53-05-03-210-835

MECH

INSP

1. INTERNAL - GENERAL VISUAL: PASSENGER COMPARTMENT FLOOR STRUCTURE - WET**AREA**

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-032

- (1) Special Access:

Number Name/Location

S2004 Passenger Compartment Floor Structure - Wet Area Inspection

NOTE: Galleys and lavs removal is not required. Remove galley kick-plates and any other easily removable panels that may help inspect areas under galleys and lavs.

SUBTASK 53-05-03-210-035

- (2) Do a General Visual inspection of the galley and lav attach fittings and any other easily visible portions of the floor structure in wet area (within approximately 20 inches from galleys and lavs, and the floor structure below the door to, and including, the door side of the first frame from the door in both the forward and aft directions).

SUBTASK 53-05-03-910-049

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-804.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	
		D633A109-AKS 53-390-00-01	Page 2 of 7 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-390-00-01
				MECH INSP
TASK 51-05-01-210-804				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-043				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-044				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-045				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-046				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-111				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-048				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	
		D633A109-AKS 53-390-00-01	Page 3 of 7 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-390-00-01				
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MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

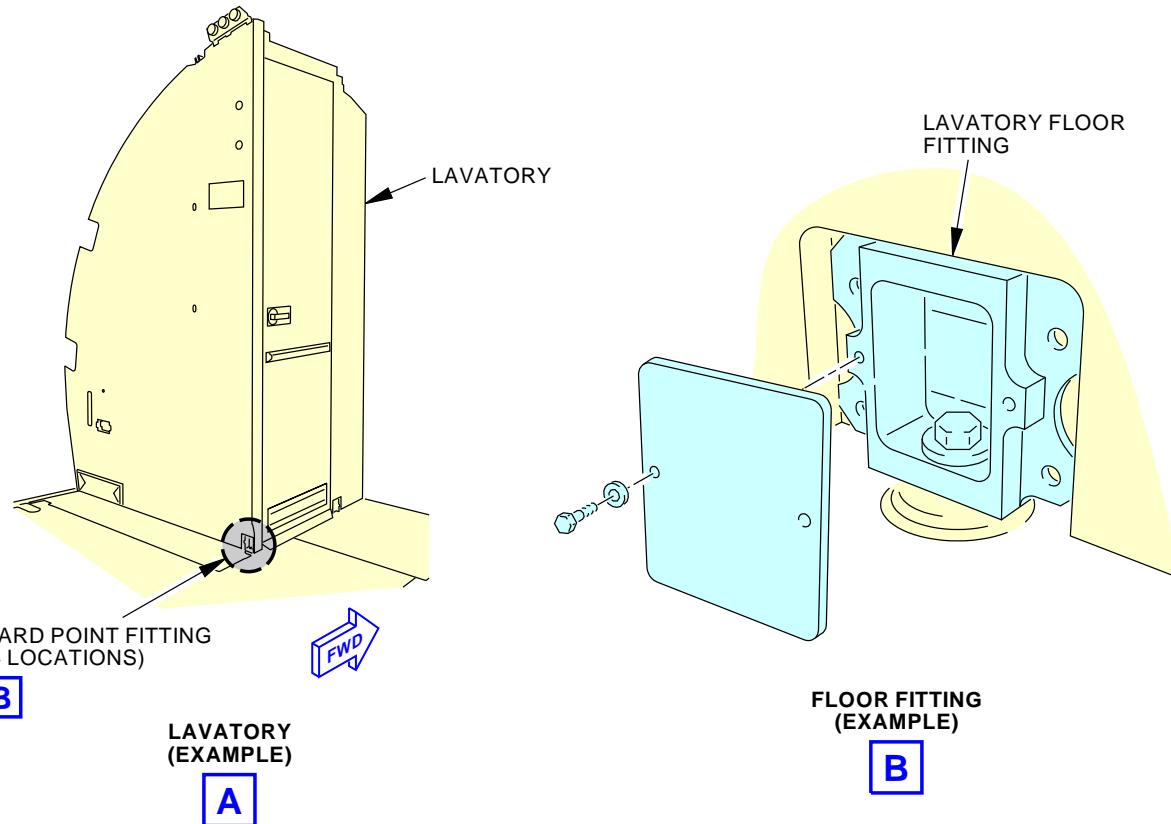
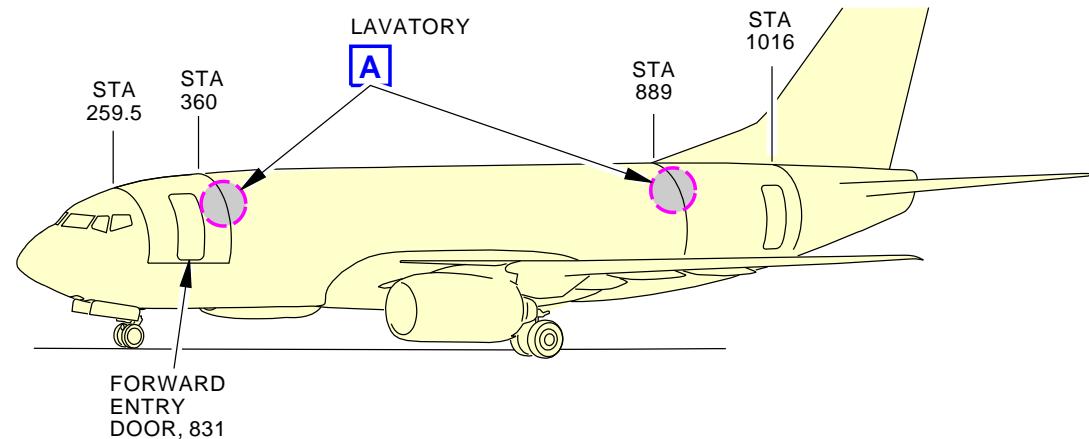
- (7) CPCP Basic Task Item 7 is not applicable.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	
		D633A109-AKS 53-390-00-01	Page 4 of 7 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-390-00-01

MPD ITEM
53-390-00

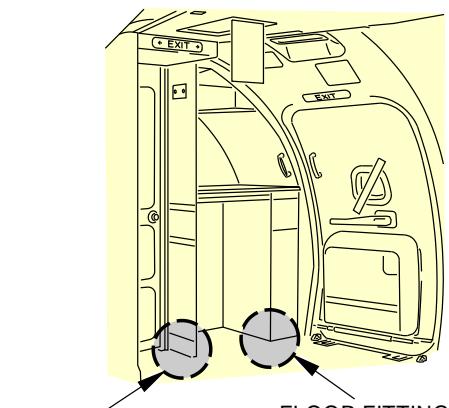
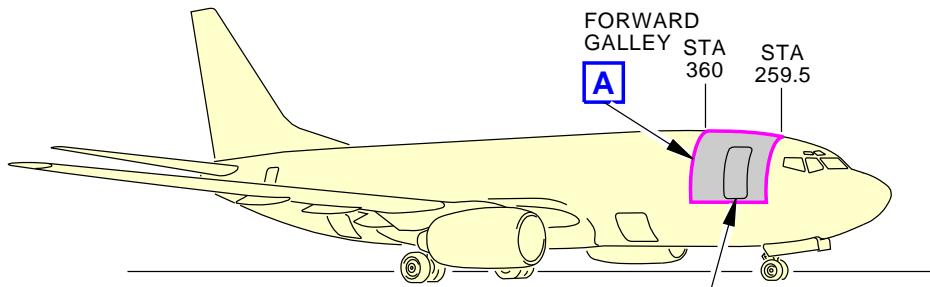
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**Passenger Compartment Floor Structure - Wet Area
Figure 1 (Sheet 1 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA
		D633A109-AKS 53-390-00-01

AKS737-600/700/800/900
TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-390-00-01



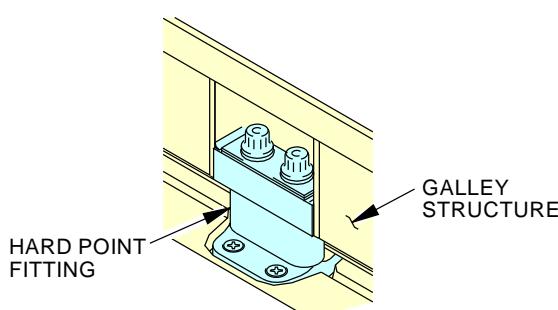
FORWARD GALLEY (G1)



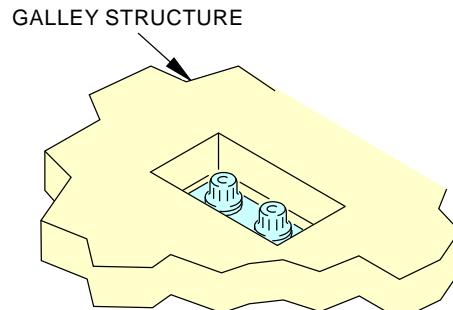
FORWARD SERVICE DOOR, 841



FORWARD GALLEY (G2)



FLOOR FITTING (EXAMPLE)

MPD ITEM
53-390-00

FLOOR FITTING (EXAMPLE)



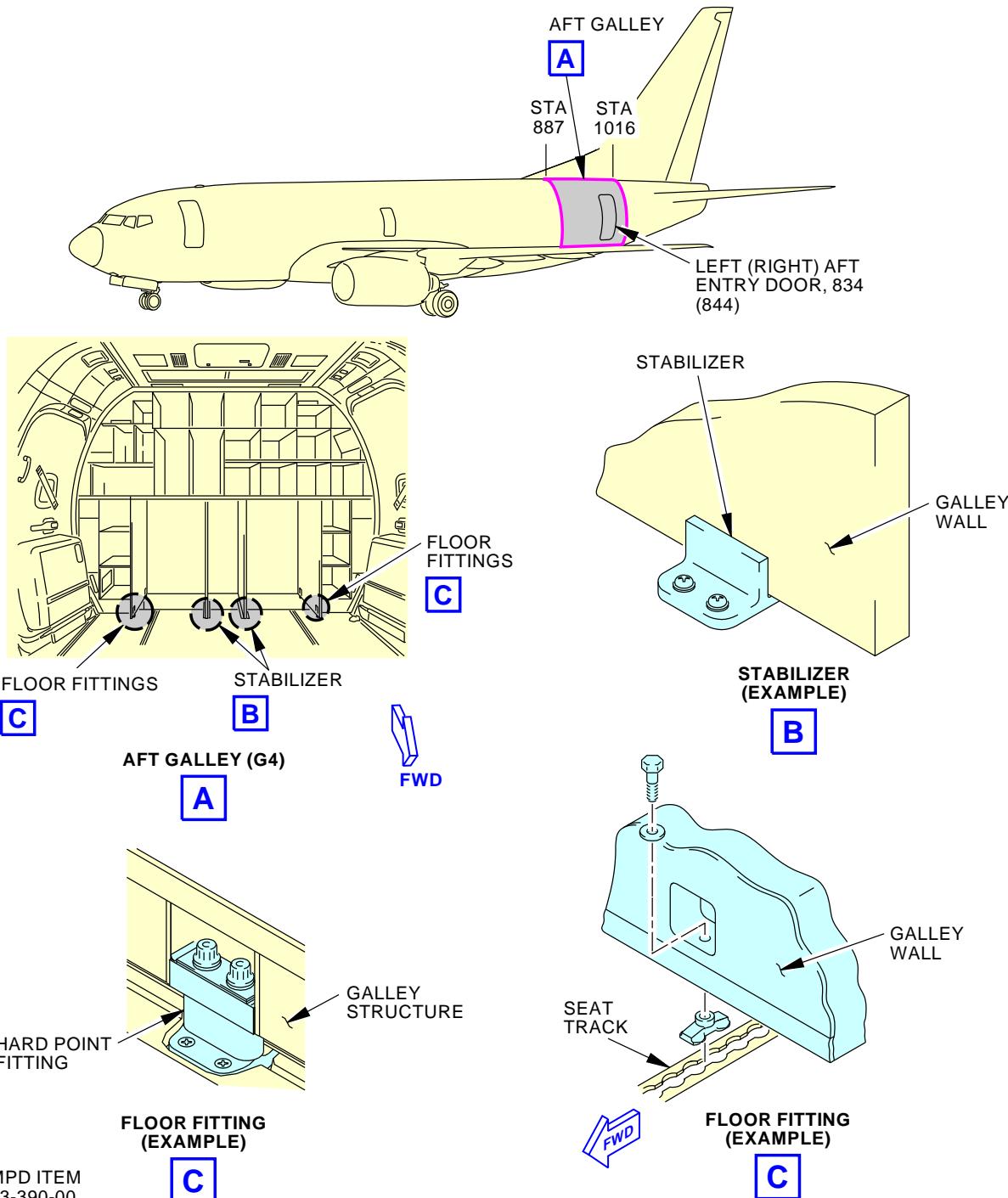
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Passenger Compartment Floor Structure - Wet Area
Figure 1 (Sheet 2 of 3)

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	
		D633A109-AKS 53-390-00-01	Page 6 of 7 Oct 15/2015

AKS737-600/700/800/900
TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-390-00-01

Passenger Compartment Floor Structure - Wet Area
Figure 1 (Sheet 3 of 3)

2341784 S0000533818_V2

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT FLOOR STRUCTURE - WET AREA	
		D633A109-AKS 53-390-00-01	Page 7 of 7 Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD PASSENGER COMPARTMENT, STA 360 TO 663.75			BOEING CARD NO. 53-400-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC	AIRPLANE ALL ENGINE ALL
		ACCESS 833 843 S2301			ZONE 231 232
		NOTE			

Inspect passenger compartment from STA 360 to 663.75, including: 1. Skin panels (skins, frames and stringers), longitudinal lap splices, circumferential skin and stringer splices; 2. Window belt structure; 3. Overwing emergency exit cutout structure; 4. Forward cargo door cutout surround structure (portion in upper lobe); 5. STA 540 and 663 bulkheads and splices; 6. Overwing frames and stub beams.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove cabin interior as required. Remove/displace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT, STA 360 TO 663.75
		D633A109-AKS 53-400-00-01

Page 1 of 7
Jun 15/2015

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-400-00-01						
				MECH INSP						
TASK 53-05-03-210-836										
1. INTERNAL - GENERAL VISUAL: FORWARD PASSENGER COMPARTMENT, STA 360 to 663.75										
(Figure 1)										
A. Inspection										
SUBTASK 53-05-03-010-033										
(1) Open these access panels:										
<table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>833</td> <td>Emergency Exit</td> </tr> <tr> <td>843</td> <td>Emergency Exit</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	833	Emergency Exit	843	Emergency Exit	
<u>Number</u>	<u>Name/Location</u>									
833	Emergency Exit									
843	Emergency Exit									
Special Access:										
<table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>S2301</td> <td>Forward Passenger Compartment STA 360 to 663.75</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	S2301	Forward Passenger Compartment STA 360 to 663.75			
<u>Number</u>	<u>Name/Location</u>									
S2301	Forward Passenger Compartment STA 360 to 663.75									
<u>NOTE:</u> Remove cabin interior as required. Remove/displace insulation blankets as required.										
SUBTASK 53-05-03-210-036										
(2) Do a General Visual inspection of the passenger compartment from STA 360 to 663.75, including:										
<ol style="list-style-type: none"> 1. Skin panels (skins, frames and stringers), longitudinal lap splices, circumferential skin and stringer splices. 2. Window belt structure. 3. Overwing emergency exit cutout structure. 4. Forward cargo door cutout surround structure (portion in upper lobe). 5. STA 540 and 663 bulkheads and splices. 6. Overwing frames and stub beams. 										
SUBTASK 53-05-03-910-050										
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.										
SUBTASK 53-05-03-410-033										
(4) Close these access panels:										
<table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>833</td> <td>Emergency Exit</td> </tr> <tr> <td>843</td> <td>Emergency Exit</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	833	Emergency Exit	843	Emergency Exit	
<u>Number</u>	<u>Name/Location</u>									
833	Emergency Exit									
843	Emergency Exit									
— END OF TASK —										

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT, STA 360 TO 663.75
		D633A109-AKS 53-400-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-400-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-075				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

**EFFECTIVITY
AKS ALL** **SOURCE
MRB** **FORWARD PASSENGER COMPARTMENT, STA 360 TO 663.75**

D633A109-AKS **Page 3 of 7**
53-400-00-01 **Oct 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-400-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT, STA 360 TO 663.75
		D633A109-AKS 53-400-00-01

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Oct 15/2014

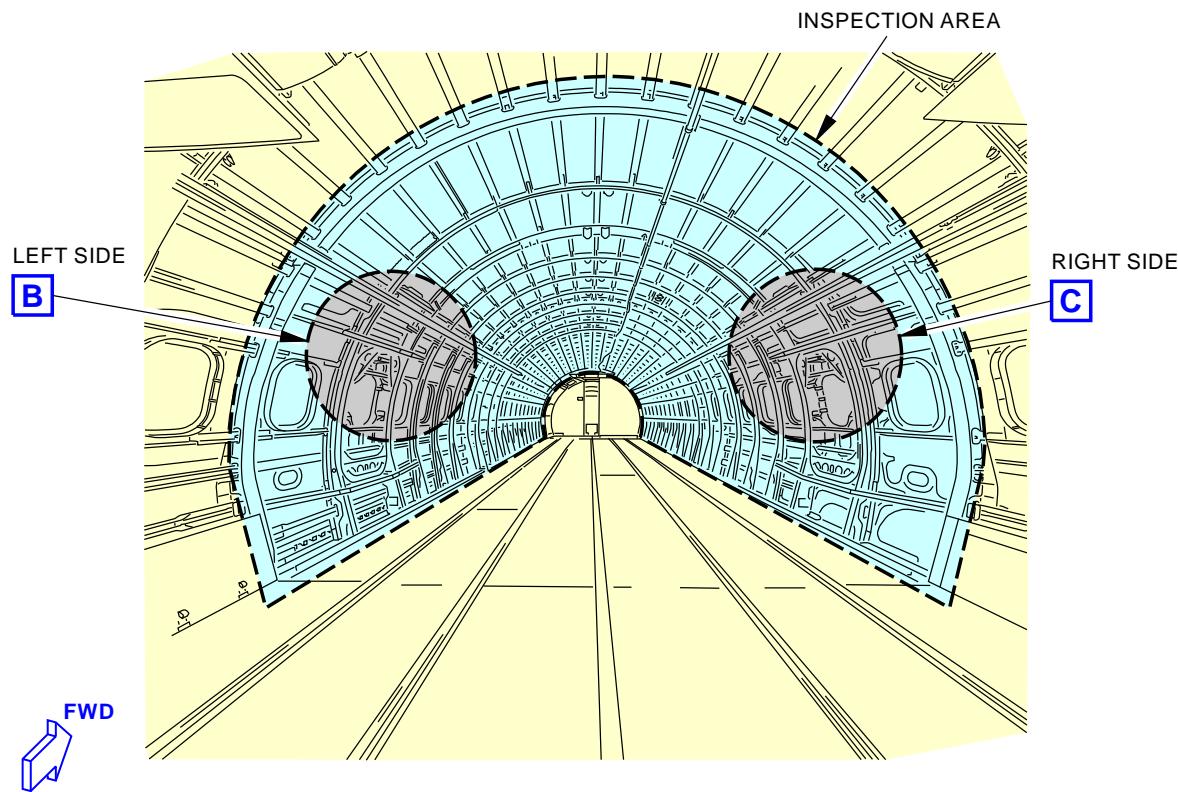
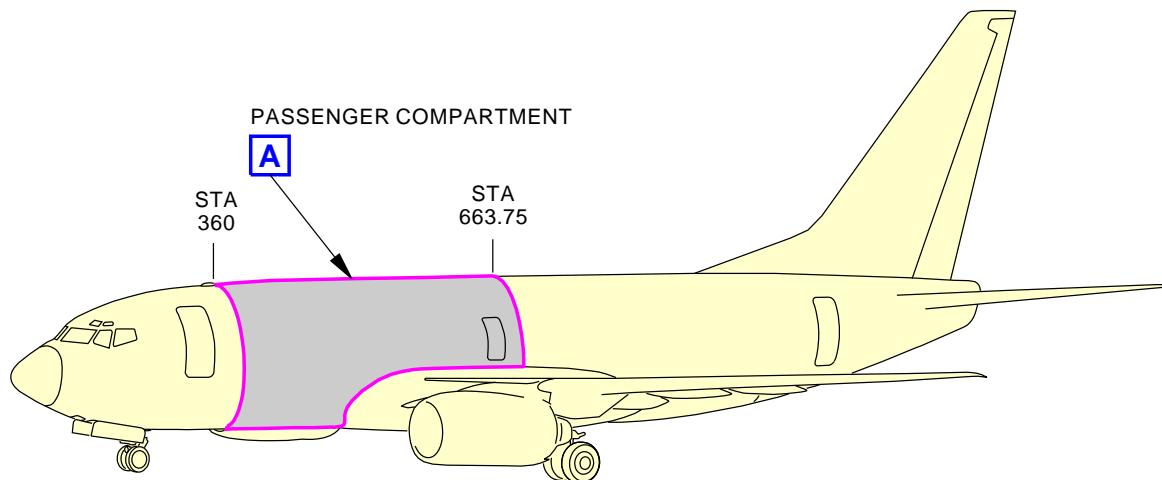
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

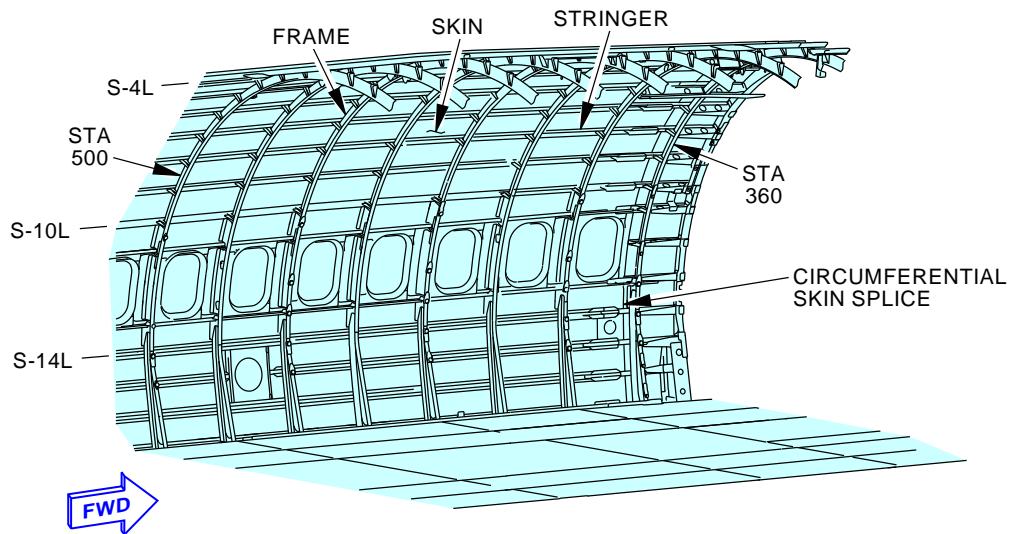
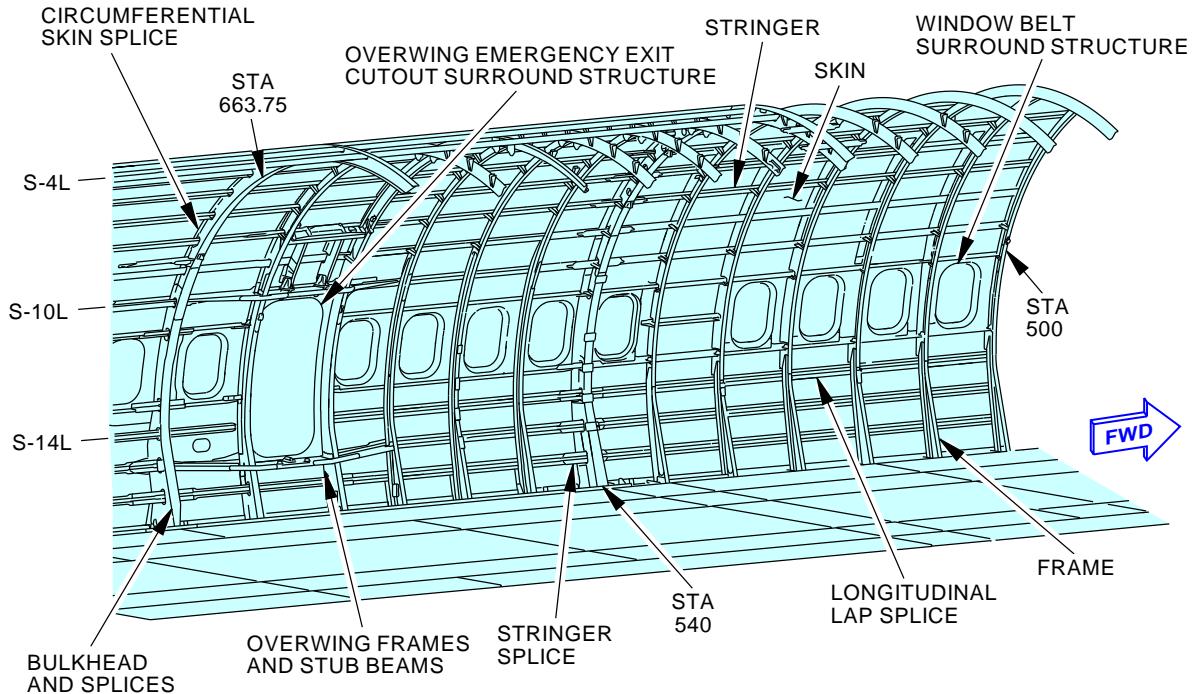
BOEING CARD NO.
53-400-00-01MPD ITEM
53-400-00**A**

2089687 S0000441136_V2

INTERNAL-GENERAL VISUAL: FORWARD PASSENGER COMPARTMENT, STA 360 TO STA 663.75
Figure 1 (Sheet 1 of 3)EFFECTIVITY
AKS ALLSOURCE
MRB**FORWARD PASSENGER COMPARTMENT, STA 360 TO 663.75****D633A109-AKS**
53-400-00-01**Page 5 of 7**
Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-400-00-01
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**LEFT SIDE - PASSENGER COMPARTMENT**MPD ITEM
53-400-00**B**

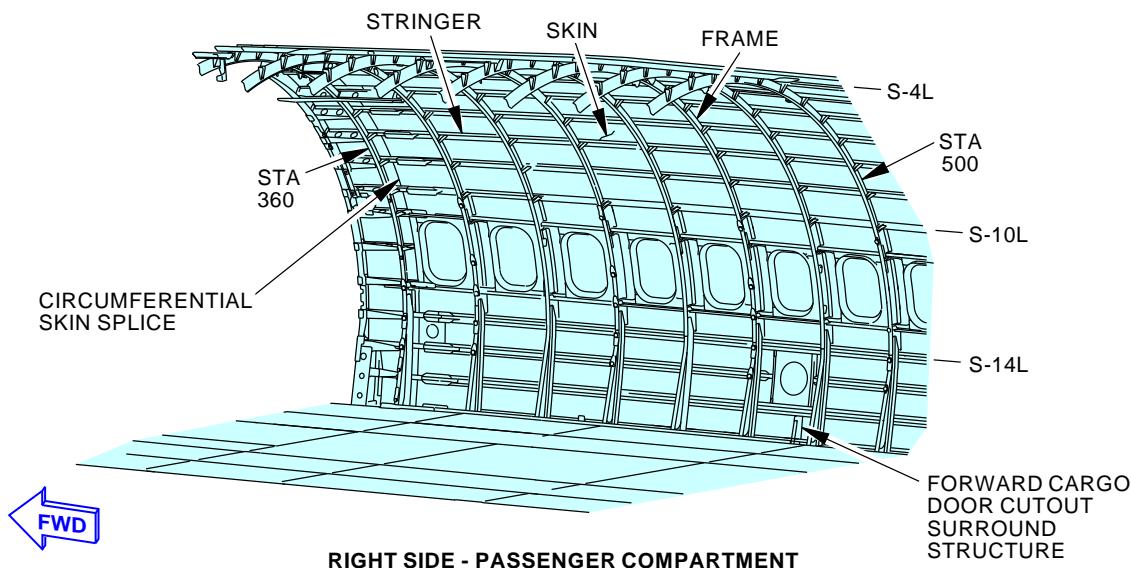
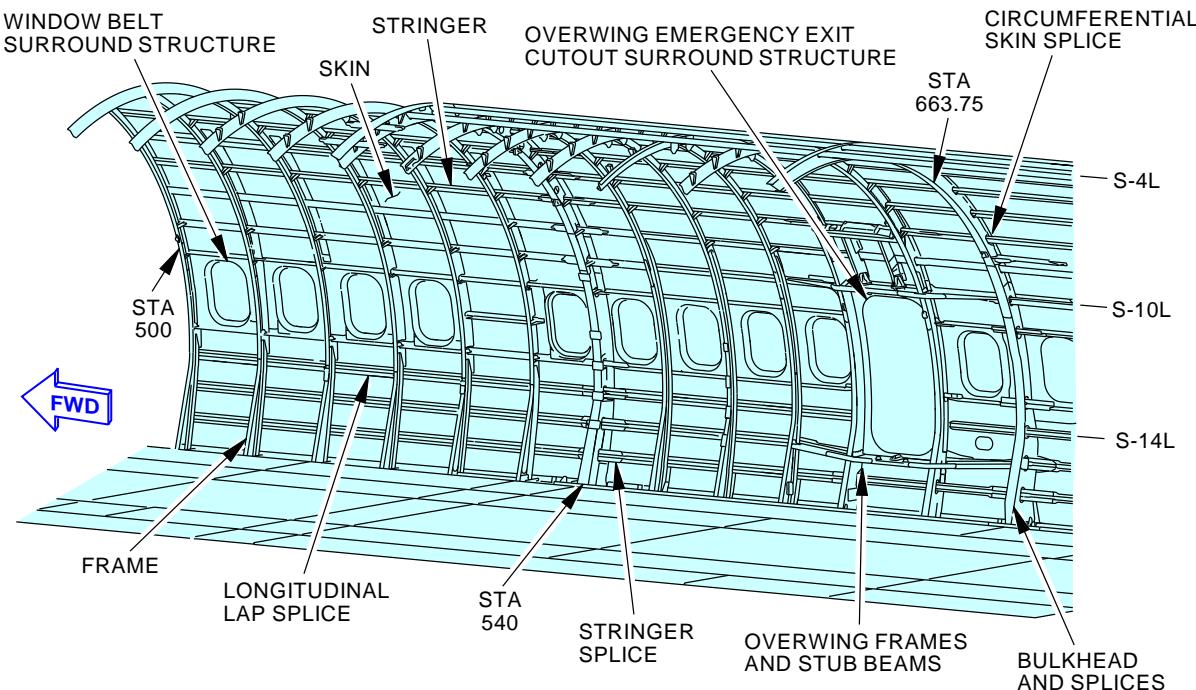
2093204 S0000441137_V2

INTERNAL-GENERAL VISUAL: FORWARD PASSENGER COMPARTMENT, STA 360 TO STA 663.75
Figure 1 (Sheet 2 of 3)

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT, STA 360 TO 663.75
		D633A109-AKS 53-400-00-01

AKS737-600/700/800/900
TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-400-00-01
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MPD ITEM
53-400-00**C**

2094904 S0000441138_V2

INTERNAL-GENERAL VISUAL: FORWARD PASSENGER COMPARTMENT, STA 360 TO STA 663.75
Figure 1 (Sheet 3 of 3)

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT, STA 360 TO 663.75
		D633A109-AKS 53-400-00-01

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Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT PASSENGER COMPARTMENT, STA 663.75 TO 1016			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-410-00-01
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC	AIRPLANE ALL
		NOTE			ENGINE ALL
		ACCESS S2401			NOTE
					ZONE 241 242

Inspect passenger compartment from STA 663.75 to 1016 (except areas around door cutouts), including: 1. Skin panels (skins, frames and stringers), longitudinal lap splices, circumferential skin and stringer splices (note: inspection includes the circumferential skin and stringer splice at Sta 727I for the -900 models); 2. Window belt structure; 3. STA 663 bulkhead and splices; 4. STA 727 bulkhead; 5. Side strut support frame at STA 706; 6. Main landing gear support frames at STA 695 and 716; 7. Wheel well frame at STA 685; 8. Aft cargo door cutout surround structure (portion in upper lobe); 9. Forward side of STA 1016 bulkhead (chords, pressure web, stiffeners, chord/web attachments), including vertical fin front spar fittings; 10. Stringer splice fittings and tension bolts at STA 1016.

INTERVAL NOTE: Whichever comes first.

AIRPLANE NOTE: Task not applicable to -900ER and -800 with Flat Pressure Bulkhead installed.

ACCESS NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/replace insulation blankets as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AFT PASSENGER COMPARTMENT, STA 663.75 TO 1016	
		D633A109-AKS 53-410-00-01	Page 1 of 8 Jun 15/2015

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-410-00-01				
				MECH INSP				
TASK 53-05-03-210-837								
1. INTERNAL - GENERAL VISUAL: AFT PASSENGER COMPARTMENT, STA 663.75 to 1016								
(Figure 1)								
A. Inspection								
SUBTASK 53-05-03-010-034								
(1) Special Access:								
<table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>S2401</td> <td>AFT Passenger Compartment STA 663.75 to 1016</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	S2401	AFT Passenger Compartment STA 663.75 to 1016	
<u>Number</u>	<u>Name/Location</u>							
S2401	AFT Passenger Compartment STA 663.75 to 1016							
NOTE: Remove galleys/lavs. Remove cabin interior as required. Remove/replace insulation blankets as required.								
SUBTASK 53-05-03-210-037								
(2) Do a General Visual inspection of the passenger compartment from STA 663.75 to 1016 (except areas around door cutouts), including:								
<ol style="list-style-type: none"> 1. Skin panels (skins, frames and stringers), longitudinal lap splices, circumferential skin and stringer splices (note: inspection includes the circumferential skin and stringer splice at Sta 727I for the -900 models). 2. Window belt structure. 3. STA 663 bulkhead and splices. 4. STA 727 bulkhead. 5. Side strut support frame at STA 706. 6. Main landing gear support frames at STA 695 and 716. 7. Wheel well frame at STA 685. 8. Aft cargo door cutout surround structure (portion in upper lobe). 9. Forward side of STA 1016 bulkhead (chords, pressure web, stiffeners, chord/web attachments), including vertical fin front spar fittings. 10. Stringer splice fittings and tension bolts at STA 1016. 								
SUBTASK 53-05-03-910-051								
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-808.								
———— END OF TASK ————								

EFFECTIVITY AKS ALL	SOURCE MRB	AFT PASSENGER COMPARTMENT, STA 663.75 TO 1016
		D633A109-AKS 53-410-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-410-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				
EFFECTIVITY AKS ALL	SOURCE MRB	AFT PASSENGER COMPARTMENT, STA 663.75 TO 1016		
		D633A109-AKS 53-410-00-01	Page 3 of 8 Oct 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-410-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
- (a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

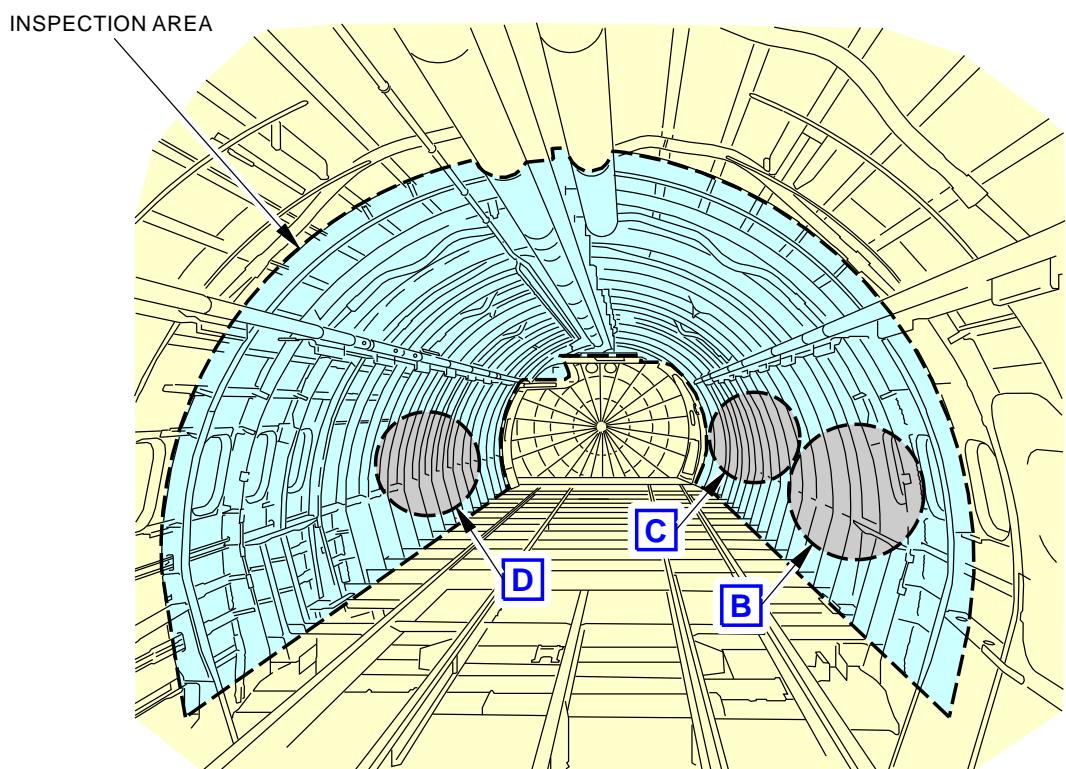
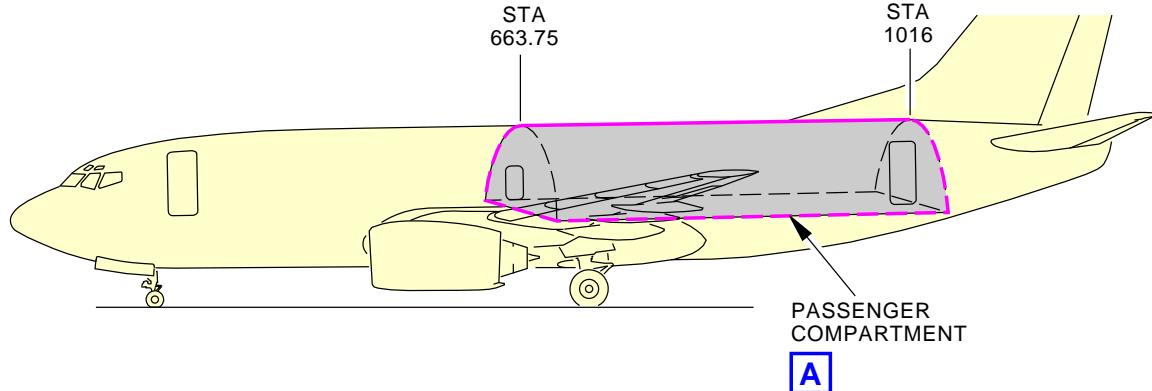
END OF TASK

EFFECTIVITY AKS ALL	SOURCE MRB	AFT PASSENGER COMPARTMENT, STA 663.75 TO 1016
		D633A109-AKS 53-410-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-410-00-01

MPD ITEM
53-410-00

2101150 S0000442502_V2

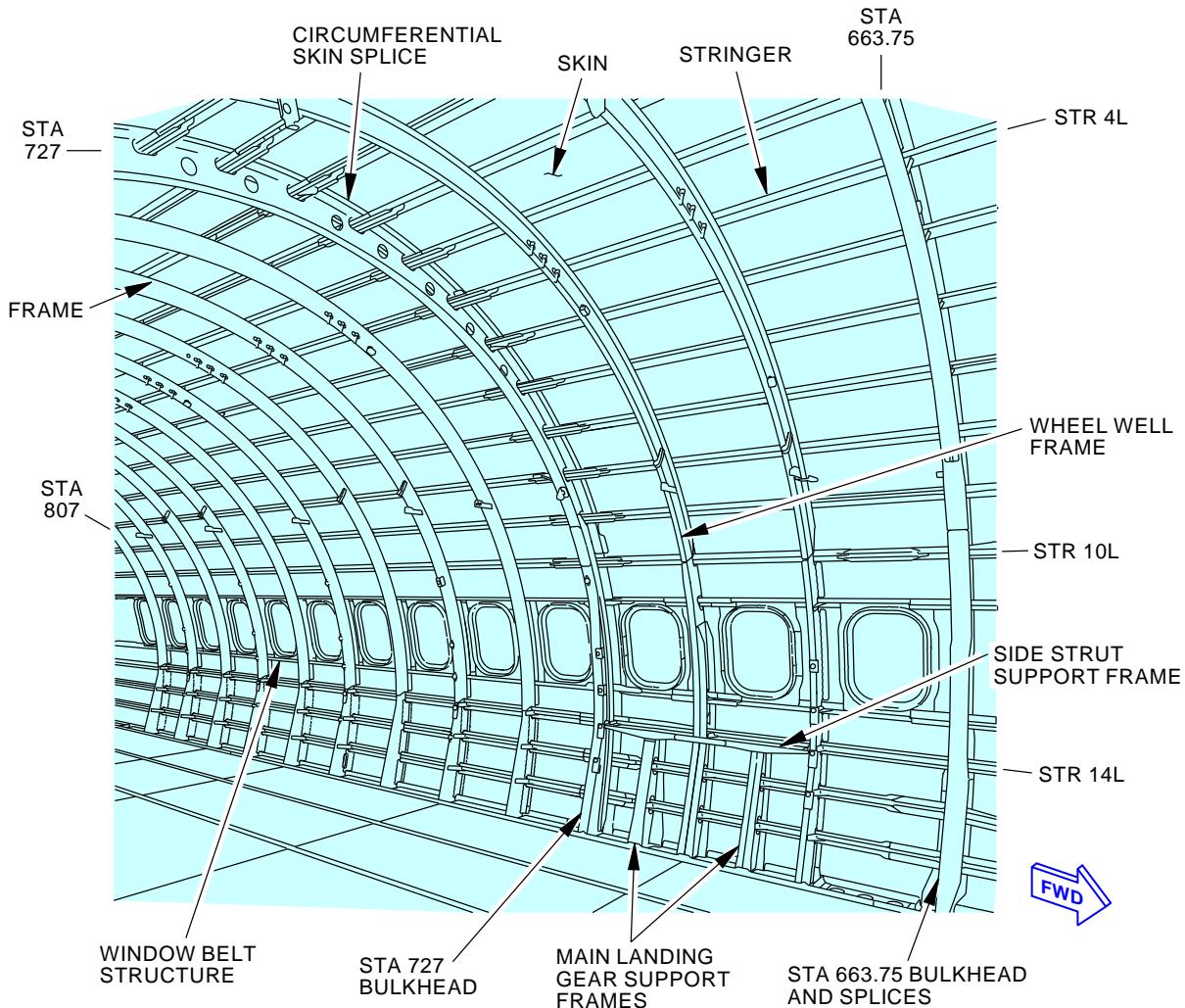
INTERNAL-GENERAL VISUAL: AFT PASSENGER COMPARTMENT, STA 663.75 TO STA 1016
Figure 1 (Sheet 1 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	AFT PASSENGER COMPARTMENT, STA 663.75 TO 1016
		D633A109-AKS 53-410-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-410-00-01
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**AFT PASSENGER COMPARTMENT
(STA 663.75 TO 807)**
(LEFT SIDE IS SHOWN, RIGHT SIDE IS ALMOST THE SAME)

MPD ITEM
53-410-00

2101205 S0000442500_V2

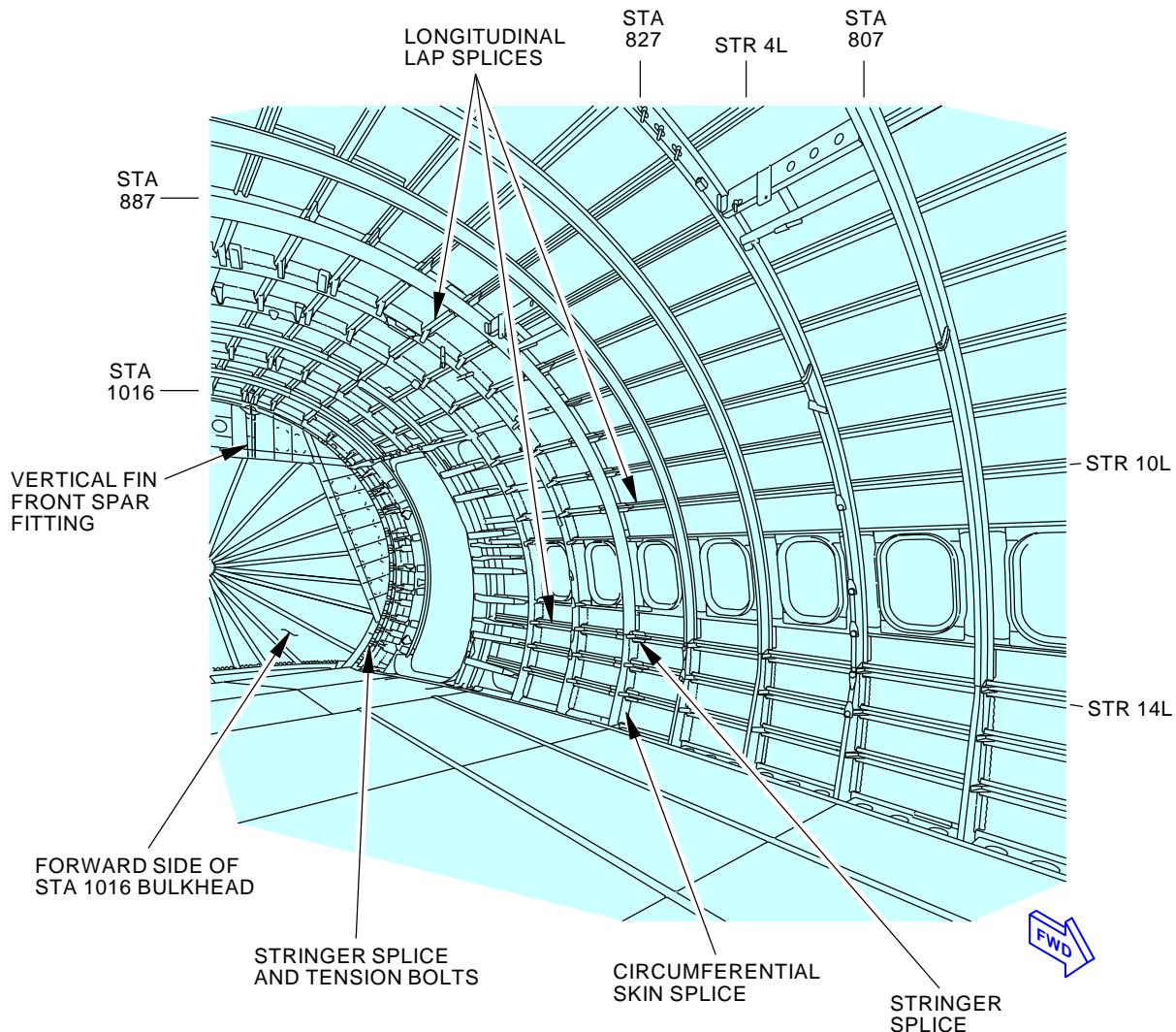
INTERNAL-GENERAL VISUAL: AFT PASSENGER COMPARTMENT, STA 663.75 TO STA 1016
Figure 1 (Sheet 2 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	AFT PASSENGER COMPARTMENT, STA 663.75 TO 1016
		D633A109-AKS 53-410-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-410-00-01



**AFT PASSENGER COMPARTMENT
(STA 807 TO 1016)**
(LEFT SIDE IS SHOWN, RIGHT SIDE IS ALMOST THE SAME)

MPD ITEM
53-410-00

2101274 S0000442503_V2

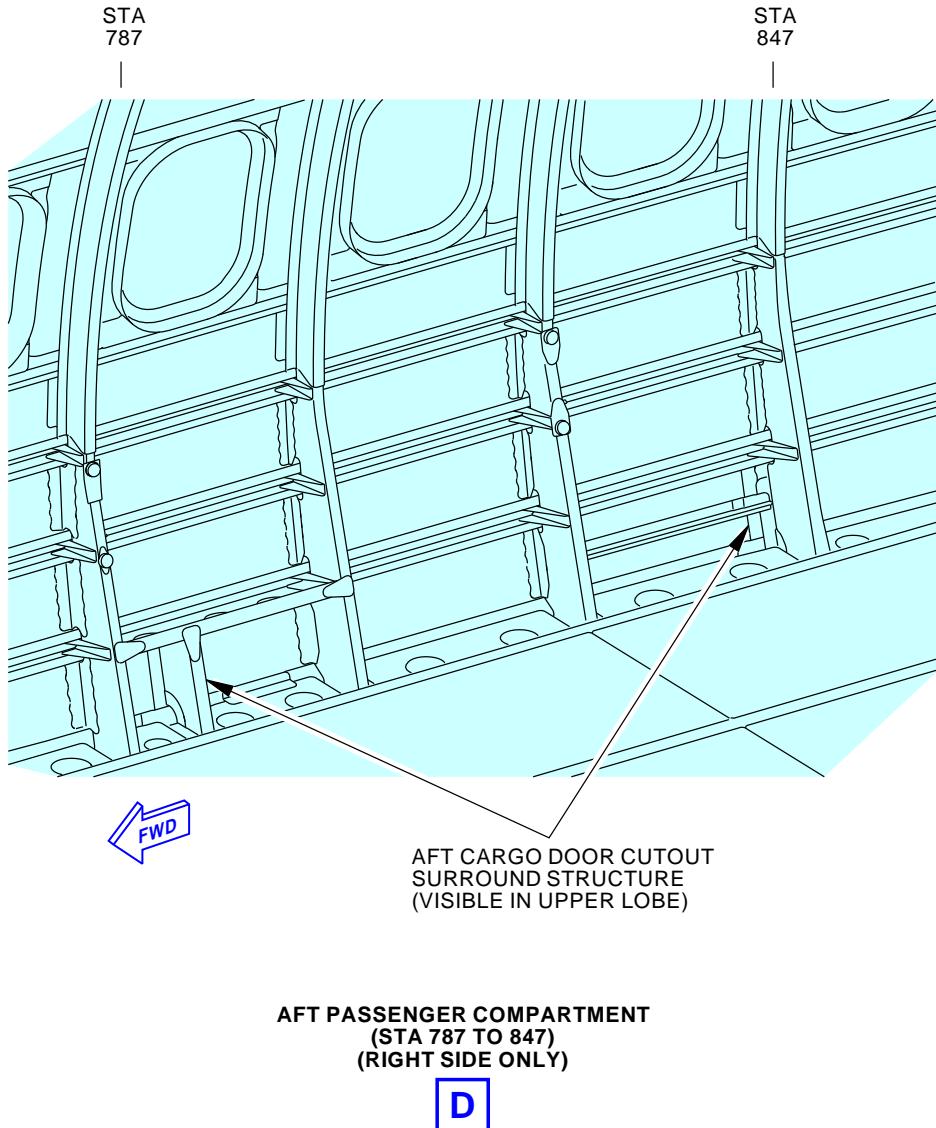
INTERNAL-GENERAL VISUAL: AFT PASSENGER COMPARTMENT, STA 663.75 TO STA 1016
Figure 1 (Sheet 3 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	AFT PASSENGER COMPARTMENT, STA 663.75 TO 1016
		D633A109-AKS 53-410-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-410-00-01

MPD ITEM
53-410-00

2101287 S0000442504_V2

INTERNAL-GENERAL VISUAL: AFT PASSENGER COMPARTMENT, STA 663.75 TO STA 1016
Figure 1 (Sheet 4 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	AFT PASSENGER COMPARTMENT, STA 663.75 TO 1016
		D633A109-AKS 53-410-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA AFT OF STA 1016 BULKHEAD			BOEING CARD NO.	
DATE	TASK GENERAL VISUAL				53-420-00-01	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY AIRPLANE ALL ENGINE ALL NOTE	
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC		
		NOTE ACCESS 311BL		ZONE 311 312		

Inspect area aft of STA 1016 pressure bulkhead to STA 1088, including: 1. Skin panels (skins, frames and stringers), longitudinal lap splices, circumferential skin and stringer splices; 2. Aft side of STA 1016 bulkhead (chords, pressure web, stiffeners, chord/web attachments); 3. Stringer splice fittings and tension bolts at STA 1016; 4. STA 1088 bulkhead, including vertical fin rear spar fittings and horizontal stabilizer center section jackscrew fitting lugs and bolts.

INTERVAL NOTE: Whichever comes first.

AIRPLANE NOTE: Task not applicable to -900ER and -800 with Flat Pressure Bulkhead installed.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF STA 1016 BULKHEAD
		D633A109-AKS 53-420-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-420-00-01	MECH	INSP

TASK 53-05-03-210-838**1. INTERNAL - GENERAL VISUAL: AREA AFT OF STA 1016 BULKHEAD**

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-035

- (1) Open this access panel:

Number Name/Location

311BL Stabilizer Trim Access Door

SUBTASK 53-05-03-210-038

- (2) Do a General Visual inspection of the area aft of STA 1016 pressure bulkhead to STA 1088, including:
1. Skin panels (skins, frames and stringers), longitudinal lap splices, circumferential skin and stringer splices.
 2. Aft side of STA 1016 bulkhead (chords, pressure web, stiffeners, chord/web attachments).
 3. Stringer splice fittings and tension bolts at STA 1016.
 4. STA 1088 bulkhead, including vertical fin rear spar fittings and horizontal stabilizer center section jackscrew fitting lugs and bolts.

SUBTASK 53-05-03-910-067

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.

SUBTASK 53-05-03-410-035

- (4) Close this access panel:

Number Name/Location

311BL Stabilizer Trim Access Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF STA 1016 BULKHEAD
		D633A109-AKS 53-420-00-01

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Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-420-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF STA 1016 BULKHEAD D633A109-AKS 53-420-00-01	Page 3 of 8 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-420-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-063

(7) CPCP Basic Task Item 7 is not applicable.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF STA 1016 BULKHEAD
		D633A109-AKS 53-420-00-01

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Oct 15/2014

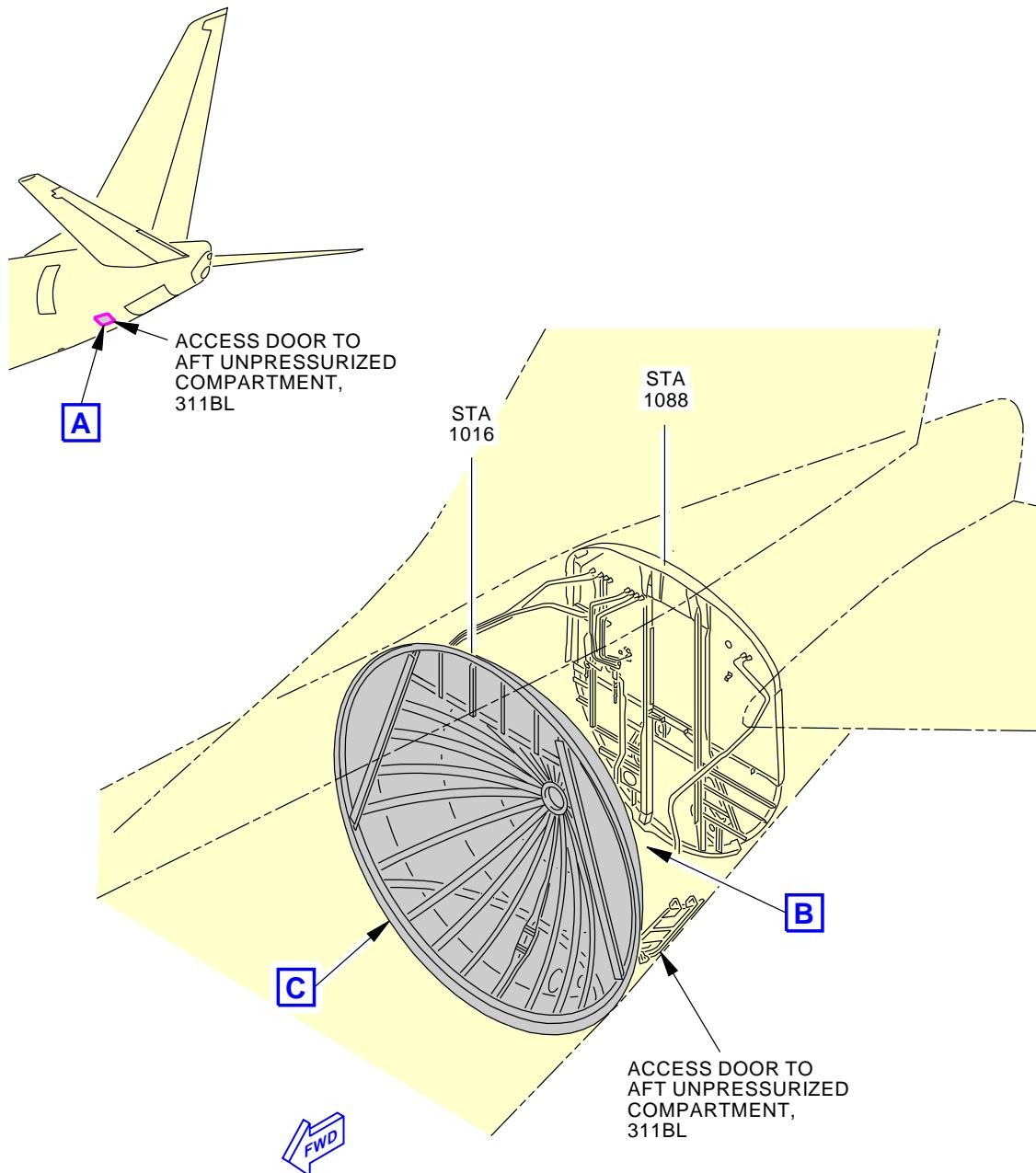
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-420-00-01MPD ITEM
53-420-00**A**

2100609 S0000445724_V2

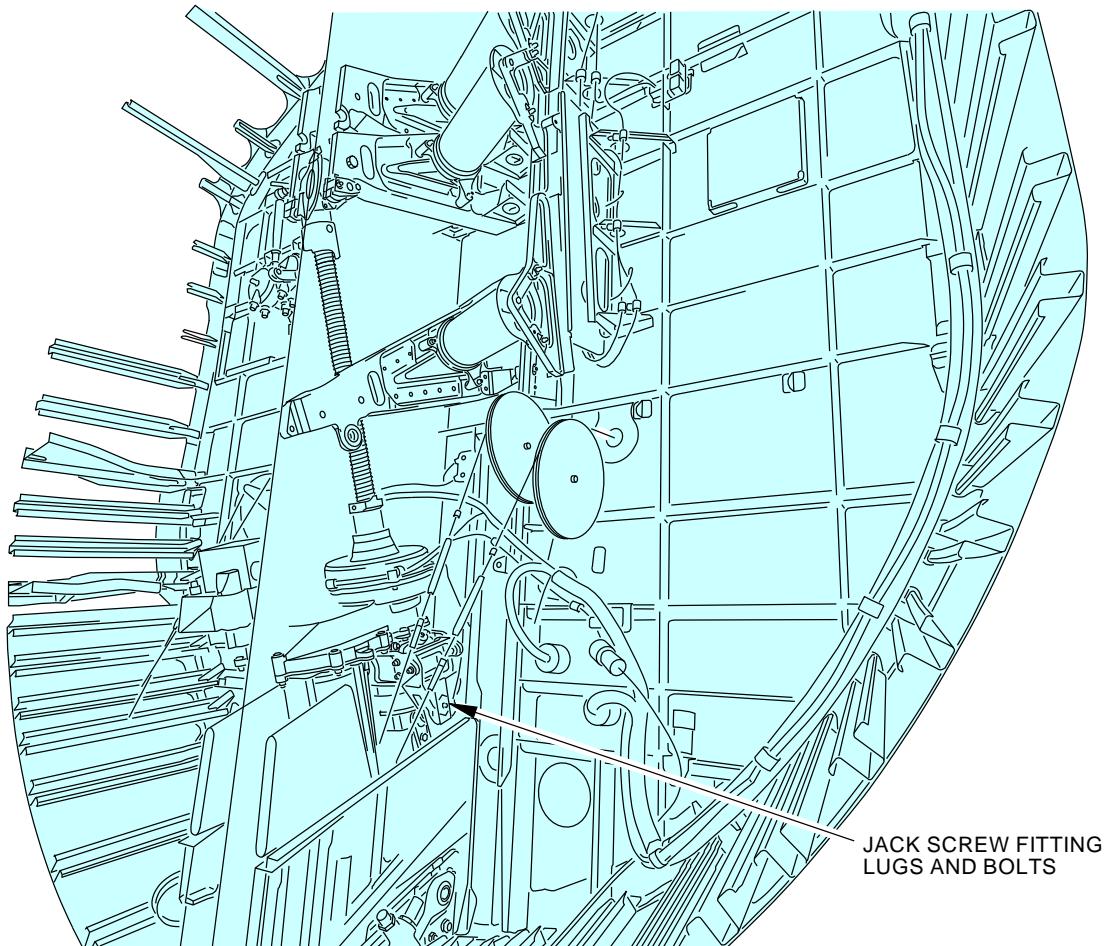
INTERNAL-GENERAL VISUAL: AREA AFT OF STA 1016 BULKHEAD
Figure 1 (Sheet 1 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF STA 1016 BULKHEAD
		D633A109-AKS 53-420-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-420-00-01



VIEW FROM BACK

MPD ITEM
53-420-00

2093789 S0000441634_V2

INTERNAL-GENERAL VISUAL: AREA AFT OF STA 1016 BULKHEAD
Figure 1 (Sheet 2 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF STA 1016 BULKHEAD
		D633A109-AKS 53-420-00-01

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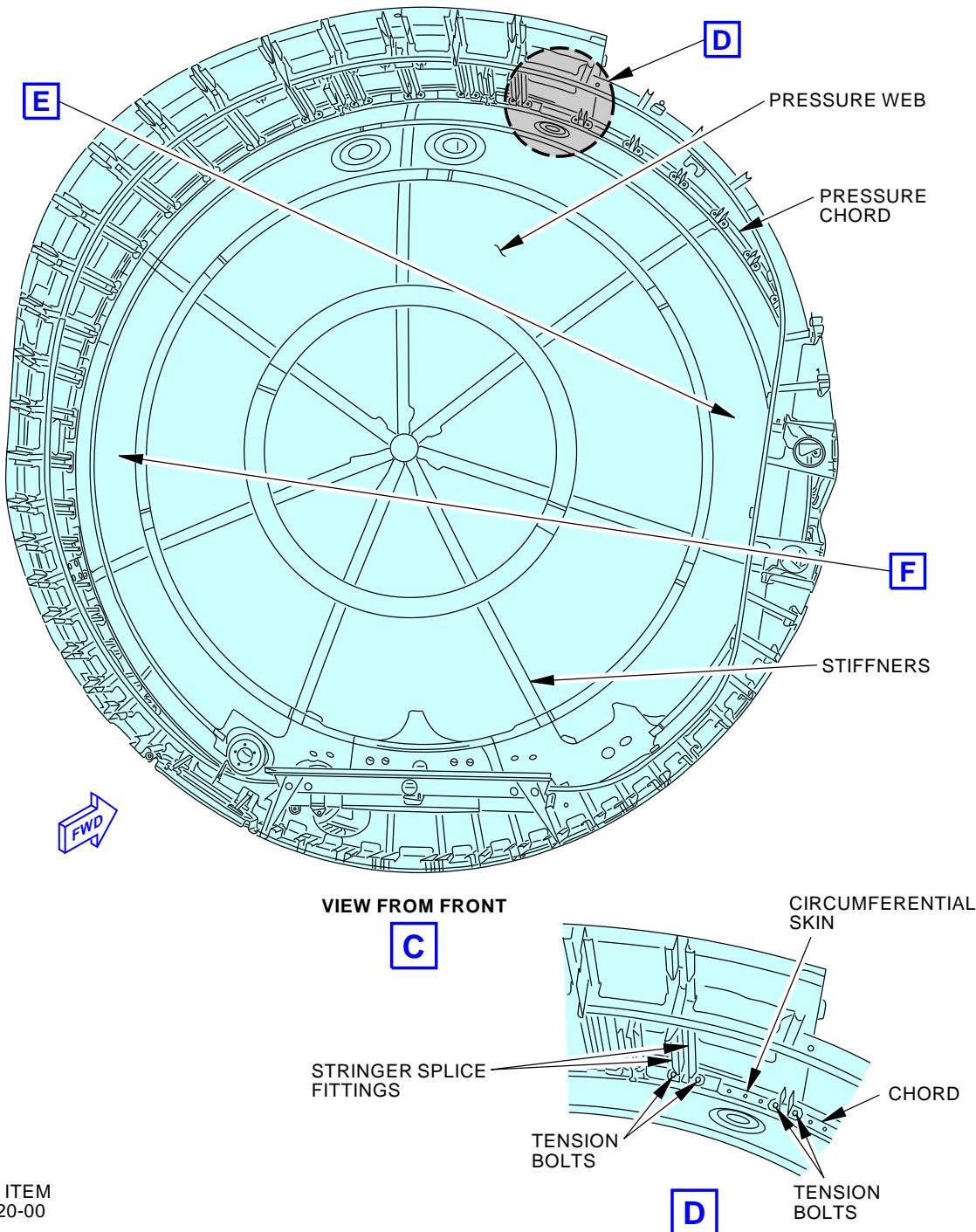
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-420-00-01MPD ITEM
53-420-00

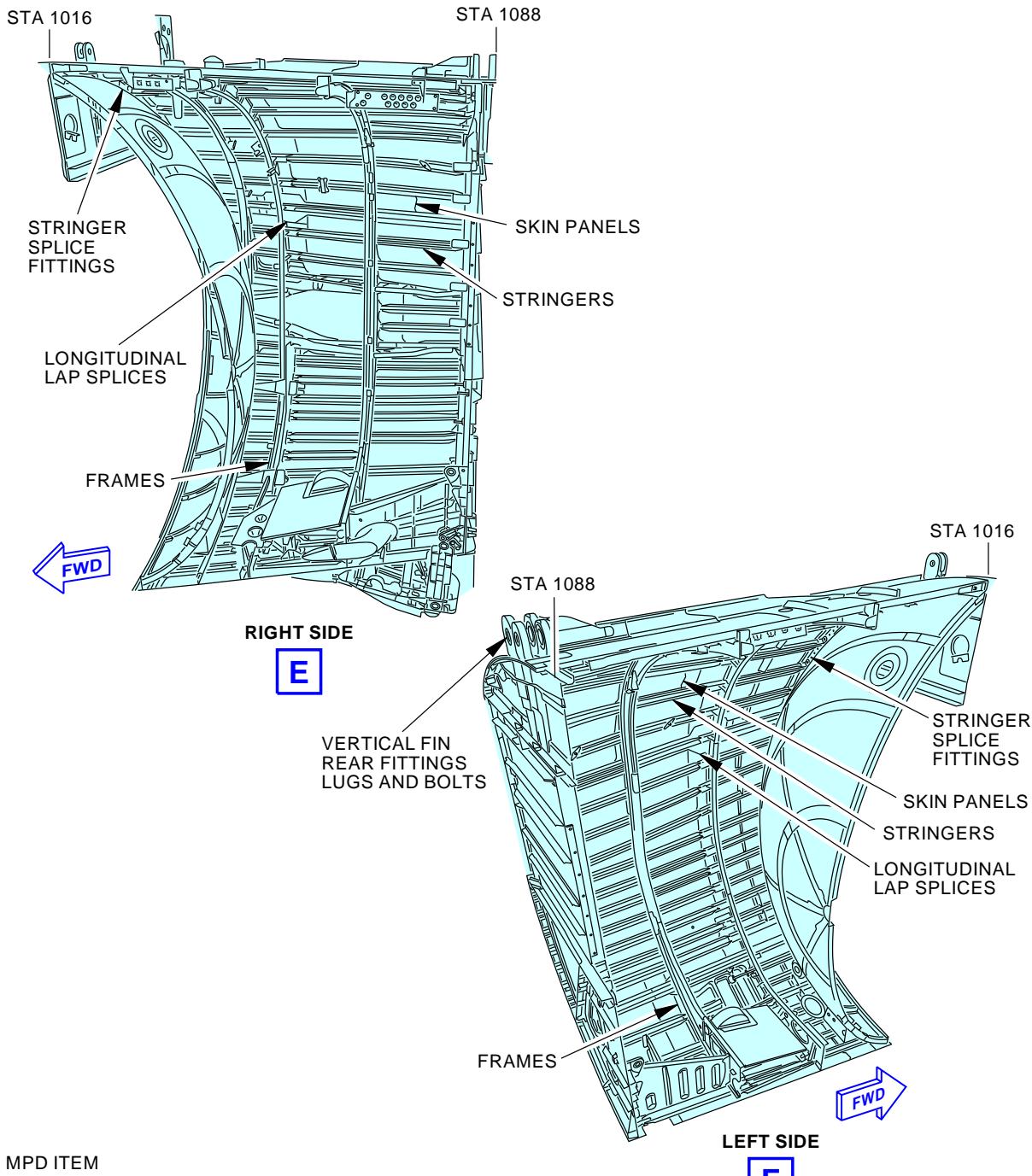
2098538 S0000444916_V2

INTERNAL-GENERAL VISUAL: AREA AFT OF STA 1016 BULKHEAD
Figure 1 (Sheet 3 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF STA 1016 BULKHEAD
		D633A109-AKS 53-420-00-01

AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-420-00-01
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MPD ITEM
53-420-00

VIEW FROM FRONT
INTERNAL-GENERAL VISUAL: AREA AFT OF STA 1016 BULKHEAD
Figure 1 (Sheet 4 of 4)

2097088 S0000441632_V3

EFFECTIVITY AKS ALL	SOURCE MRB	AREA AFT OF STA 1016 BULKHEAD
		D633A109-AKS 53-420-00-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE STABILIZER TORSION BOX COMPARTMENT AND APU COMPARTMENT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-430-00-01
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2	36000 FC	24000 FC	AIRPLANE ALL ENGINE ALL
		ACCESS 311BL 315A 331A 332AB 332AT 333AB 333AT 341A 342AB 342AT 343AB 343AT S3101			ZONE 313 314 315 316
		NOTE			

Inspect stabilizer torsion box compartment and APU compartment, including: 1. Skin panels (skins, frames and stringers), longitudinal lap splices; 2. STA 1088 bulkhead, including vertical fin rear spar fittings; 3. Forward side of STA 1156 bulkhead, including horizontal stabilizer hinge fittings and bolts; 4. Upper horizontal deck (at stringer 6) and lower horizontal deck (at stringer 12).

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: For area below stringer 12, remove APU and firewalls; remove APU plenum as required. For area above stringer 12, adjust stabilizer trim as required. For access to Sta 1156 horizontal stabilizer hinge fitting lugs and bolts, remove gap seal and horizontal stabilizer rear spar sliding seal as required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	STABILIZER TORSION BOX COMPARTMENT AND APU COMPARTMENT
		D633A109-AKS 53-430-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-430-00-01																						
				MECH INSP																						
TASK 53-05-03-210-839																										
1. INTERNAL - GENERAL VISUAL: STABILIZER TORSION BOX COMPARTMENT AND APU COMPARTMENT																										
(Figure 1)																										
A. Inspection																										
SUBTASK 53-05-03-010-036																										
(1) Open these access panels:																										
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>311BL</td><td>Stabilizer Trim Access Door</td></tr><tr><td>315A</td><td>APU Cowl Door</td></tr><tr><td>331A</td><td>Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body</td></tr><tr><td>332AB</td><td>Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body</td></tr><tr><td>332AT</td><td>Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body</td></tr><tr><td>341A</td><td>Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body</td></tr><tr><td>342AB</td><td>Horizontal Stabilizer, Gap Cover - H. Stab. to Body</td></tr><tr><td>342AT</td><td>Gap Cover, Horizontal Stabilizer</td></tr><tr><td>343AB</td><td>Horizontal Stabilizer, Gap Cover - H. Stab. to Body</td></tr><tr><td>343AT</td><td>Horizontal Stabilizer, Gap Cover - H. Stab. to Body</td></tr></tbody></table>					Number	Name/Location	311BL	Stabilizer Trim Access Door	315A	APU Cowl Door	331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body	332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body	332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body	341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body	342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body	342AT	Gap Cover, Horizontal Stabilizer	343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body	343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
Number	Name/Location																									
311BL	Stabilizer Trim Access Door																									
315A	APU Cowl Door																									
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body																									
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body																									
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body																									
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body																									
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body																									
342AT	Gap Cover, Horizontal Stabilizer																									
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body																									
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body																									
Special Access:																										
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>S3101</td><td>Stabilizer Torsion Box Compartment and APU Compartment Inspection</td></tr></tbody></table>					Number	Name/Location	S3101	Stabilizer Torsion Box Compartment and APU Compartment Inspection																		
Number	Name/Location																									
S3101	Stabilizer Torsion Box Compartment and APU Compartment Inspection																									
NOTE: For area below stringer 12, remove APU and firewalls; remove APU plenum as required. For area above stringer 12, adjust stabilizer trim as required. For access to Sta 1156 horizontal stabilizer hinge fitting lugs and bolts, remove gap seal and horizontal stabilizer rear spar sliding seal as required.																										
SUBTASK 53-05-03-210-039																										
(2) Do a General Visual inspection of the stabilizer torsion box compartment and APU compartment, including:																										
1. Skin panels (skins, frames and stringers), longitudinal lap splices. 2. STA 1088 bulkhead, including vertical fin rear spar fittings. 3. Forward side of STA 1156 bulkhead, including horizontal stabilizer hinge fittings and bolts. 4. Upper horizontal deck (at stringer 6) and lower horizontal deck (at stringer 12).																										
SUBTASK 53-05-03-910-068																										
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-804.																										
SUBTASK 53-05-03-410-072																										
(4) Close this access panel:																										
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>343AB</td><td>Horizontal Stabilizer, Gap Cover - H. Stab. to Body</td></tr></tbody></table>					Number	Name/Location	343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body																		
Number	Name/Location																									
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body																									
(a) Make sure that the blade seal is installed correctly into the forward track channel.																										

EFFECTIVITY AKS ALL	SOURCE MRB	STABILIZER TORSION BOX COMPARTMENT AND APU COMPARTMENT
		D633A109-AKS 53-430-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-430-00-01
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SUBTASK 53-05-03-410-036

- (5) Close these access panels:

Number Name/Location

311BL	Stabilizer Trim Access Door
315A	APU Cowl Door
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	STABILIZER TORSION BOX COMPARTMENT AND APU COMPARTMENT
		D633A109-AKS 53-430-00-01

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Jun 15/2016

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-430-00-01
				MECH INSP
TASK 51-05-01-210-804				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-043				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-044				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-045				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-046				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-111				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-048				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY
AKS ALLSOURCE
MRB**STABILIZER TORSION BOX COMPARTMENT AND APU
COMPARTMENT****D633A109-AKS
53-430-00-01****Page 4 of 6
Oct 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-430-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	STABILIZER TORSION BOX COMPARTMENT AND APU COMPARTMENT
		D633A109-AKS 53-430-00-01

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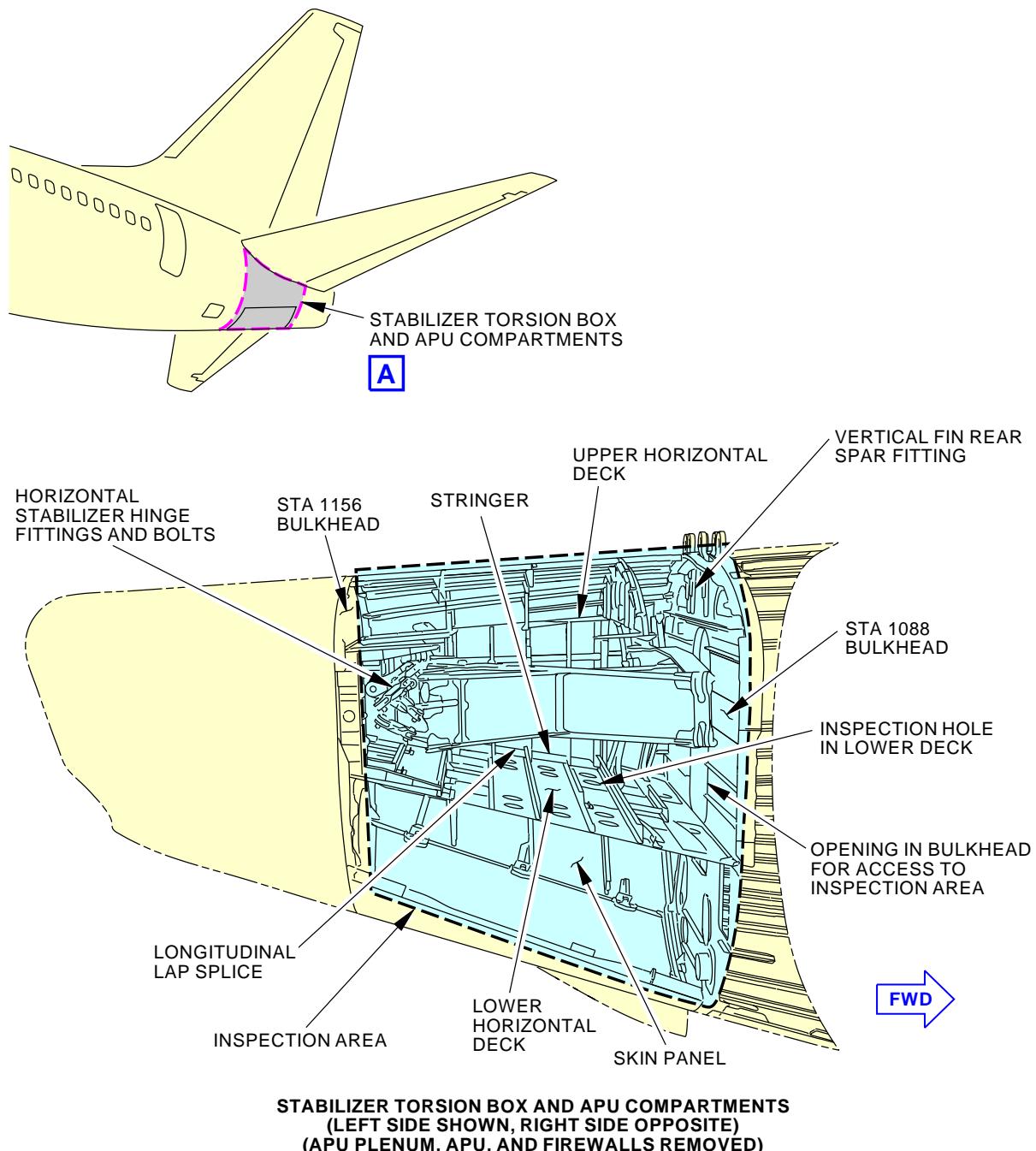
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-430-00-01

2097200 S0000441491_V2

EFFECTIVITY AKS ALL	SOURCE MRB	STABILIZER TORSION BOX COMPARTMENT AND APU COMPARTMENT
		D633A109-AKS 53-430-00-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE STA 1156 BULKHEAD			BOEING CARD NO. 53-440-00-01	
DATE	TASK GENERAL VISUAL				RELATED CARD	
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY AIRPLANE ALL ENGINE ALL	
STATION	SKILL AIRPL	1.2 NOTE	36000 FC	24000 FC		
		ACCESS 318BR				
			ZONE 317 318			

Inspect aft side of STA 1156 bulkhead.

INTERVAL NOTE: Whichever comes first.**A. References**

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	STA 1156 BULKHEAD D633A109-AKS 53-440-00-01	Page 1 of 5 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-440-00-01
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TASK 53-05-03-210-840

MECH

INSP

1. INTERNAL - GENERAL VISUAL: STA 1156 BULKHEAD

(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-037

- (1) Open this access panel:

Number Name/Location

318BR Tailcone Access Door

SUBTASK 53-05-03-210-040

- (2) Do a General Visual inspection of the aft side of STA 1156 bulkhead.

SUBTASK 53-05-03-910-069

- (3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.

SUBTASK 53-05-03-410-037

- (4) Close this access panel:

Number Name/Location

318BR Tailcone Access Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	STA 1156 BULKHEAD
		D633A109-AKS 53-440-00-01

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Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-440-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	STA 1156 BULKHEAD D633A109-AKS 53-440-00-01	Page 3 of 5 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-440-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-063

(7) CPCP Basic Task Item 7 is not applicable.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	STA 1156 BULKHEAD
		D633A109-AKS 53-440-00-01

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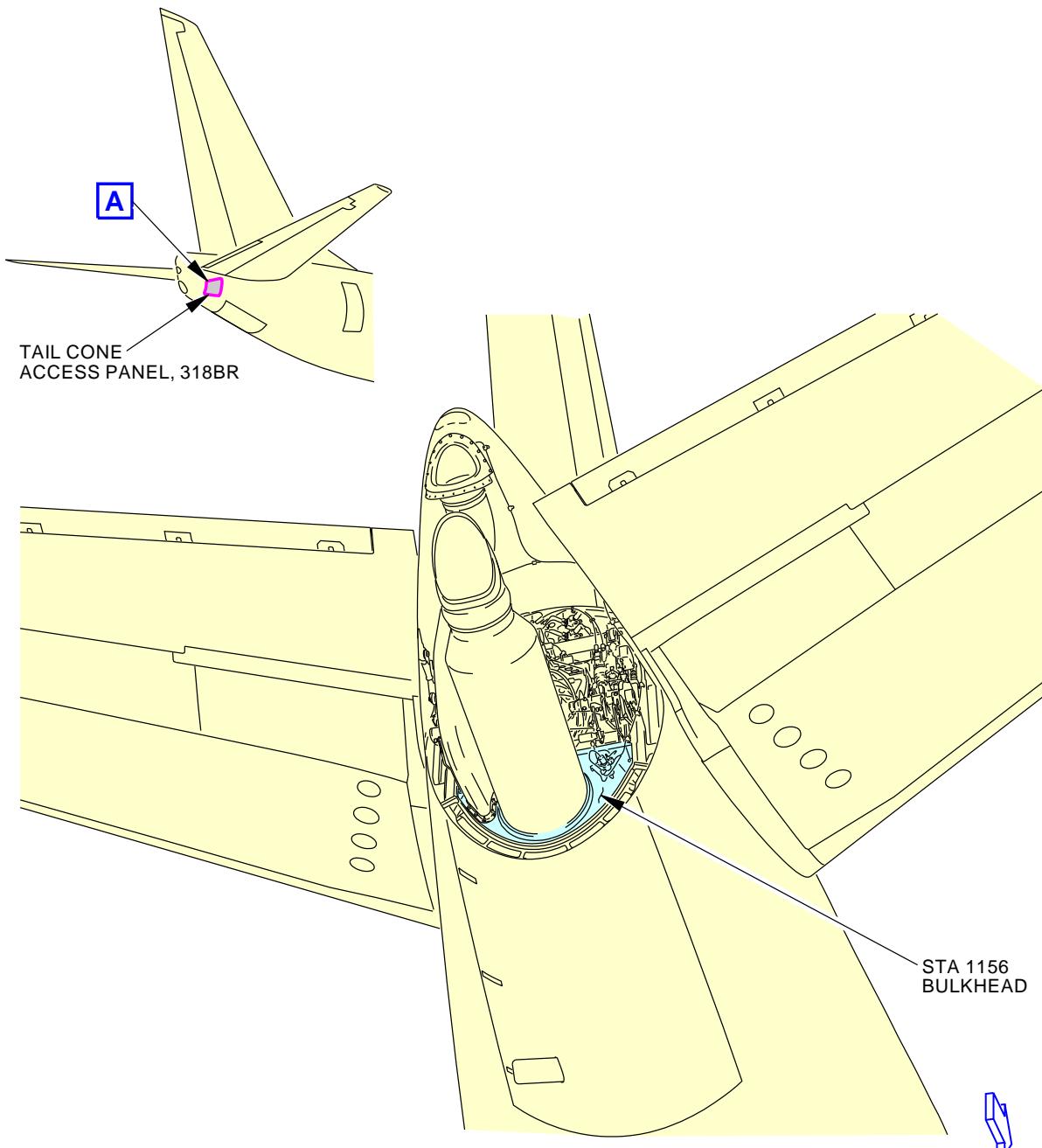
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-440-00-01MPD ITEM
53-440-00

2081370 S0000437527_V3

**INTERNAL-GENERAL VISUAL: STA 1156 BULKHEAD
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**STA 1156 BULKHEAD****D633A109-AKS
53-440-00-01****Page 5 of 5
Oct 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FUSELAGE SKIN UNDER DORSAL FIN			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-450-00-01
TAIL NUMBER	WORK AREA VERT STABILIZER	VERSION 1.1 1.2	THRESHOLD 9 YR 24000 FC	REPEAT 8 YR 24000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 321A			ZONE 321

Inspect fuselage skin under dorsal fin and aft to Sta 1016, including circumferential splice.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE SKIN UNDER DORSAL FIN	
		D633A109-AKS 53-450-00-01	Page 1 of 5 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-450-00-01
				MECH INSP
TASK 53-05-03-210-841				
1. INTERNAL - GENERAL VISUAL: FUSELAGE SKIN UNDER DORSAL FIN				
(Figure 1)				
A. Inspection				
SUBTASK 53-05-03-010-038				
(1) Open this access panel:				
Number Name/Location				
321A Vertical Fin, Dorsal Fin				
SUBTASK 53-05-03-210-041				
(2) Do a General Visual inspection of the fuselage skin under dorsal fin and aft to Sta 1016, including circumferential splice.				
SUBTASK 53-05-03-910-070				
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.				
SUBTASK 53-05-03-410-038				
(4) Close this access panel:				
Number Name/Location				
321A Vertical Fin, Dorsal Fin				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE SKIN UNDER DORSAL FIN	
		D633A109-AKS 53-450-00-01	Page 2 of 5 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-450-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE SKIN UNDER DORSAL FIN	
		D633A109-AKS 53-450-00-01	Page 3 of 5 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-450-00-01				
				<table border="1"><thead><tr><th>MECH</th><th>INSP</th></tr></thead><tbody><tr><td></td><td></td></tr></tbody></table>	MECH	INSP		
MECH	INSP							

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:
Water displacing / anti-corrosion compounds should not be applied in the following areas:
 - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
 - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
 - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
 - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
 - Areas with electrical arc potential.
 - Interior materials, including cargo liners (change of flammability properties).
 - Fiber-glass ducts where temperature exceeds 220 degrees F.
 - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-063

- (7) CPCP Basic Task Item 7 is not applicable.

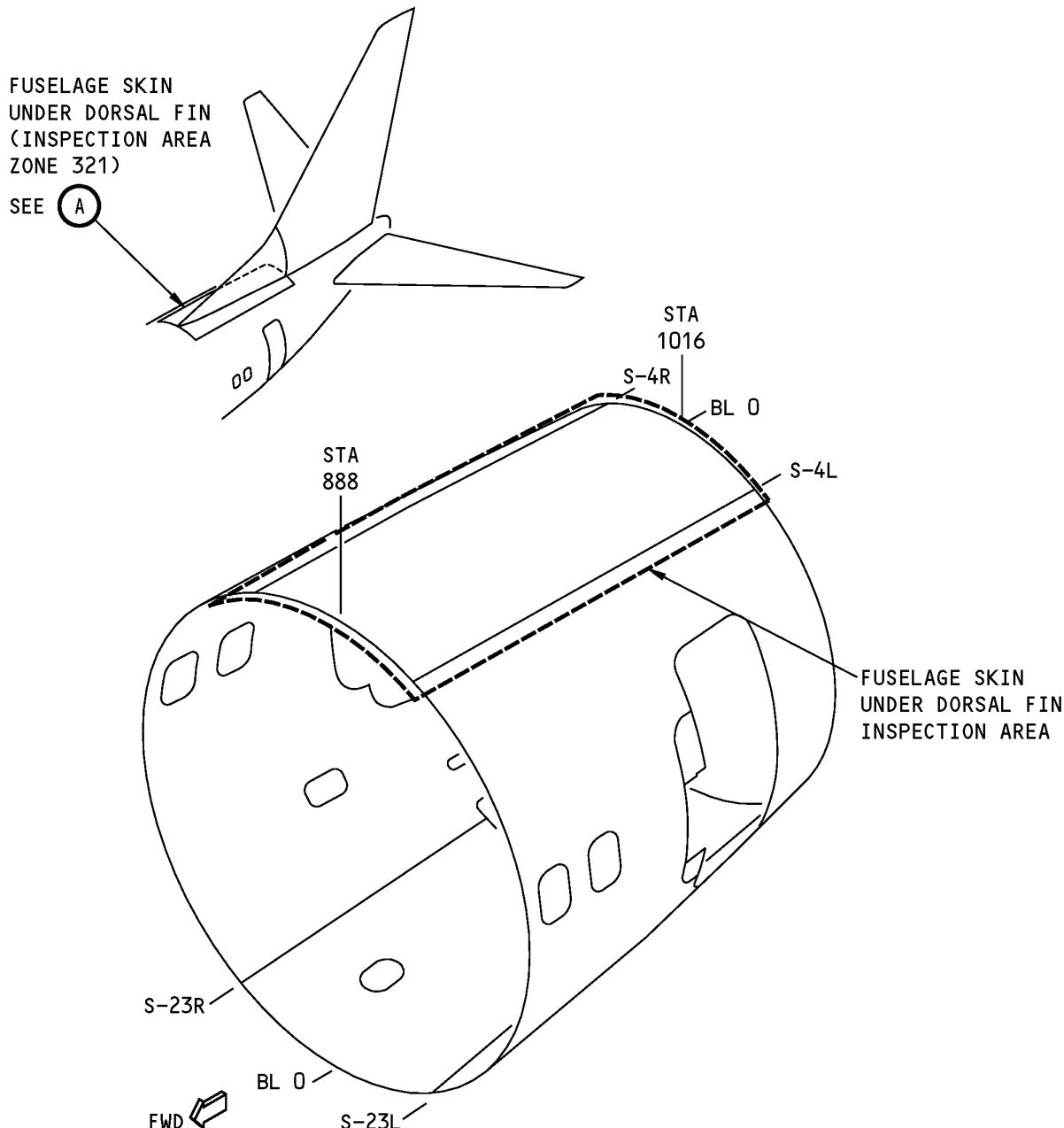
———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE SKIN UNDER DORSAL FIN
		D633A109-AKS 53-450-00-01

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Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-450-00-01

**FUSELAGE SKIN UNDER DORSAL FIN****(A)****Internal-General Visual: Fuselage Skin Under Dorsal Fin
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	FUSELAGE SKIN UNDER DORSAL FIN
		D633A109-AKS 53-450-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE VERTICAL FIN FRONT SPAR FITTING			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-460-00-01
TAIL NUMBER	WORK AREA VERT STABLIZER	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	NOTE			
		ACCESS 323AL 323AR 323BL 323BR NOTE			ZONE 322

Inspect vertical fin front spar fitting lugs and bolts (Sta 1016).

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Pin removal is not required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-460-00-01

Page 1 of 8
Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-460-00-01
TASK 53-05-03-210-842				MECH INSP

1. INTERNAL - GENERAL VISUAL: VERTICAL FIN FRONT SPAR FITTING
(Figure 1)

A. Inspection

SUBTASK 53-05-03-010-039

(1) Open these access panels:

<u>Number</u>	<u>Name/Location</u>
323AL	Vertical Fin, Front Spar Access Door
323AR	Vertical Fin, Front Spar Access Door
323BL	Vertical Fin, Forward Fin Access Door
323BR	Vertical Fin, Forward Fin Access Door

NOTE: Pin removal is not required.

SUBTASK 53-05-03-210-042

(2) Do a General Visual inspection of the vertical fin front spar fitting lugs and bolts (Sta 1016).

SUBTASK 53-05-03-910-071

(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.

SUBTASK 53-05-03-410-039

(4) Close these access panels:

<u>Number</u>	<u>Name/Location</u>
323AL	Vertical Fin, Front Spar Access Door
323AR	Vertical Fin, Front Spar Access Door
323BL	Vertical Fin, Forward Fin Access Door
323BR	Vertical Fin, Forward Fin Access Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING	
		D633A109-AKS 53-460-00-01	Page 2 of 8 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-460-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING D633A109-AKS 53-460-00-01	Page 3 of 8 Oct 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-460-00-01	
			<p>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</p> <p>2) Not applicable</p> <p>3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:</p> <p>Water displacing / anti-corrosion compounds should not be applied in the following areas:</p> <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.	MECH	INSP

SUBTASK 51-05-01-210-063

(7) CPCP Basic Task Item 7 is not applicable.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-460-00-01

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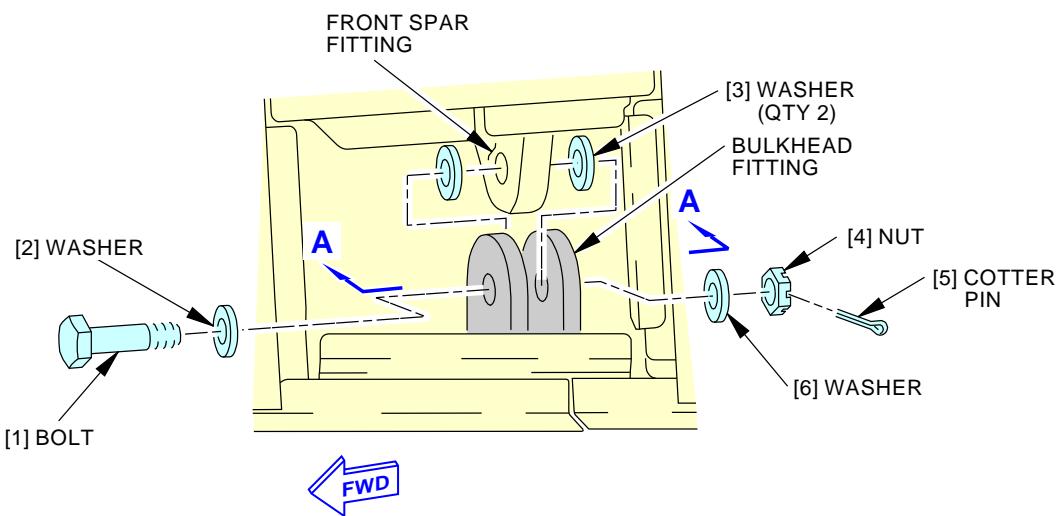
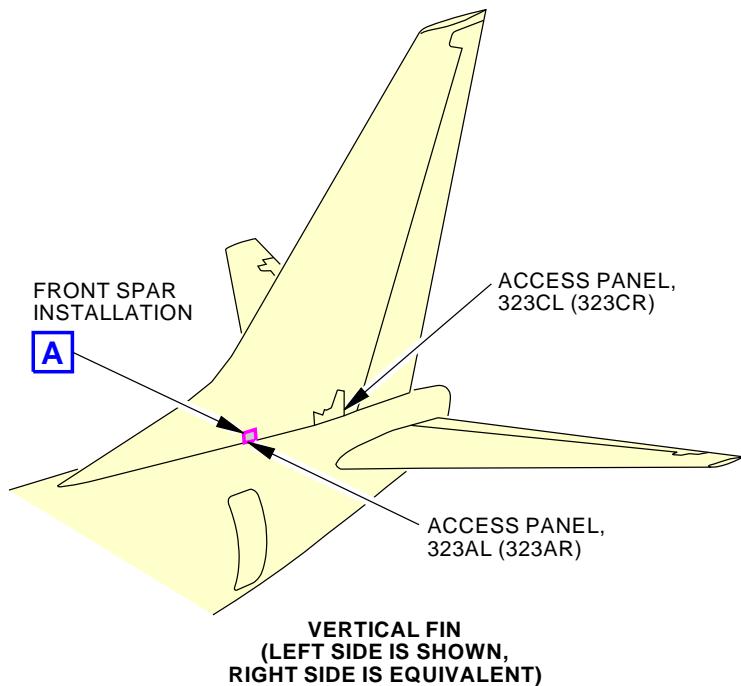
AKS737-600/700/800/900
TASK CARDS

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-460-00-01MPD ITEM
53-460-00**A**

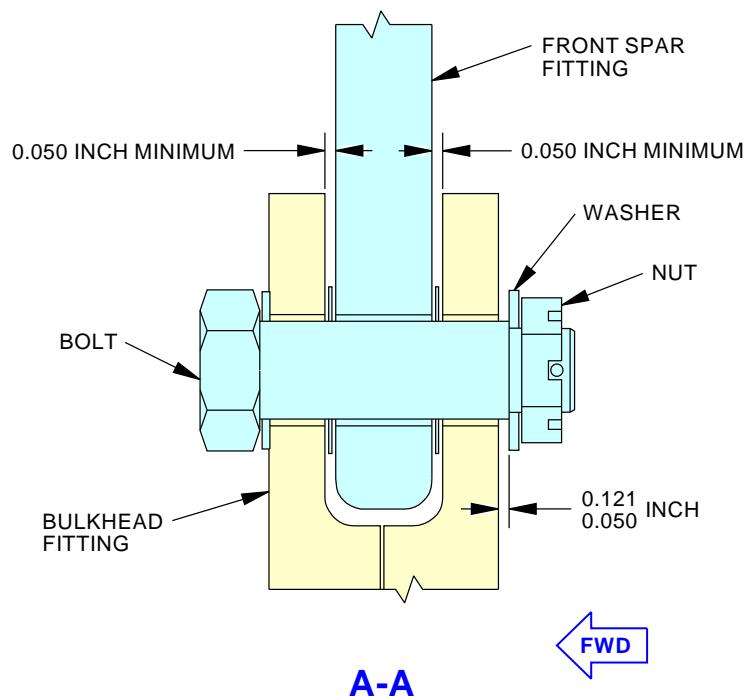
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INTERNAL-GENERAL VISUAL: VERTICAL FIN FRONT SPAR FITTING
Figure 1 (Sheet 1 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-460-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-460-00-01
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MPD ITEM
53-460-00

2087696 S0000439340_V2

INTERNAL-GENERAL VISUAL: VERTICAL FIN FRONT SPAR FITTING
Figure 1 (Sheet 2 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-460-00-01

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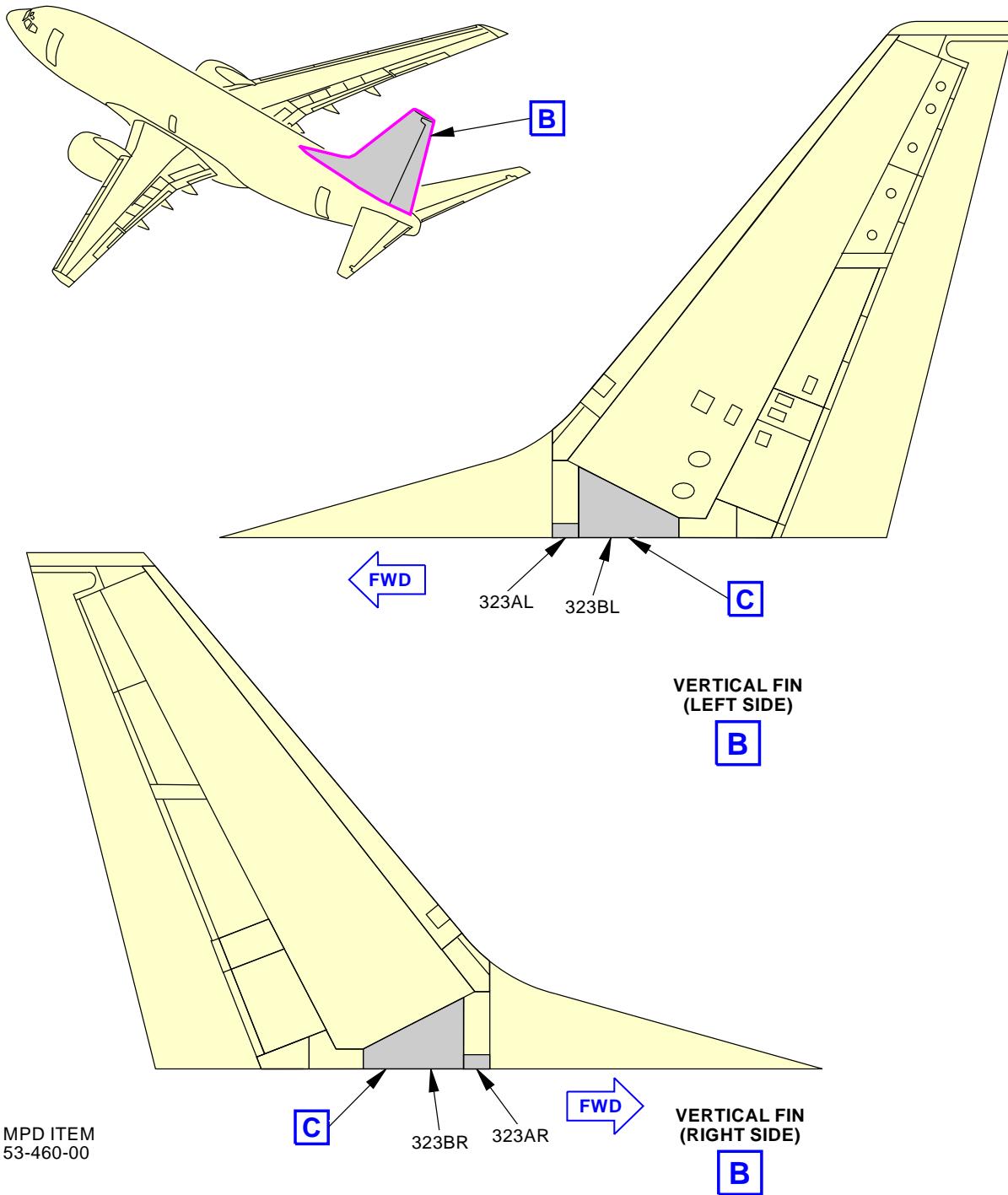
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-460-00-01

2087710 S0000439341_V2

INTERNAL-GENERAL VISUAL: VERTICAL FIN FRONT SPAR FITTING
Figure 1 (Sheet 3 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-460-00-01

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Oct 15/2015

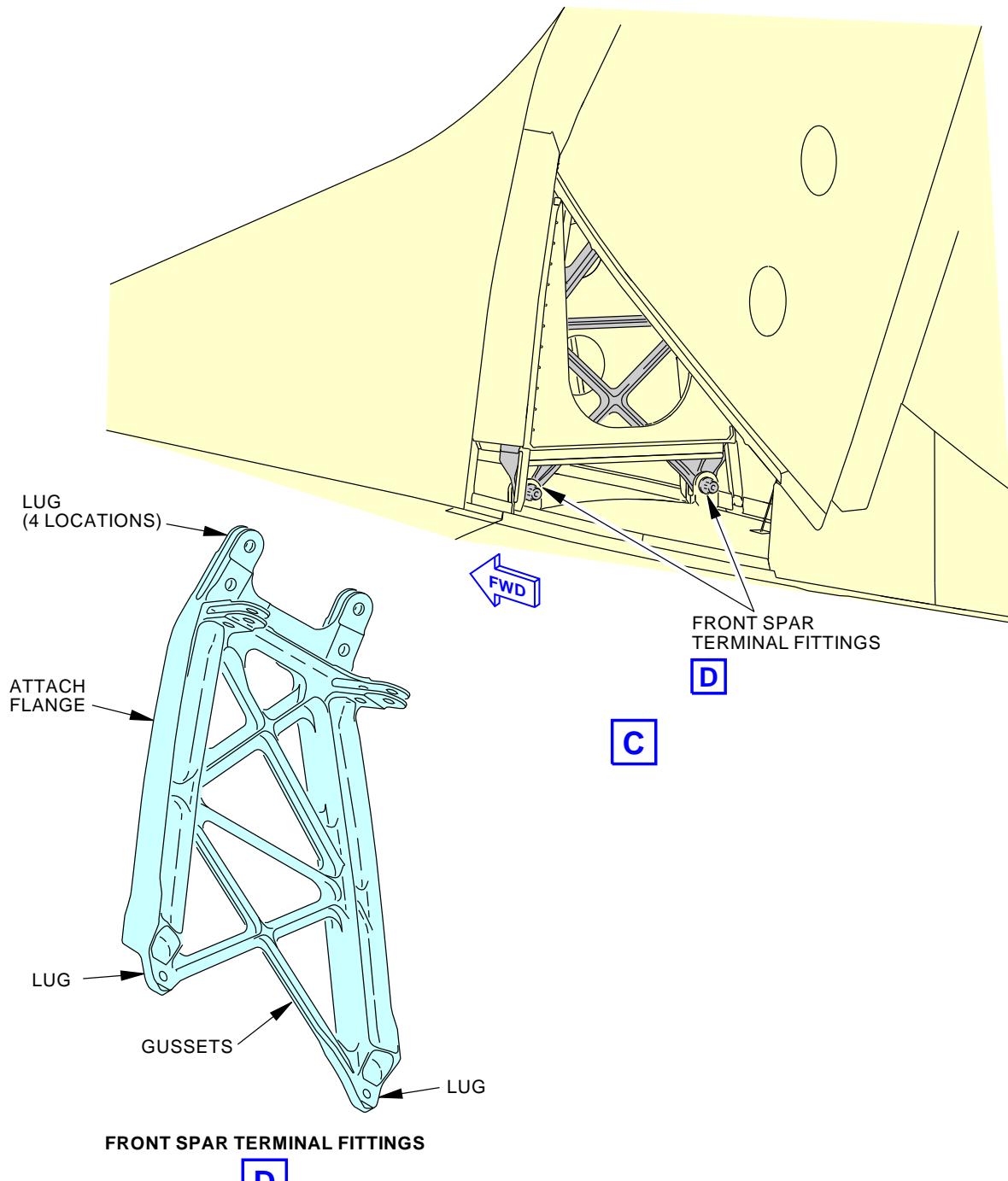
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-460-00-01MPD ITEM
53-460-00

2087737 S0000439342_V2

INTERNAL-GENERAL VISUAL: VERTICAL FIN FRONT SPAR FITTING
Figure 1 (Sheet 4 of 4)

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-460-00-01

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Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE VERTICAL FIN FRONT SPAR FITTING			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-470-00-01
TAIL NUMBER	WORK AREA VERT STABILIZER	VERSION 1.1	THRESHOLD 20 YR	REPEAT 8 YR	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS 323AL 323AR 323BL 323BR			ZONE 322
		NOTE			

Inspect vertical fin front spar fitting lugs and bolts (STA 1016).

ACCESS NOTE: Pin removal is required. Remove only one pin at a time.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING	
		D633A109-AKS 53-470-00-01	Page 1 of 6 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-470-00-01										
				MECH INSP										
TASK 53-05-03-210-843														
1. INTERNAL - GENERAL VISUAL: VERTICAL FIN FRONT SPAR FITTING														
(Figure 1)														
A. Inspection														
SUBTASK 53-05-03-010-040														
(1) Open these access panels:														
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>323AL</td><td>Vertical Fin, Front Spar Access Door</td></tr><tr><td>323AR</td><td>Vertical Fin, Front Spar Access Door</td></tr><tr><td>323BL</td><td>Vertical Fin, Forward Fin Access Door</td></tr><tr><td>323BR</td><td>Vertical Fin, Forward Fin Access Door</td></tr></tbody></table>					Number	Name/Location	323AL	Vertical Fin, Front Spar Access Door	323AR	Vertical Fin, Front Spar Access Door	323BL	Vertical Fin, Forward Fin Access Door	323BR	Vertical Fin, Forward Fin Access Door
Number	Name/Location													
323AL	Vertical Fin, Front Spar Access Door													
323AR	Vertical Fin, Front Spar Access Door													
323BL	Vertical Fin, Forward Fin Access Door													
323BR	Vertical Fin, Forward Fin Access Door													
NOTE: Pin removal is required. Remove only one pin at a time.														
SUBTASK 53-05-03-210-043														
(2) Do a General Visual inspection of the vertical fin front spar fitting lugs and bolts (STA 1016).														
SUBTASK 53-05-03-910-072														
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-804.														
SUBTASK 53-05-03-410-040														
(4) Close these access panels:														
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>323AL</td><td>Vertical Fin, Front Spar Access Door</td></tr><tr><td>323AR</td><td>Vertical Fin, Front Spar Access Door</td></tr><tr><td>323BL</td><td>Vertical Fin, Forward Fin Access Door</td></tr><tr><td>323BR</td><td>Vertical Fin, Forward Fin Access Door</td></tr></tbody></table>					Number	Name/Location	323AL	Vertical Fin, Front Spar Access Door	323AR	Vertical Fin, Front Spar Access Door	323BL	Vertical Fin, Forward Fin Access Door	323BR	Vertical Fin, Forward Fin Access Door
Number	Name/Location													
323AL	Vertical Fin, Front Spar Access Door													
323AR	Vertical Fin, Front Spar Access Door													
323BL	Vertical Fin, Forward Fin Access Door													
323BR	Vertical Fin, Forward Fin Access Door													
<hr style="width: 20%; margin-left: auto; margin-right: 0;"/> END OF TASK <hr style="width: 20%; margin-left: 0; margin-right: auto;"/>														

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-470-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-470-00-01
				MECH INSP
TASK 51-05-01-210-804				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-043				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-044				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-045				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-046				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-111				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-048				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING	
		D633A109-AKS 53-470-00-01	Page 3 of 6 Oct 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-470-00-01			
				<table border="1"><tr><td>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</td><td>MECH</td><td>INSP</td></tr></table>	1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP
1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP					

- 1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.
- 2) Not applicable
- 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

Water displacing / anti-corrosion compounds should not be applied in the following areas:

- Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
- Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
- Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
- Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
- Areas with electrical arc potential.
- Interior materials, including cargo liners (change of flammability properties).
- Fiber-glass ducts where temperature exceeds 220 degrees F.
- Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

- (7) CPCP Basic Task Item 7 is not applicable.

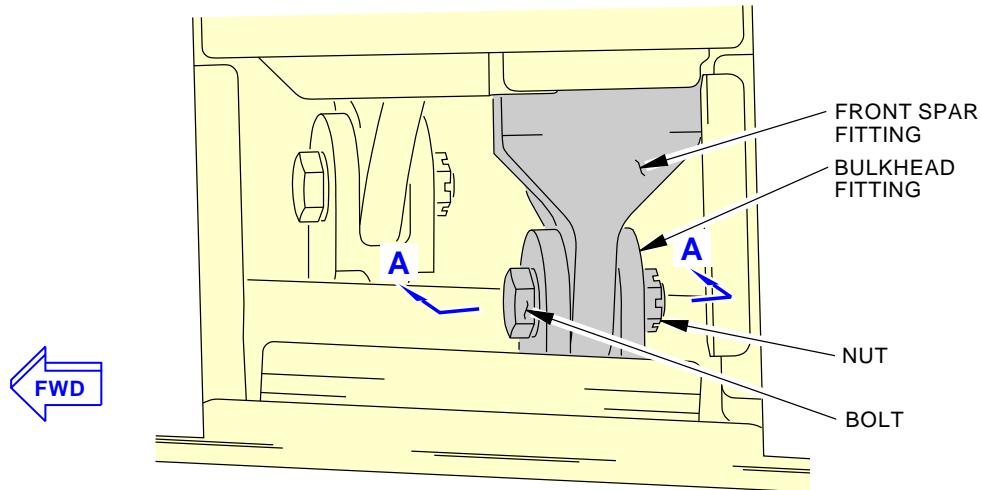
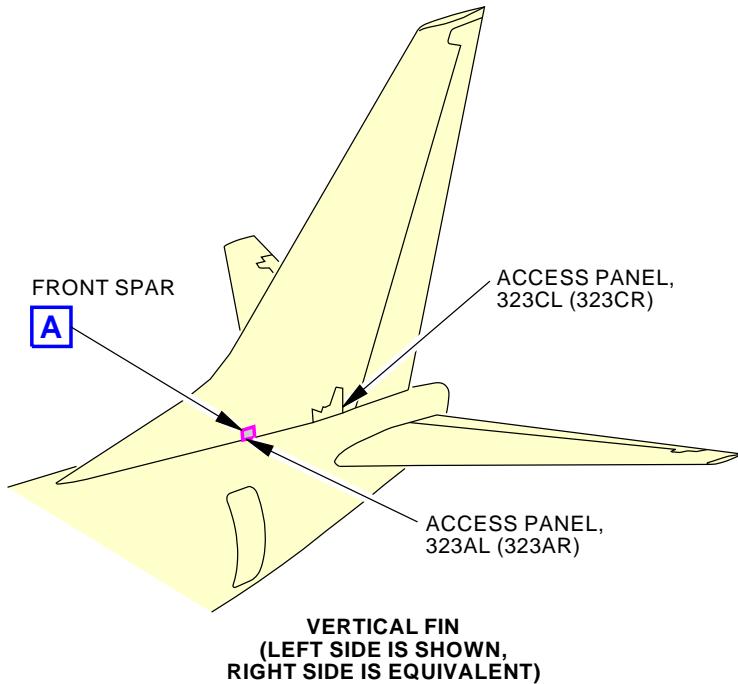
———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-470-00-01

Page 4 of 6
Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-470-00-01

MPD ITEM
53-470-00

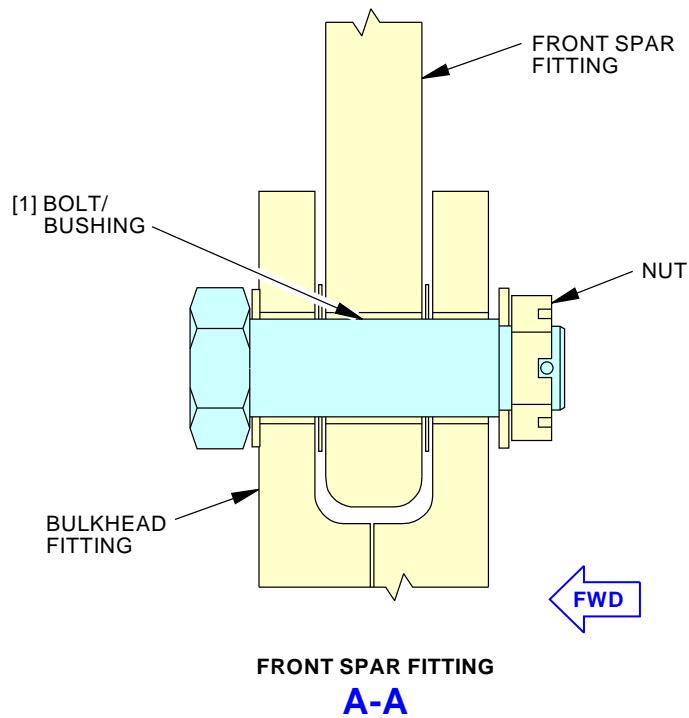
2072432 S0000432636_V2

**Vertical Fin Front Spar Fitting General Visual Inspection (Internal)
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-470-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-470-00-01

MPD ITEM
53-470-00

2072440 S0000432646_V2

**Vertical Fin Front Spar Fitting General Visual Inspection (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN FRONT SPAR FITTING
		D633A109-AKS 53-470-00-01

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE VERTICAL FIN REAR SPAR FITTING			BOEING CARD NO. 53-480-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA VERT STABILIZER	VERSION 1.1 1.2	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL	NOTE			ENGINE ALL
		ACCESS 323AL 323AR 323BL 323BR 323CL 323CR			ZONE 323

Inspect vertical fin rear spar fitting lugs and bolts at Sta 1088. Inspect fuselage skin under vertical fin from Sta 1016 to 1088.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN REAR SPAR FITTING
		D633A109-AKS 53-480-00-01

**Page 1 of 6
Jun 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-480-00-01														
				MECH INSP														
TASK 53-05-03-210-844																		
1. INTERNAL - GENERAL VISUAL: VERTICAL FIN REAR SPAR FITTING																		
(Figure 1)																		
A. Inspection																		
SUBTASK 53-05-03-010-041																		
(1) Open these access panels:																		
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>323AL</td><td>Vertical Fin, Front Spar Access Door</td></tr><tr><td>323AR</td><td>Vertical Fin, Front Spar Access Door</td></tr><tr><td>323BL</td><td>Vertical Fin, Forward Fin Access Door</td></tr><tr><td>323BR</td><td>Vertical Fin, Forward Fin Access Door</td></tr><tr><td>323CL</td><td>Vertical Fin, Rear Spar Access Door</td></tr><tr><td>323CR</td><td>Vertical Fin, Rear Spar Access Door</td></tr></tbody></table>					Number	Name/Location	323AL	Vertical Fin, Front Spar Access Door	323AR	Vertical Fin, Front Spar Access Door	323BL	Vertical Fin, Forward Fin Access Door	323BR	Vertical Fin, Forward Fin Access Door	323CL	Vertical Fin, Rear Spar Access Door	323CR	Vertical Fin, Rear Spar Access Door
Number	Name/Location																	
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323AR	Vertical Fin, Front Spar Access Door																	
323BL	Vertical Fin, Forward Fin Access Door																	
323BR	Vertical Fin, Forward Fin Access Door																	
323CL	Vertical Fin, Rear Spar Access Door																	
323CR	Vertical Fin, Rear Spar Access Door																	
SUBTASK 53-05-03-210-044																		
(2) Do a General Visual inspection of the vertical fin rear spar fitting lugs and bolts at Sta 1088. Inspect fuselage skin under vertical fin from Sta 1016 to 1088.																		
SUBTASK 53-05-03-910-073																		
(3) 737-6789 Basic Task Description, AMM Task 51-05-01-210-806.																		
SUBTASK 53-05-03-410-041																		
(4) Close these access panels:																		
<table><thead><tr><th>Number</th><th>Name/Location</th></tr></thead><tbody><tr><td>323AL</td><td>Vertical Fin, Front Spar Access Door</td></tr><tr><td>323AR</td><td>Vertical Fin, Front Spar Access Door</td></tr><tr><td>323BL</td><td>Vertical Fin, Forward Fin Access Door</td></tr><tr><td>323BR</td><td>Vertical Fin, Forward Fin Access Door</td></tr><tr><td>323CL</td><td>Vertical Fin, Rear Spar Access Door</td></tr><tr><td>323CR</td><td>Vertical Fin, Rear Spar Access Door</td></tr></tbody></table>					Number	Name/Location	323AL	Vertical Fin, Front Spar Access Door	323AR	Vertical Fin, Front Spar Access Door	323BL	Vertical Fin, Forward Fin Access Door	323BR	Vertical Fin, Forward Fin Access Door	323CL	Vertical Fin, Rear Spar Access Door	323CR	Vertical Fin, Rear Spar Access Door
Number	Name/Location																	
323AL	Vertical Fin, Front Spar Access Door																	
323AR	Vertical Fin, Front Spar Access Door																	
323BL	Vertical Fin, Forward Fin Access Door																	
323BR	Vertical Fin, Forward Fin Access Door																	
323CL	Vertical Fin, Rear Spar Access Door																	
323CR	Vertical Fin, Rear Spar Access Door																	
— END OF TASK —																		

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN REAR SPAR FITTING	
		D633A109-AKS 53-480-00-01	Page 2 of 6 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-480-00-01
				MECH INSP
TASK 51-05-01-210-806				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-057				
(1) CPCP Basic Task Item 1 is not applicable.				
SUBTASK 51-05-01-210-058				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-059				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-060				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-114				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-062				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN REAR SPAR FITTING
		D633A109-AKS 53-480-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-480-00-01			
				<table border="1"><tr><td>1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.</td><td>MECH</td><td>INSP</td></tr></table>	1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP
1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.	MECH	INSP					

SUBTASK 51-05-01-210-063

- (7) CPCP Basic Task Item 7 is not applicable.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN REAR SPAR FITTING
		D633A109-AKS 53-480-00-01

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Oct 15/2014

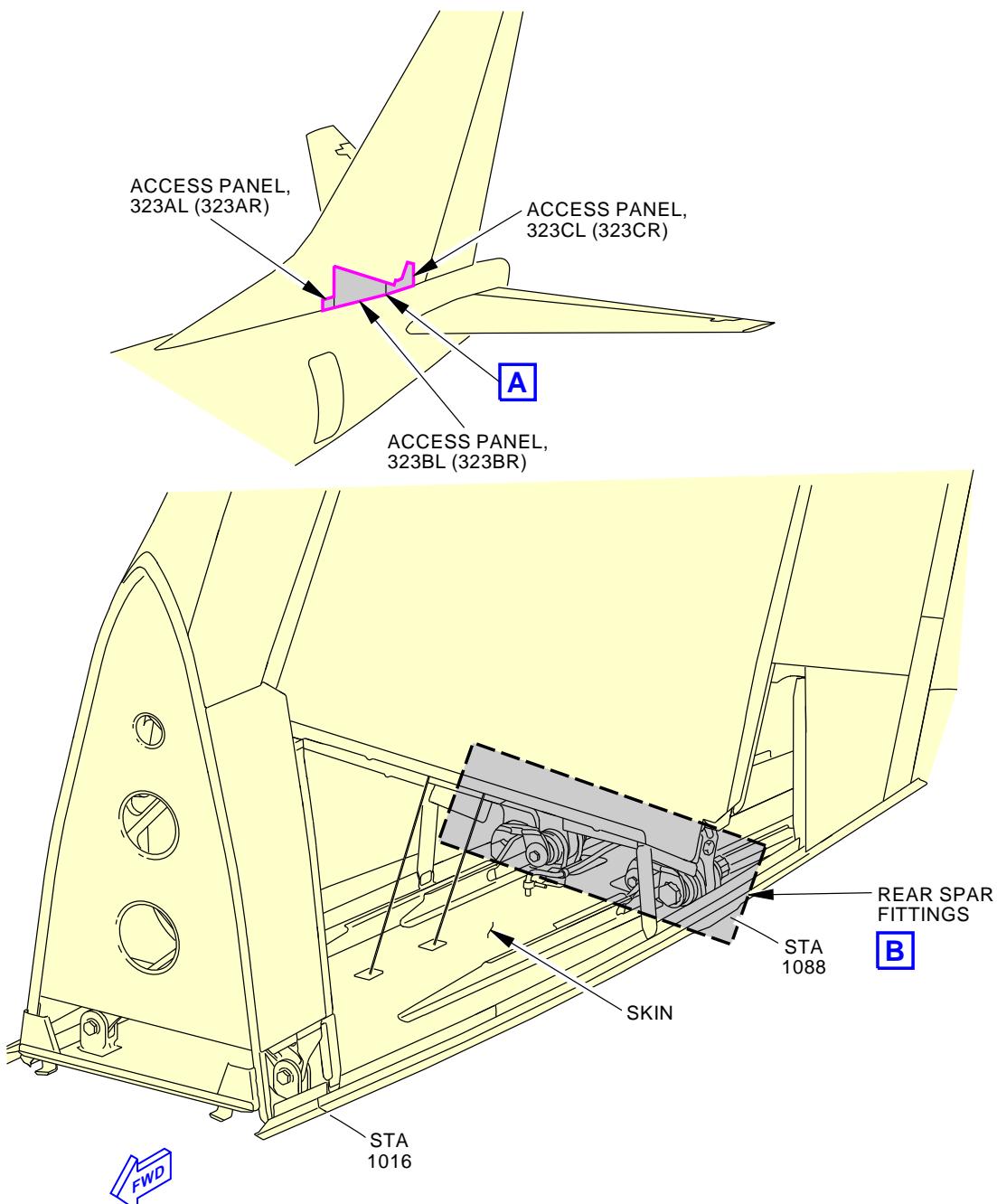
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-480-00-01MPD ITEM
53-480-00

2084332 S0000438778_V2

INTERNAL-GENERAL VISUAL: VERTICAL FIN REAR SPAR FITTING
Figure 1 (Sheet 1 of 2)**A**

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN REAR SPAR FITTING
		D633A109-AKS 53-480-00-01

Page 5 of 6
Oct 15/2015

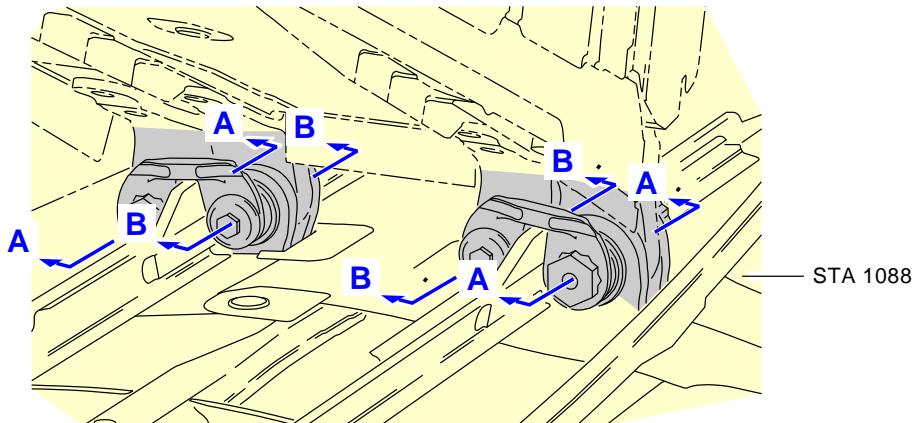
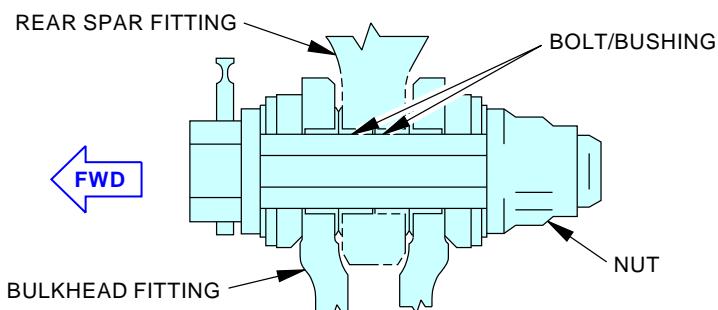
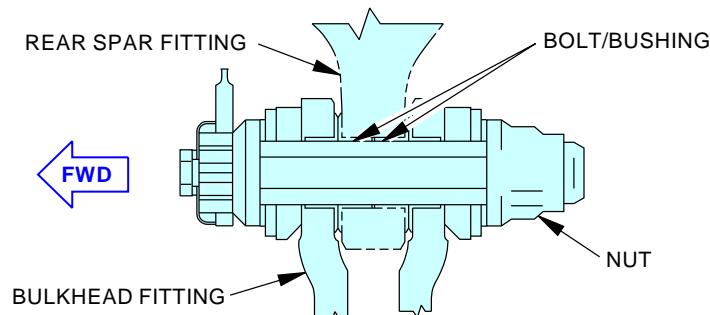
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-480-00-01**REAR SPAR FITTINGS****OUTBOARD SPAR FITTING****INBOARD SPAR FITTING**MPD ITEM
53-480-00

2084337 S0000438779_V2

**INTERNAL-GENERAL VISUAL: VERTICAL FIN REAR SPAR FITTING
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	VERTICAL FIN REAR SPAR FITTING
		D633A109-AKS 53-480-00-01

Page 6 of 6
Oct 15/2015

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE CROWN SKIN PANEL			BOEING CARD NO. 53-510-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 9 YR	REPEAT 9 YR	APPLICABILITY AIRPLANE 800 900ER ENGINE ALL NOTE
STATION	SKILL AIRPL	1.2 NOTE	24000 FC	24000 FC	
		ACCESS NOTE			ZONE 241 242

Inspect the skin and lugs under the antenna base plate Sta. 727D to 727H+5, S-4L to S-4R.

INTERVAL NOTE: Whichever comes first.

AIRPLANE NOTE: Applicable to airplanes with a KU antenna radome installed at Sta. 727D to 727H+5, S-4L to S-4R (737-800 and 737-900ER Only).

ACCESS NOTE: Antenna radome and adapter plate removal required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS
AMM 51-05-01-210-806	737-6789 Basic Task Description (P/B 201)
AMM 53-54-00-000-801	Broadband Radome Removal (P/B 401)
AMM 53-54-00-400-801	Broadband Radome Installation (P/B 401)

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

EFFECTIVITY AKS 012-999	SOURCE MRB	CROWN SKIN PANEL D633A109-AKS 53-510-00-01	Page 1 of 5 Feb 15/2016
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AKS



737-600/700/800/900 TASK CARDS

EFFECTIVITY AKS 012-999	SOURCE MRB	CROWN SKIN PANEL
		D633A109-AKS 53-510-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-510-00-01
				MECH INSP
TASK 51-05-01-210-808				
2. 737-6789 Basic Task Description				
A. CPCP Basic Task				
SUBTASK 51-05-01-210-071				
(1) Do the CPCP Basic Task Item 1 as follows:				
(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.				
SUBTASK 51-05-01-210-072				
(2) Do the CPCP Basic Task Item 2 as follows:				
(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.				
SUBTASK 51-05-01-210-073				
(3) Do the CPCP Basic Task Item 3 as follows:				
(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.				
SUBTASK 51-05-01-210-074				
(4) Do the CPCP Basic Task item 4 as follows:				
(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.				
SUBTASK 51-05-01-210-117				
(5) Do the CPCP Basic Task Item 5 as follows:				
(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.				
SUBTASK 51-05-01-210-076				
(6) Do the CPCP Basic Task item 6 as follows:				
(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.				

EFFECTIVITY
AKS 012-999SOURCE
MRB**CROWN SKIN PANEL****D633A109-AKS
53-510-00-01****Page 3 of 5
Feb 15/2016**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-510-00-01		
					MECH	INSP
				1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed. 2) Not applicable 3) List of areas / items where water displacing/anti-corrosion compounds should not be applied: Water displacing / anti-corrosion compounds should not be applied in the following areas: <ul style="list-style-type: none">• Cables, pulleys, wiring, plastics, elastomers, oxygen systems.• Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).• Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).• Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).• Areas with electrical arc potential.• Interior materials, including cargo liners (change of flammability properties).• Fiber-glass ducts where temperature exceeds 220 degrees F.• Selected areas noted in baseline program.		

SUBTASK 51-05-01-210-077

- (7) Do the CPCP Basic Task Item 7 as follows:
(a) Dry wet insulation blankets prior to re-installing, or replace with new, as applicable.

— END OF TASK —

EFFECTIVITY AKS 012-999	SOURCE MRB	CROWN SKIN PANEL
		D633A109-AKS 53-510-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-510-00-01

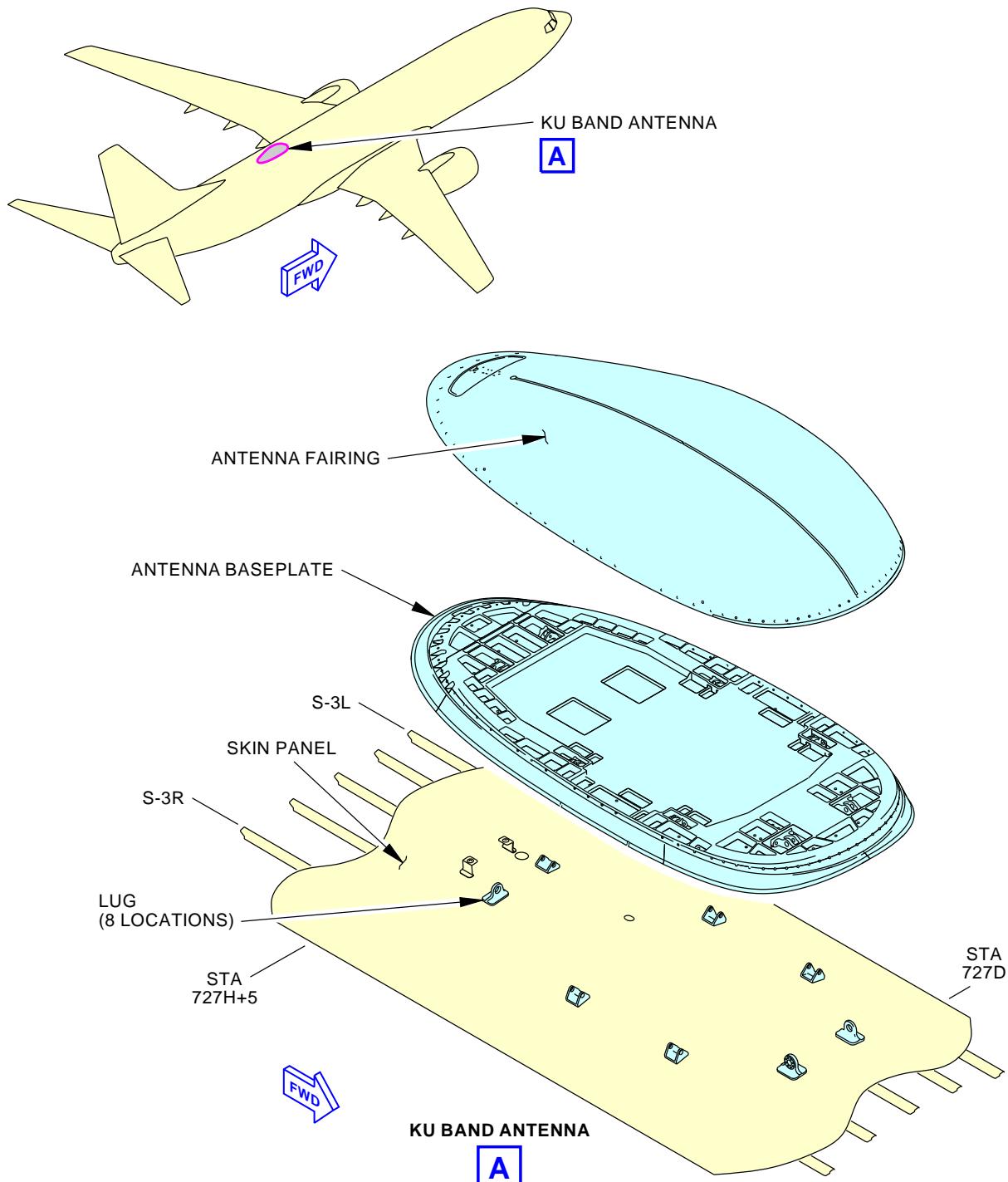


Figure 1

EFFECTIVITY AKS 012-999	SOURCE MRB	CROWN SKIN PANEL
		D633A109-AKS 53-510-00-01

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 221

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 259.5 to STA 360.

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-600-00-01	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-600-00-01	
					MECH INSP
TASK 53-05-02-250-801					
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE					
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-001</p> <p>(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 259.5 to STA 360.</p> <p>See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.</p> <p style="text-align: center;">———— END OF TASK ————</p>					
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT LONGITUDINAL LAP SPLICE</p> <p>D633A109-AKS 53-600-00-01</p>					

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL
STATION	SKILL AIRPL				800 900 900ER
		ACCESS			ZONE 222

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 259.5 to STA 360.

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-600-00-02	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-600-00-02

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-20-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 221

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 259.5 to STA 360.

See Doc. D626A001-DTR, DTR check form 53-10-03-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-600-20-01

AKS



737-600/700/800/900 TASK CARDS

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-600-20-01

AKS



737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-20-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 222

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 259.5 to STA 360.

See Doc. D626A001-DTR, DTR check form 53-10-03-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-600-20-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-600-20-02
TASK 53-05-02-250-803				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-003				
(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 259.5 to STA 360. See Doc. D626A001-DTR, DTR check form 53-10-03-2, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-600-20-02	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-30-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 221

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringer S-14L (from STA 259.5 to STA 294.5, and from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277, and from STA 344 to STA 360).

See Doc D626A001-DTR, DTR check form 53-10-03-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-600-30-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-600-30-01
TASK 53-05-02-250-805				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-005				
(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringer S-14L (from STA 259.5 to STA 294.5, and from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277, and from STA 344 to STA 360). See Doc D626A001-DTR, DTR check form 53-10-03-3 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-600-30-01		
		Page 2 of 2 Feb 15/2015		

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-30-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 222

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringer S-14L (from STA 259.5 to STA 294.5, and from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277, and from STA 344 to STA 360).

See Doc D626A001-DTR, DTR check form 53-10-03-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-600-30-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-600-30-02
TASK 53-05-02-250-805				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-005				
(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringer S-14L (from STA 259.5 to STA 294.5, and from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277, and from STA 344 to STA 360). See Doc D626A001-DTR, DTR check form 53-10-03-3 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-600-30-02	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-40-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 221

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringer S-14L (from STA 259.5 to STA 294.5 and from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277 and from STA 344 to STA 360).

See Doc D626A001-DTR, DTR check form 53-10-03-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-600-40-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-600-40-01
TASK 53-05-02-250-806				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-006				
(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringer S-14L (from STA 259.5 to STA 294.5 and from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277 and from STA 344 to STA 360). See Doc D626A001-DTR, DTR check form 53-10-03-4 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-600-40-01	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-40-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 222

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringer S-14L (from STA 259.5 to STA 294.5 and from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277 and from STA 344 to STA 360).

See Doc D626A001-DTR, DTR check form 53-10-03-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-600-40-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-600-40-02
TASK 53-05-02-250-806				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-006				
(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringer S-14L (from STA 259.5 to STA 294.5 and from STA 350 to STA 360) and at stringer S-14R (from STA 259.5 to STA 277 and from STA 344 to STA 360). See Doc D626A001-DTR, DTR check form 53-10-03-4 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-600-40-02	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-50-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 113 117

Inspect (Low Frequency Eddy Current) the upper skin along the upper fastener row at stringer S-24L (from STA 259.5 to STA 334) and at stringer S-24R (from STA 259.5 to STA 360).

See Doc D626A001-DTR, DTR check form 53-10-03-5 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-600-50-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-600-50-01
TASK 53-05-02-250-807				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-007				
(1) Do a Low Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringer S-24L (from STA 259.5 to STA 334) and at stringer S-24R (from STA 259.5 to STA 360). See Doc D626A001-DTR, DTR check form 53-10-03-5 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-600-50-01		
		Page 2 of 2 Jun 15/2015		

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-600-50-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 114 118

Inspect (Low Frequency Eddy Current) the upper skin along the upper fastener row at stringer S-24L (from STA 259.5 to STA 334) and at stringer S-24R (from STA 259.5 to STA 360).

See Doc D626A001-DTR, DTR check form 53-10-03-5 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-600-50-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-600-50-02
TASK 53-05-02-250-807				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-007				
(1) Do a Low Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringer S-24L (from STA 259.5 to STA 334) and at stringer S-24R (from STA 259.5 to STA 360). See Doc D626A001-DTR, DTR check form 53-10-03-5 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-600-50-02		
		Page 2 of 2 Jun 15/2015		

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE CUTOUT - CREW CAB WINDOWS			BOEING CARD NO. 53-601-01-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 12000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS			ZONE 211 212

Inspect (High Frequency Eddy Current) the AB post structure, from inside the aircraft, between the upper to lower sills.

See Doc D626A001-DTR, DTR check form 53-10-04-1a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-75.

AIRPLANE NOTE: Applicable to airplanes L/N 1389 and on.

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT - CREW CAB WINDOWS
		D633A109-AKS 53-601-01-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-01-01
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TASK 53-05-02-250-809

1. INTERNAL - SPECIAL DETAILED: CUTOUT - CREW CAB WINDOWS (LN 1389 & ON)

A. Inspection

SUBTASK 53-05-02-250-009

- (1) Do a High Frequency Eddy Current inspection of the AB post structure, from inside the aircraft, from the upper to lower sills and the post flanges aft of fastener locations.

See Doc D626A001-DTR, DTR check form 53-10-04-1a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-75.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT - CREW CAB WINDOWS
		D633A109-AKS 53-601-01-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT CUTOUT - CREW CAB WINDOWS			BOEING CARD NO.
DATE	TASK DETAILED				53-601-21-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS			ZONE 211

Inspect (Detailed) the CD post, from outside the aircraft, along the entire post length on both the left and right sides.

See Doc D626A001-DTR, DTR check form 53-10-04-2a for alternative inspections.

AIRPLANE NOTE: Applicable to airplanes L/N 1389 and on.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CUTOUT - CREW CAB WINDOWS
		D633A109-AKS 53-601-21-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-21-01
TASK 53-05-02-211-802				MECH INSP
1. EXTERNAL - DETAILED: CUTOUT - CREW CAB WINDOWS (LN 1389 & ON)				
A. Inspection				
SUBTASK 53-05-02-211-002				
(1) Do a Detailed inspection of the CD post, from outside the aircraft, along the entire post length on both the left and right sides.				
See Doc D626A001-DTR, DTR check form 53-10-04-2a for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CUTOUT - CREW CAB WINDOWS		
		D633A109-AKS 53-601-21-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT CUTOUT - CREW CAB WINDOWS			BOEING CARD NO.
DATE	TASK DETAILED				53-601-21-02
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS			ZONE 212

Inspect (Detailed) the CD post, from outside the aircraft, along the entire post length on both the left and right sides.

See Doc D626A001-DTR, DTR check form 53-10-04-2a for alternative inspections.

AIRPLANE NOTE: Applicable to airplanes L/N 1389 and on.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CUTOUT - CREW CAB WINDOWS
		D633A109-AKS 53-601-21-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-21-02
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TASK 53-05-02-211-802

MECH

INSP

1. EXTERNAL - DETAILED: CUTOUT - CREW CAB WINDOWS (LN 1389 & ON)**A. Inspection**

SUBTASK 53-05-02-211-002

- (1) Do a Detailed inspection of the CD post, from outside the aircraft, along the entire post length on both the left and right sides.

See Doc D626A001-DTR, DTR check form 53-10-04-2a for alternative inspections.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CUTOUT - CREW CAB WINDOWS
		D633A109-AKS 53-601-21-02

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT CUTOUT - CREW CAB WINDOWS			BOEING CARD NO.
DATE	TASK DETAILED				53-601-30-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 211

Inspect (Detailed) the EF post, from outside the aircraft, along the entire post length on both the left and right sides.

See Doc D626A001-DTR, DTR check form 53-10-04-3 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CUTOUT - CREW CAB WINDOWS
		D633A109-AKS 53-601-30-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-30-01
TASK 53-05-02-211-803				MECH INSP
1. EXTERNAL - DETAILED: CUTOUT - CREW CAB WINDOWS				
A. Inspection				
SUBTASK 53-05-02-211-003				
(1) Do a Detailed inspection of the EF post, from outside the aircraft, along the entire post length on both the left and right sides.				
See Doc D626A001-DTR, DTR check form 53-10-04-3 for alternative inspections.				
———— END OF TASK ——				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CUTOUT - CREW CAB WINDOWS		
		D633A109-AKS 53-601-30-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT CUTOUT - CREW CAB WINDOWS			BOEING CARD NO.
DATE	TASK DETAILED				53-601-30-02
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY
		ACCESS			AIRPLANE ALL ENGINE ALL
					ZONE 212

Inspect (Detailed) the EF post, from outside the aircraft, along the entire post length on both the left and right sides.

See Doc D626A001-DTR, DTR check form 53-10-04-3 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CUTOUT - CREW CAB WINDOWS
		D633A109-AKS 53-601-30-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-30-02
TASK 53-05-02-211-803				MECH INSP
1. EXTERNAL - DETAILED: CUTOUT - CREW CAB WINDOWS				
A. Inspection				
SUBTASK 53-05-02-211-003				
(1) Do a Detailed inspection of the EF post, from outside the aircraft, along the entire post length on both the left and right sides.				
See Doc D626A001-DTR, DTR check form 53-10-04-3 for alternative inspections.				
———— END OF TASK ——				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT CUTOUT - CREW CAB WINDOWS				
D633A109-AKS 53-601-30-02				
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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT CUTOUT - CREW CABIN WINDOWS			BOEING CARD NO. 53-601-41-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS			ZONE 211

Inspect (Low Frequency Eddy Current) both rows of fasteners attaching the skin to the BD Sill, from outside the aircraft, between LBL 8 and LBL 26.5. Repeat the process between RBL 8 and RBL 26.5.

See Doc D626A001-DTR, DTR check form 53-10-04-4a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-76.

AIRPLANE NOTE: Applicable to airplanes L/N 1389 and on.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CUTOUT - CREW CABIN WINDOWS
		D633A109-AKS 53-601-41-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-41-01
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TASK 53-05-02-250-811

1. EXTERNAL - SPECIAL DETAILED: CUTOUT - CREW CAB WINDOWS (LN 1389 & ON)

A. Inspection

SUBTASK 53-05-02-250-011

- (1) Do a Low Frequency Eddy Current inspection on both rows of fasteners attaching the skin to the BD Sill, from outside the aircraft, between LBL 8 and LBL 26.5. Repeat the process between RBL 8 and RBL 26.5.

See Doc D626A001-DTR, DTR check form 53-10-04-4a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-76.

— END OF TASK —

EFFECTIVITY
AKS ALL

SOURCE
AWL

LEFT CUTOUT - CREW CABIN WINDOWS

D633A109-AKS
53-601-41-01

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AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT CUTOUT - CREW CABIN WINDOWS			BOEING CARD NO. 53-601-41-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL				
		ACCESS			ZONE 212

Inspect (Low Frequency Eddy Current) both rows of fasteners attaching the skin to the BD Sill, from outside the aircraft, between LBL 8 and LBL 26.5. Repeat the process between RBL 8 and RBL 26.5.

See Doc D626A001-DTR, DTR check form 53-10-04-4a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-76.

AIRPLANE NOTE: Applicable to airplanes L/N 1389 and on.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CUTOUT - CREW CABIN WINDOWS
		D633A109-AKS 53-601-41-02

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-41-02
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TASK 53-05-02-250-811

1. EXTERNAL - SPECIAL DETAILED: CUTOUT - CREW CAB WINDOWS (LN 1389 & ON)

A. Inspection

SUBTASK 53-05-02-250-011

- (1) Do a Low Frequency Eddy Current inspection on both rows of fasteners attaching the skin to the BD Sill, from outside the aircraft, between LBL 8 and LBL 26.5. Repeat the process between RBL 8 and RBL 26.5.

See Doc D626A001-DTR, DTR check form 53-10-04-4a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-76.

END OF TASK

EFFECTIVITY
AKS ALL

SOURCE
AWL

BIG CUTOUT - CREW CABIN WINDOWS

D633A109-AKS
53-601-41-02

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT CUTOUT - CREW CABIN WINDOWS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-601-50-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 211

Inspect (High Frequency Eddy Current) the flanges of the Point "D" Fitting, from the CD post inboard to the second fastener common to the BD sill, on both the left and right sides.

See Doc D626A001-DTR, DTR check form 53-10-04-5 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-72.

ACCESS NOTE: Remove glareshield as required to perform inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CUTOUT - CREW CABIN WINDOWS D633A109-AKS 53-601-50-01	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-50-01	MECH	INSP
TASK 53-05-02-250-812						
1. INTERNAL - SPECIAL DETAILED: CUTOUT - CREW CABIN WINDOWS						
A. Inspection						
NOTE: Remove glareshield as required to perform the inspection.						
SUBTASK 53-05-02-250-012						
(1) Do a High Frequency Eddy Current inspection on the flanges of the Point D Fitting, from the CD post inboard to the second fastener common to the BD sill, on both the left and right sides.						
See Doc D626A001-DTR, DTR check form 53-10-04-5 for alternative inspections.						
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-72.						
<hr/> — END OF TASK —						

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CUTOUT - CREW CABIN WINDOWS
		D633A109-AKS 53-601-50-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT CUTOUT - CREW CABIN WINDOWS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-601-50-02 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 212

Inspect (High Frequency Eddy Current) the flanges of the Point "D" Fitting, from the CD post inboard to the second fastener common to the BD sill, on both the left and right sides.

See Doc D626A001-DTR, DTR check form 53-10-04-5 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-72.

ACCESS NOTE: Remove glareshield as required to perform inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CUTOUT - CREW CABIN WINDOWS D633A109-AKS 53-601-50-02	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-50-02		
					MECH	INSP
TASK 53-05-02-250-812						
1. INTERNAL - SPECIAL DETAILED: CUTOUT - CREW CABIN WINDOWS						
<p>A. Inspection</p> <p><u>NOTE:</u> Remove glareshield as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-012</p> <p>(1) Do a High Frequency Eddy Current inspection on the flanges of the Point D Fitting, from the CD post inboard to the second fastener common to the BD sill, on both the left and right sides.</p> <p>See Doc D626A001-DTR, DTR check form 53-10-04-5 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-72.</p>						
<hr style="width: 20%; margin: auto;"/> END OF TASK <hr style="width: 20%; margin: auto;"/>						

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CUTOUT - CREW CABIN WINDOWS
		D633A109-AKS 53-601-50-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT CUTOUT - CREW CABIN WINDOWS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-601-60-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 211

Inspect (High Frequency Eddy Current) the Inconel angle around the seven fasteners that join the angle to the B-D Sill Web and Point "D" Fitting.

See Doc D626A001-DTR, DTR check form 53-10-04-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-71.

ACCESS NOTE: Remove glareshield if/as required to perform inspection. There are three (3) fasteners that join the angle, web and fitting. There are four (4) fasteners that join the angle and the fitting.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CUTOUT - CREW CABIN WINDOWS D633A109-AKS 53-601-60-01	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-60-01
				MECH INSP
TASK 53-05-02-250-813				
1. INTERNAL - SPECIAL DETAILED: CUTOUT - CREW CABIN WINDOWS				
<p>A. Inspection</p> <p><u>NOTE:</u> Remove glareshield if/as required to perform inspection.</p> <p>There are three (3) fasteners that join the angle, web and fitting.</p> <p>There are four (4) fasteners that join the angle and the fitting.</p>				
SUBTASK 53-05-02-250-013				
<p>(1) Do a High Frequency Eddy Current inspection of the Inconel angle around the (7) fasteners that join the angle to the B-D Sill Web and Point "D" Fitting.</p> <p>See Doc D626A001-DTR, DTR check form 53-10-04-6 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-71.</p>				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CUTOUT - CREW CABIN WINDOWS
		D633A109-AKS 53-601-60-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT CUTOUT - CREW CABIN WINDOWS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-601-60-02 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 212

Inspect (High Frequency Eddy Current) the Inconel angle around the seven fasteners that join the angle to the B-D Sill Web and Point "D" Fitting.

See Doc D626A001-DTR, DTR check form 53-10-04-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-71.

ACCESS NOTE: Remove glareshield if/as required to perform inspection. There are three (3) fasteners that join the angle, web and fitting. There are four (4) fasteners that join the angle and the fitting.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CUTOUT - CREW CABIN WINDOWS
		D633A109-AKS 53-601-60-02

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-601-60-02
				MECH INSP
TASK 53-05-02-250-813				
1. INTERNAL - SPECIAL DETAILED: CUTOUT - CREW CABIN WINDOWS				
<p>A. Inspection</p> <p><u>NOTE:</u> Remove glareshield if/as required to perform inspection.</p> <p>There are three (3) fasteners that join the angle, web and fitting.</p> <p>There are four (4) fasteners that join the angle and the fitting.</p>				
SUBTASK 53-05-02-250-013				
<p>(1) Do a High Frequency Eddy Current inspection of the Inconel angle around the (7) fasteners that join the angle to the B-D Sill Web and Point "D" Fitting.</p> <p>See Doc D626A001-DTR, DTR check form 53-10-04-6 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-71.</p>				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CUTOUT - CREW CABIN WINDOWS
		D633A109-AKS 53-601-60-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT CROWN SKIN PANEL - FWD OF STA 259.5			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-602-10-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 211

Inspect (Low Frequency Eddy Current) the subsurface of the first row of fasteners, on the left and right side, of BL 0.0 between the cab window cutout and STA 259.5 panel splice.

See Doc D626A001-DTR, DTR check form 53-10-05-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-57.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT CROWN SKIN PANEL - FWD OF STA 259.5
		D633A109-AKS 53-602-10-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-602-10-01
				MECH INSP
TASK 53-05-02-250-814				
1. EXTERNAL - SPECIAL DETAILED: CROWN SKIN PANEL - FWD OF STA 259.5				
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-014</p> <p>(1) Do a Low Frequency Eddy Current inspection of the subsurface of the first row of fasteners, on the left and right side, of BL 0.0 between the cab window cutout and STA 259.5 panel splice.</p> <p>See Doc D626A001-DTR, DTR check form 53-10-05-1 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT CROWN SKIN PANEL - FWD OF STA 259.5</p> <p>D633A109-AKS 53-602-10-01</p>				

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT CROWN SKIN PANEL - FWD OF STA 259.5			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-602-10-02 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 212

Inspect (Low Frequency Eddy Current) the subsurface of the first row of fasteners, on the left and right side, of BL 0.0 between the cab window cutout and STA 259.5 panel splice.

See Doc D626A001-DTR, DTR check form 53-10-05-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-57.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CROWN SKIN PANEL - FWD OF STA 259.5
		D633A109-AKS 53-602-10-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-602-10-02
TASK 53-05-02-250-814				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: CROWN SKIN PANEL - FWD OF STA 259.5				
A. Inspection				
SUBTASK 53-05-02-250-014				
(1) Do a Low Frequency Eddy Current inspection of the subsurface of the first row of fasteners, on the left and right side, of BL 0.0 between the cab window cutout and STA 259.5 panel splice.				
See Doc D626A001-DTR, DTR check form 53-10-05-1 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT CROWN SKIN PANEL - FWD OF STA 259.5		
		D633A109-AKS 53-602-10-02	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CROWN SKIN PANEL - STA 259.5 TO 360			BOEING CARD NO. 53-602-20-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS		ZONE 221 222		

Inspect (Detailed) the skin around all of the fastener locations from stringer S-10L to S-10R, from STA 259.5 to STA 360, except at the lap splices and antennas. (53-10-08-1).

See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL - STA 259.5 TO 360
		D633A109-AKS 53-602-20-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-602-20-01
TASK 53-05-02-211-862				MECH INSP
1. EXTERNAL - DETAILED: CROWN SKIN PANEL STA 259.5 TO STA 360				
A. Inspection				
SUBTASK 53-05-02-211-062				
(1) Do a detailed inspection of the skin around all of the fastener locations from stringer S-10L to S-10R, from Sta 259.5 to Sta 360, except at the lap splices and antennas. (53-10-08-1). See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL				
SOURCE AWL				
CROWN SKIN PANEL - STA 259.5 TO 360				
D633A109-AKS 53-602-20-01				
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737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CUTOUT - ELECTRONIC EQUIPMENT DOOR			BOEING CARD NO. 53-605-10-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 117 118	

Inspect (Detailed) the skin at the edge of the E/E door cutout and around the two rows of fasteners adjacent to the edge. Inspection is performed along the entire perimeter of the E/E door cutout and bounded by STA 323.7 and STA 351.2, and RBL 15.47 and LBL 6.74.

See Doc D626A001-DTR, DTR check form 53-10-13-1 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT - ELECTRONIC EQUIPMENT DOOR
		D633A109-AKS 53-605-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-605-10-01
TASK 53-05-02-211-804				MECH INSP
1. EXTERNAL - DETAILED: CUTOUT - ELECTRONIC EQUIPMENT DOOR				
A. Inspection				
SUBTASK 53-05-02-211-004				
(1) Do a Detailed inspection of the skin at the edge of the E/E door cutout and around the two rows of fasteners adjacent to the edge. Inspection is performed along the entire perimeter of the E/E door cutout and bounded by Sta 323.7 and Sta 351.2, and RBL 15.47 and LBL 6.74.				
See Doc D626A001-DTR, DTR check form 53-10-13-1 for alternative inspections.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL				
SOURCE AWL				
CUTOUT - ELECTRONIC EQUIPMENT DOOR				
D633A109-AKS 53-605-10-01				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CUTOUT - FORWARD ENTRY DOOR FRAME			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-606-40-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 221

Inspect (High Frequency Eddy Current) the two rows of fasteners common to the forward edge frame and skin at STA 303.9 from stringers S-11L and S-12L.

See Doc. D626A001-DTR, DTR check form 53-10-14-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-89.

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT - FORWARD ENTRY DOOR FRAME
		D633A109-AKS 53-606-40-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-606-40-01
				MECH INSP
TASK 53-05-02-250-816				
1. EXTERNAL - SPECIAL DETAILED: CUTOUT - FORWARD ENTRY DOOR FRAME				
A. Inspection				
SUBTASK 53-05-02-250-016				
(1) Do a High Frequency Eddy Current inspection on the two rows of fasteners common to the forward edge frame and skin at STA 303.9 from stringers S-11L and S-12L. See Doc. D626A001-DTR, DTR check form 53-10-14-4 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-89.				
— END OF TASK —				
EFFECTIVITY AKS ALL				
SOURCE AWL				
CUTOUT - FORWARD ENTRY DOOR FRAME				
D633A109-AKS 53-606-40-01				
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737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CUTOUT - FORWARD ENTRY DOOR FRAME			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-606-50-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 221

Inspect (High Frequency Eddy Current) around the two rows of fasteners common to the forward edge frame and skin at STA 303.9 from stringers S-7L to S-11L and stringers S-12L to S-13L.

See Doc. D626A001-DTR, DTR check form 53-10-14-5 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-89.

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT - FORWARD ENTRY DOOR FRAME
		D633A109-AKS 53-606-50-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-606-50-01
				MECH INSP
TASK 53-05-02-250-817				
1. EXTERNAL - SPECIAL DETAILED: CUTOUT - FORWARD ENTRY DOOR FRAME				
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-017</p> <p>(1) Do a High Frequency Eddy Current inspection around the two rows of fasteners common to the forward edge frame and skin at STA 303.9 from stringers S-7L to S-11L and stringers S-12L to S-13L.</p> <p>See Doc. D626A001-DTR, DTR check form 53-10-14-5 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-89.</p>				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT - FORWARD ENTRY DOOR FRAME
		D633A109-AKS 53-606-50-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FWD ENTRY DOOR, FWD EDGE FRAME STOPS			BOEING CARD NO. 53-606-70-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 221

Inspect (Detailed) around the fasteners in the inboard flange of the stop fittings from stringers S-7 through S-16.

See Doc. D626A001-DTR, DTR check form 53-10-14-7 for alternative inspections.

ACCESS NOTE: Remove interior panels as required to perform inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR, FWD EDGE FRAME STOPS
		D633A109-AKS 53-606-70-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-606-70-01
				MECH INSP
TASK 53-05-02-211-805				
1. INTERNAL - DETAILED: FWD ENTRY DOOR, FWD EDGE FRAME STOPS				
A. Inspection				
NOTE: Remove interior panels as required to perform inspection.				
SUBTASK 53-05-02-211-005				
(1) Do a Detailed inspection around the fasteners in the inboard flange of the stop fittings from stringer S-7 thru S-16.				
See Doc. D626A001-DTR, DTR check form 53-10-14-7 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR, FWD EDGE FRAME STOPS		
		D633A109-AKS 53-606-70-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FWD ENTRY DOOR FWD EDGE FRAME STOPS			BOEING CARD NO. 53-618-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 30000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 221

Inspect (High Frequency Eddy Current) around the four fasteners on each stop strap at stringer S-7 thru S-14.

See Doc. D626A001-DTR, DTR check form 53-10-14-8 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-01.

ACCESS NOTE: Remove interior panels as required to perform inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR FWD EDGE FRAME STOPS
		D633A109-AKS 53-618-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-618-00-01
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TASK 53-05-02-250-818

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: FWD ENTRY DOOR FWD EDGE FRAME STOPS**A. Inspection**

NOTE: Remove interior panels as required to perform inspection.

SUBTASK 53-05-02-250-018

- (1) Do a High Frequency Eddy Current inspection around the four fasteners on each stop strap at stringer S-7 thru S-14.

See Doc. D626A001-DTR, DTR check form 53-10-14-8 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-01.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR FWD EDGE FRAME STOPS
		D633A109-AKS 53-618-00-01

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737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FWD ENTRY DOOR, FWD EDGE FRAME STOPS			BOEING CARD NO. 53-619-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS NOTE		ZONE 221	
<p>Inspect (High Frequency Eddy Current) around the first two (2) fasteners in the necked down section of the stop straps at S-15 and S-16.</p> <p>See Doc. D626A001-DTR, DTR check form 53-10-14-9 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-01.</p> <p>ACCESS NOTE: Remove interior panels as required to perform inspection.</p>					
EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR, FWD EDGE FRAME STOPS D633A109-AKS 53-619-00-01			
				Page 1 of 2 Oct 15/2014	

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-619-00-01
TASK 53-05-02-250-819				MECH INSP
1. INTERNAL - SPECIAL DETAILED: FWD ENTRY DOOR FWD EDGE FRAME STOPS				
A. Inspection				
NOTE: Remove interior panels as required to perform inspection				
SUBTASK 53-05-02-250-019				
(1) Do a High Frequency Eddy Current inspection around the first two (2) fasteners in the necked down section of the stop straps at S-15 and S-16.				
See Doc. D626A001-DTR, DTR check form 53-10-14-9 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-01.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR, FWD EDGE FRAME STOPS		
		D633A109-AKS 53-619-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FWD ENTRY DOOR, AFT EDGE FRAME STOPS			BOEING CARD NO.
DATE	TASK DETAILED				53-620-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 800 900 900ER ENGINE ALL
		ACCESS NOTE			ZONE 221

Inspect (Detailed) the S-15 and S-16 tension straps at the fastener holes on either side of STA 351.2 frame.

See Doc. D626A001-DTR, DTR check form 53-10-14-10 for alternative inspections.

ACCESS NOTE: Open FWD Entry Door. Removal of interior panel is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR, AFT EDGE FRAME STOPS
		D633A109-AKS 53-620-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-620-00-01
TASK 53-05-02-211-807				MECH INSP
1. INTERNAL - DETAILED: FWD ENTRY DOOR AFT EDGE FRAME STOPS				
A. Inspection				
SUBTASK 53-05-02-211-007				
NOTE: Open FWD Entry Door. Removal of interior panel is required to perform the inspection.				
(1) Do a Detailed inspection of the S-15 and S-16 tension straps at the fastener holes on either side of Sta 351.2 frame.				
See Doc. D626A001-DTR, DTR check form 53-10-14-10 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR, AFT EDGE FRAME STOPS		
		D633A109-AKS 53-620-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FWD ENTRY DOOR, AFT EDGE FRAME STOPS			BOEING CARD NO. 53-621-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE			ZONE 221	

Inspect (Detailed) the Aft frame stops at the inner flange holes near STA 348.2 from stringer S-7 to S-14.

See Doc. D626A001-DTR, DTR check form 53-10-14-11 for alternative inspections.

ACCESS NOTE: Open FWD Entry Door. Removal of interior panel is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR, AFT EDGE FRAME STOPS
		D633A109-AKS 53-621-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-621-00-01
TASK 53-05-02-211-808				MECH INSP
1. INTERNAL - DETAILED: FWD ENTRY DOOR AFT EDGE FRAME STOPS				
A. Inspection				
NOTE: Open FWD Entry Door. Removal of interior panel is required to perform the inspection.				
SUBTASK 53-05-02-211-008				
(1) Do a Detailed inspection of the Aft frame stops at the inner flange holes near Sta 348.2 from stringer S-7 to S-14.				
See Doc. D626A001-DTR, DTR check form 53-10-14-11 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR, AFT EDGE FRAME STOPS		
		D633A109-AKS 53-621-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FWD ENTRY DOOR CUTOUT			BOEING CARD NO. 53-622-00-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE			ZONE 117 221	

Inspect (High Frequency Eddy Current) the skin around the fastener holes and along the edge of the cutout hidden by the scuff plates from STA 303 to STA 350.

See Doc. D626A001-DTR, DTR check form 53-10-14-12 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-43.

ACCESS NOTE: Removal of scuff plate is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR CUTOUT D633A109-AKS 53-622-00-01	Page 1 of 2 Oct 15/2014
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-622-00-01
				MECH INSP
TASK 53-05-02-250-820				
1. INTERNAL - SPECIAL DETAILED: FWD ENTRY DOOR CUTOUT				
A. Inspection				
NOTE: Removal of scuff plate is required to perform the inspection.				
SUBTASK 53-05-02-250-020				
(1) Do a High Frequency Eddy Current inspection of the skin around the fastener holes and along the edge of the cutout hidden by the scuff plates from STA 303 to STA 350.				
See Doc. D626A001-DTR, DTR check form 53-10-14-12 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-43.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	FWD ENTRY DOOR CUTOUT		
		D633A109-AKS 53-622-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FWD GALLEY DOOR CUTOUT			BOEING CARD NO. 53-623-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			
			ZONE 118 222		

Inspect (High Frequency Eddy Current) the skin around the fastener holes and along the edge of the cutout hidden by the scuff plates from STA 303 to STA 350. (53-10-15).

See Doc. D626A001-DTR, DTR check form 53-10-14-12 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-43.

ACCESS NOTE: Removal of scuff plate is required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FWD GALLEY DOOR CUTOUT
		D633A109-AKS 53-623-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-623-00-01
				MECH INSP
TASK 53-05-02-250-931				
1. INTERNAL - SPECIAL DETAILED: FWD GALLEY DOOR CUTOUT				
A. Inspection				
NOTE: Removal of scuff plate is required to perform the inspection.				
SUBTASK 53-05-02-250-131				
(1) Do a High Frequency Eddy Current inspection of the skin around the fastener holes and along the edge of the cutout hidden by the scuff plates from STA 303 to STA 350. (53-10-15). See Doc. D626A001-DTR, DTR check form 53-10-14-12 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-43.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	FWD GALLEY DOOR CUTOUT		
		D633A109-AKS 53-623-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD GALLEY DOOR SURROUND STRUCTURE, DOOR STOP BACKUP STRUCTURE			BOEING CARD NO. 53-624-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 222

Inspect (High Frequency Eddy Current) the four (4) fastener at each stop location common to the intercostal tension strap at the forward edge frame at stops #1, #2, #5, #6.

See Doc. D626A001-DTR, DTR check form 53-10-15-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-98.

ACCESS NOTE: Remove interior panels as required to perform inspection. Door stops are numbered from the bottom up.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD GALLEY DOOR SURROUND STRUCTURE, DOOR STOP BACKUP STRUCTURE
		D633A109-AKS 53-624-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-624-00-01
				MECH INSP
TASK 53-05-02-250-821				
1. INTERNAL - SPECIAL DETAILED: FORWARD GALLEY DOOR SURROUND STRUCTURE DOOR STOP BACKUP STRUCTURE				
A. Inspection				
<p><u>NOTE:</u> Remove interior panels as required to perform inspection. Door stops are numbered from the bottom up.</p> <p>SUBTASK 53-05-02-250-021</p> <p>(1) Do a High Frequency Eddy Current inspection of the four (4) fastener at each stop location common to the intercostal tension strap at the forward edge frame at stops #1, #2, #5, #6.</p> <p>See Doc. D626A001-DTR, DTR check form 53-10-15-2 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-98.</p> <p style="text-align: center;">— END OF TASK —</p>				
EFFECTIVITY AKS ALL		SOURCE AWL	FORWARD GALLEY DOOR SURROUND STRUCTURE, DOOR STOP BACKUP STRUCTURE	
			D633A109-AKS 53-624-00-01	Page 2 of 2 Jun 15/2016

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE CUTOUT - FWD GALLEY DOOR			BOEING CARD NO. 53-625-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 222

Inspect (High Frequency Eddy Current) around fasteners common to the skin and chords between stringers S-8R and S-14R at the forward and aft edge frames at STA 291.5 and STA 328.5.

See Doc. D626A001-DTR, DTR check form 53-10-15-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-55.

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT - FWD GALLEY DOOR
		D633A109-AKS 53-625-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-625-00-01
				MECH INSP
TASK 53-05-02-250-822				
1. EXTERNAL - SPECIAL DETAILED: CUTOUT - FWD GALLEY DOOR				
A. Inspection				
SUBTASK 53-05-02-250-022				
(1) Do a High Frequency Eddy Current inspection around fasteners common to the skin and chords between stringers S-8R and S-14R at the forward and aft edge frames at STA 291.5 and STA 328.5.				
See Doc. D626A001-DTR, DTR check form 53-10-15-4 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-55.				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT - FWD GALLEY DOOR
		D633A109-AKS 53-625-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT BULKHEAD, STA 294.5			BOEING CARD NO. 53-626-00-01			
DATE	TASK SPECIAL DETAILED				RELATED CARD			
TAIL NUMBER	WORK AREA NOSE W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ALL			
STATION	SKILL AIRPL				ENGINE ALL			
		ACCESS 117A	NOTE		ZONE 115 117			
<p>Inspect (High Frequency Eddy Current) around the fasteners in the WL 172 beam just outboard of LBL 17 and RBL 17 at the Nose Wheel Well AFT bulkhead.</p> <p>See Doc. D626A001-DTR, DTR check form 53-10-18-1 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-99.</p> <p>ACCESS NOTE: Access through E/E Bay Aft of Nose Wheel Well.</p>								
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT BULKHEAD, STA 294.5 D633A109-AKS 53-626-00-01					
Page 1 of 2 Oct 15/2014								

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-626-00-01				
				MECH INSP				
TASK 53-05-02-250-823								
1. INTERNAL - SPECIAL DETAILED: BULKHEAD, STA 294.5								
A. Inspection								
SUBTASK 53-05-02-010-046								
(1) Open this access panel:								
<table> <thead> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>117A</td> <td>Electronic Equipment Access Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	117A	Electronic Equipment Access Door	
<u>Number</u>	<u>Name/Location</u>							
117A	Electronic Equipment Access Door							
<u>NOTE:</u> Access through E/E Bay Aft of Nose Wheel Well.								
SUBTASK 53-05-02-250-023								
(2) Do a High Frequency Eddy Current inspection around the fasteners in the WL 172 beam just outboard of LBL 17 and RBL 17 at the nose wheel well Aft bulkhead.								
See Doc. D626A001-DTR, DTR check form 53-10-18-1 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-99.								
SUBTASK 53-05-02-410-044								
(3) Close this access panel:								
<table> <thead> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>117A</td> <td>Electronic Equipment Access Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	117A	Electronic Equipment Access Door	
<u>Number</u>	<u>Name/Location</u>							
117A	Electronic Equipment Access Door							
———— END OF TASK ————								

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT BULKHEAD, STA 294.5
		D633A109-AKS 53-626-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT BULKHEAD, STA 294.5			BOEING CARD NO. 53-626-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA NOSE W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 117A			
		NOTE			
ZONE 116 118					

Inspect (High Frequency Eddy Current) around the fasteners in the WL 172 beam just outboard of LBL 17 and RBL 17 at the Nose Wheel Well AFT bulkhead.

See Doc. D626A001-DTR, DTR check form 53-10-18-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-99.

ACCESS NOTE: Access through E/E Bay Aft of Nose Wheel Well.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT BULKHEAD, STA 294.5
		D633A109-AKS 53-626-00-02

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-626-00-02				
				MECH INSP				
TASK 53-05-02-250-823								
1. INTERNAL - SPECIAL DETAILED: BULKHEAD, STA 294.5								
A. Inspection								
SUBTASK 53-05-02-010-046								
(1) Open this access panel:								
<table> <thead> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>117A</td> <td>Electronic Equipment Access Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	117A	Electronic Equipment Access Door	
<u>Number</u>	<u>Name/Location</u>							
117A	Electronic Equipment Access Door							
NOTE: Access through E/E Bay Aft of Nose Wheel Well.								
SUBTASK 53-05-02-250-023								
(2) Do a High Frequency Eddy Current inspection around the fasteners in the WL 172 beam just outboard of LBL 17 and RBL 17 at the nose wheel well Aft bulkhead.								
See Doc. D626A001-DTR, DTR check form 53-10-18-1 for alternative inspections.								
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-99.								
SUBTASK 53-05-02-410-044								
(3) Close this access panel:								
<table> <thead> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>117A</td> <td>Electronic Equipment Access Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	117A	Electronic Equipment Access Door	
<u>Number</u>	<u>Name/Location</u>							
117A	Electronic Equipment Access Door							
———— END OF TASK ————								

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT BULKHEAD, STA 294.5
		D633A109-AKS 53-626-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT NOSE WHEEL WELL SIDE AND TOP PANELS			BOEING CARD NO.
DATE	TASK DETAILED				53-627-00-01
TAIL NUMBER	WORK AREA NOSE W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 800 900 900ER ENGINE ALL
		ACCESS 113BW			ZONE 113 115
		NOTE			

Inspect (Detailed) the AFT access cutout forward vertical beam at STA 260, from WL 170 to WL 184.

See Doc. D626A001-DTR, DTR check form 53-10-19-4 for alternative inspections.

ACCESS NOTE: Access through Left Aft Nose Wheel Well Panel.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT NOSE WHEEL WELL SIDE AND TOP PANELS
		D633A109-AKS 53-627-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-627-00-01	MECH	INSP
TASK 53-05-02-211-809						
1. INTERNAL - DETAILED: NOSE WHEEL WELL SIDE AND TOP PANELS						
A. Inspection						
SUBTASK 53-05-02-010-014						
(1) Open this access panel on the Left side:						
Number Name/Location						
113BW Forward Nose Wheel Well Panel						
Open this access panel on the Right side:						
Number Name/Location						
114BW Forward Nose Wheel Well Panel						
NOTE: Access through Left Aft Nose Wheel Well Panel.						
Access through Right Aft Nose Wheel Well Panel.						
SUBTASK 53-05-02-211-009						
(2) Do a Detailed inspection of the Aft access cutout forward vertical beam at Sta 260, from WL 170 to WL 184.						
See Doc. D626A001-DTR, DTR check form 53-10-19-4, for alternative inspections.						
SUBTASK 53-05-02-410-012						
(3) Close this access panel on the Left side:						
Number Name/Location						
113BW Forward Nose Wheel Well Panel						
Close this access panel on the Right side:						
Number Name/Location						
114BW Forward Nose Wheel Well Panel						
———— END OF TASK ————						

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT NOSE WHEEL WELL SIDE AND TOP PANELS	
		D633A109-AKS 53-627-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT NOSE WHEEL WELL SIDE AND TOP PANELS			BOEING CARD NO. 53-627-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA NOSE W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 800 900 900ER ENGINE ALL
		ACCESS 114BW			ZONE 114 116
		NOTE			

Inspect (Detailed) the AFT access cutout forward vertical beam at STA 260, from WL 170 to WL 184.

See Doc. D626A001-DTR, DTR check form 53-10-19-4 for alternative inspections.

ACCESS NOTE: Access through Right Aft Nose Wheel Well Panel.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT NOSE WHEEL WELL SIDE AND TOP PANELS
		D633A109-AKS 53-627-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-627-00-02
				MECH INSP
TASK 53-05-02-211-809				
1. INTERNAL - DETAILED: NOSE WHEEL WELL SIDE AND TOP PANELS				
A. Inspection				
SUBTASK 53-05-02-010-014				
(1) Open this access panel on the Left side:				
Number Name/Location				
113BW Forward Nose Wheel Well Panel				
Open this access panel on the Right side:				
Number Name/Location				
114BW Forward Nose Wheel Well Panel				
NOTE: Access through Left Aft Nose Wheel Well Panel.				
Access through Right Aft Nose Wheel Well Panel.				
SUBTASK 53-05-02-211-009				
(2) Do a Detailed inspection of the Aft access cutout forward vertical beam at Sta 260, from WL 170 to WL 184.				
See Doc. D626A001-DTR, DTR check form 53-10-19-4, for alternative inspections.				
SUBTASK 53-05-02-410-012				
(3) Close this access panel on the Left side:				
Number Name/Location				
113BW Forward Nose Wheel Well Panel				
Close this access panel on the Right side:				
Number Name/Location				
114BW Forward Nose Wheel Well Panel				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT NOSE WHEEL WELL SIDE AND TOP PANELS
		D633A109-AKS 53-627-00-02

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT NOSE LANDING GEAR SUPPORT FITTINGS			BOEING CARD NO. 53-628-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA NOSE W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 17000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL				
		ACCESS 113AW 113BW			ZONE 113 115
		NOTE			

Inspect (Detailed) the inboard and outboard drag brace fittings around the perimeter of the bushings at STA 262, BL 16, WL 189.3.

See Doc. D626A001-DTR, DTR check form 53-10-20-2 for alternative inspections.

ACCESS NOTE: For Direction 1, removal of Drag Brace is required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT NOSE LANDING GEAR SUPPORT FITTINGS
		D633A109-AKS 53-628-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-628-00-01
TASK 53-05-02-211-810				MECH INSP
1. INTERNAL - DETAILED: NOSE LANDING GEAR SUPPORT FITTINGS				
A. Inspection				
SUBTASK 53-05-02-010-015				
(1) Open these access panels on the Left side:				
Number Name/Location				
113AW Forward Nose Wheel Well Panel				
113BW Forward Nose Wheel Well Panel				
Open these access panels on the Right side:				
Number Name/Location				
114AW Forward Nose Wheel Well Panel				
114BW Forward Nose Wheel Well Panel				
NOTE: For Direction 1, removal of Drag Brace is required.				
SUBTASK 53-05-02-211-010				
(2) Do a Detailed inspection of the inboard and outboard drag brace fittings around the perimeter of the bushings at Sta 262, BL 16, WL 189.3.				
See Doc. D626A001-DTR, DTR check form 53-10-20-2 for alternative inspections.				
SUBTASK 53-05-02-410-013				
(3) Close these access panels on the Left side:				
Number Name/Location				
113AW Forward Nose Wheel Well Panel				
113BW Forward Nose Wheel Well Panel				
Close these access panels on the Right side:				
Number Name/Location				
114AW Forward Nose Wheel Well Panel				
114BW Forward Nose Wheel Well Panel				
END OF TASK				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT NOSE LANDING GEAR SUPPORT FITTINGS	
		D633A109-AKS 53-628-00-01	Page 2 of 2 Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT NOSE LANDING GEAR SUPPORT FITTINGS			BOEING CARD NO. 53-628-00-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA NOSE W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 17000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 114AW 114BW			ZONE 114 116	
		NOTE				

Inspect (Detailed) the inboard and outboard drag brace fittings around the perimeter of the bushings at STA 262, BL 16, WL 189.3.

See Doc. D626A001-DTR, DTR check form 53-10-20-2 for alternative inspections.

ACCESS NOTE: For Direction 1, removal of Drag Brace is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT NOSE LANDING GEAR SUPPORT FITTINGS
		D633A109-AKS 53-628-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-628-00-02
TASK 53-05-02-211-810				MECH INSP

1. INTERNAL - DETAILED: NOSE LANDING GEAR SUPPORT FITTINGS

A. Inspection

SUBTASK 53-05-02-010-015

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
113AW	Forward Nose Wheel Well Panel
113BW	Forward Nose Wheel Well Panel

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
114AW	Forward Nose Wheel Well Panel
114BW	Forward Nose Wheel Well Panel

NOTE: For Direction 1, removal of Drag Brace is required.

SUBTASK 53-05-02-211-010

(2) Do a Detailed inspection of the inboard and outboard drag brace fittings around the perimeter of the bushings at Sta 262, BL 16, WL 189.3.
See Doc. D626A001-DTR, DTR check form 53-10-20-2 for alternative inspections.

SUBTASK 53-05-02-410-013

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
113AW	Forward Nose Wheel Well Panel
113BW	Forward Nose Wheel Well Panel

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
114AW	Forward Nose Wheel Well Panel
114BW	Forward Nose Wheel Well Panel

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT NOSE LANDING GEAR SUPPORT FITTINGS
		D633A109-AKS 53-628-00-02

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT NOSE LANDING GEAR TRUNNION SUPPORT FITTINGS			BOEING CARD NO. 53-629-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA NOSE W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS 113BW 117A			ZONE 113 115

Inspect (Detailed) the inboard and outboard fitting segments of the Trunnion Support Fitting around the pin socket at BS 294.5, WL 156.1, and BL 16.

See Doc. D626A001-DTR, DTR check form 53-10-20-3 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT NOSE LANDING GEAR TRUNNION SUPPORT FITTINGS
		D633A109-AKS 53-629-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-629-00-01
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TASK 53-05-02-211-811

1. INTERNAL - DETAILED: NOSE LANDING GEAR TRUNNION SUPPORT FITTING

A. Inspection

SUBTASK 53-05-02-010-016

- (1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
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113BW Forward Nose Wheel Well Panel

117A Electronic Equipment Access Doc

Open these access panels on

Number Name/Location

114BW Forward Nose Wheel Well Panel

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- (2) Do a Detailed inspection of the inboard and outboard fitting segments of the Trunnion

Support fitting around the pin socket at BG 254.3, WE 158.1, and BE 18.

- (3) Close these access panels on the Left side:

Number Name/I section

Number **Name/Location**

115BW Forward Nose Wheel Well and
117A Electronic Equipment Access Door

Close these access panels or

Number Name/Section

Number **Name/Location**

END OF TASK

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT NOSE LANDING GEAR TRUNNION SUPPORT FITTINGS
		D633A109-AKS 53-629-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LANDING GEAR TRUNNION SUPPORT FITTINGS			BOEING CARD NO. 53-629-00-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA NOSE W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 114BW 117A			ZONE 114 116	

Inspect (Detailed) the inboard and outboard fitting segments of the Trunnion Support Fitting around the pin socket at BS 294.5, WL 156.1, and BL 16.

See Doc. D626A001-DTR, DTR check form 53-10-20-3 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LANDING GEAR TRUNNION SUPPORT FITTINGS
		D633A109-AKS 53-629-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-629-00-02
TASK 53-05-02-211-811				MECH INSP

1. INTERNAL - DETAILED: NOSE LANDING GEAR TRUNNION SUPPORT FITTING

A. Inspection

SUBTASK 53-05-02-010-016

(1) Open these access panels on the Left side:

Number	Name/Location
113BW	Forward Nose Wheel Well Panel
117A	Electronic Equipment Access Door

Open these access panels on the Right side:

Number	Name/Location
114BW	Forward Nose Wheel Well Panel
117A	Electronic Equipment Access Door

SUBTASK 53-05-02-211-011

(2) Do a Detailed inspection of the inboard and outboard fitting segments of the Trunnion Support Fitting around the pin socket at BS 294.5, WL 156.1, and BL 16.

See Doc. D626A001-DTR, DTR check form 53-10-20-3 for alternative inspections.

SUBTASK 53-05-02-410-014

(3) Close these access panels on the Left side:

Number	Name/Location
113BW	Forward Nose Wheel Well Panel
117A	Electronic Equipment Access Door

Close these access panels on the Right side:

Number	Name/Location
114BW	Forward Nose Wheel Well Panel
117A	Electronic Equipment Access Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LANDING GEAR TRUNNION SUPPORT FITTINGS
		D633A109-AKS 53-629-00-02

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AIR STAIR DOOR CUTOUT			BOEING CARD NO. 53-630-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA E/E COMPARTMENT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS 117A			
		NOTE			
					ZONE 117

Inspect (Detailed) the upper and lower outer sill chords from STA 303.8 to STA 348.

See Doc. D626A001-DTR, DTR check form 53-10-21-1 for alternative inspections.

AIRPLANE NOTE: Applicable to airplanes with airstairs installed.

ACCESS NOTE: Opening of Air Stair Door is required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AIR STAIR DOOR CUTOUT
		D633A109-AKS 53-630-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-630-00-01
				MECH INSP
TASK 53-05-02-211-812				
1. INTERNAL - DETAILED: AIR STAIR DOOR CUTOUT				
A. Inspection				
SUBTASK 53-05-02-010-017				
(1) Open this access panel:				
Number Name/Location				
117A Electronic Equipment Access Door				
<u>NOTE:</u> Opening of Air Stair Door is required to perform this inspection.				
SUBTASK 53-05-02-211-012				
(2) Do a Detailed inspection of the upper and lower outer sill chords from Sta 303.8 to Sta 348.				
See Doc. D626A001-DTR, DTR check form 53-10-21-1 for alternative inspections.				
SUBTASK 53-05-02-410-015				
(3) Close this access panel:				
Number Name/Location				
117A Electronic Equipment Access Door				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	AIR STAIR DOOR CUTOUT
		D633A109-AKS 53-630-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AIR STAIR DOOR CUTOUT			BOEING CARD NO. 53-631-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA E/E COMPARTMENT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS 117A 117BL			
		NOTE			
			ZONE 117		

Inspect (Low Frequency Eddy Current) the forward and aft edge frame webs around the fasteners common to the corner clips at STA 303.9 and STA 351.2.

See Doc. D626A001-DTR, DTR check form 53-10-21-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-27.

AIRPLANE NOTE: Applicable to airplanes with airstairs installed.

ACCESS NOTE: Opening of Air Stair Door is required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AIR STAIR DOOR CUTOUT D633A109-AKS 53-631-00-01	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-631-00-01
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TASK 53-05-02-250-824

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: AIR STAIR DOOR CUTOUT**A. Inspection**

SUBTASK 53-05-02-010-175

- (1) Open these access panels:

Number Name/Location

117A Electronic Equipment Access Door
117BL Forward Airstair Door

NOTE: Opening of Air Stair Door is required to perform this inspection.

SUBTASK 53-05-02-250-024

- (2) Do a Low Frequency Eddy Current inspection of the forward and aft edge frame webs around the fasteners common to the corner clips at STA 303.9 and STA 351.2.

See Doc. D626A001-DTR, DTR check form 53-10-21-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-27.

SUBTASK 53-05-02-410-166

- (3) Close these doors:

Number Name/Location

117A Electronic Equipment Access Door
117BL Forward Airstair Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	AIR STAIR DOOR CUTOUT
		D633A109-AKS 53-631-00-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE CROWN SKIN PANEL, STA 360 TO 540			BOEING CARD NO.
DATE	TASK DETAILED				53-633-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 231 232

Inspect (Detailed) the skin around all of the fastener locations from stringers S-10L to S-10R, from STA 360 to STA 540, except at the lap splices and antennas.

See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL, STA 360 TO 540
		D633A109-AKS 53-633-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-633-00-01
TASK 53-05-02-211-813				MECH INSP
1. EXTERNAL - DETAILED: CROWN SKIN PANEL STA 360 TO 540				
A. Inspection				
SUBTASK 53-05-02-211-013				
(1) Do a Detailed inspection of the skin around all of the fastener locations from stringer S-10L to S-10R, from Sta 360 to Sta 540, except at the lap splices and antennas. See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL, STA 360 TO 540		
		D633A109-AKS 53-633-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CROWN SKIN PANEL, STA 360 TO 540			BOEING CARD NO.
DATE	TASK DETAILED				53-634-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS NOTE			ZONE 231 232

Inspect (Detailed) the exterior surface of the skin under the TCAS Antenna at STA 385, the ATC Antenna at STA 430, and the GPS Antenna at STA 500A.

See Doc D626A001-DTR, DTR check form 53-30-01-4 for alternative inspections.

AIRPLANE NOTE: For the 737-600, GPS Antennas are located at STA 482A.

ACCESS NOTE: Removal of external antenna fairings and base plates are required

EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL, STA 360 TO 540
		D633A109-AKS 53-634-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-634-00-01
				MECH INSP
TASK 53-05-02-211-814				
1. INTERNAL - DETAILED: CROWN SKIN PANEL STA 360 TO 540				
A. Inspection				
NOTE: Removal of external antenna fairings and base plates are required.				
SUBTASK 53-05-02-211-014				
(1) Do a Detailed inspection of the exterior surface of the skin under the TCAS Antenna at Sta 385, the ATC Antenna at Sta 430, and the GPS Antenna at Sta 500A.				
See Doc D626A001-DTR, DTR check form 53-30-01-4 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL, STA 360 TO 540		
		D633A109-AKS 53-634-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT SATCOM AERO - H ANTENNA INSTALLATION			BOEING CARD NO.
DATE	TASK DETAILED				53-635-00-01
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 800 NOTE ENGINE ALL
		ACCESS NOTE			ZONE 231

Inspect (Detailed) the skin near the fastener locations around the antenna cutout, stringers, and antenna nutplates on both the left and right sides of the aircraft at STA 500 between stringers S-6 and S-7.

See Doc D626A001-DTR, DTR check form 53-30-01-5 for alternative inspections.

AIRPLANE NOTE: For the 737-600 GPS Antennas are located at Sta 482A + 5, LBL 5 and RBL 5.

ACCESS NOTE: Removal of antenna is required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT SATCOM AERO - H ANTENNA INSTALLATION
		D633A109-AKS 53-635-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-635-00-01
				MECH INSP
TASK 53-05-02-211-815				
1. <u>INTERNAL - DETAILED: SATCOM AERO-H ANTENNA INSTALLATION</u>				
A. General				
B. Inspection				
NOTE: Removal of antenna is required.				
SUBTASK 53-05-02-211-015				
(1) Do a Detailed inspection of the skin near the fastener locations around the antenna cutout, stringers, and antenna nutplates on both the left and right sides of the aircraft at Sta 500 between stringers S-6 and S-7.				
See Doc D626A001-DTR, DTR check form 53-30-01-5 for alternative inspections.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT SATCOM AERO - H ANTENNA INSTALLATION				
D633A109-AKS 53-635-00-01				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT SATCOM AERO - H ANTENNA INSTALLATION			BOEING CARD NO.
DATE	TASK DETAILED				53-635-00-02
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 800 NOTE ENGINE ALL
		ACCESS NOTE			ZONE 232

Inspect (Detailed) the skin near the fastener locations around the antenna cutout, stringers, and antenna nutplates on both the left and right sides of the aircraft at STA 500 between stringers S-6 and S-7.

See Doc D626A001-DTR, DTR check form 53-30-01-5 for alternative inspections.

AIRPLANE NOTE: For the 737-600 GPS Antennas are located at Sta 482A + 5, LBL 5 and RBL 5.

ACCESS NOTE: Removal of antenna is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT SATCOM AERO - H ANTENNA INSTALLATION
		D633A109-AKS 53-635-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-635-00-02
TASK 53-05-02-211-815				MECH INSP
1. INTERNAL - DETAILED: SATCOM AERO-H ANTENNA INSTALLATION				
A. General				
B. Inspection				
NOTE: Removal of antenna is required.				
SUBTASK 53-05-02-211-015				
(1) Do a Detailed inspection of the skin near the fastener locations around the antenna cutout, stringers, and antenna nutplates on both the left and right sides of the aircraft at Sta 500 between stringers S-6 and S-7.				
See Doc D626A001-DTR, DTR check form 53-30-01-5 for alternative inspections.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT SATCOM AERO - H ANTENNA INSTALLATION				
D633A109-AKS 53-635-00-02				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT FUSELAGE SIDE SKIN PANELS			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-636-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 191AL 195AL 195BL 195CL			ZONE 191 195 231
		NOTE			

Inspect (General Visual) the skin from STA 360 to STA 540 between stringers S-14 to S-17.

See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FUSELAGE SIDE SKIN PANELS
		D633A109-AKS 53-636-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-636-00-01
TASK 53-05-02-210-801				MECH INSP
1. EXTERNAL - GENERAL VISUAL: FUSELAGE SIDE SKIN PANELS				
A. Inspection				
SUBTASK 53-05-02-010-009				
(1) Open these access panels on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
195AL Wing To Body Fairing - Left Side				
195BL Wing To Body Fairing - Left Side				
195CL Wing To Body Fairing - Left Side				
Open these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
195AR Wing To Body Fairing - Right Side				
195BR Wing To Body Fairing - Right Side				
195CR Wing To Body Fairings - Right Side				
NOTE: Remove or displace aft wing to body fairings as required.				
SUBTASK 53-05-02-210-001				
(2) Do a General Visual inspection of the skin from Sta 360 to Sta 540 between stringers S-14 to S-17.				
See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.				
SUBTASK 53-05-02-410-007				
(3) Close these access panels on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
195AL Wing To Body Fairing - Left Side				
195BL Wing To Body Fairing - Left Side				
195CL Wing To Body Fairing - Left Side				
Close these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
195AR Wing To Body Fairing - Right Side				
195BR Wing To Body Fairing - Right Side				
195CR Wing To Body Fairings - Right Side				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FUSELAGE SIDE SKIN PANELS
		D633A109-AKS 53-636-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT FUSELAGE SIDE SKIN PANELS			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-636-00-02 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 191AR 195AR 195BR 195CR			ZONE 191 196 232
		NOTE			

Inspect (General Visual) the skin from STA 360 to STA 540 between stringers S-14 to S-17.

See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT FUSELAGE SIDE SKIN PANELS
		D633A109-AKS 53-636-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-636-00-02
TASK 53-05-02-210-801				MECH INSP

1. EXTERNAL - GENERAL VISUAL: FUSELAGE SIDE SKIN PANELS

A. Inspection

SUBTASK 53-05-02-010-009

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
195AL	Wing To Body Fairing - Left Side
195BL	Wing To Body Fairing - Left Side
195CL	Wing To Body Fairing - Left Side

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
195AR	Wing To Body Fairing - Right Side
195BR	Wing To Body Fairing - Right Side
195CR	Wing To Body Fairings - Right Side

NOTE: Remove or displace aft wing to body fairings as required.

SUBTASK 53-05-02-210-001

(2) Do a General Visual inspection of the skin from Sta 360 to Sta 540 between stringers S-14 to S-17.

See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.

SUBTASK 53-05-02-410-007

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
195AL	Wing To Body Fairing - Left Side
195BL	Wing To Body Fairing - Left Side
195CL	Wing To Body Fairing - Left Side

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
195AR	Wing To Body Fairing - Right Side
195BR	Wing To Body Fairing - Right Side
195CR	Wing To Body Fairings - Right Side

———— END OF TASK ——

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT FUSELAGE SIDE SKIN PANELS
		D633A109-AKS 53-636-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING			BOEING CARD NO. 53-636-10-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 8000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 191AL 195AL			ZONE 191 195	
		NOTE				

Inspect (Detailed) the fuselage skin panels under the Wing to Body Fairing from STA 360 to STA 540.

See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING
		D633A109-AKS 53-636-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-636-10-01
				MECH INSP
TASK 53-05-02-211-816				
1. EXTERNAL - DETAILED: FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING				
A. Inspection				
SUBTASK 53-05-02-010-018				
(1) Open this access panel on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
195AL Wing To Body Fairing - Left Side				
Open these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
195AR Wing To Body Fairing - Right Side				
NOTE: Remove or displace wing to body fairings as required to perform this inspection.				
SUBTASK 53-05-02-211-016				
(2) Do a Detailed inspection of the fuselage skin panels under the Wing to Body Fairing from Sta 360 to Sta 540.				
See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.				
SUBTASK 53-05-02-410-016				
(3) Close these access panels on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
195AL Wing To Body Fairing - Left Side				
Close these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
195AR Wing To Body Fairing - Right Side				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING	D633A109-AKS 53-636-10-01	Page 2 of 2 Feb 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING			BOEING CARD NO. 53-636-10-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 8000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 191AR 195AR				ZONE 191 196
		NOTE				

Inspect (Detailed) the fuselage skin panels under the Wing to Body Fairing from STA 360 to STA 540.

See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING
		D633A109-AKS 53-636-10-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-636-10-02
				MECH INSP
TASK 53-05-02-211-816				
1. EXTERNAL - DETAILED: FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING				
A. Inspection				
SUBTASK 53-05-02-010-018				
(1) Open this access panel on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
195AL Wing To Body Fairing - Left Side				
Open these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
195AR Wing To Body Fairing - Right Side				
NOTE: Remove or displace wing to body fairings as required to perform this inspection.				
SUBTASK 53-05-02-211-016				
(2) Do a Detailed inspection of the fuselage skin panels under the Wing to Body Fairing from Sta 360 to Sta 540.				
See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.				
SUBTASK 53-05-02-410-016				
(3) Close these access panels on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
195AL Wing To Body Fairing - Left Side				
Close these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
195AR Wing To Body Fairing - Right Side				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING
		D633A109-AKS 53-636-10-02

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-637-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE 600 700 700IGW ALL 800 900 900ER
		ACCESS			ZONE 231

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 360 to STA 540. (PSE 53-30-04-1).

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-637-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-637-00-01
				MECH INSP
TASK 53-05-02-250-932				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE				
A. Inspection				
SUBTASK 53-05-02-250-132				
<p>(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 360 to STA 540 (PSE 53-30-04-1). See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.</p>				
<hr/> <hr/> <hr/>				
END OF TASK				

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-637-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 232

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 360 to STA 540. (PSE 53-30-04-1).

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-637-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-637-00-02
TASK 53-05-02-250-932				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-132				
(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 360 to STA 540 (PSE 53-30-04-1). See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-637-00-02	Page 2 of 2 Feb 15/2015	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-638-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL
STATION	SKILL AIRPL				800 900 900ER
		ACCESS			ZONE 231

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-638-00-01	Page 1 of 2 Jun 15/2015
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-638-00-01
				MECH INSP
TASK 53-05-02-250-826				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE				
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-026</p> <p>(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 360 to STA 540.</p> <p>See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.</p> <p style="text-align: center;">— END OF TASK —</p>				
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT LONGITUDINAL LAP SPLICE</p> <p>D633A109-AKS 53-638-00-01</p>				

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-638-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 232

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-638-00-02	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-638-00-02
TASK 53-05-02-250-826				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-026				
(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 360 to STA 540. See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-638-00-02	Page 2 of 2 Jun 15/2015	

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737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL SKIN SPLICING			BOEING CARD NO. 53-639-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 600 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 231

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL SKIN SPLIC
		D633A109-AKS 53-639-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-639-00-01	
					MECH INSP
TASK 53-05-02-250-828					
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL SKIN SPLICE					
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-028</p> <p>(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 360 to STA 540.</p> <p>See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.</p> <p style="text-align: center;">— END OF TASK —</p>					
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT LONGITUDINAL SKIN SPLICE</p> <p>D633A109-AKS 53-639-00-01</p>					

AKS



737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL SKIN SPLICE			BOEING CARD NO. 53-639-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 232

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL SKIN SPLIC
		D633A109-AKS 53-639-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-639-00-02
TASK 53-05-02-250-828				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL SKIN SPLICE				
A. Inspection				
SUBTASK 53-05-02-250-028				
(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 360 to STA 540. See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.				
— END OF TASK —				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT LONGITUDINAL SKIN SPLICE				
D633A109-AKS 53-639-00-02				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO.
DATE	TASK DETAILED				53-640-00-01
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE 600 700 700IGW 800 900 900ER ENGINE ALL
		ACCESS NOTE			ZONE 231

Inspect (Detailed) the lower skin along the lower fastener row at stringers S-10L and S-10R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-4, for alternative inspections.

ACCESS NOTE: Removal or displacement of interior sidewall panels and insulation blankets are required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-640-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-640-00-01
TASK 53-05-02-211-817				MECH INSP
1. INTERNAL - DETAILED: LONGITUDINAL LAP SPLICE				
A. Inspection				
<p><u>NOTE:</u> Removal or displacement of interior sidewall panels and insulation blankets are required.</p> <p>SUBTASK 53-05-02-211-017</p> <p>(1) Do a Detailed inspection of the lower skin along the lower fastener row at stringers S-10L and S-10R from Sta 360 to Sta 540.</p> <p>See Doc. D626A001-DTR, DTR check form 53-30-04-4, for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT LONGITUDINAL LAP SPLICE				
D633A109-AKS 53-640-00-01				
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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-640-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 700IGW 800 900 900ER ENGINE ALL
		ACCESS NOTE			ZONE 232

Inspect (Detailed) the lower skin along the lower fastener row at stringers S-10L and S-10R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-4, for alternative inspections.

ACCESS NOTE: Removal or displacement of interior sidewall panels and insulation blankets are required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-640-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-640-00-02
TASK 53-05-02-211-817				MECH INSP
1. INTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
<p><u>NOTE:</u> Removal or displacement of interior sidewall panels and insulation blankets are required.</p> <p>SUBTASK 53-05-02-211-017</p> <p>(1) Do a Detailed inspection of the lower skin along the lower fastener row at stringers S-10L and S-10R from Sta 360 to Sta 540.</p> <p>See Doc. D626A001-DTR, DTR check form 53-30-04-4, for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT LONGITUDINAL LAP SPLICE				
D633A109-AKS 53-640-00-02				
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Feb 15/2015				

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-641-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 700IGW 800 900 900ER ENGINE ALL
		ACCESS			ZONE 231

Inspect (Detailed) the upper skin along the upper fastener row at stringers S-14L and S-14R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-5, for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-641-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-641-00-01
TASK 53-05-02-211-819				MECH INSP
1. EXTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-211-019				
(1) Do a Detailed inspection of the upper skin along the upper fastener row at stringers S-14L and S-14R from Sta 360 to Sta 540. See Doc. D626A001-DTR, DTR check form 53-30-04-5, for alternative inspections.				
———— END OF TASK ——				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-641-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO.
DATE	TASK DETAILED				53-641-00-02
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY
					AIRPLANE 600 700 700IGW 800 900 900ER
		ACCESS			ENGINE ALL
					ZONE 232

Inspect (Detailed) the upper skin along the upper fastener row at stringers S-14L and S-14R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-5, for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-641-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-641-00-02
TASK 53-05-02-211-819				MECH INSP
1. EXTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-211-019				
(1) Do a Detailed inspection of the upper skin along the upper fastener row at stringers S-14L and S-14R from Sta 360 to Sta 540. See Doc. D626A001-DTR, DTR check form 53-30-04-5, for alternative inspections.				
———— END OF TASK ——				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-641-00-02	Page 2 of 2 Feb 15/2015	

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737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-642-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL
STATION	SKILL AIRPL				800 900 900ER
		ACCESS			ZONE 231

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-642-00-01	Page 1 of 2 Jun 15/2015
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AKS



737-600/700/800/900 TASK CARDS

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC	
		D633A109-AKS 53-642-00-01	Page 2 of 2 Jun 15/2015

AKS



737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-642-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 232

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 360 to STA 540.

See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-642-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-642-00-02
TASK 53-05-02-250-830				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-030				
(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 360 to STA 540. See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-642-00-02	Page 2 of 2 Jun 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO.			
DATE	TASK SPECIAL DETAILED				53-643-00-01 RELATED CARD			
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY			
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL			
		ACCESS NOTE			ZONE 123 124 125 126 191			
Inspect (Low Frequency Eddy Current) the upper (inner) skin along the upper fastener row at stringers S-24L and S-24R from STA 360 to STA 540, except at the cargo door cutout.								
See Doc. D626A001-DTR, DTR check form 53-30-04-7, for alternative inspections.								
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.								
ACCESS NOTE: Remove Wing to Body Fairing as required to perform this inspection.								
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC					
			D633A109-AKS 53-643-00-01					
Page 1 of 2 Jun 15/2015								

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-643-00-01
				MECH INSP
TASK 53-05-02-250-832				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
NOTE: Remove Wing to Body Fairing as required to perform this inspection.				
SUBTASK 53-05-02-250-032				
(1) Do a Low Frequency Eddy Current inspection of the upper (inner) skin along the upper fastener row at stringers S-24L and S-24R from STA 360 to STA 540, except at the cargo door cutout.				
See Doc. D626A001-DTR, DTR check form 53-30-04-7, for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-643-00-01	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-643-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 122 124 126 191

Inspect (Low Frequency Eddy Current) the upper (inner) skin along the upper fastener row at stringers S-24L and S-24R from STA 360 to STA 540, except at the cargo door cutout.

See Doc. D626A001-DTR, DTR check form 53-30-04-7, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

ACCESS NOTE: Remove Wing to Body Fairing as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-643-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-643-00-02
				MECH INSP
TASK 53-05-02-250-832				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
NOTE: Remove Wing to Body Fairing as required to perform this inspection.				
SUBTASK 53-05-02-250-032				
(1) Do a Low Frequency Eddy Current inspection of the upper (inner) skin along the upper fastener row at stringers S-24L and S-24R from STA 360 to STA 540, except at the cargo door cutout.				
See Doc. D626A001-DTR, DTR check form 53-30-04-7, for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-643-00-02	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-644-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 121 123 125 191

Inspect (Detailed) the lower (outer) skin along the lower fastener row at stringers S-24L and S-24R from STA 360 to STA 540, except at the cargo door cutout.

See Doc. D626A001-DTR, DTR check form 53-30-04-8, for alternative inspections.

ACCESS NOTE: Remove Wing to Body Fairing as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-644-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-644-00-01
TASK 53-05-02-211-821				MECH INSP
1. EXTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
NOTE: Remove Wing to Body Fairing as required to perform this inspection.				
SUBTASK 53-05-02-211-021				
(1) Do a Detailed inspection of the lower (outer) skin along the lower fastener row at stringers S-24L and S-24R from Sta 360 to Sta 540, except at the cargo door cutout.				
See Doc. D626A001-DTR, DTR check form 53-30-04-8, for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-644-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-644-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 122 124 126 191

Inspect (Detailed) the lower (outer) skin along the lower fastener row at stringers S-24L and S-24R from STA 360 to STA 540, except at the cargo door cutout.

See Doc. D626A001-DTR, DTR check form 53-30-04-8, for alternative inspections.

ACCESS NOTE: Remove Wing to Body Fairing as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-644-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-644-00-02
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TASK 53-05-02-211-821

MECH

INSP

1. EXTERNAL - DETAILED: LONGITUDINAL LAP SPLIC**A. Inspection**

NOTE: Remove Wing to Body Fairing as required to perform this inspection.

SUBTASK 53-05-02-211-021

- (1) Do a Detailed inspection of the lower (outer) skin along the lower fastener row at stringers S-24L and S-24R from Sta 360 to Sta 540, except at the cargo door cutout.

See Doc. D626A001-DTR, DTR check form 53-30-04-8, for alternative inspections.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-644-00-02

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WINDOW BELT STA 360 TO 540			BOEING CARD NO. 53-645-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE			ZONE 231	

Inspect (Detailed) the window frames around each window from STA 360 to STA 540. (PSE 53-30-05).

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets is required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 360 TO 540
		D633A109-AKS 53-645-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-645-00-01
TASK 53-05-02-211-863				MECH INSP
1. INTERNAL - DETAILED: WINDOW BELT STA 360 TO STA 540				
A. Inspection				
<p><u>NOTE:</u> Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets is required.</p> <p>SUBTASK 53-05-02-211-063</p> <p>(1) Do a Detailed inspection of the window frames around each window from Sta 360 to Sta 540. (PSE 53-30-05).</p> <p>See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT WINDOW BELT STA 360 TO 540				
D633A109-AKS 53-645-00-01				
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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT WINDOW BELT STA 360 TO 540			BOEING CARD NO.
DATE	TASK DETAILED				53-645-00-02 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 232

Inspect (Detailed) the window frames around each window from STA 360 to STA 540. (PSE 53-30-05).

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 360 TO 540
		D633A109-AKS 53-645-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-645-00-02
TASK 53-05-02-211-863				MECH INSP
1. INTERNAL - DETAILED: WINDOW BELT STA 360 TO STA 540				
A. Inspection				
<p><u>NOTE:</u> Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets is required.</p> <p>SUBTASK 53-05-02-211-063</p> <p>(1) Do a Detailed inspection of the window frames around each window from Sta 360 to Sta 540. (PSE 53-30-05).</p> <p>See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT WINDOW BELT STA 360 TO 540				
D633A109-AKS 53-645-00-02				
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Feb 15/2015				

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WINDOW BELT STA 360 TO STA 540			BOEING CARD NO. 53-645-01-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 231	

Inspect (Detailed) the window frames around each window from STA 360 to STA 540. (PSE 53-30-05).

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 360 TO STA 540
		D633A109-AKS 53-645-01-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-645-01-01
				MECH INSP
TASK 53-05-02-211-864				
1. EXTERNAL - DETAILED: WINDOW BELT STA 360 TO STA 540				
A. Inspection				
SUBTASK 53-05-02-211-064				
(1) Do a Detailed inspection of the window frames around each window from Sta 360 to Sta 540. (PSE 53-30-05). See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 360 TO STA 540		
		D633A109-AKS 53-645-01-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WINDOW BELT STA 360 TO STA 540			BOEING CARD NO. 53-645-01-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 232	

Inspect (Detailed) the window frames around each window from STA 360 to STA 540. (PSE 53-30-05).

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 360 TO STA 540
		D633A109-AKS 53-645-01-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-645-01-02
				MECH INSP
TASK 53-05-02-211-864				
1. EXTERNAL - DETAILED: WINDOW BELT STA 360 TO STA 540				
A. Inspection				
SUBTASK 53-05-02-211-064				
(1) Do a Detailed inspection of the window frames around each window from Sta 360 to Sta 540. (PSE 53-30-05). See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 360 TO STA 540		
		D633A109-AKS 53-645-01-02	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-646-00-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 821			ZONE 122 124	

Inspect (High Frequency Eddy Current) the exposed edge of the bearstrap at both the forward and aft edge of the door at STA 440 and STA 492.4 from stringers S-18R to S-25R. (PSE 53-30-08).

See Doc D626A001-DTR, DTR check form 53-60-08-8 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-21.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-00-01
TASK 53-05-02-250-934				MECH INSP

1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE

A. Inspection

SUBTASK 53-05-02-010-112

(1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
821	Forward Cargo Door

SUBTASK 53-05-02-250-134

(2) Do a High Frequency Eddy Current inspection of the exposed edge of the bearstrap at both the forward and aft edge of the door at STA 440 and STA 492.4 from stringers S-18R to S-25R. (PSE 53-30-08).

See Doc D626A001-DTR, DTR check form 53-60-08-8 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-21.

SUBTASK 53-05-02-410-110

(3) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
821	Forward Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND, FORWARD AND AFT EDGE FRAMES AT STA 440 AND 492.4			BOEING CARD NO. 53-646-10-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 821			ZONE 122 124
		NOTE			

Inspect (High Frequency Eddy Current) the outboard portion of the frame web for damage between stringers S-17R and S-26R, except at the door stops and sill locations.

See Doc D626A001-DTR, DTR check form 53-30-08-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-23.

ACCESS NOTE: Perform inspection with door open. Remove or displace cargo liners as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND, FORWARD AND AFT EDGE FRAMES AT STA 440 AND 492.4
		D633A109-AKS 53-646-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-10-01
				MECH INSP
TASK 53-05-02-250-833				
1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT STA 440 AND 492.4				
A. Inspection				
SUBTASK 53-05-02-010-048				
(1) Open this access panel:				
Number Name/Location				
821 Forward Cargo Door				
NOTE: Perform inspection with door open. Remove or displace cargo liners as required to perform this inspection.				
SUBTASK 53-05-02-250-033				
(2) Do a High Frequency Eddy Current inspection of the outboard portion of the frame web for damage between stringers S-17R and S-26R, except at the door stops and sill locations.				
See Doc D626A001-DTR, DTR check form 53-30-08-1 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-23.				
SUBTASK 53-05-02-410-046				
(3) Close this access panel:				
Number Name/Location				
821 Forward Cargo Door				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND, FORWARD AND AFT EDGE FRAMES AT STA 440 AND 492.4		
		D633A109-AKS 53-646-10-01	Page 2 of 2 Jun 15/2015	

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737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND, FORWARD AND AFT EDGE FRAMES AT STA 440 AND 492.4			BOEING CARD NO. 53-646-20-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 821			ZONE 122 124
		NOTE			

Inspect (Ultrasonic) the outboard portion of the frame web for damage under all door stop fittings and sill clips.

See Doc D626A001-DTR, DTR check form 53-30-08-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 53-10-06.

ACCESS NOTE: Perform inspection with door open. Remove or displace cargo liners as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND, FORWARD AND AFT EDGE FRAMES AT STA 440 AND 492.4
		D633A109-AKS 53-646-20-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-20-01									
TASK 53-05-02-130-801					MECH INSP								
1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT STA 440 AND 492.4													
<p>A. Inspection</p> <p>SUBTASK 53-05-02-010-003</p> <p>(1) Open this access panel:</p> <table style="margin-left: 20px; border-collapse: collapse;"> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> <tr> <td style="padding-left: 10px;">821</td> <td style="padding-left: 10px;">Forward Cargo Door</td> </tr> </table> <p><u>NOTE:</u> Perform inspection with door open. Remove or displace cargo liners as required.</p> <p>SUBTASK 53-05-02-130-001</p> <p>(2) Do an Ultrasonic inspection of the outboard portion of the frame web for damage under all door stop fittings and sill clips</p> <p>See Doc D626A001-DTR, DTR check form 53-30-08-2 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 53-10-06.</p> <p>SUBTASK 53-05-02-410-001</p> <p>(3) Close this access panel:</p> <table style="margin-left: 20px; border-collapse: collapse;"> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> <tr> <td style="padding-left: 10px;">821</td> <td style="padding-left: 10px;">Forward Cargo Door</td> </tr> </table>						<u>Number</u>	<u>Name/Location</u>	821	Forward Cargo Door	<u>Number</u>	<u>Name/Location</u>	821	Forward Cargo Door
<u>Number</u>	<u>Name/Location</u>												
821	Forward Cargo Door												
<u>Number</u>	<u>Name/Location</u>												
821	Forward Cargo Door												
———— END OF TASK ————													
EFFECTIVITY AKS ALL		SOURCE AWL	FORWARD CARGO DOOR SURROUND, FORWARD AND AFT EDGE FRAMES AT STA 440 AND 492.4										
			D633A109-AKS 53-646-20-01										

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND			BOEING CARD NO. 53-646-30-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 821			ZONE 122 124
		NOTE			

Inspect (High Frequency Eddy Current) the forward and aft edge frame inner chords between stringers S-18R and S-26R.

See Doc D626A001-DTR, DTR check form 53-30-08-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-90.

ACCESS NOTE: Perform inspection with door open. Remove sealer at door stops as required.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND
		D633A109-AKS 53-646-30-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-30-01
				MECH INSP
TASK 53-05-02-250-834				
1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND				
A. Inspection				
SUBTASK 53-05-02-010-049				
(1) Open this access panel:				
Number Name/Location				
821 Forward Cargo Door				
NOTE: Perform inspection with door open. Remove sealer at door stops as required.				
SUBTASK 53-05-02-250-034				
(2) Do a High Frequency Eddy Current inspection of the forward and aft edge frame inner chords between stringers S-18R and S-26R.				
See Doc D626A001-DTR, DTR check form 53-30-08-3 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-90.				
SUBTASK 53-05-02-410-047				
(3) Close this access panel:				
Number Name/Location				
821 Forward Cargo Door				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND
		D633A109-AKS 53-646-30-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-646-40-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 821			ZONE 122
		NOTE			

Inspect (High Frequency Eddy Current) the forward and aft edge frame inner chords between stringers S-17R and S-18R.

See Doc D626A001-DTR, DTR check form 53-30-08-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-87.

ACCESS NOTE: Remove cargo liners as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-40-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-40-01
				MECH INSP
TASK 53-05-02-250-835				
1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-050				
(1) Open this access panel:				
Number Name/Location				
821 Forward Cargo Door				
NOTE: Remove cargo liners as required to perform the inspection.				
SUBTASK 53-05-02-250-035				
(2) Do a High Frequency Eddy Current inspection of the forward and aft edge frame inner chords between stringers S-17R and S-18R.				
See Doc D626A001-DTR, DTR check form 53-30-08-4 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-87.				
SUBTASK 53-05-02-410-048				
(3) Close this access panel:				
Number Name/Location				
821 Forward Cargo Door				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE		
		D633A109-AKS 53-646-40-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-646-50-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 821			ZONE 122	
		NOTE				

Inspect (Low Frequency Eddy Current) the forward and aft edge frame inner chords at stringer S-18R.

See Doc D626A001-DTR, DTR check form 53-30-08-5 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-80.

ACCESS NOTE: Remove cargo liners as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-50-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-50-01
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TASK 53-05-02-250-836

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-051

- (1) Open this access door:

Number Name/Location

821 Forward Cargo Door

NOTE: Remove cargo liners as required to perform the inspection.

SUBTASK 53-05-02-250-036

- (2) Do a Low Frequency Eddy Current inspection of the forward and aft edge frame inner chords at stringer S-18R.

See Doc D626A001-DTR, DTR check form 53-30-08-5 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-80.

SUBTASK 53-05-02-410-049

- (3) Close this access door:

Number Name/Location

821 Forward Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-50-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-646-60-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 821		ZONE 122 124		

Inspect (High Frequency Eddy Current) the bearstrap for two inches on each side of stringer S-24R at STA 440 and STA 492.4.

See Doc D626A001-DTR, DTR check form 53-30-08-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-91.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-60-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-60-01
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TASK 53-05-02-250-837

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-052

- (1) Open this access door:

Number Name/Location

821 Forward Cargo Door

SUBTASK 53-05-02-250-037

- (2) Do a High Frequency Eddy Current inspection of the bearstrap for two inches on each side of stringer S-24R at STA 440 and STA 492.4.

See Doc D626A001-DTR, DTR check form 53-30-08-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-91.

SUBTASK 53-05-02-410-050

- (3) Close this access door:

Number Name/Location

821 Forward Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-60-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-646-61-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 821			

Inspect (Ultrasonic) the bearstrap for hidden damage under the stop backup fitting at stringer S-24R at STA 440 and STA 492.4.

See Doc D626A001-DTR, DTR check form 53-30-08-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 53-10-07.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-61-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-61-01				
TASK 53-05-02-130-802				MECH INSP				
1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE								
A. Inspection								
SUBTASK 53-05-02-010-004								
(1) Open this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>821</td><td>Forward Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	821	Forward Cargo Door
<u>Number</u>	<u>Name/Location</u>							
821	Forward Cargo Door							
SUBTASK 53-05-02-130-002								
(2) Do an Ultrasonic inspection of the bearstrap for hidden damage under the stop backup fitting at stringer S-24R at Sta 440 and Sta 492.4.								
See Doc D626A001-DTR, DTR check form 53-30-08-6 for alternative inspections.								
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 53-10-07.								
SUBTASK 53-05-02-410-002								
(3) Close this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>821</td><td>Forward Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	821	Forward Cargo Door
<u>Number</u>	<u>Name/Location</u>							
821	Forward Cargo Door							
<hr style="width: 20%; margin-left: auto; margin-right: 0;"/> END OF TASK <hr style="width: 20%; margin-left: 0; margin-right: auto;"/>								

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-61-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK DETAILED				53-646-62-01
TAIL NUMBER	WORK AREA FWD CARGO	VERSION 1.1	THRESHOLD 34000 FC	REPEAT 18000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS 821 NOTE			ZONE 122 124

Inspect (Detailed) the skin around the entire edge of the scuff plates at all four corners (upper/lower/fwd/aft) of the cargo door. (PSE 53-30-08-9).

See Doc. D626A001-DTR, DTR check form 53-60-08-9, for alternative inspections.

ACCESS NOTE: Forward cargo door must be open to perform this inspection. Scuff plate removal required.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-62-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-62-01
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TASK 53-05-02-211-865

MECH

INSP

1. EXTERNAL - DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-121

- (1) Open this access panel:

Number Name/Location

821 Forward Cargo Door

NOTE: Forward cargo door must be open to perform this inspection. Scuff plate removal required.

SUBTASK 53-05-02-211-065

- (2) Do a Detailed inspection of the skin around the entire edge of the scuff plates at all four corners (upper/lower/fwd/aft) of the cargo door. (PSE 53-30-08-9).

See Doc. D626A001-DTR, DTR check form 53-60-08-9, for alternative inspections.

SUBTASK 53-05-02-410-119

- (3) Close this access panel:

Number Name/Location

821 Forward Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-62-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-646-63-01 RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO	VERSION 1.1	THRESHOLD 34000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 821			
		NOTE			
					ZONE 122 124

Inspect (General Visual) the bearstrap at all four corners (upper/lower/fwd/aft) of the cargo door cutout. (PSE 53-30-08-9).

See Doc. D626A001-DTR, DTR check form 53-60-08-9, for alternative inspections.

ACCESS NOTE: Corner casting removal is required.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-63-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-63-01
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TASK 53-05-02-210-808

MECH

INSP

1. INTERNAL - GENERAL VISUAL: FORWARD CARGO DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-013

- (1) Open this access panel:

Number Name/Location

821 Forward Cargo Door

NOTE: Corner casting removal is required.

SUBTASK 53-05-02-210-008

- (2) Do a General Visual inspection of the bearstrap at all four corners (upper/lower/fwd/aft) of the cargo door cutout. (PSE 53-30-08-9).

See Doc. D626A001-DTR, DTR check form 53-60-08-9, for alternative inspections.

SUBTASK 53-05-02-410-011

- (3) Close this access panel:

Number Name/Location

821 Forward Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-63-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-646-70-01
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	RELATED CARD
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 821			

Inspect (High Frequency Eddy Current) the outer chord around the fasteners common to the chord and bearstrap.

See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-48.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-70-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-70-01
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TASK 53-05-02-250-838

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-053

- (1) Open this access door:

Number Name/Location

821 Forward Cargo Door

SUBTASK 53-05-02-250-038

- (2) Do a High Frequency Eddy Current inspection of the outer chord around the fasteners common to the chord and bearstrap.

See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-48.

SUBTASK 53-05-02-410-051

- (3) Close this access door:

Number Name/Location

821 Forward Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-70-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO.		
DATE	TASK SPECIAL DETAILED				53-646-71-01	RELATED CARD	
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ALL		ENGINE ALL
STATION	SKILL AIRPL						
		ACCESS 821			ZONE 122 124		

Inspect (High Frequency Eddy Current) the bearstrap along the upper edge of the forward cargo door.

See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-48.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-71-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-71-01
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TASK 53-05-02-250-839

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-054

- (1) Open this access door:

Number Name/Location

821 Forward Cargo Door

SUBTASK 53-05-02-250-039

- (2) Do a High Frequency Eddy Current inspection of the bearstrap along the upper edge of the forward cargo door.

See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-48.

SUBTASK 53-05-02-410-052

- (3) Close this access door:

Number Name/Location

821 Forward Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-71-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-646-75-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 6000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 800 900 900ER ENGINE ALL
		ACCESS 821			ZONE 122 124

Inspect (Detailed) the upper sill inner chord. (PSE 53-30-08-11).

See Doc. D626A001-DTR, DTR check form 53-60-08-11, for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-75-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-75-01
TASK 53-05-02-211-806				MECH INSP
1. INTERNAL - DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-120				
(1) Open this access panel:				
Number Name/Location				
821 Forward Cargo Door				
SUBTASK 53-05-02-211-006				
(2) Do a Detailed inspection of the upper sill inner chord. (PSE 53-30-08-11). See Doc. D626A001-DTR, DTR check form 53-60-08-11, for alternative inspections.				
SUBTASK 53-05-02-410-118				
(3) Close this access panel:				
Number Name/Location				
821 Forward Cargo Door				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE		
		D633A109-AKS 53-646-75-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-646-80-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 821			ZONE 122 124
		NOTE			

Inspect (Low Frequency Eddy Current) around the fasteners common to the web at the lower main sill chords between STA 421 and 438 (for -600) and STA 461 and STA 478 (for -700/-800).

See Doc. D626A001-DTR, DTR check form 53-30-08-12, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-33.

ACCESS NOTE: Removal of forward cargo door scuff plate is required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-80-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-646-80-01									
					MECH	INSP							
TASK 53-05-02-250-840													
1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR SURROUND STRUCTURE													
<p>A. Inspection</p> <p>SUBTASK 53-05-02-010-055</p> <p>(1) Open this access panel:</p> <table style="margin-left: 20px; border-collapse: collapse;"> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> <tr> <td style="padding-left: 10px;">821</td> <td style="padding-left: 10px;">Forward Cargo Door</td> </tr> </table> <p><u>NOTE:</u> Removal of forward cargo door scuff plate is required to perform this inspection.</p> <p>SUBTASK 53-05-02-250-040</p> <p>(2) Do a Low Frequency Eddy Current inspection around the fasteners common to the web at the lower main sill chords between STA 421 and 438 (for -600) and STA 461 and STA 478 (for -700/-800).</p> <p>See Doc. D626A001-DTR, DTR check form 53-30-08-12, for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-33.</p> <p>SUBTASK 53-05-02-410-053</p> <p>(3) Close this access panel:</p> <table style="margin-left: 20px; border-collapse: collapse;"> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> <tr> <td style="padding-left: 10px;">821</td> <td style="padding-left: 10px;">Forward Cargo Door</td> </tr> </table>						<u>Number</u>	<u>Name/Location</u>	821	Forward Cargo Door	<u>Number</u>	<u>Name/Location</u>	821	Forward Cargo Door
<u>Number</u>	<u>Name/Location</u>												
821	Forward Cargo Door												
<u>Number</u>	<u>Name/Location</u>												
821	Forward Cargo Door												
———— END OF TASK ————													

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-646-80-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR FITTINGS AND STOPS AT THE FORWARD AND AFT EDGE FRAMES			BOEING CARD NO. 53-647-00-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 821				ZONE 122
		NOTE				

Inspect (High Frequency Eddy Current) the intercostal web for cracks adjacent to rivets and fastener holes (five locations at the forward and aft edge frames) common to the backup fitting and intercostal.

See Doc D626A001-DTR, DTR check form 53-30-09-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-02.

ACCESS NOTE: Remove cargo liners as required.

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR FITTINGS AND STOPS AT THE FORWARD AND AFT EDGE FRAMES
		D633A109-AKS 53-647-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-647-00-01
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TASK 53-05-02-250-841

1. INTERNAL - SPECIAL DETAILED: FORWARD CARGO DOOR FITTINGS AND STOPS AT THE FORWARD AND AFT EDGE FRAMES

A. Inspection

SUBTASK 53-05-02-010-056

- (1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
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821 Forward Cargo Door

NOTE: Remove cargo liners as required

SUBTASK 53-05-02-250-041

- (2) Do a High Frequency Eddy Current inspection of the intercostal web for cracks adjacent to rivets and fastener holes (five locations at the forward and aft edge frames) common to the backup fitting and intercostal.

See Doc D626A001-DTR, DTR check form 53-30-09-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-02.

SUBTASK 53-05-02-410-054

- (3) Close this access panel:

Number Name/Location

821 Forward Cargo Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	FORWARD CARGO DOOR FITTINGS AND STOPS AT THE FORWARD AND AFT EDGE FRAMES
		D633A109-AKS 53-647-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WING TO BODY LOWER DRAG ANGLE			BOEING CARD NO. 53-648-00-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 191AL 191FL			ZONE 125 131 191	
		NOTE				

Inspect (High Frequency Eddy Current) around all fasteners in the angle (9 inches forward and 12 inches aft) of the wing to body intersection (STA 536) between STA 518 to STA 555, above stringer 24.

See Doc D626A001-DTR, DTR check form 53-30-11-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings is required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WING TO BODY LOWER DRAG ANGLE	
		D633A109-AKS 53-648-00-01	Page 1 of 2 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-648-00-01
TASK 53-05-02-250-842				MECH INSP

1. INTERNAL - SPECIAL DETAILED: WING TO BODY LOWER DRAG ANGLE

A. Inspection

SUBTASK 53-05-02-010-057

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

SUBTASK 53-05-02-010-058

(2) Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

NOTE: Removal of wing to body fairings is required.

SUBTASK 53-05-02-250-042

(3) Do a High Frequency Eddy Current inspection around all fasteners in the angle (9 inches forward and 12 inches aft) of the wing to body intersection (STA 536) between STA 518 to STA 555, above stringer 24.

See Doc D626A001-DTR, DTR check form 53-30-11-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-47.

SUBTASK 53-05-02-410-055

(4) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

SUBTASK 53-05-02-410-056

(5) Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WING TO BODY LOWER DRAG ANGLE	
		D633A109-AKS 53-648-00-01	Page 2 of 2 Jun 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WING TO BODY LOWER DRAG ANGLE			BOEING CARD NO. 53-648-00-02	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FWD CARGO DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 191AR 191FR	NOTE		ZONE 126 132 191	

Inspect (High Frequency Eddy Current) around all fasteners in the angle (9 inches forward and 12 inches aft) of the wing to body intersection (STA 536) between STA 518 to STA 555, above stringer 24.

See Doc D626A001-DTR, DTR check form 53-30-11-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WING TO BODY LOWER DRAG ANGLE	
		D633A109-AKS 53-648-00-02	Page 1 of 2 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-648-00-02
TASK 53-05-02-250-842				MECH INSP

1. INTERNAL - SPECIAL DETAILED: WING TO BODY LOWER DRAG ANGLE

A. Inspection

SUBTASK 53-05-02-010-057

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

SUBTASK 53-05-02-010-058

(2) Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

NOTE: Removal of wing to body fairings is required.

SUBTASK 53-05-02-250-042

(3) Do a High Frequency Eddy Current inspection around all fasteners in the angle (9 inches forward and 12 inches aft) of the wing to body intersection (STA 536) between STA 518 to STA 555, above stringer 24.

See Doc D626A001-DTR, DTR check form 53-30-11-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-47.

SUBTASK 53-05-02-410-055

(4) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

SUBTASK 53-05-02-410-056

(5) Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WING TO BODY LOWER DRAG ANGLE
		D633A109-AKS 53-648-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WING TO BODY UPPER DRAG ANGLE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-649-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 191AL 191FL			ZONE 125 131 191
		NOTE			

Inspect (High Frequency Eddy Current) eight inches FWD and AFT of STA 536 along the edge of the inboard angle adjacent to the fuselage (FWD of STA 536) and the lower wing skin (AFT of STA 536).

See Doc D626A001-DTR, DTR check form 53-30-11-02 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings, duct located aft of STA 536 and sealant along edge of angle to body contour and lower wing skin is required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WING TO BODY UPPER DRAG ANGLE
		D633A109-AKS 53-649-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-649-00-01
TASK 53-05-02-250-843				MECH INSP

1. INTERNAL - SPECIAL DETAILED: WING TO BODY UPPER DRAG ANGLE

A. Inspection

SUBTASK 53-05-02-010-059

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

NOTE: Removal of wing to body fairings, duct located aft of STA 536 and sealant along edge of angle to body contour and lower wing skin is required.

SUBTASK 53-05-02-250-043

(2) Do a High Frequency Eddy Current inspection, eight inches FWD and AFT of STA 536 along the edge of the inboard angle adjacent to the fuselage (FWD of STA 536) and the lower wing skin (AFT of STA 536).

See Doc D626A001-DTR, DTR check form 53-30-11-02 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

SUBTASK 53-05-02-410-058

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WING TO BODY UPPER DRAG ANGLE	
		D633A109-AKS 53-649-00-01	Page 2 of 2 Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WING TO BODY UPPER DRAG ANGLE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-649-00-02 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 191AR 191FR			ZONE 126 132 191
		NOTE			

Inspect (High Frequency Eddy Current) eight inches FWD and AFT of STA 536 along the edge of the inboard angle adjacent to the fuselage (FWD of STA 536) and the lower wing skin (AFT of STA 536).

See Doc D626A001-DTR, DTR check form 53-30-11-02 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings, duct located aft of STA 536 and sealant along edge of angle to body contour and lower wing skin is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WING TO BODY UPPER DRAG ANGLE
		D633A109-AKS 53-649-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-649-00-02
TASK 53-05-02-250-843				MECH INSP

1. INTERNAL - SPECIAL DETAILED: WING TO BODY UPPER DRAG ANGLE

A. Inspection

SUBTASK 53-05-02-010-059

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

NOTE: Removal of wing to body fairings, duct located aft of STA 536 and sealant along edge of angle to body contour and lower wing skin is required.

SUBTASK 53-05-02-250-043

(2) Do a High Frequency Eddy Current inspection, eight inches FWD and AFT of STA 536 along the edge of the inboard angle adjacent to the fuselage (FWD of STA 536) and the lower wing skin (AFT of STA 536).

See Doc D626A001-DTR, DTR check form 53-30-11-02 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

SUBTASK 53-05-02-410-058

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WING TO BODY UPPER DRAG ANGLE
		D633A109-AKS 53-649-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WING TO BODY UPPER DRAG ANGLE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-649-01-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 191AL 191FL			ZONE 125 131 191
		NOTE			

Inspect (High Frequency Eddy Current) eight inches FWD and AFT of STA 536 along the inboard angle at the angle to fuselage interface (FWD of STA 536) and the angle to lower wing skin interface (AFT of STA 536).

See Doc D626A001-DTR, DTR check form 53-30-11-02 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings, duct located aft of STA 536 and sealant along edge of angle to body contour and lower wing skin is required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WING TO BODY UPPER DRAG ANGLE
		D633A109-AKS 53-649-01-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-649-01-01
				MECH INSP
TASK 53-05-02-250-844				
1. INTERNAL - SPECIAL DETAILED: WING TO BODY UPPER DRAG ANGLE				
A. Inspection				
SUBTASK 53-05-02-010-061				
(1) Open these access panels on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
191FL Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet				
Open these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
191FR Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet				
NOTE: Removal of wing to body fairings, duct located aft of STA 536 and sealant along edge of angle to body contour and lower wing skin is required.				
SUBTASK 53-05-02-250-044				
(2) Do a High Frequency Eddy Current inspection, eight inches FWD and AFT of STA 536 along the inboard angle at the angle to fuselage interface (FWD of STA 536) and the angle to lower wing skin interface (AFT of STA 536).				
See Doc D626A001-DTR, DTR check form 53-30-11-02 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.				
SUBTASK 53-05-02-410-059				
(3) Close these access panels on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
191FL Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet				
Close these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
191FR Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WING TO BODY UPPER DRAG ANGLE	
		D633A109-AKS 53-649-01-01	Page 2 of 2 Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WING TO BODY UPPER DRAG ANGLE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-649-01-02 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 191AR 191FR			ZONE 126 132 191
		NOTE			

Inspect (High Frequency Eddy Current) eight inches FWD and AFT of STA 536 along the inboard angle at the angle to fuselage interface (FWD of STA 536) and the angle to lower wing skin interface (AFT of STA 536).

See Doc D626A001-DTR, DTR check form 53-30-11-02 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings, duct located aft of STA 536 and sealant along edge of angle to body contour and lower wing skin is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WING TO BODY UPPER DRAG ANGLE
		D633A109-AKS 53-649-01-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-649-01-02
TASK 53-05-02-250-844				MECH INSP

1. INTERNAL - SPECIAL DETAILED: WING TO BODY UPPER DRAG ANGLE

A. Inspection

SUBTASK 53-05-02-010-061

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

NOTE: Removal of wing to body fairings, duct located aft of STA 536 and sealant along edge of angle to body contour and lower wing skin is required.

SUBTASK 53-05-02-250-044

(2) Do a High Frequency Eddy Current inspection, eight inches FWD and AFT of STA 536 along the inboard angle at the angle to fuselage interface (FWD of STA 536) and the angle to lower wing skin interface (AFT of STA 536).

See Doc D626A001-DTR, DTR check form 53-30-11-02 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

SUBTASK 53-05-02-410-059

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WING TO BODY UPPER DRAG ANGLE
		D633A109-AKS 53-649-01-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WING TO BODY UPPER DRAG ANGLE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-650-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 191AL 191FL			ZONE 125 131 191
		NOTE			

Inspect (High Frequency Eddy Current) around the fasteners in the inboard and outboard angles eight inches forward and aft of STA 536.

See Doc. D626A001-DTR, DTR check form 53-30-11-03, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings and duct located aft of STA 536 is required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WING TO BODY UPPER DRAG ANGLE
		D633A109-AKS 53-650-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-650-00-01
				MECH INSP
TASK 53-05-02-250-845				
1. INTERNAL - SPECIAL DETAILED: WING TO BODY UPPER DRAG ANGLE				
A. Inspection				
SUBTASK 53-05-02-010-064				
(1) Open these access panels on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
191FL Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet				
Open these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
191FR Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet				
NOTE: Removal of wing to body fairings and duct located aft of STA 536 is required.				
SUBTASK 53-05-02-250-045				
(2) Do a High Frequency Eddy Current inspection around the fasteners in the inboard and outboard angles eight inches forward and aft of STA 536.				
See Doc. D626A001-DTR, DTR check form 53-30-11-03, for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.				
SUBTASK 53-05-02-410-061				
(3) Close these access panels on the Left side:				
Number Name/Location				
191AL Forward Wing To Body Fairing Panel - Upper				
191FL Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet				
SUBTASK 53-05-02-410-062				
(4) Close these access panels on the Right side:				
Number Name/Location				
191AR Forward Wing To Body Fairing Panel - Upper				
191FR Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WING TO BODY UPPER DRAG ANGLE	
		D633A109-AKS 53-650-00-01	Page 2 of 2 Oct 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WING TO BODY UPPER DRAG ANGLE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-650-00-02 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 191AR 191FR			ZONE 126 132 191
		NOTE			

Inspect (High Frequency Eddy Current) around the fasteners in the inboard and outboard angles eight inches forward and aft of STA 536.

See Doc. D626A001-DTR, DTR check form 53-30-11-03, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

ACCESS NOTE: Removal of wing to body fairings and duct located aft of STA 536 is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WING TO BODY UPPER DRAG ANGLE
		D633A109-AKS 53-650-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-650-00-02
TASK 53-05-02-250-845				MECH INSP

1. INTERNAL - SPECIAL DETAILED: WING TO BODY UPPER DRAG ANGLE

A. Inspection

SUBTASK 53-05-02-010-064

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

NOTE: Removal of wing to body fairings and duct located aft of STA 536 is required.

SUBTASK 53-05-02-250-045

(2) Do a High Frequency Eddy Current inspection around the fasteners in the inboard and outboard angles eight inches forward and aft of STA 536.

See Doc. D626A001-DTR, DTR check form 53-30-11-03, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-47.

SUBTASK 53-05-02-410-061

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
191AL	Forward Wing To Body Fairing Panel - Upper
191FL	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

SUBTASK 53-05-02-410-062

(4) Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
191AR	Forward Wing To Body Fairing Panel - Upper
191FR	Forward Wing To Body Fairing Panel - Mid Fairing, Above Ram Air Inlet

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WING TO BODY UPPER DRAG ANGLE
		D633A109-AKS 53-650-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CROWN SKIN PANEL - STA 540 TO 727			BOEING CARD NO. 53-652-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 231 232 241 242	

Inspect (Detailed) the skin around all of the fastener locations from stringer S-10L to S-10R, from STA 540 to STA 727, except at the lap splices and antennas. (53-40-01-1).

See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL - STA 540 TO 727
		D633A109-AKS 53-652-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-652-00-01
TASK 53-05-02-211-866				MECH INSP
1. EXTERNAL - DETAILED: CROWN SKIN PANEL STA 540 TO STA 727				
A. Inspection				
SUBTASK 53-05-02-211-066				
(1) Do a Detailed inspection of the skin around all of the fastener locations from stringer S-10L to S-10R, from STA 540 to STA 727, except at the lap splices and antennas. (53-40-01-1).				
See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL				
SOURCE AWL				
CROWN SKIN PANEL - STA 540 TO 727				
D633A109-AKS 53-652-00-01				
Page 2 of 2 Feb 15/2015				

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-653-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 231 241

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 540 to STA 727. (PSE 53-40-03-1).

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-653-00-01	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-653-00-01	
					MECH INSP
TASK 53-05-02-250-935					
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE					
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-135</p> <p>(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 540 to STA 727. (PSE 53-40-03-1). See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.</p> <p style="text-align: center;">— END OF TASK —</p>					
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT LONGITUDINAL LAP SPLICE</p> <p>D633A109-AKS 53-653-00-01</p>					

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-653-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 232 242

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 540 to STA 727. (PSE 53-40-03-1).

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-653-00-02	Page 1 of 2 Oct 15/2014
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-653-00-02
TASK 53-05-02-250-935				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-135				
(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 540 to STA 727. (PSE 53-40-03-1). See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-653-00-02	Page 2 of 2 Feb 15/2015	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-654-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 231 241

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 540 to STA 727. (53-40-03-2).

See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-654-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-654-00-01	MECH	INSP
TASK 53-05-02-250-937						
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE						
A. Inspection						
SUBTASK 53-05-02-250-137						
(1)	Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 540 to STA 727. (53-40-03-2). See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.					
<hr style="width: 20%; margin-left: auto; margin-right: 0;"/> END OF TASK <hr style="width: 20%; margin-left: 0; margin-right: auto;"/>						

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC	
		D633A109-AKS 53-654-00-01	Page 2 of 2 Jun 15/2015

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-654-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 232 242

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 540 to STA 727. (53-40-03-2).

See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-654-00-02	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-654-00-02
				MECH INSP
TASK 53-05-02-250-937				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-137				
(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 540 to STA 727. (53-40-03-2). See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-654-00-02	Page 2 of 2 Jun 15/2015	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-655-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 231 241

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 540 to STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-03-3, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-655-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-655-00-01	
					MECH INSP
TASK 53-05-02-250-847					
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL SKIN SPLICE					
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-047</p> <p>(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 540 to STA 727.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-03-3, for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-41.</p> <p style="text-align: center;">— END OF TASK —</p>					
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT LONGITUDINAL LAP SPLICE</p> <p>D633A109-AKS 53-655-00-01</p>					

AKS



737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-655-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 232 242

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 540 to STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-03-3, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-655-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-655-00-02
TASK 53-05-02-250-847				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL SKIN SPLICE				
A. Inspection				
SUBTASK 53-05-02-250-047				
(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 540 to STA 727. See Doc. D626A001-DTR, DTR check form 53-40-03-3, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-41.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-655-00-02	Page 2 of 2 Jun 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WINDOW BELT, STA 540 TO 727			BOEING CARD NO.
DATE	TASK DETAILED				53-656-00-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 231 241

Inspect (Detailed) stringers S-11 and S-13 from STA 540 to 727.

See Doc. D626A001-DTR, DTR check form 53-40-04-1, for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls and insulation blankets is required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT, STA 540 TO 727
		D633A109-AKS 53-656-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-656-00-01
TASK 53-05-02-211-822				MECH INSP
1. INTERNAL - DETAILED: WINDOW BELT, STA 540 TO 727				
A. Inspection				
<p><u>NOTE:</u> Removal and/or displacement of passenger cabin sidewalls and insulation blankets is required.</p> <p>SUBTASK 53-05-02-211-022</p> <p>(1) Do a Detailed inspection of the stringers S-11 and S-13 from STA 540 to 727. See Doc. D626A001-DTR, DTR check form 53-40-04-1, for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT, STA 540 TO 727		
		D633A109-AKS 53-656-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT WINDOW BELT, STA 540 TO 727			BOEING CARD NO.
DATE	TASK DETAILED				53-656-00-02 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 232 242

Inspect (Detailed) stringers S-11 and S-13 from STA 540 to 727.

See Doc. D626A001-DTR, DTR check form 53-40-04-1, for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls and insulation blankets is required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT, STA 540 TO 727
		D633A109-AKS 53-656-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-656-00-02
TASK 53-05-02-211-822				MECH INSP
1. INTERNAL - DETAILED: WINDOW BELT, STA 540 TO 727				
A. Inspection				
<p><u>NOTE:</u> Removal and/or displacement of passenger cabin sidewalls and insulation blankets is required.</p> <p>SUBTASK 53-05-02-211-022</p> <p>(1) Do a Detailed inspection of the stringers S-11 and S-13 from STA 540 to 727. See Doc. D626A001-DTR, DTR check form 53-40-04-1, for alternative inspections.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT, STA 540 TO 727		
		D633A109-AKS 53-656-00-02	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT WINDOW BELT, STA 540 TO 727			BOEING CARD NO. 53-657-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 231 241	

Inspect (Detailed) the skin from stringers S-11 to S-13 between the windows from STA 540 to STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-04-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT, STA 540 TO 727
		D633A109-AKS 53-657-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-657-00-01
TASK 53-05-02-211-823				MECH INSP
1. EXTERNAL - DETAILED: WINDOW BELT, STA 540 TO 727				
A. Inspection				
SUBTASK 53-05-02-211-023				
(1) Do a Detailed inspection of the skin from stringers S-11 to S-13 between the windows from STA 540 to STA 727.				
See Doc. D626A001-DTR, DTR check form 53-40-04-2 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT, STA 540 TO 727		
		D633A109-AKS 53-657-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WINDOW BELT, STA 540 TO 727			BOEING CARD NO. 53-657-00-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS		ZONE 232 242		

Inspect (Detailed) the skin from stringers S-11 to S-13 between the windows from STA 540 to STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-04-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT, STA 540 TO 727
		D633A109-AKS 53-657-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-657-00-02
TASK 53-05-02-211-823				MECH INSP
1. EXTERNAL - DETAILED: WINDOW BELT, STA 540 TO 727				
A. Inspection				
SUBTASK 53-05-02-211-023				
(1) Do a Detailed inspection of the skin from stringers S-11 to S-13 between the windows from STA 540 to STA 727.				
See Doc. D626A001-DTR, DTR check form 53-40-04-2 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT, STA 540 TO 727		
		D633A109-AKS 53-657-00-02	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT STRINGER SPLICE, STA 663			BOEING CARD NO.
DATE	TASK DETAILED				53-658-00-01
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 231 241

Inspect (Detailed) stringers S-11 to S-13 at a distance of 10 inches forward and aft of STA 663.

See Doc. D626A001-DTR, DTR check form 53-40-07-2 for alternative inspections.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall panels and insulation blankets.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT STRINGER SPLICE, STA 663
		D633A109-AKS 53-658-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-658-00-01
TASK 53-05-02-211-824				MECH INSP
1. INTERNAL - DETAILED: STRINGER SPLICE, STA 663				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewall panels and insulation blankets.</p> <p>SUBTASK 53-05-02-211-024</p> <p>(1) Do a Detailed inspection of the stringers S-11 to S-13 at a distance of 10 inches forward and aft of STA 663.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-07-2 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT STRINGER SPLICE, STA 663		
		D633A109-AKS 53-658-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT STRINGER SPLICE, STA 663			BOEING CARD NO. 53-658-00-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE		ZONE 232 242		

Inspect (Detailed) stringers S-11 to S-13 at a distance of 10 inches forward and aft of STA 663.

See Doc. D626A001-DTR, DTR check form 53-40-07-2 for alternative inspections.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall panels and insulation blankets.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT STRINGER SPLICE, STA 663	
		D633A109-AKS 53-658-00-02	Page 1 of 2 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-658-00-02
TASK 53-05-02-211-824				MECH INSP
1. INTERNAL - DETAILED: STRINGER SPLICE, STA 663				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewall panels and insulation blankets.</p> <p>SUBTASK 53-05-02-211-024</p> <p>(1) Do a Detailed inspection of the stringers S-11 to S-13 at a distance of 10 inches forward and aft of STA 663.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-07-2 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT STRINGER SPLICE, STA 663		
		D633A109-AKS 53-658-00-02	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT KEEL BEAM CHORDS, STIFFENERS AND SPLICE			BOEING CARD NO.
DATE	TASK DETAILED				53-659-00-01
TAIL NUMBER	WORK AREA KEEL BEAM	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS 192CL			ZONE 133 139 192

Inspect (Detailed) the keel beam side panel webs from STA 540 to STA 663.

See Doc. D626A001-DTR, DTR check form 53-40-08-1 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT KEEL BEAM CHORDS, STIFFENERS AND SPLICE
		D633A109-AKS 53-659-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-659-00-01
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TASK 53-05-02-211-825

MECH

INSP

1. INTERNAL - DETAILED: KEEL BEAM CHORDS, STIFFENERS AND SPLICE**A. Inspection**

SUBTASK 53-05-02-010-019

- (1) Open this access panel on the Left side:

Number Name/Location

192CL ECS Access Door

Open this access panel on the Right side:

Number Name/Location

192CR ECS Access Door

SUBTASK 53-05-02-211-025

- (2) Do a Detailed inspection of the keel beam side panel webs from STA 540 to STA 663.

See Doc. D626A001-DTR, DTR check form 53-40-08-1 for alternative inspections.

SUBTASK 53-05-02-410-017

- (3) Close this access panel on the Left side:

Number Name/Location

192CL ECS Access Door

Close this access panel on the Right side:

Number Name/Location

192CR ECS Access Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT KEEL BEAM CHORDS, STIFFENERS AND SPLICE
		D633A109-AKS 53-659-00-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT KEEL BEAM CHORDS, STIFFENERS AND SPLICE			BOEING CARD NO. 53-659-00-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA KEEL BEAM	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 192CR		ZONE 134 139 193		

Inspect (Detailed) the keel beam side panel webs from STA 540 to STA 663.

See Doc. D626A001-DTR, DTR check form 53-40-08-1 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT KEEL BEAM CHORDS, STIFFENERS AND SPLICE
		D633A109-AKS 53-659-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-659-00-02
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TASK 53-05-02-211-825

MECH

INSP

1. INTERNAL - DETAILED: KEEL BEAM CHORDS, STIFFENERS AND SPLICE**A. Inspection**

SUBTASK 53-05-02-010-019

- (1) Open this access panel on the Left side:

Number Name/Location

192CL ECS Access Door

Open this access panel on the Right side:

Number Name/Location

192CR ECS Access Door

SUBTASK 53-05-02-211-025

- (2) Do a Detailed inspection of the keel beam side panel webs from STA 540 to STA 663.

See Doc. D626A001-DTR, DTR check form 53-40-08-1 for alternative inspections.

SUBTASK 53-05-02-410-017

- (3) Close this access panel on the Left side:

Number Name/Location

192CL ECS Access Door

Close this access panel on the Right side:

Number Name/Location

192CR ECS Access Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT KEEL BEAM CHORDS, STIFFENERS AND SPLICE
		D633A109-AKS 53-659-00-02

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT STIFFENER ATTACHMENT TO FLOOR BEAM, STA 727, LBL 45, WL 202.6			BOEING CARD NO. 53-660-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA WHEELWELL	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 133

Inspect (High Frequency Eddy Current) the fastener row connecting the pressure bulkhead stiffener to the stiffener attachment fitting that joins the stiffener to the floor beam (at five locations) around the fastener/collar on the outboard side at LBL and RBL 45 and WL 202.6.

See Doc. D626A001-DTR, DTR check form 53-40-10-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-29.

ACCESS NOTE: Removal and/or displacement of aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT STIFFENER ATTACHMENT TO FLOOR BEAM, STA 727, LBL 45, WL 202.6
		D633A109-AKS 53-660-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-660-00-01
				MECH INSP
TASK 53-05-02-250-848				
1. INTERNAL - SPECIAL DETAILED: STIFFENER ATTACHMENT TO FLOOR BEAM, STA 727				
<p>A. Inspection</p> <p>NOTE: Removal and/or displacement of aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-048</p> <p>(1) Do a High Frequency Eddy Current inspection of the fastener row connecting the pressure bulkhead stiffener to the stiffener attachment fitting that joins the stiffener to the floor beam (at five locations) around the fastener/collar on the outboard side at LBL and RBL 45 and WL 202.6.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-10-1 for alternative inspections.</p> <p style="text-align: center;">END OF TASK</p>				
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT STIFFENER ATTACHMENT TO FLOOR BEAM, STA 727, LBL 45, WL 202.6</p> <p>D633A109-AKS 53-660-00-01</p>				

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT STIFFENER ATTACHMENT TO FLOOR BEAM, STA 727, RBL 45, WL 202.6			BOEING CARD NO. 53-660-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA WHEELWELL	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 134

Inspect (High Frequency Eddy Current) the fastener row connecting the pressure bulkhead stiffener to the stiffener attachment fitting that joins the stiffener to the floor beam (at five locations) around the fastener/collar on the outboard side at LBL and RBL 45 and WL 202.6.

See Doc. D626A001-DTR, DTR check form 53-40-10-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-29.

ACCESS NOTE: Removal and/or displacement of aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT STIFFENER ATTACHMENT TO FLOOR BEAM, STA 727, RBL 45, WL 202.6
		D633A109-AKS 53-660-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-660-00-02
TASK 53-05-02-250-848				MECH INSP
1. INTERNAL - SPECIAL DETAILED: STIFFENER ATTACHMENT TO FLOOR BEAM, STA 727				
A. Inspection				
<p><u>NOTE:</u> Removal and/or displacement of aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-048</p> <p>(1) Do a High Frequency Eddy Current inspection of the fastener row connecting the pressure bulkhead stiffener to the stiffener attachment fitting that joins the stiffener to the floor beam (at five locations) around the fastener/collar on the outboard side at LBL and RBL 45 and WL 202.6.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-10-1 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT STIFFENER ATTACHMENT TO FLOOR BEAM, STA 727, RBL 45, WL 202.6				
D633A109-AKS 53-660-00-02				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT STRINGER 18 STRAP SIDE OF BODY			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-661-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 195CL			ZONE 133 141

Inspect (Low Frequency Eddy Current) the skin under the strap at stringer S-18 between the fasteners common to the strap and skin from STA 717 to STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-11-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-49.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT STRINGER 18 STRAP SIDE OF BODY
		D633A109-AKS 53-661-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-661-00-01
				MECH INSP
TASK 53-05-02-250-849				
1. INTERNAL - SPECIAL DETAILED: STRINGER 18 STRAP SIDE OF BODY				
A. Inspection				
SUBTASK 53-05-02-010-137				
(1) Open this access panel on the Left side:				
Number Name/Location				
195CL Wing To Body Fairing - Left Side				
Open this access panel on the Right side:				
Number Name/Location				
195CR Wing To Body Fairings - Right Side				
SUBTASK 53-05-02-250-049				
(2) Do a Low Frequency Eddy Current inspection of the skin under the strap at stringer S-18 between the fasteners common to the strap and skin from STA 717 to STA 727.				
See Doc. D626A001-DTR, DTR check form 53-40-11-1 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-49.				
SUBTASK 53-05-02-410-127				
(3) Close this access panel on the Left side:				
Number Name/Location				
195CL Wing To Body Fairing - Left Side				
Close this access panel on the Right side:				
Number Name/Location				
195CR Wing To Body Fairings - Right Side				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT STRINGER 18 STRAP SIDE OF BODY	
		D633A109-AKS 53-661-00-01	Page 2 of 2 Jun 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT STRINGER 18 STRAP SIDE OF BODY			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-661-00-02 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 195CR			ZONE 134 142

Inspect (Low Frequency Eddy Current) the skin under the strap at stringer S-18 between the fasteners common to the strap and skin from STA 717 to STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-11-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-49.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT STRINGER 18 STRAP SIDE OF BODY
		D633A109-AKS 53-661-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-661-00-02
TASK 53-05-02-250-849				MECH INSP

1. INTERNAL - SPECIAL DETAILED: STRINGER 18 STRAP SIDE OF BODY

A. Inspection

SUBTASK 53-05-02-010-137

(1) Open this access panel on the Left side:

Number Name/Location

195CL Wing To Body Fairing - Left Side

Open this access panel on the Right side:

Number Name/Location

195CR Wing To Body Fairings - Right Side

SUBTASK 53-05-02-250-049

(2) Do a Low Frequency Eddy Current inspection of the skin under the strap at stringer S-18 between the fasteners common to the strap and skin from STA 717 to STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-11-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-49.

SUBTASK 53-05-02-410-127

(3) Close this access panel on the Left side:

Number Name/Location

195CL Wing To Body Fairing - Left Side

Close this access panel on the Right side:

Number Name/Location

195CR Wing To Body Fairings - Right Side

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT STRINGER 18 STRAP SIDE OF BODY
		D633A109-AKS 53-661-00-02

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT STRINGER 18A CHORD AND LINKS			BOEING CARD NO. 53-662-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L MAIN W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 131 133

Inspect (High Frequency Eddy Current) around the bushings on each lug, three lugs per assembly, on the upper and lower surface at STA 663.

See Doc. D626A001-DTR, DTR check form 53-40-12-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-30.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT STRINGER 18A CHORD AND LINKS
		D633A109-AKS 53-662-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-662-00-01
TASK 53-05-02-250-850				MECH INSP
1. INTERNAL - SPECIAL DETAILED: STRINGER 18A CHORD AND LINKS				
A. Inspection				
SUBTASK 53-05-02-250-050				
(1) Do a High Frequency Eddy Current inspection around the bushings on each lug, three lugs per assembly, on the upper and lower surface at STA 663. See Doc. D626A001-DTR, DTR check form 53-40-12-1 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-30.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT STRINGER 18A CHORD AND LINKS		
		D633A109-AKS 53-662-00-01	Page 2 of 2 Jun 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT STRINGER 18A CHORD AND LINKS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-662-00-02 RELATED CARD
TAIL NUMBER	WORK AREA R MAIN W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS			ZONE 134 142

Inspect (High Frequency Eddy Current) around the bushings on each lug, three lugs per assembly, on the upper and lower surface at STA 663.

See Doc. D626A001-DTR, DTR check form 53-40-12-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-30.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT STRINGER 18A CHORD AND LINKS
		D633A109-AKS 53-662-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-662-00-02
TASK 53-05-02-250-850				MECH INSP
1. INTERNAL - SPECIAL DETAILED: STRINGER 18A CHORD AND LINKS				
A. Inspection				
SUBTASK 53-05-02-250-050				
(1) Do a High Frequency Eddy Current inspection around the bushings on each lug, three lugs per assembly, on the upper and lower surface at STA 663. See Doc. D626A001-DTR, DTR check form 53-40-12-1 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-30.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT STRINGER 18A CHORD AND LINKS		
		D633A109-AKS 53-662-00-02	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT FRAME SPLICE AT STA 540			BOEING CARD NO. 53-663-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 231

Inspect (High Frequency Eddy Current) the frame inner chord at all fasteners common to the inner chord and to the inner splice plate between stringers S-8 and S-9.

See Doc. D626A001-DTR, DTR check form 53-40-14-1 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls and insulation blankets as required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FRAME SPLICE AT STA 540
		D633A109-AKS 53-663-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-663-00-01
TASK 53-05-02-250-851				MECH INSP
1. INTERNAL - SPECIAL DETAILED: FRAME SPLICE AT STA 540				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewalls and insulation blankets as required.</p> <p>SUBTASK 53-05-02-250-051</p> <p>(1) Do a High Frequency Eddy Current inspection of the frame inner chord at all fasteners common to the inner chord and to the inner splice plate between stringers S-8 and S-9. See Doc. D626A001-DTR, DTR check form 53-40-14-1 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT FRAME SPLICE AT STA 540				
D633A109-AKS 53-663-00-01				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT FRAME SPLICE AT STA 540			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-663-00-02 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 232

Inspect (High Frequency Eddy Current) the frame inner chord at all fasteners common to the inner chord and to the inner splice plate between stringers S-8 and S-9.

See Doc. D626A001-DTR, DTR check form 53-40-14-1 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls and insulation blankets as required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT FRAME SPLICE AT STA 540
		D633A109-AKS 53-663-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-663-00-02
TASK 53-05-02-250-851				MECH INSP
1. INTERNAL - SPECIAL DETAILED: FRAME SPLICE AT STA 540				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewalls and insulation blankets as required.</p> <p>SUBTASK 53-05-02-250-051</p> <p>(1) Do a High Frequency Eddy Current inspection of the frame inner chord at all fasteners common to the inner chord and to the inner splice plate between stringers S-8 and S-9. See Doc. D626A001-DTR, DTR check form 53-40-14-1 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT FRAME SPLICE AT STA 540				
D633A109-AKS 53-663-00-02				
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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BS 540 BULKHEAD			BOEING CARD NO. 53-664-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 231 232

Inspect (Detailed) the skin panels at the outer chord from stringers S-9L to S-9R, on each side of splice 540, for cracks at the frame to skin fastener holes.

See Doc. D626A001-DTR, DTR check form 53-40-14-4 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	BS 540 BULKHEAD D633A109-AKS 53-664-00-01	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

EFFECTIVITY AKS ALL	SOURCE AWL	BS 540 BULKHEAD
		D633A109-AKS 53-664-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT BS 663 BULKHEAD			BOEING CARD NO. 53-665-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 231

Inspect (Detailed) the skin on each side of STA 663, from stringers S-8 to S-11, on both the left and right sides.

See Doc. D626A001-DTR, DTR check form 53-40-15-1 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT BS 663 BULKHEAD
		D633A109-AKS 53-665-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-665-00-01	MECH	INSP
TASK 53-05-02-211-827						
1. EXTERNAL - DETAILED: BS 663 BULKHEAD						
A. Inspection						
	SUBTASK 53-05-02-211-027					
(1)	Do a Detailed inspection of the skin on each side of STA 663, from stringers S-8 to S-11, on both the left and right sides.					
	See Doc. D626A001-DTR, DTR check form 53-40-15-1 for alternative inspections.					
	———— END OF TASK ————					

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT BS 663 BULKHEAD
		D633A109-AKS 53-665-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT BS 663 BULKHEAD			BOEING CARD NO. 53-665-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 232

Inspect (Detailed) the skin on each side of STA 663, from stringers S-8 to S-11, on both the left and right sides.

See Doc. D626A001-DTR, DTR check form 53-40-15-1 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT BS 663 BULKHEAD
		D633A109-AKS 53-665-00-02

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-665-00-02	MECH	INSP
TASK 53-05-02-211-827						
1. EXTERNAL - DETAILED: BS 663 BULKHEAD						
A. Inspection						
SUBTASK 53-05-02-211-027						
(1) Do a Detailed inspection of the skin on each side of STA 663, from stringers S-8 to S-11, on both the left and right sides.						
See Doc. D626A001-DTR, DTR check form 53-40-15-1 for alternative inspections.						
———— END OF TASK ——						

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT BS 663 BULKHEAD
		D633A109-AKS 53-665-00-02

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT BS 663 BULKHEAD			BOEING CARD NO. 53-666-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL				
		ACCESS NOTE			
					ZONE 231

Inspect (High Frequency Eddy Current) the bulkhead inner chord from stringers S-10 and S-17 on both the left and right hand sides.

See Doc. D626A001-DTR, DTR check form 53-40-15-3 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls and insulation blankets as required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT BS 663 BULKHEAD
		D633A109-AKS 53-666-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-666-00-01
				MECH INSP
TASK 53-05-02-250-852				
1. INTERNAL - SPECIAL DETAILED: BS 663 BULKHEAD				
<p>A. Inspection</p> <p><u>NOTE:</u> Remove and/or displace passenger cabin sidewalls and insulation blankets as required.</p> <p>SUBTASK 53-05-02-250-052</p> <p>(1) Do a High Frequency Eddy Current inspection of the bulkhead inner chord from stringers S-10 and S-17 on both the left and right hand sides.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-15-3 for alternative inspections.</p> <p style="text-align: center;">— END OF TASK —</p>				
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT BS 663 BULKHEAD</p> <p>D633A109-AKS 53-666-00-01</p>				

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT BS 663 BULKHEAD			BOEING CARD NO. 53-666-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 232
<p>Inspect (High Frequency Eddy Current) the bulkhead inner chord from stringers S-10 and S-17 on both the left and right hand sides.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-15-3 for alternative inspections.</p> <p>ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls and insulation blankets as required.</p>					
EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT BS 663 BULKHEAD D633A109-AKS 53-666-00-02		
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-666-00-02
TASK 53-05-02-250-852				MECH INSP
1. INTERNAL - SPECIAL DETAILED: BS 663 BULKHEAD				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewalls and insulation blankets as required.</p> <p>SUBTASK 53-05-02-250-052</p> <p>(1) Do a High Frequency Eddy Current inspection of the bulkhead inner chord from stringers S-10 and S-17 on both the left and right hand sides.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-15-3 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT BS 663 BULKHEAD				
D633A109-AKS 53-666-00-02				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT AFT WHEEL WELL BULKHEAD STA 727			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-667-00-01 RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 16000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 141 143

Inspect (High Frequency Eddy Current) the fail safe angle from inside the aft cargo bay at frame 727, from stringers S-21L to S-27L and stringers S-21R to S-27R.

See Doc. D626A001-DTR, DTR check form 53-40-16-1a for alternative inspections.

ACCESS NOTE: Removal and/or displacement of aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT AFT WHEEL WELL BULKHEAD STA 727
		D633A109-AKS 53-667-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-667-00-01
TASK 53-05-02-250-854				MECH INSP
1. INTERNAL - SPECIAL DETAILED: AFT WHEEL WELL BULKHEAD, STA 727				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-054</p> <p>(1) Do a High Frequency Eddy Current inspection of the fail safe angle from inside the aft cargo bay at frame 727, from stringers S-21L to S-27L and stringers S-21R to S-27R.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-16-1a for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT AFT WHEEL WELL BULKHEAD STA 727				
D633A109-AKS 53-667-00-01				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT AFT WHEEL WELL BULKHEAD STA 727			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-667-00-02 RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 16000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 142 144

Inspect (High Frequency Eddy Current) the fail safe angle from inside the aft cargo bay at frame 727, from stringers S-21L to S-27L and stringers S-21R to S-27R.

See Doc. D626A001-DTR, DTR check form 53-40-16-1a for alternative inspections.

ACCESS NOTE: Removal and/or displacement of aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT AFT WHEEL WELL BULKHEAD STA 727
		D633A109-AKS 53-667-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-667-00-02
TASK 53-05-02-250-854				MECH INSP
1. INTERNAL - SPECIAL DETAILED: AFT WHEEL WELL BULKHEAD, STA 727				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-054</p> <p>(1) Do a High Frequency Eddy Current inspection of the fail safe angle from inside the aft cargo bay at frame 727, from stringers S-21L to S-27L and stringers S-21R to S-27R.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-16-1a for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT AFT WHEEL WELL BULKHEAD STA 727				
D633A109-AKS 53-667-00-02				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT AFT WHEEL WELL BULKHEAD STA 727			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-667-10-01 RELATED CARD
TAIL NUMBER	WORK AREA L MAIN W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 133 141 143 193

Inspect (High Frequency Eddy Current) the chord at frame 727 from stringers S-21L to S-27L and stringers S-21R to S-27R.

See Doc. D626A001-DTR, DTR check form 53-40-16-1b for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-93.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT AFT WHEEL WELL BULKHEAD STA 727
		D633A109-AKS 53-667-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-667-10-01
				MECH INSP
TASK 53-05-02-250-A98				
1. EXTERNAL - SPECIAL DETAILED: AFT WHEEL WELL BULKHEAD, STA 727				
A. Inspection				
SUBTASK 53-05-02-250-303				
(1) Do a High Frequency Eddy Current inspection of the chord at frame 727 from stringers S-21L to S-27L and stringers S-21R to S-27R.				
See Doc. D626A001-DTR, DTR check form 53-40-16-1b for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-93.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT AFT WHEEL WELL BULKHEAD STA 727		
		D633A109-AKS 53-667-10-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT AFT WHEEL WELL BULKHEAD STA 727			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-667-10-02 RELATED CARD
TAIL NUMBER	WORK AREA R MAIN W/W	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 134 142 144 193

Inspect (High Frequency Eddy Current) the chord at frame 727 from stringers S-21L to S-27L and stringers S-21R to S-27R.

See Doc. D626A001-DTR, DTR check form 53-40-16-1b for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-93.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT AFT WHEEL WELL BULKHEAD STA 727
		D633A109-AKS 53-667-10-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-667-10-02
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TASK 53-05-02-250-A98

MECH

INSP

1. EXTERNAL - SPECIAL DETAILED: AFT WHEEL WELL BULKHEAD, STA 727**A. Inspection**

SUBTASK 53-05-02-250-303

- (1) Do a High Frequency Eddy Current inspection of the chord at frame 727 from stringers S-21L to S-27L and stringers S-21R to S-27R.

See Doc. D626A001-DTR, DTR check form 53-40-16-1b for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-93.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT AFT WHEEL WELL BULKHEAD STA 727
		D633A109-AKS 53-667-10-02

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WHEEL WELL AFT BULKHEAD			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-668-10-01
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD NOTE	REPEAT	RELATED CARD
STATION	SKILL AIRPL	NOTE			APPLICABILITY
		ACCESS NOTE			AIRPLANE ALL ENGINE ALL NOTE
					ZONE 141

Inspect (Ultrasonic) the frame around the six fasteners through the inner chord and web at STA 727 and WL 201.

See Doc. D626A001-DTR, DTR check form 53-40-16-2a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 53-10-09.

INTERVAL NOTE: For 737-600 and -700 LN # 1193 and on - Threshold is 50,000 FC, Repeat is 24,000 FC.

For 737-700C/-700IGW/-800/-900 and -900ER - Threshold is 50,000 FC, Repeat is 24,000 FC.

AIRPLANE NOTE: This DTR Form is effective for all models from L/N 1193 and On.

See interval note for airplane specific threshold and repeat intervals.

ACCESS NOTE: Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WHEEL WELL AFT BULKHEAD
		D633A109-AKS 53-668-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-668-10-01
				MECH INSP
TASK 53-05-02-130-804				
1. INTERNAL - SPECIAL DETAILED: WHEEL WELL AFT BULKHEAD				
A. Inspection				
<p><u>NOTE:</u> Remove or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required.</p> <p>SUBTASK 53-05-02-130-004</p> <p>(1) Do an Ultrasonic inspection of the six fasteners through the inner chord and web at STA 727 and WL 201.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-16-2a for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 53-10-09.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT WHEEL WELL AFT BULKHEAD				
D633A109-AKS 53-668-10-01				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WHEEL WELL AFT BULKHEAD			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-668-10-02
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD NOTE	REPEAT	RELATED CARD
STATION	SKILL AIRPL	NOTE			APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
		ACCESS NOTE			ZONE 142

Inspect (Ultrasonic) the frame around the six fasteners through the inner chord and web at STA 727 and WL 201.

See Doc. D626A001-DTR, DTR check form 53-40-16-2a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 53-10-09.

INTERVAL NOTE: For 737-600 and -700 LN # 1193 and on - Threshold is 50,000 FC, Repeat is 24,000 FC.

For 737-700C/-700IGW/-800/-900 and -900ER - Threshold is 50,000 FC, Repeat is 24,000 FC.

AIRPLANE NOTE: This DTR Form is effective for all models from L/N 1193 and On.

See interval note for airplane specific threshold and repeat intervals.

ACCESS NOTE: Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WHEEL WELL AFT BULKHEAD	
		D633A109-AKS 53-668-10-02	Page 1 of 2 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-668-10-02
TASK 53-05-02-130-804				MECH INSP
1. INTERNAL - SPECIAL DETAILED: WHEEL WELL AFT BULKHEAD				
A. Inspection				
<p><u>NOTE:</u> Remove or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required.</p> <p>SUBTASK 53-05-02-130-004</p> <p>(1) Do an Ultrasonic inspection of the six fasteners through the inner chord and web at STA 727 and WL 201.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-16-2a for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 53-10-09.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WHEEL WELL AFT BULKHEAD		
		D633A109-AKS 53-668-10-02	Page 2 of 2 Jun 15/2015	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT WHEEL WELL AFT BULKHEAD AND PRESSURE WEB STA 727			BOEING CARD NO. 53-669-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL				
		ACCESS NOTE			
					ZONE 141

Inspect (High Frequency Eddy Current) the frame inner chord at STA 727 between stringers S-17 and S-21 on both the left and right hand sides.

See Doc. D626A001-DTR, DTR check form 53-40-16-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-25.

ACCESS NOTE: Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WHEEL WELL AFT BULKHEAD AND PRESSURE WEB STA 727
		D633A109-AKS 53-669-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-669-00-01
				MECH INSP
TASK 53-05-02-250-853				
1. INTERNAL - SPECIAL DETAILED: WHEEL WELL AFT BULKHEAD AND PRESSURE WEB STA 727				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.</p>				
<p>SUBTASK 53-05-02-250-053</p>				
<p>(1) Do a High Frequency Eddy Current inspection of the frame inner chord at STA 727 between stringers S-17 and S-21 on both the left and right hand sides.</p>				
<p>See Doc. D626A001-DTR, DTR check form 53-40-16-3 for alternative inspections.</p>				
<p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-25.</p>				
<hr/> END OF TASK <hr/>				
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT WHEEL WELL AFT BULKHEAD AND PRESSURE WEB STA 727	
			D633A109-AKS 53-669-00-01	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WHEEL WELL AFT BULKHEAD AND PRESSURE WEB STA 727			BOEING CARD NO. 53-669-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL				
		ACCESS NOTE			ZONE 142

Inspect (High Frequency Eddy Current) the frame inner chord at STA 727 between stringers S-17 and S-21 on both the left and right hand sides.

See Doc. D626A001-DTR, DTR check form 53-40-16-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-25.

ACCESS NOTE: Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WHEEL WELL AFT BULKHEAD AND PRESSURE WEB STA 727
		D633A109-AKS 53-669-00-02

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-669-00-02
				MECH INSP
TASK 53-05-02-250-853				
1. INTERNAL - SPECIAL DETAILED: WHEEL WELL AFT BULKHEAD AND PRESSURE WEB STA 727				
<p>A. Inspection</p> <p><u>NOTE:</u> Remove and/or displace aft cargo forward bulkhead, ceiling, sidewall panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-053</p> <p>(1) Do a High Frequency Eddy Current inspection of the frame inner chord at STA 727 between stringers S-17 and S-21 on both the left and right hand sides.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-16-3 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-25.</p>				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WHEEL WELL AFT BULKHEAD AND PRESSURE WEB STA 727		
		D633A109-AKS 53-669-00-02		

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT WHEEL WELL BULKHEAD AND PRESSURE WEB, STA 727			BOEING CARD NO. 53-670-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 241 242

Inspect (Detailed) the frame inner chord and web between stringers S-9L and S-9R at STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-16-5 for alternative inspections.

ACCESS NOTE: Remove and/or displace passenger cabin ceiling panels and insulation as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT WHEEL WELL BULKHEAD AND PRESSURE WEB, STA 727
		D633A109-AKS 53-670-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-670-00-01
TASK 53-05-02-211-832				MECH INSP
1. INTERNAL - DETAILED: AFT WHEEL WELL BULKHEAD AND PRESSURE WEB, STA 727				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin ceiling panels and insulation as required to perform the inspection.</p> <p>SUBTASK 53-05-02-211-032</p> <p>(1) Do a Detailed inspection of the frame inner chord and web between stringers S-9L and S-9R at STA 727.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-16-5 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL	SOURCE AWL	AFT WHEEL WELL BULKHEAD AND PRESSURE WEB, STA 727		
		D633A109-AKS 53-670-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT AFT WHEEL WELL BULKHEAD FRAME STA 727			BOEING CARD NO. 53-671-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 241

Inspect (High Frequency Eddy Current) the visible portion of the frame web above the splice angle on the forward side of the frame between stringers S-9 and S-10 on both sides of the aircraft at STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-16-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-92.

ACCESS NOTE: Remove and/or displace passenger cabin ceiling panels and insulation as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT AFT WHEEL WELL BULKHEAD FRAME STA 727
		D633A109-AKS 53-671-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-671-00-01
TASK 53-05-02-250-855				MECH INSP
1. INTERNAL - SPECIAL DETAILED: AFT WHEEL WELL BULKHEAD FRAME, STA 727				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin ceiling panels and insulation as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-055</p> <p>(1) Do a High Frequency Eddy Current inspection of the visible portion of the frame web above the splice angle on the forward side of the frame between stringers S-9 and S-10 on both sides of the aircraft at STA 727.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-16-6 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-92.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT AFT WHEEL WELL BULKHEAD FRAME STA 727		
		D633A109-AKS 53-671-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT AFT WHEEL WELL BULKHEAD FRAME STA 727			BOEING CARD NO. 53-671-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 242

Inspect (High Frequency Eddy Current) the visible portion of the frame web above the splice angle on the forward side of the frame between stringers S-9 and S-10 on both sides of the aircraft at STA 727.

See Doc. D626A001-DTR, DTR check form 53-40-16-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-92.

ACCESS NOTE: Remove and/or displace passenger cabin ceiling panels and insulation as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT AFT WHEEL WELL BULKHEAD FRAME STA 727
		D633A109-AKS 53-671-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-671-00-02
TASK 53-05-02-250-855				MECH INSP
1. INTERNAL - SPECIAL DETAILED: AFT WHEEL WELL BULKHEAD FRAME, STA 727				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin ceiling panels and insulation as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-055</p> <p>(1) Do a High Frequency Eddy Current inspection of the visible portion of the frame web above the splice angle on the forward side of the frame between stringers S-9 and S-10 on both sides of the aircraft at STA 727.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-16-6 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-92.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT AFT WHEEL WELL BULKHEAD FRAME STA 727		
		D633A109-AKS 53-671-00-02	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT SIDE STRUT SUPPORT FRAME STA 706			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-672-00-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 241

Inspect (High Frequency Eddy Current) the frame inner chord and fail-safe angle (around the fasteners common to the fail-safe angle), the forward frame web (around the fasteners common to the fail-safe angle), and the frame outer chord (around the fasteners common to the skin) between stringers S-10 and S-13.

See Doc. D626A001-DTR, DTR check form 53-40-17-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-19.

ACCESS NOTE: Remove and/or displace passenger cabin sidewalls and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT SIDE STRUT SUPPORT FRAME STA 706
		D633A109-AKS 53-672-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-672-00-01
TASK 53-05-02-250-856				MECH INSP
1. INTERNAL - SPECIAL DETAILED: SIDE STRUT SUPPORT FRAME, STA 706				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewalls and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-056</p> <p>(1) Do a High Frequency Eddy Current inspection of the frame inner chord and fail-safe angle (around the fasteners common to the fail-safe angle), the forward frame web (around the fasteners common to the fail-safe angle), and the frame outer chord (around the fasteners common to the skin) between stringers S-10 and S-13.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-17-1 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-19.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT SIDE STRUT SUPPORT FRAME STA 706		
		D633A109-AKS 53-672-00-01		
		Page 2 of 2 Jun 15/2015		

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT SIDE STRUT SUPPORT FRAME STA 706			BOEING CARD NO. 53-672-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 242

Inspect (High Frequency Eddy Current) the frame inner chord and fail-safe angle (around the fasteners common to the fail-safe angle), the forward frame web (around the fasteners common to the fail-safe angle), and the frame outer chord (around the fasteners common to the skin) between stringers S-10 and S-13.

See Doc. D626A001-DTR, DTR check form 53-40-17-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-11-19.

ACCESS NOTE: Remove and/or displace passenger cabin sidewalls and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT SIDE STRUT SUPPORT FRAME STA 706
		D633A109-AKS 53-672-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-672-00-02
TASK 53-05-02-250-856				MECH INSP
1. INTERNAL - SPECIAL DETAILED: SIDE STRUT SUPPORT FRAME, STA 706				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewalls and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-056</p> <p>(1) Do a High Frequency Eddy Current inspection of the frame inner chord and fail-safe angle (around the fasteners common to the fail-safe angle), the forward frame web (around the fasteners common to the fail-safe angle), and the frame outer chord (around the fasteners common to the skin) between stringers S-10 and S-13.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-17-1 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-19.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT SIDE STRUT SUPPORT FRAME STA 706		
		D633A109-AKS 53-672-00-02		
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT MAIN LANDING GEAR SUPPORT FRAME, STA 695			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-673-00-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 133 241

Inspect (General Visual) the upper fastener through the web.

See Doc. D626A001-DTR, DTR check form 53-40-18-2 for alternative inspections.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT MAIN LANDING GEAR SUPPORT FRAME, STA 695
		D633A109-AKS 53-673-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-673-00-01
TASK 53-05-02-210-802				MECH INSP
1. INTERNAL - GENERAL VISUAL: MAIN LANDING GEAR SUPPORT FRAME, STA 695				
A. Inspection				
<p><u>NOTE:</u> Remove or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required.</p> <p>SUBTASK 53-05-02-210-002</p> <p>(1) Do a General Visual inspection of the upper fastener through the web. See Doc. D626A001-DTR, DTR check form 53-40-18-2 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT MAIN LANDING GEAR SUPPORT FRAME, STA 695				
D633A109-AKS 53-673-00-01				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT MAIN LANDING GEAR SUPPORT FRAME, STA 695			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-673-00-02
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 134 242

Inspect (General Visual) the upper fastener through the web.

See Doc. D626A001-DTR, DTR check form 53-40-18-2 for alternative inspections.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MAIN LANDING GEAR SUPPORT FRAME, STA 695
		D633A109-AKS 53-673-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-673-00-02
TASK 53-05-02-210-802				MECH INSP
1. INTERNAL - GENERAL VISUAL: MAIN LANDING GEAR SUPPORT FRAME, STA 695				
A. Inspection				
<p><u>NOTE:</u> Remove or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required.</p> <p>SUBTASK 53-05-02-210-002</p> <p>(1) Do a General Visual inspection of the upper fastener through the web. See Doc. D626A001-DTR, DTR check form 53-40-18-2 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MAIN LANDING GEAR SUPPORT FRAME, STA 695		
		D633A109-AKS 53-673-00-02	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716			BOEING CARD NO. 53-674-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 133 241

Inspect (High Frequency Eddy Current) the frame web around the fasteners common to the stringer clip at stringer S-16 and the forward and aft flanges of the frame inner chord from 6 inches above and below stringer S-16.

See Doc. D626A001-DTR, DTR check form 53-40-19-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-40-05.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall panels, sidewall air grilles, and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716
		D633A109-AKS 53-674-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-674-00-01
TASK 53-05-02-250-857				MECH INSP
1. INTERNAL - SPECIAL DETAILED: MAIN LANDING GEAR SUPPORT FRAME, STA 716				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewall panels, sidewall air grilles, and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-057</p> <p>(1) Do a High Frequency Eddy Current inspection of the frame web around the fasteners common to the stringer clip at stringer S-16 and the forward and aft flanges of the frame inner chord from 6 inches above and below stringer S-16.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-19-1 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-40-05.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716	
			D633A109-AKS 53-674-00-01	Page 2 of 2 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716			BOEING CARD NO. 53-674-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 134 242

Inspect (High Frequency Eddy Current) the frame web around the fasteners common to the stringer clip at stringer S-16 and the forward and aft flanges of the frame inner chord from 6 inches above and below stringer S-16.

See Doc. D626A001-DTR, DTR check form 53-40-19-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-40-05.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall panels, sidewall air grilles, and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716
		D633A109-AKS 53-674-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-674-00-02
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TASK 53-05-02-250-857

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: MAIN LANDING GEAR SUPPORT FRAME, STA 716**A. Inspection**

NOTE: Remove and/or displace passenger cabin sidewall panels, sidewall air grilles, and insulation blankets as required to perform the inspection.

SUBTASK 53-05-02-250-057

- (1) Do a High Frequency Eddy Current inspection of the frame web around the fasteners common to the stringer clip at stringer S-16 and the forward and aft flanges of the frame inner chord from 6 inches above and below stringer S-16.

See Doc. D626A001-DTR, DTR check form 53-40-19-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-40-05.

———— END OF TASK ———

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716
		D633A109-AKS 53-674-00-02

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716			BOEING CARD NO. 53-675-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 133 241

Inspect (High Frequency Eddy Current) the stub beam upper chord around the two fasteners common to the crease beam inner chord.

See Doc. D626A001-DTR, DTR check form 53-40-19-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-14.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716
		D633A109-AKS 53-675-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-675-00-01
TASK 53-05-02-250-858				MECH INSP
1. INTERNAL - SPECIAL DETAILED: MAIN LANDING GEAR SUPPORT FRAME, STA 716				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-058</p> <p>(1) Do a High Frequency Eddy Current inspection of the stub beam upper chord around the two fasteners common to the crease beam inner chord.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-19-2 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-14.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716	
			D633A109-AKS 53-675-00-01	Page 2 of 2 Jun 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-675-00-02 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 134 242

Inspect (High Frequency Eddy Current) the stub beam upper chord around the two fasteners common to the crease beam inner chord.

See Doc. D626A001-DTR, DTR check form 53-40-19-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-14.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716
		D633A109-AKS 53-675-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-675-00-02
TASK 53-05-02-250-858				MECH INSP
1. INTERNAL - SPECIAL DETAILED: MAIN LANDING GEAR SUPPORT FRAME, STA 716				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-058</p> <p>(1) Do a High Frequency Eddy Current inspection of the stub beam upper chord around the two fasteners common to the crease beam inner chord.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-19-2 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-14.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716		
		D633A109-AKS 53-675-00-02	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716			BOEING CARD NO. 53-676-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 133 241

Inspect (High Frequency Eddy Current) the web of the stub beam around the fasteners common to the floor clip.

See Doc. D626A001-DTR, DTR check form 53-40-19-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716
		D633A109-AKS 53-676-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-676-00-01
TASK 53-05-02-250-859				MECH INSP
1. INTERNAL - SPECIAL DETAILED: MAIN LANDING GEAR SUPPORT FRAME, STA 716				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-059</p> <p>(1) Do a High Frequency Eddy Current inspection on the web of the stub beam around the fasteners common to the floor clip.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-19-3 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716		
		D633A109-AKS 53-676-00-01	Page 2 of 2 Jun 15/2015	

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737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-676-00-02 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 134 242

Inspect (High Frequency Eddy Current) the web of the stub beam around the fasteners common to the floor clip.

See Doc. D626A001-DTR, DTR check form 53-40-19-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716
		D633A109-AKS 53-676-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-676-00-02
TASK 53-05-02-250-859				MECH INSP
1. INTERNAL - SPECIAL DETAILED: MAIN LANDING GEAR SUPPORT FRAME, STA 716				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-059</p> <p>(1) Do a High Frequency Eddy Current inspection on the web of the stub beam around the fasteners common to the floor clip.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-19-3 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716		
		D633A109-AKS 53-676-00-02	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716			BOEING CARD NO. 53-676-10-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 15000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 133 241

Inspect (High Frequency Eddy Current) the upper side of the upper flange on both the forward and aft sides from BL 45.5 to BL 64.6 on both the right and left sides.

See Doc. D626A001-DTR, DTR check form 53-40-19-3a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716
		D633A109-AKS 53-676-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-676-10-01
TASK 53-05-02-250-860				MECH INSP
1. INTERNAL - SPECIAL DETAILED: MAIN LANDING GEAR SUPPORT FRAME, STA 716				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-060</p> <p>(1) Do a High Frequency Eddy Current inspection on the upper side of the upper flange on both the forward and aft sides from BL 45.5 to BL 64.6 on both the right and left sides.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-19-3a for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT MAIN LANDING GEAR SUPPORT FRAME STA 716	
			D633A109-AKS 53-676-10-01	Page 2 of 2 Jun 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-676-10-02 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 15000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 134 242

Inspect (High Frequency Eddy Current) the upper side of the upper flange on both the forward and aft sides from BL 45.5 to BL 64.6 on both the right and left sides.

See Doc. D626A001-DTR, DTR check form 53-40-19-3a for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09.

ACCESS NOTE: Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716
		D633A109-AKS 53-676-10-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-676-10-02
TASK 53-05-02-250-860				MECH INSP
1. INTERNAL - SPECIAL DETAILED: MAIN LANDING GEAR SUPPORT FRAME, STA 716				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewall air grilles, floor panels and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-060</p> <p>(1) Do a High Frequency Eddy Current inspection on the upper side of the upper flange on both the forward and aft sides from BL 45.5 to BL 64.6 on both the right and left sides.</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-19-3a for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-09.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT MAIN LANDING GEAR SUPPORT FRAME STA 716				
D633A109-AKS 53-676-10-02				
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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT WHEEL WELL FRAME, STA 685			BOEING CARD NO.
DATE	TASK DETAILED				53-677-00-01
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 133 241

Inspect (Detailed) the frame inner chord from stringers S-13 to S-15.

See Doc. D626A001-DTR, DTR check form 53-40-21-2 for alternative inspections.

ACCESS NOTE: Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WHEEL WELL FRAME, STA 685
		D633A109-AKS 53-677-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-677-00-01
TASK 53-05-02-211-833				MECH INSP
1. INTERNAL - DETAILED: WHEEL WELL FRAME, STA 685				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-211-033</p> <p>(1) Do a Detailed inspection of the frame inner chord from stringers S-13 to S-15. See Doc. D626A001-DTR, DTR check form 53-40-21-2 for alternative inspections.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WHEEL WELL FRAME, STA 685		
		D633A109-AKS 53-677-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WHEEL WELL FRAME, STA 685			BOEING CARD NO. 53-677-00-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE		ZONE 134 242		

Inspect (Detailed) the frame inner chord from stringers S-13 to S-15.

See Doc. D626A001-DTR, DTR check form 53-40-21-2 for alternative inspections.

ACCESS NOTE: Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WHEEL WELL FRAME, STA 685
		D633A109-AKS 53-677-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-677-00-02
TASK 53-05-02-211-833				MECH INSP
1. INTERNAL - DETAILED: WHEEL WELL FRAME, STA 685				
A. Inspection				
<p><u>NOTE:</u> Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.</p> <p>SUBTASK 53-05-02-211-033</p> <p>(1) Do a Detailed inspection of the frame inner chord from stringers S-13 to S-15. See Doc. D626A001-DTR, DTR check form 53-40-21-2 for alternative inspections.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT WHEEL WELL FRAME, STA 685				
D633A109-AKS 53-677-00-02				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WHEEL WELL FRAME STA 685			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-678-00-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 133 241

Inspect (High Frequency Eddy Current) the frame inner chord flange and around accessible fasteners common to the inner chord and stringer clips from stringers S-17 to S-14.

See Doc. D626A001-DTR, DTR check form 53-40-21-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-17.

ACCESS NOTE: Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WHEEL WELL FRAME STA 685
		D633A109-AKS 53-678-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-678-00-01
				MECH INSP
TASK 53-05-02-250-861				
1. INTERNAL - SPECIAL DETAILED: WHEEL WELL FRAME, STA 685				
A. Inspection				
<u>SUBTASK 53-05-02-250-061</u>				
<p>NOTE: Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.</p>				
<p>(1) Do a High Frequency Eddy Current inspection of the frame inner chord flange and around accessible fasteners common to the inner chord and stringer clips from stringers S-17 to S-14.</p>				
<p>See Doc. D626A001-DTR, DTR check form 53-40-21-3 for alternative inspections.</p>				
<p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-17.</p>				
<hr style="display: inline-block; width: 40%; vertical-align: middle;"/> END OF TASK <hr style="display: inline-block; width: 40%; vertical-align: middle;"/>				

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WHEEL WELL FRAME STA 685			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-678-00-02 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 134 242

Inspect (High Frequency Eddy Current) the frame inner chord flange and around accessible fasteners common to the inner chord and stringer clips from stringers S-17 to S-14.

See Doc. D626A001-DTR, DTR check form 53-40-21-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-17.

ACCESS NOTE: Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WHEEL WELL FRAME STA 685
		D633A109-AKS 53-678-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-678-00-02
TASK 53-05-02-250-861				MECH INSP
1. INTERNAL - SPECIAL DETAILED: WHEEL WELL FRAME, STA 685				
A. Inspection				
SUBTASK 53-05-02-250-061				
NOTE: Remove and/or displace passenger cabin sidewalls, sidewall air grilles, and insulation blankets as required to perform the inspection.				
(1) Do a High Frequency Eddy Current inspection of the frame inner chord flange and around accessible fasteners common to the inner chord and stringer clips from stringers S-17 to S-14.				
See Doc. D626A001-DTR, DTR check form 53-40-21-3 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-17.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WHEEL WELL FRAME STA 685		
		D633A109-AKS 53-678-00-02	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT WHEEL WELL FRAME AT STA 685			BOEING CARD NO. 53-679-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 133 241

Inspect (High Frequency Eddy Current) the stub beam upper chord from two inches inside the skin to a distance of twelve inches inboard and around any fasteners through the upper web and chord in this area.

See Doc. D626A001-DTR, DTR check form 53-40-21-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-06.

ACCESS NOTE: Remove and/or displace passenger cabin floor panels as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WHEEL WELL FRAME AT STA 685
		D633A109-AKS 53-679-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-679-00-01
TASK 53-05-02-250-862				MECH INSP
1. INTERNAL - SPECIAL DETAILED: WHEEL WELL FRAME AT STA 685				
A. Inspection				
SUBTASK 53-05-02-250-062				
NOTE: Remove and/or displace passenger cabin floor panels as required to perform the inspection.				
(1) Do a High Frequency Eddy Current inspection of the stub beam upper chord from two inches inside the skin to a distance of twelve inches inboard and around any fasteners through the upper web and chord in this area.				
See Doc. D626A001-DTR, DTR check form 53-40-21-4 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-06.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WHEEL WELL FRAME AT STA 685		
		D633A109-AKS 53-679-00-01	Page 2 of 2 Jun 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WHEEL WELL FRAME AT STA 685			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-679-00-02 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 134 242

Inspect (High Frequency Eddy Current) the stub beam upper chord from two inches inside the skin to a distance of twelve inches inboard and around any fasteners through the upper web and chord in this area.

See Doc. D626A001-DTR, DTR check form 53-40-21-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-06.

ACCESS NOTE: Remove and/or displace passenger cabin floor panels as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WHEEL WELL FRAME AT STA 685
		D633A109-AKS 53-679-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-679-00-02
TASK 53-05-02-250-862				MECH INSP
1. INTERNAL - SPECIAL DETAILED: WHEEL WELL FRAME AT STA 685				
A. Inspection				
SUBTASK 53-05-02-250-062				
NOTE: Remove and/or displace passenger cabin floor panels as required to perform the inspection.				
(1) Do a High Frequency Eddy Current inspection of the stub beam upper chord from two inches inside the skin to a distance of twelve inches inboard and around any fasteners through the upper web and chord in this area.				
See Doc. D626A001-DTR, DTR check form 53-40-21-4 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-06.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WHEEL WELL FRAME AT STA 685		
		D633A109-AKS 53-679-00-02	Page 2 of 2 Jun 15/2015	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD			BOEING CARD NO. 53-682-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER NOTE
		ACCESS 832			ZONE 231
					ENGINE ALL

Inspect (Low Frequency Eddy Current) the doublers around the fasteners common to the STA 578 cutout forward edge frame outer chord from stringers S-10 to S-13.

See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

NOTE: This inspection must be work in conjunction with either fatigue task 53-682-01 or 53-682-03 to meet DTR requirements..

AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-682-00-01
				MECH INSP
TASK 53-05-02-250-867				
1. EXTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD				
A. Inspection				
SUBTASK 53-05-02-010-080				
(1) Open this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Open this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
SUBTASK 53-05-02-250-067				
(2) Do a Low Frequency Eddy Current inspection of the doublers around the fasteners common to the STA 578 cutout forward edge frame outer chord from stringers S-10 to S-13.				
See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.				
NOTE:				
NOTE: This inspection must be work in conjunction with either fatigue task 53-682-01 or 53-682-03 to meet DTR requirements.				
SUBTASK 53-05-02-410-078				
(3) Close this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Close this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-00-01

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD			BOEING CARD NO. 53-682-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE 800 900 900ER NOTE
STATION	SKILL AIRPL				ENGINE ALL
		ACCESS 842			ZONE 232

Inspect (Low Frequency Eddy Current) the doublers around the fasteners common to the STA 578 cutout forward edge frame outer chord from stringers S-10 to S-13.

See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

NOTE: This inspection must be work in conjunction with either fatigue task 53-682-01 or 53-682-03 to meet DTR requirements..

AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-682-00-02	MECH	INSP
TASK 53-05-02-250-867						

1. EXTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD

A. Inspection

SUBTASK 53-05-02-010-080

(1) Open this access panel on the Left side:

Number Name/Location
832 Emergency Exit

Open this access panel on the Right side:

Number Name/Location
842 Emergency Exit

SUBTASK 53-05-02-250-067

(2) Do a Low Frequency Eddy Current inspection of the doublers around the fasteners common to the STA 578 cutout forward edge frame outer chord from stringers S-10 to S-13.

See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

NOTE:

NOTE: This inspection must be work in conjunction with either fatigue task 53-682-01 or 53-682-03 to meet DTR requirements.

SUBTASK 53-05-02-410-078

(3) Close this access panel on the Left side:

Number Name/Location
832 Emergency Exit

Close this access panel on the Right side:

Number Name/Location
842 Emergency Exit

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD			BOEING CARD NO. 53-682-01-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL NOTE
		ACCESS 832 NOTE			ZONE 231

Inspect (High Frequency Eddy Current) the edges of the doublers from stringers S-11 to S-13.

See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirements.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes.

ACCESS NOTE: Emergency Exit Door must be open to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-01-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-682-01-01
				MECH INSP
TASK 53-05-02-250-868				
1. EXTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD				
A. Inspection				
SUBTASK 53-05-02-010-081				
(1) Open this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Open this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
NOTE: Emergency Exit Door must be open to perform this inspection.				
SUBTASK 53-05-02-250-068				
(2) Do a High Frequency Eddy Current inspection on the edges of the doublers from stringers S-11 to S-13.				
See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.				
NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirements.				
SUBTASK 53-05-02-410-079				
(3) Close this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Close this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD	D633A109-AKS 53-682-01-01	Page 2 of 2 Jun 15/2015
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD			BOEING CARD NO. 53-682-01-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL NOTE
		ACCESS 842			ZONE 232
		NOTE			

Inspect (High Frequency Eddy Current) the edges of the doublers from stringers S-11 to S-13.

See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirements.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes.

ACCESS NOTE: Emergency Exit Door must be open to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-01-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-682-01-02
				MECH INSP
TASK 53-05-02-250-868				
1. EXTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD				
A. Inspection				
SUBTASK 53-05-02-010-081				
(1) Open this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Open this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
NOTE: Emergency Exit Door must be open to perform this inspection.				
SUBTASK 53-05-02-250-068				
(2) Do a High Frequency Eddy Current inspection on the edges of the doublers from stringers S-11 to S-13.				
See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.				
NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirements.				
SUBTASK 53-05-02-410-079				
(3) Close this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Close this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-01-02

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD			BOEING CARD NO. 53-682-03-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER NOTE
		ACCESS 832			ZONE 231
		NOTE			

Inspect (High Frequency Eddy Current) the inner doubler from stringers S-11 to S-13.

See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirements.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes.

ACCESS NOTE: Emergency Exit Door must be open to perform this inspection. Seal removal or displacement is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-03-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-682-03-01
				MECH INSP
TASK 53-05-02-250-869				
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD				
A. Inspection				
SUBTASK 53-05-02-010-082				
(1) Open this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Open this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
NOTE: Emergency Exit Door must be open to perform this inspection. Seal removal or displacement is required to perform the inspection.				
SUBTASK 53-05-02-250-069				
(2) Do a High Frequency Eddy Current inspection of the inner doubler from stringers S-11 to S-13.				
See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.				
NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirements.				
SUBTASK 53-05-02-410-080				
(3) Close this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Close this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-03-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD			BOEING CARD NO. 53-682-03-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL NOTE
		ACCESS 842			ZONE 232
		NOTE			

Inspect (High Frequency Eddy Current) the inner doubler from stringers S-11 to S-13.

See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.

NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirements.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

AIRPLANE NOTE: Applicable to 737-800 (line number 9 and on) and 737-900 airplanes.

ACCESS NOTE: Emergency Exit Door must be open to perform this inspection. Seal removal or displacement is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD
		D633A109-AKS 53-682-03-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-682-03-02
				MECH INSP
TASK 53-05-02-250-869				
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD				
A. Inspection				
SUBTASK 53-05-02-010-082				
(1) Open this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Open this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
NOTE: Emergency Exit Door must be open to perform this inspection. Seal removal or displacement is required to perform the inspection.				
SUBTASK 53-05-02-250-069				
(2) Do a High Frequency Eddy Current inspection of the inner doubler from stringers S-11 to S-13.				
See Doc. D626A001-DTR, DTR check form 53-40-22-3 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.				
NOTE: This inspection may be optional. See DTR 53-40-22-3 for inspection requirements.				
SUBTASK 53-05-02-410-080				
(3) Close this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
Close this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY & FRAME OUTER CHORD		
		D633A109-AKS 53-682-03-02	Page 2 of 2 Jun 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-685-00-01 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 6000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 832 833			
		NOTE			
					ZONE 231

Inspect (High Frequency Eddy Current) the edges of the doublers, on the upper edge, at stringer S-11(from STA 578 to STA 601 and from STA 616 to STA 639).

See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.

NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS D633A109-AKS 53-685-00-01	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-685-00-01
				MECH INSP
TASK 53-05-02-250-874				
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS				
A. Inspection				
SUBTASK 53-05-02-010-085				
(1) Open these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
NOTE: Emergency Exit door must be open to perform the inspection.				
SUBTASK 53-05-02-250-074				
(2) Do a High Frequency Eddy Current inspection on the edges of the doublers, on the upper edge, at stringer S-11(from STA 578 to STA 601 and from STA 616 to STA 639).				
See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.				
NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler.				
SUBTASK 53-05-02-410-083				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS	D633A109-AKS 53-685-00-01	Page 2 of 2 Jun 15/2015
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-685-00-02 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 6000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 842 843			
		NOTE			
					ZONE 232

Inspect (High Frequency Eddy Current) the edges of the doublers, on the upper edge, at stringer S-11(from STA 578 to STA 601 and from STA 616 to STA 639).

See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.

NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS D633A109-AKS 53-685-00-02	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-685-00-02	MECH	INSP
TASK 53-05-02-250-874						
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS						
A. Inspection						
SUBTASK 53-05-02-010-085						
(1) Open these access panels on the Left side:						
Number Name/Location						
832 Emergency Exit						
833 Emergency Exit						
Open these access panels on the Right side:						
Number Name/Location						
842 Emergency Exit						
843 Emergency Exit						
NOTE: Emergency Exit door must be open to perform the inspection.						
SUBTASK 53-05-02-250-074						
(2) Do a High Frequency Eddy Current inspection on the edges of the doublers, on the upper edge, at stringer S-11(from STA 578 to STA 601 and from STA 616 to STA 639).						
See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.						
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.						
NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler.						
SUBTASK 53-05-02-410-083						
(3) Close these access panels on the Left side:						
Number Name/Location						
832 Emergency Exit						
833 Emergency Exit						
Close these access panels on the Right side:						
Number Name/Location						
842 Emergency Exit						
843 Emergency Exit						
———— END OF TASK ————						

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS	D633A109-AKS 53-685-00-02	Page 2 of 2 Jun 15/2015
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-685-01-01 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 6000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 832 833			
		NOTE			
					ZONE 231

Inspect (High Frequency Eddy Current) the inner doubler between the seal retainer and the frames and sills, on the upper edge, at stringer S-11 (from STA 578 to STA 601 and from STA 616 to STA 639).

See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.

NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform this inspection. Remove or displace passenger cabin sidewall lining as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS D633A109-AKS 53-685-01-01	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-685-01-01
				MECH INSP
TASK 53-05-02-250-875				
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS				
A. Inspection				
SUBTASK 53-05-02-010-087				
(1) Open these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
NOTE: Emergency Exit door must be open to perform this inspection. Remove or displace passenger cabin sidewall lining if/as required to perform this inspection.				
SUBTASK 53-05-02-250-075				
(2) Do a High Frequency Eddy Current inspection of the inner doubler between the seal retainer and the frames and sills, on the upper edge, at stringer S-11 (from STA 578 to STA 601 and from STA 616 to STA 639).				
See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.				
NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler.				
SUBTASK 53-05-02-410-085				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS	
		D633A109-AKS 53-685-01-01	Page 2 of 2 Jun 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-685-01-02 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 6000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 842 843			
		NOTE			
					ZONE 232

Inspect (High Frequency Eddy Current) the inner doubler between the seal retainer and the frames and sills, on the upper edge, at stringer S-11 (from STA 578 to STA 601 and from STA 616 to STA 639).

See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.

NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform this inspection. Remove or displace passenger cabin sidewall lining as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS D633A109-AKS 53-685-01-02	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-685-01-02
				MECH INSP
TASK 53-05-02-250-875				
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS				
A. Inspection				
SUBTASK 53-05-02-010-087				
(1) Open these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open this access panel on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
NOTE: Emergency Exit door must be open to perform this inspection. Remove or displace passenger cabin sidewall lining if/as required to perform this inspection.				
SUBTASK 53-05-02-250-075				
(2) Do a High Frequency Eddy Current inspection of the inner doubler between the seal retainer and the frames and sills, on the upper edge, at stringer S-11 (from STA 578 to STA 601 and from STA 616 to STA 639).				
See Doc. D626A001-DTR, DTR check form 53-40-22-6 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.				
NOTE: Doors with external doublers at the upper forward corner refer to DTR 53-40-22-22 for area covered by the doubler.				
SUBTASK 53-05-02-410-085				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS
		D633A109-AKS 53-685-01-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT CUTOUT			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-686-00-01 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS 832 833			
		NOTE			
			ZONE 231		

Inspect (Ultrasonic) the edge frames outer chord under the stop backup fittings at stringers S-11 and S-12.

See Doc. D626A001-DTR, DTR check form 53-40-22-7 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 53-10-10.

AIRPLANE NOTE: For the 737-600 and -700, STA 616 and STA 639.

For the 737-800, STA 578 and STA 639.

ACCESS NOTE: Emergency Exit door must be open to perform the inspection. Remove or displace passenger cabin sidewall lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT	
		D633A109-AKS 53-686-00-01	Page 1 of 2 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-686-00-01
				MECH INSP
TASK 53-05-02-130-805				
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT				
A. Inspection				
SUBTASK 53-05-02-010-005				
(1) Open these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
NOTE: Emergency Exit door must be open to perform the inspection. Remove or displace passenger cabin sidewall lining as required to perform the inspection.				
SUBTASK 53-05-02-130-005				
(2) Do an Ultrasonic inspection of the edge frames outer chord under the stop backup fittings at stringers S-11 and S-12.				
See Doc. D626A001-DTR, DTR check form 53-40-22-7 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 53-10-10.				
SUBTASK 53-05-02-410-003				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT
		D633A109-AKS 53-686-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT CUTOUT			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-686-00-02 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS 842 843			
		NOTE			
					ZONE 232

Inspect (Ultrasonic) the edge frames outer chord under the stop backup fittings at stringers S-11 and S-12.

See Doc. D626A001-DTR, DTR check form 53-40-22-7 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 53-10-10.

AIRPLANE NOTE: For the 737-600 and -700, STA 616 and STA 639.

For the 737-800, STA 578 and STA 639.

ACCESS NOTE: Emergency Exit door must be open to perform the inspection. Remove or displace passenger cabin sidewall lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT
		D633A109-AKS 53-686-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-686-00-02
TASK 53-05-02-130-805				MECH INSP
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT				
A. Inspection				
SUBTASK 53-05-02-010-005				
(1) Open these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
NOTE: Emergency Exit door must be open to perform the inspection. Remove or displace passenger cabin sidewall lining as required to perform the inspection.				
SUBTASK 53-05-02-130-005				
(2) Do an Ultrasonic inspection of the edge frames outer chord under the stop backup fittings at stringers S-11 and S-12.				
See Doc. D626A001-DTR, DTR check form 53-40-22-7 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 53-10-10.				
SUBTASK 53-05-02-410-003				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT
		D633A109-AKS 53-686-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT CUTOUT			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-688-00-01 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 832 833			
		NOTE			
					ZONE 231

Inspect (High Frequency Eddy Current) the edges of the cutout doublers at all four lower corners (intersection of lower sill and edge frames) at stringer S-14 (from STA 578 to STA 607 and from STA 616 to STA 639).

See Doc. D626A001-DTR, DTR check form 53-40-22-9 for alternative inspections.

NOTE: Doors with external doublers at the lower door corner refer to DTR 53-40-22-22 for area covered by the doubler.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT	
		D633A109-AKS 53-688-00-01	Page 1 of 2 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-688-00-01
				MECH INSP
TASK 53-05-02-250-878				
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT				
A. Inspection				
SUBTASK 53-05-02-010-089				
(1) Open these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
NOTE: Emergency Exit door must be open to perform the inspection.				
SUBTASK 53-05-02-250-078				
(2) Do a High Frequency Eddy Current inspection on the edges of the cutout doublers at all four lower corners (intersection of lower sill and edge frames) at stringer S-14 (from STA 578 to STA 607 and from STA 616 to STA 639).				
See Doc. D626A001-DTR, DTR check form 53-40-22-9 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.				
NOTE: Doors with external doublers at the lower door corner refer to DTR 53-40-22-22 for area covered by the doubler.				
SUBTASK 53-05-02-410-087				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT	
		D633A109-AKS 53-688-00-01	Page 2 of 2 Jun 15/2016

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-688-00-02 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 842 843			
		NOTE			
					ZONE 232

Inspect (High Frequency Eddy Current) the edges of the cutout doublers at all four lower corners (intersection of lower sill and edge frames) at stringer S-14 (from STA 578 to STA 607 and from STA 616 to STA 639).

See Doc. D626A001-DTR, DTR check form 53-40-22-9 for alternative inspections.

NOTE: Doors with external doublers at the lower door corner refer to DTR 53-40-22-22 for area covered by the doubler.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.

ACCESS NOTE: Emergency Exit door must be open to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS D633A109-AKS 53-688-00-02	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-688-00-02
				MECH INSP
TASK 53-05-02-250-878				
1. INTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT				
A. Inspection				
SUBTASK 53-05-02-010-089				
(1) Open these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
NOTE: Emergency Exit door must be open to perform the inspection.				
SUBTASK 53-05-02-250-078				
(2) Do a High Frequency Eddy Current inspection on the edges of the cutout doublers at all four lower corners (intersection of lower sill and edge frames) at stringer S-14 (from STA 578 to STA 607 and from STA 616 to STA 639).				
See Doc. D626A001-DTR, DTR check form 53-40-22-9 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-15.				
NOTE: Doors with external doublers at the lower door corner refer to DTR 53-40-22-22 for area covered by the doubler.				
SUBTASK 53-05-02-410-087				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT - DOOR CUTOUT CORNERS
		D633A109-AKS 53-688-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD			BOEING CARD NO. 53-690-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 832 833		ZONE 231	
<p>Inspect (Low Frequency Eddy Current) the doublers around the fasteners common to the edge frame at STA 578 (from stringers S-13 to S-15) and at STAs 601, 616 and 639 (from stringers S-10 to S-15).</p> <p>See Doc. D626A001-DTR, DTR check form 53-40-22-11 for alternative inspections.</p> <p>NOTE: Doors with external doublers at the lower door corners refer to DTR 53-40-22-22 for area covered by doubler.</p>					
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD		
			D633A109-AKS 53-690-00-01		
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-690-00-01
TASK 53-05-02-250-880				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT SKIN ASSY AND FRAME OUTER CHORD				
A. Inspection				
SUBTASK 53-05-02-010-091				
(1) Open this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
SUBTASK 53-05-02-250-080				
(2) Do a Low Frequency Eddy Current inspection of the doublers around the fasteners common to the edge frame at STA 578 (from stringers S-13 to S-15) and at STAs 601, 616 and 639 (from stringers S-10 to S-15).				
See Doc. D626A001-DTR, DTR check form 53-40-22-11 for alternative inspections.				
NOTE: Doors with external doublers at the lower door corners refer to DTR 53-40-22-22 for area covered by doubler.				
SUBTASK 53-05-02-410-089				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD	D633A109-AKS 53-690-00-01	Page 2 of 2 Feb 15/2015
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD			BOEING CARD NO. 53-690-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 842 843			ZONE 232

Inspect (Low Frequency Eddy Current) the doublers around the fasteners common to the edge frame at STA 578 (from stringers S-13 to S-15) and at STAs 601, 616 and 639 (from stringers S-10 to S-15).

See Doc. D626A001-DTR, DTR check form 53-40-22-11 for alternative inspections.

NOTE: Doors with external doublers at the lower door corners refer to DTR 53-40-22-22 for area covered by doubler.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD
		D633A109-AKS 53-690-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-690-00-02
TASK 53-05-02-250-880				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: OVERWING EMERGENCY EXIT CUTOUT SKIN ASSY AND FRAME OUTER CHORD				
A. Inspection				
SUBTASK 53-05-02-010-091				
(1) Open this access panel on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
SUBTASK 53-05-02-250-080				
(2) Do a Low Frequency Eddy Current inspection of the doublers around the fasteners common to the edge frame at STA 578 (from stringers S-13 to S-15) and at STAs 601, 616 and 639 (from stringers S-10 to S-15).				
See Doc. D626A001-DTR, DTR check form 53-40-22-11 for alternative inspections.				
NOTE: Doors with external doublers at the lower door corners refer to DTR 53-40-22-22 for area covered by doubler.				
SUBTASK 53-05-02-410-089				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT SKIN ASSEMBLY AND FRAME OUTER CHORD
		D633A109-AKS 53-690-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT CUTOUT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-692-00-01 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 832 833			
		NOTE			
					ZONE 231

Inspect (General Visual) the width and thickness of the lower sill inner splice strap and around the five fasteners at STAs 578, 601, 616 and 639.

Note: Fastener location is three FWD and two AFT at STAs 578/616 and two FWD and three AFT at STAs 601/639.

See Doc. D626A001-DTR, DTR check form 53-40-22-13 for alternative inspections.

ACCESS NOTE: Emergency Exit Door must be open to perform the inspection. Remove or displace passenger cabin sidewall lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT	
		D633A109-AKS 53-692-00-01	Page 1 of 2 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-692-00-01
TASK 53-05-02-210-803				MECH INSP

1. INTERNAL - GENERAL VISUAL: OVERWING EMERGENCY EXIT CUTOUT

A. Inspection

SUBTASK 53-05-02-010-010

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
832	Emergency Exit
833	Emergency Exit

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
842	Emergency Exit
843	Emergency Exit

NOTE: Emergency Exit Door must be open to perform the inspection. Remove or displace passenger cabin sidewall lining as required to perform the inspection.

SUBTASK 53-05-02-210-003

(2) Do a General Visual inspection on the width and thickness of the lower sill inner splice strap and around the five fasteners at STAs 578, 601, 616 and 639.

NOTE: Fastener location is three FWD and two AFT at STAs 578/616 and two FWD and three AFT at STAs 601/639.

See Doc. D626A001-DTR, DTR check form 53-40-22-13 for alternative inspections.

SUBTASK 53-05-02-410-008

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
832	Emergency Exit
833	Emergency Exit

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
842	Emergency Exit
843	Emergency Exit

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT CUTOUT	
		D633A109-AKS 53-692-00-01	Page 2 of 2 Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT CUTOUT			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-692-00-02 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 800 900 900ER ENGINE ALL
		ACCESS 842 843			
		NOTE			
					ZONE 232

Inspect (General Visual) the width and thickness of the lower sill inner splice strap and around the five fasteners at STAs 578, 601, 616 and 639.

Note: Fastener location is three FWD and two AFT at STAs 578/616 and two FWD and three AFT at STAs 601/639.

See Doc. D626A001-DTR, DTR check form 53-40-22-13 for alternative inspections.

ACCESS NOTE: Emergency Exit Door must be open to perform the inspection. Remove or displace passenger cabin sidewall lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT	
		D633A109-AKS 53-692-00-02	Page 1 of 2 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-692-00-02
TASK 53-05-02-210-803				MECH INSP
1. INTERNAL - GENERAL VISUAL: OVERWING EMERGENCY EXIT CUTOUT				
A. Inspection				
SUBTASK 53-05-02-010-010				
(1) Open these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Open these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
NOTE: Emergency Exit Door must be open to perform the inspection. Remove or displace passenger cabin sidewall lining as required to perform the inspection.				
SUBTASK 53-05-02-210-003				
(2) Do a General Visual inspection on the width and thickness of the lower sill inner splice strap and around the five fasteners at STAs 578, 601, 616 and 639.				
NOTE: Fastener location is three FWD and two AFT at STAs 578/616 and two FWD and three AFT at STAs 601/639.				
See Doc. D626A001-DTR, DTR check form 53-40-22-13 for alternative inspections.				
SUBTASK 53-05-02-410-008				
(3) Close these access panels on the Left side:				
Number Name/Location				
832 Emergency Exit				
833 Emergency Exit				
Close these access panels on the Right side:				
Number Name/Location				
842 Emergency Exit				
843 Emergency Exit				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT CUTOUT
		D633A109-AKS 53-692-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS			BOEING CARD NO. 53-695-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 832 833				ZONE 231
		NOTE				

Inspect (Detailed) the door stops attached to the forward and aft edge frames, six fittings per door.

See Doc. D626A001-DTR, DTR check form 53-40-23-1 for alternative inspections.

ACCESS NOTE: Emergency Exit Door must be open to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS D633A109-AKS 53-695-00-01
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-695-00-01
TASK 53-05-02-211-835				MECH INSP

1. INTERNAL - DETAILED: OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS

A. Inspection

SUBTASK 53-05-02-010-021

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
832	Emergency Exit
833	Emergency Exit

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
842	Emergency Exit
843	Emergency Exit

NOTE: Emergency Exit Door must be open to perform the inspection.

SUBTASK 53-05-02-211-035

(2) Do a Detailed inspection of the door stops attached to the forward and aft edge frames, six fittings per door.

See Doc. D626A001-DTR, DTR check form 53-40-23-1 for alternative inspections.

SUBTASK 53-05-02-410-019

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
832	Emergency Exit
833	Emergency Exit

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
842	Emergency Exit
843	Emergency Exit

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS
		D633A109-AKS 53-695-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS			BOEING CARD NO. 53-695-00-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 842 843				ZONE 232
		NOTE				

Inspect (Detailed) the door stops attached to the forward and aft edge frames, six fittings per door.

See Doc. D626A001-DTR, DTR check form 53-40-23-1 for alternative inspections.

ACCESS NOTE: Emergency Exit Door must be open to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS
		D633A109-AKS 53-695-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-695-00-02
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TASK 53-05-02-211-835

MECH

INSP

1. INTERNAL - DETAILED: OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS**A. Inspection**

SUBTASK 53-05-02-010-021

- (1) Open these access panels on the Left side:

Number Name/Location

832	Emergency Exit
833	Emergency Exit

Open these access panels on the Right side:

Number Name/Location

842	Emergency Exit
843	Emergency Exit

NOTE: Emergency Exit Door must be open to perform the inspection.

SUBTASK 53-05-02-211-035

- (2) Do a Detailed inspection of the door stops attached to the forward and aft edge frames, six fittings per door.

See Doc. D626A001-DTR, DTR check form 53-40-23-1 for alternative inspections.

SUBTASK 53-05-02-410-019

- (3) Close these access panels on the Left side:

Number Name/Location

832	Emergency Exit
833	Emergency Exit

Close these access panels on the Right side:

Number Name/Location

842	Emergency Exit
843	Emergency Exit

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS
		D633A109-AKS 53-695-00-02

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS			BOEING CARD NO.
DATE	TASK DETAILED				53-696-00-01 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE 800 900 900ER ALL
		ACCESS 832 833			ZONE 231
		NOTE			

Inspect (Detailed) the door stop intercostals (three locations) between STA 601 to STA 616.

See Doc. D626A001-DTR, DTR check form 53-40-23-2 for alternative inspections.

ACCESS NOTE: Emergency Exit Door must be open to perform this inspection. Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS D633A109-AKS 53-696-00-01
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-696-00-01		
					MECH	INSP

TASK 53-05-02-211-836

1. INTERNAL - DETAILED: OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS

A. Inspection

SUBTASK 53-05-02-010-022

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
832	Emergency Exit
833	Emergency Exit

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
842	Emergency Exit
843	Emergency Exit

NOTE: Emergency Exit Door must be open to perform this inspection. Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform the inspection.

SUBTASK 53-05-02-211-036

(2) Do a Detailed inspection of the door stop intercostals (three locations) between STA 601 to STA 616.

See Doc. D626A001-DTR, DTR check form 53-40-23-2 for alternative inspections.

SUBTASK 53-05-02-410-020

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
832	Emergency Exit
833	Emergency Exit

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
842	Emergency Exit
843	Emergency Exit

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS	D633A109-AKS 53-696-00-01	Page 2 of 2 Feb 15/2015
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS			BOEING CARD NO.
DATE	TASK DETAILED				53-696-00-02
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	RELATED CARD
STATION	SKILL AIRPL			APPLICABILITY AIRPLANE 800 900 900ER ENGINE ALL	
		ACCESS 842 843		ZONE 232	
		NOTE			

Inspect (Detailed) the door stop intercostals (three locations) between STA 601 to STA 616.

See Doc. D626A001-DTR, DTR check form 53-40-23-2 for alternative inspections.

ACCESS NOTE: Emergency Exit Door must be open to perform this inspection. Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS D633A109-AKS 53-696-00-02	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-696-00-02
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TASK 53-05-02-211-836

1. INTERNAL - DETAILED: OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS

A. Inspection

SUBTASK 53-05-02-010-022

- (1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
---------------	----------------------

832 Emergency Exit

833 Emergency Exit

Open these access panels on

<u>Number</u>	<u>Name/Location</u>
---------------	----------------------

842 Emergency Exit

843 Emergency Exit

NOTE: Emergency Exit Door must be open to perform this inspection. Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform the inspection.

SUBTASK 53-05-02-211-036

- (2) Do a Detailed inspection of the door stop intercostals (three locations) between STA 601 to STA 616.

See Doc. D626A001-DTR, DTR check form 53-40-23-2 for alternative inspections.

SUBTASK 53-05-02-410-020

- (3) Close these access panels on the Left side:

Number **Name/Location**

832 Emergency Exit

833 Emergency Exit

Close these access panels on

<u>Number</u>	<u>Name/Location</u>
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842 Emergency Exit

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS	D633A109-AKS 53-696-00-02	Page 2 of 2 Feb 15/2015
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-697-00-01 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 800 900 900ER NOTE ENGINE ALL
		ACCESS 832 833			
		NOTE			
			ZONE 231		

Inspect (High Frequency Eddy Current) the window frame edge inboard of the fasteners common to the door stop backup fitting attachment at stringers S-11 and S-12.

See Doc. D626A001-DTR, DTR check form 53-40-23-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-24.

AIRPLANE NOTE: For the 737-600 and -700, STA 616.

For the 737-800, STA 578.

ACCESS NOTE: Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS D633A109-AKS 53-697-00-01	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-697-00-01
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TASK 53-05-02-250-884

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: OVERWING EXIT DOOR STOPS AND FITTINGS**A. Inspection**

SUBTASK 53-05-02-010-095

- (1) Open these access panels on the Left side:

Number Name/Location

832	Emergency Exit
833	Emergency Exit
842	Emergency Exit
843	Emergency Exit

Open these access panels on the Right side:

Number Name/Location

842	Emergency Exit
843	Emergency Exit

NOTE: Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform the inspection.

SUBTASK 53-05-02-250-084

- (2) Do a High Frequency Eddy Current inspection of the window frame edge inboard of the fasteners common to the door stop backup fitting attachment at stringers S-11 and S-12.

See Doc. D626A001-DTR, DTR check form 53-40-23-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-24.

SUBTASK 53-05-02-410-093

- (3) Close these access panels on the Left side:

Number Name/Location

832	Emergency Exit
833	Emergency Exit

Close these access panels on the Right side:

Number Name/Location

842	Emergency Exit
843	Emergency Exit

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS D633A109-AKS 53-697-00-01
		Page 2 of 2 Jun 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-697-00-02 RELATED CARD
TAIL NUMBER	WORK AREA EMERGENCY EXIT	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 800 900 900ER NOTE ENGINE ALL
		ACCESS 842 843			
		NOTE			
			ZONE 232		

Inspect (High Frequency Eddy Current) the window frame edge inboard of the fasteners common to the door stop backup fitting attachment at stringers S-11 and S-12.

See Doc. D626A001-DTR, DTR check form 53-40-23-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-24.

AIRPLANE NOTE: For the 737-600 and -700, STA 616.

For the 737-800, STA 578.

ACCESS NOTE: Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS D633A109-AKS 53-697-00-02	Page 1 of 2 Jun 15/2015
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-697-00-02
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TASK 53-05-02-250-884

1. INTERNAL - SPECIAL DETAILED: OVERWING EXIT DOOR STOPS AND FITTINGS

A. Inspection

SUBTASK 53-05-02-010-095

- (1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
832	Emergency Exit
833	Emergency Exit
842	Emergency Exit
843	Emergency Exit

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
842	Emergency Exit
843	Emergency Exit

NOTE: Removal of passenger cabin sidewall lining between Emergency Exit Doors is required to perform the inspection.

SUBTASK 53-05-02-250-084

- (2) Do a High Frequency Eddy Current inspection of the window frame edge inboard of the fasteners common to the door stop backup fitting attachment at stringers S-11 and S-12

See Doc. D626A001-DTR, DTR check form 53-40-23-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-24.

SUBTASK 53-05-02-410-093

- (3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
832	Emergency Exit
833	Emergency Exit

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
842	Emergency Exit
843	Emergency Exit

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT OVERWING EMERGENCY EXIT DOOR STOPS AND FITTINGS	D633A109-AKS 53-697-00-02	Page 2 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE PRESSURE DECK ATTACHMENTS TO REAR SPAR EXTENSION			BOEING CARD NO. 53-697-10-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA WHEELWELL	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 133 134

Inspect (General Visual) the angle between the rear spar extension and the pressure deck from the AFT or FWD side, including the bend radius.

Note: Either the AFT or FWD side inspection may be performed.

See Doc. D626A001-DTR, DTR check form 53-40-24-1 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	PRESSURE DECK ATTACHMENTS TO REAR SPAR EXTENSION D633A109-AKS 53-697-10-01	Page 1 of 2 Jun 15/2016
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-697-10-01
TASK 53-05-02-210-837				MECH INSP

1. INTERNAL - GENERAL VISUAL: PRESSURE DECK ATTACHMENTS TO REAR SPAR EXTENSION

A. Inspection

SUBTASK 53-05-02-210-037

(1) Do a General Visual inspection of the angle between the rear spar extension and the pressure deck from the AFT or FWD side, including the bend radius.

NOTE: Either the AFT or FWD side inspection may be performed.

See Doc. D626A001-DTR, DTR check form 53-40-24-1 for alternative inspections.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	PRESSURE DECK ATTACHMENTS TO REAR SPAR EXTENSION	
		D633A109-AKS 53-697-10-01	Page 2 of 2 Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CROWN SKIN PANEL - STA 727 TO 887			BOEING CARD NO. 53-698-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE			ZONE 241 242	

Inspect (Detailed) the skin around all of the fastener locations from stringer S-10L to S-10R, from STA 727 to STA 887, except at the lap splices and antennas. (PSE 53-60-01-2).

See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.

ACCESS NOTE: Remove Dorsal Fin as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL - STA 727 TO 887
		D633A109-AKS 53-698-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-698-00-01	MECH	INSP
TASK 53-05-02-211-867						
1. EXTERNAL - DETAILED: CROWN SKIN PANEL STA 727 TO STA 887						
A. Inspection						
NOTE: Remove Dorsal Fin as required to perform the inspection.						
SUBTASK 53-05-02-211-067						
(1) Do a Detailed inspection of the skin around all of the fastener locations from stringer S-10L to S-10R, from STA 727 to STA 887, except at the lap splices and antennas. (PSE 53-60-01-2).						
See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.						
———— END OF TASK ————						

EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL - STA 727 TO 887
		D633A109-AKS 53-698-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CROWN SKIN PANEL STA 727 TO 887			BOEING CARD NO. 53-699-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE		ZONE 241 242		

Inspect (Detailed) the crown skin panel at the ADF Antenna cutout (STA 727+9, RBL 5) and the SATCOM Antenna cutout (STA 747, stringer S-1).

See Doc. D626A001-DTR, DTR check form 53-60-01-4 for alternative inspections.

ACCESS NOTE: Removal of antenna, fairing and base plate as required to expose the skin to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL STA 727 TO 887
		D633A109-AKS 53-699-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-699-00-01
TASK 53-05-02-211-837				MECH INSP
1. INTERNAL - DETAILED: CROWN SKIN PANEL STA 727 TO 887				
A. Inspection				
<p><u>NOTE:</u> Removal of ADT Antenna, fairing and base plate as required to expose the skin to perform the inspection.</p> <p>SUBTASK 53-05-02-211-037</p> <p>(1) Do a Detailed inspection of the crown skin panel at the ADF Antenna cutout (STA 727+9, RBL 5) and the SATCOM Antenna cutout (STA 747, stringer S-1).</p> <p>See Doc. D626A001-DTR, DTR check form 53-60-01-4 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
CROWN SKIN PANEL STA 727 TO 887				
D633A109-AKS 53-699-00-01				
Page 2 of 2 Feb 15/2015				

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE SATCOM ANTENNA INSTALLATION (CANADIAN MARCONI AND HONEYWELL)			BOEING CARD NO. 53-700-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 241 242

Inspect (Low Frequency Eddy Current) the skin for hidden cracks between the adaptor plate and stringers S-1 and S-2L.

See Doc. D626A001-DTR, DTR check form 53-60-01-5 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin ceiling panels and air conditioning duct as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	SATCOM ANTENNA INSTALLATION (CANADIAN MARCONI AND HONEYWELL) D633A109-AKS 53-700-00-01
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-700-00-01
				MECH INSP
TASK 53-05-02-250-885				
1. INTERNAL - SPECIAL DETAILED: SATCOM ANTENNA INSTALLATION (CANADIAN MARCONI AND HONEYWELL)				
<p>A. Inspection</p> <p><u>NOTE:</u> Remove or displace passenger cabin ceiling panels and air conditioning duct as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-085</p> <p>(1) Do a Low Frequency Eddy Current inspection of the skin for hidden cracks between the adaptor plate and stringers S-1 and S-2L.</p> <p>See Doc. D626A001-DTR, DTR check form 53-60-01-5 for alternative inspections.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	SATCOM ANTENNA INSTALLATION (CANADIAN MARCONI AND HONEYWELL)		
		D633A109-AKS 53-700-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE SATCOM ANTENNA INSTALLATION (CANADIAN MARCONI AND HONEYWELL)			BOEING CARD NO.
DATE	TASK DETAILED				53-701-00-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 241 242

Inspect (Detailed) the skin under the antenna adaptor plate from stringers S-1 to S-2L at STA 767.

See Doc. D626A001-DTR, DTR check form 53-60-01-5 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin ceiling panels and air conditioning duct as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	SATCOM ANTENNA INSTALLATION (CANADIAN MARCONI AND HONEYWELL)
		D633A109-AKS 53-701-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-701-00-01
				MECH INSP
TASK 53-05-02-211-838				
1. INTERNAL - DETAILED: SATCOM ANTENNA INSTALLATION (CANADIAN MARCONI AND HONEYWELL)				
<p>A. Inspection</p> <p><u>NOTE:</u> Remove or displace passenger cabin ceiling panels and air conditioning duct as required to perform the inspection.</p> <p>SUBTASK 53-05-02-211-038</p> <p>(1) Do a Detailed inspection of the skin under the antenna adaptor plate from stringers S-1 to S-2L at STA 767.</p> <p>See Doc. D626A001-DTR, DTR check form 53-60-01-5 for alternative inspections.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	SATCOM ANTENNA INSTALLATION (CANADIAN MARCONI AND HONEYWELL)		
		D633A109-AKS 53-701-00-01		
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE SATCOM ANTENNA INSTALLATION (ALL EXCEPT THOSE COVERED BY PSE 53-60-01-5)			BOEING CARD NO. 53-702-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS NOTE			ZONE 241 242

Inspect (Low Frequency Eddy Current) the skin for hidden cracks between the adaptor plate and stringers S-1 and S-2L.

See Doc. D626A001-DTR, DTR check form 53-60-01-6 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin ceiling panels and air conditioning duct as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	SATCOM ANTENNA INSTALLATION (ALL EXCEPT THOSE COVERED BY PSE 53-60-01-5)
		D633A109-AKS 53-702-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-702-00-01
				MECH INSP
TASK 53-05-02-250-886				
1. INTERNAL - SPECIAL DETAILED: SATCOM ANTENNA INSTALLATION (ALL EXCEPT THOSE COVERED BY PSE 53-60-01-5)				
A. Inspection				
NOTE: Remove or displace passenger cabin ceiling panels and air conditioning duct as required to perform the inspection				
SUBTASK 53-05-02-250-086				
(1) Do a Low Frequency Eddy Current inspection of the skin for hidden cracks between the adaptor plate and stringers S-1 and S-2L.				
See Doc. D626A001-DTR, DTR check form 53-60-01-6 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	SATCOM ANTENNA INSTALLATION (ALL EXCEPT THOSE COVERED BY PSE 53-60-01-5)		
		D633A109-AKS 53-702-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE SATCOM ANTENNA INSTALLATION (ALL EXCEPT THOSE COVERED BY PSE 53-60-01-5)			BOEING CARD NO. 53-702-01-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 12000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE		ZONE 241 242		

Inspect (Detailed) the skin under the antenna adaptor plate from stringers S-1 to S-2L at STA 767.

See Doc. D626A001-DTR, DTR check form 53-60-01-6 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin ceiling panels and air conditioning duct as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	SATCOM ANTENNA INSTALLATION (ALL EXCEPT THOSE COVERED BY PSE 53-60-01-5)
		D633A109-AKS 53-702-01-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-702-01-01
				MECH INSP
TASK 53-05-02-211-839				
1. INTERNAL - DETAILED: SATCOM ANTENNA INSTALLATION (ALL EXCEPT THOSE COVERED BY PSE 53-60-01-5)				
A. Inspection				
NOTE: Remove or displace passenger cabin ceiling panels and air conditioning duct as required to perform the inspection.				
SUBTASK 53-05-02-211-039				
(1) Do a Detailed inspection of the skin under the antenna adaptor plate from stringers S-1 to S-2L at STA 767.				
See Doc. D626A001-DTR, DTR check form 53-60-01-6 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	SATCOM ANTENNA INSTALLATION (ALL EXCEPT THOSE COVERED BY PSE 53-60-01-5)		
		D633A109-AKS 53-702-01-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT FUSELEAGE SIDE SKIN PANELS			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-703-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 194AL 194FL 194GR			ZONE 141 194 241
		NOTE			

Inspect (General Visual) the skin from STA 727 to STA 887 between stringers S-14 to S-17. (PSE 53-60-02-1).

See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FUSELEAGE SIDE SKIN PANELS
		D633A109-AKS 53-703-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-703-00-01										
				MECH INSP										
TASK 53-05-02-210-805														
1. EXTERNAL - GENERAL VISUAL: FUSELAGE SIDE SKIN PANELS														
A. Inspection														
SUBTASK 53-05-02-010-012														
(1) Open these access panels:														
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>194AL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194FL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GR</td><td>Aft Wing To Body Fairing Panel</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	194AL	Aft Wing To Body Fairing Panel	194FL	Aft Wing To Body Fairing Panel	194GL	Aft Wing To Body Fairing Panel	194GR	Aft Wing To Body Fairing Panel
<u>Number</u>	<u>Name/Location</u>													
194AL	Aft Wing To Body Fairing Panel													
194FL	Aft Wing To Body Fairing Panel													
194GL	Aft Wing To Body Fairing Panel													
194GR	Aft Wing To Body Fairing Panel													
<u>NOTE:</u> Remove or displace wing to body fairings as required to perform the inspection.														
SUBTASK 53-05-02-210-005														
(2) Do a General Visual inspection of the skin from STA 727 to STA 887 between stringers S-14 to S-17. (PSE 53-60-02-1).														
See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.														
SUBTASK 53-05-02-410-010														
(3) Close these access panels:														
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>194AL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194FL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GR</td><td>Aft Wing To Body Fairing Panel</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	194AL	Aft Wing To Body Fairing Panel	194FL	Aft Wing To Body Fairing Panel	194GL	Aft Wing To Body Fairing Panel	194GR	Aft Wing To Body Fairing Panel
<u>Number</u>	<u>Name/Location</u>													
194AL	Aft Wing To Body Fairing Panel													
194FL	Aft Wing To Body Fairing Panel													
194GL	Aft Wing To Body Fairing Panel													
194GR	Aft Wing To Body Fairing Panel													
———— END OF TASK ————														

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FUSELAGE SIDE SKIN PANELS
		D633A109-AKS 53-703-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT FUSELEAGE SIDE SKIN PANELS			BOEING CARD NO. 53-703-00-02
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 194AL 194FL 194GL			ZONE 142 194 242
		NOTE			

Inspect (General Visual) the skin from STA 727 to STA 887 between stringers S-14 to S-17. (PSE 53-60-02-1).

See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT FUSELEAGE SIDE SKIN PANELS
		D633A109-AKS 53-703-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-703-00-02										
TASK 53-05-02-210-805				MECH INSP										
1. EXTERNAL - GENERAL VISUAL: FUSELAGE SIDE SKIN PANELS														
A. Inspection														
SUBTASK 53-05-02-010-012														
(1) Open these access panels:														
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>194AL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194FL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GR</td><td>Aft Wing To Body Fairing Panel</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	194AL	Aft Wing To Body Fairing Panel	194FL	Aft Wing To Body Fairing Panel	194GL	Aft Wing To Body Fairing Panel	194GR	Aft Wing To Body Fairing Panel
<u>Number</u>	<u>Name/Location</u>													
194AL	Aft Wing To Body Fairing Panel													
194FL	Aft Wing To Body Fairing Panel													
194GL	Aft Wing To Body Fairing Panel													
194GR	Aft Wing To Body Fairing Panel													
NOTE: Remove or displace wing to body fairings as required to perform the inspection.														
SUBTASK 53-05-02-210-005														
(2) Do a General Visual inspection of the skin from STA 727 to STA 887 between stringers S-14 to S-17. (PSE 53-60-02-1).														
See Doc D626A001-DTR, DTR check form 53-30-02-1 for alternative inspections.														
SUBTASK 53-05-02-410-010														
(3) Close these access panels:														
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>194AL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194FL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GR</td><td>Aft Wing To Body Fairing Panel</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	194AL	Aft Wing To Body Fairing Panel	194FL	Aft Wing To Body Fairing Panel	194GL	Aft Wing To Body Fairing Panel	194GR	Aft Wing To Body Fairing Panel
<u>Number</u>	<u>Name/Location</u>													
194AL	Aft Wing To Body Fairing Panel													
194FL	Aft Wing To Body Fairing Panel													
194GL	Aft Wing To Body Fairing Panel													
194GR	Aft Wing To Body Fairing Panel													
———— END OF TASK ————														

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT FUSELAGE SIDE SKIN PANELS
		D633A109-AKS 53-703-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING			BOEING CARD NO.
DATE	TASK DETAILED				53-703-10-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 8000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 194AL 194FL 194GL			ZONE 194
		NOTE			

Inspect (Detailed) the fuselage skin panels under the Wing to Body Fairing from STA 727 to STA 887. (53-60-02-4).

See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING	
		D633A109-AKS 53-703-10-01	Page 1 of 2 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-703-10-01
				MECH INSP
TASK 53-05-02-211-868				
1. EXTERNAL - DETAILED: FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING				
A. Inspection				
SUBTASK 53-05-02-010-037				
(1) Open these access panels on the Left side:				
Number Name/Location				
194AL Aft Wing To Body Fairing Panel				
194FL Aft Wing To Body Fairing Panel				
194GL Aft Wing To Body Fairing Panel				
Open these access panels on the Right side:				
Number Name/Location				
194AR Aft Wing To Body Fairing Panel				
194FR Aft Wing To Body Fairing Panel				
194GR Aft Wing To Body Fairing Panel				
NOTE: Remove or displace wing to body fairings as required to perform this inspection.				
SUBTASK 53-05-02-211-068				
(2) Do a Detailed inspection of the fuselage skin panels under the Wing to Body Fairing from STA 727 to STA 887. (53-60-02-4).				
See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.				
SUBTASK 53-05-02-410-035				
(3) Close these access panels on the Left side:				
Number Name/Location				
194AL Aft Wing To Body Fairing Panel				
194FL Aft Wing To Body Fairing Panel				
194GL Aft Wing To Body Fairing Panel				
Close these access panels on the Right side:				
Number Name/Location				
194AR Aft Wing To Body Fairing Panel				
194FR Aft Wing To Body Fairing Panel				
194GR Aft Wing To Body Fairing Panel				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING	D633A109-AKS 53-703-10-01	Page 2 of 2 Feb 15/2015
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING			BOEING CARD NO.
DATE	TASK DETAILED				53-703-10-02 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 8000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 194AR 194FR 194GR			ZONE 194
		NOTE			

Inspect (Detailed) the fuselage skin panels under the Wing to Body Fairing from STA 727 to STA 887. (53-60-02-4).

See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.

ACCESS NOTE: Remove or displace wing to body fairings as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING
		D633A109-AKS 53-703-10-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-703-10-02
TASK 53-05-02-211-868				MECH INSP

1. EXTERNAL - DETAILED: FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING

A. Inspection

SUBTASK 53-05-02-010-037

(1) Open these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
194AL	Aft Wing To Body Fairing Panel
194FL	Aft Wing To Body Fairing Panel
194GL	Aft Wing To Body Fairing Panel

Open these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
194AR	Aft Wing To Body Fairing Panel
194FR	Aft Wing To Body Fairing Panel
194GR	Aft Wing To Body Fairing Panel

NOTE: Remove or displace wing to body fairings as required to perform this inspection.

SUBTASK 53-05-02-211-068

(2) Do a Detailed inspection of the fuselage skin panels under the Wing to Body Fairing from STA 727 to STA 887. (53-60-02-4).

See Doc D626A001-DTR, DTR check form 53-30-02-4 for alternative inspections.

SUBTASK 53-05-02-410-035

(3) Close these access panels on the Left side:

<u>Number</u>	<u>Name/Location</u>
194AL	Aft Wing To Body Fairing Panel
194FL	Aft Wing To Body Fairing Panel
194GL	Aft Wing To Body Fairing Panel

Close these access panels on the Right side:

<u>Number</u>	<u>Name/Location</u>
194AR	Aft Wing To Body Fairing Panel
194FR	Aft Wing To Body Fairing Panel
194GR	Aft Wing To Body Fairing Panel

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT FUSELAGE SIDE SKIN PANELS UNDER THE WING-TO-BODY FAIRING
		D633A109-AKS 53-703-10-02

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-704-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL
STATION	SKILL AIRPL				800 900 900ER
		ACCESS			ZONE 241

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 727 to STA 887. (PSE 53-60-04-1).

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-704-00-01	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-704-00-01
				MECH INSP
TASK 53-05-02-250-940				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE				
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-140</p> <p>(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 727 to STA 887. (PSE 53-60-04-1). See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.</p>				
———— END OF TASK ————				
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE	
			D633A109-AKS 53-704-00-01	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-704-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 242

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 727 to STA 887. (PSE 53-60-04-1).

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-704-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-704-00-02
TASK 53-05-02-250-940				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-140				
(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 727 to STA 887. (PSE 53-60-04-1). See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-704-00-02	Page 2 of 2 Feb 15/2015	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-705-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL
STATION	SKILL AIRPL				800 900 900ER
		ACCESS			ZONE 241

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 727 to STA 887. (PSE 53-60-04-2).

See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-705-00-01	Page 1 of 2 Jun 15/2015
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-705-00-01		
					MECH	INSP
TASK 53-05-02-250-942						
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE						
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-142</p> <p>(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 727 to STA 887. (PSE 53-60-04-2). See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.</p>						
———— END OF TASK ————						

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-705-00-01

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-705-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 242

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 727 to STA 887. (PSE 53-60-04-2).

See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-705-00-02	Page 1 of 2 Jun 15/2015
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-705-00-02	
				MECH	INSP
TASK 53-05-02-250-942					
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE					
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-142</p> <p>(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 727 to STA 887. (PSE 53-60-04-2). See Doc. D626A001-DTR, DTR check form 53-30-04-2, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.</p>					
———— END OF TASK ————					

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-705-00-02

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLICER, STRINGER S-10L			BOEING CARD NO. 53-706-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL
STATION	SKILL AIRPL				800 900 900ER
		ACCESS			ZONE 241

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-3)

See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICING, STRINGER S-10L
		D633A109-AKS 53-706-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-706-00-01
				MECH INSP
TASK 53-05-02-250-944				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-144				
(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-3) See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.				
— END OF TASK —				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT LONGITUDINAL LAP SPLICE, STRINGER S-10L				
D633A109-AKS 53-706-00-01				
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AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLICER, STRINGER S-10R			BOEING CARD NO. 53-706-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 242

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-3)

See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICING, STRINGER S-10R
		D633A109-AKS 53-706-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-706-00-02
				MECH INSP
TASK 53-05-02-250-944				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-144				
(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-3) See Doc. D626A001-DTR, DTR check form 53-30-04-3, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.				
— END OF TASK —				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT LONGITUDINAL LAP SPLIC, STRINGER S-10R				
D633A109-AKS 53-706-00-02				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO.
DATE	TASK DETAILED				53-707-00-01
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE 600 700 700IGW 800 900 900ER ENGINE ALL
		ACCESS NOTE			ZONE 241

Inspect (Detailed) the lower skin along the lower fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-4)

See Doc. D626A001-DTR, DTR check form 53-30-04-4, for alternative inspections.

ACCESS NOTE: Removal or displacement of interior sidewall panels and insulation blankets are required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-707-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-707-00-01
TASK 53-05-02-211-869				MECH INSP
1. INTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
<p><u>NOTE:</u> Removal or displacement of interior sidewall panels and insulation blankets are required.</p> <p>SUBTASK 53-05-02-211-069</p> <p>(1) Do a Detailed inspection of the lower skin along the lower fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-4)</p> <p>See Doc. D626A001-DTR, DTR check form 53-30-04-4, for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT LONGITUDINAL LAP SPLICE				
D633A109-AKS 53-707-00-01				
Page 2 of 2				
Feb 15/2015				

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO.
DATE	TASK DETAILED				53-707-00-02
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY AIRPLANE 600 700 700IGW 800 900 900ER ENGINE ALL
		ACCESS NOTE			ZONE 242

Inspect (Detailed) the lower skin along the lower fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-4)

See Doc. D626A001-DTR, DTR check form 53-30-04-4, for alternative inspections.

ACCESS NOTE: Removal or displacement of interior sidewall panels and insulation blankets are required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-707-00-02 Page 1 of 2 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-707-00-02
TASK 53-05-02-211-869				MECH INSP
1. INTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
<p><u>NOTE:</u> Removal or displacement of interior sidewall panels and insulation blankets are required.</p> <p>SUBTASK 53-05-02-211-069</p> <p>(1) Do a Detailed inspection of the lower skin along the lower fastener row at stringers S-10L and S-10R from STA 727 to STA 887. (PSE 53-60-04-4)</p> <p>See Doc. D626A001-DTR, DTR check form 53-30-04-4, for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT LONGITUDINAL LAP SPLICE				
D633A109-AKS 53-707-00-02				
Page 2 of 2 Feb 15/2015				

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-708-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 700IGW 800 900 900ER ENGINE ALL
		ACCESS			ZONE 241

Inspect (Detailed) the upper skin along the upper fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-5)

See Doc. D626A001-DTR, DTR check form 53-30-04-5, for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-708-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-708-00-01
TASK 53-05-02-211-871				MECH INSP
1. EXTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-211-071				
(1) Do a Detailed inspection of the upper skin along the upper fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-5) See Doc. D626A001-DTR, DTR check form 53-30-04-5, for alternative inspections.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-708-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-708-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 700IGW 800 900 900ER ENGINE ALL
		ACCESS			ZONE 242

Inspect (Detailed) the upper skin along the upper fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-5)

See Doc. D626A001-DTR, DTR check form 53-30-04-5, for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-708-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-708-00-02
TASK 53-05-02-211-871				MECH INSP
1. EXTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-211-071				
(1) Do a Detailed inspection of the upper skin along the upper fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-5) See Doc. D626A001-DTR, DTR check form 53-30-04-5, for alternative inspections.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-708-00-02	Page 2 of 2 Feb 15/2015	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-709-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 241

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-6)

See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-709-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-709-00-01
				MECH INSP
TASK 53-05-02-250-946				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-146				
(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-6). See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL				
SOURCE AWL				
LEFT LONGITUDINAL LAP SPLICE				
D633A109-AKS 53-709-00-01				
Page 2 of 2 Jun 15/2015				

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-709-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 242

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-6)

See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-709-00-02	Page 1 of 2 Jun 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-709-00-02
TASK 53-05-02-250-946				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-146				
(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-14L and S-14R from STA 727 to STA 887. (PSE 53-60-04-6). See Doc. D626A001-DTR, DTR check form 53-30-04-6, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-709-00-02	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-710-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 141
<p>Inspect ((Low Frequency Eddy Current) the upper (inner) skin along the upper fastener row at stringers S-23L and S-23R from STA 727 to STA 887, except at the cargo door cutout.</p> <p>See Doc. D626A001-DTR, DTR check form 53-60-04-7, for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.</p>					
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE D633A109-AKS 53-710-00-01		
Page 1 of 2 Jun 15/2016					

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-710-00-01
				MECH INSP
TASK 53-05-02-250-887				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE				
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-887</p> <p>(1) Do a Low Frequency Eddy Current inspection of the upper (inner) skin along the upper fastener row at stringers S-23L and S-23R from STA 727 to STA 887, except at the cargo door cutout.</p> <p>See Doc. D626A001-DTR, DTR check form 53-60-04-7, for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.</p> <p style="text-align: center;">— END OF TASK —</p>				
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT LONGITUDINAL LAP SPLICE</p> <p>D633A109-AKS 53-710-00-01</p>				

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-710-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 142
<p>Inspect ((Low Frequency Eddy Current) the upper (inner) skin along the upper fastener row at stringers S-23L and S-23R from STA 727 to STA 887, except at the cargo door cutout.</p> <p>See Doc. D626A001-DTR, DTR check form 53-60-04-7, for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.</p>					
EFFECTIVITY AKS ALL		SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE D633A109-AKS 53-710-00-02		
Page 1 of 2 Jun 15/2016					

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-710-00-02	MECH	INSP
TASK 53-05-02-250-887						
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE						
A. Inspection						
SUBTASK 53-05-02-250-087						
(1) Do a Low Frequency Eddy Current inspection of the upper (inner) skin along the upper fastener row at stringers S-23L and S-23R from STA 727 to STA 887, except at the cargo door cutout.						
See Doc. D626A001-DTR, DTR check form 53-60-04-7, for alternative inspections.						
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.						
———— END OF TASK ————						

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-710-00-02	Page 2 of 2 Jun 15/2015
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WINDOW BELT STA 727 TO 888			BOEING CARD NO.
DATE	TASK DETAILED				53-711-00-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 241

Inspect (Detailed) the window frames around each window from STA 727 to STA 888.

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets as required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 727 TO 888
		D633A109-AKS 53-711-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-711-00-01
				MECH INSP
TASK 53-05-02-211-840				
1. INTERNAL - DETAILED: WINDOW BELT STA 727 TO STA 888				
A. Inspection				
NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets as required.				
SUBTASK 53-05-02-211-040				
(1) Do a Detailed inspection of the window frames around each window from STA 727 to STA 888.				
See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 727 TO 888		
		D633A109-AKS 53-711-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT WINDOW BELT STA 727 TO 888			BOEING CARD NO.
DATE	TASK DETAILED				53-711-00-02
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	RELATED CARD
STATION	SKILL AIRPL				APPLICABILITY
		ACCESS NOTE			AIRPLANE ALL ENGINE ALL
					ZONE 242

Inspect (Detailed) the window frames around each window from STA 727 to STA 888.

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets as required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 727 TO 888
		D633A109-AKS 53-711-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-711-00-02
				MECH INSP
TASK 53-05-02-211-840				
1. INTERNAL - DETAILED: WINDOW BELT STA 727 TO STA 888				
A. Inspection				
NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets as required.				
SUBTASK 53-05-02-211-040				
(1) Do a Detailed inspection of the window frames around each window from STA 727 to STA 888.				
See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 727 TO 888		
		D633A109-AKS 53-711-00-02	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT WINDOW BELT STA 727 TO 888			BOEING CARD NO. 53-711-01-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 241

Inspect (Detailed) the window frames around each window from STA 727 to STA 888.

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 727 TO 888
		D633A109-AKS 53-711-01-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-711-01-01
				MECH INSP
TASK 53-05-02-211-841				
1. EXTERNAL - DETAILED: WINDOW BELT STA 727 TO STA 888				
A. Inspection				
SUBTASK 53-05-02-211-041				
(1) Do a Detailed inspection of the window frames around each window from STA 727 to STA 888.				
See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 727 TO 888
		D633A109-AKS 53-711-01-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT WINDOW BELT STA 727 TO 888			BOEING CARD NO. 53-711-01-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS		ZONE 242		

Inspect (Detailed) the window frames around each window from STA 727 to STA 888.

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 727 TO 888
		D633A109-AKS 53-711-01-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-711-01-02
TASK 53-05-02-211-841				MECH INSP
1. EXTERNAL - DETAILED: WINDOW BELT STA 727 TO STA 888				
A. Inspection				
SUBTASK 53-05-02-211-041				
(1) Do a Detailed inspection of the window frames around each window from STA 727 to STA 888. See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 727 TO 888		
		D633A109-AKS 53-711-01-02	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-712-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 822			
		NOTE			
ZONE 142 144					

Inspect (Low Frequency Eddy Current) around the fasteners common to the web at the lower main sill chords between STA 807 and STA 827. (PSE 53-60-08).

See Doc. D626A001-DTR, DTR check form 53-30-08-12, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-33.

ACCESS NOTE: Removal of aft cargo door scuff plate is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-712-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-712-00-01				
				MECH INSP				
TASK 53-05-02-250-948								
1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND STRUCTURE								
A. Inspection								
SUBTASK 53-05-02-010-113								
(1) Open this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
NOTE: Removal of aft cargo door scuff plate is required to perform the inspection.								
SUBTASK 53-05-02-250-148								
(2) Do a Low Frequency Eddy Current inspection around the fasteners common to the web at the lower main sill chords between STA 807 and STA 827. (PSE 53-60-08).								
See Doc. D626A001-DTR, DTR check form 53-30-08-12, for alternative inspections.								
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-33.								
SUBTASK 53-05-02-410-111								
(3) Close this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
— END OF TASK —								

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-712-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-713-00-01 RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 822			
		NOTE			
			ZONE 142 144		

Inspect (High Frequency Eddy Current) the upper sill outer chord around the fasteners common to the chord and bearstrap. (PSE 53-60-08).

See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections.

ACCESS NOTE: Remove or displace aft cargo door lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-713-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-713-00-01				
				MECH INSP				
TASK 53-05-02-250-949								
1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND STRUCTURE								
<p>A. Inspection</p> <p>SUBTASK 53-05-02-010-115</p> <p>(1) Open this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>822</td> <td>Aft Cargo Door</td> </tr> </tbody> </table> <p><u>NOTE:</u> Remove or displace aft cargo door lining as required to perform the inspection.</p>				<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door	
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
<p>SUBTASK 53-05-02-250-149</p> <p>(2) Do a High Frequency Eddy Current inspection of the upper sill outer chord around the fasteners common to the chord and bearstrap. (PSE 53-60-08).</p> <p>See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections.</p>								
<p>SUBTASK 53-05-02-410-113</p> <p>(3) Close this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>822</td> <td>Aft Cargo Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door	
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
———— END OF TASK ————								

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-713-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-713-01-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS 822			
		NOTE			

Inspect (High Frequency Eddy Current) the bearstrap along the upper edge of the aft cargo door. (53-60-08).

See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections.

ACCESS NOTE: Remove or displace aft cargo door lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-713-01-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-713-01-01				
				MECH INSP				
TASK 53-05-02-250-974								
1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND STRUCTURE								
<p>A. Inspection</p> <p>SUBTASK 53-05-02-010-117</p> <p>(1) Open this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>822</td> <td>Aft Cargo Door</td> </tr> </tbody> </table> <p><u>NOTE:</u> Remove or displace aft cargo door lining as required to perform the inspection.</p>				<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door	
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
<p>SUBTASK 53-05-02-250-174</p> <p>(2) Do a High Frequency Eddy Current inspection of the bearstrap along the upper edge of the aft cargo door. (53-60-08).</p> <p>See Doc. D626A001-DTR, DTR check form 53-30-08-10, for alternative inspections.</p>								
<p>SUBTASK 53-05-02-410-115</p> <p>(3) Close this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>822</td> <td>Aft Cargo Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door	
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
———— END OF TASK ————								

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-713-01-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-714-00-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 822			ZONE 142 144	
EFFECTIVITY AKS ALL		SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE D633A109-AKS 53-714-00-01			
						Page 1 of 2 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-714-00-01				
				MECH INSP				
TASK 53-05-02-250-950								
1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND STRUCTURE								
A. Inspection								
SUBTASK 53-05-02-010-114								
(1) Open this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
SUBTASK 53-05-02-250-150								
(2) Do a High Frequency Eddy Current inspection of the bearstrap for two inches on each side of stringer S-24R at STA 794.4 and STA 847. (PSE 53-60-08)								
See Doc D626A001-DTR, DTR check form 53-30-08-6 for alternative inspections.								
SUBTASK 53-05-02-410-112								
(3) Close this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
— END OF TASK —								
EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE						
		D633A109-AKS 53-714-00-01	Page 2 of 2 Feb 15/2015					

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-714-01-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 822			
			ZONE 142 144		

Inspect (Ultrasonic) the bearstrap for hidden damage under the stop backup fitting at stringer S-24R at STA 794.4 and STA 847. (PSE 53-60-08)

See Doc D626A001-DTR, DTR check form 53-30-08-6 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-714-01-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-714-01-01				
				MECH INSP				
TASK 53-05-02-130-809								
1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND STRUCTURE								
A. Inspection								
SUBTASK 53-05-02-010-008								
(1) Open this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
SUBTASK 53-05-02-130-009								
(2) Do an Ultrasonic inspection of the bearstrap for hidden damage under the stop backup fitting at stringer S-24R at STA 794.4 and STA 847. (PSE 53-60-08).								
See Doc D626A001-DTR, DTR check form 53-30-08-6 for alternative inspections.								
SUBTASK 53-05-02-410-006								
(3) Close this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
— END OF TASK —								
EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE						
		D633A109-AKS 53-714-01-01	Page 2 of 2 Feb 15/2015					

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-715-00-01 RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 822			
		NOTE			
			ZONE 142		

Inspect (High Frequency Eddy Current) the frame inner chord between the web and failsafe strap from stringers S-18R to S-26R.

See Doc D626A001-DTR, DTR check form 53-60-08-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-88.

ACCESS NOTE: Cargo Door must be open to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-715-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-715-00-01	MECH	INSP
TASK 53-05-02-250-888						

1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-096

- (1) Open this access panel:

Number Name/Location

822 Aft Cargo Door

NOTE: Cargo Door must be open to perform the inspection.

SUBTASK 53-05-02-250-088

- (2) Do a High Frequency Eddy Current inspection of the frame inner chord between the web and failsafe strap from stringers S-18R to S-26R.

See Doc D626A001-DTR, DTR check form 53-60-08-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-88.

SUBTASK 53-05-02-410-094

- (3) Close this access panel:

Number Name/Location

822 Aft Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE	
		D633A109-AKS 53-715-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-716-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS 822			
		NOTE			
			ZONE 142		

Inspect (High Frequency Eddy Current) the forward and aft edge frame inner chord fail-safe straps between stringers S-17R and S-18R.

See Doc D626A001-DTR, DTR check form 53-60-08-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-88.

ACCESS NOTE: Cargo Door must be open to perform the inspection. Remove or displace aft cargo sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-716-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-716-00-01				
				MECH INSP				
TASK 53-05-02-250-889								
1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND STRUCTURE								
A. Inspection								
SUBTASK 53-05-02-010-097								
(1) Open this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
NOTE: Cargo Door must be open to perform the inspection. Remove or displace aft cargo sidewall and ceiling lining as required to perform the inspection.								
SUBTASK 53-05-02-250-089								
(2) Do a High Frequency Eddy Current inspection of the forward and aft edge frame inner chord fail-safe straps between stringers S-17R and S-18R.								
See Doc D626A001-DTR, DTR check form 53-60-08-2 for alternative inspections.								
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-88.								
SUBTASK 53-05-02-410-095								
(3) Close this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
— END OF TASK —								

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-716-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-717-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS 822			
		NOTE			
					ZONE 142 144

Inspect (Low Frequency Eddy Current) both the forward and aft edge frame inner chords at stringer S-18R.

See Doc D626A001-DTR, DTR check form 53-60-08-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-81.

ACCESS NOTE: Remove or displace aft cargo sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-717-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-717-00-01				
				MECH INSP				
TASK 53-05-02-250-890								
1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND STRUCTURE								
A. Inspection								
SUBTASK 53-05-02-010-098								
(1) Open this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
<u>NOTE:</u> Remove or displace aft cargo sidewall and ceiling lining as required to perform the inspection.								
SUBTASK 53-05-02-250-090								
(2) Do a Low Frequency Eddy Current inspection of the forward and aft edge frame inner chords at stringer S-18R.								
See Doc D626A001-DTR, DTR check form 53-60-08-3 for alternative inspections.								
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-81.								
SUBTASK 53-05-02-410-096								
(3) Close this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>822</td><td>Aft Cargo Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
— END OF TASK —								

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-717-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT BS 794.37 AND 847			BOEING CARD NO. 53-718-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 822			ZONE 142 144
		NOTE			

Inspect (High Frequency Eddy Current) the outboard portion of the web on the outer chord between stringers S-16R and S-26R at STA 794.37 and STA 847 (except at door stops and sills location).

See Doc D626A001-DTR, DTR check form 53-60-08-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-22.

ACCESS NOTE: Aft cargo door must be open to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT BS 794.37 AND 847
		D633A109-AKS 53-718-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-718-00-01
				MECH INSP
TASK 53-05-02-250-891				
1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT BS 794.37 AND BS 847				
A. Inspection				
SUBTASK 53-05-02-010-099				
(1) Open this access panel:				
Number Name/Location				
822 Aft Cargo Door				
<u>NOTE:</u> Aft cargo door must be open to perform the inspection.				
SUBTASK 53-05-02-250-091				
(2) Do a High Frequency Eddy Current inspection of the outboard portion of the web on the outer chord between stringers S-16R and S-26R at STA 794.37 and STA 847 (except at door stops and sills location).				
See Doc D626A001-DTR, DTR check form 53-60-08-4 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-22.				
SUBTASK 53-05-02-410-097				
(3) Close this access panel:				
Number Name/Location				
822 Aft Cargo Door				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT BS 794.37 AND 847		
		D633A109-AKS 53-718-00-01	Page 2 of 2 Jun 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT BS 794.37 AND 847			BOEING CARD NO. 53-719-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 822			ZONE 142 144
		NOTE			

Inspect (Ultrasonic) the outboard portion of the frame web under the door stop fittings and sill clips at STA 794.37 and STA 847.

See Doc D626A001-DTR, DTR check form 53-60-08-5 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 53-10-08.

ACCESS NOTE: Aft cargo door must be open to perform the inspection. Remove or displace aft cargo lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT BS 794.37 AND 847
		D633A109-AKS 53-719-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-719-00-01
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TASK 53-05-02-130-806

- ## **1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT BS 794.37 AND BS 847**

A. Inspection

SUBTASK 53-05-02-010-006

- (1) Open this access panel:

Number **Name/Location**

822 Aft Cargo Door

NOTE: Aft cargo door must be open to perform the inspection. Remove or displace aft cargo lining as required to perform the inspection.

SUBTASK 53-05-02-130-006

- (2) Do an Ultrasonic inspection of the outboard portion of the frame web under the door stop fittings and sill clips at STA 794.37 and STA 847.

See Doc D626A001-DTR. DTR check form 53-60-08-5 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 53-10-08.

SUBTASK 53-05-02-410-004

- (3) Close this access panel:

Number Name/Location

822 Aft Cargo Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND FORWARD AND AFT EDGE FRAMES AT BS 794.37 AND 847
		D633A109-AKS 53-719-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-720-00-01 RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 822			
			ZONE 142 144		

Inspect (High Frequency Eddy Current) the exposed edge of the bearstrap at both the forward and aft edge of the door at STA 794.4 and STA 847 from stringers S-18R to S-25R.

See Doc D626A001-DTR, DTR check form 53-60-08-8 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-21.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-720-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-720-00-01
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TASK 53-05-02-250-892

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-100

- (1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
822	Aft Cargo Door

SUBTASK 53-05-02-250-092

- (2) Do a High Frequency Eddy Current inspection of the exposed edge of the bearstrap at both the forward and aft edge of the door at STA 794.4 and STA 847 from stringers S-18R to S-25R.

See Doc D626A001-DTR, DTR check form 53-60-08-8 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-21.

SUBTASK 53-05-02-410-098

- (3) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
822	Aft Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE	
		D633A109-AKS 53-720-00-01	Page 2 of 2 Jun 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-721-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 34000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 822			ZONE 142	
		NOTE				

Inspect (Detailed) the skin at all four corners (upper/lower/FWD/AFT) of the cargo door cutout.

See Doc. D626A001-DTR, DTR check form 53-60-08-9, for alternative inspections.

ACCESS NOTE: Scuff plate removal required.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-721-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-721-00-01
				MECH INSP
TASK 53-05-02-211-842				
1. EXTERNAL - DETAILED: AFT CARGO DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-023				
(1) Open this access panel:				
Number Name/Location				
822 Aft Cargo Door				
NOTE: Scuff plate removal required.				
SUBTASK 53-05-02-211-042				
(2) Do a Detailed inspection of the skin at all four corners (upper/lower/FWD/AFT) of the cargo door cutout.				
See Doc. D626A001-DTR, DTR check form 53-60-08-9, for alternative inspections.				
SUBTASK 53-05-02-410-021				
(3) Close this access panel:				
Number Name/Location				
822 Aft Cargo Door				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE		
		D633A109-AKS 53-721-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-721-10-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 34000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS 822			
		NOTE			

Inspect (General Visual) the bearstrap at all four corners (upper/lower/fwd/aft) of the cargo door cutout.

See Doc. D626A001-DTR, DTR check form 53-60-08-9, for alternative inspections.

ACCESS NOTE: Corner casting removal is required.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-721-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-721-10-01
				MECH INSP
TASK 53-05-02-210-804				
1. INTERNAL - GENERAL VISUAL: AFT CARGO DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-011				
(1) Open this access panel:				
Number Name/Location				
822 Aft Cargo Door				
NOTE: Corner casting removal is required.				
SUBTASK 53-05-02-210-004				
(2) Do a General Visual inspection of the bearstrap at all four corners (upper/lower/fwd/aft) of the cargo door cutout.				
See Doc. D626A001-DTR, DTR check form 53-60-08-9, for alternative inspections.				
SUBTASK 53-05-02-410-009				
(3) Close this access panel:				
Number Name/Location				
822 Aft Cargo Door				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE		
		D633A109-AKS 53-721-10-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-721-20-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 6000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE 600 700 800 900 900ER ENGINE ALL
		ACCESS 822			ZONE 142 144

Inspect (DET) the upper sill inner chord.

See Doc. D626A001-DTR, DTR check form 53-60-08-11, for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-721-20-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-721-20-01
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TASK 53-05-02-211-828

MECH

INSP

1. INTERNAL - DETAILED: AFT CARGO DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-142

- (1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
822	Aft Cargo Door

SUBTASK 53-05-02-211-028

- (2) Do a Detailed inspection of the of the upper sill inner chord.

See Doc. D626A001-DTR, DTR check form 53-60-08-11, for alternative inspections.

SUBTASK 53-05-02-410-134

- (3) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
822	Aft Cargo Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR SURROUND STRUCTURE
		D633A109-AKS 53-721-20-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR FITTINGS AND STOPS AT THE FORWARD AND AFT EDGE FRAMES			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-722-00-01 RELATED CARD
TAIL NUMBER	WORK AREA AFT CARGO	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 822			
		NOTE			
			ZONE 142		

Inspect (High Frequency Eddy Current) the intercostal web for cracks adjacent to the rivet and fastener holes. Five door stops locations on both the forward and aft edge frames.

See Doc. D626A001-DTR, DTR check form 53-60-09-3, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-82.

ACCESS NOTE: Aft cargo door must be open to perform the inspection. Remove or displace aft cargo sidewall lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT CARGO DOOR FITTINGS AND STOPS AT THE FORWARD AND AFT EDGE FRAMES
		D633A109-AKS 53-722-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-722-00-01									
					MECH INSP								
TASK 53-05-02-250-893													
1. INTERNAL - SPECIAL DETAILED: AFT CARGO DOOR FITTINGS AND STOPS AT THE FORWARD AND AFT EDGE FRAMES													
<p>A. Inspection</p> <p>SUBTASK 53-05-02-010-101</p> <p>(1) Open this access panel:</p> <table style="margin-left: 20px; border-collapse: collapse;"> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> <tr> <td style="padding-left: 10px;">822</td> <td style="padding-left: 10px;">Aft Cargo Door</td> </tr> </table> <p><u>NOTE:</u> Aft cargo door must be open to perform the inspection. Remove or displace aft cargo sidewall lining as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-093</p> <p>(2) Do a High Frequency Eddy Current inspection of the intercostal web for cracks adjacent to the rivet and fastener holes. Five door stops locations on both the forward and aft edge frames.</p> <p>See Doc. D626A001-DTR, DTR check form 53-60-09-3, for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-82.</p> <p>SUBTASK 53-05-02-410-099</p> <p>(3) Close this access panel:</p> <table style="margin-left: 20px; border-collapse: collapse;"> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> <tr> <td style="padding-left: 10px;">822</td> <td style="padding-left: 10px;">Aft Cargo Door</td> </tr> </table>						<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door	<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door
<u>Number</u>	<u>Name/Location</u>												
822	Aft Cargo Door												
<u>Number</u>	<u>Name/Location</u>												
822	Aft Cargo Door												
— END OF TASK —													

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-723-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 241

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 887 to STA 1016. (PSE 53-70-03-1).

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-723-00-01	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-723-00-01
				MECH INSP
TASK 53-05-02-250-951				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE				
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-151</p> <p>(1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 887 to STA 1016. (PSE 53-70-03-1). See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL		SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE	
			D633A109-AKS 53-723-00-01	

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-723-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE 600 700 700IGW ENGINE ALL 800 900 900ER
STATION	SKILL AIRPL	ACCESS			ZONE 242

Inspect (High Frequency Eddy Current) the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 887 to STA 1016. (PSE 53-70-03-1).

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC	D633A109-AKS 53-723-00-02	Page 1 of 2 Oct 15/2014
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-723-00-02

TASK 53-05-02-250-951

1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC

A. Inspection

SUBTASK 53-05-02-250-151

- (1) Do a High Frequency Eddy Current inspection of the upper skin along the upper fastener row at stringers S-4L and S-4R from STA 887 to STA 1016. (PSE 53-70-03-1).

See Doc. D626A001-DTR, DTR check form 53-10-03-1, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-41.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-723-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-724-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS			ZONE 241

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 887 to STA 1016.

See Doc. D626A001-DTR, DTR check form 53-70-03-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-724-00-01

AKS



737-600/700/800/900 TASK CARDS

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-724-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-724-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 242

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 887 to STA 1016.

See Doc. D626A001-DTR, DTR check form 53-70-03-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-724-00-02

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-724-00-02
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TASK 53-05-02-250-894

1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC

A. Inspection

SUBTASK 53-05-02-250-094

- (1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringers S-4L and S-4R from STA 887 to STA 1016.

See Doc. D626A001-DTR, DTR check form 53-70-03-2, for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-724-00-02

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-725-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 241

Inspect (Detailed) the upper skin along the upper fastener row at stringer S-14L (from STA 888 to STA 947, and from STA 1006 to STA 1016) and at stringer S-14R (from STA 888 to STA 947, and from STA 996 to STA 1016).

See Doc D626A001-DTR, DTR check form 53-70-03-3 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-725-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-725-00-01
				MECH INSP
TASK 53-05-02-211-845				
1. EXTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-211-045				
(1) Do a Detailed inspection of the upper skin along the upper fastener row at stringer S-14L (from STA 888 to STA 947, and from STA 1006 to STA 1016) and at stringer S-14R (from STA 888 to STA 947, and from STA 996 to STA 1016).				
See Doc D626A001-DTR, DTR check form 53-70-03-3 for alternative inspections.				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-725-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-725-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS		ZONE 242	

Inspect (Detailed) the upper skin along the upper fastener row at stringer S-14L (from STA 888 to STA 947, and from STA 1006 to STA 1016) and at stringer S-14R (from STA 888 to STA 947, and from STA 996 to STA 1016).

See Doc D626A001-DTR, DTR check form 53-70-03-3 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-725-00-02

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-725-00-02
				MECH INSP
TASK 53-05-02-211-845				
1. EXTERNAL - DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-211-045				
(1) Do a Detailed inspection of the upper skin along the upper fastener row at stringer S-14L (from STA 888 to STA 947, and from STA 1006 to STA 1016) and at stringer S-14R (from STA 888 to STA 947, and from STA 996 to STA 1016).				
See Doc D626A001-DTR, DTR check form 53-70-03-3 for alternative inspections.				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLIC
		D633A109-AKS 53-725-00-02

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT LONGITUDINAL LAP SPLIC			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-726-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS			ZONE 241

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringer S-14L (from STA 888 to STA 947 and from STA 1006 to STA 1016) and at stringer S-14R (from STA 888 to STA 947 and from STA 996 to STA 1016).

See Doc D626A001-DTR, DTR check form 53-70-03-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-726-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-726-00-01
				MECH INSP
TASK 53-05-02-250-895				
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLICE				
<p>A. Inspection</p> <p>SUBTASK 53-05-02-250-095</p> <p>(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringer S-14L (from STA 888 to STA 947 and from STA 1006 to STA 1016) and at stringer S-14R (from STA 888 to STA 947 and from STA 996 to STA 1016).</p> <p>See Doc D626A001-DTR, DTR check form 53-70-03-4 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.</p> <p style="text-align: center;">— END OF TASK —</p>				
<p>EFFECTIVITY AKS ALL</p> <p>SOURCE AWL</p> <p>LEFT LONGITUDINAL LAP SPLICE</p> <p>D633A109-AKS 53-726-00-01</p>				

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT LONGITUDINAL LAP SPLIC			BOEING CARD NO. 53-726-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS			ZONE 242

Inspect (Low Frequency Eddy Current) the lower skin along the lower fastener row at stringer S-14L (from STA 888 to STA 947 and from STA 1006 to STA 1016) and at stringer S-14R (from STA 888 to STA 947 and from STA 996 to STA 1016).

See Doc D626A001-DTR, DTR check form 53-70-03-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE
		D633A109-AKS 53-726-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-726-00-02
TASK 53-05-02-250-895				MECH INSP
1. EXTERNAL - SPECIAL DETAILED: LONGITUDINAL LAP SPLIC				
A. Inspection				
SUBTASK 53-05-02-250-095				
(1) Do a Low Frequency Eddy Current inspection of the lower skin along the lower fastener row at stringer S-14L (from STA 888 to STA 947 and from STA 1006 to STA 1016) and at stringer S-14R (from STA 888 to STA 947 and from STA 996 to STA 1016). See Doc D626A001-DTR, DTR check form 53-70-03-4 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-30-50.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT LONGITUDINAL LAP SPLICE		
		D633A109-AKS 53-726-00-02	Page 2 of 2 Jun 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CROWN SKIN PANEL - STA 887 TO 1016			BOEING CARD NO. 53-727-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE		ZONE 241 242		

Inspect (Detailed) the skin around all of the fastener locations from stringer S-10L to S-10R, from STA 887 to STA 1016, except at the lap splices and antennas. (PSE 53-70-04-1).

See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.

ACCESS NOTE: Remove Dorsal Fin as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	CROWN SKIN PANEL - STA 887 TO 1016
		D633A109-AKS 53-727-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-727-00-01
TASK 53-05-02-211-873				MECH INSP
1. EXTERNAL - DETAILED: CROWN SKIN PANEL STA 887 TO STA 1016				
A. Inspection				
NOTE: Remove Dorsal Fin as required to perform the inspection.				
SUBTASK 53-05-02-211-073				
(1) Do a Detailed inspection of the skin around all of the fastener locations from stringer S-10L to S-10R, from STA 887 to STA 1016, except at the lap splices and antennas. (PSE 53-70-04-1).				
See Doc D626A001-DTR, DTR check form 53-30-01-2 for alternative inspections.				
———— END OF TASK ——				
EFFECTIVITY AKS ALL				
SOURCE AWL				
CROWN SKIN PANEL - STA 887 TO 1016				
D633A109-AKS 53-727-00-01				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT ENTRY DOOR, FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE			BOEING CARD NO. 53-728-00-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FWD ENTRY DOOR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL	
STATION	SKILL AIRPL					
		ACCESS NOTE		ZONE 241		

Inspect (High Frequency Eddy Current) the four fastener locations at the #1, #2, #6 and #7 stop locations.

See Doc D626A001-DTR, DTR check form 53-70-07-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-35.

ACCESS NOTE: Remove or displace interior sidewall lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR, FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE	D633A109-AKS 53-728-00-01	Page 1 of 2 Oct 15/2014
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-728-00-01
				MECH INSP
TASK 53-05-02-250-896				
1. INTERNAL - SPECIAL DETAILED: AFT ENTRY DOOR, FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE				
A. Inspection <u>NOTE:</u> Remove or displace interior sidewall lining as required to perform the inspection. SUBTASK 53-05-02-250-096 (1) Do a High Frequency Eddy Current inspection of the four fastener locations at the #1, #2, #6 and #7 stop locations. See Doc D626A001-DTR, DTR check form 53-70-07-3 for alternative inspections. The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-35.				
— END OF TASK —				
EFFECTIVITY AKS ALL		SOURCE AWL	AFT ENTRY DOOR, FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE	
			D633A109-AKS 53-728-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT ENTRY DOOR, AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS			BOEING CARD NO. 53-729-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 241

Inspect (High Frequency Eddy Current) the door stop intercostals along the inner chord and around fasteners common to the web and doublers at the #1, #2, #6 and #7 stop locations.

See Doc D626A001-DTR, DTR check form 53-70-07-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-36.

ACCESS NOTE: Remove or displace interior sidewall and door lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR, AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS
		D633A109-AKS 53-729-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-729-00-01
				MECH INSP
TASK 53-05-02-250-897				
1. INTERNAL - SPECIAL DETAILED: AFT ENTRY DOOR, AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS				
<p>A. Inspection</p> <p><u>NOTE:</u> Remove or displace interior sidewall and door lining as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-097</p> <p>(1) Do a High Frequency Eddy Current inspection of the door stop intercostals along the inner chord and around fasteners common to the web and doublers at the #1, #2, #6 and #7 stop locations.</p> <p>See Doc D626A001-DTR, DTR check form 53-70-07-4 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-36.</p> <p style="text-align: center;">— END OF TASK —</p>				
EFFECTIVITY AKS ALL		SOURCE AWL	AFT ENTRY DOOR, AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS	
			D633A109-AKS 53-729-00-01	
Page 2 of 2 Feb 15/2015				

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT ENTRY DOOR, AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS			BOEING CARD NO.
DATE	TASK GENERAL VISUAL				53-729-01-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 241

Inspect (General Visual) the door stop intercostals along the inner chord and around fasteners common to the web and doublers at the #1, #2, #6 and #7 stop locations.

See Doc D626A001-DTR, DTR check form 53-70-07-4 for alternative inspections.

ACCESS NOTE: Remove or displace interior sidewall and door lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR, AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS
		D633A109-AKS 53-729-01-01

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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-729-01-01		
					MECH	INSP
TASK 53-05-02-210-842						
1. INTERNAL - GENERAL VISUAL: AFT ENTRY DOOR, AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS						
A. Inspection						
SUBTASK 53-05-02-210-042						
NOTE: Remove or displace interior sidewall and door lining as required to perform the inspection.						
(1) Do a General Visual inspection of the door stop intercostals along the inner chord and around fasteners common to the web and doublers at the #1, #2, #6 and #7 stop locations.						
See Doc D626A001-DTR, DTR check form 53-70-07-4 for alternative inspections.						
———— END OF TASK ————						
EFFECTIVITY AKS ALL		SOURCE AWL		AFT ENTRY DOOR, AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS		
				D633A109-AKS 53-729-01-01		

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CUTOUT, AFT ENTRY DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-730-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 834			ZONE 241	
		NOTE				

Inspect (Detailed) the perimeter of the cutout and around the fasteners common to the edge frames and upper sill outer chords.

See Doc D626A001-DTR, DTR check form 53-70-07-5 for alternative inspections.

ACCESS NOTE: Remove or displace interior sidewall and door lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT, AFT ENTRY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-730-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-730-00-01
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TASK 53-05-02-211-846

MECH

INSP

1. EXTERNAL - DETAILED: CUTOUT, AFT ENTRY DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-025

- (1) Open this access panel:

Number Name/Location

834 Aft Entry Door

NOTE: Remove or displace interior sidewall and door lining as required to do the inspection.

SUBTASK 53-05-02-211-046

- (2) Do a Detailed inspection on the perimeter of the cutout and around the fasteners common to the edge frames and upper sill outer chords.

See Doc D626A001-DTR, DTR check form 53-70-07-5 for alternative inspections.

SUBTASK 53-05-02-410-023

- (3) Close this access panel:

Number Name/Location

834 Aft Entry Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT, AFT ENTRY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-730-00-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT ENTRY DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-731-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA AFT ENTRY DOOR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 834				ZONE 145 241

Inspect (Detailed) the skin around the edges of the scuff plates.

See Doc D626A001-DTR, DTR check form 53-70-07-6 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-731-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-731-00-01
				MECH INSP
TASK 53-05-02-211-847				
1. EXTERNAL - DETAILED: AFT ENTRY DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-026				
(1) Open this access panel:				
Number Name/Location				
834 Aft Entry Door				
SUBTASK 53-05-02-211-047				
(2) Do a Detailed inspection of the skin around the edges of the scuff plates.				
See Doc D626A001-DTR, DTR check form 53-70-07-6 for alternative inspections.				
SUBTASK 53-05-02-410-024				
(3) Close this access panel:				
Number Name/Location				
834 Aft Entry Door				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE		
		D633A109-AKS 53-731-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT ENTRY DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-731-01-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT ENTRY DOOR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 834			ZONE 145 241
		NOTE			

Inspect (High Frequency Eddy Current) the skin around the fastener holes hidden by the scuff plate.

See Doc D626A001-DTR, DTR check form 53-70-07-6 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-34.

ACCESS NOTE: Remove scuff plate.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-731-01-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-731-01-01
				MECH INSP
TASK 53-05-02-250-898				
1. EXTERNAL - SPECIAL DETAILED: AFT ENTRY DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-103				
(1) Open this access panel:				
Number Name/Location				
834 Aft Entry Door				
NOTE: Remove scuff plate.				
SUBTASK 53-05-02-250-098				
(2) Do a High Frequency Eddy Current inspection of the skin around the fastener holes hidden by the scuff plate.				
See Doc D626A001-DTR, DTR check form 53-70-07-6 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-30-34.				
SUBTASK 53-05-02-410-101				
(3) Close this access panel:				
Number Name/Location				
834 Aft Entry Door				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-731-01-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT ENTRY DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-732-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 22000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 834			ZONE 241
		NOTE			

Inspect (Detailed) the inner chord and web along the upper main sill from STA 951 to STA 1006.

See Doc D626A001-DTR, DTR check form 53-70-07-11 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-732-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-732-00-01
				MECH INSP
TASK 53-05-02-211-848				
1. INTERNAL - DETAILED: AFT ENTRY DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-027				
(1) Open this access panel:				
Number Name/Location				
834 Aft Entry Door				
NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to do the inspection.				
SUBTASK 53-05-02-211-048				
(2) Do a Detailed inspection of the inner chord and web along the upper main sill from STA 951 to STA 1006.				
See Doc D626A001-DTR, DTR check form 53-70-07-11 for alternative inspections.				
SUBTASK 53-05-02-410-167				
(3) Close this access panel:				
Number Name/Location				
834 Aft Entry Door				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE		
		D633A109-AKS 53-732-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT ENTRY DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK DETAILED				53-732-01-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 22000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS 834 NOTE			ZONE 241

Inspect (Detailed) the inner chord strap near the edge frames from STA 951 to STA 1006.

See Doc D626A001-DTR, DTR check form 53-70-07-11 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-732-01-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-732-01-01
				MECH INSP
TASK 53-05-02-211-981				
1. INTERNAL - DETAILED: AFT ENTRY DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-176				
(1) Open this access panel:				
Number Name/Location				
834 Aft Entry Door				
NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.				
B. Procedure				
SUBTASK 53-05-02-211-181				
(1) Do a Detailed inspection of the inner chord strap near the edge frames from STA 951 to STA 1006.				
See Doc D626A001-DTR, DTR check form 53-70-07-11 for alternative inspections.				
SUBTASK 53-05-02-410-168				
(2) Close this access panel:				
Number Name/Location				
834 Aft Entry Door				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-732-01-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT ENTRY DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-732-10-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 834			
		NOTE			
ZONE 241					

Inspect (High Frequency Eddy Current) the first five fasteners on the upper flange of the lower main sill outer chord, aft of the edge frame.

See Doc D626A001-DTR, DTR check form 53-70-07-12 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-20.

ACCESS NOTE: Remove scuff plates as required for access to the outer chord.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-732-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-732-10-01
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TASK 53-05-02-250-997

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: AFT ENTRY DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-172

- (1) Open this access panel:

Number Name/Location

834 Aft Entry Door

NOTE: Remove scuff plates as required for access to the outer chord.

SUBTASK 53-05-02-250-197

- (2) Do a High Frequency Eddy Current inspection of the first five fasteners on the upper flange of the lower main sill outer chord, aft of the edge frame.

See Doc D626A001-DTR, DTR check form 53-70-07-12 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-20.

SUBTASK 53-05-02-410-163

- (3) Close this access panel:

Number Name/Location

834 Aft Entry Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	AFT ENTRY DOOR SURROUND STRUCTURE	
		D633A109-AKS 53-732-10-01	Page 2 of 2 Jun 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT GALLEY DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-733-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 22000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 844			ZONE 242	
		NOTE				

Inspect (Detailed) the inner chord and web along the upper main sill from STA 951 to STA 1006. (PSE 53-70-08).

See Doc D626A001-DTR, DTR check form 53-70-07-11 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-733-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-733-00-01
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TASK 53-05-02-211-874

MECH

INSP

1. INTERNAL - DETAILED: AFT GALLEY DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-038

- (1) Open this access panel:

Number Name/Location

844 Aft Galley Service Door

NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required.

SUBTASK 53-05-02-211-074

- (2) Do a Detailed inspection of the inner chord and web along the upper main sill from STA 951 to STA 1006. (PSE 53-70-08).

See Doc D626A001-DTR, DTR check form 53-70-07-11 for alternative inspections.

SUBTASK 53-05-02-410-036

- (3) Close this access panel:

Number Name/Location

844 Aft Galley Service Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE	
		D633A109-AKS 53-733-00-01	Page 2 of 2 Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT GALLEY DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-733-01-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 22000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 844			ZONE 242	
		NOTE				

Inspect (Detailed) the inner chord strap near the edge frames from STA 951 to STA 1006. (PSE 53-70-08).

See Doc D626A001-DTR, DTR check form 53-70-07-11 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-733-01-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-733-01-01				
				MECH INSP				
TASK 53-05-02-211-982								
1. INTERNAL - DETAILED: AFT GALLEY DOOR SURROUND STRUCTURE								
<p>A. Inspection</p> <p>SUBTASK 53-05-02-010-177</p> <p>(1) Open this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>844</td> <td>Aft Galley Service Door</td> </tr> </tbody> </table> <p><u>NOTE:</u> Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.</p>				<u>Number</u>	<u>Name/Location</u>	844	Aft Galley Service Door	
<u>Number</u>	<u>Name/Location</u>							
844	Aft Galley Service Door							
<p>SUBTASK 53-05-02-211-183</p> <p>(2) Do a Detailed inspection of the inner chord strap near the edge frames from STA 951 to STA 1006. (PSE 53-70-08).</p> <p>See Doc D626A001-DTR, DTR check form 53-70-07-11 for alternative inspections.</p>								
<p>SUBTASK 53-05-02-410-170</p> <p>(3) Close this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>844</td> <td>Aft Galley Service Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	844	Aft Galley Service Door	
<u>Number</u>	<u>Name/Location</u>							
844	Aft Galley Service Door							
———— END OF TASK ————								

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-733-01-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT GALLEY DOOR FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-734-00-01 RELATED CARD
TAIL NUMBER	WORK AREA AFT SERVICE DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 844			
		NOTE			
			ZONE 242		

Inspect (Low Frequency Eddy Current) the four fastener locations at the #1, #2, #5 and #6 stop locations.

See Doc D626A001-DTR, DTR check form 53-70-08-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-83.

ACCESS NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE
		D633A109-AKS 53-734-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-734-00-01
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TASK 53-05-02-250-899

**1. INTERNAL - SPECIAL DETAILED: AFT GALLEY DOOR FORWARD EDGE FRAME DOOR STOP
BACKUP STRUCTURE**

A. Inspection

SUBTASK 53-05-02-010-102

- (1) Open this access panel:

Number **Name/Location**

844 Aft Galley Service Door

NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

SUBTASK 53-05-02-250-099

- (2) Do a Low Frequency Eddy Current inspection on the four fastener locations at the #1, #2, #5 and #6 stop locations.

See Doc D626A001-DTR. DTR check form 53-70-08-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-83.

SUBTASK 53-05-02-410-100

- (3) Close this access panel:

Number Name/Location

844 Aft Galley Service Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE
		D633A109-AKS 53-734-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT GALLEY DOOR FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE			BOEING CARD NO. 53-734-01-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT SERVICE DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 12000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 844			ZONE 242
		NOTE			

Inspect (High Frequency Eddy Current) the four fastener locations at the #1, #2, #5 and #6 stop locations. For door stop #5, there are two locations in the strap hidden by the bracket.

See Doc D626A001-DTR, DTR check form 53-70-08-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-83.

ACCESS NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE
		D633A109-AKS 53-734-01-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-734-01-01
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TASK 53-05-02-250-900

- ## **1. INTERNAL - SPECIAL DETAILED: AFT GALLEY DOOR FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE**

A. Inspection

SUBTASK 53-05-02-010-104

- (1) Open this access panel:

Number **Name/Location**

844 Aft Galley Service Door

NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

SUBTASK 53-05-02-250-100

- (2) Do a High Frequency Eddy Current inspection on the four fastener locations at the #1, #2, #5 and #6 stop locations. For door stop #5, there are two locations in the strap hidden by the bracket.

See Doc D626A001-DTR, DTR check form 53-70-08-3 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-83.

SUBTASK 53-05-02-410-102

- (3) Close this access panel:

Number Name/Location

844 Aft Galley Service Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR FORWARD EDGE FRAME DOOR STOP BACKUP STRUCTURE
		D633A109-AKS 53-734-01-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT GALLEY DOOR AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS			BOEING CARD NO. 53-735-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA AFT SERVICE DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 844			
		NOTE			
			ZONE 242		

Inspect (High Frequency Eddy Current) the door stop intercostals along the inner chord and around fasteners common to the web and doubler at the #1, #2, #5, and #6 stop locations.

See Doc D626A001-DTR, DTR check form 53-70-08-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-84.

ACCESS NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS
		D633A109-AKS 53-735-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-735-00-01								
				MECH INSP								
TASK 53-05-02-250-901												
1. INTERNAL - SPECIAL DETAILED: AFT GALLEY DOOR AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS												
<p>A. Inspection</p> <p>SUBTASK 53-05-02-010-105</p> <p>(1) Open this access panel:</p> <table style="margin-left: 20px; border-collapse: collapse;"> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> <tr> <td style="padding-left: 10px;">844</td> <td style="padding-left: 10px;">Aft Galley Service Door</td> </tr> </table> <p><u>NOTE:</u> Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-101</p> <p>(2) Do a High Frequency Eddy Current inspection of the door stop intercostals along the inner chord and around fasteners common to the web and doubler at the #1, #2, #5, and #6 stop locations.</p> <p>See Doc D626A001-DTR, DTR check form 53-70-08-4 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-84.</p> <p>SUBTASK 53-05-02-410-103</p> <p>(3) Close this access panel:</p> <table style="margin-left: 20px; border-collapse: collapse;"> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> <tr> <td style="padding-left: 10px;">844</td> <td style="padding-left: 10px;">Aft Galley Service Door</td> </tr> </table>					<u>Number</u>	<u>Name/Location</u>	844	Aft Galley Service Door	<u>Number</u>	<u>Name/Location</u>	844	Aft Galley Service Door
<u>Number</u>	<u>Name/Location</u>											
844	Aft Galley Service Door											
<u>Number</u>	<u>Name/Location</u>											
844	Aft Galley Service Door											
<hr style="display: inline-block; width: 40%; vertical-align: middle;"/> END OF TASK <hr style="display: inline-block; width: 40%; vertical-align: middle;"/>												

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT GALLEY DOOR AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS			BOEING CARD NO. 53-735-01-01	
DATE	TASK GENERAL VISUAL				RELATED CARD	
TAIL NUMBER	WORK AREA AFT SERVICE DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 844			ZONE 242	
		NOTE				

Inspect (General Visual) the door stop intercostals along the inner chord and around fasteners common to the web and doubler at the #1, #2, #5, and #6 stop locations.

See Doc D626A001-DTR, DTR check form 53-70-08-4 for alternative inspections.

ACCESS NOTE: Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS	D633A109-AKS 53-735-01-01	Page 1 of 2 Jun 15/2016
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-735-01-01							
				MECH INSP							
TASK 53-05-02-210-843											
1. INTERNAL - GENERAL VISUAL: AFT GALLEY DOOR AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS											
<p>A. Inspection</p> <p>SUBTASK 53-05-02-410-171</p> <p>(1) Open this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>844</td> <td>Aft Galley Service Door</td> </tr> </tbody> </table> <p><u>NOTE:</u> Remove or displace passenger cabin sidewall and ceiling lining as required to perform the inspection.</p> <p>SUBTASK 53-05-02-210-043</p> <p>(2) Do a General Visual inspection of the door stop intercostals along the inner chord and around fasteners common to the web and doubler at the #1, #2, #5, and #6 stop locations.</p> <p>See Doc D626A001-DTR, DTR check form 53-70-08-4 for alternative inspections.</p> <p>SUBTASK 53-05-02-410-172</p> <p>(3) Close this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>844</td> <td>Aft Galley Service Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	844	Aft Galley Service Door	<u>Number</u>	<u>Name/Location</u>	844	Aft Galley Service Door
<u>Number</u>	<u>Name/Location</u>										
844	Aft Galley Service Door										
<u>Number</u>	<u>Name/Location</u>										
844	Aft Galley Service Door										
— END OF TASK —											
EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR AFT EDGE FRAME DOOR STOP BACKUP INTERCOSTALS									
D633A109-AKS 53-735-01-01											
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE CUTOUT, AFT GALLEY DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK DETAILED				53-736-00-01 RELATED CARD
TAIL NUMBER	WORK AREA AFT SERVICE DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 844			ZONE 242

Inspect (Detailed) the perimeter of the cutout and around the fasteners common to the edge frames and upper sill outer chords.

See Doc D626A001-DTR, DTR check form 53-70-08-5 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT, AFT GALLEY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-736-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-736-00-01
				MECH INSP
TASK 53-05-02-211-849				
1. EXTERNAL - DETAILED: CUTOUT, AFT GALLEY DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-028				
(1) Open this access panel:				
Number Name/Location				
844 Aft Galley Service Door				
SUBTASK 53-05-02-211-049				
(2) Do a Detailed inspection of the perimeter of the cutout and around the fasteners common to the edge frames and upper sill outer chords.				
See Doc D626A001-DTR, DTR check form 53-70-08-5 for alternative inspections.				
SUBTASK 53-05-02-410-026				
(3) Close this access panel:				
Number Name/Location				
844 Aft Galley Service Door				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	CUTOUT, AFT GALLEY DOOR SURROUND STRUCTURE		
		D633A109-AKS 53-736-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT GALLEY DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-737-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA AFT SERVICE DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 844			ZONE 146 242	

Inspect (Detailed) the skin around the edges of the scuff plates. (PSE 53-70-08-6).

See Doc D626A001-DTR, DTR check form 53-70-07-6 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-737-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-737-00-01
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TASK 53-05-02-211-875

MECH

INSP

1. EXTERNAL - DETAILED: AFT GALLEY DOOR SURROUND STRUCTURE**A. Inspection**

SUBTASK 53-05-02-010-039

- (1) Open this access panel:

Number Name/Location

844 Aft Galley Service Door

SUBTASK 53-05-02-211-075

- (2) Do a Detailed inspection of the skin around the edges of the scuff plates. (PSE 53-70-08-6).

See Doc D626A001-DTR, DTR check form 53-70-07-6 for alternative inspections.

SUBTASK 53-05-02-410-037

- (3) Close this access panel:

Number Name/Location

844 Aft Galley Service Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE	
		D633A109-AKS 53-737-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT GALLEY DOOR SURROUND STRUCTURE			BOEING CARD NO. 53-737-01-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA AFT SERVICE DR	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 844			ZONE 146 242	
		NOTE				

Inspect (High Frequency Eddy Current) the skin around the fastener holes hidden by the scuff plate. (PSE 53-70-08-6).

See Doc D626A001-DTR, DTR check form 53-70-07-6 for alternative inspections.

ACCESS NOTE: Remove scuff plate.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-737-01-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-737-01-01				
				MECH INSP				
TASK 53-05-02-250-953								
1. INTERNAL - SPECIAL DETAILED: AFT GALLEY DOOR SURROUND STRUCTURE								
<p>A. Inspection</p> <p>SUBTASK 53-05-02-010-116</p> <p>(1) Open this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>844</td> <td>Aft Galley Service Door</td> </tr> </tbody> </table> <p><u>NOTE:</u> Remove scuff plate.</p>				<u>Number</u>	<u>Name/Location</u>	844	Aft Galley Service Door	
<u>Number</u>	<u>Name/Location</u>							
844	Aft Galley Service Door							
<p>SUBTASK 53-05-02-250-153</p> <p>(2) Do a High Frequency Eddy Current inspection on the skin around the fastener holes hidden by the scuff plate. (PSE 53-70-08-6).</p> <p>See Doc D626A001-DTR, DTR check form 53-70-07-6 for alternative inspection</p>								
<p>SUBTASK 53-05-02-410-114</p> <p>(3) Close this access panel:</p> <table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>844</td> <td>Aft Galley Service Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	844	Aft Galley Service Door	
<u>Number</u>	<u>Name/Location</u>							
844	Aft Galley Service Door							
———— END OF TASK ————								

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-737-01-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT GALLEY DOOR SURROUND STRUCTURE			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-737-10-01 RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 844			
		NOTE			
ZONE 242					

Inspect (High Frequency Eddy Current) the first five fasteners on the upper flange of the lower main sill outer chord, aft of the edge frame. (53-70-08-12).

See Doc D626A001-DTR, DTR check form 53-70-07-12 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-20.

ACCESS NOTE: Remove scuff plates as required for access to the outer chord.

EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE
		D633A109-AKS 53-737-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-737-10-01
				MECH INSP
TASK 53-05-02-250-A79				
1. INTERNAL - SPECIAL DETAILED: AFT GALLEY DOOR SURROUND STRUCTURE				
A. Inspection				
SUBTASK 53-05-02-010-173				
(1) Open this access panel:				
Number Name/Location				
844 Aft Galley Service Door				
NOTE: Remove scuff plates as required for access to the outer chord.				
SUBTASK 53-05-02-250-279				
(2) Do a High Frequency Eddy Current inspection on the first five fasteners on the upper flange of the lower main sill outer chord, aft of the edge frame. (53-70-08-12).				
See Doc D626A001-DTR, DTR check form 53-70-07-12 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-20.				
SUBTASK 53-05-02-410-164				
(3) Close this access panel:				
Number Name/Location				
844 Aft Galley Service Door				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	AFT GALLEY DOOR SURROUND STRUCTURE		
		D633A109-AKS 53-737-10-01	Page 2 of 2 Jun 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LEFT WINDOW BELT STA 888 TO 927			BOEING CARD NO. 53-738-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 241

Inspect (Detailed) the window frames around each window from STA 888 to STA 927. (PSE 53-70-09)

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets as required.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 888 TO 927
		D633A109-AKS 53-738-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-738-00-01
				MECH INSP
TASK 53-05-02-211-876				
1. INTERNAL - DETAILED: WINDOW BELT STA 888 to 927				
A. Inspection				
NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets as required.				
SUBTASK 53-05-02-211-076				
(1) Do a Detailed inspection of the window frames around each window from STA 888 to STA 927. (PSE 53-70-09)				
See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 888 TO 927		
		D633A109-AKS 53-738-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT WINDOW BELT STA 888 TO 927			BOEING CARD NO. 53-738-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 242

Inspect (Detailed) the window frames around each window from STA 888 to STA 927. (PSE 53-70-09)

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

ACCESS NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets as required.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 888 TO 927
		D633A109-AKS 53-738-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-738-00-02
				MECH INSP
TASK 53-05-02-211-876				
1. INTERNAL - DETAILED: WINDOW BELT STA 888 to 927				
A. Inspection				
NOTE: Removal and/or displacement of passenger cabin sidewalls or sidewall window assemblies and insulation blankets as required.				
SUBTASK 53-05-02-211-076				
(1) Do a Detailed inspection of the window frames around each window from STA 888 to STA 927. (PSE 53-70-09)				
See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.				
———— END OF TASK ——				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 888 TO 927		
		D633A109-AKS 53-738-00-02	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT WINDOW BELT STA 888 TO 927			BOEING CARD NO. 53-738-01-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 241	

Inspect (Detailed) the window frames around each window from STA 888 to STA 927. (PSE 53-70-09)

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT WINDOW BELT STA 888 TO 927
		D633A109-AKS 53-738-01-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-738-01-01
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TASK 53-05-02-211-877

1. EXTERNAL - DETAILED: WINDOW BELT STA 888 to 927

A. Inspection

SUBTASK 53-05-02-211-077

- (1) Do a Detailed inspection of the window frames around each window from STA 888 to STA 927. (PSE 53-70-09)

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

END OF TASK

MECH INSP

53-738-01-01

	MECH	INSP
<p>TASK 53-05-02-211-877</p> <p>1. EXTERNAL - DETAILED: WINDOW BELT STA 888 to 927</p> <p>A. Inspection</p> <p>SUBTASK 53-05-02-211-077</p> <p>(1) Do a Detailed inspection of the window frames around each window from STA 888 to STA 927. (PSE 53-70-09)</p> <p>See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>		

EFFECTIVITY
AKS ALL

SOURCE
AWL

LEFT WINDOW BELT STA 888 TO 927

D633A109-AKS
53-738-01-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT WINDOW BELT STA 888 TO 927			BOEING CARD NO. 53-738-01-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 4000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 242	

Inspect (Detailed) the window frames around each window from STA 888 to STA 927. (PSE 53-70-09)

See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 888 TO 927
		D633A109-AKS 53-738-01-02

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-738-01-02	MECH	INSP
TASK 53-05-02-211-877						
1. EXTERNAL - DETAILED: WINDOW BELT STA 888 to 927						
A. Inspection						
SUBTASK 53-05-02-211-077						
(1) Do a Detailed inspection of the window frames around each window from STA 888 to STA 927. (PSE 53-70-09)						
See Doc D626A001-DTR, DTR check form 53-60-05-2 for alternative inspections.						
———— END OF TASK ————						

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT WINDOW BELT STA 888 TO 927
		D633A109-AKS 53-738-01-02

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-739-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL	ACCESS 311BL			
					ZONE 311 312

Inspect (Detailed) the pressure dome webs between the stiffeners and tear straps.

See Doc D626A001-DTR, DTR check form 53-80-01-2 for alternative inspections.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-739-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-739-00-01
				MECH INSP
TASK 53-05-02-211-850				
1. INTERNAL - DETAILED: BULKHEAD STA 1016				
A. Inspection				
SUBTASK 53-05-02-010-029				
(1) Open this access panel:				
Number Name/Location				
311BL Stabilizer Trim Access Door				
SUBTASK 53-05-02-211-050				
(2) Do a Detailed inspection of the pressure dome webs between the stiffeners and tear straps.				
See Doc D626A001-DTR, DTR check form 53-80-01-2 for alternative inspections.				
SUBTASK 53-05-02-410-027				
(3) Close this access panel:				
Number Name/Location				
311BL Stabilizer Trim Access Door				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016	
		D633A109-AKS 53-739-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-740-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL	ACCESS 311BL			
					ZONE 311 312

Inspect (Detailed) the pressure dome web lap splices along the fastener rows adjacent to the radial stiffeners.

See Doc D626A001-DTR, DTR check form 53-80-01-3 for alternative inspections.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-740-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-740-00-01
TASK 53-05-02-211-851				MECH INSP

1. INTERNAL - DETAILED: BULKHEAD STA 1016**A. Inspection**

SUBTASK 53-05-02-010-030

- (1) Open this access panel:

Number Name/Location

311BL Stabilizer Trim Access Door

SUBTASK 53-05-02-211-051

- (2) Do a Detailed inspection of the pressure dome web lap splices along the fastener rows adjacent to the radial stiffeners.

See Doc D626A001-DTR, DTR check form 53-80-01-3 for alternative inspections.

SUBTASK 53-05-02-410-028

- (3) Close this access panel:

Number Name/Location

311BL Stabilizer Trim Access Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-740-00-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-741-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS NOTE			ZONE 145 146 241 242 311 312

Inspect (High Frequency Eddy Current) the forward side of the pressure dome web along the aft fastener row attaching the web to the pressure chord. Inspect at the edge of each stiffener/clip and around the two fasteners on each side of the stiffener.

See Doc D626A001-DTR, DTR check form 53-80-01-4 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-11.

AIRPLANE NOTE: For L/N 721 and on, the inspection also includes the web adjacent to the stiffeners between stringers S-1 and S-3, S-3 and S-5, and S-5 and S-7 on both sides of the aircraft. Applicable to all models except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin and aft cargo interiors as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-741-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-741-00-01
TASK 53-05-02-250-902				MECH INSP
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1016				
A. Inspection				
<p><u>NOTE:</u> Remove necessary passenger cabin and aft cargo interiors as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-102</p> <p>(1) Do a High Frequency Eddy Current inspection of the forward side of the pressure dome web along the aft fastener row attaching the web to the pressure chord. Inspect at the edge of each stiffener/clip and around the two fasteners on each side of the stiffener.</p> <p>See Doc D626A001-DTR, DTR check form 53-80-01-4 for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-11.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL		SOURCE AWL	BULKHEAD STA 1016	
			D633A109-AKS 53-741-00-01	Page 2 of 2 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-742-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 12000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS NOTE			ZONE 241 242 311 312

Inspect (High Frequency Eddy Current) the pressure dome webs along the aft fastener row attaching the web to the pressure chord and between the stiffener locations from stringers S-5L to S-7L and S-5R to S-9R.

See Doc D626A001-DTR, DTR check form 53-80-01-5A for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-38.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-742-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-742-00-01
TASK 53-05-02-250-903				MECH INSP
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1016				
A. Inspection				
<p><u>NOTE:</u> Remove necessary passenger cabin interiors as required to perform the inspection.</p> <p>SUBTASK 53-05-02-250-103</p> <p>(1) Do a High Frequency Eddy Current inspection of the pressure dome webs along the aft fastener row attaching the web to the pressure chord and between the stiffener locations from stringers S-5L to S-7L and S-5R to S-9R.</p> <p>See Doc D626A001-DTR, DTR check form 53-80-01-5A for alternative inspections.</p> <p>The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-38.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL		SOURCE AWL	BULKHEAD STA 1016	
			D633A109-AKS 53-742-00-01	Page 2 of 2 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-743-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 145 146 241 242 311 312

Inspect (High Frequency Eddy Current) the pressure dome webs along the aft fastener row attaching the web to the pressure chord and between the stiffener locations outside of stringers S-5L to S-7L and S-5R to S-9R.

See Doc D626A001-DTR, DTR check form 53-80-01-5B for alternative inspections.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform this inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-743-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-743-00-01
TASK 53-05-02-250-904				MECH INSP
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1016				
A. Inspection				
<p><u>NOTE:</u> Remove necessary passenger cabin interiors if/as required to perform this inspection.</p> <p>SUBTASK 53-05-02-250-104</p> <p>(1) Do a High Frequency Eddy Current inspection of the pressure dome webs along the aft fastener row attaching the web to the pressure chord and between the stiffener locations outside of stringers S-5L to S-7L and S-5R to S-9R.</p> <p>See Doc D626A001-DTR, DTR check form 53-80-01-5B for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
BULKHEAD STA 1016				
D633A109-AKS 53-743-00-01				
Page 2 of 2 Feb 15/2015				

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-744-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 145 146 241 242 311 312

Inspect (Low Frequency Eddy Current) around the fasteners common to the pressure chord splices between stringers S-2 and S-3 and S-16 and S-17A.

See Doc D626A001-DTR, DTR check form 53-80-01-6 for alternative inspections.

AIRPLANE NOTE: For L/N 1057 and on, this inspection applies only to the splices between stringer S-16 and S-17A. Applicable to all models except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-744-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-744-00-01	MECH	INSP
TASK 53-05-02-250-905						
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1016						

A. Inspection

NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.

SUBTASK 53-05-02-250-105

(1) Do a Low Frequency Eddy Current inspection around the fasteners common to the pressure chord splices between stringers S-2 and S-3 and S-16 and S-17A.

See Doc D626A001-DTR, DTR check form 53-80-01-6 for alternative inspections.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-744-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-745-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL	ACCESS NOTE			ZONE 145 146 241 242 311 312

Inspect (Low Frequency Eddy Current) the forward side of the pressure dome web around the fasteners common to the lap splice and the stiffeners.

See Doc D626A001-DTR, DTR check form 53-80-01-7 for alternative inspections.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-745-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-745-00-01	MECH	INSP
TASK 53-05-02-250-906						
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1016						

A. Inspection

NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.

SUBTASK 53-05-02-250-106

(1) Do a Low Frequency Eddy Current inspection on the forward side of the pressure dome web around the fasteners common to the lap splice and the stiffeners.

See Doc D626A001-DTR, DTR check form 53-80-01-7 for alternative inspections.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-745-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-746-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS NOTE			ZONE 311 312

Inspect (High Frequency Eddy Current) the forward side of the pressure dome web around the fasteners common to the tear strap.

See Doc D626A001-DTR, DTR check form 53-80-01-8 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-97.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-746-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-746-00-01
TASK 53-05-02-250-907				MECH INSP
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1016				
A. Inspection				
NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.				
SUBTASK 53-05-02-250-107				
(1) Do a High Frequency Eddy Current inspection on the forward side of the pressure dome web around the fasteners common to the tear strap.				
See Doc D626A001-DTR, DTR check form 53-80-01-8 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-97.				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016		
		D633A109-AKS 53-746-00-01		
		Page 2 of 2 Feb 15/2015		

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-747-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 9000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS 311BL			ZONE 311 312

Inspect (Low Frequency Eddy Current) the aft side of the pressure dome web at the intersection of the tear straps and lap splice next to the stiffeners.

See Doc D626A001-DTR, DTR check form 53-80-01-9 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-08.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-747-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-747-00-01				
				MECH INSP				
TASK 53-05-02-250-908								
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1016								
A. Inspection								
SUBTASK 53-05-02-010-141								
(1) Open this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>311BL</td><td>Stabilizer Trim Access Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	311BL	Stabilizer Trim Access Door
<u>Number</u>	<u>Name/Location</u>							
311BL	Stabilizer Trim Access Door							
SUBTASK 53-05-02-250-108								
(2) Do a Low Frequency Eddy Current inspection on the aft side of the pressure dome web at the intersection of the tear straps and lap splice next to the stiffeners.								
See Doc D626A001-DTR, DTR check form 53-80-01-9 for alternative inspections.								
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-08.								
SUBTASK 53-05-02-410-132								
(3) Close this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>311BL</td><td>Stabilizer Trim Access Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	311BL	Stabilizer Trim Access Door
<u>Number</u>	<u>Name/Location</u>							
311BL	Stabilizer Trim Access Door							
<hr style="width: 20%; margin-left: auto; margin-right: 0;"/> END OF TASK <hr style="width: 20%; margin-left: 0; margin-right: auto;"/>								

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016	
		D633A109-AKS 53-747-00-01	Page 2 of 2 Jun 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-748-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS NOTE			ZONE 311 312

Inspect (High Frequency Eddy Current) the forward side of the pressure dome webs at the junction of the radial stiffeners/lap splices and the tear straps.

See Doc D626A001-DTR, DTR check form 53-80-01-11 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-07.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-748-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-748-00-01
				MECH INSP
TASK 53-05-02-250-909				
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1016				
A. Inspection				
<u>NOTE:</u> Remove necessary passenger cabin interiors as required to perform the inspection.				
SUBTASK 53-05-02-250-109				
(1) Do a High Frequency Eddy Current inspection on the forward side of the pressure dome webs at the junction of the radial stiffeners/lap splices and the tear straps.				
See Doc D626A001-DTR, DTR check form 53-80-01-11 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-07.				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-748-00-01

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-749-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA PASS CABIN	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS NOTE			ZONE 311 312

Inspect (Low Frequency Eddy Current) the forward side of the pressure dome web around the fasteners common to the doubler, Y-Chord and the tear strap between stringers S-1 and S-3.

See Doc D626A001-DTR, DTR check form 53-80-01-13 for alternative inspections.

AIRPLANE NOTE: Applicable to all L/N 1057 an on, except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-749-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-749-00-01	MECH	INSP
TASK 53-05-02-250-910						
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1016						
A. Inspection						
NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.						
SUBTASK 53-05-02-250-110						
(1) Do a Low Frequency Eddy Current inspection on the forward side of the pressure dome web around the fasteners common to the doubler, Y-Chord and the tear strap between stringers S-1 and S-3.						
See Doc D626A001-DTR, DTR check form 53-80-01-13 for alternative inspections.						
———— END OF TASK ————						

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016	
		D633A109-AKS 53-749-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STA 1016			BOEING CARD NO. 53-749-10-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 36000 FC	REPEAT 10000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL	ACCESS 311BL			
		ZONE 311 312			

Inspect (Detailed) the aft side of STA 1016 bulkhead web for oil cans.

Note: Refer to Structural Repair Manual, Section 53-80-08, for definition of oil can.

No DTR form available for PSE 53-80-01-14.

AIRPLANE NOTE: All aircraft L/N 1756 and on, except -800FPB and -900ER.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-749-10-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-749-10-01
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TASK 53-05-02-211-888

MECH

INSP

1. INTERNAL - DETAILED: BULKHEAD STA 1016**A. Inspection**

SUBTASK 53-05-02-010-071

- (1) Open this access panel:

Number Name/Location

311BL Stabilizer Trim Access Door

SUBTASK 53-05-02-211-088

- (2) Do a Detailed inspection of the aft side of STA 1016 bulkhead web for oil cans.

NOTE: Refer to Structural Repair Manual, Section 53-80-08, for definition of oil can.

No DTR form available for PSE 53-80-01-14.

SUBTASK 53-05-02-410-069

- (3) Close this access panel:

Number Name/Location

311BL Stabilizer Trim Access Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STA 1016
		D633A109-AKS 53-749-10-01

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Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE VERTICAL FIN FRONT SPAR FITTING BULKHEAD ATTACHMENT - STA 1016			BOEING CARD NO. 53-750-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 33000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL				ZONE 241 242 321
		ACCESS NOTE			

Inspect (Detailed) the fittings on both sides of the bulkhead at STA 1016.

See Doc D626A001-DTR, DTR check form 53-80-02-1 for alternative inspections.

AIRPLANE NOTE: Applicable to all models except -800FPB and -900ER.

ACCESS NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	VERTICAL FIN FRONT SPAR FITTING BULKHEAD ATTACHMENT - STA 1016
		D633A109-AKS 53-750-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-750-00-01
				MECH INSP
TASK 53-05-02-211-852				
1. INTERNAL - DETAILED: VERTICAL FIN FRONT SPAR FITTING BULKHEAD ATTACHMENT - STA 1016				
A. Inspection				
NOTE: Remove necessary passenger cabin interiors as required to perform the inspection.				
SUBTASK 53-05-02-211-052				
(1) Do a Detailed inspection of the fittings on both sides of the bulkhead at STA 1016. See Doc D626A001-DTR, DTR check form 53-80-02-1 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	VERTICAL FIN FRONT SPAR FITTING BULKHEAD ATTACHMENT - STA 1016		
		D633A109-AKS 53-750-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE VERTICAL FIN FRONT SPAR FITTING - STA 1016			BOEING CARD NO. 53-751-00-01	
DATE	TASK SPECIAL DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA VERT STABILIZER	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 323AL 323BL		ZONE 241 242 322		

Inspect (High Frequency Eddy Current) the exposed forward and aft surfaces of the fitting lugs. Bolt removal is not required.

See Doc D626A001-DTR, DTR check form 53-80-02-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-78.

EFFECTIVITY AKS ALL	SOURCE AWL	VERTICAL FIN FRONT SPAR FITTING - STA 1016	
		D633A109-AKS 53-751-00-01	Page 1 of 2 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-751-00-01
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TASK 53-05-02-250-912

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: VERTICAL FIN FRONT SPAR FITTINGS - STA 1016**A. Inspection**

SUBTASK 53-05-02-010-174

- (1) Open these access panels:

Number Name/Location

323AL Vertical Fin, Front Spar Access Door
323BL Vertical Fin, Forward Fin Access Door

SUBTASK 53-05-02-250-112

- (2) Do a High Frequency Eddy Current inspection of the exposed forward and aft surfaces of the fitting lugs. Bolt removal is not required.

See Doc D626A001-DTR, DTR check form 53-80-02-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-78.

SUBTASK 53-05-02-410-165

- (3) Close these access panels:

Number Name/Location

323AL Vertical Fin, Front Spar Access Door
323BL Vertical Fin, Forward Fin Access Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	VERTICAL FIN FRONT SPAR FITTING - STA 1016	
		D633A109-AKS 53-751-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE STRINGER SPLICE FITTINGS			BOEING CARD NO. 53-752-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL	ACCESS 311BL			
		NOTE			
			ZONE 241 242 311 312		

Inspect (High Frequency Eddy Current) the stringer splice fittings from stringer S-9L to S-9R at the first two fastener locations forward and aft of the STA 1016 bulkhead.

See Doc D626A001-DTR, DTR check form 53-80-03-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-79.

ACCESS NOTE: Remove or displace passenger cabin interior as required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	STRINGER SPLICE FITTINGS D633A109-AKS 53-752-00-01	Page 1 of 2 Oct 15/2014
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-752-00-01
				MECH INSP
TASK 53-05-02-250-913				
1. INTERNAL - SPECIAL DETAILED: STRINGER SPLICE FITTINGS				
A. Inspection				
SUBTASK 53-05-02-010-144				
(1) Open this access panel:				
Number Name/Location				
311BL Stabilizer Trim Access Door				
NOTE: Remove or displace passenger cabin interior as required to perform the inspection.				
SUBTASK 53-05-02-250-113				
(2) Do a High Frequency Eddy Current inspection of the stringer splice fittings from stringer S-9L to S-9R at the first two fastener locations forward and aft of the STA 1016 bulkhead. See Doc D626A001-DTR, DTR check form 53-80-03-1 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-79.				
SUBTASK 53-05-02-410-135				
(3) Close this access panel:				
Number Name/Location				
311BL Stabilizer Trim Access Door				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	STRINGER SPLICE FITTINGS	
		D633A109-AKS 53-752-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE JACKSCREW FITTING LUGS			BOEING CARD NO. 53-753-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 311BL			
		NOTE			
			ZONE 311 312		

Inspect (High Frequency Eddy Current) both primary jackscrew fitting lugs on both sides around the bushing at the STA 1088 bulkhead.

See Doc D626A001-DTR, DTR check form 53-80-05-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-95.

ACCESS NOTE: Remove access panel as required. Remove/disconnect the jackscrew from the fitting and move aside for access to the lugs.

EFFECTIVITY AKS ALL	SOURCE AWL	JACKSCREW FITTING LUGS
		D633A109-AKS 53-753-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-753-00-01
				MECH INSP
TASK 53-05-02-250-914				
1. INTERNAL - SPECIAL DETAILED: JACKSCREW FITTING LUGS				
A. Inspection				
SUBTASK 53-05-02-010-106				
(1) Open this access panel:				
Number Name/Location				
311BL Stabilizer Trim Access Door				
NOTE: Remove access panel as required. Remove/disconnect the jackscrew from the fitting and move aside for access to the lugs.				
SUBTASK 53-05-02-250-114				
(2) Do a High Frequency Eddy Current inspection of both primary jackscrew fitting lugs on both sides around the bushing at the STA 1088 bulkhead.				
See Doc D626A001-DTR, DTR check form 53-80-05-1 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-95.				
SUBTASK 53-05-02-410-104				
(3) Close this access panel:				
Number Name/Location				
311BL Stabilizer Trim Access Door				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE AWL	JACKSCREW FITTING LUGS
		D633A109-AKS 53-753-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE VERTICAL FIN REAR SPAR ATTACHMENT FITTINGS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-754-00-01 RELATED CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS 311BL			
		NOTE			
					ZONE 311 312 313 314 323

Inspect (Ultrasonic) the top two fasteners in the outboard primary fittings common to the STA 1088 bulkhead.

See Doc D626A001-DTR, DTR check form 53-80-06-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 53-80-01.

ACCESS NOTE: Access fittings inside the tailcone on the forward and aft side of the STA 1088 Bulkhead. The top of the fittings are sandwiched between the splice angles and the bulkhead.

EFFECTIVITY AKS ALL	SOURCE AWL	VERTICAL FIN REAR SPAR ATTACHMENT FITTINGS
		D633A109-AKS 53-754-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-754-00-01
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TASK 53-05-02-130-807

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: VERTICAL FIN REAR SPAR ATTACHMENT FITTINGS**A. Inspection**

SUBTASK 53-05-02-010-007

- (1) Open this access panel:

Number Name/Location

311BL Stabilizer Trim Access Door

NOTE: Access fittings inside the tailcone on the forward and aft side of the Sta 1088 Bulkhead. The top of the fittings are sandwiched between the splice angles and the bulkhead.

SUBTASK 53-05-02-130-007

- (2) Do an Ultrasonic inspection of the top two fasteners in the outboard primary fittings common to the STA 1088 bulkhead.

See Doc D626A001-DTR, DTR check form 53-80-06-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 53-80-01.

SUBTASK 53-05-02-410-005

- (3) Close this access panel:

Number Name/Location

311BL Stabilizer Trim Access Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	VERTICAL FIN REAR SPAR ATTACHMENT FITTINGS	
		D633A109-AKS 53-754-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE VERTICAL FIN REAR SPAR ATTACHMENT FITTINGS			BOEING CARD NO. 53-755-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA VERT STABILIZER	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 311 312 313 314 323

Inspect (Detailed) the four primary fitting lugs, from the inside of the lugs, at STA 1088.

See Doc D626A001-DTR, DTR check form 53-80-06-2 for alternative inspections.

ACCESS NOTE: Remove vertical fin including primary and fail-safe bolts.

EFFECTIVITY AKS ALL	SOURCE AWL	VERTICAL FIN REAR SPAR ATTACHMENT FITTINGS
		D633A109-AKS 53-755-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-755-00-01	MECH	INSP
TASK 53-05-02-211-853						
1. INTERNAL - DETAILED: VERTICAL FIN REAR SPAR ATTACHMENT FITTINGS						
A. Inspection						
NOTE: Remove vertical fin including primary and fail-safe bolts.						
SUBTASK 53-05-02-211-053						
(1) Do a Detailed inspection of the four primary fitting lugs, from the inside of the lugs, at STA 1088.						
See Doc D626A001-DTR, DTR check form 53-80-06-2 for alternative inspections.						
———— END OF TASK ————						

EFFECTIVITY AKS ALL	SOURCE AWL	VERTICAL FIN REAR SPAR ATTACHMENT FITTINGS
	D633A109-AKS 53-755-00-01	Page 2 of 2 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE BULKHEAD STATION 1156			BOEING CARD NO. 53-756-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL NOTE
		ACCESS 318BR			ZONE 313 314 317 318

Inspect (High Frequency Eddy Current) the edge of the outer chord and the web above and below the stab pivot line.

See Doc D626A001-DTR, DTR check form 53-80-07-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-85.

AIRPLANE NOTE: Applicable to L/N 1 to 1198.

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STATION 1156
		D633A109-AKS 53-756-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-756-00-01				
				MECH INSP				
TASK 53-05-02-250-915								
1. INTERNAL - SPECIAL DETAILED: BULKHEAD STA 1156								
A. Inspection								
SUBTASK 53-05-02-010-107								
(1) Open this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>318BR</td><td>Tailcone Access Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	318BR	Tailcone Access Door
<u>Number</u>	<u>Name/Location</u>							
318BR	Tailcone Access Door							
SUBTASK 53-05-02-250-115								
(2) Do a High Frequency Eddy Current inspection on the edge of the outer chord and the web above and below the stab pivot line.								
See Doc D626A001-DTR, DTR check form 53-80-07-1 for alternative inspections.								
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-85.								
SUBTASK 53-05-02-410-105								
(3) Close this access panel:								
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>318BR</td><td>Tailcone Access Door</td></tr></tbody></table>					<u>Number</u>	<u>Name/Location</u>	318BR	Tailcone Access Door
<u>Number</u>	<u>Name/Location</u>							
318BR	Tailcone Access Door							
<hr style="width: 20%; margin-left: auto; margin-right: 0;"/> END OF TASK <hr style="width: 20%; margin-left: 0; margin-right: auto;"/>								

EFFECTIVITY AKS ALL	SOURCE AWL	BULKHEAD STATION 1156
		D633A109-AKS 53-756-00-01

AKS



737-600/700/800/900 TASK CARDS

AIRLINE CARD NO		TITLE INTEGRATED BULKHEAD STA 1156			BOEING CARD NO. 53-757-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL				
		ACCESS 318BR NOTE			ZONE 313 314 317 318

Inspect (High Frequency Eddy Current) the web inboard along the failsafe strap from the top of the stabilizer attach fitting down 16 inches vertically.

See Doc D626A001-DTR, DTR check form 53-80-07-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-33.

AIRPLANE NOTE: Applicable to all aircrafts from L/N 1199 and on.

ACCESS NOTE: Enter aircraft through the tail cone access panel.

EFFECTIVITY AKS ALL	SOURCE AWL	INTEGRATED BULKHEAD STA 1156
		D633A109-AKS 53-757-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-757-00-01	MECH	INSP

TASK 53-05-02-250-916

1. INTERNAL - SPECIAL DETAILED: INTEGRATED BULKHEAD STA 1156

A. Inspection

SUBTASK 53-05-02-010-108

(1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
318BR	Tailcone Access Door

NOTE: Enter aircraft through the tail cone access panel.

SUBTASK 53-05-02-250-116

(2) Do a High Frequency Eddy Current inspection of the web inboard along the failsafe strap from the top of the stabilizer attach fitting down 16 inches vertically.

See Doc D626A001-DTR, DTR check form 53-80-07-2 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 53-11-33.

SUBTASK 53-05-02-410-106

(3) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
318BR	Tailcone Access Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE AWL	INTEGRATED BULKHEAD STA 1156
		D633A109-AKS 53-757-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT HORIZONTAL STABILIZER PIVOT FITTING			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-758-00-01 RELATED CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 313

Inspect (High Frequency Eddy Current) the pivot fitting beams around the pivot pins at the STA 1156 hinge beam.

See Doc D626A001-DTR, DTR check form 53-80-08-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-86.

ACCESS NOTE: Remove sliding seals for access.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT HORIZONTAL STABILIZER PIVOT FITTING
		D633A109-AKS 53-758-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-758-00-01
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TASK 53-05-02-250-917

MECH

INSP

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER PIVOT FITTING**A. Inspection**NOTE: Remove sliding seals for access.

SUBTASK 53-05-02-250-117

- (1) Do a High Frequency Eddy Current inspection of the pivot fitting beams around the pivot pins at the STA 1156 hinge beam.

See Doc D626A001-DTR, DTR check form 53-80-08-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-86.

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT HORIZONTAL STABILIZER PIVOT FITTING
		D633A109-AKS 53-758-00-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT HORIZONTAL STABILIZER PIVOT FITTING			BOEING CARD NO. 53-758-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 314

Inspect (High Frequency Eddy Current) the pivot fitting beams around the pivot pins at the STA 1156 hinge beam.

See Doc D626A001-DTR, DTR check form 53-80-08-1 for alternative inspections.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-86.

ACCESS NOTE: Remove sliding seals for access.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT HORIZONTAL STABILIZER PIVOT FITTING D633A109-AKS 53-758-00-02	Page 1 of 2 Oct 15/2014
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-758-00-02
TASK 53-05-02-250-917				MECH INSP
1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER PIVOT FITTING				
A. Inspection				
NOTE: Remove sliding seals for access.				
SUBTASK 53-05-02-250-117				
(1) Do a High Frequency Eddy Current inspection of the pivot fitting beams around the pivot pins at the STA 1156 hinge beam.				
See Doc D626A001-DTR, DTR check form 53-80-08-1 for alternative inspections.				
The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 53-10-86.				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT HORIZONTAL STABILIZER PIVOT FITTING	
		D633A109-AKS 53-758-00-02	Page 2 of 2 Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT HORIZONTAL STABILIZER PIVOT PINS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-759-00-01 RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 14000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 313

Inspect (Dye Penetrant) both the inner and outer pivot pins at STA 1156.

See Doc D626A001-DTR, DTR check form 53-80-08-2 for alternative inspections.

ACCESS NOTE: Removal and separation of pivot pins is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT HORIZONTAL STABILIZER PIVOT PINS
		D633A109-AKS 53-759-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-759-00-01
TASK 53-05-02-230-801				MECH INSP
1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER PIVOT PINS				
A. Inspection				
<p><u>NOTE:</u> Removal and separation of pivot pins is required to perform the inspection.</p> <p>SUBTASK 53-05-02-230-001</p> <p>(1) Do a Dye Penetrant of both the inner and outer pivot pins at STA 1156.</p> <p>See Doc D626A001-DTR, DTR check form 53-80-08-2 for alternative inspections.</p>				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT HORIZONTAL STABILIZER PIVOT PINS		
		D633A109-AKS 53-759-00-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE RIGHT HORIZONTAL STABILIZER PIVOT PINS			BOEING CARD NO.
DATE	TASK SPECIAL DETAILED				53-759-00-02 RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 14000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ENGINE ALL ALL
		ACCESS NOTE			ZONE 314

Inspect (Dye Penetrant) both the inner and outer pivot pins at STA 1156.

See Doc D626A001-DTR, DTR check form 53-80-08-2 for alternative inspections.

ACCESS NOTE: Removal and separation of pivot pins is required to perform the inspection.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT HORIZONTAL STABILIZER PIVOT PINS
		D633A109-AKS 53-759-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-759-00-02
TASK 53-05-02-230-801				MECH INSP
1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER PIVOT PINS				
A. Inspection				
<p><u>NOTE:</u> Removal and separation of pivot pins is required to perform the inspection.</p> <p>SUBTASK 53-05-02-230-001</p> <p>(1) Do a Dye Penetrant of both the inner and outer pivot pins at STA 1156. See Doc D626A001-DTR, DTR check form 53-80-08-2 for alternative inspections.</p> <p style="text-align: center;">———— END OF TASK ————</p>				
EFFECTIVITY AKS ALL				
SOURCE AWL				
RIGHT HORIZONTAL STABILIZER PIVOT PINS				
D633A109-AKS 53-759-00-02				
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LEFT SECTION 48 SKIN PANELS, STA 1088 TO 1156			BOEING CARD NO. 53-760-00-01	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 313	

Inspect (Detailed) the skin panels around the STA 1138 cutout.

See Doc D626A001-DTR, DTR check form 53-80-10-1 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	LEFT SECTION 48 SKIN PANELS, STA 1088 TO 1156
		D633A109-AKS 53-760-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-760-00-01
				MECH INSP
TASK 53-05-02-211-854				
1. EXTERNAL - DETAILED: SECTION 48 SKIN PANELS, STA 1088 TO STA 1156				
A. Inspection				
SUBTASK 53-05-02-211-054				
(1) Do a Detailed inspection of the skin panels around the STA 1138 cutout. See Doc D626A001-DTR, DTR check form 53-80-10-1 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	LEFT SECTION 48 SKIN PANELS, STA 1088 TO 1156		
		D633A109-AKS 53-760-00-01	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RIGHT SECTION 48 SKIN PANELS, STA 1088 TO 1156			BOEING CARD NO. 53-760-00-02	
DATE	TASK DETAILED				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 50000 FC	REPEAT 18000 FC	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 314	

Inspect (Detailed) the skin panels around the STA 1138 cutout.

See Doc D626A001-DTR, DTR check form 53-80-10-1 for alternative inspections.

EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT SECTION 48 SKIN PANELS, STA 1088 TO 1156
		D633A109-AKS 53-760-00-02

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-760-00-02
TASK 53-05-02-211-854				MECH INSP
1. EXTERNAL - DETAILED: SECTION 48 SKIN PANELS, STA 1088 TO STA 1156				
A. Inspection				
SUBTASK 53-05-02-211-054				
(1) Do a Detailed inspection of the skin panels around the STA 1138 cutout. See Doc D626A001-DTR, DTR check form 53-80-10-1 for alternative inspections.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE AWL	RIGHT SECTION 48 SKIN PANELS, STA 1088 TO 1156		
		D633A109-AKS 53-760-00-02	Page 2 of 2 Feb 15/2015	

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE LOWER FUSELAGE			BOEING CARD NO. 53-800-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 120 DY	REPEAT 120 DY	APPLICABILITY	
STATION	SKILL AIRPL	NOTE			AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 110 120 130 140	

Perform an external zonal inspection (GV) of the lower fuselage. The nose landing gear wheel well and main landing gear wheel well are also included. Inspection is accomplished from the ground, without the use of stands or ladders. No additional access panels required. (EZAP)

INTERVAL NOTE: The EZAP inspection requirement with interval 5500 FC/30 MO is satisfied by this zonal inspection.

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER FUSELAGE	
		D633A109-AKS 53-800-00-01	Page 1 of 6 Jun 15/2016

AKS

737-600/700/800/900

TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-800-00-01
				MECH INSP
► EWIS				
TASK 05-41-01-210-801				
1. EXTERNAL - ZONAL (GV): Lower Fuselage				
(Figure 1)				
A. General				
(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection				
SUBTASK 05-41-01-210-048				
(1) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.				
(2) Perform an external zonal inspection (GV) of the lower fuselage. The nose landing gear wheel well and main landing gear wheel well are also included. Inspection is accomplished from the ground, without the use of stands or ladders. No additional access panels required. (EZAP)				
SUBTASK 05-41-01-910-021				
(3) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER FUSELAGE	
		D633A109-AKS 53-800-00-01	Page 2 of 6 Feb 15/2015

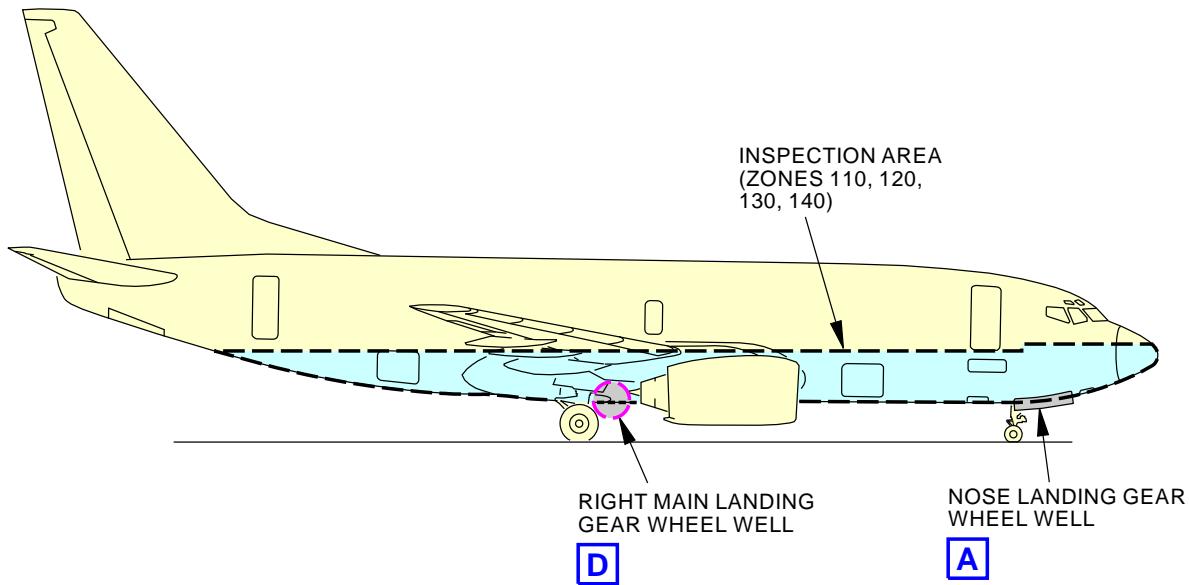
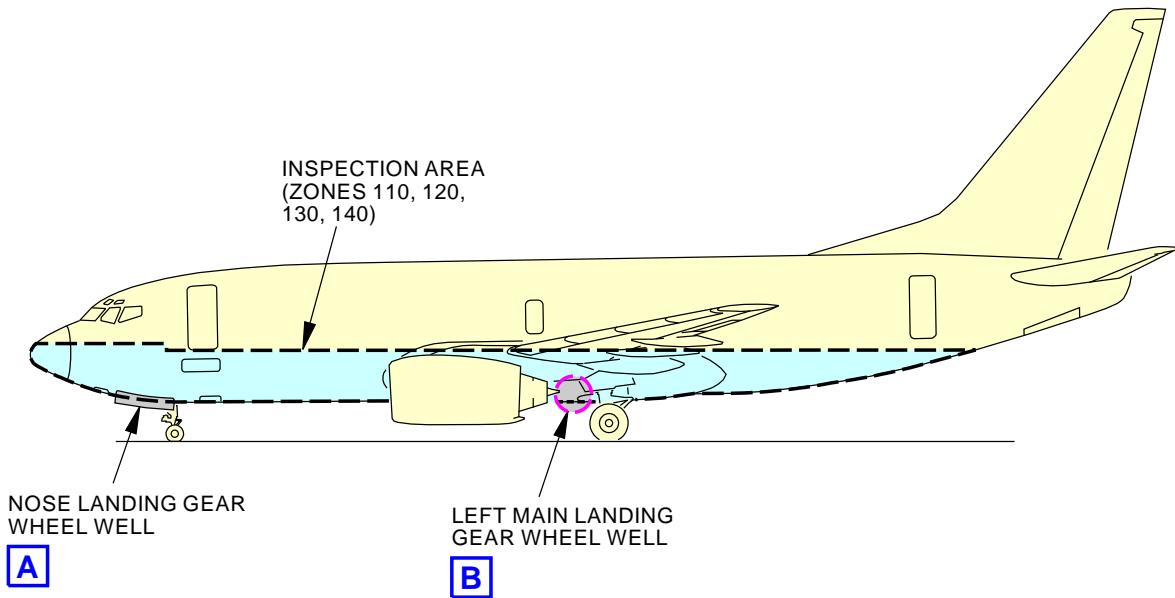
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

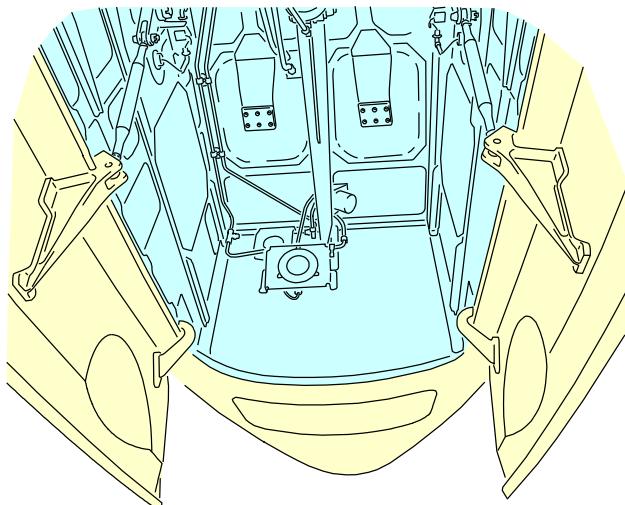
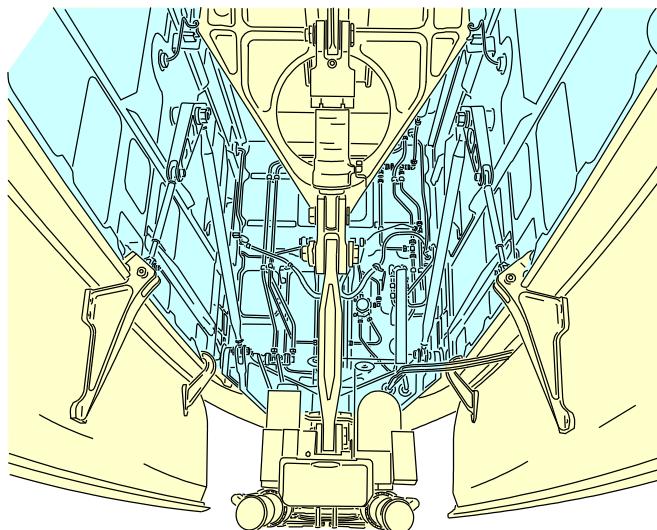
BOEING CARD NO.
53-800-00-01

K73171 S0006584030_V2

**Lower Half of Fuselage General Visual (External)
Figure 1 (Sheet 1 of 4)**EFFECTIVITY
AKS ALLSOURCE
MRB**LOWER FUSELAGE****D633A109-AKS
53-800-00-01****Page 3 of 6
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-800-00-01

**NOSE LANDING GEAR WHEEL WELL
(VIEW IN THE FORWARD DIRECTION)****A****NOSE LANDING GEAR WHEEL WELL
(VIEW IN THE AFT DIRECTION)****A****Lower Half of Fuselage General Visual (External)
Figure 1 (Sheet 2 of 4)**

K75955 S0006584031_V2

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER FUSELAGE
		D633A109-AKS 53-800-00-01

**Page 4 of 6
Feb 15/2015**

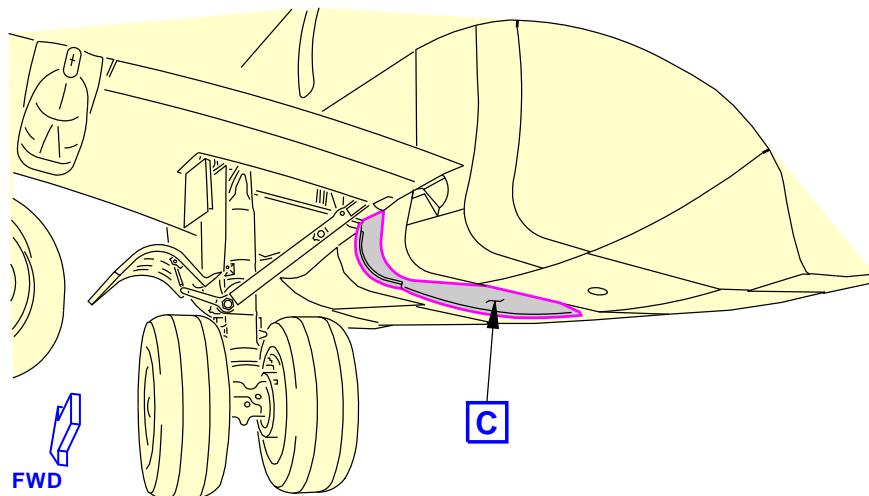
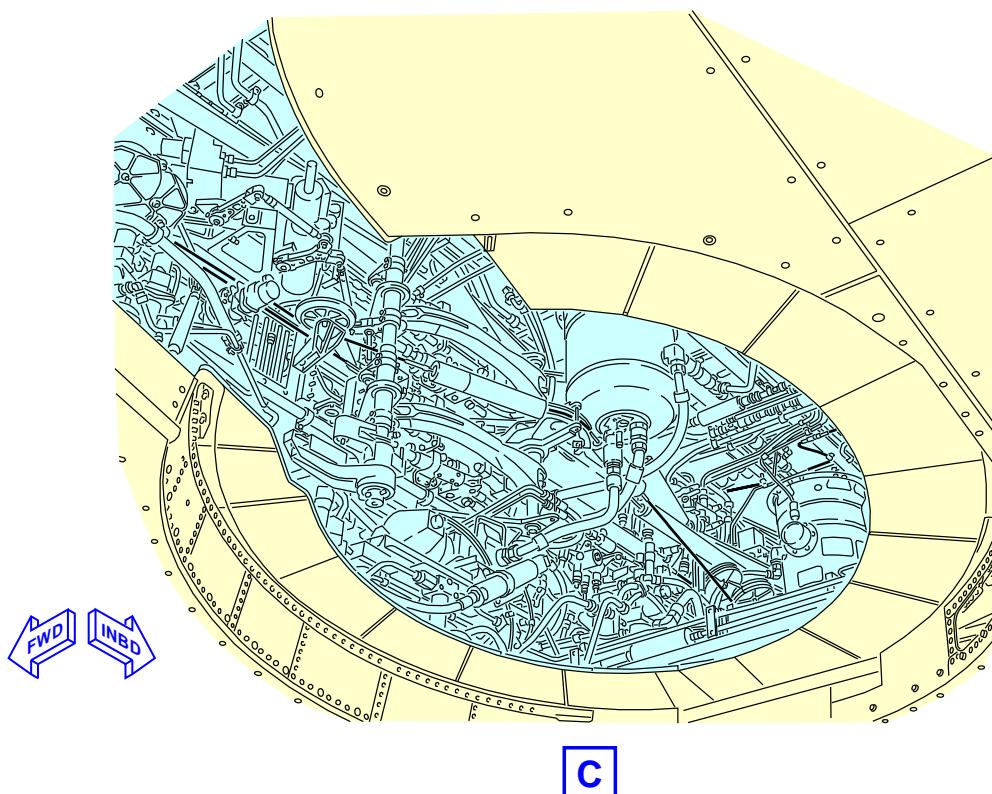
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-800-00-01**B****C**

1341201 S0006584032_V3

**Lower Half of Fuselage General Visual (External)
Figure 1 (Sheet 3 of 4)**EFFECTIVITY
AKS ALLSOURCE
MRB**LOWER FUSELAGE****D633A109-AKS
53-800-00-01****Page 5 of 6
Feb 15/2015**

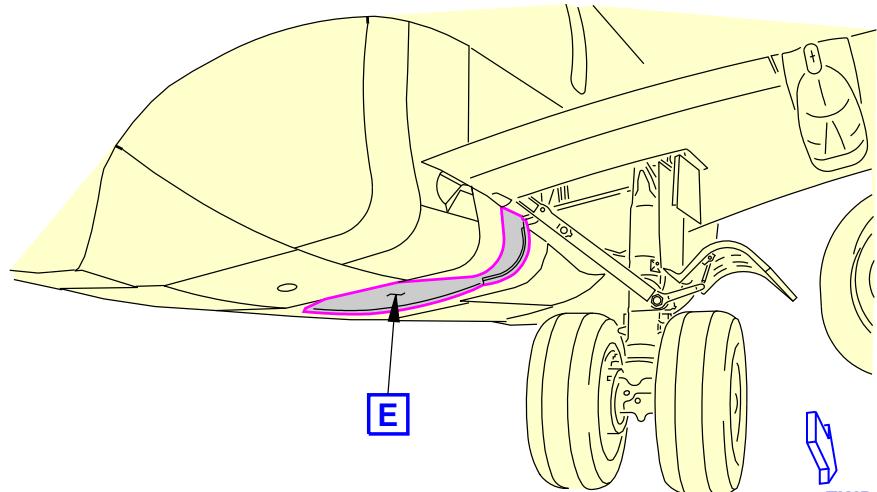
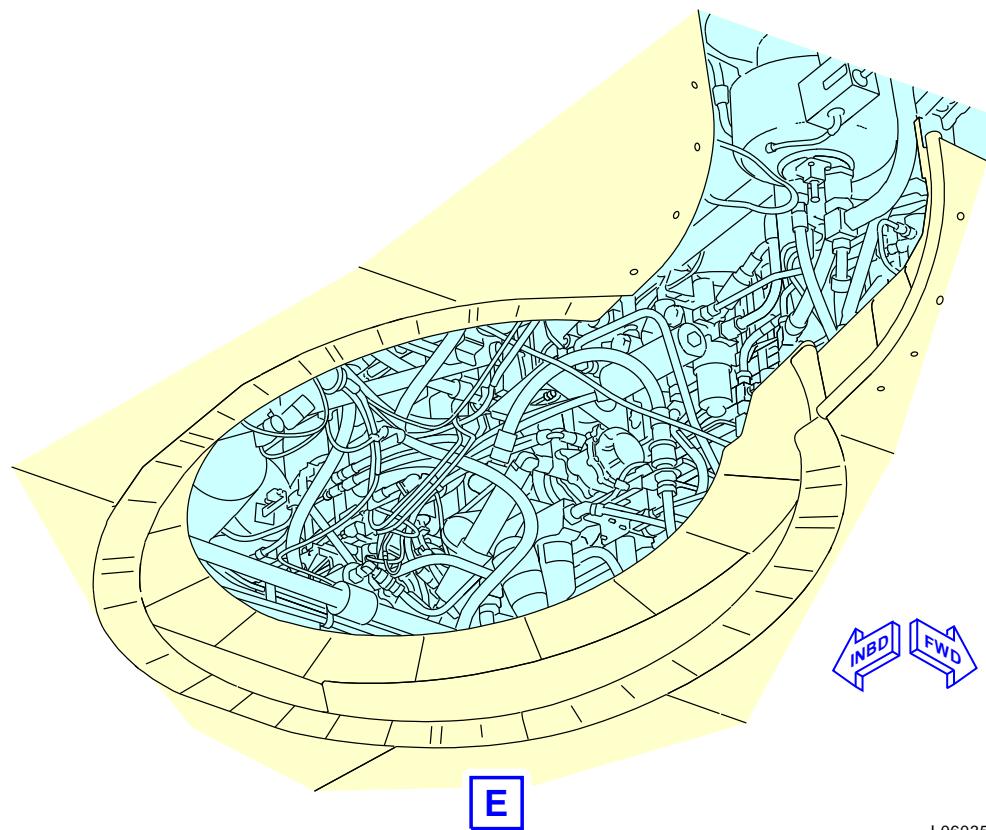
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-800-00-01**RIGHT MAIN LANDING GEAR WHEEL WELL****Lower Half of Fuselage General Visual (External)
Figure 1 (Sheet 4 of 4)**

L06035 S0006584033_V3

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER FUSELAGE
		D633A109-AKS 53-800-00-01

**Page 6 of 6
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE RADOME			BOEING CARD NO. 53-802-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	ACCESS 111			ZONE 111

Perform an internal zonal inspection (GV) of the radome.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 53-52-00-010-802	Nose Radome - Open (P/B 201)
AMM 53-52-00-410-802	Nose Radome - Close (P/B 201)

EFFECTIVITY AKS ALL	SOURCE MRB	RADOME D633A109-AKS 53-802-00-01	Page 1 of 3 Jun 15/2016
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AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-802-00-01				
				MECH INSP				
TASK 05-41-01-210-802								
1. INTERNAL - ZONAL (GV): RADOME								
(Figure 1)								
A. Zonal Inspection								
SUBTASK 05-41-01-010-002								
(1) Open this access panel:								
<table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>111</td> <td>Radome</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	111	Radome	
<u>Number</u>	<u>Name/Location</u>							
111	Radome							
(a) Do this task: Nose Radome - Open, AMM TASK 53-52-00-010-802.								
SUBTASK 05-41-01-210-002								
(2) Do a General Visual inspection of the radome.								
SUBTASK 05-41-01-410-002								
(3) Close this access panel:								
<table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>111</td> <td>Radome</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	111	Radome	
<u>Number</u>	<u>Name/Location</u>							
111	Radome							
(a) Do this task: Nose Radome - Close, AMM TASK 53-52-00-410-802.								
———— END OF TASK ——								

EFFECTIVITY AKS ALL	SOURCE MRB	RADOME	D633A109-AKS 53-802-00-01	Page 2 of 3 Jun 15/2016
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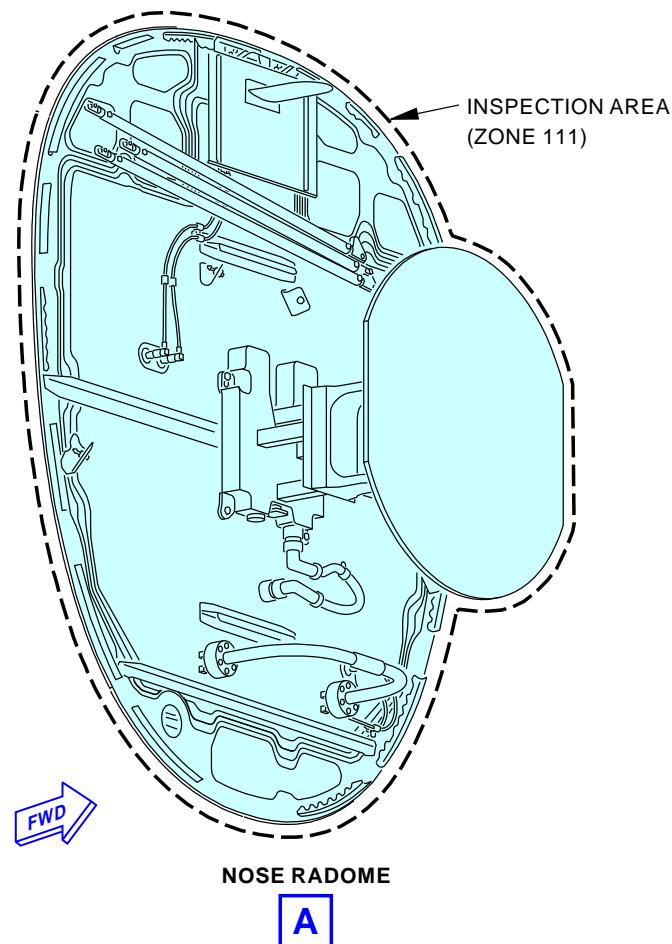
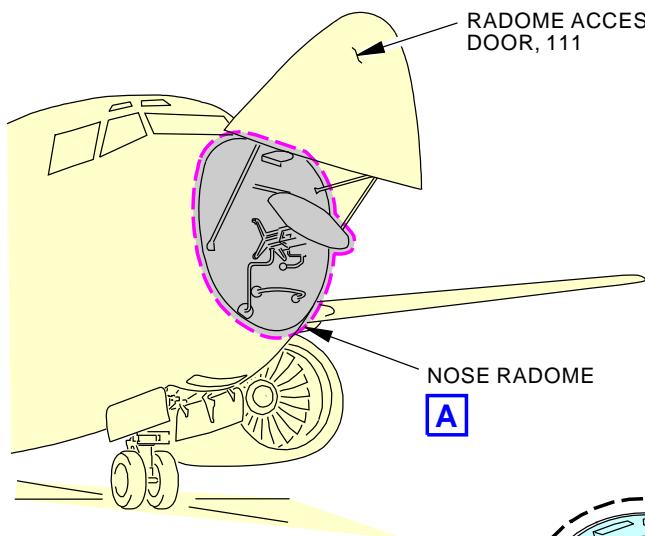
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-802-00-01

Nose Radome General Visual (Internal)
Figure 1

K94746 S0006584035_V2

EFFECTIVITY AKS ALL	SOURCE MRB	RADOME D633A109-AKS 53-802-00-01
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Page 3 of 3
Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA FORWARD OF NOSE WHEEL WELL			BOEING CARD NO.
DATE	TASK ZONAL (GV)				53-804-00-01
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 112A			ENGINE ALL
					ZONE 112

Perform an internal zonal inspection (GV) of the area forward of the nose wheel well - Section 41, Sta 178 to Sta 224.8. (EZAP)

INTERVAL NOTE: Whichever comes first. This EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

EFFECTIVITY AKS ALL	SOURCE MRB	AREA FORWARD OF NOSE WHEEL WELL
		D633A109-AKS 53-804-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-804-00-01
				MECH INSP
► EWIS TASK 05-41-01-210-803				
1. INTERNAL - ZONAL (GV): Area Forward of Nose Wheel Well (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection SUBTASK 05-41-01-010-003 (1) Open this access panel: Number Name/Location 112A Forward Access Door				
SUBTASK 05-41-01-210-003 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the area forward of the nose wheel well - Section 41, Sta 178 to Sta 224.8. (EZAP)				
SUBTASK 05-41-01-910-002 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-01-410-003 (5) Close this access panel: Number Name/Location 112A Forward Access Door				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA FORWARD OF NOSE WHEEL WELL	
		D633A109-AKS 53-804-00-01	Page 2 of 4 Feb 15/2015

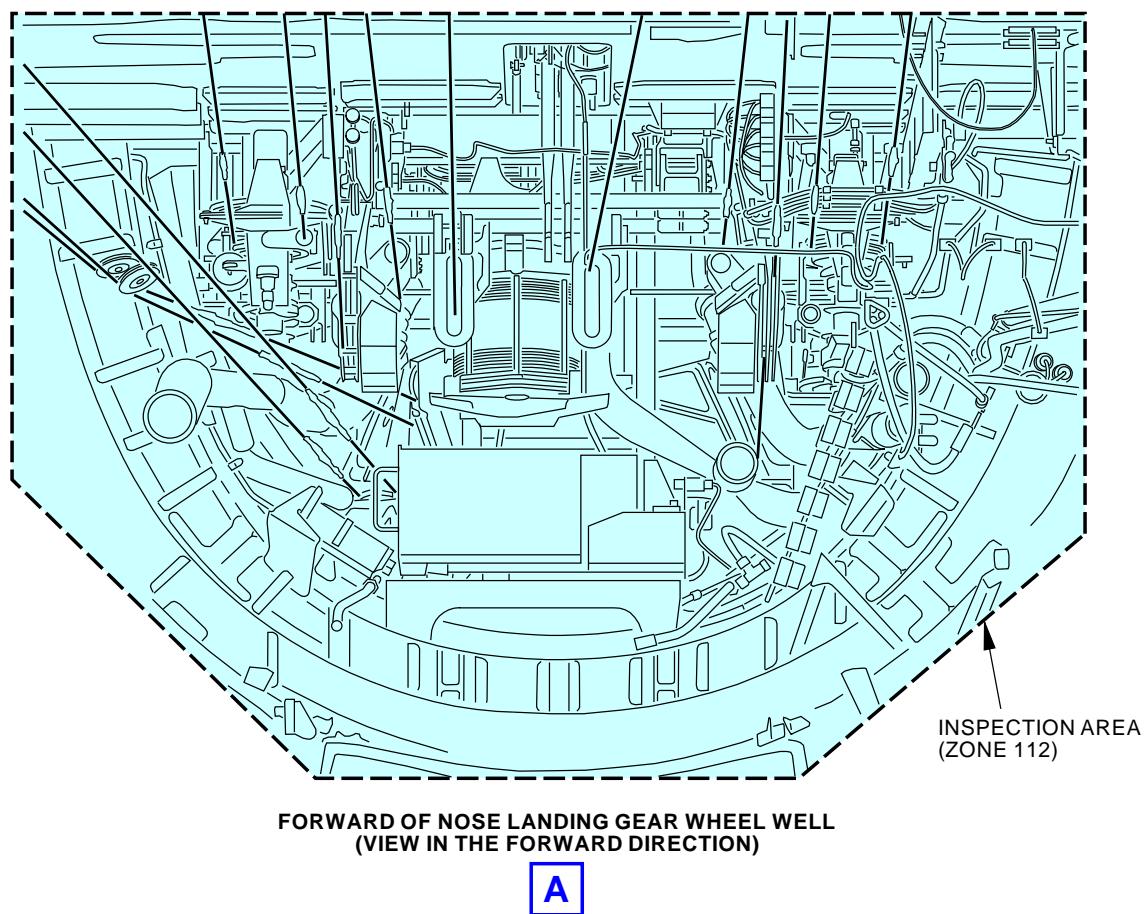
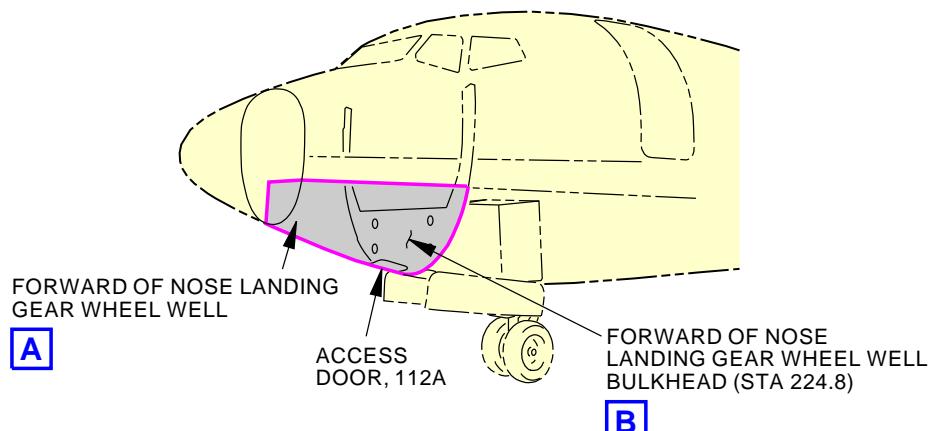
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-804-00-01

Forward of Nose Landing Gear Wheel Well General Visual (Internal)
Figure 1 (Sheet 1 of 2)

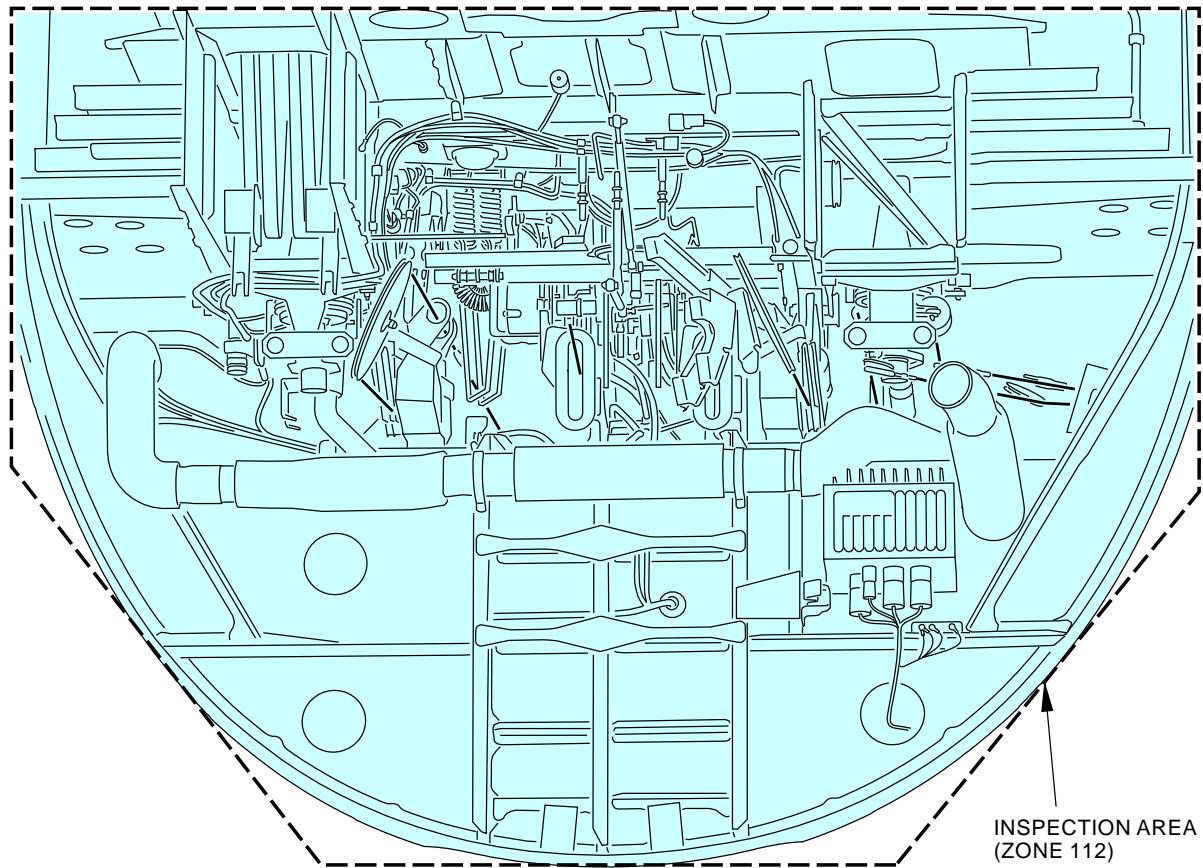
K94804 S0006584037_V2

EFFECTIVITY AKS ALL	SOURCE MRB	AREA FORWARD OF NOSE WHEEL WELL
		D633A109-AKS 53-804-00-01

Page 3 of 4
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-804-00-01
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**FORWARD OF NOSE LANDING GEAR WHEEL WELL BULKHEAD (STA 224.8)
(VIEW IN THE AFT DIRECTION)**

B

K94821 S0006584038_V2

**Forward of Nose Landing Gear Wheel Well General Visual (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA FORWARD OF NOSE WHEEL WELL
		D633A109-AKS 53-804-00-01

**Page 4 of 4
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL			BOEING CARD NO. 53-806-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 113AC 113AW 113BW 114AC 114AW 114BW			ZONE	ENGINE ALL
					113 114	

Perform an internal zonal inspection (GV) of the area above and outboard of the nose wheel well. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL
		D633A109-AKS 53-806-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-806-00-01
------	-------------	---------	------------------	--

► EWIS**TASK 05-41-01-210-804****1. INTERNAL - ZONAL (GV): Area Above and Outboard of Nose Wheel Well**

(Figure 1)

A. General

- (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.

B. Zonal Inspection

SUBTASK 05-41-01-010-004

- (1) Open these access panels:

<u>Number</u>	<u>Name/Location</u>
113AC	Fwd Nose Wheel Well Upper Access Panel
113AW	Forward Nose Wheel Well Panel
113BW	Forward Nose Wheel Well Panel
114AC	Fwd Nose Wheel Well Upper Access Panel
114AW	Forward Nose Wheel Well Panel
114BW	Forward Nose Wheel Well Panel

SUBTASK 05-41-01-210-004

- (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.
(3) Perform an internal zonal inspection (GV) of the area above and outboard of the nose wheel well. (EZAP)

SUBTASK 05-41-01-910-001

- (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.

SUBTASK 05-41-01-410-004

- (5) Close these access panels:

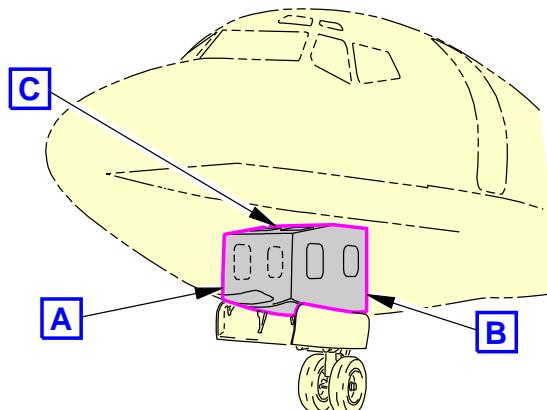
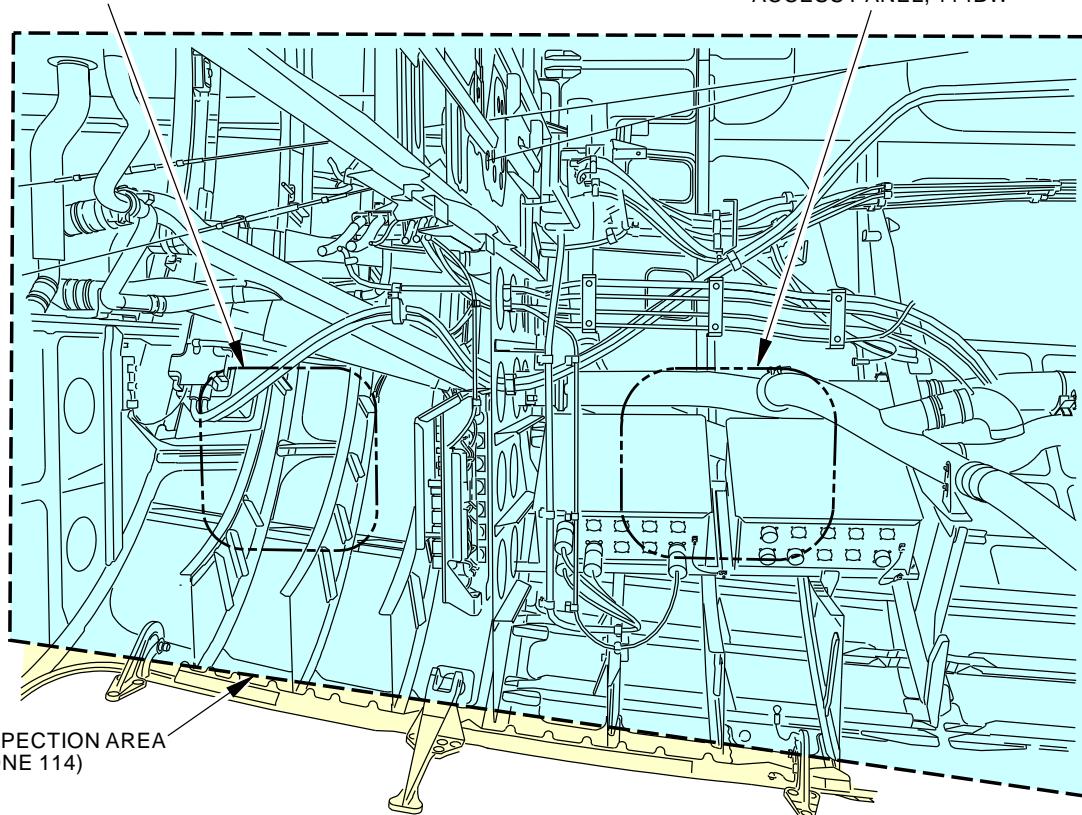
<u>Number</u>	<u>Name/Location</u>
113AC	Fwd Nose Wheel Well Upper Access Panel
113AW	Forward Nose Wheel Well Panel
113BW	Forward Nose Wheel Well Panel
114AC	Fwd Nose Wheel Well Upper Access Panel
114AW	Forward Nose Wheel Well Panel
114BW	Forward Nose Wheel Well Panel

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL	
		D633A109-AKS 53-806-00-01	Page 2 of 5 Feb 15/2015

AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-806-00-01
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FORWARD NOSE WHEEL WELL
ACCESS PANEL, 114AWAFT NOSE WHEEL WELL
ACCESS PANEL, 114BWNOSE LANDING GEAR WHEEL WELL
(OUTBOARD, RIGHT SIDE)**A**

K94456 S0006584040_V2

**Above and Outboard of the Nose Landing Gear Wheel Well General Visual (Internal)
Figure 1 (Sheet 1 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL
		D633A109-AKS 53-806-00-01

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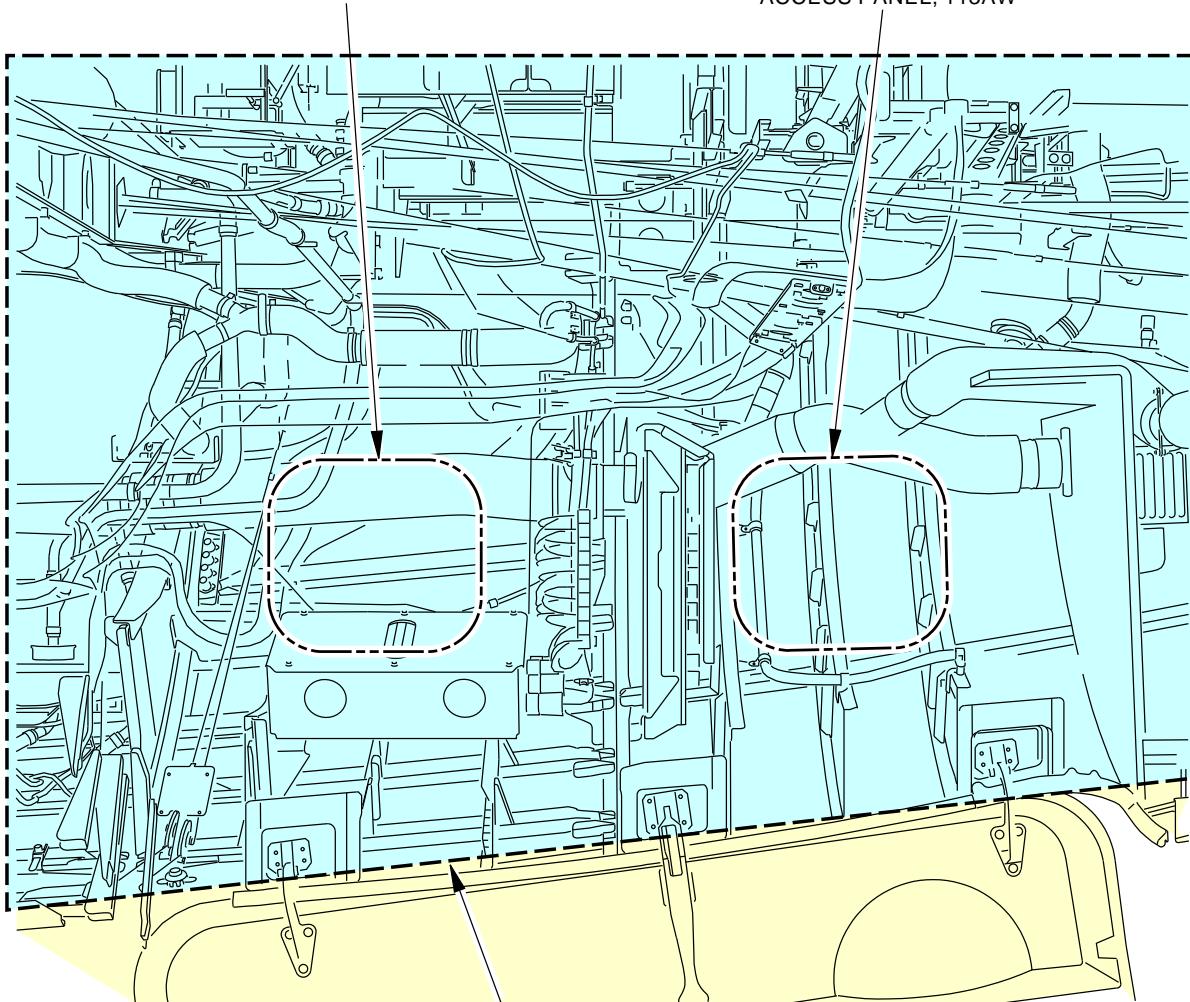
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

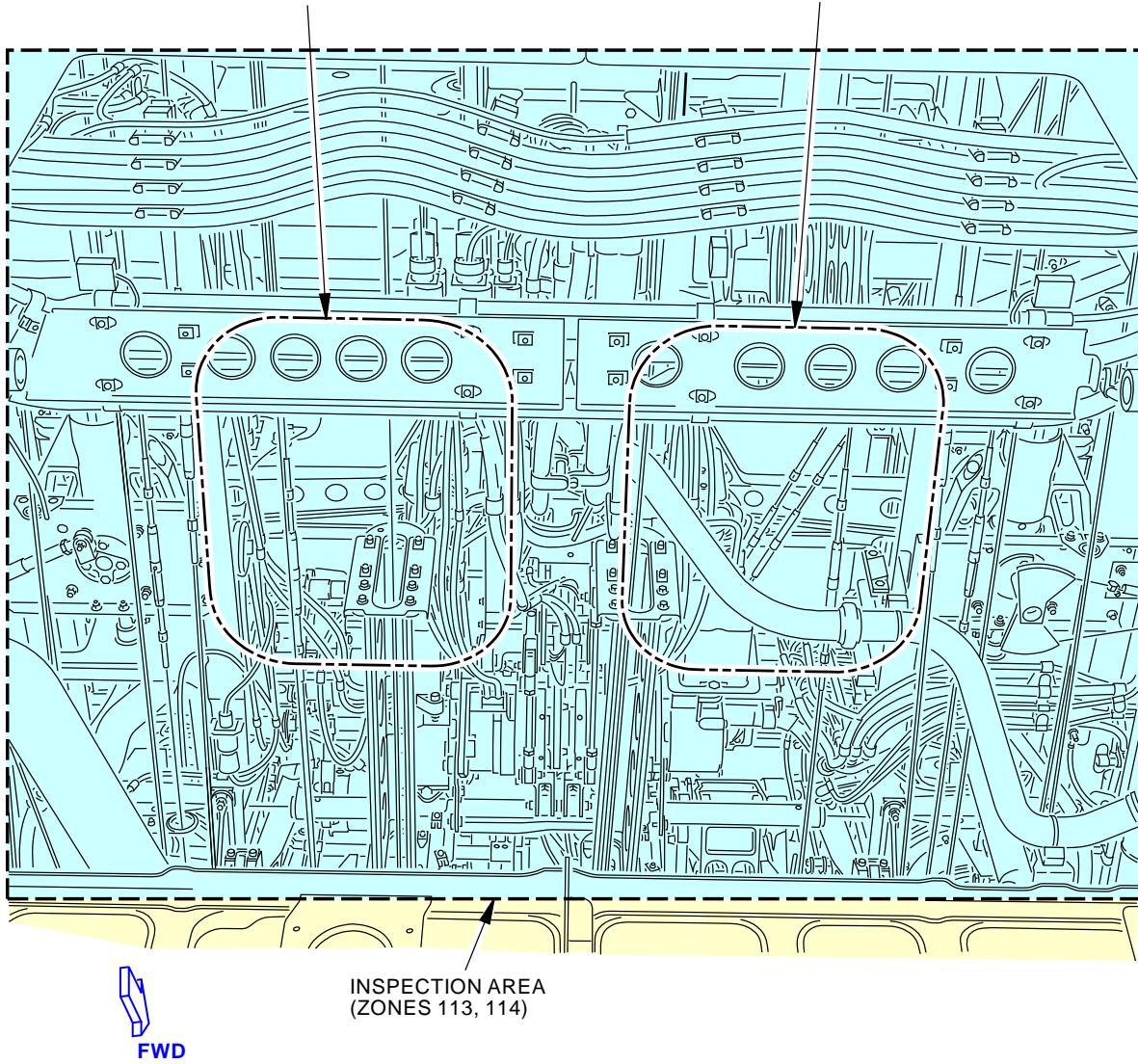
BOEING CARD NO.
53-806-00-01**AFT NOSE WHEEL WELL
ACCESS PANEL, 113BW****FORWARD NOSE WHEEL WELL
ACCESS PANEL, 113AW****NOSE LANDING GEAR WHEEL WELL
(OUTBOARD, LEFT SIDE)****B**

K94473 S0006584041_V3

**Above and Outboard of the Nose Landing Gear Wheel Well General Visual (Internal)
Figure 1 (Sheet 2 of 3)**EFFECTIVITY
AKS ALLSOURCE
MRB**AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL****D633A109-AKS
53-806-00-01****Page 4 of 5
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-806-00-01

UPPER NOSE WHEEL WELL
ACCESS PANEL, 113ACUPPER NOSE WHEEL WELL
ACCESS PANEL, 114ACABOVE THE NOSE LANDING GEAR WHEEL WELL
(VIEW IN THE UP DIRECTION)

K94475 S0006584042_V3

**Above and Outboard of the Nose Landing Gear Wheel Well General Visual (Internal)
Figure 1 (Sheet 3 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE AND OUTBOARD OF NOSE WHEEL WELL
		D633A109-AKS 53-806-00-01

**Page 5 of 5
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE ELECTRICAL AND ELECTRONICS COMPARTMENT			BOEING CARD NO. 53-808-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 117A 117BL NOTE				ENGINE ALL
		ACCESS 117A 117BL NOTE			ZONE 117 118	

Perform an internal zonal inspection (GV) of the electrical and electronics compartment. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

ACCESS NOTE: Forward airstair drip pan access panel must be removed if airstair installed. Access panel 117BL is only for airplanes with airstairs installed.

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT
		D633A109-AKS 53-808-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-808-00-01
				MECH INSP
► EWIS TASK 05-41-01-210-805				
1. INTERNAL - ZONAL (GV): Electrical and Electronics Compartment (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection SUBTASK 05-41-01-010-005 (1) Open these access panels: Number Name/Location 117A Electronic Equipment Access Door 117BL Forward Airstair Door NOTE: Forward airstair drip pan access panel must be removed if airstair installed. Access panel 117BL is only for airplanes with airstairs installed.				
SUBTASK 05-41-01-210-005 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the electrical and electronics compartment. (EZAP)				
SUBTASK 05-41-01-910-004 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-01-410-005 (5) Close these access panels: Number Name/Location 117A Electronic Equipment Access Door 117BL Forward Airstair Door				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT	
		D633A109-AKS 53-808-00-01	Page 2 of 4 Feb 15/2015

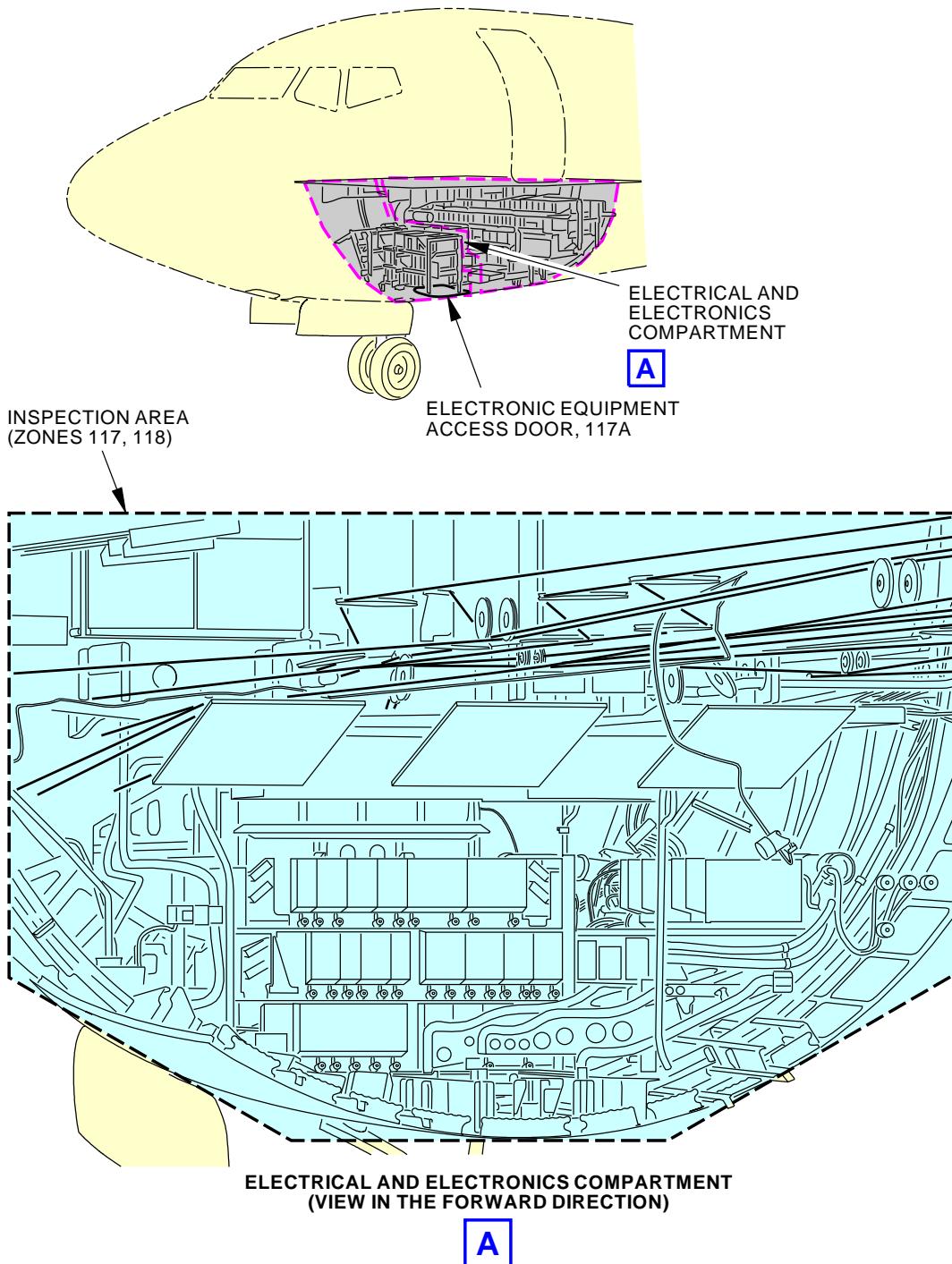
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

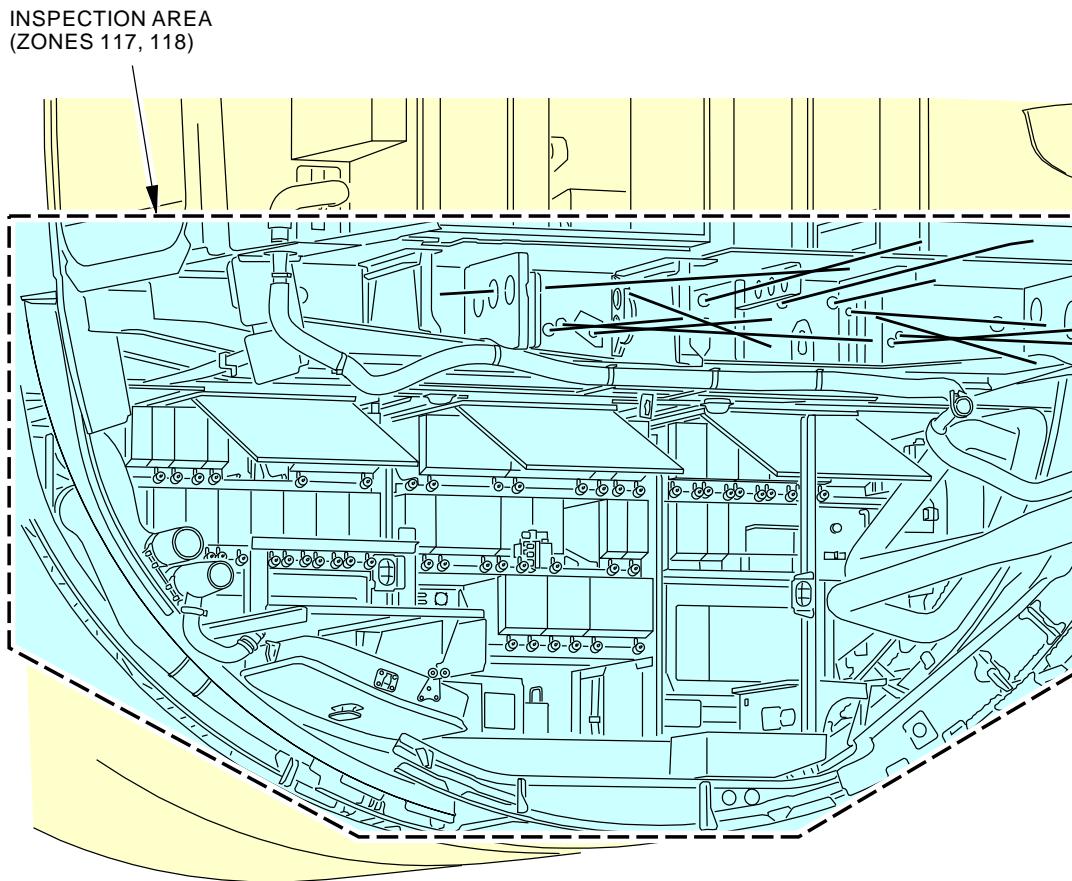
BOEING CARD NO.
53-808-00-01

K94950 S0006584044_V3

**Electrical and Electronics Compartment General Visual (Internal)
Figure 1 (Sheet 1 of 2)**EFFECTIVITY
AKS ALLSOURCE
MRB**ELECTRICAL AND ELECTRONICS COMPARTMENT****D633A109-AKS
53-808-00-01****Page 3 of 4
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-808-00-01



ELECTRICAL AND ELECTRONICS COMPARTMENT
(VIEW IN THE AFT DIRECTION)

A

K94973 S0006584045_V3

**Electrical and Electronics Compartment General Visual (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT
		D633A109-AKS 53-808-00-01

Page 4 of 4
Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE ELECTRICAL AND ELECTRONICS COMPARTMENT ACCESS DOOR			BOEING CARD NO. 53-810-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL				ENGINE ALL	
		ACCESS 117AW			ZONE 117 118	

Perform an internal zonal inspection (GV) of the electrical and electronics compartment access door.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT ACCESS DOOR D633A109-AKS 53-810-00-01	Page 1 of 3 Feb 15/2015
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-810-00-01
				MECH INSP
TASK 05-41-01-210-806				
1. INTERNAL - ZONAL (GV): ELECTRICAL AND ELECTRONICS COMPARTMENT ACCESS DOOR (Figure 1)				
A. Zonal Inspection				
SUBTASK 05-41-01-010-006				
(1) Open this access panel:				
Number Name/Location				
117AW Equipment Access Door Cover				
SUBTASK 05-41-01-210-006				
(2) Do a General Visual inspection of the electrical and electronics compartment access door.				
SUBTASK 05-41-01-410-006				
(3) Close this access panel:				
Number Name/Location				
117AW Equipment Access Door Cover				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE MRB	ELECTRICAL AND ELECTRONICS COMPARTMENT ACCESS DOOR		
		D633A109-AKS 53-810-00-01	Page 2 of 3 Feb 15/2015	

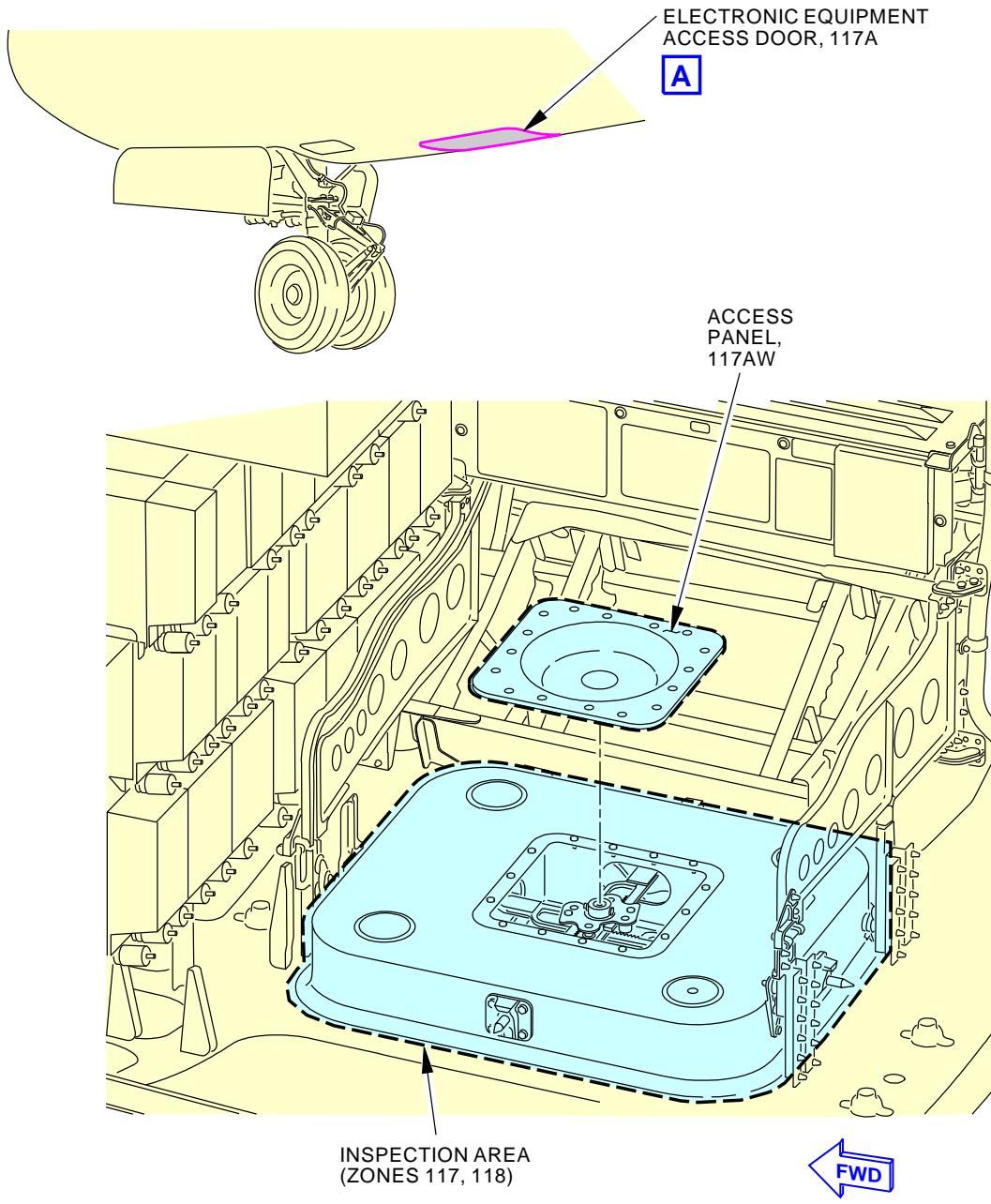
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-810-00-01**ELECTRICAL EQUIPMENT ACCESS DOOR, 117A****A**

K94845 S0006584047_V2

**Electrical Equipment Access Door General Visual (Internal)
Figure 1****EFFECTIVITY
AKS ALL****SOURCE
MRB****ELECTRICAL AND ELECTRONICS COMPARTMENT ACCESS
DOOR****D633A109-AKS
53-810-00-01****Page 3 of 3
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD CARGO COMPARTMENT			BOEING CARD NO. 53-812-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 120 DY	REPEAT 120 DY	APPLICABILITY AIRPLANE ENGINE ALL ALL NOTE
STATION	SKILL AIRPL	ACCESS 821 NOTE			ZONE 121 122

Perform an internal zonal inspection (GV) of the forward cargo compartment - section 43, STA 396 to forward cargo compartment aft bulkhead.

AIRPLANE NOTE: For aircraft equipped with optional auxiliary fuel tanks, removal of tanks not required.

ACCESS NOTE: No cargo liners removed.

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-812-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-812-00-01
TASK 05-41-01-210-807				MECH INSP

1. INTERNAL - ZONAL (GV): FORWARD CARGO COMPARTMENT
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-01-010-007

(1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
821	Forward Cargo Door

NOTE: No Cargo liners removed.

SUBTASK 05-41-01-210-007

(2) Do a General Visual inspection of the forward cargo compartment - Section 43, Sta 396 to forward cargo compartment aft bulkhead.

SUBTASK 05-41-01-410-007

(3) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
821	Forward Cargo Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-812-00-01

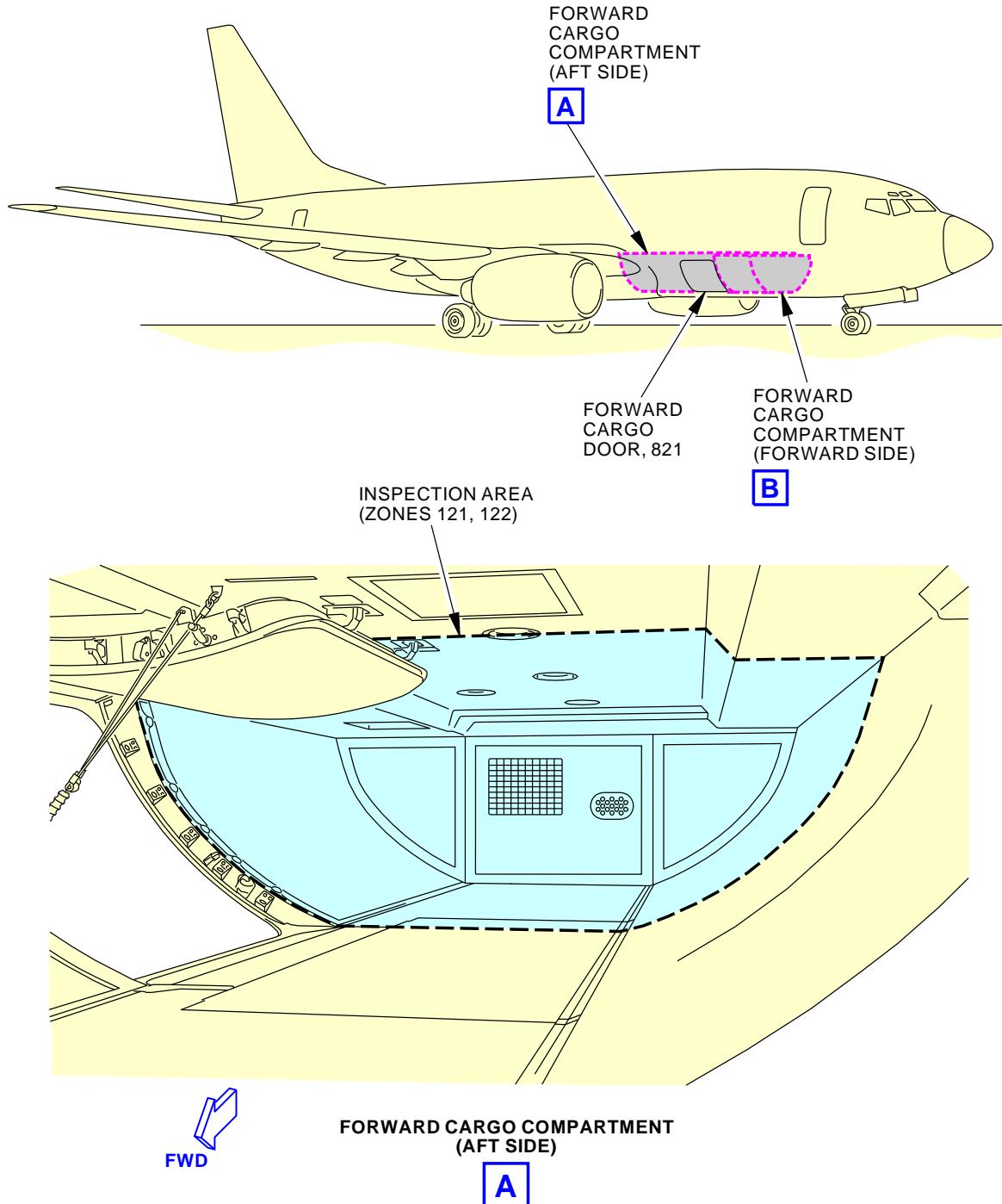
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-812-00-01

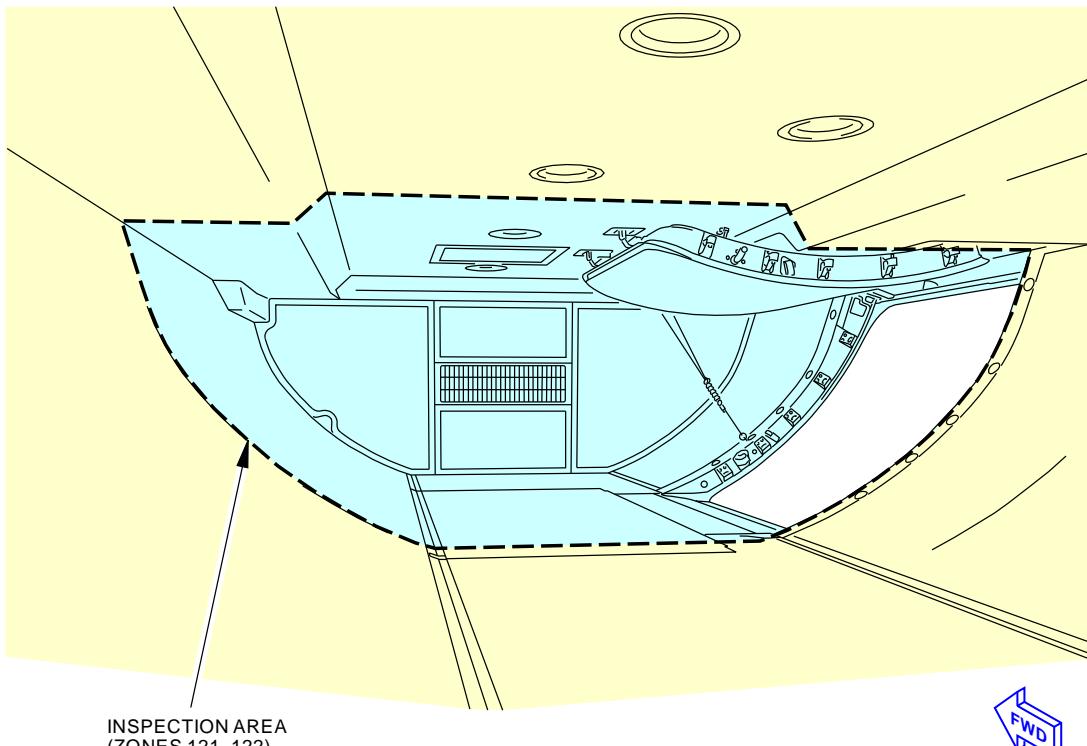
K67143 S0006584049_V2

**Forward Cargo Compartment General Visual (Internal)
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-812-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-812-00-01



K67355 S0006584050_V2

**Forward Cargo Compartment General Visual (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-812-00-01

Page 4 of 4
Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO COMPARTMENT			BOEING CARD NO. 53-814-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 13200 FC	REPEAT 13200 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL		72 MO	72 MO		
		ACCESS 821				ZONE 121 122
		NOTE				

Perform an internal zonal inspection (GV) of the area above the forward cargo compartment - section 43, STA 396 to forward cargo compartment aft bulkhead.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Ceiling panels removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-814-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-814-00-01
TASK 05-41-01-210-808				MECH INSP

1. INTERNAL - ZONAL (GV): FORWARD CARGO COMPARTMENT

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-01-010-008

- (1) Open this access panel:

Number Name/Location

821 Forward Cargo Door

NOTE: Ceiling panels removal required.

SUBTASK 05-41-01-210-008

- (2) Do a General Visual inspection of the area above the forward cargo compartment - Section 43, Sta 396 to forward cargo compartment aft bulkhead.

SUBTASK 05-41-01-410-008

- (3) Close this access panel:

Number Name/Location

821 Forward Cargo Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-814-00-01

Page 2 of 4
Feb 15/2015

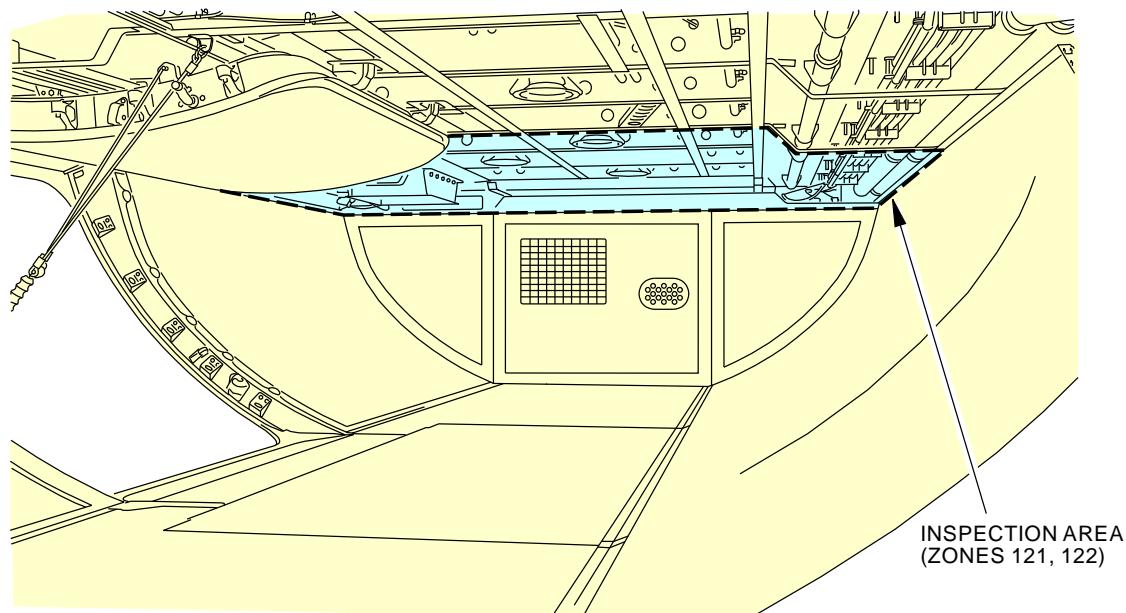
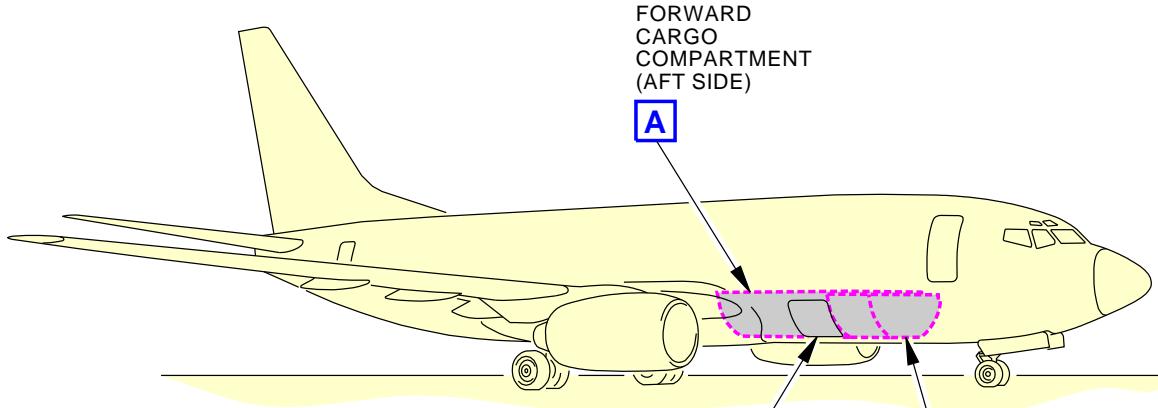
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

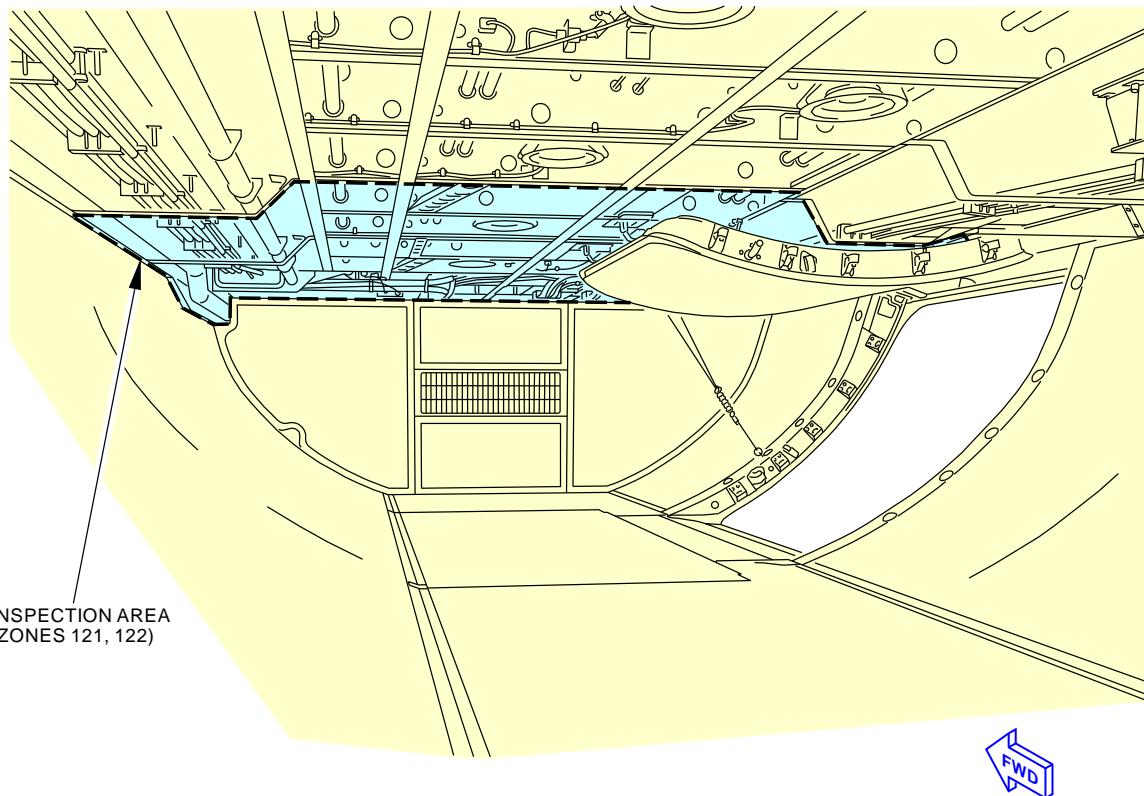
BOEING CARD NO.
53-814-00-01**FORWARD CARGO COMPARTMENT
(AFT SIDE)**

K68237 S0006584052_V2

**Forward Cargo Compartment General Visual (Internal) (Ceiling Liners Removed)
Figure 1 (Sheet 1 of 2)**EFFECTIVITY
AKS ALLSOURCE
MRB**FORWARD CARGO COMPARTMENT****D633A109-AKS
53-814-00-01****Page 3 of 4
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-814-00-01

**FORWARD CARGO COMPARTMENT
(FORWARD SIDE)****B**

K68684 S0006584053_V2

**Forward Cargo Compartment General Visual (Internal) (Ceiling Liners Removed)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-814-00-01

**Page 4 of 4
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO COMPARTMENT			BOEING CARD NO. 53-816-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 36000 FC 12 YR	REPEAT 36000 FC 12 YR	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 821 NOTE				ZONE 121 122

Perform an internal zonal inspection (GV) of the forward cargo compartment - section 43, STA 396 to forward cargo compartment aft bulkhead.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Sidewall panels removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-816-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-816-00-01
TASK 05-41-01-210-809				MECH INSP

1. INTERNAL - ZONAL (GV): FORWARD CARGO COMPARTMENT
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-01-010-009

(1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
821	Forward Cargo Door

NOTE: Sidewall panels removal required.

SUBTASK 05-41-01-210-009

(2) Do a General Visual inspection of the forward cargo compartment - Section 43, Sta 396 to forward cargo compartment aft bulkhead.

SUBTASK 05-41-01-410-009

(3) Close this access panel:

<u>Number</u>	<u>Name/Location</u>
821	Forward Cargo Door

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-816-00-01

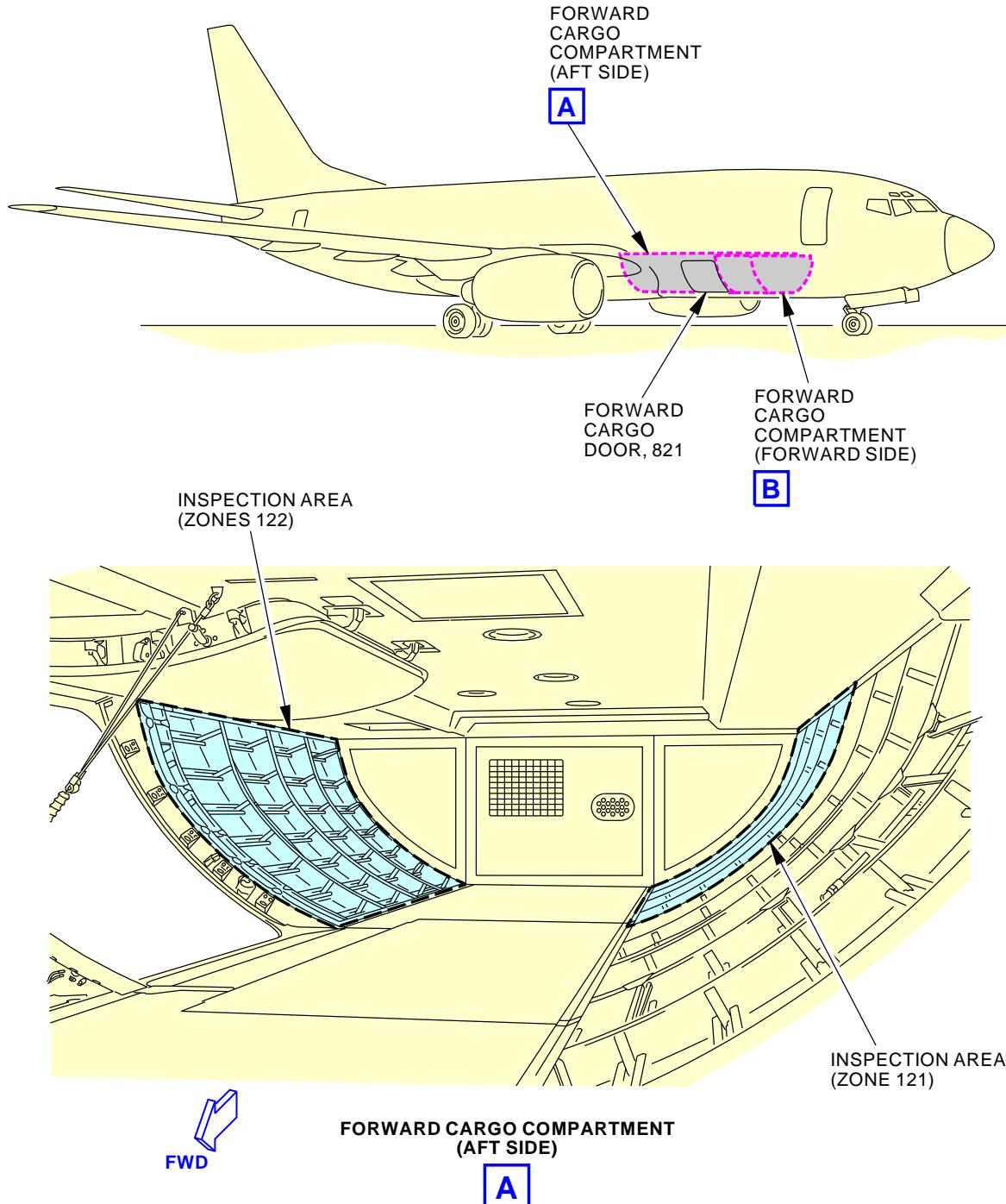
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-816-00-01

K69996 S0006584055_V2

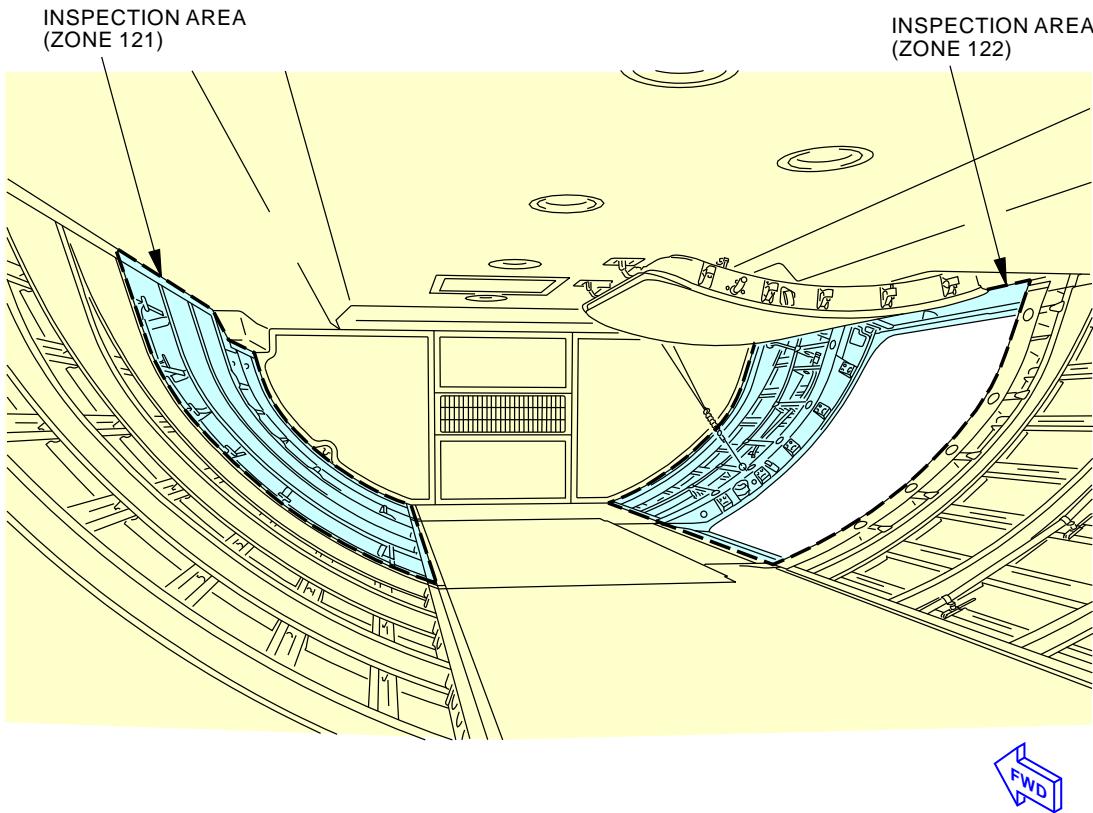
**Forward Cargo Compartment General Visual (Internal) (Sidewall Liners and Insulation Removed)
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-816-00-01

Page 3 of 4
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-816-00-01

**FORWARD CARGO COMPARTMENT
(FORWARD SIDE)****B**

K70175 S0006584056_V2

**Forward Cargo Compartment General Visual (Internal) (Sidewall Liners and Insulation Removed)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-816-00-01

Page 4 of 4
Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD CARGO DOOR SURROUND STRUCTURE FITTINGS AND STOPS			BOEING CARD NO. 53-818-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 2000 FC 240 DY	REPEAT 2000 FC 240 DY	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 821			ZONE 122	

Perform an external zonal inspection (GV) of the forward cargo door surround structure fittings and stops - section 43, STA 460.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR SURROUND STRUCTURE FITTINGS AND STOPS
		D633A109-AKS 53-818-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-818-00-01	MECH	INSP
TASK 05-41-01-210-810						
1. EXTERNAL - ZONAL (GV): FORWARD CARGO DOOR SURROUND STRUCTURE FITTINGS AND STOPS						
(Figure 1)						
A. Zonal Inspection						
SUBTASK 05-41-01-010-010						
(1) Open this access panel:						
Number Name/Location						
821 Forward Cargo Door						
SUBTASK 05-41-01-210-010						
(2) Do a General Visual inspection of the forward cargo door surround structure fittings and stops - Section 43, Sta 460.						
SUBTASK 05-41-01-410-010						
(3) Close this access panel:						
Number Name/Location						
821 Forward Cargo Door						
———— END OF TASK ————						
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR SURROUND STRUCTURE FITTINGS AND STOPS				
		D633A109-AKS				
		53-818-00-01				

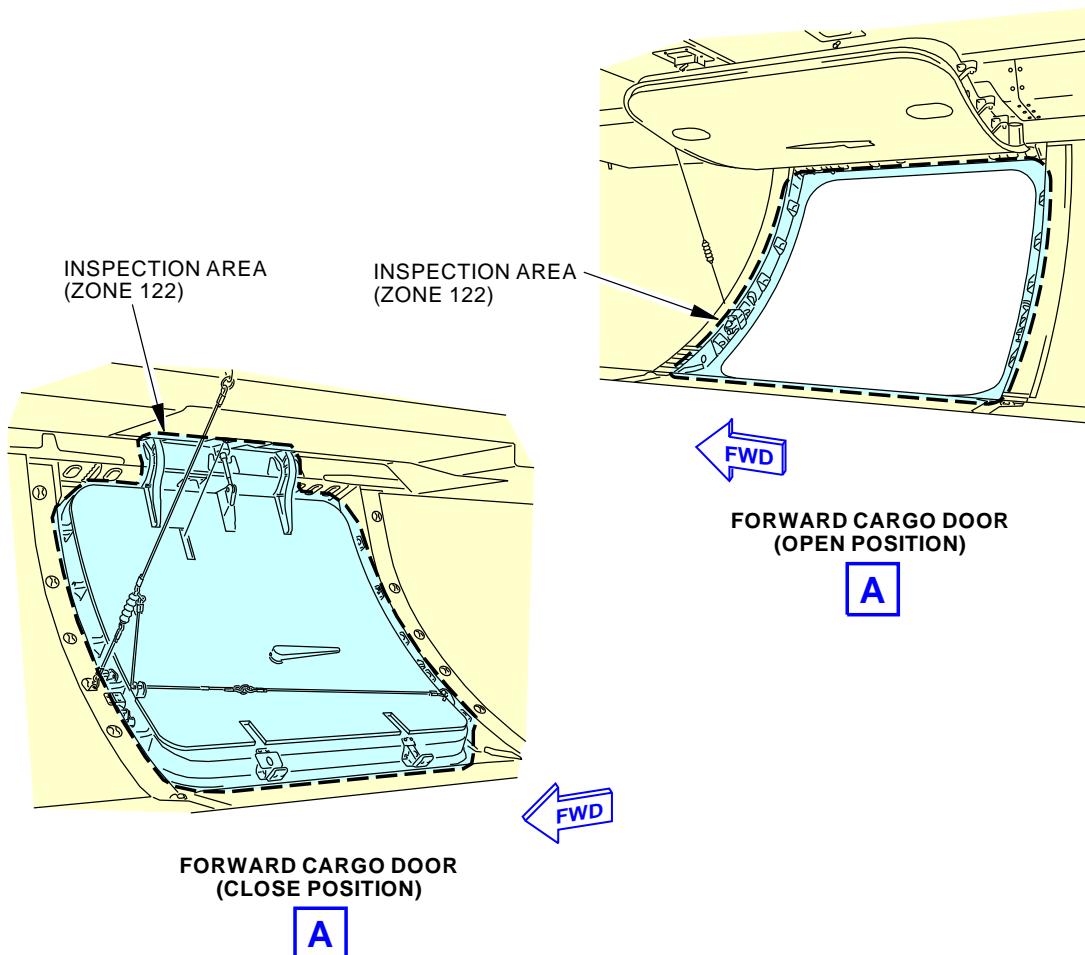
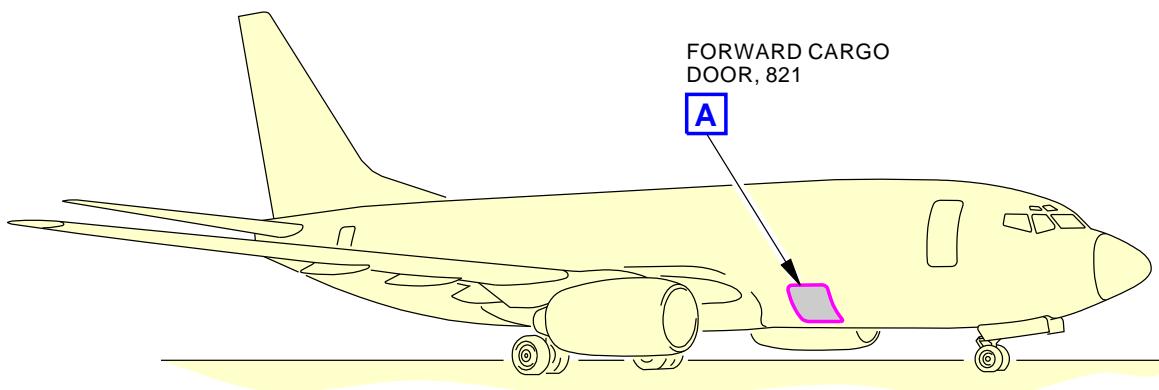
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-818-00-01

L06002 S0006584058_V2

**Fittings and Stops - Forward Cargo Door Surround Structure General Visual (External)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD CARGO DOOR SURROUND STRUCTURE FITTINGS AND STOPS
		D633A109-AKS 53-818-00-01

Page 3 of 3
Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA BELOW FORWARD CARGO COMPARTMENT			BOEING CARD NO. 53-820-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 13200 FC 72 MO	REPEAT 13200 FC 72 MO	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 821 NOTE			ZONE 123 124	

Perform an internal zonal inspection (GV) of the area below the forward cargo compartment - Section 43, Sta 396 to forward cargo compartment aft bulkhead. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

ACCESS NOTE: Center floor panels removal required. Cargo loading system removed/displaced as required.

EFFECTIVITY AKS ALL	SOURCE MRB	AREA BELOW FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-820-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-820-00-01
				MECH INSP
► EWIS TASK 05-41-01-210-811				
1. INTERNAL - ZONAL (GV): Area Below Forward Cargo Compartment (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection SUBTASK 05-41-01-010-011 (1) Open this access panel: Number Name/Location 821 Forward Cargo Door NOTE: Center floor panels removal required. Cargo loading system removed/displaced as required.				
SUBTASK 05-41-01-210-011 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the area below the forward cargo compartment - Section 43, Sta 396 to forward cargo compartment aft bulkhead. (EZAP)				
SUBTASK 05-41-01-910-005 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-01-410-011 (5) Close this access panel: Number Name/Location 821 Forward Cargo Door				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA BELOW FORWARD CARGO COMPARTMENT	
		D633A109-AKS 53-820-00-01	Page 2 of 4 Feb 15/2015

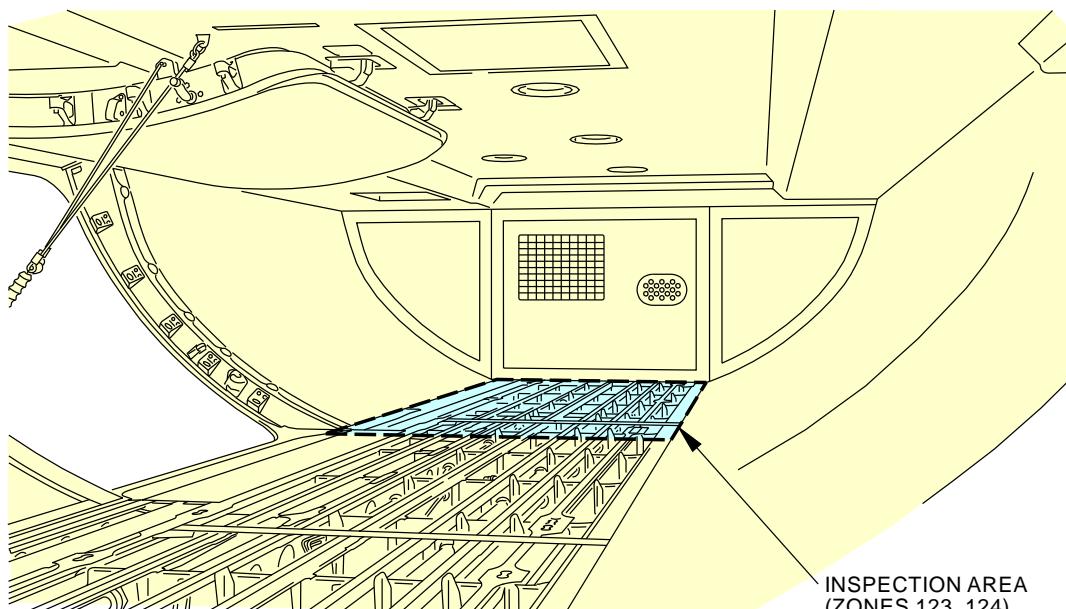
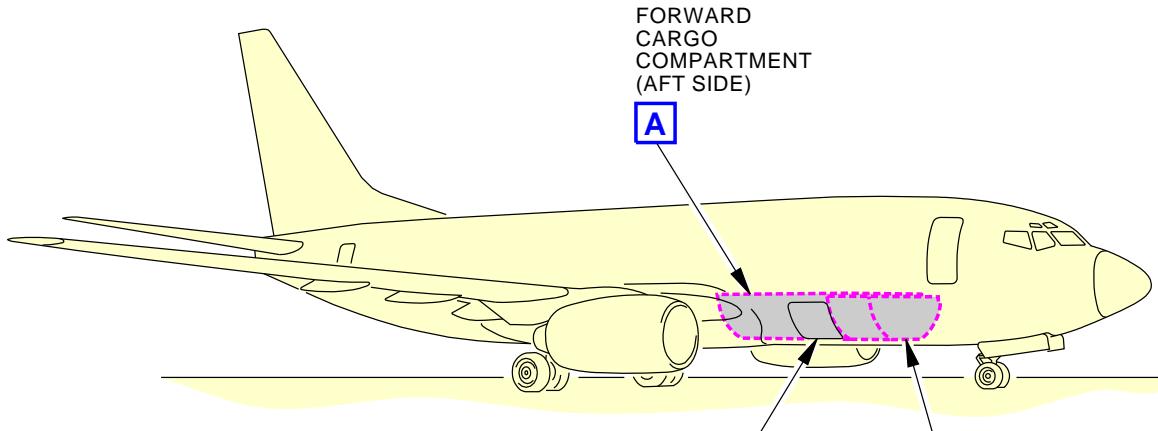
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

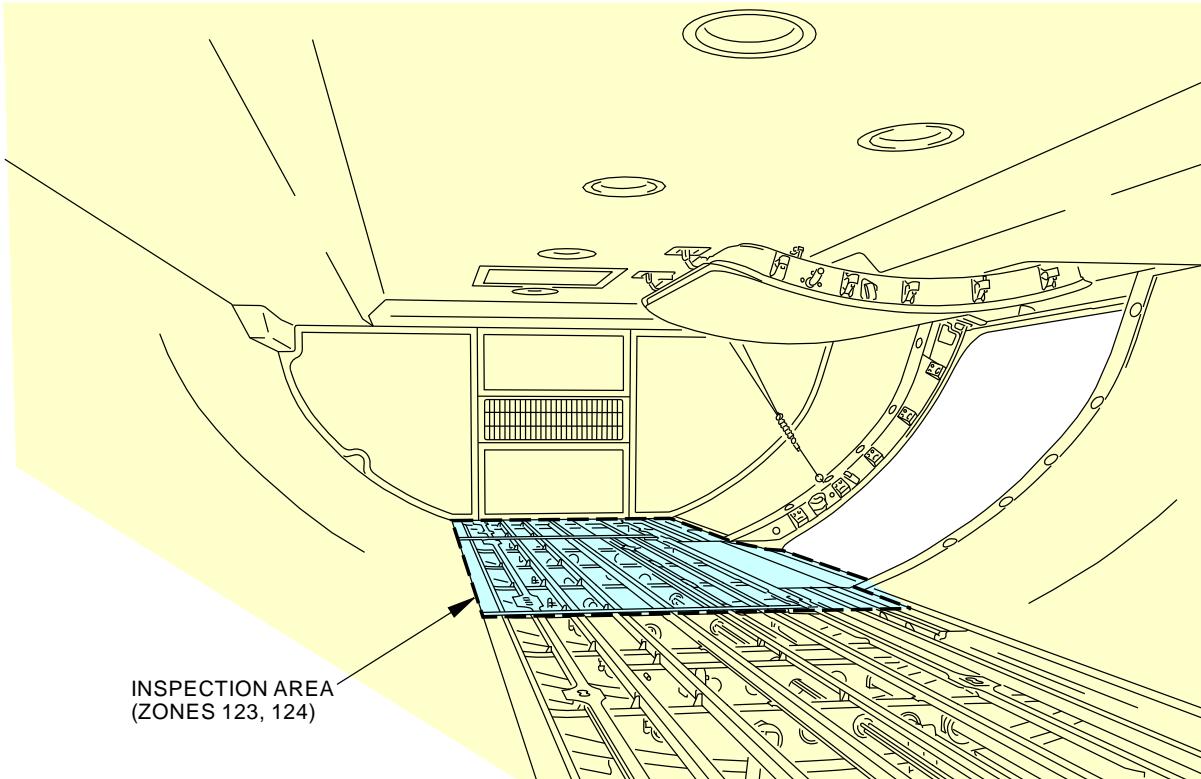
BOEING CARD NO.
53-820-00-01 **FWD****FORWARD CARGO COMPARTMENT
(AFT SIDE)**

K70466 S0006584060_V2

**Below the Forward Cargo Compartment General Visual (Internal) (Floor Panels Removed)
Figure 1 (Sheet 1 of 2)**EFFECTIVITY
AKS ALLSOURCE
MRB**AREA BELOW FORWARD CARGO COMPARTMENT****D633A109-AKS
53-820-00-01****Page 3 of 4
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-820-00-01

**FORWARD CARGO COMPARTMENT
(FORWARD SIDE)****B**

K70472 S0006584061_V2

**Below the Forward Cargo Compartment General Visual (Internal) (Floor Panels Removed)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA BELOW FORWARD CARGO COMPARTMENT
		D633A109-AKS 53-820-00-01

**Page 4 of 4
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AIR CONDITIONING DISTRIBUTION BAY			BOEING CARD NO. 53-822-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	
STATION	SKILL AIRPL	ACCESS 821 NOTE			AIRPLANE ALL	ENGINE ALL
					ZONE 125 126	

Perform an internal zonal inspection (GV) of the air conditioning distribution bay - Section 43, forward cargo compartment bulkhead to Sta 540. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: Forward cargo compartment aft bulkhead center panel removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	AIR CONDITIONING DISTRIBUTION BAY
		D633A109-AKS 53-822-00-01

AKS

737-600/700/800/900

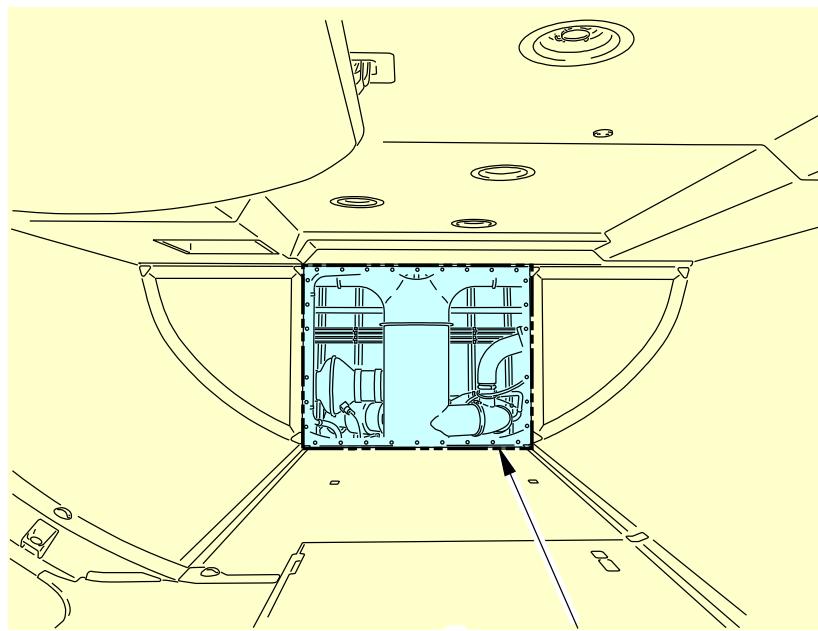
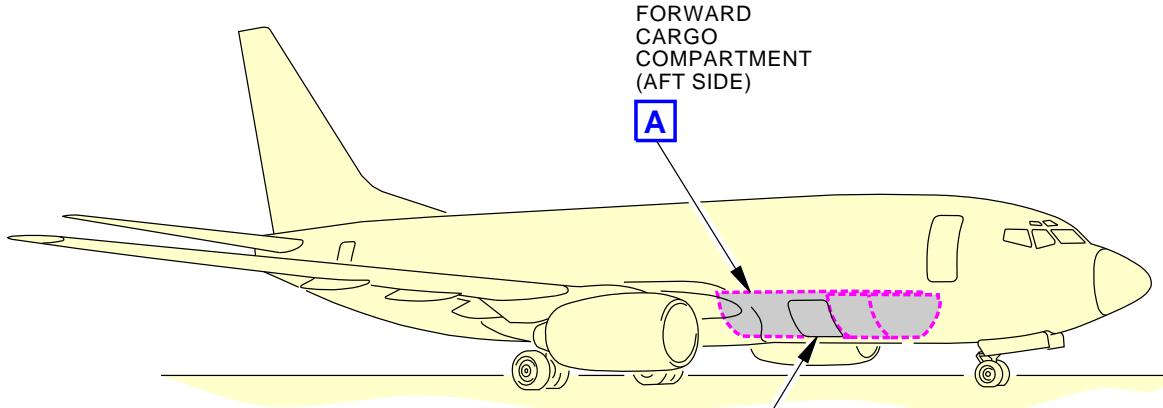
TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-822-00-01
				MECH INSP
► EWIS				
TASK 05-41-01-210-812				
1. INTERNAL - ZONAL (GV): Air Conditioning Distribution Bay (Figure 1)				
A. General				
(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection				
SUBTASK 05-41-01-010-012				
(1) Open this access panel:				
Number Name/Location				
821 Forward Cargo Door				
NOTE: Forward cargo compartment aft bulkhead center panel removal required.				
SUBTASK 05-41-01-210-012				
(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.				
(3) Perform an internal zonal inspection (GV) of the air conditioning distribution bay - Section 43, forward cargo compartment bulkhead to Sta 540. (EZAP)				
SUBTASK 05-41-01-910-006				
(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-01-410-012				
(5) Close this access panel:				
Number Name/Location				
821 Forward Cargo Door				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	AIR CONDITIONING DISTRIBUTION BAY
		D633A109-AKS 53-822-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-822-00-01
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K72789 S0006584063_V2

**Aft of Forward Cargo Compartment General Visual (Internal) (Aft Bulkhead Center Panel Removed)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	AIR CONDITIONING DISTRIBUTION BAY
		D633A109-AKS 53-822-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA ABOVE CENTER SECTION WING BOX			BOEING CARD NO. 53-824-01-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 36000 FC 12 YR	REPEAT 36000 FC 12 YR	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS NOTE				ENGINE ALL
					ZONE 135 136	

Perform an internal zonal inspection (GV) of the area above the center section wing box - section 44, STA 540 to STA 663.75.

Note: Access for this task is also provided when accomplishing tasks 53-878-00-01 and 53-880-00-01.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Floor panels removal required. Insulation as required.

EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE CENTER SECTION WING BOX	
		D633A109-AKS 53-824-01-01	Page 1 of 2 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-824-01-01
				MECH INSP
TASK 05-41-01-210-813				
1. INTERNAL - ZONAL (GV): AREA ABOVE CENTER SECTION WING BOX				
A. Zonal Inspection				
NOTE: Floor panels removal required. Insulation as required.				
SUBTASK 05-41-01-210-013				
(1) Do a General Visual inspection of the area above the center section wing box - Section 44, Sta 540 to Sta 663.75.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE MRB	AREA ABOVE CENTER SECTION WING BOX		
		D633A109-AKS 53-824-01-01	Page 2 of 2 Feb 15/2015	

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PRESSURE DECK ABOVE MAIN LANDING GEAR WHEEL WELL			BOEING CARD NO. 53-826-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 13200 FC 72 MO	REPEAT 13200 FC 72 MO	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS NOTE				ENGINE ALL
					ZONE 137 138	

Perform an internal zonal inspection (GV) of the pressure deck above the main landing gear wheel well - Section 44, Sta 663.75 to Sta 727. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: Floor panels removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	PRESSURE DECK ABOVE MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-826-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-826-00-01
------	-------------	---------	------------------	--

► EWIS**TASK 05-41-01-210-814****1. INTERNAL - ZONAL (GV): Pressure Deck Above Main Landing Gear Wheel Well**

(Figure 1)

A. General

- (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.

B. Zonal Inspection

NOTE: Floor panels removal required.

SUBTASK 05-41-01-210-014

- (1) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.
- (2) Perform an internal zonal inspection (GV) of the pressure deck above the main landing gear wheel well - Section 44, Sta 663.75 to Sta 727. (EZAP)

SUBTASK 05-41-01-910-007

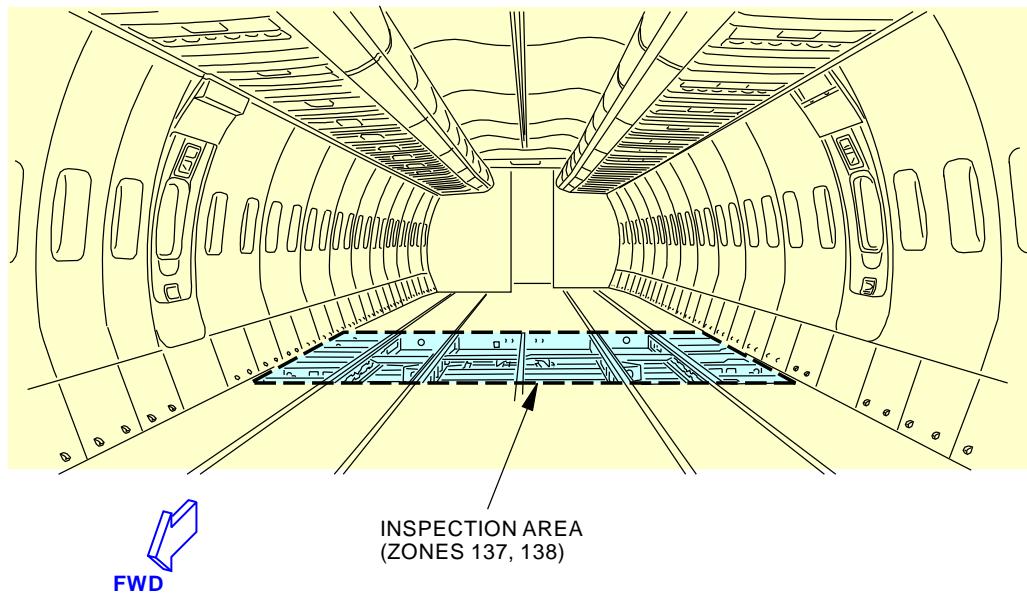
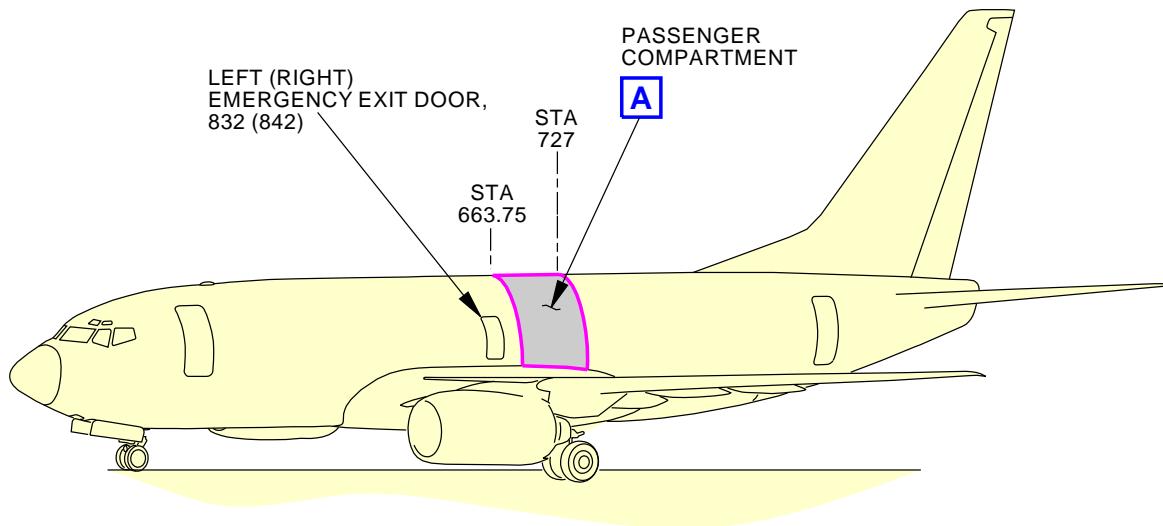
- (3) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	PRESSURE DECK ABOVE MAIN LANDING GEAR WHEEL WELL	
		D633A109-AKS 53-826-00-01	Page 2 of 3 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-826-00-01



Passenger Compartment (Sta 663.75-727) General Visual (Internal) (Floor Panels and Insulation Removed)
Figure 1

L07042 S0006584066_V2

EFFECTIVITY AKS ALL	SOURCE MRB	PRESSURE DECK ABOVE MAIN LANDING GEAR WHEEL WELL
		D633A109-AKS 53-826-00-01

Page 3 of 3
Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE KEEL BEAM (PART) STA 540 TO 727			BOEING CARD NO. 53-828-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 192CL 192CR			ZONE 139	

Perform an internal zonal inspection (GV) of the keel beam (part) Sta 540 to 727 - Section 44. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM (PART) STA 540 TO 727
		D633A109-AKS 53-828-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-828-00-01						
				MECH INSP						
► EWIS TASK 05-41-01-210-815										
1. INTERNAL - ZONAL (GV): Keel Beam (Part) Sta 540 to 727 (Figure 1)										
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.										
B. Zonal Inspection SUBTASK 05-41-01-010-013 (1) Open these access panels: <table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>192CL</td><td>ECS Access Door</td></tr><tr><td>192CR</td><td>ECS Access Door</td></tr></tbody></table>				<u>Number</u>	<u>Name/Location</u>	192CL	ECS Access Door	192CR	ECS Access Door	
<u>Number</u>	<u>Name/Location</u>									
192CL	ECS Access Door									
192CR	ECS Access Door									
SUBTASK 05-41-01-210-015 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the keel beam (part) Sta 540 to 727 - Section 44. (EZAP)										
SUBTASK 05-41-01-910-008 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.										
SUBTASK 05-41-01-410-013 (5) Close these access panels: <table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>192CL</td><td>ECS Access Door</td></tr><tr><td>192CR</td><td>ECS Access Door</td></tr></tbody></table>				<u>Number</u>	<u>Name/Location</u>	192CL	ECS Access Door	192CR	ECS Access Door	
<u>Number</u>	<u>Name/Location</u>									
192CL	ECS Access Door									
192CR	ECS Access Door									
— END OF TASK —										

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM (PART) STA 540 TO 727	
		D633A109-AKS 53-828-00-01	Page 2 of 5 Feb 15/2015

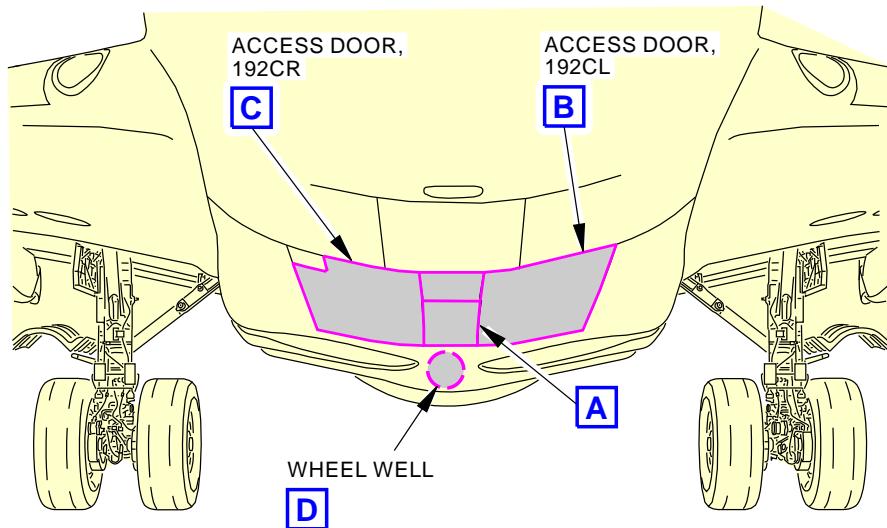
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

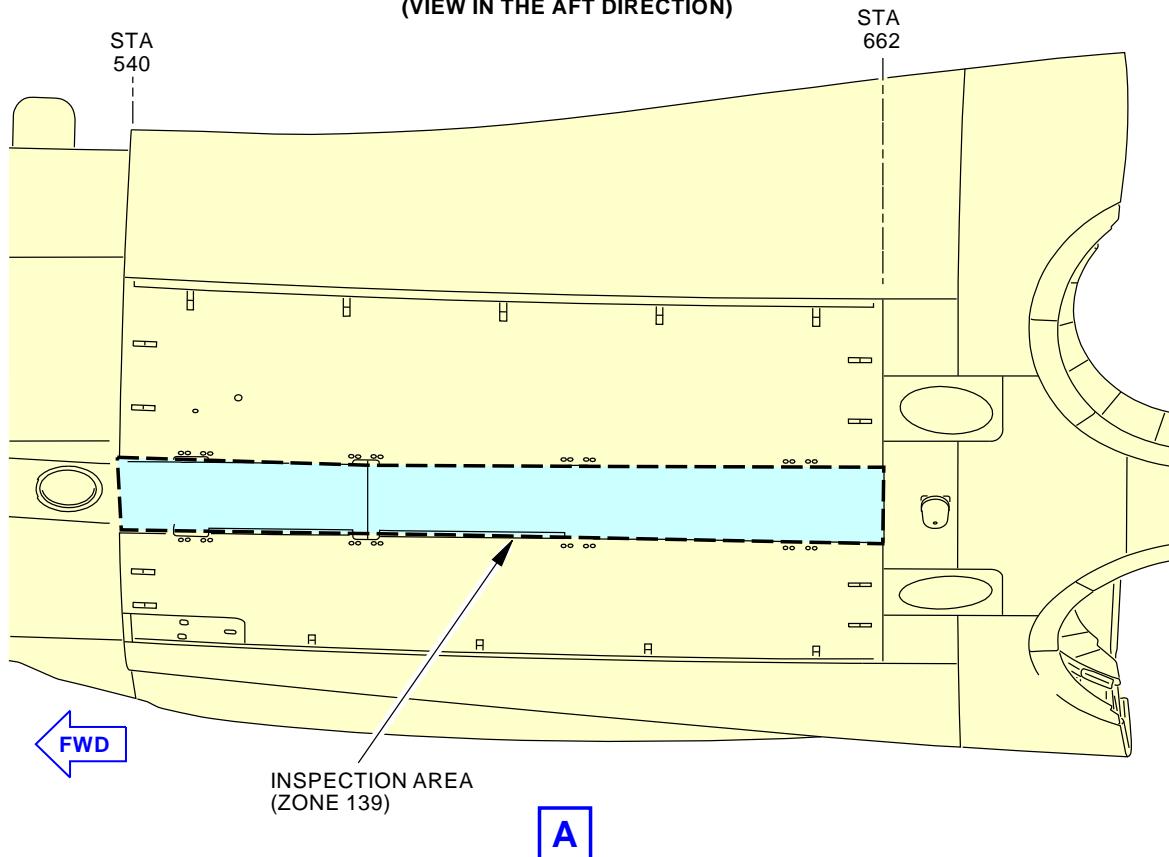
TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-828-00-01

(VIEW IN THE AFT DIRECTION)



K80115 S0006584068_V2

**Keel Beam (Sta 540-727) General Visual (Internal)
Figure 1 (Sheet 1 of 3)**EFFECTIVITY
AKS ALLSOURCE
MRB**KEEL BEAM (PART) STA 540 TO 727****D633A109-AKS
53-828-00-01****Page 3 of 5
Feb 15/2015**

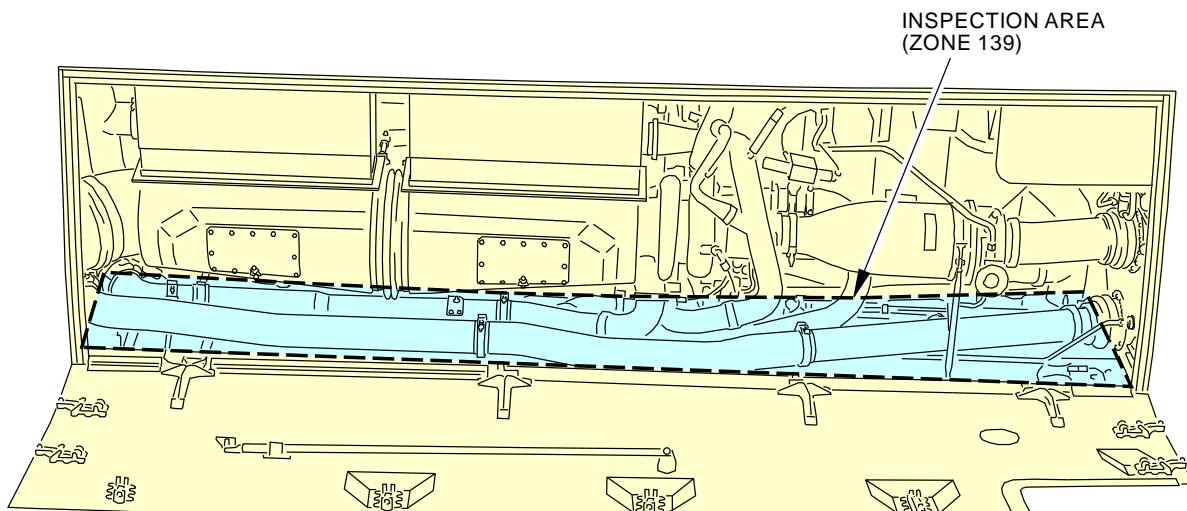
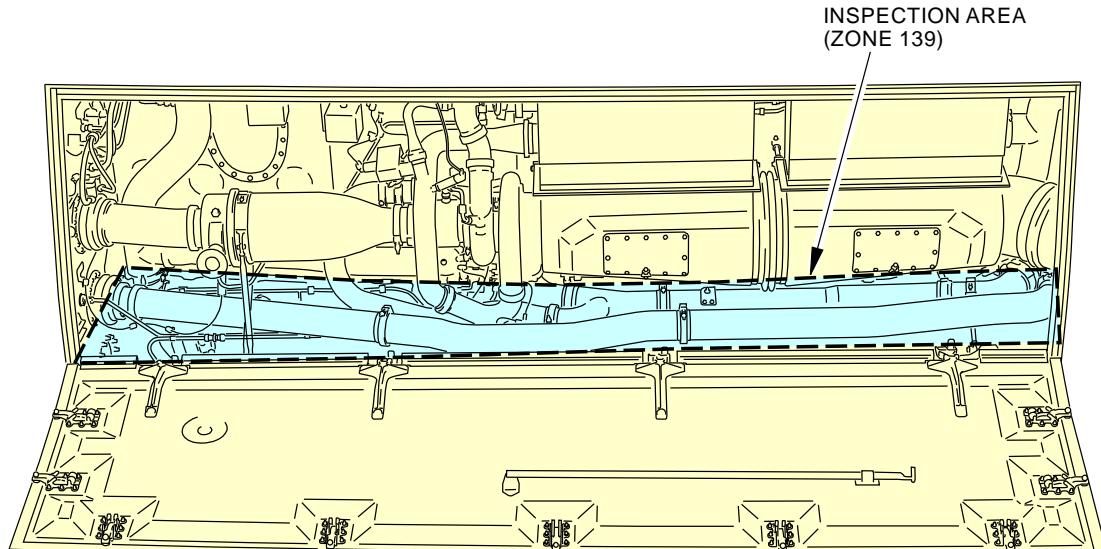
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-828-00-01

Keel Beam (Sta 540-727) General Visual (Internal)
Figure 1 (Sheet 2 of 3)

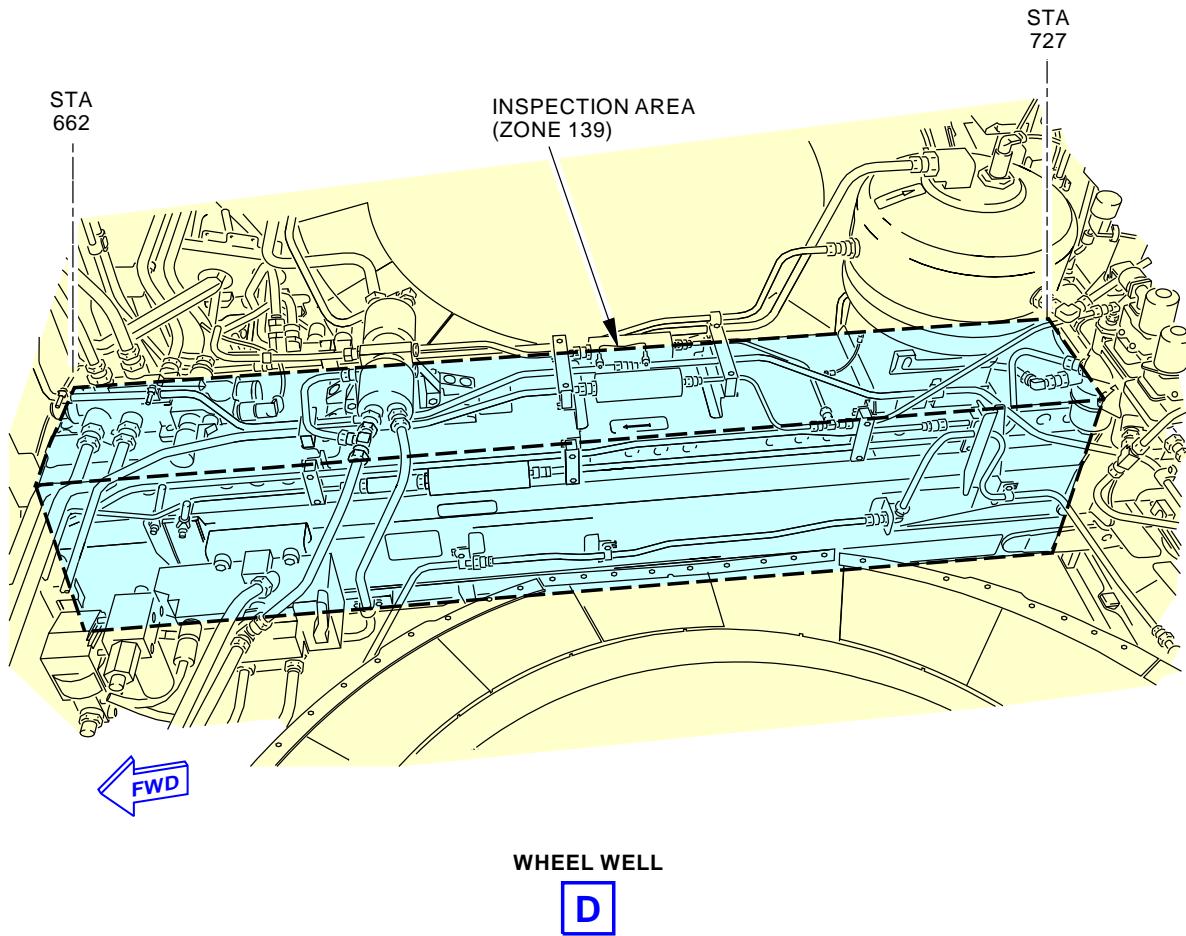
K81569 S0006584069_V2

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM (PART) STA 540 TO 727
		D633A109-AKS 53-828-00-01

Page 4 of 5
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-828-00-01



K93707 S0006584070_V2

**Keel Beam (Sta 540-727) General Visual (Internal)
Figure 1 (Sheet 3 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	KEEL BEAM (PART) STA 540 TO 727
		D633A109-AKS 53-828-00-01

**Page 5 of 5
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO COMPARTMENT			BOEING CARD NO. 53-830-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 120 DY	REPEAT 120 DY	APPLICABILITY	ENGINE
STATION	SKILL AIRPL				AIRPLANE ALL NOTE	ENGINE ALL
		ACCESS 822				ZONE 141 142
		NOTE				

Perform an internal zonal inspection (GV) of the aft cargo compartment - section 46 and 47 (part), sta 727 to sta 947.5.

AIRPLANE NOTE: For aircraft equipped with optional fuel tanks, removal of tanks not required.

ACCESS NOTE: No cargo liners removed.

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-830-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-830-00-01
				MECH INSP
TASK 05-41-01-210-817				
1. INTERNAL - ZONAL (GV): AFT CARGO COMPARTMENT				
(Figure 1)				
A. Zonal Inspection				
SUBTASK 05-41-01-010-015				
(1) Open this access panel:				
Number Name/Location				
822 Aft Cargo Door				
NOTE: No cargo liners removed.				
SUBTASK 05-41-01-210-017				
(2) Do a General Visual inspection of the aft cargo compartment - Section 46 and 47 (part), Sta 727 to Sta 947.5.				
SUBTASK 05-41-01-410-015				
(3) Close this access panel:				
Number Name/Location				
822 Aft Cargo Door				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-830-00-01

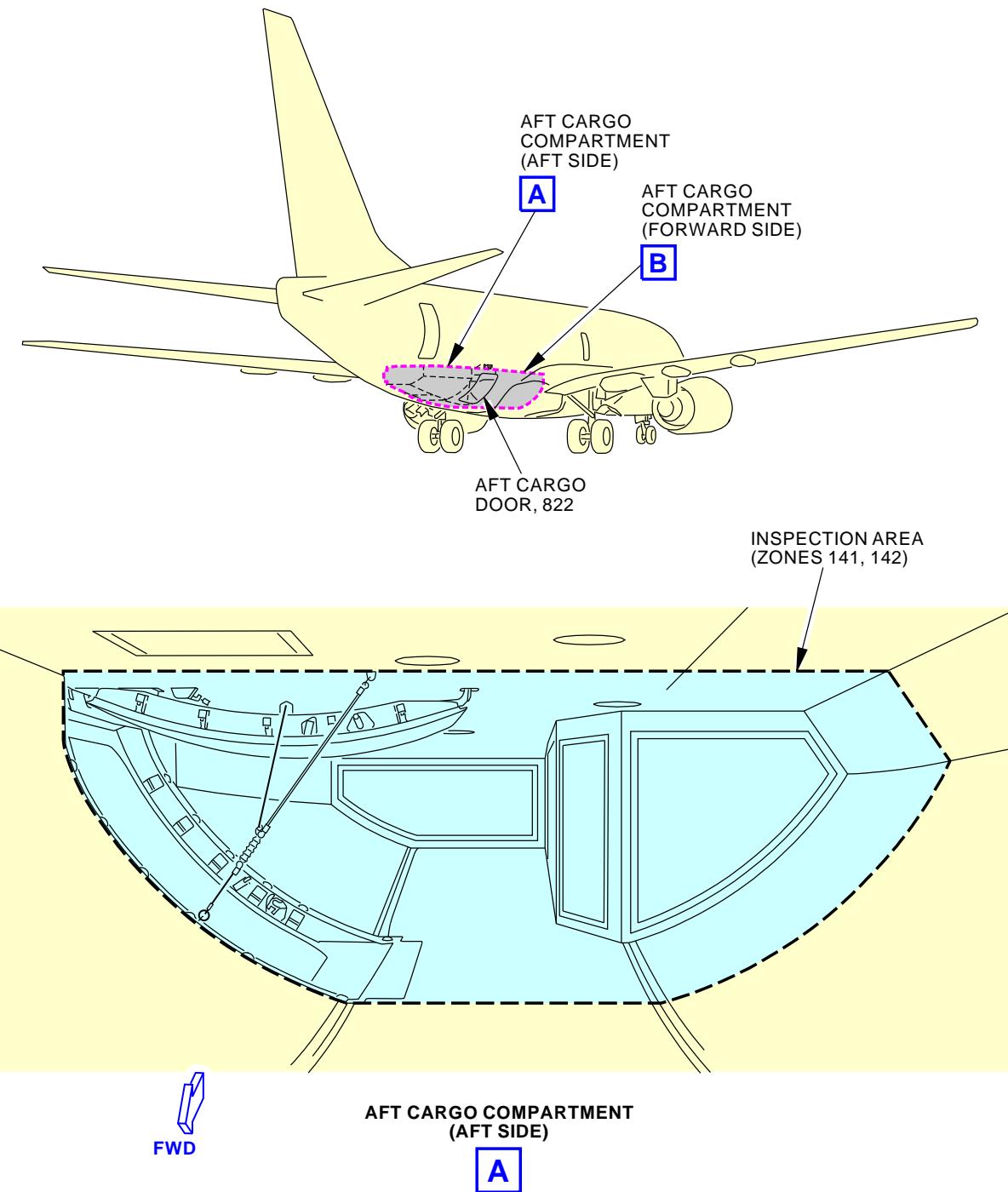
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-830-00-01

K77931 S0006584075_V2

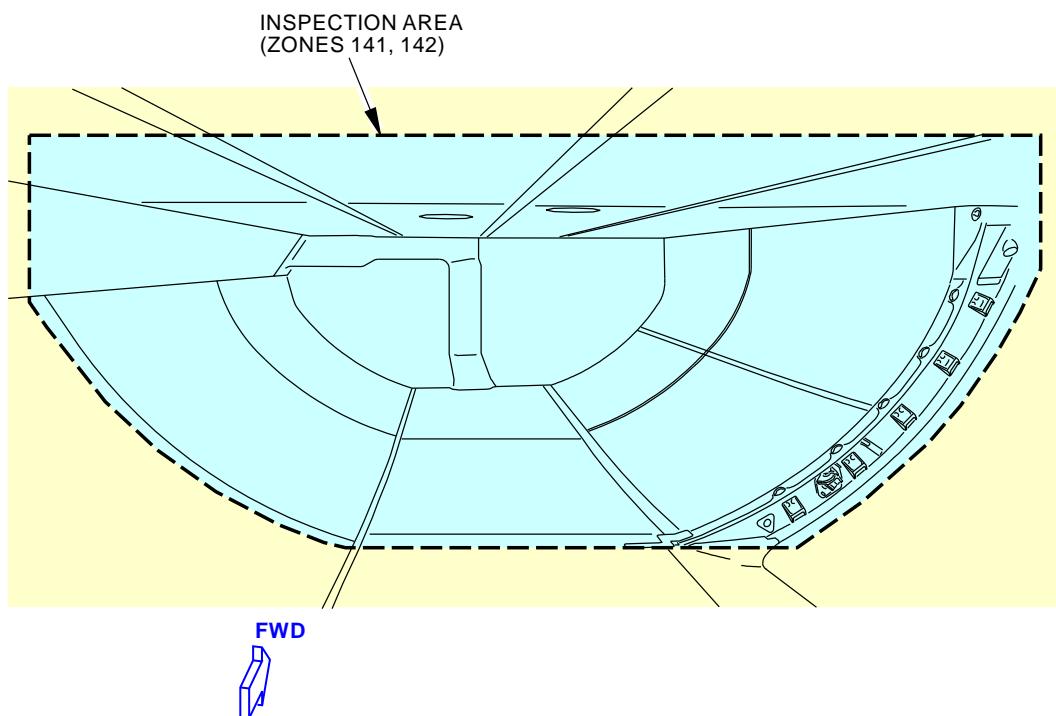
**Aft Cargo Compartment General Visual (Internal)
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-830-00-01

Page 3 of 4
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-830-00-01

**AFT CARGO COMPARTMENT
(FORWARD SIDE)****B**

K77978 S0006584076_V2

**Aft Cargo Compartment General Visual (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-830-00-01

Page 4 of 4
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO COMPARTMENT			BOEING CARD NO. 53-832-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 13200 FC 72 MO	REPEAT 13200 FC 72 MO	APPLICABILITY	ENGINE
STATION	SKILL AIRPL	ACCESS 822 NOTE				ZONE 141 142

Perform an internal zonal inspection (GV) of the aft cargo compartment - section 46 and 47 (part), sta 727 to sta 947.5.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Upper angled sidewall and ceiling panels removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-832-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-832-00-01				
				MECH INSP				
TASK 05-41-01-210-818								
1. INTERNAL - ZONAL (GV): AFT CARGO COMPARTMENT								
(Figure 1)								
A. Zonal Inspection								
SUBTASK 05-41-01-010-016								
(1) Open this access panel:								
<table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>822</td> <td>Aft Cargo Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door	
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
NOTE: Upper angled sidewall and ceiling panels removal required.								
SUBTASK 05-41-01-210-018								
(2) Do a General Visual inspection of the aft cargo compartment - Section 46 and 47 (part), Sta 727 to Sta 947.5.								
SUBTASK 05-41-01-410-016								
(3) Close this access panel:								
<table> <thead> <tr> <th><u>Number</u></th> <th><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>822</td> <td>Aft Cargo Door</td> </tr> </tbody> </table>				<u>Number</u>	<u>Name/Location</u>	822	Aft Cargo Door	
<u>Number</u>	<u>Name/Location</u>							
822	Aft Cargo Door							
———— END OF TASK ————								

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-832-00-01

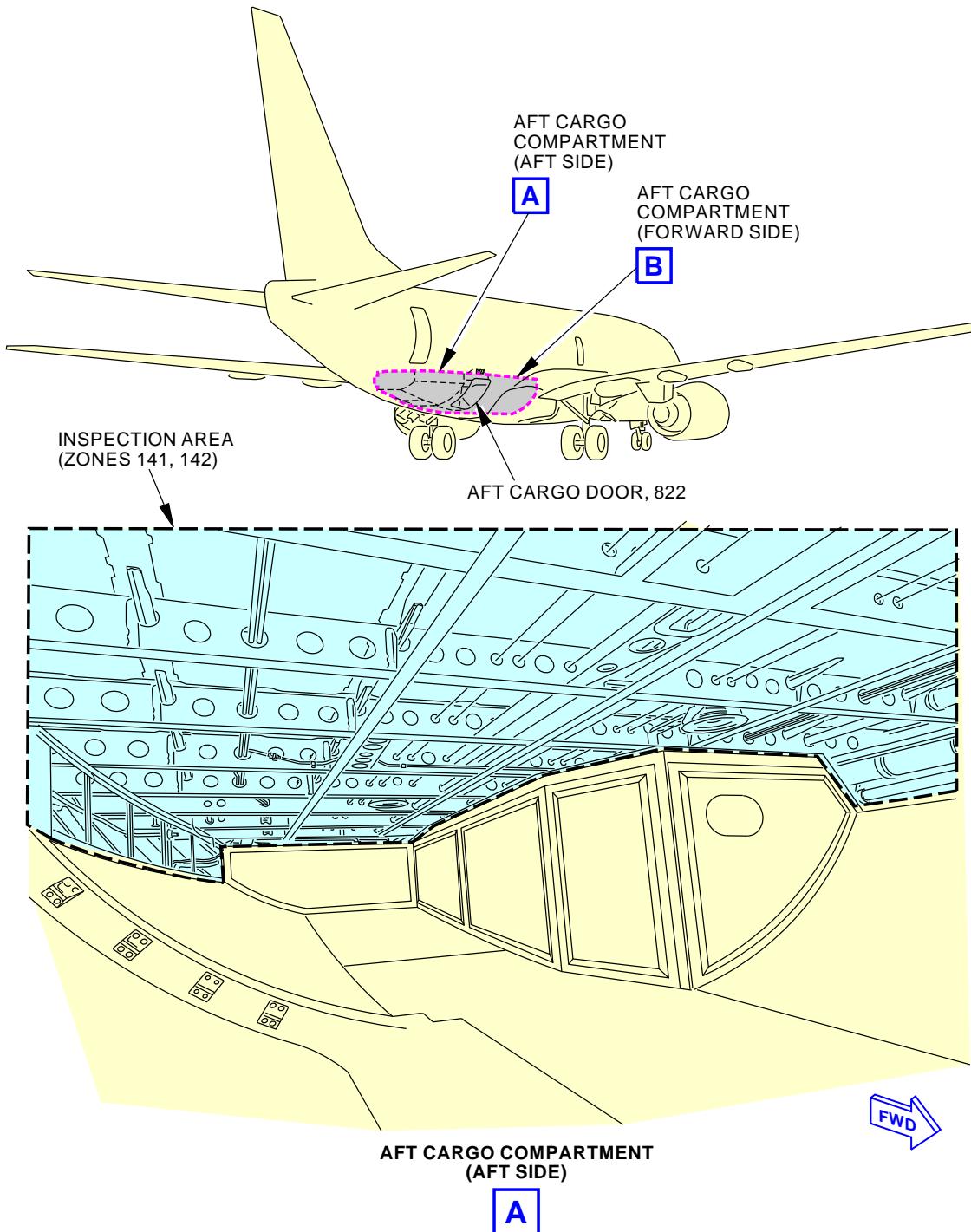
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-832-00-01

K97443 S0006584078_V2

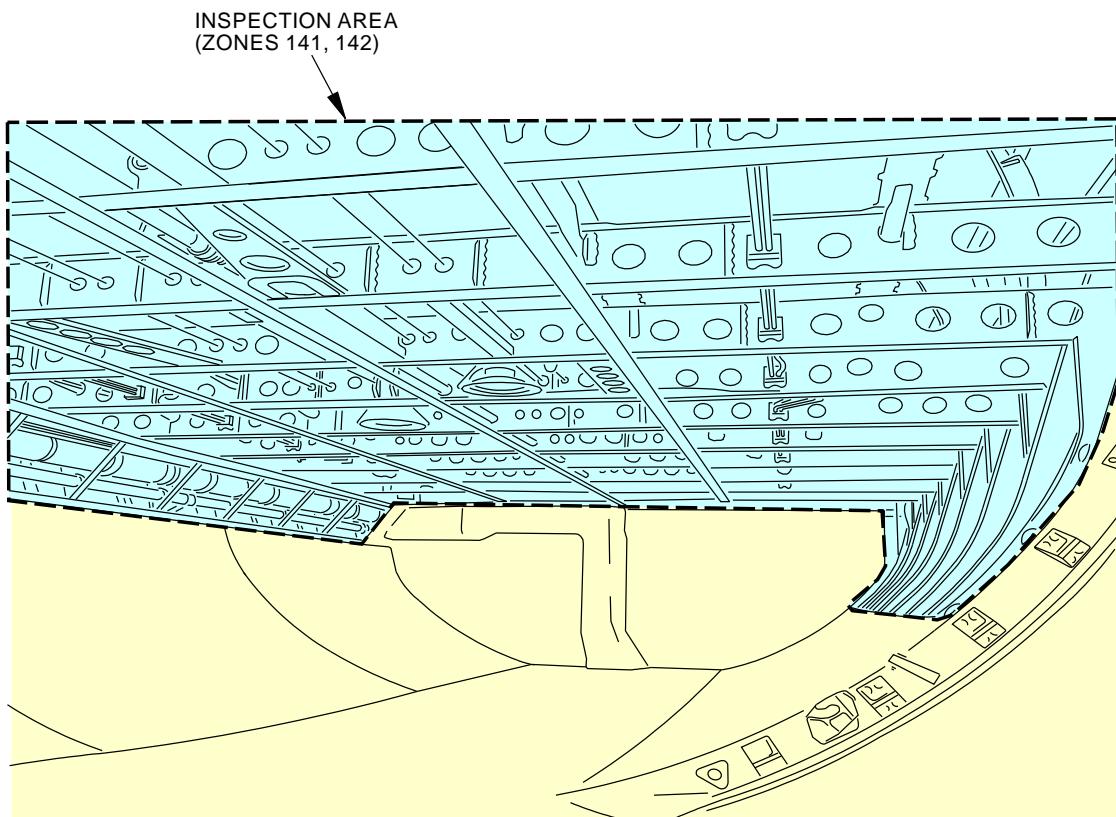
**Aft Cargo Compartment General Visual (Internal) (Upper Angled Sidewall and Ceiling Panels Removed)
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-832-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-832-00-01

**AFT CARGO COMPARTMENT
(FORWARD SIDE)**

K97453 S0006584079_V2
**Aft Cargo Compartment General Visual (Internal) (Upper Angled Sidewall and Ceiling Panels Removed)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-832-00-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT CARGO COMPARTMENT			BOEING CARD NO. 53-834-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 36000 FC	REPEAT 36000 FC	APPLICABILITY	
STATION	SKILL AIRPL	1.2	12 YR	12 YR	AIRPLANE ALL	ENGINE ALL
		ACCESS 822				ZONE 141 142
		NOTE				

Perform an internal zonal inspection (GV) of the aft cargo compartment - section 46 and 47 (part), sta 727 to sta 947.5.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Sidewall panels removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-834-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-834-00-01
				MECH INSP
TASK 05-41-01-210-819				
1. INTERNAL - ZONAL (GV): AFT CARGO COMPARTMENT				
(Figure 1)				
A. Zonal Inspection				
SUBTASK 05-41-01-010-017				
(1) Open this access panel:				
Number Name/Location				
822 Aft Cargo Door				
NOTE: Sidewall panels removal required.				
SUBTASK 05-41-01-210-019				
(2) Do a General Visual inspection of the aft cargo compartment - Section 46 and 47 (part), Sta 727 to Sta 947.5.				
SUBTASK 05-41-01-410-017				
(3) Close this access panel:				
Number Name/Location				
822 Aft Cargo Door				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-834-00-01

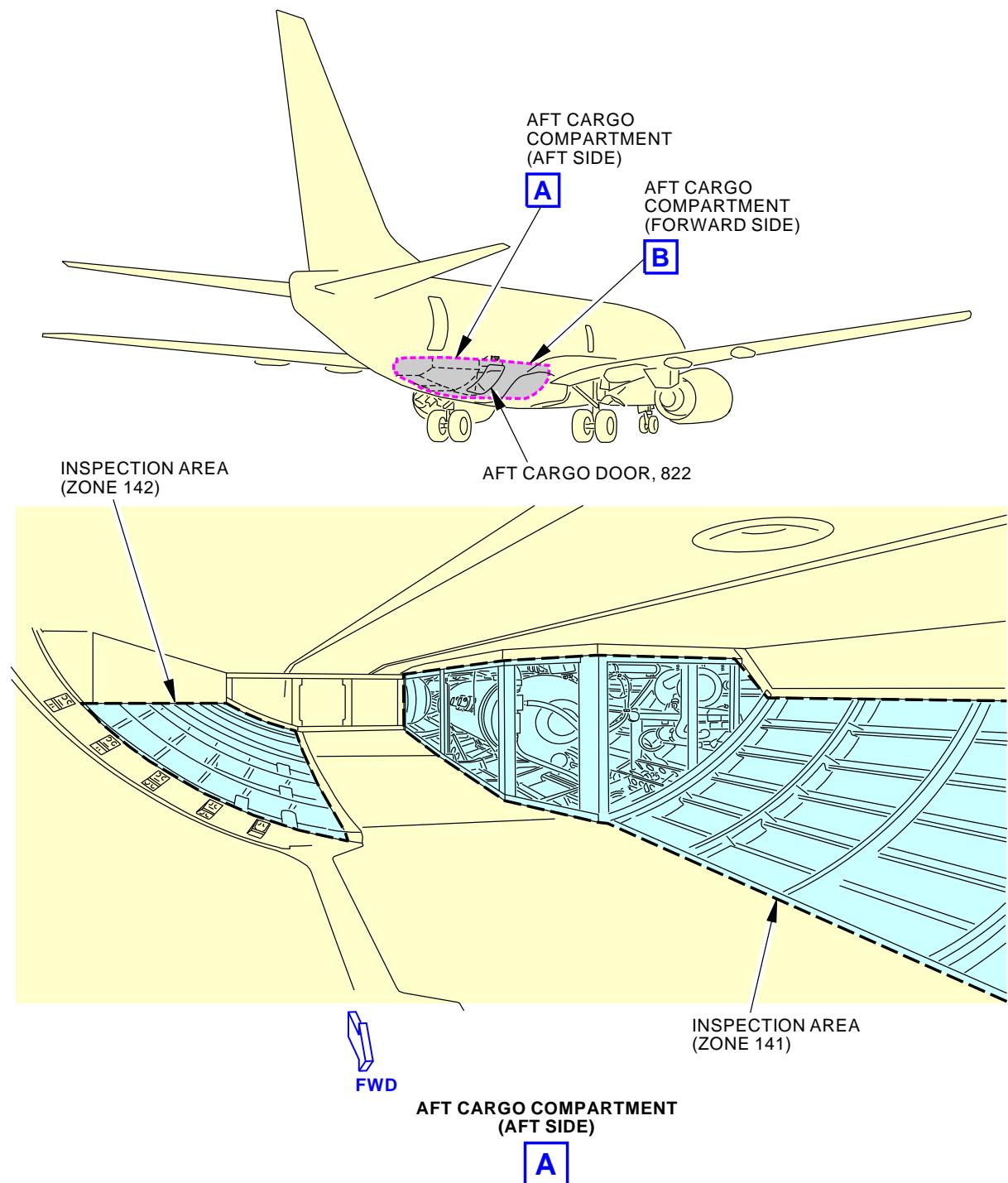
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-834-00-01

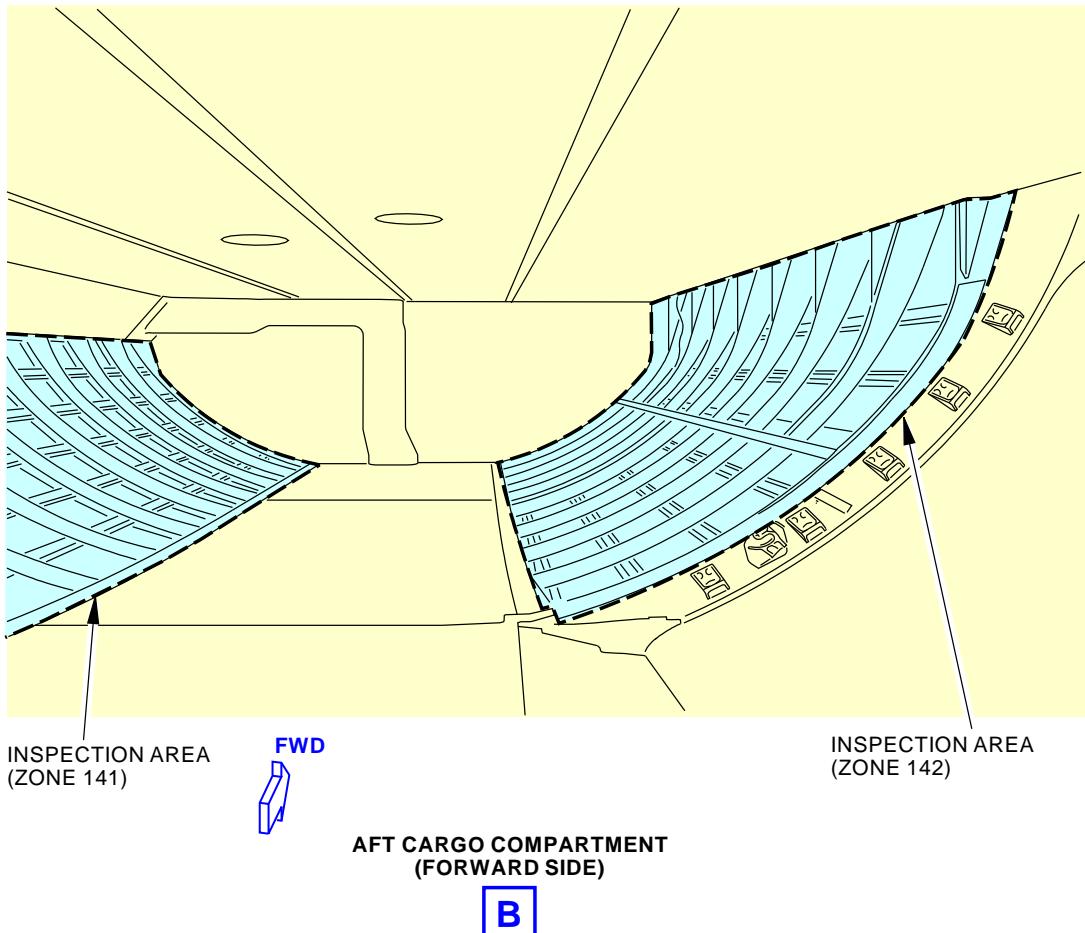
K78591 S0006584081_V3

**Aft Cargo Compartment General Visual (Internal) (Sidewall Liners Removed)
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-834-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-834-00-01



K80000 S0006584082_V3
Aft Cargo Compartment General Visual (Internal) (Sidewall Liners Removed)
Figure 1 (Sheet 2 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT
		D633A109-AKS 53-834-00-01

Page 4 of 4
Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO DOOR SURROUND STRUCTURE FITTINGS AND STOPS			BOEING CARD NO. 53-836-00-01		
DATE	TASK ZONAL (GV)				RELATED CARD		
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 2000 FC 240 DY	REPEAT 2000 FC 240 DY	APPLICABILITY	AIRPLANE ALL	ENGINE ALL
		ACCESS 822			ZONE 142		

Perform an external zonal inspection (GV) of the aft cargo door surround structure fittings and stops - section 46, sta 827.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO DOOR SURROUND STRUCTURE FITTINGS AND STOPS
		D633A109-AKS 53-836-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-836-00-01
				MECH INSP
TASK 05-41-01-210-820				
1. EXTERNAL - ZONAL (GV): AFT CARGO DOOR SURROUND STRUCTURE FITTINGS AND STOPS				
(Figure 1)				
A. Zonal Inspection				
SUBTASK 05-41-01-010-018				
(1) Open this access panel:				
Number Name/Location				
822 Aft Cargo Door				
SUBTASK 05-41-01-210-020				
(2) Do a General Visual inspection of the aft cargo door surround structure fittings and stops - Section 46, Sta 827.				
SUBTASK 05-41-01-410-018				
(3) Close this access panel:				
Number Name/Location				
822 Aft Cargo Door				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO DOOR SURROUND STRUCTURE FITTINGS AND STOPS		
		D633A109-AKS 53-836-00-01		
				Page 2 of 3 Feb 15/2015

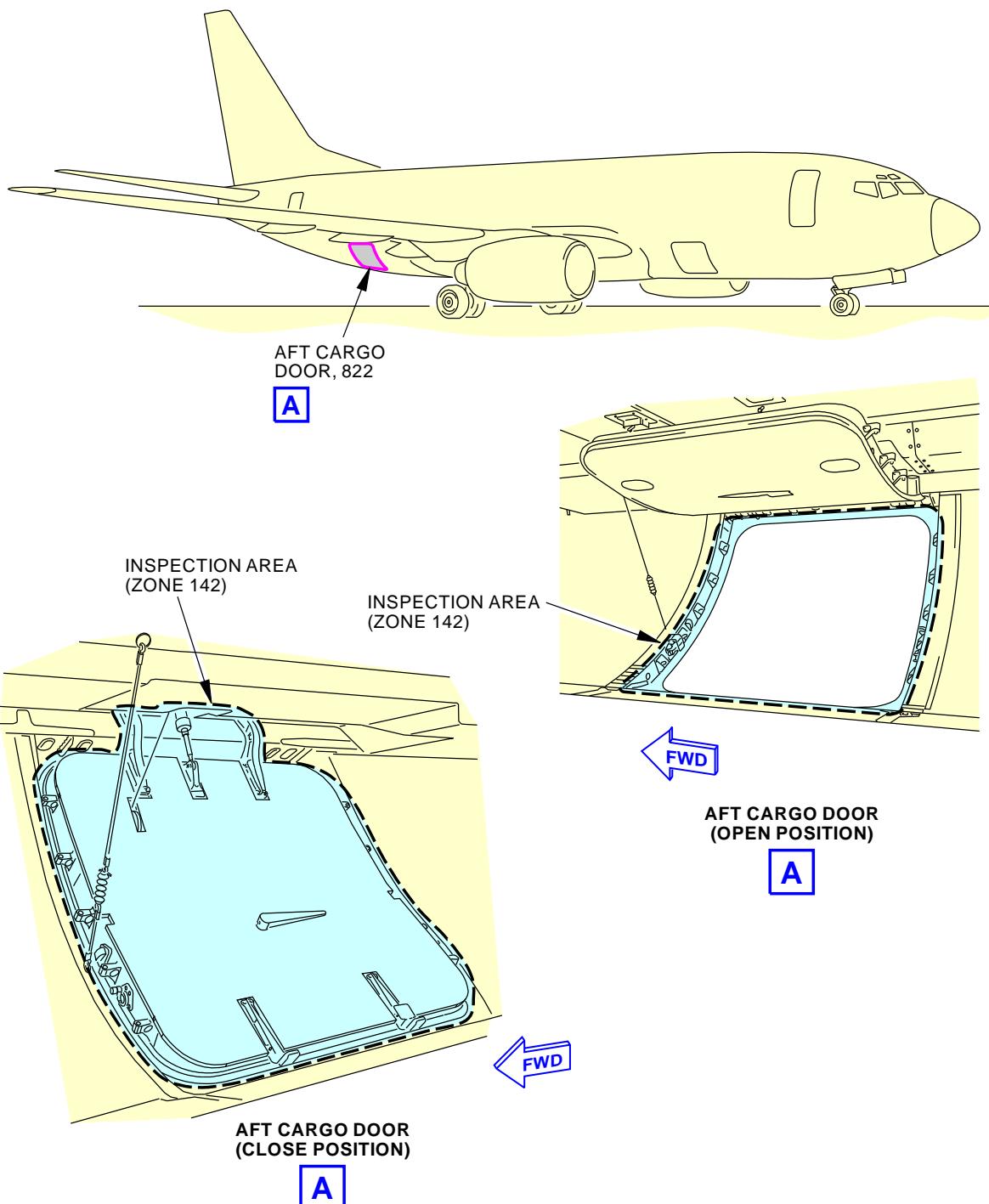
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-836-00-01

K76836 S0006584084_V2

**Fittings and Stops - Aft Cargo Door Surround Structure General Visual (External)
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**AFT CARGO DOOR SURROUND STRUCTURE FITTINGS AND
STOPS****D633A109-AKS
53-836-00-01****Page 3 of 3
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO COMPARTMENT VACUUM WASTE COMPARTMENT			BOEING CARD NO. 53-838-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 822 NOTE				ENGINE ALL
					ZONE 141	

Perform an internal zonal inspection (GV) of the aft cargo compartment vacuum waste compartment.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Vacuum waste compartment panels removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT VACUUM WASTE COMPARTMENT
		D633A109-AKS 53-838-00-01

AKS



737-600/700/800/900 TASK CARDS

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT VACUUM WASTE COMPARTMENT	D633A109-AKS 53-838-00-01	Page 2 of 3 Feb 15/2015
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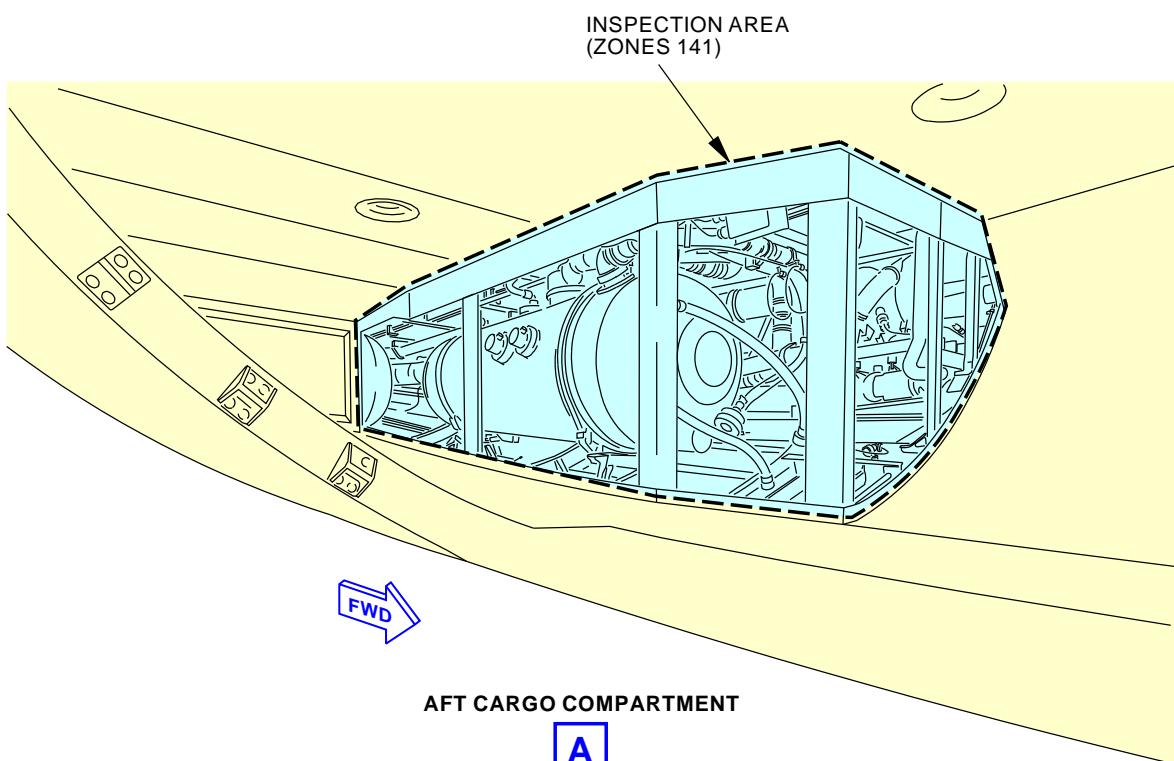
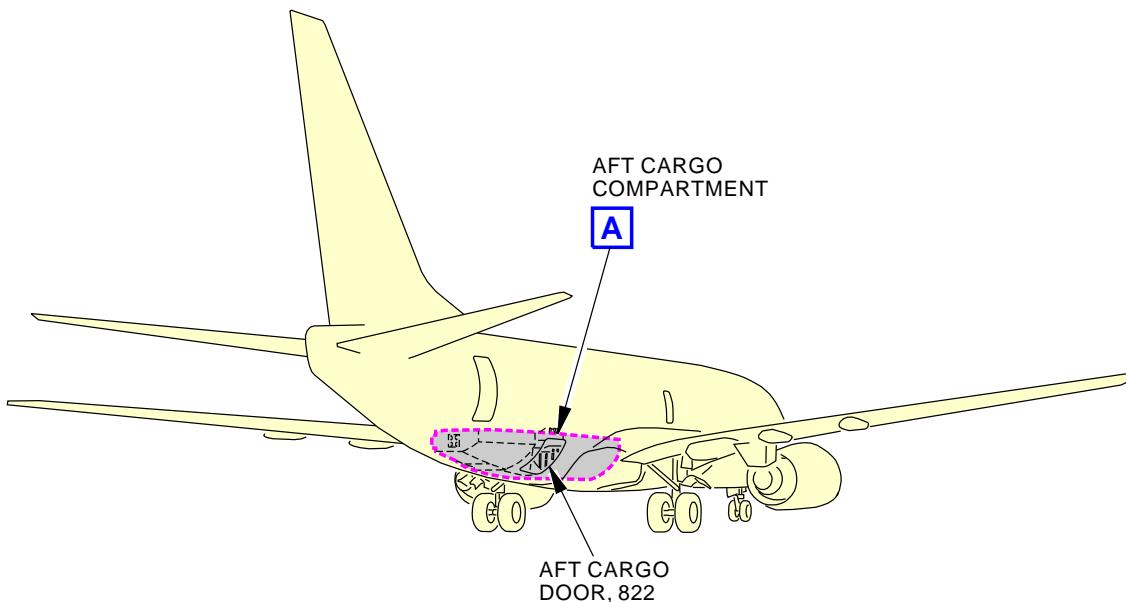
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-838-00-01

K97138 S0006584086_V2

**Aft Cargo Compartment - Vacuum Waste Compartment General Visual (Internal)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT VACUUM WASTE COMPARTMENT
		D633A109-AKS 53-838-00-01

Page 3 of 3
Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA BELOW AFT CARGO COMPARTMENT			BOEING CARD NO. 53-840-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 13200 FC 72 MO	REPEAT 13200 FC 72 MO	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 822 NOTE			ZONE	ENGINE ALL
					143 144	

Perform an internal zonal inspection (GV) of the area below the aft cargo compartment - Section 46 and 47 (part), Sta 727 to Sta 947.5. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

ACCESS NOTE: Center floor panels removal required. Cargo loading system removed/displaced as required.

EFFECTIVITY AKS ALL	SOURCE MRB	AREA BELOW AFT CARGO COMPARTMENT
		D633A109-AKS 53-840-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-840-00-01
				MECH INSP
► EWIS TASK 05-41-01-210-822				
1. INTERNAL - ZONAL (GV): Area Below Aft Cargo Compartment (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection SUBTASK 05-41-01-010-020 (1) Open this access panel: Number Name/Location 822 Aft Cargo Door <u>NOTE:</u> Center floor panels removal required. Cargo loading system removed/displaced as required.				
SUBTASK 05-41-01-210-022 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the area below the aft cargo compartment - Section 46 and 47 (part), Sta 727 to Sta 947.5. (EZAP)				
SUBTASK 05-41-01-910-009 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-01-410-020 (5) Close this access panel: Number Name/Location 822 Aft Cargo Door				
———— END OF TASK ——				

EFFECTIVITY AKS ALL	SOURCE MRB	AREA BELOW AFT CARGO COMPARTMENT
		D633A109-AKS 53-840-00-01

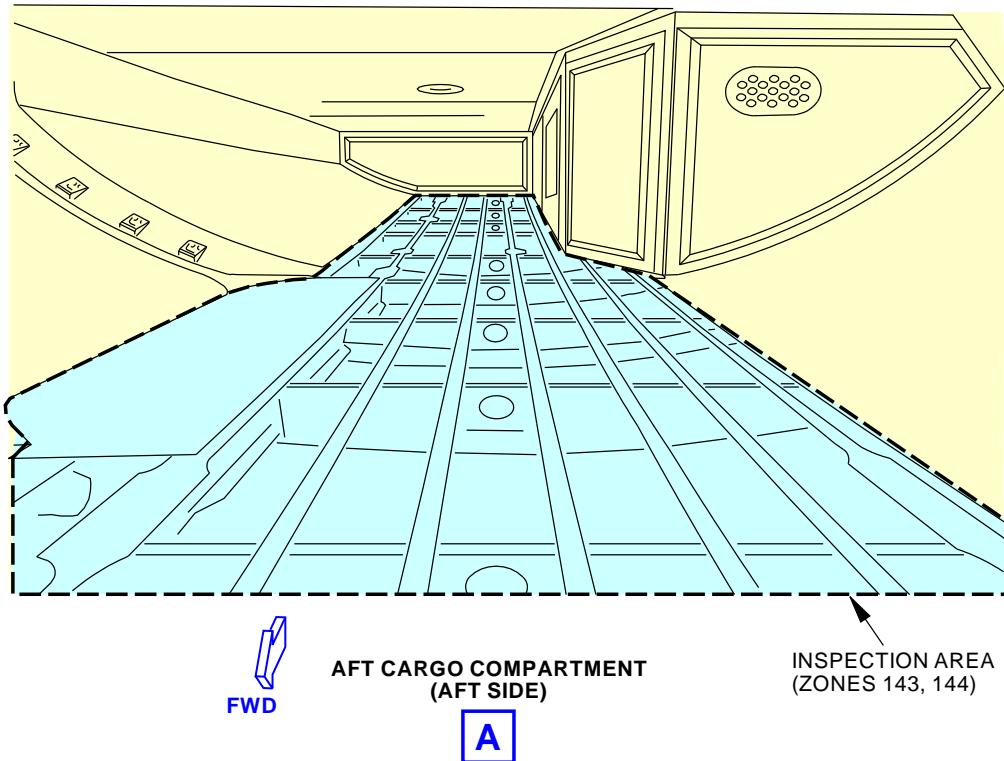
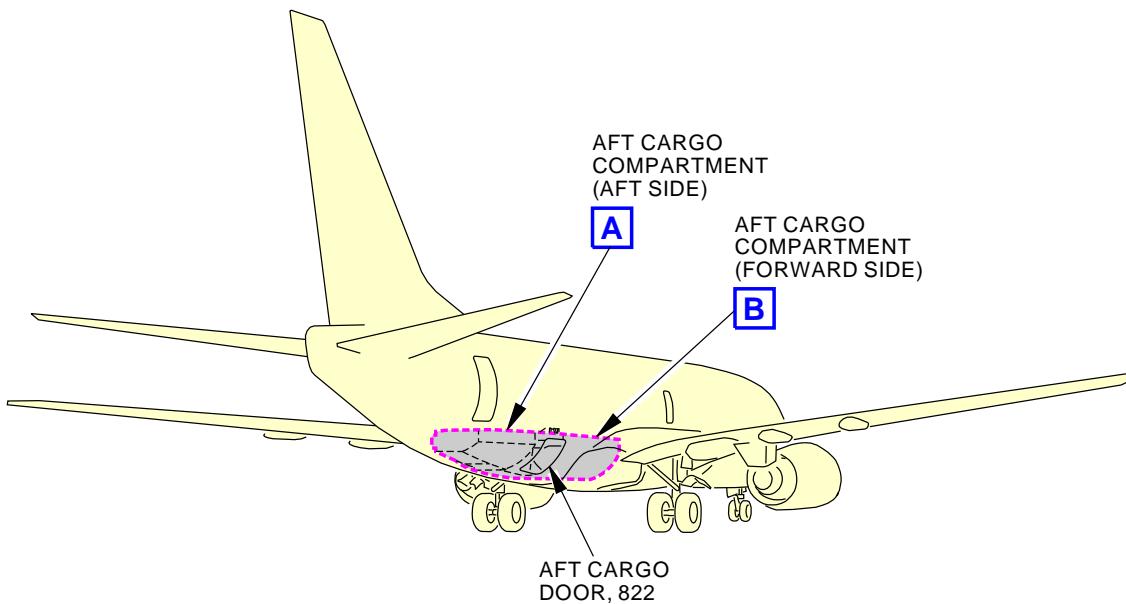
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-840-00-01

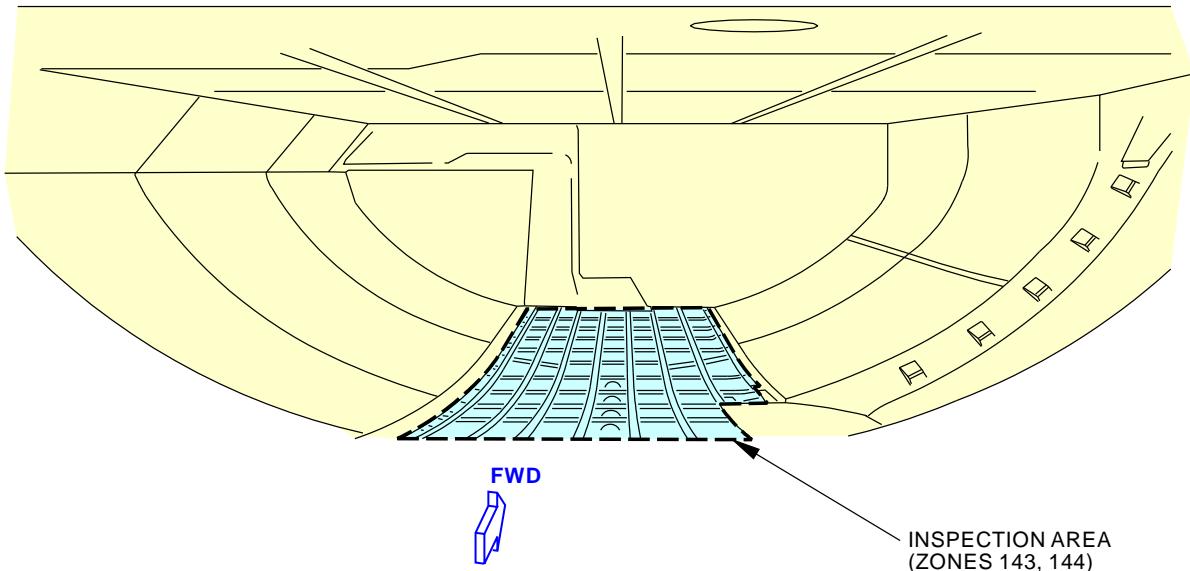
K97728 S0006584088_V2

Below the Aft Cargo Compartment General Visual (Internal) (Floor Panels Removed)
Figure 1 (Sheet 1 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	AREA BELOW AFT CARGO COMPARTMENT
		D633A109-AKS 53-840-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-840-00-01

**AFT CARGO COMPARTMENT
(FORWARD SIDE)**

K97467 S0006584089_V2

**Below the Aft Cargo Compartment General Visual (Internal) (Floor Panels Removed)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	AREA BELOW AFT CARGO COMPARTMENT
		D633A109-AKS 53-840-00-01

**Page 4 of 4
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT CARGO COMPARTMENT EQUIPMENT BAY			BOEING CARD NO. 53-842-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 822 NOTE				ENGINE ALL
					ZONE 145 146	

Perform an internal zonal inspection (GV) of the aft cargo compartment equipment bay - Section 47, Sta 947.5. to aft pressure bulkhead. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

ACCESS NOTE: Aft cargo panels at Sta 947 bulkhead removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT EQUIPMENT BAY
		D633A109-AKS 53-842-00-01

AKS

737-600/700/800/900

TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-842-00-01
				MECH INSP
EWIS				
TASK 05-41-01-210-823				
1. INTERNAL - ZONAL (GV): Aft Cargo Compartment Equipment Bay (Figure 1)				
A. General				
(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection				
SUBTASK 05-41-01-010-021				
(1) Open this access panel:				
Number Name/Location				
822 Aft Cargo Door				
NOTE: Aft cargo panels at Sta 947 bulkhead removal required.				
SUBTASK 05-41-01-210-023				
(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.				
(3) Perform an internal zonal inspection (GV) of the aft cargo compartment equipment bay - Section 47, Sta 947.5. to aft pressure bulkhead. (EZAP)				
SUBTASK 05-41-01-910-010				
(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-01-410-021				
(5) Close this access panel:				
Number Name/Location				
822 Aft Cargo Door				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	AFT CARGO COMPARTMENT EQUIPMENT BAY
		D633A109-AKS 53-842-00-01

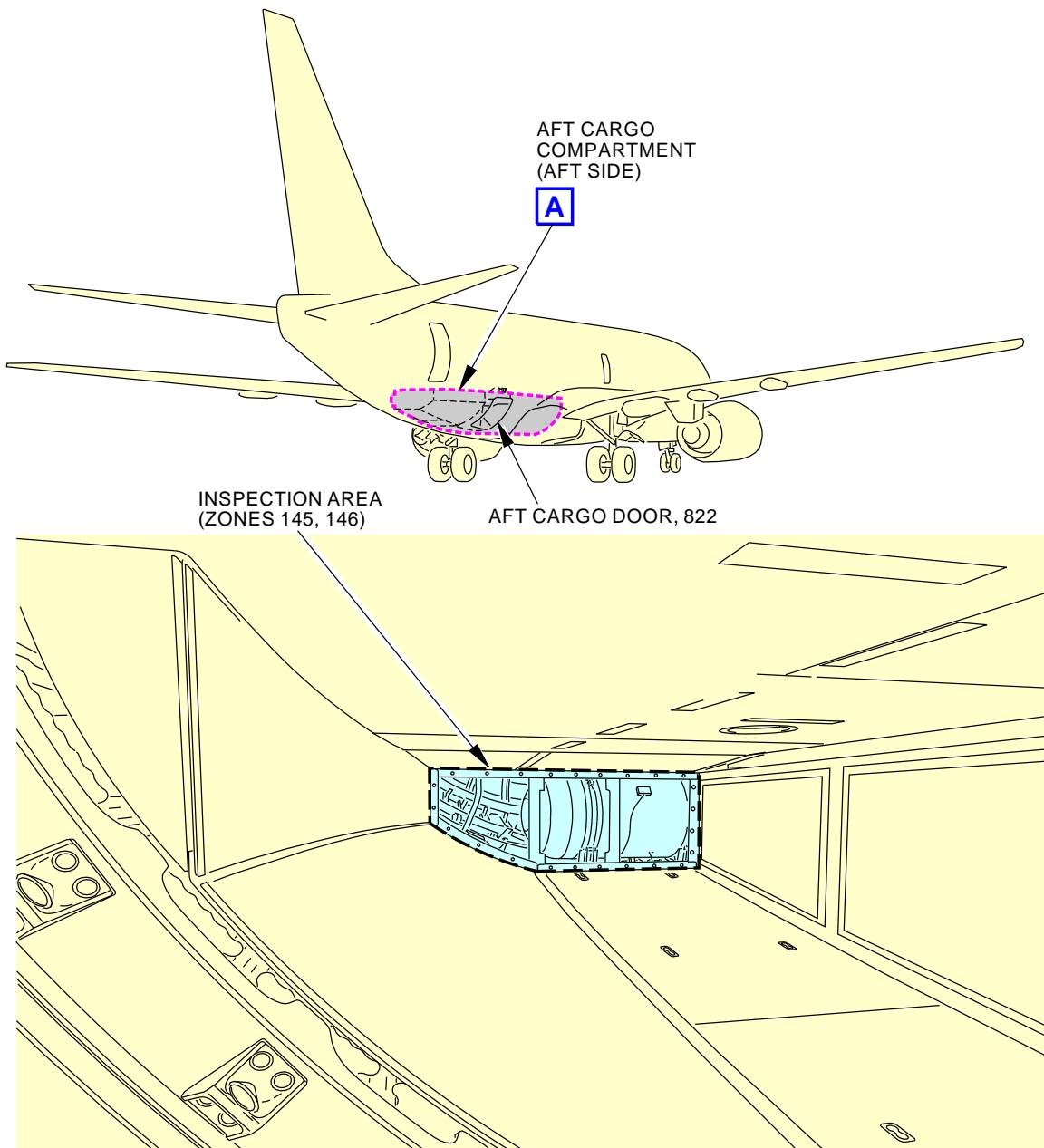
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-842-00-01

K97844 S0006584091_V2

**Aft Cargo Compartment - Equipment Bay General Visual (Internal) (Endwall Liners Removed)
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**AFT CARGO COMPARTMENT EQUIPMENT BAY****D633A109-AKS
53-842-00-01****Page 3 of 3
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE WING TO BODY FAIRING			BOEING CARD NO. 53-844-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 120 DY	REPEAT 120 DY	APPLICABILITY AIRPLANE ENGINE ALL ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 190

Perform an external zonal inspection (GV) of the wing to body fairing. Inspection is accomplished from the ground, without the use of stands or ladders. No additional access panels required.

EFFECTIVITY AKS ALL	SOURCE MRB	WING TO BODY FAIRING
		D633A109-AKS 53-844-00-01

Page 1 of 4
Jun 15/2016

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-844-00-01	MECH	INSP
TASK 05-41-01-210-824						
1. EXTERNAL - ZONAL (GV): WING TO BODY FAIRING						
(Figure 1)						
A. Zonal Inspection						
SUBTASK 05-41-01-210-024						
(1) Do a General Visual inspection of the wing to body fairing. Inspection is accomplished from the ground, without the use of stands or ladders. No additional access panels required.						
———— END OF TASK ——						

EFFECTIVITY AKS ALL	SOURCE MRB	WING TO BODY FAIRING
		D633A109-AKS 53-844-00-01

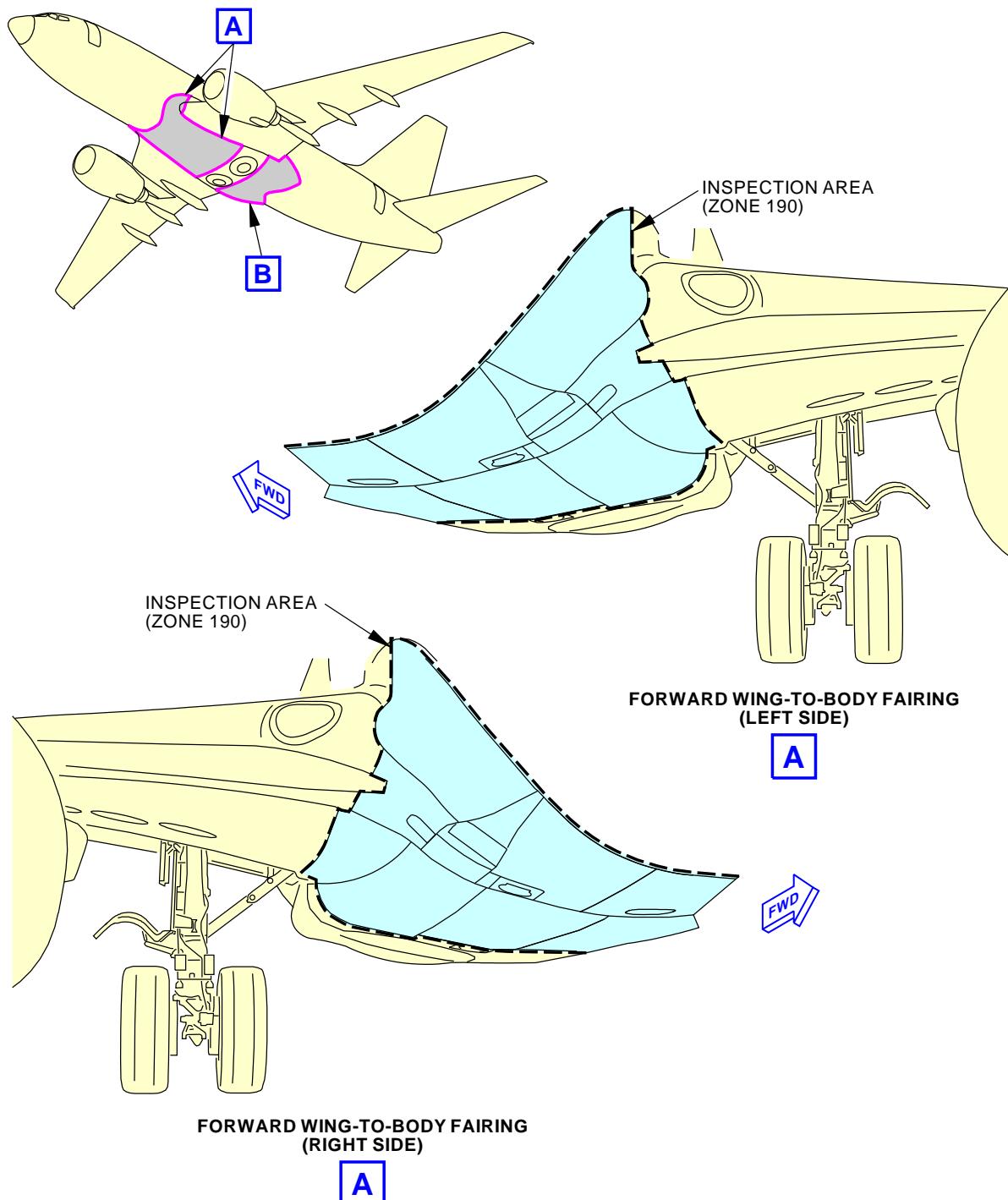
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-844-00-01

K68795 S0006584093_V2

**Wing-to-Body Fairing General Visual (External)
Figure 1 (Sheet 1 of 2)**EFFECTIVITY
AKS ALLSOURCE
MRB**WING TO BODY FAIRING****D633A109-AKS
53-844-00-01****Page 3 of 4
Feb 15/2015**

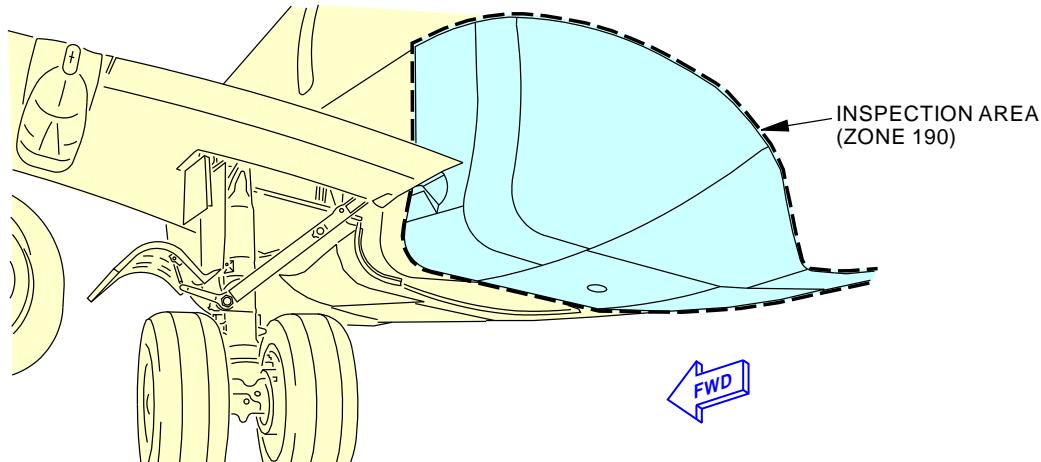
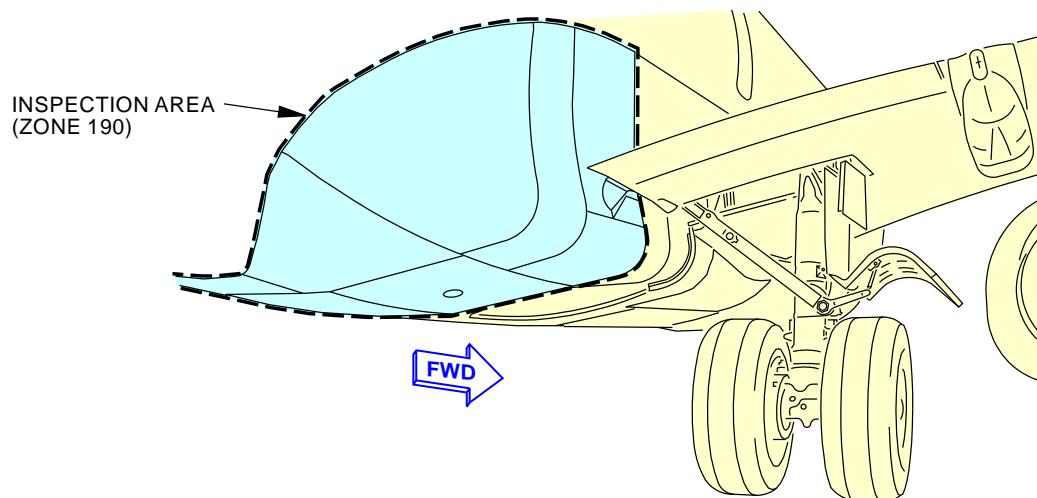
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-844-00-01**AFT WING-TO-BODY FAIRING
(LEFT SIDE)****B****AFT WING-TO-BODY FAIRING
(RIGHT SIDE)****B**

K68797 S0006584094_V2

**Wing-to-Body Fairing General Visual (External)
Figure 1 (Sheet 2 of 2)**EFFECTIVITY
AKS ALLSOURCE
MRB**WING TO BODY FAIRING****D633A109-AKS
53-844-00-01****Page 4 of 4
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LOWER WING TO BODY FAIRING - FORWARD OF WING BOX			BOEING CARD NO. 53-846-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLICABILITY AIRPLANE ALL	
STATION	SKILL AIRPL		36 MO	36 MO		
		ACCESS 191GL 191GR 192CL 192CR				ZONE 191
		NOTE				

Perform an internal zonal inspection (GV) of the lower wing to body fairing - forward of wing box. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: Through access provided.

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - FORWARD OF WING BOX
		D633A109-AKS 53-846-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-846-00-01																				
 EWIS																								
TASK 05-41-01-210-825																								
<p>1. INTERNAL - ZONAL (GV): Lower Wing to Body Fairing - Forward of Wing Box</p> <p>(Figure 1)</p> <p>A. General</p> <p class="list-item-l1">(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.</p> <p>B. Zonal Inspection</p> <p>SUBTASK 05-41-01-010-022</p> <p class="list-item-l1">(1) Open these access panels:</p> <table border="0" style="width: 100%;"> <thead> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>191GL</td> <td>Ram Air Actuator Panel - Forward</td> </tr> <tr> <td>191GR</td> <td>Ram Air Actuator Panel - Forward</td> </tr> <tr> <td>192CL</td> <td>ECS Access Door</td> </tr> <tr> <td>192CR</td> <td>ECS Access Door</td> </tr> </tbody> </table> <p><u>NOTE:</u> Through access provided.</p> <p>SUBTASK 05-41-01-210-025</p> <p class="list-item-l1">(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.</p> <p class="list-item-l1">(3) Perform an internal zonal inspection (GV) of the lower wing to body fairing - forward of wing box. (EZAP)</p> <p>SUBTASK 05-41-01-910-017</p> <p class="list-item-l1">(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.</p> <p>SUBTASK 05-41-01-410-022</p> <p class="list-item-l1">(5) Close these access panels:</p> <table border="0" style="width: 100%;"> <thead> <tr> <th style="text-align: left;"><u>Number</u></th> <th style="text-align: left;"><u>Name/Location</u></th> </tr> </thead> <tbody> <tr> <td>191GL</td> <td>Ram Air Actuator Panel - Forward</td> </tr> <tr> <td>191GR</td> <td>Ram Air Actuator Panel - Forward</td> </tr> <tr> <td>192CL</td> <td>ECS Access Door</td> </tr> <tr> <td>192CR</td> <td>ECS Access Door</td> </tr> </tbody> </table> <p style="text-align: center; margin-top: 10px;">— END OF TASK —</p>					<u>Number</u>	<u>Name/Location</u>	191GL	Ram Air Actuator Panel - Forward	191GR	Ram Air Actuator Panel - Forward	192CL	ECS Access Door	192CR	ECS Access Door	<u>Number</u>	<u>Name/Location</u>	191GL	Ram Air Actuator Panel - Forward	191GR	Ram Air Actuator Panel - Forward	192CL	ECS Access Door	192CR	ECS Access Door
<u>Number</u>	<u>Name/Location</u>																							
191GL	Ram Air Actuator Panel - Forward																							
191GR	Ram Air Actuator Panel - Forward																							
192CL	ECS Access Door																							
192CR	ECS Access Door																							
<u>Number</u>	<u>Name/Location</u>																							
191GL	Ram Air Actuator Panel - Forward																							
191GR	Ram Air Actuator Panel - Forward																							
192CL	ECS Access Door																							
192CR	ECS Access Door																							
EFFECTIVITY AKS ALL		SOURCE MRB	LOWER WING TO BODY FAIRING - FORWARD OF WING BOX																					
			D633A109-AKS 53-846-00-01																					
Page 2 of 4 Feb 15/2015																								

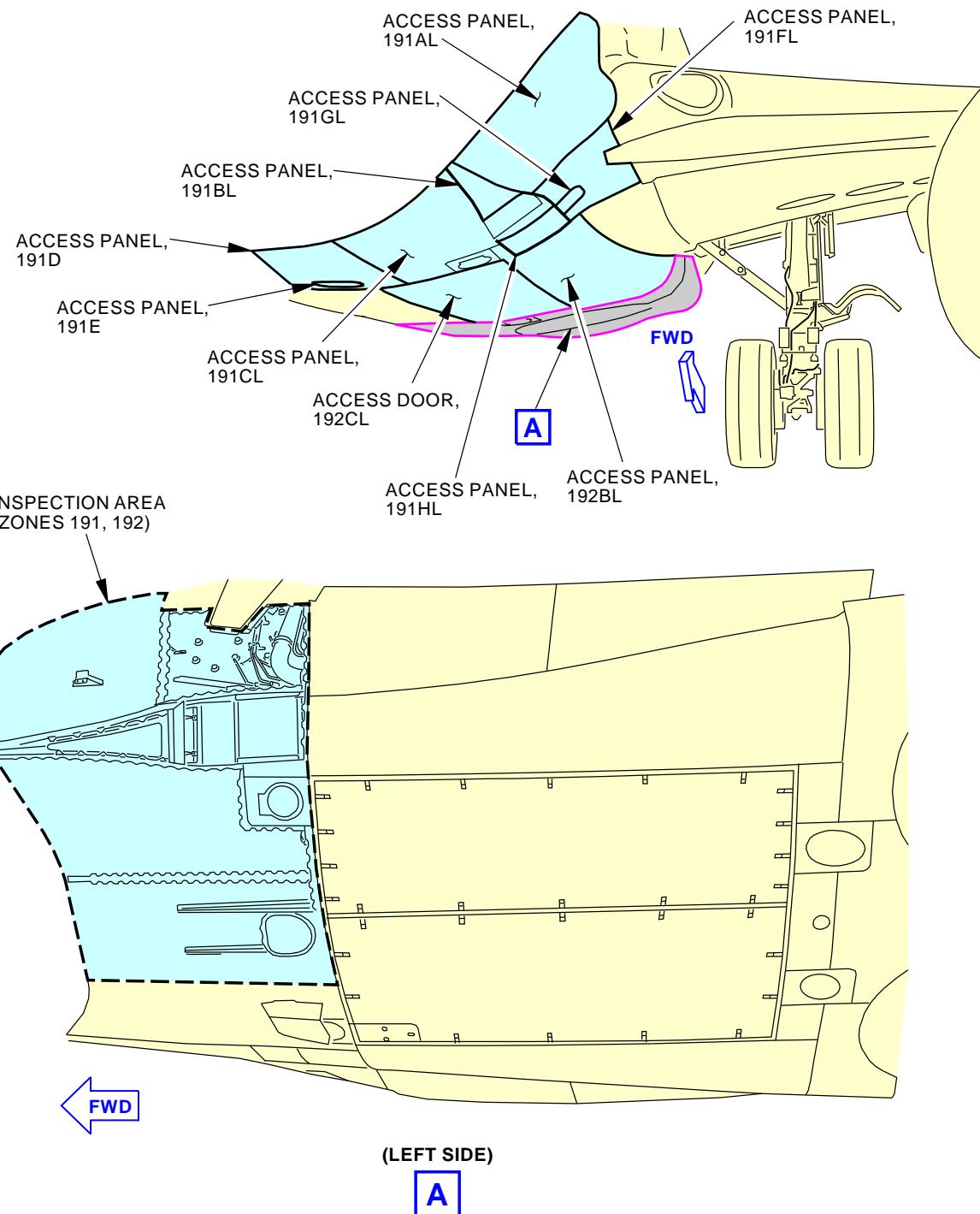
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

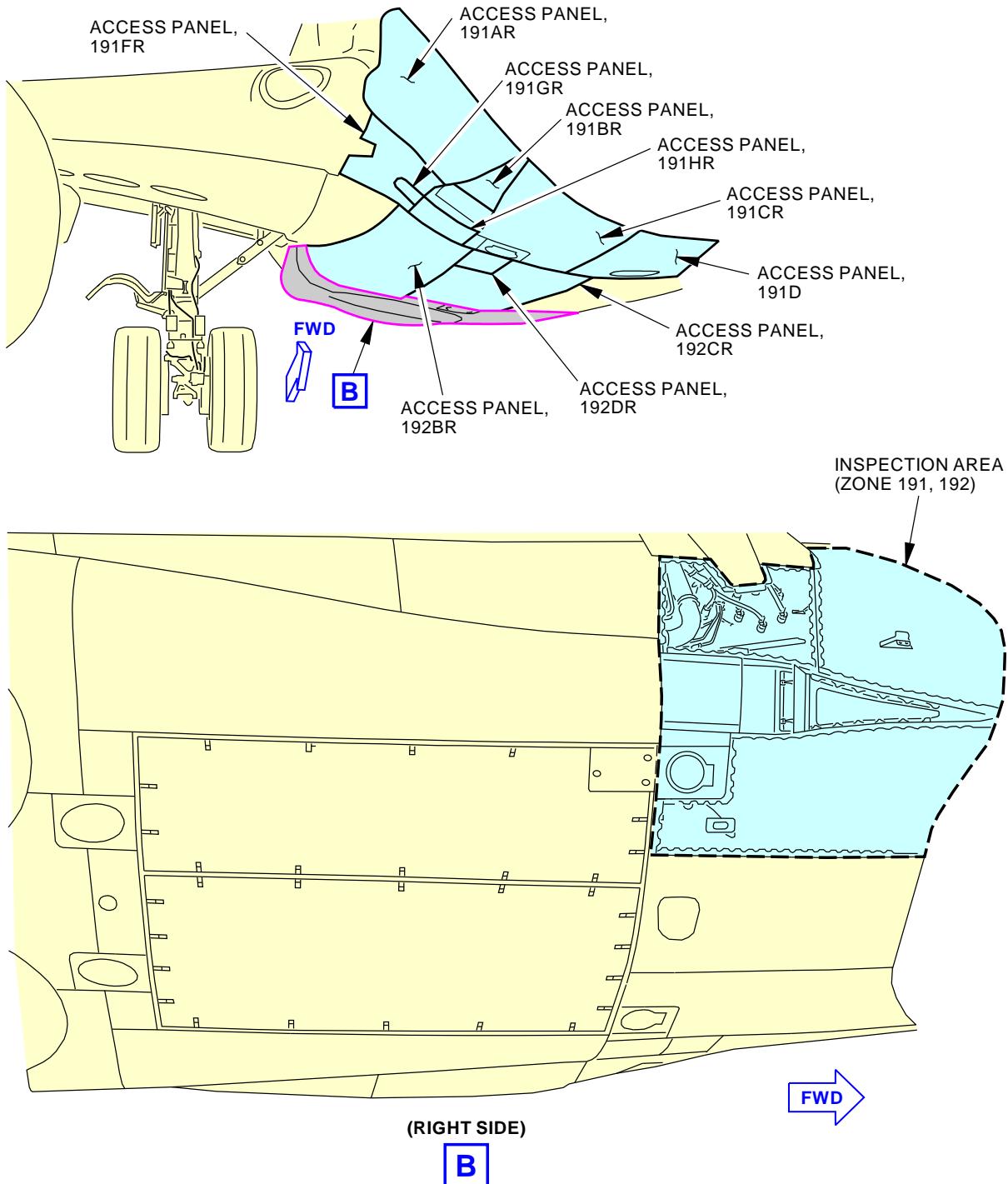
BOEING CARD NO.
53-846-00-01

K76495 S0006584096_V3

**Forward Wing-to-Body Fairing General Visual (Internal)
Figure 1 (Sheet 1 of 2)**EFFECTIVITY
AKS ALLSOURCE
MRB**LOWER WING TO BODY FAIRING - FORWARD OF WING BOX****D633A109-AKS
53-846-00-01****Page 3 of 4
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-846-00-01



Forward Wing-to-Body Fairing General Visual (Internal)
Figure 1 (Sheet 2 of 2)

K76903 S0006584097_V4

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - FORWARD OF WING BOX
		D633A109-AKS 53-846-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LOWER WING TO BODY FAIRING - UNDER WING BOX			BOEING CARD NO.
DATE	TASK ZONAL (GV)				53-848-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 192BL 192BR 192CL 192CR 192DR NOTE			ENGINE ALL
					ZONE 192

Perform an internal zonal inspection (GV) of the lower wing to body fairing - under the wing box. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

ACCESS NOTE: Through access provided.

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - UNDER WING BOX
		D633A109-AKS 53-848-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-848-00-01												
				MECH INSP												
► EWIS TASK 05-41-01-210-826																
1. INTERNAL - ZONAL (GV): Lower Wing to Body Fairing - Under Wing Box (Figure 1)																
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.																
B. Zonal Inspection SUBTASK 05-41-01-010-023 (1) Open these access panels:																
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>192BL</td><td>ECS Ram Air Inlet Mixing Duct Panel - Forward</td></tr><tr><td>192BR</td><td>ECS Ram Air Inlet Mixing Duct Panel - Forward</td></tr><tr><td>192CL</td><td>ECS Access Door</td></tr><tr><td>192CR</td><td>ECS Access Door</td></tr><tr><td>192DR</td><td>ECS High Pressure Access Door</td></tr></tbody></table> <u>NOTE:</u> Through access provided.				<u>Number</u>	<u>Name/Location</u>	192BL	ECS Ram Air Inlet Mixing Duct Panel - Forward	192BR	ECS Ram Air Inlet Mixing Duct Panel - Forward	192CL	ECS Access Door	192CR	ECS Access Door	192DR	ECS High Pressure Access Door	
<u>Number</u>	<u>Name/Location</u>															
192BL	ECS Ram Air Inlet Mixing Duct Panel - Forward															
192BR	ECS Ram Air Inlet Mixing Duct Panel - Forward															
192CL	ECS Access Door															
192CR	ECS Access Door															
192DR	ECS High Pressure Access Door															
SUBTASK 05-41-01-210-026 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.																
 (3) Perform an internal zonal inspection (GV) of the lower wing to body fairing - under the wing box. (EZAP)																
SUBTASK 05-41-01-910-018 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.																
SUBTASK 05-41-01-410-023 (5) Close these access panels:																
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>192BL</td><td>ECS Ram Air Inlet Mixing Duct Panel - Forward</td></tr><tr><td>192BR</td><td>ECS Ram Air Inlet Mixing Duct Panel - Forward</td></tr><tr><td>192CL</td><td>ECS Access Door</td></tr><tr><td>192CR</td><td>ECS Access Door</td></tr><tr><td>192DR</td><td>ECS High Pressure Access Door</td></tr></tbody></table>				<u>Number</u>	<u>Name/Location</u>	192BL	ECS Ram Air Inlet Mixing Duct Panel - Forward	192BR	ECS Ram Air Inlet Mixing Duct Panel - Forward	192CL	ECS Access Door	192CR	ECS Access Door	192DR	ECS High Pressure Access Door	
<u>Number</u>	<u>Name/Location</u>															
192BL	ECS Ram Air Inlet Mixing Duct Panel - Forward															
192BR	ECS Ram Air Inlet Mixing Duct Panel - Forward															
192CL	ECS Access Door															
192CR	ECS Access Door															
192DR	ECS High Pressure Access Door															
— END OF TASK —																

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - UNDER WING BOX	
		D633A109-AKS 53-848-00-01	Page 2 of 4 Feb 15/2015

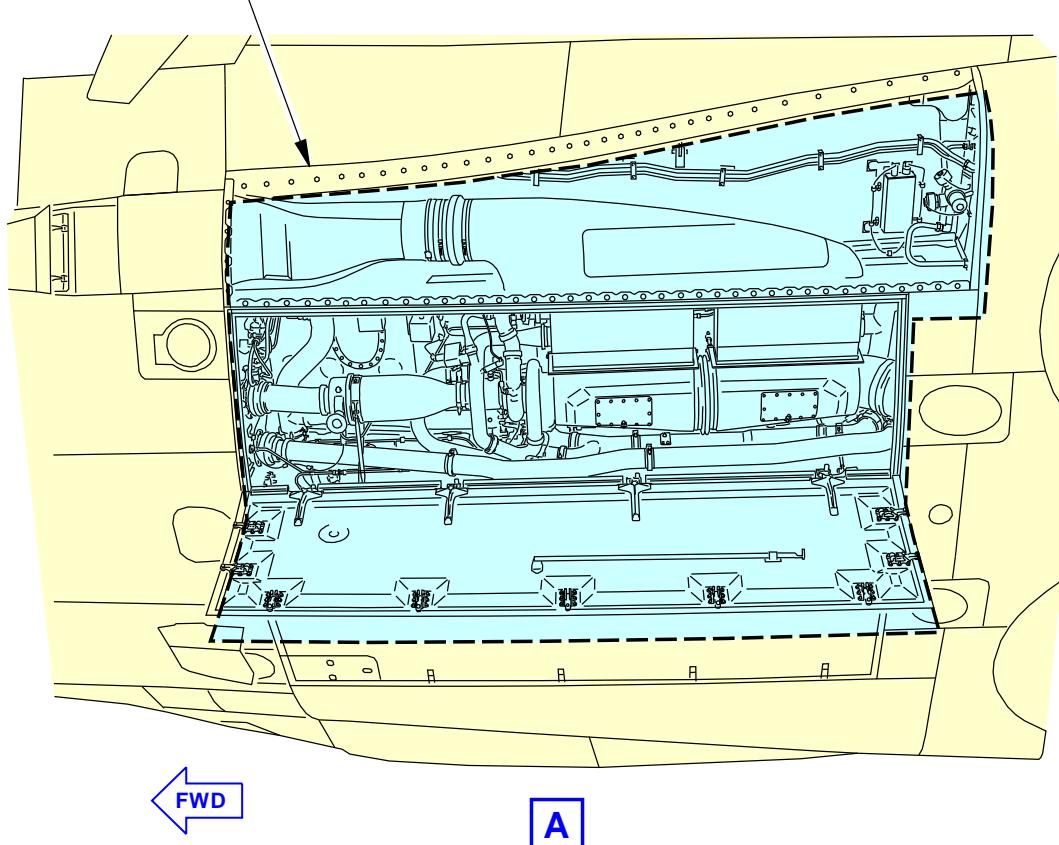
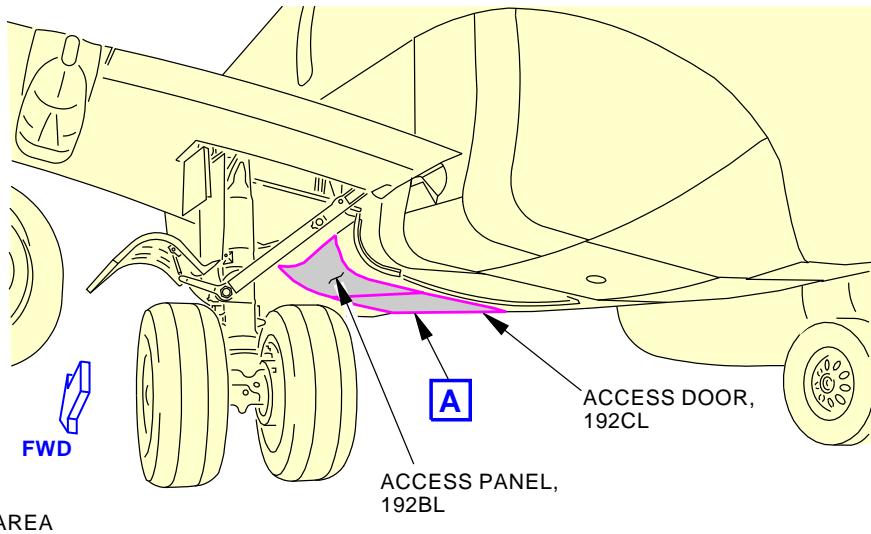
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-848-00-01

K66050 S0006584099_V2

**Underwing - Wing-to-Body Fairing General Visual (Internal)
Figure 1 (Sheet 1 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - UNDER WING BOX
		D633A109-AKS 53-848-00-01

Page 3 of 4
Feb 15/2015

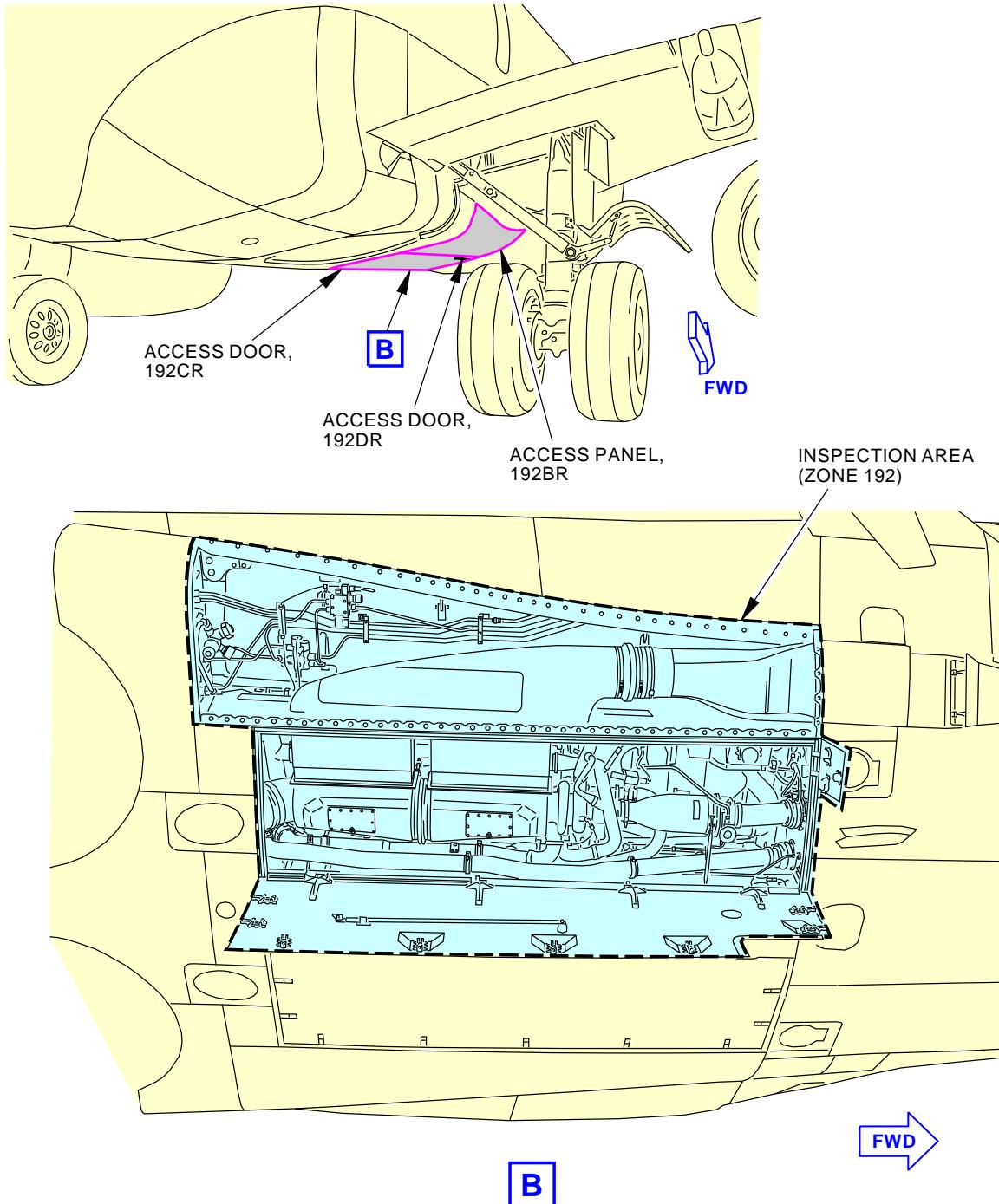
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-848-00-01

K75572 S0006584100_V2

**Underwing - Wing-to-Body Fairing General Visual (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - UNDER WING BOX
		D633A109-AKS 53-848-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL			BOEING CARD NO. 53-850-00-01		
DATE	TASK ZONAL (GV)				RELATED CARD		
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 24000 FC 9 YR	REPEAT 24000 FC 8 YR	APPLICABILITY	AIRPLANE ALL	ENGINE ALL
STATION	SKILL AIRPL	ACCESS 194AL 194AR 194BL 194BR 194CL 194CR 194DL 194DR 194E 194FL 194FR 194GL 194GR 194HL 194HR			ZONE 194		

Perform an internal zonal inspection (GV) of the lower wing to body fairing - aft of wheel well. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL	
		D633A109-AKS 53-850-00-01	Page 1 of 5 Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-850-00-01																																
EWIS				MECH INSP																																
TASK 05-41-01-210-827																																				
1. INTERNAL - ZONAL (GV): Lower Wing to Body Fairing - Aft of Wheel Well																																				
(Figure 1)																																				
A. General																																				
(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.																																				
B. Zonal Inspection																																				
SUBTASK 05-41-01-010-024																																				
(1) Open these access panels:																																				
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>194AL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194AR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194BL</td><td>Flap Track Lubrication Panel - Aft</td></tr><tr><td>194BR</td><td>Flap Track Lubrication Panel - Aft</td></tr><tr><td>194CL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194CR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194DL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194DR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194E</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194FL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194FR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194GR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194HL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194HR</td><td>Aft Wing To Body Fairing Panel</td></tr></tbody></table>				<u>Number</u>	<u>Name/Location</u>	194AL	Aft Wing To Body Fairing Panel	194AR	Aft Wing To Body Fairing Panel	194BL	Flap Track Lubrication Panel - Aft	194BR	Flap Track Lubrication Panel - Aft	194CL	Aft Wing To Body Fairing Panel	194CR	Aft Wing To Body Fairing Panel	194DL	Aft Wing To Body Fairing Panel	194DR	Aft Wing To Body Fairing Panel	194E	Aft Wing To Body Fairing Panel	194FL	Aft Wing To Body Fairing Panel	194FR	Aft Wing To Body Fairing Panel	194GL	Aft Wing To Body Fairing Panel	194GR	Aft Wing To Body Fairing Panel	194HL	Aft Wing To Body Fairing Panel	194HR	Aft Wing To Body Fairing Panel	
<u>Number</u>	<u>Name/Location</u>																																			
194AL	Aft Wing To Body Fairing Panel																																			
194AR	Aft Wing To Body Fairing Panel																																			
194BL	Flap Track Lubrication Panel - Aft																																			
194BR	Flap Track Lubrication Panel - Aft																																			
194CL	Aft Wing To Body Fairing Panel																																			
194CR	Aft Wing To Body Fairing Panel																																			
194DL	Aft Wing To Body Fairing Panel																																			
194DR	Aft Wing To Body Fairing Panel																																			
194E	Aft Wing To Body Fairing Panel																																			
194FL	Aft Wing To Body Fairing Panel																																			
194FR	Aft Wing To Body Fairing Panel																																			
194GL	Aft Wing To Body Fairing Panel																																			
194GR	Aft Wing To Body Fairing Panel																																			
194HL	Aft Wing To Body Fairing Panel																																			
194HR	Aft Wing To Body Fairing Panel																																			
SUBTASK 05-41-01-210-027																																				
(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.																																				
(3) Perform an internal zonal inspection (GV) of the lower wing to body fairing - aft of wheel well. (EZAP)																																				
SUBTASK 05-41-01-910-019																																				
(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.																																				
SUBTASK 05-41-01-410-024																																				
(5) Close these access panels:																																				
<table><thead><tr><th><u>Number</u></th><th><u>Name/Location</u></th></tr></thead><tbody><tr><td>194AL</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194AR</td><td>Aft Wing To Body Fairing Panel</td></tr><tr><td>194BL</td><td>Flap Track Lubrication Panel - Aft</td></tr><tr><td>194BR</td><td>Flap Track Lubrication Panel - Aft</td></tr><tr><td>194CL</td><td>Aft Wing To Body Fairing Panel</td></tr></tbody></table>				<u>Number</u>	<u>Name/Location</u>	194AL	Aft Wing To Body Fairing Panel	194AR	Aft Wing To Body Fairing Panel	194BL	Flap Track Lubrication Panel - Aft	194BR	Flap Track Lubrication Panel - Aft	194CL	Aft Wing To Body Fairing Panel																					
<u>Number</u>	<u>Name/Location</u>																																			
194AL	Aft Wing To Body Fairing Panel																																			
194AR	Aft Wing To Body Fairing Panel																																			
194BL	Flap Track Lubrication Panel - Aft																																			
194BR	Flap Track Lubrication Panel - Aft																																			
194CL	Aft Wing To Body Fairing Panel																																			
EFFECTIVITY AKS ALL				SOURCE MRB																																
LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL				D633A109-AKS 53-850-00-01																																
				Page 2 of 5 Feb 15/2015																																

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-850-00-01
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(Continued)

Number Name/Location

194CR	Aft Wing To Body Fairing Panel
194DL	Aft Wing To Body Fairing Panel
194DR	Aft Wing To Body Fairing Panel
194E	Aft Wing To Body Fairing Panel
194FL	Aft Wing To Body Fairing Panel
194FR	Aft Wing To Body Fairing Panel
194GL	Aft Wing To Body Fairing Panel
194GR	Aft Wing To Body Fairing Panel
194HL	Aft Wing To Body Fairing Panel
194HR	Aft Wing To Body Fairing Panel

MECH

INSP

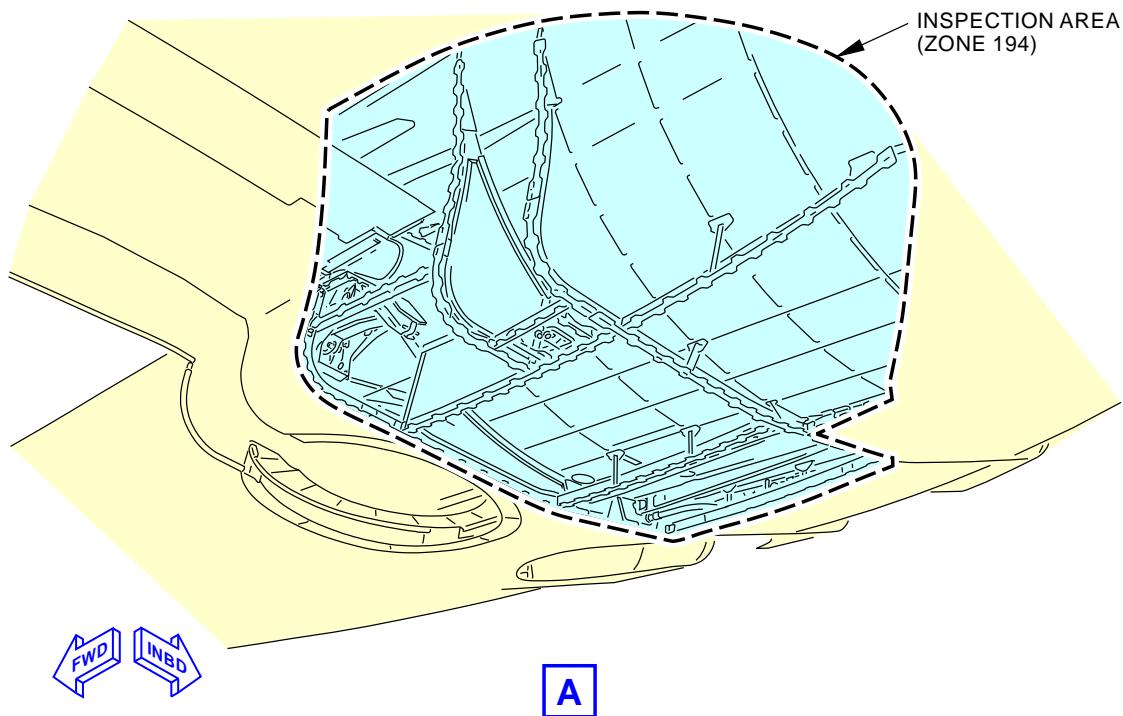
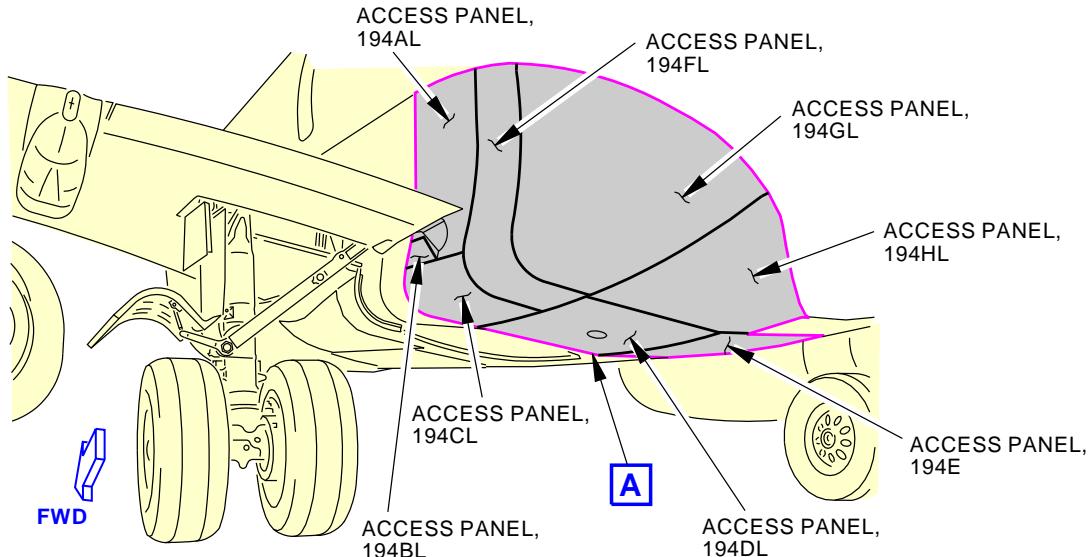
— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL
		D633A109-AKS 53-850-00-01

**Page 3 of 5
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-850-00-01
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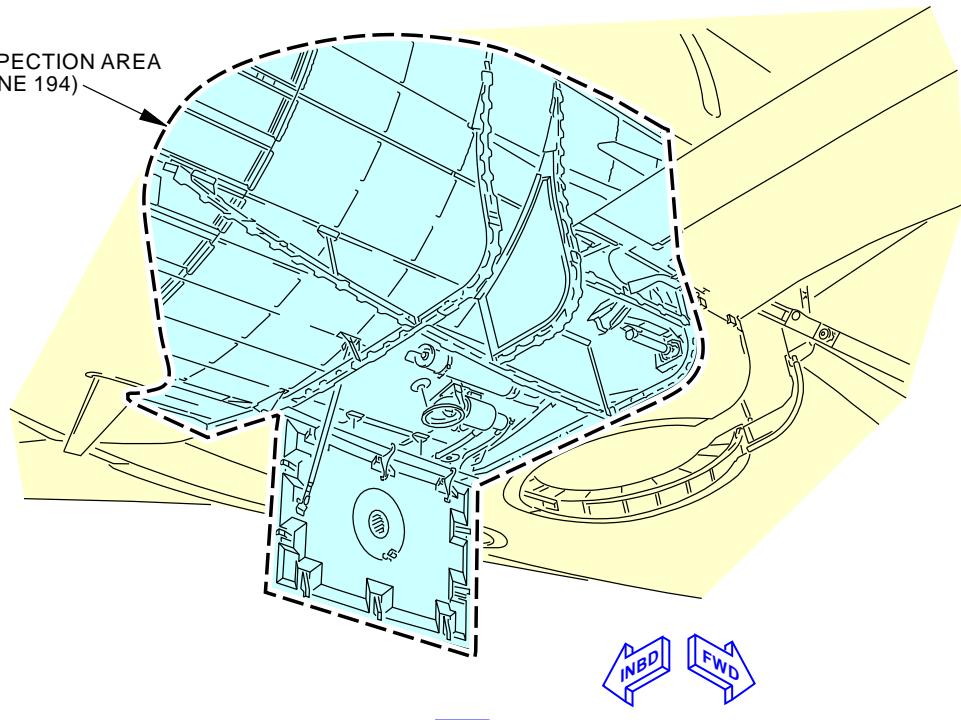
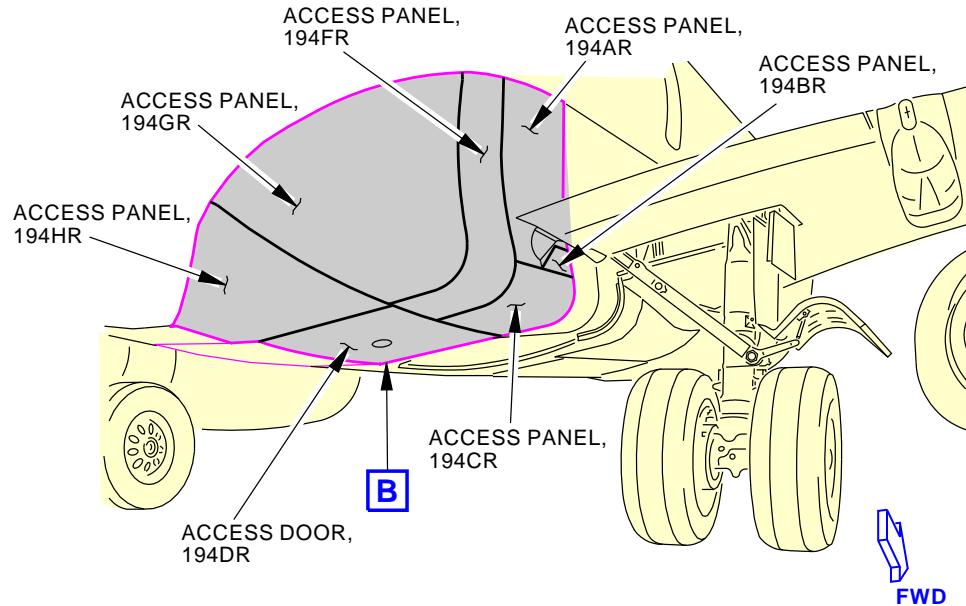


K66154 S0006584102_V2
Aft of Wheel Well - Wing-to-Body Fairing General Visual (Internal)
Figure 1 (Sheet 1 of 2)

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL
		D633A109-AKS 53-850-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-850-00-01
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K66156 S0006584103_V2

**Aft of Wheel Well - Wing-to-Body Fairing General Visual (Internal)
Figure 1 (Sheet 2 of 2)**

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL
		D633A109-AKS 53-850-00-01

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Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL			BOEING CARD NO.		
DATE	TASK ZONAL (GV)				53-852-00-01		
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY AIRPLANE ALL		
STATION	SKILL AIRPL	ACCESS 194DR NOTE			ENGINE ALL		
					ZONE 194		

Perform an internal zonal inspection (GV) of the lower wing to body fairing - aft of wheel well.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Through access provided.

EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL
		D633A109-AKS 53-852-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-852-00-01
TASK 05-41-01-210-828				MECH INSP
1. INTERNAL - ZONAL (GV): LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL (Figure 1)				
A. Zonal Inspection				
SUBTASK 05-41-01-010-025				
(1) Open this access panel:				
Number Name/Location				
194DR Aft Wing To Body Fairing Panel				
NOTE: Through access provided.				
SUBTASK 05-41-01-210-028				
(2) Do a General Visual inspection of the lower wing to body fairing - aft of wheel well.				
SUBTASK 05-41-01-410-025				
(3) Close this access panel:				
Number Name/Location				
194DR Aft Wing To Body Fairing Panel				
———— END OF TASK ————				
EFFECTIVITY AKS ALL	SOURCE MRB	LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL		
		D633A109-AKS 53-852-00-01	Page 2 of 3 Feb 15/2015	

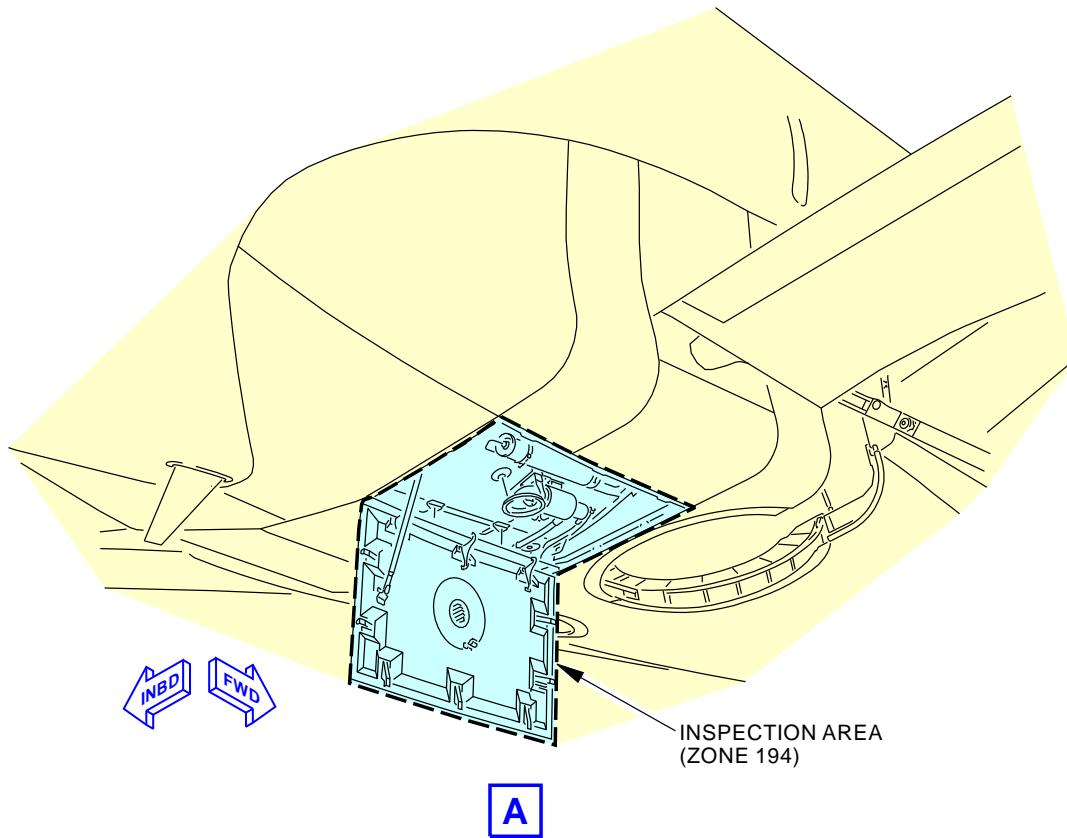
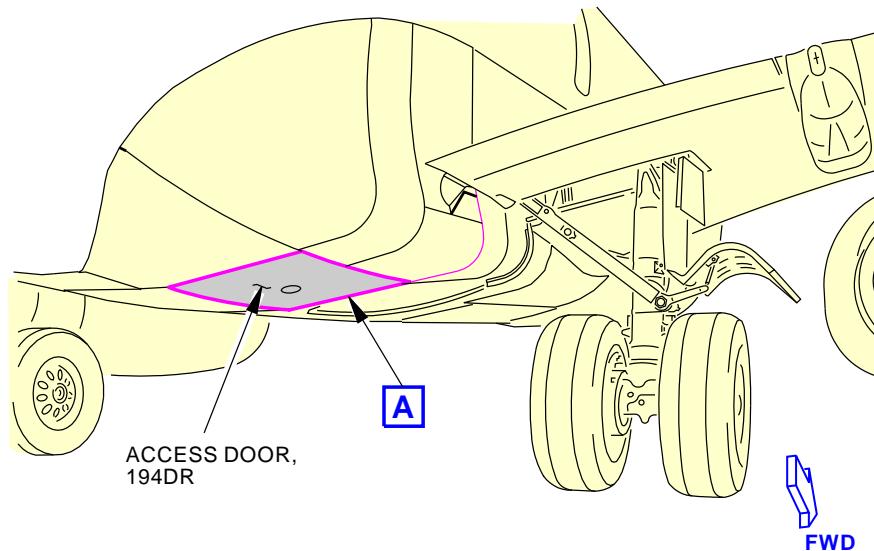
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-852-00-01

K76876 S0006584105_V2

**Right Aft of Wheel Well - Wing-to-Body Fairing General Visual (Internal)
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**LOWER WING TO BODY FAIRING - AFT OF WHEEL WELL****D633A109-AKS
53-852-00-01****Page 3 of 3
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE ABOVE WING, WING TO BODY FAIRING - RIGHT			BOEING CARD NO. 53-854-02-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 24000 FC 9 YR	REPEAT 24000 FC 8 YR	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 195AR 195BR 195CR			ZONE 196	ENGINE ALL

Perform an internal zonal inspection (GV) of the above wing, wing to body fairing - right side.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	ABOVE WING, WING TO BODY FAIRING - RIGHT
		D633A109-AKS 53-854-02-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-854-02-01
TASK 05-41-01-210-829				MECH INSP

1. INTERNAL - ZONAL (GV): ABOVE WING, WING TO BODY FAIRING - RIGHT
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-01-010-026

(1) Open these access panels:

<u>Number</u>	<u>Name/Location</u>
195AR	Wing To Body Fairing - Right Side
195BR	Wing To Body Fairing - Right Side
195CR	Wing To Body Fairings - Right Side

SUBTASK 05-41-01-210-029

(2) Do a General Visual inspection of the above wing, wing to body fairing - right side.

SUBTASK 05-41-01-410-026

(3) Close these access panels:

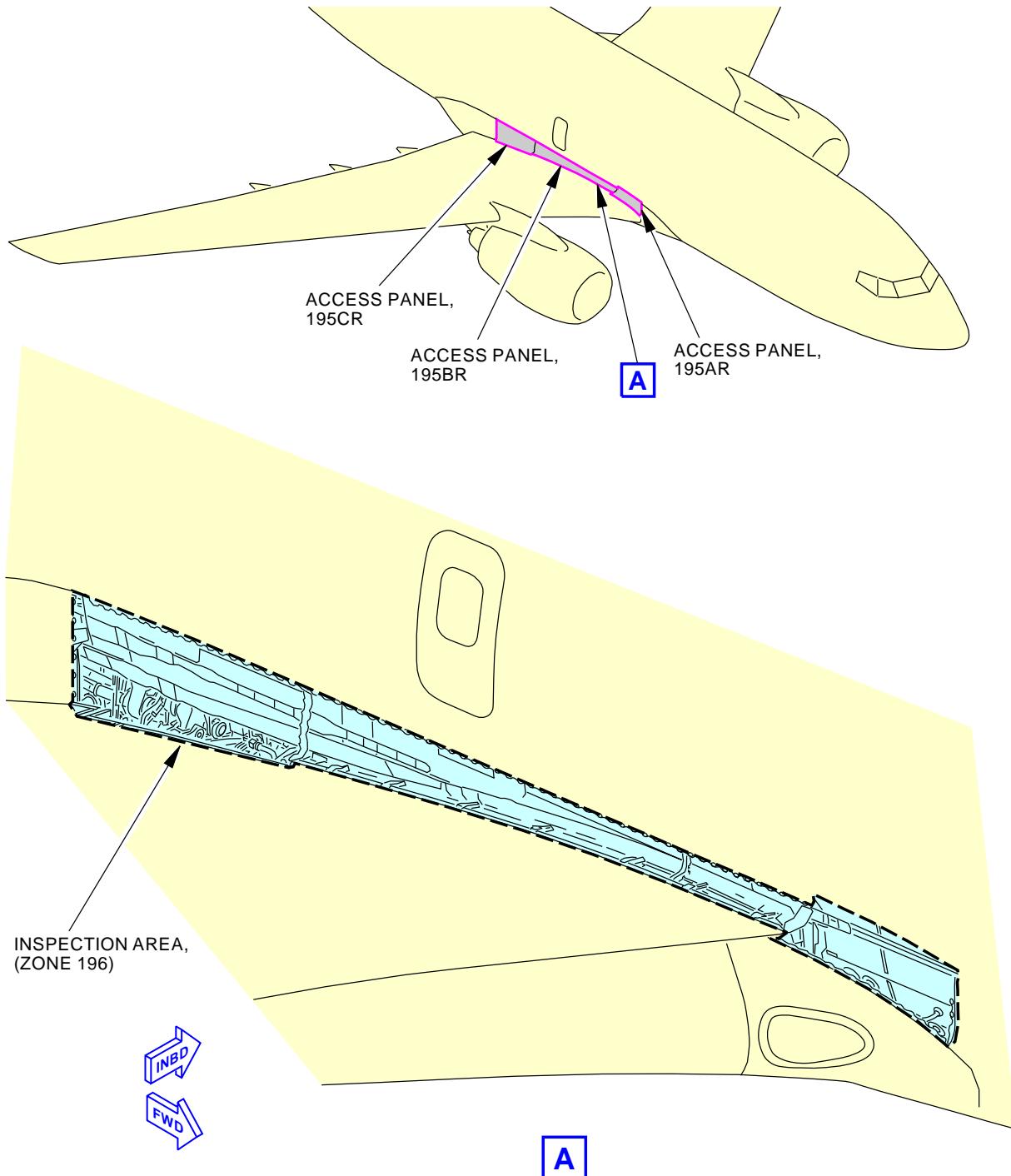
<u>Number</u>	<u>Name/Location</u>
195AR	Wing To Body Fairing - Right Side
195BR	Wing To Body Fairing - Right Side
195CR	Wing To Body Fairings - Right Side

— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	ABOVE WING, WING TO BODY FAIRING - RIGHT
		D633A109-AKS 53-854-02-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-854-02-01



K78579 S0006584107_V4

**Above the Right Wing - Wing-to-Body Fairing General Visual (Internal)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	ABOVE WING, WING TO BODY FAIRING - RIGHT
		D633A109-AKS 53-854-02-01

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Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE ABOVE WING, WING TO BODY FAIRING - LEFT			BOEING CARD NO. 53-856-01-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 24000 FC 9 YR	REPEAT 24000 FC 8 YR	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 195AL 195BL 195CL			ZONE 195	ENGINE ALL

Perform an internal zonal inspection (GV) of the above wing, wing to body fairing - left side.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	ABOVE WING, WING TO BODY FAIRING - LEFT
		D633A109-AKS 53-856-01-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-856-01-01
TASK 05-41-01-210-830				MECH INSP

1. INTERNAL - ZONAL (GV): ABOVE WING, WING TO BODY FAIRING - LEFT

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-01-010-027

- (1) Open these access panels:

Number Name/Location

195AL	Wing To Body Fairing - Left Side
195BL	Wing To Body Fairing - Left Side
195CL	Wing To Body Fairing - Left Side

SUBTASK 05-41-01-210-030

- (2) Do a General Visual inspection of the above wing, wing to body fairing - left side.

SUBTASK 05-41-01-410-027

- (3) Close these access panels:

Number Name/Location

195AL	Wing To Body Fairing - Left Side
195BL	Wing To Body Fairing - Left Side
195CL	Wing To Body Fairing - Left Side

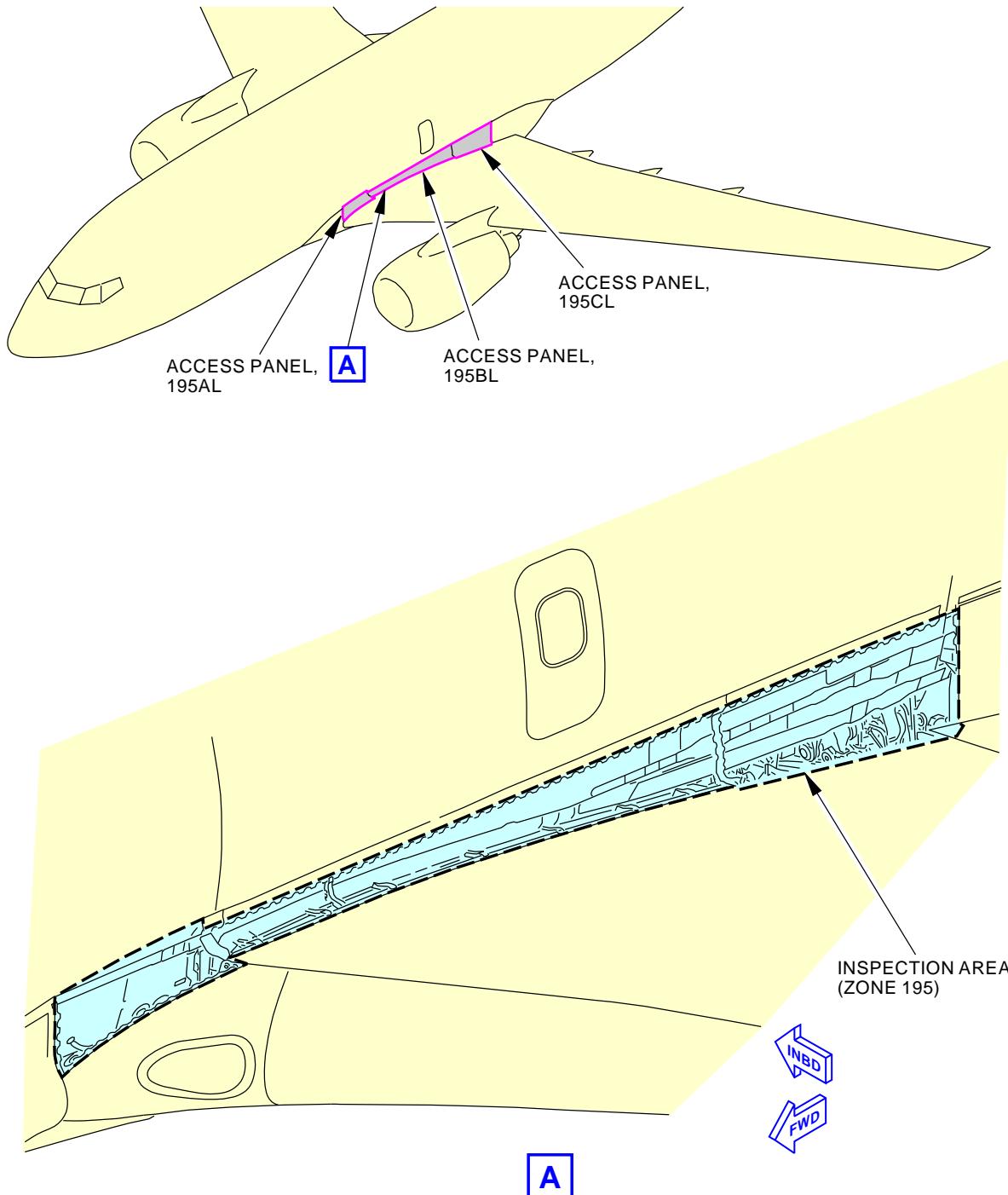
———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	ABOVE WING, WING TO BODY FAIRING - LEFT
		D633A109-AKS 53-856-01-01

Page 2 of 3
Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-856-01-01



L03910 S0006584109_V3

**Above the Left Wing - Wing-to-Body Fairing General Visual (Internal)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	ABOVE WING, WING TO BODY FAIRING - LEFT
		D633A109-AKS 53-856-01-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FLIGHT CONTROL COMPARTMENT			BOEING CARD NO. 53-858-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	ACCESS			ZONE 211 212

Perform an external zonal inspection (GV) of the flight control compartment - section 41, STA 178 to STA 270.

INTERVAL NOTE: Whichever comes first.

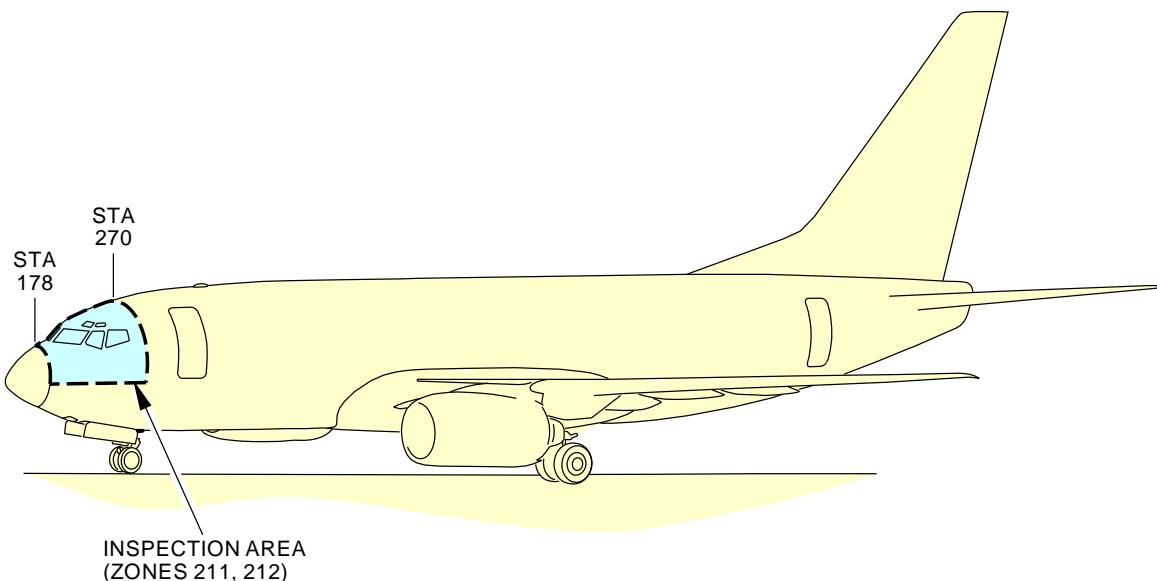
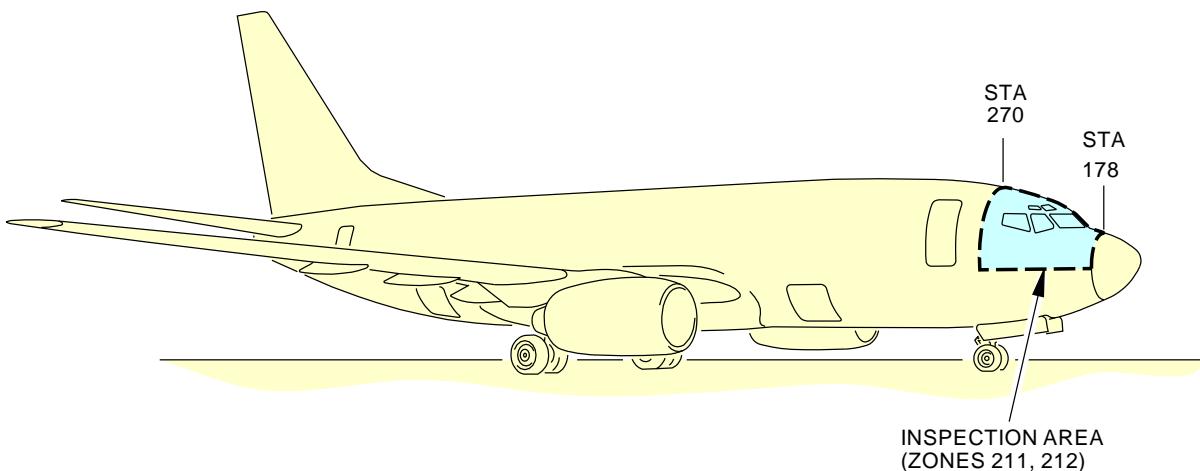
EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT CONTROL COMPARTMENT
		D633A109-AKS 53-858-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-858-00-01
				MECH INSP
TASK 05-41-02-210-801				
1. EXTERNAL - ZONAL (GV): FLIGHT CONTROL COMPARTMENT				
(Figure 1)				
A. Zonal Inspection				
SUBTASK 05-41-02-210-001				
(1) Do a General Visual inspection of the flight control compartment - Section 41, Sta 178 to Sta 270.				
———— END OF TASK ——				
EFFECTIVITY AKS ALL		SOURCE MRB	FLIGHT CONTROL COMPARTMENT	
			D633A109-AKS 53-858-00-01	
				Page 2 of 3 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-858-00-01



L05706 S0006584113_V3

**Flight Compartment (Sta 178-270) General Visual (External)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT CONTROL COMPARTMENT
		D633A109-AKS 53-858-00-01

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Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FLIGHT CONTROL COMPARTMENT			BOEING CARD NO. 53-860-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 5500 FC 24 MO	REPEAT 5500 FC 24 MO	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS NOTE			ENGINE ALL
					ZONE 211 212

Perform an internal zonal inspection (GV) of the flight control compartment - section 41, sta 178 to sta 270.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Control stand access panels 211A and 212A removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT CONTROL COMPARTMENT
		D633A109-AKS 53-860-00-01

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-860-00-01	MECH	INSP
TASK 05-41-02-210-802						
1. INTERNAL - ZONAL (GV): FLIGHT CONTROL COMPARTMENT						
(Figure 1)						
A. Zonal Inspection						
NOTE: Control stand access panels 211A and 212A removal required.						
SUBTASK 05-41-02-210-002						
(1) Do a General Visual inspection of the flight control compartment - Section 41, Sta 178 to Sta 270.						
———— END OF TASK ——						

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT CONTROL COMPARTMENT
		D633A109-AKS 53-860-00-01

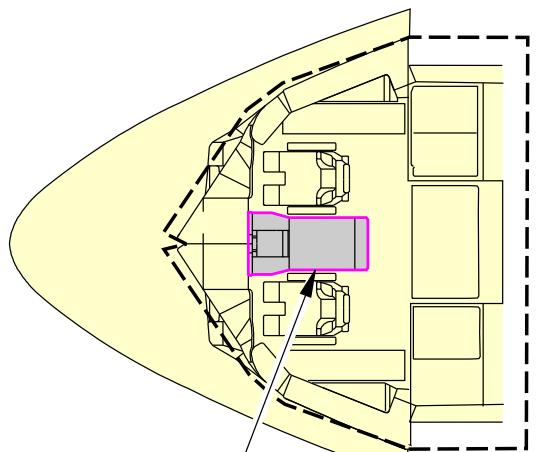
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

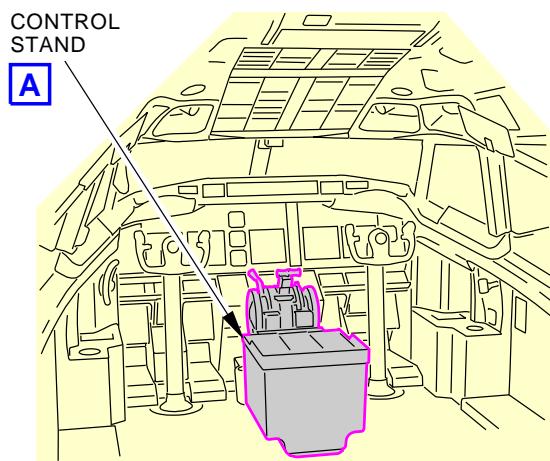
AIRLINE CARD NO.

BOEING CARD NO.
53-860-00-01

CONTROL STAND

A

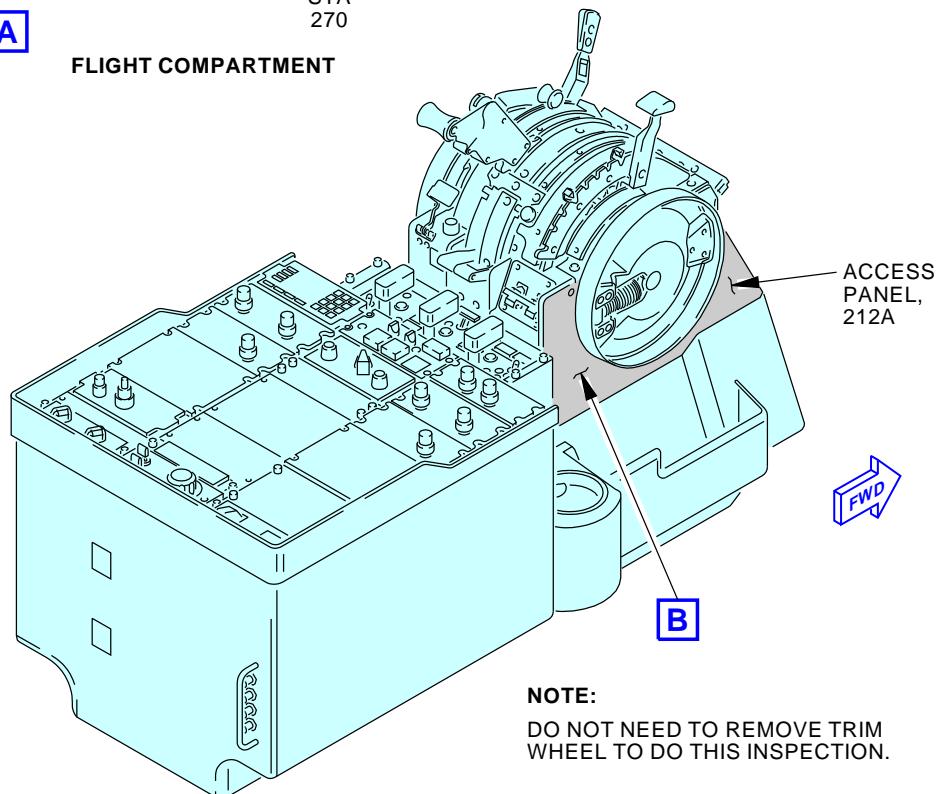
FLIGHT COMPARTMENT



CONTROL STAND

A

FLIGHT COMPARTMENT

ACCESS PANEL,
212A

FWD

B

NOTE:

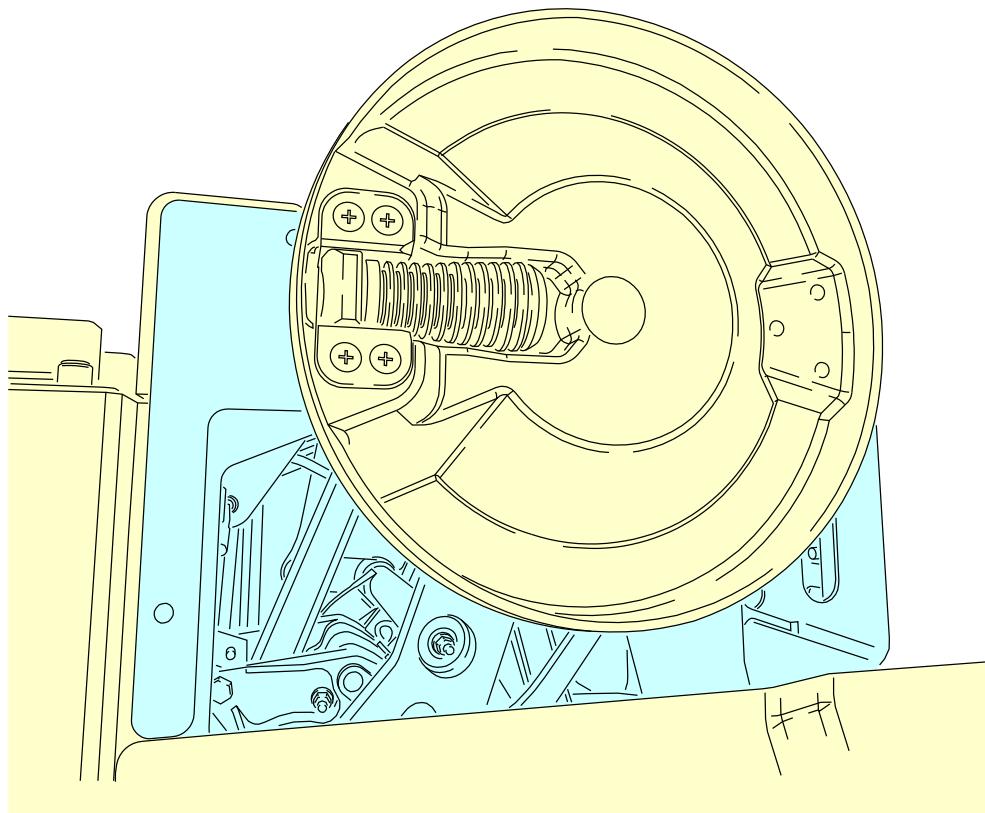
DO NOT NEED TO REMOVE TRIM
WHEEL TO DO THIS INSPECTION.CONTROL STAND
(RIGHT SIDE)**A**

K78493 S0006584115_V4

**Control Stand - Flight Compartment General Visual (Internal)
Figure 1 (Sheet 1 of 3)**EFFECTIVITY
AKS ALLSOURCE
MRB**FLIGHT CONTROL COMPARTMENT****D633A109-AKS
53-860-00-01****Page 3 of 5
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AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-860-00-01

**B**

K78574 S0006584116_V4

**Control Stand - Flight Compartment General Visual (Internal)
Figure 1 (Sheet 2 of 3)**

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT CONTROL COMPARTMENT
		D633A109-AKS 53-860-00-01

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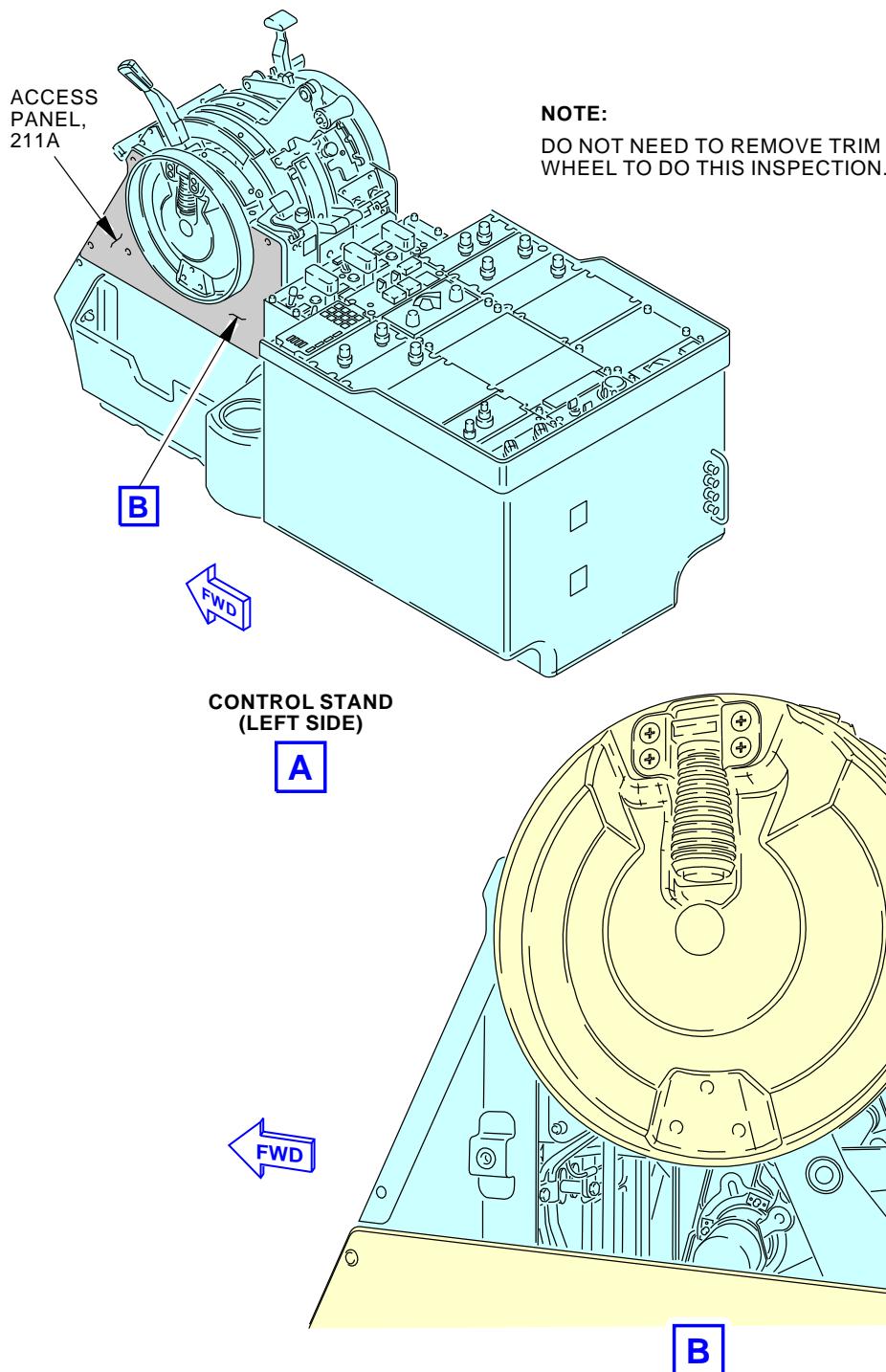
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-860-00-01

K78601 S0006584117_V4

**Control Stand - Flight Compartment General Visual (Internal)
Figure 1 (Sheet 3 of 3)**EFFECTIVITY
AKS ALLSOURCE
MRB**FLIGHT CONTROL COMPARTMENT****D633A109-AKS
53-860-00-01****Page 5 of 5
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FLIGHT CONTROL COMPARTMENT			BOEING CARD NO. 53-862-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 36000 FC 10 YR	REPEAT 36000 FC 10 YR	APPLICABILITY	
STATION	SKILL AIRPL	ACCESS 211AF NOTE			AIRPLANE ALL	ENGINE ALL
		ACCESS 211AF NOTE			ZONE 211 212	

Perform an internal zonal inspection (GV) of the flight control compartment - section 41, STA 178 to STA 270. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: With access provided. Seats removed. Control stand access panels, overhead and sidewall panels, and floor panel removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT CONTROL COMPARTMENT
		D633A109-AKS 53-862-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-862-00-01
				MECH INSP
► EWIS TASK 05-41-02-210-803				
1. INTERNAL - ZONAL (GV): Flight Control Compartment (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection SUBTASK 05-41-02-010-001 (1) Open this access panel: Number Name/Location 211AF Panel Assy - Flight Compartment Floor NOTE: With access provided. Seats removed. Control stand access panels, overhead and sidewall panels, and floor panel.				
SUBTASK 05-41-02-210-003 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the flight control compartment - section 41, STA 178 to STA 270. (EZAP)				
SUBTASK 05-41-02-410-001 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-02-410-001 (5) Close this access panel: Number Name/Location 211AF Panel Assy - Flight Compartment Floor				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT CONTROL COMPARTMENT
		D633A109-AKS 53-862-00-01

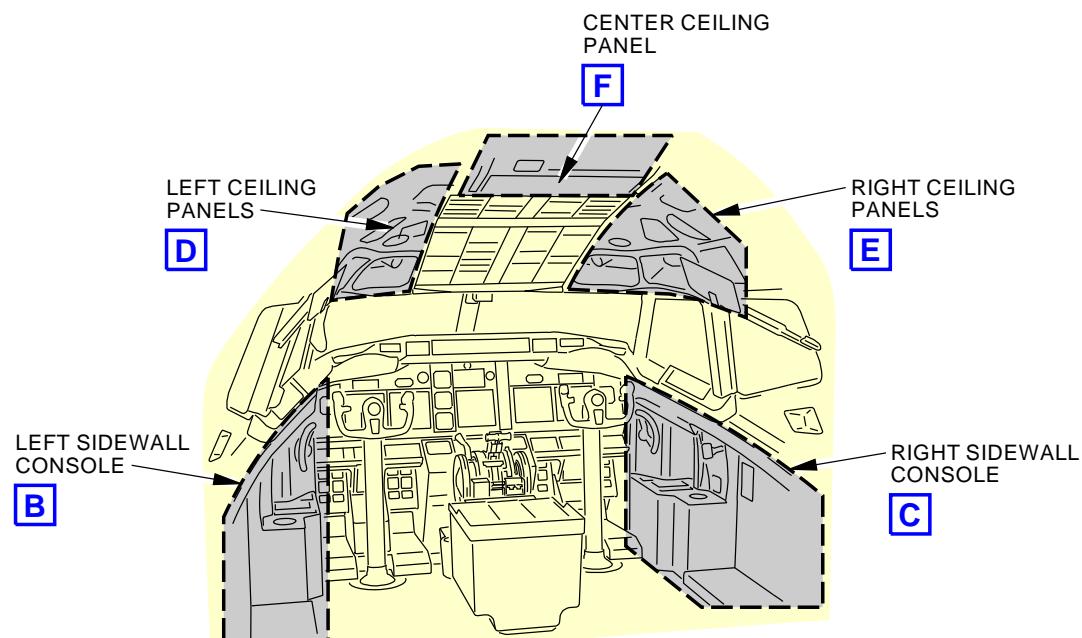
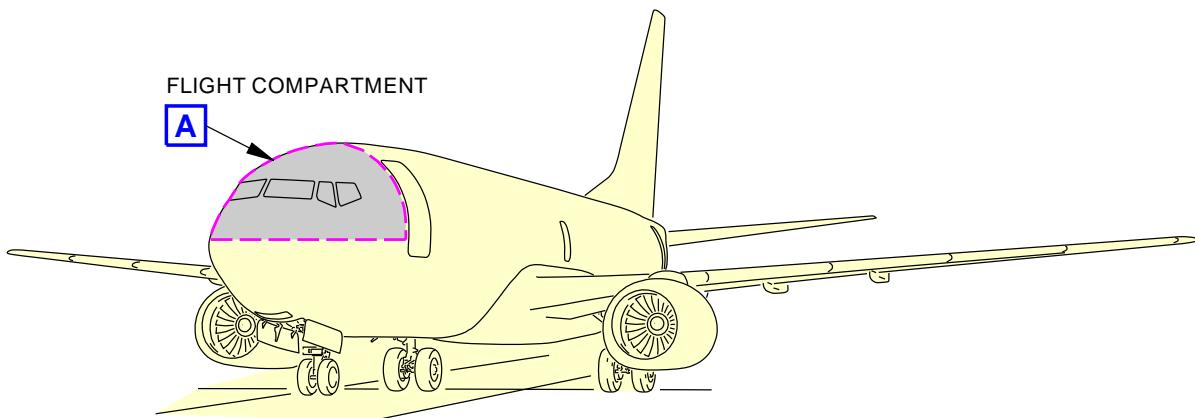
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-862-00-01**FLIGHT COMPARTMENT**

K96666 S0006584119_V2

Ceiling Panels and Sidewall Console - Flight Compartment General Visual (Internal)
Figure 1 (Sheet 1 of 6)

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT CONTROL COMPARTMENT
		D633A109-AKS 53-862-00-01

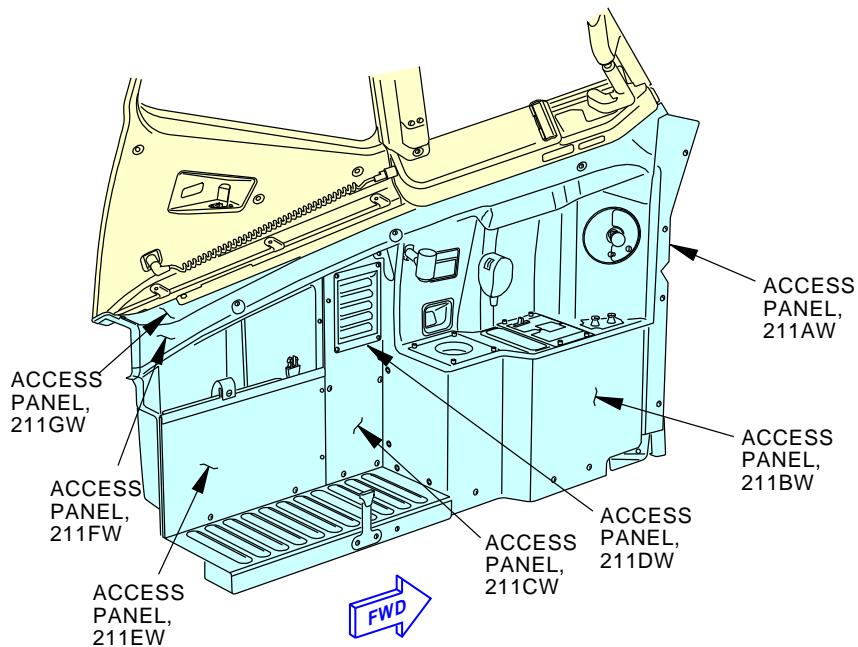
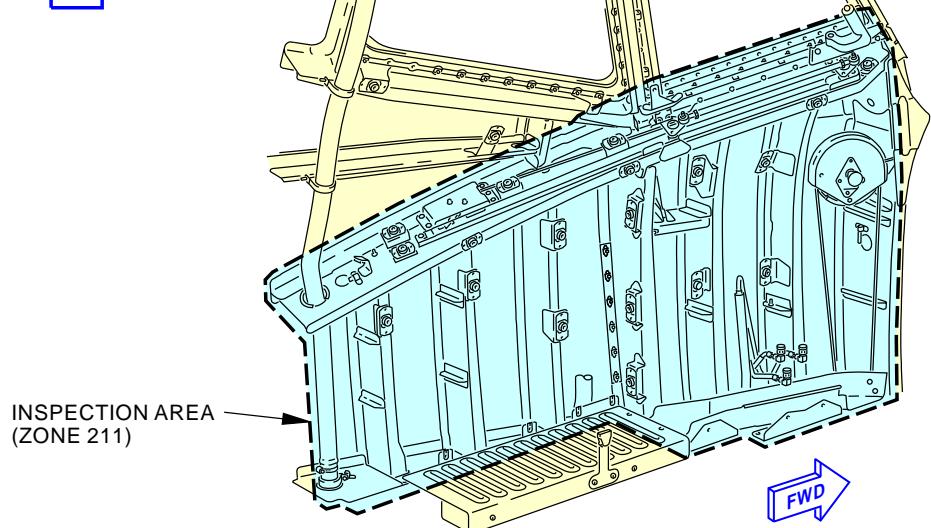
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-862-00-01**B****B**

Ceiling Panels and Sidewall Console - Flight Compartment General Visual (Internal)
Figure 1 (Sheet 2 of 6)

K96802 S0006584120_V2

EFFECTIVITY AKS ALL	SOURCE MRB	FLIGHT CONTROL COMPARTMENT
		D633A109-AKS 53-862-00-01

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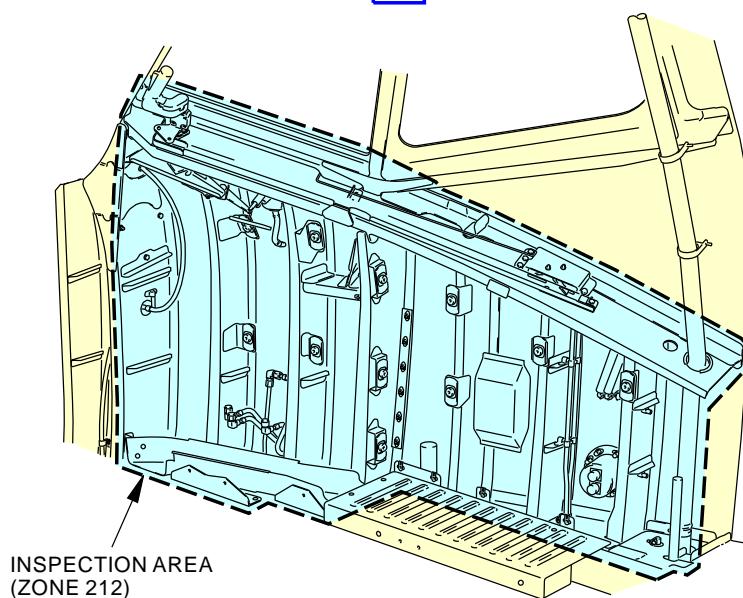
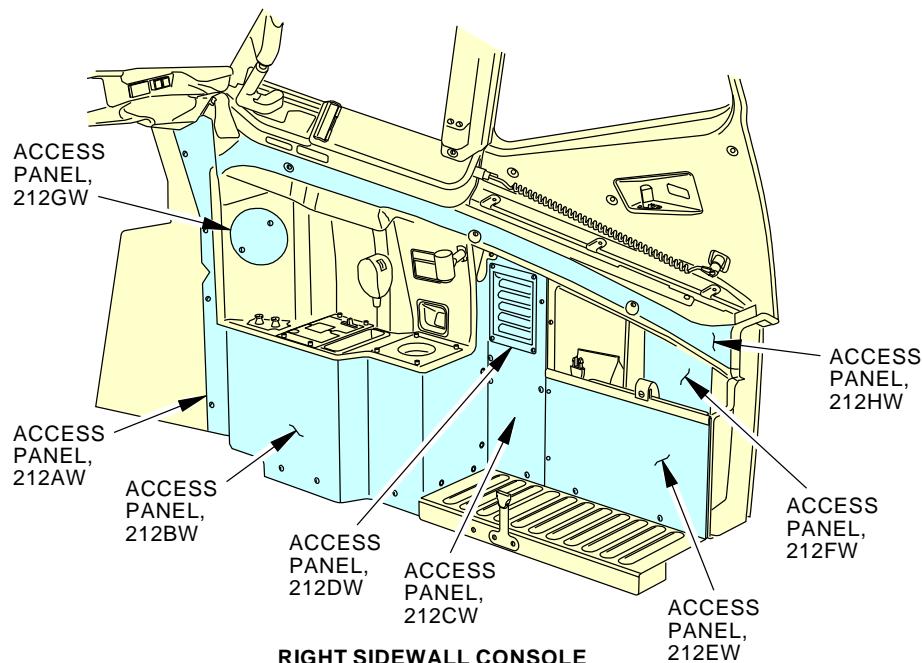
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-862-00-01

K97349 S0006584121_V2

Ceiling Panels and Sidewall Console - Flight Compartment General Visual (Internal)
Figure 1 (Sheet 3 of 6)EFFECTIVITY
AKS ALLSOURCE
MRB**FLIGHT CONTROL COMPARTMENT****D633A109-AKS
53-862-00-01****Page 5 of 8
Feb 15/2015**

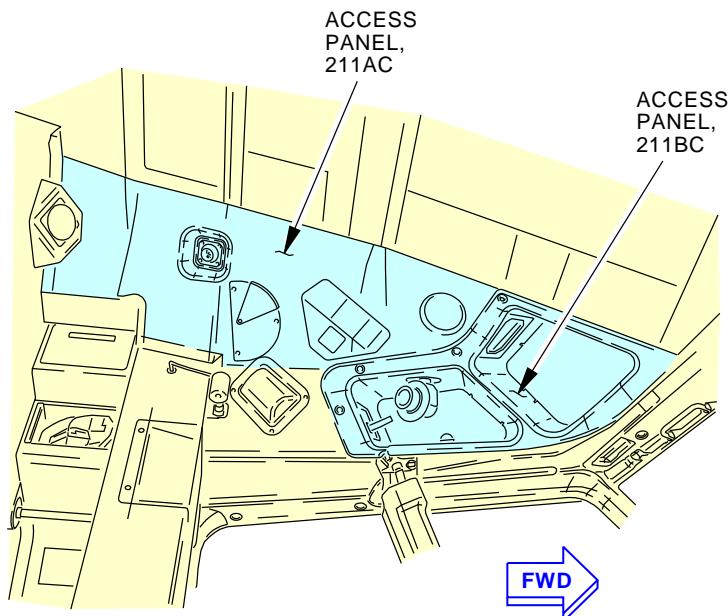
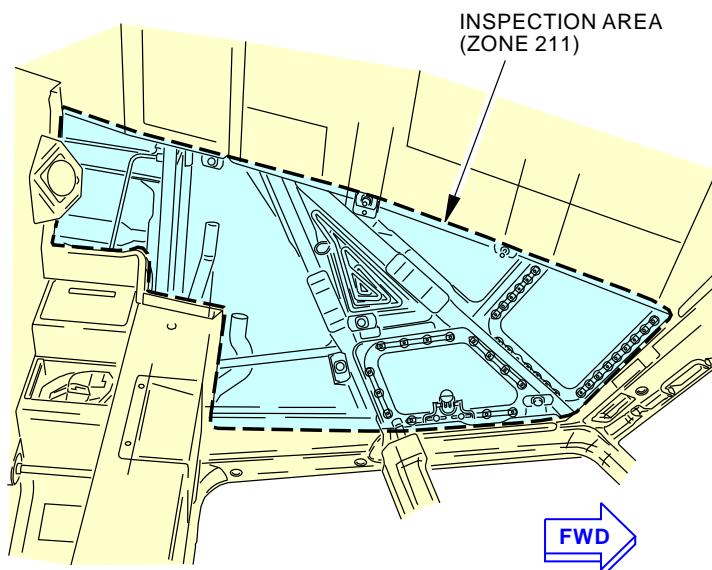
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-862-00-01**LEFT CEILING PANELS****D****LEFT CEILING PANELS
(PANELS REMOVED)****D**

K97386 S0006584122_V2

**Ceiling Panels and Sidewall Console - Flight Compartment General Visual (Internal)
Figure 1 (Sheet 4 of 6)**EFFECTIVITY
AKS ALLSOURCE
MRB**FLIGHT CONTROL COMPARTMENT****D633A109-AKS
53-862-00-01****Page 6 of 8
Feb 15/2015**

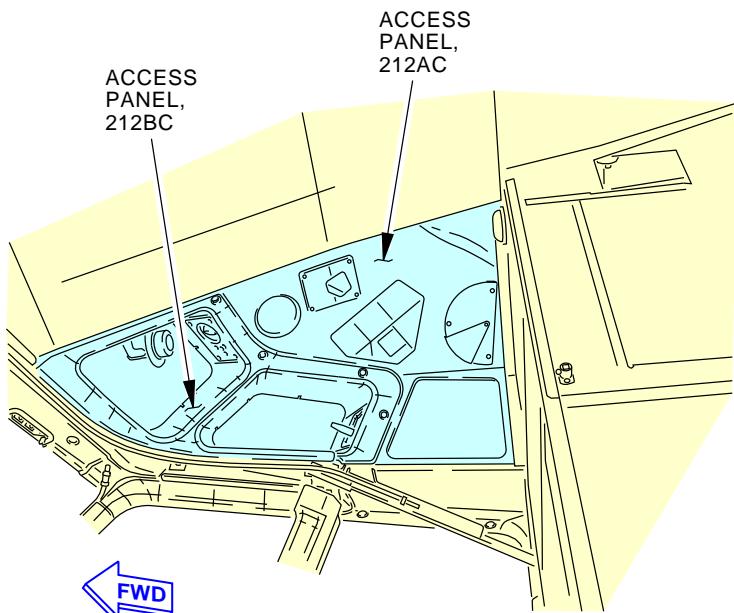
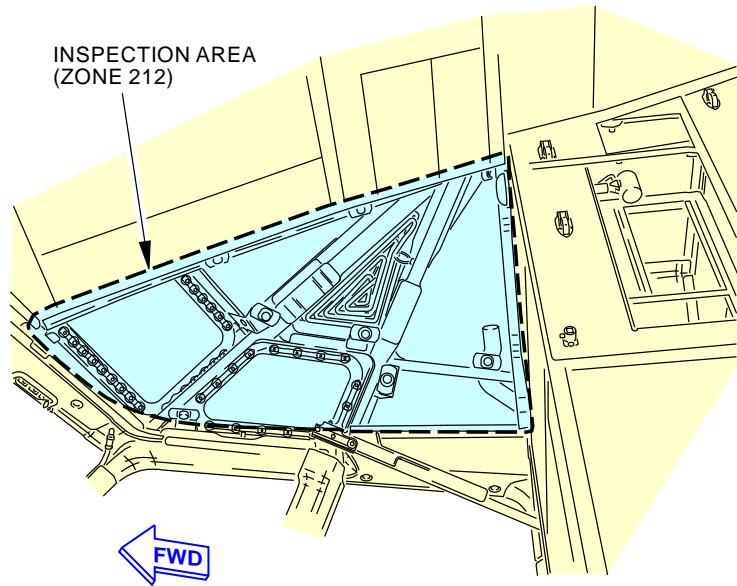
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-862-00-01**RIGHT CEILING PANELS****E****RIGHT CEILING PANELS
(PANELS REMOVED)****E**

K97747 S0006584123_V3

**Ceiling Panels and Sidewall Console - Flight Compartment General Visual (Internal)
Figure 1 (Sheet 5 of 6)**EFFECTIVITY
AKS ALLSOURCE
MRB**FLIGHT CONTROL COMPARTMENT****D633A109-AKS
53-862-00-01****Page 7 of 8
Feb 15/2015**

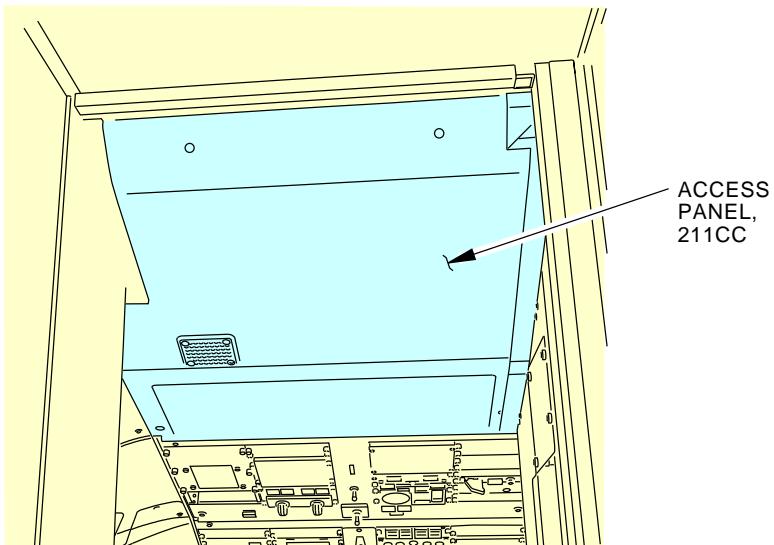
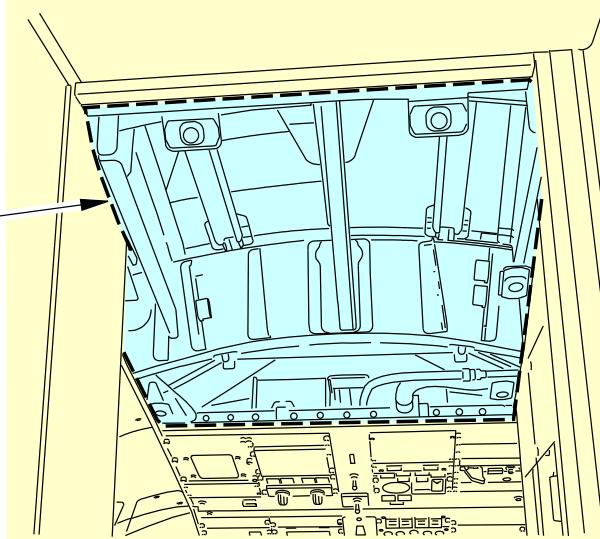
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-862-00-01
FWD**CENTER CEILING PANEL****F**
**INSPECTION AREA
(ZONE 211)**
FWD**CENTER CEILING PANEL
(PANEL REMOVED)****F**

K97751 S0006584124_V2

**Ceiling Panels and Sidewall Console - Flight Compartment General Visual (Internal)
Figure 1 (Sheet 6 of 6)**EFFECTIVITY
AKS ALLSOURCE
MRB**FLIGHT CONTROL COMPARTMENT****D633A109-AKS
53-862-00-01****Page 8 of 8
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT - AFT OF CONTROL COMPARTMENT TO FWD ENTRY DOOR			BOEING CARD NO. 53-864-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLICABILITY	
STATION	SKILL AIRPL	1.2	36 MO	36 MO	AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 221 222	

Perform an external zonal inspection (GV) of the passenger compartment - aft of the control compartment to forward entry door - left and right - section 41, sta 270 to sta 360.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT - AFT OF CONTROL COMPARTMENT TO FWD ENTRY DOOR
		D633A109-AKS 53-864-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-864-00-01
TASK 05-41-02-210-804				MECH INSP

**1. EXTERNAL - ZONAL (GV): PASSENGER COMPARTMENT - AFT OF CONTROL
COMPARTMENT TO FWD ENTRY DOOR**

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-02-210-004

- (1) Do a General Visual inspection of the passenger compartment - aft of the control compartment to forward entry door - left and right - section 41, sta 270 to sta 360.

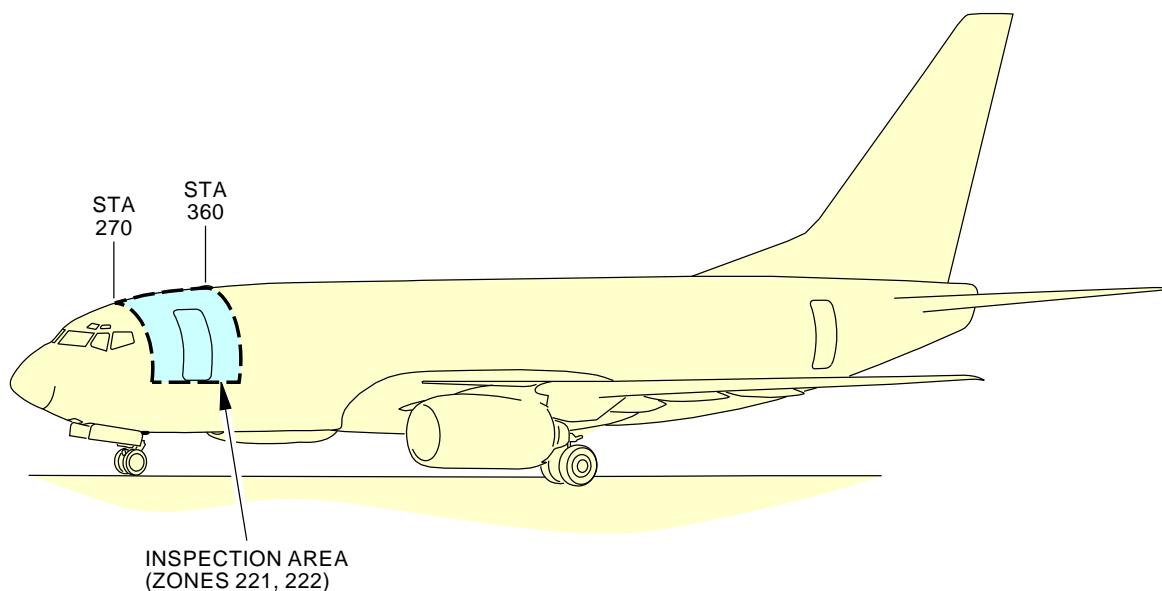
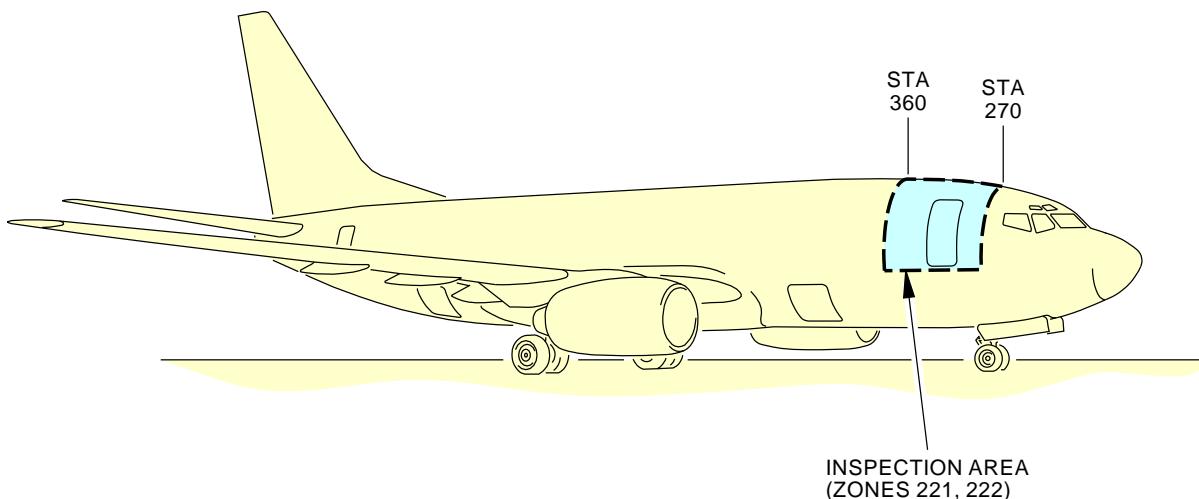
— END OF TASK —

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT - AFT OF CONTROL COMPARTMENT TO FWD ENTRY DOOR
		D633A109-AKS 53-864-00-01

**Page 2 of 3
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-864-00-01



K67409 S0006584126_V3
Passenger Compartment (Sta 270-360) General Visual (External)
Figure 1

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT - AFT OF CONTROL COMPARTMENT TO FWD ENTRY DOOR
		D633A109-AKS 53-864-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT - SECTION 41, STA 270 TO STA 360			BOEING CARD NO. 53-866-00-01		
DATE	TASK ZONAL (GV)				RELATED CARD		
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 4800 FC 24 MO	REPEAT 4800 FC 24 MO	APPLICABILITY	AIRPLANE ALL	ENGINE ALL
STATION	SKILL AIRPL	ACCESS			ZONE	221 222	

Perform an internal zonal inspection (GV) of the passenger compartment - left and right - Section 41, STA 270 to STA 360.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT - SECTION 41, STA 270 TO STA 360
		D633A109-AKS 53-866-00-01

AKS



**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-866-00-01	MECH	INSP
TASK 05-41-02-210-805						
1. INTERNAL - ZONAL (GV): PASSENGER COMPARTMENT - SECTION 41, STA 270 TO STA 360						
(Figure 1)						
A. Zonal Inspection						
SUBTASK 05-41-02-210-005						
(1) Do a General Visual inspection of the passenger compartment - left and right - Section 41, STA 270 to STA 360.						
———— END OF TASK ————						
EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT - SECTION 41, STA 270 TO STA 360				
		D633A109-AKS 53-866-00-01				
		Page 2 of 4 Feb 15/2015				

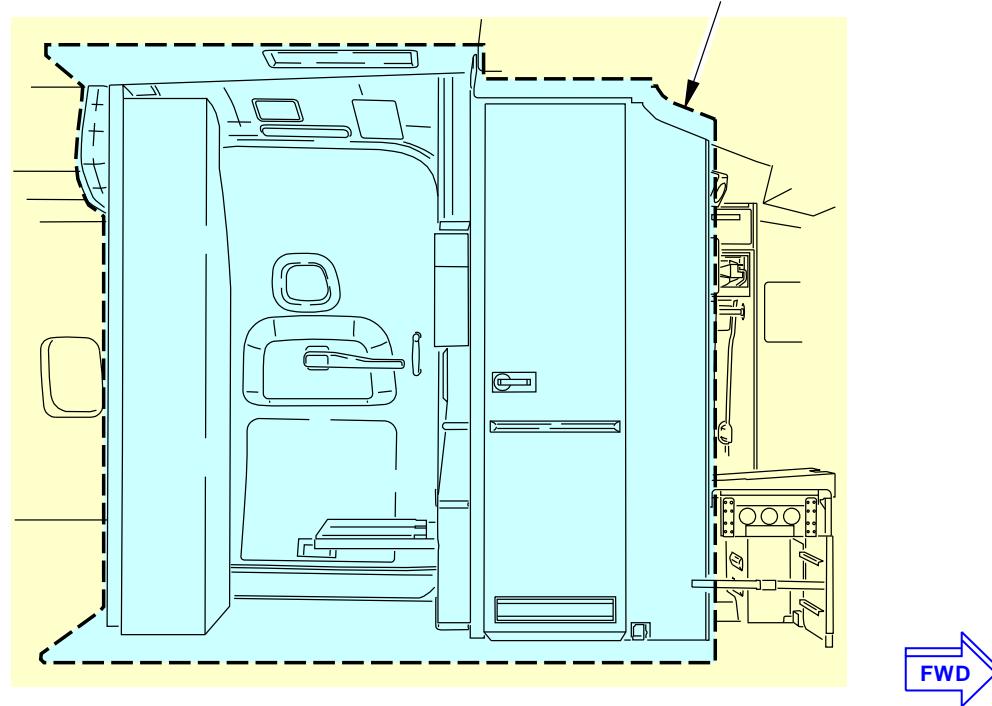
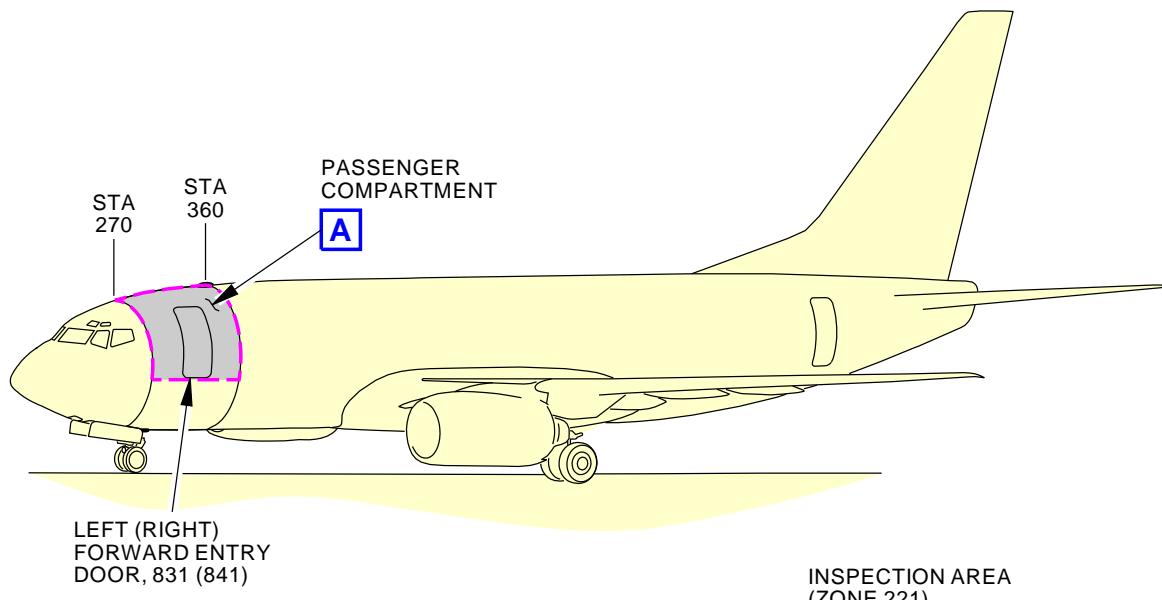
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-866-00-01

Passenger Compartment (Sta 270-360) General Visual (Internal)
Figure 1 (Sheet 1 of 2)

K99232 S0006584128_V3

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT - SECTION 41, STA 270 TO STA 360
		D633A109-AKS 53-866-00-01

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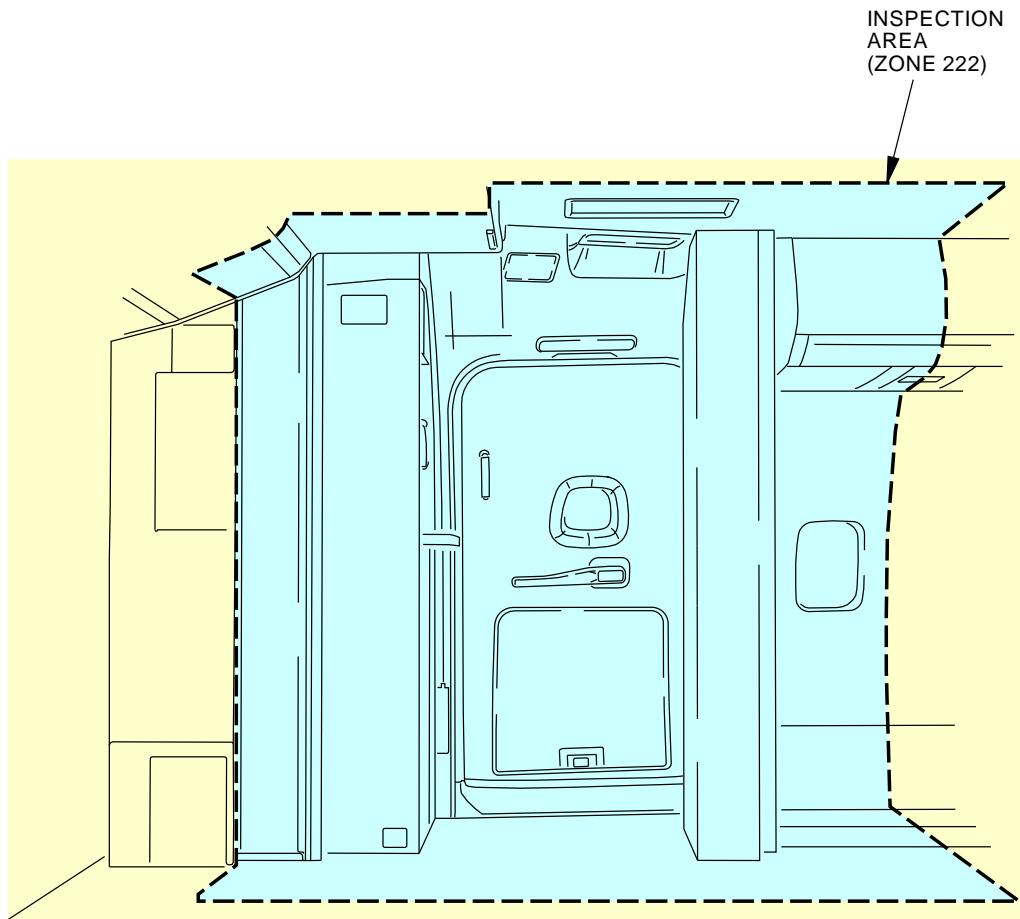
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-866-00-01**PASSENGER COMPARTMENT
(RIGHT SIDE)****A**

K99528 S0006584129_V3

**Passenger Compartment (Sta 270-360) General Visual (Internal)
Figure 1 (Sheet 2 of 2)****EFFECTIVITY
AKS ALL****SOURCE
MRB****PASSENGER COMPARTMENT - SECTION 41, STA 270 TO STA
360****D633A109-AKS
53-866-00-01****Page 4 of 4
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE PASSENGER COMPARTMENT - AFT OF CONTROL COMPARTMENT TO FWD ENTRY DOOR			BOEING CARD NO. 53-868-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 36000 FC 9 YR	REPEAT 36000 FC 8 YR	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL				ENGINE	ALL
		ACCESS NOTE			ZONE	221 222

Perform an internal zonal inspection (GV) of the passenger compartment - aft of the control compartment to forward entry door - left and right - Section 41, Sta 270 to Sta 360. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: With access provided. Galleys and lavs removed. Floor panels, sidewall panels, and ceiling panels removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT - AFT OF CONTROL COMPARTMENT TO FWD ENTRY DOOR
		D633A109-AKS 53-868-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-868-00-01
				MECH INSP
► EWIS TASK 05-41-02-210-806				
1. INTERNAL - ZONAL (GV): Passenger Compartment - Aft of Control Compartment to Fwd Entry Door (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection <u>NOTE:</u> With access provided. Galleys and lavs removed. Floor panels, sidewall panels, and ceiling panels removal required.				
SUBTASK 05-41-02-210-006 (1) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (2) Perform an internal zonal inspection (GV) of the passenger compartment - aft of the control compartment to forward entry door - left and right - Section 41, Sta 270 to Sta 360. (EZAP)				
SUBTASK 05-41-02-910-001 (3) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
———— END OF TASK ——				

EFFECTIVITY AKS ALL	SOURCE MRB	PASSENGER COMPARTMENT - AFT OF CONTROL COMPARTMENT TO FWD ENTRY DOOR
		D633A109-AKS 53-868-00-01

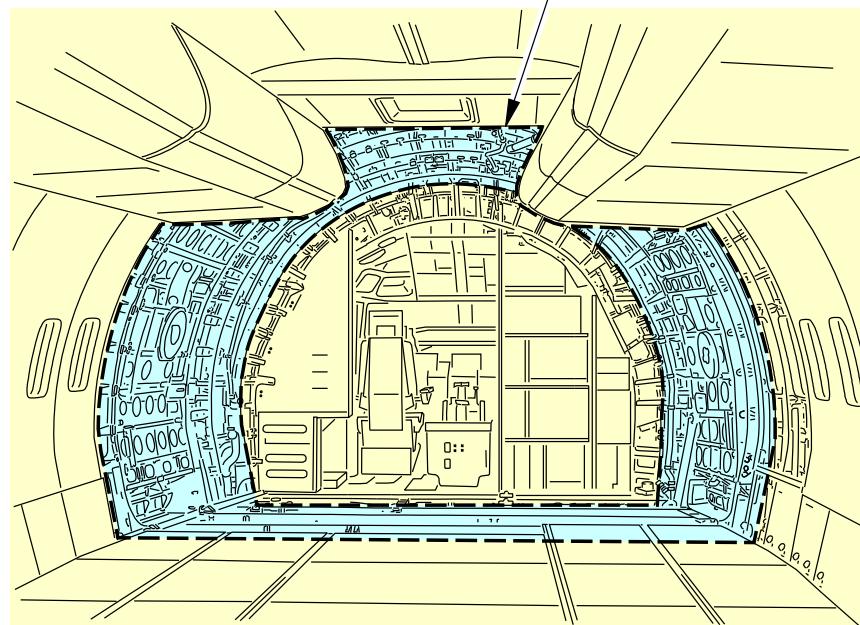
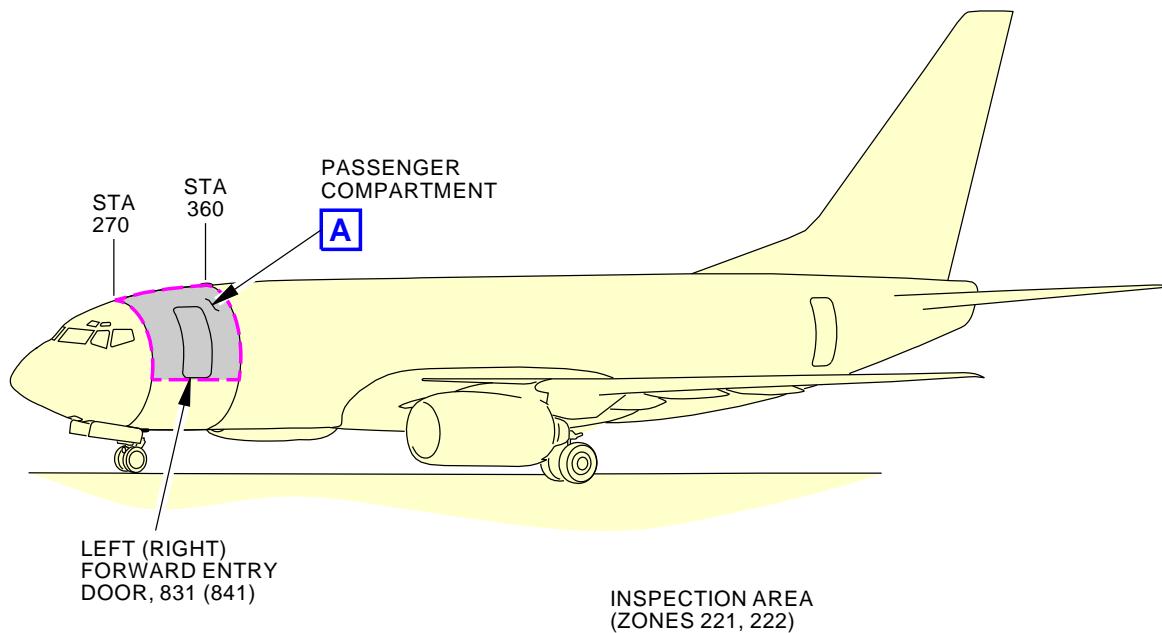
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-868-00-01**PASSENGER COMPARTMENT**

L01220 S0006584131_V4

**Passenger Compartment (Sta 270-360) General Visual (Internal) (Panels Removed)
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**PASSENGER COMPARTMENT - AFT OF CONTROL
COMPARTMENT TO FWD ENTRY DOOR****D633A109-AKS
53-868-00-01****Page 3 of 3
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE FORWARD PASSENGER ENTRY DOOR STOPS, LATCHES AND HINGES			BOEING CARD NO. 53-870-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1	THRESHOLD 2000 FC	REPEAT 2000 FC	APPLICABILITY	
STATION	SKILL AIRPL	1.2	240 DY	240 DY	AIRPLANE ALL	ENGINE ALL
		NOTE 831				ZONE 221

Perform an external zonal inspection (GV) of the forward passenger entry door stops, latches and hinges - section 41, sta 345.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER ENTRY DOOR STOPS, LATCHES AND HINGES
		D633A109-AKS 53-870-00-01

AKS

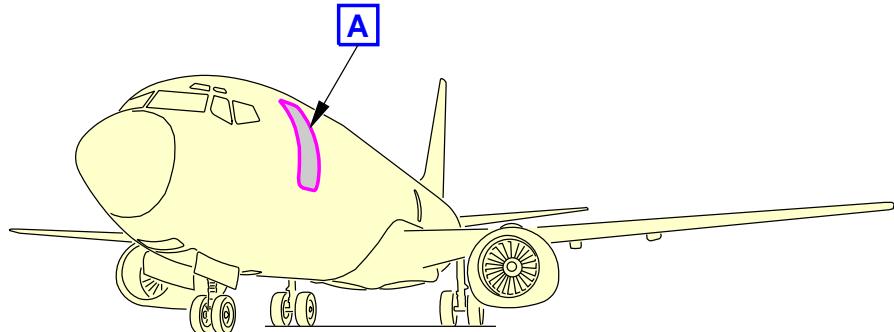
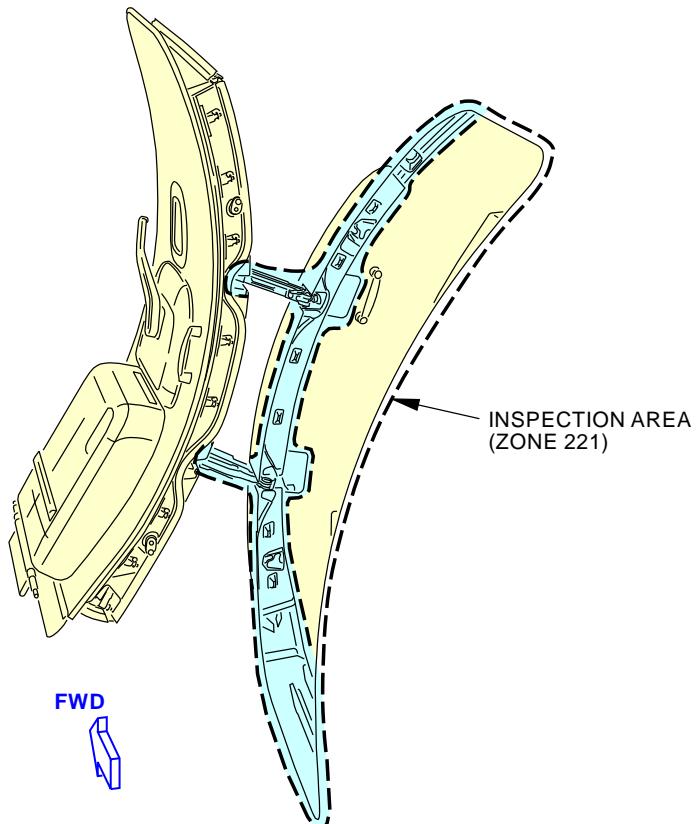


737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-870-00-01	MECH	INSP
TASK 05-41-02-210-807						
1. EXTERNAL - ZONAL (GV): FORWARD PASSENGER ENTRY DOOR STOPS, LATCHES, AND HINGES						
(Figure 1)						
A. Zonal Inspection						
SUBTASK 05-41-02-010-002						
(1) Open this access panel:						
Number Name/Location						
831 Forward Entry Door						
SUBTASK 05-41-02-210-007						
(2) Do a General Visual inspection of the forward passenger entry door stops, latches and hinges - Section 41, Sta 345.						
SUBTASK 05-41-02-410-002						
(3) Close this access panel:						
Number Name/Location						
831 Forward Entry Door						
———— END OF TASK ————						
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER ENTRY DOOR STOPS, LATCHES AND HINGES				
		D633A109-AKS 53-870-00-01				

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-870-00-01
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FORWARD ENTRY
DOOR, 831INSPECTION AREA
(ZONE 221)

FORWARD ENTRY DOOR



K68063 S0006584133_V2

**Door Stops, Latches and Hinges - Forward Entry Door General Visual (External)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER ENTRY DOOR STOPS, LATCHES AND HINGES
		D633A109-AKS 53-870-00-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES			BOEING CARD NO.
DATE	TASK ZONAL (GV)				53-872-00-01 RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 2000 FC 240 DY	REPEAT 2000 FC 240 DY	APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	ACCESS 841			ZONE 222

Perform an external zonal inspection (GV) of the forward galley service door stops, latches, and hinges - section 41, sta 340.

INTERVAL NOTE: Whichever comes first.

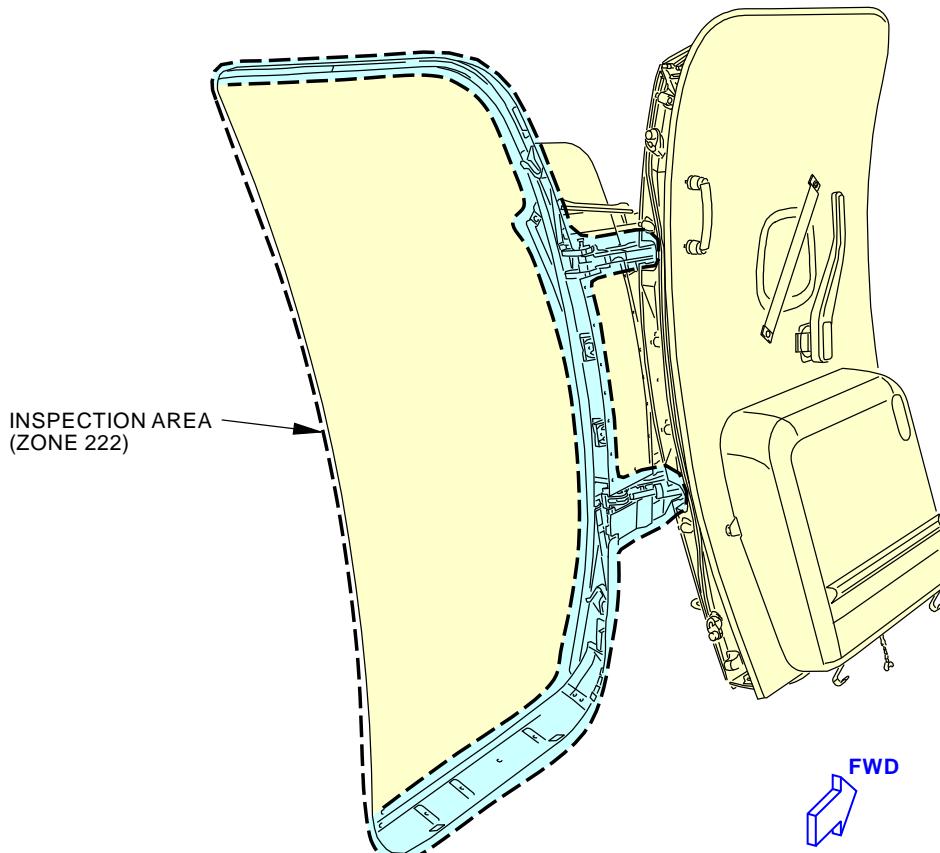
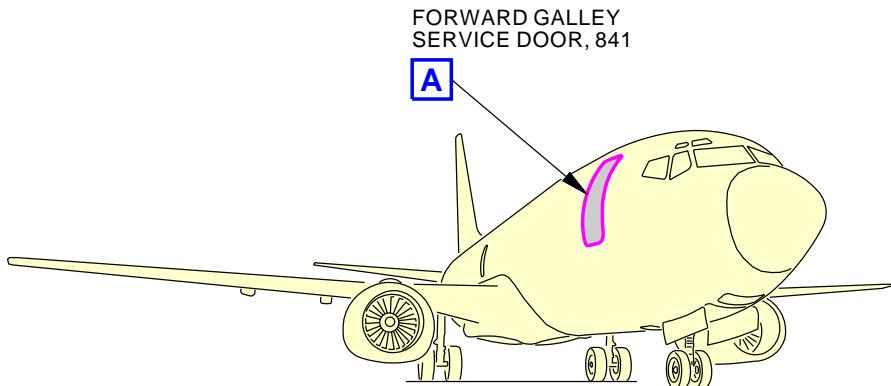
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES D633A109-AKS 53-872-00-01
		Page 1 of 3 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-872-00-01			
				MECH INSP			
TASK 05-41-02-210-808							
1. EXTERNAL - ZONAL (GV): FORWARD GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES							
(Figure 1)							
A. Zonal Inspection							
SUBTASK 05-41-02-010-003							
(1) Open this access panel:							
Number Name/Location							
841 Forward Galley Service Door							
SUBTASK 05-41-02-210-008							
(2) Do a General Visual inspection of the forward galley service door stops, latches, and hinges - Section 41, Sta 340.							
SUBTASK 05-41-02-410-003							
(3) Close this access panel:							
Number Name/Location							
841 Forward Galley Service Door							
— END OF TASK —							
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES					
		D633A109-AKS					
		53-872-00-01					
Page 2 of 3							
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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-872-00-01



Door Stops, Latches and Hinges - Forward Galley Service Door General Visual (External)
Figure 1

K68622 S0006584135_V2

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES
		D633A109-AKS 53-872-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75			BOEING CARD NO. 53-874-00-01		
DATE	TASK ZONAL (GV)				RELATED CARD		
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	AIRPLANE ALL	ENGINE ALL
STATION	SKILL AIRPL	ACCESS			ZONE	231 232	

Perform an external zonal inspection (GV) of the forward passenger compartment - sta 360 to sta 663.75 - left and right - section 43 and 44 (part).

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75	D633A109-AKS 53-874-00-01	Page 1 of 3 Feb 15/2015
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AKS

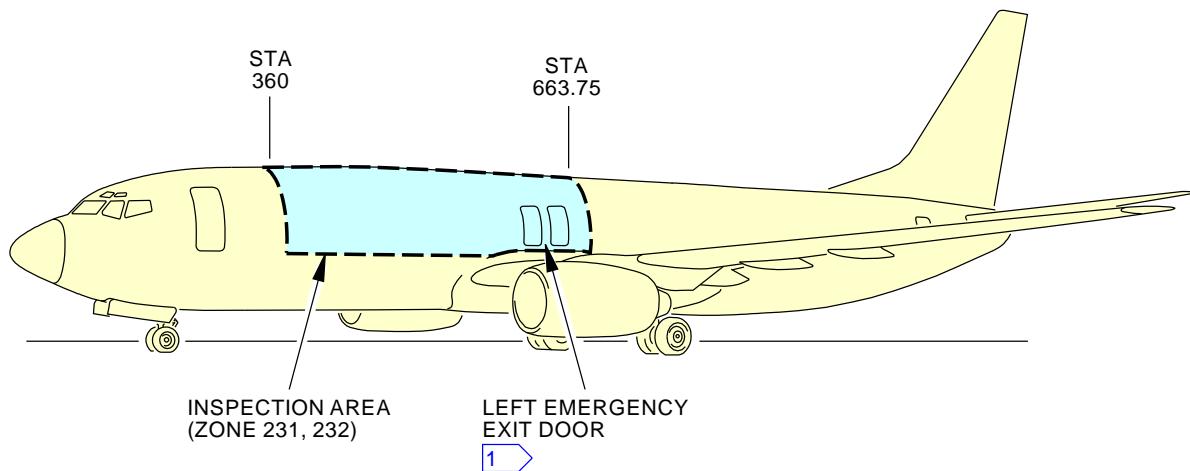
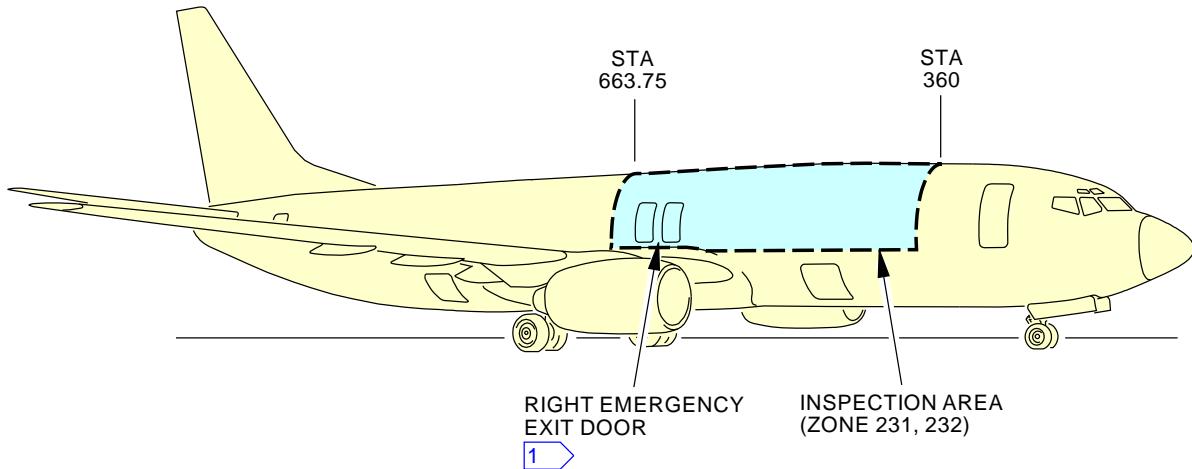


737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-874-00-01	
TASK 05-41-02-210-810					MECH
1. EXTERNAL - ZONAL (GV): FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75 (Figure 1)					INSP
<p>A. Zonal Inspection</p> <p>SUBTASK 05-41-02-210-010</p> <p>(1) Do a General Visual inspection of the forward passenger compartment - Sta 360 to Sta 663.75 - left and right - Section 43 and 44 (part).</p> <p style="text-align: center;">———— END OF TASK ————</p>					
EFFECTIVITY AKS ALL		SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75		
			D633A109-AKS 53-874-00-01		

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-874-00-01
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**NOTE:**

- 1 TWO EMERGENCY EXIT DOORS ON 800 AND 900 ONLY

K76414 S0006584138_V3

**Passenger Compartment (Sta 360-663.75) General Visual (External)
Figure 1**

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75
		D633A109-AKS 53-874-00-01

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Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75			BOEING CARD NO. 53-876-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 4800 FC 24 MO	REPEAT 4800 FC 24 MO	APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	ACCESS			ZONE 231 232

Perform an internal zonal inspection (GV) of the forward passenger compartment - sta 360 to sta 663.75 - left and right - section 43 and 44 (part).

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75
		D633A109-AKS 53-876-00-01

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AKS

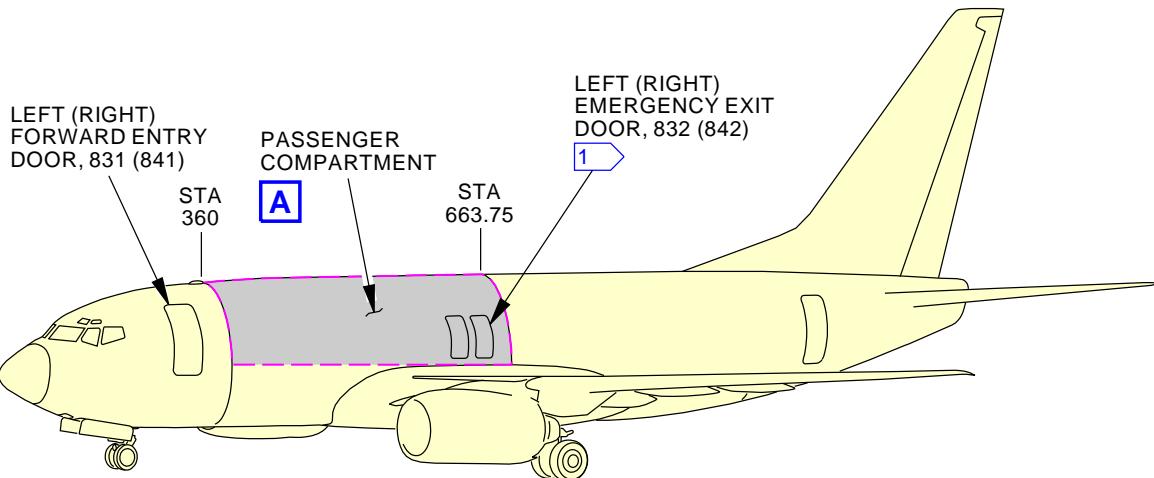


737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-876-00-01
				MECH INSP
TASK 05-41-02-210-811				
1. INTERNAL - ZONAL (GV): FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75				
(Figure 1)				
A. Zonal Inspection				
SUBTASK 05-41-02-210-011				
(1) Do a General Visual inspection of the forward passenger compartment - Sta 360 to Sta 663.75 - left and right - Section 43 and 44 (part).				
———— END OF TASK ——				
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75		
		D633A109-AKS 53-876-00-01		

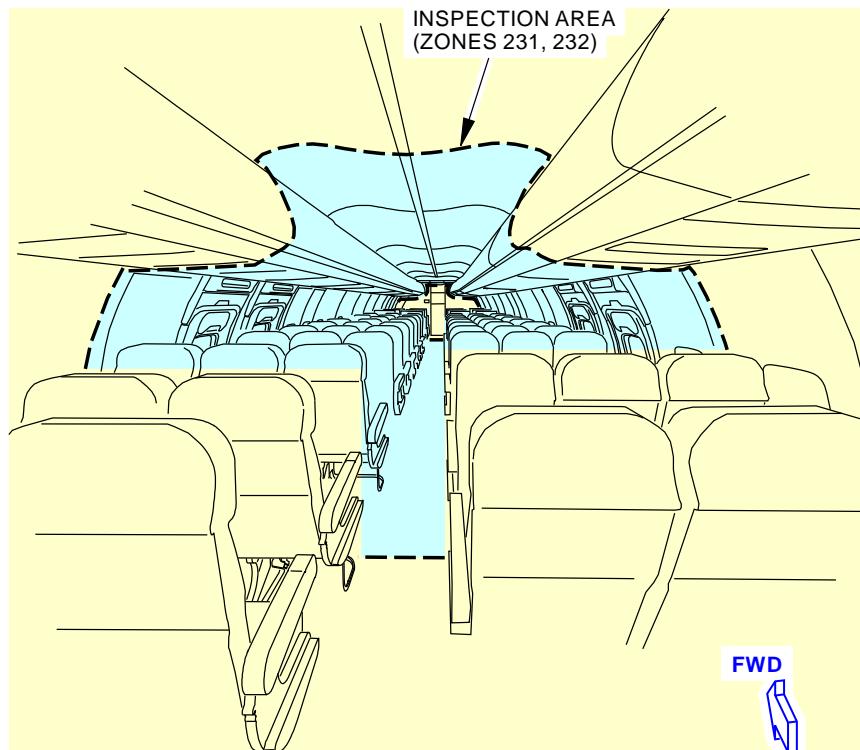
AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-876-00-01



NOTE:

TWO EMERGENCY EXIT DOORS ON 800 AND 900 ONLY



Passenger Compartment (Sta 360-663.75) General Visual (Internal)
Figure 1

L03717 S0006584140_V3

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75
		D633A109-AKS 53-876-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75 WET AREAS			BOEING CARD NO. 53-878-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 36000 FC 8 YR	REPEAT 36000 FC 8 YR	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL				ENGINE	ALL
		ACCESS NOTE			ZONE	231 232

Perform an internal zonal inspection (GV) of the forward passenger compartment - Sta 360 to Sta 663.75 wet areas (within approximately 20 inches from galley or lav) - left and right - Section 43 and 44 (part). (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: Galleys and lavs removed. Floor panels, sidewall panels, and ceiling panels removal required in areas where galleys and lavs are located.

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75 WET AREAS	
		D633A109-AKS 53-878-00-01	Page 1 of 3 Oct 15/2014

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-878-00-01
				MECH INSP
► EWIS TASK 05-41-02-210-812				
1. INTERNAL - ZONAL (GV): Forward Passenger Compartment - Sta 360 to Sta 663.75 Wet Areas				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection <u>NOTE:</u> Galleys and lavs removed. Floor panels, sidewall panels, and ceiling panels removal required in areas where galleys and lavs are located.				
SUBTASK 05-41-02-210-012 (1) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (2) Perform an internal zonal inspection (GV) of the forward passenger compartment - Sta 360 to Sta 663.75 wet areas (within approximately 20 inches from galley or lav) - left and right - Section 43 and 44 (part). (EZAP)				
SUBTASK 05-41-02-910-002 (3) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75 WET AREAS D633A109-AKS 53-878-00-01		
				Page 2 of 3 Feb 15/2015

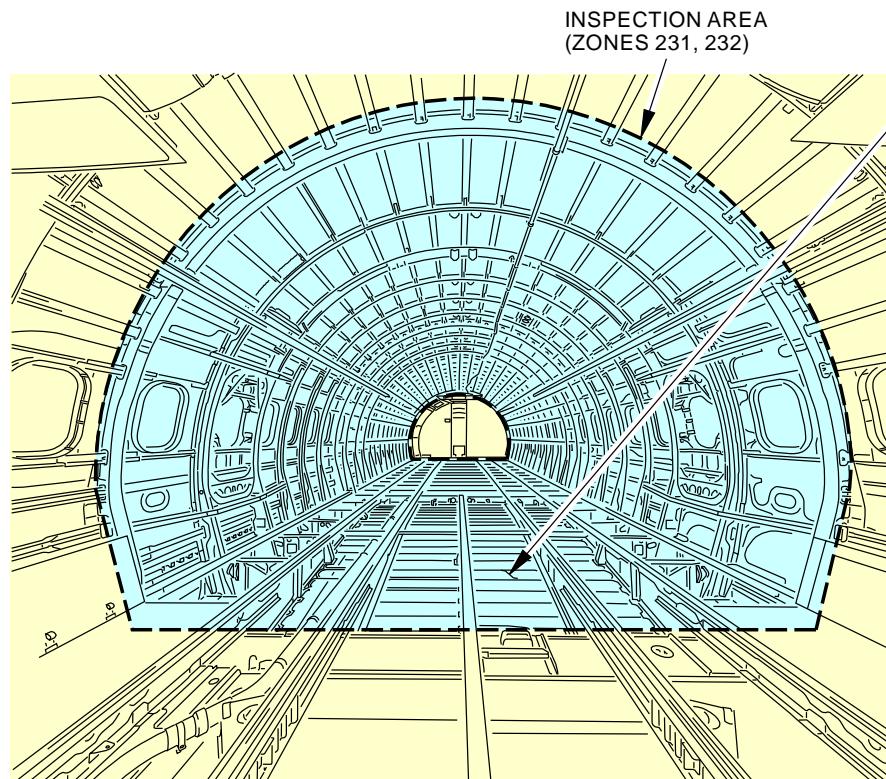
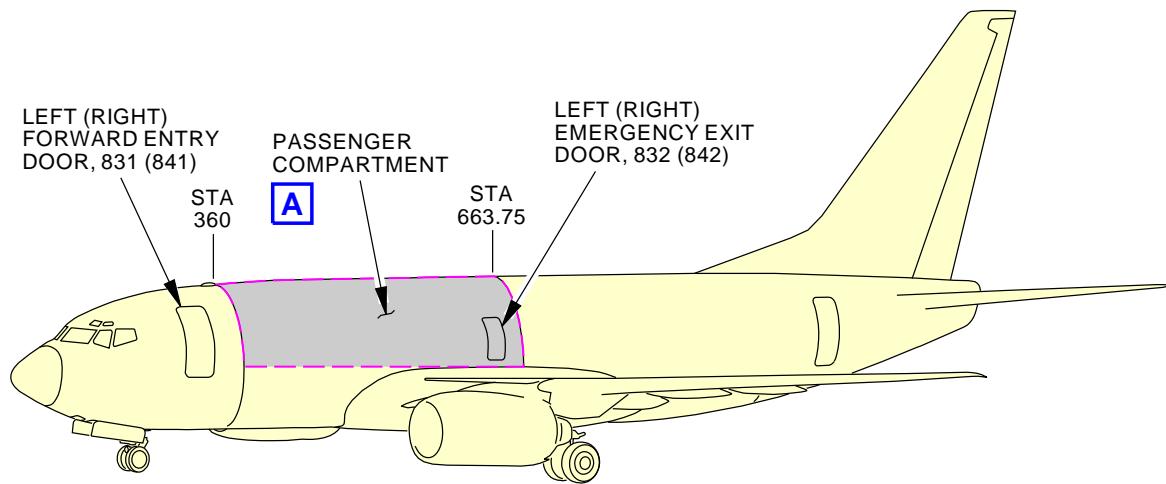
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-878-00-01

2210747 S0000493441_V2

**Passenger Compartment (Sta 360-663.75) General Visual (Internal) (Panels Removed)
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**FORWARD PASSENGER COMPARTMENT - STA 360 TO STA
663.75 WET AREAS****D633A109-AKS
53-878-00-01****Page 3 of 3
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75 DRY AREAS			BOEING CARD NO. 53-880-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 36000 FC 12 YR	REPEAT 36000 FC 12 YR	APPLICABILITY	AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	ACCESS NOTE			ZONE	231 232

Perform an internal zonal inspection (GV) of the forward passenger compartment - Sta 360 to Sta 663.75 dry area (away from doors, galleys, and lavs) - left and right - Section 43 and 44 (part). (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: Floor panels, sidewall panels, and ceiling panels removal required.

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75 DRY AREAS
		D633A109-AKS 53-880-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-880-00-01
				MECH INSP
► EWIS TASK 05-41-02-210-813				
1. INTERNAL - ZONAL (GV): Forward Passenger Compartment - Sta 360 to Sta 663.75 Dry Areas (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection <u>NOTE:</u> Floor panels, sidewall panels, and ceiling panels removal required. SUBTASK 05-41-02-210-013 (1) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (2) Perform an internal zonal inspection (GV) of the forward passenger compartment - Sta 360 to Sta 663.75 dry area (away from doors, galleys, and lavs) - left and right - Section 43 and 44 (part). (EZAP) SUBTASK 05-41-02-910-003 (3) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
— END OF TASK —				
EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75 DRY AREAS D633A109-AKS 53-880-00-01		
				Page 2 of 3 Feb 15/2015

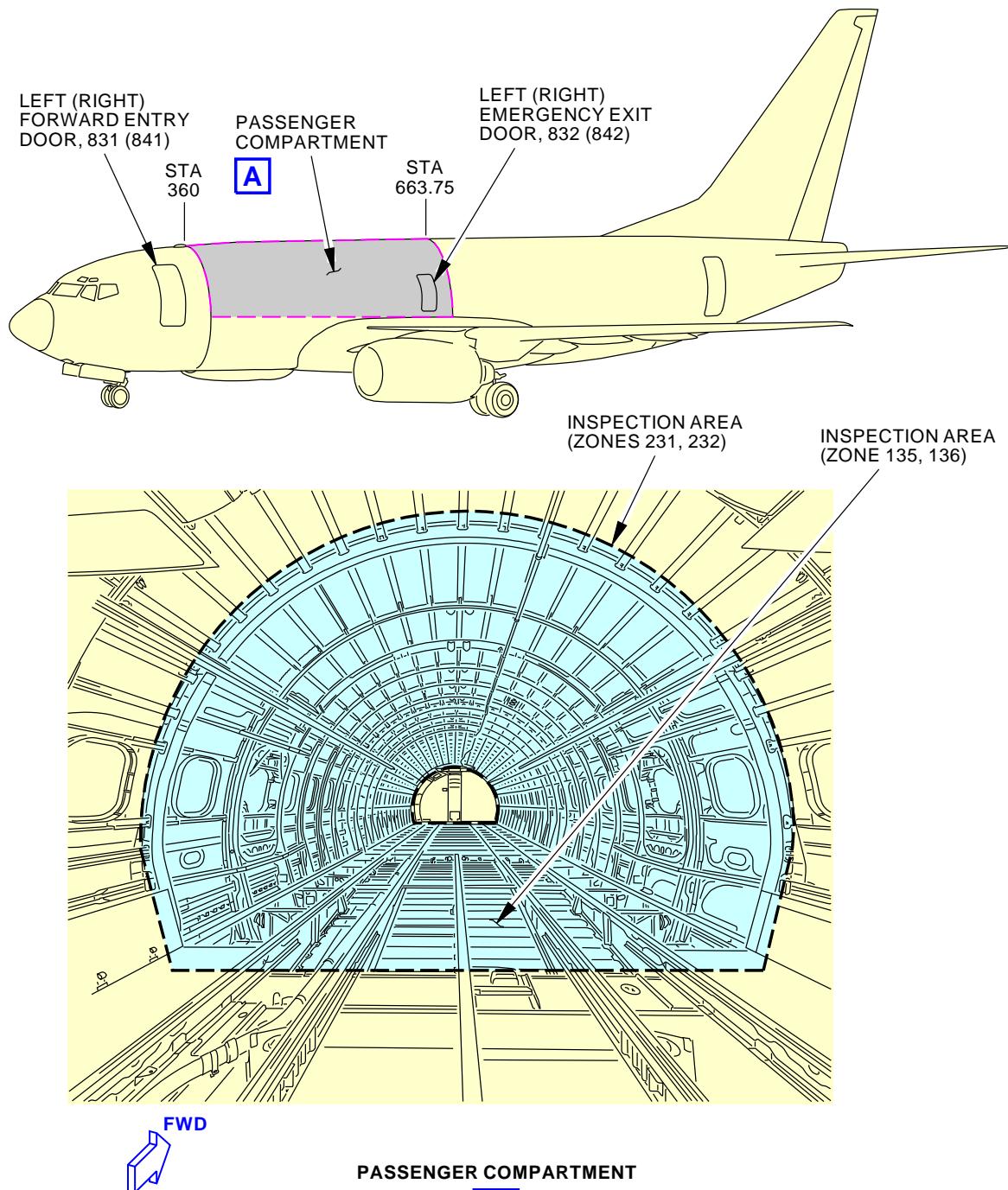
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-880-00-01

Passenger Compartment (Sta 360-663.75) General Visual (Internal) (Panels Removed)
Figure 1

L03458 S0006584143_V2

EFFECTIVITY AKS ALL	SOURCE MRB	FORWARD PASSENGER COMPARTMENT - STA 360 TO STA 663.75 DRY AREAS
		D633A109-AKS 53-880-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD			BOEING CARD NO. 53-882-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL				ENGINE	ALL
		ACCESS			ZONE	241 242

Perform an external zonal inspection (gv) of the aft passenger compartment - sta 663.75 to the aft pressure bulkhead - left and right - section 44 (part), 46, and 47, sta 663.75 to the aft pressure bulkhead.

NOTE: Not applicable to airplanes with flat aft pressure bulkhead.

INTERVAL NOTE: Whichever comes first.

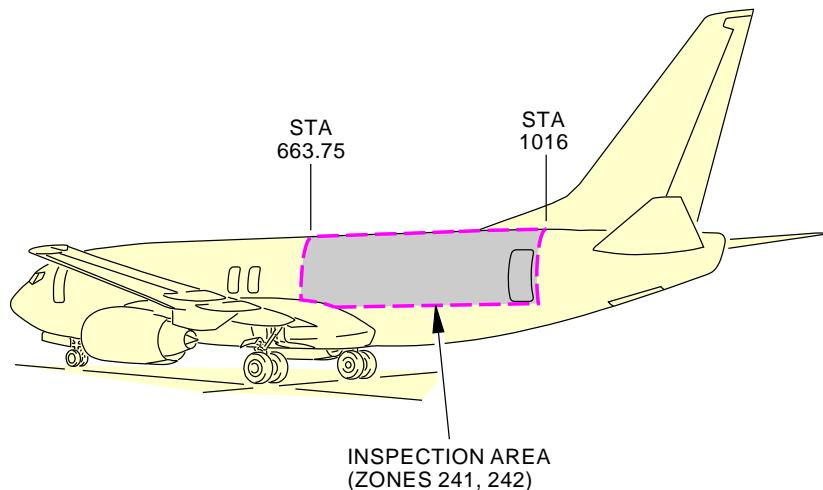
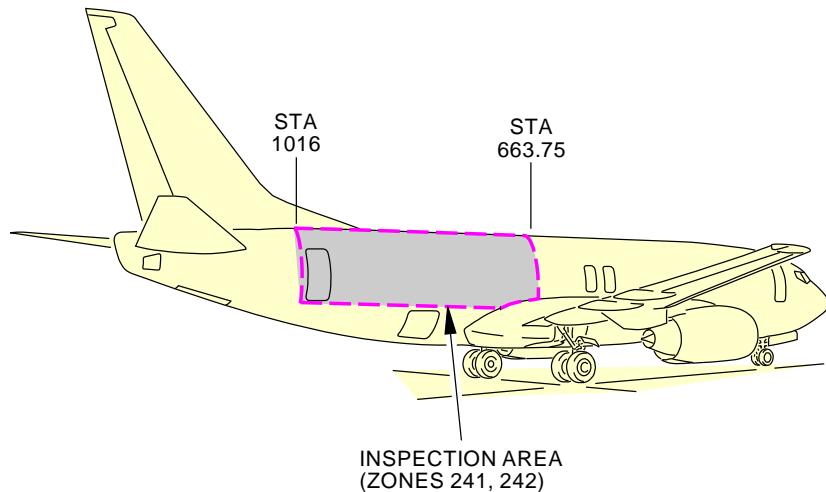
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD
		D633A109-AKS 53-882-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-882-00-01
				MECH INSP
TASK 05-41-02-210-814				
1. EXTERNAL - ZONAL (GV): AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD				
(Figure 1)				
A. Zonal Inspection				
SUBTASK 05-41-02-210-014				
(1) Do a General Visual inspection of the aft passenger compartment - Sta 663.75 to the aft pressure bulkhead - left and right - Section 44 (part), 46, and 47, Sta 663.75 to the aft pressure bulkhead.				
— END OF TASK —				
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD				SOURCE MRB
AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD				D633A109-AKS 53-882-00-01
				Page 2 of 3 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-882-00-01



K76665 S0006584145_V2
Passenger Compartment (Sta 663.75-1042) General Visual (External)
Figure 1

EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD
		D633A109-AKS 53-882-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD			BOEING CARD NO. 53-884-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 4800 FC 24 MO	REPEAT 4800 FC 24 MO	APPLICABILITY	AIRPLANE ALL
STATION	SKILL AIRPL				ENGINE	ALL
		ACCESS			ZONE	241 242

Perform an internal zonal inspection (GV) of the aft passenger compartment - STA 663.75 to the aft pressure bulkhead - left and right- section 44 (part), 46 and 47, STA 663.75 to aft pressure bulkhead.

NOTE: Not applicable to airplanes with flat aft pressure bulkhead.

INTERVAL NOTE: Whichever comes first.

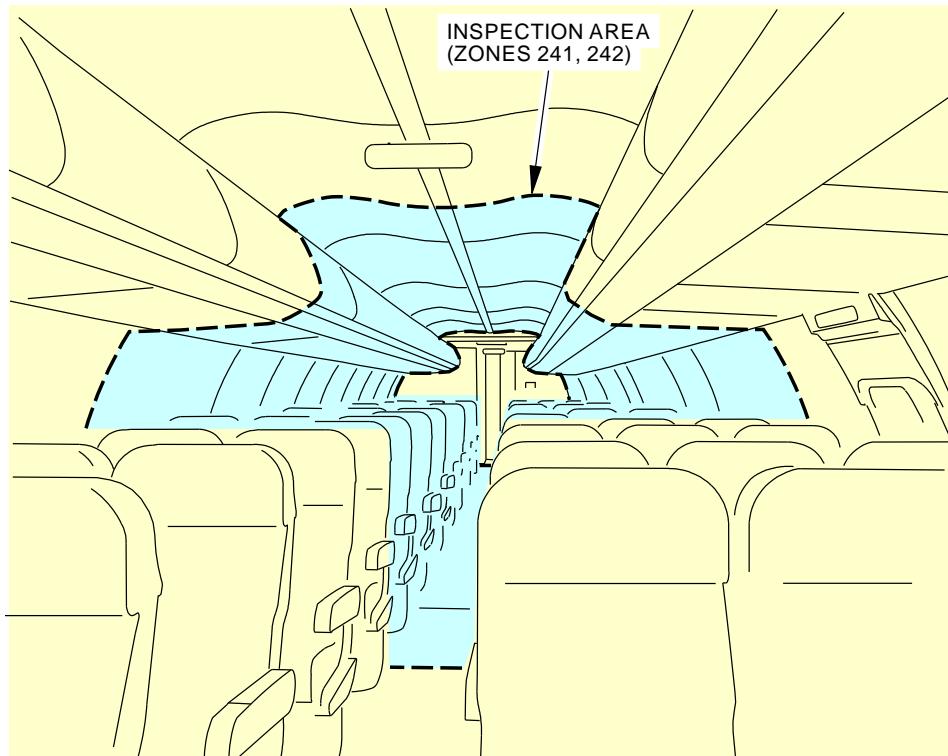
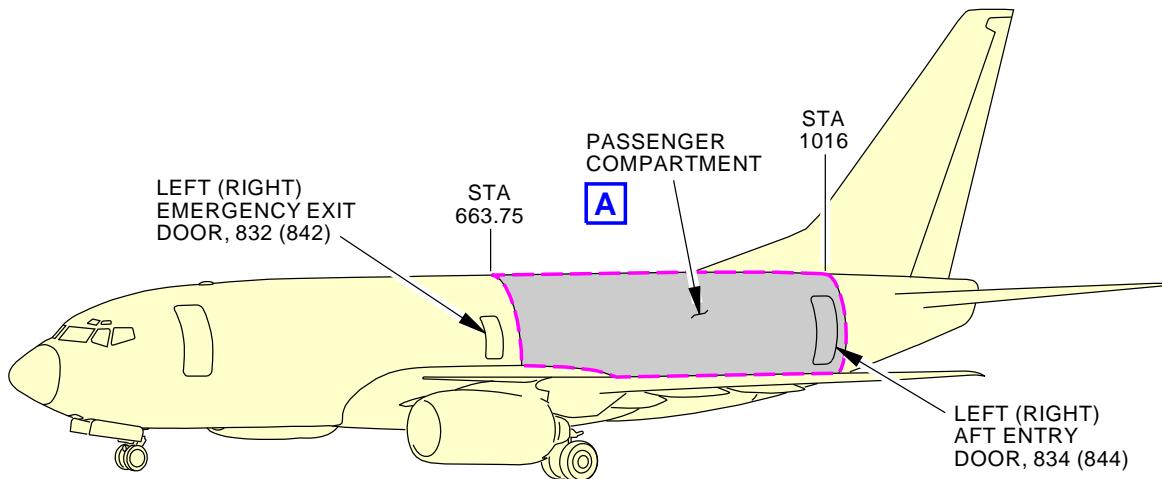
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD
		D633A109-AKS 53-884-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-884-00-01
				MECH INSP
TASK 05-41-02-210-815				
1. INTERNAL - ZONAL (GV): AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD				
(Figure 1)				
A. Zonal Inspection				
SUBTASK 05-41-02-210-015				
(1) Do a General Visual inspection of the aft passenger compartment - Sta 663.75 to the aft pressure bulkhead - left and right- Section 44 (part), 46 and 47, Sta 663.75 to aft pressure bulkhead.				
— END OF TASK —				
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD		SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD	
			D633A109-AKS 53-884-00-01	
				Page 2 of 3 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-884-00-01



PASSENGER COMPARTMENT



L04277 S0006584147_V2

**Passenger Compartment (Sta 663.75-1016) General Visual (Internal)
Figure 1**

EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD
		D633A109-AKS 53-884-00-01

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD WET AREA			BOEING CARD NO. 53-886-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 36000 FC 8 YR	REPEAT 36000 FC 8 YR	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL				ENGINE ALL
		ACCESS NOTE			ZONE 241 242

Perform an internal zonal inspection (GV) of the aft passenger compartment - Sta 663.75 to the aft pressure bulkhead wet area (within approximately 20 inches from galley service door, passenger door, galley or lav) - left and right - Section 44 (part), 46, and 47, Sta 663.75 to aft pressure bulkhead. (EZAP)

NOTE: Not applicable to airplanes with flat aft pressure bulkhead.

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: Galleys and lavs removed. Floor panels, sidewall panels, and ceiling panels removal required in areas where galleys and lavs are located.

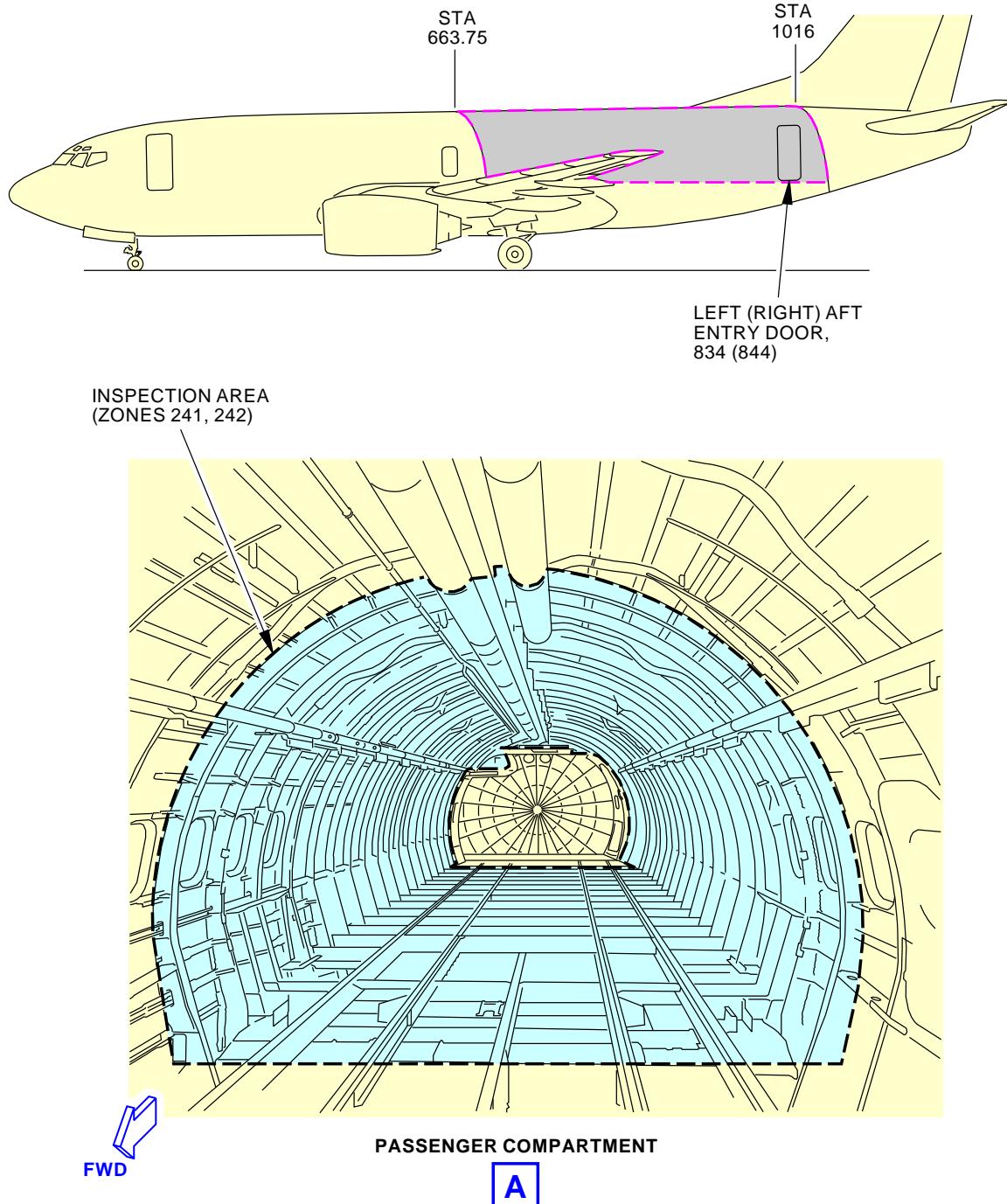
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD WET AREA
		D633A109-AKS 53-886-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-886-00-01
				MECH INSP
EWIS TASK 05-41-02-210-816				
1. INTERNAL - ZONAL (GV): Aft Passenger Compartment - Sta 663.75 to Aft Pressure Bulkhead Wet Area (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection <u>NOTE:</u> Galleys and lavs removed. Floor panels, sidewall panels, and ceiling panels removal required in areas where galleys and lavs are located.				
SUBTASK 05-41-02-210-016 (1) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (2) Perform an internal zonal inspection (GV) of the aft passenger compartment - Sta 663.75 to the aft pressure bulkhead wet area (within approximately 20 inches from galley service door, passenger door, galley or lav) - left and right - Section 44 (part), 46, and 47, Sta 663.75 to aft pressure bulkhead. (EZAP)				
SUBTASK 05-41-02-910-004 (3) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance".				
— END OF TASK —				
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD		SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD WET AREA D633A109-AKS 53-886-00-01	
				Page 2 of 3 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-886-00-01



L05805 S0006584149_V3

**Passenger Compartment (Sta 663.75-1016) General Visual (Internal) (Galley, Lavatory and Panels Removed)
Figure 1**

EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD WET AREA
		D633A109-AKS 53-886-00-01

AKS
**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD DRY AREA			BOEING CARD NO. 53-888-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 36000 FC 12 YR	REPEAT 36000 FC 12 YR	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS NOTE			ZONE 241 242	

Perform an internal zonal inspection (GV) of the aft passenger compartment - Sta 663.75 to the aft pressure bulkhead dry area (away from doors, galleys and lavs) - left and right - Section 44 (part), 46, and 47, Sta 663.75 to aft pressure bulkhead. (EZAP)

NOTE: Not applicable to airplanes with flat aft pressure bulkhead.

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

ACCESS NOTE: Floor panels, sidewall panels, and ceiling panels removal required.

EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD DRY AREA
		D633A109-AKS 53-888-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-888-00-01
				MECH INSP
► EWIS TASK 05-41-02-210-817				
1. INTERNAL - ZONAL (GV): Passenger Compartment - Sta 663.75 to Aft Pressure Bulkhead Dry Area (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection <u>NOTE:</u> Floor panels, sidewall panels, and ceiling panels removal required. SUBTASK 05-41-02-210-031 (1) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (2) Perform an internal zonal inspection (GV) of the aft passenger compartment - Sta 663.75 to the aft pressure bulkhead dry area (away from doors, galleys and lavs) - left and right - Section 44 (part), 46, and 47, Sta 663.75 to aft pressure bulkhead. (EZAP) SUBTASK 05-41-02-910-006 (3) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
— END OF TASK —				
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT PRESSURE BULKHEAD DRY AREA D633A109-AKS 53-888-00-01		
				Page 2 of 3 Feb 15/2015

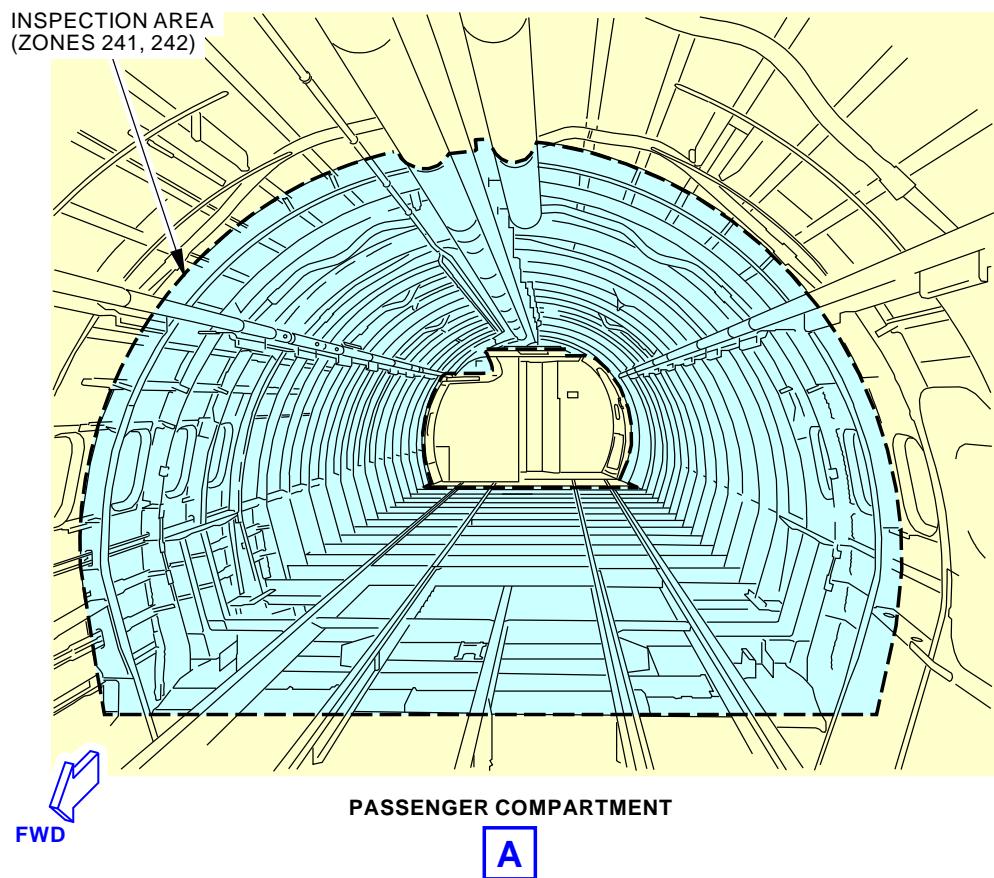
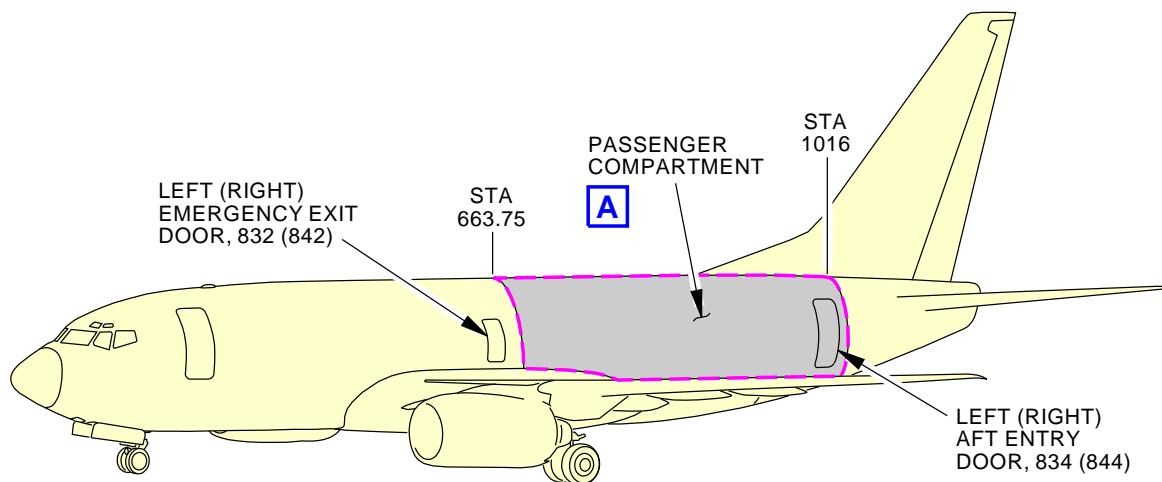
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-888-00-01

L05822 S0006584151_V2

**Passenger Compartment (Sta 663.75-1016) General Visual (Internal) (Panels Removed)
Figure 1****EFFECTIVITY
AKS ALL; AIRPLANES WITH A CURVED AFT
PRESSURE BULKHEAD****SOURCE
MRB****AFT PASSENGER COMPARTMENT - STA 663.75 TO AFT
PRESSURE BULKHEAD DRY AREA****D633A109-AKS
53-888-00-01****Page 3 of 3
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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT PASSENGER ENTRY DOOR STOPS, LATCHES, AND HINGES			BOEING CARD NO. 53-890-00-01		
DATE	TASK ZONAL (GV)				RELATED CARD		
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 2000 FC 240 DY	REPEAT 2000 FC 240 DY	APPLICABILITY	AIRPLANE ALL	ENGINE ALL
STATION	SKILL AIRPL	ACCESS 834				ZONE 241	

Perform an external zonal inspection (GV) of the aft passenger entry door stops, latches, and hinges - section 47, sta 980.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	AFT PASSENGER ENTRY DOOR STOPS, LATCHES, AND HINGES	
		D633A109-AKS 53-890-00-01	Page 1 of 3 Feb 15/2015

AKS



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-890-00-01
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TASK 05-41-02-210-818

1. EXTERNAL - ZONAL (GV): AFT PASSENGER ENTRY DOOR STOPS, LATCHES, AND HINGES
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-02-010-005

- (1) Open this access panel:

<u>Number</u>	<u>Name/Location</u>
---------------	----------------------

834 Aft Entry Door

SUBTASK 05-41-02-210-018

- (2) Do a General Visual inspection of the aft passenger entry door stops, latches, and hinges
- Section 47, Sta 980.

SUBTASK 05-41-02-410-005

- (3) Close this access panel:

Number Name/Location

834 Aft Entry Door

— END OF TASK —

EFFECTIVITY
AKS ALL

SOURCE
MRB

AFT PASSENGER ENTRY DOOR STOPS, LATCHES, AND HINGES

D633A109-AKS
53-890-00-01

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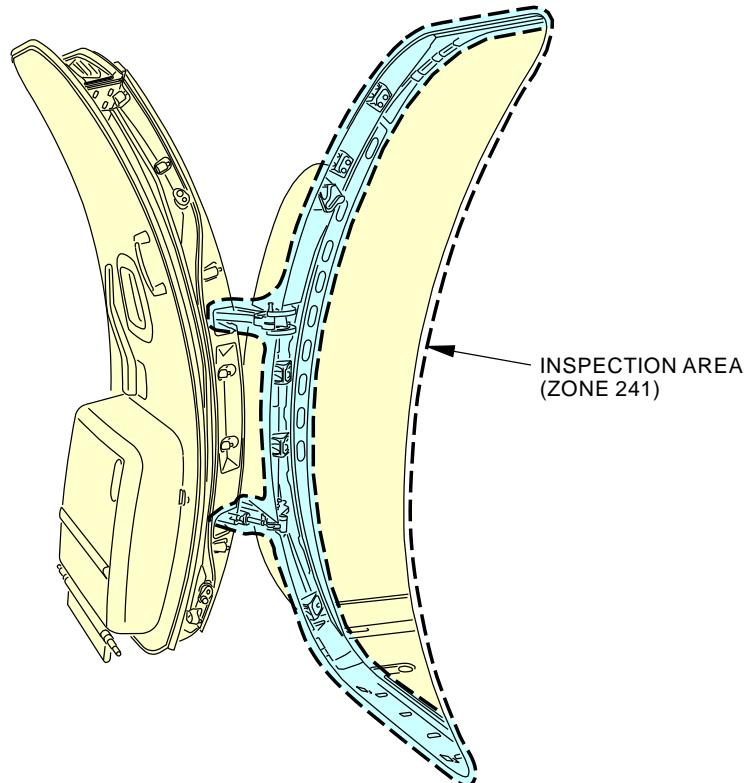
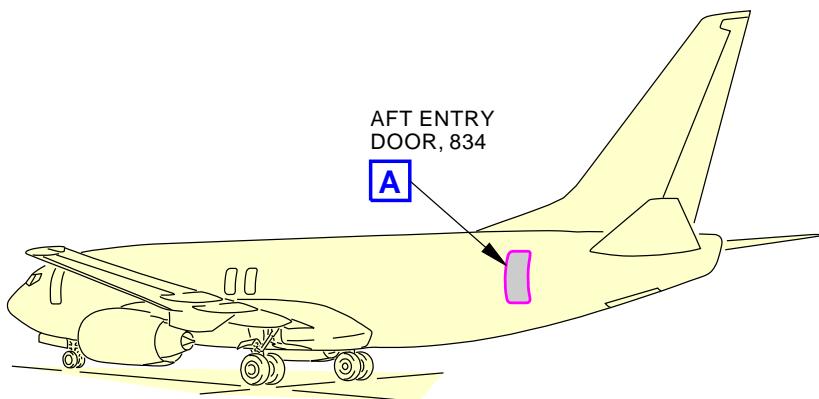
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-890-00-01**AFT ENTRY DOOR**

K68960 S0006584153_V2

**Door Stops, Latches and Hinges - Aft Entry Door General Visual (External)
Figure 1****EFFECTIVITY
AKS ALL****SOURCE
MRB****AFT PASSENGER ENTRY DOOR STOPS, LATCHES, AND
HINGES****D633A109-AKS
53-890-00-01****Page 3 of 3
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AFT GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES			BOEING CARD NO. 53-892-00-01		
DATE	TASK ZONAL (GV)				RELATED CARD		
TAIL NUMBER	WORK AREA FUSELAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 2000 FC 240 DY	REPEAT 2000 FC 240 DY	APPLICABILITY	AIRPLANE ALL	ENGINE ALL
STATION	SKILL AIRPL	ACCESS 844				ZONE 242	

Perform an external zonal inspection (GV) of the aft galley service door stops, latches, and hinges - section 47, sta 980.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY AKS ALL	SOURCE MRB	AFT GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES
		D633A109-AKS 53-892-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-892-00-01
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TASK 05-41-02-210-819

MECH

INSP

1. EXTERNAL - ZONAL (GV): AFT GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-02-010-006

- (1) Open this access panel:

Number Name/Location

844 Aft Galley Service Door

SUBTASK 05-41-02-210-019

- (2) Do a General Visual inspection of the aft galley service door stops, latches, and hinges - Section 47, Sta 980.

SUBTASK 05-41-02-410-006

- (3) Close this access panel:

Number Name/Location

844 Aft Galley Service Door

———— END OF TASK ————

EFFECTIVITY AKS ALL	SOURCE MRB	AFT GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES
		D633A109-AKS 53-892-00-01

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Feb 15/2015

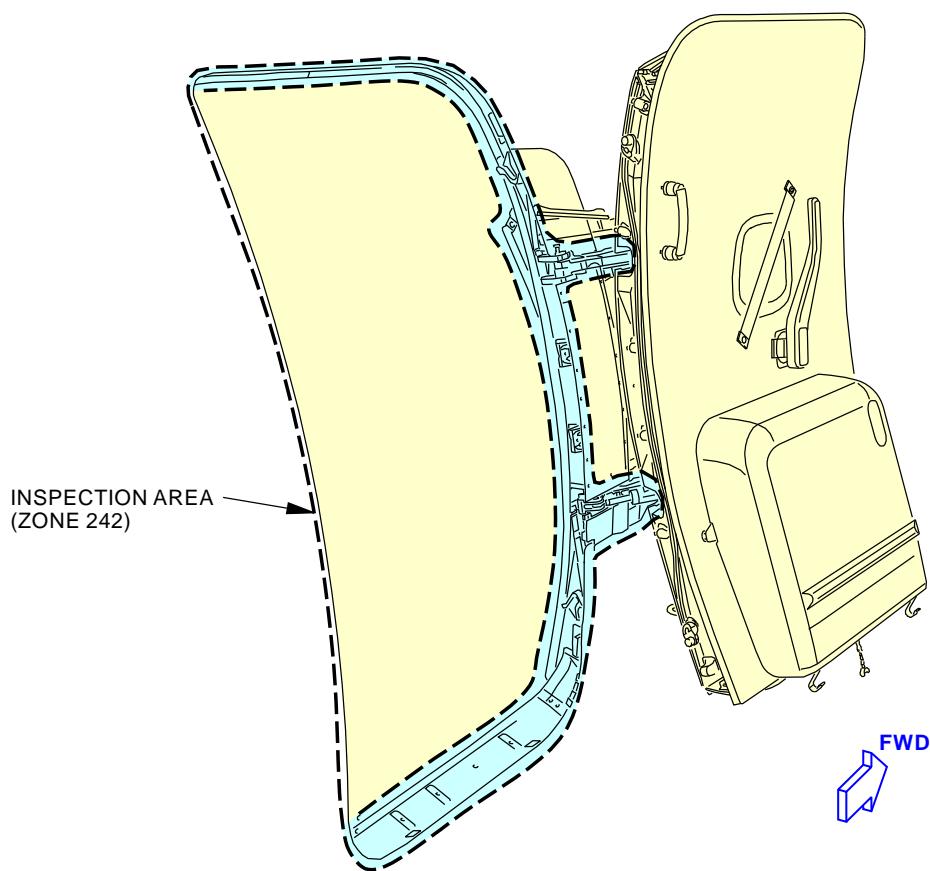
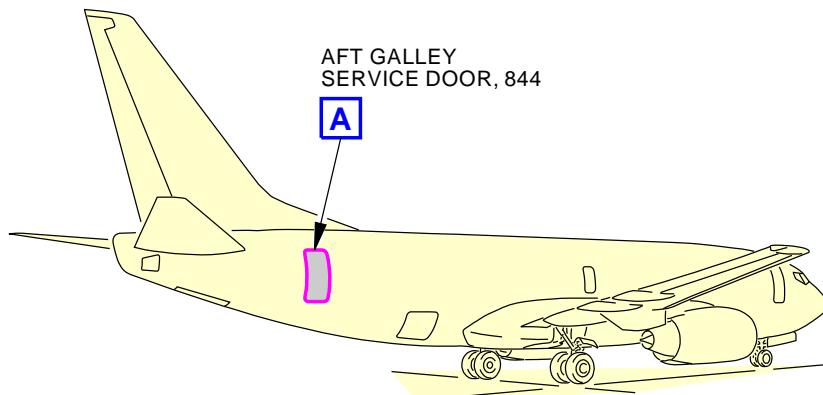
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-892-00-01**AFT GALLEY SERVICE DOOR****A**

K68993 S0006584155_V2

**Door Stops, Latches and Hinges - Aft Galley Service Door General Visual (External)
Figure 1**EFFECTIVITY
AKS ALLSOURCE
MRB**AFT GALLEY SERVICE DOOR STOPS, LATCHES, AND HINGES****D633A109-AKS
53-892-00-01****Page 3 of 3
Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA AFT OF THE PRESSURE BULKHEAD			BOEING CARD NO. 53-894-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 120 DY	REPEAT 120 DY	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS			ZONE 310	

Perform an external zonal inspection (GV) of the area aft of the pressure bulkhead. Inspection is accomplished from the ground, without the use of stands or ladders. No additional access panels required.

NOTE: Not applicable to airplanes with flat aft pressure bulkhead.

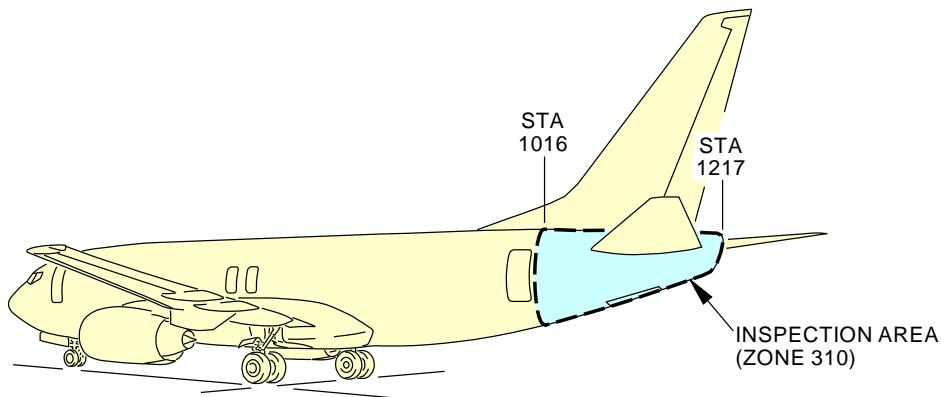
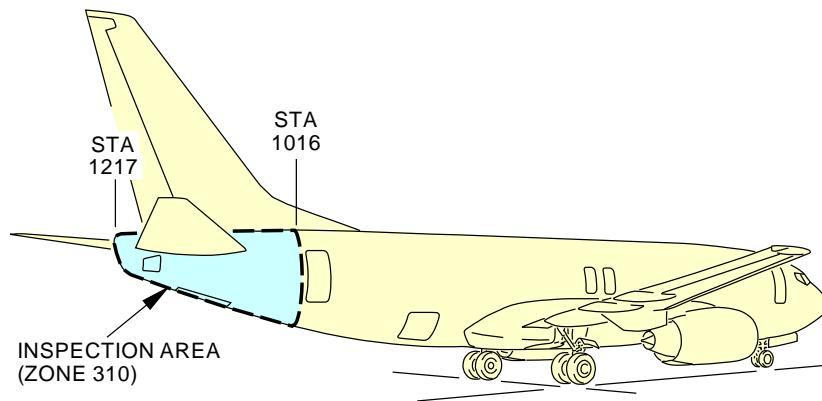
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AREA AFT OF THE PRESSURE BULKHEAD
		D633A109-AKS 53-894-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-894-00-01
TASK 05-41-03-210-801				MECH INSP
1. EXTERNAL - ZONAL (GV): AREA AFT OF THE PRESSURE BULKHEAD (Figure 1)				
A. Zonal Inspection SUBTASK 05-41-03-210-001 (1) Do a General Visual inspection of the area aft of the pressure bulkhead. Inspection is accomplished from the ground, without the use of stands or ladders. No additional access panels required.				
— END OF TASK —				
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AREA AFT OF THE PRESSURE BULKHEAD D633A109-AKS 53-894-00-01		
				Page 2 of 3 Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-894-00-01



K68967 S0006584159_V2
Passenger Compartment (Sta 1016-1217) General Visual (External)
Figure 1

EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AREA AFT OF THE PRESSURE BULKHEAD
		D633A109-AKS 53-894-00-01

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE AREA AFT OF PRESSURE BULKHEAD			BOEING CARD NO. 53-896-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 311BL			ZONE 311 312	

Perform an internal zonal inspection (GV) of the area aft of pressure bulkhead - Section 48, aft of aft pressure bulkhead to Sta 1088. (EZAP)

NOTE: Not applicable to airplanes with flat aft pressure bulkhead.

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AREA AFT OF PRESSURE BULKHEAD
		D633A109-AKS 53-896-00-01

AKS

737-600/700/800/900

TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-896-00-01
				MECH INSP
EWIS				
TASK 05-41-03-210-802				
1. INTERNAL - ZONAL (GV): Area Aft of Pressure Bulkhead				
(Figure 1)				
A. General				
(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection				
SUBTASK 05-41-03-010-001				
(1) Open this access panel:				
Number Name/Location				
311BL Stabilizer Trim Access Door				
SUBTASK 05-41-03-210-033				
(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.				
(3) Perform an internal zonal inspection (GV) of the area aft of pressure bulkhead - Section 48, aft of aft pressure bulkhead to Sta 1088. (EZAP)				
SUBTASK 05-41-03-910-005				
(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-03-410-001				
(5) Close this access panel:				
Number Name/Location				
311BL Stabilizer Trim Access Door				
— END OF TASK —				
EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AREA AFT OF PRESSURE BULKHEAD		
		D633A109-AKS 53-896-00-01		Page 2 of 5 Feb 15/2015

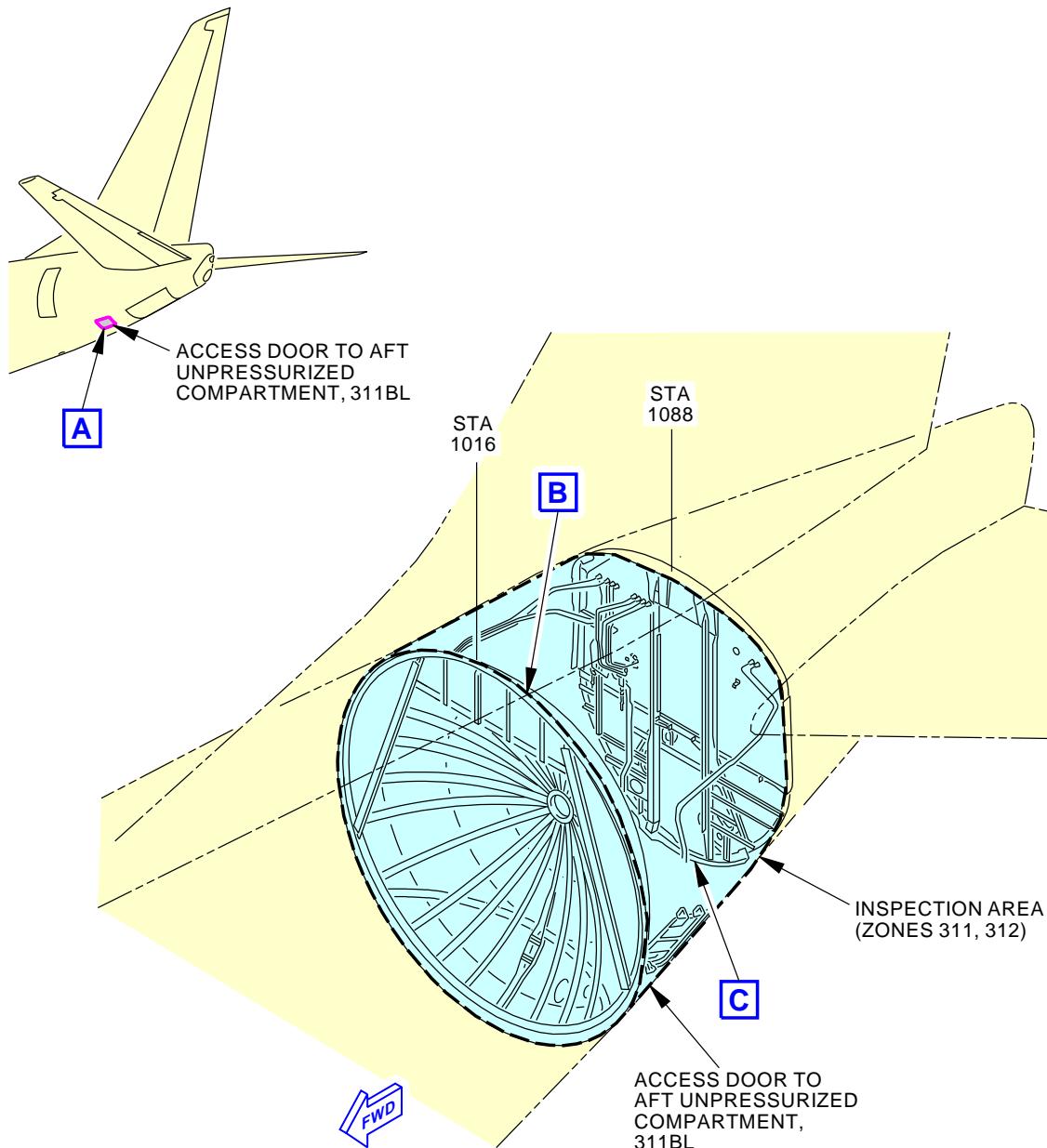
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

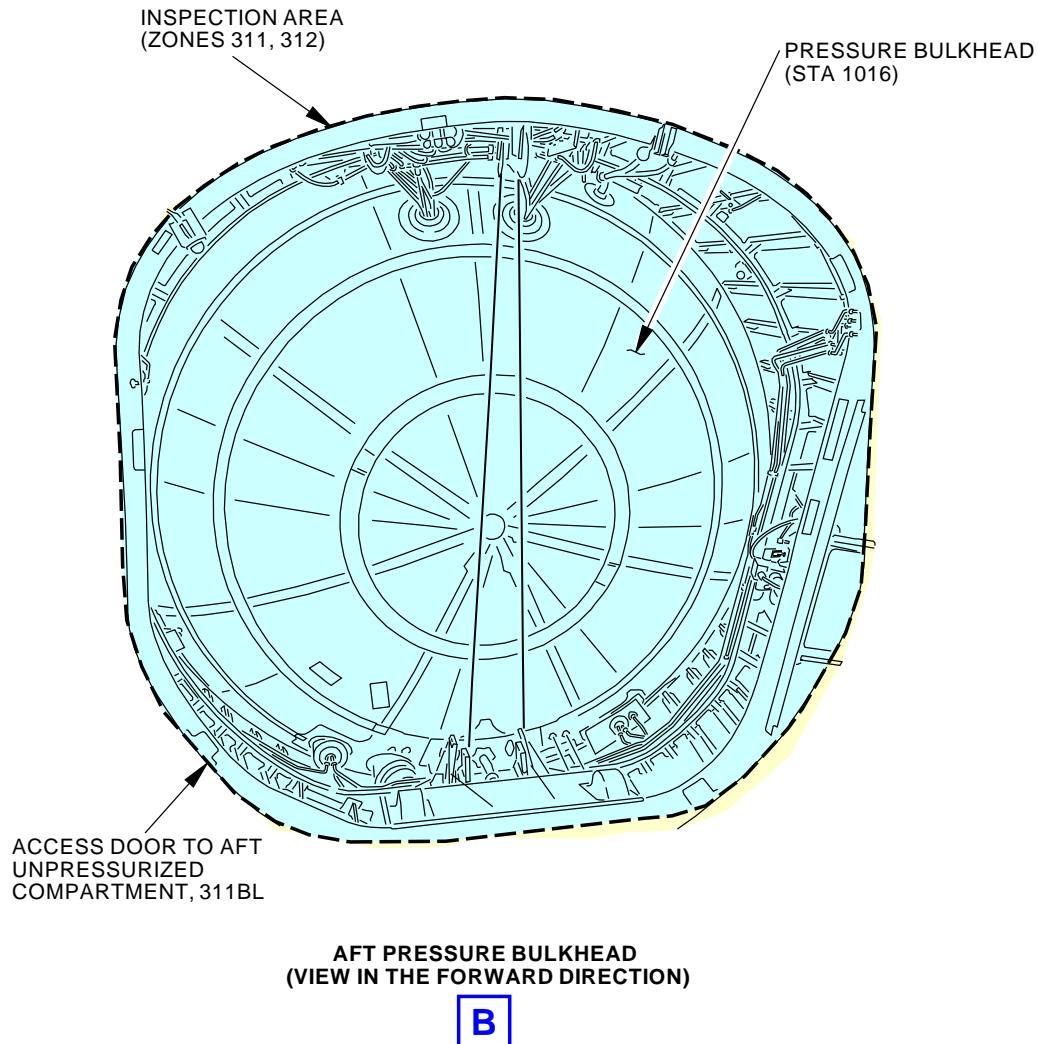
BOEING CARD NO.
53-896-00-01**A**

L04248 S0006584161_V2

**Aft Pressure Bulkhead General Visual (Internal)
Figure 1 (Sheet 1 of 3)****EFFECTIVITY
AKS ALL; AIRPLANES WITH A CURVED AFT
PRESSURE BULKHEAD****SOURCE
MRB****AREA AFT OF PRESSURE BULKHEAD****D633A109-AKS
53-896-00-01****Page 3 of 5
Feb 15/2015**

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-896-00-01
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Aft Pressure Bulkhead General Visual (Internal)
Figure 1 (Sheet 2 of 3)

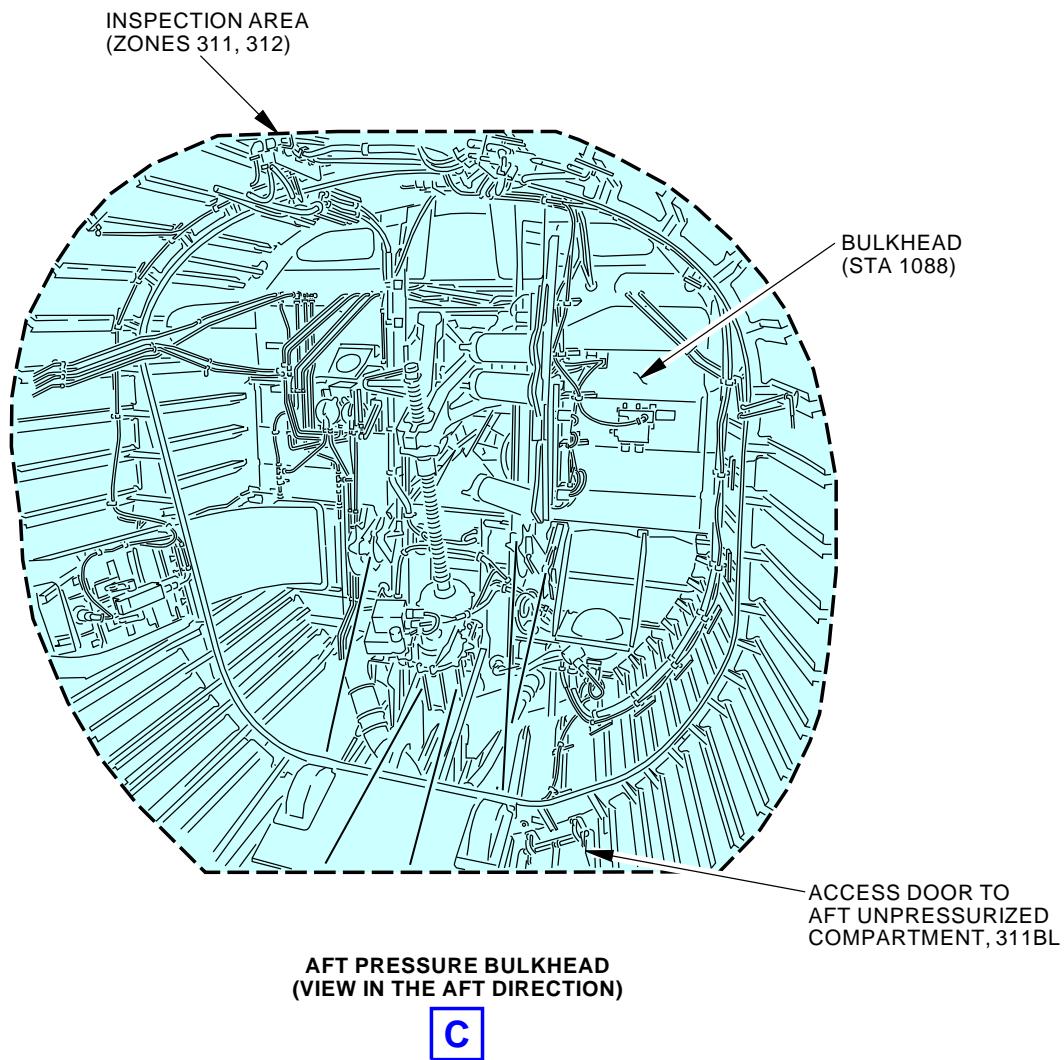
L04388 S0006584162_V2

EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AREA AFT OF PRESSURE BULKHEAD
		D633A109-AKS 53-896-00-01

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AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				53-896-00-01



L07041 S0006584163_V2

**Aft Pressure Bulkhead General Visual (Internal)
Figure 1 (Sheet 3 of 3)**

EFFECTIVITY AKS ALL; AIRPLANES WITH A CURVED AFT PRESSURE BULKHEAD	SOURCE MRB	AREA AFT OF PRESSURE BULKHEAD
		D633A109-AKS 53-896-00-01

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Feb 15/2015**

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE STABILIZER TORSION BOX COMPARTMENT			BOEING CARD NO. 53-898-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY	
STATION	SKILL AIRPL				AIRPLANE ALL	ENGINE ALL
		ACCESS 311BL			ZONE 313 314	

Perform an internal zonal inspection (GV) of the stabilizer torsion box compartment - Section 48, Sta 1088 to Sta 1156. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

EFFECTIVITY AKS ALL	SOURCE MRB	STABILIZER TORSION BOX COMPARTMENT
		D633A109-AKS 53-898-00-01

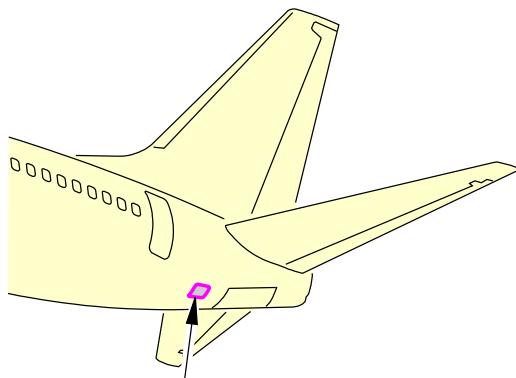
AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-898-00-01
				MECH INSP
► EWIS TASK 05-41-03-210-803				
1. INTERNAL - ZONAL (GV): Stabilizer Torsion Box Compartment (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection SUBTASK 05-41-03-010-002 (1) Open this access panel: Number Name/Location 311BL Stabilizer Trim Access Door				
SUBTASK 05-41-03-210-035 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the stabilizer torsion box compartment - Section 48, Sta 1088 to Sta 1156. (EZAP)				
SUBTASK 05-41-03-910-007 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-03-410-002 (5) Close this access panel: Number Name/Location 311BL Stabilizer Trim Access Door				
———— END OF TASK ——				

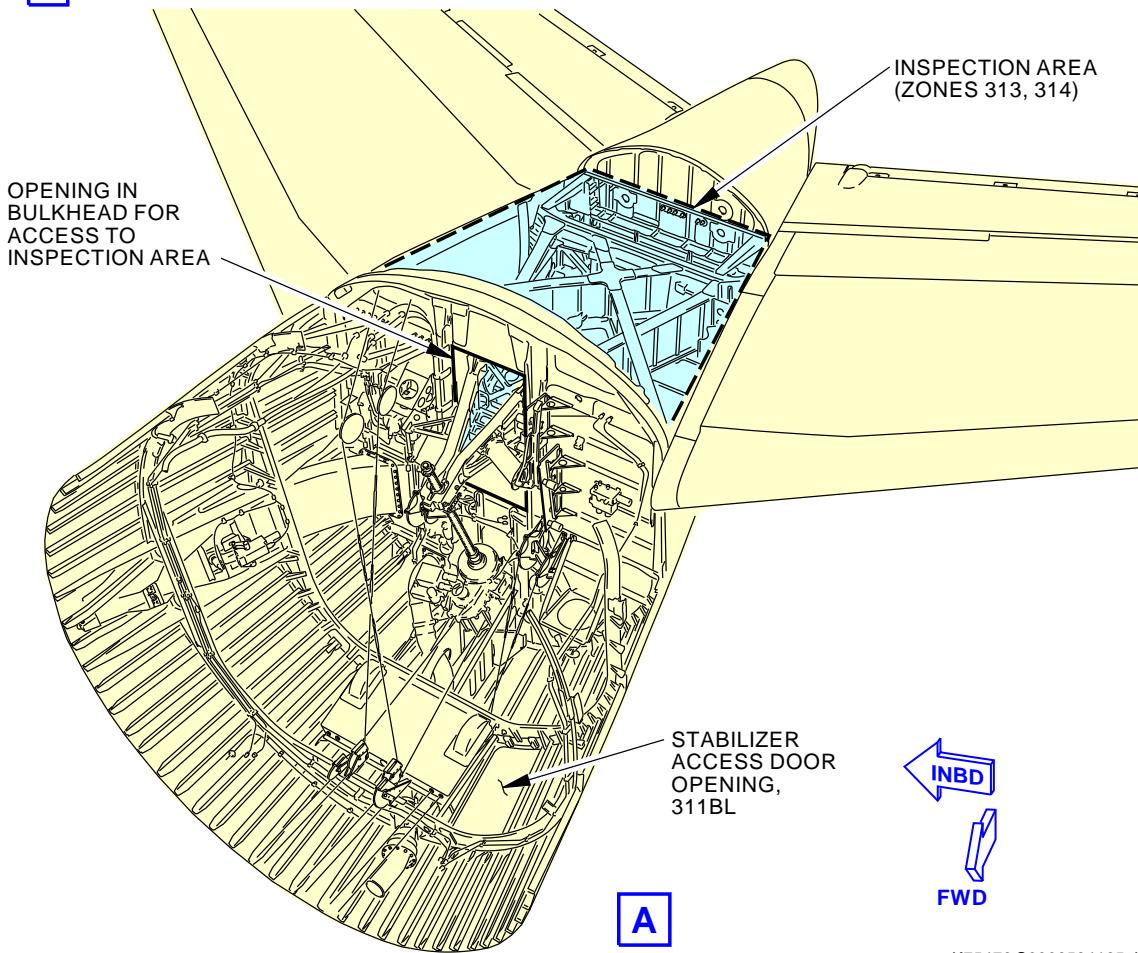
EFFECTIVITY AKS ALL	SOURCE MRB	STABILIZER TORSION BOX COMPARTMENT
		D633A109-AKS 53-898-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-898-00-01
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STABILIZER ACCESS DOOR, 311BL

A**Stabilizer Torsion Box Compartment General Visual (Internal)
Figure 1**

K75476 S0006584165_V2

EFFECTIVITY AKS ALL	SOURCE MRB	STABILIZER TORSION BOX COMPARTMENT
		D633A109-AKS 53-898-00-01

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AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE APU COMPARTMENT			BOEING CARD NO. 53-900-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 5500 FC 30 MO	REPEAT 5500 FC 30 MO	APPLICABILITY AIRPLANE ALL
STATION	SKILL AIRPL	ACCESS 315A			ZONE 315 316

Perform an internal zonal inspection (GV) of the APU compartment - Section 48, Sta 1088 to Sta 1156. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 5500 FC/30 MO is satisfied by this zonal inspection.

EFFECTIVITY AKS ALL	SOURCE MRB	APU COMPARTMENT
		D633A109-AKS 53-900-00-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-900-00-01
				MECH INSP
► EWIS TASK 05-41-03-210-804				
1. INTERNAL - ZONAL (GV): APU Compartment (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection SUBTASK 05-41-03-010-003 (1) Open this access panel: Number Name/Location 315A APU Cowl Door				
SUBTASK 05-41-03-210-036 (2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804. (3) Perform an internal zonal inspection (GV) of the APU compartment - Section 48, Sta 1088 to Sta 1156. (EZAP)				
SUBTASK 05-41-03-910-008 (4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-03-410-003 (5) Close this access panel: Number Name/Location 315A APU Cowl Door				
———— END OF TASK ————				

EFFECTIVITY AKS ALL	SOURCE MRB	APU COMPARTMENT D633A109-AKS 53-900-00-01	Page 2 of 3 Feb 15/2015
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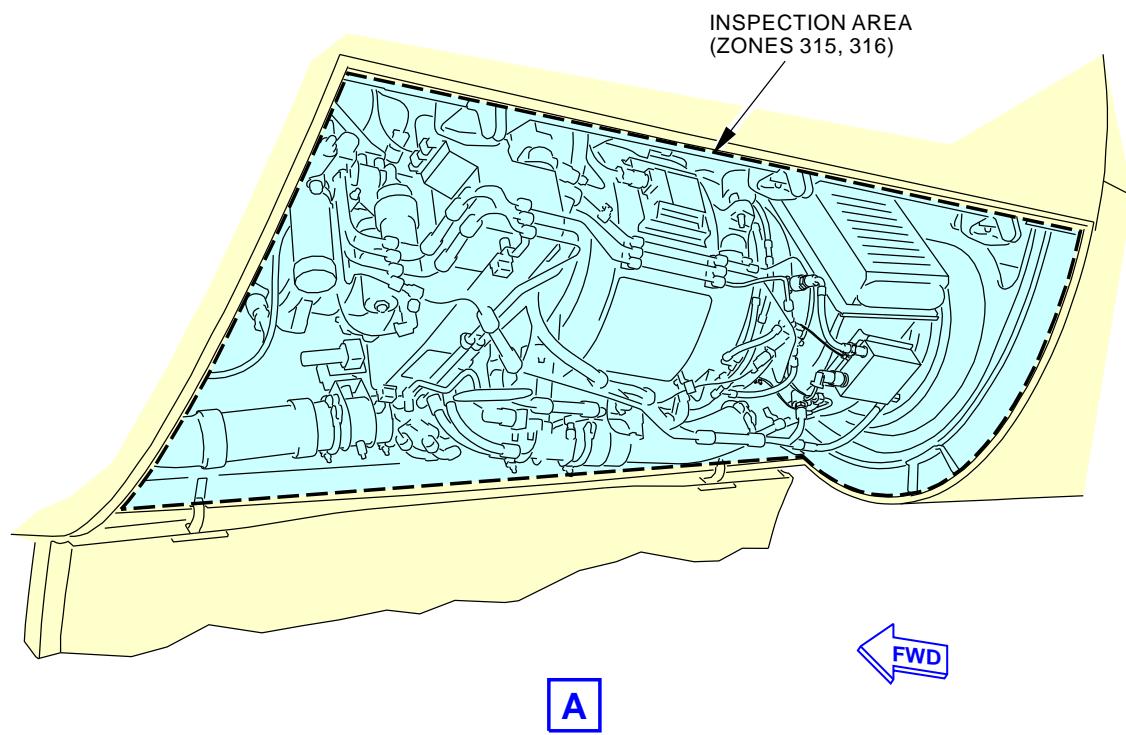
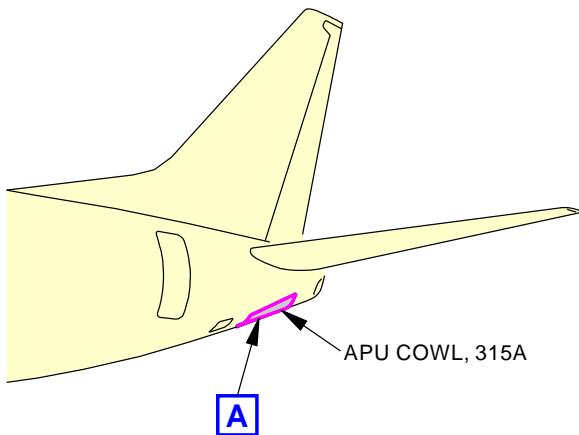
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-900-00-01

APU Compartment - General Visual (Internal)
Figure 1

K69242 S0006584167_V2

EFFECTIVITY AKS ALL	SOURCE MRB	APU COMPARTMENT
		D633A109-AKS 53-900-00-01

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Feb 15/2015

AKS**737-600/700/800/900
TASK CARDS**

AIRLINE CARD NO		TITLE TAIL CONE			BOEING CARD NO. 53-902-00-01	
DATE	TASK ZONAL (GV)				RELATED CARD	
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 36 MO	APPLICABILITY	
STATION	SKILL AIRPL	VERSION 1.2	THRESHOLD 6600 FC	REPEAT 36 MO	AIRPLANE ALL	ENGINE ALL
		ACCESS 318BR			ZONE 317 318	

Perform an internal zonal inspection (GV) of the tail cone - Section 48, Sta 1156 to Sta 1217. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 18000 FC/6 YR is satisfied by this zonal inspection.

EFFECTIVITY AKS ALL	SOURCE MRB	TAIL CONE D633A109-AKS 53-902-00-01	Page 1 of 3 Feb 15/2015
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AKS

737-600/700/800/900

TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-902-00-01
				MECH INSP
EWIS				
TASK 05-41-03-210-805				
1. INTERNAL - ZONAL (GV): Tail Cone				
(Figure 1)				
A. General				
(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection				
SUBTASK 05-41-03-010-004				
(1) Open this access panel:				
Number Name/Location				
318BR Tailcone Access Door				
SUBTASK 05-41-03-210-037				
(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.				
(3) Perform an internal zonal inspection (GV) of the tail cone - Section 48, Sta 1156 to Sta 1217. (EZAP)				
SUBTASK 05-41-03-910-009				
(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-03-410-004				
(5) Close this access panel:				
Number Name/Location				
318BR Tailcone Access Door				
— END OF TASK —				

EFFECTIVITY AKS ALL	SOURCE MRB	TAIL CONE D633A109-AKS 53-902-00-01	Page 2 of 3 Feb 15/2015
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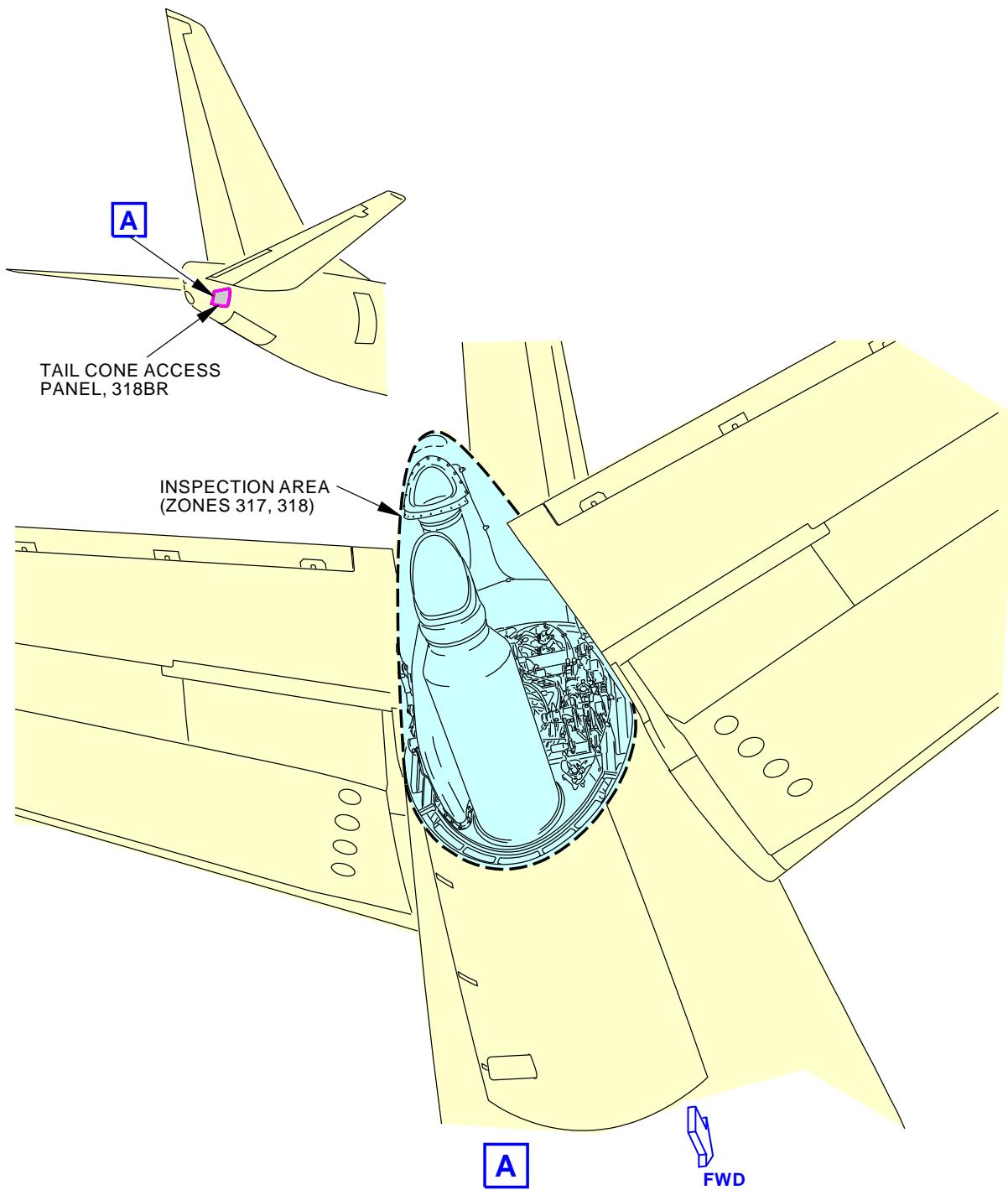
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-902-00-01**Tail Cone - General Visual (Internal)
Figure 1**

K77042 S0006584169_V2

EFFECTIVITY AKS ALL	SOURCE MRB	TAIL CONE D633A109-AKS 53-902-00-01
		Page 3 of 3 Feb 15/2015

AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FLAP SUPPORT NO. 4 - LEFT WING			BOEING CARD NO.
DATE	TASK ZONAL (GV)				53-904-01-01
TAIL NUMBER	WORK AREA L MAIN W/W	VERSION 1.1 1.2 NOTE	THRESHOLD 5500 FC 30 MO	REPEAT 5500 FC 30 MO	RELATED CARD APPLICABILITY AIRPLANE ALL ENGINE ALL
STATION	SKILL AIRPL	ACCESS 194BL NOTE			ZONE 133 194 541

Perform an internal zonal inspection (GV) of the flap support no. 4 - left wing - Section 46, Sta 727. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 5500 FC/30 MO is satisfied by this zonal inspection.

ACCESS NOTE: Flaps extended.

A. References

<u>Reference</u>	<u>Title</u>
AMM 27-51-00-860-803	Extend the Trailing Edge Flaps (P/B 201)
AMM 27-51-00-860-804	Retract the Trailing Edge Flaps (P/B 201)

EFFECTIVITY AKS ALL	SOURCE MRB	FLAP SUPPORT NO. 4 - LEFT WING
		D633A109-AKS 53-904-01-01

Page 1 of 3
Jun 15/2016

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-904-01-01
				MECH INSP
► EWIS TASK 05-41-05-210-820				
1. INTERNAL - ZONAL (GV): Flap Support No. 4 - Left Wing (Figure 1)				
A. General (1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection SUBTASK 05-41-05-210-047 (1) Extend the trailing edge flaps to the 40-unit position. To extend them, do this task:Extend the Trailing Edge Flaps, AMM TASK 27-51-00-860-803.				
SUBTASK 05-41-05-010-008 (2) Open this access panel: Number Name/Location 194BL Flap Track Lubrication Panel - Aft				
SUBTASK 05-41-05-210-039 (3) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.				
(4) Perform an internal zonal inspection (GV) of the flap support no. 4 - left wing - Section 46, Sta 727. (EZAP)				
SUBTASK 05-41-05-910-008 (5) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-05-410-008 (6) Close this access panel: Number Name/Location 194BL Flap Track Lubrication Panel - Aft				
SUBTASK 05-41-05-210-048 (7) Retract the trailing edge flaps to the UP position. To retract them, do this task:Retract the Trailing Edge Flaps, AMM TASK 27-51-00-860-804.				
———— END OF TASK ——				

EFFECTIVITY AKS ALL	SOURCE MRB	FLAP SUPPORT NO. 4 - LEFT WING
		D633A109-AKS 53-904-01-01

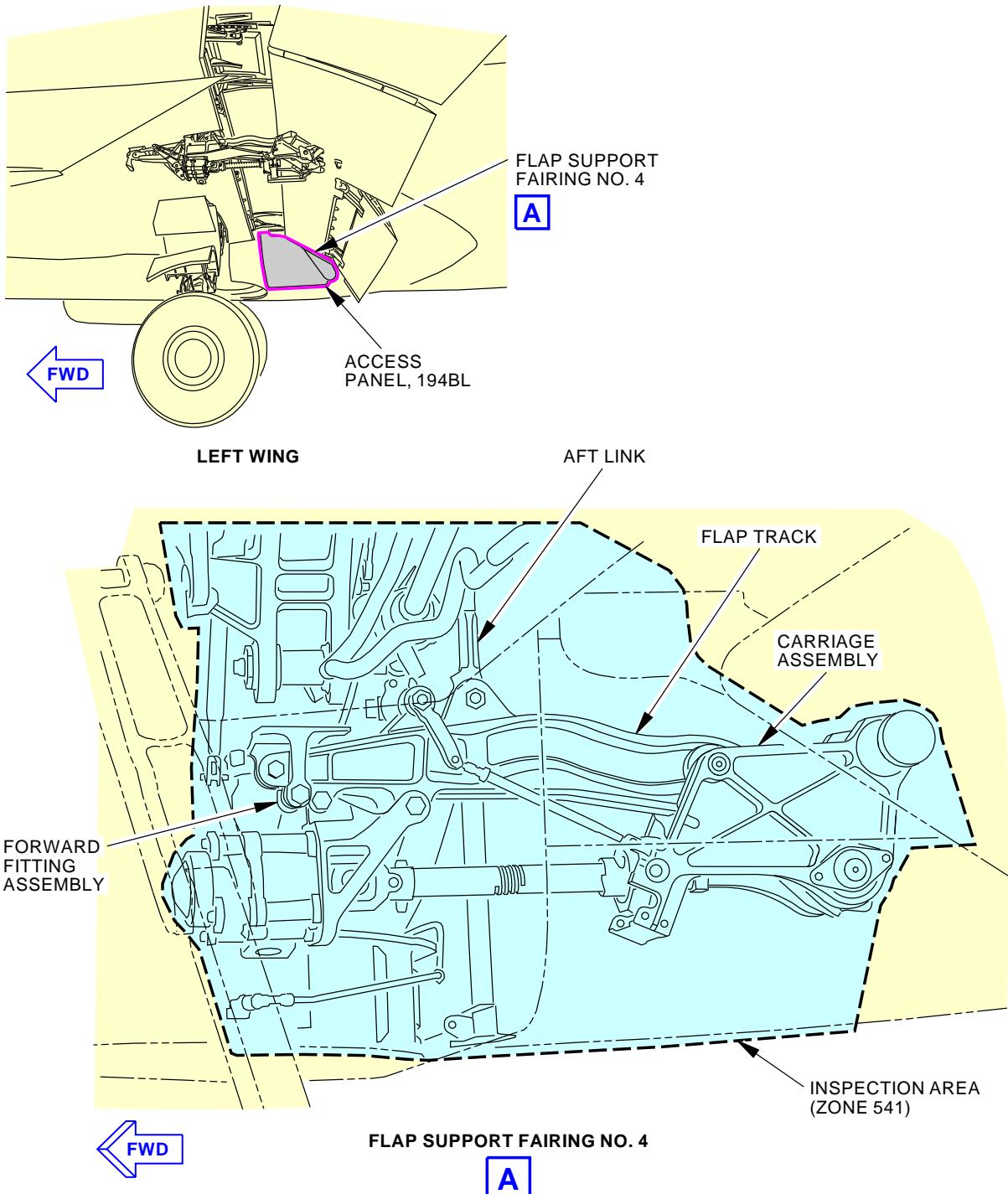
AKS**737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-904-01-01

**Flap Support Fairing No. 4 General Visual (Internal)
Figure 1**

L02791 S0006584315_V4

EFFECTIVITY AKS ALL	SOURCE MRB	FLAP SUPPORT NO. 4 - LEFT WING
		D633A109-AKS 53-904-01-01

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AKS

737-600/700/800/900

TASK CARDS

AIRLINE CARD NO		TITLE FLAP SUPPORT NO. 5 - RIGHT WING			BOEING CARD NO.		
DATE	TASK ZONAL (GV)				53-906-02-01		
TAIL NUMBER	WORK AREA R MAIN W/W	VERSION 1.1 1.2 NOTE	THRESHOLD 5500 FC 30 MO	REPEAT 5500 FC 30 MO	APPLICABILITY AIRPLANE ALL		
STATION	SKILL AIRPL	ACCESS 194BR NOTE			ENGINE ALL		
					ZONE 134 194 641		

Perform an internal zonal inspection (GV) of the flap support no. 5 - right wing - Section 46, Sta 727. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 5500 FC/30 MO is satisfied by this zonal inspection.

ACCESS NOTE: Flaps extended.

A. References

<u>Reference</u>	<u>Title</u>
AMM 27-51-00-860-803	Extend the Trailing Edge Flaps (P/B 201)
AMM 27-51-00-860-804	Retract the Trailing Edge Flaps (P/B 201)

EFFECTIVITY AKS ALL	SOURCE MRB	FLAP SUPPORT NO. 5 - RIGHT WING
		D633A109-AKS 53-906-02-01

AKS**737-600/700/800/900
TASK CARDS**

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO. 53-906-02-01
				MECH INSP
► EWIS				
TASK 05-41-06-210-820				
1. INTERNAL - ZONAL (GV): Flap Support No. 5 - Right Wing				
(Figure 1)				
A. General				
(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.				
B. Zonal Inspection				
SUBTASK 05-41-06-210-047				
(1) Extend the trailing edge flaps to the 40-unit position. To extend them, do this task:Extend the Trailing Edge Flaps, AMM TASK 27-51-00-860-803.				
SUBTASK 05-41-06-010-008				
(2) Open this access panel:				
Number Name/Location				
194BR Flap Track Lubrication Panel - Aft				
SUBTASK 05-41-06-210-039				
(3) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.				
(4) Perform an internal zonal inspection (GV) of the flap support fairing no. 5 - right wing - Section 46, Sta 727. (EZAP)				
SUBTASK 05-41-06-910-008				
(5) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.				
SUBTASK 05-41-06-410-008				
(6) Close this access panel:				
Number Name/Location				
194BR Flap Track Lubrication Panel - Aft				
SUBTASK 05-41-06-210-048				
(7) Retract the trailing edge flaps to the UP position. To retract them, do this task: Retract the Trailing Edge Flaps, AMM TASK 27-51-00-860-804.				
———— END OF TASK ——				

EFFECTIVITY AKS ALL	SOURCE MRB	FLAP SUPPORT NO. 5 - RIGHT WING	
		D633A109-AKS 53-906-02-01	Page 2 of 3 Jun 15/2016

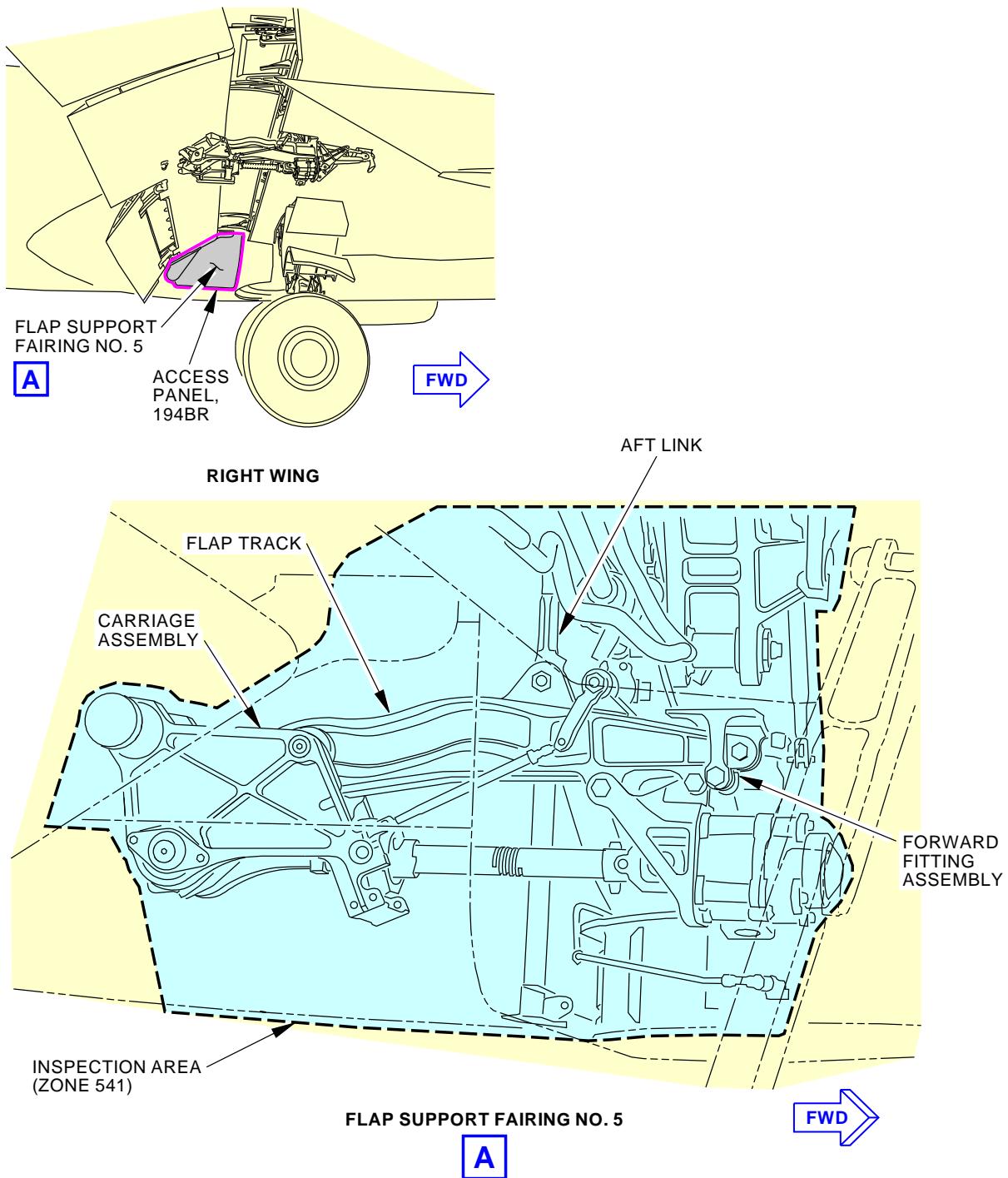
AKS**BOEING****737-600/700/800/900
TASK CARDS**

DATE

TAIL NUMBER

STATION

AIRLINE CARD NO.

BOEING CARD NO.
53-906-02-01

**Flap Support Fairing No. 5 General Visual (Internal)
Figure 1**

L02787 S0006584401_V4

EFFECTIVITY AKS ALL	SOURCE MRB	FLAP SUPPORT NO. 5 - RIGHT WING
		D633A109-AKS 53-906-02-01

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Feb 15/2015