CHAPTER

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STABILIZERS

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55-641-00-01	FAT	
1	Oct 15/2014	
R 2	Jun 15/2016	
55-641-00-02	FAT	
1	Oct 15/2014	
R 2	Jun 15/2016	
55-642-00-01	FAT	
1	Oct 15/2014	
R 2	Jun 15/2016	
55-642-00-02	FAT	
1	Oct 15/2014	
R 2	Jun 15/2016	
55-643-00-01	FAT	
1	Oct 15/2014	
2	Feb 15/2015	
55-643-00-02	FAT	
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2	Feb 15/2015	
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2	Feb 15/2015	
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55-645-00-01	FAT	
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2	Feb 15/2015	
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2	Feb 15/2015	
55-646-00-01	FAT	
1	Oct 15/2014	
2	Feb 15/2015	
55-646-00-02	FAT	
1	Oct 15/2014	
2	Feb 15/2015	
55-647-00-01	FAT	
1	Jun 15/2015	
2	Jun 15/2015	
3	Jun 15/2015	
55-647-00-02	FAT	
1	Jun 15/2015	
2	Jun 15/2015	
3	Jun 15/2015	
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1	Jun 15/2015	
R 2	Jun 15/2016	
A 3	Jun 15/2016	
55-648-00-02	FAT	
1	Jun 15/2015	
R 2	Jun 15/2016	
A 3	Jun 15/2016	
55-649-00-01	FAT	
1	Oct 15/2014	
2	Feb 15/2015	
55-649-00-02	FAT	
1	Oct 15/2014	
2	Feb 15/2015	
55-652-00-01	FAT	
1	Oct 15/2014	
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1	Oct 15/2014	
2	Feb 15/2015	
55-662-00-01	FAT	
1	Oct 15/2014	
2	Feb 15/2015	
3	Feb 15/2015	
55-663-00-01	FAT	
1	Oct 15/2014	
2	Feb 15/2015	
55-664-00-01	FAT	
1	Oct 15/2014	
2	Feb 15/2015	
55-665-00-01	FAT	
1	Jun 15/2015	
2	Jun 15/2015	

A = Added, R = Revised, D = Deleted, O = Overflow, C = Customer Originated Change

55-EFFECTIVE PAGES

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CHAPTER 55
STABILIZERS

Subject/Page	Date	COC
55-666-00-01	FAT	
1	Oct 15/2014	
2	Feb 15/2015	
55-800-00-01	ZON	
R 1	Jun 15/2016	
2	Feb 15/2015	
3	Feb 15/2015	
55-802-00-01	ZON	
1	Feb 15/2015	
2	Feb 15/2015	
3	Feb 15/2015	
55-804-00-01	ZON	
1	Feb 15/2015	
2	Feb 15/2015	
3	Oct 15/2015	
55-806-00-01	ZON	
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2	Feb 15/2015	
3	Feb 15/2015	
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3	Feb 15/2015	
4	Feb 15/2015	
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2	Feb 15/2015	
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4	Jun 15/2015	
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1	Oct 15/2014	
2	Feb 15/2015	
3	Feb 15/2015	
55-814-00-01	ZON	
1	Oct 15/2014	
2	Feb 15/2015	
3	Feb 15/2015	
55-814-00-01	ZON (cont)	
4	Feb 15/2015	
55-816-00-01	ZON	
1	Feb 15/2015	
2	Feb 15/2015	
3	Feb 15/2015	
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3	Feb 15/2015	
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R 2	Jun 15/2016	
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2	Feb 15/2015	
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2	Feb 15/2015	
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55-848-02-01	ZON	
1	Feb 15/2015	
2	Feb 15/2015	
3	Feb 15/2015	

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55-EFFECTIVE PAGES

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER CENTER SECTION			55-010-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1 1.2 NOTE	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 311 312

Inspect horizontal stabilizer center section jackscrew fitting.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-010-00-01

MECH	INSP

TASK 55-05-03-210-801

1. INTERNAL - GENERAL VISUAL: HORIZONTAL STABILIZER CENTER SECTION

(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-001

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

SUBTASK 55-05-03-210-001

(2) Do a General Visual inspection of the horizontal stabilizer center section jackscrew fitting.

SUBTASK 55-05-03-910-001

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–809.

SUBTASK 55-05-03-410-001

(4) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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MECH	INSP

TASK 51-05-01-210-809

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-078

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-079

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-080

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-081

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-082

(5) CPCP Basic Task Item 5 is not applicable.

SUBTASK 51-05-01-210-100

(6) Do the CPCP Basic Task Item 6 (Not Applicable)

SUBTASK 51-05-01-210-084

(7) CPCP Basic Task Item 7 is not applicable.

END OF TASK

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-010-00-01

MPD ITEM
55-010-00

2085645 S0000439464_V2

INTERNAL-GENERAL VISUAL: HORIZONTAL STABILIZER CENTER SECTION
Figure 1

BOEING PROPRIETARY - Copyright © Unpublished Work - See title page for details

AIRLINE CARD NO		TITLE			BOEING CARD NO.
		STABILIZER TORSION BOX COMPARTMENT			55-020-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA TAIL CONE	VERSION 1.1 1.2 NOTE	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL S3102 NOTE			ZONE 313 314

Inspect horizontal stabilizer center section front spar, rear spar, pivot fittings, and primary and secondary thrust beams.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Adjust stabilizer trim as required. Remove gap seal and rear spar sliding seal to inspect pivot fittings.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-020-00-01

MECH	INSP

TASK 55-05-03-210-802

1. INTERNAL - GENERAL VISUAL: STABILIZER TORSION BOX COMPARTMENT

(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-002

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

Special Access:

Number	Name/Location
S3102	Stabilizer Torsion Box Compartment Inspection

NOTE: Adjust stabilizer trim as required. Remove gap seal and rear spar sliding seal to inspect pivot fittings.

SUBTASK 55-05-03-210-002

(2) Do a General Visual inspection of the horizontal stabilizer center section front spar, rear spar, pivot fittings, and primary and secondary thrust beams.

SUBTASK 55-05-03-910-002

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–804.

SUBTASK 55-05-03-410-002

(4) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-020-00-01

MECH	INSP

TASK 51-05-01-210-804

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-043

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-044

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-045

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-046

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-111

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-048

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-020-00-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-020-00-01

MPD ITEM
55-020-00

2087220 S0000439467_V2

INTERNAL-GENERAL VISUAL: STABILIZER TORSION BOX COMPARTMENT
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-020-00-01

HORIZONTAL STABILIZER CENTER SECTION

B

MPD ITEM
55-020-00

2087239 S0000439468_V2

INTERNAL-GENERAL VISUAL: STABILIZER TORSION BOX COMPARTMENT
Figure 1 (Sheet 2 of 2)

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AIRLINE CARD NO		TITLE **VERTICAL FIN LEADING EDGE**			BOEING CARD NO. **55-030-00-01**
DATE	TASK **GENERAL VISUAL**				RELATED CARD
TAIL NUMBER	WORK AREA **VERT STABLIZER**	VERSION **1.1** **1.2**	THRESHOLD **12 YR** **36000 FC**	REPEAT **8 YR** **24000 FC**	APPLICABILITY AIRPLANE **ALL** ENGINE **ALL**
STATION	SKILL **AIRPL**	**NOTE**			
		ACCESS **322B 322C 323AL 323AR 323BL 323BR**			ZONE **322**

Inspect the forward side of the vertical fin front spar, including front spar chords, webs, and terminal fittings.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-030-00-01

MECH	INSP

TASK 55-05-03-210-803

1. INTERNAL - GENERAL VISUAL: VERTICAL FIN LEADING EDGE

(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-003

(1) Open these access panels:

Number	Name/Location
322B	Vertical Fin, Removable Leading Edge
322C	Vertical Fin, Removable Leading Edge
323AL	Vertical Fin, Front Spar Access Door
323AR	Vertical Fin, Front Spar Access Door
323BL	Vertical Fin, Forward Fin Access Door
323BR	Vertical Fin, Forward Fin Access Door

SUBTASK 55-05-03-210-003

(2) Do a General Visual inspection of the forward side of the vertical fin front spar, including front spar chords, webs, and terminal fittings.

SUBTASK 55-05-03-910-003

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–806.

SUBTASK 55-05-03-410-003

(4) Close these access panels:

Number	Name/Location
322B	Vertical Fin, Removable Leading Edge
322C	Vertical Fin, Removable Leading Edge
323AL	Vertical Fin, Front Spar Access Door
323AR	Vertical Fin, Front Spar Access Door
323BL	Vertical Fin, Forward Fin Access Door
323BR	Vertical Fin, Forward Fin Access Door

——— END OF TASK ———

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MECH	INSP

TASK 51-05-01-210-806

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-057

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-058

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-059

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-060

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-114

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-062

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-030-00-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-063

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-030-00-01

MPD ITEM
55-030-00

2087274 S0000439604_V2

INTERNAL-GENERAL VISUAL: VERTICAL FIN LEADING EDGE
Figure 1 (Sheet 1 of 3)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-030-00-01

VERTICAL FIN FRONT SPAR
(LEADING EDGE SKIN REMOVED)

B

MPD ITEM
55-030-00

2087343 S0000439605_V2

INTERNAL-GENERAL VISUAL: VERTICAL FIN LEADING EDGE
Figure 1 (Sheet 2 of 3)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-030-00-01

TERMINAL FITTING ASSEMBLY

C

MPD ITEM
55-030-00

2087424 S0000439606_V2

INTERNAL-GENERAL VISUAL: VERTICAL FIN LEADING EDGE
Figure 1 (Sheet 3 of 3)

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AIRLINE CARD NO		TITLE **VERTICAL FIN**			BOEING CARD NO. **55-050-00-01**
DATE	TASK **GENERAL VISUAL**				RELATED CARD
TAIL NUMBER	WORK AREA **VERT STABLIZER**	VERSION **1.1** **1.2**	THRESHOLD **12 YR** **36000 FC**	REPEAT **8 YR** **24000 FC**	APPLICABILITY AIRPLANE **ALL** ENGINE **ALL**
STATION	SKILL **AIRPL**	**NOTE**			
		ACCESS **322A 322AL 322AR 322B 322C 323AL 323AR 323BL 323BR 323CL 323CR 323DL 323EL 323FL 323GL 324AL 324AR 324BL 324BR 324DL 324DR 324JL 324KL 324LL 324ML 324NL S3232** **NOTE**			ZONE **323**

Inspect vertical fin from front spar to rear spar, including spar chords and webs, inspar skins, lower closure rib, and rear spar terminal fittings.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove upper removable web. Pin removal is not required for terminal fitting inspection.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-050-00-01

MECH	INSP

TASK 55-05-03-210-804

1. INTERNAL - GENERAL VISUAL: VERTICAL FIN

(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-004

(1) Open these access panels:

Number	Name/Location
322A	Vertical Fin, Fixed Leading Edge
322AL	Vertical Fin, Fixed Leading Edge
322AR	Vertical Fin, Fixed Leading Edge
322B	Vertical Fin, Removable Leading Edge
322C	Vertical Fin, Removable Leading Edge
323AL	Vertical Fin, Front Spar Access Door
323AR	Vertical Fin, Front Spar Access Door
323BL	Vertical Fin, Forward Fin Access Door
323BR	Vertical Fin, Forward Fin Access Door
323CL	Vertical Fin, Rear Spar Access Door
323CR	Vertical Fin, Rear Spar Access Door
323DL	Vertical Fin, Access
323EL	Vertical Fin, Access
323FL	Vertical Fin, Access
323GL	Vertical Fin, Access
324AL	Vertical Fin, Aft Fin Access Door
324AR	Vertical Fin, Aft Fin Access Door
324BL	Vertical Fin, Trailing Edge Access
324BR	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324DR	Vertical Fin, Trailing Edge Access
324JL	Vertical Fin, Access
324KL	Vertical Fin, Access
324LL	Vertical Fin, Access
324ML	Vertical Fin, Access
324NL	Vertical Fin, Access

Special Access:

Number	Name/Location
S3232	Vertical Fin Inspection

NOTE: Remove upper removable web. Pin removal is not required for terminal fitting inspection.

SUBTASK 55-05-03-210-004

(2) Do a General Visual inspection of the vertical fin from front spar to rear spar, including spar chords and webs, inspar skins, lower closure rib, and rear spar terminal fittings.

SUBTASK 55-05-03-910-004

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–804.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-050-00-01

MECH	INSP

SUBTASK 55-05-03-410-004

(4) Close these access panels:

Number	Name/Location
322A	Vertical Fin, Fixed Leading Edge
322AL	Vertical Fin, Fixed Leading Edge
322AR	Vertical Fin, Fixed Leading Edge
322B	Vertical Fin, Removable Leading Edge
322C	Vertical Fin, Removable Leading Edge
323AL	Vertical Fin, Front Spar Access Door
323AR	Vertical Fin, Front Spar Access Door
323BL	Vertical Fin, Forward Fin Access Door
323BR	Vertical Fin, Forward Fin Access Door
323CL	Vertical Fin, Rear Spar Access Door
323CR	Vertical Fin, Rear Spar Access Door
323DL	Vertical Fin, Access
323EL	Vertical Fin, Access
323FL	Vertical Fin, Access
323GL	Vertical Fin, Access
324AL	Vertical Fin, Aft Fin Access Door
324AR	Vertical Fin, Aft Fin Access Door
324BL	Vertical Fin, Trailing Edge Access
324BR	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324DR	Vertical Fin, Trailing Edge Access
324JL	Vertical Fin, Access
324KL	Vertical Fin, Access
324LL	Vertical Fin, Access
324ML	Vertical Fin, Access
324NL	Vertical Fin, Access

END OF TASK

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-050-00-01

MECH	INSP

TASK 51-05-01-210-804

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-043

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-044

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-045

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-046

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-111

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-048

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-050-00-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-050-00-01

VERTICAL FIN
(LEFT SIDE)
A

VERTICAL FIN
(RIGHT SIDE)
A

MPD ITEM
55-050-00

2089274 S0000440578_V2

INTERNAL-GENERAL VISUAL: VERTICAL FIN
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-050-00-01

INTERNAL-GENERAL VISUAL: VERTICAL FIN
Figure 1 (Sheet 2 of 2)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN TRAILING EDGE			55-060-00-01
DATE	TASK				RELATED CARD
	GENERAL VISUAL				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	VERT STABLIZER	1.1 1.2	12 YR 36000 FC	8 YR 24000 FC	AIRPLANE: ALL ENGINE: ALL
STATION	SKILL	NOTE			
	AIRPL				
		ACCESS			ZONE
		323CL 324BL 324DL 324JL 324KL 324LL 324ML 324NL			324

Inspect aft side of vertical fin rear spar, including rear spar chords and webs, and rudder hinge ribs.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-060-00-01

MECH	INSP

TASK 55-05-03-210-805

1. INTERNAL - GENERAL VISUAL: VERTICAL FIN TRAILING EDGE

Figure 1

A. Inspection

SUBTASK 55-05-03-010-018

(1) Open these access panels:

Number	Name/Location
323CL	Vertical Fin, Rear Spar Access Door
324BL	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324JL	Vertical Fin, Access
324KL	Vertical Fin, Access
324LL	Vertical Fin, Access
324ML	Vertical Fin, Access
324NL	Vertical Fin, Access

SUBTASK 55-05-03-210-005

(2) Do a General Visual inspection of the aft side of vertical fin rear spar, including rear spar chords and webs, and rudder hinge ribs.

SUBTASK 55-05-03-910-005

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–806.

SUBTASK 55-05-03-410-018

(4) Close these access panels:

Number	Name/Location
323CL	Vertical Fin, Rear Spar Access Door
324BL	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324JL	Vertical Fin, Access
324KL	Vertical Fin, Access
324LL	Vertical Fin, Access
324ML	Vertical Fin, Access
324NL	Vertical Fin, Access

——— END OF TASK ———

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MECH	INSP

TASK 51-05-01-210-806

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-057

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-058

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-059

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-060

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-114

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-062

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-060-00-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-063

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-060-00-01

VERTICAL FIN (LEFT SIDE)

AFT SIDE OF VERTICAL FIN REAR SPAR (ACCESS AND SKIN PANELS NOT SHOWN)

A

MPD ITEM
55-060-00

2089247 S0000440581_V3

INTERNAL-GENERAL VISUAL: VERTICAL FIN TRAILING EDGE
Figure 1

BOEING PROPRIETARY - Copyright © Unpublished Work - See title page for details

AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER LEADING EDGE			55-070-01-01
DATE	TASK				RELATED CARD
	GENERAL VISUAL				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	12 YR	8 YR	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL	1.2	36000 FC	24000 FC	
	AIRPL	NOTE			
		ACCESS			ZONE
		331A 331AZ 331B S3311			331
		NOTE			

Inspect forward side of front spar, including front spar chords and webs, terminal fittings, and center section front spar lugs.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Bolt removal is required to inspect terminal fittings and center section front spar lugs.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
SPL-2035	Sling - Horizontal Stabilizer Part #: C55007-1 Supplier: 81205 Part #: C55007-29 Supplier: 81205

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MECH	INSP

TASK 55-05-03-210-806

1. INTERNAL - GENERAL VISUAL: LEFT HORIZONTAL STABILIZER LEADING EDGE
(Figure 1)

A. Prepare for the Inspection

SUBTASK 55-05-03-480-001

(1) Install the sling, sling, SPL-2035.

(a) Remove the bolts [1] and bolts [2] from the stabilizer top surface.

WARNING: MAKE SURE YOU INSTALL THE HOISTING SLING CORRECTLY WHEN YOU REMOVE THE BOLTS. IF YOU DO NOT INSTALL THE HOISTING SLING CORRECTLY, YOU CAN CAUSE DAMAGE TO EQUIPMENT OR INJURY TO PERSONNEL.

(b) Attach the sling, SPL-2035 to the horizontal stabilizer with fasteners.

(c) Attach a hoist to the sling, SPL-2035.

CAUTION: YOU MUST CAREFULLY LIFT THE HOISTING SLING. A SUDDEN MOVEMENT OF THE HOISTING SLING CAN CAUSE DAMAGE TO THE HORIZONTAL STABILIZER.

(d) Apply sufficient tension to the sling to lift the weight of the horizontal stabilizer.

B. Inspection

SUBTASK 55-05-03-010-005

(1) Open these access panels:

Number	Name/Location
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
331AZ	Horizontal Stabilizer, Access Panel, Inboard L.E. Closure Rib
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Special Access:

Number	Name/Location
S3311	Left Horizontal Stabilizer Leading Edge Inspection

SUBTASK 55-05-03-020-001

(2) At the horizontal stabilizer front spar, remove and keep the pins [7], nuts [6], washers [5], shims [8], washers [3] and bolts [4].

SUBTASK 55-05-03-210-006

(3) Do a general visual inspection of the forward side of front spar, including front spar chords and webs, terminal fittings, and center section front spar lugs.

SUBTASK 55-05-03-910-006

(4) Do this task: 737-6789 Basic Task Description, TASK 51-05-01-210-804.

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MECH	INSP

C. Put the Airplane Back to its Usual Condition

SUBTASK 55-05-03-420-001

(1) At the horizontal stabilizer front spar, install the pins [7], nuts [6], washers [5], shims [8], washers [3] and bolts [4].

SUBTASK 55-05-03-410-019

(2) Close these access panels:

Number	Name/Location
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
331AZ	Horizontal Stabilizer, Access Panel, Inboard L.E. Closure Rib
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Special Access:

Number	Name/Location
S3311	Left Horizontal Stabilizer Leading Edge Inspection

SUBTASK 55-05-03-080-001

(3) Remove the sling, SPL-2035.

(a) Release tension applied to the sling, SPL-2035.

(b) Remove the hoist to the sling, SPL-2035.

(c) Remove the fasteners that attach the sling, SPL-2035.

(d) Install bolts [1] and bolts [2] at the stabilizer top surface.

——— END OF TASK ———

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MECH	INSP

TASK 51-05-01-210-804

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-043

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-044

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-045

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-046

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-111

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-048

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-01-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-01-01

MPD ITEM
55-070-01

2082194 S0000437284_V2

INTERNAL-GENERAL VISUAL: LEFT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 1 of 5)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-01-01

(FIXED LEADING EDGE IS REMOVED)

C

MPD ITEM
55-070-01

2082193 S0000437285_V3

INTERNAL-GENERAL VISUAL: LEFT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 2 of 5)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-01-01

1 THE FORWARD ATTACH (BLUE) FITTING IS ATTACHED TO THE STABILIZER FRONT SPAR (LEADING EDGE ATTACH BOLTS).

TWO BOLTS ARE USED ON THE INBOARD SIDE OF THE LEADING EDGE RIB (STA 114.55) AND TWO BOLTS ARE USED ON THE OUTBOARD SIDE OF THE LEADING EDGE RIB (STA 114.55).

2 THE AFT OUTBOARD ATTACH (RED) FITTING IS ATTACHED TO THE ELEVATOR HINGE SUPPORT RIBS, FOUR BOLTS ARE USED AT STA 176.64 AND FOUR BOLTS ARE USED AT STA 149.12.

3 THE AFT INBOARD ATTACH (GREEN) FITTING IS ATTACHED TO THE ELEVATOR HINGE SUPPORT RIBS, SIX BOLTS ARE USED AT STA 66.54 AND FOUR BOLTS ARE USED AT STA 39.02.

2438621 S0000565964_V1

INTERNAL-GENERAL VISUAL: LEFT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 3 of 5)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-01-01

2438627 S0000565965_V1

INTERNAL-GENERAL VISUAL: LEFT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 4 of 5)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-01-01

4 BEFORE INSTALLING NUT, COAT THREADS OF BOLT AND NUT WITH BMS 3-28 ANTI-SEIZE COMPOUND.

5 TORQUE NUT TO AN INITIAL TORQUE OF 50 POUND-INCHES (5.6 NEWTON-METERS)

6 AFTER INITIAL TORQUE, WASHER [5] OR SHIM WASHER [11]/[12] MUST TOUCH STEPPED PART OF BOLT [4]. IF THE WASHER [5] OR SHIM WASHER [11]/[12] DOES NOT CONTACT STEPPED PART OF BOLT [4], REMOVE LAMINATED SHIMS [8] UNTIL IT DOES.

7 THE GAP "X" BETWEEN THE SHIMS [8] AND WASHER [5] IS MORE THAN 0.002 INCH (0.05 mm), ADD SHIMS [8] UNTIL THE GAP "X" IS BETWEEN 0.002-0.000 INCH (0.05-0.00 mm).

8 WHEN SHIM [8] ADJUSTMENTS ARE COMPLETE, TORQUE NUT TO 950-1400 POUND-INCHES (107.4-158.2 NEWTON-METERS).

2438674 S0000565966_V1

INTERNAL-GENERAL VISUAL: LEFT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 5 of 5)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER LEADING EDGE			55-070-02-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1 1.2 NOTE	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341A 341AZ 341B S3411 NOTE			ZONE 341

Inspect forward side of front spar, including front spar chords and webs, terminal fittings, and center section front spar lugs.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Bolt removal is required to inspect terminal fittings and center section front spar lugs.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
SPL-2035	Sling - Horizontal Stabilizer Part #: C55007-1 Supplier: 81205 Part #: C55007-29 Supplier: 81205

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-02-01

MECH	INSP

TASK 55-05-03-210-807

1. INTERNAL - GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER LEADING EDGE
(Figure 1)

A. Prepare for the Inspection

SUBTASK 55-05-03-480-002

(1) Install the sling, SPL-2035.

(a) Remove the bolts [1] and bolts [2] from the stabilizer top surface.

WARNING: MAKE SURE YOU INSTALL THE HOISTING SLING CORRECTLY WHEN YOU REMOVE THE BOLTS. IF YOU DO NOT INSTALL THE HOISTING SLING CORRECTLY, YOU CAN CAUSE DAMAGE TO EQUIPMENT OR INJURY TO PERSONNEL.

(b) Attach the sling, SPL-2035 to the horizontal stabilizer with fasteners.

(c) Attach a hoist to the sling, SPL-2035.

CAUTION: YOU MUST CAREFULLY LIFT THE HOISTING SLING. A SUDDEN MOVEMENT OF THE HOISTING SLING CAN CAUSE DAMAGE TO THE HORIZONTAL STABILIZER.

(d) Apply sufficient tension to the sling to lift the weight of the horizontal stabilizer.

B. Inspection

SUBTASK 55-05-03-010-006

(1) Open these access panels:

Number	Name/Location
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
341AZ	Horizontal Stabilizer, Access Panel - Inbd L.E. Closure Rib
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

Special Access:

Number	Name/Location
S3411	Right Horizontal Stabilizer Leading Edge Inspection

SUBTASK 55-05-03-020-002

(2) At the horizontal stabilizer front spar, remove and keep the pins [7], nuts [6], washers [5], shims [8], washers [7] and bolts [4].

SUBTASK 55-05-03-210-007

(3) Do a general visual inspection of the forward side of front spar, including front spar chords and webs, terminal fittings, and center section front spar lugs.

SUBTASK 55-05-03-910-007

(4) Do this task: 737-6789 Basic Task Description, TASK 51-05-01-210-804.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-02-01

MECH	INSP

C. Put the Airplane Back to its Usual Condition

SUBTASK 55-05-03-400-001

(1) At the horizontal stabilizer front spar, install the pins [7], nuts [6], washers [5], shims [8], washers [7] and bolts [4].

SUBTASK 55-05-03-410-006

(2) Close these access panels:

Number	Name/Location
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
341AZ	Horizontal Stabilizer, Access Panel - Inbd L.E. Closure Rib
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

Special Access:

Number	Name/Location
S3411	Right Horizontal Stabilizer Leading Edge Inspection

SUBTASK 55-05-03-080-002

(3) Remove the sling, SPL-2035.

(a) Release tension applied to the sling, SPL-2035.

(b) Remove the hoist to the sling, SPL-2035.

(c) Remove the fasteners that attach the sling, SPL-2035.

(d) Install bolts [1] and bolts [2] at the stabilizer top surface.

——— END OF TASK ———

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TASK 51-05-01-210-804

MECH	INSP

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-043

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-044

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-045

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-046

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-111

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-048

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-02-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-02-01

MPD ITEM
55-070-02

2082191 S0000437290_V2

INTERNAL-GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 1 of 5)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-02-01

MPD ITEM
55-070-02

2082192 S0000437291_V3

INTERNAL-GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 2 of 5)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-02-01

1 THE FORWARD ATTACH (BLUE) FITTING IS ATTACHED TO THE STABILIZER FRONT SPAR (LEADING EDGE ATTACH BOLTS).

TWO BOLTS ARE USED ON THE INBOARD SIDE OF THE LEADING EDGE RIB (STA 114.55) AND TWO BOLTS ARE USED ON THE OUTBOARD SIDE OF THE LEADING EDGE RIB (STA 114.55).

2 THE AFT OUTBOARD ATTACH (RED) FITTING IS ATTACHED TO THE ELEVATOR HINGE SUPPORT RIBS, FOUR BOLTS ARE USED AT STA 176.64 AND FOUR BOLTS ARE USED AT STA 149.12.

3 THE AFT INBOARD ATTACH (GREEN) FITTING IS ATTACHED TO THE ELEVATOR HINGE SUPPORT RIBS, SIX BOLTS ARE USED AT STA 66.54 AND FOUR BOLTS ARE USED AT STA 39.02.

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INTERNAL-GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 3 of 5)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-02-01

INTERNAL-GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 4 of 5)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-070-02-01

4 BEFORE INSTALLING NUT, COAT THREADS OF BOLT AND NUT WITH BMS 3-28 ANTI-SEIZE COMPOUND.

5 TORQUE NUT TO AN INITIAL TORQUE OF 50 POUND-INCHES (5.6 NEWTON-METERS)

6 AFTER INITIAL TORQUE, WASHER [5] OR SHIM WASHER [11]/[12] MUST TOUCH STEPPED PART OF BOLT [4]. IF THE WASHER [5] OR SHIM WASHER [11]/[12] DOES NOT CONTACT STEPPED PART OF BOLT [4], REMOVE LAMINATED SHIMS [8] UNTIL IT DOES.

7 THE GAP "X" BETWEEN THE SHIMS [8] AND WASHER [5] IS MORE THAN 0.002 INCH (0.05 mm), ADD SHIMS [8] UNTIL THE GAP "X" IS BETWEEN 0.002-0.000 INCH (0.05-0.00 mm).

8 WHEN SHIM [8] ADJUSTMENTS ARE COMPLETE, TORQUE NUT TO 950-1400 POUND-INCHES (107.4-158.2 NEWTON-METERS).

2438709 S0000565972_V1

INTERNAL-GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER LEADING EDGE
Figure 1 (Sheet 5 of 5)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER LEADING EDGE			55-080-01-01
DATE	TASK				RELATED CARD
	GENERAL VISUAL				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	8 YR	4 YR	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		331A			331
		NOTE			

Inspect left horizontal stabilizer front spar terminal fittings and center section front spar lugs.

ACCESS NOTE: Bolt removal is not required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-080-01-01

MECH	INSP

TASK 55-05-03-210-808

1. INTERNAL - GENERAL VISUAL: LEFT HORIZONTAL STABILIZER LEADING EDGE
(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-007

(1) Open this access panel:

Number	Name/Location
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body

NOTE: Bolt removal is not required.

SUBTASK 55-05-03-210-008

(2) Do a General Visual inspection of the left horizontal stabilizer front spar terminal fittings and center section front spar lugs.

SUBTASK 55-05-03-910-008

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–801.

SUBTASK 55-05-03-410-007

(4) Close this access panel:

Number	Name/Location
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body

——— END OF TASK ———

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TASK 51-05-01-210-801

MECH	INSP

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-022

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-023

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-024

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-025

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-026

(5) CPCP Basic Task Item 5 is not applicable.

SUBTASK 51-05-01-210-027

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

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MECH	INSP

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

Water displacing / anti-corrosion compounds should not be applied in the following areas:

- Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
- Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
- Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
- Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
- Areas with electrical arc potential.
- Interior materials, including cargo liners (change of flammability properties).
- Fiber-glass ducts where temperature exceeds 220 degrees F.
- Selected areas noted in baseline program.

SUBTASK 51-05-01-210-028

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-080-01-01

Left Horizontal Stabilizer Leading Edge General Visual (Internal)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER LEADING EDGE			55-080-02-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 8 YR	REPEAT 4 YR	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341A			ZONE 341
		NOTE			

Inspect right horizontal stabilizer front spar terminal fittings and center section front spar lugs.

ACCESS NOTE: Bolt removal is not required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-080-02-01

MECH	INSP

TASK 55-05-03-210-809

1. INTERNAL - GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER LEADING EDGE
(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-008

(1) Open this access panel:

Number	Name/Location
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body

NOTE: Bolt removal is not required.

SUBTASK 55-05-03-210-009

(2) Do a General Visual inspection of the right horizontal stabilizer front spar terminal fittings and center section front spar lugs.

SUBTASK 55-05-03-910-009

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–801.

SUBTASK 55-05-03-410-008

(4) Close this access panel:

Number	Name/Location
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-080-02-01

MECH	INSP

TASK 51-05-01-210-801

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-022

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-023

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-024

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-025

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-026

(5) CPCP Basic Task Item 5 is not applicable.

SUBTASK 51-05-01-210-027

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-080-02-01

MECH	INSP

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

Water displacing / anti-corrosion compounds should not be applied in the following areas:

- Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
- Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
- Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
- Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
- Areas with electrical arc potential.
- Interior materials, including cargo liners (change of flammability properties).
- Fiber-glass ducts where temperature exceeds 220 degrees F.
- Selected areas noted in baseline program.

SUBTASK 51-05-01-210-028

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-080-02-01

Right Horizontal Stabilizer Leading Edge General Visual (Internal)
Figure 1

BOEING PROPRIETARY - Copyright © Unpublished Work - See title page for details

AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER			55-100-01-01
DATE	TASK				RELATED CARD
	GENERAL VISUAL				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	12 YR	8 YR	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL	1.2	36000 FC	24000 FC	
	AIRPL	NOTE			
		ACCESS			ZONE
		331B 331BZ 331CZ 331DZ 332AB 332AT 332BB 332CB 332DB 332EB 332FB 332GB 332HB 332JB 332KB 332LB 332MB 332NB 335A			332

Inspect left horizontal stabilizer from front spar to rear spar including spar chords, webs, terminal fittings, upper and lower inspar skins.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-01-01

MECH	INSP

TASK 55-05-03-210-810

1. INTERNAL - GENERAL VISUAL: LEFT HORIZONTAL STABILIZER

(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-009

(1) Open these access panels:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ
331BZ	Horizontal Stabilizer Front Spar Removal Web Panel
331CZ	Horizontal Stabilizer Front Spar Removal Web Panel
331DZ	Horizontal Stabilizer Front Spar Removal Web Panel
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332BB	Horizontal Stabilizer, Access Door
332CB	Horizontal Stabilizer, Access Door
332DB	Horizontal Stabilizer, Access Door
332EB	Horizontal Stabilizer, Access Door
332FB	Horizontal Stabilizer, Access Door
332GB	Horizontal Stabilizer, Access Door
332HB	Horizontal Stabilizer, Access Door
332JB	Horizontal Stabilizer, Access Door
332KB	Horizontal Stabilizer, Access Door
332LB	Horizontal Stabilizer, Access Door
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
335A	Horizontal Stabilizer, Removable Stabilizer Tip

SUBTASK 55-05-03-210-010

(2) Do a General Visual inspection of the left horizontal stabilizer from front spar to rear spar including spar chords, webs, terminal fittings, upper and lower inspar skins.

SUBTASK 55-05-03-910-010

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–806.

SUBTASK 55-05-03-410-009

(4) Close these access panels:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ
331BZ	Horizontal Stabilizer Front Spar Removal Web Panel
331CZ	Horizontal Stabilizer Front Spar Removal Web Panel
331DZ	Horizontal Stabilizer Front Spar Removal Web Panel
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-01-01

MECH	INSP

(Continued)

Number	Name/Location
332BB	Horizontal Stabilizer, Access Door
332CB	Horizontal Stabilizer, Access Door
332DB	Horizontal Stabilizer, Access Door
332EB	Horizontal Stabilizer, Access Door
332FB	Horizontal Stabilizer, Access Door
332GB	Horizontal Stabilizer, Access Door
332HB	Horizontal Stabilizer, Access Door
332JB	Horizontal Stabilizer, Access Door
332KB	Horizontal Stabilizer, Access Door
332LB	Horizontal Stabilizer, Access Door
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
335A	Horizontal Stabilizer, Removable Stabilizer Tip

——— END OF TASK ———

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TASK 51-05-01-210-806

MECH	INSP

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-057

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-058

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-059

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-060

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-114

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-062

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-01-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-063

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-01-01

(BOTTOM VIEW)

A

(INSPAR SKINS NOT SHOWN)

MPD ITEM
55-100-01

B

2084317 S0000438551_V2

INTERNAL-GENERAL VISUAL: LEFT HORIZONTAL STABILIZER
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-01-01

MPD ITEM
55-100-01

2084318 S0000438552_V2

INTERNAL-GENERAL VISUAL: LEFT HORIZONTAL STABILIZER
Figure 1 (Sheet 2 of 2)

BOEING PROPRIETARY - Copyright © Unpublished Work - See title page for details

AIRLINE CARD NO		TITLE RIGHT HORIZONTAL STABILIZER			BOEING CARD NO. 55-100-02-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	36000 FC	24000 FC	
		ACCESS 341B 341BZ 341CZ 341DZ 342AB 342AT 342BB 342CB 342DB 342EB 342FB 342GB 342HB 342JB 342KB 342LB 342MB 342NB 345A			ZONE 342

Inspect right horizontal stabilizer from front spar to rear spar including spar chords, webs, terminal fittings, upper and lower inspar skins.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-02-01

MECH	INSP

TASK 55-05-03-210-811

1. INTERNAL - GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER

(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-010

(1) Open these access panels:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ
341BZ	Horizontal Stabilizer Front Spar Removal Web Panel
341CZ	Horizontal Stabilizer Front Spar Removal Web Panel
341DZ	Horizontal Stabilizer Front Spar Removal Web Panel
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer
342BB	Horizontal Stabilizer, Access Door
342CB	Horizontal Stabilizer, Access Door
342DB	Horizontal Stabilizer, Access Door
342EB	Horizontal Stabilizer, Access Door
342FB	Horizontal Stabilizer, Access Door
342GB	Horizontal Stabilizer, Access Door
342HB	Horizontal Stabilizer, Access Door
342JB	Horizontal Stabilizer, Access Door
342KB	Horizontal Stabilizer, Access Door
342LB	Horizontal Stabilizer, Access Door
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
345A	Horizontal Stabilizer, Removable Stabilizer Tip

SUBTASK 55-05-03-210-011

(2) Do a General Visual inspection of the right horizontal stabilizer from front spar to rear spar including spar chords, webs, terminal fittings, upper and lower inspar skins.

SUBTASK 55-05-03-910-011

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–806.

SUBTASK 55-05-03-410-010

(4) Close these access panels:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ
341BZ	Horizontal Stabilizer Front Spar Removal Web Panel
341CZ	Horizontal Stabilizer Front Spar Removal Web Panel
341DZ	Horizontal Stabilizer Front Spar Removal Web Panel
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-02-01

MECH	INSP

(Continued)

Number	Name/Location
342BB	Horizontal Stabilizer, Access Door
342CB	Horizontal Stabilizer, Access Door
342DB	Horizontal Stabilizer, Access Door
342EB	Horizontal Stabilizer, Access Door
342FB	Horizontal Stabilizer, Access Door
342GB	Horizontal Stabilizer, Access Door
342HB	Horizontal Stabilizer, Access Door
342JB	Horizontal Stabilizer, Access Door
342KB	Horizontal Stabilizer, Access Door
342LB	Horizontal Stabilizer, Access Door
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
345A	Horizontal Stabilizer, Removable Stabilizer Tip

——— END OF TASK ———

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MECH	INSP

TASK 51-05-01-210-806

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-057

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-058

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-059

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-060

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-114

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-062

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-02-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-063

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-02-01

(BOTTOM VIEW)

A

(INSPAR SKINS NOT SHOWN)

B

MPD ITEM
55-100-02

2084315 S0000438553_V2

INTERNAL-GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-100-02-01

MPD ITEM
55-100-02

2084319 S0000438554_V2

INTERNAL-GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER
Figure 1 (Sheet 2 of 2)

BOEING PROPRIETARY - Copyright © Unpublished Work - See title page for details

AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER TRAILING EDGE AND ELEVATOR TAB LEADING EDGE			55-110-01-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1 1.2 NOTE	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333AB 333AT 333BB 333CB 333DB 333EB 333FB 334AB 334BB 334CB 334DB 334EB 334FB 335A S3331 NOTE			ZONE 333

Inspect aft side of left horizontal stabilizer rear spar, including spar chords and webs, terminal fittings, elevator tab actuator support fitting, center section rear spar lugs, elevator hinge ribs, and elevator tab leading edge.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Disconnect balance panels in balance bays. Bolt removal is required to inspect terminal fittings and center section rear spar lugs, except at rear spar upper lugs and clevis. Remove tab hinge covers to inspect elevator tab leading edge.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
SPL-2030	Equipment - Pin Removal, Horizontal Stabilizer Part #: C55005-19 Supplier: 81205 Opt Part #: C55005-11 Supplier: 81205
SPL-2035	Sling - Horizontal Stabilizer Part #: C55007-1 Supplier: 81205 Part #: C55007-29 Supplier: 81205

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MECH	INSP

TASK 55-05-03-210-812

1. **INTERNAL - GENERAL VISUAL: LEFT HORIZONTAL STABILIZER TRAILING EDGE AND ELEVATOR TAB LEADING EDGE**

(Figure 1)

A. Prepare for the Inspection

SUBTASK 55-05-03-480-003

(1) Install the sling, SPL-2035.

(a) Remove the bolts [1] and bolts [2] from the stabilizer top surface.

WARNING: MAKE SURE YOU INSTALL THE HOISTING SLING CORRECTLY WHEN YOU REMOVE THE BOLTS. IF YOU DO NOT INSTALL THE HOISTING SLING CORRECTLY, YOU CAN CAUSE DAMAGE TO EQUIPMENT OR INJURY TO PERSONNEL.

(b) Attach the sling, SPL-2035 to the horizontal stabilizer with fasteners.

(c) Attach a hoist to the sling, SPL-2035.

CAUTION: YOU MUST CAREFULLY LIFT THE HOISTING SLING. A SUDDEN MOVEMENT OF THE HOISTING SLING CAN CAUSE DAMAGE TO THE HORIZONTAL STABILIZER.

(d) Apply sufficient tension to the sling to lift the weight of the horizontal stabilizer.

B. Inspection

SUBTASK 55-05-03-010-011

(1) Open these access panels:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334BB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334CB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334DB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334EB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
335A	Horizontal Stabilizer, Removable Stabilizer Tip

Special Access:

Number	Name/Location
S3331	Left Hrizontal Stabilizer Trailing Edge Inspection

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MECH	INSP

SUBTASK 55-05-03-020-003

(2) Disconnect the balance panels in the balance bays.

SUBTASK 55-05-03-010-019

(3) Remove tab hinge covers to inspect elevator tab leading edge.

SUBTASK 55-05-03-020-004

(4) Remove the horizontal stabilizer rear spar bolts:

NOTE: It is not required to remove the bolts at the upper lugs and clevis at the rear spar.

NOTE: If the bolts are frozen, use equipment, SPL-2030 to assist in removal.

(a) Remove and keep the nuts [5] and washers [4] from bolts [8] and bolts [9]. Keep a record of where each part came from.

(b) Remove and keep bolts [8], bolts [9], washers [7], and washers [3]. Keep a record of where each part came from.

(c) Remove and keep the pins [6] and anti-rotation locks [11]. Keep a record of where each part came from.

NOTE: If necessary, adjust the tension of the sling, SPL-2035 to aid removal of the pins [6].

SUBTASK 55-05-03-210-012

(5) Do a general visual inspection of the aft side of left horizontal stabilizer rear spar, including spar chords and webs, terminal fittings, elevator tab actuator support fitting, center section rear spar lugs, elevator hinge ribs, and elevator tab leading edge.

SUBTASK 55-05-03-910-012

(6) Do this task: 737-6789 Basic Task Description, TASK 51-05-01-210-804.

C. Put the Airplane Back to its Usual Condition

SUBTASK 55-05-03-400-002

(1) Install the pins [6] and anti-rotation locks [11].

SUBTASK 55-05-03-400-003

(2) Install bolt [8], bolt [9], washers [7], and washers [3].

SUBTASK 55-05-03-400-004

(3) Install the nuts [5] and washers [4] on bolt [8] and bolt [9].

SUBTASK 55-05-03-420-002

(4) Replace the tab hinge covers.

SUBTASK 55-05-03-420-003

(5) Reconnect the balance panels in the balance bay.

SUBTASK 55-05-03-410-021

(6) Close this access panel:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

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SUBTASK 55-05-03-410-011

(7) Close these access panels:

Number	Name/Location
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334BB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334CB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334DB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334EB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
335A	Horizontal Stabilizer, Removable Stabilizer Tip

Special Access:

Number	Name/Location
S3331	Left Hrizontal Stabilizer Trailing Edge Inspection

SUBTASK 55-05-03-080-003

(8) Remove the sling, SPL-2035:

(a) Release tension applied to the sling, SPL-2035.

(b) Remove the hoist to the sling, SPL-2035.

(c) Remove the fasteners that attach the sling, SPL-2035.

(d) Install bolts [1] and bolts [2] at the stabilizer top surface.

——— END OF TASK ———

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MECH	INSP

TASK 51-05-01-210-804

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-043

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-044

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-045

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-046

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-111

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-048

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

LEFT ELEVATOR
(LOWER TRAILING EDGE PANELS REMOVED)

MPD ITEM
55-110-01

A

2077097 S0000434967_V3

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 1 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 2 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 3 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

2077200 S0000434980_V3

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 4 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

MPD ITEM
55-110-01

H

MPD ITEM
55-110-01

2077205 S0000434982_V3

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 5 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

HINGE 1 AND 2
(COVER REMOVED FOR CLARITY)

I

HINGE 3
(HINGE 4, 5 AND 6 ARE EQUIVALENT)

MPD ITEM
55-110-01

J

MPD ITEM
55-110-01

2077208 S0000434983_V3

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 6 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

LEFT HORIZONTAL STABILIZER REAR SPAR CHORDS, WEB, TERMINAL FITTINGS AND TAB ACTUATOR FITTING (PANELS REMOVED FOR CLARITY)

MPD ITEM
55-110-01

K

MPD ITEM
55-110-01

2077326 S0000434985_V3

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 7 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

1 THE FORWARD ATTACH (BLUE) FITTING IS ATTACHED TO THE STABILIZER FRONT SPAR (LEADING EDGE ATTACH BOLTS).

TWO BOLTS ARE USED ON THE INBOARD SIDE OF THE LEADING EDGE RIB (STA 114.55) AND TWO BOLTS ARE USED ON THE OUTBOARD SIDE OF THE LEADING EDGE RIB (STA 114.55).

2 THE AFT OUTBOARD ATTACH (RED) FITTING IS ATTACHED TO THE ELEVATOR HINGE SUPPORT RIBS, FOUR BOLTS ARE USED AT STA 176.64 AND FOUR BOLTS ARE USED AT STA 149.12.

3 THE AFT INBOARD ATTACH (GREEN) FITTING IS ATTACHED TO THE ELEVATOR HINGE SUPPORT RIBS, SIX BOLTS ARE USED AT STA 66.54 AND FOUR BOLTS ARE USED AT STA 39.02.

2438800 S0000565960_V1

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 8 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

M

REAR SPAR CONNECTION
(ACCESS PANELS AND GAP COVERS REMOVED)

M

2438817 S0000565961_V1

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 9 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-01-01

NOTE:

COMBINED GAPS BETWEEN BUSHINGS AT A1 AND A2 SHOULD BE 0.046 INCH (1.17 mm), OR LESS.

N

NOTE:

COMBINED GAPS BETWEEN BUSHINGS AT A1 AND A2 SHOULD BE 0.032 INCH (0.81 mm), OR LESS.

O

4 BOLT HEAD DIRECTION OPTIONAL (WASHER NEEDS TO BE INSTALLED AT BOLT HEAD LOCATION).

2438844 S0000565962_V1

Left Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 10 of 10)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER TRAILING EDGE AND ELEVATOR TAB LEADING EDGE			55-110-02-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1 1.2 NOTE	THRESHOLD 12 YR 36000 FC	REPEAT 8 YR 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343AB 343AT 343BB 343CB 343DB 343EB 343FB 344AB 344BB 344CB 344DB 344EB 344FB 345A S3431 NOTE			ZONE 343

Inspect aft side of right horizontal stabilizer rear spar, including spar chords and webs, terminal fittings, elevator tab actuator support fitting, center section rear spar lugs, elevator hinge ribs, and elevator tab leading edge.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Disconnect balance panels in balance bays. Bolt removal is required to inspect terminal fittings and center section rear spar lugs, except at rear spar upper lugs and clevis. Remove tab hinge covers to inspect elevator tab leading edge.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

C. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
SPL-2030	Equipment - Pin Removal, Horizontal Stabilizer Part #: C55005-19 Supplier: 81205 Opt Part #: C55005-11 Supplier: 81205
SPL-2035	Sling - Horizontal Stabilizer Part #: C55007-1 Supplier: 81205 Part #: C55007-29 Supplier: 81205

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MECH	INSP

TASK 55-05-03-210-813

1. **INTERNAL - GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER TRAILING EDGE AND ELEVATOR TAB LEADING EDGE**

(Figure 1)

A. Prepare for the Inspection

SUBTASK 55-05-03-480-004

(1) Install the sling, SPL-2035.

(a) Remove the bolts [1] and bolts [2] from the stabilizer top surface.

WARNING: MAKE SURE YOU INSTALL THE HOISTING SLING CORRECTLY WHEN YOU REMOVE THE BOLTS. IF YOU DO NOT INSTALL THE HOISTING SLING CORRECTLY, YOU CAN CAUSE DAMAGE TO EQUIPMENT OR INJURY TO PERSONNEL.

(b) Attach the sling, SPL-2035 to the horizontal stabilizer with fasteners.

(c) Attach a hoist to the sling, SPL-2035.

CAUTION: YOU MUST CAREFULLY LIFT THE HOISTING SLING. A SUDDEN MOVEMENT OF THE HOISTING SLING CAN CAUSE DAMAGE TO THE HORIZONTAL STABILIZER.

(d) Apply sufficient tension to the sling to lift the weight of the horizontal stabilizer.

B. Inspection

SUBTASK 55-05-03-010-012

(1) Open these access panels:

Number	Name/Location
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334BB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334CB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334DB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334EB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344BB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344CB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344DB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344EB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
345A	Horizontal Stabilizer, Removable Stabilizer Tip

Special Access:

Number	Name/Location
S3431	Right Horizontal Stabilizer Trailing Edge

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SUBTASK 55-05-03-020-005

(2) Disconnect the balance panels in the balance bays.

SUBTASK 55-05-03-010-020

(3) Remove tab hinge covers to inspect elevator tab leading edge.

SUBTASK 55-05-03-020-006

(4) Remove the horizontal stabilizer rear spar bolts:

NOTE: It is not required to remove the bolts at the upper lugs and clevis at the rear spar.

NOTE: If the bolts are frozen, use equipment, SPL-2030 to assist in removal.

(a) Remove and keep the nuts [5] and washers [4] from bolts [8] and bolts [9]. Keep a record of where each part came from.

(b) Remove and keep bolts [8], bolts [9], washers [7], and washers [3]. Keep a record of where each part came from.

(c) Remove and keep the pins [6] and anti-rotation locks [11]. Keep a record of where each part came from.

NOTE: If necessary, adjust the tension of the sling, SPL-2035 to aid removal of the pins [6].

SUBTASK 55-05-03-210-013

(5) Do a general visual inspection of the aft side of right horizontal stabilizer rear spar, including spar chords and webs, terminal fittings, elevator tab actuator support fitting, center section rear spar lugs, elevator hinge ribs, and elevator tab leading edge.

SUBTASK 55-05-03-910-013

(6) Do this task: 737-6789 Basic Task Description, TASK 51-05-01-210-804.

C. Put the Airplane Back to its Usual Condition

SUBTASK 55-05-03-400-005

(1) Install the pins [6] and anti-rotation locks [11].

SUBTASK 55-05-03-400-006

(2) Install bolt [8], bolt [9], washers [7], and washers [3].

SUBTASK 55-05-03-400-007

(3) Install the nuts [5] and washers [4] on bolt [8] and bolt [9].

SUBTASK 55-05-03-420-004

(4) Replace the tab hinge covers.

SUBTASK 55-05-03-420-005

(5) Reconnect the balance panels in the balance bay.

SUBTASK 55-05-03-410-012

(6) Close these access panels:

Number	Name/Location
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334BB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334CB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

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(Continued)

Number	Name/Location
334DB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334EB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344BB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344CB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344DB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344EB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
345A	Horizontal Stabilizer, Removable Stabilizer Tip

Special Access:

Number	Name/Location
S3431	Right Horizontal Stabilizer Trailing Edge

SUBTASK 55-05-03-080-004

(7) Remove the sling, SPL-2035:

(a) Release tension applied to the sling, SPL-2035.

(b) Remove the hoist to the sling, SPL-2035.

(c) Remove the fasteners that attach the sling, SPL-2035.

(d) Install bolts [1] and bolts [2] at the stabilizer top surface.

——— END OF TASK ———

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MECH	INSP

TASK 51-05-01-210-804

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-043

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-044

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-045

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-046

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-111

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-048

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

RIGHT ELEVATOR
(LOWER TRAILING EDGE PANELS REMOVED)

MPD ITEM
55-110-02

A

2077679 S0000434990_V3

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 1 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

2077680 S0000434991_V3

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 2 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

2077681 S0000434992_V3

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 3 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 4 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

MPD ITEM
55-110-02

H

MPD ITEM
55-110-02

2077684 S0000434994_V3

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 5 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

HINGE 1 AND 2
(COVER REMOVED FOR CLARITY)

I

HINGE 3
(HINGE 4, 5 AND 6 ARE EQUIVALENT)

J

MPD ITEM
55-110-02

MPD ITEM
55-110-02

2077688 S0000434995_V3

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 6 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

LEFT HORIZONTAL STABILIZER REAR SPAR CHORDS, WEB, TERMINAL FITTINGS AND TAB ACTUATOR FITTING (PANELS REMOVED FOR CLARITY)

MPD ITEM
55-110-02

K

MPD ITEM
55-110-02

2077690 S0000434996_V3

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 7 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

1 THE FORWARD ATTACH (BLUE) FITTING IS ATTACHED TO THE STABILIZER FRONT SPAR (LEADING EDGE ATTACH BOLTS).

TWO BOLTS ARE USED ON THE INBOARD SIDE OF THE LEADING EDGE RIB (STA 114.55) AND TWO BOLTS ARE USED ON THE OUTBOARD SIDE OF THE LEADING EDGE RIB (STA 114.55).

2 THE AFT OUTBOARD ATTACH (RED) FITTING IS ATTACHED TO THE ELEVATOR HINGE SUPPORT RIBS, FOUR BOLTS ARE USED AT STA 176.64 AND FOUR BOLTS ARE USED AT STA 149.12.

3 THE AFT INBOARD ATTACH (GREEN) FITTING IS ATTACHED TO THE ELEVATOR HINGE SUPPORT RIBS, SIX BOLTS ARE USED AT STA 66.54 AND FOUR BOLTS ARE USED AT STA 39.02.

2438883 S0000565930_V1

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 8 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

REAR SPAR CONNECTION
(ACCESS PANELS AND GAP COVERS REMOVED)

M

2438884 S0000565931_V1

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 9 of 10)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-110-02-01

NOTE: COMBINED GAPS BETWEEN BUSHINGS AT A1 AND A2 SHOULD BE 0.032 INCH (0.81 mm), OR LESS.

O

NOTE: COMBINED GAPS BETWEEN BUSHINGS AT A1 AND A2 SHOULD BE 0.046 INCH (1.17 mm), OR LESS.

N

4 BOLT HEAD DIRECTION OPTIONAL (WASHER NEEDS TO BE INSTALLED AT BOLT HEAD LOCATION).

2439034 S0000565932_V1

Right Elevator Hinge, Actuator, and Tab Mast Arm Fittings and Weight Support Structure - General Visual (Internal)
Figure 1 (Sheet 10 of 10)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT ELEVATOR TAB SUPT FTGS ON FRONT SPAR AND TAB SPAR AT LEADING EDGE			55-115-01-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L ELEVATOR	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2 NOTE	36000 FC	24000 FC	AIRPLANE: ALL ENGINE: ALL
		ACCESS NOTE			ZONE 334 344

Inspect left elevator tab mechanism support fittings on elevator front spar and elevator tab spar at leading edge cutouts.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove upper or lower horizontal stabilizer trailing edge seal and elevator inboard hinge cover panel to inspect support fittings. Remove tab hinge covers to locally inspect forward face of spar.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-01-01

MECH	INSP

TASK 55-05-03-211-801

1. **INTERNAL - DETAILED: LEFT ELEVATOR TAB SUPT FTGS ON FRONT SPAR AND TAB SPAR AT LEADING EDGE**

(Figure 1)

A. Inspection

NOTE: Remove upper or lower horizontal stabilizer trailing edge seal and elevator inboard hinge cover panel to inspect support fittings. Remove tab hinge covers to locally inspect forward face of spar.

SUBTASK 55-05-03-211-001

(1) Do a Detailed inspection of the left elevator tab mechanism support fittings on elevator front spar and elevator tab spar at leading edge cutouts.

SUBTASK 55-05-03-910-014

(2) 737–6789 Basic Task Description, AMM Task 51–05–01–210–804.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-01-01

MECH	INSP

TASK 51-05-01-210-804

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-043

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-044

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-045

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-046

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-111

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-048

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-01-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-01-01

2087570 S0000439782_V2

INTERNAL-DETAILED: LEFT ELEVATOR TAB SUPT FTGS ON FRONT SPAR & TAB SPAR AT LEADING EDGE CUTOUTS

Figure 1 (Sheet 1 of 4)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-01-01

ELEVATOR TAB MECHANISM SUPPORT FITTINGS

B

MPD ITEM
55-115-01

2087556 S0000439787_V2

INTERNAL-DETAILED: LEFT ELEVATOR TAB SUPT FTGS ON FRONT SPAR & TAB SPAR AT LEADING EDGE CUTOUTS

Figure 1 (Sheet 2 of 4)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-01-01

MPD ITEM
55-115-01

C

2087561 S0000439789_V2

INTERNAL-DETAILED: LEFT ELEVATOR TAB SUPT FTGS ON FRONT SPAR & TAB SPAR AT LEADING EDGE CUTOUTS

Figure 1 (Sheet 3 of 4)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-01-01

HINGE 1 AND 2
(COVER REMOVED FOR CLARITY)

D

HINGE 3
(HINGE 4, 5 AND 6 ARE EQUIVALENT)

E

MPD ITEM
55-115-01

2087545 S0000439790_V2

INTERNAL-DETAILED: LEFT ELEVATOR TAB SUPT FTGS ON FRONT SPAR & TAB SPAR AT LEADING EDGE CUTOUTS

Figure 1 (Sheet 4 of 4)

BOEING PROPRIETARY - Copyright © Unpublished Work - See title page for details

AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT ELEVATOR TAB SUPT FTGS ON FRONT SPAR AND TAB SPAR AT LEADING EDGE			55-115-02-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R ELEVATOR	VERSION 1.1	THRESHOLD 12 YR	REPEAT 8 YR	APPLICABILITY
STATION	SKILL AIRPL	1.2 NOTE	36000 FC	24000 FC	AIRPLANE: ALL ENGINE: ALL
		ACCESS NOTE			ZONE 334 344

Inspect right elevator tab mechanism support fittings on elevator front spar and elevator tab spar at leading edge cutouts.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: Remove upper or lower horizontal stabilizer trailing edge seal and elevator inboard hinge cover panel to inspect support fittings. Remove tab hinge covers to locally inspect forward face of spar.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-02-01

MECH	INSP

TASK 55-05-03-211-802

1. INTERNAL - DETAILED: RIGHT ELEVATOR TAB SUPT FTGS ON FRONT SPAR AND TAB SPAR AT LEADING EDGE

(Figure 1)

A. Inspection

NOTE: Remove upper or lower horizontal stabilizer trailing edge seal and elevator inboard hinge cover panel to inspect support fittings. Remove tab hinge covers to locally inspect forward face of spar.

SUBTASK 55-05-03-211-002

(1) Do a Detailed inspection of the right elevator tab mechanism support fittings on elevator front spar and elevator tab spar at leading edge cutouts.

SUBTASK 55-05-03-910-015

(2) 737–6789 Basic Task Description, AMM Task 51–05–01–210–804.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-02-01

MECH	INSP

TASK 51-05-01-210-804

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-043

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-044

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-045

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-046

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-111

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-048

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-049

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-02-01

INTERNAL-DETAILED: RIGHT ELEVATOR TAB SUPT FTGS ON FRONT SPAR & TAB SPAR AT LEADING EDGE CUTOUTS

Figure 1 (Sheet 1 of 4)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-02-01

ELEVATOR TAB MECHANISM SUPPORT FITTINGS

B

MPD ITEM
55-115-02

2087903 S0000439793_V2

INTERNAL-DETAILED: RIGHT ELEVATOR TAB SUPT FTGS ON FRONT SPAR & TAB SPAR AT LEADING EDGE CUTOUTS

Figure 1 (Sheet 2 of 4)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-02-01

MPD ITEM
55-115-02

H

2087944 S0000439794_V2

INTERNAL-DETAILED: RIGHT ELEVATOR TAB SUPT FTGS ON FRONT SPAR & TAB SPAR AT LEADING EDGE CUTOUTS

Figure 1 (Sheet 3 of 4)

EFFECTIVITY	SOURCE	
AKS ALL	**MRB**	**RIGHT ELEVATOR TAB SUPT FTGS ON FRONT SPAR AND TAB SPAR AT LEADING EDGE**
		D633A109-AKS **Page 7 of 8**
		55-115-02-01 **Oct 15/2015**

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-115-02-01

HINGE 1 AND 2
(COVER REMOVED FOR CLARITY)

D

HINGE 3
(HINGE 4, 5 AND 6 ARE EQUIVALENT)

E

MPD ITEM
55-115-02

2087984 S0000439795_V2

INTERNAL-DETAILED: RIGHT ELEVATOR TAB SUPT FTGS ON FRONT SPAR & TAB SPAR AT LEADING EDGE CUTOUTS

Figure 1 (Sheet 4 of 4)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER TRAILING EDGE			55-120-01-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 8 YR	REPEAT 4 YR	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333AB 333AT			ZONE 333
		NOTE			

Inspect left horizontal stabilizer rear spar terminal fittings and center section rear spar lugs.

ACCESS NOTE: Bolt removal is not required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-120-01-01

MECH	INSP

TASK 55-05-03-210-814

1. INTERNAL - GENERAL VISUAL: LEFT HORIZONTAL STABILIZER TRAILING EDGE
(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-013

(1) Open these access panels:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

NOTE: Bolt removal is not required.

SUBTASK 55-05-03-210-014

(2) Do a General Visual inspection of the left horizontal stabilizer rear spar terminal fittings and center section rear spar lugs.

SUBTASK 55-05-03-910-016

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–801.

SUBTASK 55-05-03-410-013

(4) Close this access panel:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(5) Close this access panel:

Number	Name/Location
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

——— END OF TASK ———

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TASK 51-05-01-210-801

MECH	INSP

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-022

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-023

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-024

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-025

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-026

(5) CPCP Basic Task Item 5 is not applicable.

SUBTASK 51-05-01-210-027

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-120-01-01

MECH	INSP

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

Water displacing / anti-corrosion compounds should not be applied in the following areas:

- Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
- Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
- Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
- Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
- Areas with electrical arc potential.
- Interior materials, including cargo liners (change of flammability properties).
- Fiber-glass ducts where temperature exceeds 220 degrees F.
- Selected areas noted in baseline program.

SUBTASK 51-05-01-210-028

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-120-01-01

(BOTTOM VIEW)

(TOP VIEW)

Left Horizontal Stabilizer Trailing Edge General Visual (Internal)
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-120-01-01

Left Horizontal Stabilizer Trailing Edge General Visual (Internal)
Figure 1 (Sheet 2 of 2)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER TRAILING EDGE			55-120-02-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 8 YR	REPEAT 4 YR	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343AB 343AT			ZONE 343
		NOTE			

Inspect right horizontal stabilizer rear spar terminal fittings and center section rear spar lugs.

ACCESS NOTE: Bolt removal is not required.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-120-02-01

MECH	INSP

TASK 55-05-03-210-815

1. INTERNAL - GENERAL VISUAL: RIGHT HORIZONTAL STABILIZER TRAILING EDGE
(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-014

(1) Open these access panels:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

NOTE: Bolt removal is not required.

SUBTASK 55-05-03-210-015

(2) Do a General Visual inspection of the right horizontal stabilizer rear spar terminal fittings and center section rear spar lugs.

SUBTASK 55-05-03-910-017

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–801.

SUBTASK 55-05-03-410-014

(4) Close this access panel:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(5) Close this access panel:

Number	Name/Location
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-120-02-01

MECH	INSP

TASK 51-05-01-210-801

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-022

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-023

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-024

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-025

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-026

(5) CPCP Basic Task Item 5 is not applicable.

SUBTASK 51-05-01-210-027

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-120-02-01

MECH	INSP

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

Water displacing / anti-corrosion compounds should not be applied in the following areas:

- Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
- Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
- Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
- Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
- Areas with electrical arc potential.
- Interior materials, including cargo liners (change of flammability properties).
- Fiber-glass ducts where temperature exceeds 220 degrees F.
- Selected areas noted in baseline program.

SUBTASK 51-05-01-210-028

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-120-02-01

Right Horizontal Stabilizer Trailing Edge General Visual (Internal)
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-120-02-01

487844 S0000146453_V2

Right Horizontal Stabilizer Trailing Edge General Visual (Internal)
Figure 1 (Sheet 2 of 2)

BOEING PROPRIETARY - Copyright © Unpublished Work - See title page for details

AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RUDDER, ELEVATOR AND ELEVATOR TAB ATTACH FITTINGS			55-130-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 12 YR 36000 FC	REPEAT 10 YR 30000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 324AAL 324ABL 324ACL 324ADL 324AEL 324AFL 324AGL 324AHL 324AJL 324AKL 324ALL 324AXL 324QL 324RL 324SL 324TL 324UL 324VL 324WL 324XL 324YL 324ZL 333AB 333BB 333CB 333DB 333EB 333FB 334AB 334PT 343AB 343BB 343CB 343DB 343EB 343FB 344AB 344PT S3001			ZONE 324 325 334 344

Inspect rudder, elevator and elevator tab skin panels, rudder and elevator spars, rudder and elevator inspar ribs, rudder and elevator leading edge skins, rudder and elevator leading edge ribs, and rudder and elevator leading edge spars.

INTERVAL NOTE: Whichever comes first.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

B. Consumable Materials

Reference	Description	Specification
C00174	Compound - Corrosion Preventive, Solvent Cutback, Cold Application	MIL-PRF-16173 (Supersedes MIL-C-16173)
C00915	Compound - Organic Corrosion Inhibiting, Advanced	BMS3-29
G00009	Compound - Organic Corrosion Inhibiting	BMS3-23

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-130-00-01

MECH	INSP

TASK 55-05-03-210-816

1. **INTERNAL - GENERAL VISUAL: RUDDER, ELEVATOR AND ELEVATOR TAB ATTACH FITTINGS**

(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-015

(1) Open these access panels:

Number	Name/Location
324AAL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324ABL	Panel Assy - Trailing Edge, Beam Seal, Vertical Fin
324ACL	Rudder Nose Cover/Rudder Hinge No. 8 At Rudder Station 276.24
324ADL	Rudder Nose Cover/Rudder Hinge No. 7 At Rudder Station 257.92
324AEL	Rudder Nose Cover/Rudder Hinge No. 6 At Rudder Station 239.61
324AFL	Rudder Nose Cover/Rudder Hinge No. 5 At Rudder Station 184.67
324AGL	Upper Rudder Gap Cover At Rudder Station 164.90
324AHL	Rudder Nose Fairing/Rudder Hinge No. 4 At Rudder Station 129.74.
324AJL	Rudder Nose Fairing/Rudder Hinge No. 3 At Rudder Station 70.65
324AKL	Rudder Nose Fairing/Rudder Hinge No. 2 At Rudder Station 53.48
324ALL	Rudder Nose Cover/Rudder Hinge No. 1 At Rudder Station 6.96
324AXL	Lower Rudder Gap Cover
324QL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324RL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324SL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324TL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324UL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324VL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324WL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324XL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324YL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324ZL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334PT	Horizontal Stabilizer, Tab Control Rod Fairing
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

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(Continued)

Number	Name/Location
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344PT	Horizontal Stabilizer, Tab Control Rod Fairing, Elevator Sta 34.0

Special Access:

Number	Name/Location
S3001	Rudder, Elevator, and Elevator Tab Attach Fittings

SUBTASK 55-05-03-210-016

(2) Do a General Visual inspection of the rudder, elevator and elevator tab skin panels, rudder and elevator spars, rudder and elevator inspar ribs, rudder and elevator leading edge skins, rudder and elevator leading edge ribs, and rudder and elevator leading edge spars.

SUBTASK 55-05-03-910-018

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–806.

SUBTASK 55-05-03-410-020

(4) Close this access panel:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(5) Close this access panel:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 55-05-03-410-015

(6) Close these access panels:

Number	Name/Location
324AAL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324ABL	Panel Assy - Trailing Edge, Beam Seal, Vertical Fin
324ACL	Rudder Nose Cover/Rudder Hinge No. 8 At Rudder Station 276.24
324ADL	Rudder Nose Cover/Rudder Hinge No. 7 At Rudder Station 257.92
324AEL	Rudder Nose Cover/Rudder Hinge No. 6 At Rudder Station 239.61
324AFL	Rudder Nose Cover/Rudder Hinge No. 5 At Rudder Station 184.67
324AGL	Upper Rudder Gap Cover At Rudder Station 164.90
324AHL	Rudder Nose Fairing/Rudder Hinge No. 4 At Rudder Station 129.74.
324AJL	Rudder Nose Fairing/Rudder Hinge No. 3 At Rudder Station 70.65
324AKL	Rudder Nose Fairing/Rudder Hinge No. 2 At Rudder Station 53.48
324ALL	Rudder Nose Cover/Rudder Hinge No. 1 At Rudder Station 6.96
324AXL	Lower Rudder Gap Cover
324QL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324RL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324SL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-130-00-01

MECH	INSP

(Continued)

Number	Name/Location
324TL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324UL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324VL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324WL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324XL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324YL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
324ZL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334PT	Horizontal Stabilizer, Tab Control Rod Fairing
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344PT	Horizontal Stabilizer, Tab Control Rod Fairing, Elevator Sta 34.0

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-130-00-01

MECH	INSP

TASK 51-05-01-210-806

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-057

(1) CPCP Basic Task Item 1 is not applicable.

SUBTASK 51-05-01-210-058

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-059

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-060

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-114

(5) Do the CPCP Basic Task Item 5 as follows:

(a) Clear any blocked holes or gaps that may hinder drainage, as applicable.

SUBTASK 51-05-01-210-062

(6) Do the CPCP Basic Task item 6 as follows:

(a) Apply suitable approved water displacing / anti-corrosion compound as necessary.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-130-00-01

MECH	INSP

1) The minimum requirement for all areas (except as noted in CPCP Basic Task 6 (a) 3)) is single coat of water displacing / anti-corrosion compound that penetrates faying surfaces and displaces moisture, e.g. a single coat of compound, C00915 BMS 3-29 or corrosion inhibiting compound, G00009 BMS 3-23, where the initial or previous coat has been disturbed or removed.

2) Not applicable

3) List of areas / items where water displacing/anti-corrosion compounds should not be applied:

   Water displacing / anti-corrosion compounds should not be applied in the following areas:

   - Cables, pulleys, wiring, plastics, elastomers, oxygen systems.
   - Lubricated or Teflon surfaces (E.g. greased joints, sealed bearings).
   - Over compound, C00174 Cosmoline 1058 (or Equivalent per MIL-C-16173 Grade 1).
   - Adjacent to tears / holes in insulation blankets (water repelling characteristics are diminished).
   - Areas with electrical arc potential.
   - Interior materials, including cargo liners (change of flammability properties).
   - Fiber-glass ducts where temperature exceeds 220 degrees F.
   - Selected areas noted in baseline program.

SUBTASK 51-05-01-210-063

(7) CPCP Basic Task Item 7 is not applicable.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-130-00-01

MPD ITEM
55-130-00

2085420 S0000439304_V2

INTERNAL-GENERAL VISUAL: RUDDER, ELEVATOR & ELEVATOR TAB ATTACH FITTINGS
Figure 1 (Sheet 1 of 4)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-130-00-01

2085423 S0000439305_V2

INTERNAL-GENERAL VISUAL: RUDDER, ELEVATOR & ELEVATOR TAB ATTACH FITTINGS
Figure 1 (Sheet 2 of 4)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-130-00-01

MPD ITEM
55-130-00

2085426 S0000439306_V2

INTERNAL-GENERAL VISUAL: RUDDER, ELEVATOR & ELEVATOR TAB ATTACH FITTINGS
Figure 1 (Sheet 3 of 4)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-130-00-01

ELEVATOR
(LEFT SIDE IS SHOWN, RIGHT SIDE IS OPPOSITE)

D

MPD ITEM
55-130-00

2085530 S0000439307_V2

INTERNAL-GENERAL VISUAL: RUDDER, ELEVATOR & ELEVATOR TAB ATTACH FITTINGS
Figure 1 (Sheet 4 of 4)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT ELEVATOR HINGE, ACTUATOR, AND TAB MAST ARM FTGS AND BALANCE WT SUPPT STRUCT.			55-135-01-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L ELEVATOR	VERSION 1.1 1.2 NOTE	THRESHOLD 8 YR 18000 FC	REPEAT 8 YR 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333BB 333CB 333DB 333EB 333FB 334GB 334HB 334JB 334KB 334LB 334MB 334NB NOTE			ZONE 334

Inspect the left elevator hinge fittings, left elevator actuator fittings, left elevator balance weight support structure, left elevator tab mast arm fitting and left elevator tab hinge fittings.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: After removal of access panels, disconnect balance bays. Alternative - Remove elevator from horizontal stabilizer. For elevator tab mast arm fittings, remove fairing on upper surface of elevator to gain access.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-01-01

MECH	INSP

TASK 55-05-03-211-807

1. **INTERNAL - DETAILED: LEFT ELEVATOR HINGE, ACTUATOR, AND TAB MAST ARM FTGS AND BALANCE WT SUPPT STRUC**

(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-016

(1) Open these access panels:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334HB	Horizontal Stabilizer, Elevator Hinge Cover
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover

NOTE: Remove lower trailing edge panels and disconnect balance bay panels. Remove elevator hinge cover plates. Alternative - Remove elevator from horizontal stabilizer. For elevator tab mast arm fittings, remove fairing on upper surface of elevator to gain access.

SUBTASK 55-05-03-211-007

(2) Do a Detailed inspection of the left elevator hinge fittings, left elevator actuator fittings, left elevator balance weight support structure, left elevator tab mast arm fitting and left elevator tab hinge fittings.

SUBTASK 55-05-03-910-019

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–803.

SUBTASK 55-05-03-410-016

(4) Close these access panels:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334HB	Horizontal Stabilizer, Elevator Hinge Cover
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-01-01

MECH	INSP

(Continued)

Number	Name/Location
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover

END OF TASK

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MECH	INSP

TASK 51-05-01-210-803

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-036

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-037

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-038

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-039

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-040

(5) CPCP Basic Task Item 5 is not applicable.

SUBTASK 51-05-01-210-110

(6) Do the CPCP Basic Task Item 6 (Not Applicable)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-01-01

MECH	INSP

SUBTASK 51-05-01-210-042

(7) CPCP Basic Task Item 7 is not applicable.

END OF TASK

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-01-01

LEFT ELEVATOR
(LOWER TRAILING EDGE PANELS REMOVED)

A

MPD ITEM
55-135-01

D46120 S0000158663_V4

Lft Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 1 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-01-01

Lft Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 2 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-01-01

MPD ITEM
55-135-01

D46124 S0000158665_V3

Lft Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 3 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-01-01

Lft Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 4 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-01-01

MPD ITEM
55-135-01

H

D46127 S0000158667_V4

Lft Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 5 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-01-01

HINGE 1 AND 2
(COVER REMOVED FOR CLARITY)

I

HINGE 3
(HINGE 4, 5 AND 6 ARE EQUIVALENT)

MPD ITEM
55-135-01

J

D46129 S0000158668_V4

Lft Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 6 of 6)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT ELEVATOR HINGE, ACTUATOR, AND TAB MAST ARM FTGS AND BALANCE WT SUPPT STRUCT.			55-135-02-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R ELEVATOR	VERSION 1.1	THRESHOLD 8 YR	REPEAT 8 YR	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	18000 FC	18000 FC	
		ACCESS 343BB 343CB 343DB 343EB 343FB 344GB 344HB 344JB 344KB 344LB 344MB 344NB NOTE			ZONE 344

Inspect the right elevator hinge fittings, right elevator actuator fittings, right elevator balance weight support structure, right elevator tab mast arm fitting and right elevator tab hinge fittings.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: After removal of access panels, disconnect balance bays. Alternative - Remove elevator from horizontal stabilizer. For elevator tab mast arm fittings, remove fairing on upper surface of elevator to gain access.

A. References

Reference	Title
AMM 51-00-58	STANDARD TREATMENT METHODS

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-02-01

MECH	INSP

TASK 55-05-03-211-808

1. **INTERNAL - DETAILED: RIGHT ELEVATOR HINGE, ACTUATOR, AND TAB MAST ARM FTGS AND BALANCE WT SUPPT STRUCT**

(Figure 1)

A. Inspection

SUBTASK 55-05-03-010-017

(1) Open these access panels:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344HB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 66.54
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

NOTE: Remove lower trailing edge panels and disconnect balance bay panels. Remove elevator hinge cover plates. Alternative - Remove elevator from horizontal stabilizer. For elevator tab mast arm fittings, remove fairing on upper surface of elevator to gain access.

SUBTASK 55-05-03-211-008

(2) Do a Detailed inspection of the right elevator hinge fittings, right elevator actuator fittings, right elevator balance weight support structure, right elevator tab mast arm fitting and right elevator tab hinge fittings.

SUBTASK 55-05-03-910-020

(3) 737–6789 Basic Task Description, AMM Task 51–05–01–210–803.

SUBTASK 55-05-03-410-017

(4) Close these access panels:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344HB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 66.54
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-02-01

MECH	INSP

(Continued)

Number	Name/Location
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

END OF TASK

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MECH	INSP

TASK 51-05-01-210-803

2. 737-6789 Basic Task Description

A. CPCP Basic Task

SUBTASK 51-05-01-210-036

(1) Do the CPCP Basic Task Item 1 as follows:

(a) Remove all systems, equipment and interior furnishings, etc. (e.g. toilets, galleys, linings, installation) as necessary to accomplish CPCP Basic Task Item 3. It is not necessary to remove bushings unless specified in the Task Description, or if there is an indication of corrosion, or that the bushing has migrated.

SUBTASK 51-05-01-210-037

(2) Do the CPCP Basic Task Item 2 as follows:

(a) Prior to inspection clean the area as required to accomplish CPCP Basic Task Item 3. It is not necessary to remove normal amounts of sealant/leveling compound unless it has deteriorated to the point where moisture can penetrate down to the metal. A light uniform film of Corrosion Inhibiting Compound (CIC) that has not accumulated dirt or debris, will normally allow adequate inspection of the structure without removal. CIC may require removal if there are multiple layers and/or accumulations of dirt or debris.

SUBTASK 51-05-01-210-038

(3) Do the CPCP Basic Task Item 3 as follows:

(a) Visually inspect all structure listed in the task description. The inspection method is as specified in each task description. Use Additional non-destructive inspections or visual inspections following partial disassembly if there are indications of hidden corrosion, such as bulging skins or corrosion running into splices, or under fittings, etc. In the task area, check the integrity of any sealant/leveling compound to determine if removal is required, and any corrosion inhibiting compound, particularly at faying surfaces, to determine if additional application is required per CPCP Basic Task Item 6.

SUBTASK 51-05-01-210-039

(4) Do the CPCP Basic Task item 4 as follows:

(a) Remove all corrosion, evaluate damage and repair or replace all discrepant structure as required, including application of protective finishes per STANDARD TREATMENT METHODS, AMM SUBJECT 51-00-58, or 737 Structural Repair Manual (SRM) D634A200, (-600), D634A201 (-700), D634A210 (-800), D634A211(-900), D634A333 (BBJ), or related service bulletin, as appropriate. Surface oxidation of ferrous metal fasteners may be handled by normal or existing maintenance practices.

SUBTASK 51-05-01-210-040

(5) CPCP Basic Task Item 5 is not applicable.

SUBTASK 51-05-01-210-110

(6) Do the CPCP Basic Task Item 6 (Not Applicable)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-02-01

MECH	INSP

SUBTASK 51-05-01-210-042

(7) CPCP Basic Task Item 7 is not applicable.

END OF TASK

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-02-01

RIGHT ELEVATOR
(LOWER TRAILING EDGE PANELS REMOVED)

MPD ITEM
55-135-02

A

D46151 S0000158670_V4

Rt. Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 1 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-02-01

D46183 S0000158671_V3

Rt. Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 2 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-02-01

D46207 S0000158672_V3

Rt. Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 3 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-02-01

Rt. Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 4 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-02-01

MPD ITEM
55-135-02

D46375 S0000158674_V4

Rt. Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 5 of 6)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-135-02-01

HINGE 1 AND 2
(COVER REMOVED FOR CLARITY)

I

HINGE 3
(HINGE 4, 5 AND 6 ARE EQUIVALENT)

J

MPD ITEM
55-135-02

D46393 S0000158675_V4

Rt. Ele. Hinge, Actuator, and Tab Mast Arm Ftgs and Balance WT Suppt
Figure 1 (Sheet 6 of 6)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-600-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	56000 FC	4000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		311BL			313
		NOTE			

Inspect (Ultrasonic) the spar chord between the web and shear tie at BL 1.3 on the left and right hand sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-1, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-06.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-600-00-01

MECH	INSP

TASK 55-05-02-130-801

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-001

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-001

(2) Do an Ultrasonic inspection of the spar chord between the web and shear tie at BL 1.3 on the left and right hand sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-1, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 4, Subject 55–10–06.

SUBTASK 55-05-02-410-001

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-600-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (Ultrasonic) the spar chord between the web and shear tie at BL 1.3 on the left and right hand sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-1, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-06.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-600-00-02

MECH	INSP

TASK 55-05-02-130-801

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-001

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-001

(2) Do an Ultrasonic inspection of the spar chord between the web and shear tie at BL 1.3 on the left and right hand sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-1, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 4, Subject 55–10–06.

SUBTASK 55-05-02-410-001

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-601-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (High Frequency Eddy Current) the upper chord around the fasteners common to the web at BL 2.7 and BL 6.7 on the left and right hand sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-16.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-601-00-01

MECH	INSP

TASK 55-05-02-250-801

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-018

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-001

(2) Do a High Frequency Eddy Current inspection of the upper chord and around the fasteners common to the web at BL 2.7 and BL 6.7 on the left and right hand sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 6, Subject 55–10–16.

SUBTASK 55-05-02-410-018

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-601-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (High Frequency Eddy Current) the upper chord around the fasteners common to the web at BL 2.7 and BL 6.7 on the left and right hand sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-16.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-601-00-02

MECH	INSP

TASK 55-05-02-250-801

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-018

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-001

(2) Do a High Frequency Eddy Current inspection of the upper chord and around the fasteners common to the web at BL 2.7 and BL 6.7 on the left and right hand sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 6, Subject 55–10–16.

SUBTASK 55-05-02-410-018

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-602-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar upper chord from BL 8.1 to BL 19.7 on the left and right sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-22.

ACCESS NOTE: Removal of fasteners from both LBL and RBL 8.1 to 19.7 is required.

Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-602-00-01

MECH	INSP

TASK 55-05-02-250-802

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-019

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Removal of fasteners from both LBL and RBL 8.1 to 19.7 is required. Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-002

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord from BL 8.1 to BL 19.7 on the left and right sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-3, for alternative inspection.

The NDT method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 6, Subject 55–10–22.

SUBTASK 55-05-02-410-019

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-602-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar upper chord from BL 8.1 to BL 19.7 on the left and right sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-22.

ACCESS NOTE: Removal of fasteners from both LBL and RBL 8.1 to 19.7 is required.

Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-602-00-02

MECH	INSP

TASK 55-05-02-250-802

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-019

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Removal of fasteners from both LBL and RBL 8.1 to 19.7 is required. Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-002

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord from BL 8.1 to BL 19.7 on the left and right sides.

See Doc. D626A001 - DTR, DTR check form 55-10-01-3, for alternative inspection.

The NDT method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 6, Subject 55–10–22.

SUBTASK 55-05-02-410-019

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-603-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (Ultrasonic) the rear spar forward flange at the four Texas Star attachment points.

See Doc. D626A001 - DTR, DTR check form 55-10-01-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-07.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-603-00-01

MECH	INSP

TASK 55-05-02-130-802

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-002

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-002

(2) Do an Ultrasonic inspection of the rear spar forward flange at the four Texas Star attachment points.

See Doc. D626A001 - DTR, DTR check form 55-10-01-4, for alternative inspection.

The NDT method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 4, Subject 55–10–07.

SUBTASK 55-05-02-410-002

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-603-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (Ultrasonic) the rear spar forward flange at the four Texas Star attachment points.

See Doc. D626A001 - DTR, DTR check form 55-10-01-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-07.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-603-00-02

MECH	INSP

TASK 55-05-02-130-802

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-002

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-002

(2) Do an Ultrasonic inspection of the rear spar forward flange at the four Texas Star attachment points.

See Doc. D626A001 - DTR, DTR check form 55-10-01-4, for alternative inspection.

The NDT method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 4, Subject 55–10–07.

SUBTASK 55-05-02-410-002

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION PIVOT FITTING			55-604-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	56000 FC	10000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		311BL			313
		NOTE			

Inspect (Ultrasonic) the pivot fittings on the horizontal stabilizer center section hinge house pivot lug.

See Doc. D626A001 - DTR, DTR check form 55-10-02-1, for alternative inspection.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-604-00-01

MECH	INSP

TASK 55-05-02-130-804

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PIVOT FITTING

A. Inspection

SUBTASK 55-05-02-010-072

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-004

(2) Do an Ultrasonic inspection of the pivot fittings on the horizontal stabilizer center section hinge house pivot lug.

See Doc. D626A001 - DTR, DTR check form 55-10-02-1, for alternative inspection.

SUBTASK 55-05-02-410-070

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION PIVOT FITTING			55-604-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 10000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (Ultrasonic) the pivot fittings on the horizontal stabilizer center section hinge house pivot lug.

See Doc. D626A001 - DTR, DTR check form 55-10-02-1, for alternative inspection.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-604-00-02

MECH	INSP

TASK 55-05-02-130-804

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PIVOT FITTING

A. Inspection

SUBTASK 55-05-02-010-072

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-004

(2) Do an Ultrasonic inspection of the pivot fittings on the horizontal stabilizer center section hinge house pivot lug.

See Doc. D626A001 - DTR, DTR check form 55-10-02-1, for alternative inspection.

SUBTASK 55-05-02-410-070

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION PIVOT FITTING			55-605-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (Ultrasonic) the pivot fitting plates at the lower hinge housing lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-02-2, for alternative inspection.

ACCESS NOTE: Removal of gap covers and sliding seals is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-605-00-01

MECH	INSP

TASK 55-05-02-130-824

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PIVOT FITTING

A. Inspection

SUBTASK 55-05-02-130-026

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Removal of gap covers and sliding seals is required.

SUBTASK 55-05-02-130-027

(2) Do an Ultrasonic inspection of the pivot fitting plates at the lower hinge housing lugs. See Doc. D626A001 - DTR, DTR check form 55-10-02-2, for alternative inspection.

SUBTASK 55-05-02-130-028

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION PIVOT FITTING			55-605-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (Ultrasonic) the pivot fitting plates at the lower hinge housing lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-02-2, for alternative inspection.

ACCESS NOTE: Removal of gap covers and sliding seals is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-605-00-02

MECH	INSP

TASK 55-05-02-130-824

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PIVOT FITTING

A. Inspection

SUBTASK 55-05-02-130-026

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Removal of gap covers and sliding seals is required.

SUBTASK 55-05-02-130-027

(2) Do an Ultrasonic inspection of the pivot fitting plates at the lower hinge housing lugs. See Doc. D626A001 - DTR, DTR check form 55-10-02-2, for alternative inspection.

SUBTASK 55-05-02-130-028

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION THRUST LINK FITTINGS			55-606-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	56000 FC	1600 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		311BL			313
		NOTE			

Inspect (Ultrasonic) the fitting lugs on the horizontal stabilizer center section thrust link fittings. There are 4 lugs per fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-02-3, for alternative inspection.

ACCESS NOTE: Remove fittings to inspect the fitting lugs, 4 lugs per fitting.

Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-606-00-01

MECH	INSP

TASK 55-05-02-130-819

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST LINK FITTINGS

A. Inspection

SUBTASK 55-05-02-010-105

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Remove fittings to inspect the fitting lugs, 4 lugs per fitting. Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-023

(2) Do an Ultrasonic inspection of the fitting lugs on the horizontal stabilizer center section thrust link fittings. There are 4 lugs per fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-02-3, for alternative inspection.

SUBTASK 55-05-02-410-103

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION THRUST LINK FITTINGS			55-606-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 1600 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (Ultrasonic) the fitting lugs on the horizontal stabilizer center section thrust link fittings. There are 4 lugs per fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-02-3, for alternative inspection.

ACCESS NOTE: Remove fittings to inspect the fitting lugs, 4 lugs per fitting.

Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-606-00-02

MECH	INSP

TASK 55-05-02-130-819

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST LINK FITTINGS

A. Inspection

SUBTASK 55-05-02-010-105

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Remove fittings to inspect the fitting lugs, 4 lugs per fitting. Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-023

(2) Do an Ultrasonic inspection of the fitting lugs on the horizontal stabilizer center section thrust link fittings. There are 4 lugs per fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-02-3, for alternative inspection.

SUBTASK 55-05-02-410-103

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION JACKSCREW FITTING			55-607-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 311
		NOTE			

Inspect (Ultrasonic) the horizontal stabilizer center section jackscrew fitting lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-03-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-04.

ACCESS NOTE: Removal of bolts and bushings required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-607-00-01

MECH	INSP

TASK 55-05-02-130-823

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION JACKSCREW FITTING

A. Inspection

SUBTASK 55-05-02-010-123

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Removal of bolts and bushings required.

SUBTASK 55-05-02-130-024

(2) Do an Ultrasonic inspection of the horizontal stabilizer center section jackscrew fitting lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-03-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-04.

SUBTASK 55-05-02-410-121

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION JACKSCREW FITTING			55-607-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 312
		NOTE			

Inspect (Ultrasonic) the horizontal stabilizer center section jackscrew fitting lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-03-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-04.

ACCESS NOTE: Removal of bolts and bushings required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-607-00-02

MECH	INSP

TASK 55-05-02-130-823

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION JACKSCREW FITTING

A. Inspection

SUBTASK 55-05-02-010-123

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Removal of bolts and bushings required.

SUBTASK 55-05-02-130-024

(2) Do an Ultrasonic inspection of the horizontal stabilizer center section jackscrew fitting lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-03-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-04.

SUBTASK 55-05-02-410-121

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-608-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
					332 333

Inspect (High Frequency Eddy Current) the visible portion of the lower chord between stabilizer STA 129.5 and stabilizer STA 212.3.

See Doc. D626A001 - DTR, DTR check form 55-10-04-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-27.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-608-00-01

MECH	INSP

TASK 55-05-02-250-806

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-250-006

(1) Do a High Frequency Eddy Current inspection of the visible portion of the lower chord between stabilizer STA 129.5 and stabilizer STA 212.3.

See Doc. D626A001 - DTR, DTR check form 55-10-04-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 6, Subject 55–10–27.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-608-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (High Frequency Eddy Current) the visible portion of the lower chord between stabilizer STA 129.5 and stabilizer STA 212.3.

See Doc. D626A001 - DTR, DTR check form 55-10-04-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-27.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-608-00-02

MECH	INSP

TASK 55-05-02-250-806

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-250-006

(1) Do a High Frequency Eddy Current inspection of the visible portion of the lower chord between stabilizer STA 129.5 and stabilizer STA 212.3.

See Doc. D626A001 - DTR, DTR check form 55-10-04-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6–37239). The inspection procedures are contained in Part 6, Subject 55–10–27.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-609-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
					332 333

Inspect (Low Frequency Eddy Current) both fastener rows in the FWD flange at, and between, the ribs at stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-29.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-609-00-01

MECH	INSP

TASK 55-05-02-250-807

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-250-007

(1) Do a Low Frequency Eddy Current inspection of both fastener rows in the FWD flange at, and between, the ribs at stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-29.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-609-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (Low Frequency Eddy Current) both fastener rows in the FWD flange at, and between, the ribs at stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-29.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-609-00-02

MECH	INSP

TASK 55-05-02-250-807

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-250-007

(1) Do a Low Frequency Eddy Current inspection of both fastener rows in the FWD flange at, and between, the ribs at stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-29.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-610-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		333DB 333EB 333FB			332 333
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar lower chord AFT flange, at and between the ribs, from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-3 for alternative inspection.

ACCESS NOTE: Inspection requires removal of the Trailing Edge (TE) panel and fasteners.
The TE panels are attached by nutplates.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-610-00-01

MECH	INSP

TASK 55-05-02-250-805

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-068

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Inspection requires removal of the Trailing Edge (TE) panel and fasteners. The TE panels are attached by nutplates.

SUBTASK 55-05-02-250-005

(2) Do a High Frequency Eddy Current inspection of the rear spar lower chord AFT flange, at and between the ribs, from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-3 for alternative inspection.

SUBTASK 55-05-02-410-066

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-610-00-02
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	R HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		343DB 343EB 343FB			342 343
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar lower chord AFT flange, at and between the ribs, from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-3 for alternative inspection.

ACCESS NOTE: Inspection requires removal of the Trailing Edge (TE) panel and fasteners.
The TE panels are attached by nutplates.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-610-00-02

MECH	INSP

TASK 55-05-02-250-805

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-068

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Inspection requires removal of the Trailing Edge (TE) panel and fasteners. The TE panels are attached by nutplates.

SUBTASK 55-05-02-250-005

(2) Do a High Frequency Eddy Current inspection of the rear spar lower chord AFT flange, at and between the ribs, from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-3 for alternative inspection.

SUBTASK 55-05-02-410-066

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333DB 333EB 333FB			ZONE 332 333

Inspect (Ultrasonic) the rear spar lower chord web flange at the Trailing Edge (TE) ribs and stiffeners from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-4, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-00-01

MECH	INSP

TASK 55-05-02-130-803

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-070

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-130-003

(2) Do an Ultrasonic inspection of the rear spar lower chord web flange at the Trailing Edge (TE) ribs and stiffeners from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-4, for alternative inspection.

SUBTASK 55-05-02-410-068

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343DB 343EB 343FB			ZONE 342 343

Inspect (Ultrasonic) the rear spar lower chord web flange at the Trailing Edge (TE) ribs and stiffeners from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-4, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-00-02

MECH	INSP

TASK 55-05-02-130-803

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-070

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-130-003

(2) Do an Ultrasonic inspection of the rear spar lower chord web flange at the Trailing Edge (TE) ribs and stiffeners from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-4, for alternative inspection.

SUBTASK 55-05-02-410-068

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-10-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333DB 333EB 333FB			ZONE 332 333

Inspect (High Frequency Eddy Current) the rear spar lower chord web flange between the Trailing Edge (TE) ribs from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-5, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-10-01

MECH	INSP

TASK 55-05-02-250-845

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-077

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-045

(2) Do a High Frequency Eddy Current inspection of the rear spar lower chord web flange between the Trailing Edge (TE) ribs from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-5, for alternative inspection.

SUBTASK 55-05-02-410-075

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-10-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343DB 343EB 343FB			ZONE 342 343

Inspect (High Frequency Eddy Current) the rear spar lower chord web flange between the Trailing Edge (TE) ribs from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-5, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-10-02

MECH	INSP

TASK 55-05-02-250-845

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-077

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-045

(2) Do a High Frequency Eddy Current inspection of the rear spar lower chord web flange between the Trailing Edge (TE) ribs from stabilizer STA 212.3 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-5, for alternative inspection.

SUBTASK 55-05-02-410-075

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-12-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	24000 FC	AIRPLANE: ALL; ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
					332 333

Inspect (Low Frequency Eddy Current) both the fastener rows on the rear spar lower chord FWD flange, at and between the ribs, from stabilizer STA 310.54 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-56.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-12-01

MECH	INSP

TASK 55-05-02-250-846

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-250-046

(1) Do a Low Frequency Eddy Current inspection of both the fastener rows on the rear spar lower chord FWD flange, at and between the ribs, from stabilizer STA 310.54 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-56.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-12-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (Low Frequency Eddy Current) both the fastener rows on the rear spar lower chord FWD flange, at and between the ribs, from stabilizer STA 310.54 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-56.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-12-02

MECH	INSP

TASK 55-05-02-250-846

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-250-046

(1) Do a Low Frequency Eddy Current inspection of both the fastener rows on the rear spar lower chord FWD flange, at and between the ribs, from stabilizer STA 310.54 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-56.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-14-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	15000 FC	AIRPLANE: ALL; ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
					332 333

Inspect (Low Frequency Eddy Current) the rear spar lower chord aft flange, at and between the ribs, from stabilizer STA 310.54 to outboard tip.

See Doc. D626A001-DTR, DTR check form 55-10-04-7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-55.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-14-01

MECH	INSP

TASK 55-05-02-250-815

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-250-015

(1) Do a Low Frequency Eddy Current inspection of the rear spar lower chord aft flange, at and between the ribs, from stabilizer STA 310.54 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04–7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-55.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-14-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (Low Frequency Eddy Current) the rear spar lower chord aft flange, at and between the ribs, from stabilizer STA 310.54 to outboard tip.

See Doc. D626A001-DTR, DTR check form 55-10-04-7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-55.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-14-02

MECH	INSP

TASK 55-05-02-250-815

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-250-015

(1) Do a Low Frequency Eddy Current inspection of the rear spar lower chord aft flange, at and between the ribs, from stabilizer STA 310.54 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04–7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-55.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-16-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	24000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		332MB 332NB 332PB			332 333

Inspect (Low Frequency Eddy Current) the rear spar lower chord from stabilizer STA 310.54 to outboard tip web flange, at and between, the ribs.

See Doc. D626A001 - DTR, DTR check form 55-10-04-8, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-50.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-16-01

MECH	INSP

TASK 55-05-02-250-847

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-079

(1) Open these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Open these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

SUBTASK 55-05-02-250-047

(2) Do a Low Frequency Eddy Current inspection of the rear spar lower chord from stabilizer STA 310.54 to outboard tip web flange, at and between, the ribs.

See Doc. D626A001–DTR, DTR check form 55–10–04–8, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55–10–50.

SUBTASK 55-05-02-410-077

(3) Close these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Close these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-16-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342MB 342NB 342PB			ZONE 342 343

Inspect (Low Frequency Eddy Current) the rear spar lower chord from stabilizer STA 310.54 to outboard tip web flange, at and between, the ribs.

See Doc. D626A001 - DTR, DTR check form 55-10-04-8, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-50.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-16-02

MECH	INSP

TASK 55-05-02-250-847

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-079

(1) Open these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Open these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

SUBTASK 55-05-02-250-047

(2) Do a Low Frequency Eddy Current inspection of the rear spar lower chord from stabilizer STA 310.54 to outboard tip web flange, at and between, the ribs.

See Doc. D626A001–DTR, DTR check form 55–10–04–8, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55–10–50.

SUBTASK 55-05-02-410-077

(3) Close these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Close these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-18-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		333EB 333FB			332 333

Inspect (High Frequency Eddy Current) the rear spar lower chord at Bay 21- AFT flange at the hinge, from stabilizer STA 242 to stabilizer STA 247, and elevator STA 197 to elevator STA 203.

See Doc. D626A001 - DTR, DTR check form 55-10-04-9, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-52.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-18-01

MECH	INSP

TASK 55-05-02-250-848

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-081

(1) Open these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-082

(2) Do a High Frequency Eddy Current inspection of the rear spar lower chord at Bay 21-AFT flange at the hinge, from stabilizer STA 242 to stabilizer STA 247, and elevator STA 197 to elevator STA 203.

See Doc. D626A001– DTR, DTR check form 55–10–04–9, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-52.

SUBTASK 55-05-02-410-079

(3) Close these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-611-18-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343EB 343FB			ZONE 342 343

Inspect (High Frequency Eddy Current) the rear spar lower chord at Bay 21- AFT flange at the hinge, from stabilizer STA 242 to stabilizer STA 247, and elevator STA 197 to elevator STA 203.

See Doc. D626A001 - DTR, DTR check form 55-10-04-9, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-52.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-18-02

MECH	INSP

TASK 55-05-02-250-848

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-081

(1) Open these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-082

(2) Do a High Frequency Eddy Current inspection of the rear spar lower chord at Bay 21-AFT flange at the hinge, from stabilizer STA 242 to stabilizer STA 247, and elevator STA 197 to elevator STA 203.

See Doc. D626A001– DTR, DTR check form 55–10–04–9, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-52.

SUBTASK 55-05-02-410-079

(3) Close these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-20-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 332 333

Inspect (High Frequency Eddy Current) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to all four bolt locations common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-27.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-20-01

MECH	INSP

TASK 55-05-02-250-849

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY

A. Inspection

SUBTASK 55-05-02-250-049

(1) Do a High Frequency Eddy Current inspection of the rear spar upper chord between the terminal fitting fork.

NOTE: Inspection is applicable to all four bolt locations common to the chord and terminal fitting.

(2) See Doc. D626A001 - DTR, DTR check form 55-10-04-10A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-27.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-20-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (High Frequency Eddy Current) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to all four bolt locations common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-27.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-20-02

MECH	INSP

TASK 55-05-02-250-849

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY

A. Inspection

SUBTASK 55-05-02-250-049

(1) Do a High Frequency Eddy Current inspection of the rear spar upper chord between the terminal fitting fork.

NOTE: Inspection is applicable to all four bolt locations common to the chord and terminal fitting.

(2) See Doc. D626A001 - DTR, DTR check form 55-10-04-10A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-27.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-22-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3550 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS NOTE			ZONE 332 333

Inspect (High Frequency Eddy Current) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting.

See Doc. D626A001-DTR, DTR check form 55-10-04-10B, for alternative inspection.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Sealant removal is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-22-01

MECH	INSP

TASK 55-05-02-250-804

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY

A. Inspection

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Sealant removal is required.

SUBTASK 55-05-02-250-004

(1) Do a High Frequency Eddy Current inspection of the rear spar upper chord between the terminal fitting fork.

NOTE: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting.

See Doc. D626A001-DTR, DTR check form 55-10-04-10B, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-22-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3550 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS NOTE			ZONE 342 343

Inspect (High Frequency Eddy Current) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting.

See Doc. D626A001-DTR, DTR check form 55-10-04-10B, for alternative inspection.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Sealant removal is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-22-02

MECH	INSP

TASK 55-05-02-250-804

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY

A. Inspection

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Sealant removal is required.

SUBTASK 55-05-02-250-004

(1) Do a High Frequency Eddy Current inspection of the rear spar upper chord between the terminal fitting fork.

NOTE: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting.

See Doc. D626A001-DTR, DTR check form 55-10-04-10B, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-23-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3550 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332EB 333BB			ZONE 332 333
		NOTE			

Inspect (Ultrasonic) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10B, for alternative inspection.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Sealant removal is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-23-01

MECH	INSP

TASK 55-05-02-130-814

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY

A. Inspection

SUBTASK 55-05-02-010-085

(1) Open these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Sealant removal is required.

SUBTASK 55-05-02-130-025

(2) Do an Ultrasonic inspection of the rear spar upper chord between the terminal fitting fork.

NOTE: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10B, for alternative inspection.

SUBTASK 55-05-02-410-083

(3) Close these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-23-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3550 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342EB 343BB			ZONE 342 343
		NOTE			

Inspect (Ultrasonic) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10B, for alternative inspection.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Sealant removal is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-23-02

MECH	INSP

TASK 55-05-02-130-814

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY

A. Inspection

SUBTASK 55-05-02-010-085

(1) Open these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Sealant removal is required.

SUBTASK 55-05-02-130-025

(2) Do an Ultrasonic inspection of the rear spar upper chord between the terminal fitting fork.

NOTE: Inspection is applicable to the outboard three bolt locations common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10B, for alternative inspection.

SUBTASK 55-05-02-410-083

(3) Close these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-24-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 12000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 332 333

Inspect (High Frequency Eddy Current) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to the inboard bolt location common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10C, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-24-01

MECH	INSP

TASK 55-05-02-250-850

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE OF BODY

A. Inspection

SUBTASK 55-05-02-250-050

(1) Do a High Frequency Eddy Current inspection of the rear spar upper chord between the terminal fitting fork.

NOTE: Inspection is applicable to the inboard bolt location common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10C, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-24-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 12000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (High Frequency Eddy Current) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to the inboard bolt location common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10C, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-24-02

MECH	INSP

TASK 55-05-02-250-850

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE OF BODY

A. Inspection

SUBTASK 55-05-02-250-050

(1) Do a High Frequency Eddy Current inspection of the rear spar upper chord between the terminal fitting fork.

NOTE: Inspection is applicable to the inboard bolt location common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10C, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-25-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 12000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 332 333

Inspect (Ultrasonic) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to the inboard bolt location common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10C, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-25-01

MECH	INSP

TASK 55-05-02-130-816

1. **EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY**

A. Inspection

SUBTASK 55-05-02-130-016

(1) Do an Ultrasonic inspection of the rear spar upper chord between the terminal fitting fork.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10C, for alternative inspection.

NOTE: Inspection is applicable to the inboard bolt location common to the chord and terminal fitting.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY			55-611-25-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 12000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (Ultrasonic) the rear spar upper chord between the terminal fitting fork.

Note: Inspection is applicable to the inboard bolt location common to the chord and terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10C, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-25-02

MECH	INSP

TASK 55-05-02-130-816

1. **EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD AT SIDE-OF-BODY**

A. Inspection

SUBTASK 55-05-02-130-016

(1) Do an Ultrasonic inspection of the rear spar upper chord between the terminal fitting fork.

See Doc. D626A001 - DTR, DTR check form 55-10-04-10C, for alternative inspection.

NOTE: Inspection is applicable to the inboard bolt location common to the chord and terminal fitting.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-26-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 332 333

Inspect (Low Frequency Eddy Current) both the forward and aft rear spar upper chord flanges from stabilizer STA 67.78 to stabilizer STA 203.10.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11A, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-26-01

MECH	INSP

TASK 55-05-02-250-851

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-051

(1) Do a Low Frequency Eddy Current inspection of both the forward and aft rear spar upper chord flanges from stabilizer STA 67.78 to stabilizer STA 203.10.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11A, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-26-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (Low Frequency Eddy Current) both the forward and aft rear spar upper chord flanges from stabilizer STA 67.78 to stabilizer STA 203.10.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11A, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-26-02

MECH	INSP

TASK 55-05-02-250-851

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-051

(1) Do a Low Frequency Eddy Current inspection of both the forward and aft rear spar upper chord flanges from stabilizer STA 67.78 to stabilizer STA 203.10.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11A, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-27-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	3500 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
					332 333

Inspect (High Frequency Eddy Current) the rear spar upper cord from stabilizer STA 67.78 to stabilizer STA 203.10.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11A, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-27-01

MECH	INSP

TASK 55-05-02-250-852

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-052

(1) Do a High Frequency Eddy Current inspection of the rear spar upper cord from stabilizer STA 67.78 to stabilizer STA 203.10.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11A, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-27-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3500 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (High Frequency Eddy Current) the rear spar upper cord from stabilizer STA 67.78 to stabilizer STA 203.10.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11A, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-27-02

MECH	INSP

TASK 55-05-02-250-852

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-052

(1) Do a High Frequency Eddy Current inspection of the rear spar upper cord from stabilizer STA 67.78 to stabilizer STA 203.10.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11A, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-28-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 332 333

Inspect (Low Frequency Eddy Current) both the forward and aft rear spar upper chord skin flanges from stabilizer STA 203.10 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11B, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-28-01

MECH	INSP

TASK 55-05-02-250-853

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-053

(1) Do a Low Frequency Eddy Current inspection of both the forward and aft rear spar upper chord skin flanges from stabilizer STA 203.10 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11B, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-28-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (Low Frequency Eddy Current) both the forward and aft rear spar upper chord skin flanges from stabilizer STA 203.10 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11B, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-28-02

MECH	INSP

TASK 55-05-02-250-853

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-053

(1) Do a Low Frequency Eddy Current inspection of both the forward and aft rear spar upper chord skin flanges from stabilizer STA 203.10 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11B, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-30-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	18000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		333DB 333EB 333FB			332 333

Inspect (Ultrasonic) the rear spar upper chord web flange at the rib and stiffener locations from stabilizer STA 203.1 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11C, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-30-01

MECH	INSP

TASK 55-05-02-130-817

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-083

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-130-017

(2) Do an Ultrasonic inspection of the rear spar upper chord web flange at the rib and stiffener locations from stabilizer STA 203.1 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11C, for alternative inspection.

SUBTASK 55-05-02-410-081

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-30-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343DB 343EB 343FB			ZONE 342 343

Inspect (Ultrasonic) the rear spar upper chord web flange at the rib and stiffener locations from stabilizer STA 203.1 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11C, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-30-02

MECH	INSP

TASK 55-05-02-130-817

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-083

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-130-017

(2) Do an Ultrasonic inspection of the rear spar upper chord web flange at the rib and stiffener locations from stabilizer STA 203.1 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11C, for alternative inspection.

SUBTASK 55-05-02-410-081

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-32-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333DB 333EB 333FB			ZONE 332 333

Inspect (High Frequency Eddy Current) the rear spar upper chord web flange between the ribs and stiffeners from stabilizer STA 203.10 to stabilizer STA 258.28.

Note: Remove sealant in excess of .20" on either side of the fastener head or collar.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11D, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-32-01

MECH	INSP

TASK 55-05-02-250-854

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-087

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-054

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord web flange between the ribs and stiffeners from stabilizer STA 203.10 to stabilizer STA 258.28.

NOTE: Remove sealant in excess of .20" on either side of the fastener head or collar.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11D, for alternative inspection.

SUBTASK 55-05-02-410-085

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-32-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343DB 343EB 343FB			ZONE 342 343

Inspect (High Frequency Eddy Current) the rear spar upper chord web flange between the ribs and stiffeners from stabilizer STA 203.10 to stabilizer STA 258.28.

Note: Remove sealant in excess of .20" on either side of the fastener head or collar.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11D, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-32-02

MECH	INSP

TASK 55-05-02-250-854

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-087

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-054

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord web flange between the ribs and stiffeners from stabilizer STA 203.10 to stabilizer STA 258.28.

NOTE: Remove sealant in excess of .20" on either side of the fastener head or collar.

See Doc. D626A001 - DTR, DTR check form 55-10-04-11D, for alternative inspection.

SUBTASK 55-05-02-410-085

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-34-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 332 333

Inspect (Low Frequency Eddy Current) the rear spar upper chord, at the forward flange of the chord, from stabilizer STA 258.28 to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-34-01

MECH	INSP

TASK 55-05-02-250-855

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-055

(1) Do a Low Frequency Eddy Current inspection of the rear spar upper chord, at the forward flange of the chord, from stabilizer STA 258.28 to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-34-02
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	R HORZ STAB	1.1	56000 FC	15000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
					342 343

Inspect (Low Frequency Eddy Current) the rear spar upper chord, at the forward flange of the chord, from stabilizer STA 258.28 to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-34-02

MECH	INSP

TASK 55-05-02-250-855

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-055

(1) Do a Low Frequency Eddy Current inspection of the rear spar upper chord, at the forward flange of the chord, from stabilizer STA 258.28 to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-12A, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-36-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 21000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 332 333

Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-36-01

MECH	INSP

TASK 55-05-02-250-856

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-056

(1) Do a Low Frequency Eddy Current inspection of the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-36-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 21000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342 343

Inspect (Low Frequency Eddy Current) the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-36-02

MECH	INSP

TASK 55-05-02-250-856

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-056

(1) Do a Low Frequency Eddy Current inspection of the rear spar upper cord, at the aft flange of the cord, from stabilizer STA 258.28 to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-12B, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-38-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	18000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		333EB 333FB			332 333

Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-38-01

MECH	INSP

TASK 55-05-02-130-818

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-089

(1) Open these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-130-018

(2) Do an Ultrasonic inspection of the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.

SUBTASK 55-05-02-410-087

(3) Close these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-38-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343EB 343FB			ZONE 342 343

Inspect (Ultrasonic) the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-38-02

MECH	INSP

TASK 55-05-02-130-818

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-089

(1) Open these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-130-018

(2) Do an Ultrasonic inspection of the rear spar upper cord web flange, at the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Section 55-10-19.

SUBTASK 55-05-02-410-087

(3) Close these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-40-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333EB 333FB			ZONE 332 333

Inspect (High Frequency Eddy Current) the rear spar upper chord web flange, between the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-59.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-40-01

MECH	INSP

TASK 55-05-02-250-871

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-091

(1) Open these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-071

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord web flange, between the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-59.

SUBTASK 55-05-02-410-089

(3) Close these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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BOEING

737-600/700/800/900
TASK CARDS

AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-40-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343EB 343FB			ZONE 342 343

Inspect (High Frequency Eddy Current) the rear spar upper chord web flange, between the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-59.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-40-02

MECH	INSP

TASK 55-05-02-250-871

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-091

(1) Open these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-071

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord web flange, between the rib and stiffener locations, from stabilizer STA 258.28 to stabilizer STA 310.54.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-59.

SUBTASK 55-05-02-410-089

(3) Close these access panels on the Left side:

Number	Name/Location
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-42-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332MB 332NB 332PB			ZONE 332 333
		NOTE			

Inspect (Low Frequency Eddy Current) the rear spar upper cord web flange, at and between the ribs/stiffeners, from stabilizer STA 310.54 to tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13C, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-60.

ACCESS NOTE: Access is through the lower inspar skin access hole. Subsurface inspection of the chord is through the web or through the web and stiffener.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-42-01

MECH	INSP

TASK 55-05-02-250-857

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-093

(1) Open these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Open these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

NOTE: Access is through the lower inspar skin access hole. Subsurface inspection of the chord is through the web or through the web and stiffener.

SUBTASK 55-05-02-250-057

(2) Do a Low Frequency Eddy Current inspection of the rear spar upper cord web flange, at and between the ribs/stiffeners, from stabilizer STA 310.54 to tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13C, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-60.

SUBTASK 55-05-02-410-091

(3) Close these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Close these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD			55-611-42-02
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	R HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		342MB 342NB 342PB			342 343
		NOTE			

Inspect (Low Frequency Eddy Current) the rear spar upper cord web flange, at and between the ribs/stiffeners, from stabilizer STA 310.54 to tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13C, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-60.

ACCESS NOTE: Access is through the lower inspar skin access hole. Subsurface inspection of the chord is through the web or through the web and stiffener.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-42-02

MECH	INSP

TASK 55-05-02-250-857

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-093

(1) Open these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Open these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

NOTE: Access is through the lower inspar skin access hole. Subsurface inspection of the chord is through the web or through the web and stiffener.

SUBTASK 55-05-02-250-057

(2) Do a Low Frequency Eddy Current inspection of the rear spar upper cord web flange, at and between the ribs/stiffeners, from stabilizer STA 310.54 to tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-13C, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-60.

SUBTASK 55-05-02-410-091

(3) Close these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Close these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-44-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333BB 333CB 333DB			ZONE 332 333
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar web at the lower edge of the fail-safe chord, at and between the stiffeners, from stabilizer STA 83.5 to stabilizer STA 184.7.

See Doc. D626A001 - DTR, DTR check form 55-10-04-20, for alternative inspection.

ACCESS NOTE: Access is through the removable trailing edge lower skin panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-44-01

MECH	INSP

TASK 55-05-02-250-858

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-095

(1) Open these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Access is through the removable trailing edge lower skin panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

SUBTASK 55-05-02-250-058

(2) Do a High Frequency Eddy Current inspection of the rear spar web at the lower edge of the fail-safe chord, at and between the stiffeners, from stabilizer STA 83.5 to stabilizer STA 184.7.

See Doc. D626A001 - DTR, DTR check form 55-10-04-20, for alternative inspection.

SUBTASK 55-05-02-410-093

(3) Close these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-44-02
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	R HORZ STAB	1.1	56000 FC	18000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		343BB 343CB 343DB			342 343
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar web at the lower edge of the fail-safe chord, at and between the stiffeners, from stabilizer STA 83.5 to stabilizer STA 184.7.

See Doc. D626A001 - DTR, DTR check form 55-10-04-20, for alternative inspection.

ACCESS NOTE: Access is through the removable trailing edge lower skin panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-44-02

MECH	INSP

TASK 55-05-02-250-858

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-095

(1) Open these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Access is through the removable trailing edge lower skin panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

SUBTASK 55-05-02-250-058

(2) Do a High Frequency Eddy Current inspection of the rear spar web at the lower edge of the fail-safe chord, at and between the stiffeners, from stabilizer STA 83.5 to stabilizer STA 184.7.

See Doc. D626A001 - DTR, DTR check form 55-10-04-20, for alternative inspection.

SUBTASK 55-05-02-410-093

(3) Close these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-46-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333DB 333EB 333FB			ZONE 332 333
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar web at the lower edge of the upper chord, at and between the stiffeners, from stabilizer STA 184.7 to stabilizer STA 285.9.

See Doc. D626A001 - DTR, DTR check form 55-10-04-21, for alternative inspection.

ACCESS NOTE: Access is through the removable Trailing Edge Lower Skin Panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-46-01

MECH	INSP

TASK 55-05-02-250-859

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-097

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-059

NOTE: Access is through the removable Trailing Edge Lower Skin Panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

(2) Do a High Frequency Eddy Current inspection of the rear spar web at the lower edge of the upper chord, at and between the stiffeners, from stabilizer STA 184.7 to stabilizer STA 285.9.

See Doc. D626A001 - DTR, DTR check form 55-10-04-21, for alternative inspection.

SUBTASK 55-05-02-410-095

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-46-02
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	R HORZ STAB	1.1	56000 FC	18000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		343DB 343EB 343FB			342 343
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar web at the lower edge of the upper chord, at and between the stiffeners, from stabilizer STA 184.7 to stabilizer STA 285.9.

See Doc. D626A001 - DTR, DTR check form 55-10-04-21, for alternative inspection.

ACCESS NOTE: Access is through the removable Trailing Edge Lower Skin Panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-46-02

MECH	INSP

TASK 55-05-02-250-859

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-097

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 55-05-02-250-059

NOTE: Access is through the removable Trailing Edge Lower Skin Panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

(2) Do a High Frequency Eddy Current inspection of the rear spar web at the lower edge of the upper chord, at and between the stiffeners, from stabilizer STA 184.7 to stabilizer STA 285.9.

See Doc. D626A001 - DTR, DTR check form 55-10-04-21, for alternative inspection.

SUBTASK 55-05-02-410-095

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-48-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333FB			ZONE 332 333
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar web at the upper and lower chord edges, at and between the stiffeners, from stabilizer STA 285.9 to stabilizer STA 310.5.

See Doc. D626A001 - DTR, DTR check form 55-10-04-22, for alternative inspection.

ACCESS NOTE: Access is through the removable Trailing Edge Panel Lower Skin Panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-48-01

MECH	INSP

TASK 55-05-02-250-860

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-099

(1) Open this access panel on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open this access panel on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Access is through the removable Trailing Edge Panel Lower Skin Panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

SUBTASK 55-05-02-250-060

(2) Do a High Frequency Eddy Current inspection of the rear spar web at the upper and lower chord edges, at and between the stiffeners, from stabilizer STA 285.9 to stabilizer STA 310.5.

See Doc. D626A001 - DTR, DTR check form 55-10-04-22, for alternative inspection.

SUBTASK 55-05-02-410-097

(3) Close this access panel on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close this access panel on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-48-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343FB			ZONE 342 343
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar web at the upper and lower chord edges, at and between the stiffeners, from stabilizer STA 285.9 to stabilizer STA 310.5.

See Doc. D626A001 - DTR, DTR check form 55-10-04-22, for alternative inspection.

ACCESS NOTE: Access is through the removable Trailing Edge Panel Lower Skin Panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-48-02

MECH	INSP

TASK 55-05-02-250-860

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-099

(1) Open this access panel on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open this access panel on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Access is through the removable Trailing Edge Panel Lower Skin Panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

SUBTASK 55-05-02-250-060

(2) Do a High Frequency Eddy Current inspection of the rear spar web at the upper and lower chord edges, at and between the stiffeners, from stabilizer STA 285.9 to stabilizer STA 310.5.

See Doc. D626A001 - DTR, DTR check form 55-10-04-22, for alternative inspection.

SUBTASK 55-05-02-410-097

(3) Close this access panel on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close this access panel on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-50-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332MB 332NB 332PB			ZONE 332 333
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar web at the upper and lower web to chord fastener locations, between the stiffeners, from stabilizer STA 310.5 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-23, for alternative inspection.

ACCESS NOTE: Access is through the inspar lower skin access panels. The DTR curve assumes no cap seal. Remove any cap seals present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-50-01

MECH	INSP

TASK 55-05-02-250-861

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-101

(1) Open these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Open these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

NOTE: Access is through the inspar lower skin access panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

SUBTASK 55-05-02-250-061

(2) Do a High Frequency Eddy Current inspection of the rear spar web at the upper and lower web to chord fastener locations, between the stiffeners, from stabilizer STA 310.5 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-23, for alternative inspection.

SUBTASK 55-05-02-410-099

(3) Close these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Close these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-50-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342MB 342NB 342PB			ZONE 342 343
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar web at the upper and lower web to chord fastener locations, between the stiffeners, from stabilizer STA 310.5 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-23, for alternative inspection.

ACCESS NOTE: Access is through the inspar lower skin access panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-50-02

MECH	INSP

TASK 55-05-02-250-861

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-101

(1) Open these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Open these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

NOTE: Access is through the inspar lower skin access panels. The DTR curve assumes no fillet seal. Remove any fillet seals present.

SUBTASK 55-05-02-250-061

(2) Do a High Frequency Eddy Current inspection of the rear spar web at the upper and lower web to chord fastener locations, between the stiffeners, from stabilizer STA 310.5 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-23, for alternative inspection.

SUBTASK 55-05-02-410-099

(3) Close these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Close these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-52-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332MB 332NB 332PB			ZONE 332 333
		NOTE			

Inspect (Ultrasonic) the rear spar web at the stiffeners common to both the upper and lower chords from stabilizer STA 310.5 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-24, for alternative inspection.

ACCESS NOTE: Access is through the inspar lower skin access panels.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-52-01

MECH	INSP

TASK 55-05-02-130-820

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-103

(1) Open these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Open these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

NOTE: Access is through the inspar lower skin access panels.

SUBTASK 55-05-02-130-020

(2) Do an Ultrasonic inspection of the rear spar web at the stiffeners common to both the upper and lower chords from stabilizer STA 310.5 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-24, for alternative inspection.

SUBTASK 55-05-02-410-101

(3) Close these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Close these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

END OF TASK

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR WEB			55-611-52-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342MB 342NB 342PB			ZONE 342 343
		NOTE			

Inspect (Ultrasonic) the rear spar web at the stiffeners common to both the upper and lower chords from stabilizer STA 310.5 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-24, for alternative inspection.

ACCESS NOTE: Access is through the inspar lower skin access panels.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-611-52-02

MECH	INSP

TASK 55-05-02-130-820

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-103

(1) Open these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Open these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

NOTE: Access is through the inspar lower skin access panels.

SUBTASK 55-05-02-130-020

(2) Do an Ultrasonic inspection of the rear spar web at the stiffeners common to both the upper and lower chords from stabilizer STA 310.5 to outboard tip.

See Doc. D626A001 - DTR, DTR check form 55-10-04-24, for alternative inspection.

SUBTASK 55-05-02-410-101

(3) Close these access panels on the Left side:

Number	Name/Location
332MB	Horizontal Stabilizer, Access Door
332NB	Horizontal Stabilizer, Access Door
332PB	Horizontal Stabilizer, Access Door

Close these access panels on the Right side:

Number	Name/Location
342MB	Horizontal Stabilizer, Access Door
342NB	Horizontal Stabilizer, Access Door
342PB	Horizontal Stabilizer, Access Door

END OF TASK

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER AND FAILSAFE CHORD LUGS			55-612-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332AB 333AB 333AT			ZONE 332 333
		NOTE			

Inspect (Ultrasonic) the rear spar upper and failsafe chord lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-05-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-15.

ACCESS NOTE: Removal of gap covers is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-612-00-01

MECH	INSP

TASK 55-05-02-130-821

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER AND FAILSAFE CHORD LUGS

A. Inspection

SUBTASK 55-05-02-010-106

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

NOTE: Removal of gap covers is required.

SUBTASK 55-05-02-130-021

(2) Do an Ultrasonic inspection of the rear spar upper and failsafe chord lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-05-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-15.

SUBTASK 55-05-02-410-159

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 55-05-02-410-104

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-612-00-01

MECH	INSP

(Continued)

Number	Name/Location
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

END OF TASK

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER AND FAILSAFE CHORD LUGS			55-612-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342AB 343AB 343AT			ZONE 342 343
		NOTE			

Inspect (Ultrasonic) the rear spar upper and failsafe chord lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-05-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-15.

ACCESS NOTE: Removal of gap covers is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-612-00-02

MECH	INSP

TASK 55-05-02-130-821

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER AND FAILSAFE CHORD LUGS

A. Inspection

SUBTASK 55-05-02-010-106

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

NOTE: Removal of gap covers is required.

SUBTASK 55-05-02-130-021

(2) Do an Ultrasonic inspection of the rear spar upper and failsafe chord lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-05-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-15.

SUBTASK 55-05-02-410-159

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 55-05-02-410-104

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-612-00-02

MECH	INSP

(Continued)

Number	Name/Location
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

END OF TASK

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING			55-613-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3550 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332EB 333BB			ZONE 332 333
		NOTE			

Inspect (Ultrasonic) the forward and aft sides of the terminal fitting around the three inboard fasteners common to the spar chord.

Note: Bolts and bushings should remain installed for the inspection. Remove any sealant present.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2A, for alternative inspection.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the inspar lower skin. Sealant, if present, must be removed for inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-613-00-01

MECH	INSP

TASK 55-05-02-130-822

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-108

(1) Open these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the inspar lower skin. Sealant, if present, must be removed for inspection.

SUBTASK 55-05-02-130-022

(2) Do an Ultrasonic inspection of the forward and aft sides of the terminal fitting around the three inboard fasteners common to the spar chord.

NOTE: Bolts and bushings should remain installed for the inspection. Remove any sealant present.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2A, for alternative inspection.

SUBTASK 55-05-02-410-106

(3) Close these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING			55-613-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3550 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342EB 343BB			ZONE 342 343
		NOTE			

Inspect (Ultrasonic) the forward and aft sides of the terminal fitting around the three inboard fasteners common to the spar chord.

Note: Bolts and bushings should remain installed for the inspection. Remove any sealant present.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2A, for alternative inspection.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the inspar lower skin. Sealant, if present, must be removed for inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-613-00-02

MECH	INSP

TASK 55-05-02-130-822

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-108

(1) Open these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the inspar lower skin. Sealant, if present, must be removed for inspection.

SUBTASK 55-05-02-130-022

(2) Do an Ultrasonic inspection of the forward and aft sides of the terminal fitting around the three inboard fasteners common to the spar chord.

NOTE: Bolts and bushings should remain installed for the inspection. Remove any sealant present.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2A, for alternative inspection.

SUBTASK 55-05-02-410-106

(3) Close these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING			55-613-01-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	3550 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		332EB 333BB			332 333
		NOTE			

Inspect (High Frequency Eddy Current) the forward and aft sides of the terminal fitting around the three inboard fasteners common to the spar chord.

Note: Bolts and bushings should remain installed for the inspection. Remove any sealant present.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-18.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the inspar lower skin. Sealant, if present, must be removed for inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-613-01-01

MECH	INSP

TASK 55-05-02-250-812

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-073

(1) Open these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the inspar lower skin. Sealant, if present, must be removed for inspection.

SUBTASK 55-05-02-250-012

(2) Do a High Frequency Eddy Current inspection of the forward and aft sides of the terminal fitting around the three inboard fasteners common to the spar chord.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-18.

NOTE: Bolts and bushings should remain installed for the inspection. Remove any sealant present.

SUBTASK 55-05-02-410-071

(3) Close these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING			55-613-01-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 3550 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342EB 343BB			ZONE 342 343
		NOTE			

Inspect (High Frequency Eddy Current) the forward and aft sides of the terminal fitting around the three inboard fasteners common to the spar chord.

Note: Bolts and bushings should remain installed for the inspection. Remove any sealant present.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-18.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the inspar lower skin. Sealant, if present, must be removed for inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-613-01-02

MECH	INSP

TASK 55-05-02-250-812

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-073

(1) Open these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the inspar lower skin. Sealant, if present, must be removed for inspection.

SUBTASK 55-05-02-250-012

(2) Do a High Frequency Eddy Current inspection of the forward and aft sides of the terminal fitting around the three inboard fasteners common to the spar chord.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-18.

NOTE: Bolts and bushings should remain installed for the inspection. Remove any sealant present.

SUBTASK 55-05-02-410-071

(3) Close these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING			55-613-10-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332EB 333BB NOTE			ZONE 332 333

Inspect (Detailed) the forward and aft sides of the terminal fitting at the outboard attach bolt location.

Note: Bolts and bushings should remain installed for the inspection. Sealant removal is required.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2B, for alternative inspection.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Use of a Bore scope is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-613-10-01

MECH	INSP

TASK 55-05-02-211-813

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-112

(1) Open these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Use of a Bore scope is required.

SUBTASK 55-05-02-211-013

(2) Do a Detailed inspection of the forward and aft sides of the terminal fitting at the outboard attach bolt location.

NOTE: Bolts and bushings should remain installed for the inspection. Sealant removal is required.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2B, for alternative inspection.

SUBTASK 55-05-02-410-110

(3) Close these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING			55-613-10-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342EB 343BB			ZONE 342 343
		NOTE			

Inspect (Detailed) the forward and aft sides of the terminal fitting at the outboard attach bolt location.

Note: Bolts and bushings should remain installed for the inspection. Sealant removal is required.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2B, for alternative inspection.

ACCESS NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Use of a Bore scope is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-613-10-02

MECH	INSP

TASK 55-05-02-211-813

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-112

(1) Open these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The aft side is accessed from the trailing edge. The forward side is accessed through the access panel in the lower inspar skin. Use of a Bore scope is required.

SUBTASK 55-05-02-211-013

(2) Do a Detailed inspection of the forward and aft sides of the terminal fitting at the outboard attach bolt location.

NOTE: Bolts and bushings should remain installed for the inspection. Sealant removal is required.

See Doc. D626A001 - DTR, DTR check form 55-10-05-2B, for alternative inspection.

SUBTASK 55-05-02-410-110

(3) Close these access panels on the Left side:

Number	Name/Location
332EB	Horizontal Stabilizer, Access Door
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342EB	Horizontal Stabilizer, Access Door
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR TERMINAL FITTING			55-614-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	9000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		332AB 332EB 333AB 333AT 333BB			332 333
		NOTE			

Inspect (Penetrant) the upper, lower and failsafe pins at the side of body on the horizontal stabilizer rear spar terminal fitting for both the left and right sides.

See Doc. D626A001 - DTR, DTR check form 55-10-05-3, for alternative inspection.

ACCESS NOTE: Removal of pins is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-00-01

MECH	INSP

TASK 55-05-02-230-801

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-030

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of pins is required.

SUBTASK 55-05-02-230-001

(2) Do a Dye Penetrant inspection of the upper, lower and failsafe pins at the side of body on the horizontal stabilizer rear spar terminal fitting for both the left and right sides.

See Doc. D626A001 - DTR, DTR check form 55-10-05-3, for alternative inspection.

SUBTASK 55-05-02-410-157

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 55-05-02-410-028

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-00-01

MECH	INSP

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR TERMINAL FITTING			55-614-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342AB 342EB 343AB 343AT 343BB			ZONE 342 343
		NOTE			

Inspect (Penetrant) the upper, lower and failsafe pins at the side of body on the horizontal stabilizer rear spar terminal fitting for both the left and right sides.

See Doc. D626A001 - DTR, DTR check form 55-10-05-3, for alternative inspection.

ACCESS NOTE: Removal of pins is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-00-02

MECH	INSP

TASK 55-05-02-230-801

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-030

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of pins is required.

SUBTASK 55-05-02-230-001

(2) Do a Dye Penetrant inspection of the upper, lower and failsafe pins at the side of body on the horizontal stabilizer rear spar terminal fitting for both the left and right sides.

See Doc. D626A001 - DTR, DTR check form 55-10-05-3, for alternative inspection.

SUBTASK 55-05-02-410-157

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 55-05-02-410-028

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-00-02

MECH	INSP

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR TERMINAL FITTING (L3)			55-614-10-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 21000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332AB 332EB 333AB 333AT 333BB			ZONE 332 333
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar upper chord terminal fitting on the upper side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

ACCESS NOTE: Removal of the tension bolt is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-10-01

MECH	INSP

TASK 55-05-02-250-810

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING (L3)

A. Inspection

SUBTASK 55-05-02-010-149

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of the tension bolt is required.

SUBTASK 55-05-02-250-010

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord terminal fitting on the upper side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative repeat inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

SUBTASK 55-05-02-410-156

(3) Close this access panel on the Left Side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-10-01

MECH	INSP

SUBTASK 55-05-02-410-147

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR TERMINAL FITTING (L3)			55-614-10-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 21000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342AB 342EB 343AB 343AT 343BB			ZONE 342 343
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar upper chord terminal fitting on the upper side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

ACCESS NOTE: Removal of the tension bolt is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-10-02

MECH	INSP

TASK 55-05-02-250-810

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING (L3)

A. Inspection

SUBTASK 55-05-02-010-149

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of the tension bolt is required.

SUBTASK 55-05-02-250-010

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord terminal fitting on the upper side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative repeat inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

SUBTASK 55-05-02-410-156

(3) Close this access panel on the Left Side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-10-02

MECH	INSP

SUBTASK 55-05-02-410-147

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR TERMINAL FITTING (L4)			55-614-11-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 21000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332AB 332EB 333AB 333AT 333BB			ZONE 332 333
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar upper chord terminal fitting on the lower side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

ACCESS NOTE: Removal of the tension bolt is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-11-01

MECH	INSP

TASK 55-05-02-250-883

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING (L4)

A. Inspection

SUBTASK 55-05-02-010-151

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of the tension bolt is required.

SUBTASK 55-05-02-250-085

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord terminal fitting on the upper side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative repeat inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

SUBTASK 55-05-02-410-161

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-11-01

MECH	INSP

SUBTASK 55-05-02-410-149

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR TERMINAL FITTING (L4)			55-614-11-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 21000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342AB 342EB 343AB 343AT 343BB			ZONE 342 343
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar upper chord terminal fitting on the lower side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

ACCESS NOTE: Removal of the tension bolt is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-11-02

MECH	INSP

TASK 55-05-02-250-883

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR UPPER CHORD TERMINAL FITTING (L4)

A. Inspection

SUBTASK 55-05-02-010-151

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of the tension bolt is required.

SUBTASK 55-05-02-250-085

(2) Do a High Frequency Eddy Current inspection of the rear spar upper chord terminal fitting on the upper side of the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-4, for alternative repeat inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-49.

SUBTASK 55-05-02-410-161

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-11-02

MECH	INSP

SUBTASK 55-05-02-410-149

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-614-12-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332AB 332EB 333AB 333AT 333BB			ZONE 332 333
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar lower chord at the side of body, common to the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-48.

ACCESS NOTE: The inspection requires the removal of the bolt.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-12-01

MECH	INSP

TASK 55-05-02-250-863

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-119

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The inspection requires the removal of the bolt.

SUBTASK 55-05-02-250-063

(2) Do a High Frequency Eddy Current inspection of the rear spar lower chord at the side of body, common to the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-48.

SUBTASK 55-05-02-410-160

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 55-05-02-410-117

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-12-01

MECH	INSP

(Continued)

Number	Name/Location
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER REAR SPAR LOWER CHORD			55-614-12-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342AB 342EB 343AB 343AT 343BB			ZONE 342 343
		NOTE			

Inspect (High Frequency Eddy Current) the rear spar lower chord at the side of body, common to the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-48.

ACCESS NOTE: The inspection requires the removal of the bolt.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-12-02

MECH	INSP

TASK 55-05-02-250-863

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER REAR SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-119

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: The inspection requires the removal of the bolt.

SUBTASK 55-05-02-250-063

(2) Do a High Frequency Eddy Current inspection of the rear spar lower chord at the side of body, common to the Side of Body (SOB) rib tension bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-05-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-48.

SUBTASK 55-05-02-410-160

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 55-05-02-410-117

(5) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332EB	Horizontal Stabilizer, Access Door

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-614-12-02

MECH	INSP

(Continued)

Number	Name/Location
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342EB	Horizontal Stabilizer, Access Door
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD			55-615-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 331 332

Inspect (Low Frequency Eddy Current) the front spar upper chord from the side of body to stabilizer STA 66.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-28.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-615-00-01

MECH	INSP

TASK 55-05-02-250-808

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-008

(1) Do a Low Frequency Eddy Current inspection of the front spar upper chord from the side of body to stabilizer STA 66.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-28.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD			55-615-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 341 342

Inspect (Low Frequency Eddy Current) the front spar upper chord from the side of body to stabilizer STA 66.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-28.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-615-00-02

MECH	INSP

TASK 55-05-02-250-808

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-250-008

(1) Do a Low Frequency Eddy Current inspection of the front spar upper chord from the side of body to stabilizer STA 66.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-28.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, AFT FLANGE			55-616-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 331 332

Inspect (Low Frequency Eddy Current) the front spar upper chord aft flange from stabilizer STA 66.5 to stabilizer tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-32.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-00-01

MECH	INSP

TASK 55-05-02-250-803

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, AFT FLANGE

A. Inspection

SUBTASK 55-05-02-250-003

(1) Do a Low Frequency Eddy Current inspection of the front spar upper chord aft flange from stabilizer STA 66.5 to stabilizer tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-32.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, AFT FLANGE			55-616-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 341 342

Inspect (Low Frequency Eddy Current) the front spar upper chord aft flange from stabilizer STA 66.5 to stabilizer tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-32.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-00-02

MECH	INSP

TASK 55-05-02-250-803

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, AFT FLANGE

A. Inspection

SUBTASK 55-05-02-250-003

(1) Do a Low Frequency Eddy Current inspection of the front spar upper chord aft flange from stabilizer STA 66.5 to stabilizer tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-32.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-05-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (Low Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 75.0 to Stab Sta 175.5, except at web splice, stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4C, for alternative inspection.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-05-01

MECH	INSP

TASK 55-05-02-250-885

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-153

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-087

(2) Do a Low Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 75.0 to stabilizer STA 175.5 except at web splice, stabilizer STA 111.1.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4C, for alternative inspection.

SUBTASK 55-05-02-410-151

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-05-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (Low Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 75.0 to Stab Sta 175.5, except at web splice, stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4C, for alternative inspection.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-05-02

MECH	INSP

TASK 55-05-02-250-885

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-153

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-087

(2) Do a Low Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 75.0 to stabilizer STA 175.5 except at web splice, stabilizer STA 111.1.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4C, for alternative inspection.

SUBTASK 55-05-02-410-151

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-06-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 75.0 to Stab Sta 175.5, except at web splice, stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4C, for alternative inspection.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-06-01

MECH	INSP

TASK 55-05-02-250-886

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-154

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-088

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 75.0 to stabilizer STA 175.5 except at web splice, stabilizer STA 111.1.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4C, for alternative inspection.

SUBTASK 55-05-02-410-152

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-06-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 75.0 to Stab Sta 175.5, except at web splice, stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4C, for alternative inspection.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-06-02

MECH	INSP

TASK 55-05-02-250-886

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-154

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-088

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 75.0 to stabilizer STA 175.5 except at web splice, stabilizer STA 111.1.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4C, for alternative inspection.

SUBTASK 55-05-02-410-152

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-07-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (Low Frequency Eddy Current) the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-07-01

MECH	INSP

TASK 55-05-02-250-887

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-155

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-089

(2) Do a Low Frequency Eddy Current inspection of the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection.

SUBTASK 55-05-02-410-153

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-07-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (Low Frequency Eddy Current) the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-07-02

MECH	INSP

TASK 55-05-02-250-887

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-155

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-089

(2) Do a Low Frequency Eddy Current inspection of the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection.

SUBTASK 55-05-02-410-153

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-08-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-08-01

MECH	INSP

TASK 55-05-02-250-888

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-156

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-090

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection.

SUBTASK 55-05-02-410-154

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-08-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-08-02

MECH	INSP

TASK 55-05-02-250-888

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-156

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-090

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord web flange at web splice from STA 110.24 to STA 111.96.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4D, for alternative inspection.

SUBTASK 55-05-02-410-154

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-09-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 258.28 to Outboard Tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4E, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-54.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-09-01

MECH	INSP

TASK 55-05-02-250-884

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-152

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-086

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 258.28 to outboard tip.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4E, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-54.

SUBTASK 55-05-02-410-150

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, WEB FLANGE			55-616-09-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 258.28 to Outboard Tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4E, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-54.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-09-02

MECH	INSP

TASK 55-05-02-250-884

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-152

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-086

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 258.28 to outboard tip.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4E, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-54.

SUBTASK 55-05-02-410-150

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, FORWARD FLANGE			55-616-10-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord forward flange from stabilizer STA 66.5 to stabilizer tip.

Note: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-46.

ACCESS NOTE: Removal of the horizontal stabilizer removable leading edge is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-10-01

MECH	INSP

TASK 55-05-02-250-864

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-115

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the horizontal stabilizer removable leading edge is required.

SUBTASK 55-05-02-250-064

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord forward flange from stabilizer STA 66.5 to stabilizer tip.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-46.

SUBTASK 55-05-02-410-113

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, FORWARD FLANGE			55-616-10-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord forward flange from stabilizer STA 66.5 to stabilizer tip.

Note: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-46.

ACCESS NOTE: Removal of the horizontal stabilizer removable leading edge is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-10-02

MECH	INSP

TASK 55-05-02-250-864

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-115

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the horizontal stabilizer removable leading edge is required.

SUBTASK 55-05-02-250-064

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord forward flange from stabilizer STA 66.5 to stabilizer tip.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-46.

SUBTASK 55-05-02-410-113

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, FORWARD FLANGE			55-616-11-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (Low Frequency Eddy Current) the front spar upper chord forward flange from stabilizer STA 66.5 to stabilizer tip.

Note: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-3, for alternative inspection.

ACCESS NOTE: Removal of the horizontal stabilizer removable leading edge is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-11-01

MECH	INSP

TASK 55-05-02-250-889

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-157

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the horizontal stabilizer removable leading edge is required.

SUBTASK 55-05-02-250-091

(2) Do a Low Frequency Eddy Current inspection of the front spar upper chord forward flange from stabilizer STA 66.5 to stabilizer tip.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-3, for alternative inspection.

SUBTASK 55-05-02-410-155

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, FORWARD FLANGE			55-616-11-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 2000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (Low Frequency Eddy Current) the front spar upper chord forward flange from stabilizer STA 66.5 to stabilizer tip.

Note: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-3, for alternative inspection.

ACCESS NOTE: Removal of the horizontal stabilizer removable leading edge is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-11-02

MECH	INSP

TASK 55-05-02-250-889

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD, FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-157

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the horizontal stabilizer removable leading edge is required.

SUBTASK 55-05-02-250-091

(2) Do a Low Frequency Eddy Current inspection of the front spar upper chord forward flange from stabilizer STA 66.5 to stabilizer tip.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-3, for alternative inspection.

SUBTASK 55-05-02-410-155

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE			55-616-12-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (Low Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 66.5 to stabilizer STA 75.0.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-45.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-12-01

MECH	INSP

TASK 55-05-02-250-865

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-116

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-065

(2) Do a Low Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 66.5 to stabilizer STA 75.0.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-45.

SUBTASK 55-05-02-410-114

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE			55-616-12-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 6000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (Low Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 66.5 to stabilizer STA 75.0.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-45.

ACCESS NOTE: Removal of the Leading Edge is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-12-02

MECH	INSP

TASK 55-05-02-250-865

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-116

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the Leading Edge is required for inspection access.

SUBTASK 55-05-02-250-065

(2) Do a Low Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 66.5 to stabilizer STA 75.0.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-45.

SUBTASK 55-05-02-410-114

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE			55-616-14-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B 331CZ			ZONE 331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 175.5 to stabilizer STA 258.28.

Note: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-47.

ACCESS NOTE: Removal of the horizontal stabilizer leading edge and the removable front spar web is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-14-01

MECH	INSP

TASK 55-05-02-250-868

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-124

(1) Open these access panels on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ
331CZ	Horizontal Stabilizer Front Spar Removal Web Panel

Open these access panels on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ
341CZ	Horizontal Stabilizer Front Spar Removal Web Panel

NOTE: Removal of the horizontal stabilizer leading edge and the removable front spar web is required for inspection access.

SUBTASK 55-05-02-250-068

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 175.5 to stabilizer STA 258.28.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-47.

SUBTASK 55-05-02-410-122

(3) Close these access panels on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ
331CZ	Horizontal Stabilizer Front Spar Removal Web Panel

Close these access panels on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ
341CZ	Horizontal Stabilizer Front Spar Removal Web Panel

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE			55-616-14-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B 341CZ			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar upper chord web flange from stabilizer STA 175.5 to stabilizer STA 258.28.

Note: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-47.

ACCESS NOTE: Removal of the horizontal stabilizer leading edge and the removable front spar web is required for inspection access.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-14-02

MECH	INSP

TASK 55-05-02-250-868

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD - WEB FLANGE

A. Inspection

SUBTASK 55-05-02-010-124

(1) Open these access panels on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ
331CZ	Horizontal Stabilizer Front Spar Removal Web Panel

Open these access panels on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ
341CZ	Horizontal Stabilizer Front Spar Removal Web Panel

NOTE: Removal of the horizontal stabilizer leading edge and the removable front spar web is required for inspection access.

SUBTASK 55-05-02-250-068

(2) Do a High Frequency Eddy Current inspection of the front spar upper chord web flange from stabilizer STA 175.5 to stabilizer STA 258.28.

NOTE: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-4B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-47.

SUBTASK 55-05-02-410-122

(3) Close these access panels on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ
331CZ	Horizontal Stabilizer Front Spar Removal Web Panel

Close these access panels on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ
341CZ	Horizontal Stabilizer Front Spar Removal Web Panel

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD - FORWARD FLANGE			55-616-16-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (Detailed) the front spar lower chord forward flange from stabilizer STA 66.50 to stabilizer STA 175.50.

See Doc. D626A001 - DTR, DTR check form 55-10-06-6A, for alternative inspection.

ACCESS NOTE: Removal of the leading edge panels is required to gain access to the lower surface of the forward flange.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-16-01

MECH	INSP

TASK 55-05-02-211-804

1. INTERNAL - DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD - FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-126

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the leading edge panels is required to gain access to the lower surface of the forward flange.

SUBTASK 55-05-02-211-004

(2) Do a Detailed inspection of the front spar lower chord forward flange from stabilizer STA 66.50 to stabilizer STA 175.50.

See Doc. D626A001 - DTR, DTR check form 55-10-06-6A, for alternative inspection.

SUBTASK 55-05-02-410-124

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD - FORWARD FLANGE			55-616-16-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (Detailed) the front spar lower chord forward flange from stabilizer STA 66.50 to stabilizer STA 175.50.

See Doc. D626A001 - DTR, DTR check form 55-10-06-6A, for alternative inspection.

ACCESS NOTE: Removal of the leading edge panels is required to gain access to the lower surface of the forward flange.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-16-02

MECH	INSP

TASK 55-05-02-211-804

1. INTERNAL - DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD - FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-126

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the leading edge panels is required to gain access to the lower surface of the forward flange.

SUBTASK 55-05-02-211-004

(2) Do a Detailed inspection of the front spar lower chord forward flange from stabilizer STA 66.50 to stabilizer STA 175.50.

See Doc. D626A001 - DTR, DTR check form 55-10-06-6A, for alternative inspection.

SUBTASK 55-05-02-410-124

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE			55-616-18-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 331 332

Inspect (Low Frequency Eddy Current) the front spar lower chord aft flange from stabilizer STA 66.50 to stabilizer STA 175.50.

See Doc. D626A001 - DTR, DTR check form 55-10-06-6B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-53.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-18-01

MECH	INSP

TASK 55-05-02-250-869

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD - AFT FLANGE

A. Inspection

SUBTASK 55-05-02-250-069

(1) Do a Low Frequency Eddy Current inspection of the front spar lower chord aft flange from stabilizer STA 66.50 to stabilizer STA 175.50.

See Doc. D626A001 - DTR, DTR check form 55-10-06-6B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-53.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE			55-616-18-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 341 342

Inspect (Low Frequency Eddy Current) the front spar lower chord aft flange from stabilizer STA 66.50 to stabilizer STA 175.50.

See Doc. D626A001 - DTR, DTR check form 55-10-06-6B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-53.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-18-02

MECH	INSP

TASK 55-05-02-250-869

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD - AFT FLANGE

A. Inspection

SUBTASK 55-05-02-250-069

(1) Do a Low Frequency Eddy Current inspection of the front spar lower chord aft flange from stabilizer STA 66.50 to stabilizer STA 175.50.

See Doc. D626A001 - DTR, DTR check form 55-10-06-6B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-53.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, FORWARD FLANGE			55-616-20-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar lower chord forward flange from stabilizer STA 175.50 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-7A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51.

ACCESS NOTE: Removal of the leading edge panel is required to gain access to the lower surface of the forward flange.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-20-01

MECH	INSP

TASK 55-05-02-250-870

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-128

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the leading edge panel is required to gain access to the lower surface of the forward flange.

SUBTASK 55-05-02-250-070

(2) Do a High Frequency Eddy Current inspection of the front spar lower chord forward flange from stabilizer STA 175.50 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-7A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51.

SUBTASK 55-05-02-410-126

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, FORWARD FLANGE			55-616-20-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar lower chord forward flange from stabilizer STA 175.50 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-7A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51.

ACCESS NOTE: Removal of the leading edge panel is required to gain access to the lower surface of the forward flange.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-20-02

MECH	INSP

TASK 55-05-02-250-870

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-128

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Removal of the leading edge panel is required to gain access to the lower surface of the forward flange.

SUBTASK 55-05-02-250-070

(2) Do a High Frequency Eddy Current inspection of the front spar lower chord forward flange from stabilizer STA 175.50 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-7A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51.

SUBTASK 55-05-02-410-126

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE			55-616-22-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 331 332

Inspect (Low Frequency Eddy Current) the front spar lower chord aft flange from stabilizer STA 175.50 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-7B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-57.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-22-01

MECH	INSP

TASK 55-05-02-250-872

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE

A. Inspection

SUBTASK 55-05-02-250-072

(1) Do a Low Frequency Eddy Current inspection of the front spar lower chord aft flange from stabilizer STA 175.50 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-7B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-57.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE			55-616-22-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 341 342

Inspect (Low Frequency Eddy Current) the front spar lower chord aft flange from stabilizer STA 175.50 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-7B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-57.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-22-02

MECH	INSP

TASK 55-05-02-250-872

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE

A. Inspection

SUBTASK 55-05-02-250-072

(1) Do a Low Frequency Eddy Current inspection of the front spar lower chord aft flange from stabilizer STA 175.50 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-7B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-57.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, FORWARD FLANGE			55-616-24-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar lower chord forward flange from stabilizer STA 258.28 to stabilizer BL 281.81.

See Doc. D626A001 - DTR, DTR check form 55-10-06-8A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51.

ACCESS NOTE: Leading edge skin assembly removal is required for the inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-24-01

MECH	INSP

TASK 55-05-02-250-874

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-130

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Leading edge skin assembly removal is required for the inspection.

SUBTASK 55-05-02-250-074

(2) Do a High Frequency Eddy Current inspection of the front spar lower chord forward flange from stabilizer STA 258.28 to stabilizer BL 281.81.

See Doc. D626A001 - DTR, DTR check form 55-10-06-8A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51.

SUBTASK 55-05-02-410-128

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, FORWARD FLANGE			55-616-24-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar lower chord forward flange from stabilizer STA 258.28 to stabilizer BL 281.81.

See Doc. D626A001 - DTR, DTR check form 55-10-06-8A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51.

ACCESS NOTE: Leading edge skin assembly removal is required for the inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-24-02

MECH	INSP

TASK 55-05-02-250-874

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, FORWARD FLANGE

A. Inspection

SUBTASK 55-05-02-010-130

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: Leading edge skin assembly removal is required for the inspection.

SUBTASK 55-05-02-250-074

(2) Do a High Frequency Eddy Current inspection of the front spar lower chord forward flange from stabilizer STA 258.28 to stabilizer BL 281.81.

See Doc. D626A001 - DTR, DTR check form 55-10-06-8A, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-51.

SUBTASK 55-05-02-410-128

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE			55-616-26-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 331 332

Inspect (Low Frequency Eddy Current) the front spar lower chord aft flange from stabilizer STA 258.28 to stabilizer BL 281.81 (tip).

See Doc. D626A001 - DTR, DTR check form 55-10-06-8B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-57.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-26-01

MECH	INSP

TASK 55-05-02-250-873

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE

A. Inspection

SUBTASK 55-05-02-250-073

(1) Do a Low Frequency Eddy Current inspection of the front spar lower chord aft flange from stabilizer STA 258.28 to stabilizer BL 281.81 (tip).

See Doc. D626A001 - DTR, DTR check form 55-10-06-8B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-57.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE			55-616-26-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 341 342

Inspect (Low Frequency Eddy Current) the front spar lower chord aft flange from stabilizer STA 258.28 to stabilizer BL 281.81 (tip).

See Doc. D626A001 - DTR, DTR check form 55-10-06-8B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-57.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-26-02

MECH	INSP

TASK 55-05-02-250-873

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD, AFT FLANGE

A. Inspection

SUBTASK 55-05-02-250-073

(1) Do a Low Frequency Eddy Current inspection of the front spar lower chord aft flange from stabilizer STA 258.28 to stabilizer BL 281.81 (tip).

See Doc. D626A001 - DTR, DTR check form 55-10-06-8B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-57.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-28-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	18000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		331B			331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 66.5 to stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-28-01

MECH	INSP

TASK 55-05-02-250-875

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-132

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

SUBTASK 55-05-02-250-075

(2) Do a High Frequency Eddy Current inspection of the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 66.5 to stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

SUBTASK 55-05-02-410-130

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-28-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 66.5 to stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-28-02

MECH	INSP

TASK 55-05-02-250-875

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-132

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

SUBTASK 55-05-02-250-075

(2) Do a High Frequency Eddy Current inspection of the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 66.5 to stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

SUBTASK 55-05-02-410-130

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-30-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	15000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		331B			331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web splice, at the upper and lower web edges at stabilizer STA 111.1.

Note: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-WS, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-30-01

MECH	INSP

TASK 55-05-02-250-876

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-134

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

SUBTASK 55-05-02-250-076

(2) Do a High Frequency Eddy Current inspection of the front spar web splice, at the upper and lower web edges at stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-WS, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

NOTE: Inspection requires probe placement on primed metal surface.

SUBTASK 55-05-02-410-132

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-30-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 15000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web splice, at the upper and lower web edges at stabilizer STA 111.1.

Note: Inspection requires probe placement on primed metal surface.

See Doc. D626A001 - DTR, DTR check form 55-10-06-WS, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-30-02

MECH	INSP

TASK 55-05-02-250-876

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-134

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

SUBTASK 55-05-02-250-076

(2) Do a High Frequency Eddy Current inspection of the front spar web splice, at the upper and lower web edges at stabilizer STA 111.1.

See Doc. D626A001 - DTR, DTR check form 55-10-06-WS, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

NOTE: Inspection requires probe placement on primed metal surface.

SUBTASK 55-05-02-410-132

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-32-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	12000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		331B			331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 111.1 to stabilizer STA 175.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any sealant which exceeds .30" on either side of a fastener head or collar.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-32-01

MECH	INSP

TASK 55-05-02-211-814

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-136

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any sealant which exceeds .30" on either side of a fastener head or collar.

SUBTASK 55-05-02-211-014

(2) Do a High Frequency Eddy Current inspection of the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 111.1 to stabilizer STA 175.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

SUBTASK 55-05-02-410-134

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-32-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 12000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 111.1 to stabilizer STA 175.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any sealant which exceeds .30" on either side of a fastener head or collar.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-32-02

MECH	INSP

TASK 55-05-02-211-814

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-136

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any sealant which exceeds .30" on either side of a fastener head or collar.

SUBTASK 55-05-02-211-014

(2) Do a High Frequency Eddy Current inspection of the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 111.1 to stabilizer STA 175.5.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

SUBTASK 55-05-02-410-134

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-34-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 175.5 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

Remove any cap seal, that is present, on nut plated BACB30NM fasteners.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-34-01

MECH	INSP

TASK 55-05-02-250-878

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-138

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any cap seal, that is present, on nut plated BACB30NM fasteners.

SUBTASK 55-05-02-250-078

(2) Do a High Frequency Eddy Current inspection of the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 175.5 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

SUBTASK 55-05-02-410-136

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-34-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 175.5 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

Remove any cap seal, that is present, on nut plated BACB30NM fasteners.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-34-02

MECH	INSP

TASK 55-05-02-250-878

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-138

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any cap seal, that is present, on nut plated BACB30NM fasteners.

SUBTASK 55-05-02-250-078

(2) Do a High Frequency Eddy Current inspection of the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 175.5 to stabilizer STA 258.28.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

SUBTASK 55-05-02-410-136

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-36-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331B			ZONE 331 332
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 258.28 to stabilizer tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

Remove any cap sealant that is present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-36-01

MECH	INSP

TASK 55-05-02-250-879

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-140

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any cap sealant that is present.

SUBTASK 55-05-02-250-079

(2) Do a High Frequency Eddy Current inspection of the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 258.28 to stabilizer tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

SUBTASK 55-05-02-410-138

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR WEB			55-616-36-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341B			ZONE 341 342
		NOTE			

Inspect (High Frequency Eddy Current) the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 258.28 to stabilizer tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

ACCESS NOTE: The inspection requires removal of the horizontal stabilizer leading edge.

Remove any cap sealant that is present.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-616-36-02

MECH	INSP

TASK 55-05-02-250-879

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR WEB

A. Inspection

SUBTASK 55-05-02-010-140

(1) Open this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Open this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

NOTE: The inspection requires removal of the horizontal stabilizer leading edge. Remove any cap sealant that is present.

SUBTASK 55-05-02-250-079

(2) Do a High Frequency Eddy Current inspection of the front spar web, at the upper and lower spar chord attachments, from stabilizer STA 258.28 to stabilizer tip.

See Doc. D626A001 - DTR, DTR check form 55-10-06-W4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-10-58.

SUBTASK 55-05-02-410-138

(3) Close this access panel on the Left side:

Number	Name/Location
331B	Horizontal Stabilizer - 331B is the Removable leading edge that includes individual panels 331BZ, 331CZ, 331DZ

Close this access panel on the Right side:

Number	Name/Location
341B	Horizontal Stabilizer - 341B is the Removable leading edge that includes individual panels 341BZ, 341CZ, 341DZ

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD			55-617-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 331 332

Inspect (Detailed) the exposed surface of the front spar upper chord between the upper skin and Leading Edge (LE) skin.

See Doc. D626A001 - DTR, DTR check form 55-10-07-2, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-617-00-01

MECH	INSP

TASK 55-05-02-211-803

1. EXTERNAL - DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-211-003

(1) Do a Detailed inspection of the exposed surface of the front spar upper chord between the upper skin and Leading Edge (LE) skin.

See Doc. D626A001 - DTR, DTR check form 55-10-07-2, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD			55-617-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 341 342

Inspect (Detailed) the exposed surface of the front spar upper chord between the upper skin and Leading Edge (LE) skin.

See Doc. D626A001 - DTR, DTR check form 55-10-07-2, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-617-00-02

MECH	INSP

TASK 55-05-02-211-803

1. EXTERNAL - DETAILED: HORIZONTAL STABILIZER FRONT SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-211-003

(1) Do a Detailed inspection of the exposed surface of the front spar upper chord between the upper skin and Leading Edge (LE) skin.

See Doc. D626A001 - DTR, DTR check form 55-10-07-2, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR TERMINAL FITTING			55-618-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	12000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		332AB 332AT			331 332
		NOTE			

Inspect (Penetrant) the upper and lower Side of Body (SOB) spar bolts on the horizontal stabilizer front spar terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-07-3, for alternative inspection.

ACCESS NOTE: Removal of upper and lower spar bolts is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-618-00-01

MECH	INSP

TASK 55-05-02-230-802

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-017

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer

NOTE: Removal of upper and lower spar bolts is required.

SUBTASK 55-05-02-230-002

(2) Do a Dye Penetrant inspection of the upper and lower Side of Body (SOB) spar bolts on the horizontal stabilizer front spar terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-07-3, for alternative inspection.

SUBTASK 55-05-02-410-017

(3) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR TERMINAL FITTING			55-618-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 12000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342AB 342AT			ZONE 341 342
		NOTE			

Inspect (Penetrant) the upper and lower Side of Body (SOB) spar bolts on the horizontal stabilizer front spar terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-07-3, for alternative inspection.

ACCESS NOTE: Removal of upper and lower spar bolts is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-618-00-02

MECH	INSP

TASK 55-05-02-230-802

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR TERMINAL FITTING

A. Inspection

SUBTASK 55-05-02-010-017

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer

NOTE: Removal of upper and lower spar bolts is required.

SUBTASK 55-05-02-230-002

(2) Do a Dye Penetrant inspection of the upper and lower Side of Body (SOB) spar bolts on the horizontal stabilizer front spar terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-10-07-3, for alternative inspection.

SUBTASK 55-05-02-410-017

(3) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD			55-618-10-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332AB 332AT			ZONE 331 332
		NOTE			

Inspect (High frequency Eddy Current) the lower chord at the end rib tension fitting installation bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-07-4, for alternative inspection.

ACCESS NOTE: The inspection requires the removal of the bolt.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-618-10-01

MECH	INSP

TASK 55-05-02-250-880

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-142

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer

NOTE: The inspection requires the removal of the bolt.

SUBTASK 55-05-02-250-080

(2) Do a High frequency Eddy Current inspection of the lower chord at the end rib tension fitting installation bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-07-4, for alternative inspection.

SUBTASK 55-05-02-410-140

(3) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD			55-618-10-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342AB 342AT			ZONE 341 342
		NOTE			

Inspect (High frequency Eddy Current) the lower chord at the end rib tension fitting installation bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-07-4, for alternative inspection.

ACCESS NOTE: The inspection requires the removal of the bolt.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-618-10-02

MECH	INSP

TASK 55-05-02-250-880

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER FRONT SPAR LOWER CHORD

A. Inspection

SUBTASK 55-05-02-010-142

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer

NOTE: The inspection requires the removal of the bolt.

SUBTASK 55-05-02-250-080

(2) Do a High frequency Eddy Current inspection of the lower chord at the end rib tension fitting installation bolt hole.

See Doc. D626A001 - DTR, DTR check form 55-10-07-4, for alternative inspection.

SUBTASK 55-05-02-410-140

(3) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
342AT	Gap Cover, Horizontal Stabilizer

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-619-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333CB 333DB 333EB 334JB 334KB			ZONE 333 334
		NOTE			

Inspect (High Frequency Eddy Current) all of the holes on the hinge plate, around the bore, at elevator STA 121 and STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-08-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-35.

ACCESS NOTE: Removal of hinge plate from assembly is required prior to inspection. The spherical bearing should not be removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-619-00-01

MECH	INSP

TASK 55-05-02-250-816

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-024

(1) Open these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64

NOTE: Removal of hinge plate from assembly is required prior to inspection. The spherical bearing should not be removed.

SUBTASK 55-05-02-250-016

(2) Do a High Frequency Eddy Current inspection of all holes on the hinge plate, around the bore, at elevator STA 121 and STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-08-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-35.

SUBTASK 55-05-02-010-025

(3) Close these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-619-00-01

MECH	INSP

(Continued)

Number	Name/Location
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64

END OF TASK

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-619-00-02
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	R HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		343CB 343DB 343EB 344JB 344KB			343 344
		NOTE			

Inspect (High Frequency Eddy Current) all of the holes on the hinge plate, around the bore, at elevator STA 121 and STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-08-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-35.

ACCESS NOTE: Removal of hinge plate from assembly is required prior to inspection. The spherical bearing should not be removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-619-00-02

MECH	INSP

TASK 55-05-02-250-816

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-024

(1) Open these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64

NOTE: Removal of hinge plate from assembly is required prior to inspection. The spherical bearing should not be removed.

SUBTASK 55-05-02-250-016

(2) Do a High Frequency Eddy Current inspection of all holes on the hinge plate, around the bore, at elevator STA 121 and STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-08-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-35.

SUBTASK 55-05-02-010-025

(3) Close these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-619-00-02

MECH	INSP

(Continued)

Number	Name/Location
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64

END OF TASK

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-620-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		333FB 334LB 334MB 334MT 334NB 334NT			333 334
		NOTE			

Inspect (High Frequency Eddy Current) all of the holes on the hinge plate, around the bore, at elevator STA 213 and STA 250.

See Doc. D626A001 - DTR, DTR check form 55-10-08-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-35.

ACCESS NOTE: Removal of hinge plate from assembly is required prior to inspection. The spherical bearing should not be removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-620-00-01

MECH	INSP

TASK 55-05-02-250-817

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-026

(1) Open these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334MT	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344MT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

NOTE: Removal of hinge plate from assembly is required prior to inspection. The spherical bearing should not be removed.

SUBTASK 55-05-02-250-017

(2) Do a High Frequency Eddy Current inspection of all of the holes on the hinge plate, around the bore, at elevator STA 213 and STA 250.

See Doc. D626A001 - DTR, DTR check form 55-10-08-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-35.

SUBTASK 55-05-02-410-024

(3) Close these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334MT	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-620-00-01

MECH	INSP

(Continued)

Number	Name/Location
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344MT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-620-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343FB 344LB 344MB 344MT 344NB 344NT			ZONE 343 344
		NOTE			

Inspect (High Frequency Eddy Current) all of the holes on the hinge plate, around the bore, at elevator STA 213 and STA 250.

See Doc. D626A001 - DTR, DTR check form 55-10-08-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-35.

ACCESS NOTE: Removal of hinge plate from assembly is required prior to inspection. The spherical bearing should not be removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-620-00-02

MECH	INSP

TASK 55-05-02-250-817

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-026

(1) Open these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334MT	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344MT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

NOTE: Removal of hinge plate from assembly is required prior to inspection. The spherical bearing should not be removed.

SUBTASK 55-05-02-250-017

(2) Do a High Frequency Eddy Current inspection of all of the holes on the hinge plate, around the bore, at elevator STA 213 and STA 250.

See Doc. D626A001 - DTR, DTR check form 55-10-08-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-35.

SUBTASK 55-05-02-410-024

(3) Close these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334MT	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-620-00-02

MECH	INSP

(Continued)

Number	Name/Location
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344MT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER TRAILING EDGE HINGE PLATE LOWER FORWARD LUG			55-621-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 334NB 334NT			ZONE 333 334
		NOTE			

Inspect (High Frequency Eddy Current) all of the holes on the hinge plates around the bore at elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-08-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-36.

ACCESS NOTE: Removal of hinge plate is required. The spherical bearing should not be removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-621-00-01

MECH	INSP

TASK 55-05-02-250-818

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER TRAILING EDGE HINGE PLATE LOWER FORWARD LUG

A. Inspection

SUBTASK 55-05-02-010-027

(1) Open these access panels on the Left side:

Number	Name/Location
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

NOTE: Removal of hinge plate is required. The spherical bearing should not be removed.

SUBTASK 55-05-02-250-018

(2) Do a High Frequency Eddy Current inspection of all the holes on the hinge plates around the bore at elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-08-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-36.

SUBTASK 55-05-02-410-025

(3) Close these access panels on the Left side:

Number	Name/Location
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER TRAILING EDGE HINGE PLATE LOWER FORWARD LUG			55-621-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 344NB 344NT			ZONE 343 344
		NOTE			

Inspect (High Frequency Eddy Current) all of the holes on the hinge plates around the bore at elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-08-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-36.

ACCESS NOTE: Removal of hinge plate is required. The spherical bearing should not be removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-621-00-02

MECH	INSP

TASK 55-05-02-250-818

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER TRAILING EDGE HINGE PLATE LOWER FORWARD LUG

A. Inspection

SUBTASK 55-05-02-010-027

(1) Open these access panels on the Left side:

Number	Name/Location
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

NOTE: Removal of hinge plate is required. The spherical bearing should not be removed.

SUBTASK 55-05-02-250-018

(2) Do a High Frequency Eddy Current inspection of all the holes on the hinge plates around the bore at elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-08-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-36.

SUBTASK 55-05-02-410-025

(3) Close these access panels on the Left side:

Number	Name/Location
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-622-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333CB 334AT 334GB 334GT			ZONE 333 334
		NOTE			

Inspect (High Frequency Eddy Current) all holes on the hinge plate at elevator STA 66.

See Doc. D626A001 - DTR, DTR check form 55-10-08-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-36.

ACCESS NOTE: Removal of hinge plate is required. The spherical bearing should not be removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-622-00-01

MECH	INSP

TASK 55-05-02-250-819

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-028

(1) Open these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

NOTE: Removal of hinge plate is required. The spherical bearing should not be removed.

SUBTASK 55-05-02-250-019

(2) Do a High Frequency Eddy Current inspection of all holes on the hinge plate at elevator STA 66.

See Doc. D626A001 - DTR, DTR check form 55-10-08-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-36.

SUBTASK 55-05-02-410-026

(3) Close these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-622-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343CB 344AT 344GB 344GT			ZONE 343 344
		NOTE			

Inspect (High Frequency Eddy Current) all holes on the hinge plate at elevator STA 66.

See Doc. D626A001 - DTR, DTR check form 55-10-08-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-36.

ACCESS NOTE: Removal of hinge plate is required. The spherical bearing should not be removed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-622-00-02

MECH	INSP

TASK 55-05-02-250-819

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-028

(1) Open these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

NOTE: Removal of hinge plate is required. The spherical bearing should not be removed.

SUBTASK 55-05-02-250-019

(2) Do a High Frequency Eddy Current inspection of all holes on the hinge plate at elevator STA 66.

See Doc. D626A001 - DTR, DTR check form 55-10-08-4, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-36.

SUBTASK 55-05-02-410-026

(3) Close these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS TRAILING EDGE CLEVIS			55-623-01-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL NOTE
STATION	SKILL AIRPL				
		ACCESS 333CB 333DB 333EB 334JB 334KB			ZONE 333 334

Inspect (High Frequency Eddy Current) around the outer edge of the bushing flange on both the upper and lower trailing edge clevis lugs at elevator STA 121 and elevator STA 176.

See Doc. D626A001-DTR, DTR check form 55-10-08-5B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

AIRPLANE NOTE: This inspection applies to LN # 216 and on with hinge rib clevis lug thickness of t=0.25 inch.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-623-01-01

MECH	INSP

TASK 55-05-02-250-821

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS TRAILING EDGE CLEVIS

A. Inspection

SUBTASK 55-05-02-010-032

(1) Open these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64

SUBTASK 55-05-02-250-021

(2) Do a High Frequency Eddy Current inspection around the outer edge of the bushing flange on both the upper and lower trailing edge clevis lugs at elevator STA 121 and elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-08-5B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

SUBTASK 55-05-02-410-030

(3) Close these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-623-01-01

MECH	INSP

(Continued)

Number	Name/Location
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64

END OF TASK

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS TRAILING EDGE CLEVIS			55-623-01-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				NOTE
		ACCESS 343CB 343DB 343EB 344JB 344KB			ZONE 343 344

Inspect (High Frequency Eddy Current) around the outer edge of the bushing flange on both the upper and lower trailing edge clevis lugs at elevator STA 121 and elevator STA 176.

See Doc. D626A001-DTR, DTR check form 55-10-08-5B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

AIRPLANE NOTE: This inspection applies to LN # 216 and on with hinge rib clevis lug thickness of t=0.25 inch.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-623-01-02

MECH	INSP

TASK 55-05-02-250-821

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS TRAILING EDGE CLEVIS

A. Inspection

SUBTASK 55-05-02-010-032

(1) Open these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64

SUBTASK 55-05-02-250-021

(2) Do a High Frequency Eddy Current inspection around the outer edge of the bushing flange on both the upper and lower trailing edge clevis lugs at elevator STA 121 and elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-08-5B, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

SUBTASK 55-05-02-410-030

(3) Close these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-623-01-02

MECH	INSP

(Continued)

Number	Name/Location
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64

END OF TASK

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-624-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		333CB 334AT 334GB 334GT			333 334

Inspect (High Frequency Eddy Current) around the bushing flanges on both the upper and lower clevis lugs at elevator STA 66.

See Doc. D626A001-DTR, DTR check form 55-10-08-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-624-00-01

MECH	INSP

TASK 55-05-02-250-822

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-033

(1) Open these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

SUBTASK 55-05-02-250-022

(2) Do a High Frequency Eddy Current inspection around the bushing flanges on both the upper and lower clevis lugs at elevator STA 66.

See Doc. D626A001 - DTR, DTR check form 55-10-08-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55–10–23.

SUBTASK 55-05-02-410-031

(3) Close these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-624-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343CB 344AT 344GB 344GT			ZONE 343 344

Inspect (High Frequency Eddy Current) around the bushing flanges on both the upper and lower clevis lugs at elevator STA 66.

See Doc. D626A001-DTR, DTR check form 55-10-08-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-624-00-02

MECH	INSP

TASK 55-05-02-250-822

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-033

(1) Open these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

SUBTASK 55-05-02-250-022

(2) Do a High Frequency Eddy Current inspection around the bushing flanges on both the upper and lower clevis lugs at elevator STA 66.

See Doc. D626A001 - DTR, DTR check form 55-10-08-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55–10–23.

SUBTASK 55-05-02-410-031

(3) Close these access panels on the Left side:

Number	Name/Location
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-625-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333FB 334LB 334MB 334NB 334NT			ZONE 333 334

Inspect (High Frequency Eddy Current) around the bushing flanges at both the upper and lower clevis lugs at elevator STA 213 and elevator STA 250.

See Doc. D626A001 - DTR, DTR check form 55-10-08-7, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-625-00-01

MECH	INSP

TASK 55-05-02-250-823

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-034

(1) Open these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

(2) For the Right side, open these access panels:

Number	Name/Location
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

SUBTASK 55-05-02-250-023

(3) Do a High Frequency Eddy Current inspection around the bushing flanges at both the upper and lower clevis lugs at elevator STA 213 and elevator STA 250.

See Doc. D626A001 - DTR, DTR check form 55-10-08-7, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

SUBTASK 55-05-02-410-032

(4) For the Left side, close these access panels:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

(5) For the Right side, close these access panels:

Number	Name/Location
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-625-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 334FB 344LB 344MB 344NB 344NT			ZONE 343 344

Inspect (High Frequency Eddy Current) around the bushing flanges at both the upper and lower clevis lugs at elevator STA 213 and elevator STA 250.

See Doc. D626A001 - DTR, DTR check form 55-10-08-7, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-625-00-02

MECH	INSP

TASK 55-05-02-250-823

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-034

(1) Open these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

(2) For the Right side, open these access panels:

Number	Name/Location
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

SUBTASK 55-05-02-250-023

(3) Do a High Frequency Eddy Current inspection around the bushing flanges at both the upper and lower clevis lugs at elevator STA 213 and elevator STA 250.

See Doc. D626A001 - DTR, DTR check form 55-10-08-7, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

SUBTASK 55-05-02-410-032

(4) For the Left side, close these access panels:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

(5) For the Right side, close these access panels:

Number	Name/Location
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-626-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	18000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		334NB 334NT			333 334

Inspect (High Frequency Eddy Current) around the edge of the bushing at both the upper and lower clevis lugs at elevator STA 265.

Note: Bushing removal not required.

See Doc. D626A001 - DTR, DTR check form 55-10-08-8, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-626-00-01

MECH	INSP

TASK 55-05-02-250-824

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-035

(1) Open these access panels on the Left side:

Number	Name/Location
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

SUBTASK 55-05-02-250-024

(2) Do a High Frequency Eddy Current inspection around the edge of the bushing at both the upper and lower clevis lugs at elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-08-8, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

NOTE: Bushing removal not required.

SUBTASK 55-05-02-410-033

(3) Close these access panels on the Left side:

Number	Name/Location
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-626-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 344NB 344NT			ZONE 343 344

Inspect (High Frequency Eddy Current) around the edge of the bushing at both the upper and lower clevis lugs at elevator STA 265.

Note: Bushing removal not required.

See Doc. D626A001 - DTR, DTR check form 55-10-08-8, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-626-00-02

MECH	INSP

TASK 55-05-02-250-824

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-035

(1) Open these access panels on the Left side:

Number	Name/Location
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

SUBTASK 55-05-02-250-024

(2) Do a High Frequency Eddy Current inspection around the edge of the bushing at both the upper and lower clevis lugs at elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-08-8, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-23.

NOTE: Bushing removal not required.

SUBTASK 55-05-02-410-033

(3) Close these access panels on the Left side:

Number	Name/Location
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER TRAILING EDGE ELEVATOR HINGE RIB			55-627-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 12000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333BB 333CB 333DB			ZONE 333 334
		NOTE			

Inspect (High Frequency Eddy Current) the elevator hinge rib chord around the fasteners common to the upper and lower spar chords at elevator STA 66 and elevator STA 121.

See Doc. D626A001 - DTR, DTR check form 55-10-08-9, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

ACCESS NOTE: Removal of lower composite skin panels is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-627-00-01

MECH	INSP

TASK 55-05-02-250-825

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER TRAILING EDGE ELEVATOR HINGE RIB

A. Inspection

SUBTASK 55-05-02-010-075

(1) Open these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of lower composite skin panels is required.

SUBTASK 55-05-02-250-025

(2) Do a High Frequency Eddy Current inspection of the elevator hinge rib chord around the fasteners common to the upper and lower spar chords at elevator STA 66 and elevator STA 121.

See Doc. D626A001 - DTR, DTR check form 55-10-08-9, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

SUBTASK 55-05-02-410-073

(3) Close these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER TRAILING EDGE ELEVATOR HINGE RIB			55-627-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 12000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343BB 343CB 343DB			ZONE 343 344
		NOTE			

Inspect (High Frequency Eddy Current) the elevator hinge rib chord around the fasteners common to the upper and lower spar chords at elevator STA 66 and elevator STA 121.

See Doc. D626A001 - DTR, DTR check form 55-10-08-9, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

ACCESS NOTE: Removal of lower composite skin panels is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-627-00-02

MECH	INSP

TASK 55-05-02-250-825

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER TRAILING EDGE ELEVATOR HINGE RIB

A. Inspection

SUBTASK 55-05-02-010-075

(1) Open these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of lower composite skin panels is required.

SUBTASK 55-05-02-250-025

(2) Do a High Frequency Eddy Current inspection of the elevator hinge rib chord around the fasteners common to the upper and lower spar chords at elevator STA 66 and elevator STA 121.

See Doc. D626A001 - DTR, DTR check form 55-10-08-9, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

SUBTASK 55-05-02-410-073

(3) Close these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER, TRAILING EDGE HINGE RIB TO REAR SPAR ATTACHMENT			55-628-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333DB 333EB			ZONE 333 334
		NOTE			

Inspect (High Frequency Eddy Current) the rib chord around the fasteners common to the stabilizer upper and lower rear spar chord at elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-8-10, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

ACCESS NOTE: Removal of lower trailing edge panels is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-628-00-01

MECH	INSP

TASK 55-05-02-250-826

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER, TRAILING EDGE HINGE RIB TO REAR SPAR ATTACHMENT

A. Inspection

SUBTASK 55-05-02-010-036

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of lower trailing edge panels is required.

SUBTASK 55-05-02-250-026

(2) Do a High Frequency Eddy Current inspection of the rib chord around the fasteners common to the stabilizer upper and lower rear spar chord at Elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-8-10, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

SUBTASK 55-05-02-410-042

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER, TRAILING EDGE HINGE RIB TO REAR SPAR ATTACHMENT			55-628-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343DB 343EB			ZONE 343 344
		NOTE			

Inspect (High Frequency Eddy Current) the rib chord around the fasteners common to the stabilizer upper and lower rear spar chord at elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-8-10, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

ACCESS NOTE: Removal of lower trailing edge panels is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-628-00-02

MECH	INSP

TASK 55-05-02-250-826

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER, TRAILING EDGE HINGE RIB TO REAR SPAR ATTACHMENT

A. Inspection

SUBTASK 55-05-02-010-036

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of lower trailing edge panels is required.

SUBTASK 55-05-02-250-026

(2) Do a High Frequency Eddy Current inspection of the rib chord around the fasteners common to the stabilizer upper and lower rear spar chord at Elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-8-10, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

SUBTASK 55-05-02-410-042

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-629-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 30000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333FB 334FB			ZONE 333 334
		NOTE			

Inspect (High Frequency Eddy Current) the hinge rib to rear spar attachment, on both the upper and lower chords, at elevator STA 213, elevator STA 250 and elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-8-11, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

ACCESS NOTE: Removal of elevator and cover panel is required. No fastener removal is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-629-00-01

MECH	INSP

TASK 55-05-02-250-827

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-043

(1) Open these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

Open these access panels on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

NOTE: Removal of elevator and cover panel is required. No fastener removal is required.

SUBTASK 55-05-02-250-027

(2) Do a High Frequency Eddy Current inspection of the hinge rib to rear spar attachment, on both the upper and lower chords, at elevator STA 213, elevator STA 250 and elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-8-11, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

SUBTASK 55-05-02-410-040

(3) Close these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

Close these access panels on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-629-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 30000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343FB 344FB			ZONE 343 344
		NOTE			

Inspect (High Frequency Eddy Current) the hinge rib to rear spar attachment, on both the upper and lower chords, at elevator STA 213, elevator STA 250 and elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-8-11, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

ACCESS NOTE: Removal of elevator and cover panel is required. No fastener removal is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-629-00-02

MECH	INSP

TASK 55-05-02-250-827

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-043

(1) Open these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

Open these access panels on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

NOTE: Removal of elevator and cover panel is required. No fastener removal is required.

SUBTASK 55-05-02-250-027

(2) Do a High Frequency Eddy Current inspection of the hinge rib to rear spar attachment, on both the upper and lower chords, at elevator STA 213, elevator STA 250 and elevator STA 265.

See Doc. D626A001 - DTR, DTR check form 55-10-8-11, alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-37.

SUBTASK 55-05-02-410-040

(3) Close these access panels on the Left side:

Number	Name/Location
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

Close these access panels on the Right side:

Number	Name/Location
343FB	Horizontal Stabilizer, Access Panel - T.E. Area
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-630-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333BB 333CB 333DB			ZONE 333 334
		NOTE			

Inspect (High Frequency Eddy Current) around the fastener holes that are used to attach the composite skin panels to the rib and the upper chord and lower chord fastener locations between the rear spar and trailing edge beam at elevator STA 66 and elevator STA 121.

See Doc. D626A001 - DTR, DTR check form 55-10-8-13, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-33.

ACCESS NOTE: Removal of lower composite skin panels is required. No fastener removal required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-630-00-01

MECH	INSP

TASK 55-05-02-250-828

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-041

(1) Open these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of lower composite skin panels is required. No fastener removal required.

SUBTASK 55-05-02-250-028

(2) Do a High Frequency Eddy Current inspection around the fastener holes that are used to attach the composite skin panels to the rib and the upper chord and lower chord fastener locations between the rear spar and trailing edge beam at Elevator STA 66 and elevator STA 121.

See Doc. D626A001 - DTR, DTR check form 55-10-8-13, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-33.

SUBTASK 55-05-02-410-038

(3) Close these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-630-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343BB 343CB 343DB			ZONE 343 344
		NOTE			

Inspect (High Frequency Eddy Current) around the fastener holes that are used to attach the composite skin panels to the rib and the upper chord and lower chord fastener locations between the rear spar and trailing edge beam at elevator STA 66 and elevator STA 121.

See Doc. D626A001 - DTR, DTR check form 55-10-8-13, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-33.

ACCESS NOTE: Removal of lower composite skin panels is required. No fastener removal required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-630-00-02

MECH	INSP

TASK 55-05-02-250-828

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-041

(1) Open these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of lower composite skin panels is required. No fastener removal required.

SUBTASK 55-05-02-250-028

(2) Do a High Frequency Eddy Current inspection around the fastener holes that are used to attach the composite skin panels to the rib and the upper chord and lower chord fastener locations between the rear spar and trailing edge beam at Elevator STA 66 and elevator STA 121.

See Doc. D626A001 - DTR, DTR check form 55-10-8-13, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-33.

SUBTASK 55-05-02-410-038

(3) Close these access panels on the Left side:

Number	Name/Location
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-631-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	18000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		333DB 333EB			333 334
		NOTE			

Inspect (High Frequency Eddy Current) both upper and lower rib chords around the fastener holes common to the trailing edge panels between the stabilizer rear spar and the trailing edge beam at elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-8-14, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-33.

ACCESS NOTE: Removal of the lower trailing edge skin panels is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-631-00-01

MECH	INSP

TASK 55-05-02-250-829

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-039

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of the lower trailing edge skin panels is required.

SUBTASK 55-05-02-250-029

(2) Do a High Frequency Eddy Current inspection of both upper and lower rib chords around the fastener holes common to the trailing edge panels between the stabilizer rear spar and the trailing edge beam at elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-08-14, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-33.

SUBTASK 55-05-02-410-036

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER ELEVATOR HINGE RIBS			55-631-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 343DB 343EB			ZONE 343 344
		NOTE			

Inspect (High Frequency Eddy Current) both upper and lower rib chords around the fastener holes common to the trailing edge panels between the stabilizer rear spar and the trailing edge beam at elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-8-14, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-33.

ACCESS NOTE: Removal of the lower trailing edge skin panels is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-631-00-02

MECH	INSP

TASK 55-05-02-250-829

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER ELEVATOR HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-039

(1) Open these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge

Open these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area

NOTE: Removal of the lower trailing edge skin panels is required.

SUBTASK 55-05-02-250-029

(2) Do a High Frequency Eddy Current inspection of both upper and lower rib chords around the fastener holes common to the trailing edge panels between the stabilizer rear spar and the trailing edge beam at elevator STA 176.

See Doc. D626A001 - DTR, DTR check form 55-10-08-14, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-33.

SUBTASK 55-05-02-410-036

(3) Close these access panels on the Left side:

Number	Name/Location
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge

Close these access panels on the Right side:

Number	Name/Location
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER UPPER INSPAR SKINS			55-631-10-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 332

Inspect (High Frequency Eddy Current) around all fastener locations in the upper inspar skin between the SOB rib and the rib at stabilizer STA. 157.1.

See Doc. D626A001 - DTR, DTR check form 55-10-09-1, alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-631-10-01

MECH	INSP

TASK 55-05-02-250-809

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER UPPER INSPAR SKIN

A. Inspection

SUBTASK 55-05-02-250-009

(1) Do a High Frequency Eddy Current inspection around all fastener locations in the upper inspar skin between the SOB rib and the rib at stabilizer STA. 157.1.

See Doc. D626A001 - DTR, DTR check form 55-10-09-1, alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER UPPER INSPAR SKIN			55-631-10-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342

Inspect (High Frequency Eddy Current) around all fastener locations in the upper inspar skin between the SOB rib and the rib at stabilizer STA. 157.1.

See Doc. D626A001 - DTR, DTR check form 55-10-09-1, alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-631-10-02

MECH	INSP

TASK 55-05-02-250-809

1. EXTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER UPPER INSPAR SKIN

A. Inspection

SUBTASK 55-05-02-250-009

(1) Do a High Frequency Eddy Current inspection around all fastener locations in the upper inspar skin between the SOB rib and the rib at stabilizer STA. 157.1.

See Doc. D626A001 - DTR, DTR check form 55-10-09-1, alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER LOWER INSPAR SKIN			55-633-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 332

Inspect (Detailed) the horizontal stabilizer lower inspar skin from side-of-body to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-10-1, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-633-00-01

MECH	INSP

TASK 55-05-02-211-805

1. EXTERNAL - DETALED: HORIZONTAL STABILIZER LOWER INSPAR SKIN

A. Inspection

SUBTASK 55-05-02-211-005

(1) Do a Detailed inspection of the horizontal stabilizer lower inspar skin from side-of-body to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-10-1, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER LOWER INSPAR SKIN			55-633-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 342

Inspect (Detailed) the horizontal stabilizer lower inspar skin from side-of-body to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-10-1, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-633-00-02

MECH	INSP

TASK 55-05-02-211-805

1. EXTERNAL - DETALED: HORIZONTAL STABILIZER LOWER INSPAR SKIN

A. Inspection

SUBTASK 55-05-02-211-005

(1) Do a Detailed inspection of the horizontal stabilizer lower inspar skin from side-of-body to the tip.

See Doc. D626A001 - DTR, DTR check form 55-10-10-1, for alternative inspection.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION FRONT SPAR UPPER CHORD			55-634-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 331A 331AZ			ZONE 313
		NOTE			

Inspect (Detailed) all four (4) front spar upper side of body clevis lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-11-1, for alternative inspection.

ACCESS NOTE: Removal of leading edge and gap seals is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-634-00-01

MECH	INSP

TASK 55-05-02-211-806

1. INTERNAL - DETAILED: HORIZONTAL STABILIZER CENTER SECTION FRONT SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-029

(1) Open these access panels on the Left side:

Number	Name/Location
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
331AZ	Horizontal Stabilizer, Access Panel, Inboard L.E. Closure Rib

Open these access panels on the Right side:

Number	Name/Location
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
341AZ	Horizontal Stabilizer, Access Panel - Inbd L.E. Closure Rib

NOTE: Removal of leading edge and gap seals is required.

SUBTASK 55-05-02-211-006

(2) Do a Detailed inspection of all four (4) front spar upper side of body clevis lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-11-1, for alternative inspection.

SUBTASK 55-05-02-410-027

(3) Close these access panels on the Left side:

Number	Name/Location
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
331AZ	Horizontal Stabilizer, Access Panel, Inboard L.E. Closure Rib

Close these access panels on the Right side:

Number	Name/Location
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
341AZ	Horizontal Stabilizer, Access Panel - Inbd L.E. Closure Rib

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION FRONT SPAR UPPER CHORD			55-634-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 341A 341AZ			ZONE 314
		NOTE			

Inspect (Detailed) all four (4) front spar upper side of body clevis lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-11-1, for alternative inspection.

ACCESS NOTE: Removal of leading edge and gap seals is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-634-00-02

MECH	INSP

TASK 55-05-02-211-806

1. INTERNAL - DETAILED: HORIZONTAL STABILIZER CENTER SECTION FRONT SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-029

(1) Open these access panels on the Left side:

Number	Name/Location
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
331AZ	Horizontal Stabilizer, Access Panel, Inboard L.E. Closure Rib

Open these access panels on the Right side:

Number	Name/Location
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
341AZ	Horizontal Stabilizer, Access Panel - Inbd L.E. Closure Rib

NOTE: Removal of leading edge and gap seals is required.

SUBTASK 55-05-02-211-006

(2) Do a Detailed inspection of all four (4) front spar upper side of body clevis lugs.

See Doc. D626A001 - DTR, DTR check form 55-10-11-1, for alternative inspection.

SUBTASK 55-05-02-410-027

(3) Close these access panels on the Left side:

Number	Name/Location
331A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
331AZ	Horizontal Stabilizer, Access Panel, Inboard L.E. Closure Rib

Close these access panels on the Right side:

Number	Name/Location
341A	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer To Body
341AZ	Horizontal Stabilizer, Access Panel - Inbd L.E. Closure Rib

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR LUGS			55-635-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		332AT 333AT			313 332 333
		NOTE			

Inspect (Ultrasonic) all four lower pivot fitting clevis lugs for hidden damage at the lug bore.

See Doc. D626A001 - DTR, DTR check form 55-10-12-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-05.

ACCESS NOTE: Removal of the gap covers and sliding seals is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-635-00-01

MECH	INSP

TASK 55-05-02-130-805

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR LUGS

A. Inspection

SUBTASK 55-05-02-010-003

(1) Open these access panels on the Left side:

Number	Name/Location
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AT	Gap Cover, Horizontal Stabilizer
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

NOTE: Removal of the gap covers and sliding seals is required.

SUBTASK 55-05-02-130-005

(2) Do an Ultrasonic inspection of all four (4) Lower Pivot Fitting Clevis Lugs for hidden damage at the lug bore.

See Doc. D626A001 - DTR, DTR check form 55-10-12-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-05.

SUBTASK 55-05-02-410-003

(3) Close these access panels on the Left side:

Number	Name/Location
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AT	Gap Cover, Horizontal Stabilizer
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR LUGS			55-635-00-02
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	R HORZ STAB	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		342AT 343AT			314 342 343
		NOTE			

Inspect (Ultrasonic) all four lower pivot fitting clevis lugs for hidden damage at the lug bore.

See Doc. D626A001 - DTR, DTR check form 55-10-12-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-05.

ACCESS NOTE: Removal of the gap covers and sliding seals is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-635-00-02

MECH	INSP

TASK 55-05-02-130-805

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR LUGS

A. Inspection

SUBTASK 55-05-02-010-003

(1) Open these access panels on the Left side:

Number	Name/Location
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AT	Gap Cover, Horizontal Stabilizer
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

NOTE: Removal of the gap covers and sliding seals is required.

SUBTASK 55-05-02-130-005

(2) Do an Ultrasonic inspection of all four (4) Lower Pivot Fitting Clevis Lugs for hidden damage at the lug bore.

See Doc. D626A001 - DTR, DTR check form 55-10-12-1, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-05.

SUBTASK 55-05-02-410-003

(3) Close these access panels on the Left side:

Number	Name/Location
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AT	Gap Cover, Horizontal Stabilizer
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-636-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 332AB 333AT			ZONE 313 332 333
		NOTE			

Inspect (Ultrasonic) the side of body lug bore on the horizontal stabilizer center section rear spar upper chord.

See Doc. D626A001 - DTR, DTR check form 55-10-12-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-08.

ACCESS NOTE: Removal of gap covers is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-636-00-01

MECH	INSP

TASK 55-05-02-130-806

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-004

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

NOTE: Removal of gap covers is required.

SUBTASK 55-05-02-130-006

(2) Do an Ultrasonic inspection of the side of body lug bore on the horizontal stabilizer center section rear spar upper chord.

See Doc. D626A001 - DTR, DTR check form 55-10-12-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-08.

SUBTASK 55-05-02-410-004

(3) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD			55-636-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R HORZ STAB	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342AB 343AT			ZONE 314 342 343
		NOTE			

Inspect (Ultrasonic) the side of body lug bore on the horizontal stabilizer center section rear spar upper chord.

See Doc. D626A001 - DTR, DTR check form 55-10-12-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-08.

ACCESS NOTE: Removal of gap covers is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-636-00-02

MECH	INSP

TASK 55-05-02-130-806

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION REAR SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-004

(1) Open these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Open these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

NOTE: Removal of gap covers is required.

SUBTASK 55-05-02-130-006

(2) Do an Ultrasonic inspection of the side of body lug bore on the horizontal stabilizer center section rear spar upper chord.

See Doc. D626A001 - DTR, DTR check form 55-10-12-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-08.

SUBTASK 55-05-02-410-004

(3) Close these access panels on the Left side:

Number	Name/Location
332AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

Close these access panels on the Right side:

Number	Name/Location
342AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION FRONT SPAR UPPER CHORD			55-637-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313

Inspect (Detailed) the thrust beam to spar joint on the horizontal stabilizer center section front spar upper chord.

See Doc. D626A001 - DTR, DTR check form 55-10-13-2, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-637-00-01

MECH	INSP

TASK 55-05-02-211-807

1. INTERNAL - DETAILED: HORIZONTAL STABILIZER CENTER SECTION FRONT SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-012

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

SUBTASK 55-05-02-211-007

(2) Do a Detailed inspection of the thrust beam to spar joint on the horizontal stabilizer center section front spar upper chord.

See Doc. D626A001 - DTR, DTR check form 55-10-13-2, for alternative inspection.

SUBTASK 55-05-02-410-012

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION FRONT SPAR UPPER CHORD			55-637-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314

Inspect (Detailed) the thrust beam to spar joint on the horizontal stabilizer center section front spar upper chord.

See Doc. D626A001 - DTR, DTR check form 55-10-13-2, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-637-00-02

MECH	INSP

TASK 55-05-02-211-807

1. INTERNAL - DETAILED: HORIZONTAL STABILIZER CENTER SECTION FRONT SPAR UPPER CHORD

A. Inspection

SUBTASK 55-05-02-010-012

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

SUBTASK 55-05-02-211-007

(2) Do a Detailed inspection of the thrust beam to spar joint on the horizontal stabilizer center section front spar upper chord.

See Doc. D626A001 - DTR, DTR check form 55-10-13-2, for alternative inspection.

SUBTASK 55-05-02-410-012

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS			55-638-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (Ultrasonic) the upper and lower chords on the primary beam at the front and rear spar joints.

See Doc. D626A001 - DTR, DTR check form 55-10-14-1, for alternative inspection.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-638-00-01

MECH	INSP

TASK 55-05-02-130-807

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-005

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-007

(2) Do an Ultrasonic inspection of the upper and lower chords on the primary beam at the front and rear spar joints.

See Doc. D626A001 - DTR, DTR check form 55-10-14-1, for alternative inspection.

SUBTASK 55-05-02-410-005

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS			55-638-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (Ultrasonic) the upper and lower chords on the primary beam at the front and rear spar joints.

See Doc. D626A001 - DTR, DTR check form 55-10-14-1, for alternative inspection.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-638-00-02

MECH	INSP

TASK 55-05-02-130-807

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-005

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-007

(2) Do an Ultrasonic inspection of the upper and lower chords on the primary beam at the front and rear spar joints.

See Doc. D626A001 - DTR, DTR check form 55-10-14-1, for alternative inspection.

SUBTASK 55-05-02-410-005

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS			55-639-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (High Frequency Eddy Current) the upper and lower chord splice members of the upper primary beam at the front and rear spar joint plates and angles.

See Doc. D626A001 - DTR, DTR check form 55-10-14-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-38.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-639-00-01

MECH	INSP

TASK 55-05-02-250-830

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-038

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-030

(2) Do a High Frequency Eddy Current inspection of the upper and lower chord splice members of the upper primary beam at the front and rear spar joint plates and angles.

See Doc. D626A001 - DTR, DTR check form 55-10-14-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-38.

SUBTASK 55-05-02-410-035

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS			55-639-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (High Frequency Eddy Current) the upper and lower chord splice members of the upper primary beam at the front and rear spar joint plates and angles.

See Doc. D626A001 - DTR, DTR check form 55-10-14-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-38.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-639-00-02

MECH	INSP

TASK 55-05-02-250-830

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-038

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-030

(2) Do a High Frequency Eddy Current inspection of the upper and lower chord splice members of the upper primary beam at the front and rear spar joint plates and angles.

See Doc. D626A001 - DTR, DTR check form 55-10-14-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-38.

SUBTASK 55-05-02-410-035

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS			55-640-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (Ultrasonic) the upper primary beam chord for damage under the thrust beam chord around fasteners common to the two chords and splice angle.

See Doc. D626A001 - DTR, DTR check form 55-10-14-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-14.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-640-00-01

MECH	INSP

TASK 55-05-02-130-808

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-006

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-008

(2) Do an Ultrasonic inspection of the upper primary beam chord for damage under the thrust beam chord around fasteners common to the two chords and splice angle.

See Doc. D626A001 - DTR, DTR check form 55-10-14-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-14.

SUBTASK 55-05-02-410-006

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS			55-640-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (Ultrasonic) the upper primary beam chord for damage under the thrust beam chord around fasteners common to the two chords and splice angle.

See Doc. D626A001 - DTR, DTR check form 55-10-14-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-14.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-640-00-02

MECH	INSP

TASK 55-05-02-130-808

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PRIMARY AND THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-006

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-008

(2) Do an Ultrasonic inspection of the upper primary beam chord for damage under the thrust beam chord around fasteners common to the two chords and splice angle.

See Doc. D626A001 - DTR, DTR check form 55-10-14-3, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-14.

SUBTASK 55-05-02-410-006

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS			55-641-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 21000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (Ultrasonic) the thrust beam upper and lower chords around the first row of fasteners common to the splice members at both the front and rear spar joints.

See Doc. D626A001 - DTR, DTR check form 55-10-14-4, for alternative inspection.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-641-00-01

MECH	INSP

TASK 55-05-02-130-809

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-007

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-009

(2) Do an Ultrasonic inspection of the thrust beam upper and lower chords around the first row of fasteners common to the splice members at both the front and rear spar joints.

See Doc. D626A001 - DTR, DTR check form 55-10-14-4, for alternative inspection.

SUBTASK 55-05-02-410-007

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS			55-641-00-02
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	56000 FC	21000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		311BL			314
		NOTE			

Inspect (Ultrasonic) the thrust beam upper and lower chords around the first row of fasteners common to the splice members at both the front and rear spar joints.

See Doc. D626A001 - DTR, DTR check form 55-10-14-4, for alternative inspection.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-641-00-02

MECH	INSP

TASK 55-05-02-130-809

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-007

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-009

(2) Do an Ultrasonic inspection of the thrust beam upper and lower chords around the first row of fasteners common to the splice members at both the front and rear spar joints.

See Doc. D626A001 - DTR, DTR check form 55-10-14-4, for alternative inspection.

SUBTASK 55-05-02-410-007

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS			55-642-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	56000 FC	34000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		311BL			313
		NOTE			

Inspect (High Frequency Eddy Current) the upper and lower thrust beam chords around the first row of fasteners common to the splice members at both the front and rear spars.

See Doc. D626A001 - DTR, DTR check form 55-10-14-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-39.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-642-00-01

MECH	INSP

TASK 55-05-02-250-831

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-037

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-031

(2) Do a High Frequency Eddy Current inspection of the upper and lower thrust beam chords around the first row of fasteners common to the splice members at both the front and rear spars.

See Doc. D626A001 - DTR, DTR check form 55-10-14-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-39.

SUBTASK 55-05-02-410-034

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS			55-642-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 34000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (High Frequency Eddy Current) the upper and lower thrust beam chords around the first row of fasteners common to the splice members at both the front and rear spars.

See Doc. D626A001 - DTR, DTR check form 55-10-14-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-39.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-642-00-02

MECH	INSP

TASK 55-05-02-250-831

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-037

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-031

(2) Do a High Frequency Eddy Current inspection of the upper and lower thrust beam chords around the first row of fasteners common to the splice members at both the front and rear spars.

See Doc. D626A001 - DTR, DTR check form 55-10-14-5, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-39.

SUBTASK 55-05-02-410-034

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS			55-643-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	56000 FC	14000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		311BL			313
		NOTE			

Inspect (Ultrasonic) upper and lower thrust beam chords forward and aft of the intersection joint (4 places).

See Doc. D626A001 - DTR, DTR check form 55-10-14-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-09.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-643-00-01

MECH	INSP

TASK 55-05-02-130-810

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-008

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-010

(2) Do an Ultrasonic inspection of the upper and lower thrust beam chords forward and aft of the intersection joint (4 places).

See Doc. D626A001 - DTR, DTR check form 55-10-14-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-09.

SUBTASK 55-05-02-410-008

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS			55-643-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 14000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (Ultrasonic) upper and lower thrust beam chords forward and aft of the intersection joint (4 places).

See Doc. D626A001 - DTR, DTR check form 55-10-14-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-09.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-643-00-02

MECH	INSP

TASK 55-05-02-130-810

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-008

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-010

(2) Do an Ultrasonic inspection of the upper and lower thrust beam chords forward and aft of the intersection joint (4 places).

See Doc. D626A001 - DTR, DTR check form 55-10-14-6, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-09.

SUBTASK 55-05-02-410-008

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAM INTERSECTION			55-644-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 27000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (High Frequency Eddy Current) upper and lower thrust beam splice plates.

See Doc. D626A001 - DTR, DTR check form 55-10-14-7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-20.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-644-00-01

MECH	INSP

TASK 55-05-02-250-832

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAM INTERSECTION

A. Inspection

SUBTASK 55-05-02-010-046

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-032

(2) Do a High Frequency Eddy Current inspection of the upper and lower thrust beam splice plates.

See Doc. D626A001 - DTR, DTR check form 55-10-14-7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-20.

SUBTASK 55-05-02-410-044

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAM INTERSECTION			55-644-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 27000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (High Frequency Eddy Current) upper and lower thrust beam splice plates.

See Doc. D626A001 - DTR, DTR check form 55-10-14-7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-20.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-644-00-02

MECH	INSP

TASK 55-05-02-250-832

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAM INTERSECTION

A. Inspection

SUBTASK 55-05-02-010-046

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-032

(2) Do a High Frequency Eddy Current inspection of the upper and lower thrust beam splice plates.

See Doc. D626A001 - DTR, DTR check form 55-10-14-7, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-20.

SUBTASK 55-05-02-410-044

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAM			55-645-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 13000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (Ultrasonic) the upper thrust beam chord for damage around the four fasteners common to the primary beam chord.

See Doc. D626A001 - DTR, DTR check form 55-10-14-8, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-10.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-645-00-01

MECH	INSP

TASK 55-05-02-130-811

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-009

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-011

(2) Do an Ultrasonic inspection of the upper thrust beam chord for damage around the four fasteners common to the primary beam chord.

See Doc. D626A001 - DTR, DTR check form 55-10-14-8, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-10.

SUBTASK 55-05-02-410-009

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION THRUST BEAM			55-645-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 13000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (Ultrasonic) the upper thrust beam chord for damage around the four fasteners common to the primary beam chord.

See Doc. D626A001 - DTR, DTR check form 55-10-14-8, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-10.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-645-00-02

MECH	INSP

TASK 55-05-02-130-811

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION THRUST BEAMS

A. Inspection

SUBTASK 55-05-02-010-009

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-130-011

(2) Do an Ultrasonic inspection of the upper thrust beam chord for damage around the four fasteners common to the primary beam chord.

See Doc. D626A001 - DTR, DTR check form 55-10-14-8, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 4, Subject 55-10-10.

SUBTASK 55-05-02-410-009

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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BOEING

737-600/700/800/900
TASK CARDS

AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT HORIZONTAL STABILIZER CENTER SECTION PRIMARY BEAM CHORD TO WEB ATTACHMENT			55-646-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 8000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 313
		NOTE			

Inspect (High Frequency Eddy Current) the upper chord web flange around the fasteners common to the web between the front and rear spars from STA 1216 to STA 1242.

See Doc. D626A001 - DTR, DTR check form 55-10-14-9, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-25.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-646-00-01

MECH	INSP

TASK 55-05-02-250-833

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PRIMARY BEAM CHORD TO WEB ATTACHMENT

A. Inspection

SUBTASK 55-05-02-010-047

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-033

(2) Do a High Frequency Eddy Current inspection of the upper chord web flange around the fasteners common to the web between the front and rear spars from STA 1216 to STA 1242.

See Doc. D626A001 - DTR, DTR check form 55-10-14-9, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-25.

SUBTASK 55-05-02-410-045

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT HORIZONTAL STABILIZER CENTER SECTION PRIMARY BEAM CHORD TO WEB ATTACHMENT			55-646-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 8000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 311BL			ZONE 314
		NOTE			

Inspect (High Frequency Eddy Current) the upper chord web flange around the fasteners common to the web between the front and rear spars from STA 1216 to STA 1242.

See Doc. D626A001 - DTR, DTR check form 55-10-14-9, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-25.

ACCESS NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-646-00-02

MECH	INSP

TASK 55-05-02-250-833

1. INTERNAL - SPECIAL DETAILED: HORIZONTAL STABILIZER CENTER SECTION PRIMARY BEAM CHORD TO WEB ATTACHMENT

A. Inspection

SUBTASK 55-05-02-010-047

(1) Open this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

NOTE: Access horizontal stabilizer center section through opening in center of 1088 bulkhead.

SUBTASK 55-05-02-250-033

(2) Do a High Frequency Eddy Current inspection of the upper chord web flange around the fasteners common to the web between the front and rear spars from STA 1216 to STA 1242.

See Doc. D626A001 - DTR, DTR check form 55-10-14-9, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-25.

SUBTASK 55-05-02-410-045

(3) Close this access panel:

Number	Name/Location
311BL	Stabilizer Trim Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT ELEVATOR HINGE FITTINGS			55-647-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 334GB 334GT 334HB 334JB 334JT 334KB 334LB 334MB 334MT 334NB 334NT			ZONE 334

Inspect (High Frequency Eddy Current) all hinge fitting clevis lugs on the lug face around the circumference of the bushing.

See Doc. D626A001 - DTR, DTR check form 55-20-05, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-07.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-647-00-01

MECH	INSP

TASK 55-05-02-250-834

1. INTERNAL - SPECIAL DETAILED: ELEVATOR HINGE FITTINGS

A. Inspection

SUBTASK 55-05-02-010-067

(1) Open these access panels on the Left side:

Number	Name/Location
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover
334HB	Horizontal Stabilizer, Elevator Hinge Cover
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334JT	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334MT	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02
344HB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 66.54
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344JT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344MT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

SUBTASK 55-05-02-250-034

(2) Do a High Frequency Eddy Current inspection of all hinge fitting clevis lugs on lug face around the circumference of the bushing.

See Doc. D626A001 - DTR, DTR check form 55-20-05, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-07.

SUBTASK 55-05-02-410-065

(3) Close these access panels on the Left side:

Number	Name/Location
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-647-00-01

MECH	INSP

(Continued)

Number	Name/Location
334HB	Horizontal Stabilizer, Elevator Hinge Cover
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334JT	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334MT	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02
344HB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 66.54
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344JT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344MT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT ELEVATOR HINGE FITTINGS			55-647-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 344GB 344GT 344HB 344JB 344JT 344KB 344LB 344MB 344MT 344NB 344NT			ZONE 344

Inspect (High Frequency Eddy Current) all hinge fitting clevis lugs on the lug face around the circumference of the bushing.

See Doc. D626A001 - DTR, DTR check form 55-20-05, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-07.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-647-00-02

MECH	INSP

TASK 55-05-02-250-834

1. INTERNAL - SPECIAL DETAILED: ELEVATOR HINGE FITTINGS

A. Inspection

SUBTASK 55-05-02-010-067

(1) Open these access panels on the Left side:

Number	Name/Location
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover
334HB	Horizontal Stabilizer, Elevator Hinge Cover
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334JT	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334MT	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02
344HB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 66.54
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344JT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344MT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

SUBTASK 55-05-02-250-034

(2) Do a High Frequency Eddy Current inspection of all hinge fitting clevis lugs on lug face around the circumference of the bushing.

See Doc. D626A001 - DTR, DTR check form 55-20-05, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-07.

SUBTASK 55-05-02-410-065

(3) Close these access panels on the Left side:

Number	Name/Location
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-647-00-02

MECH	INSP

(Continued)

Number	Name/Location
334HB	Horizontal Stabilizer, Elevator Hinge Cover
334JB	Horizontal Stabilizer, Elevator Hinge Cover
334JT	Horizontal Stabilizer, Elevator Hinge Cover
334KB	Horizontal Stabilizer, Elevator Hinge Cover
334LB	Horizontal Stabilizer, Elevator Hinge Cover
334MB	Horizontal Stabilizer, Elevator Hinge Cover
334MT	Horizontal Stabilizer, Elevator Hinge Cover
334NB	Horizontal Stabilizer, Elevator Hinge Cover
334NT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02
344HB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 66.54
344JB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344JT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 121.59
344KB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 176.64
344LB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 213.32
344MB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344MT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 250.04
344NB	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45
344NT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 265.45

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT ELEVATOR MAST ARM ASSEMBLY (ELEVATOR ACTUATOR FITTING)			55-648-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 333AB 334AB 334GB 334GT			ZONE 334
		NOTE			

Inspect (High Frequency Eddy Current) each lug face of the elevator actuator fittings.

See Doc. D626A001 - DTR, DTR check form 55-20-06, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-09.

ACCESS NOTE: Remove the actuator rod assembly as required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-648-00-01

MECH	INSP

TASK 55-05-02-250-835

1. INTERNAL - SPECIAL DETAILED: ELEVATOR MAST ARM ASSEMBLY (ELEVATOR ACTUATOR FITTING)

A. Inspection

SUBTASK 55-05-02-010-048

(1) Open these access panels on the Left side::

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

NOTE: Remove the actuator rod assembly as required.

SUBTASK 55-05-02-250-035

(2) Do a High Frequency Eddy Current inspection of each lug face of the elevator actuator fittings.

See Doc. D626A001 - DTR, DTR check form 55-20-06, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-09.

SUBTASK 55-05-02-410-158

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 55-05-02-410-046

(5) Close these access panels on the Left side:

Number	Name/Location
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-648-00-01

MECH	INSP

(Continued)

Number	Name/Location
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT ELEVATOR MAST ARM ASSEMBLY (ELEVATOR ACTUATOR FITTING)			55-648-00-02
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	R ELEVATOR	1.1	56000 FC	24000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		343AB 344AB 344GB 344GT			344
		NOTE			

Inspect (High Frequency Eddy Current) each lug face of the elevator actuator fittings.

See Doc. D626A001 - DTR, DTR check form 55-20-06, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-09.

ACCESS NOTE: Remove the actuator rod assembly as required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-648-00-02

MECH	INSP

TASK 55-05-02-250-835

1. INTERNAL - SPECIAL DETAILED: ELEVATOR MAST ARM ASSEMBLY (ELEVATOR ACTUATOR FITTING)

A. Inspection

SUBTASK 55-05-02-010-048

(1) Open these access panels on the Left side::

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

NOTE: Remove the actuator rod assembly as required.

SUBTASK 55-05-02-250-035

(2) Do a High Frequency Eddy Current inspection of each lug face of the elevator actuator fittings.

See Doc. D626A001 - DTR, DTR check form 55-20-06, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-09.

SUBTASK 55-05-02-410-158

(3) Close this access panel on the Left side:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

(4) Close this access panel on the Right side:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 55-05-02-410-046

(5) Close these access panels on the Left side:

Number	Name/Location
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-648-00-02

MECH	INSP

(Continued)

Number	Name/Location
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT ELEVATOR HORN BALANCE WEIGHT SUPPORT STRUCTURE			55-649-00-01
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA L ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 334ET 334FB			ZONE 334
		NOTE			

Inspect (General Visual) the elevator horn balance weight support structure at elevator STA 260.

See Doc. D626A001 - DTR, DTR check form 55-20-07, for alternative inspection.

ACCESS NOTE: Removal of the elevator horn balance weight fairing (183A7400) is required.
Outboard rib is inspected through hole in inboard rib.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-649-00-01

MECH	INSP

TASK 55-05-02-210-801

1. INTERNAL - GENERAL VISUAL: ELEVATOR HORN BALANCE WEIGHT SUPPORT STRUCTURE

A. Inspection

SUBTASK 55-05-02-010-011

(1) Open these access panels on the Left side:

Number	Name/Location
334ET	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

Open these access panels on the Right side:

Number	Name/Location
344ET	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

NOTE: Removal of the elevator horn balance weight fairing (183A7400) is required. Outboard rib is inspected through hole in inboard rib.

SUBTASK 55-05-02-210-001

(2) Do a General Visual inspection of the elevator horn balance weight support structure at elevator STA 260.

See Doc. D626A001 - DTR, DTR check form 55-20-07, for alternative inspection.

SUBTASK 55-05-02-410-011

(3) Close these access panels on the Left side:

Number	Name/Location
334ET	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

Close these access panels on the Right side:

Number	Name/Location
344ET	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT ELEVATOR HORN BALANCE WEIGHT SUPPORT STRUCTURE			55-649-00-02
DATE	TASK GENERAL VISUAL				RELATED CARD
TAIL NUMBER	WORK AREA R ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 344ET 344FB			ZONE 344
		NOTE			

Inspect (General Visual) the elevator horn balance weight support structure at elevator STA 260.

See Doc. D626A001 - DTR, DTR check form 55-20-07, for alternative inspection.

ACCESS NOTE: Removal of the elevator horn balance weight fairing (183A7400) is required.
Outboard rib is inspected through hole in inboard rib.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-649-00-02

MECH	INSP

TASK 55-05-02-210-801

1. INTERNAL - GENERAL VISUAL: ELEVATOR HORN BALANCE WEIGHT SUPPORT STRUCTURE

A. Inspection

SUBTASK 55-05-02-010-011

(1) Open these access panels on the Left side:

Number	Name/Location
334ET	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

Open these access panels on the Right side:

Number	Name/Location
344ET	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

NOTE: Removal of the elevator horn balance weight fairing (183A7400) is required. Outboard rib is inspected through hole in inboard rib.

SUBTASK 55-05-02-210-001

(2) Do a General Visual inspection of the elevator horn balance weight support structure at elevator STA 260.

See Doc. D626A001 - DTR, DTR check form 55-20-07, for alternative inspection.

SUBTASK 55-05-02-410-011

(3) Close these access panels on the Left side:

Number	Name/Location
334ET	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

Close these access panels on the Right side:

Number	Name/Location
344ET	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344FB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT 183A8700 HINGE/MAST ARM FITTING UPPER FLANGE			55-652-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	L ELEVATOR	1.1	56000 FC	18000 FC	AIRPLANE: ALL NOTE; ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		334PT 334RB 334RT			334

Inspect (High Frequency Eddy Current) around the circumference of the washer at bolt hole 2 and 3 on the mast arm fitting upper flange.

See Doc. D626A001 - DTR, DTR check form 55-20-08-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-15.

AIRPLANE NOTE: Applicable to L/N 596 and ALL 1175 and on and L/N 1 to 1174 that have incorporated SB 737-55A1080 or SB 737-55-1081.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-652-00-01

MECH	INSP

TASK 55-05-02-250-838

1. INTERNAL - SPECIAL DETAILED: 183A8700 HINGE/MAST ARM FITTING UPPER FLANGE

A. Inspection

SUBTASK 55-05-02-010-054

(1) Open this access panel on the Left side:

Number	Name/Location
334PT	Horizontal Stabilizer, Tab Control Rod Fairing

Open this access panel on the Right side:

Number	Name/Location
344PT	Horizontal Stabilizer, Tab Control Rod Fairing, Elevator Sta 34.0

SUBTASK 55-05-02-250-038

(2) Do a High Frequency Eddy Current inspection around the circumference of the washer at bolt hole 2 and 3 on the mast arm fitting upper flange.

See Doc. D626A001 - DTR, DTR check form 55-20-08-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-15.

SUBTASK 55-05-02-410-052

(3) Close this access panel on the Left side:

Number	Name/Location
334PT	Horizontal Stabilizer, Tab Control Rod Fairing

Close this access panel on the Right side:

Number	Name/Location
344PT	Horizontal Stabilizer, Tab Control Rod Fairing, Elevator Sta 34.0

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT 183A8700 HINGE/MAST ARM FITTING UPPER FLANGE			55-652-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 18000 FC	APPLICABILITY AIRPLANE: ALL NOTE ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 344PT 344RB 344RT			ZONE 344

Inspect (High Frequency Eddy Current) around the circumference of the washer at bolt hole 2 and 3 on the mast arm fitting upper flange.

See Doc. D626A001 - DTR, DTR check form 55-20-08-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-15.

AIRPLANE NOTE: Applicable to L/N 596 and ALL 1175 and on and L/N 1 to 1174 that have incorporated SB 737-55A1080 or SB 737-55-1081.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-652-00-02

MECH	INSP

TASK 55-05-02-250-838

1. INTERNAL - SPECIAL DETAILED: 183A8700 HINGE/MAST ARM FITTING UPPER FLANGE

A. Inspection

SUBTASK 55-05-02-010-054

(1) Open this access panel on the Left side:

Number	Name/Location
334PT	Horizontal Stabilizer, Tab Control Rod Fairing

Open this access panel on the Right side:

Number	Name/Location
344PT	Horizontal Stabilizer, Tab Control Rod Fairing, Elevator Sta 34.0

SUBTASK 55-05-02-250-038

(2) Do a High Frequency Eddy Current inspection around the circumference of the washer at bolt hole 2 and 3 on the mast arm fitting upper flange.

See Doc. D626A001 - DTR, DTR check form 55-20-08-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-10-15.

SUBTASK 55-05-02-410-052

(3) Close this access panel on the Left side:

Number	Name/Location
334PT	Horizontal Stabilizer, Tab Control Rod Fairing

Close this access panel on the Right side:

Number	Name/Location
344PT	Horizontal Stabilizer, Tab Control Rod Fairing, Elevator Sta 34.0

——— **END OF TASK** ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT 183A8300 BRIDGE FITTING UPPER/LOWER FLANGE			55-655-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL NOTE ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 334

Inspect (X-ray) the upper and lower flanges of hinges 3, 4, 5, and 6.

See Doc. D626A001 - DTR, DTR check form 55-20-09-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 2, Subject 55-10-01.

AIRPLANE NOTE: Applicable to L/N 596 and ALL 1175 and On, and L/N 1 - 1174 that have incorporated SB 737-55A1080 or SB 737-55-1081.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-655-00-01

MECH	INSP

TASK 55-05-02-260-802

1. EXTERNAL - SPECIAL DETAILED: 183A8300 BRIDGE FITTING UPPER/LOWER FLANGE

A. Inspection

SUBTASK 55-05-02-260-002

(1) Do an X-ray inspection of the upper and lower flanges of hinges 3, 4, 5, and 6.

See Doc. D626A001 - DTR, DTR check form 55-20-09-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 2, Subject 55-10-01.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT 183A8300 BRIDGE FITTING UPPER/LOWER FLANGE			55-655-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL NOTE ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 344

Inspect (X-ray) the upper and lower flanges of hinges 3, 4, 5, and 6.

See Doc. D626A001 - DTR, DTR check form 55-20-09-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 2, Subject 55-10-01.

AIRPLANE NOTE: Applicable to L/N 596 and ALL 1175 and On, and L/N 1 - 1174 that have incorporated SB 737-55A1080 or SB 737-55-1081.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-655-00-02

MECH	INSP

TASK 55-05-02-260-802

1. EXTERNAL - SPECIAL DETAILED: 183A8300 BRIDGE FITTING UPPER/LOWER FLANGE

A. Inspection

SUBTASK 55-05-02-260-002

(1) Do an X-ray inspection of the upper and lower flanges of hinges 3, 4, 5, and 6.

See Doc. D626A001 - DTR, DTR check form 55-20-09-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 2, Subject 55-10-01.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT 183A1400 - BRACKET ASSY - TAB CONTROL MECHANISM, ELEVATOR			55-657-00-01
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA L ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL NOTE
STATION	SKILL AIRPL				
		ACCESS 334AB 334AT 334GB 334GT NOTE			ZONE 334

Inspect (High Frequency Eddy Current) the surface of the fitting and doublers of each bracket assembly around the lug bore.

See Doc. D626A001 - DTR, DTR check form 55-20-12-1, for alternative inspection.

AIRPLANE NOTE: Applicable to L/N 596 and ALL 1175 and on, and L/N 1 - 1174 that have incorporated SB 737-55A1080 or SB 737-55-1081.

ACCESS NOTE: Remove upper or lower horizontal stabilizer trailing edge seal and elevator inboard hinge cover panel. Bushing removal is not required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-657-00-01

MECH	INSP

TASK 55-05-02-250-841

1. INTERNAL - SPECIAL DETAILED: 183A1400 BRACKET ASSEMBLY - TAB CONTROL MECHANISM, ELEVATOR

A. Inspection

SUBTASK 55-05-02-010-060

(1) Open these access panels on the Left Side:

Number	Name/Location
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

NOTE: Remove upper or lower horizontal stabilizer trailing edge seal and elevator inboard hinge cover panel. Bushing removal is not required.

SUBTASK 55-05-02-250-041

(2) Do a High Frequency Eddy Current inspection of the surface of the fitting and doublers of each bracket assembly around the lug bore.

See Doc. D626A001 - DTR, DTR check form 55-20-12-1, for alternative inspection.

SUBTASK 55-05-02-410-058

(3) Close these access panels on the Left Side:

Number	Name/Location
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT 183A1400 - BRACKET ASSY - TAB CONTROL MECHANISM, ELEVATOR			55-657-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA R ELEVATOR	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL NOTE
STATION	SKILL AIRPL				
		ACCESS 344AB 344AT 344GB 344GT NOTE			ZONE 344

Inspect (High Frequency Eddy Current) the surface of the fitting and doublers of each bracket assembly around the lug bore.

See Doc. D626A001 - DTR, DTR check form 55-20-12-1, for alternative inspection.

AIRPLANE NOTE: Applicable to L/N 596 and ALL 1175 and on, and L/N 1 - 1174 that have incorporated SB 737-55A1080 or SB 737-55-1081.

ACCESS NOTE: Remove upper or lower horizontal stabilizer trailing edge seal and elevator inboard hinge cover panel. Bushing removal is not required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-657-00-02

MECH	INSP

TASK 55-05-02-250-841

1. INTERNAL - SPECIAL DETAILED: 183A1400 BRACKET ASSEMBLY - TAB CONTROL MECHANISM, ELEVATOR

A. Inspection

SUBTASK 55-05-02-010-060

(1) Open these access panels on the Left Side:

Number	Name/Location
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Open these access panels on the Right side:

Number	Name/Location
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

NOTE: Remove upper or lower horizontal stabilizer trailing edge seal and elevator inboard hinge cover panel. Bushing removal is not required.

SUBTASK 55-05-02-250-041

(2) Do a High Frequency Eddy Current inspection of the surface of the fitting and doublers of each bracket assembly around the lug bore.

See Doc. D626A001 - DTR, DTR check form 55-20-12-1, for alternative inspection.

SUBTASK 55-05-02-410-058

(3) Close these access panels on the Left Side:

Number	Name/Location
334AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
334GB	Horizontal Stabilizer, Elevator Hinge Cover
334GT	Horizontal Stabilizer, Elevator Hinge Cover

Close these access panels on the Right side:

Number	Name/Location
344AB	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344AT	Horizontal Stabilizer, Seal, Trailing Edge to Elevator
344GB	Horizontal Stabilizer, Hinge Cover, Elevator Station 24.09
344GT	Horizontal Stabilizer, Elevator Hinge Cover, Elevator Sta 39.02

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT VERTICAL FIN - REAR SPAR			55-658-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	VERT STABLIZER	1.1	56000 FC	24000 FC	
STATION	SKILL				AIRPLANE / ENGINE
	AIRPL				600 700 800 900 900ER / ALL
		ACCESS			ZONE
					323

Inspect (High Frequency Eddy Current) the exposed chord between the inspar and trailing edge skin from vertical fin STA 73.4 to vertical fin tip.

See Doc. D626A001 - DTR, DTR check form 55-30-03-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-30-05.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-658-00-01

MECH	INSP

TASK 55-05-02-250-842

1. EXTERNAL - SPECIAL DETAILED: VERTICAL FIN - REAR SPAR

A. Inspection

SUBTASK 55-05-02-250-042

(1) Do a High Frequency Eddy Current inspection of the exposed chord between the inspar and trailing edge skin from Vertical Fin STA 73.4 to Vertical Fin tip.

See Doc. D626A001 - DTR, DTR check form 55-30-03-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-30-05.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT VERTICAL FIN - REAR SPAR			55-658-00-02
DATE	TASK SPECIAL DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA VERT STABLIZER	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE: 600 700 800 900 900ER ENGINE: ALL
		ACCESS			ZONE 324

Inspect (High Frequency Eddy Current) the exposed chord between the inspar and trailing edge skin from vertical fin STA 73.4 to vertical fin tip.

See Doc. D626A001 - DTR, DTR check form 55-30-03-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-30-05.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-658-00-02

MECH	INSP

TASK 55-05-02-250-842

1. EXTERNAL - SPECIAL DETAILED: VERTICAL FIN - REAR SPAR

A. Inspection

SUBTASK 55-05-02-250-042

(1) Do a High Frequency Eddy Current inspection of the exposed chord between the inspar and trailing edge skin from Vertical Fin STA 73.4 to Vertical Fin tip.

See Doc. D626A001 - DTR, DTR check form 55-30-03-2, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Subject 55-30-05.

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		LEFT VERTICAL FIN - CLOSURE RIB			55-660-00-01
DATE	TASK				RELATED CARD
	DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	VERT STABLIZER	1.1	56000 FC	36000 FC	
STATION	SKILL				AIRPLANE / ENGINE
	AIRPL				ALL / ALL
		ACCESS			ZONE
		323BL 323CL			323 324

Inspect (Detailed) the chords lower flange and land-up at the forward end fastener location common to rear spar terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-30-05-1, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-660-00-01

MECH	INSP

TASK 55-05-02-211-808

1. INTERNAL - DETAILED: VERTICAL FIN - CLOSURE RIB

A. Inspection

SUBTASK 55-05-02-010-013

(1) Open these access panels on the Left side:

Number	Name/Location
323BL	Vertical Fin, Forward Fin Access Door
323CL	Vertical Fin, Rear Spar Access Door

Open these access panels on the Right side:

Number	Name/Location
323BR	Vertical Fin, Forward Fin Access Door
323CR	Vertical Fin, Rear Spar Access Door

SUBTASK 55-05-02-211-008

(2) Do a Detailed inspection of the chords lower flange and land-up at the forward end fastener location common to rear spar terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-30-05–1, for alternative inspection.

SUBTASK 55-05-02-410-013

(3) Close these access panels on the Left side:

Number	Name/Location
323BL	Vertical Fin, Forward Fin Access Door
323CL	Vertical Fin, Rear Spar Access Door

Close these access panels on the Right side:

Number	Name/Location
323BR	Vertical Fin, Forward Fin Access Door
323CR	Vertical Fin, Rear Spar Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RIGHT VERTICAL FIN - CLOSURE RIB			55-660-00-02
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA VERT STABLIZER	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL / ENGINE ALL
		ACCESS 323BR 323CR			ZONE 323 324

Inspect (Detailed) the chords lower flange and land-up at the forward end fastener location common to rear spar terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-30-05-1, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-660-00-02

MECH	INSP

TASK 55-05-02-211-808

1. INTERNAL - DETAILED: VERTICAL FIN - CLOSURE RIB

A. Inspection

SUBTASK 55-05-02-010-013

(1) Open these access panels on the Left side:

Number	Name/Location
323BL	Vertical Fin, Forward Fin Access Door
323CL	Vertical Fin, Rear Spar Access Door

Open these access panels on the Right side:

Number	Name/Location
323BR	Vertical Fin, Forward Fin Access Door
323CR	Vertical Fin, Rear Spar Access Door

SUBTASK 55-05-02-211-008

(2) Do a Detailed inspection of the chords lower flange and land-up at the forward end fastener location common to rear spar terminal fitting.

See Doc. D626A001 - DTR, DTR check form 55-30-05–1, for alternative inspection.

SUBTASK 55-05-02-410-013

(3) Close these access panels on the Left side:

Number	Name/Location
323BL	Vertical Fin, Forward Fin Access Door
323CL	Vertical Fin, Rear Spar Access Door

Close these access panels on the Right side:

Number	Name/Location
323BR	Vertical Fin, Forward Fin Access Door
323CR	Vertical Fin, Rear Spar Access Door

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RUDDER HINGE RIBS			55-662-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA VERT STABLIZER	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 24000 FC	APPLICABILITY
STATION	SKILL AIRPL				AIRPLANE ALL ENGINE ALL
		ACCESS 324ACL 324ADL 324AEL 324AFL 324AHL 324AJL 324AKL 324AL 324ALL 324BL 324DL 324JL 324KL 324LL 324ML 324NL 324PL			ZONE 324

Inspect (Detailed) the hinge fitting lugs at the rib attachment for hinges #1, #2, #3, #4, #5, #6, #7, #7A and #8.

See Doc. D626A001 - DTR, DTR check form 55-30-06-1, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-662-00-01

MECH	INSP

TASK 55-05-02-211-809

1. INTERNAL - DETAILED: RUDDER HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-014

(1) Open these access panels:

Number	Name/Location
324ACL	Rudder Nose Cover/Rudder Hinge No. 8 At Rudder Station 276.24
324ADL	Rudder Nose Cover/Rudder Hinge No. 7 At Rudder Station 257.92
324AEL	Rudder Nose Cover/Rudder Hinge No. 6 At Rudder Station 239.61
324AFL	Rudder Nose Cover/Rudder Hinge No. 5 At Rudder Station 184.67
324AHL	Rudder Nose Fairing/Rudder Hinge No. 4 At Rudder Station 129.74.
324AJL	Rudder Nose Fairing/Rudder Hinge No. 3 At Rudder Station 70.65
324AKL	Rudder Nose Fairing/Rudder Hinge No. 2 At Rudder Station 53.48
324AL	Vertical Fin, Aft Fin Access Door
324ALL	Rudder Nose Cover/Rudder Hinge No. 1 At Rudder Station 6.96
324BL	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324JL	Vertical Fin, Access
324KL	Vertical Fin, Access
324LL	Vertical Fin, Access
324ML	Vertical Fin, Access
324NL	Vertical Fin, Access
324PL	Vertical Fin, Access

SUBTASK 55-05-02-211-009

(2) Do a Detailed inspection of the hinge fitting lugs at rib attachment hinge #1, #2, #3, #4, #5, #6, #7, #7A and #8.

See Doc. D626A001 - DTR, DTR check form 55–30–06–1, for alternative inspection.

SUBTASK 55-05-02-410-014

(3) Close these access panels:

Number	Name/Location
324ACL	Rudder Nose Cover/Rudder Hinge No. 8 At Rudder Station 276.24
324ADL	Rudder Nose Cover/Rudder Hinge No. 7 At Rudder Station 257.92
324AEL	Rudder Nose Cover/Rudder Hinge No. 6 At Rudder Station 239.61
324AFL	Rudder Nose Cover/Rudder Hinge No. 5 At Rudder Station 184.67
324AHL	Rudder Nose Fairing/Rudder Hinge No. 4 At Rudder Station 129.74.
324AJL	Rudder Nose Fairing/Rudder Hinge No. 3 At Rudder Station 70.65
324AKL	Rudder Nose Fairing/Rudder Hinge No. 2 At Rudder Station 53.48
324AL	Vertical Fin, Aft Fin Access Door
324ALL	Rudder Nose Cover/Rudder Hinge No. 1 At Rudder Station 6.96
324BL	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324JL	Vertical Fin, Access
324KL	Vertical Fin, Access

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-662-00-01

MECH	INSP

(Continued)

Number	Name/Location
324LL	Vertical Fin, Access
324ML	Vertical Fin, Access
324NL	Vertical Fin, Access
324PL	Vertical Fin, Access

END OF TASK

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RUDDER HINGE RIBS			55-663-00-01
DATE	TASK				RELATED CARD
	DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	VERT STABLIZER	1.1	56000 FC	36000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		323CL 324AL 324BL 324DL 324JL 324KL 324LL 324ML 324NL 324PL			324
		NOTE			

Inspect (Detailed) rudder hinge ribs 1 through 7A at attachment to both rear spar chords.

See Doc. D626A001 - DTR, DTR check form 55-30-06-2, for alternative inspection.

ACCESS NOTE: Removal of the skin panels aft of the fin spar is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-663-00-01

MECH	INSP

TASK 55-05-02-211-810

1. INTERNAL - DETAILED: RUDDER HINGE RIBS

A. Inspection

SUBTASK 55-05-02-010-015

(1) Open these access panels:

Number	Name/Location
323CL	Vertical Fin, Rear Spar Access Door
324AL	Vertical Fin, Aft Fin Access Door
324BL	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324JL	Vertical Fin, Access
324KL	Vertical Fin, Access
324LL	Vertical Fin, Access
324ML	Vertical Fin, Access
324NL	Vertical Fin, Access
324PL	Vertical Fin, Access

NOTE: Removal of the skin panels aft of the fin spar is required.

SUBTASK 55-05-02-211-010

(2) Do a Detailed inspection of the Rudder Hinge Ribs 1 through 7A at attachment to both Rear Spar Chords.

See Doc. D626A001 - DTR, DTR check form 55–30–06-2, for alternative inspection.

SUBTASK 55-05-02-410-015

(3) Close these access panels:

Number	Name/Location
323CL	Vertical Fin, Rear Spar Access Door
324AL	Vertical Fin, Aft Fin Access Door
324BL	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324JL	Vertical Fin, Access
324KL	Vertical Fin, Access
324LL	Vertical Fin, Access
324ML	Vertical Fin, Access
324NL	Vertical Fin, Access
324PL	Vertical Fin, Access

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RUDDER HINGE FITTINGS			55-664-00-01
DATE	TASK DETAILED				RELATED CARD
TAIL NUMBER	WORK AREA RUDDER	VERSION 1.1	THRESHOLD 56000 FC	REPEAT 9000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 324ACL 324ACR 324ADL 324AEL 324AFL 324AGL 324AHL 324AJL 324AKL 324ALL			ZONE 325

Inspect (Detailed) all the rudder hinge fitting clevis lugs and the hinge fitting attachments to the skin and the spar.

See Doc. D626A001 - DTR, DTR check form 55-40-05, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-664-00-01

MECH	INSP

TASK 55-05-02-211-811

1. INTERNAL - DETAILED: RUDDER HINGE FITTINGS

A. Inspection

SUBTASK 55-05-02-010-016

(1) Open these access panels:

Number	Name/Location
324ACL	Rudder Nose Cover/Rudder Hinge No. 8 At Rudder Station 276.24
324ACR	Upper Rudder Gap Cover At Rudder Station 164.90
324ADL	Rudder Nose Cover/Rudder Hinge No. 7 At Rudder Station 257.92
324AEL	Rudder Nose Cover/Rudder Hinge No. 6 At Rudder Station 239.61
324AFL	Rudder Nose Cover/Rudder Hinge No. 5 At Rudder Station 184.67
324AGL	Upper Rudder Gap Cover At Rudder Station 164.90
324AHL	Rudder Nose Fairing/Rudder Hinge No. 4 At Rudder Station 129.74.
324AJL	Rudder Nose Fairing/Rudder Hinge No. 3 At Rudder Station 70.65
324AKL	Rudder Nose Fairing/Rudder Hinge No. 2 At Rudder Station 53.48
324ALL	Rudder Nose Cover/Rudder Hinge No. 1 At Rudder Station 6.96

SUBTASK 55-05-02-211-011

(2) Do an a Detailed inspection of all the rudder hinge fitting clevis lugs and the hinge fitting attachments to the skin and the spar.

See Doc. D626A001 - DTR, DTR check from 55–40–05, for alternative inspection.

SUBTASK 55-05-02-410-016

(3) Close these access panels:

Number	Name/Location
324ACL	Rudder Nose Cover/Rudder Hinge No. 8 At Rudder Station 276.24
324ACR	Upper Rudder Gap Cover At Rudder Station 164.90
324ADL	Rudder Nose Cover/Rudder Hinge No. 7 At Rudder Station 257.92
324AEL	Rudder Nose Cover/Rudder Hinge No. 6 At Rudder Station 239.61
324AFL	Rudder Nose Cover/Rudder Hinge No. 5 At Rudder Station 184.67
324AGL	Upper Rudder Gap Cover At Rudder Station 164.90
324AHL	Rudder Nose Fairing/Rudder Hinge No. 4 At Rudder Station 129.74.
324AJL	Rudder Nose Fairing/Rudder Hinge No. 3 At Rudder Station 70.65
324AKL	Rudder Nose Fairing/Rudder Hinge No. 2 At Rudder Station 53.48
324ALL	Rudder Nose Cover/Rudder Hinge No. 1 At Rudder Station 6.96

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RUDDER ACTUATOR FITTINGS			55-665-00-01
DATE	TASK				RELATED CARD
	SPECIAL DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	VERT STABLIZER	1.1	56000 FC	18000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		324AJL 324AKL			325
		NOTE			

Inspect (High Frequency Eddy Current) the rudder actuator fitting clevis lugs at rudder STA 60.85 and rudder STA 70.65, as well as visually inspect the hinge fitting attachments to the skin and the spar.

See Doc. D626A001 - DTR, DTR check form 55-40-06, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-08.

ACCESS NOTE: Removal of the actuator rod for access to the inner lug faces is required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-665-00-01

MECH	INSP

TASK 55-05-02-250-844

1. INTERNAL - SPECIAL DETAILED: RUDDER ACTUATOR FITTINGS

A. Inspection

SUBTASK 55-05-02-010-062

(1) Open these access panels:

Number	Name/Location
324AJL	Rudder Nose Fairing/Rudder Hinge No. 3 At Rudder Station 70.65
324AKL	Rudder Nose Fairing/Rudder Hinge No. 2 At Rudder Station 53.48

NOTE: Removal of the actuator rod for access to the inner lug faces is required.

SUBTASK 55-05-02-250-044

(2) Do a High Frequency Eddy Current inspection of the rudder actuator fitting clevis lugs at rudder STA 60.85 and rudder STA 70.65, as well as visually inspect the hinge fitting attachments to the skin and the spar.

See Doc. D626A001 - DTR, DTR check form 55–40–06, for alternative inspection.

The NDI method(s) necessary to accomplish the intent of this inspection is contained in the 737 Nondestructive Test Manual (D6-37239). The inspection procedures are contained in Part 6, Section 55-50-08.

SUBTASK 55-05-02-410-060

(3) Close these access panels:

Number	Name/Location
324AJL	Rudder Nose Fairing/Rudder Hinge No. 3 At Rudder Station 70.65
324AKL	Rudder Nose Fairing/Rudder Hinge No. 2 At Rudder Station 53.48

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		737NG RUDDER BALANCE ARM			55-666-00-01
DATE	TASK				RELATED CARD
	DETAILED				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	RUDDER	1.1	56000 FC	18000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL				
	AIRPL				
		ACCESS			ZONE
		324ACR 324ADR 324AGL 324AXL 324WL			325

Inspect (Detailed) the casting in the area around the doubler as well as the balance arm assembly attachment to the rudder front spar.

See Doc. D626A001 - DTR, DTR check form 55-40-07, for alternative inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-666-00-01

MECH	INSP

TASK 55-05-02-211-812

1. INTERNAL - DETAILED: RUDDER BALANCE ARM

A. Inspection

SUBTASK 55-05-02-010-148

(1) Open these access panels:

Number	Name/Location
324ACR	Upper Rudder Gap Cover At Rudder Station 164.90
324ADR	Lower Rudder Gap Cover
324AGL	Upper Rudder Gap Cover At Rudder Station 164.90
324AXL	Lower Rudder Gap Cover
324WL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin

SUBTASK 55-05-02-211-012

(2) Do a Detailed inspection of the casting in the area around the doubler as well as the balance arm assembly attachment to the rudder front spar.

See Doc. D626A001 - DTR, DTR check form 55–40–07, for alternative inspection.

SUBTASK 55-05-02-410-146

(3) Close these access panels:

Number	Name/Location
324ACR	Upper Rudder Gap Cover At Rudder Station 164.90
324ADR	Lower Rudder Gap Cover
324AGL	Upper Rudder Gap Cover At Rudder Station 164.90
324AXL	Lower Rudder Gap Cover
324WL	Panel Assy - Trailing Edge, Beam Seal, Vert Fin

——— END OF TASK ———

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN AND HORIZONTAL STABILIZER			55-800-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 120 DY	REPEAT 120 DY	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 320 330 340

Perform an external zonal inspection (GV) of the vertical fin and horizontal stabilizer. Inspection is accomplished from the ground, without the use of stands or ladders. No additional access panel required.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-800-00-01

MECH	INSP

TASK 05-41-03-210-806

1. EXTERNAL - ZONAL (GV): VERTICAL FIN AND HORIZONTAL STABILIZER

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-006

(1) Do a General Visual inspection of the vertical fin and horizontal stabilizer. Inspection is accomplished from the ground, without the use of stands or ladders. No additional access panel required.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-800-00-01

Vertical Fin and Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN - DORSAL FIN			55-802-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLICABILITY
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	AIRPLANE: ALL ENGINE: ALL
		ACCESS			ZONE 321

Perform an external zonal inspection (GV) of the vertical fin - dorsal fin.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-802-00-01

MECH	INSP

TASK 05-41-03-210-807

1. EXTERNAL - ZONAL (GV): VERTICAL FIN - DORSAL FIN

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-007

(1) Do a General Visual inspection of the vertical fin - dorsal fin.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-802-00-01

K93952 S0006584173_V2

Dorsal Fin - Vertical Fin General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN - LEADING EDGE (ONLY IF HF SYSTEM INSTALLED)			55-804-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY AIRPLANE: ALL ENGINE: ALL NOTE
STATION	SKILL AIRPL				
		ACCESS 322AL			ZONE 322

Perform an internal zonal inspection (GV) of the vertical fin - leading edge (only if HF system installed).

INTERVAL NOTE: Whichever comes first.

AIRPLANE NOTE: Only if HF system installed.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-804-00-01

MECH	INSP

TASK 05-41-03-210-808

1. INTERNAL - ZONAL (GV): VERTICAL FIN - LEADING EDGE (ONLY IF HF SYSTEM INSTALLED)
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-010-005

(1) Open this access panel:

Number	Name/Location
322AL	Vertical Fin, Fixed Leading Edge

SUBTASK 05-41-03-210-008

(2) Do a General Visual inspection of the vertical fin - leading edge (only if HF system installed).

SUBTASK 05-41-03-410-005

(3) Close this access panel:

Number	Name/Location
322AL	Vertical Fin, Fixed Leading Edge

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-804-00-01

K94001 S0006584175_V2

Leading Edge - Vertical Fin General Visual (Internal)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN - LEADING EDGE			55-806-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLICABILITY
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	AIRPLANE: ALL ENGINE: ALL
		ACCESS			ZONE 322

Perform an external zonal inspection (GV) of the vertical fin - leading edge.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-806-00-01

MECH	INSP

TASK 05-41-03-210-809

1. **EXTERNAL - ZONAL (GV): VERTICAL FIN - LEADING EDGE**
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-009

(1) Do a General Visual inspection of the vertical fin - leading edge.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-806-00-01

K94933 S0006584177_V2

Leading Edge - Vertical Fin General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN - LEADING EDGE			55-808-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 36000 FC	REPEAT 36000 FC	APPLICABILITY
STATION	SKILL AIRPL	1.2 NOTE	12 YR	12 YR	AIRPLANE: ALL ENGINE: ALL
		ACCESS 322B			ZONE 322

Perform an internal zonal inspection (GV) of the vertical fin - leading edge.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-808-00-01

MECH	INSP

TASK 05-41-03-210-810

1. INTERNAL - ZONAL (GV): VERTICAL FIN - LEADING EDGE

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-010-006

(1) Open this access panel:

Number	Name/Location
322B	Vertical Fin, Removable Leading Edge

SUBTASK 05-41-03-210-010

(2) Do a General Visual inspection of the vertical fin - leading edge.

SUBTASK 05-41-03-410-006

(3) Close this access panel:

Number	Name/Location
322B	Vertical Fin, Removable Leading Edge

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-808-00-01

K94837 S0006584179_V2

Leading Edge - Vertical Fin General Visual (Internal)
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-808-00-01

A

K94839 S0006584180_V2

Leading Edge - Vertical Fin General Visual (Internal)
Figure 1 (Sheet 2 of 2)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN - FRONT SPAR TO REAR SPAR			55-810-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 4800 FC	REPEAT 4800 FC	APPLICABILITY
STATION	SKILL AIRPL	1.2 NOTE	24 MO	24 MO	AIRPLANE: ALL ENGINE: ALL
		ACCESS 323AL 323BL 323CL 323DL 323EL 323FL 323GL			ZONE 323

Perform an internal zonal inspection (GV) of the vertical fin - front spar to rear spar. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-810-00-01

MECH	INSP

EWIS

TASK 05-41-03-210-811

1. **INTERNAL - ZONAL (GV): Vertical Fin - Front Spar to Rear Spar**
(Figure 1)

A. General

(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.

B. Zonal Inspection

SUBTASK 05-41-03-010-007

(1) Open these access panels:

Number	Name/Location
323AL	Vertical Fin, Front Spar Access Door
323BL	Vertical Fin, Forward Fin Access Door
323CL	Vertical Fin, Rear Spar Access Door
323DL	Vertical Fin, Access
323EL	Vertical Fin, Access
323FL	Vertical Fin, Access
323GL	Vertical Fin, Access

SUBTASK 05-41-03-210-011

(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.

(3) Perform an internal zonal inspection (GV) of the vertical fin - front spar to rear spar. (EZAP)

SUBTASK 05-41-03-910-001

(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.

SUBTASK 05-41-03-410-007

(5) Close these access panels:

Number	Name/Location
323AL	Vertical Fin, Front Spar Access Door
323BL	Vertical Fin, Forward Fin Access Door
323CL	Vertical Fin, Rear Spar Access Door
323DL	Vertical Fin, Access
323EL	Vertical Fin, Access
323FL	Vertical Fin, Access
323GL	Vertical Fin, Access

END OF TASK

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-810-00-01

K94809 S0006584182_V2

Front Spar to Rear Spar - Vertical Fin General Visual (Internal)
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-810-00-01

K94817 S0006584183_V3

Front Spar to Rear Spar - Vertical Fin General Visual (Internal)
Figure 1 (Sheet 2 of 2)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN - FRONT SPAR TO REAR SPAR			55-812-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 5500 FC	REPEAT 5500 FC	APPLICABILITY
STATION	SKILL AIRPL	1.2 NOTE	30 MO	30 MO	AIRPLANE: ALL ENGINE: ALL
		ACCESS			ZONE 323

Perform an external zonal inspection (GV) of the vertical fin - front spar to rear spar.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-812-00-01

MECH	INSP

TASK 05-41-03-210-812

1. **EXTERNAL - ZONAL (GV): VERTICAL FIN - FRONT SPAR TO REAR SPAR**
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-012

(1) Do a General Visual inspection of the vertical fin - front spar to rear spar.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-812-00-01

K86258 S0006584185_V2

Front Spar to Rear Spar - Vertical Fin General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN - REAR SPAR TO TRAILING EDGE			55-814-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 4800 FC	REPEAT 4800 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	24 MO	24 MO	
		ACCESS 324AL 324BL 324DL 324HL			ZONE 324

Perform an internal zonal inspection (GV) of the vertical fin - rear spar to trailing edge. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-814-00-01

MECH	INSP

EWIS

TASK 05-41-03-210-813

1. **INTERNAL - ZONAL (GV): Vertical Fin - Rear Spar to Trailing Edge**
(Figure 1)

A. General

(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.

B. Zonal Inspection

SUBTASK 05-41-03-010-008

(1) Open these access panels:

Number	Name/Location
324AL	Vertical Fin, Aft Fin Access Door
324BL	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324HL	Vertical Fin, Access

SUBTASK 05-41-03-210-013

(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.

(3) Perform an internal zonal inspection (GV) of the vertical fin - rear spar to trailing edge. (EZAP)

SUBTASK 05-41-03-910-002

(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.

SUBTASK 05-41-03-410-008

(5) Close these access panels:

Number	Name/Location
324AL	Vertical Fin, Aft Fin Access Door
324BL	Vertical Fin, Trailing Edge Access
324DL	Trailing Edge Access
324HL	Vertical Fin, Access

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-814-00-01

K84534 S0006584187_V2

Rear Spar to Trailing Edge - Vertical Fin General Visual (Internal)
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-814-00-01

K84499 S0006584188_V2

Rear Spar to Trailing Edge - Vertical Fin General Visual (Internal)
Figure 1 (Sheet 2 of 2)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN - REAR SPAR TO TRAILING EDGE			55-816-00-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	
		ACCESS			ZONE 324

Perform an external zonal inspection (GV) of the vertical fin - rear spar to trailing edge.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-816-00-01

MECH	INSP

TASK 05-41-03-210-814

1. **EXTERNAL - ZONAL (GV): VERTICAL FIN - REAR SPAR TO TRAILING EDGE**
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-014

(1) Do a General Visual inspection of the vertical fin - rear spar to trailing edge.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-816-00-01

(LEFT SIDE)

A

(RIGHT SIDE)

A

K86580 S0006584190_V2

Rear Spar to Trailing Edge - Vertical Fin General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		RUDDER			55-818-00-01
DATE	TASK				RELATED CARD
	ZONAL (GV)				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	5500 FC	5500 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL	1.2	30 MO	30 MO	
	AIRPL	NOTE			
		ACCESS			ZONE
					325

Perform an external zonal inspection (GV) of the rudder.

INTERVAL NOTE: Whichever comes first.

EFFECTIVITY	SOURCE	
AKS ALL	MRB	RUDDER

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-818-00-01

MECH	INSP

TASK 05-41-03-210-815

1. EXTERNAL - ZONAL (GV): RUDDER

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-015

(1) Do a General Visual inspection of the rudder.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-818-00-01

(LEFT SIDE)

A

(RIGHT SIDE)

A

K75582 S0006584192_V2

Rudder General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		VERTICAL FIN TIP			55-820-00-01
DATE	TASK				RELATED CARD
	ZONAL (GV)				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	6600 FC	6600 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL	1.2	36 MO	36 MO	
	AIRPL	NOTE			
		ACCESS			ZONE
					326

Perform an external zonal inspection (GV) of the vertical fin tip.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-820-00-01

MECH	INSP

TASK 05-41-03-210-816

1. EXTERNAL - ZONAL (GV): VERTICAL FIN TIP

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-016

(1) Do a General Visual inspection of the vertical fin tip.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-820-00-01

K75021 S0006584194_V2

Vertical Fin Tip General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - LEADING EDGE - LEFT			55-822-01-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 331

Perform an external zonal inspection (GV) of the horizontal stabilizer - leading edge - left.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-822-01-01

MECH	INSP

TASK 05-41-03-210-817

1. **EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - LEADING EDGE - LEFT**
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-017

(1) Do a General Visual inspection of the horizontal stabilizer - leading edge - left.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-822-01-01

(BOTTOM VIEW)

(TOP VIEW)

K64926 S0006584196_V2

Leading Edge - Left Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - FRONT SPAR TO REAR SPAR - LEFT			55-824-01-01
DATE	TASK				RELATED CARD
	ZONAL (GV)				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	4000 FC	4000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL	1.2	18 MO	18 MO	
	AIRPL	NOTE			
		ACCESS			ZONE
					332

Perform an external zonal inspection (GV) of the horizontal stabilizer - front spar to rear spar - left.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-824-01-01

MECH	INSP

TASK 05-41-03-210-818

1. **EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - FRONT SPAR TO REAR SPAR - LEFT**
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-018

(1) Do a General Visual inspection of the horizontal stabilizer - front spar to rear spar - left.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-824-01-01

(BOTTOM VIEW)

(TOP VIEW)

K64947 S0006584198_V2

Front Spar to Rear Spar - Left Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - FRONT SPAR TO REAR SPAR - LEFT			55-826-01-01
DATE	TASK				RELATED CARD
	ZONAL (GV)				
TAIL NUMBER	WORK AREA	VERSION	THRESHOLD	REPEAT	APPLICABILITY
	EMPENNAGE	1.1	4000 FC	4000 FC	AIRPLANE: ALL / ENGINE: ALL
STATION	SKILL	1.2	18 MO	18 MO	
	AIRPL	NOTE			
		ACCESS			ZONE
		332AT			332
		NOTE			

Perform an internal zonal inspection (GV) of the inboard horizontal stabilizer - front spar to rear spar - left.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: As an alternate to the access provided, access panel 332AB can be removed for inspection from lower side of stabilizer.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-826-01-01

MECH	INSP

TASK 05-41-03-210-819

1. INTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - FRONT SPAR TO REAR SPAR - LEFT
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-010-009

(1) Open this access panel:

Number	Name/Location
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

NOTE: As an alternate to the access provided, access panel 332AB can be removed for inspection from lower side of stabilizer.

SUBTASK 05-41-03-210-019

(2) Do a General Visual inspection of the inboard horizontal stabilizer - front spar to rear spar - left.

SUBTASK 05-41-03-410-009

(3) Close this access panel:

Number	Name/Location
332AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-826-01-01

Front Spar to Rear Spar - Left Horizontal Stabilizer General Visual (Internal)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - REAR SPAR TO TRAILING EDGE - LEFT			55-828-01-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 4800 FC	REPEAT 4800 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	24 MO	24 MO	
		ACCESS 333AB 333AT 333BB 333CB 333DB 333EB 333FB			ZONE 333

Perform an internal zonal inspection (GV) of the horizontal stabilizer - rear spar to trailing edge - left. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-828-01-01

MECH	INSP

EWIS

TASK 05-41-03-210-820

1. **INTERNAL - ZONAL (GV): Horizontal Stabilizer - Rear Spar to Trailing Edge - Left**
(Figure 1)

A. General

(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.

B. Zonal Inspection

SUBTASK 05-41-03-010-010

(1) Open these access panels:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

SUBTASK 05-41-03-210-020

(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.

(3) Perform an internal zonal inspection (GV) of the horizontal stabilizer - rear spar to trailing edge - left. (EZAP)

SUBTASK 05-41-03-910-004

(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.

SUBTASK 05-41-03-410-015

(5) Close this access panel:

Number	Name/Location
333AB	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 05-41-03-410-010

(6) Close these access panels:

Number	Name/Location
333AT	Horizontal Stabilizer, Gap Cover, Horizontal Stabilizer to Body
333BB	Horizontal Stabilizer, Access Panel, Trailing Edge
333CB	Horizontal Stabilizer, Access Panel, Trailing Edge
333DB	Horizontal Stabilizer, Access Panel, Trailing Edge
333EB	Horizontal Stabilizer, Access Panel, Trailing Edge
333FB	Horizontal Stabilizer, Access Panel, Trailing Edge

END OF TASK

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-828-01-01

(BOTTOM VIEW)

A

L04717 S0006584202_V4

Rear Spar to Trailing Edge - Left Horizontal Stabilizer General Visual (Internal)
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-828-01-01

Rear Spar to Trailing Edge - Left Horizontal Stabilizer General Visual (Internal)
Figure 1 (Sheet 2 of 2)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - REAR SPAR TO TRAILING EDGE - LEFT			55-830-01-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 4800 FC 24 MO	REPEAT 4800 FC 24 MO	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 333

Perform an external zonal inspection (GV) of the horizontal stabilizer - rear spar to trailing edge - left.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-830-01-01

MECH	INSP

TASK 05-41-03-210-821

1. **EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - REAR SPAR TO TRAILING EDGE - LEFT**

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-021

(1) Do a General Visual inspection of the horizontal stabilizer - rear spar to trailing edge - left.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-830-01-01

L04719 S0006584205_V2

Rear Spar to Trailing Edge - Left Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - ELEVATOR - LEFT			55-832-01-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	
		ACCESS			ZONE 334

Perform an external zonal inspection (GV) of the horizontal stabilizer - elevator - left.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-832-01-01

MECH	INSP

TASK 05-41-03-210-822

1. **EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - ELEVATOR - LEFT**
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-022

(1) Do a General Visual inspection of the horizontal stabilizer - elevator - left.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-832-01-01

L04722 S0006584207_V2

Elevator - Left Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - STABILIZER TIP - LEFT			55-834-01-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 335

Perform an external zonal inspection (GV) of the horizontal stabilizer - stabilizer tip -left.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-834-01-01

MECH	INSP

TASK 05-41-03-210-823

1. **EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - STABILIZER TIP - LEFT**
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-023

(1) Do a General Visual inspection of the horizontal stabilizer - stabilizer tip -left.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-834-01-01

L04724 S0006584209_V2

Stabilizer Tip - Left Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - LEADING EDGE - RIGHT			55-836-02-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 6600 FC 36 MO	REPEAT 6600 FC 36 MO	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS			ZONE 341

Perform an external zonal inspection (GV) of the horizontal stabilizer - leading edge - right.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-836-02-01

MECH	INSP

TASK 05-41-03-210-824

1. EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - LEADING EDGE - RIGHT

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-024

(1) Do a General Visual inspection of the horizontal stabilizer - leading edge - right.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-836-02-01

K64944 S0006584211_V2

Leading Edge - Right Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - FRONT SPAR TO REAR SPAR - RIGHT			55-838-02-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 4000 FC	REPEAT 4000 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	18 MO	18 MO	
		ACCESS			ZONE 342

Perform an external zonal inspection (GV) of the horizontal stabilizer - front spar to rear spar - right.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-838-02-01

MECH	INSP

TASK 05-41-03-210-825

1. **EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - FRONT SPAR TO REAR SPAR - RIGHT**
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-025

(1) Do a General Visual inspection of the horizontal stabilizer - front spar to rear spar - right.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-838-02-01

Front Spar to Rear Spar - Right Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - FRONT SPAR TO REAR SPAR - RIGHT			55-840-02-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1 1.2 NOTE	THRESHOLD 4000 FC 18 MO	REPEAT 4000 FC 18 MO	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL				
		ACCESS 342AT NOTE			ZONE 342

Perform an internal zonal inspection (GV) of the inboard horizontal stabilizer - front spar to rear spar - right.

INTERVAL NOTE: Whichever comes first.

ACCESS NOTE: With access provided. Access panel 342AB can be used as an alternate to inspect from the lower side of the stabilizer.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-840-02-01

MECH	INSP

TASK 05-41-03-210-826

1. INTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - FRONT SPAR TO REAR SPAR - RIGHT
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-010-011

(1) Open this access panel:

Number	Name/Location
342AT	Gap Cover, Horizontal Stabilizer

NOTE: With access provided. Access panel 342AB can be used as an alternate to inspect from the lower side of the stabilizer.

SUBTASK 05-41-03-210-026

(2) Do a General Visual inspection of the inboard horizontal stabilizer - front spar to rear spar - right.

SUBTASK 05-41-03-410-011

(3) Close this access panel:

Number	Name/Location
342AT	Gap Cover, Horizontal Stabilizer

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-840-02-01

K67168 S0006584215_V2

Front Spar to Rear Spar - Right Horizontal Stabilizer General Visual (Internal)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - REAR SPAR TO TRAILING EDGE - RIGHT			55-842-02-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 4800 FC	REPEAT 4800 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	24 MO	24 MO	
		ACCESS 343AB 343AT 343BB 343CB 343DB 343EB 343FB			ZONE 343

Perform an internal zonal inspection (GV) of the horizontal stabilizer - rear spar to trailing edge - right. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 36000 FC/12 YR is satisfied by this zonal inspection.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-842-02-01

MECH	INSP

EWIS

TASK 05-41-03-210-827

1. **INTERNAL - ZONAL (GV): Horizontal Stabilizer - Rear Spar to Trailing Edge - Right**
(Figure 1)

A. General

(1) This Zonal inspection procedure satisfies the required EZAP-derived Zonal inspection requirement for this zone.

B. Zonal Inspection

SUBTASK 05-41-03-010-012

(1) Open these access panels:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

SUBTASK 05-41-03-210-027

(2) Do the zonal inspection following the procedures in AMM 05-00-00-910-804.

(3) Perform an internal zonal inspection (GV) of the horizontal stabilizer - rear spar to trailing edge - right. (EZAP)

SUBTASK 05-41-03-910-003

(4) Refer to 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to EWIS during maintenance.

SUBTASK 05-41-03-410-014

(5) Close this access panel:

Number	Name/Location
343AB	Horizontal Stabilizer, Gap Cover - H. Stab. to Body

(a) Make sure that the blade seal is installed correctly into the forward track channel.

SUBTASK 05-41-03-410-012

(6) Close these access panels:

Number	Name/Location
343AT	Horizontal Stabilizer, Gap Cover - H. Stab. to Body
343BB	Horizontal Stabilizer, Access Panel - T.E. Area
343CB	Horizontal Stabilizer, Access Panel - T.E. Area
343DB	Horizontal Stabilizer, Access Panel - T.E. Area
343EB	Horizontal Stabilizer, Access Panel - T.E. Area
343FB	Horizontal Stabilizer, Access Panel - T.E. Area

END OF TASK

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-842-02-01

L04718 S0006584217_V4

Rear Spar to Trailing Edge - Right Horizontal Stabilizer General Visual (Internal)
Figure 1 (Sheet 1 of 2)

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-842-02-01

L16872 S0006584218_V3

Rear Spar to Trailing Edge - Right Horizontal Stabilizer General Visual (Internal)
Figure 1 (Sheet 2 of 2)

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - REAR SPAR TO TRAILING EDGE - RIGHT			55-844-02-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 4800 FC	REPEAT 4800 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	24 MO	24 MO	
		ACCESS			ZONE 343

Perform an external zonal inspection (GV) of the horizontal stabilizer - rear spar to trailing edge - right.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-844-02-01

MECH	INSP

TASK 05-41-03-210-828

1. **EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - REAR SPAR TO TRAILING EDGE - RIGHT**

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-028

(1) Do a General Visual inspection of the horizontal stabilizer - rear spar to trailing edge - right.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-844-02-01

L04721 S0006584220_V2

Rear Spar to Trailing Edge - Right Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - ELEVATOR - RIGHT			55-846-02-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	
		ACCESS			ZONE 344

Perform an external zonal inspection (GV) of the horizontal stabilizer - elevator - right.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-846-02-01

MECH	INSP

TASK 05-41-03-210-829

1. **EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - ELEVATOR - RIGHT**
(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-029

(1) Do a General Visual inspection of the horizontal stabilizer - elevator - right.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-846-02-01

L04723 S0006584222_V2

Elevator - Right Horizontal Stabilizer General Visual (External)
Figure 1

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AIRLINE CARD NO		TITLE			BOEING CARD NO.
		HORIZONTAL STABILIZER - STABILIZER TIP - RIGHT			55-848-02-01
DATE	TASK ZONAL (GV)				RELATED CARD
TAIL NUMBER	WORK AREA EMPENNAGE	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLICABILITY AIRPLANE: ALL ENGINE: ALL
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	
		ACCESS			ZONE 345

Perform an external zonal inspection (GV) of the horizontal stabilizer - stabilizer tip - right.

INTERVAL NOTE: Whichever comes first.

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-848-02-01

MECH	INSP

TASK 05-41-03-210-830

1. EXTERNAL - ZONAL (GV): HORIZONTAL STABILIZER - STABILIZER TIP - RIGHT

(Figure 1)

A. Zonal Inspection

SUBTASK 05-41-03-210-030

(1) Do a General Visual inspection of the horizontal stabilizer - stabilizer tip - right.

——— END OF TASK ———

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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				55-848-02-01

L04725 S0006584224_V2

Stabilizer Tip - Right Horizontal Stabilizer General Visual (External)
Figure 1

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