

HYDRAULIC MACHINES

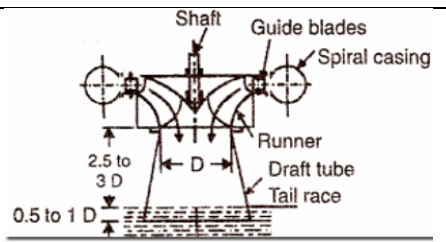
Pelton Turbine	Francis Turbine	Kaplan Turbine
Similarity Law	Cavitation	

1. PELTON TURBINE:

$H_g = H + H_f$ $H = \left(\frac{P}{\rho g} + \frac{V^2}{2g} \right)_{\text{inlet of nozzle}} = \frac{V^2}{2g} + H_{fn}$ $V = \sqrt{2g(H - H_{fn})} = C_v \sqrt{2gH}$			H_g = Gross Head H = Head Available at the inlet of the nozzle H_f = Head Loss in the Penstock H_{fn} = Head Loss in the Nozzle C_v = Coefficient of velocity (0.9-0.98) α = Vane Angle = Angle between \vec{u} & \vec{V} β = Blade Angle = Angle between \vec{u} & \vec{V}_r \vec{V} = Absolute Velocity of Water \vec{V}_r = Relative Velocity of Water w.r.t. Bucket \vec{u} = Bucket Velocity K = Bucket Friction Coefficient (0.9-0.98)
Force exerted by jet on the bucket: $F = \left(\frac{\text{Mass Striking on the}}{\text{Bucket per unit time}} \right) \left(\frac{\text{Decreasing in the vel.}}{\text{in direction of force}} \right)$			
$\vec{V} = \vec{u} + \vec{V}_r$	$V_{r1} = V - u$	$V_{r2} = K(V - u)$	
$F = \dot{m}(\vec{V}_{r1} - \vec{V}_{r2}) = \dot{m}[(V - u) + V_{r2} \cos \beta_2]$ $F = \dot{m}(V - u)[1 + K \cos \beta_2]$			
Torque $T = FR = \dot{m}(V - u)[1 + K \cos \beta_2]R$ Power Generated by Runner: $RP = T\omega = Fu$			Mass Striking, For Single Bucket, $\dot{m} = \rho a(V - u)$ For Multiple Bucket, $\dot{m} = \rho aV$
Wheel Efficiency: $\eta_w = \frac{RP}{KE} = \frac{2(V - u)[1 + K \cos \beta_2]}{V^2} = f(u)$ Condition For maximum η_w , $u = V/2$			$KE = \frac{1}{2} \dot{m} V^2$ = Kinetic Energy per unit time, $\eta_{w,max} = \frac{1 - K \cos \beta_2}{2}$
N_R = Run Away Speed = Maximum Possible Speed of Turbine Bucket (It can be Theoretical or Actual) Turbines are designed at N_R for Safety Aspect. Because of No load condition Turbines have max. Stress at N_R			

Nozzle Efficiency	Hydraulic Efficiency	Mechanical Efficiency	$HP = \rho g QH$ = Hydraulic Power RP = Runner Power SP = Shaft Power
$\eta_n = \frac{KE}{HP} = C_v^2$	$\eta_h = \frac{RP}{HP} = \eta_n \eta_w$	$\eta_m = \frac{RP}{SP}$	
Overall Efficiency: $\eta_o = SP/HP = \eta_h \eta_m = \eta_n \eta_w \eta_m$			
Jet Ratio		Speed Ratio	No of Buckets on Wheel
$m = \frac{\text{Wheel Diameter}}{\text{Jet Diameter}} = \frac{D}{d}$		$K_u = \frac{u}{V_{th}} = \frac{u}{\sqrt{2gH}} \cong \frac{u}{V}$	$Z = 0.5m + 15$ It's Tygun's Formula
Relation between frequency of AC & Runner Speed: 1 Rev. = 1 Cycle of AC		$f = \frac{PN}{60}$, Where P = No. of Poles.	

2. FRANCIS TURBINE:

Spiral Casing: Distribute water equally to all guide vanes. Guide Vanes: It guides water for smooth entry of water. It acts like nozzle (Converts pressure partially into KE). It Controls discharge through the runner. Runner: To generate torque/ Power output. Pressure decreases (Nozzle Effect & Centrifugal Force) and KE is also decreases. Old Turbine: Radial Entry & Radial Exit. Morden Turbine: Radial Entry & Axial Exit.	
\vec{V}_w = Velocity of Whirl = Component of \vec{V} along \vec{u} \vec{V}_f = Velocity of Flow = Component of \vec{V} Perpendicular to \vec{u}	α = Guide Vane = Angle between \vec{u} & \vec{V} β = Runner Blade = Angle between \vec{u} & \vec{V}_r
Euler's Turbine Equation:	
$\left(\frac{\text{Torque Generated}}{\text{By Runner}} \right) = \left(\frac{\text{Angular Momentum lost}}{\text{By Water Per unit time}} \right)$ $T = (\vec{r} \times \dot{m} \vec{V})_1 - (\vec{r} \times \dot{m} \vec{V})_2$	$T = r_1 \dot{m} \vec{V}_{w1} - r_2 \dot{m} \vec{V}_{w2}$ $T = \rho Q (r_1 \vec{V}_{w1} - r_2 \vec{V}_{w2})$ Euler eq. for Turbo M/c. $T = \rho Q (r_2 \vec{V}_{w2} - r_1 \vec{V}_{w1})$ Euler eq. for Pumps.

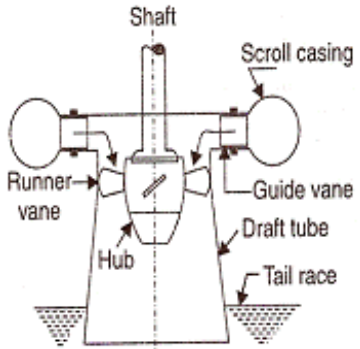
Inertial Frame has Zero acceleration.

Discharge Through Francis Turbine:	$Q_i = \pi D_1 B_1 V_{f1}$	$Q_e = \pi D_2 B_2 V_{f2}$
Torque Generated by Runner:	$T = \rho Q [r_1 V_{w1}]$	
Power Generated by Runner:	$RP = T\omega = \rho Q V_{w1} u_1$	

Various losses in Francis Turbine:

Hydraulic Efficiency	Mechanical Efficiency	Overall Efficiency
$\eta_h = \frac{RP}{HP} = \frac{V_{w1}u_1}{gH}$	$\eta_m = \frac{RP}{SP}$	$\eta_o = \frac{SP}{HP} = \eta_h\eta_m$
Speed Ratio (0.75-0.82)	Flow Ratio (0.15-0.30)	Head Developed by Turbine
$K_u = \frac{u_i}{\sqrt{2gH}} = f \left(\begin{matrix} \text{Geometric} \\ \text{Shape} \end{matrix} \right)$	$K_f = \frac{V_{fi}}{\sqrt{2gH}} = f \left(\begin{matrix} \text{Geometric} \\ \text{Shape} \end{matrix} \right)$	$H_e = \frac{V_{w1}u_1}{g}$

3. KAPLAN TURBINE:

<div>1. Spiral Casing: Same as Francis Turbine</div> <div>2. Guide Vanes: Same as Francis Turbine</div> <div>3. Swirl/ Whirl Chamber: At exit of Guide Vanes, 2 Component of velocities are present in the flow. 1) Radial 2) Tangential (Responsible for the whirling of the flow). Hence, Angular momentum of fluid partial is conserved in swirl chamber. $V_w r = Const.$ (Free Vortex Flow).</div> <div>4. Runner: $u_r = r\omega$ hence, velocities are changing w.r.t radius. So, velocity diagram will be different at different radial location. Here, Runner blades are adjustable. So, it's very costly.</div> <div>Propeller Turbines has fixed blades.</div> <div>5. Draft Tube:</div>	
<div><div>$u_1 = u_2 = r\omega$</div><div>$V_f = V_{f1} = V_{f2} = \frac{Q}{A_{f1}} = \frac{Q}{A_{f2}}$</div><div>$A_{f1} = A_{f2} = \frac{\pi}{4}(D_t^2 - D_h^2)$</div></div> <div><div>$D_t = \text{Blade Tip diameter}$</div><div>$D_h = \text{Hub diameter}$</div></div>	
<div>Discharge Through Kaplan Turbine:</div> <div>$Q = A_{f1}V_{f1} = A_{f2}V_{f2}$</div>	
<div>Torque Generated by Runner:</div> <div>$T = \rho Q [r_1 V_{w1}]$</div>	
<div>Power Generated by Runner:</div> <div>$RP = T\omega = \rho Q V_{w1}u_1$</div>	
<div>Hydraulic Efficiency:</div> <div>$\eta_h = \frac{RP}{HP} = \frac{V_{w1}u_1}{gH}$</div>	

Torque, RP, Hydraulic Efficiency for any radial location will remains exactly same.

Various losses in Kaplan Turbine:

Hydraulic Efficiency	Mechanical Efficiency	Overall Efficiency
$\eta_h = \frac{RP}{HP} = \frac{V_{w1}u_1}{gH}$	$\eta_m = \frac{RP}{SP}$	$\eta_o = \frac{SP}{HP} = \eta_h\eta_m$
Speed Ratio (1.3-2.3)	Flow Ratio (0.35-0.75)	Head Developed by Turbine
$K_u = \frac{u_{tip}}{\sqrt{2gH}} = f \left(\begin{matrix} \text{Geometric} \\ \text{Shape} \end{matrix} \right)$	$K_f = \frac{V_f}{\sqrt{2gH}} = f \left(\begin{matrix} \text{Geometric} \\ \text{Shape} \end{matrix} \right)$	$H_e = \frac{V_{w1}u_1}{g}$

4. SIMILARITY LAW:

Valid under homologous condition:		$d \propto D$	$Velocity \propto \sqrt{H}$
<ol style="list-style-type: none"> Model & Prototype are geometrically similar. Corresponding velocity triangles are also similar. η of the Hydraulic Machine depends on Velocity triangle.		$L_r = \frac{D_m}{D_p} = \frac{d_m}{d_p}$	$\frac{(V_{r1})_m}{(V_{r1})_p} = \frac{(u_1)_m}{(u_1)_p} = \frac{V_m}{V_p} = \sqrt{\frac{H_m}{H_p}}$
$u = \pi DN$		$ND \propto \sqrt{H}$	$C_H = \text{Head Coefficient}$
$Q = aV = (\pi/4)D^2\sqrt{2gH}$		$Q \propto D^2\sqrt{H}$	$C_Q = \text{Discharge Coefficient}$
$P = \eta_o \rho gQH$		$P \propto D^2H^{3/2}$	$C_P = \text{Power Coefficient}$

Important π Terms Valid for all Hydraulic Machines:

$\frac{Q}{ND^3}$	$\frac{gH}{N^2D^2}$	$Re = \frac{\rho ND^2}{\mu}$	$\frac{P}{\rho N^3D^5}$	$\frac{E/\rho}{N^2D^2}$
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SPECIFIC SPEED (N_s):

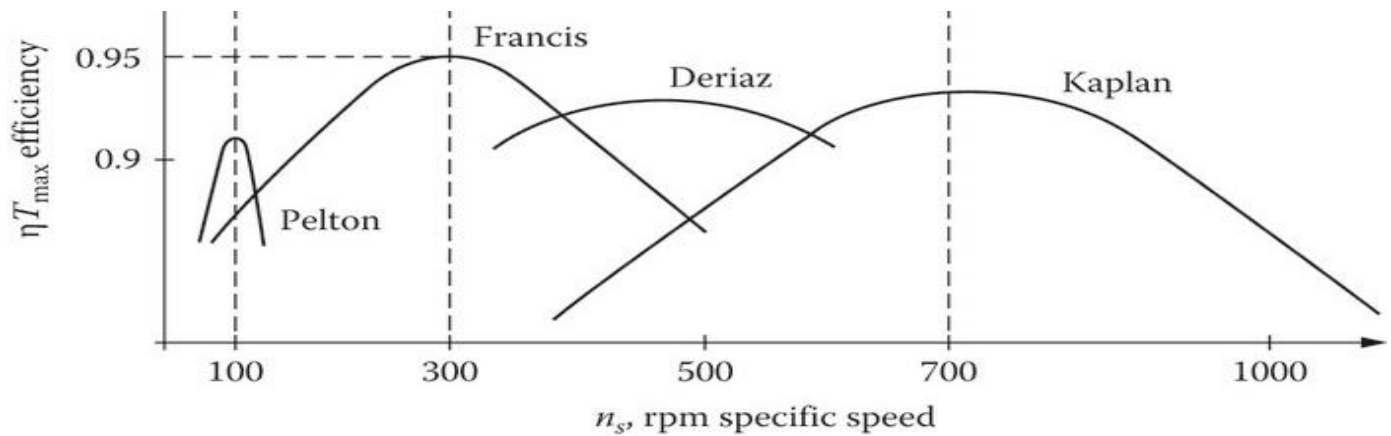
$C_H, C_Q, C_P, C_S = f \left(\begin{matrix} \text{Geometric Shape} \\ \text{Shape of Velocity Triangle} \end{matrix} \right)$	$\text{Speed Coefficient} = C_S = C_H\sqrt{C_P} = \frac{N\sqrt{P}}{H^{5/4}}$
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Specific Speed is the speed coefficient at maximum efficiency. Specific Speed is function of geometric shape only.

Selection of turbine based on specific Speed:

P = Based upon quantity of water available or power requirement.	H = Based upon head available.
N = Based upon gearbox & type of generator used.	

Based on the graph of Efficiency Vs. Specific Speed, the suitable turbine is selected.



TURBINE	TYPE	FLOW DIRECTION	$N_s (rpm \sqrt{KW}/m^{5/4})$	HEAD (M)	DISCHARGE
Pelton	Impulse	Tangential	8-30	High (50-1500)	Low
Francis	Reaction	Radially inward	40-400	Medium (25-350)	Medium
Kaplan	Reaction	Axial	300-900	Low (2-40)	High

Note:

1. For Pelton Turbine Power per jet is considered in case of MultiJet turbine.
2. Power is always considered in “KW”

DRAFT TUBE:

1. Allows installation of turbine above tail-race without effective loss in net head.
2. Recovers KE loss at exit partially.

Head Developed by Turbine $H_T = B_{1'} - B_{2'} \cong \frac{P_{1'} - P_{2'}}{\rho g}$	From the Bernoulli Eq. (2'-3), $\frac{P_{2'}}{\rho g} + \frac{V_{2'}^2}{2g} + (H_S + y) = \left(\frac{P_{atm}}{\rho g} + y\right) + \frac{V_3^2}{2g} + 0 + h_{fd}$
H_S = Recovery of lost Head, $\frac{V_{2'}^2 - V_3^2}{2g}$ = Partial Recovery of KE lost,	$\frac{P_{2'}}{\rho g} = \frac{P_{atm}}{\rho g} - \left(H_S + \frac{V_{2'}^2 - V_3^2}{2g} - h_{fd}\right) \leq \frac{P_{atm}}{\rho g}$

EFFICIENCY OF DRAFT TUBE (η_d):

$\eta_d = \frac{\text{Actual Recovery of KE head of draft Tube}}{\text{KE head at inlet of draft Tube}} = \frac{\frac{V_{2'}^2 - V_3^2}{2g} - h_{fd}}{\frac{V_{2'}^2}{2g}}$	$\eta_d = \frac{\text{Actual Recovery of KE head}}{\text{Theoretical Recovery}} = \frac{\frac{V_{2'}^2 - V_3^2}{2g} - h_{fd}}{\frac{V_{2'}^2 - V_3^2}{2g}}$
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5. CAVITATION: $P_{min} < P_{Vapour}$ Cavitation starts.

For turbines, Cavitation starts after turbine exit/ before draft tube entry if $P_{2'} < (P_{Vapour})_{water}$.

TOMA'S CAVITATION FACTOR (σ):

$\sigma = \frac{H_a - H_S - H_V}{H}$ Critical Cavitation Factor(σ_c) is given by manufacturer. For Cavitation not to happen, $\sigma < \sigma_c$.	H_a = Head of atmospheric pressure, H_S = Height of Draft tube, H_V = Head of Vapour Pressure, H = Net head on the turbine,
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