

1. CRYSTAL STRUCTURE

Solid Materials	
Crystalline Materials	Non-Crystalline Materials (Amorphous)
Eg. All metals, Some Ceramics, Quartz.	Eg. Plastic, Rubber, Glass (Ceramics)
High Strength	Low Strength
Anisotropic/ Un-isotropic	Isotropic
Sharp & High melting point	Low & Range of melting point.

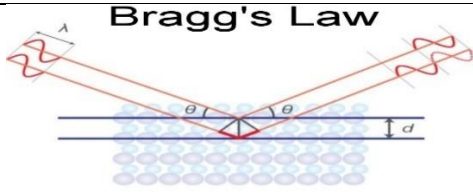
Bravis Lattice System (Types of Unit cells):

Total Number of Lattice System = 14	Basic or Primary Lattice System = 7
Cubic System	Triclinic System
$a = b = c$, And $\alpha = \beta = \gamma = 90^\circ$, Number of parameter required to describe system (n) = 1	$a \neq b \neq c$, And $\alpha \neq \beta \neq \gamma \neq 90^\circ$, Number of parameter required to describe system (n) = 6, Most Complex Structure

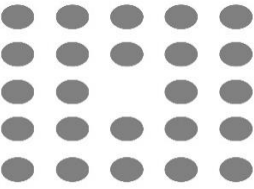
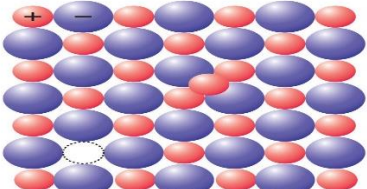
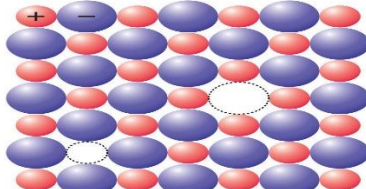
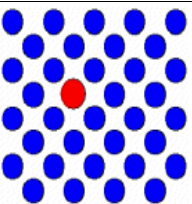
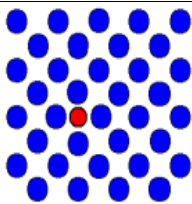
(1) Number of Atoms Required to form Cell (N): (2) Number of Atoms Inside the unit Cell (n): (3) Side length given by parameters (a, b, c): (4) Bond Length: Minimum Distance between two atoms. (5) Atomic Packing Factor (Packing Density): Total Volume of atom / Volume of Unit cell	(6) % of Empty Space in Cell: (7) Coordination Number: the number of atoms or ions immediately surrounding a central atom in a complex or crystal.
--	---

	Simple Cubic Structure	Body Centred Structure	Face Centred Cubic Structure	Diamond Cubic Structure/ Face centred Tetrahedral	Hexagonal Closed Packed Structure (HCP)
1	8	9	14	18	17
2	1 Atom	2 Atoms	4 Atoms	8 Atoms	6 Atoms
3	$a = 2R$	$\sqrt{3}a = 4R$	$\sqrt{2}a = 4R$	$\sqrt{3}a = 8R$	$\frac{c}{a} = 1.634$ & $a = 2R$
4	$2R$	$2R$	$a/\sqrt{2}$	$8R/\sqrt{3}a$	$\frac{2R}{a} / \frac{c}{1.634}$
5	0.52	0.68	0.74	0.34	0.74
6	48 %	32 %	26 %	66 %	26 %
7	6	8	12	4	12
Eg.	Plutonium	W, Iron (α -Fe), Cr, Na	Ni, Al, Au, Ag	Diamond, Si, Ge	Zn, Co, Mg, Ti, Be, Zr

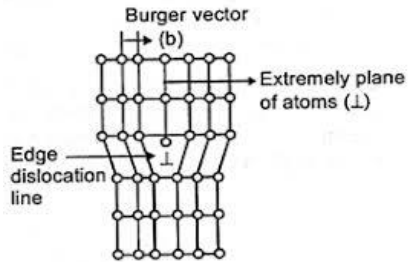
Crystallographic Directions: Line Drawn within unit-cells (Simple Vector).	Single Direction = $[x, y, z]$
Miller Indices: Integer denotation of Reciprocal of Plane Fraction.	Family of Direction = $\langle x, y, z \rangle$
Crystallographic Plane: Procedure: 1) Locate Origin in 3d, 2) Find plane fraction, 3) Reciprocal And convert in to integer number by multiplication.	Single Plane = $[miller\ indices]$ Family of Direction = $\{miller\ indices\}$
$Linear\ Density = \frac{Mass}{Volume} = \frac{No.\ of\ Atoms\ in\ a\ Direction}{Length\ of\ direction}$	
$Planner\ Density = \frac{Mass}{Volume} = \frac{No.\ of\ Atoms\ in\ a\ Plane}{Area\ of\ Plane}$	
$Inter\ Planer\ Distance\ (d) = \frac{1}{\sqrt{\left(\frac{h}{a}\right)^2 + \left(\frac{k}{b}\right)^2 + \left(\frac{l}{c}\right)^2}}$	Where, (h, k, l) = Miller Indices, a, b, c = Lattice Parameter.

X-ray Diffraction Method (Brag's Method 1912): $E \propto \frac{1}{\lambda}$ $AB + BC = n\lambda$	 <p style="text-align: center;">Bragg's Law</p> <p style="text-align: center;">$n\lambda = 2d \cdot \sin\theta$</p>	n = Order of reflection EG. 1,2, ... λ = Wave Length of X-ray, θ = Brag's Angle, 2θ = Deflection Angle, d = Inter Planer Distance.
--	--	--

IMPERFECTIONS IN SOLIDS/ DEFECTS			
Point Defect (0 D)		Line Defect (1 D)	Surface Defect (2 D)
Intrinsic Defect	Extrinsic Defect	1. Edge Dislocation 2. Screw Dislocation	1. Grain Boundary 2. Tile Boundary 3. Twin Boundary 4. Stacking Fault
1. Vacancy 2. Frenkel 3. Schottky	1. Substitutional 2. Interstitial		

Point Defect: Missing or Miss Placing of Atom.		
Intrinsic Defect: At high Temperature.		Extrinsic Defect: Adding Impurity.
Intrinsic Defect		
Vacancy Defect: Missing Atom	Frenkel Defect: Misplacing of ion (Mostly observed in Ionic Crystal NaCl)	Schottky Defect: Missing of pair of atoms Atom. Electrically balanced effect
		
Extrinsic Defect		
Substitutional Defects	Interstitial defect	 Substitutional foreign atom  Interstitial foreign atom
<ul style="list-style-type: none"> Adding foreign Atom in place of host material atom Size of Foragn atom is similar to Host atom. Strenght is not changing And other properties changes. Eg, Cr + Steel (Corrosion Resistance increase) 	<ul style="list-style-type: none"> Adding Impurity atoms at intersitional position. Size of Foragn atom <<<< Host atom. Strenght And other properties changes. Eg. C + Fe (Strength Increases) 	

Defects Are not bad.

Line Defect (1 D)		
Edge Dislocation: <ol style="list-style-type: none">Dislocated atoms move parallel to F.Only Translatory direction.Burger vector is parallel to F and perpendicular to dislocation	Screw Dislocation: <ol style="list-style-type: none">Dislocated atoms move perpendicular to F.Initial is translatory direction and followed by rotational/ Screw motion)	
Burger's Vector: Gives direction and magnitude of dislocation.		

Hall Petch Equation: Relation Between Strenght and Grain Size. $\sigma = \sigma_0 + K/\sqrt{d}$	σ = Strenght of material, d = Grain Size, σ_0, K = Constants for material.
--	--

GRAIN AND TYPES OF GRAIN MATERIAL:

Atoms are arranged in periodic order And separated by Grain Boundary.

At the grain Boundary:

More Bond Length => Low Bond Energy => Weak Region (Other Atom comes and shows effects Eg. Corrosion)

TYPES OF GRAIN MATERIAL	
FINE GRAIN MATERIAL	COARSE GRAIN MATERIAL
Smaller Grain Size	Larger Grain Size
More Grain Boundaries	Less Grain Boundaries
Obtained by Fast Cooling	Obtained by Slow Cooling
At room Temp (27° C): High Strenght (Eg. Deformation decreases at each grain boundary)	At room Temp (27° C): Low Strenght
At High Temp: Low Strenght	At High Temp: High Strenght

ASTM Grain Number (n): The number of grains per unit area is measured by ASTM Grain Size number (n) at Magnification (M) of 100X.	$\frac{\text{No. of grains}}{\text{in}^2} (N_x) = \frac{2^{n-1}}{(M/100)^2}$
--	--

SLIP SYSTEM

SLIP: It's Dislocation moment in a easy manner that requires low energy. **Closely packed Direction.**

SLIP SYSTEM: It's a combination of slip direction and slip plane. Eg. $\{<1,1,0>, \{1,1,1\}\}$ **Closely packed Plane.**

Ductility \propto No. of Slip System	Structure	No. of Slip System
Note: FCC is more ductile than BCC Because Active number of slip system are more. HCP Are brittle because it has less slip system.	HCP	3
	FCC	12
	BCC	48

2. MATERIAL PROPERTIES AND TESTING

Theoretical Density: It's measured by considering unit cell of a material.	n = No. of Atoms inside unitcell AW = Atomic Weight (g/ mol) AN = Avogadro's No. (1mol = 6.023*10 ²³) VUC = Volume of unit cell
$\text{Theoretical Density} = \left(\frac{\text{Mass}}{\text{Volume}} \right)_{\text{unitCell}} = \frac{n \text{ AW}}{\text{VUC}} = \frac{n * \text{AW}}{\text{AN} * \text{VUC}}$	

Elasticity	
Line Elastic Material	Non- Linear Elastic Material
Hook's Law Valid up to proportionality limit. By considering slop of Stress Strain curve we can find (Y).	

Engineering Stress $\sigma_E = \text{Load Applied} / \text{Initial Area}$	True Stress $\sigma_T = \text{Load Applied} / \text{Instantaneous Area}$
$\sigma_T = \frac{F}{A_f} = \sigma_E \frac{A_i}{A_f} = \sigma_E \frac{d_i^2}{d_f^2} = \sigma_E \frac{l_f}{l_i} = \sigma_E (1 + \epsilon_E) \left(\because \sigma_E = \frac{F}{A_i}, A_i l_i = A_f l_f \right)$	

POWER LAW OF STRAIN HARDENING

STRAIN HARDENING: It's one of the strengthening mechanisms, in this process, strength and hardness is increased by arresting dislocation motion. Strain hardened material is more brittle due to accumulation of dislocations at grain boundary.									
Strength of Rolled Component > Strength of Input Component									
The power law gives the relationship between true stress and true strain prior to the necking. It's valid up to ultimate point (Necking).									
$\sigma_T = K \epsilon_T^n$ Where, σ_T = True Stress, ϵ_T = True Strain, K = Constant of Material, n = Strain/ Work Hardening Exponent. $0 < n < 1$									
For Some materials at necking point,									
$\epsilon_T = n \text{ hence, } \sigma_T = K \epsilon_T^n$ Cold Workability \propto Strain hardening Component $(CW)_{FCC} > (CW)_{BCC} > (CW)_{HCP}$	<table> <tr> <th>Structure</th><th>n</th></tr> <tr> <td>HCP</td><td>0.05</td></tr> <tr> <td>BCC</td><td>0.25</td></tr> <tr> <td>FCC</td><td>0.5</td></tr> </table>	Structure	n	HCP	0.05	BCC	0.25	FCC	0.5
Structure	n								
HCP	0.05								
BCC	0.25								
FCC	0.5								

DUCTILITY AND MALLEABILITY

DUCTILITY: It's Ability of material, that deform plastically up to the failure point by applying tensile load. Making wires.	MALLEABILITY: It's Ability of material, that deform plastically in lateral direction by applying compressive load up to the failure point. Making Thin Sheets.
---	---

Measure of Ductility/ Malleability:

- % Increase/Decrease in Length = $\pm \left(\frac{l_f - l_i}{l_f} \right) * 100 = \epsilon * 100$
- % Increase/Decrease in C/S Area = $\pm \left(\frac{A_i - A_f}{A_i} \right) * 100 = \left(1 - \frac{d_f^2}{d_i^2} \right) * 100$

FCC MATEIALS DO NOT DISPLAY DUCTILE TO BRITTLE TRANSFORMATION PHENOMENON. AND HENCE THESE METALS ARE HAVING TOUGHNESS AT LOW TEMPERATURE AND HENCE MATERIALS ARE PREFERABLE FOR LOW TEMPERATURE FOR SUDDEN LOAD APPLICATION.	
---	--

TOUGHNESS:

It's ability of material that can absorb impact energy or energy up to the **failure point**.

Toughness is an **extrinsic** property of material (Depending on dimensions).

Modulus of toughness is an **intrinsic** property of material (In-Dependent of dimensions).

$$\text{Modulus of Toughness} = \text{Area of Stress Strain Curve Up to failure point} = \frac{\text{Toughness}}{\text{Volume}}$$

Toughness of Brittle materials are measurement	
Izod Impact test (Cantilever Beam Method)	Charpy Test (Simply Supported Beam Method)
Uniformly Varying Load	Uniformly Distributed Load

Notch is provided to initiate crack (Stress Concentration).

Charpy test is more accurate in comparison with Izod Impact test because of SSB hence UDL is applied.

NOTE: The Toughness of a ductile materials are measured by are method and toughness of brittle material are measured by impact test.

RESILIENCE:

It's ability of material that can absorb energy or energy up to the **Elastic Limit**.

$$\text{Modulus of Resilience} = \text{Area of Stress Strain Curve Up to Elastic Limit} = \frac{\text{Resilience}}{\text{Volume}}$$

HARDNESS	STRENGTH	STIFFNESS
<ul style="list-style-type: none"> It's the ability of material that can resist scratch or indentation or penetration on the surface of the material. Hardness is Surface property. 	<ul style="list-style-type: none"> It's the ability of material that can resist failure. It's Volumetric property of material. (Eg. $\sigma = 3.5BHN$) 	<ul style="list-style-type: none"> It's the ability of material that can resist elastic deformation. It's an extrinsic property of the material.
	Strength \propto Hardness	Stiffness \propto Young's Modulus

HARDNESS MEASUREMENT			
TEST	BRINAL HARDNESS TEST	VIECKER'S HARDNESS TEST	ROCKWELL HARDNESS TEST
APPLICABILITY	Low & Medium hardness material. Eg. Al, Cu, Brass, Mild Steel.	Very High hardness materials. Eg. SiC, WC, Cubic Boron Nitride (CBN), Si ₃ N ₄ , HCP	All (Low, Medium, High) Eg. L-CS, M-CS, H-CS, Alloy Steel, Etc...
INDENTATION TOOL	Hardened steel ball	Pyramid Shape Diamond Tool. Angle = 120°	Conical Shape Diamond tool. Cone angle = 136°
EQUATION	Area of penetration $A_p = \frac{\pi}{2} D \left[D - \sqrt{D^2 - d^2} \right]$ $BHN = \frac{P}{A_p}$	$D_{avg} = \frac{D_1 + D_2}{2}$ $VHN = \frac{1.854P}{D_{avg}^2}$	$RHN \propto \frac{1}{D_p}$
Where,	D = Indenter tool Diameter d = Indentation Diameter P = Applied Force	D ₁ , D ₂ = Indentation Diagonals P = Applied Force	D _p = Depth of Penetration.
TEST	SHORE TEST	BARCOL TEST	KNOOP TEST
APPLICABILITY	Thermoplastic materials. Soft Polymer. Eg. PVC, Nylon	Thermosetting Materials. Hard Polymer. Eg. Bakelite (Switch board)	Micro material.

CREEP FAILURE: It's the deformation of a material with respect to time at constant load and at high temperature (mainly). It's Also known as progressive deformation.

Creep Rate initially decreases and steady state and finally increasing (Transient) (Due to Neck formation)	
It's Bad and Creep resistance is increased by, <ol style="list-style-type: none"> Using high Strength and Young's Modulus material. Using Fine/Corse grain materials at room/High temperature. Adding Allowing element. Eg. Refractory Materials (High Melting point) (W, Mo, Nb, Ta >2500°C). 	

FATIGUE: It's a type of failure due to **dynamic** load or **cyclic** loads. And Failure are Catastrophic/ Sudden in nature.
Eg. Bridge Failure, Aeroplane Wings

Types of Dynamic loading: 1) Completely Reversible Loading, 2) Alternative Loads, 3) Repeated load, 4) Fluctuation Load

FRACTURE: Simple Fracture is separation of body or component into two or more pieces by Applying constant load (Static Load) at lower temperature (< Melting point).

Fracture Mechanism	<ol style="list-style-type: none"> 1. Crack Initiation or formation 2. Crack Propagation
---------------------------	--

TYPES OF FRACTURE:

1. BRITTLE FRACTURE

- It takes place without appreciable deformation in the material.
- Brittle Fracture is nearly perpendicular to applied load.
- Brittle Fracture in amorphous material (like ceramics & glass) is shiny and smooth.

2. INTER-GRANULAR FRACTURE

- The Crack propagates along the grain boundary.

3. TRANS-GRANULAR FRACTURE

- The crack propagation is passing through the grain in a specific crystallographic direction.

GRIFFITH THEORY:

Based on this theory, the crack will propagate, when a decrease in elastic strain energy is at least equal to the critical energy required to propagate crack.

$\sigma_c = \sqrt{\frac{2E\gamma_s}{\pi a}}$	<p>Where, σ_c = Critical Stress required to propagate crack, E = Young's Modulus, γ_s = Specific surface Energy, a = Half Crack Length.</p>	
--	---	--