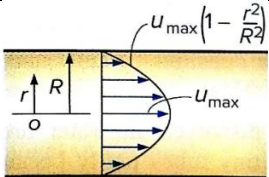


# 8. LAMINAR FLOW

**LAMINAR FLOW:** A flow in which layers (Laminas) of fluid slides over one another with relative velocities by the virtue of viscous forces.

<b>CHARACTERISTICS OF LAMINAR FLOW:</b> <ol style="list-style-type: none"> <li>1. <math>Re</math> &amp; <math>KE</math> of low is relatively lower.</li> <li>2. No intermixing of fluid particles between the layers.</li> <li>3. No random fluctuation of velocity of fluid particles with respect time.</li> <li>4. Newtons law of viscosity is enough to calculate shear stress.</li> <li>5. Flow is rotational.</li> <li>6. Surface roughness of pipe does not affect losses in laminar flow.</li> </ol>	<b>EXAMPLES OF LAMINAR FLOW:</b> <ol style="list-style-type: none"> <li>1. Flow past dust particles settling of impurities.</li> <li>2. Capillary flow in soil.</li> <li>3. Flow of blood in veins.</li> <li>4. Flow through pipe.</li> <li>5. Flow between parallel plates.</li> <li>6. Open channel flow: <math>Re \leq 500</math></li> <li>7. Flow past sphere (stokes law): <math>Re \leq 1</math></li> </ol>
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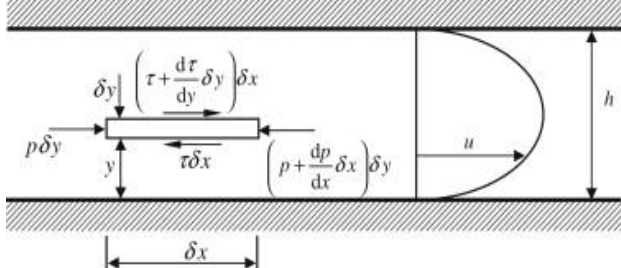
<b>LAMINAR FLOW THROUGH PIPE: <math>Re \leq 2000</math></b>		
1. Local Velocity: Here, $y = R - r \Rightarrow dy = -dr$		
$\tau = \mu \frac{du}{dy} = \mu \left( \frac{-du}{dr} \right) = \left( \frac{-dP}{dx} \right) \frac{r}{2} \Rightarrow du = \frac{1}{2\mu} \left( \frac{dP}{dx} \right) r dr \text{ (From Previous Chp.)}$		
$u = \frac{1}{4\mu} \left( \frac{dP}{dx} \right) r^2 + C$	$C = \frac{1}{4\mu} \left( \frac{-dP}{dx} \right) R^2 (\because BC @ r = R, u = 0)$	$u = \frac{1}{4\mu} \left( \frac{-dP}{dx} \right) (R^2 - r^2)$
This is equation of local velocity in laminar flow through pipes.		
<b>OBSERVATIONS:</b> 1. $u = f(r)$ . 2. Local Velocity decreases parabolically with respect to “r”. 3. Maximum Local Velocity is attaining at centreline. $r = 0$		$u_{max} = \frac{1}{4\mu} \left( \frac{-dP}{dx} \right) R^2$ $\therefore \frac{u}{u_{max}} = 1 - \frac{r^2}{R^2}$
2. Discharge:		
$dQ = u dA = u_{max} \left( 1 - \frac{r^2}{R^2} \right) 2\pi r dr$	$Q = \pi R^2 \frac{u_{max}}{2}$	
Average velocity: $V_{avg.} = Q/A$	$V_{avg.} = u_{max}/2$	
<b>NOTE:</b> At $u = V_{avg.}$ (Local Velocity = Avg. Velocity), $R = \sqrt{2}r$ .		
3. Pressure Drop Equation (Hagen-Poiseuille Equation):		
$V_{avg.} = \frac{u_{max}}{2} = \frac{1}{8\mu} \left( \frac{-dP}{dx} \right) R^2 = \frac{1}{8\mu} \left( \frac{\Delta P}{L} \right) R^2$	$\therefore \frac{-dP}{dx} = \frac{\Delta P}{L}$	$\Delta P = \frac{32\mu V_{avg.} L}{D^2} = \frac{128\mu Q L}{\pi D^4}$
4. Head Loss:	5. Equation of Power Loss:	
$h = \frac{\Delta P}{\gamma} = \frac{32\mu V_{avg.} L}{\gamma D^2} = \frac{128\mu Q L}{\gamma \pi D^4}$	$P = h\gamma Q = \frac{128\mu Q^2 L}{\pi D^4}$	
6. Fanning's Friction coefficient( $f'$ ):		
$f' = \frac{\tau_0}{(1/2)\rho V^2} = \frac{16}{Re}$	$\because \tau_0 = \left( \frac{-dP}{dx} \right) \frac{R}{2} \text{ \& } V_{avg.} = \frac{1}{8\mu} \left( \frac{-dP}{dx} \right) R^2$	
7. Darcy Friction Factor( $f$ ): $f = 4f' = 64/Re$		

## SUMMARY

1. Local Velocity	2. Avg. Velocity ( $V_{avg.} = u_{max}/2$ )	3. Discharge ( $Q = V_{avg.} A$ )
4. Drop in Pressure ( $-dP/dx = \Delta P/L$ )	5. Head Loss ( $h = \Delta P/\gamma$ )	6. Power Loss ( $P = h\gamma Q$ )
7. $f' = 16/Re$	8. $f = 4f' = 64/Re$	

**NOTE:** If pipe is inclined, Replace Pressure with piezometric pressure.  $P = P^*$  &  $x$  is axis of flow.

## FLOW BETWEEN TWO FIXED PARALLEL PLATES: $Re \leq 1000$

From continuity equation, $V_1 = V_2$ From momentum conservation equation, $PdyW - (P + dP)dyW + (\tau + d\tau)dxW - \tau dxW = 0$ $\frac{dP}{dx} = \frac{d\tau}{dy} \text{ (Equation Governing The Flow)}$ $\tau = \mu \frac{du}{dy} \Rightarrow \frac{d\tau}{dy} = \frac{dP}{dx} = \frac{d^2u}{dy^2}$		
$u = \frac{1}{2\mu} \left( \frac{dP}{dx} \right) y^2 + C_1 y + C_2$	$C_1 = \frac{1}{2\mu} \left( \frac{-dP}{dx} \right) B \left( \because BC @ y = B/2, u = u_{max} \Rightarrow du/dy = 0 \right)$	
$C_2 = 0 \left( \because BC @ y = 0, u = 0 \right)$	$u_{max} = \frac{1}{8\mu} \left( -\frac{dP}{dx} \right) B^2 \left( \because u = u_{max} @ y = \frac{B}{2} \right)$	
Discharge:		
$dQ = u dA = \frac{1}{2\mu} \left( -\frac{dP}{dx} \right) (By - y^2) dyW$	$Q = \frac{1}{12\mu} \left( -\frac{dP}{dx} \right) B^3 W$	
Average velocity: $V_{avg.} = Q/A$	$V_{avg.} = (2/3) u_{max}$	
Pressure Drop:		
$\Delta P = \frac{12 \mu V_{avg.} L}{B^2} = \frac{12 \mu Q L}{B^3 W} \left( \because \text{From } V_{avg.} \text{ Formula} \right)$	$\therefore \frac{-dP}{dx} = \frac{\Delta P}{L}$	
Head Loss:		Power Loss:
$h = \frac{\Delta P}{\gamma} = \frac{12 \mu V_{avg.} L}{\gamma B^2} = \frac{12 \mu Q L}{\gamma B^3 W}$		$P = h \gamma Q = \frac{12 \mu Q^2 L}{B^3 W}$
Shear Stress:		Wall Shear Stress: $\tau = \tau_0 @ y = 0$
$\tau = \mu \frac{du}{dy} = \left( -\frac{dP}{dx} \right) \left( \frac{B - 2y}{2} \right) \text{ (Linear Profile)}$		$\tau_0 = \left( -\frac{dP}{dx} \right) \left( \frac{B}{2} \right)$

**NOTE:** Here,  $y$  is distance measured from wall not centre.

### SUMMARY:

1. Local Velocity	2. Avg. Velocity $[V_{avg.} = (2/3) u_{max}]$	3. Discharge $(Q = V_{avg.} A)$
4. Drop in Pressure $(-dP/dx = \Delta P/L)$	5. Head Loss $(h = \Delta P/\gamma)$	6. Power Loss $(P = h \gamma Q)$
7. Shear Stress		

**NOTE:** If Plates are inclined, Replace Pressure with piezometric pressure.  $P = P^*$  &  $x$  is axis of flow.

**COUETTE FLOW:** Laminar Flow Between two parallel plates such that one plate is moving relative to the other.

TYPES OF COUETTE FLOW				
$\frac{dP}{dx} > 0$ (Parabolic)	$\frac{dP}{dx} = 0$ (Linear)	$\frac{dP}{dx} < 0$ (Parabolic)		
Backward Couette Flow: Laminar Flow is in the direction opposite to the moving Plate.	Simple/ Pure/ Plane Couette Flow	Forward Couette Flow: Laminar Flow is in the direction of the moving Plate.		
$u = \frac{1}{2\mu} \left( \frac{dP}{dx} \right) y^2 + C_1 y + C_2$	$\frac{u}{U} = \frac{y}{B}$	$u = \frac{1}{2\mu} \left( \frac{dP}{dx} \right) y^2 + C_1 y + C_2$		
<p><math>P &lt; -1</math></p> <p><math>\frac{dp}{dx} \gg 0</math></p> <p>Pressure increases in the direction of upper-plate motion</p>	<p><math>P &lt; 0</math></p> <p><math>\frac{dp}{dx} &gt; 0</math></p>	<p><math>P = 0</math></p> <p><math>\frac{dp}{dx} = 0</math></p> <p>Zero pressure gradient</p>	<p><math>P &gt; 0</math></p> <p><math>\frac{dp}{dx} &lt; 0</math></p>	<p><math>P &gt; 1</math></p> <p><math>\frac{dp}{dx} \ll 0</math></p> <p>Pressure decreases in the direction of upper-plate motion</p>
$V_{CF} = V_{Simple Flow} + V_{Flow Btw.2fixed plate}$	$Q_{CF} = Q_{Simple Flow} + Q_{Flow Btw.2fixed plate}$			
$V_{CF} = U \left( \frac{y}{B} \right) + \frac{1}{2\mu} \left( -\frac{dP}{dx} \right) (By - y^2)$	$Q_{CF} = \frac{1}{2} UBW + \frac{1}{12\mu} \left( -\frac{dP}{dx} \right) B^3 W$	$V_{avg.} = \frac{Q}{A} = \frac{Q}{WB}$		
$\tau_{CF} = \tau_{Simple Flow} + \tau_{Flow Btw.2fixed plate}$	$\tau_{CF} = \mu \frac{dV_{CF}}{dy} = \mu \left( \frac{U}{B} \right) + \left( -\frac{dP}{dx} \right) \left( \frac{B - 2y}{2} \right)$			
$u_{max} = u _{\frac{du}{dy}=0}$				