

High Altitude Long Endurance Unmanned Aerial Vehicles HALE-UAVs

A Brief Introduction

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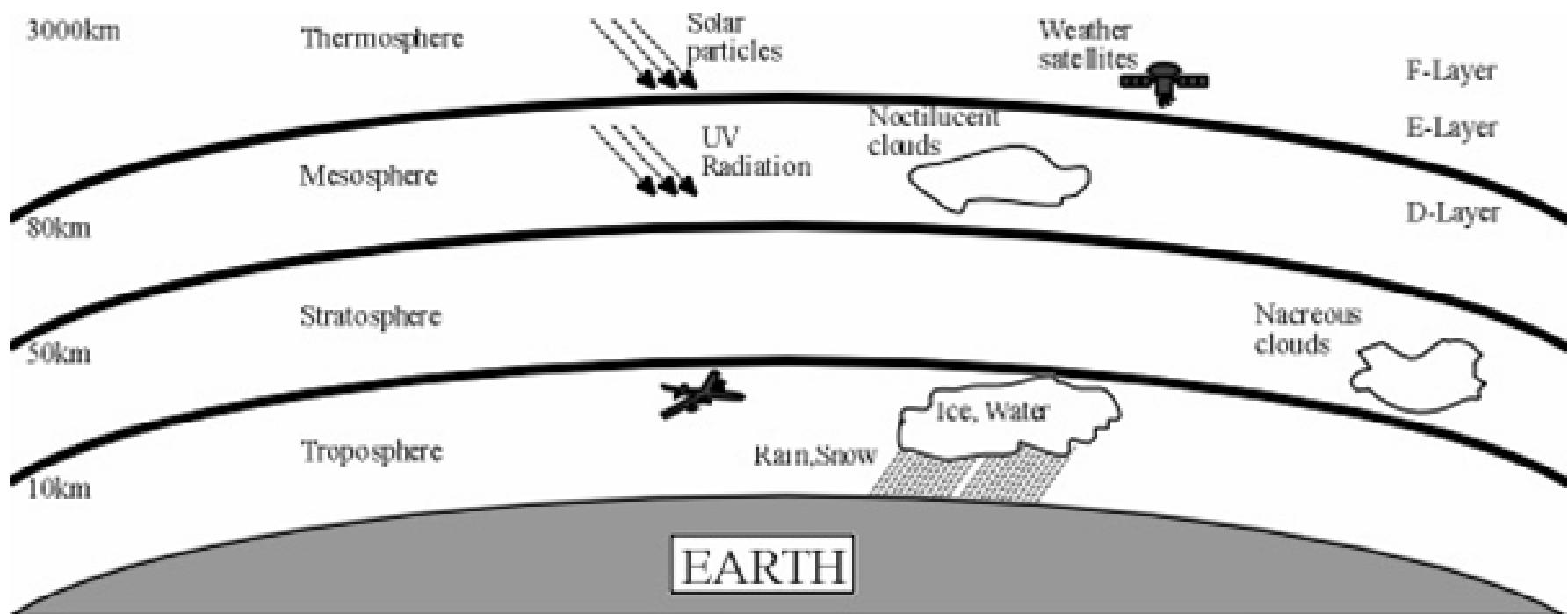
2012-15

High Altitude

How high is high?

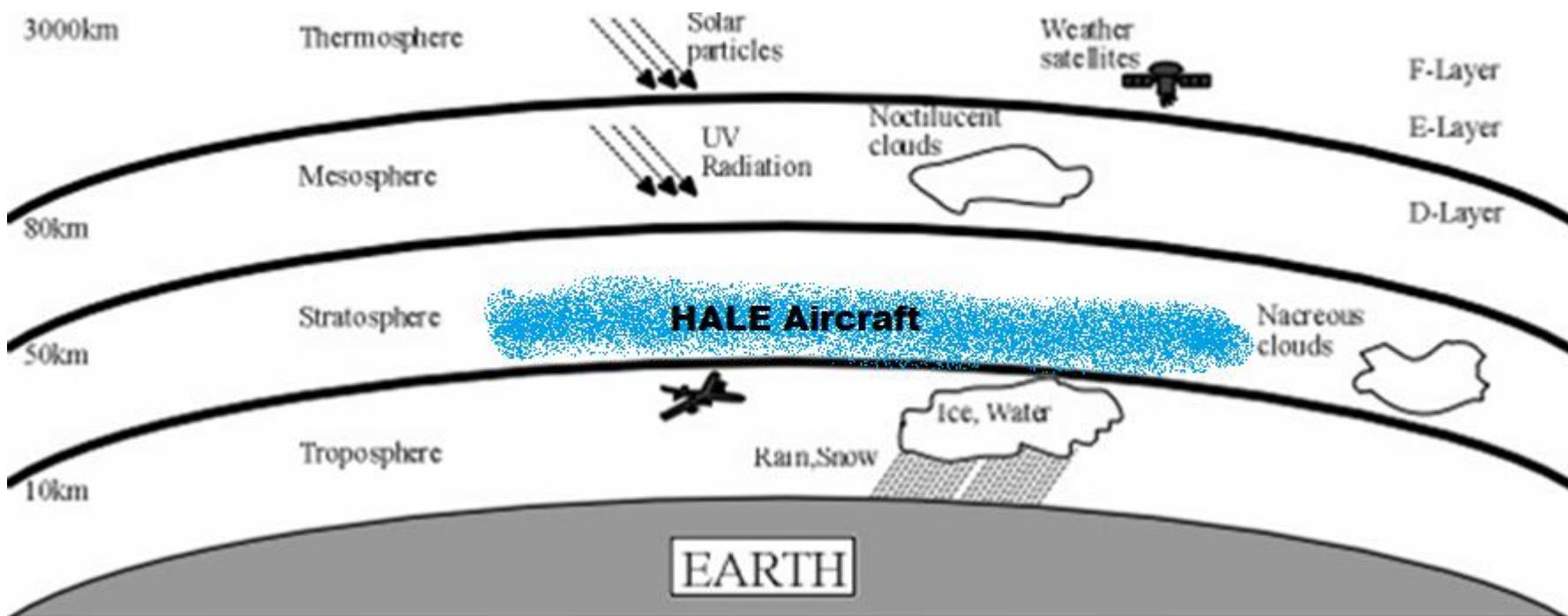
High Altitude

How high is high?



High Altitude

How high is high?



Long Endurance

How long is long?

- Flight durations from Mumbai



Mumbai To	Travel Time (at 10 km altitude, ~ 0.8 M)
Delhi	2 hours
Dubai, UAE	3 hours
Paris, France	9 hours
Newark, USA	16 hours
Sydney to DFW, Texas	17 hours

Long Endurance

- Around the world



- Virgin Atlantic Global Flyer designed by Burt Rutan
- Around the world = Absolute World Record for the fastest nonstop unrefueled circumnavigation: 67 hours 1 minute.

Applications

Earth Science



Pseudo Satellites



Surveillance

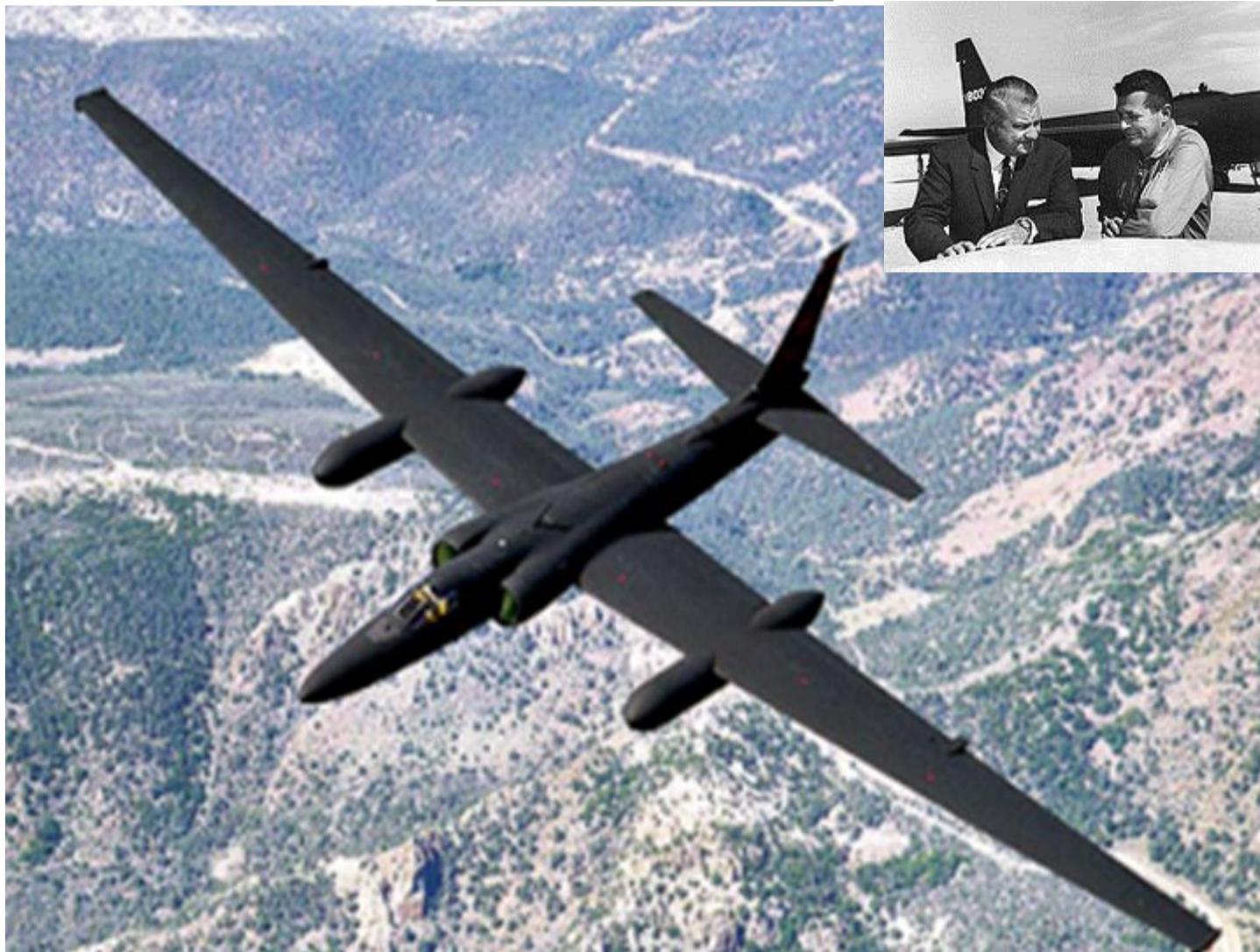


Key Mission Features

- Flies above
 - Weather
 - Interception
 - Commercial flights
- Payloads cover larger area of the earth at a time
 - SAR RADAR
 - Satellite Communication

Legacy HALE Aircraft

Lockheed U-2



Existing HALE Aircraft Designs

Northrop Grumman's
Global Hawk RQ-4



AeroVironment Global
Observer



Boeing Phantom Eye



Qinetiq Zephyr



Odysseus



NASA Helios



GLOBAL HAWK CASE STUDY

RQ 4A (2000)

RQ 4B (2007)

EuroHawk (2007)



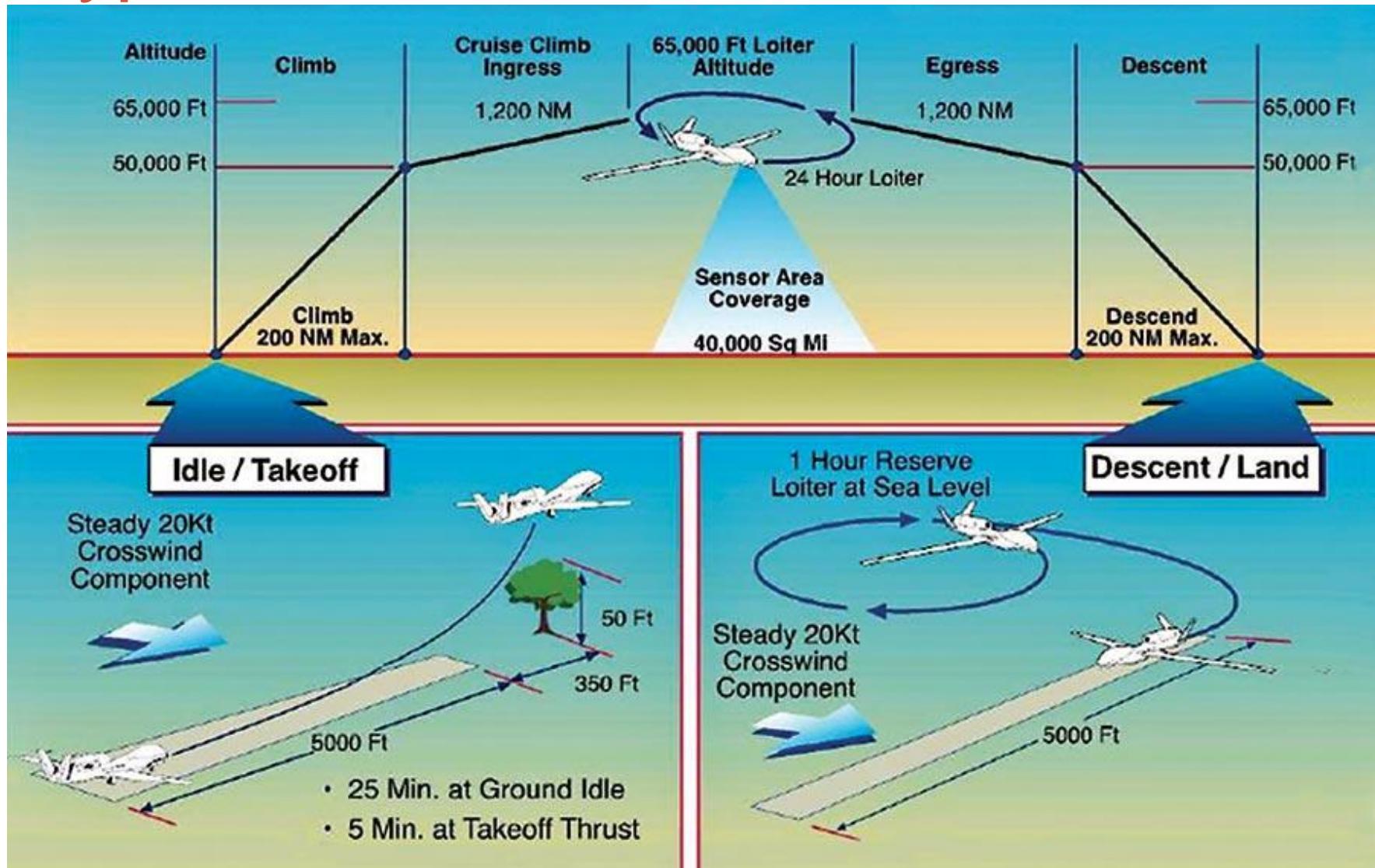
RQ-4B Aircraft Specifications

Endurance	32+ hours
(Ferry) Range	12,300 nm (~22,780 km)
Service Ceiling	60,000 feet (18.3 km)
Cruise Speed	310 knots (574 kmph) ~0.6 Mach
Flight Crew	3 remote pilots
Wingspan	39.9 m
Length	14.5 m
Engine	Turbo-fan, Rolls Royce F137-RR-100
Payload	Suite of RADAR and SIGINT
GTOW	14628 kg
Payload Weight	1360 kg (~9%)
Fuel Weight	6500 kg (~45%)
Empty Weight	6740 kg (~46%)

Source: RQ-4 Block 40 Global Hawk

http://www.northropgrumman.com/Capabilities/GlobalHawk/Documents/Datasheet_GH_Block_40.pdf

Typical Global Hawk Mission

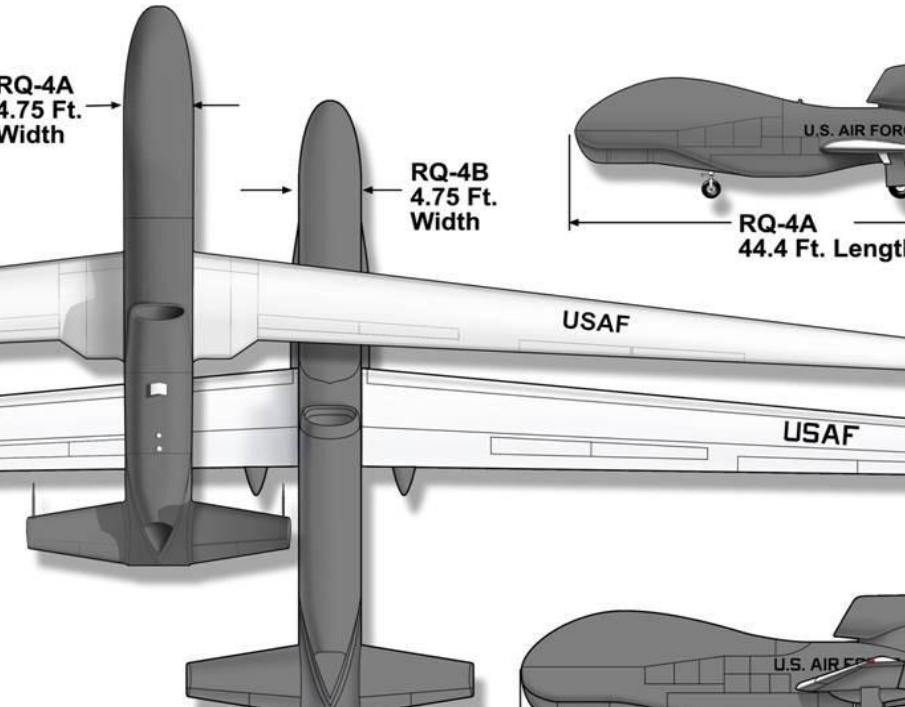


Features of Configuration?

RQ-4A

Gross Take-off Weight	26,700 lbs (12,110.9kg)
Payload	2,000 lbs (907.2 kg)
Ferry Range	12,000 nm (22,236km)
Maximum Altitude	More than 65,000 ft (19.8 km)
Loiter Velocity	343 kts TAS
On-Station Endurance at 1,200 nm	24 Hours
Maximum Endurance	35 hours

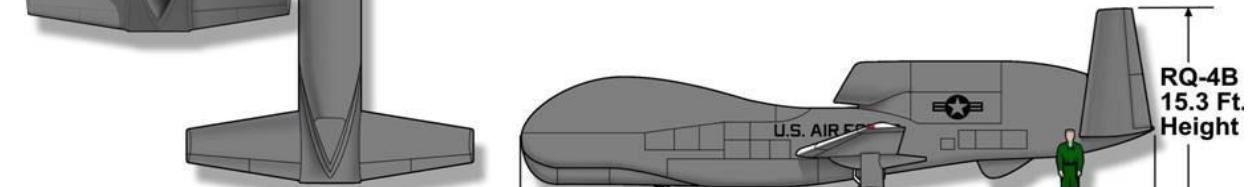
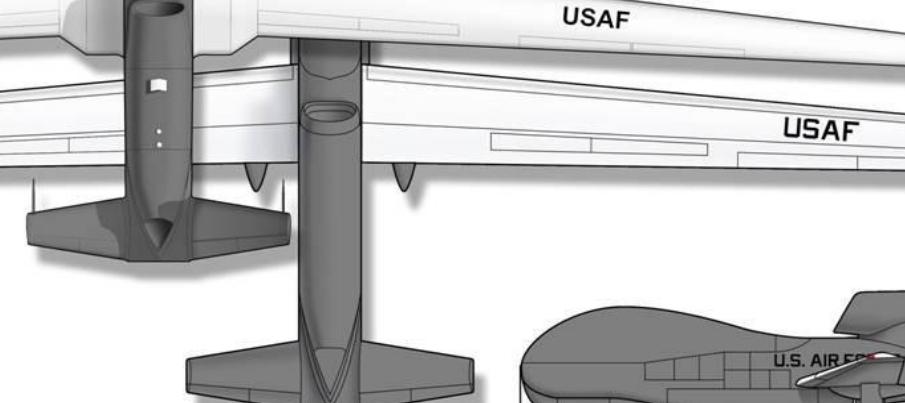
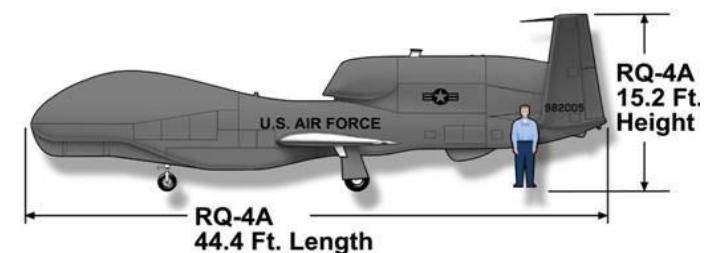
RQ-4A
4.75 Ft.
Width



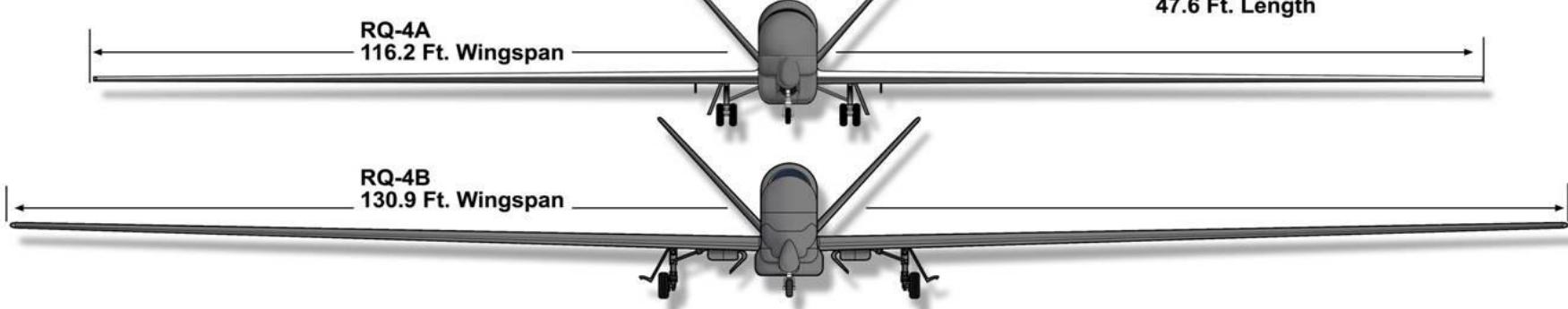
RQ-4B

Gross Take-off Weight	32,500 lbs (14,628kg)
Payload	3,000 lbs (1,360 kg)
Ferry Range	12,300 nm (22,780km)
Maximum Altitude	More than 60,000 ft (18.3 km)
Loiter Velocity	343 kts TAS
On-Station Endurance at 1,200 nm	24 Hours
Maximum Endurance	36 hours

RQ-4B
4.75 Ft.
Width



RQ-4A
116.2 Ft. Wingspan



Crew

Monoplane/Biplane

High/mid/low wing

Engine type

Number of engines

Tail type

Landing Gear Configuration

??

Design Features

- High Aspect Ratio Unswept wings



$$C_D = C_{D0} + \frac{C_l^2}{\pi e AR}$$

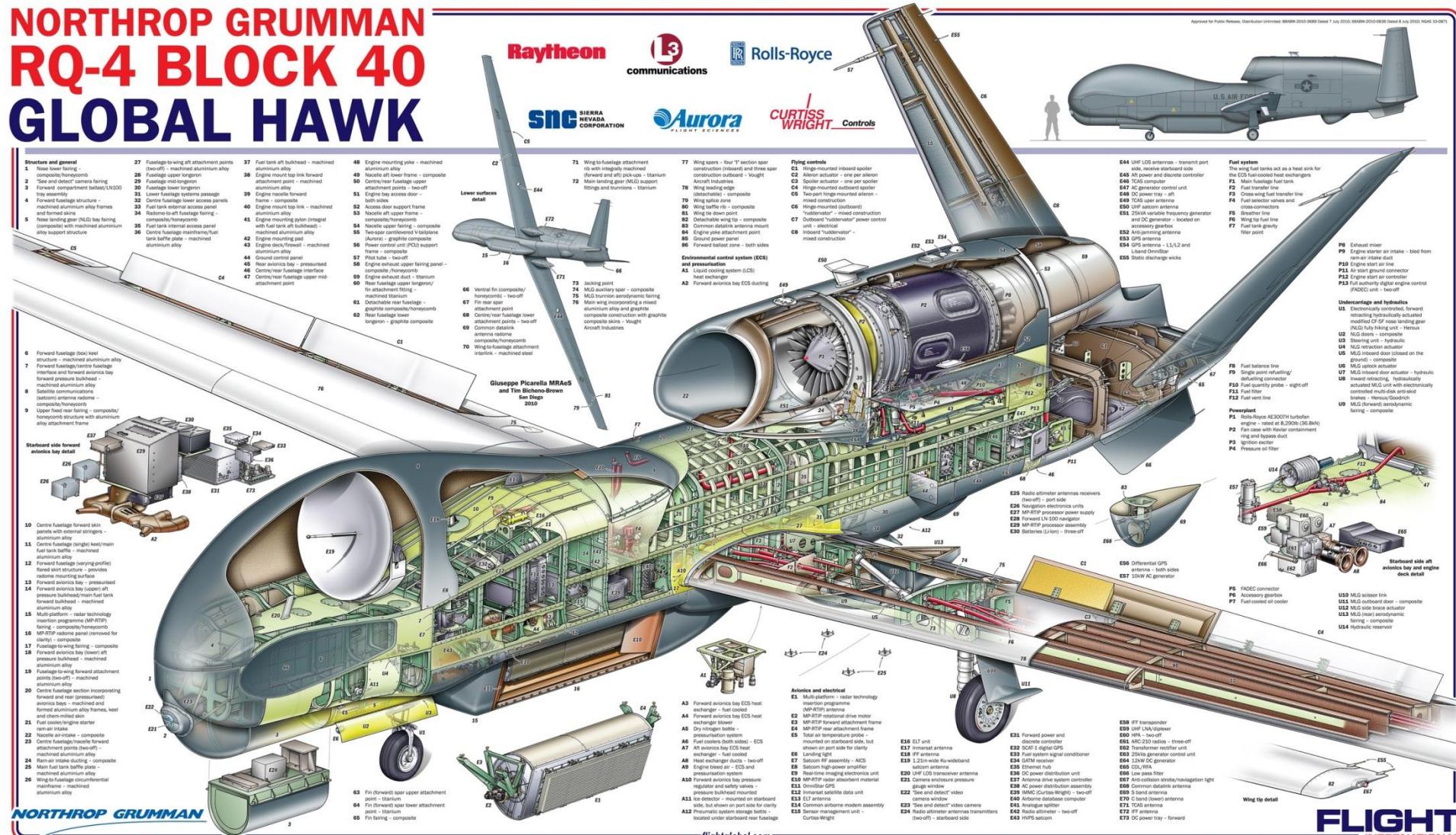
Design Features

- Bulging Beluga-whale nose



Inside the Global Hawk

NORTHROP GRUMMAN RQ-4 BLOCK 40 GLOBAL HAWK



Inside the Global Hawk Operations Centre



Sizing Method for a HALE UAV*

Step 1: Estimation of MTOW

- Empty Weight Fraction table*

UAV	Empty Weight Fraction
Predator – C	0.390
U-2S	0.445
Grob Stratos 2C	0.500
Global Hawk	0.460

- Crew weight = 0 (unmanned)

*Lloyd R. Jenkinson and James F. Marchman III, Aircraft Design Projects for Engineering Students, ISBN: 978-0-7506-5772-3, Elsevier Ltd., 2003

Sizing Method

- Mission Segment Weight Fractions:
 - As usual (Refer Global Hawk mission profile)
- Cruise segment weight fraction from Breguet Endurance equation:
 - SFC(cruise) for a medium bypass ratio turbofan engine = 0.55 lb/lb-hr*
 - For a 2020 engine, take cruise SFC = 0.47 – 0.50**
 - Reserve fuel = 10%
 - $L/D_{cruise} = L/D_{max}$ (Condition for Jet Aircraft Endurance)

* Lloyd R. Jenkinson and James F. Marchman III, Aircraft Design Projects for Engineering Students, ISBN: 978-0-7506-5772-3, Elsevier Ltd., 2003

** suggested by Dr. Scott Eberhardt, in Prof. Pant's lecture notes

Suggested values for max L/D

	L/D_{\max} Range	Average L/D_{\max}
Propeller Personal/Utility	9.6–14.2	12.1
Propeller Commercial Transport	13.8–18.5	16.3
Business Jet	13.0–15.6	14.3
Commercial Jet Transport	15.0–18.2	14.4
Military Transport/Bomber	17.5–20.5	18.9
Military Fighter (subsonic cruise)	9.2–13.9	11.0

Global Hawk* **33 to 34**

* from *Flight International*, UAVs, page 28, 5 / 1/01. Source: Chaput

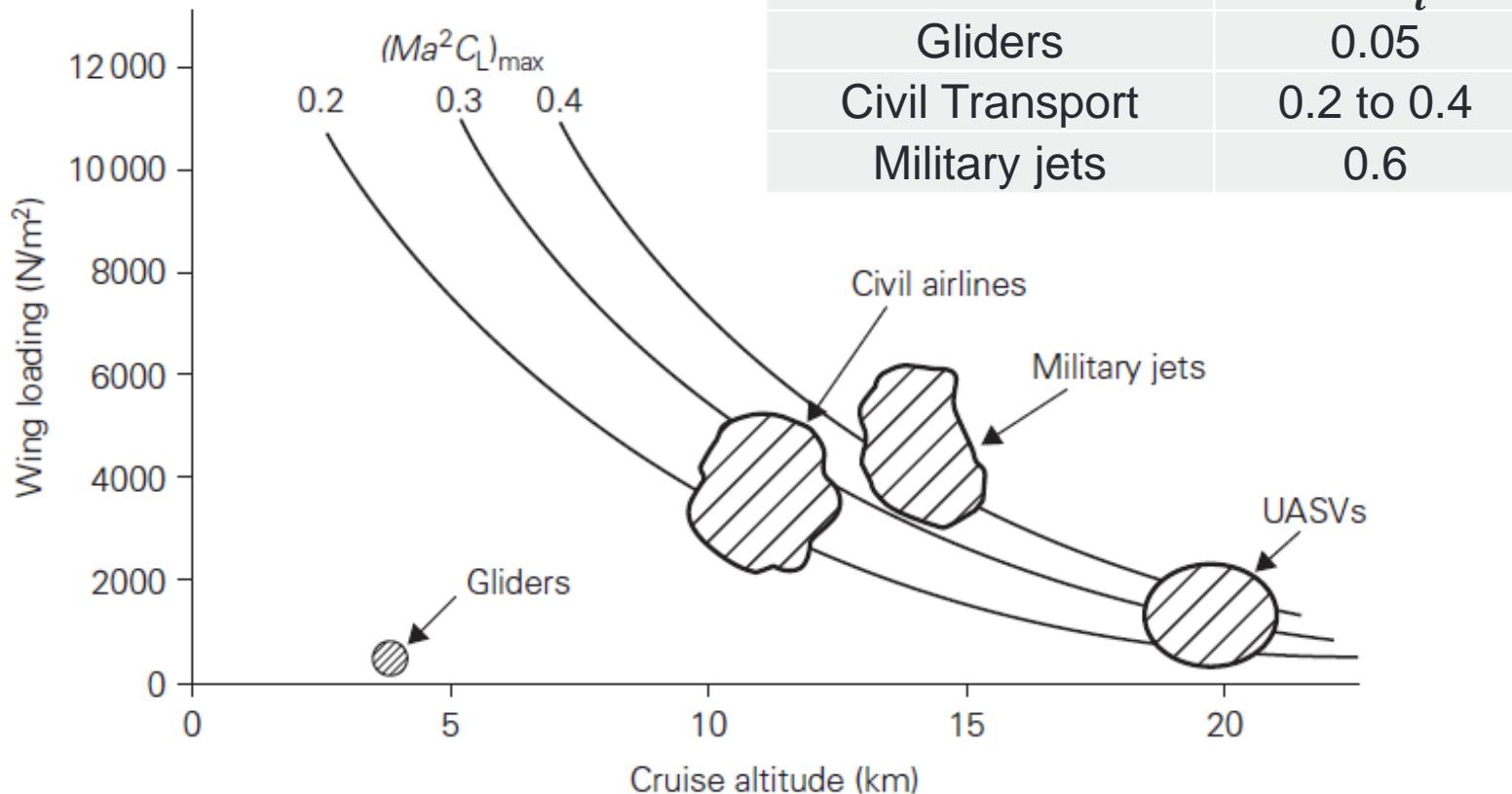
Sizing Method

- GTOW Estimation:

$$W_0 = \frac{W_{crew} + W_{payload}}{1 - (WF_e + WF_f)}$$

Sizing Method

- Wing-loading Calculation, knowing $M^2 C_{Lmax}$ gives wing area



Wing Planform Design

- High Aspect Ratio

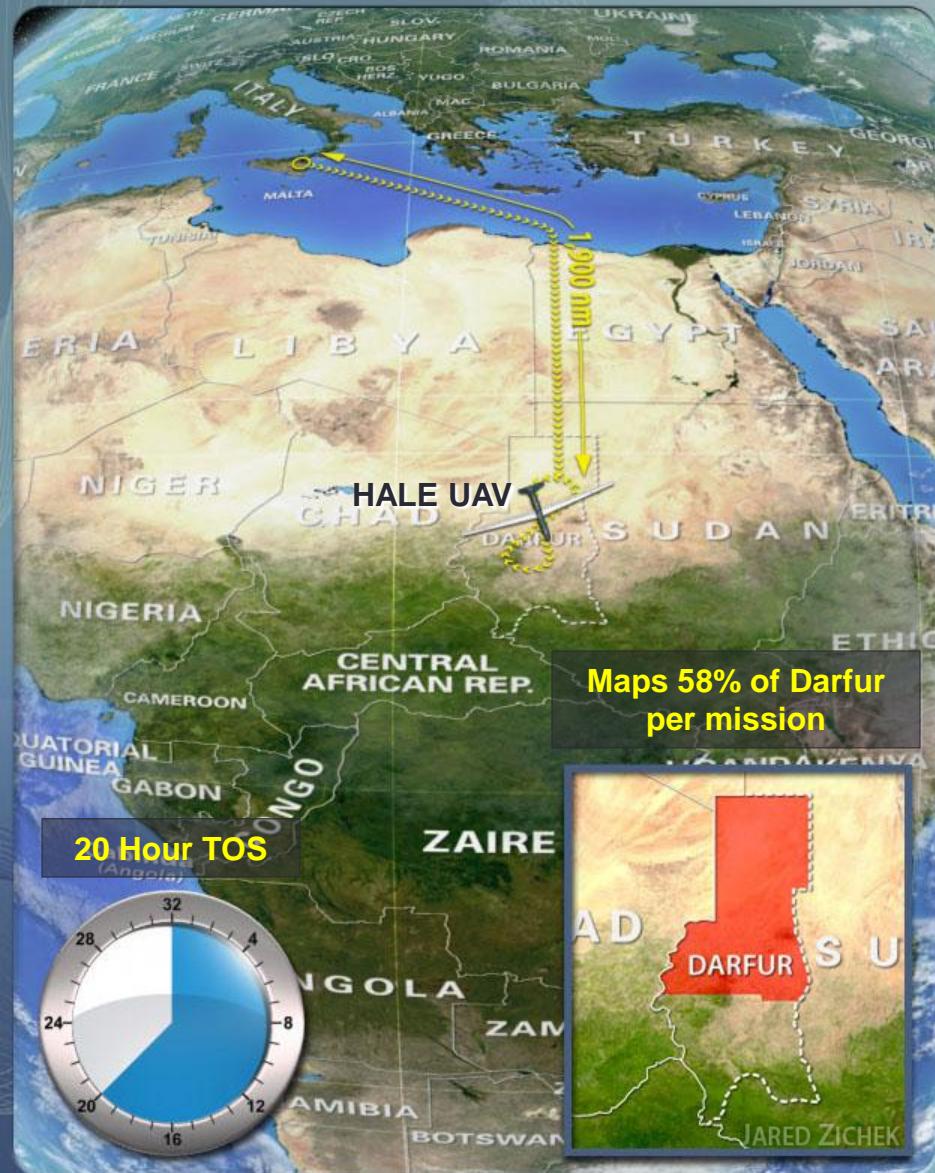
Aircraft	L/D
Transport	7 - 9
Gliders	20 – 30

- Wetted area from Wing loading
- Calculate wingspan
- Taper/sweep = 0
- Estimation of C_L and C_{D0} as usual

Points to Ponder

- Should HALE aircraft be manned?
- Infinite energy sources?

Manned vs Unmanned



JARED ZICHEK

Air to Air Re-fuelling



NASA Helios – Solar HALE Aircraft



NASA Dryden Flight Research Center Photo Collection

<http://www.dfrc.nasa.gov/gallery/photo/index.html>

NASA Photo: ED01-0209-3 Date: July 14, 2001 Photo by: Nick Galante/PMRF

The Helios Prototype flying wing is shown over the Pacific Ocean during its first test flight on solar power from the U.S. Navy's Pacific Missile Range Facility in Hawaii.

QUESTIONS?

