

Milestone Review Flysheet 2017-2018

Institution	Maharishi School	Milestone	PDR			
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Vehicle Properties		Motor Properties	
Total Length (in)	92.4	Motor Brand/Designation	Aerotech K680
Diameter (in)	6.1	Max/Average Thrust (lb.)	180 lbs
Gross Lift Off Weigh (lb.)		Total Impulse (lbf-s)	
Airframe Material(s)	Fiberglass	Mass Before/After Burn (lb.)	
Fin Material and Thickness (in)	Fiberglass, 0.118	Liftoff Thrust (lb.)	140 lbs
Coupler Length/Shoulder Length(s) (in)		Motor Retention Method	

Stability Analysis		Ascent Analysis	
Center of Pressure (in from nose)	72.4 inches	Maximum Velocity (ft/s)	
Center of Gravity (in from nose)	60.2 inches	Maximum Mach Number	1
Static Stability Margin (on pad)	2.01	Maximum Acceleration (ft/s^2)	
Static Stability Margin (at rail exit)	2.01	Predicted Apogee (From Sim.) (ft)	5,283
Thrust-to-Weight Ratio			
Rail Size/Type and Length (in)			
Rail Exit Velocity (ft/s)			

Recovery System Properties		Recovery System Properties				
Drogue Parachute		Main Parachute				
Manufacturer/Model		Manufacturer/Model				
Size/Diameter (in or ft)		Size/Diameter (in or ft)				
Altitude at Deployment (ft)		Altitude at Deployment (ft)				
Velocity at Deployment (ft/s)		Velocity at Deployment (ft/s)				
Terminal Velocity (ft/s)		Terminal Velocity (ft/s)				
Recovery Harness Material		Recovery Harness Material				
Recovery Harness Size/Thickness (in)		Recovery Harness Size/Thickness (in)				
Recovery Harness Length (ft)		Recovery Harness Length (ft)				
Harness/Airframe Interfaces		Harness/Airframe Interfaces				
		Kinetic Energy of Each Section (Ft-lbs)	Section 1	Section 2	Section 3	Section 4

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Recovery Electronics					Recovery Electronics					
Altimeter(s)/Timer(s) (Make/Model)					Rocket Locators (Make/Model)					
Redundancy Plan and Backup Deployment Settings					Transmitting Frequencies (all - vehicle and payload)		***Required by CDR***			
Pad Stay Time (Launch Configuration)					Ejection System Energetics (ex. Black Powder)					
					Energetics Mass - Drogue Chute (grams)		Primary			
							Backup			
					Energetics Mass - Main Chute (grams)		Primary			
							Backup			
					Energetics Masses - Other (grams) - If Applicable		Primary			
							Backup			
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Payload										
Payload 1 (official payload)	Overview									
	Engineering payload will function through the unravelling of a shock cord through the use of a motor/winch when the arduino board and altimeter work									
Payload 2 (non-scored payload)	Overview									

Test Plans, Status, and Results	
Ejection Charge Tests	Planned to do by Nov 30th
Sub-scale Test Flights	Planned to do 2 sub-scale flights by Dec 14th
Full-scale Test Flights	Planned two full-scale flights by March 16th
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Additional Comments	

