

- 2 A satellite S of mass m is in a stable circular orbit at an altitude of $2R$ above the surface of a planet of mass M and radius R , as shown in Fig. 2.1.

Assume the planet has no atmosphere and that all its mass is concentrated at its centre.

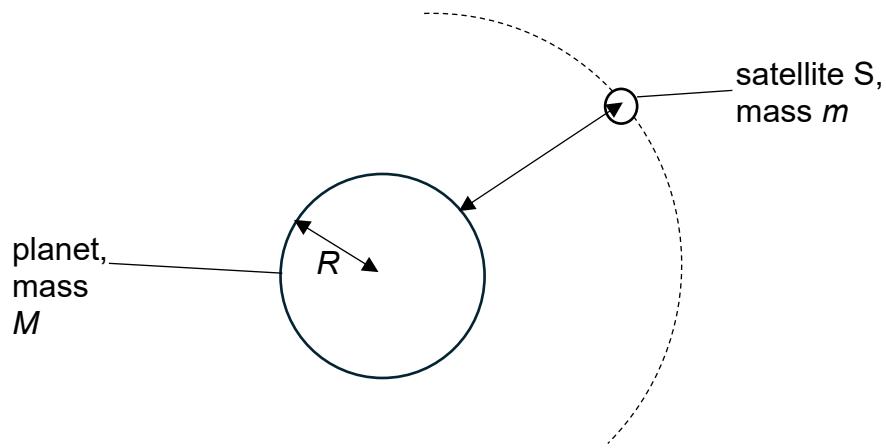


Fig. 2.1

- (a) Show that the kinetic energy E_k of the satellite S in orbit is given by the expression:

$$E_k = \frac{GMm}{6R}$$

where G is the gravitational constant.

[2]

- (b) The planet has mass 4.5×10^{24} kg and radius of 5.5×10^3 km. The satellite has a mass of 1500 kg.

Determine the total energy of satellite S in orbit.

total energy = J [2]

- (c) A second satellite P is launched into orbits from the surface of the same planet with an initial kinetic energy of 2.0×10^{10} J. It rises to a distance of $4R$ from the centre of the planet.

On the axes provided in Fig 2.2, sketch a graph to show how the satellite's orbital kinetic energy varies with distance from the centre of the planet as it moves from R to $4R$.

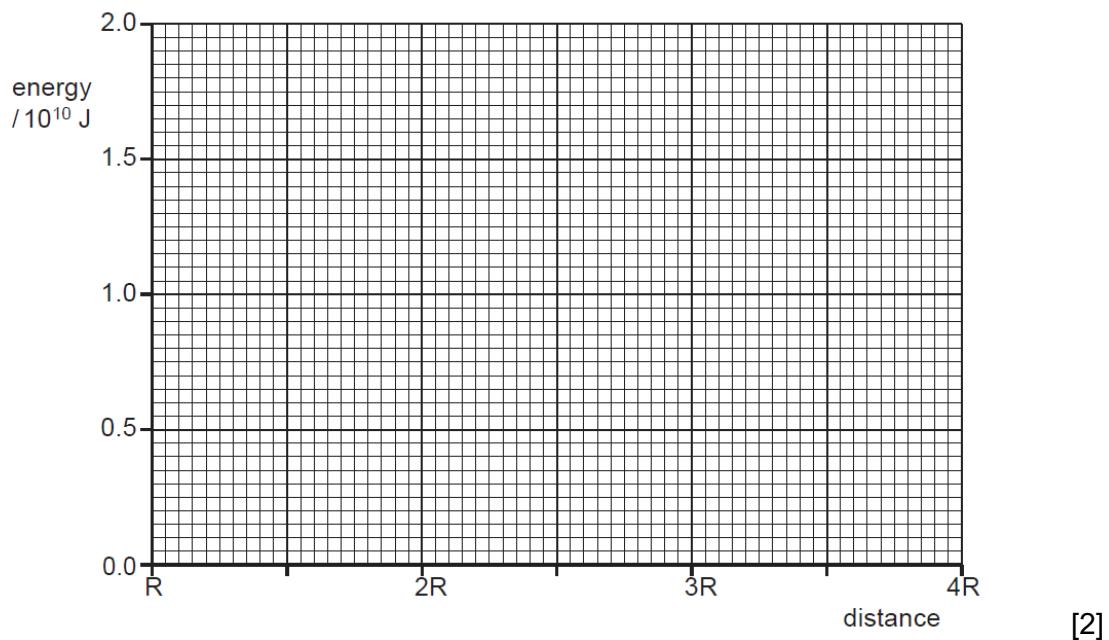


Fig 2.2

- (d) A third satellite Q is to be launched vertically from the surface of the same planet. Determine the minimum speed that satellite Q must be given at the surface to escape the planet's gravitational field.

$$\text{minimum speed} = \dots \text{ m s}^{-1} \quad [2]$$