

# Solutions for SPhO 2023 Theory Paper 2

## Question 1: Projectile Motion

### (a) Stationary Target

The motion of the projectile is described by the following equations:

$$x(t) = (v_0 \cos \theta)t$$

$$y(t) = y_0 + (v_0 \sin \theta)t - \frac{1}{2}gt^2$$

Given values are  $v_0 = 180 \text{ m/s}$ ,  $y_0 = 200 \text{ m}$ ,  $x = 2500 \text{ m}$ , and  $g = 9.81 \text{ m/s}^2$ . The shell hits the target at sea level, so the final vertical position is  $y = 0$ .

From the horizontal motion equation, the time of flight  $t$  is:

$$t = \frac{x}{v_0 \cos \theta} = \frac{2500}{180 \cos \theta}$$

Substitute this into the vertical motion equation:

$$\begin{aligned} 0 &= y_0 + (v_0 \sin \theta) \left( \frac{x}{v_0 \cos \theta} \right) - \frac{1}{2}g \left( \frac{x}{v_0 \cos \theta} \right)^2 \\ 0 &= y_0 + x \tan \theta - \frac{gx^2}{2v_0^2} \sec^2 \theta \end{aligned}$$

Using the identity  $\sec^2 \theta = 1 + \tan^2 \theta$ :

$$0 = y_0 + x \tan \theta - \frac{gx^2}{2v_0^2} (1 + \tan^2 \theta)$$

Let  $u = \tan \theta$ . This gives a quadratic equation in  $u$ :

$$\left( \frac{gx^2}{2v_0^2} \right) u^2 - xu - \left( y_0 - \frac{gx^2}{2v_0^2} \right) = 0$$

Calculating the coefficient  $\frac{gx^2}{2v_0^2}$ :

$$\frac{(9.81)(2500)^2}{2(180)^2} = 946.18 \text{ m}$$

The equation becomes:

$$\begin{aligned} 946.18u^2 - 2500u - (200 - 946.18) &= 0 \\ 946.18u^2 - 2500u + 746.18 &= 0 \end{aligned}$$

Solving for  $u$  using the quadratic formula:

$$u = \frac{2500 \pm \sqrt{(-2500)^2 - 4(946.18)(746.18)}}{2(946.18)} = \frac{2500 \pm \sqrt{3425816}}{1892.36} = \frac{2500 \pm 1850.9}{1892.36}$$

This yields two possible values for  $u = \tan \theta$ :

$$u_1 = \frac{4350.9}{1892.36} \approx 2.30 \implies \theta_1 = \arctan(2.30) \approx 66.5^\circ$$

$$u_2 = \frac{649.1}{1892.36} \approx 0.343 \implies \theta_2 = \arctan(0.343) \approx 18.9^\circ$$

**Answer:** The angle of elevation can be **66.5°** or **18.9°**.

### (b) Moving Target

The target moves with speed  $v_T = 10 \text{ m/s}$ . We use the smaller angle,  $\theta = 18.9^\circ$ . Let the new muzzle speed be  $v'_0$  and the new time of flight be  $t'$ . At time  $t'$ , the horizontal positions must match:

$$(v'_0 \cos \theta)t' = 2500 + v_T t'$$

The vertical position of the shell is:

$$0 = y_0 + (v'_0 \sin \theta)t' - \frac{1}{2}g(t')^2$$

$$0 = y_0 + (v'_0 \cos \theta \tan \theta)t' - \frac{1}{2}g(t')^2$$

Substitute  $(v'_0 \cos \theta)t'$  from the horizontal equation into the vertical one:

$$0 = y_0 + (2500 + v_T t') \tan \theta - \frac{1}{2}g(t')^2$$

$$\frac{1}{2}g(t')^2 - (v_T \tan \theta)t' - (y_0 + 2500 \tan \theta) = 0$$

Substitute the values:  $\tan(18.9^\circ) \approx 0.343$ :

$$\frac{1}{2}(9.81)(t')^2 - (10 \times 0.343)t' - (200 + 2500 \times 0.343) = 0$$

$$4.905(t')^2 - 3.43t' - 1057.5 = 0$$

Solving for  $t'$ :

$$t' = \frac{3.43 \pm \sqrt{(-3.43)^2 - 4(4.905)(-1057.5)}}{2(4.905)} = \frac{3.43 \pm 144.08}{9.81}$$

Since  $t' > 0$ , we take the positive root:  $t' = \frac{147.51}{9.81} \approx 15.04 \text{ s}$ . Now, we find  $v'_0$ :

$$v'_0 = \frac{2500 + v_T t'}{t' \cos \theta} = \frac{2500 + 10(15.04)}{15.04 \cos(18.9^\circ)} = \frac{2650.4}{14.23} \approx 186.2 \text{ m/s}$$

The required change in muzzle speed is  $\Delta v_0 = v'_0 - v_0 = 186.2 - 180 = 6.2 \text{ m/s}$ . **Answer:** The muzzle speed must be increased by approximately **6.2 m/s**.

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## Question 2: Orbital Mechanics and Wave Interference

### (a) Satellites in Orbit

For a satellite in a circular orbit, gravitational force equals centripetal force:

$$\frac{GM_E m}{r^2} = m\omega^2 r \implies \omega = \sqrt{\frac{GM_E}{r^3}}$$

Using  $g = \frac{GM_E}{R_E^2}$ , we get  $\omega = R_E \sqrt{\frac{g}{r^3}}$ . The radii are:  $r_X = R_E + h_X = 6.36 \times 10^6 + 0.5 \times 10^6 = 6.86 \times 10^6 \text{ m}$   $r_Y = R_E + h_Y = 6.36 \times 10^6 + 1.0 \times 10^6 = 7.36 \times 10^6 \text{ m}$  The angular velocities are:

$$\omega_X = (6.36 \times 10^6) \sqrt{\frac{9.81}{(6.86 \times 10^6)^3}} \approx 1.135 \times 10^{-3} \text{ rad/s}$$

$$\omega_Y = (6.36 \times 10^6) \sqrt{\frac{9.81}{(7.36 \times 10^6)^3}} \approx 1.002 \times 10^{-3} \text{ rad/s}$$

The satellites are aligned again when the difference in their angular displacement is  $2\pi$ :

$$(\omega_X - \omega_Y)t = 2\pi$$

$$t = \frac{2\pi}{\omega_X - \omega_Y} = \frac{2\pi}{(1.135 - 1.002) \times 10^{-3}} = \frac{2\pi}{0.133 \times 10^{-3}} \approx 47242 \text{ s}$$

In minutes,  $t = \frac{47242}{60} \approx 787.4 \text{ minutes}$ . **Answer:** The value of  $t$  is approximately **787.4 minutes**.