

14 - CR3BP Continued, Lagrange Points, the Interplanetary Transport Network, and N-Body Integration

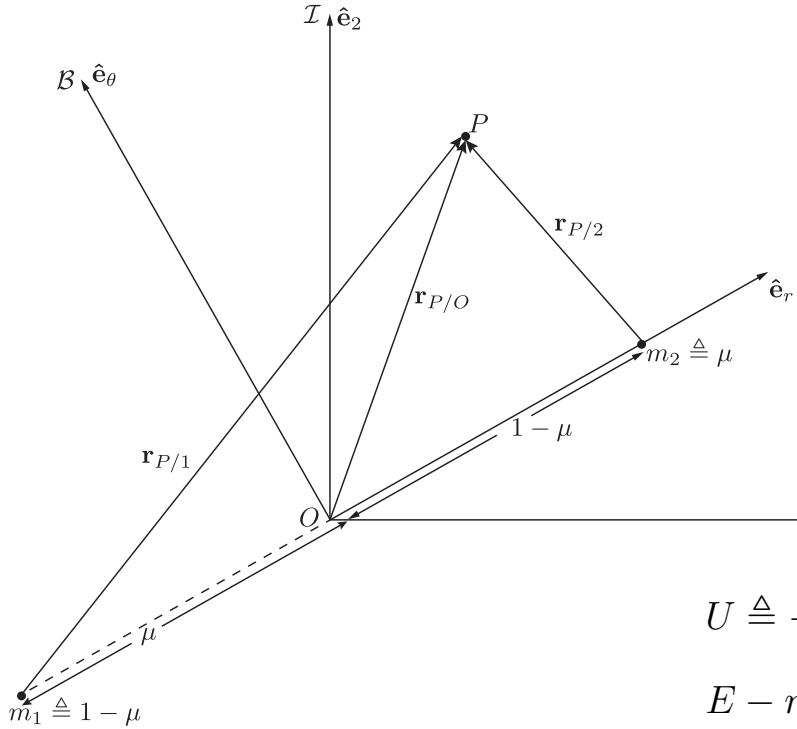
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Recall the Circular Restricted 3-Body Problem (CR3BP)



- ▶ $m_1, m_2 \gg m_p$
- ▶ m_1 and m_2 in mutual circular orbits, unaffected by m_p
- ▶ μ is the mass of the smaller major body (m_2)

$$\ddot{x} = -\frac{\partial U}{\partial x} + 2\dot{y}$$

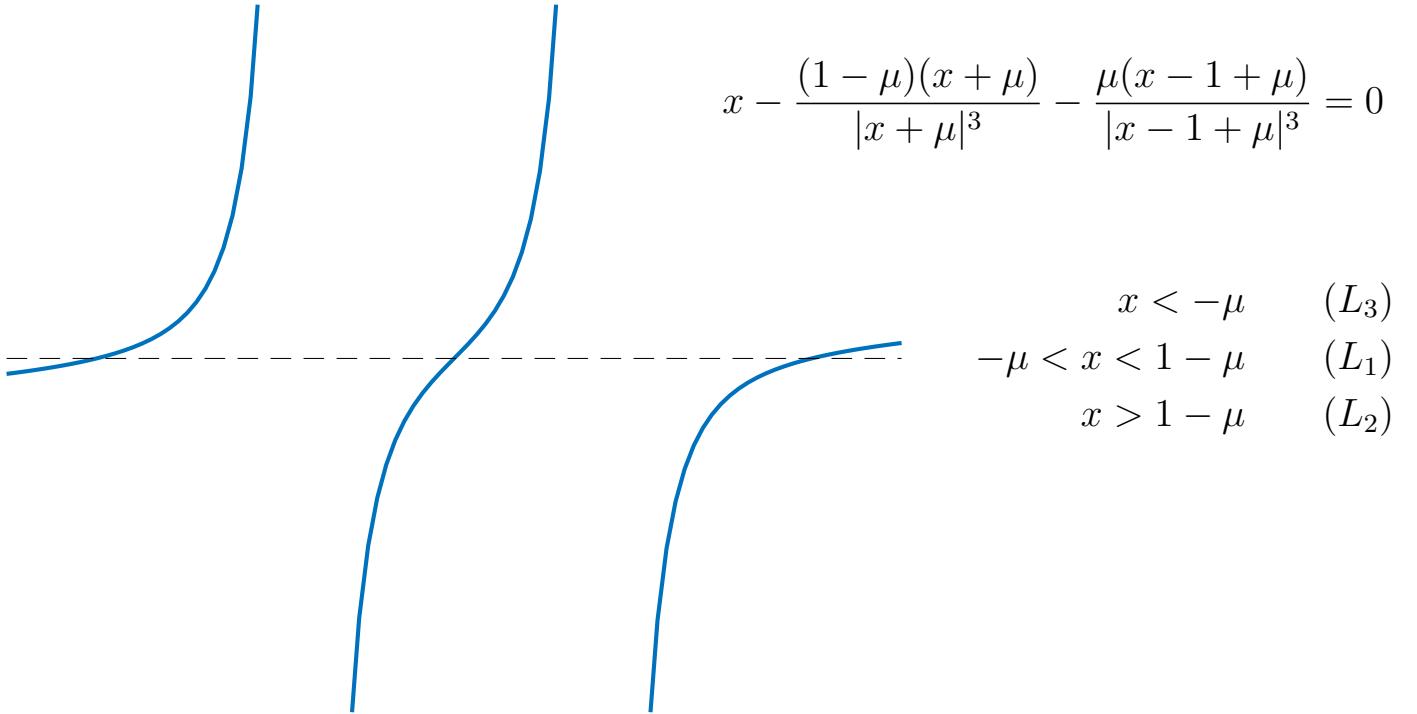
$$\ddot{y} = -\frac{\partial U}{\partial y} - 2\dot{x}$$

$$\ddot{z} = -\frac{\partial U}{\partial z}$$

$$U \triangleq -\frac{1}{2}(x^2 + y^2) - \left(\frac{1-\mu}{r_1} + \frac{\mu}{r_2} \right)$$

$$E - nh \cos(I) = C$$

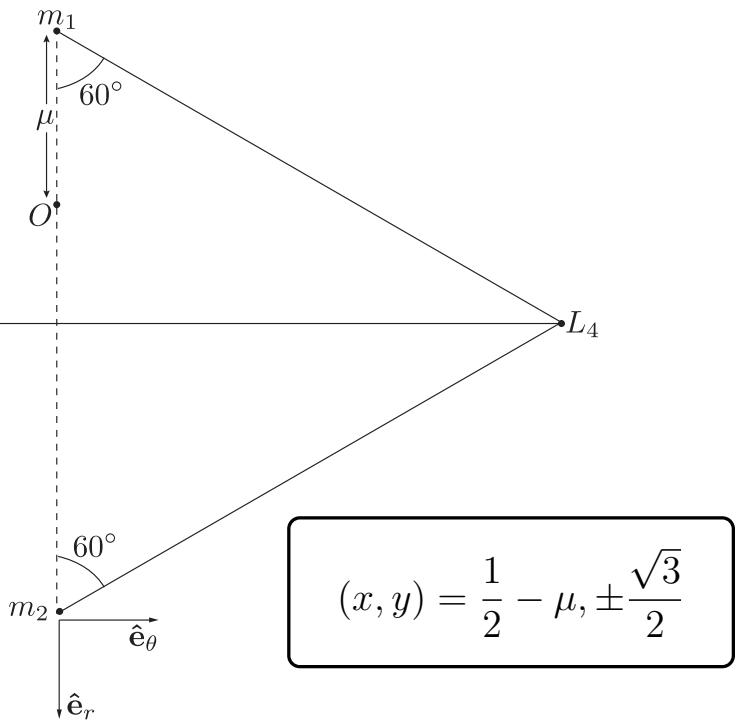
$y = 0$: On-Axis Equilibrium Points



$y \neq 0$: Off-Axis Equilibrium Points

$$\begin{aligned}\frac{1-\mu}{r_1^3} &= \frac{1-\mu}{r_2^3} \Rightarrow r_1 = r_2 \\ 1 - \frac{1-\mu}{r_1^3} - \frac{\mu}{r_2^3} &= 0 \Rightarrow r_1 = r_2 = 1\end{aligned}$$

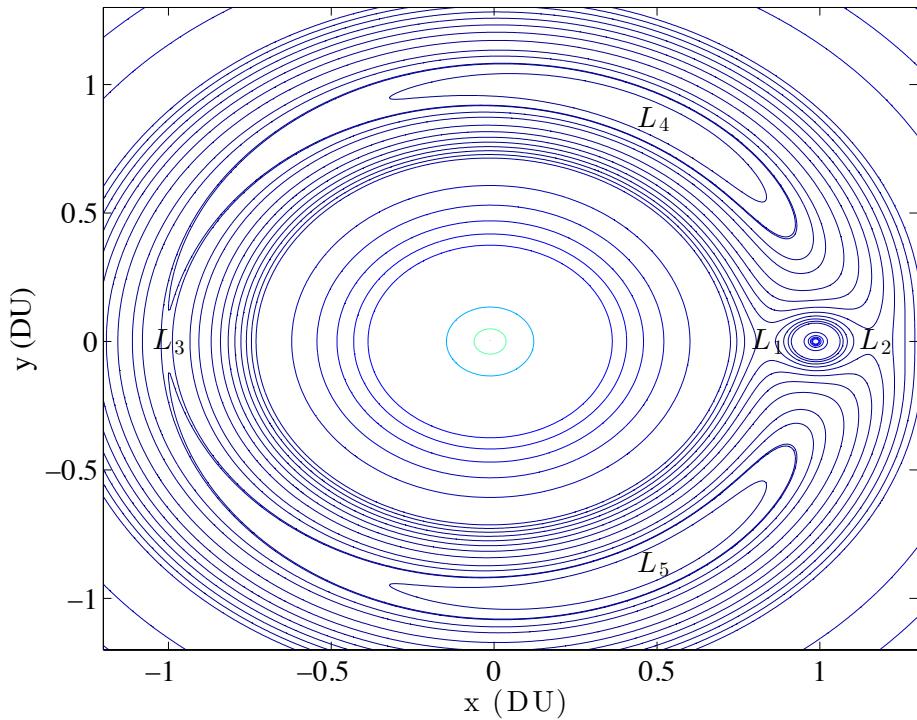
L_5



$$C_4 = C_5$$

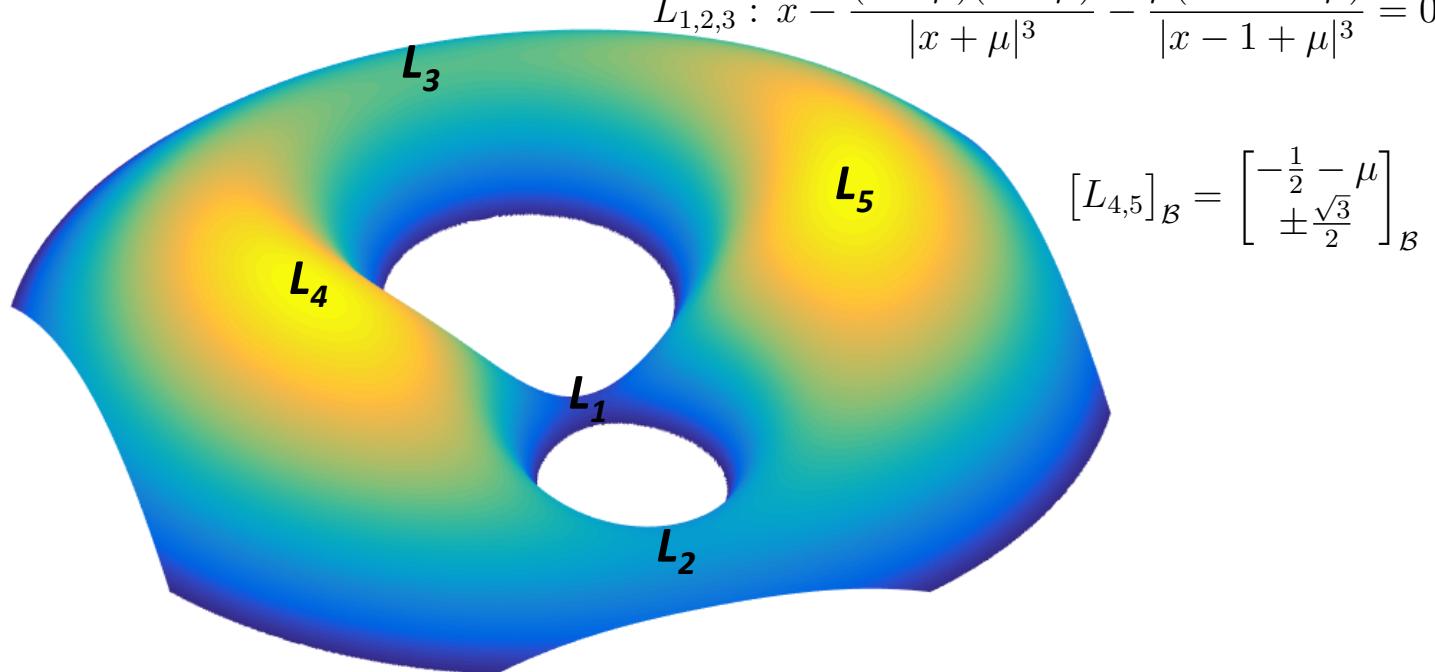
$$\begin{aligned}&= U\left(x = \frac{1}{2} - \mu, y = \pm \frac{\sqrt{3}}{2}, z = 0\right) \\ &= -\frac{1}{2} (3 + \mu - \mu^2)\end{aligned}$$

Earth/Moon Hill Curves ($\mu = 1/81$)



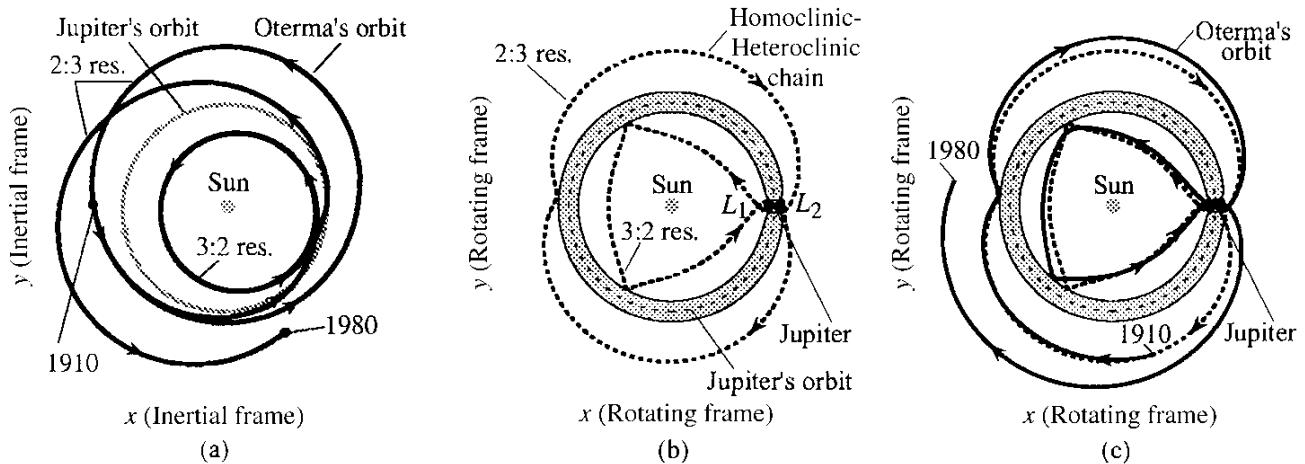
The Lagrange Points

$$L_{1,2,3} : x - \frac{(1-\mu)(x+\mu)}{|x+\mu|^3} - \frac{\mu(x-1+\mu)}{|x-1+\mu|^3} = 0$$



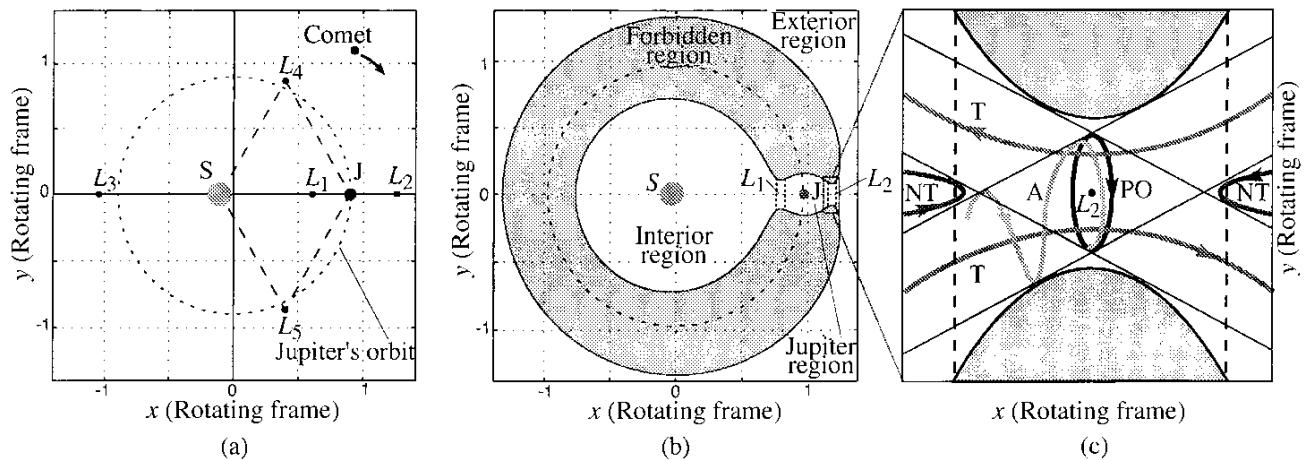
$$[L_{4,5}]_{\mathcal{B}} = \begin{bmatrix} -\frac{1}{2} - \mu \\ \pm \frac{\sqrt{3}}{2} \end{bmatrix}_{\mathcal{B}}$$

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Koon et al. (2001) Fig. 1

Flows about Equilibrium Points



Koon et al. (2001) Fig. 2

Perturbation of $L_{4/5}$ Points

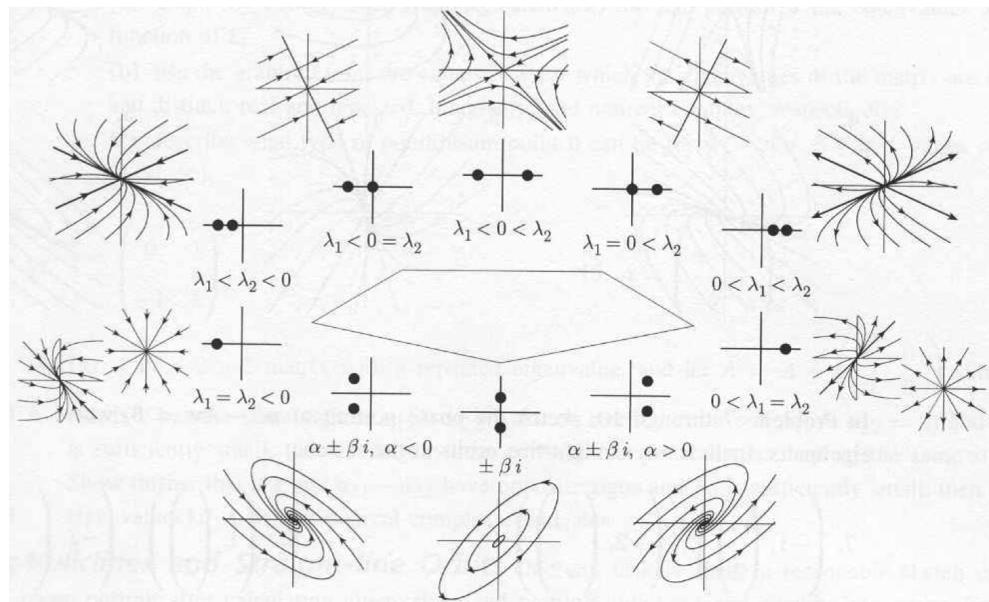
$$\ddot{x} = -\frac{\partial U}{\partial x} + 2\dot{y} \quad \ddot{y} = -\frac{\partial U}{\partial y} - 2\dot{x} \quad \ddot{z} = -\frac{\partial U}{\partial z}$$

$$U \triangleq -\frac{1}{2}(x^2+y^2) - \left(\frac{1-\mu}{r_1} + \frac{\mu}{r_2} \right) \Leftrightarrow \frac{\partial U}{\partial x} = \frac{\partial U}{\partial x} \Big|_{L_i} + \alpha \frac{\partial^2 U}{\partial x^2} \Big|_{L_i} + \beta \frac{\partial^2 U}{\partial x \partial y} \Big|_{L_i} + \dots$$

$$\begin{aligned} \frac{\partial U}{\partial x} \Big|_{L_{4/5}} &\approx -\left(\frac{3\alpha}{4} + \frac{3\sqrt{3}}{4}(1-2\mu)\beta\right) \\ \frac{\partial U}{\partial y} \Big|_{L_{4/5}} &\approx -\left(\frac{9\beta}{4} + \frac{3\sqrt{3}}{4}(1-2\mu)\alpha\right) \end{aligned} \quad \begin{aligned} \ddot{x} - 2\dot{y} &= -\frac{\partial U}{\partial x} = \frac{3\alpha}{4} + \frac{3\sqrt{3}}{4}(1-2\mu)\beta \\ \ddot{y} + 2\dot{x} &= -\frac{\partial U}{\partial y} = \frac{9\beta}{4} + \frac{3\sqrt{3}}{4}(1-2\mu)\alpha \end{aligned}$$

$$\begin{aligned} \alpha &\triangleq Ae^{\lambda t} \\ \beta &\triangleq Be^{\lambda t} \end{aligned} \quad \begin{aligned} A\lambda^2 - 2B\lambda &= \frac{3A}{4} + \frac{3\sqrt{3}}{4}(1-2\mu)B \\ B\lambda^2 + 2A\lambda &= \frac{9B}{4} + \frac{3\sqrt{3}}{4}(1-2\mu)A \end{aligned} \quad \begin{aligned} \lambda^4 + \lambda^2 + \frac{27}{4}\mu(1-\mu) &= 0 \\ \lambda^2 &= -\frac{1}{2} \pm \frac{1}{2}\sqrt{1-27\mu(1-\mu)} \end{aligned}$$

System Behavior About Equilibrium Points



From: Hollis (2002)

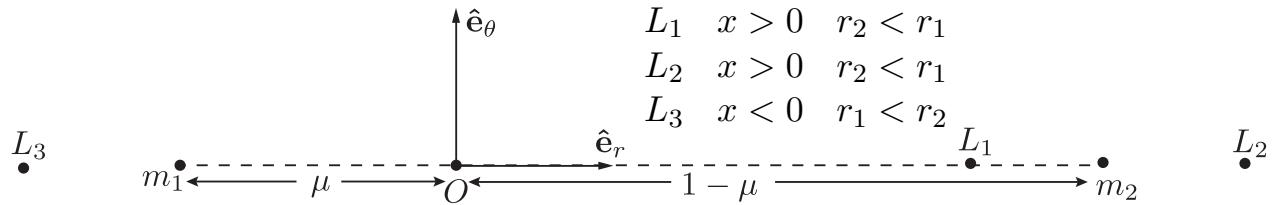
Perturbation of $L_{1\dots 3}$ Points

$$\ddot{\alpha} - 2\dot{\beta} = -\frac{\partial U}{\partial x}\Big|_{L_{1\dots 3}} = \alpha(1 + 2D)$$

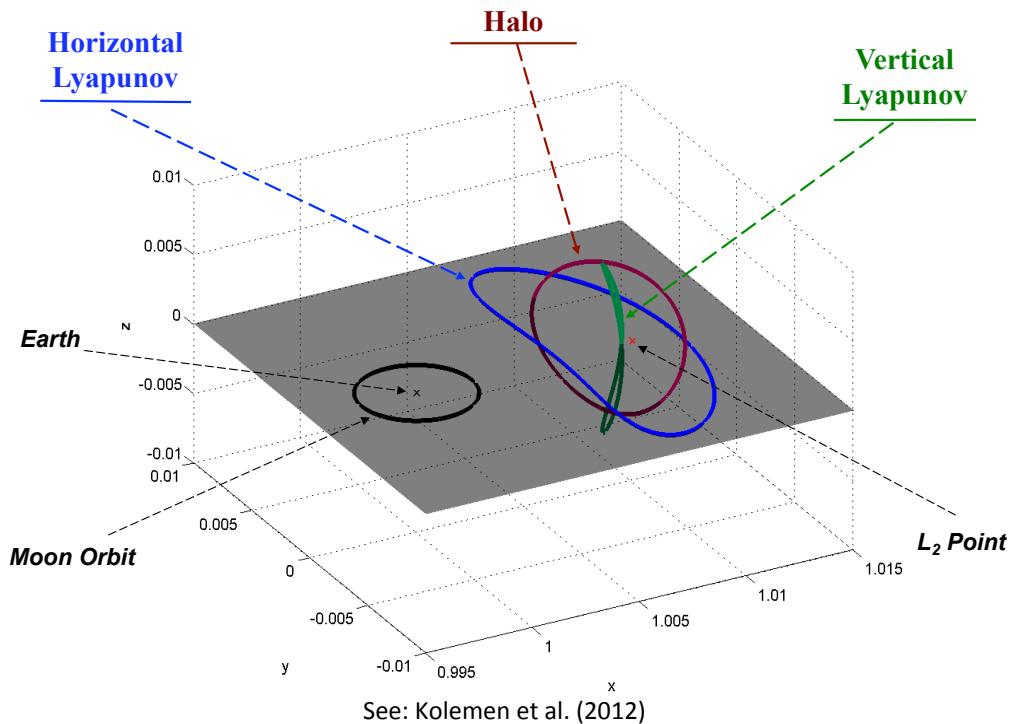
$$\ddot{\beta} + 2\dot{\alpha} = -\frac{\partial U}{\partial y}\Big|_{L_{1\dots 3}} = \beta(1 - D) \quad D \triangleq \frac{1 - \mu}{r_1^3} + \frac{\mu}{r_2^3}$$

$$\left. \begin{array}{l} \alpha \triangleq Ae^{\lambda t} \\ \beta \triangleq Be^{\lambda t} \end{array} \right\} \begin{array}{l} \lambda^4 + (2 - D)\lambda^2 + (1 + 2D)(1 - D) = 0 \\ \lambda^2 = \left(\frac{D}{2} - 1\right) \pm \frac{1}{2}\sqrt{D(9D - 8)} \end{array} \Rightarrow$$

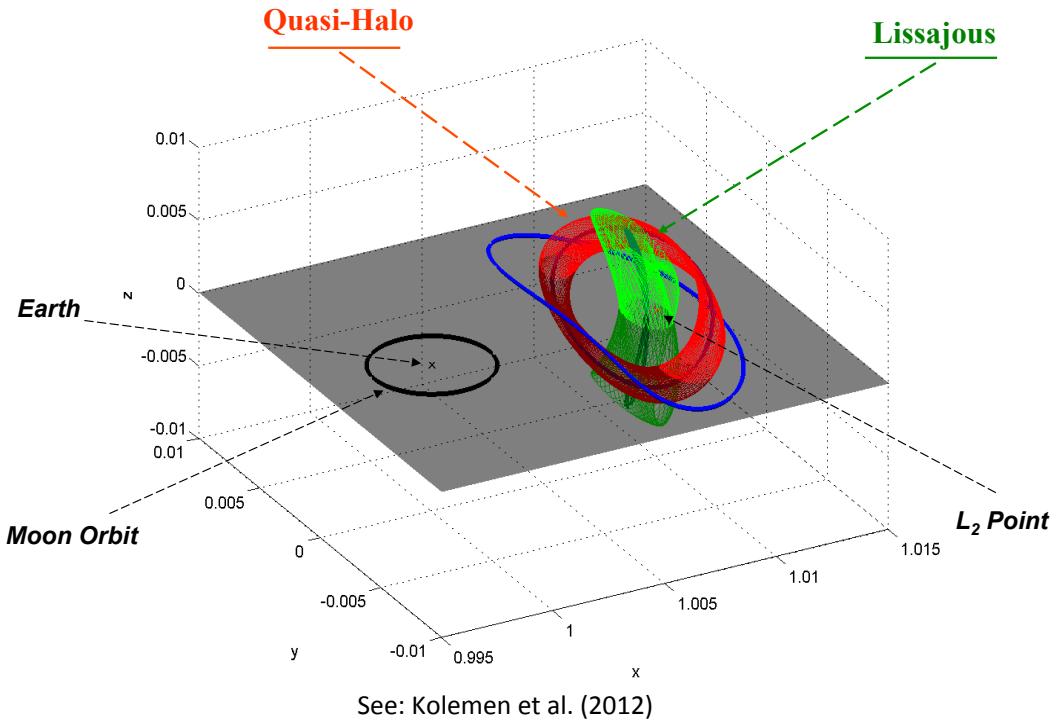
$$x - \frac{(1 - \mu)(x + \mu)}{|x + \mu|^3} - \frac{\mu(x - 1 + \mu)}{|x - 1 + \mu|^3} = 0 \Rightarrow 1 - D = \frac{\mu(1 - \mu)}{x} \left(\frac{1}{r_1^3} - \frac{1}{r_2^3}\right)$$



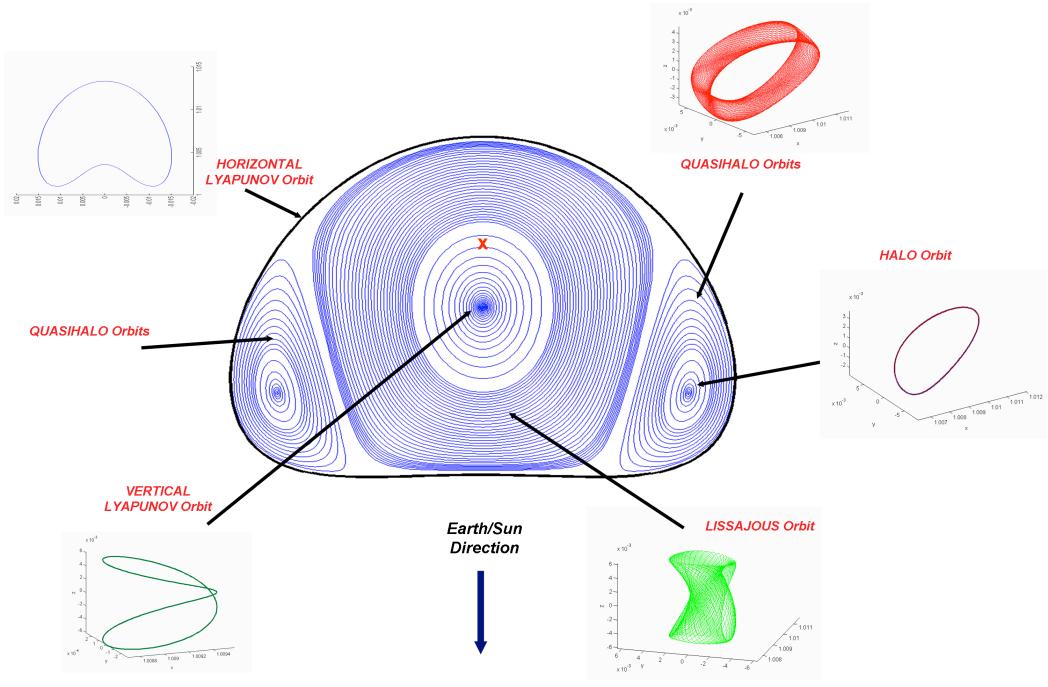
Stable Structures About L_2



Stable Structures About L_2

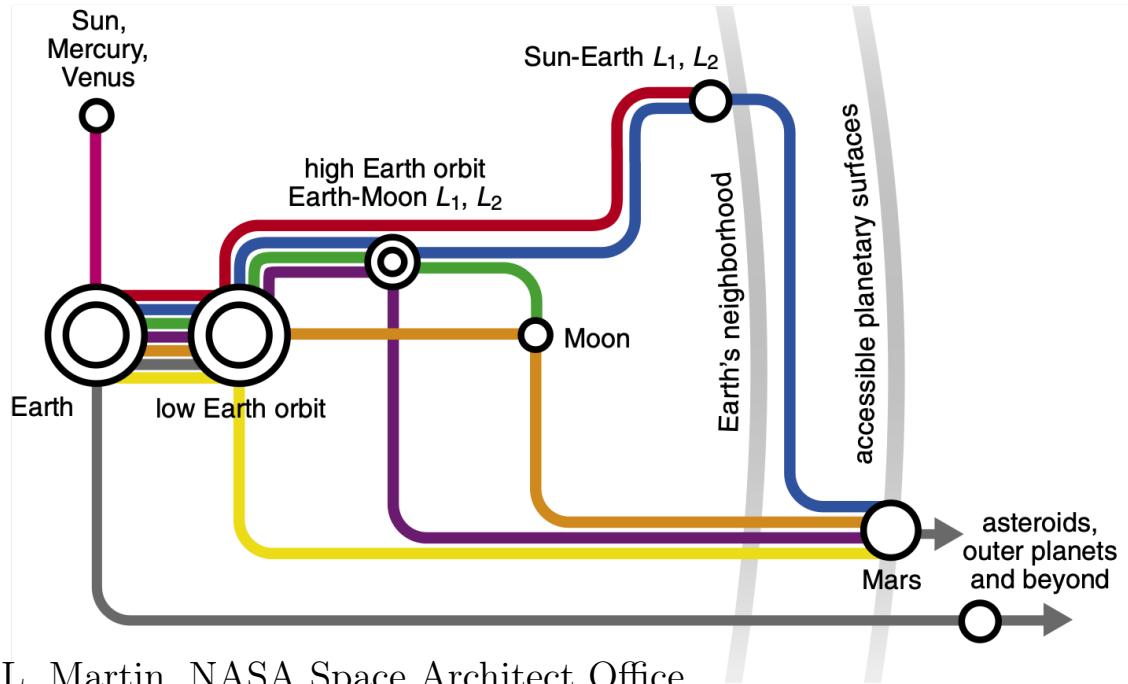


Stable Structures About L_2



See: Kolemen et al. (2012)

The Interplanetary Transport Network



By: Gary L. Martin, NASA Space Architect Office

The Tisserand Criterion

$$\frac{1}{2} \left(\mathcal{B}_{\mathbf{v}_{P/O}} \cdot \underbrace{\mathcal{B}_{\mathbf{v}_{P/O}}}_{\mathcal{B}_{\mathbf{v}_{P/O}} = \mathcal{I}_{\mathbf{v}_{P/O}} - \mathbf{n} \times \mathbf{r}_{P/O}} \right) - \frac{x^2 + y^2}{2} - \left(\frac{1 - \mu}{r_1} + \frac{\mu}{r_2} \right) = C$$

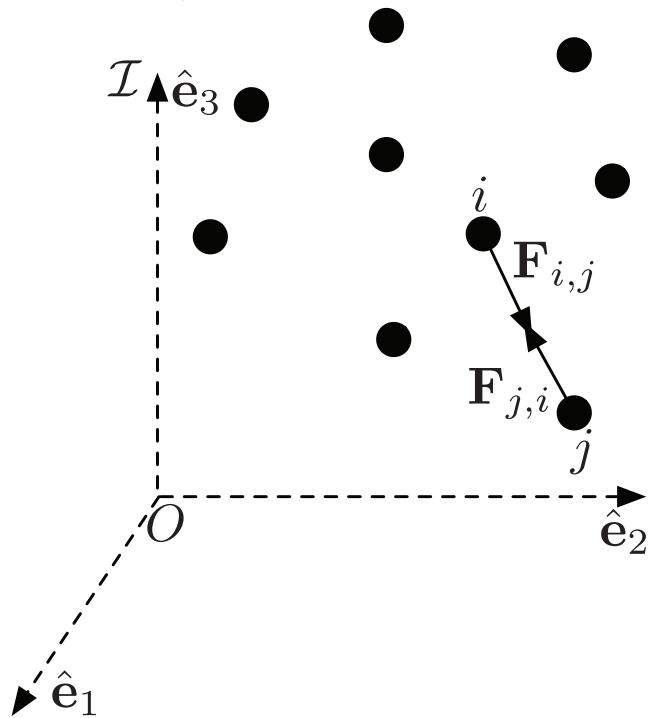
$$C = \frac{1}{2} (\mathcal{I}_{\mathbf{v}_{P/O}} \cdot \mathcal{I}_{\mathbf{v}_{P/O}}) - \hat{\mathbf{e}}_3 \cdot \mathcal{I}_{\mathbf{h}_P} - \left(\frac{1 - \mu}{r_1} + \frac{\mu}{r_2} \right)$$

$$\frac{1}{2} (\mathcal{I}_{\mathbf{v}_{P/1}} \cdot \mathcal{I}_{\mathbf{v}_{P/1}}) = \frac{1}{r_1} - \frac{1}{2a} \quad \hat{\mathbf{e}}_3 \cdot \mathcal{I}_{\mathbf{h}_P} = \sqrt{a(1 - e^2)} \cos(I)$$

$$\frac{1}{a} + 2\sqrt{a(1 - e^2)} \cos(I) + 2\mu \underbrace{\left(\frac{1}{r_2} - \frac{1}{r_1} \right)}_{\text{small}} = -2C$$

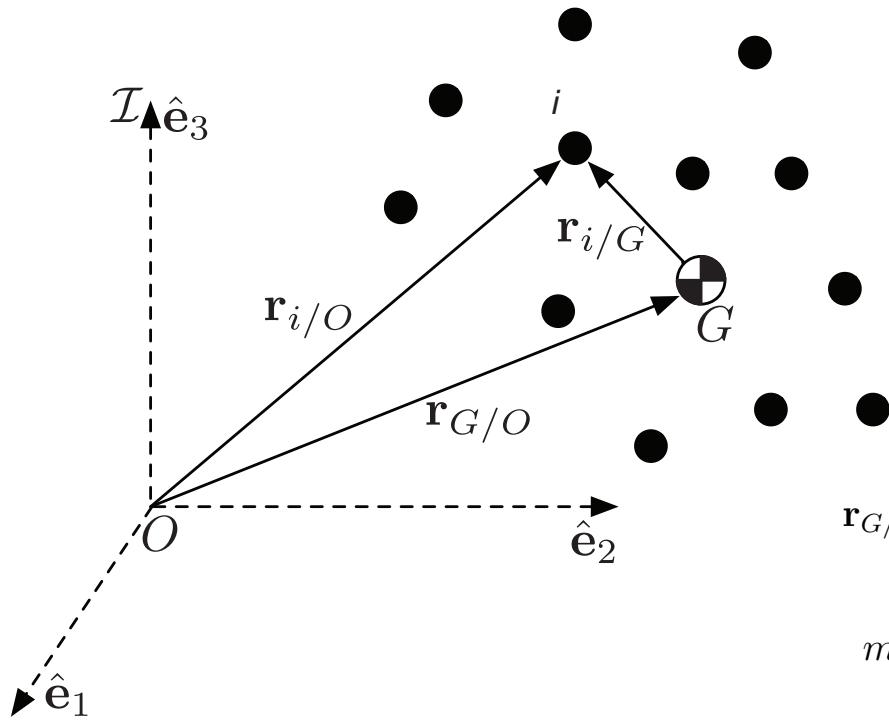
$$T \triangleq \frac{1}{a} + 2\sqrt{a(1 - e^2)} \cos(I) \approx -2C$$

The N-Body Problem



$$\frac{d^2}{dt^2} \mathbf{r}_{i/O} = -G \sum_{\substack{j=1 \\ j \neq i}}^N \frac{m_j}{\|\mathbf{r}_{i/j}\|^3} \mathbf{r}_{i/j}$$

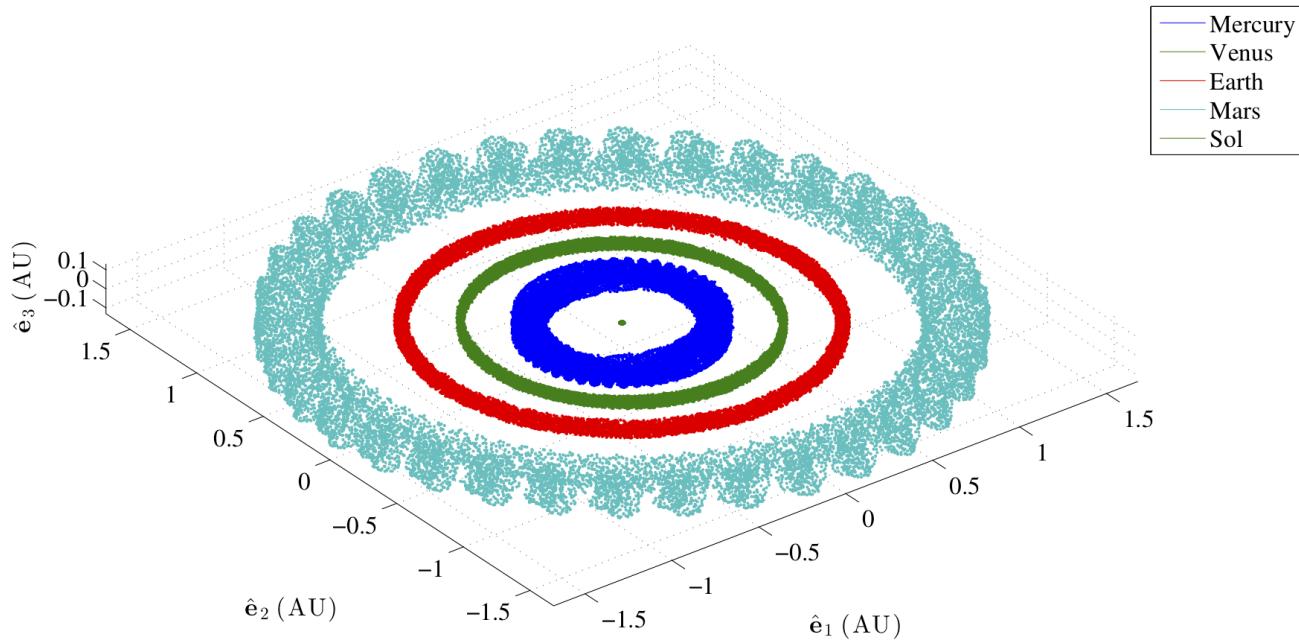
Center of Mass



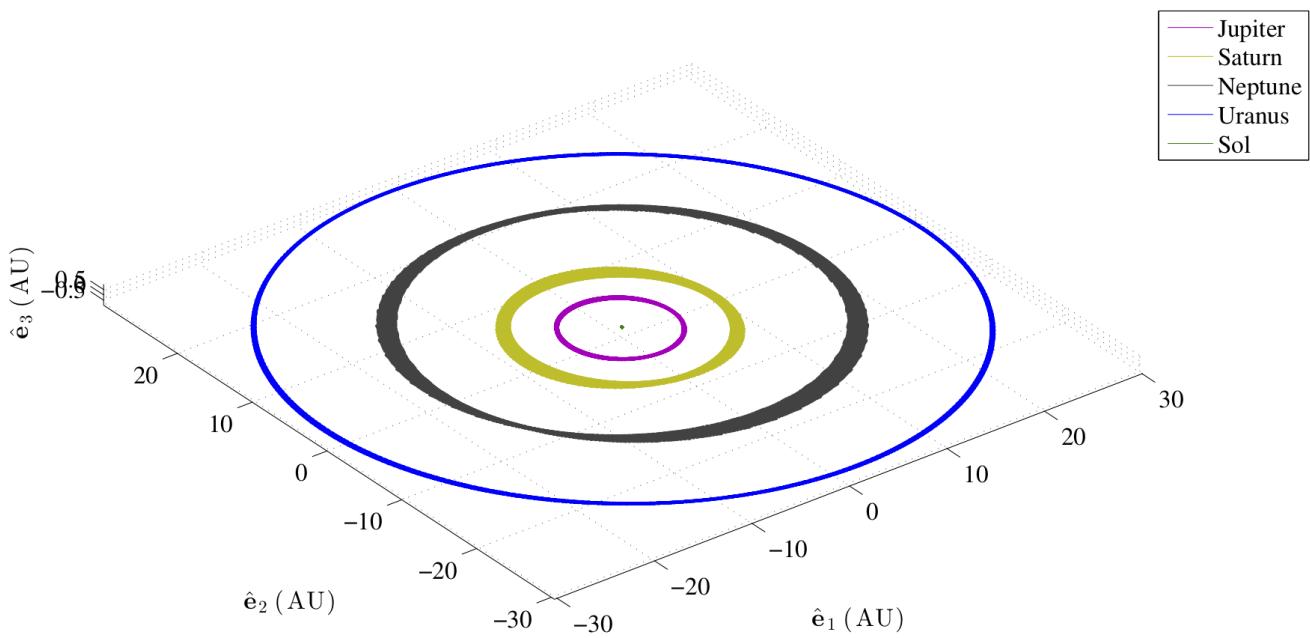
$$\mathbf{r}_{G/O} \triangleq \frac{1}{m_G} \sum_{i=1}^N m_i \mathbf{r}_{i/O}$$

$$m_G \triangleq \sum_{i=1}^N m_i$$

Inner Solar System over 100,000 Years



Outer Solar System over 100,000 Years



Earth Orbital Elements over 100,000 Years

