

NAME: Prachi Singhroha Roll No: 101903545 Group: 2CO21

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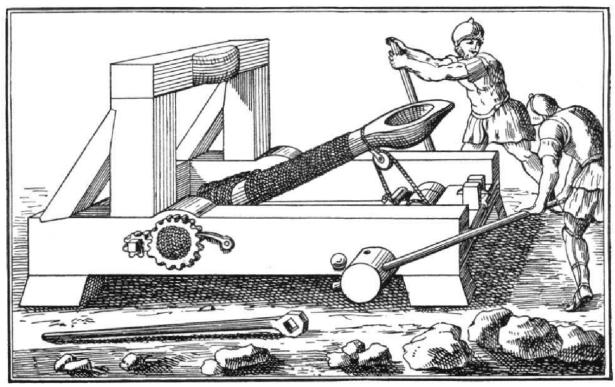


MECHANICAL ENGINEERING DEPARTMENT

Thapar Institute of Engineering and Technology, Patiala

ASSIGNMENT - 4. **DESIGN AGAINST FAILURE UNDER DYNAMIC ACTIONS**

UTA013 Engineering Design Project-I





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ASSIGNMENT - 4. STRUCTURAL ENGINEERING COMPONENT DESIGN AGAINST FAILURE UNDER DYNAMIC ACTIONS

The following tasks have been based on the lecture by Dr. T K Bera on designing against structural failure under *dynamic* loads. Complete the following **individually, copying will be dealt with severely.**

Notes:

- 1. Excel spreadsheet to be created for Q1, Q2 (a) and Q3 and evaluated by end of 2 hour class.
- 2. The print of this word document with **graphs** (with Name and Roll No in text box) and hand written conclusion, name and roll number on every page, stapled together, is to be submitted in next Tutorial class (if it is a holiday, then as instructed). Submit your documents on time. No extensions will be granted.

Despite this list, try and enjoy the assignment and try to think around the subject as much as possible and take from it any tips that you might use with your own Catapult.

When you have built your own mangonel, with your own choice of rotating arm, L2 part (ie spoon: material, diameter and length) and having measured the rotational velocity on impact using the electronic component of this project, then the procedures in Assignments 3 and 4 should allow you to make a reasonable prediction as to whether your chosen arm is likely to fail statically when fully loaded or dynamically when the missile is released. It would clearly be desirable to avoid an unexpected structural failure of any part during the competition!

Marking Scheme: Assignment 4 (10Marks) = 10% Evaluation at end of 2 Hours Tutorial: 5 Marks Evaluation from printout submission: 5 Marks



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TUTORIAL CLASS EVALUATION Q1-Q3. [5 Marks]

Q1. A dowel of 0.006mdiameter (d), a beam span of 0.3m, fails at a static failure load of 47N. Calculate the static failure stressin Excel sheet.

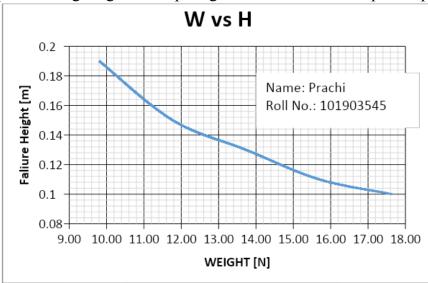
(Evaluated at the end of the Tutorial class)

Measured		Calculated			Actual strength	
				y=d/		
Span	Dia	Failure Force	M=PL/4	2	I=πd^4/64	σ=y*M/I
L mm	d mm	PN	Nmm	mm	mm^4	MPa(N/mm^2)
300.0						
0	6.00	47.00	3525.00	3.00	63.62	166.23

Q2. (a) A series of dynamic tests were performed where weights of different magnitude were dropped onto the dowel span from different heights. The following table was produced;

Weight (N)	Height Failure (m)	Strike Velocity m/s
9.81	0.19	1.93
11.77	0.15	1.72
13.73	0.13	1.60
15.69	0.11	1.47
17.65	0.10	1.40

Insert a plot of weight against drop height to failure for the impact experiment.



(Table and Plot Evaluated at the end of the Tutorial class)

(b) Comment on the shape of the plot and the magnitude of the values to failure when compared to the static failure load.



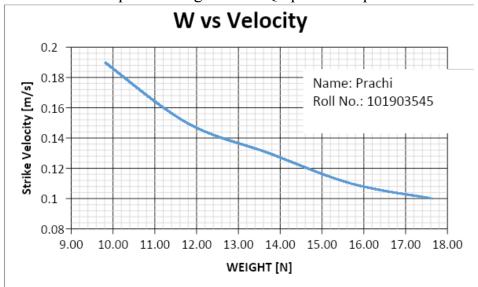
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2.	The state of the s
92)6	With the increase in boads and decrease
	in neight, the graph shows linear
	decrement. This implies the fact that
	when the force or load reaches a
	particular neight upon constant incre-
	ment it hads to breaking of wooden
	static.

Q3. (a) From the tabulated the theoretical velocity on impact for the masses dropped from their respective heights from Q2 produce a plot.



(Plot Evaluated at the end of the Tutorial class)

(b) Comment on this plot in comparison with the plot in Q2 above.

nel	b) both the graphs show decrement upon	
Ψο	incurate in load and decrease in bo	Ch
	neight and velocity but the shape of	0
	graph is different bleause Vis direct porportional to SH	d

Q4. Using *Scenario 4: Case 1* from the lecture 3 and 4supplementary notes, assuming a Dynamic Magnification Factor of 2, calculate the approximate maximum dynamic force that might be applied to the beam of Q1 inducing a stress equal to the static failure stress.



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01)	DMF=2 1 X 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
94)	Fo = 47 N
	DMF = Fd
	AMO I KEEPER MADELINE (MICHAEL)
	n Fd = DMFXfo
	= 47×2
	= 94N

Q5. Using *Scenario 4: Case 2* in the lecture 3 and 4supplementary notes, calculate the mass density, γ, (in units of kg/m³) of the timber dowel beam, the mass per unit length, **m**., (in kg/m) and the load per unit length, ω, (in N/m). The mass of the dowel was measured to be m=4.7g, 6mm diameter and the total length equal to L=0.3m

[3]



Q6. Using this value for \mathbf{m} , and selecting an overhang for the arm of 0.2m (see Figure 3(b) in the lecture notes and slide 7 of lecture), calculate the theoretical deflection of this cantilever of length L_2 ,6mm diameter, under a static point load equivalent to its own weight when in fully cocked state of the Mangonel arm. The value of the Young's modulus of elasticity, E, can be assumed from the lecture notes.

	THE PROPERTY OF THE PROPERTY O				
96)	Sb=PXl2 = mgxLxL3				
	3XEXI 3XEXI				
	where I = nd4 = n (6×10-3)4 = 63.617×10-12				
	64 64				
	$85 = mg \times LXL^{3} = 1.56 \times 10^{-12} \times 9.8 \times (0.2)^{4}$ $3 \times 6 \times I$ $3 \times 10^{10} \times 63.617 \times 10^{-12}$				
	8b=644.2 ×10-6 × 0-2				
	= 128.856 × 10-6				



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Q7. Due to the dynamic effect of a rotating cantilever, assumed equivalent to a drop height of h, calculate the Dynamic Magnification Factor for a variety of realistic impact velocities, using equation (4) in the lecture supplementary notes. You should use here the impact velocities of Q2.

δstatic = 1.28 * 10^-3	Height h (m)	Velocity m/s	DMF
Weight of L2= mass/length x 9.81 x length = 15.67 x 9.81 x 0.2	0.19	1.93	55.50
= 30.738	0.15	1.72	49.42
	0.13	1.60	46.08
	0.11	1.47	42.47
	0.10	1.40	40.54

Q8. Takethe velocity corresponding to the drop height of 0.25m (giving rise to a corresponding DMF) and check that this velocity on impact will not cause the cantilever of L₂=0.2m to fail, taking failure stress from Q1, remembering that the dynamic stress can be approximated to $\sigma_{\text{dynamic}} = \sigma_{\text{static}} x$ DMF, where σ_{static} is from last equation in lecture notes.



NAME: Prachi Singhroha Roll No: 101903545 Group: 2CO21 Pright = 0.25 m. V = J2gh L2 = 0.2 m. = 2.21 m S $= \frac{16 \times 153.58 \times 10^{-3} \times (0.2)^{2}}{3.14 \times (6 \times 10^{-3})^{3}}$ $= \frac{98.29 \times 10^{-3}}{678.24 \times 10^{-9}}$ 0 s = 1.449 x 105 N/m2 DMF= 1+ 1+2h Ss = 0.12 31 × 10-3 DMF= 1+ 1+ 0.5 0.1231×10-3 $= 1 + 1 + 4.0617 \times 10^{3}$ = 1 + 63.7= 64.7 od = DMF xos = 64.7 × 1.449 × 105 = 93.7 × 105 N/m² = 9.37 × 105 N/m² Faliure Stress = 166.2285 × 10° N m2 which is to greater than od,