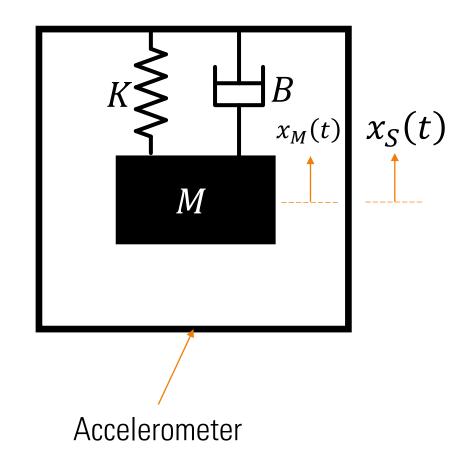
Transducers & Instrumentation

Module 03 - 02

Measuring Movements

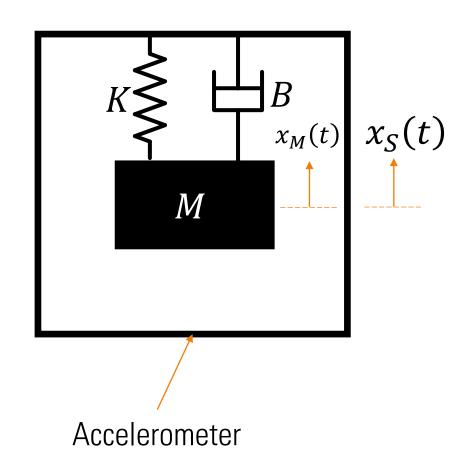


 $x_S(t)$ is the movement of the accelerometer, and $\ddot{x}_S(t)$ is its acceleration.

Acceleration of the accelerometer results in movements of the mass M, $x_M(t)$.

We are interested in the relative movement of the mass within the accelerometer,

$$y(t) = x_S(t) - x_M(t)$$



$$M\ddot{x}_{M}(t) = K \cdot \left(x_{S}(t) - x_{M}(t)\right) + B \cdot \left(\dot{x}_{S}(t) - \dot{x}_{M}(t)\right)$$

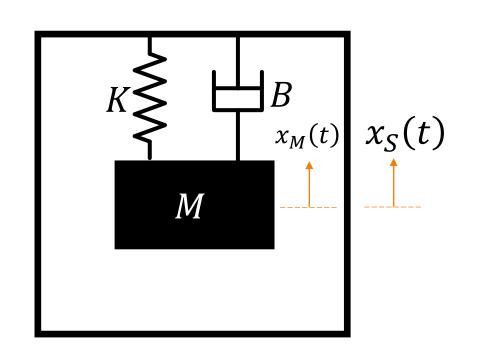
$$Ms^{2}X_{M}(s) = K \cdot \left(X_{S}(s) - X_{M}(s)\right) + sB \cdot \left(X_{S}(s) - X_{M}(s)\right)$$

$$X_{M}(s) = \frac{sB + K}{s^{2}M + sB + K}X_{S}(s)$$

$$\downarrow$$

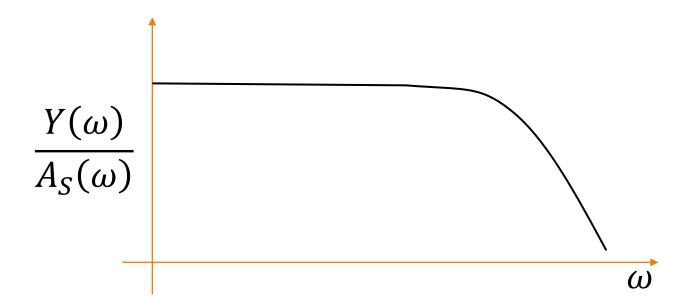
$$Y(s) = X_{S}(s) - X_{M}(s) = \frac{s^{2}M}{s^{2}M + sB + K}X_{S}(s)$$

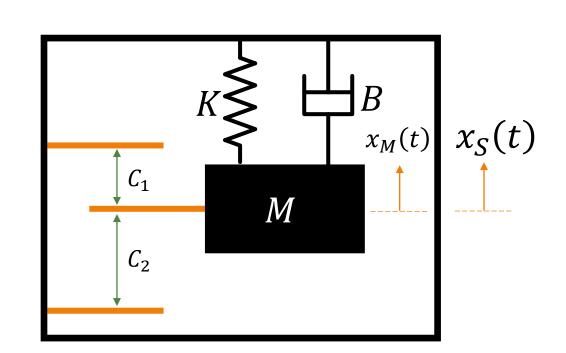
$$Y(s) = \frac{M}{s^2M + sB + K}A_S(s)$$



$$Y(s) = \frac{1}{s^2 + s\frac{B}{M} + \frac{K}{M}} A_S(s)$$

Natural Frequency $\omega_n = \sqrt{\frac{K}{M}}$ Static Sensitivity $= \frac{M}{K}$

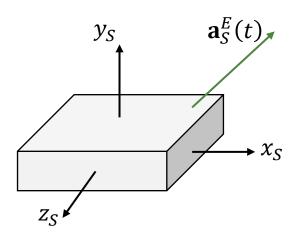




$$Y(s) = \frac{1}{s^2 + s\frac{B}{M} + \frac{K}{M}} A_S(s)$$

Capacitive sensing mechanisms for y(t).

Multi-axis Accelerometers



Signal measured by the accelerometer will:

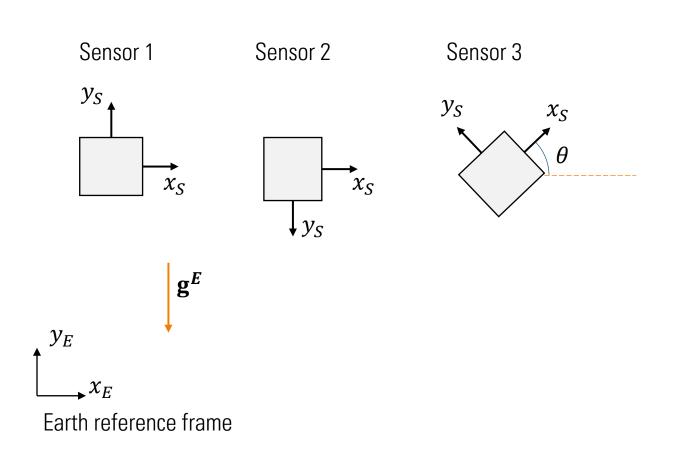
- 1) Contain acceleration due to gravity, and
- 2) Depend on the orientation of the accelerometer with respect to an inertial reference frame of interest.

$$\mathbf{g}^{E}$$
 y_{E}

$$z_{E}$$
Earth reference frame

$$\mathbf{a}_{S}^{S}(t) = \mathbf{R}_{E}^{S}(t) \cdot \left(\mathbf{a}_{S}^{E}(t) - g^{E}\right)$$

Two-axis Accelerometers



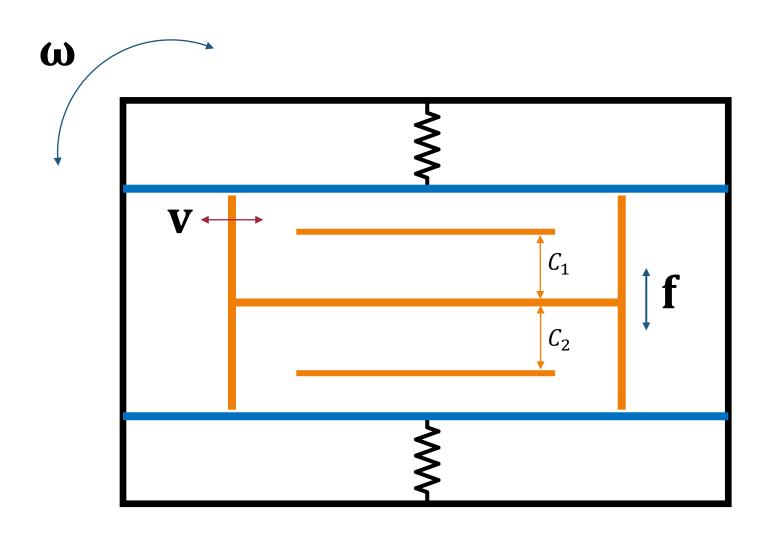
$$\mathbf{g}^{E} = \begin{bmatrix} 0 \\ -9.81 \end{bmatrix} ms^{-2}$$

Sensor 1
$$\mathbf{a}_S^S =$$

Sensor 2
$$\mathbf{a}_S^S =$$

Sensor 3
$$\mathbf{a}_S^S =$$

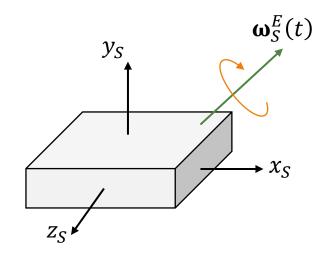
Gyroscopes



 C_1 and C_2 are determined by the Coriolis force.

 ${f v}$ is known, which can be used to compute ${f \omega}$

Multi-axis Gyroscopes



$$\mathbf{\omega}_{S}^{S}(t) = \mathbf{R}_{E}^{S}(t) \cdot \mathbf{\omega}_{S}^{E}(t)$$

