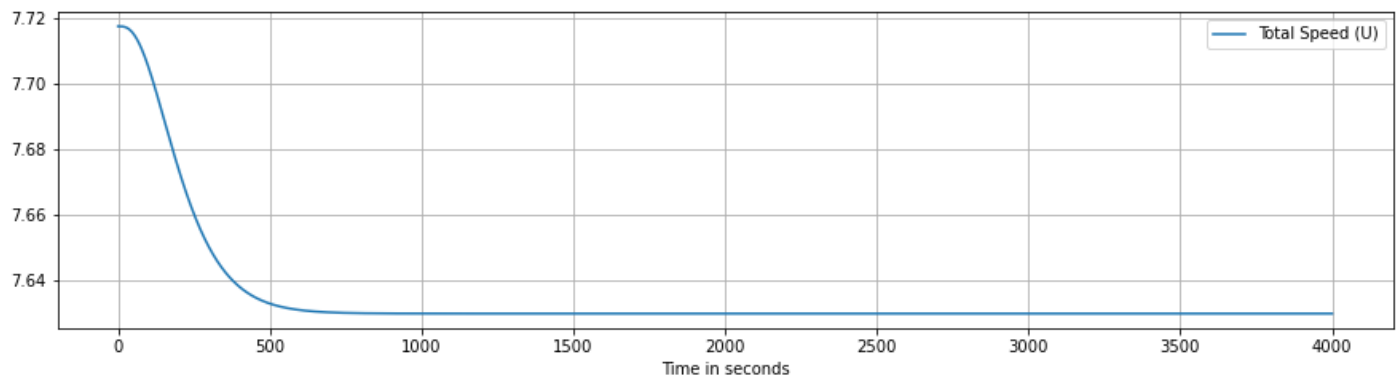
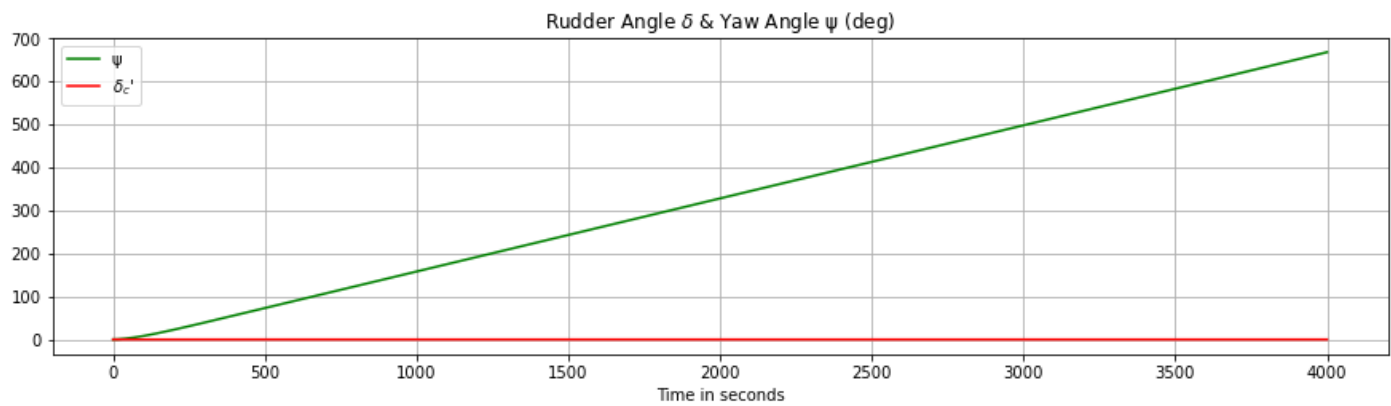
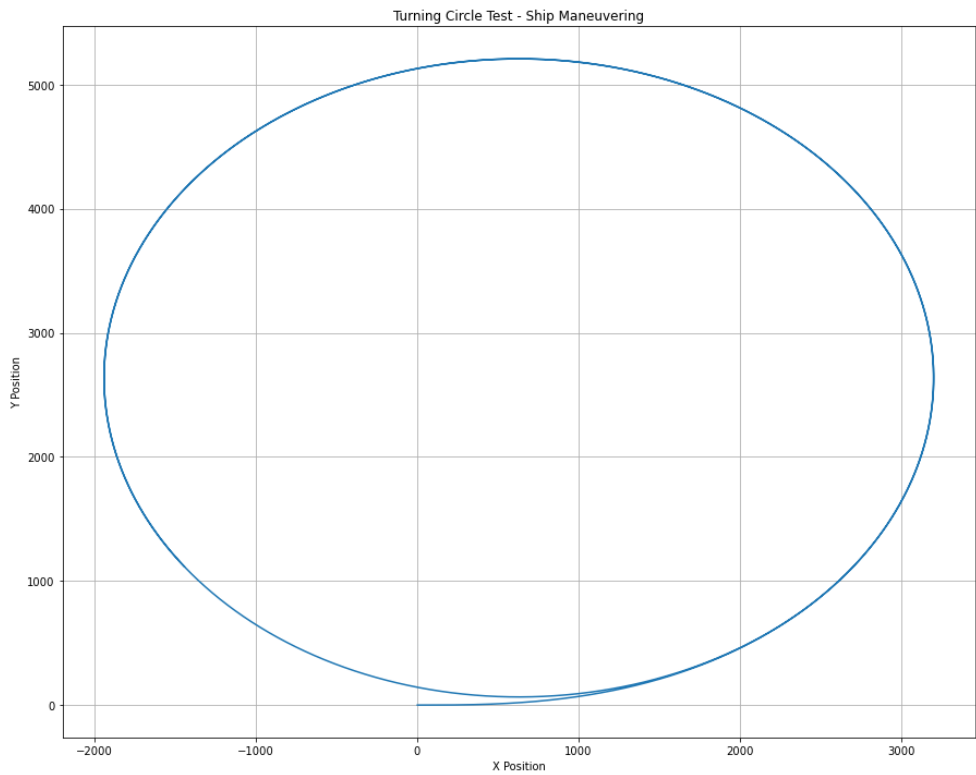
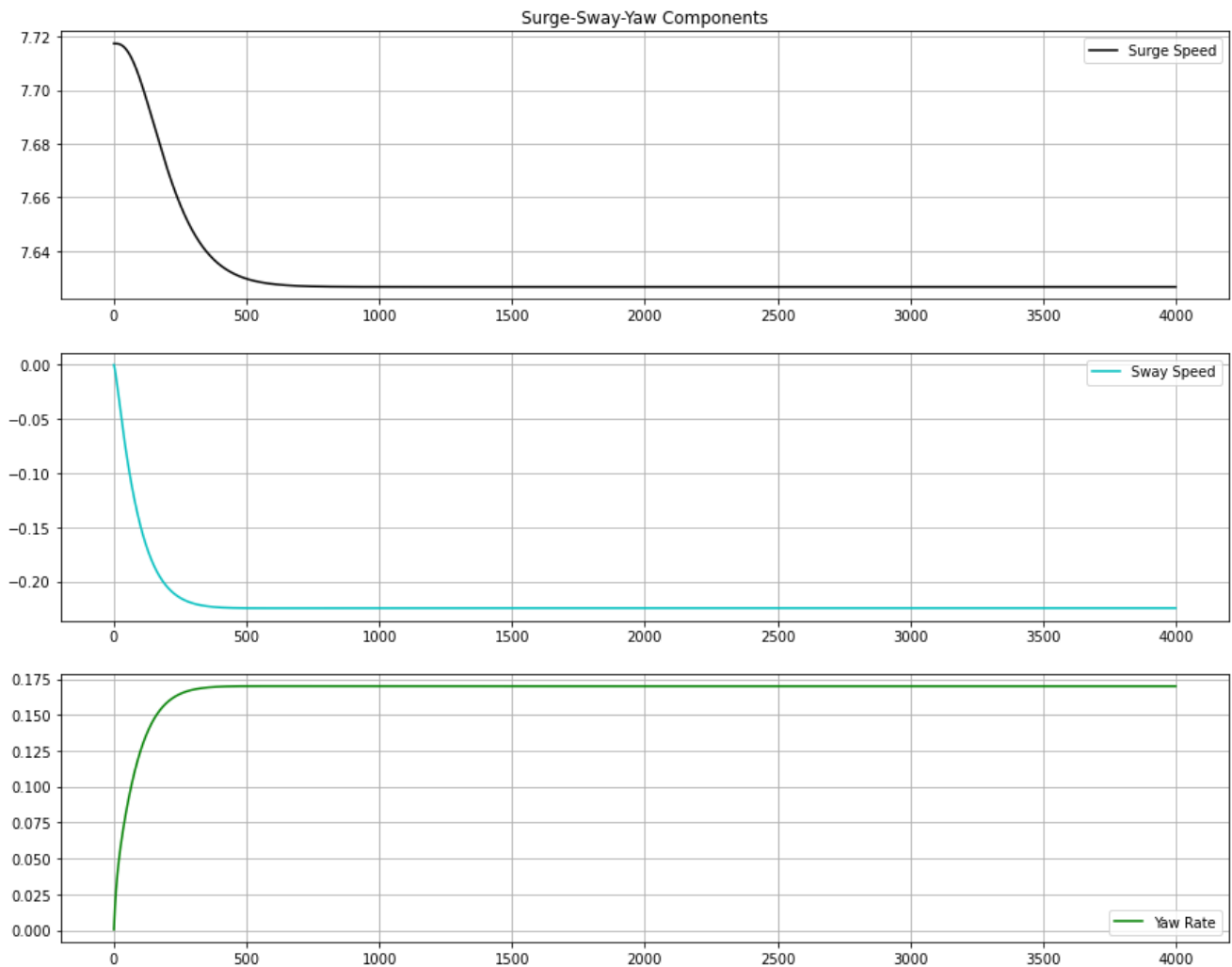


## Data Simulation for Turning Circle Test

Rudder execute (x-coordinate)	: 770.8916672223172
Steady turning radius	: 3535.453460772008
Maximum transfer	: 5207.825225755647
Maximum advance	: 2429.873905270763
Transfer at 90 (deg) heading	: [2560.86926142]
Advance at 90 (deg) heading	: [2428.66435758]





#### Code:

```
#####
### Simulate Data #####
#####

import numpy as np
import matplotlib.pyplot as plt
import Turning_circle

"""
Generates the zigzag maneuver for given different type of ship

Author:   Sivaraman Sivaraj, Suresh Rajendran
Date:     01 December 2020

Reference Author : Thor I. Fossen
Date:           23th July 2001

"""

Req_simulation_time = 4000 #Total simulation time (sec)
t_rudderexecute = 100      #time for rudder is executed at particular angle(sec)
h = 0.1                   #sampling time (sec)

print("Turning Circle test for given ship model is about to start...")
```

```

xt = np.zeros((7,1)) #x = [ u v r x y psi delta ]' (initial values)
ui = 0;

t,u,v,r,x,y,psi,U,delta,D = Turning_circle.activate('mariner',xt,ui,Req_simulation_time,t_
rudderexecute,h)

t_a = np.array(t)
u_a = np.array(u)
v_a = np.array(v)
r_a = np.array(r)
x_a = np.array(x)
y_a = np.array(y)
psi_a = np.array(psi)
U_a = np.array(U)
delta_a = np.array(delta)

t = t.tolist()

v = v.tolist()
r = r.tolist()
x = x.tolist()
y = y.tolist()
psi = psi.tolist()
U = U.tolist()
delta = delta.tolist()

Turning_circle.plot_components_psi_delta_U(t,psi,delta,U)
Turning_circle.plot_components_xy(x,y)

def Plot_simulated_Data1():
    plt.figure(figsize=(15,12))

    plt.subplot(311)
    plt.plot(t[:len(t)-2],7.7175+u[:len(u)-2],'k',label= "Surge Speed")
    plt.grid(b=0.1)
    plt.legend(loc="best")
    plt.title("Surge-Sway-Yaw Components")

    plt.subplot(312)
    plt.plot(t[:len(t)-2],v[:len(v)-2],'c',label= "Sway Speed")
    plt.grid(b=0.1)
    plt.legend(loc="best")

    plt.subplot(313)
    plt.plot(t[:len(t)-2],r[:len(r)-2],'g',label= "Yaw Rate")
    plt.grid(b=0.1)
    plt.legend(loc="best")
    plt.show()

def Plot_simulated_Data2():

```

```

plt.figure(figsize=(15,8))

plt.subplot(211)
plt.grid(b=0.1)
plt.plot(t[:len(t)-2],delta[:len(t)-2],'-r')
plt.plot(t[:len(t)-2],psi[:len(t)-2],'-b')

plt.subplot(212)
plt.plot(t[:len(t)-2],U[:len(U)-2], 'm',label= "Total Speed")
plt.grid()
plt.legend(loc="best")
plt.show()

Plot_simulated_Data1()
Plot_simulated_Data2()

du = u_a[2:]-u_a[1:len(u)-1]
dv = v_a[2:]-v_a[1:len(v)-1]
dr = r_a[2:]-r_a[1:len(r)-1]
output1=[du,dv,dr]

u = u_a + 7.7175
u = u.tolist()
output_turning_circle = np.asarray([t,u,v,r,psi,U,delta])
np.savetxt("5000_sec_turning_circle.csv", output_turning_circle.T, delimiter=",")

#####
##### Turning Circle #####
#####

import numpy as np
import matplotlib.pyplot as plt
import mariner, straight_mariner

def euler_integration(xdot,x,h):
    """
    Integrate a system of ordinary differential equations using
    Euler's 2nd-order method.

    x_next = euler_integration(xdot,x,h)

    Parameters
    -----
    xdot : dx/dt(k) = f(x(k))
    x : x(k)
    h      - step size

    Returns
    -----
    x_next - x(k+1)

    """
    a = np.array(x)
    b = np.array(xdot)

```

```

return a + (h*b)

def activate(ship,x,ui,Req_simulation_time,t_rudderexecute,h,maneuver="ccw"):
    """
    It performs the zig-zag maneuver

    Input Variables
    -----
    ship      : ship model. Compatible with the models under ../gnc/VesselModels/
    x         : initial state vector for ship model
    ui        : given rudder angle
    t_final   : final simulation time

    t_rudderexecute : rudder's time control input is activated

    h           : sampling time

    maneuver : [rudder angle, heading angle]. Default 20-
    20 deg that is: maneuver = [20, 20]
                rudder is changed to maneuver(1) when heading angle is larger than maneuver
    (2)

    Returns
    -----
    t                = time vector
    u,v,r,x,y,psi,U,delta_c = time series
    X_out = as matrix
    D = [advance,transfer,tactical]

    """
    N = round(Req_simulation_time/h)           #Number of samples
    xout = np.zeros((N+1,9))                  #Empty Allocation
    T_var1, T_var2 = 1,1                      #Terminate Variable
    print("Simulating the Maneuver data.....")

    u_ship=ui

    for i in range(N):

        time = (i-1)*h
        if round(float(x[5])*180/np.pi, 3)>= 90 and T_var1 == 1:
            transfer = x[4]    #transfer at 90 deg
            advance = x[3]     #advance at 90 deg
            T_var1 = 0

        if round(float(x[5])*180/np.pi,3) >= 180 and T_var2 == 1:
            tactical=x[4]    %% tactical diameter at 180 deg
            T_var2 = 0

        u_ship = ui;
        if round(time) < t_rudderexecute:
            u_ship = 0

```

```

x_dot,U = mariner.activate(x,u_ship)
x = euler_integration(x_dot,x,h)

#####
temp = list()
temp.append(time)
for j in range(6):
    temp.append(x[j])
temp.append(U[0])
temp.append(u_ship)
xout[i,:] = temp
# print(temp)
#####
# print(i)

#Declassification
t      = xout[:,0]
u      = xout[:,1]
v      = xout[:,2]
r      = xout[:,3]*180/np.pi
x      = xout[:,4]
y      = xout[:,5]
psi    = xout[:,6]*180/np.pi
U      = xout[:,7]
delta_c = xout[:,8]*180/np.pi
D = [advance,transfer,tactical]

Nrudder = round(t_rudderexecute/h)
print('Rudder execute (x-coordinate)           : ',abs(x[Nrudder]))
print('Steady turning radius                   : ',U[Nrudder]/abs(r[Nrudder]*np.pi/180)
)
print('Maximum transfer                         : ',abs(max(abs(y))))
print('Maximum advance                         : ',abs(max(abs(x))-x[Nrudder]))
print('Transfer at 90 (deg) heading            : ',abs(transfer))
print('Advance at 90 (deg) heading             : ',abs(advance-x[Nrudder]))

return t,u,v,r,x,y,psi,U,delta_c,D

def plot_components_xy(x,y):
    plt.figure(figsize=(15,12))
    plt.grid()
    plt.plot(x[:len(x)-2],y[:len(y)-2])
    plt.xlabel("X Position")
    plt.ylabel("Y Position")
    plt.title("Turning Circle Test - Ship Maneuvering")
    plt.show()

def plot_components_psi_delta_U(t,psi,delta_c,U):
    plt.figure(figsize=(15,8))

    plt.subplot(211)
    plt.plot(t[:len(psi)-2],psi[:len(psi)-2],"g",label="ψ")

```

```

plt.plot(t[:len(U)-2],delta_c[:len(U)-2],'r',label = "$ \delta_c $'")
plt.grid()
plt.legend(loc="best")
plt.xlabel("Time in seconds")
plt.title("Rudder Angle $ \delta $ & Yaw Angle  $\psi$  (deg)")

plt.subplot(212)
plt.plot(t[:len(U)-2],U[:len(U)-2], label ="Total Speed (U)")
plt.grid()
plt.legend(loc = "best")
plt.xlabel("Time in seconds")
plt.show()

#####
##### Mariner #####
#####

import numpy as np
import matplotlib.pyplot as plt
import math

def activate(x,ui,U0 = 7.7175):
    """
    Parameters
    -----
    x : [ u v r x y psi delta]
    ui : commanded rudder angle (rad)
    U0 : nominal speed (optionally). Default value is U0 = 7.7175 m/s = 15 knots.

    Returns
    -----
    xdot : Time derivative of the state vector
    U : speed

    Descriptions:
        for the Mariner class vessel L = 160.93 m, where
        u      = pertubed surge velocity about Uo (m/s)
        v      = pertubed sway velocity about zero (m/s)
        r      = pertubed yaw velocity about zero (rad/s)
        x      = position in x-direction (m)
        y      = position in y-direction (m)
        psi    = pertubed yaw angle about zero (rad)
        delta  = actual rudder angle (rad)

    Reference: M.S. Chislett and J. Stroem-Tejsen (1965). Planar Motion Mechanism Tests
    and Full-
    Scale Steering and Maneuvering Predictions for a Mariner Class Vessel,
    Technical Report Hy-5, Hydro- and Aerodynamics Laboratory, Lyngby, Denmark.

```

Author: Trygve Lauvdal

Date: 12th May 1994

Revisions: 19th July 2001 (Thor I. Fossen): added input/output U0 and U, changed order of x-vector

20th July 2001 (Thor I. Fossen): replaced inertia matrix with correct values

11th July 2003 (Thor I. Fossen): max rudder is changed from 30 deg to 40 deg to satisfy IMO regulations for 35 deg rudder execute

"""

# Normalization variables

L = 160.93

u1 = U0+x[0]

U = np.sqrt((u1\*\*2) +(x[1]\*\*2))

#Non-dimensional states and inputs

delta\_c = -ui    #% delta\_c = -ui such that positive delta\_c -> positive r

u     = x[0]/U

v     = x[1]/U

r     = x[2]\*L/U

psi   = x[5]

delta = x[6]

#Parameters, hydrodynamic derivatives and main dimensions

delta\_max = 40                    #max rudder angle       (deg)

Ddelta\_max = 5                    #max rudder derivative (deg/s)

m = 798e-5

Iz = 39.2e-5

xG = -0.023

[Xudot,Xu,Xuu,Xuuu,Xvv,Xrr,Xdd,Xudd,Xrv,Xvd,Xuvd] = [-42e-5,-184e-5,-110e-5,-215e-5,-899e-5,  
18e-5,-95e-5,-190e-5,798e-5,93e-5,93e-5]

[Yvdot,Yrdot,Yv,Yr,Yvvv,Yvvr,Yvu,Yru,  
Yd,Yddd,Yud,Yuud,Yvdd,Yvvd,Y0,Y0u,Y0uu] = [-748e-5,-9.354e-5,-1160e-5,-499e-5,-8078e-5,15356e-5,  
-1160e-5,-499e-5,278e-5,-90e-5,556e-5,278e-5,-4e-5,  
1190e-5,-4e-5,-8e-5,-4e-5]

[Nvdot,Nrdot,Nv,Nr,Nvvv,Nvvr,Nvu,Nru,  
Nd,Nddd,Nud,Nuud,Nvdd,Nvvd,N0,N0u,N0uu] = [4.646e-5,-43.8e-5,-264e-5,-166e-5,1636e-5,  
-5483e-5,-264e-5,-166e-5,-139e-5,45e-5,-278e-5,  
-139e-5, 13e-5,-489e-5,3e-5,6e-5,3e-5]

# Masses and moments of inertia



```

m11 = m-Xudot
m22 = m-Yvdot
m23 = m*xG-Yrdot
m32 = m*xG-Nvdot
m33 = Iz-Nrdot

#Rudder saturation and dynamics
if abs(delta_c) >= (delta_max*np.pi)/180:
    delta_c = np.sign(delta_c)*delta_max*np.pi/180

delta_dot = delta_c - delta

if abs(delta_dot) >= Ddelta_max*np.pi/180:
    delta_dot = np.sign(delta_dot)*Ddelta_max*np.pi/180

# Forces and Moments

X = Xu*u + Xuu*(u**2) + Xuuu*(u**3) + Xvv*(v**2) + Xrr*(r**2) + Xrv*r*v + Xdd*(delta**
2)+\
    Xudd*u*(delta**2) + Xvd*v*delta + Xuvd*u*v*delta

Y = Yv*v + Yr*r + Yvvv*(v**3) + Yvvr*(v**2)*r + Yvu*v*u + Yru*r*u + Yd*delta+\
    Yddd*(delta**3) + Yud*u*delta + Yuud*(u**2)*delta + Yvdd*v*(delta**2) +\
    Yvvd*(v**2)*delta + (Y0 + Y0u*u + Y0uu*(u**2))

N = Nv*v + Nr*r + Nvvv*(v**3) + Nvvr*(v**2)*r + Nvu*v*u + Nru*r*u + Nd*delta +\
    Nddd*(delta**3) + Nud*u*delta + Nuud*(u**2)*delta + Nvdd*v*(delta**2) +\
    Nvvd*(v**2)*delta + (N0 + N0u*u + N0uu*(u**2))

# Dimensional state derivative
detM22 = m22*m33-m23*m32

xdot = [X*((U**2)/L)/m11 ,
        -(-m33*Y+m23*N)*((U**2)/L)/detM22,
        (-m32*Y+m22*N)*((U**2)/(L**2))/detM22,
        (np.cos(psi)*(U0/U+u)-np.sin(psi)*v)*U,
        (np.sin(psi)*(U0/U+u)+np.cos(psi)*v)*U,
        r*(U/L),
        delta_dot]

return xdot,U

# x =np.array([0.8,0.5,0.3,100,100,40,30])
# d = [0.8,0.5,0.3,100,100,40,30]
# ui = -30

# aa,b = activate(x, ui)
# aa1,b1 = activate(d, ui)

# a = np.array(aa)
# print(a.T)
# print(a.shape)
# print(b)

```

```
# print(aa1)
# print(b1)
```