Newtonian gravitation from scratch, for C++ programmers

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Abstract

This paper is a short introduction to Newtonian gravitation. The focus is on some C++ code.

1 Typedefs

In our code, we leave the real number type up to the coder. For instance, we can use long doubles:

```
typedef long double real_type;
```

Otherise, we can use the Boost multiprecision library just as easily:

2 Constants

The following constants will be used in this tutorial:

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```
const real_type pi = 4.0 * atan(1.0);
const real_type G = 6.67430e-11; // Newton's constant
const real_type c = 299792458; // Speed of light in vacuum
const real_type c2 = c * c;
const real_type c3 = c * c * c;
const real_type c4 = c * c * c * c;
const real_type h = 6.62607015e-34; // Planck's constant
const real_type hbar = h / (2.0 * pi);
const real_type k = 1.380649e-23; // Boltzmann's constant
```

3 Brute force: integer field line count

The main idea behind this tutorial is that there is a finite number of field lines extending out from the gravitating body. We use the field line intersection count to signify field strength, from which we can obtain the gradient.

Where r is the receiver radius, R is the distance from the centre of the emitter, β is the get intersecting line count function, and n is the field line count, the gradient is:

$$\alpha = \frac{\beta(R+\epsilon) - \beta(R)}{\epsilon}.$$
 (1)

The gradient strength is:

$$g = \frac{-\alpha}{r^2}. (2)$$

```
cout << "Allocating memory for field lines" << endl;</pre>
vector<vector_3> unit_vectors(n);
for (size_t i = 0; i < n; i++)
        unit_vectors[i] = RandomUnitVector();
        static const size_t output_mod = 10000;
        if (i % output_mod = 0)
                cout << "Getting pseudorandom locations: "</pre>
                << static_cast < float > (i) / n << endl;
}
string filename = "newton.txt";
ofstream out_file(filename.c_str());
out_file << setprecision(30);
const real_type start_distance = 10.0;
const real_type end_distance = 100.0;
const size_t distance_res = 1000;
const real_type distance_step_size =
        (end_distance - start_distance)
        / (distance_{res} - 1);
for (size_t step_index = 0; step_index < distance_res; step_index++)
        const real_type r =
                start_distance +
                step_index * distance_step_size;
        const vector_3 receiver_pos(r, 0, 0);
        const real_type receiver_radius = 1.0;
        const real_type epsilon = 1.0;
        vector_3 receiver_pos_plus = receiver_pos;
        receiver_pos_plus.x += epsilon;
        const long long signed int collision_count_plus =
                get_intersecting_line_count(
                         unit_vectors,
                        receiver_pos_plus,
                        receiver_radius);
        const long long signed int collision_count =
                get_intersecting_line_count(
                        unit_vectors,
                        receiver_pos,
                        receiver_radius);
        const real_type gradient =
```

While this method works, it is both memory and processor intensive. This method is meant to be a stepping stone for the next section. See Fig 1.

4 Heuristic: real field line count, where $R \gg 1$

Rather than allocating gigabytes of RAM to store some unit vectors, we can instead use a heuristic approach to solve the problem from the previous section. This heuristic solution instead uses basic geometry to obtain the intersection count.

Where r is the receiver radius, R is the distance from the centre of the emitter, β is the get intersecting line count function, and n is the field line count, the gradient is:

$$\alpha = \frac{\beta(R+\epsilon) - \beta(R)}{\epsilon}.$$
 (3)

Here we assume that the maximum number of field lines is given by the holographic principle:

$$n = \frac{Akc^3}{4G\hbar \log 2}. (4)$$

The gradient strength is:

$$g = \frac{-\alpha}{r^2} \approx \frac{n}{2R^3}. (5)$$

From this we can get the Newtonian gradient, in terms of either q, n, A, or GM:

$$g_N = \frac{gRc\hbar \log 2}{k2\pi M} = \frac{nc\hbar \log 2}{k4\pi MR^2} = \frac{Ac^4}{16\pi GMR^2} = \frac{GM}{R^2}.$$
 (6)

We will use $g_N = GM/R^2$ in the next section.

```
const real_type sphere_radius)
        const real_type big_area =
                4 * pi * sphere_location.x * sphere_location.x;
        const real_type small_area =
                pi * sphere_radius * sphere_radius;
        const real_type ratio =
                small_area / big_area;
        return n * ratio;
int main(int argc, char** argv)
        const real_type emitter_radius = 1.0;
        const real_type emitter_area =
                4.0 * pi * emitter_radius * emitter_radius;
        // Field line count
        // re: holographic principle:
        const real_type n =
                (k * c3 * emitter_area)
                / (\log (2.0) * 4.0 * G * hbar);
        const real_type emitter_mass = c2 * emitter_radius / (2.0 * G);
        // 2.39545e47 is the 't Hooft-Susskind constant:
        // the number of field lines for a black hole of
        // unit Schwarzschild radius
        //const\ real\_type\ G_{-}=
               (c3 * pi)
                / (log(2.0) * hbar * 2.39545e47);
        const string filename = "newton.txt";
        ofstream out_file(filename.c_str());
        out_file << setprecision (30);
        const real_type start_distance = 10.0;
        const real_type end_distance = 100.0;
        const size_t distance_res = 1000;
        const real_type distance_step_size =
                (end_distance - start_distance)
                / (distance_res - 1);
        for (size_t step_index = 0; step_index < distance_res; step_index++)
                const real_type r =
                        start_distance + step_index * distance_step_size;
```

```
const vector_3 receiver_pos(r, 0, 0);
const real_type receiver_radius = 1.0;
const real_type epsilon = 1.0;
vector_3 receiver_pos_plus = receiver_pos;
receiver_pos_plus.x += epsilon;
// https://en.wikipedia.org/wiki/Directional_derivative
const real_type collision_count_plus =
        get_intersecting_line_count(
                receiver_pos_plus,
                receiver_radius);
const real_type collision_count =
        get_intersecting_line_count(
                receiver_pos,
                receiver_radius);
const real_type gradient =
        (collision_count_plus - collision_count)
        / epsilon;
real_type gradient_strength =
        -gradient
        / (receiver_radius * receiver_radius);
const real_type gradient_strength_ =
        n / (2.0 * pow(receiver_pos.x, 3.0));
const real_type newton_strength =
        n * c * hbar * log(2.0)
        (k * pow(receiver_pos.x, 2.0)
                * emitter_mass * 4.0 * pi);
const real_type newton_strength_ =
        c4 * emitter_area
        / (16.0 * pi * G)
                * pow(receiver_pos.x, 2.0) * emitter_mass);
const real_type newton_strength__ =
        gradient_strength_ * receiver_pos.x
        * c * hbar * log(2)
        / (k * 2 * pi * emitter_mass);
const real_type newton_strength___ =
        G * emitter_mass / pow(receiver_pos.x, 2.0);
//cout \ll newton\_strength\_\_ / newton\_strength\_\_ \ll endl;
cout << "r:" << r << " gradient strength: "
```

This method is faster and less memory intensive when compared to the integer field count method. This method is meant to be a stepping stone for the next section.

For reference, if you know n, and you wish to know the emitter radius from that, then the equation is:

$$r_{emitter} = \sqrt{\frac{nG\hbar \log 2}{kc^3\pi}}. (7)$$

Using this radius, one can ensure that the results from this section match the results of the previous section, where n is relatively small anyway.

5 Application: modeling Mercury's orbit using numerical integration

In essence, the numerical calculation of the Newtonian orbit of Mercury is as follows:

- Place Mercury at the aphelion to start
- Calculate the orbit path by repeatedly taking steps in time

The constant time slice is:

```
\mathbf{const} real type \mathrm{dt} = 10000; \ // \ 2.777777 \ hours
```

The initial conditions are:

```
vector_3 Mercury_pos(0, 69817079000.0, 0); // Aphelion location
vector_3 Mercury_vel(-38860, 0, 0); // Aphelion velocity
```

The orbit code is as follows. Here we use Eq. 6 (e.g. $g_N = GM/R^2$) to calculate the acceleration from Newtonian gravitation:

```
vector_3 Newtonian_acceleration(
    const real_type emitter_mass,
    const vector_3& pos, // Receiver pos
    const real_type G)
{
    // Sun's position is fixed at the origin
    vector_3 grav_dir = vector_3(0, 0, 0) - pos;
    const real_type distance = grav_dir.length();
    grav_dir.normalize();

vector_3 accel = grav_dir * G * emitter_mass / pow(distance, 2.0);
```

```
return accel;
}
```

Here we show the Euler integration, which is extremely simple. The acceleration is calculated, then it is added (e.g. integrated) to the velocity. Once that's done, the velocity is added to the position.

The passage of time is computed whenever the window manager (e.g. OpenGL/GLUT) is not busy drawing or processing input:

```
void idle_func(void)
{
         proceed_Euler(Mercury_pos, Mercury_vel, G, dt);
}
```

On the other hand, rather than using Euler integration, the order-4 symplectic integration does a better job at conserving energy, but at a speed cost:

```
void proceed_symplectic_order_4 (
        vector_3& pos,
        vector_3& vel,
        real_type G,
        real_type dt)
        static const real_type cr2 =
                pow(2.0, 1.0 / 3.0);
        static const real_type c[4] =
                1.0 / (2.0 * (2.0 - cr2)),
                (1.0 - cr2) / (2.0 * (2.0 - cr2)),
                (1.0 - cr2) / (2.0 * (2.0 - cr2)),
                1.0 / (2.0 * (2.0 - cr2))
        };
        static const real_type d[4] =
                1.0 / (2.0 - cr2),
                -cr2 / (2.0 - cr2),
                1.0 / (2.0 - cr2),
```

```
0.0
};
pos += vel * c[0] * dt;
vel += Newtonian_acceleration (
                emitter_mass,
                pos,
                G) * d[0] * dt;
pos += vel * c[1] * dt;
vel += Newtonian_acceleration (
                emitter_mass,
                pos,
                G) * d[1] * dt;
pos += vel * c[2] * dt;
vel += Newtonian_acceleration(
                emitter_mass,
                pos,
                G) * d[2] * dt;
pos += vel * c[3] * dt;
// last element d[3] is always 0
```

See Fig. 2.

6 Final code

A final code, which models the orbit of Mercury, is at: https://github.com/sjhalayka/mercury_orbit_glut

References

- [1] Misner et al. Gravitation. (1970)
- [2] 't Hooft. Dimensional reduction in quantum gravity. (1993)
- [3] Susskind. The World as a Hologram. (1994)
- [4] Fiedler. Fix Your Timestep! (2004)
- [5] Fiedler. Integration Basics. (2004)
- [6] Williams. NASA Mercury Fact Sheet. (2024)

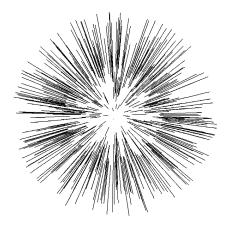


Figure 1: Example of an isotropic emitter. The emitter is spherical. The field line starting locations are placed pseudorandomly on a 2-sphere, and the normals (e.g. field line directions) are calculated using the same sphere.

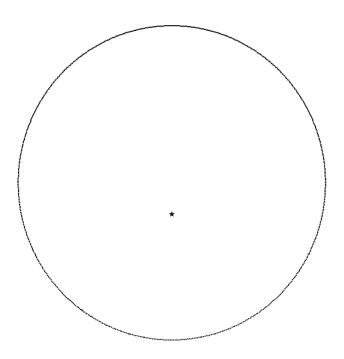


Figure 2: Mercury in orbit around the Sun.