Newtonian gravitation from scratch, for C++ programmers

S. Halayka*

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Abstract

This paper contains a short introduction to Newtonian gravitation. The main focus is on some C++ code.

1 Introduction

In this paper, we cover the following subjects:

- Typedefs that can be used to specify the precision of the floating point variables.
- A custom 3D vector class that is used to encapsulate the data and member functions using object-oriented paradigms.
- Some constants that are used throughout the paper.
- A brute force integer field line intersection count function.
- A heuristic real field line intersection count function.
- An application is demonstrated, where we model Mercury's orbit by using numerical integration. A final code is given.

Some Figs. are provided.

The main goal is to acquaint the coder with the basic mathematics behind Mercury's orbit due to Newtonian gravitation.

^{*}sjhalayka@gmail.com

2 Typedefs

In this tutorial, we leave the real number type up to the coder. For instance, we can use quad-precision long doubles (e.g. 16-byte floating point variables on Ubuntu):

```
typedef long double real_type;
```

On the other hand, we can use the Boost multiprecision library just as easily. Here we can use oct-precision variables (e.g. 32-byte floating point variables on all platforms):

```
#include <boost/multiprecision/cpp_bin_float.hpp>
using namespace boost::multiprecision;
typedef number<
        backends::cpp_bin_float <
                 237,
                backends::digit_base_2,
                void,
                std :: int32_t,
                 -262142, // 2^18 = 262144
                262143 >,
        et_off > cpp_bin_float_oct;
        // 237 significand bits
        // + 18 exponent bits
        // + 1 sign bit =
        // 256 bits (32 bytes)
typedef cpp_bin_float_oct real_type;
```

3 Custom 3D vector class

The tutorial also makes use of a 3D vector class:

```
class vector_3
{
    public:
        real_type x, y, z;

        // Overloaded operators go here

        real_type dot(const vector_3& rhs) const
        {
            return x*rhs.x + y*rhs.y + z*rhs.z;
        }

        real_type self_dot(void) const
        {
            return x*x + y*y + z*z;
        }

        real_type length(void) const
        {
            real_type length(void) const
        }
        }
}
```

```
return sqrt(self_dot());
}

vector_3& normalize(void)
{
    const real_type len = length();

    if(len != 0)
    {
        x /= len;
        y /= len;
        z /= len;
        z /= len;
    }

return *this;
}
```

4 Constants

The following constants will be used in this tutorial:

```
const real_type pi = 4.0 * atan(1.0);
const real_type G = 6.67430e-11; // Newton's constant
const real_type c = 299792458; // Speed of light in vacuum
const real_type c2 = c * c;
const real_type c3 = c * c * c;
const real_type c4 = c * c * c * c;
const real_type h = 6.62607015e-34; // Planck's constant
const real_type hbar = h / (2.0 * pi);
const real_type k = 1.380649e-23; // Boltzmann's constant
```

5 Brute force: integer field line count

The main idea behind this tutorial is that there is a finite number of field lines extending out from a gravitating body. See Fig. 1. We use the field line intersection count to signify field strength, from which we can obtain the gradient.

Where r is the receiver radius, R is the distance from the centre of the emitter, β is the get intersecting line count function, and n is the field line count, the gradient (e.g. directional derivative) is:

$$\alpha = \frac{\beta(R+\epsilon) - \beta(R)}{\epsilon}.$$
 (1)

The gradient strength is:

$$g = \frac{-\alpha}{r^2}. (2)$$

```
long long unsigned int get_intersecting_line_count(
    const vector<vector_3>& unit_vectors,
    const vector_3& sphere_location,
```

```
const real_type sphere_radius)
        long long unsigned int count = 0;
        // Get cross-section edge direction
        vector_3 cross_section_edge_dir(sphere_location.x, sphere_radius, 0);
        cross_section_edge_dir.normalize();
        // Get receiver direction
        vector_3 receiver_dir(sphere_location.x, 0, 0);
        receiver_dir.normalize();
        // The minimum threshold for intersection
        const real_type min_dot = cross_section_edge_dir.dot(receiver_dir);
        for (size_t i = 0; i < unit_vectors.size(); i++)
                if (unit_vectors[i].dot(receiver_dir) >= min_dot)
                        count++; // Intersection occurred
        return count;
int main(int argc, char** argv)
        // Field line count
        const size_t = 1000000000;
        cout << "Allocating memory for field lines" << endl;
        vector<vector_3> unit_vectors(n);
        for (size_t i = 0; i < n; i++)
                unit_vectors[i] = pseudorandom_unit_vector();
                static const size_t output_mod = 10000;
                if (i % output_mod = 0)
                        cout << "Getting pseudorandom locations: "</pre>
                        << static_cast<float>(i) / n << endl;</pre>
        }
        string filename = "newton.txt";
        ofstream out_file(filename.c_str());
        out_file << setprecision(30);
        const real_type start_distance = 10.0;
        const real_type end_distance = 100.0;
        const size_t distance_res = 1000;
        const real_type distance_step_size =
                (end_distance - start_distance)
                / (distance_{res} - 1);
        for (size_t step_index = 0; step_index < distance_res; step_index++)
```

```
const real_type r =
                start_distance +
                step_index * distance_step_size;
        const vector_3 receiver_pos(r, 0, 0);
        const real_type receiver_radius = 1.0;
        const real_type epsilon = 1.0;
        vector_3 receiver_pos_plus = receiver_pos;
        receiver_pos_plus.x += epsilon;
        const long long signed int collision_count_plus =
                get_intersecting_line_count(
                         unit_vectors,
                         receiver_pos_plus,
                         receiver_radius);
        const long long signed int collision_count =
                get_intersecting_line_count(
                         unit_vectors,
                         receiver_pos,
                         receiver_radius);
        const real_type gradient =
                static_cast < real_type >
                (collision_count_plus - collision_count)
                / epsilon;
        const real_type gradient_strength =
                -gradient
                / (receiver_radius * receiver_radius);
        cout << "r: " << r << " gradient strength: "
        << gradient_strength << endl;</pre>
        out_file << r << " " << gradient_strength << endl;
}
out_file.close();
return 0;
```

While this method works, it is both memory and processor intensive. This method is meant to be a stepping stone for the next section.

6 Heuristic: real field line count

Rather than allocating gigabytes of RAM to store some unit vectors, we can instead use a heuristic approach to solve the problem from the previous section. This heuristic solution

instead uses basic geometry to obtain the intersection count.

Where r is the receiver radius, R is the distance from the centre of the emitter, β is the get intersecting line count function, and n is the field line count, the gradient is:

$$\alpha = \frac{\beta(R+\epsilon) - \beta(R)}{\epsilon}.$$
 (3)

Here we assume that the number of field lines is given by the holographic principle:

$$n = \frac{Akc^3}{4G\hbar \log 2}. (4)$$

The gradient strength is:

$$g = \frac{-\alpha}{r^2} \approx \frac{n}{2R^3}. (5)$$

From this we can get the Newtonian gradient, in terms of either n, g, A, or GM:

$$g_N = \frac{nc\hbar \log 2}{k4\pi M R^2} = \frac{gRc\hbar \log 2}{k2\pi M} = \frac{Ac^4}{16\pi GMR^2} = \frac{GM}{R^2}.$$
 (6)

We will use $g_N = GM/R^2$ in the next section.

```
// 2.39545e47 is the 't Hooft-Susskind constant:
// the number of field lines for a black hole of
// unit Schwarzschild radius
//const\ real_type\ G_- =
        (k * c3 * pi)
       / (log(2.0) * hbar * 2.39545e47);
const string filename = "newton.txt";
ofstream out_file(filename.c_str());
out_file << setprecision (30);
const real_type start_distance = 10.0;
const real_type end_distance = 100.0;
const size_t distance_res = 1000;
const real_type distance_step_size =
        (end_distance - start_distance)
        / (distance_res - 1);
for (size_t step_index = 0; step_index < distance_res; step_index++)
        const real_type r =
                start_distance + step_index * distance_step_size;
        const vector_3 receiver_pos(r, 0, 0);
        const real_type receiver_radius = 1.0;
        const real_type epsilon = 1.0;
        vector_3 receiver_pos_plus = receiver_pos;
        receiver_pos_plus.x += epsilon;
        // https://en.wikipedia.org/wiki/Directional_derivative
        const real_type collision_count_plus =
                get_intersecting_line_count(
                        receiver_pos_plus,
                        receiver_radius);
        const real_type collision_count =
                get_intersecting_line_count(
                        receiver_pos,
                        receiver_radius);
        const real_type gradient =
                (collision_count_plus - collision_count)
                / epsilon;
        real_type gradient_strength =
                -gradient
                / (receiver_radius * receiver_radius);
```

This method is faster and less memory intensive when compared to the integer field count method. This method is meant to be a stepping stone for the next section.

For reference, if you know n, and you wish to know the emitter radius from that, then the equation is:

$$r_{emitter} = \sqrt{\frac{nG\hbar \log 2}{kc^3\pi}}. (7)$$

Using this radius, one can ensure that the results from this section match the results of the previous section, where n is relatively small anyway (e.g. $n = 10^8$).

7 Application: modeling Mercury's orbit using numerical integration

In essence, the numerical calculation of the Newtonian orbit of Mercury is as follows:

- Place Mercury at the aphelion to start
- Calculate the orbit path by repeatedly taking steps in time

The constant time slice is:

```
\mathbf{const} real type \mathrm{dt} = 10000; \ // \ 2.77777 \ hours
```

The initial conditions are:

```
vector_3 Mercury_pos(0, 69817079000.0, 0); // Aphelion location vector_3 Mercury_vel(-38860, 0, 0); // Aphelion velocity
```

The orbit code is as follows. Here we use Eq. 6 (e.g. $g_N = GM/R^2$) to calculate the acceleration from Newtonian gravitation:

```
vector_3 Newtonian_acceleration(
    const real_type emitter_mass,
    const vector_3& pos, // Receiver pos
    const real_type G)

{
    // Sun's position is fixed at the origin
    vector_3 grav_dir = vector_3(0, 0, 0) - pos;
    const real_type distance = grav_dir.length();
    grav_dir.normalize();

    vector_3 accel = grav_dir * G * emitter_mass / pow(distance, 2.0);

    return accel;
}
```

Here we show the Euler integration, which is extremely simple. The acceleration is calculated, then it is added (e.g. integrated) to the velocity. Once that's done, the velocity is added to the position. See Figs. 2 and 3.

The passage of time is computed whenever the window manager (e.g. OpenGL/GLUT) is not busy drawing or processing input:

```
void idle_func(void)
{
         proceed_Euler(Mercury_pos, Mercury_vel, G, dt);
}
```

On the other hand, rather than using Euler integration, the order-4 symplectic integration does a better job at conserving energy, but at a speed cost:

```
pow(2.0, 1.0 / 3.0);
static const real_type c[4] =
        1.0 / (2.0 * (2.0 - cr2)),
        (1.0 - cr2) / (2.0 * (2.0 - cr2)),
        (1.0 - cr2) / (2.0 * (2.0 - cr2)),
        1.0 / (2.0 * (2.0 - cr2))
};
static const real_type d[4] =
        1.0 / (2.0 - cr2),
        -cr2 / (2.0 - cr2),
        1.0 / (2.0 - cr2),
        0.0
};
pos += vel * c[0] * dt;
vel += Newtonian_acceleration (
                emitter_mass,
                pos,
                G) * d[0] * dt;
pos += vel * c[1] * dt;
vel += Newtonian_acceleration(
                emitter_mass,
                pos,
                G) * d[1] * dt;
pos += vel * c[2] * dt;
vel += Newtonian_acceleration(
                emitter_mass,
                pos,
                G) * d[2] * dt;
pos += vel * c[3] * dt;
// last element d/3 is always 0
```

See Fig. 4.

8 Final code

A final code, which models the orbit of Mercury, is at: https://github.com/sjhalayka/mercury_orbit_glut

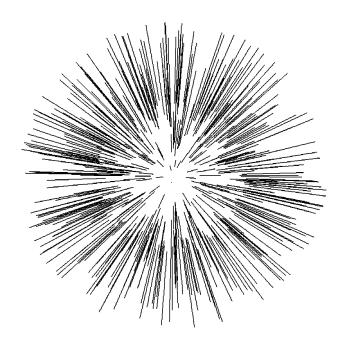


Figure 1: The emitter is spherical. The field line starting positions are placed pseudorandomly on a 2-sphere, and the normals (e.g. field line directions) are calculated using the same positions.

References

- [1] Misner et al. Gravitation. (1970)
- [2] 't Hooft. Dimensional reduction in quantum gravity. (1993)
- [3] Susskind. The World as a Hologram. (1994)
- [4] Fiedler. Fix Your Timestep! (2004)
- $[5]\,$ Fiedler. Integration Basics. (2004)
- [6] Williams. NASA Mercury Fact Sheet. (2024)

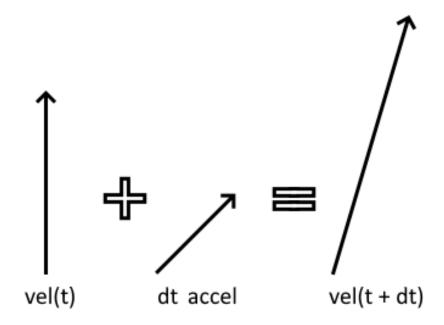


Figure 2: A diagram of the Euler integration of velocity.

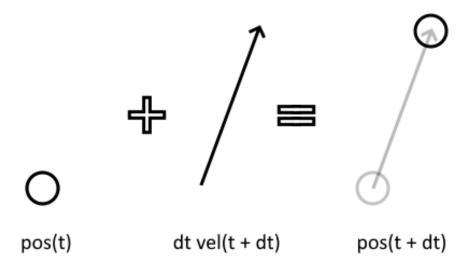


Figure 3: A diagram of the Euler integration of position.

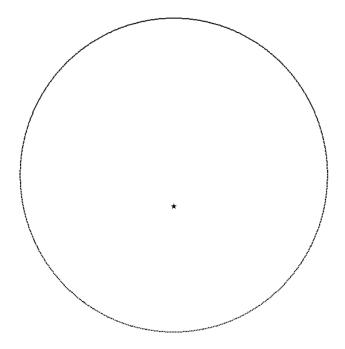


Figure 4: Mercury in orbit around the Sun.