

## MAE3134: Homework 3

Due date: 20 February 2018

**Problem 1.** A Boeing 747 is in Straight and Level Unaccelerated Flight (SLUF) at 40 000 ft, 871 ft/s. The linearized equations of longitudinal motion are given by

$$\begin{bmatrix} s + 0.0828 & -0.0215 & 0.5589 \\ 0.0573 & 15.3348s + 5.9633 & -15.0685s \\ 0.0057 & 0.1425s + 1.6165 & s^2 + 0.0438s \end{bmatrix} \begin{bmatrix} \delta u \\ \delta \alpha \\ \delta \theta \end{bmatrix} = \begin{bmatrix} 0 \\ -0.3226 \\ -1.2124 \end{bmatrix} \delta e, \quad (1)$$

where  $\delta u, \delta \theta, \delta e, \delta \alpha$  are the deviations of forward velocity in ft/s, pitch in degrees, elevation deflection in degree, and deviation of the angle of attack in degrees, respectively.

We can define a transfer function relating the input elevator deflection to the output of the pitch deviations as

$$\frac{\theta(s)}{\delta e(s)} = \frac{-18.5(s + 0.36)(s + 0.08)}{(s + 0.47 \pm 1.24j)(s + 0.06)(s + 0.02)}. \quad (2)$$

(a) Find the pitch response  $\theta(t)$  if the elevation is deflected a constant  $-0.01^\circ$ .

It is typically difficult to analyze the short period motion, due to the complex poles, and the phugoid motion, due to the real poles. As a result, it is possible to separate the dynamics into two separate transfer functions:

$$\frac{\delta \alpha}{\delta e} = \frac{-0.021(s^2 + 56.7s)}{s(s + 0.283 \pm 1.24j)} \quad (3)$$

$$\frac{\delta \theta}{\delta e} = \frac{-1.21(s + 0.36)}{s(s + 0.283 \pm 1.24j)} \quad (4)$$

**Problem 2.** Insert a tikz picture of a missile in flight A missile in flight, as shown in Fig whatever, is subject to several forces, such as thrust, lift, drag, and gravity. The missile flies at an angle of attack,  $\alpha$ , with respect to the velocity vector, creating lift. For steering, the body angle from vertical,  $\phi$ , is controlled by rotating the thrust vector at the tail. The transfer function relating the body angle,  $\phi$ , to the angular displacement,  $\delta$ , of the engine is of the form

$$\frac{\phi(s)}{\delta(s)} = \frac{K_a s + K_b}{K_3 s^3 + K_s s^2 + K_1 s + K_0} \quad (5)$$

**Problem 3.** For each second order system below, find  $\zeta, \omega_n, T_s, T_p, T_r$ , and %OS. You will need to estimate the rise time from a plot of the unit step response.

(a)

$$T(s) = \frac{16}{s^2 + 3s + 16}$$

(b)

$$T(s) = \frac{0.04}{s^2 + 0.02s + 0.04}$$

(c)

$$T(s) = \frac{1.05 \times 10^7}{s^2 + 1.6 \times 10^3 s + 1.05 \times 10^7}$$

**Problem 4.** For each pair of second order system specifications, find the location of the second order poles. Write the transfer function that will satisfy the specifications.

(a)  $\%OS = 12\%, T_s = 0.6 \text{ s}$

(b)  $\%OS = 10\%, T_p = 5 \text{ s}$

(c)  $T_s = 7 \text{ s}, T_p = 3 \text{ s}$