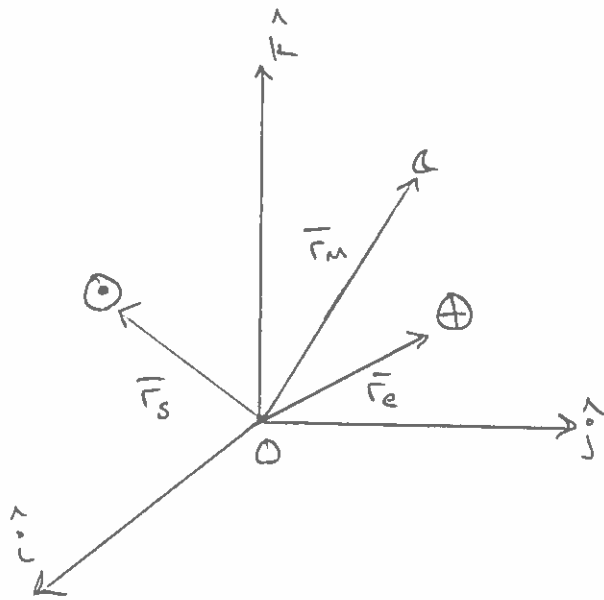


N-BODY EXAMPLE

CONSIDER THREE BODIES: \odot , \oplus , \triangle

SUN EARTH MOON



WE WILL ASSUME THAT
OUR ORIGIN = SUN !

CENTER OF MASS

$$\left(\sum_{i=1}^3 m_i \right) \vec{r}_{\text{com}} = \sum_{i=1}^3 m_i \vec{r}_i$$

$$\vec{r}_{\text{com}} = \frac{m_s \vec{r}_s + m_e \vec{r}_e + m_m \vec{r}_m}{m_s + m_e + m_m}$$

← PLUG IN VALUES
FROM CONSTANTS
SHEET

FOR OUR SOLAR SYSTEM + CONSTANTS SHEET

$\vec{r}_s = \vec{0} \Rightarrow$ FOR THIS PROBLEM WE'LL TREAT PT O,
OUR ORIGIN AS THE SUN

\Rightarrow THIS MEANS \vec{r}_{com} WILL BE A VECTOR FROM
THE SUN/ORIGIN TO THE CENTER OF MASS

EQUATIONS OF MOTION OF \triangle WRT CENTER OF MASS

$$m_c \ddot{\vec{r}}_c = -G \sum_{j=1}^2 \frac{m_c m_j}{r_{jc}^3} \vec{r}_{jc}$$

WHERE $j = \text{SUN, EARTH}$

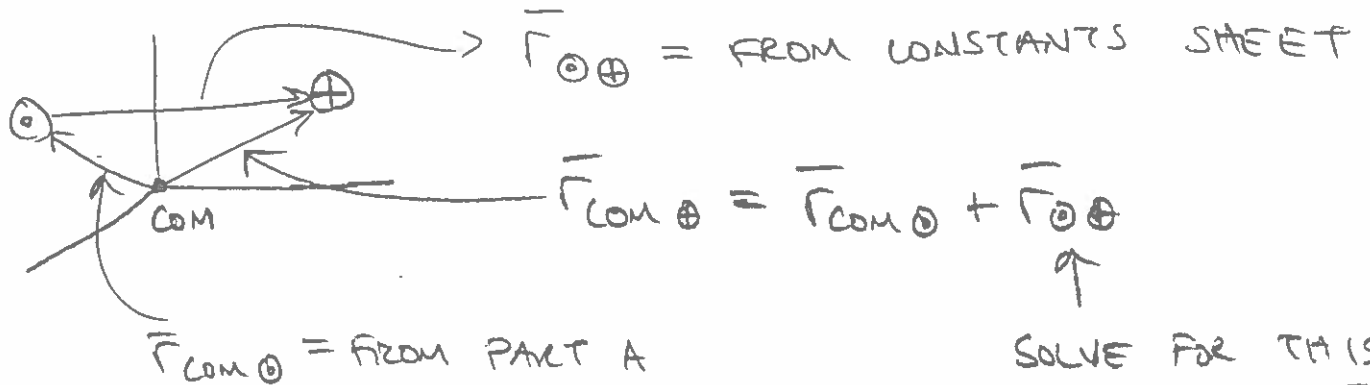
CENTER OF MASS IS AN INERTIALLY FIXED POINT
FROM THE CONSERVATION OF LINEAR MOMENTUM

$$M_A \ddot{\vec{r}}_{\text{com}A} = -\frac{G M_A M_\odot}{r_{\odot A}^3} \vec{r}_{\odot A} - \frac{G M_A M_\oplus}{r_{\oplus A}^3} \vec{r}_{\oplus A} \quad (*)$$

↑
CENTRE OF
MASS TO A

↑
NEED TO
FIND VECTOR FROM SUN TO MOON/EARTH

LET'S FIND THE POSITION OF EACH BODY WRT COM.



↑
SOLVE FOR THIS
PART. TO USE
IN (*)

QUESTION TO ASK YOURSELF

1. IS THERE A DIFFERENCE IN THESE TWO EQUATIONS

$$\vec{r}_{\odot\oplus} = \vec{r}_{\oplus A} - \vec{r}_{\odot A} \rightarrow \left(\begin{matrix} \text{MOON TO} \\ \text{EARTH} \end{matrix} \right) - \left(\begin{matrix} \text{MOON TO} \\ \text{SUN} \end{matrix} \right)$$

$$\vec{r}_{\odot\oplus} = \vec{r}_{\text{COM}\oplus} - \vec{r}_{\text{COM}\odot} \rightarrow \left(\begin{matrix} \text{COM TO} \\ \text{EARTH} \end{matrix} \right) - \left(\begin{matrix} \text{COM TO} \\ \text{SUN} \end{matrix} \right)$$