

```
% Devon Doud, Nathan Sanders, Sam Weston
% MSU Spring, 2019
% EMEC 303 Project 1
```

```
clear; clc
close all
```

```
% Set step size and length of time in seconds
```

```
h = 1;
Lt = 6000;
N = round(Lt/h);
```

```
% Preallocation
```

```
height = zeros(1,N);
den = zeros(1,N);
t = zeros(1,N);
phi = zeros(1,N);
force_r = zeros(1,N);
force_theta = zeros(1,N);
m = zeros(1,N);
```

```
% Define initial conditions and constants
```

```
t_m = 1.1045; % Thrust multiplier
t(1) = 0; % Initial time
den(1) = rho(0); % Starting density
r(1) = 6.378e6; % Radius of earth
vr(1) = 0; % Initial radial velocity
th(1) = 0; % Initial angle theta
w(1) = 7.29e-5; % Angular velocity of earth
phi(1) = deg2rad(73.8); % Launch angle relative to horizontal
height(1) = 0; % Initial heigh above ground
```

```
% Call force function to determine additional initial conditions
```

```
[force_r(1),force_theta(1),m(1)] = ...
    force(t(1),den(1),phi(1),vr(1),w(1),r(1),t_m);
```

```
% Store dependent variables into y
```

```
y = [r,vr,th,w];
```

```
% ODEs describing motion of rocket
```

```
f = @(t,y,force_r,force_theta,m) ...
```

```
[y(2), force_r/m + y(1)*y(4)^2,...  
y(4),(force_theta/m - 2*y(2)*y(4))/y(1)];
```

```
for i = 1:N-1
```

```
    % Update height and density
```

```
    height(i+1) = y(i,1)-r(1);
```

```
    den(i+1) = rho(height(i));
```

```
    % Update time
```

```
    t(i+1) = t(i) + h;
```

```
    % Update phi based on height
```

```
    if height(i) <= 275000
```

```
        phi(i+1) = sqrt(phi(1)-phi(1)*(height(i)/275000)^2);
```

```
    else
```

```
        phi(i+1) = 0;
```

```
    end
```

```
    % Call force function to update force in r and theta directions
```

```
    [force_r(i+1),force_theta(i+1),m(i+1)] = ...
```

```
        force(t(i),den(i),phi(i),y(i,2),y(i,4),y(i,1),t_m);
```

```
    % Euler update for dependent variables
```

```
    y(i+1,:) = y(i,:) + h*f(t(i),y(i,:),force_r(i),force_theta(i),m(i));
```

```
end
```

```
% Plot results
```

```
figure
```

```
polarplot((y(:,3)),y(:,1))
```

```
hold on
```

```
polarplot(0:.01:2*pi,ones(size(0:.01:2*pi))*r(1))
```

```
title('Rocket Trajectory in Polar Coordinates')
```

```
word1={'Radius (m)'};
```

```
text(0,3000000,word1)
```

```
figure
```

```
subplot(2,2,1)
```

```
plot(height,phi)
```

```
title ('Rocket Angle Relative to Horizontal')
```

```
ylabel('Angle (rad)')  
xlabel('Height (m)')
```

```
subplot(2,2,2)  
plot(t,height)  
title ('Rocket Height vs Time')  
ylabel('Height (m)')  
xlabel('Time (s)')
```

```
subplot(2,2,3)  
plot(t,y(:,2))  
title ('Radial Velocity vs Time')  
ylabel('Speed (m/s)')  
xlabel('Time (s)')
```

```
subplot(2,2,4)  
plot(t,y(:,4).*y(:,1))  
title ('Velocity in the Theta Direction vs Time')  
ylabel('Speed (m/s)')  
xlabel('Time (s)')
```