Optimum design of laminated composites for minimum thickness by a self-adaptative genetic algorithm

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1. Classic Lamination Theory

$$\begin{bmatrix} N_{x} \\ N_{y} \\ N_{xy} \end{bmatrix} = \begin{bmatrix} A_{11} & A_{12} & A_{16} \\ A_{12} & A_{22} & A_{26} \\ A_{16} & A_{26} & A_{66} \end{bmatrix} \begin{bmatrix} \varepsilon_{x}^{0} \\ \varepsilon_{y}^{0} \\ \gamma_{xy}^{0} \end{bmatrix}$$

$$+ \begin{bmatrix} B_{11} & B_{12} & B_{16} \\ B_{11} & B_{12} & B_{16} \\ B_{16} & B_{26} & B_{66} \end{bmatrix} \begin{bmatrix} k_{x} \\ k_{y} \\ k_{xy} \end{bmatrix}$$

$$= \begin{bmatrix} B_{11} & B_{12} & B_{16} \\ B_{12} & B_{22} & B_{26} \\ B_{16} & B_{26} & B_{66} \end{bmatrix} \begin{bmatrix} \varepsilon_{x}^{0} \\ \varepsilon_{y}^{0} \\ k_{xy} \end{bmatrix}$$

$$+ \begin{bmatrix} D_{11} & D_{12} & D_{16} \\ D_{11} & D_{12} & D_{16} \\ D_{16} & D_{26} & D_{66} \end{bmatrix} \begin{bmatrix} k_{x} \\ k_{y} \\ k_{xy} \end{bmatrix}$$

2. Failure Theory

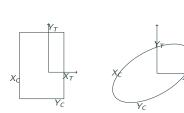


Figure 2: Schematic failure surfaces for maximum stress and quadratic failure criteria

Maximum stress failure

$$SF_{MS}^{k} = \min \text{ of } \begin{cases} SF_{X}^{k} = \begin{cases} \frac{X_{t}}{\sigma_{11}}, \text{ if } \sigma_{11} > 0\\ \frac{X_{c}}{\sigma_{11}}, \text{ if } \sigma_{11} < 0 \end{cases} \\ SF_{Y}^{k} = \begin{cases} \frac{Y_{t}}{\sigma_{22}}, \text{ if } \sigma_{22} > 0\\ \frac{Y_{c}}{\sigma_{22}}, \text{ if } \sigma_{22} < 0 \end{cases} \\ SF_{S}^{k} = \begin{cases} \frac{S}{|\tau_{12}|} \end{cases}$$

• Tsai-wu failure theory

$$\begin{split} H_1\sigma_1 + H_2\sigma_2 + H_6\tau_{12} + H_{11}\sigma_1^2 + H_{22}\sigma_2^2 \\ + H_{66}\tau_{12}^2 + 2H_{12}\sigma_1\sigma_2 < 1 \end{split}$$

3. Self-adaptative GA

- Modifying selection strategy: in order to handle the constraint search
- Self-adaptative mutation direction of fiber orientation and laminate thickness: random change the length, and the angle in the laminate.
- The self-adaptative parameters don't refer to parent's proportion, mutation probability.

3. Self-adaptative GA: selection operator

- acitve group: individual is used to increase the diversity of the population
- potential group: individual doesn't fulfill constraint
- proper group: individual meet constraint

3. Self-adaptative GA: mutation operator

$$\mathsf{md} = [\mathit{CT}_1, \cdots, \mathit{CT}_{n-1}, \mathit{CT}_n] - [\mathit{ICV}_0, \cdots, \mathit{ICV}_{n-1}, \mathit{ICV}_n]$$

- md means mutation direction.
- CT_i denotes the i-th constraint, such as weight, safety factor.
- ICV_i denotes individual's i-th constraint value, such as, weight, safety factor of current individual.

4. Self-adaptative GA: mutation operator

length mutation =

$$\begin{cases} LMC * [0, \sum_{i=1}^{N} md_i] & \text{if } \sum_{i=1}^{N} md_i > 0 \\ LMC * [\sum_{i=1}^{N} md_i, 0] & \text{if } \sum_{i=1}^{N} md_i < 0 \end{cases}$$

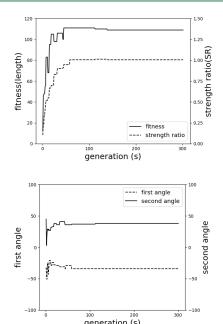
LMC stands for length mutation coefficient, it's a positive integer.

• angle mutation =

$$\begin{cases} AMC * [0, \sum_{i=1}^{N} md_i] & \text{if } \sum_{i=1}^{N} md_i > 0 \\ AMC * [\sum_{i=1}^{N} md_i, 0] & \text{if } \sum_{i=1}^{N} md_i < 0 \end{cases}$$

AMC stands for angle mutation coefficient, it's sign is unclear.

5. Result: Loading $N_x = 10$, $N_y = 5$ MPa m



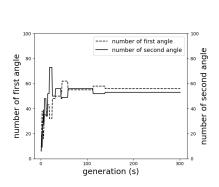


Figure 3: Two distinct angles in the laminate

6. Comparison with related research

Table 1: Comparison with the results of DSA

Loading	Akbulut and Sonmez's Study				Present Study			
$N_x/N_y/N_{xy}$ (MPa m)	Optimum lay-up sequences	laminate thickness	TW	MS	Optimum lay-up sequences	laminate thickness	TW	MS
10/5/0	[37 ₂₇ /-37 ₂₇] _s	108	1.0068	1.0277	[33 ₂₉ /-39 ₂₅ /-39] _s	109	1.0074	1.0246
20/5/0	[31 ₂₃ /-31 ₂₃] _s	92	1.0208	1.1985	[33 ₂₂ /-31 ₂₄] _s	92	1.0055	1.2065
40/5/0	[26 ₂₀ /-26 ₂₀] _s	80	1.0190	1.5381	$[29_{18}/-21_{23}/-\bar{2}1]_s$	83	1.0034	1.7350
80/5/0	[21 ₂₅ /-19 ₂₈] _s	106	1.0113	1.2213	[-20 ₂₇ /21 ₂₅ /25] _s	105	1.0029	1.2063
120/5/0	[17 ₃₅ /-17 ₃₅] _s	140	1.0030	1.0950	[-18 ₃₄ /17 ₃₆] _s	140	1.0000	1.0898