

Perturbation in hierarchical system.

first → Kozai:

Lets remember that we can write the Hamiltonian including a small perturbing term R (perturbing function) as:

$$(1) \quad H = -\frac{M^2}{2L^2} - R$$

where the Delaunay elements are:

$$(2) \quad \begin{cases} h = \mathcal{E} \\ g = \omega \\ l = h(t - \tau) \approx M \end{cases} \quad \begin{cases} H = \sqrt{\mu a(1-e^2)} \cos i \\ G = \sqrt{\mu a(1-e^2)} \\ L = \sqrt{\mu a M} \end{cases} \quad \begin{matrix} \leftarrow \text{character} \\ \text{of the} \\ \text{angular} \\ \text{momentum} \end{matrix}$$

for 3-body problem we will limit ourselves to a hierarchical system:

(i) - internal

(e) - external

The full Hamiltonian is:

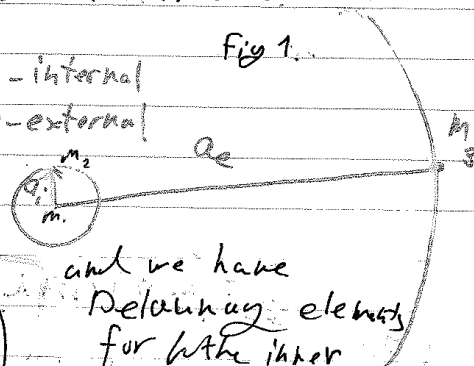
$$(3) \quad H = -\frac{M^2}{2L_i^2} - \frac{m^2}{2L_e^2} + Gm_3 \left(\frac{m_1 m_2}{r_{12}} - \frac{m_1}{r_{13}} - \frac{m_2}{r_{23}} \right)$$

and we have

Delaunay elements for the inner and outer orbits.

and the Hamiltonian eqs. of motions are

$$\dot{h} = (h, H) \quad \dot{m} = (m, H)$$



$$(4) \quad \dot{L}_{i,e} = \frac{\partial H}{\partial l_{i,e}} \quad \dot{l}_{i,e} = \frac{\partial H}{\partial L_{i,e}}$$

$$(5) \quad \dot{G}_{i,e} = -\frac{\partial H}{\partial g_{i,e}} \quad \dot{g}_{i,e} = \frac{\partial H}{\partial G_{i,e}}$$

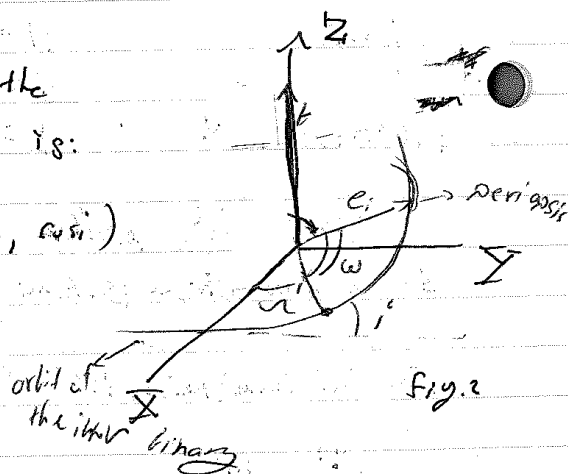
$$(6) \quad \dot{H}_{i,e} = -\frac{\partial H}{\partial h_{i,e}} \quad \dot{h}_{i,e} = \frac{\partial H}{\partial H_{i,e}}$$

Now Note that $\vec{G}_{i,e}$ is the angular momentum of the inner/external orbits and the total angular momentum \vec{G}_{tot} is simply

$$(7) \quad \vec{G}_{tot} = \vec{G}_i + \vec{G}_e$$

So the projection on the frame where $\vec{G}_{tot} \parallel \hat{z}$ is:

$$(8) \quad \vec{G}_{tot} = G(\sin i \sin h_i, -\sin i \cos h_i, \cos i)$$



$$(9) \quad \vec{G}_{tot} = (G_i \sin i \sin h_i + G_e \sin i \sin h_e, -G_i \sin i \cos h_i - G_e \sin i \cos h_e, H_i + H_e)$$

remember: $H = G \cos i$

Elimination of nodes: We can simplify the Hamiltonian by choosing an optimal coordinate system. We choose the invariable plane which is perpendicular to the total angular momentum.

So in this coordinate system the total angular momentum is:

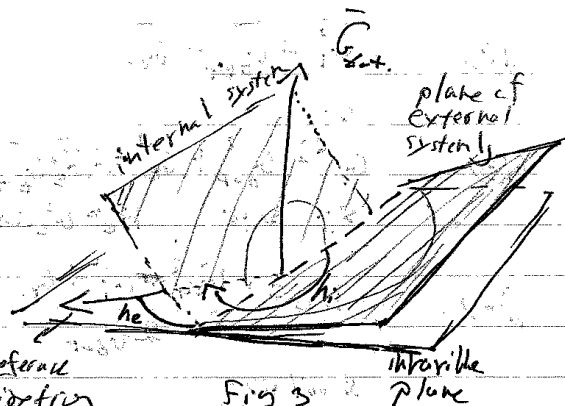
$$(10) \quad \vec{G}_{tot} = (0, 0, G_{tot})$$

new system

remains - longitude of ascending nodes is defined with respect to the reference direction

So in the invariable plane

h_e and h_i are different by amount of π .



$$(11) \quad h_i + \pi = h_e$$

So from (10) and (11)

$$(12) \quad G_i \sin i \sin h_i = -G_e \sin i \sin h_e \quad \rightarrow \text{the same as}$$

$$(13) \quad G_i^2 - H_i^2 = G_e^2 - H_e^2$$

$$(14) \quad *G_i \sin i \cos h_i = G_e \sin i \cos h_i *$$

$$(15) \quad \vec{G}_i + \vec{G}_e = \vec{G}_{tot}$$

$$(16) \quad G_{tot}^2 = |\vec{G}_i + \vec{G}_e|^2 = G_i^2 + G_e^2 + 2\vec{G}_i \cdot \vec{G}_e = G_i^2 + G_e^2 + 2G_i G_e \cos i$$

where i is the angle between \vec{G}_i and \vec{G}_e which is the inclination angle.

So eq. (16) is

$$(17) \quad \boxed{G_{tot}^2 = G_1^2 + G_e^2 - 2G_1 G_e \cos i}$$

from eq. (15)

$$(18) \quad \bar{G}_i = \bar{G}_{tot} - \bar{G}_e$$

so

$$(19) \quad G_i^2 = G_{tot}^2 + G_e^2 - 2\bar{G}_e \cdot \bar{G}_{tot} = G_{tot}^2 + G_e^2 - 2G_e G_{tot} \cos i_e$$

so; since $H_e = G_e \cos i_e$

$$(20) \quad \boxed{H_e = \frac{G_{tot}^2 - G_i^2 + G_e^2}{2G_{tot}}}$$

similarly:

$$(21) \quad \boxed{H_i = \frac{G_{tot}^2 - G_e^2 + G_i^2}{2G_{tot}}}$$

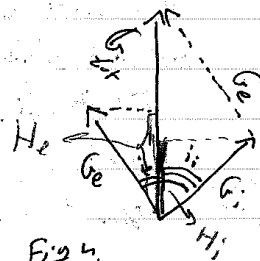


Fig 4.

$$\boxed{H_e + H_i = G_{tot}}$$

So the Hamiltonian depends only on a combination of $h_i - h_e$. To see that lets remember the Hamiltonian.

Let's take a look at the Hamiltonian

$$(22) \quad H = T + V$$

and

$$(23) \quad V = -\frac{Gm_1 m_2}{r} - \frac{Gm_1 m_3}{r_{13}} - \frac{Gm_2 m_3}{r_{23}} \quad (\text{remember Fig. 1})$$

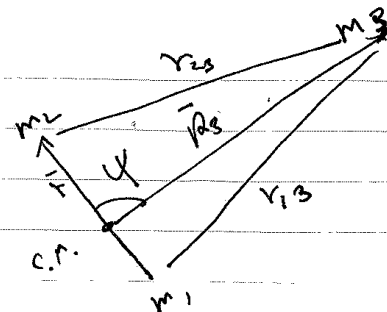
The Hamiltonian is:

$$(24) \quad H = -\frac{\mu^2}{2L_1^2} - \frac{\mu^2}{2L_2^2} + GM_3 \left(\frac{m_1 + m_2}{R_3} - \frac{m_1}{r_{13}} - \frac{m_2}{r_{23}} \right)$$

So also note:

$$(25) \quad \vec{r}_{23} = \vec{R}_3 - \frac{m_1}{m_1 + m_2} \vec{r}$$

$$(26) \quad \vec{r}_{13} = \vec{R}_3 + \frac{m_2}{m_1 + m_2} \vec{r}$$



Let us remember that in "The Disturbing Function" chapter we had our disturbing function define as:

$$(27) \quad R = \frac{m'}{r'} \sum_{l=2}^{\infty} \left(\frac{r'}{r} \right)^l P_l(\cos \psi)$$

Remember

we had:

$$(A.1) \quad \frac{1}{|\vec{r} - \vec{r}'|} = \frac{1}{\sqrt{r^2 + r'^2 - 2rr' \cos \psi}} =$$

$$= \frac{1}{r} \left[1 + \left(\frac{r'}{r} \right)^2 - 2 \frac{r'}{r} \cos \psi \right]^{-\frac{1}{2}}$$

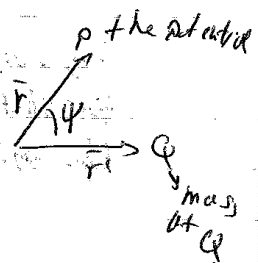
Define

$$(A.2) \quad t = \frac{r'}{r}, \text{ so}$$

$$(A.3) \quad \frac{1}{|\vec{r} - \vec{r}'|} = \frac{1}{r} \left[1 + t^2 - 2tx \right]^{-\frac{1}{2}} = \frac{1}{r} \sum_{h=0}^{\infty} t^h P_h(x)$$

We can write $P_h(x)$:

$$(A.4) \quad P_h(x) = \sum_{k=0}^{h/2} \frac{(-1)^k (2h-2k)!}{2^{2k} k! (h-k)! (h-2k)!} x^{h-2k}$$



and we have:

Legendre
polynomials

A.5 $P_0(x) = 1$

A.6 $P_1(x) = x$

A.7 $P_2(x) = \frac{1}{2}(3x^2 - 1)$

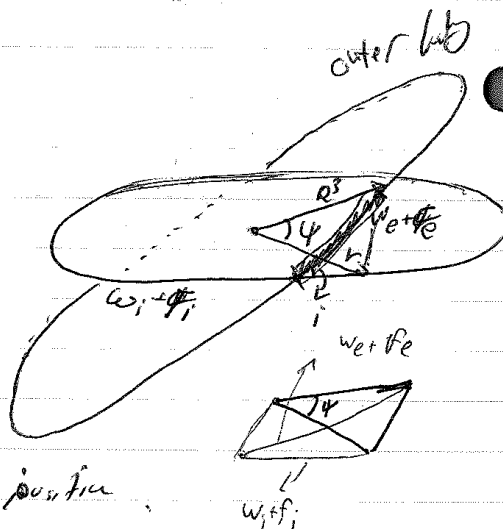
A.8 $P_3(x) = \frac{1}{2}(5x^3 - 3x)$

A.9 $P_n(x) = \frac{1}{8}(35x^4 - 30x^2 + 3)$ and so on...

So, Lets calculate R .
for that we need
the angle ψ

$\phi_e \rightarrow$ the phase (the angle)
in the plane

$\omega_i + \phi_i$ = from the right
of interaction ϕ_e , the
line of nodes is this position



So:

(28) $\cos \psi = \cos(\omega_e + \phi_e) \cos(\omega_i + \phi_i) + \sin(\omega_e + \phi_e) \sin(\omega_i + \phi_i) \cos i$

We can write V as:

$$(20) \quad V = -\frac{Gm_1 m_2}{r} - \frac{Gm_1 m_3}{R_3} \sum_{n=0}^{\infty} \left(-\frac{m_2}{m_1 + m_2} \right)^n \left(\frac{r}{R_3} \right)^n P_n(\cos \psi) \\ - \frac{Gm_2 m_3}{R_3} \sum_{n=0}^{\infty} \left(\frac{m_1}{m_1 + m_2} \right)^n \left(\frac{r}{R_3} \right)^n P_n(\cos \psi) =$$

$$= -\frac{Gm_1 m_2}{r} - \frac{Gm_1 m_3}{R_3} \left[\left(-\frac{m_2}{m_1 + m_2} \right)^0 \left(\frac{r}{R_3} \right)^0 P_0(\cos \psi) + \left(-\frac{m_2}{m_1 + m_2} \right)^1 \frac{r}{R_3} P_1(\cos \psi) \right] \\ - \frac{Gm_2 m_3}{R_3} \left[\left(\frac{m_1}{m_1 + m_2} \right)^0 \left(\frac{r}{R_3} \right)^0 P_0(\cos \psi) + \left(\frac{m_1}{m_1 + m_2} \right)^1 \frac{r}{R_3} P_1(\cos \psi) \right] \\ - \frac{Gm_1 m_3}{R_3} \sum_{n=2}^{\infty} \left(-\frac{m_2}{m_1 + m_2} \right)^n \left(\frac{r}{R_3} \right)^n P_n(\cos \psi) \\ - \frac{Gm_2 m_3}{R_3} \sum_{n=2}^{\infty} \left(\frac{m_1}{m_1 + m_2} \right)^n \left(\frac{r}{R_3} \right)^n P_n(\cos \psi)$$

$R \rightarrow$ the distance function

Quadrupole $\rightarrow n=2$

$$(30) \quad V_{(n=0)} = -\frac{Gm_1 m_2}{r} - \frac{Gm_3(m_1 - m_2)}{R_3} \\ - \frac{Gm_1 m_2 m_3}{2(m_1 + m_2) R_3} \left(\frac{r}{R_3} \right)^2 [3\cos^2 \psi - 1]$$

The first two terms are the potential of the two body motion for the inner or outer binary.

With the kinetic energy they give 1/2 the

Hamiltonian $-\frac{m^2}{2I} = -\frac{m^2}{2\mu}.$

So for the h=2 we get that

$$(31) \quad R = \frac{G m_1 m_2 m_3}{2(m_1 + m_2) R_3} \left(\frac{r}{R_3} \right)^2 (3 \cos^2 \psi - 1)$$

We need to integrate twice. Once over the outer orbit and once over the inner orbit.

So we can write R as:

$$(32) \quad R = \frac{G m_1 m_2 m_3}{2(m_1 + m_2)} dV_i dV_e dM_i dM_e$$

So from Kepler eq. $M = E - e \sin E$

We have:

$$(33) \quad dM_i = (1 - e_i \cos E) dE \quad \text{inner orbit}$$

and since

$$(34) \quad R_3 = \frac{a_e (1 - e_e^2)}{1 + e_e \cos f_e}$$

We fix (after some more manipulations)

$$(35) \quad dM_e = \left(\frac{R_3}{a_e} \right)^2 \frac{df_e}{1 - e_e^2}$$

$$(36) \quad dV_e = (3 \cos^2 \psi - 1) \frac{(1 + e_e \cos f_e)}{a_e^3 (1 - e_e^2)^{3/2}} df_e$$

$$(37) \quad \langle \dots \rangle = \frac{1}{2\pi} \int_0^{2\pi} \dots df_e \quad \leftarrow \text{so the averaging}$$

where

$$(38) \quad \begin{cases} \langle \cos f_e \rangle = 0 & \langle \cos^2 f_e \rangle = \frac{1}{2} & \langle \sin^2 f_e \rangle = \frac{1}{2} \\ \langle \sin f_e \cos f_e \rangle = 0 & e_e \langle \cos^3 f_e \rangle = 0 \\ e_e \langle \cos f_e \sin^2 f_e \rangle = 0 & e_e \langle \sin^4 f_e \cos f_e \rangle = 0 \end{cases}$$

$$(39) \quad \langle V_e \rangle = \frac{1}{2a_e^3(1-e^2)^{3/2}} [3C_1^2 + 3C_2^2 \cos^2 i - 2]$$

where

$$(40) \quad C_1 = \cos(\omega_i + f_i)$$

$$(41) \quad C_2 = \sin(\omega_i + f_i)$$

Thus,

$$(42) \quad \langle R \rangle = \frac{Gm_1 m_2 m_3}{2(m_1 m_2)} \frac{r^2}{2a_e^3(1-e^2)^{3/2}} [3(\cos^2(\omega_i + f_i) + \sin^2(\omega_i + f_i) \cdot \cos^2 i) - 2]$$

So we need now to average:

$$(43) \quad M_i = \left(\frac{r}{a_i}\right)^2 \int_0^{2\pi} [3(\cos^2(\omega_i + f_i) + \sin^2(\omega_i + f_i) \cos^2 i) - 2] dM_i$$

So:

$$(44) \quad \langle \langle R \rangle \rangle = \frac{Gm_1 m_2 m_3 a_i^2}{8m_1 a_e^3(1-e^2)^{3/2}} \int_0^{2\pi} [2 + 3e_i^2 - 3\sin^2 i [5e_i^2 \sin^2 \omega_i + 1 - e_i^2]] d\omega_i$$

Conserved parameters:

- 1) since we have integrated over $f_i, \omega_i, (l_0, l_i), (M_0, M_i)$ the corresponding canonical momenta L_i, L_e are constant (elimination of these anomalies).

$$(45) \quad L_i = \text{const}, \quad L_e = \text{const} \rightarrow a_i = \text{const}, \quad a_e = \text{const}$$

- 2) note that for this approximation the Hamiltonian does not depend on ω_e . (remark: no longer true when we do the expansion)

- So
- (46) $G_e = \text{const}$
- (47) $e_e = \text{const}$ since $1 - e_e^2 = \frac{G_e^2}{L_e^2}$
- So the outer orbit does not change in its shape and size.

3. Elimination of the nodes

The way we choose our form of reference mean that

- (48) $H_e + H_i = G_{ew}$ (eq. 20) and (21)
- that mean that only $H_i - h_e$ enters the Hamiltonian, i.e., $H_e + H_i$ enters

through

(49) $\cos i = \frac{G_{ew}^2 - G_i^2 - G_e^2}{2 G_i G_e}$ (eq. 17)

and $H_e = \frac{G_{ew}^2 - G_i^2 - G_e^2}{2 G_{ew}}$, $H_i = \frac{G_{ew}^2 - G_e^2 + G_i^2}{2 G_{ew}}$

Lets count degrees of freedom

Kepler has 6, so for each orbit $\times 2 = 12$ total

$a_{e,i}, l_{e,i}, w_{e,i}, l_{e,i}, h_e, e_e$ [12] total

from (1) $a_{e,i} = 0, l_{e,i} = 0$

$12 - 4 = 8$ total

from (2) $w_e = 0, e_e = 0$

6 total

from (3) $H_i = -h_e$

5 total

also for fix $i, i.e = i$

11 total

now we will choose $i_e = 0$

This means that we restrict our outer orbit to be in the invariable plane. This is not always realistic since if our outer orbit doesn't carry most of the angular momentum or even a lot it is not a very good assumption. Nonetheless we will assume it. ~~Thus~~

Thus,

$$(S_0) \quad H_e = G_e \cos i_e = G_e \stackrel{E}{=} \cos i \quad (46)$$

since

$$(S_1) \quad H_e + H_i = G_{tot} \quad \downarrow$$

$$(S_2) \quad H_i = \cos i$$

$$(S_3) \quad G_e \cos i_e \stackrel{i_e=0}{=} G_i \cos i = \sqrt{1-e_i^2} \cos i \stackrel{E}{=} \cos i$$

so the inner binary is in a perfect Kepler orbit so its z-component of the angular momentum is conserved

We will now describe the evolution in time of the orbital parameters.

From the hamiltonian formalism we know the evolution of e, i, Ω the other orbital parameter are

$$(54) \quad a = \frac{L^2}{\mu}$$

$$(55) \quad e = \sqrt{1 - \frac{G^2}{h^2}}$$

$$(56) \quad \cos i = \frac{H}{G}$$

So after some math and remember that

$$(57) \quad \sqrt{\frac{\mu}{a}} = n \quad \sqrt{\mu a} = a^2 n \quad \text{we have:}$$

$$(58) \quad \dot{a} = \frac{2}{na} \frac{\partial R}{\partial \mu}$$

remember the Lagrangian planetary equations

$$(59) \quad \dot{e} = -\frac{\sqrt{1-e^2}}{na^2 e} \frac{\partial R}{\partial \omega} - \frac{1-e^2}{na^2 e} \frac{\partial R}{\partial \mu}$$

$$(60) \quad \dot{i} = -\frac{1}{na^3 \sqrt{1-e^2} \sin i} \frac{\partial R}{\partial \Omega} + \frac{\cos i}{na^3 \sqrt{1-e^2} \sin i} \frac{\partial R}{\partial \omega}$$

$$(61) \quad \dot{\mu} = n - \frac{2}{na} \frac{\partial R}{\partial a} - \frac{1-e^2}{na^2 e} \frac{\partial R}{\partial e}$$

$$(62) \quad \dot{\omega} = \frac{\sqrt{1-e^2}}{na^2 e} \frac{\partial R}{\partial \Omega} - \frac{\cos i}{na^2 \sqrt{1-e^2} \sin i} \frac{\partial R}{\partial i}$$

$$(63) \quad \dot{\Omega} = \frac{1}{na^3 \sqrt{1-e^2} \sin i} \frac{\partial R}{\partial i}$$

for all systems with all the observed taken into account:

$$(64) \quad \dot{\theta} = -\frac{15}{8} \frac{e^2}{\sqrt{1-e^2}} \sinh^2 w \sinh^2 i \frac{A}{h}$$

$$(65) \quad \dot{e} = \frac{15}{8} e \sqrt{1-e^2} \sinh^2 w \sinh^2 i \frac{A}{h}$$

$$(66) \quad \dot{\omega} = \frac{3}{4} \frac{1}{\sqrt{1-e^2}} [2(1-e^2) + 5 \sinh^2 w (e^2 - \sinh^2 i)] \frac{A}{h}$$

$$(67) \quad \dot{\Omega} = -\frac{\cos i}{4\sqrt{1-e^2}} [3 + 12e^2 - 15e^2 \cosh^2 w] \frac{A}{h}$$

where,

$$(68) \quad A = \frac{G m_1 m_2 m_3}{m_3 a e^2 (1-e^2)^{3/2}}$$

we can change the variable $t \rightarrow \tau$ so

$$(69) \quad \tau = \frac{A}{h} t = \frac{G m_1 m_2 m_3}{m_3 h e^3} t \quad \text{where } h \text{ is semi-minor axis}$$

$$(70) \quad e = \frac{\sqrt{a^2 - b^2}}{a}$$

$$(71) \quad \frac{d\theta}{d\tau} = -\frac{15}{8} \frac{e^2}{\sqrt{1-e^2}} \sinh^2 w \sinh^2 i \cosh^2 i$$

$$(72) \quad \frac{de}{d\tau} = \frac{15}{8} e \sqrt{1-e^2} \sinh^2 w \sinh^2 i$$

$$(73) \quad \frac{d\omega}{d\tau} = \frac{3}{4} \frac{1}{\sqrt{1-e^2}} [2(1-e^2) + 5 \sinh^2 w (e^2 - \sinh^2 i)]$$

$$(74) \quad \frac{d\Omega}{d\tau} = -\frac{\cos i}{4\sqrt{1-e^2}} (3 + 12e^2 - 15e^2 \cosh^2 w)$$

close
ex
system
solution
is
numerical

To get some intuition we will make an approximation that the eccentricity is very small, i.e., $e \ll 1$

Thus $(e^2) \rightarrow 0$

Therefore we can write:

$$(75) \quad \frac{di}{dt} = 0 \Rightarrow \text{in const.}$$

$$(76) \quad \frac{de}{dt} = \frac{15}{8} e \sin^2 i \sin^2 \omega$$

$$(77) \quad \frac{d\omega}{dt} = \frac{3}{2} (2 - 5 \sin^2 i \sin^2 \omega)$$

$$(78) \quad \frac{di}{dt} = -\frac{3}{2} \cos i$$

Since we want to solve for the evolution of e with time we need to know how ω changes with time.

$$(79) \quad \int \frac{d\omega}{3(2 - 5 \sin^2 i \sin^2 \omega)} = \int dt$$

$$(80) \quad \int \frac{dx}{a+b \sin^2 x} = \begin{cases} \frac{1}{\sqrt{a(a+b)}} \arctan \sqrt{\frac{(a+b) \tan x}{a(a+b)}} & a(a+b) > 0 \\ \frac{1}{2\sqrt{a(a+b)}} \ln \left| \frac{(a+b) \tan x - \sqrt{a(a+b)}}{(a+b) \tan x + \sqrt{a(a+b)}} \right| & a(a+b) < 0 \end{cases}$$

here $a = 2$, $b = 5 \sin^2 i$

so we have a critical angle which differs between the two behaviours:

$$(81) \quad 2 - 5 \sin^2 i = 0$$

$$(82) \quad \sin^2 i = \frac{2}{5}$$

$$(83) \quad i = 39.23^\circ \quad // \quad 140.77$$

So:

$$(84) \quad w = \begin{cases} \operatorname{arctg} \left[\sqrt{\frac{2}{A}} \frac{e^{\frac{3}{2}\sqrt{2A}z} - 1}{e^{\frac{3}{2}\sqrt{2A}z} + 1} \right] & \sinh^2 i > 0.4, i > 39.23 \\ \operatorname{arctg} \left[\sqrt{\frac{2}{-A}} \tanh \left(\frac{3}{4}\sqrt{-2A}z \right) \right] & \sinh^2 i < 0.4, i < 39.23 \end{cases}$$

↓

$$(85) \quad w \rightarrow \begin{cases} w(z=0) = \frac{\pi}{2}, & w(z \rightarrow \infty) = \text{const} \quad i > 39.23 \\ w(z=0) = 0, & w(z \rightarrow \infty) \rightarrow \text{fluctuate. periodic} \quad i < 39.23 \end{cases}$$

So for the regime where $i > 39$ we can write

$$(86) \quad \frac{dw}{dz} = 0$$

Thus eq. 72 for the evolution of w now gives:

$$(87) \quad \boxed{8 \sinh^2 w \sinh^2 i = 2}$$

$$(88) \quad \sinh^2 w = \frac{2}{8 \sinh^2 i}$$

and

$$(89) \quad \sinh^2 w = 2 \sinh w \cosh w = 2 \sinh w \sqrt{1 - \sinh^2 w}$$

So inserting that to eq. 76 we get

$$\frac{dw}{dz} = \frac{15}{4} e^{2 \sinh w \sqrt{1 - \sinh^2 w} \sinh^2 i} = \frac{15}{8} e^{2 \sqrt{\frac{2}{8 \sinh^2 i}} \frac{1}{\sinh^2 i}}$$

$$\cdot \sqrt{1 - \frac{2}{8 \sinh^2 i}} \sinh^2 i = \frac{15}{4} e^{\sinh^2 i \sqrt{\frac{2}{8 \sinh^2 i} - \frac{2}{8}}} \frac{1}{\sinh^2 i}$$

$$(90) \quad \boxed{\frac{dw}{dz} = \frac{15}{4} e^{\frac{2}{8} (\sinh^2 i - \frac{2}{8})}} \quad \rightarrow$$

Where we take the positive value of $\sin \omega = \pm \sqrt{\frac{2}{3}} \frac{1}{\sin i}$ but $\frac{de}{d\tau}$ can be negative

The solution of eq. 90 is

$$(91) \quad \tau = 0.42 h \left(\frac{e}{e_0} \right) \sqrt{\sin^2 i - \frac{2}{3}}$$

where $e(\tau=0) = e_0$

Remember that this happens only in the asymptotic regime of ω .

But we can still have a general explanation using the angular momentum conservation

$$(92) \quad \sqrt{1-e^2} \cos i = \text{const}$$

In the regime of $i > 39^\circ$, i.e.,

$$\sin^2 i > \frac{2}{3} \Rightarrow \cos^2 i < \frac{1}{3} \text{ so this}$$

is the minimum value for i of which we get e_{\max}

(i - grow, e decreases and vice versa)

so

$$(93) \quad \sqrt{1-e_0^2} \cos i_0 = \sqrt{1-e_{\max}^2} \sqrt{\frac{3}{5}}$$

$$(94) \quad e_{\max} = \sqrt{1 - \frac{5}{3}(1-e_0^2)\cos^2 i_0} \approx \sqrt{1 - \frac{5}{3}\cos^2 i_0} \quad \text{if } e_0 \ll 1$$

if we will insert that into (91) to get an estimate for the time it takes

to get to e_{\max}

$$(95) \quad \tau = 0.42 h \frac{e_{\max}}{e_0} \sqrt{\sin^2 i_0 - \frac{3}{5}}$$

We set $i = i_0$ since in fact in this approx. (eq. 75) setting the value for τ (eq. 69) we have

$$(96) \quad t = \tau \frac{m_B}{C m_B m_B} b e^3 \sqrt{\frac{G m_B}{a^3}} h \frac{e_{\max}}{e_0}$$

Since this is an oscillatory motion e_{\max} is an q fold of e_0 , thus $h \frac{e_{\max}}{e_0} = 1$ and using $p = 2\pi \sqrt{\frac{a^3}{G m_B}}$ the inner orbit period we can write

$$(97) \quad t = \tau_0 \frac{m_B}{m_B} \frac{b e^3}{a^3} \frac{1}{2\pi} p$$

where we define

$$(98) \quad \tau_0 = 0.42 \sqrt{\sin^2 i_0 - 0.6}$$

so the spread time is

$$(99) \quad t = \tau_0 \frac{m_B}{m_B} \left(\frac{b e_0}{a}\right)^3 \frac{1}{2\pi} p \quad \leftarrow \begin{array}{l} \text{assuming} \\ b e_0 = a \text{ outer orbit} \\ \text{is close} \\ \text{to circular} \end{array}$$

This actually gives a lower estimation in general

$$(100) \quad \boxed{t_{\text{lower}} \approx \left(\frac{a e}{a}\right)^3 \frac{m_B}{m_B} p}$$

The functions that describe the evolution of θ_{osc} are actually elliptical integral, thus the period is set by the extremum values

② Let's find in general $i_{\text{max}}, e_{\text{max}}, i_{\text{min}}, e_{\text{min}}$ for a given i_0, e_0, w_0

first note that after some mathematical manipulation

multiply eq. (72) by $2e[s\sin^2 w - 2]s\sin^2 i$

eq. (74) by $[se^2\sin^2 w + 2(1-e^2)]2s\sin^2 i$

eq. (73) by $se^2\sin^2 i 2s\sin^2 w \cos w$

adding all we get:

$$(101) \quad [se^2\sin^2 w + 2(1-e^2)]2s\sin^2 i \frac{di}{dt} + se^2\sin^2 i 2s\sin^2 w \cos w \frac{dw}{dt} + 2e[s\sin^2 w - 2]s\sin^2 i \frac{de}{dt} = 0$$

and then after arranging we get

$$(102) \quad [se^2\sin^2 w + 2(1-e^2)]s\sin^2 i = \text{const.}$$

now we are looking for extremum

So:

(103) $\frac{de}{dt} = 0$, thus from eq. (65) it happens

(104) if $w = 0$, or, $\frac{\pi}{2}$ mod 2π

so if we define

$$(105) \quad X = [5e_0^2 \sinh^2 \omega_0 + 2(1-e_0^2)] \sinh^2 i_0$$

$$(106) \quad Y = (1-e_0^2) \cosh^2 i_0$$

we also know that X, Y are constant

so for

$$\omega = 0:$$

$$(107) \quad Y = (1-e_{\min}^2) \cosh^2 i_{\max}$$

$$(108) \quad X = (1-e_{\min}^2) \sinh^2 i_{\max}$$

$$(109) \quad \begin{cases} i_{\max} = \operatorname{arctg} \sqrt{\frac{X}{2Y}} \\ e_{\min}^2 = \sqrt{1 - \frac{Y}{\cosh^2 i_{\max}}} \end{cases}$$

and for $\omega = \frac{\pi}{2}$

$$(110) \quad X = [5e_{\max}^2 + 2(1-e_{\max}^2)] \sinh^2 i_{\min} = (3e_{\max}^2 + 2)(1 - \cosh^2 i_{\min})$$

$$(111) \quad Y = (1-e_{\max}^2) \cosh^2 i_{\min}$$

so e_{\max} and i_{\min} is the solution of this eqs.

$$(112) \quad 1 - 3e_{\max}^4 + (1 - 3Y + X) e_{\max}^2 + 2 - 2Y - X = 0$$