

## >>Execution

- Plotting 3 types of airfoils

### #1. NACA 0015

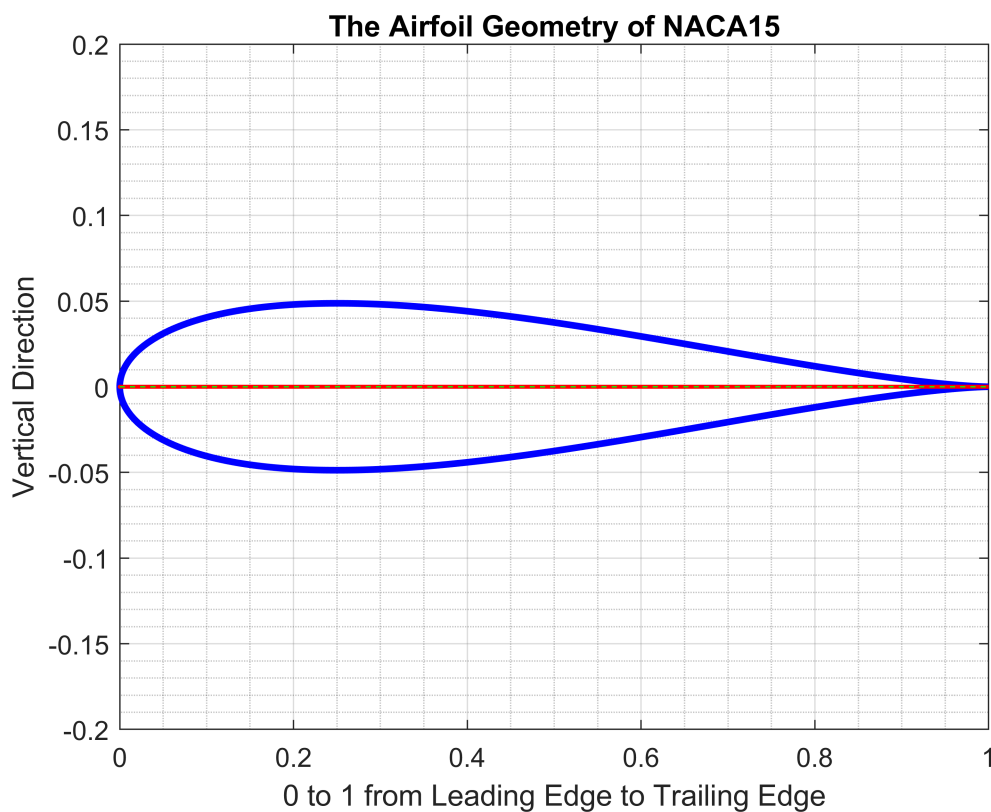
With

base shape coefficient = B = default (0)

taper exponent = P = default (0)

reflex parameter = R = default (0)

```
airfoil_plotter_func(0015,0,0,0);
```



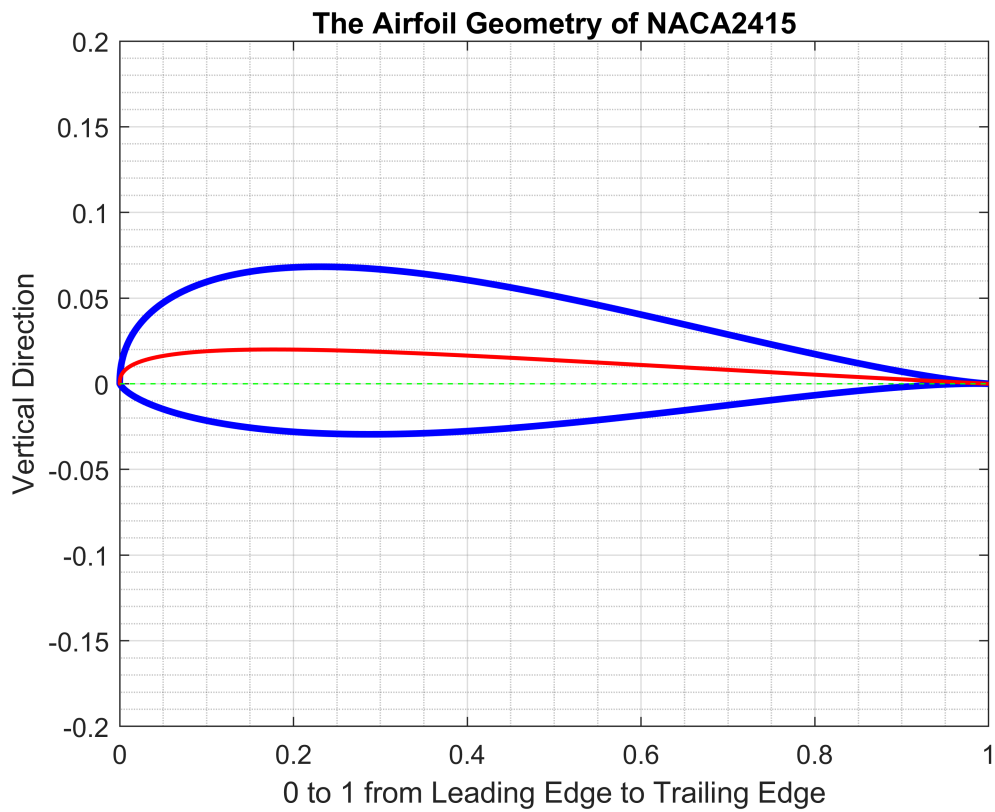
### #2. NACA 2415

base shape coefficient = B = default (0)

taper exponent = P = default (0)

reflex parameter = R = default (0)

```
airfoil_plotter_func(2415,0,0,0);
```



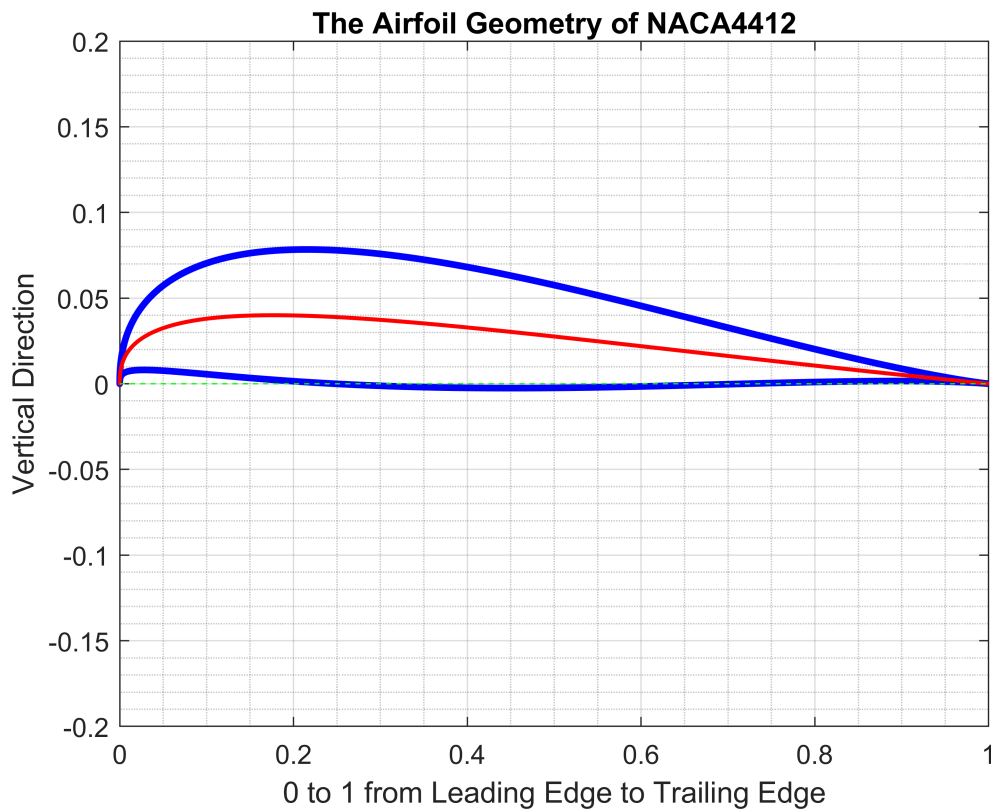
### #3. NACA 4412

base shape coefficient = B = default (0)

taper exponent = P = default (0)

reflex parameter = R = default (0)

```
airfoil_plotter_func(4412,0,0,0);
```



#### #4. NACA 2415 (NOT DEFAULT SETTINGS)

base shape coefficient =  $B = 2.5$

taper exponent =  $P = \text{default } 1.8$

reflex parameter =  $R = \text{default } 0.02$

```
airfoil_plotter_func(2415,2.5,1.5,0.01);
```

