

Note: $|\varepsilon_i| \leq 1$

Observe: 3 DOF and 4 variables

One constraint equation \Rightarrow $\mathcal{E}_{1}^{2} + \mathcal{E}_{2}^{2} + \mathcal{E}_{3}^{2} + \mathcal{E}_{4}^{2} = |$

If we know the Euler parameters describing orientation of b_i with respect to their original orientations \hat{a}_i , can determine

the equivalent single rotation $(\hat{\lambda}, \theta)$ $\lambda = \frac{\overline{\varepsilon}}{\sin(\frac{\theta}{2})} = \frac{\overline{\varepsilon}}{[1-\cos^{\frac{\theta}{2}}]^{\frac{1}{2}}} = \frac{\overline{\varepsilon}}{(1-\varepsilon^{\frac{\theta}{2}})^{\frac{1}{2}}}$ $\theta = 2\arccos(\varepsilon_{4}) \quad \lambda = \varepsilon_{1}^{\frac{\theta}{2}} + \varepsilon_{2}^{\frac{\theta}{2}} + \varepsilon_{3}^{\frac{\theta}{2}}$ $(\varepsilon_{1}^{2} + \varepsilon_{2}^{\frac{\theta}{2}} + \varepsilon_{3}^{\frac{\theta}{2}})^{\frac{\theta}{2}}$ $(\varepsilon_{1}^{2} + \varepsilon_{2}^{\frac{\theta}{2}} + \varepsilon_{3}^{\frac{\theta}{2}})^{\frac{\theta}{2}}$

What additional quantities might we want to express in terms of Euler parameters?

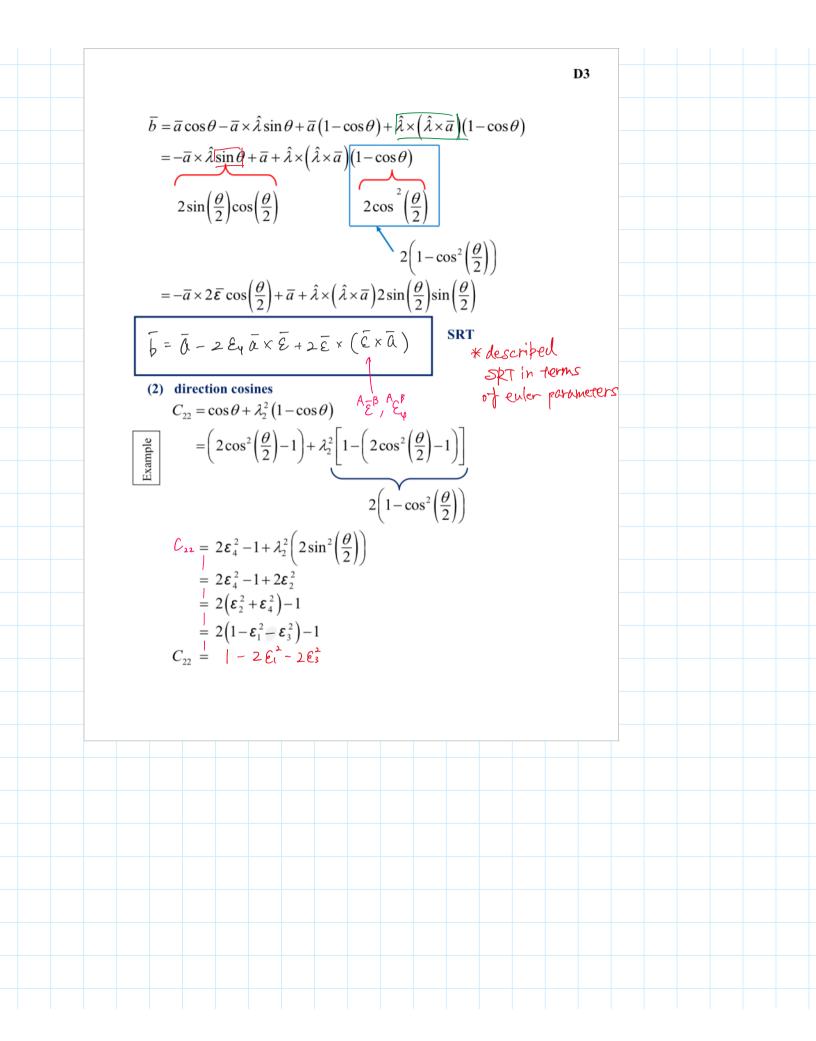
(1) SRT - Euler parameters expressed as SRT

$$\overline{b} = \overline{a}\cos\theta - \overline{a} \times \hat{\lambda}\sin\theta + \overline{a} \cdot \hat{\lambda}\hat{\lambda}(1 - \cos\theta)$$

Note: $(\overline{a} \cdot \hat{\lambda})\hat{\lambda} = (\hat{\lambda} \cdot \hat{\lambda})\overline{a} + \hat{\lambda} \times (\hat{\lambda} \times \overline{a})$ C A B A B C A B C

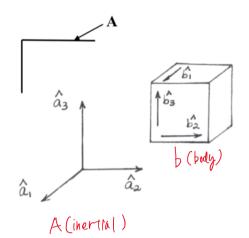
$$\overline{A} \times (\overline{B} \times \overline{C}) = (\overline{C} \cdot \overline{A}) \overline{B} - (\overline{A} \cdot \overline{B}) \overline{C}$$

$$\stackrel{\wedge}{\lambda} \times (\stackrel{\wedge}{\lambda} \times \stackrel{\wedge}{\lambda}) = (\stackrel{\wedge}{\lambda} \cdot \stackrel{\wedge}{\lambda}) \stackrel{\wedge}{\lambda} - (\stackrel{\wedge}{\lambda} \cdot \stackrel{\wedge}{\lambda}) \stackrel{\wedge}{\lambda}$$

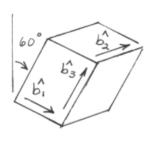


Example Assume that we have knowledge of initial and the (desired) final orientation, Determine the equivalent single rotation

Might represent, in simplified form, a "real" spacecraft
Problem: know current s/c orientation
assume you want to fire main thrusters to change orbit (main engines, no gimbal); need to reorient vehicle to fire in proper direction (change attitude with attitude control thrusters)



ρ̂λ = β̂ @ t = 30



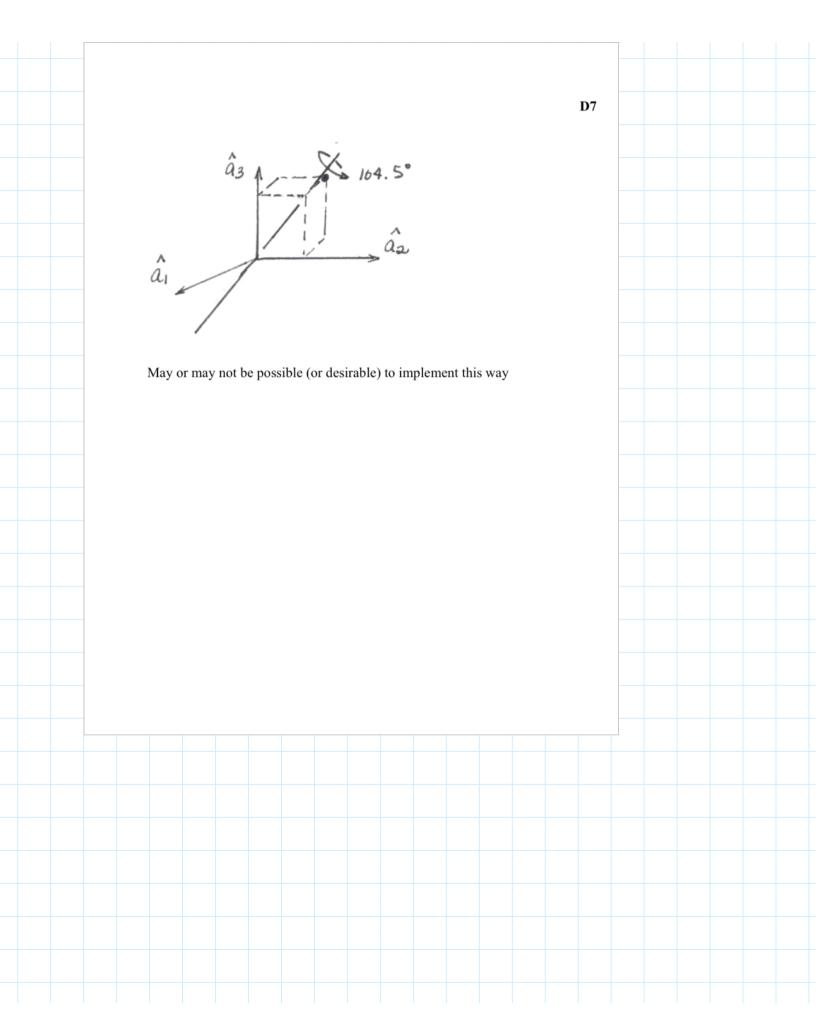
How to visualize this orientation change:

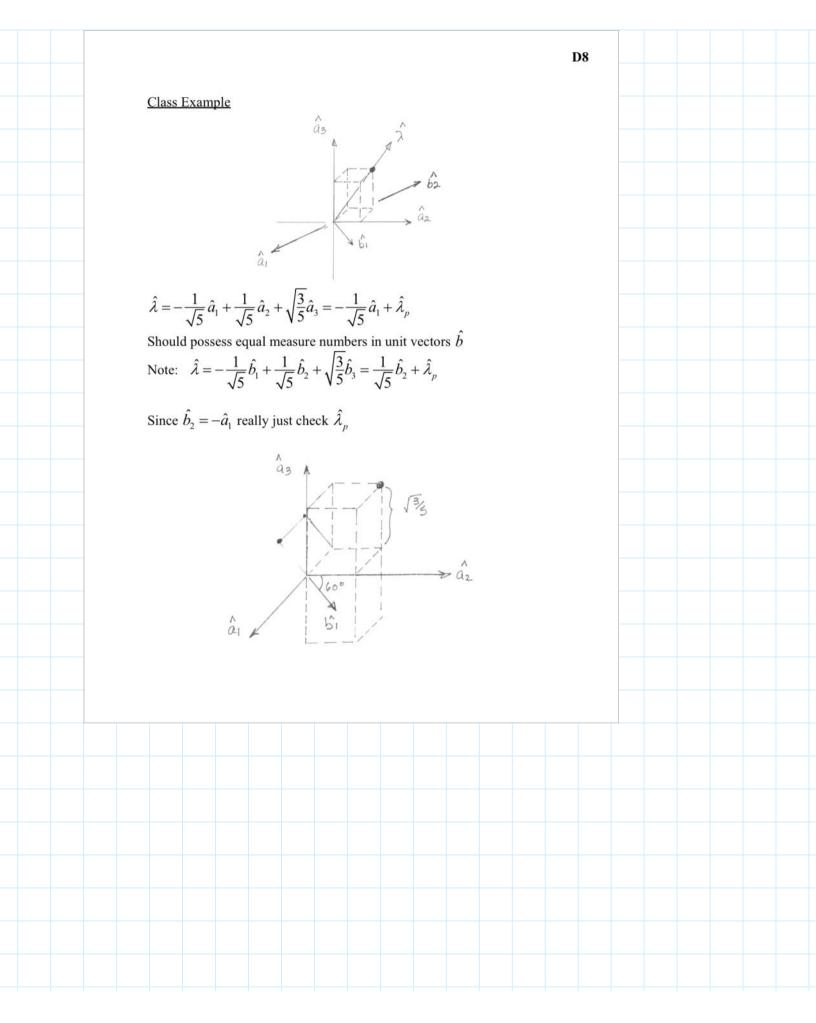
Sum of two rotations $\begin{cases} 90^{\circ} \text{ about } \hat{a}_{3} & -270^{\circ} \text{ about } \hat{a}_{3} \\ 60^{\circ} \text{ about } \hat{b}_{2} & -300^{\circ} \text{ about } \hat{b}_{1} \end{cases}$

(Great for analysis; great to visualize → will not actually want to implement change this way probably. Is it possible in one burn?)

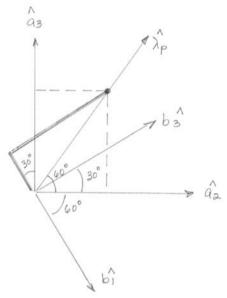
How do we describe this in terms of a cingle rotation?

Note:	Equivalent single rotation $(\hat{\lambda}, \theta)$ to do this job? Note: in this simple problem, possible to construct ${}^{A}C^{B}$ by inspection $(\hat{\lambda}, \theta)$ by $($							orthogonality			ondition ser su = 1	· Wh	Sqr	sum
$\theta = 2$	$c\cos^{-1}(\varepsilon_4)$	= (٥५,5°	,	0.2	U	j								





$$\hat{\lambda}_p = -\frac{1}{\sqrt{5}}\hat{b}_1 + \sqrt{\frac{3}{5}}\hat{b}_3 = \frac{1}{\sqrt{5}}\hat{a}_2 + \sqrt{\frac{3}{5}}\hat{a}_3$$



$$\begin{split} \hat{\lambda}_{p} &= -\frac{1}{\sqrt{5}} \hat{b}_{1} + \sqrt{\frac{3}{5}} \hat{b}_{3} \\ &= -\frac{1}{\sqrt{5}} \left(\cos 60^{\circ} \hat{a}_{2} - \sin 60^{\circ} \hat{a}_{3} \right) + \sqrt{\frac{3}{5}} \left(\cos 60^{\circ} \hat{a}_{3} + \sin 60^{\circ} \hat{a}_{2} \right) \\ &= \left(-\frac{1}{\sqrt{5}} \cos 60^{\circ} + \sqrt{\frac{3}{5}} \sin 60^{\circ} \right) \hat{a}_{2} + \left(\frac{1}{\sqrt{5}} \sin 60^{\circ} + \sqrt{\frac{3}{5}} \cos 60^{\circ} \right) \hat{a}_{3} \\ &= \frac{1}{\sqrt{5}} \hat{a}_{2} + \sqrt{\frac{3}{5}} \hat{a}_{3} \end{split}$$