HW1

Sunday, January 19, 2020

12:32 AM

AAE 440 - Spacecraft Attitude Dynamics

Problem Set 1 Due: 1/24/20

Problem 1: (a) The scalar triple product $(\overline{a} \times \overline{b}) \cdot \overline{c}$

scalar and vector product operations or by a change in the order of $\bar{a}, \bar{b}, \bar{c}$ are in cyclic order, i.e.,

$$(\overline{a} \times \overline{b}) \cdot \overline{c} = \overline{a} \cdot (\overline{b} \times \overline{c}) = \overline{b} \cdot (\overline{c} \times \overline{a}) = (\overline{c} \times \overline{a}) \cdot \overline{b}$$

Express these in subscript format to demonstrate that this statement is true.

(b) Use the permutation symbol with subscript format and prove that the following identity is also true

$$\left(\overline{a}\times\overline{b}\right)\times\left(\overline{c}\times\overline{d}\right)=\left[\left(\overline{a}\times\overline{b}\right)\bullet\overline{d}\right]\overline{c}-\left[\left(\overline{c}\times\overline{a}\right)\bullet\overline{b}\right]\overline{d}$$

Problem 2: Assume that vectors \overline{q} , \overline{r} , \overline{s} orthonormal triad \hat{n}_1 , \hat{n}_2 , \hat{n}_3

$$\overline{q} = q_i \hat{n}_i$$

$$\overline{r} = r_j \hat{n}_j$$

$$\overline{s} = s_{\alpha} \hat{n}_{\alpha}$$

(a) Evaluate the following quantities and express each in summation (subscript) format, in terms of the vector basis \hat{n}_i .

Example:
$$\overline{v} = \overline{q} \cdot \overline{r} \ \overline{q} \times \overline{s}$$

$$= q_i \hat{n}_i \cdot r_j \hat{n}_j \ q_\ell \hat{n}_\ell \times s_m \hat{n}_m$$

$$= q_i r_j \delta_{ij} \ q_\ell \ s_m \varepsilon_{\ell m p} \hat{n}_p$$

$$\overline{v} = q_i r_i \ q_\ell \ s_m \varepsilon_{\ell m p} \hat{n}_p$$

(i)
$$\overline{\overline{G}} = \overline{r} \, \overline{s} + (\overline{r} + \overline{s}) \overline{q}$$

(ii)
$$\overline{a} = \overline{r} \cdot \overline{\overline{G}} + \overline{q} \times \overline{s}$$

(iii)
$$\overline{\overline{H}} = \overline{\overline{G}} - \overline{r} \, \overline{a} \cdot \overline{s} \, \overline{q}$$

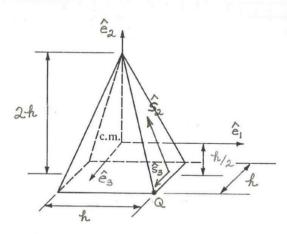
(iv)
$$\overline{\overline{R}} = \overline{\overline{G}} \times \overline{s} - \overline{\overline{H}}$$

(v)
$$c = \overline{q} \cdot \overline{\overline{R}} \cdot \overline{s} + \overline{q} \cdot \overline{r} \times \overline{\overline{G}} \cdot \overline{s}$$

(b) Let
$$\overline{q}$$
, \overline{r} , \overline{s}
 $\overline{q} = \hat{n}_2 - 2\hat{n}_3$
 $\overline{r} = 3\hat{n}_1 + 0.5\hat{n}_2 - \hat{n}_3$
 $\overline{s} = 4\hat{n}_1 - 2\hat{n}_2$

Evaluate each of the quantities in (a) using the subscript definitions.

Problem 3: The inertia matrix can be written in dyadic form which is particularly useful when inertia information is required in various vector bases. Below is a right rectangular pyramid of total mass m.



Right rectangular pyramid

- (a) Determine the inertia dyadic for the pyramid P, relative to point Q, i.e. $\bar{I}^{P/Q}$, for unit vectors $\hat{e}_1, \hat{e}_2, \hat{e}_3$
- (b) Use the similarity transformation to transform direction cosine matrix $\overline{\overline{I}}^{P/Q}$ to vector basis \hat{s} ; write the corresponding dyadic.
- (c) An inertia element in vector basis. relative to the <u>same point</u>, can also be determined through the relation

$$I_{ij} = \hat{s}_i \bullet \overline{\overline{I}} \bullet \hat{s}_j$$

Demonstrate that this expression produces the same result as the similarity transformation.