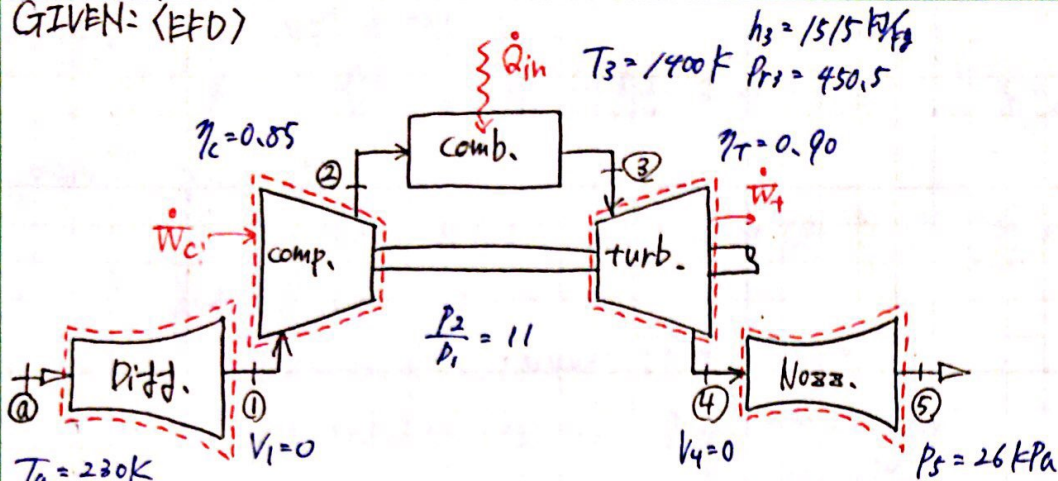


GIVEN: (EFD)



FIND: (a)  $P_i$  in kPa,  $T_i$  in K for each state  
 (b)  $\dot{Q}_{in}$  in kW  
 (c)  $V_5$  in m/s

ASSUMP: open sys, SSSF, I-DUF,  $\Delta KE = \Delta PE = 0$  (\*KE except Diff. inlet and Nozz. outlet), air standard sys, ideal gas, comb. only heat addition no pressure drop, Diff., Nozz., Comp., Turb.  $\Rightarrow \dot{Q} = 0$ , Diff. & Nozz.  $\Rightarrow \dot{W} = 0$

EQN:  $\frac{dm}{dt}_{sys} = \sum \dot{m}_i - \sum \dot{m}_e$ ,  $\frac{dE}{dt}_{sys} = \dot{Q} - \dot{W} + \sum \dot{m}_i (h_i + p_i + ke_i) - \sum \dot{m}_e (h_e + p_e + ke_e)$

SOLN:

(a)

<state 1> for state a, we can find  $h_a = 230.0 \text{ kJ/kg}$

Since Diffuser  $\dot{W} = 0$  &  $\dot{Q} = 0$

$$0 = \dot{Q}_{in} - \dot{W}_{out} + \dot{m} (h_a - h_1) + \dot{m} \frac{V_a^2}{2} \cdot \frac{1}{1000}$$

$$\therefore h_1 = h_a + \frac{V_a^2}{2000} = 230.0 \text{ kJ/kg} + \frac{220^2}{2000} = 254.2 \text{ kJ/kg}$$

from this using interpolation

$$T_1 = (254.2 - 250.0) \frac{\text{kJ/kg}}{\text{kJ/kg}} \frac{660 - 250 \text{ K}}{(260.0 - 250.0) \frac{\text{kJ/kg}}{\text{kJ/kg}}} + 250 \text{ K} \approx 254.2 \text{ K}$$

likewise

$$Pr_1 = (254.2 - 250.0) \frac{\text{kJ/kg}}{\text{kJ/kg}} \frac{0.8405 - 0.7329}{(260.0 - 250.0) \frac{\text{kJ/kg}}{\text{kJ/kg}}} + 0.7329$$

$$\approx 0.7781$$

$$T_1 = 254 \text{ K}$$

since diff. is  $\Delta S = 0$

$$P_1 = P_2 \cdot \frac{P_{r1}}{P_{r2}} = 26 \text{ kPa} \cdot \frac{0.9781}{0.5477} \approx 36.9 \text{ kPa}$$

$$P_1 = 36.9 \text{ kPa}$$

<State 2> from CR

$$P_2 = 11 P_1 = 11 (36.9 \text{ kPa}) \approx 406 \text{ kPa}$$

$$P_2 = 406 \text{ kPa}$$

and if comp. is  $\Delta S = 0$

$$P_{r2} = P_{r1} \frac{P_2}{P_1} = (0.9781) \cdot 11 \approx 8.559$$

the corresponding  $h_{2s}$  is, from interpolation

$$h_{2s} = (8.559 - 8.411) \cdot \frac{(513.6 - 503.3) \text{ kJ/kg}}{8.031 - 8.411} + 503.3 \text{ kJ/kg}$$

$$\approx 505.8 \text{ kJ/kg}$$

now with  $\eta_c$

$$h_2 = \frac{1}{\eta_c} (h_{2s} - h_1) + h_1 = \frac{1}{0.85} (505.8 - 254.2) \text{ kJ/kg} + 254.2 \text{ kJ/kg}$$

$$= 550.2 \text{ kJ/kg}$$

now the corresponding  $T_2$  with interpolation is

$$T_2 = (550.2 - 544.6) \text{ K} \cdot \frac{550 \text{ k} - 540 \text{ k}}{(555.0 - 544.6) \text{ kJ/kg}} + 540 \text{ K}$$

$$\approx 545.4 \text{ K}$$

$$T_2 = 545 \text{ K}$$

<State 3> since combustion has no p loss  $P_3 = P_2$

<State 4> since

$$P_3 = 406 \text{ kPa}$$

$$\dot{w}_T + \dot{w}_c = 0$$

$$h_3 - h_4 + h_1 - h_2 = 0$$

$$h_4 = h_1 - h_2 + h_3 = (254.2 - 550.2 + 1515) \text{ kJ/kg}$$

$$= 1219 \text{ kJ/kg}$$

interpolate to find  $T_4$

$$T_4 = (1219 - 1208) \text{ K} \cdot \frac{(1160 - 1140) \text{ K}}{(1231 - 1208) \text{ kJ/kg}} + 1140 \text{ K}$$

$$\approx 1150 \text{ K}$$

$$T_4 = 1150 \text{ K}$$

from  $\eta_T = 0.90$

$$h_{4s} = h_3 - \frac{1}{\eta_T} (h_3 - h_4) = 1515 \text{ kJ/kg} - \frac{1}{0.90} (1515 - 1219) \text{ kJ/kg}$$

$$\approx 1186 \text{ kJ/kg}$$



then  $h_{4s} \approx h|_{T=1120} \Rightarrow p_{r4} = (1126-1184) \frac{\text{kJ}}{\text{kg}} \cdot \frac{(493.1-129.7)}{(1201-1184) \frac{\text{kJ}}{\text{kg}}} + 119.7 \approx 180.8$   
 for  $h_4$  (actual)

therefore

$$p_4 = \frac{p_{r4}}{p_{r3}} p_3 = \frac{180.8}{450.5} (406 \text{ kPa}) \approx 162.9 \text{ kPa}$$

$$p_4 = 163 \text{ kPa}$$

<state 5> since Nozz.  $\Delta S = 0$ 

$$p_{r5} = \frac{p_5}{p_4} p_{r4} = \frac{26 \text{ kPa}}{163 \text{ kPa}} (180.8) \approx 28.84$$

the corresponding temp.  $T_5$ :

$$p_{r5} \approx p_r|_{T=700\text{K}} \Rightarrow h_5 = 713.3 \text{ kJ/kg}$$

$$T_5 = 700 \text{ K}$$

(b) for comb. &lt;COE&gt;

$$0 = \dot{Q}_{in} - \dot{W} + \dot{m}(h_2 - h_3)$$

$$\dot{Q}_{in} = \dot{m}(h_3 - h_2) = (25 \text{ kg/s})(1515 - 550.2) \text{ kJ/kg}$$

$$= 24120 \text{ kW}$$

$$\dot{Q}_{in} = 24.1 \times 10^3 \text{ kW}$$

(c) For Nozz. &lt;COE&gt;

$$0 = \dot{Q}_{45} - \dot{W}_{45} + \dot{m}(h_4 - h_5) - \dot{m} \cdot \frac{V_5^2}{2} \cdot \frac{1}{1000}$$

$$V_5 = \sqrt{2000(h_4 - h_5)}$$

$$= \sqrt{2000(1219 \text{ kJ/kg} - 713.3 \text{ kJ/kg})}$$

$$\approx 1005 \text{ m/s}$$