

AAE 364: Controls System Analysis

HW 6: Steady State Error Evaluation

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(1)

B-5-21. Consider the following characteristic equation:

$$s^4 + 2s^3 + (4 + K)s^2 + 9s + 25 = 0$$

Using the Routh stability criterion, determine the range of K for stability.

using Routh's Stability Criterion (RSC)

s^4	1	$4+K$	25
s^3	2	9	0
s^2	$\frac{2K-1}{2}$	25	
s^1	$\frac{18K-109}{2K-1}$	0	
s^0	25		

$$\frac{2(4+K) - 1 \times 9}{2} = \frac{2K-1}{2}$$

$$\frac{\left(\frac{2K-1}{2}\right)9 - 50}{\frac{2K-1}{2}} = \frac{18K-109}{2K-1}$$

now, the 1st column has to all be positive; thus,

$$\frac{2K-1}{2} > 0 \quad \text{and} \quad \frac{18K-109}{2K-1} > 0$$

$$K > \frac{1}{2} = 0.5$$

$$K > \frac{109}{18} \approx 6.056$$

$$K \in \left(\frac{109}{18}, \infty\right)$$

B-5-23. Consider the satellite attitude control system shown in Figure 5-80(a). The output of this system exhibits continued oscillations and is not desirable. This system can be stabilized by use of tachometer feedback, as shown in Figure 5-80(b). If $K/J = 4$, what value of K_h will yield the damping ratio to be 0.6?

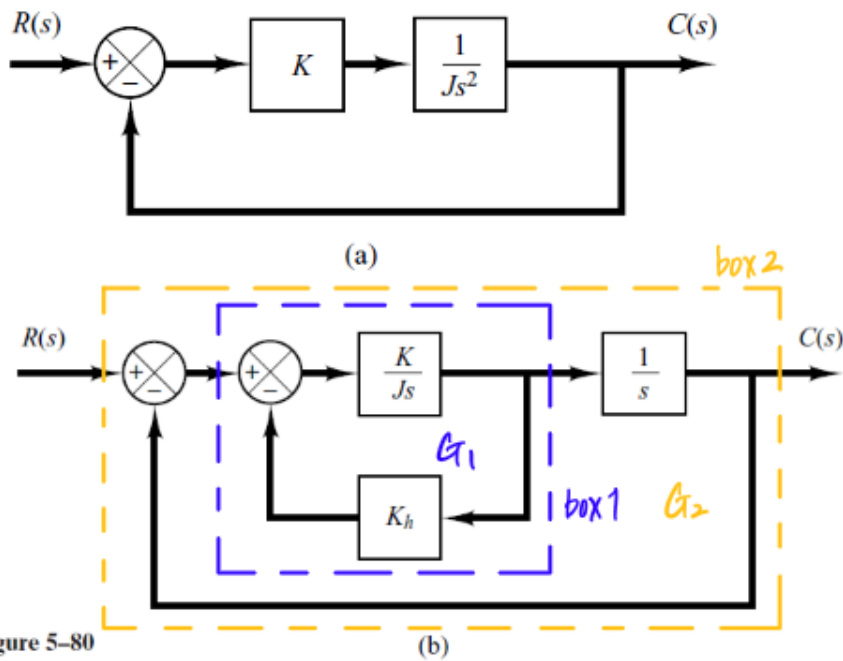


Figure 5-80
(a) Unstable satellite attitude control system;
(b) stabilized system.

from the figure above, by dissecting it into 2 boxes

for box 1

$$G_1 = \frac{\frac{K}{J s}}{1 + \left(\frac{K}{J s}\right)(K_h)} = \frac{K}{J s^2 + K K_h}$$

then for box 2

$$G_2 = \frac{G_1 \cdot \frac{1}{s}}{1 + G_1 \cdot \frac{1}{s}} = \frac{G_1}{s + G_1} = \frac{\frac{K}{J s^2 + K K_h}}{s + \frac{K}{J s^2 + K K_h}}$$

$$= \frac{K}{J s^2 + K K_h s + K}$$

thus,

$$\frac{C(s)}{R(s)} = G_2 = \frac{K}{J s^2 + K K_h s + K} = \frac{K/J}{s^2 + K_h (K/J) s + K/J}$$

since $K/J = 4$

$$\frac{C(s)}{R(s)} = \frac{4}{s^2 + 4 K_h s + 4}$$

here

$$\omega_n = \sqrt{4} = 2 \quad (>0)$$

then

$$4 K_h = 2 \zeta \omega_n$$

if $\zeta = 0.6$

$$K_h = \frac{2 \times 0.6 \times 2}{4} = 0.6$$

$$K_h = 0.6$$

B-5-26. Consider a unity-feedback control system with the closed-loop transfer function

$$\frac{C(s)}{R(s)} = \frac{Ks + b}{s^2 + as + b}$$

Determine the open-loop transfer function $G(s)$.

Show that the steady-state error in the unit-ramp response is given by

$$e_{ss} = \frac{1}{K_v} = \frac{a - K}{b}$$

the open-loop tf can be deduced from the following relation

$$\frac{G(s)}{1 + G(s)} = \frac{Ks + b}{s^2 + as + b}$$

$$G(s)(s^2 + as + b) = Ks + b + (Ks + b)G(s)$$

$$G(s)(\cancel{s^2 + as + b} - Ks - b) = Ks + b$$

$$G(s)[s^2 + (a - K)s] = Ks + b$$

$$\therefore G(s) = \frac{Ks + b}{s^2 + (a - K)s}$$

$$G(s) = \frac{Ks + b}{s(s + a - K)}$$

now

$$\begin{aligned} \frac{E(s)}{R(s)} &= \frac{R(s) - C(s)}{R(s)} = 1 - \frac{C(s)}{R(s)} = 1 - \frac{Ks + b}{s^2 + as + b} \\ &= \frac{s^2 + as + b - Ks - b}{s^2 + as + b} = \frac{s^2 + (a - K)s}{s^2 + as + b} \end{aligned}$$

thus, if $R(s) \rightarrow$ unit ramp input $\frac{1}{s^2}$

$$\begin{aligned} e_{ss} &= \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} s \left(\frac{1}{s^2} \right) \left[\frac{s^2 + (a - K)s}{s^2 + as + b} \right] \\ &= \lim_{s \rightarrow 0} \frac{s + (a - K)}{s^2 + as + b} = \frac{a - K}{b} \end{aligned}$$

(2)

2. Consider the unity-feedback control system in Figure 1 with the following open-loop transfer function:

$$G(s) = \frac{2}{(s+10)^2}$$

Compute the steady state error of the closed-loop system with respect to the step input $r(t) = 5$.

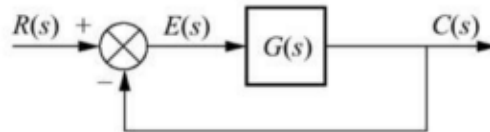


Figure 1: Unity-feedback system

The CLTF is

$$\frac{C(s)}{R(s)} = \frac{G(s)}{1 + G(s)} = \frac{\frac{2}{(s+10)^2}}{1 + \frac{2}{(s+10)^2}} = \frac{2}{s^2 + 20s + 102}$$

then

$$\frac{E(s)}{R(s)} = 1 - \frac{C(s)}{R(s)} = \frac{s^2 + 20s + 102 - 2}{s^2 + 20s + 102} = \frac{s^2 + 20s + 100}{s^2 + 20s + 102}$$

$$\text{if } r(t) = 5 \xrightarrow{\mathcal{L}} R(s) = \frac{5}{s}$$

thus

$$\begin{aligned} e_{ss} &= \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} s E(s) \\ &= \lim_{s \rightarrow 0} s \left(\frac{5}{s} \right) \left(\frac{s^2 + 20s + 100}{s^2 + 20s + 102} \right) \\ &= \lim_{s \rightarrow 0} \frac{5(s^2 + 20s + 100)}{s^2 + 20s + 102} \\ &= \frac{500}{102} \\ &\approx 4.9020 \end{aligned}$$

$$e_{ss} = 4.9020$$

(3)

3. Consider the unity-feedback control system in Figure 1 with the following open-loop transfer function:

$$G(s) = \frac{3}{s(s+1)(s+2)}$$

Compute the steady state error of the closed-loop system with respect to the ramp input $r(t) = 3t$.

The CLTF becomes

$$\begin{aligned}\frac{C(s)}{R(s)} &= \frac{G(s)}{1+G(s)} = \frac{\frac{3}{s(s+1)(s+2)}}{\frac{s(s+1)(s+2) + 3}{s(s+1)(s+2)}} = \frac{3}{s(s+1)(s+2) + 3} \\ &= \frac{3}{s^3 + 2s^2 + s^2 + 2s + 3} = \frac{3}{s^3 + 3s^2 + 2s + 3}\end{aligned}$$

$$\text{if } r(t) = 3t \xrightarrow{\mathcal{L}} R(s) = \frac{3}{s^2}$$

$$\begin{aligned}E(s) &= 1 - \frac{C(s)}{R(s)} \\ &= 1 - \frac{3}{s^3 + 3s^2 + 2s + 3} = \frac{s^3 + 3s^2 + 2s}{s^3 + 3s^2 + 2s + 3}\end{aligned}$$

thus,

$$\begin{aligned}e_{ss} &= \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} sE(s) \\ &= \lim_{s \rightarrow 0} s \left(\frac{3}{s^2} \right) \left(\frac{s^3 + 3s^2 + 2s}{s^3 + 3s^2 + 2s + 3} \right) \\ &= \lim_{s \rightarrow 0} \frac{3(s^2 + 3s + 2)}{s^3 + 3s^2 + 2s + 3} \\ &= \frac{6}{3} = 2\end{aligned}$$

$$e_{ss} = 2$$