

Orbits in Three Dimensions

Previously, we considered everything in 2D

Now, try 3D problems



some background necessary
to define an orbit in space

First, define coordinate systems (3D) to help



many available

Two basic types for us to use:

- (1) **Ecliptic System** – fundamental plane is the plane of the \oplus 's orbit about the Sun (latitude, longitude)

x_ϵ, y_ϵ

- (2) **Equatorial System** – Fundamental plane is the plane of the body's equator (right ascension, declination)

x, y

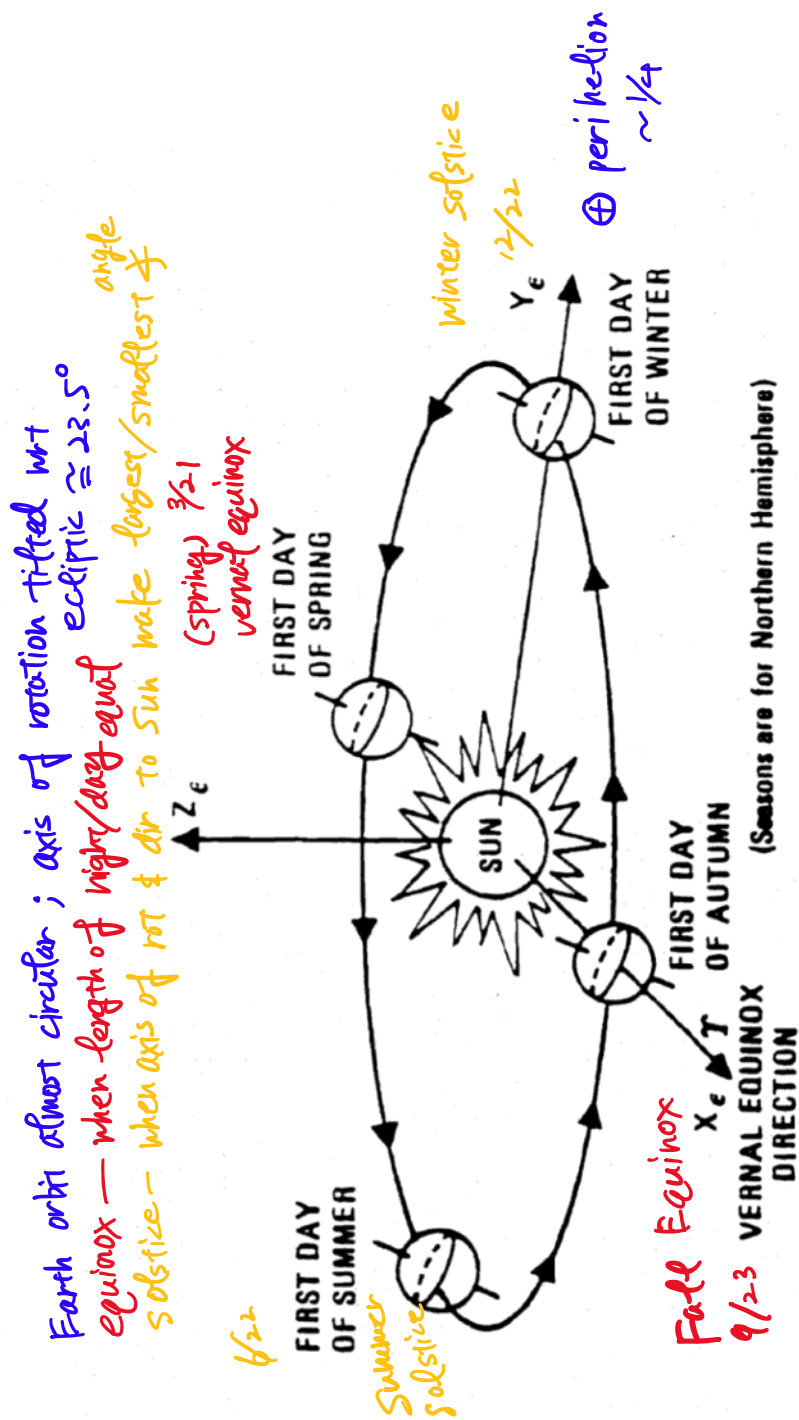
Obliquity of ecliptic (ϵ) – inclination of ecliptic with respect to equator

To effectively use a coordinate system, reference directions must be known and understood; we need a fixed reference direction in the fundamental plane from which measurements are made

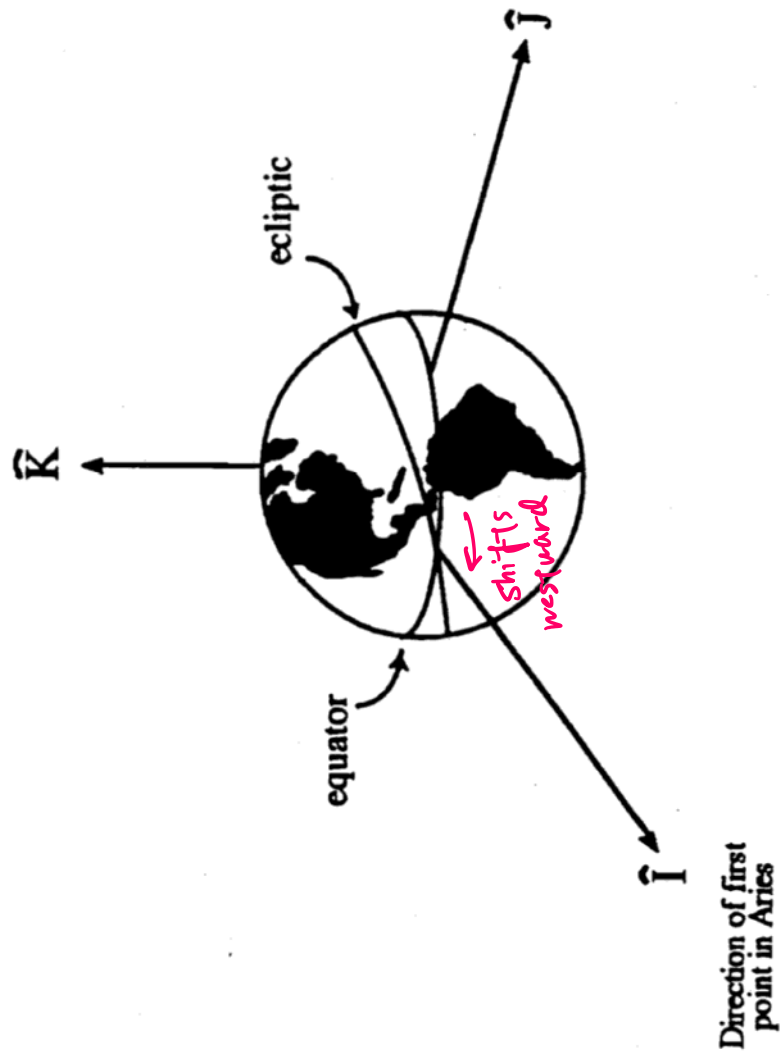
→ vernal equinox

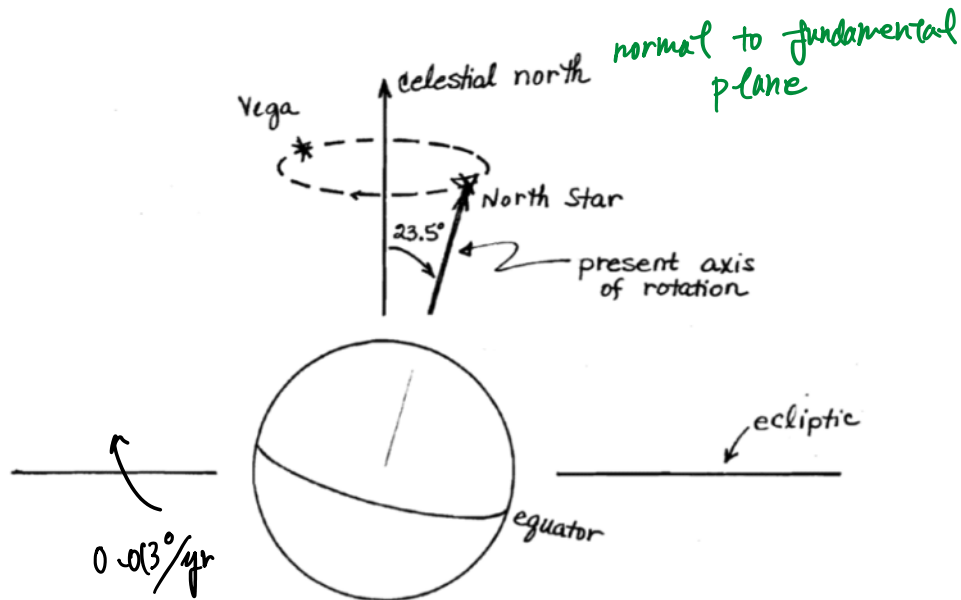
$\chi_e = \chi$ intersection of ecliptic and Earth equatorial plane

diagram



γ first point in Aries \leftarrow the ram
 ~ 4000 yrs since vernal equinox pointing in dir
 of constellation Aries
 (currently direction is toward Pisces) $\leftarrow \sim 90 B.C.$





“precession of the equinoxes” – change in direction of Earth spin axis

Caused by perturbing forces on its attitude, i.e., \odot and \sphericalangle gravity forces

These apply a precessing motion (same as the precessing motion of a spinning top or a torque-free rigid body)

Known as early as 2nd century BC to Greek astronomer Hipparchus

Time for complete precession is 26,000 years

$$\begin{aligned} \text{shift thru} &= \frac{360^\circ}{26000 \text{ yrs}} = \frac{4}{5} \frac{\text{arcmin}}{\text{yr}} \\ &= 0.13^\circ/\text{yr} \end{aligned}$$

Consequence

1. Cataloging of celestial objects must refer to a specific date \rightarrow epoch \rightarrow currently 0.0 hrs
2. We will assume Υ fixed; reasonable over the relatively short intervals of interest



Reference System

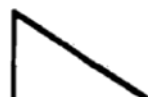
ecliptic equatorial

\hat{x} direction of the vernal equinox
 \hat{z} normal to fundamental plane; + north
 $\hat{y} = \hat{z} \times \hat{x}$

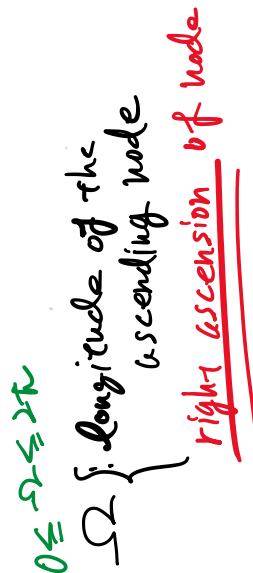
So, to locate s/c in space :

- (1) locate s/c in orbit (θ^* , E , M)
- (2) identify orientation of orbit within orbit plane (ω);
size and shape of orbit (a , e)
- (3) identify orientation of orbit plane in space (Ω , i)

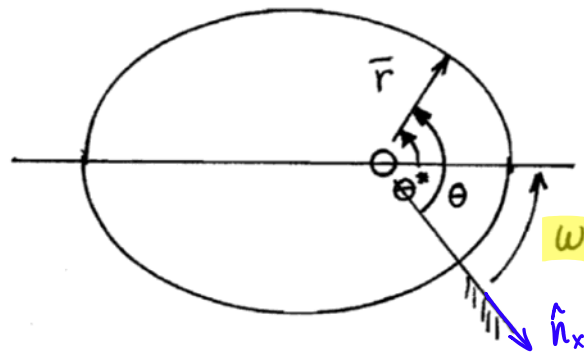
time
Argument of periaapsis
reference to some fundamental plane



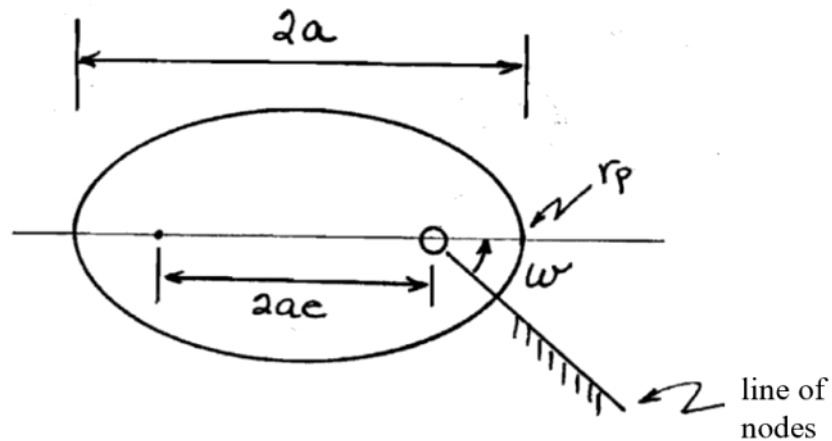
fundamental plane
ance
ne
of nodes

 λ : Indefinition $\delta \leq \lambda \leq \tau$

- (1) Locate s/c in orbit: time $\leftarrow M, E, \theta^*$ one of these

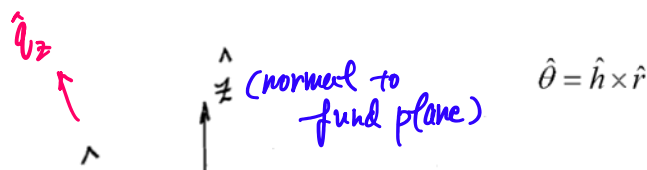


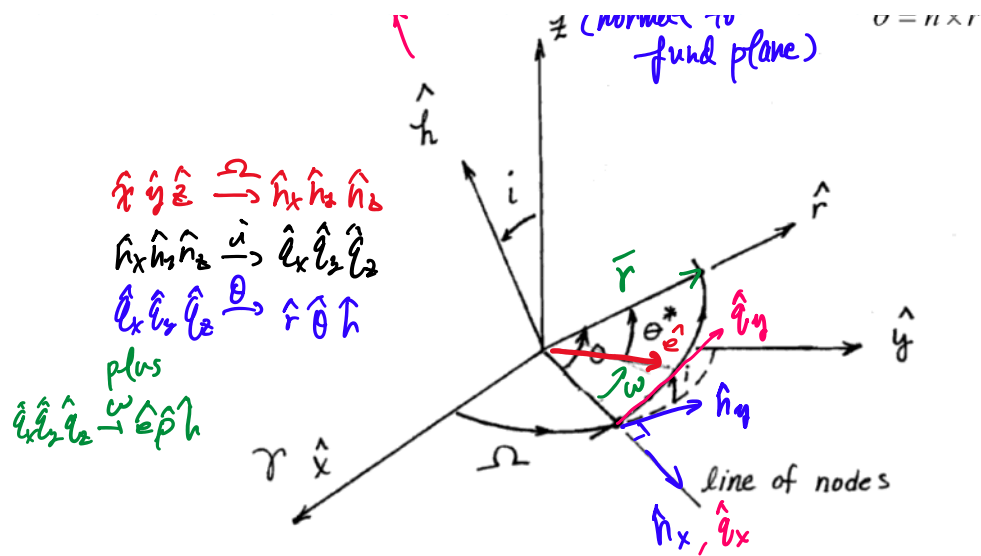
- (2) Within orbit plane : orbit size and shape (a, e)
orbit orientation within plane (ω)



- (3) orientation of orbit plane : Ω, i
orbital elements : $a, e, i, \Omega, \omega, M(x_p)$
 \uparrow
 E, θ^*

write $\bar{r}, \bar{v} \rightarrow \hat{r}, \hat{\theta} / \hat{e}, \hat{p}$ 18





$$\hat{v}_x, \hat{v}_y, \hat{v}_z \xrightarrow{\omega} \hat{e}, \hat{p}, \hat{h}$$

$$\hat{x} \hat{y} \hat{z} \xrightarrow{\Omega} \hat{x}_h \hat{y}_h \hat{z}_h$$

$$\hat{x}_h \hat{y}_h \hat{z}_h \xrightarrow{i} \hat{x}_p \hat{y}_p \hat{z}_p$$

$$\hat{x}_p \hat{y}_p \hat{z}_p \xrightarrow{\omega} \hat{r} \hat{\theta} \hat{h}$$

$$\text{plus} \\ \hat{x}_p \hat{y}_p \hat{z}_p \xrightarrow{\omega} \hat{e} \hat{p} \hat{h}$$

$$\omega \left\{ \begin{array}{l} \text{argument of periapsis} \\ 0 \leq \omega \leq 2\pi \end{array} \right.$$

$$\Omega + \omega = \varpi \left\{ \begin{array}{l} \text{longitude of periapsis} \end{array} \right.$$

$$\varpi + \theta^* = L \left\{ \begin{array}{l} \text{true longitude at epoch} \\ L = \Omega + \omega + \theta^* \end{array} \right.$$

Ω, i, θ

3-1-3 (body-two) Euler sequence \Rightarrow transformation matrix

$\hat{z} - \hat{h}_x - \hat{h}$

	\hat{r}	$\hat{\theta}$	\hat{h}
\hat{x}	$c_\Omega c_\theta - s_\Omega c_i s_\theta$	$-c_\Omega s_\theta - s_\Omega c_i c_\theta$	$s_\Omega s_i$
\hat{y}	$s_\Omega c_\theta + c_\Omega c_i s_\theta$	$-s_\Omega s_\theta + c_\Omega c_i c_\theta$	$-c_\Omega s_i$
\hat{z}	$s_i s_\theta$	$s_i c_\theta$	c_i

