

# AAE 364: Control Systems Analysis

## HW8: Controller Design and Root Locus Analysis

Dr. Sun

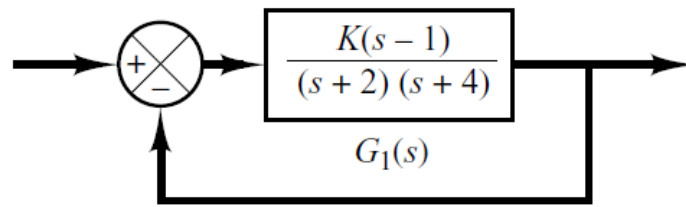
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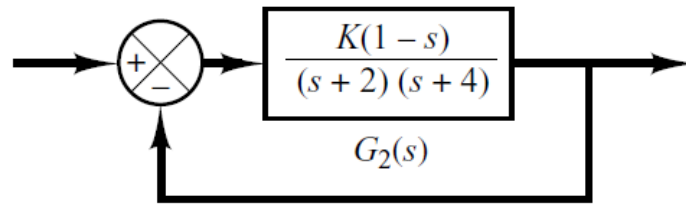
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**B-6-12.** Plot root-locus diagrams for the nonminimum-phase systems shown in Figures 6-102(a) and (b), respectively.



(a)



(b)

**Figure 6-102** (a) and (b) Nonminimum-phase systems.

(a)

This a feedback system. Then the CE becomes

$$CE := 1 + K \frac{(s-1)}{(s+2)(s+4)} \quad \text{where } L(s) = \frac{s-1}{(s+2)(s+4)}$$

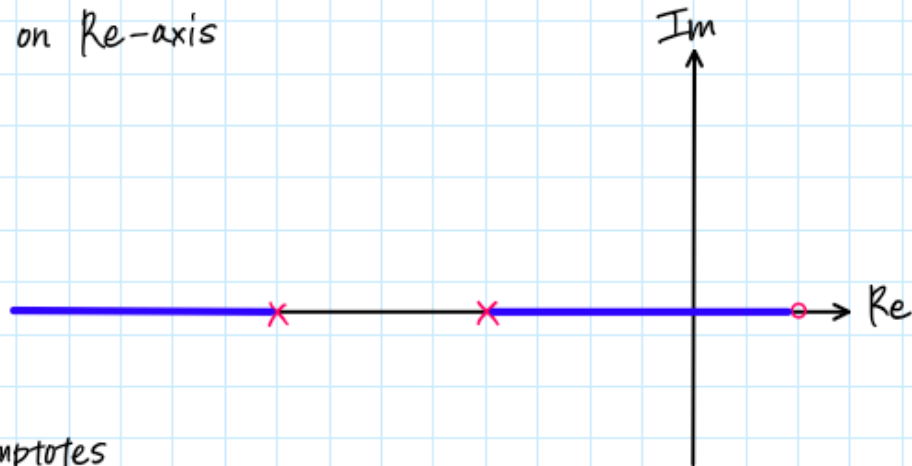
<i> Poles and Zeros

$$\text{Poles: } (s+2)(s+4) = 0 \Rightarrow s = -2, -4 \quad \therefore n = 2$$

$$\text{Zeros: } s-1 = 0 \Rightarrow s = 1 \quad \therefore m = 1$$

<ii> Symmetry TRUE

<iii> RL on Re-axis



<iv> Asymptotes

$$\theta_a = \frac{180^\circ + 360^\circ l}{n - m} = \frac{180^\circ + 360^\circ l}{2 - 1} = 180^\circ + 360^\circ l \quad (l=0)$$

$$\theta_a = 180^\circ$$

and

$$\sigma_a = \frac{\sum p_i - \sum z_i}{n - m} = \frac{(-2) + (-4) - 1}{2 - 1} = -7$$

<V> Break-in/away points

$$\frac{d}{ds} \left[ -\frac{1}{L(s)} \right] = 0 \Rightarrow \frac{d}{ds} \left[ -\frac{(s+2)(s+4)}{s-1} \right] = 0$$

$$\frac{-s^2 + 2s + 14}{(s-1)^2} = 0$$

$$\begin{cases} \hat{s}_1 = 4.8730 \\ \hat{s}_2 = -2.8730 \end{cases} \rightarrow \begin{array}{l} \text{do not reside in} \\ s \in [-2, 1] \\ s \in (-\infty, -4] \end{array}$$

$\hat{s}_1$  &  $\hat{s}_2$  are **NOT** break-in/away points  $\leftarrow$

<vi> Angle of departure

since we know that all poles and zeros are on the Re-axis departure/arrival angles are unnecessary

<vii> Intersection of RL with Im-axis

$$1 + \hat{k}L(j\hat{\omega}) = 0$$

$$1 + \hat{k} \cdot \frac{j\hat{\omega} - 1}{(j\hat{\omega} + 2)(j\hat{\omega} + 4)} = 0$$

solving this we get

$$\begin{array}{l} \text{Re: } -\hat{k} = \hat{\omega}^2 - 8 \\ \text{Im: } \hat{k}\hat{\omega} = -6\hat{\omega} \end{array}$$

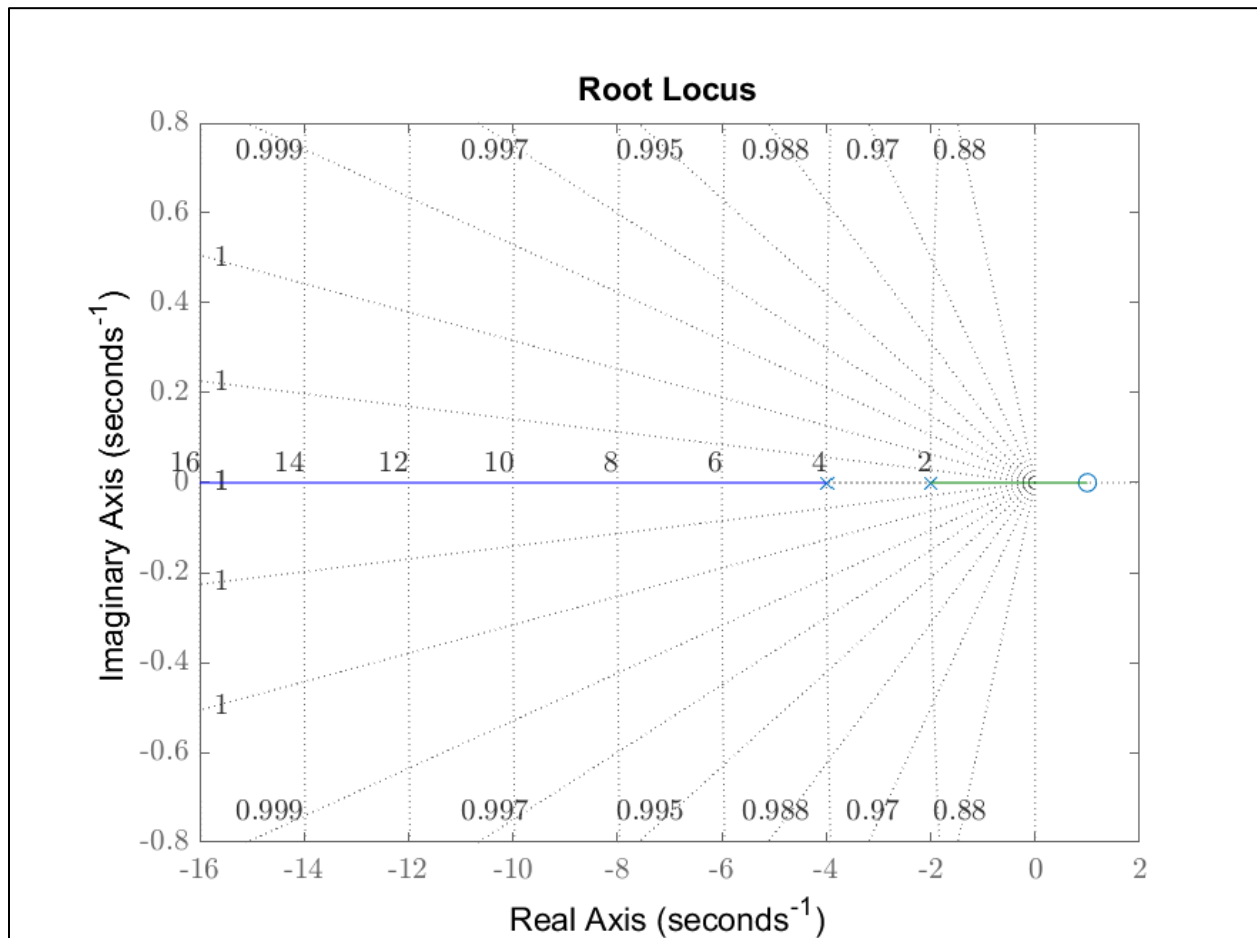
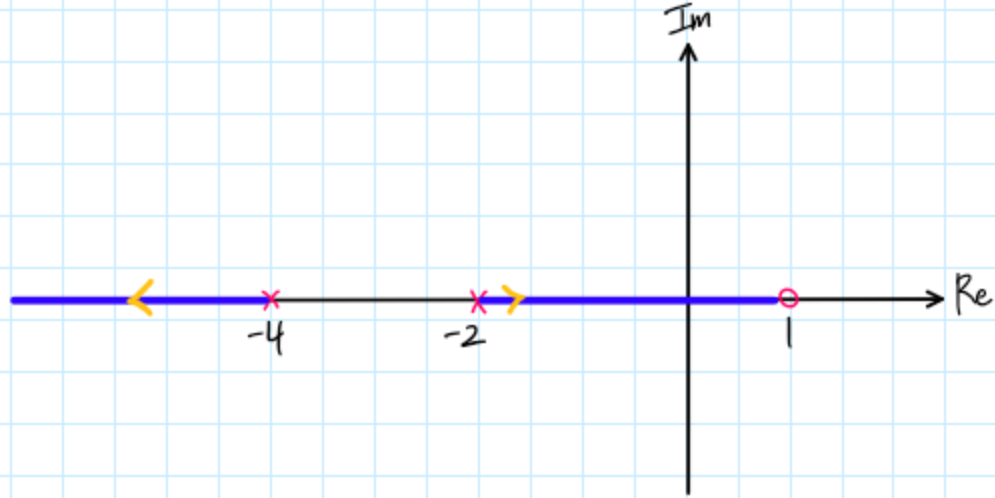
$$\hat{k} = \begin{bmatrix} -6 \\ -6 \\ 8 \end{bmatrix}$$

$$\hat{\omega} = \begin{bmatrix} -3.7417 \\ 3.7417 \\ 0 \end{bmatrix}$$

(\* $\hat{k} > 0$ )

but there is **NO** intersection with Im-axis

from  $\angle i \sim \angle vii$  **RL** becomes



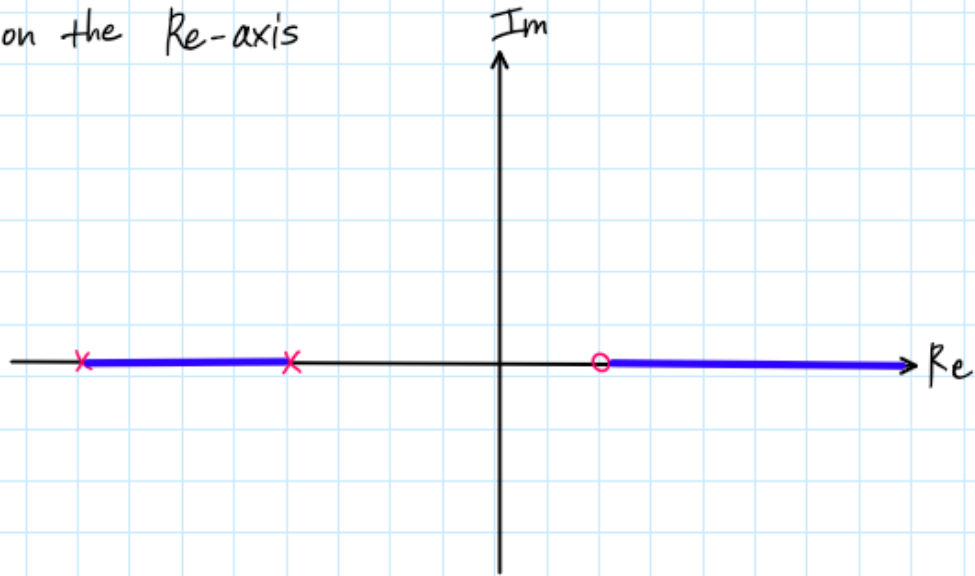
(b) This becomes a **positive** feedback

$$CE := 1 - K \frac{(s-1)}{(s+2)(s+4)} \quad \text{where} \quad L(s) = \frac{s-1}{(s+2)(s+4)}$$

<i> Poles and zeros **same as (a)**

<ii> Symmetry **TRUE**

<iii> RL on the Re-axis



<iv> Asymptotes

$$\theta_a = \frac{360^\circ l}{n-m} = 360^\circ l \quad l = 0$$

$$\text{and} \quad \sigma_a = -\frac{\sum p_i - \sum z_i}{n-m} = -1$$

<v> Break-in/away points **same as (a)**

$$\hat{s}_1 = 4.8730$$

$$\hat{s}_2 = -2.8730$$

→ resides on the blue domain on the figure above. So **both** are break-in/away points

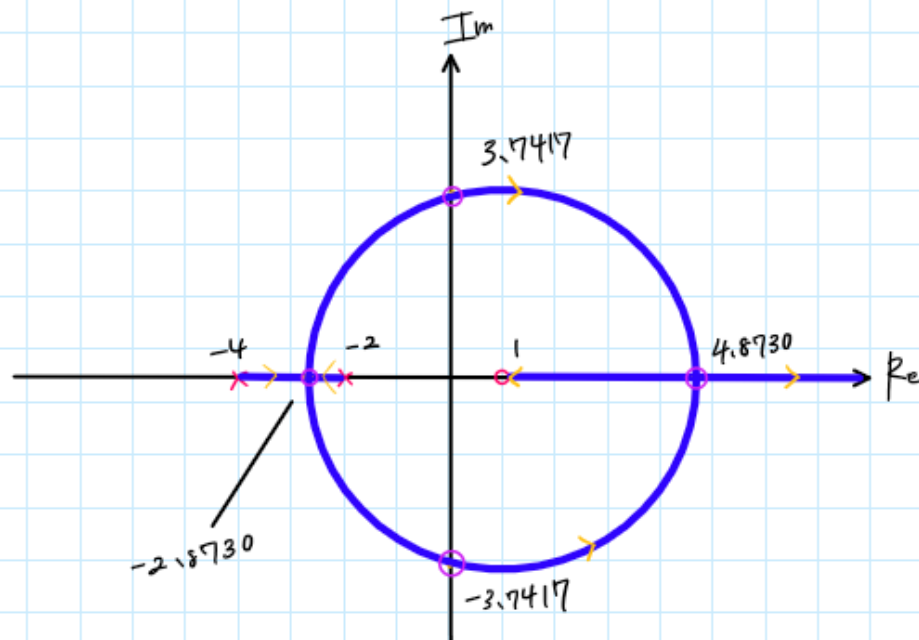
<vi> Angle of departure

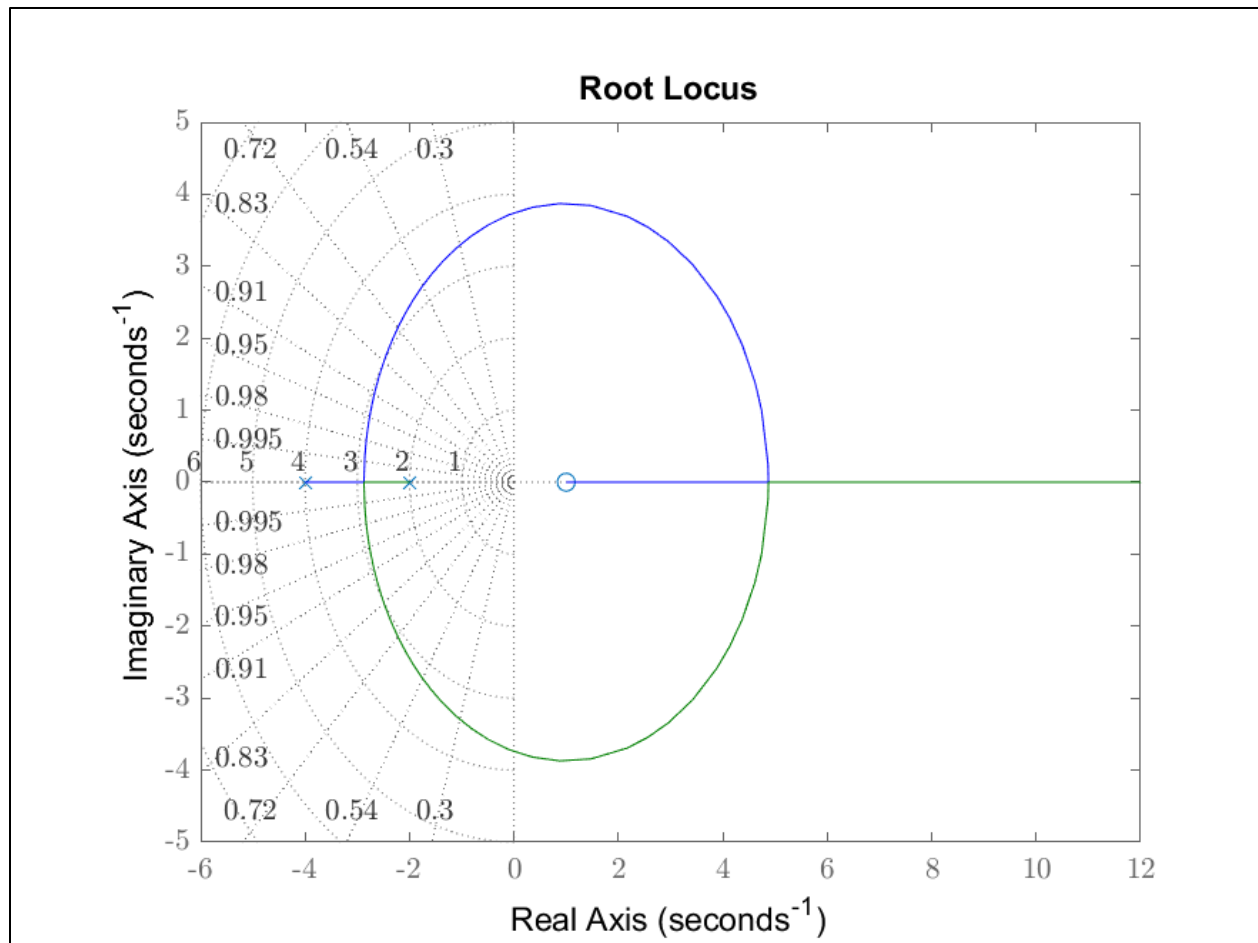
unnecessary for the same reason as (a)

<vii> Intersection of RL with Im-axis  
→ Same as (a)

$$\hat{k} = \begin{bmatrix} -6 \\ -6 \\ 8 \end{bmatrix}, \quad \hat{\omega} = \begin{bmatrix} -3.7417 \\ 3.7417 \\ 0 \end{bmatrix} \quad (*\hat{k} < 0)$$

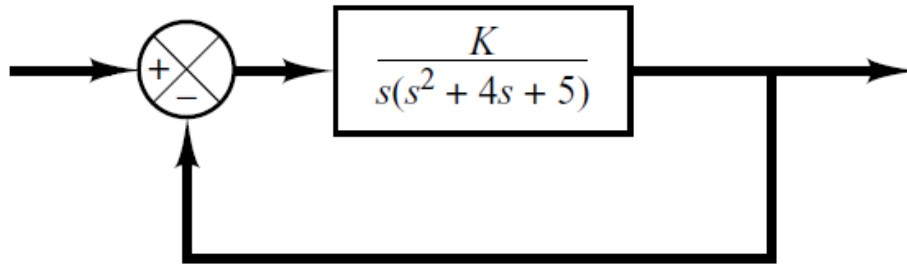
from <i> ~ <vii> RL becomes (nonminimum phase system)







**B-6-14.** Consider the system shown in Figure 6-104. Plot the root loci for the system. Determine the value of  $K$  such that the damping ratio  $\zeta$  of the dominant closed-loop poles is 0.5. Then determine all closed-loop poles. Plot the unit-step response curve with MATLAB.



**Figure 6-104** Control system.

(a) Find the RL

Since this is a negative feedback

$$CF := 1 + k \frac{1}{s(s^2+4s+5)} \quad \text{where} \quad L(s) = \frac{1}{s(s^2+4s+5)}$$

<i> Poles and Zeros

$$\text{Poles: } s(s^2+4s+5)=0 \Rightarrow s=0, -2 \pm j \Rightarrow n=3$$

$$\text{Zeros: none} \Rightarrow m=0$$

<ii> Symmetry **TRUE**

<iii> RL on Re-axis



<iv> Asymptotes

$$\theta_a = \frac{180^\circ + 360^\circ l}{n-m} = 60^\circ + 120^\circ l \quad l=0, 1, 2$$

$$\theta_a = 60^\circ, 180^\circ, 300^\circ$$

$$\text{and} \quad \sigma_a = \frac{\sum P_i - \sum Z_i}{n-m} = \frac{0 + (-2+j) + (-2-j)}{3} = -\frac{4}{3}$$

<v> Break-in/away points

$$\frac{d}{ds} \left( -\frac{1}{L(s)} \right) = -\frac{d}{ds} (s(s^2+4s+5)) = 0$$

$$-3s^2 - 8s - 5 = 0$$

$$\Rightarrow \hat{S}_1 = -1.6667, \hat{S}_2 = -1$$

<vi> Angle of departure

defining a hypothetical point  $s^d$  in the proximity of  $-2+j$

$$\angle L(s^d) = -180^\circ = 0 - \arg(-2+j-0) - \arg(-2+j-(-2-j))$$

$$- \theta_d$$

$$\Rightarrow \theta_d = -63.4349^\circ$$

<vii> Intersection of RL with Im-axis

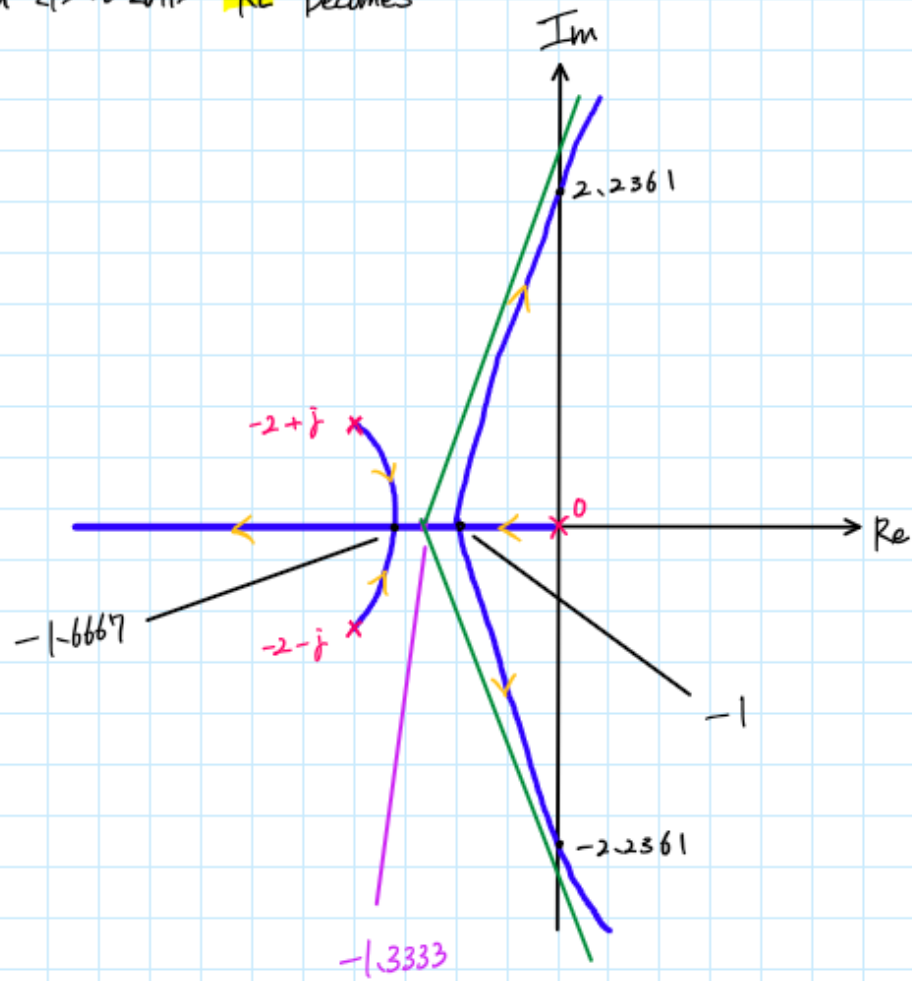
$$1 + \hat{k} L(j\hat{\omega}) = 0 \Rightarrow 1 + \hat{k} \frac{1}{j\hat{\omega}(-\hat{\omega}^2 + 4j\hat{\omega} + 5)} = 0$$

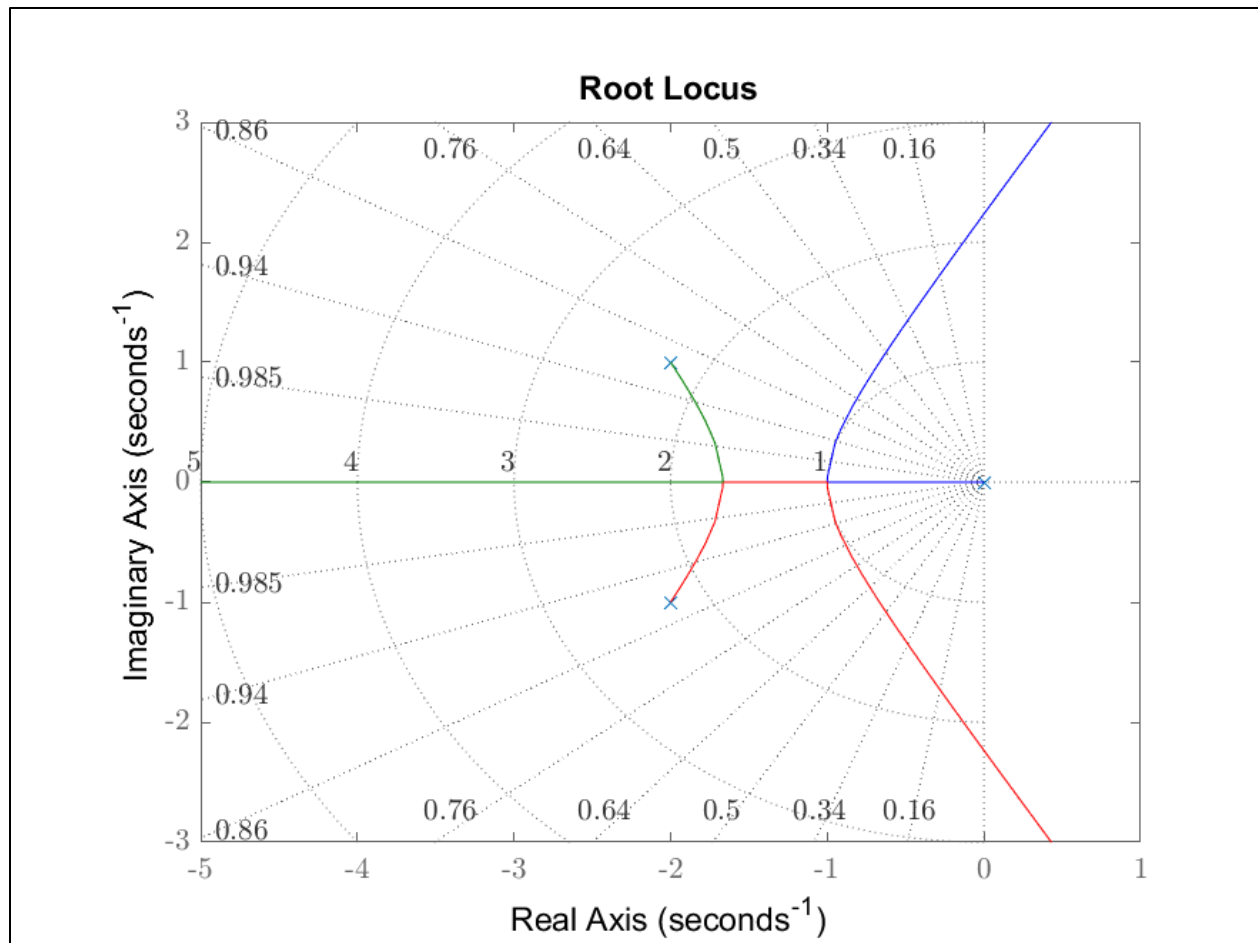
$$\Rightarrow \hat{k} = j\hat{\omega}^3 + 4\hat{\omega}^2 - 5j\hat{\omega}$$

$$\Rightarrow \begin{array}{l} \text{Im:} \quad 0 = \hat{\omega}^3 - 5\hat{\omega} \\ \text{Re:} \quad \hat{k} = 4\hat{\omega}^2 \end{array}$$

$$\hat{k} = \begin{bmatrix} 0 \\ 20 \\ 20 \end{bmatrix} \quad \hat{\omega} = \begin{bmatrix} 0 \\ -2.2361 \\ 2.2361 \end{bmatrix}$$

from  $\angle i \sim \angle vii$  **RL** becomes





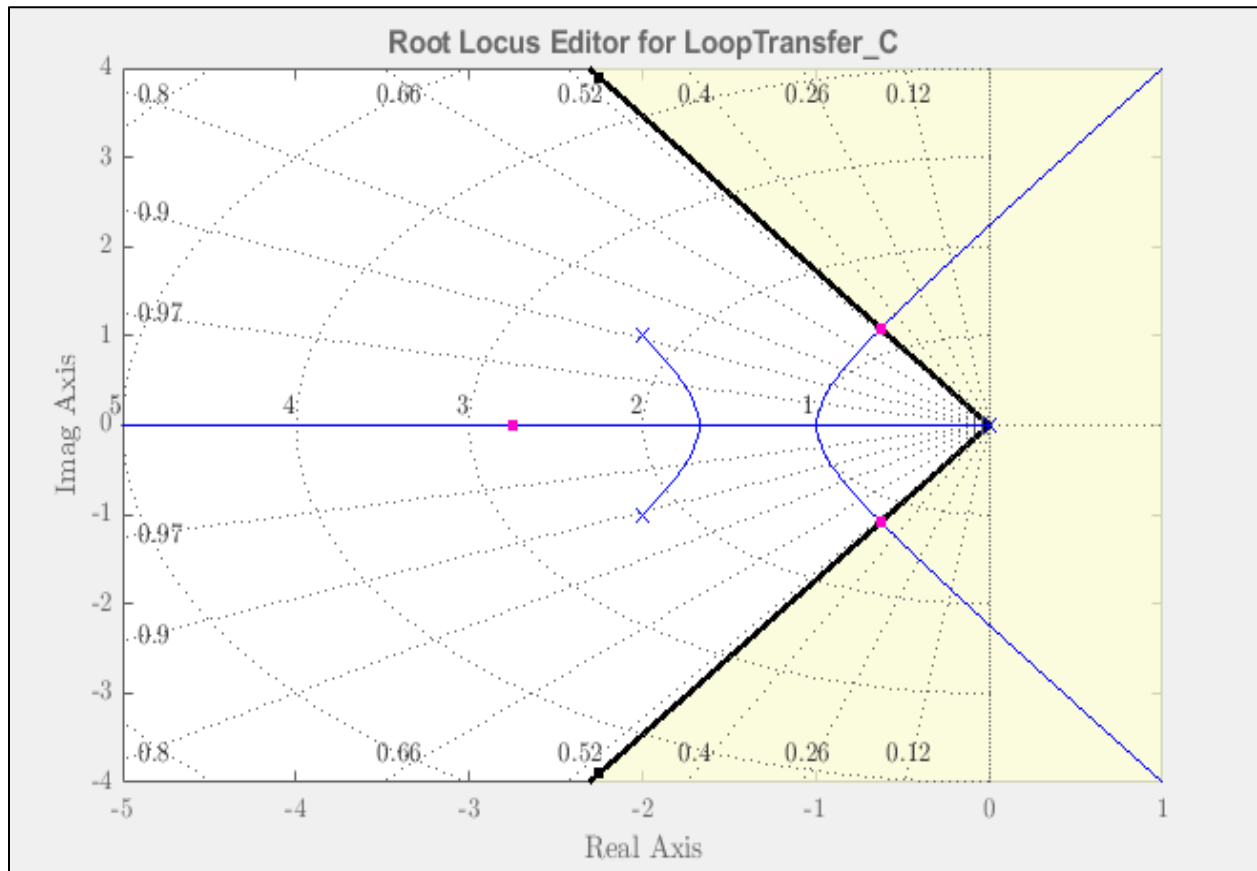
(b) Find value of K

if the damping ratio  $\zeta = 0.5$  then  $\Rightarrow \cos \theta = 0.5 \Rightarrow \theta = \pm 60^\circ$

now from this we want to find the location of the pole for this  $\zeta$  on the RL. To do this we will use MATLAB's **Control System Designer**

→ call `>> controlSystemDesigner(tf(num,den));`

→ drag the complex poles to where it satisfies  
 $\zeta = \cos \theta = 0.5$



→ for this the pole is  $p = -0.625 \pm 1.08j$   
 → then find the gains by

calling `rlocfind()` or looking at the gains corresponding to the design you have in your control system designer

now we obtain  $K = 4.2852$

Analytically we can obtain  $K$  by

→ define a pole  $p = -\sigma + \omega j$   
 → this pole must satisfy the two

$$(1) \quad 1 + KL(p) = 0$$

$$(2) \quad \omega = (\tan 60^\circ) \cdot \sigma = \sqrt{3} \sigma$$

→ solving this we get

$$1 + K \frac{1}{(-\sigma + \omega j)((-\sigma + \omega j)^2 + 4(-\sigma + \omega j) + 5)} = 0$$

$$K = \frac{-\omega^3 j + 3\omega^2 \sigma - 4\omega^2 + 3\omega \sigma^2 j - 8\omega \sigma j + 5\omega j}{-\sigma^3 + 4\sigma^2 - 5\sigma}$$

$$\Rightarrow K = 8\sigma^3 - 8\sigma^2 - 5\sigma + 5\sqrt{3}\sigma j - 8\sqrt{3}\sigma^2 j$$

$$\text{Re: } K = -8\sigma^3 + 8\sigma^2 + 5\sigma$$

$$\text{Im: } 0 = 8\sqrt{3}\sigma^2 - 5\sqrt{3}\sigma$$

$$\therefore \sigma = \begin{bmatrix} 0 \\ 0.6250 \end{bmatrix} \quad K = \begin{bmatrix} 0 \\ 4.2969 \end{bmatrix}$$

$$\omega = \begin{bmatrix} 0 \\ 1.0825 \end{bmatrix}$$

$$\begin{array}{l} K_{\text{comp}} = 4.2852 \\ K_{\text{TH}} = 4.2969 \end{array} \rightarrow \frac{4.2969}{s(s^2 + 4s + 5)} \quad \text{OLTF (open loop)}$$

The CLTF becomes

$$\frac{G(s)}{1 + G(s)} = \frac{4.2969}{s^3 + 4s^2 + 5s + 4.2969}$$

because this is a unit-step response

$$\text{input } R(s) = \frac{1}{s}$$

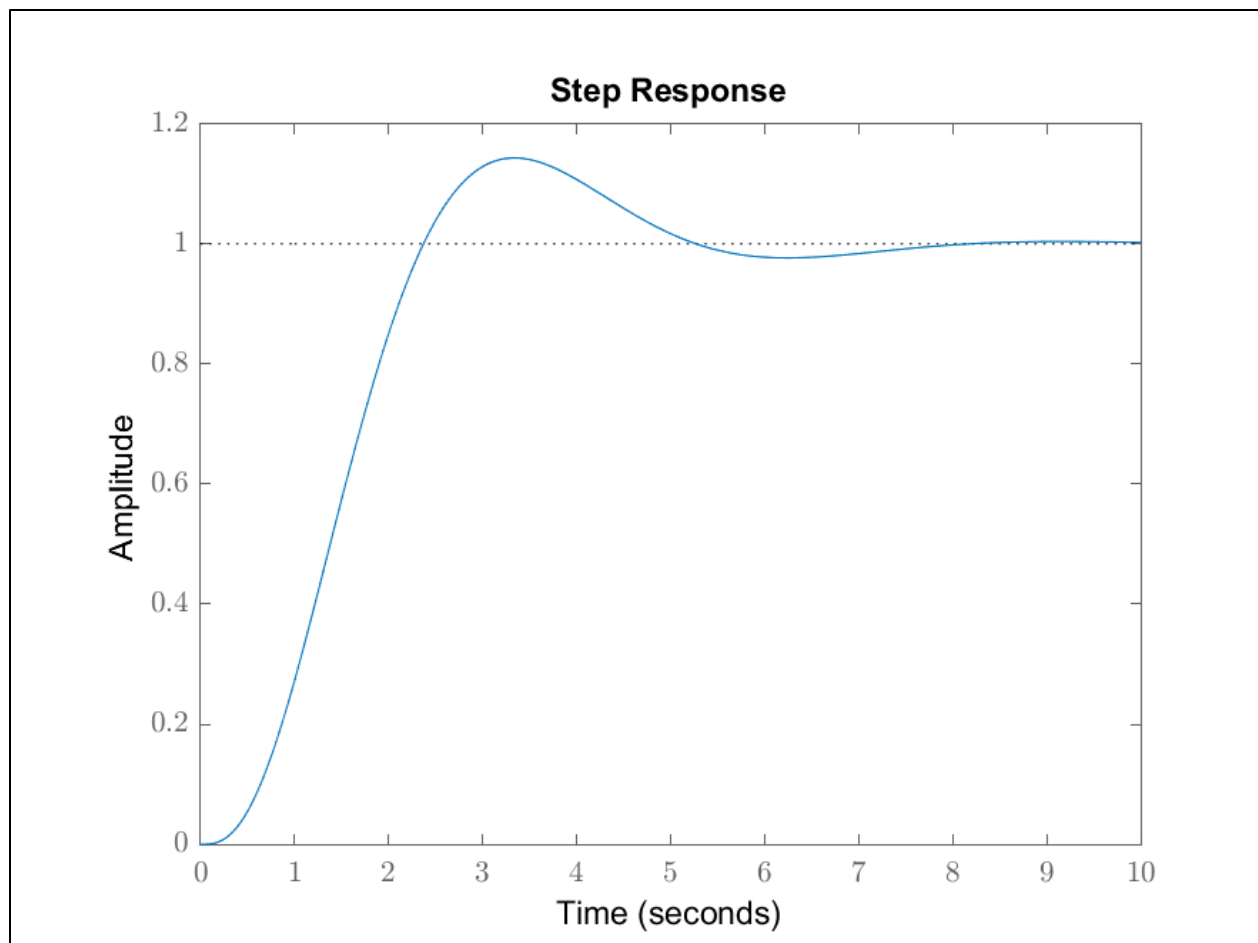
$$\Rightarrow Y(s) = \frac{4.2969}{s(s^3 + 4s^2 + 5s + 4.2969)}$$

↓

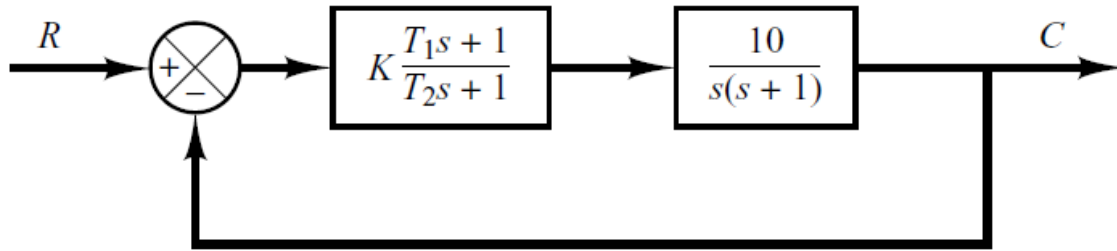
$$\mathcal{L}^{-1}[Y(s)] = 1 - \frac{66}{91} e^{-\frac{5}{8}t} \left[ \cos\left(\frac{5\sqrt{3}}{8}t\right) + \frac{8\sqrt{3}}{9} \sin\left(\frac{5\sqrt{3}}{8}t\right) \right] - \frac{25}{91} e^{-\frac{11}{4}t}$$

↓





**B-6-15.** Determine the values of  $K$ ,  $T_1$ , and  $T_2$  of the system shown in Figure 6-105 so that the dominant closed-loop poles have the damping ratio  $\zeta = 0.5$  and the undamped natural frequency  $\omega_n = 3$  rad/sec.



**Figure 6-105** Control system.

$$OLTF := G_c G = \frac{10K(T_1 s + 1)}{s(s+1)(T_2 s + 1)}$$

since this is a negative feedback loop

$$CE := 1 + G_c G = 1 + \frac{10K(T_1 s + 1)}{s(s+1)(T_2 s + 1)} = 0$$

$$\begin{aligned} \Rightarrow s(s+1)(T_2 s + 1) + 10K(T_1 s + 1) &= 0 \\ (s^2 + s)(T_2 s + 1) + 10KT_1 s + 10K &= 0 \\ T_2 s^3 + (T_2 + 1)s^2 + (10KT_1 + 1)s + 10K &= 0 \end{aligned}$$

if  $\zeta = 0.5 = \cos\theta$  and  $\omega_n = 3 \text{ rad/s}$   
the complex pole locations are

$$p = -\omega_n \zeta \pm j \omega_n \sqrt{1 - \zeta^2}$$

$$p = -1.50 \pm 2.5981j \quad CLTF \Rightarrow \frac{10}{s^2 + s + 10}$$

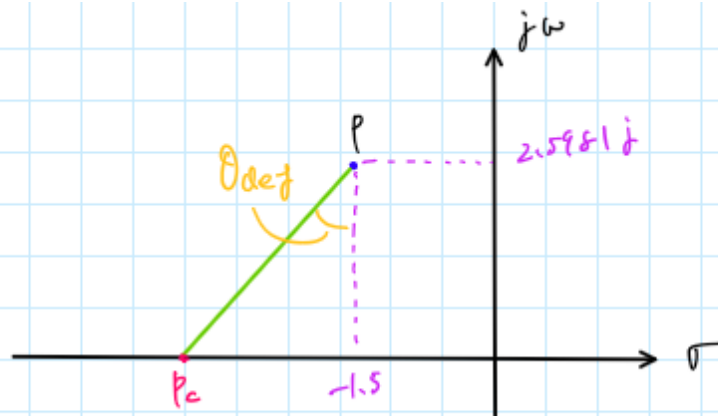
this the desired pole. from this find the angle

$$\begin{aligned} \theta_{\text{def}} &= 180 - \arg(p) - \arg(p - (-1)) \\ &= -40.8934 \end{aligned}$$

choose the zero of the compensator to be

$$z_c = -1.5$$

$$\text{now if } G_c = \frac{s - z_c}{s - p_c} \quad (*p_c \text{ and } z_c \text{ are real})$$



$$p_c = 1.5 + 2.5981 \times \tan(\theta_{deg}) = -3.750$$

thus,

$$G_c = k_c \frac{s+1.5}{s+3.75}$$

the value  $k_c$  can be determined by magnitude condition

$$k_c \left| \frac{s+1.5}{s+3.75} \cdot \frac{10}{s(s+1)} \right|_{s=p} = 1$$

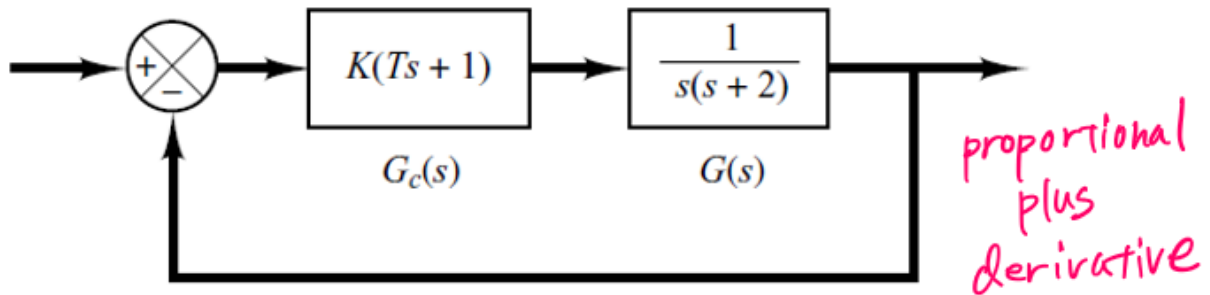
$$k_c = 1.0500$$

Thus,

$$k_c \frac{s+1.5}{s+3.75} = k \frac{T_1 s + 1}{T_2 s + 1} = \frac{k T_1 (s + \frac{1}{T_1})}{T_2 (s + \frac{1}{T_2})}$$

$$\left\{ \begin{array}{l} \frac{1}{T_1} = 1.5 \\ \frac{1}{T_2} = 3.75 \\ \frac{k T_1}{T_2} = k_c \end{array} \right. \Rightarrow \left\{ \begin{array}{l} T_1 = 0.6667 \\ T_2 = 0.2667 \\ k = 0.4200 \end{array} \right.$$

**B-6-16.** Consider the control system shown in Figure 6-106. Determine the gain  $K$  and time constant  $T$  of the controller  $G_c(s)$  such that the closed-loop poles are located at  $s = -2 \pm j2$ .



**Figure 6-106** Control system.

Similar approach to B-6-15

if  $s = -2 \pm j2$  is the desired pole  
the angle deficiency  $\theta_{def}$  is calculated by the  
following method say  $p = -2 + j2$

$$\begin{aligned}\theta_{def} &= 180 - \arg(p) - \arg(p - (-2)) \\ &= -45 \text{ deg}\end{aligned}$$

this means that

$$s = -\frac{1}{T} \text{ must contribute to } 45 \text{ deg}$$

which means if the zero for the  
compensator is  $z_c$

$$45 \text{ deg} = \arg(p - z_c)$$

$$\Rightarrow z_c = -4 \Rightarrow -\frac{1}{T} = -4$$

Hence, we have

$$\begin{aligned}K(Ts+1) &= KT(s+\frac{1}{T}) \\ &= KT(s+4)\end{aligned}$$

$T$  is

$$T = \frac{1}{4} = 0.25$$

now from magnitude condition  $K$  is

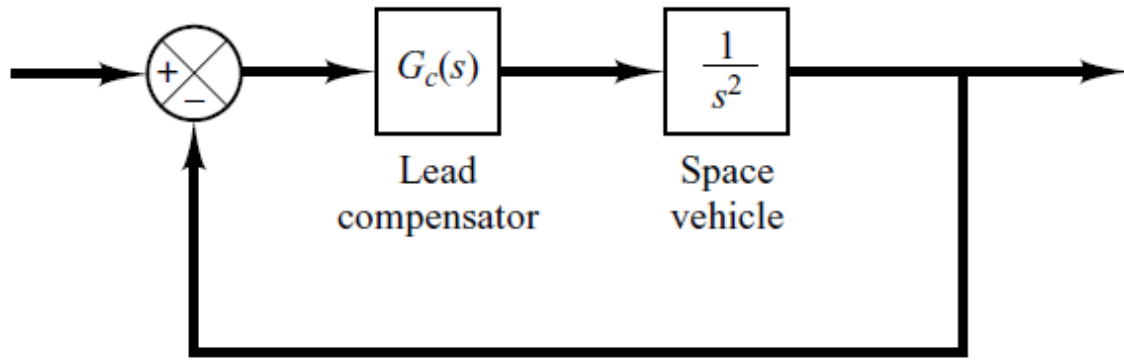
$$K \left| \frac{1}{4} (s+4) \frac{1}{s(s+2)} \right|_{s=p} = 1$$

$$\Rightarrow \frac{K}{8} = 1$$

$$K = 8$$

$$K=8, T=0.25$$

**B-6-18.** Consider the system shown in Figure 6-108. Design a compensator such that the dominant closed-loop poles are located at  $s = -1 \pm j1$ .



**Figure 6-108** Control system.



<9> Attempt #1

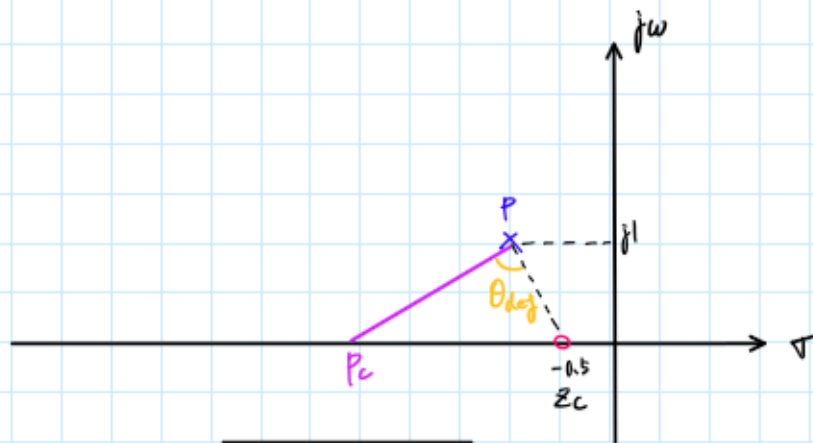
Assume the compensator  $G_c(s)$  to be in the form of

$$G_c(s) = K_c \left( \frac{s + \frac{1}{T}}{s + \frac{1}{qT}} \right) \quad (0 < q < 1)$$

we are given that the desired pole is @  $s = p = -1 \pm j$   
now, the angle deficiency will be

$$\begin{aligned} \theta_{def} &= 180^\circ - \arg(p - 0) - \arg(p - 0) \\ &= -90^\circ \end{aligned}$$

arbitrarily choose the zero of  $G_c(s)$ ,  $z_c = -0.5$ . Then the pole,  $p_c$  can be determined



$$\Rightarrow p_c = -0.5 - \sqrt{\|p + 0.5\|^2 + \|p - p_c\|^2}$$

$$\therefore z_c = 0.5$$

$$\Rightarrow p_c = -3$$

$$\text{therefore } G_c(s) = K_c \frac{s + 0.5}{s + 3}$$

$K_c$  can be determined from magnitude condition

$$K_c \left\| \frac{s + 0.5}{s + 3} \cdot \frac{1}{s^2} \right\|_{s=p} = 1$$

$$K_c \left\| \frac{-1+j+0.5}{-1+j+3} \frac{1}{(-1+j)^2} \right\| = 1$$

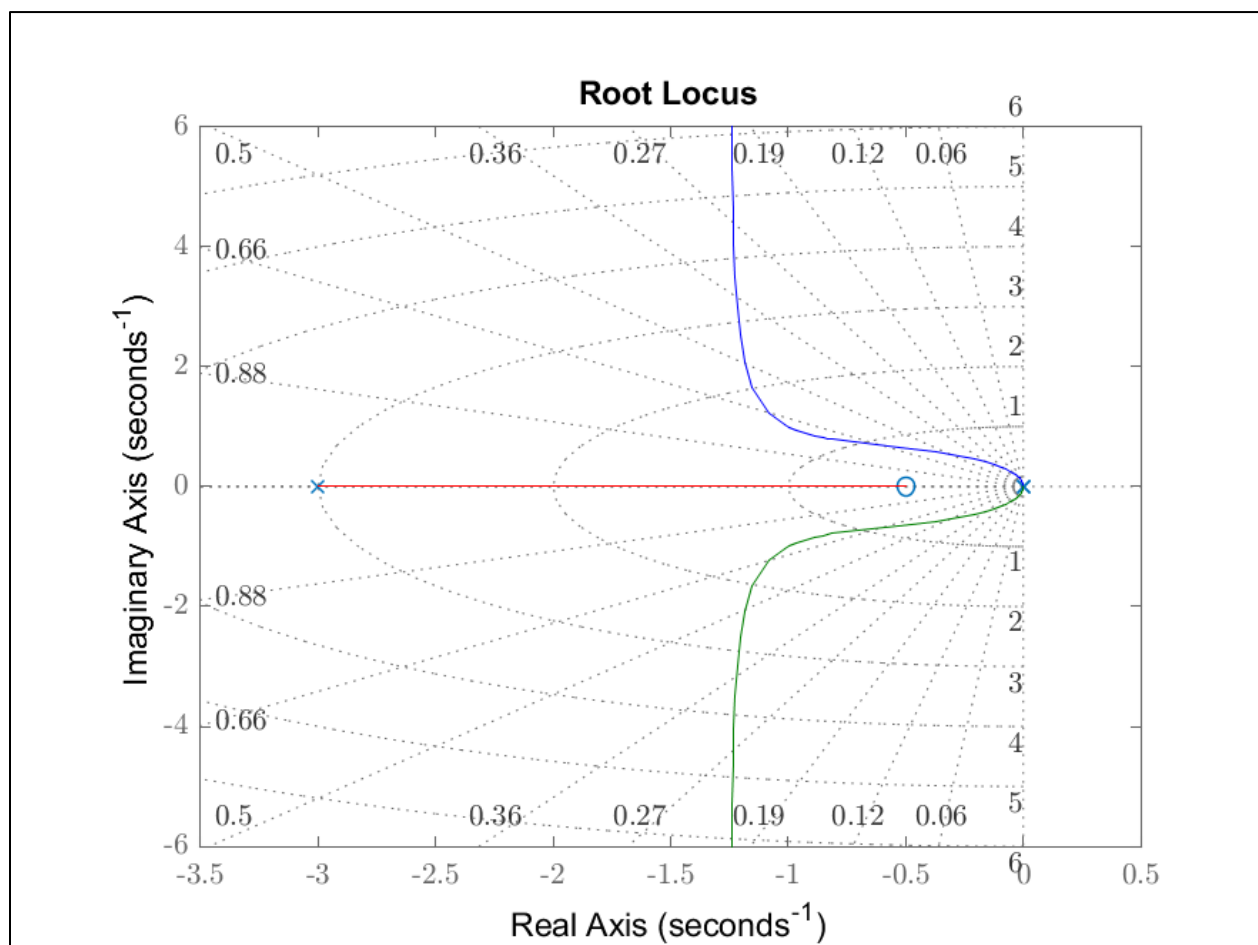
$$\therefore K_c = 4$$

thus,  
and

$$G_c(s) = 4 \frac{s+0.5}{s+3}$$

$$G_c(s)G(s) = \frac{4(s+0.5)}{s^2(s+3)}$$

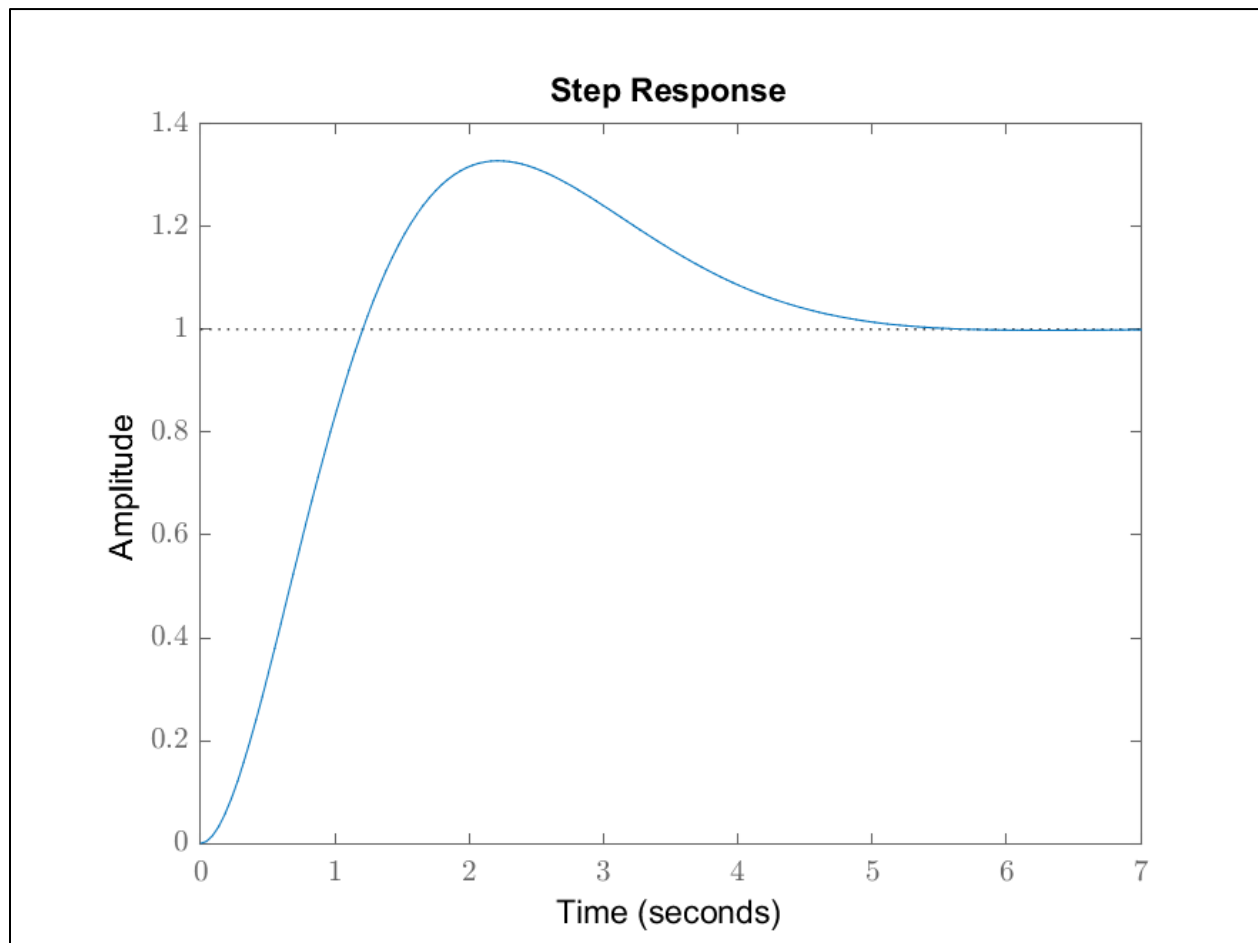
the **RL** for this becomes



and the step response becomes

$$\frac{C(s)}{R(s)} = \frac{G_c G}{1 + G_c G} = \frac{4s + 2}{s^3 + 3s^2 + 4s + 2}$$

$$\Rightarrow C(s) = \frac{4s + 2}{s(s^3 + 3s^2 + 4s + 2)}$$



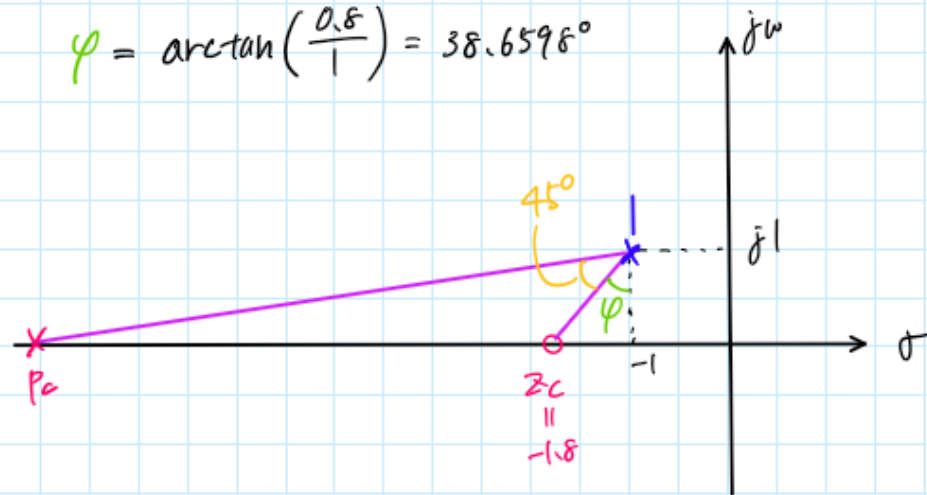
The overshoot is unacceptable so we go to the second attempt

<1> Attempt #2

We will use two lead networks using half the necessary lead angle  $\frac{90^\circ}{2} = 45^\circ$

next arbitrarily select  $z_c = -1.8$

$$\varphi = \arctan\left(\frac{0.8}{1}\right) = 38.6596^\circ$$



$$\Rightarrow p_c = -1 - \tan(45^\circ + \varphi)$$

$$= -10$$

therefore the lead compensator (2 lead networks)

$$G_c(s) = k_c \left( \frac{s - z_c}{s - p_c} \right)^2 = k_c \left( \frac{s + 1.8}{s + 10} \right)^2$$

$k_c$  can be determined from magnitude condition

$$k_c \left\| \left( \frac{s + 1.8}{s + 10} \right)^2 \cdot \frac{1}{s^2} \right\| = 1$$

$$k_c \left\| \left( \frac{-1 + j + 1.8}{-1 + j + 10} \right)^2 \frac{1}{(-1 + j)^2} \right\| = 1$$

$$\therefore K_c = 100$$

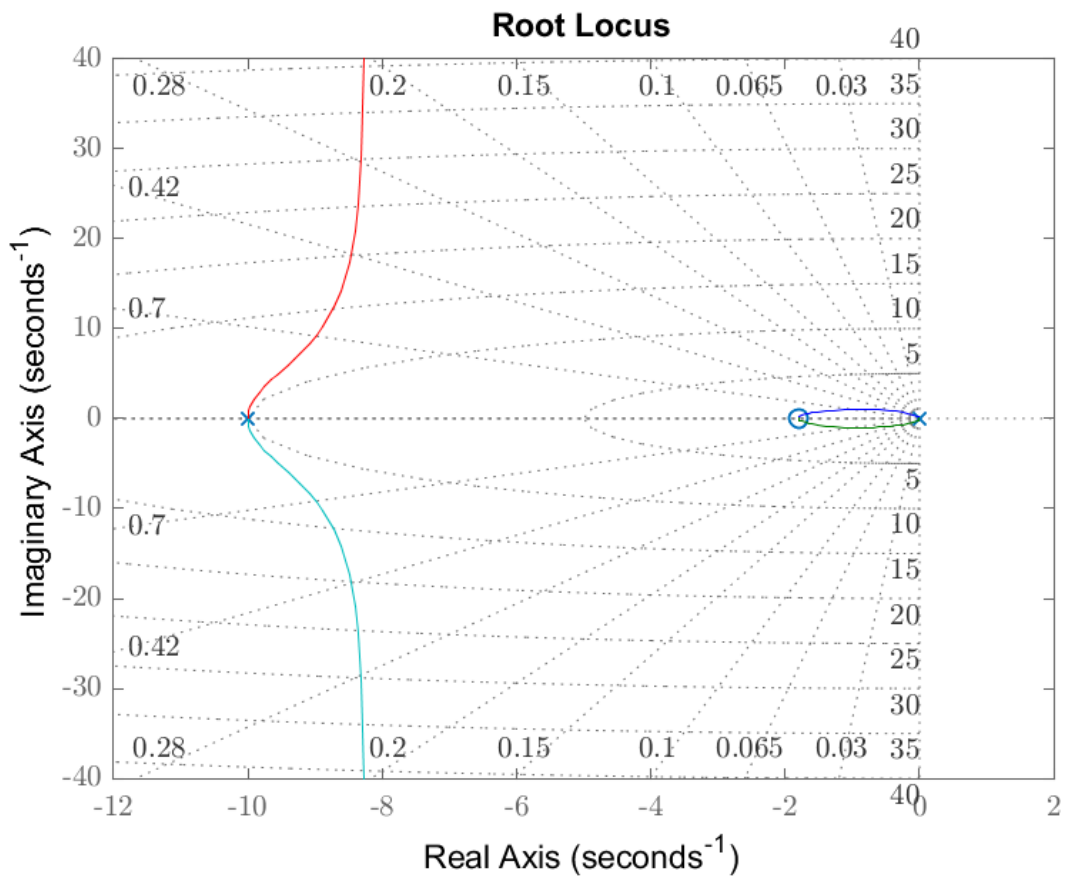
thus,

$$G_c(s) = 100 \left( \frac{s+1.8}{s+10} \right)^2$$

then

$$G_c(s)G(s) = \frac{100}{s^2} \left( \frac{s+1.8}{s+10} \right)^2$$

the **KL** becomes



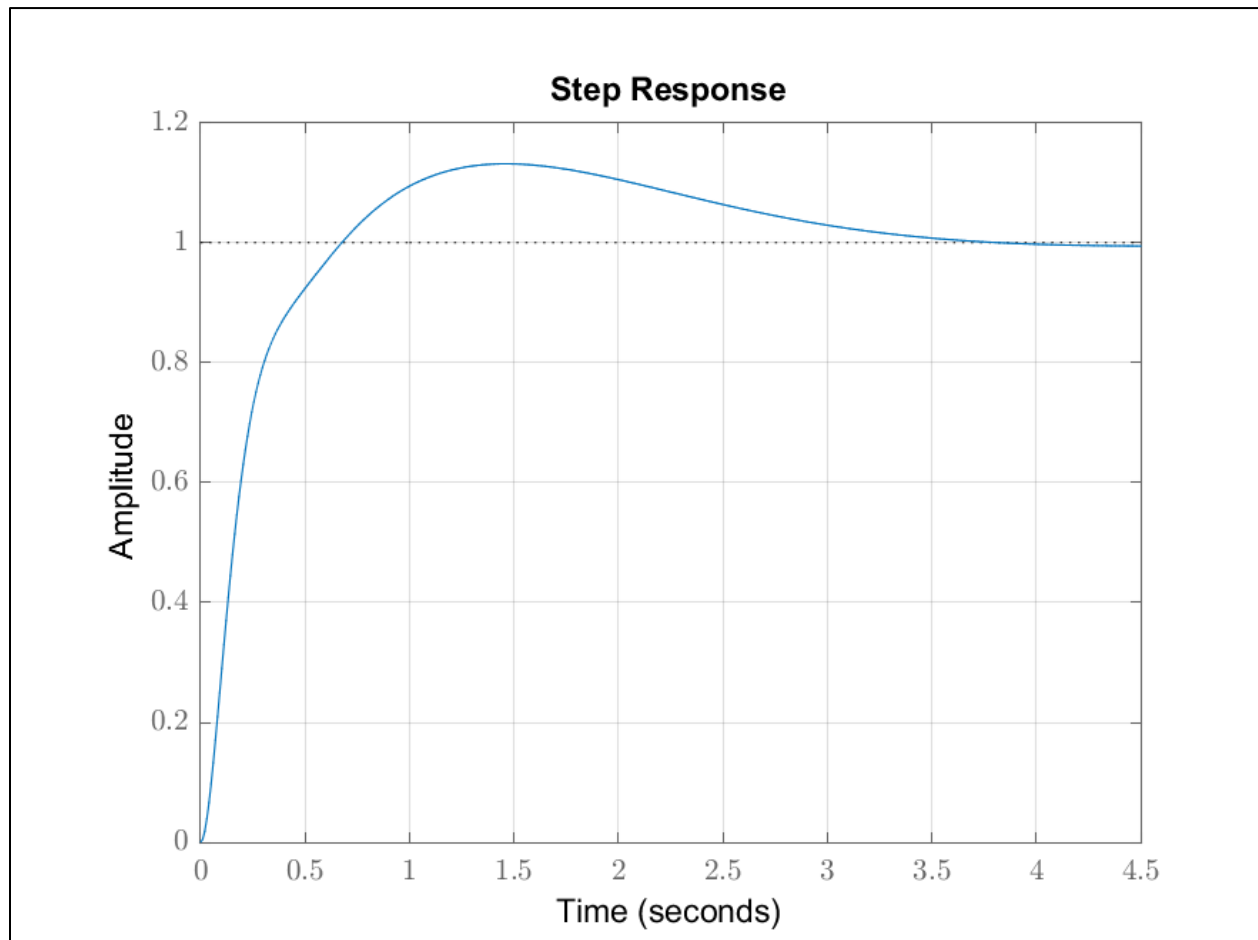
the CLTF becomes

$$G_1 = \frac{G_c G}{1 + G_c G} = \frac{100s^2 + 360s + 324}{s^4 + 20s^3 + 200s^2 + 360s + 324}$$

then

$$C(s) = G_1(s) R(s) = \frac{100s^2 + 360s + 324}{s(s^4 + 20s^3 + 200s^2 + 360s + 324)}$$

Hence the  
step response becomes



The overshoot has been improved significantly and this is acceptable

$$\therefore G_c = 100 \left( \frac{s+1.8}{s+10} \right)^2$$

## Problem 2

The following figure shows the coordinate axes and forces acting on the aircraft in the longitudinal plane of motion. Assuming that the aircraft is cruising at constant velocity and altitude.

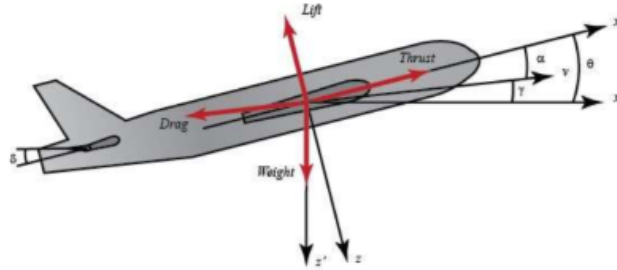


Figure 1: Forces acting on an aircraft in the Longitudinal plane.

Consider the unity-feedback system in Figure 2 with the plant  $G(s)$  representing the aircraft shown in Figure 1. Sketch the root locus of the unity-feedback system, with  $K(s) = K$ , as  $K$  varies from 0 to  $\infty$  (as accurately as you can) with the following  $G(s)$ :

1.  $G(s)$  representing the aircraft pitch angle response output to the elevator deflection input:

$$G(s) = \frac{\Theta(s)}{\Delta(s)} = \frac{1.1057s + 0.1900}{s^3 + 0.7385s^2 + 0.8008s}$$

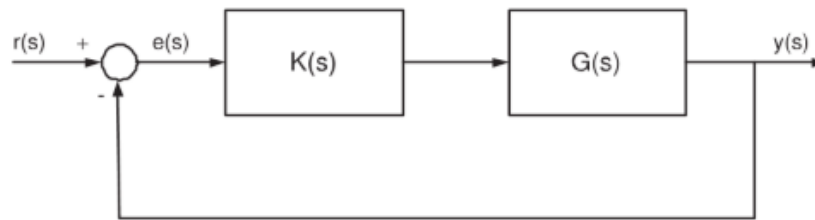


Figure 2: Unity-Feedback System with controller  $K(s)$  and plant  $G(s)$ .

2.  $G(s)$  representing the aircraft altitude response output to the elevator deflection input:

$$G(s) = \frac{H(s)}{\Delta(s)} = \frac{1.1057s - 0.1900}{s^5 + 17.95s^4 + 123.3s^3 + 366.3s^2 + 112.2s}$$



(1) When  $G(s) = \frac{H(s)}{A(s)} = \frac{1.1057s + 0.1900}{s^3 + 0.7385s^2 + 0.8008s}$

Since this is a negative feedback system

$$CE := 1 + K G(s) = 1 + K \frac{1.1057s + 0.1900}{s^3 + 0.7385s^2 + 0.8008s} = 0$$

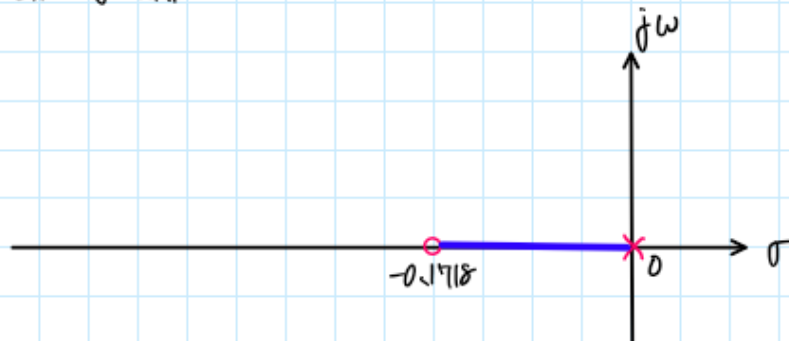
<i> Poles and Zeros

Poles:  $s^3 + 0.7385s^2 + 0.8008s = 0 \Rightarrow s = 0, -0.3693 \pm 0.8151j \Rightarrow n=3$

Zeros:  $1.1057s + 0.1900 = 0 \Rightarrow s = -0.1718 \Rightarrow m=1$

<ii> Symmetry TRUE

<iii> RL on  $\sigma$ -axis



<iv> Asymptotes

$$\theta_a = \frac{180^\circ + 360^\circ l}{n-m} = 90^\circ + 180^\circ l \quad (l = 0, 1)$$

$$\theta_a = 90^\circ, 270^\circ$$

and

$$\sigma_a = \frac{\sum P_i - \sum Z_i}{n-m}$$

$$= \frac{0 + (-0.3693 + 0.8151j) + (-0.3693 - 0.8151j) - (-0.1718)}{2}$$

$$= -0.2833$$

<v> Break-in/away points

$$\frac{d}{ds} \left( -\frac{1}{G(s)} \right) = -\frac{d}{ds} \left( \frac{s^3 + 0.7385s^2 + 0.8008s}{1.1057s + 0.1900} \right) = 0$$

$$\Rightarrow \frac{-2.2114s^3 - 1.3866s^2 - 0.2806s - 0.1522}{1.2226s^2 + 0.4202s + 0.0861} = 0$$

the real answer is

$$\times \hat{s}_1 = -0.6052 \rightarrow \text{However this not in the domain of PL on } \sigma\text{-axis}$$

<vi> Departure/Arrival angles

take a hypothetical point in the proximity of a complex pole

$$s^d = -0.3693 + 0.8151j$$

$$\begin{aligned} \angle G(s^d) &= -180^\circ = \arg(-0.3693 + 0.8151j - (-0.1718)) \\ &\quad - \theta_d - \arg(-0.3693 + 0.8151j - 0) \\ &\quad - \arg(-0.3693 + 0.8151j - (-0.3693 - 0.8151j)) \end{aligned}$$

$$\Rightarrow \theta_d = 79.2439^\circ$$

<vii> Intersection with  $j\omega$ -axis

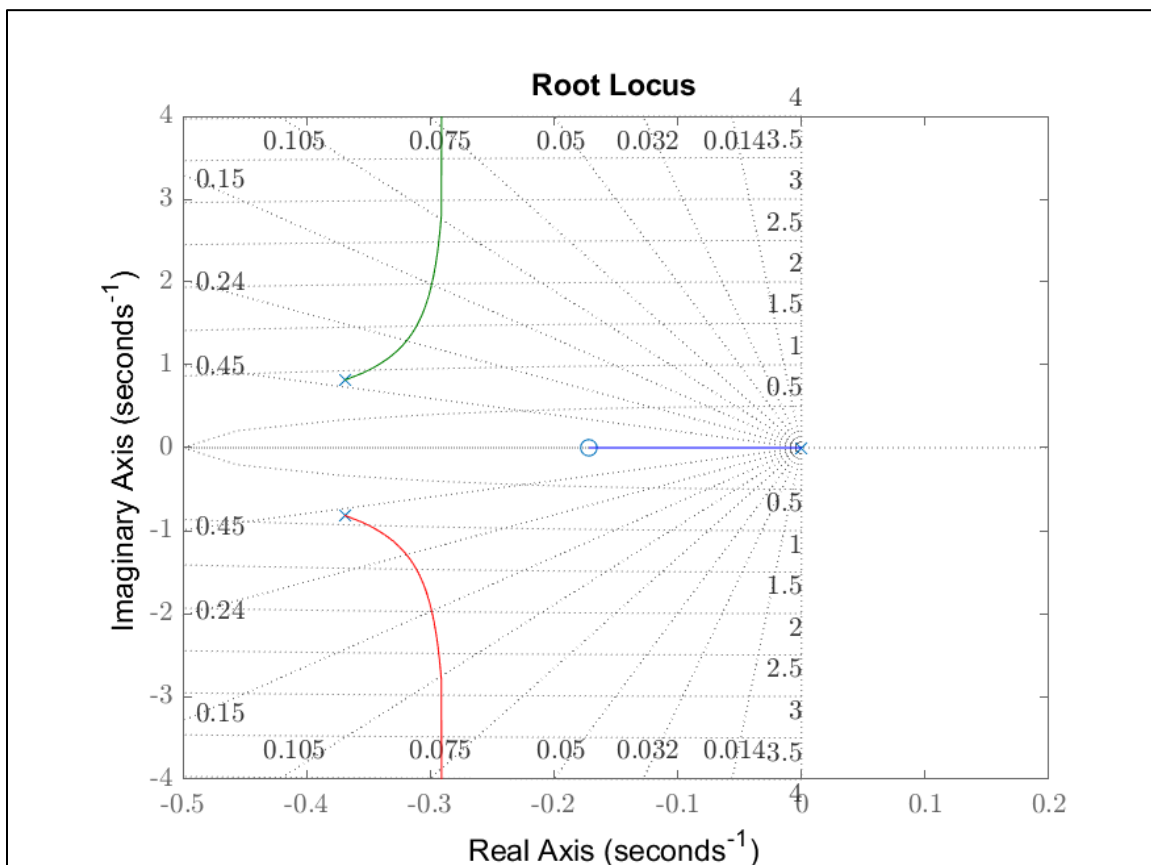
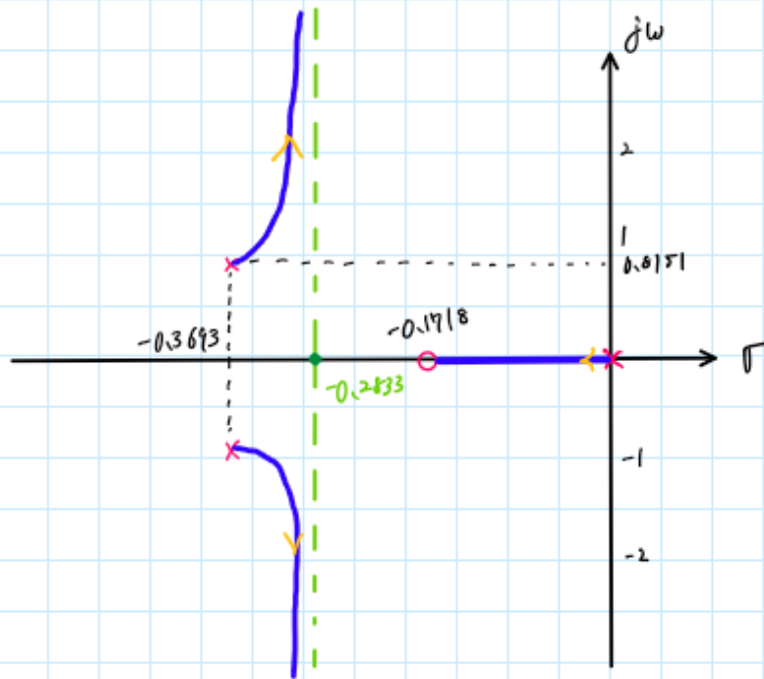
$$1 + \hat{k} G(j\hat{\omega}) = 1 + \hat{k} \frac{1.1057(j\hat{\omega}) + 0.1900}{-j\hat{\omega}^3 - 0.7385\hat{\omega}^2 + 0.8008j\hat{\omega}} = 0$$

solving this we get

$$\begin{cases} \sigma: \frac{19\hat{k}}{100} = \frac{1477}{2000}\hat{\omega}^2 \\ j\omega: \frac{11057\hat{k}}{10000}\hat{\omega} = \hat{\omega}^3 - \frac{1001}{1250}\hat{\omega} \end{cases}$$

$$\hat{k} = \begin{bmatrix} -0.9439 \\ -0.9439 \\ 0 \end{bmatrix} \quad \hat{\omega} = \begin{bmatrix} -0.4928j \\ 0.4928j \\ 0 \end{bmatrix} \quad \text{NONE}$$

therefore **RL** becomes



(2) when

$$G(s) = \frac{H(s)}{A(s)} = \frac{1.1057s - 0.1900}{s^5 + 17.95s^4 + 123.3s^3 + 366.3s^2 + 112.2s}$$

Since this is a negative feedback system

$$CE := 1 + K G(s) = 1 + K \frac{1.1057s - 0.1900}{s^5 + 17.95s^4 + 123.3s^3 + 366.3s^2 + 112.2s}$$

i) Poles and Zeros

$$\text{Poles: } s^5 + 17.95s^4 + 123.3s^3 + 366.3s^2 + 112.2s = 0$$

$$\Rightarrow p_1 = 0$$

$$p_4 = -4.7533 + 4.2012j$$

$$p_2 = -8.0992$$

$$p_5 = -4.7533 - 4.2012j$$

$$p_3 = -0.3442$$

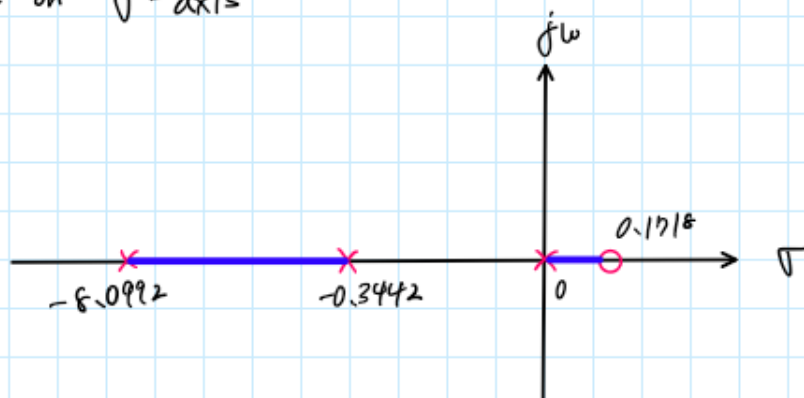
$$\Rightarrow n = 5$$

$$\text{Zeros: } 1.1057s - 0.1900 = 0$$

$$\Rightarrow z_1 = 0.1718 \Rightarrow m = 1$$

ii) Symmetry **TRUE**

iii) RL on  $\sigma$ -axis



iv) Asymptotes

$$\theta_a = \frac{(50^\circ + 360^\circ l)}{n - m} = 45^\circ + 90^\circ l \quad (l = 0, 1, 2, 3)$$

$$\theta_a = 45^\circ, 135^\circ, 225^\circ, 315^\circ$$

and

$$\sigma_a = \frac{\sum p_i - \sum z_i}{n - m}$$

$$= \frac{1}{4} [ 0 + (-8.0992) + (-0.3442) + (-4.7533 + 4.2012j) + (-4.7533 - 4.2012j) - (0.1718) ]$$

$$= -4.5305$$

<V> Break-in/away point

$$\frac{d}{ds} \left( -\frac{1}{G(s)} \right) = -\frac{d}{ds} \left( \frac{s^5 + 17.95s^4 + 123.3s^3 + 366.3s^2 + 112.2s}{1.1057s - 0.1900} \right) = 0$$

$$\Rightarrow \frac{-4.4228s^5 - 58.5919s^4 - 259.0236s^3 - 334.7369s^2 + 139.1940s + 21.3180}{1.2226s^2 - 0.4202s + 0.0361} = 0$$

$$\Rightarrow \hat{s}_1 = -3.2759 \text{ OK!}$$

$$\hat{s}_2 = 0.4189 \text{ X}$$

$$\hat{s}_3 = -0.1211 \text{ X}$$

<Vi> Departure/arrival angles

take a hypothetical point in the proximity of  $p_4$

$$s^d = -4.7533 + 4.2012j$$

$$\angle G(s^d) = -180^\circ = \arg(s^d - z_1) - \arg(s^d - p_1) - \arg(s^d - p_2) - \arg(s^d - p_3) - \theta_d - \arg(s^d - p_5)$$

$$\Rightarrow \theta_d = -96.8418^\circ$$

<vii> Intersection with  $j\omega$ -axis

$$1 + \hat{K} G(j\hat{\omega})$$

$$= 1 + \hat{K} \frac{1.1057(j\hat{\omega}) - 0.1900}{(j\hat{\omega})^5 + 17.95(j\hat{\omega})^4 + 123.3(j\hat{\omega})^3 + 366.3(j\hat{\omega})^2 + 112.2j\hat{\omega}} = 0$$

$$\Rightarrow \hat{K} (1.1057j\hat{\omega} - 0.1900)$$

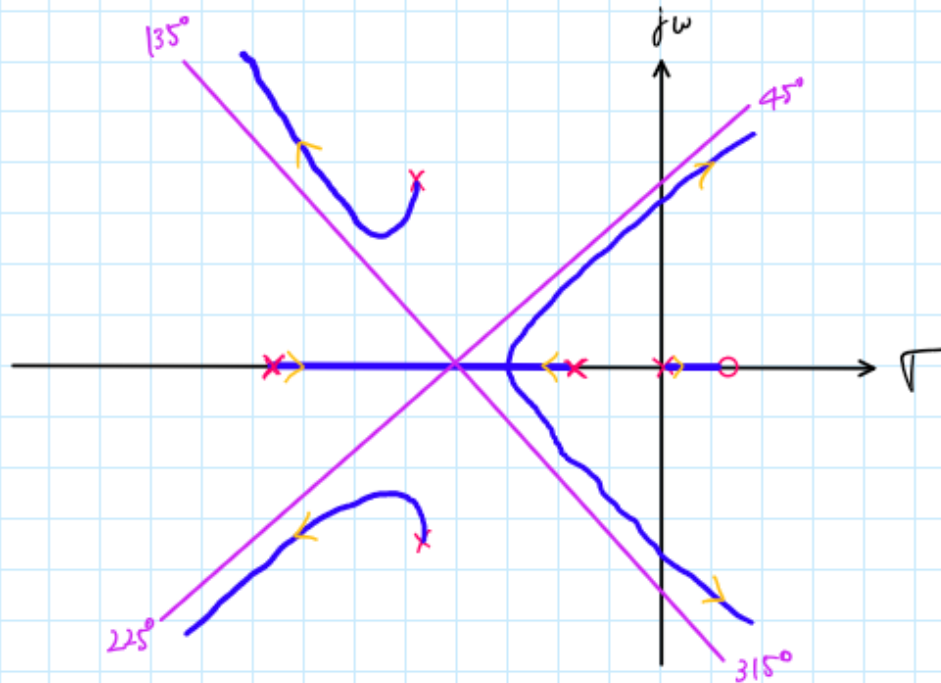
$$= -(j\hat{\omega}^5 + 17.95\hat{\omega}^4 - 123.3j\hat{\omega}^3 - 366.3\hat{\omega}^2 + 112.2j\hat{\omega})$$

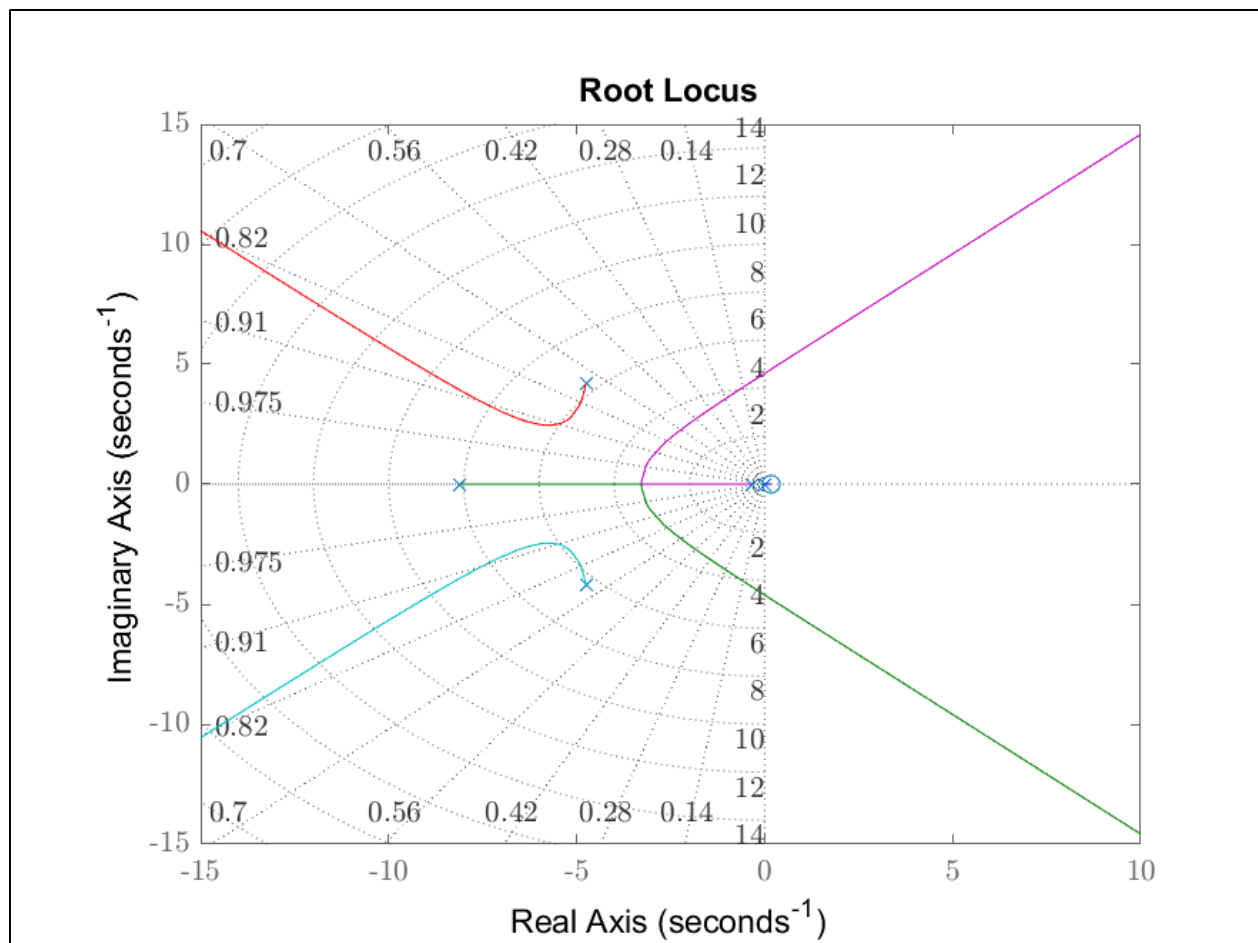
Solving this

$$\begin{cases} \sigma: -\frac{19\hat{K}}{100} = \frac{366.3}{100}\hat{\omega}^2 - \frac{359}{20}\hat{\omega}^4 \\ j\omega: \frac{11057}{10000}\hat{K}\hat{\omega} = -\hat{\omega}^5 + \frac{1233}{10}\hat{\omega}^3 - \frac{561}{5}\hat{\omega} \end{cases}$$

$$\hat{K} = \begin{bmatrix} 1.8658 \\ 1.8658 \\ -0.0959 \\ -0.0959 \\ 0 \end{bmatrix} \quad \hat{\omega} = \begin{bmatrix} -4.6187 \\ 4.6187 \\ -0.2233 \\ 0.2233 \\ 0 \end{bmatrix} \quad (*\hat{K} > 0)$$

Hence, from  $\zeta_p \sim \angle \hat{v}_i$  **RL** becomes





## Appendix

### AAE364 HW8 MATLAB CODE

```
clear all; close all; clc;
fdir = 'C:\Users\Tomo\Desktop\studies\2020-
Spring\AAE364\matlab\matlab_output\hw8';
set(groot, 'defaulttextinterpreter','latex');
set(groot, 'defaultAxesTickLabelInterpreter','latex');
set(groot, 'defaultLegendInterpreter','latex');
```

#### B-6-12

```
% (a)
num = [1 -1];
den = conv([1 2],[1 4]);
[poles,zrs,angs,sigma,bi_pt,T_P,T_Z,k,w,fig1] =
rootLocus_stepBystep_negFeedback(num,den)
saveas(fig1, fullfile(fdir,'RL_B-6-12_a.png'))
```

```
% (b)
num = -[1 -1];
den = conv([1 2],[1 4]);
[poles,zrs,angs,sigma,bi_pt,T_P,T_Z,k,w,fig1] =
rootLocus_stepBystep_posFeedback(num,den)
saveas(fig1, fullfile(fdir,'RL_B-6-12_b.png'))
```

#### B-6-14

```
num = [0 1];
den = conv([1 0],[1 4 5]);
[poles,zrs,angs,sigma,bi_pt,T_P,T_Z,k,w,fig1] =
rootLocus_stepBystep_negFeedback(num,den)
saveas(fig1, fullfile(fdir,'RL_B-6-14.png'))
```

#### % Step Response

```
% Find gain, K when zeta = 0.5
% Analytical
syms K x w s
assume(K,'real');
assume(x,'real');
```



```

assume(w, 'real');
p = -x + w*1j
RHS = (s*(s^2 + 4*s + 5))
RHS = subs(RHS,s,p)
RHS = expand(RHS)
RHS = subs(RHS,w,tand(60)*x)
eqn1 = K == -real(RHS)
eqn2 = 0 == -imag(RHS)
res = solve([eqn1 eqn2],[x K]);
sigma = double(res.x)
w = tand(60)*sigma
K_th = double(res.K)
K_th = nonzeros(K_th)

```

```

% Verify (computational)
G = tf(num,den);
% controlSystemDesigner(G) % from this we find that the pole we need is
% the following variable opz to satisfy zeta = 0.5
opz = [-0.625+1.08i, -0.625-1.08i];
[K_comp,pls] = rlocfind(G,opz)

```

```

% Plot step response
[numCL, denCL] = cloop((K_th)*num, den)
L_inv_expr = return_inverseLaplace_expression(numCL,conv(denCL,[1 0]))

```

```

fig2 = figure("Renderer","painters");
step(numCL,denCL);
grid on; grid minor; box on;
saveas(fig2,fullfile(fdir,"STEP_RES_B-6-14.png"));

```

## B-6-15

```

wn = 3; % natural frequency [rad/s]
zeta = 0.5; % damping ratio
p = -wn*zeta + 1j*wn*sqrt(1-zeta^2)

% deficiency angle
theta_def = pi + angle(10/p/(p + 1))
theta_deg_deg = rad2deg(theta_def)

% find the pole and zero for compensator
pc = real(p) - tan(2*pi - theta_def)*imag(p)

```

```

zc = real(p)

% find Kc
syms Kc
eqn = Kc*abs((p+1.5)/(p+3.75) * 10/(p*(p+1))) == 1;
K_c = double(solve(eqn,Kc))

% T1 T2 and K
T1 = 1/-zc
T2 = 1/-pc
K = K_c*T2/T1

```

## B-6-16

```

% angle deficiency
p = -2+2j;
theta_def = deg2rad(180) - angle(p) - angle(p + 2)
theta_def_deg = rad2deg(theta_def)

% find zero
syms zc
assume(zc,'real')
eqn = -theta_def == angle(p - zc)
Z_c = double(solve(eqn,zc))

syms K
eqn = K*0.25*abs((p+4)/p/(p+2)) == 1
K_p = double(solve(eqn,K))

```

## B-6-18

```

% First attempt
% angle deficiency
p = -1+1j;
theta_def = pi - angle(p - 0) - angle(p - 0)
theta_def_deg = rad2deg(theta_def)

% find pc
syms pc
eqn = pc == -0.5 - sqrt((abs(p + 0.5))^2 + (abs(p - pc))^2);
P_c = double(solve(eqn,pc))

```

```

% find Kc
syms Kc
eqn = Kc*abs((p + 0.5)/(p + 3)/p^2) == 1;
Kc = double(solve(eqn,Kc))

% Plot RL
num = Kc*[1 0.5];
den = conv([1 0 0],[1 3]);
fig1 = figure("Renderer","painters");
rlocus(tf(num,den))
sgrid
saveas(fig1, fullfile(fdir,'RL_B-6-18.png'))

```

```

% Step response
[numCL, denCL] = cloop(num,den)
fig2 = figure("Renderer","painters");
step(numCL,denCL);
grid on; grid minor; box on;
saveas(fig2,fullfile(fdir,"STEP_RES_B-6-18.png"));

```

```

% Second attempt
zc = -1.8;
phi = atan(0.8)
phi_deg = rad2deg(phi)
pc = -1 - tan(-theta_def/2 + phi)

% find Kc
syms Kc
eqn = Kc*abs(((p - zc)/(p - pc))^2/p^2) == 1;
Kc = double(solve(eqn,Kc))

```

```

% Plot RL
fig1 = figure("Renderer","painters");
num = Kc*conv([1 -zc],[1 -zc])
den = conv([1 0 0],[1 -pc]);
den = conv(den,[1 -pc])
rlocus(tf(num,den));
sgrid
saveas(fig1, fullfile(fdir,"RL_B-6-18_2.png"));

```

```

% Step response
[numCL, denCL] = cloop(num,den)

```

```

fig2 = figure("Renderer","painters");
step(numCL,denCL);
grid on; grid minor; box on;
saveas(fig2,fullfile(fdir,"STEP_RES_B-6-18_2.png"));

```

## P2

```

% (1)
num = [1.1057 0.1900];
den = [1 0.7385 0.8008 0];
[poles,zrs,angs,sigma,bi_pt,T_P,T_Z,k,w,fig1] =
rootLocus_stepBystep_negFeedback(num,den)
saveas(fig1, fullfile(fdir,'RL_P2_1.png'))

```

```

% (2)
num = [1.1057 -0.1900];
den = [1 17.95 123.3 366.3 112.2 0];
[poles,zrs,angs,sigma,bi_pt,T_P,T_Z,k,w,fig1] =
rootLocus_stepBystep_negFeedback(num,den)
saveas(fig1, fullfile(fdir,'RL_P2_2.png'))

```

```

function [poles,zrs,angs,sigma,bi_pt,T_P,T_Z,k,w,fig1] =
rootLocus_stepBystep_posFeedback(num,den)
    %{
        NAME:      ROOTLOCUS_STEPBYSTEP_NEGFEEDBACK
        AUTHOR:    TOMOKI KOIKE
        INPUTS:    (1) num:   THE NUMERATOR OF THE TRANSFER FUNCTION
                  (2) den:   THE DENOMINATOR OF THE TRANSFER FUNCTION
        OUTPUTS:  (1) poles: POLES OF THE TRANSFER FUNCTION
                  (2) zrs:   ZEROS OF THE TRANSFER FUNCTION
                  (3) angs:  ANGLES OF THE ASYMPTOTES
                  (4) sigma: INTERSECTION OF THE ASYMPTOTES
                  (5) bi_pt: BREAK-IN/AWAY POINT
                  (6) T_P:   TABLE WITH EACH POLE AND THEIR
                           DEPARTURE OR ARRIVAL ANGLES
                  (6) T_Z:   TABLE WITH EACH ZERO AND THEIR
                           DEPARTURE OR ARRIVAL ANGLES
                  (7) k:     VALUE K_HAT FOR INTERSECTION WITH IM AXIS
                  (8) w:     INTERSECTION POINT WITH THE IM AXIS
                  (9) fig1:  THE FIGURE WITH THE ROOT LOCUS PLOT
        DESCRIPTION: CONDUCTS THE 7 STEP PROCEDURE OF THE ROOT LOCUS
        ANALYSIS AND DISPLAYS THE RESULTS AS WELL AS THE PLOT FOR A POSITIVE
        FEEDBACK LOOP
    %}

```

```

% STEP1 - POLES & ZEROS
poles = roots(den);
zrs = roots(num);

% STEP2 - SYMMETRY (*TAKEN FOR GRANTED)

% STEP3 - ROOT LOCUS ON REAL AXIS (*OMMITTED)

% STEP4 - ASYMPTOTES
[angs,sigma] = RL_asymptote_posFeedback(zrs,poles);

% STEP5 - BREAK-IN/AWAY POINTS
bi_pt = break_in_away_pt(num,den);

% STEP6 - ANGLE OF DEPARTURE
[T_P, T_Z] = departure_arrival_angle_calc_posFeedback(zrs, poles);

% STEP7 - INTERSECTION WITH IMAGINARY AXIS
[k,w] = intersection_IM_axis(num,den);

% DEFINE THE TRANSFER FUNCTION
L = tf(num, den);
% PLOTTING THE ROOT LOCUS
fig1 = figure(1);
rlocus(L)
sgrid
end

```

```

function [poles,zrs,angs,sigma,bi_pt,T_P,T_Z,k,w,fig1] =
rootLocus_stepBystep_negFeedback(num,den)
%{
    NAME:      ROOTLOCUS_STEPBYSTEP_NEGFEEDBACK
    AUTHOR:    TOMOKI KOIKE
    INPUTS:    (1) num:   THE NUMERATOR OF THE TRANSFER FUNCTION
               (2) den:   THE DENOMINATOR OF THE TRANSFER FUNCTION
    OUTPUTS:   (1) poles: POLES OF THE TRANSFER FUNCTION
               (2) zrs:   ZEROS OF THE TRANSFER FUNCTION
               (3) angs:  ANGLES OF THE ASYMPTOTES
               (4) sigma: INTERSECTION OF THE ASYMPTOTES
               (5) bi_pt: BREAK-IN/AWAY POINT
               (6) T_P:   TABLE WITH EACH POLE AND THEIR
                           DEPARTURE OR ARRIVAL ANGLES
               (6) T_Z:   TABLE WITH EACH ZERO AND THEIR
                           DEPARTURE OR ARRIVAL ANGLES
               (7) k:     VALUE K_HAT FOR INTERSECTION WITH IM AXIS
               (8) w:     INTERSECTION POINT WITH THE IM AXIS
               (9) fig1:  THE FIGURE WITH THE ROOT LOCUS PLOT
%}

```

```

        DESCRIPTION: CONDUCTS THE 7 STEP PROCEDURE OF THE ROOT LOCUS
        ANALYSIS AND DISPLAYS THE RESULTS AS WELL AS THE PLOT FOR A NEGATIVE
        FEEDBACK LOOP
    %}

    % STEP1 - POLES & ZEROS
    poles = roots(den);
    zrs = roots(num);

    % STEP2 - SYMMETRY (*TAKEN FOR GRANTED)

    % STEP3 - ROOT LOCUS ON REAL AXIS (*OMMITTED)

    % STEP4 - ASYMPTOTES
    [angs,sigma] = RL_asymptote(zrs,poles);

    % STEP5 - BREAK-IN/AWAY POINTS
    bi_pt = break_in_away_pt(num,den);

    % STEP6 - ANGLE OF DEPARTURE
    [T_P, T_Z] = departure_arrival_angle_calc(zrs, poles);

    % STEP7 - INTERSECTION WITH IMAGINARY AXIS
    [k,w] = intersection_IM_axis(num,den);

    % DEFINE THE TRANSFER FUNCTION
    L = tf(num, den);
    % PLOTTING THE ROOT LOCUS
    fig1 = figure(1);
    rlocus(L)
    sgrid
end

```

```

function [angs, sigma] = RL_asymptote_posFeedback(zrs, poles)
    n = length(poles);
    m = length(zrs);
    angs = zeros([1,n-m]);
    for i = 0:(n-m)-1
        angs(i+1) = (360*i)/(n - m);
    end
    sigma = -(sum(poles) - sum(zrs))/(n - m);
end

```

```

function [angs, sigma] = RL_asymptote(zrs, poles)
    n = length(poles);
    m = length(zrs);
    angs = zeros([1,n-m]);
    for i = 0:(n-m)-1
        angs(i+1) = (180 + 360*i)/(n - m);
    end
    sigma = (sum(poles) - sum(zrs))/(n - m);
end

```

```

function [K, W] = intersection_IM_axis(num, den)
    syms k w
    n = length(den);
    m = length(num);
    f1 = 0; f2 = 0; p1 = 0; p2 = 0;

    % RHS (denominator)
    % when the largest order of s is even
    if rem(n,2) == 1
        % powers to the even numbers (real)
        for i = 1:2:n
            if rem(n-i,4) == 0
                f1 = f1 + den(i)*w^(n-i);
            else
                f1 = f1 + den(i)*w^(n-i)*(-1);
            end
        end
        % powers to the odd numbers (imaginary)
        for i = 2:2:n-1
            if rem(n-i,4) == 1
                f2 = f2 + den(i)*w^(n-i);
            else
                f2 = f2 + den(i)*w^(n-i)*(-1);
            end
        end
    end
    % when the largest order of s is odd
    elseif rem(n,2) == 0
        % powers to the even numbers (real)
        for i = 2:2:n
            if rem(n-i,4) == 0
                f1 = f1 + den(i)*w^(n-i);
            else
                f1 = f1 + den(i)*w^(n-i)*(-1);
            end
        end
        % powers to the odd numbers (imaginary)
        for i = 1:2:n-1
            if rem(n-i,4) == 1

```

```

        f2 = f2 + den(i)*w^(n-i);
    else
        f2 = f2 + den(i)*w^(n-i)*(-1);
    end
end
end

% LHS
% when the largest order of s is even
if rem(m,2) == 1
    % powers to the even numbers (real)
    for i = 1:2:m
        if rem(m-i,4) == 0
            p1 = p1 + num(i)*w^(m-i);
        else
            p1 = p1 + num(i)*w^(m-i)*(-1);
        end
    end
    % powers to the odd numbers (imaginary)
    for i = 2:2:m-1
        if rem(m-i,4) == 1
            p2 = p2 + num(i)*w^(m-i);
        else
            p2 = p2 + num(i)*w^(m-i)*(-1);
        end
    end
end
% when the largest order of s is odd
elseif rem(m,2) == 0
    % powers to the even numbers (real)
    for i = 2:2:m
        if rem(m-i,4) == 0
            p1 = p1 + num(i)*w^(m-i);
        else
            p1 = p1 + num(i)*w^(m-i)*(-1);
        end
    end
    % powers to the odd numbers (imaginary)
    for i = 1:2:m-1
        if rem(m-i,4) == 1
            p2 = p2 + num(i)*w^(m-i);
        else
            p2 = p2 + num(i)*w^(m-i)*(-1);
        end
    end
end
end

% Solving the system equations
Re = k*p1 == -f1
Im = k*p2 == -f2
a = vpasolve([Re Im], [k w]);

```



```

K = double(a.k);
W = double(a.w);
end

```

```

function [table_P, table_Z] = departure_arrival_angle_calc_posFeedback(zrs,
poles)
    %{
        NAME:      ROOTLOCUS_STEPBYSTEP
        AUTHOR:     TOMOKI KOIKE
        INPUTS:     (1) zrs:   THE ZEROS OF THE TRANSFER FUNCTION
                   (2) poles: THE POLES OF THE TRANSFER FUNCTION
        OUTPUTS:    (1) theta: THE DEPARTURE ANGLES AND ARRIVAL ANGLES FOR THE
                   TRANSFER FUNCTION
        DESCRIPTION: CALCULATES ALL THE DEPARTURE ANGLES AND ARRIVALS ANGLES
        FOR THE PROVIDED ZEROS AND POLES OF A TRANSFER FUNCTION FOR POSITIVE
        FEEDBACK LOOP
    %}

    % PREALLOCATE ARRAY THETA TO STORE ALL ANGLES FOR THE POLES
    theta_P = zeros([1,(length(poles))]);

    % ANGLE FOR A FICTICIOUS POINT CLOSE TO EACH POLE
    for n = 1:length(poles)
        obj = poles(n);
        % ANGLES FROM THE ZERO POINT TO A THE CURRENT POINT
        if not(isempty(zrs))
            for i = 1:length(zrs)
                theta_P(n) = theta_P(n) + angle(obj - zrs(i));
            end
        end
        % ANGLE FROM ANOTHER POLE TO THE CURRENT POLE
        for i = 1:length(poles)
            theta_P(n) = theta_P(n) - angle(obj - poles(i));
        end
        % THE ANGLE BECOMES
        theta_P(n) = theta_P(n) + deg2rad(0); % [rad]
    end

    % CREATING TABLE
    rad_P = reshape(theta_P,[length(theta_P),1]);
    deg_P = rad2deg(rad_P);
    table_P = table(reshape(poles,[length(poles),1]),rad_P,deg_P);
    table_P.Properties.VariableNames = {'POLES','RADIUS','DEGREES'};

    if not(isempty(zrs))

```

```

% PREALLOCATE ARRAY THETA TO STORE ALL ANGLES FOR THE POLES
theta_Z = zeros([1,(length(zrs))]);
% ANGLE FOR A FICTICIOUS POINT CLOSE TO EACH ZERO
for n = 1:length(zeros)
    obj = zrs(n);
    % ANGLES FROM THE ZERO POINT TO A THE CURRENT POINT
    if not isempty(zrs)
        for i = 1:length(zrs)
            theta_Z(n) = theta_Z(n) + angle(obj - zrs(i));
        end
    end
    % ANGLE FROM A POLE TO THE CURRENT ZERO POINT
    for i = 1:length(poles)
        theta_Z(n) = theta_Z(n) - angle(obj - poles(i));
    end
    % THE ANGLE BECOMES
    theta_Z(n) = -deg2rad(0) - theta_Z(n); % [rad]
end

% CREATING TABLE
rad_Z = reshape(theta_Z,[length(theta_Z),1]);
deg_Z = rad2deg(rad_Z);
table_Z = table(reshape(zrs,[length(zrs),1]),rad_Z,deg_Z);
table_Z.Properties.VariableNames = {'ZEROS','ANGLES','DEGREES'};
else
    table_Z = [];
end
end
end

```

```

function [table_P, table_Z] = departure_arrival_angle_calc(zrs, poles)
%{
    NAME:      ROOTLOCUS_STEPBYSTEP
    AUTHOR:    TOMOKI KOIKE
    INPUTS:    (1) zrs:   THE ZEROS OF THE TRANSFER FUNCTION
               (2) poles: THE POLES OF THE TRANSFER FUNCTION
    OUTPUTS:   (1) theta: THE DEPARTURE ANGLES AND ARRIVAL ANGLES FOR THE
                  TRANSFER FUNCTION
    DESCRIPTION: CALCULATES ALL THE DEPARTURE ANGLES AND ARRIVALS ANGLES
                  FOR THE PROVIDED ZEROS AND POLES OF A TRANSFER FUNCTION FOR NEGATIVE
                  FEEDBACK LOOP
%}

```

```

% PREALLOCATE ARRAY THETA TO STORE ALL ANGLES FOR THE POLES
theta_P = zeros([1,(length(poles))]);

% ANGLE FOR A FICTICIOUS POINT CLOSE TO EACH POLE
for n = 1:length(poles)
    obj = poles(n);
    % ANGLES FROM THE ZERO POINT TO A THE CURRENT POINT
    if not isempty(zrs)
        for i = 1:length(zrs)
            theta_P(n) = theta_P(n) + angle(obj - zrs(i));
        end
    end
    % ANGLE FROM ANOTHER POLE TO THE CURRENT POLE
    for i = 1:length(poles)
        theta_P(n) = theta_P(n) - angle(obj - poles(i));
    end
    % THE ANGLE BECOMES
    theta_P(n) = theta_P(n) + deg2rad(180); % [rad]
end

% CREATING TABLE
rad_P = reshape(theta_P,[length(theta_P),1]);
deg_P = rad2deg(rad_P);
table_P = table(reshape(poles,[length(poles),1]),rad_P,deg_P);
table_P.Properties.VariableNames = {'POLES','RADIUS','DEGREES'};

if not isempty(zrs)
    % PREALLOCATE ARRAY THETA TO STORE ALL ANGLES FOR THE POLES
    theta_Z = zeros([1,(length(zrs))]);
    % ANGLE FOR A FICTICIOUS POINT CLOSE TO EACH ZERO
    for n = 1:length(zeros)
        obj = zrs(n);
        % ANGLES FROM THE ZERO POINT TO A THE CURRENT POINT
        if not isempty(zrs)
            for i = 1:length(zrs)
                theta_Z(n) = theta_Z(n) + angle(obj - zrs(i));
            end
        end
        % ANGLE FROM A POLE TO THE CURRENT ZERO POINT
        for i = 1:length(poles)
            theta_Z(n) = theta_Z(n) - angle(obj - poles(i));
        end
        % THE ANGLE BECOMES
        theta_Z(n) = -deg2rad(180) - theta_Z(n); % [rad]
    end

    % CREATING TABLE
    rad_Z = reshape(theta_Z,[length(theta_Z),1]);
    deg_Z = rad2deg(rad_Z);

```

```

        table_Z = table(reshape(zrs,[length(zrs),1]),rad_Z,deg_Z);
        table_Z.Properties.VariableNames = {'ZEROS','ANGLES','DEGREES'};
    else
        table_Z = [];
    end
end

```

```

function rts = break_in_away_pt(num,den)
    [q, d] = polyder(-den,num)
    rts = roots(q);
    rts = rts(rts==real(rts));
end

```