

This article was downloaded by: [Laurie Rotman]

On: 09 March 2015, At: 08:32

Publisher: Taylor & Francis

Informa Ltd Registered in England and Wales Registered Number: 1072954 Registered office: Mortimer House, 37-41 Mortimer Street, London W1T 3JH, UK



International Journal of Control

Publication details, including instructions for authors and subscription information:

<http://www.tandfonline.com/loi/tcon20>

Design of unknown input observers and robust fault detection filters

JIE CHEN^a, RON J. PATTON^a & HONG-YUE ZHANG^b

^a Department of Electronic Engineering, University of Hull, Cottingham Road, Hull, HU6 7RX, U.K.

^b Department of Automatic Control, Beijing University of Aeronautics and Astronautics, Beijing, 100083, People's Republic of China

Published online: 27 Feb 2007.

To cite this article: JIE CHEN, RON J. PATTON & HONG-YUE ZHANG (1996) Design of unknown input observers and robust fault detection filters, *International Journal of Control*, 63:1, 85-105, DOI: [10.1080/00207179608921833](https://doi.org/10.1080/00207179608921833)

To link to this article: <http://dx.doi.org/10.1080/00207179608921833>

PLEASE SCROLL DOWN FOR ARTICLE

Taylor & Francis makes every effort to ensure the accuracy of all the information (the "Content") contained in the publications on our platform. However, Taylor & Francis, our agents, and our licensors make no representations or warranties whatsoever as to the accuracy, completeness, or suitability for any purpose of the Content. Any opinions and views expressed in this publication are the opinions and views of the authors, and are not the views of or endorsed by Taylor & Francis. The accuracy of the Content should not be relied upon and should be independently verified with primary sources of information. Taylor and Francis shall not be liable for any losses, actions, claims, proceedings, demands, costs, expenses, damages, and other liabilities whatsoever or howsoever caused arising directly or indirectly in connection with, in relation to or arising out of the use of the Content.

This article may be used for research, teaching, and private study purposes. Any substantial or systematic reproduction, redistribution, reselling, loan, sub-licensing, systematic supply, or distribution in any form to anyone is expressly forbidden. Terms & Conditions of access and use can be found at <http://www.tandfonline.com/page/terms-and-conditions>

Design of unknown input observers and robust fault detection filters

JIE CHEN[†], RON J. PATTON[†] and HONG-YUE ZHANG[‡]

Fault detection filters are a special class of observers that can generate directional residuals for the purpose of fault isolation. This paper proposes a new approach to design robust (in the disturbance de-coupling sense) fault detection filters which ensure that the residual vector, generated by this filter, has both robust and directional properties. This is done by combining the unknown input observer and fault detection filter principles. The paper proposes a new full-order unknown input observer, and gives necessary and sufficient conditions for its existence. After the disturbance de-coupling conditions are satisfied, the remaining design freedom can be used to make the residual have the directional property, based on the fault detection filter principle. A nonlinear jet engine system is used to illustrate the robust fault isolation approach presented. It is shown that linearization errors can be approximately treated as unknown disturbances and be de-coupled in the design of a robust fault detection filter. Simulation results show that mis-isolation of faults can be avoided using the robust scheme.

1. Introduction

The process of detecting and isolating system faults has been of considerable interest during the last two decades (Beard 1971, Jones 1973, Willsky 1976, Patton *et al.* 1989, Frank 1990, Gertler 1993). Research is still under way into the development of more effective solutions for fault detection and isolation (FDI) in automatic control systems (Patton and Chen 1994). A fault is defined as an unexpected change in a system, such as a component malfunction and variations in operating condition, which tends to degrade overall system performance. In the early developmental stages of fault detection techniques, most investigators used the term 'failure' (Beard 1971, Jones 1973, Willsky 1976). Recently, most international researchers agree that the term 'fault' is more appropriate than the term 'failure' (Patton *et al.* 1989, Frank 1990, Park and Rizzoni 1993, Patton and Chen 1994). The term failure suggests complete breakdown, whilst a fault may denote something tolerable. We use the term 'fault' rather than 'failure' to denote a tolerable malfunction rather than a catastrophe.

The purpose of fault detection is to determine that a fault has occurred in the system, whereas fault isolation procedures are used to determine the location of the fault, after detection. The basic principle of model-based FDI is to compare actual and anticipated system responses generated using mathematical models. The most effective model-based FDI approach is the observer-based approach (Beard 1971, Jones 1973,

Received 13 September 1994. Revised 7 December 1994.

[†] Department of Electronic Engineering, University of Hull, Cottingham Road, Hull HU6 7RX, U.K.

[‡] Department of Automatic Control, Beijing University of Aeronautics and Astronautics, Beijing 100083, People's Republic of China.

Willsky 1976, Patton *et al.* 1989, Frank 1990) in which the difference between actual and estimated outputs is used as a residual vector. This residual is zero when the system is normal and is non-zero when a fault occurs. Hence, the residual can be compared with a threshold for detection purposes. Fault isolation is more complicated than detection and one approach is to design a set of structured residual signals (Patton *et al.* 1989, Frank 1990, Gertler 1993, Patton and Chen 1994). 'Structured' means that each residual is designed to be sensitive to a certain group of faults, whilst insensitive to others. The sensitivity and insensitivity properties make isolation possible. Another approach to isolation is via the design of a directional residual vector, i.e. to make the residual vector lie in a fixed and fault-specific direction in the residual space in response to each fault. With directional residual vectors, the fault isolation can be achieved by determining to which of the known fault signature directions the residual vector lies closest.

The most effective way to generate direction residual vectors is to use the Beard fault detection filter (BFDF) (Beard 1971, Jones 1973, Massoumnia 1986, White and Speyer 1987, Park and Rizzoni 1993). It has to be pointed out that this class of observers has been referred to in the literature as a 'failure detection filter' (Beard 1971, Jones 1973, White and Speyer 1987) in the early development of fault (failure) diagnosis. Fault detection filters are a particular class of full-order Luenberger observers with a specially designed feedback gain matrix such that the output estimation error (residual vector) has unidirectional characteristics associated with some known fault directions. To be specific, the residual vector of a fault detection filter is fixed along with a predetermined direction for an actuator fault or lies in a specific plane for a sensor fault. Since the important information required for isolation is contained in the direction of the residual rather than in its time function, the use of a BFDF does not require the knowledge of the fault mode. The fault isolation task can be facilitated by comparing the residual direction with pre-defined fault signature directions (or planes), and only one (or the minimum number of) observer(s) required for fault isolation due to directional characteristics of the residual. This is the main and most appealing feature of fault detection filters.

Note that the robustness issues have not been considered in the context of Beard fault detection filters up to now. Hence, this approach does not account for the effects of disturbances, nonlinearities, modelling errors, parameter variations and other uncertain factors in the system. There would be false or missed alarms when this approach is directly applied to industrial systems, in which the uncertain factors are unavoidable in modelling (especially for systems such as mechanical, electromechanical, thermofluid and aircraft systems). The application of BFDFs has been obstructed by the lack of robustness. Robustness in FDI is defined as the degree to which the FDI performance is unaffected by (or remains insensitive to) uncertainties of the system. An ideal (robust) FDI scheme is such a scheme that is sensitive to faults, whilst insensitive to system uncertainties.

One of the most successful robust fault diagnosis approaches is the use of the disturbance de-coupling principle, in which the residual is designed to be insensitive to unknown disturbances, whilst sensitive to faults. This can be done by using unknown input observers (Watanabe and Himmelblau 1982, Frank and Wünnenberg 1989, Chen and Zhang 1991, Patton and Chen 1993*b*), by eigenstructure assignment (Patton *et al.* 1987, Patton and Chen, 1992, Patten and Chen 1993*a*) or using structured parity equations (Gertler 1993, Gertler and Kunwer 1993). Using the disturbance de-coupling approach, the robust FDI with respect to unknown inputs (disturbances) can

be solved. However, the robustness problem with respect to modelling errors and other uncertain factors is more difficult. To solve this problem, some investigators have suggested an approach in which modelling errors are approximately represented as unknown disturbances with an estimated distribution matrix (Patton and Chen 1992, 1993a, Patton *et al.* 1992, Gertler and Kunwer 1993, Gertler 1994). In this way, an optimally robust solution is achievable. This approximate strategy has extended the application domain of disturbance de-coupling-based robust FDI approaches, which have been favoured recently in FDI research.

This paper proposes a new strategy to design robust fault detection filters whose residuals have both disturbance de-coupling and uni-directional properties. This filter is designed by combining the unknown input observer (UIO) principle with the Beard fault detection filter principle. Using UIO principles, the residual has been made robust against unknown inputs (disturbances). The uni-directional property is achieved based on BFDF techniques. It should be pointed out that there are two approaches to the robust detection filter design. The detection filter can be designed first and then the design freedom can be used to reject disturbances. Alternatively, the UIO can be designed first, the freedom is then used to make the fault signature have uni-directional properties. The choice depends on the design emphasis and the availability of the design procedure. This paper is based upon the second approach.

The paper is organized as follows. In §2, a new full-order UIO structure is proposed. A rigorous mathematical foundation in designing such a UIO has been laid down and the necessary and sufficient existence conditions for such a UIO are presented and thoroughly proved. The existence conditions are easy to verify and the design procedure is very simple. There is still some design freedom left after the disturbance (unknown input) de-coupling conditions have been satisfied. Section 3 studies how to exploit the remaining design freedom to make the residual have uni-directional characteristics. This section discusses first the basic principles of the Beard fault detection filters. These principles are then used to make the residual, generated by an UIO, take on the required uni-directional properties. A filter that can produce disturbance de-coupled and directional residuals is called a 'disturbance de-coupled (robust) fault detection filter'. Section 3 also presents a robust fault isolation scheme using robust directional residual vectors. In §4, a real example of the isolation of faulty sensors in a jet engine system is presented. This is a nonlinear system and the linearization error can cause mis-isolation problems if the robustness issue is not properly considered. This paper presents a method to represent linearization errors as an unknown input term and the term's distribution is estimated using a least-squares procedure. This is a new and effective way to tackle the robustness problem. The paper describes the design of a robust fault detection filter for isolating faults in a jet engine system and the simulation results show that faults are correctly isolated.

2. Theory and design of unknown input observers

In this paper, we consider a class of systems in which the system uncertainty can be summarized as an *additive unknown disturbance* term in the dynamic equation, described as follows

$$\left. \begin{aligned} \dot{x}(t) &= Ax(t) + Bu(t) + Ed(t) \\ y(t) &= Cx(t) \end{aligned} \right\} \quad (1)$$

where $x(t) \in \mathcal{R}^n$ is the state vector, $y(t) \in \mathcal{R}^m$ is the output vector, $u(t) \in \mathcal{R}^r$ is the known

input vector and $d(t) \in \mathbb{R}^q$ is the unknown input (or disturbance) vector. A , B , C and E are known matrices with appropriate dimensions. There is no loss of generality in assuming that the known unknown input distribution matrix E is a full column rank matrix. Otherwise, the following rank decomposition can be applied to the matrix E

$$Ed(t) = E_1 E_2 d(t)$$

where E_1 is a full column rank matrix and $E_2 d(t)$ can now be considered as a new unknown input. The disturbance term may also appear in the output equation; however, this case is not considered here because the disturbance term in the output equation can be nulled by simply using a transformation of the output signal $y(t)$.

The term $Ed(t)$ can be used to describe additive disturbance as well as a number of different kinds of modelling uncertainties. For example, noise, interconnecting terms in large-scale systems, nonlinear terms in system dynamics (Watanabe and Himmelblau 1982, Chen and Zhang 1991), terms arising from time-varying system dynamics, linearization and model reduction errors, parameter variations can all be represented in this way. A detailed study of this problem can be found in Patton and Chen (1993a, 1993b, 1992), Patton *et al.* (1992), Gertler and Kunwer (1993). In the following, two examples of the representation of modelling errors as an additive disturbance term $Ed(t)$ are given.

The first example considers the system dynamics with parameter perturbations, represented as

$$\dot{x}(t) = (A + \Delta A)x(t) + (B + \Delta B)u(t)$$

The parameter perturbations sometimes considered in the robust control literature are approximated as

$$\Delta A \approx \sum_{i=1}^N a_i A_i \quad \Delta B \approx \sum_{i=1}^N b_i B_i$$

where A_i and B_i are known matrices with proper dimensions, a_i and b_i are scalar factors. In this case, the modelling errors can be approximated by the disturbance term as

$$\begin{aligned} Ed(t) &= \Delta A x(t) + \Delta B u(t) \\ &= [A_1 \cdots A_N B_1 \cdots B_N] \begin{bmatrix} a_1 x(t) \\ \vdots \\ a_N x(t) \\ b_1 u(t) \\ \vdots \\ b_N u(t) \end{bmatrix} \end{aligned}$$

Now, to consider the second example in which the system matrices are functions of a parameter vector $\alpha \in \mathbb{R}^g$

$$\dot{x}(t) = A(\alpha)x(t) + B(\alpha)u(t)$$

If the parameter has a perturbation around the nominal condition $\alpha = \alpha_0$, this equation can be expanded as

$$\dot{x}(t) = A(\alpha_0)x(t) + B(\alpha_0)u(t) + \sum_{i=1}^g \left\{ \frac{\partial A}{\partial \alpha_i} \delta \alpha_i x(t) + \frac{\partial B}{\partial \alpha_i} \delta \alpha_i u(t) \right\}$$

In this case, the distribution matrix and unknown disturbance vector are expressed as

$$E = \left[\frac{\partial A}{\partial \alpha_1} \frac{\partial B}{\partial \alpha_1} \cdots \frac{\partial A}{\partial \alpha_g} \frac{\partial B}{\partial \alpha_g} \right]$$

$$d(t) = [\delta \alpha_1 x(t)^T | \delta \alpha_1 u(t) | \cdots | \delta \alpha_g x(t)^T | \delta \alpha_g u(t)^T]^T$$

Definition 1—Unknown input observer (UIO): An observer is defined as an unknown input observer for the system (1), if its state estimation error vector $e(t)$ approaches zero asymptotically regardless of the presence of the unknown input (disturbance) in the system. \square

The problem of designing an observer for a linear system with both known and unknown inputs has been studied for nearly two decades (Wang *et al.* 1975). The problem is of considerable importance since, in practice, there are many situations where there are disturbances present, or when some of the system inputs are inaccessible (or unmeasurable), and therefore a conventional observer, which uses all input signals, cannot be used. It is more useful to assume no *a priori* knowledge about unknown inputs. Wang *et al.* (1975) proposed a minimal-order UIO for the system (1). The existence conditions for such an $(n-m)$ th-order observer were shown by Kudva *et al.* (1980). After the work of Wang *et al.* many approaches for designing unknown input observers have been proposed, for example, the geometric method by Bhattacharyya (1978), the inversion algorithm by Kobayashi and Nakamizo (1982), the matrix algebra method of Watanabe and Himmelblau (1982), the generalized matrix inversion approach by Miller and Mukundan (1982), and the singular value decomposition technique by Fairman *et al.* (1984). Park and Stein (1988) studied the simultaneous estimation problem for both states and unknown input observers. More recently, the problem of designing reduced-order unknown input observers has been revisited by Hou and Müller (1992) and Guan and Saif (1991) using algebraic approaches. In their studies, the state vector is divided into two parts, via a linear transformation onto the state equation; one part can be directly obtained from the measurements, and the other part has to be estimated using a reduced-order disturbance de-coupled observer.

Unlike all the above mentioned work, in which the reduced-order observer structure has been used, Kurek (1982) proposed a full-order UIO structure. Yang and Richard (1988) gave a direct design procedure of the full-order UIO for system (1) and have shown, through an example, that the reduced-order observer may restrict the convergence rate in estimation. However, the design procedure they presented is very complicated and involves some trial-and-error exercises, and the existence conditions are not very easy to verify. It has been shown that the minimal order of a UIO is $(n-m)$, any order between $(n-m)$ to n is possible for a UIO to exist. The disturbance de-coupling conditions for a full-order UIO is not very different from that for a reduced-order UIO. That is to say, there are no significant differences between full-order and reduced-order UIOs, as far as unknown input (disturbance) de-coupling is concerned. However, there is more design freedom available for a full-order UIO to achieve other required performances, such as the rate of convergence and minimum variance. This is easy to understand, since the number of free parameters will increase if the observer order is increased.

In this study, a full-order UIO structure, which is similar to the one in Kurek (1982), is used because extra design freedom is required for generating directional

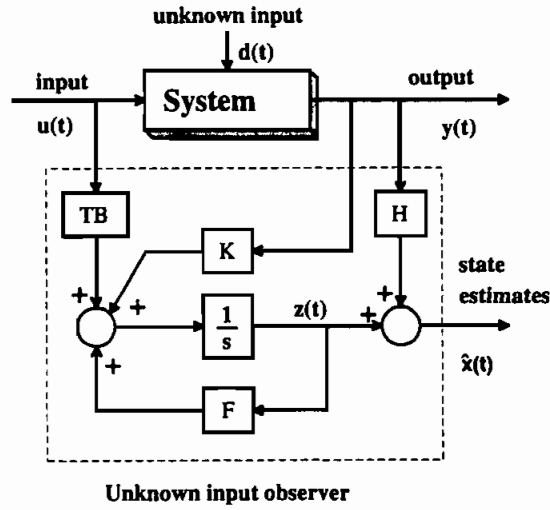


Figure 1. The structure of a full-order unknown input observer.

residuals in fault isolation. A rigorous mathematical foundation in designing a full-order UIO is presented. The necessary and sufficient conditions for this observer to exist are given and thoroughly proved in this paper. These conditions are easy to verify, the design procedure is systematic and straightforward to implement. Moreover, one of the contributions of this paper is to show that the remaining freedom can be used to make the residual have directional properties, after unknown input (or disturbance) de-coupling has been achieved.

The structure for a full-order observer is described as

$$\left. \begin{aligned} \dot{z}(t) &= Fz(t) + TBu(t) + Ky(t) \\ \hat{x}(t) &= z(t) + Hy(t) \end{aligned} \right\} \quad (2)$$

where $\hat{x} \in \mathbb{R}^n$ is the estimated state vector and $z \in \mathbb{R}^n$ is the state of this full-order observer, and F, T, K, H are matrices to be designed for achieving unknown input de-coupling and other design requirements. The observer described by (2) is illustrated in Fig. 1.

When the observer (2) is applied to the system (1), the estimation error ($e(t) = x(t) - \hat{x}(t)$) is governed by the following equation:

$$\begin{aligned} \dot{e} &= (A - HCA - K_1 C)e + [F - (A - HCA - K_1 C)]z \\ &\quad + [K_2 - (A - HCA - K_1 C)H]y \\ &\quad + [T - (I - HC)]Bu + (HC - I)Ed \end{aligned} \quad (3)$$

where

$$K = K_1 + K_2 \quad (4)$$

If one can make the following relations hold true:

$$(HC - I)E = 0 \quad (5)$$

$$T = I - HC \quad (6)$$

$$F = A - HCA - K_1 C \quad (7)$$

$$K_2 = FH \quad (8)$$

The state estimation error will then be

$$\dot{e}(t) = Fe(t) \quad (9)$$

If all eigenvalues of F are stable, $e(t)$ will approach zero asymptotically, i.e. $\hat{x} \rightarrow x$. This means that the observer (2) is an unknown input observer for the system (1) according to Definition 1. The design of this UIO involves solving (4)–(8) and ensuring that all eigenvalues of the system matrix F are stable. Before we give the necessary and sufficient conditions for existence of a UIO, two Lemmas are introduced.

Lemma 1: Equation (5) is solvable if and only if

$$\text{rank}(CE) = \text{rank}(E) \quad (10)$$

and a special solution is

$$H^* = E[(CE)^T CE]^{-1} (CE)^T \quad (11)$$

Proof: For the proof see Appendix A. \square

Lemma 2: Let

$$C_1 = \begin{bmatrix} C \\ CA \end{bmatrix} \quad (12)$$

then the detectability for the pair (C_1, A) is equivalent to that for the pair (C, A) .

Proof: For the proof see Appendix B. \square

Note that the detectability (Chen 1984) is a weaker condition than observability. A pair (C, A) is detectable when all unobservable modes for this pair are stable.

Theorem 1: Necessary and sufficient conditions for (2) to be a UIO for the system defined by (1) are:

- (i) $\text{rank}(CE) = \text{rank}(E)$
- (ii) (C, A_1) is a detectable pair, where

$$A_1 = A - E[(CE)^T CE]^{-1} (CE)^T CA \quad (13)$$

Proof:

Sufficiency. According to Lemma 1, (5) is solvable when condition (i) holds true. A special solution for H is $H^* = E[(CE)^T CE]^{-1} (CE)^T$. In this case, the system dynamics matrix is

$$F = A - HCA - K_1 C = A_1 - K_1 C$$

which can be stabilized by selecting the gain matrix K_1 due to the condition (ii). Finally, the remaining matrices of the observer described in (2) can be calculated using (4)–(8). Thus, the observer (2) is a UIO for the system (1).

Necessity. Since (2) is a UIO for (1), Equation (5) is solvable. This leads to the fact that condition (i) holds true according to Lemma 1. The general solution for the matrix H for (5) can be calculated as

$$H = E(CE)^* + H_0[I_m - CE(CE)^*]$$

where $H_0 \in \mathbb{R}^{n \times m}$ an arbitrary matrix and $(CE)^*$ is the left inverse of CE which is

$$(CE)^* = [(CE)^T CE]^{-1} (CE)^T$$

Substituting the solution for H into (7), the system dynamics matrix F is

$$\begin{aligned}
 F &= A - HCA - K_1 C \\
 &= [I_n - E(CE)^*C]A - [K_1 \ H_0] \begin{bmatrix} C \\ [I_m - CE(CE)^*]CA \end{bmatrix} \\
 &= A_1 - [K_1 \ H_0] \begin{bmatrix} C \\ CA_1 \end{bmatrix} \\
 &= A_1 - \bar{K}_1 \bar{C}_1
 \end{aligned}$$

where

$$\bar{K}_1 = [K_1 \ H_0] \quad \text{and} \quad \bar{C}_1 = \begin{bmatrix} C \\ CA_1 \end{bmatrix}$$

Since the matrix F is stable, the pair (\bar{C}_1, A_1) is detectable, and the pair (C, A_1) is also detectable according to Lemma 2. \square

It can be noted that the number of independent rows of the matrix C must not be less than the number of the independent columns of the matrix E to satisfy condition (i). That is to say, the maximum number of disturbances that can be de-coupled cannot be larger than the number of the independent measurements. Condition (ii) can be verified in terms of the structural properties of the original system. In fact, this condition is equivalent to the condition that the transmission zeros from the unknown inputs to the measurements must be stable, i.e.

$$\begin{bmatrix} sI_n - A & E \\ C & 0 \end{bmatrix}$$

is of full column rank for all s with $\text{Re}(s) \geq 0$. This can be proved as follows. It can be verified that

$$\begin{bmatrix} I_n - E(CE)^*C & sE(CE)^* \\ 0 & I_m \\ E(CE)^*C & -sE(CE)^* \end{bmatrix} \begin{bmatrix} sI_n - A & E \\ C & 0 \end{bmatrix} = \begin{bmatrix} sI_n - A_1 & 0 \\ C & 0 \\ -E(CE)^*CA & E \end{bmatrix}$$

As the first matrix in the left side of the above equation is a full column rank matrix, we have

$$\text{rank} \begin{bmatrix} sI_n - A & E \\ C & 0 \end{bmatrix} = \text{rank} \begin{bmatrix} sI_n - A_1 & 0 \\ C & 0 \\ -E(CE)^*CA & E \end{bmatrix} = \text{rank} \begin{bmatrix} sI_n - A_1 \\ C \\ -E(CE)^*CA \end{bmatrix} + \text{rank}(E)$$

We have assumed that E is a full column rank matrix. Hence, condition (ii) is equivalent to the case when the matrix of the left-hand side of the above equation is full column rank for all s with $\text{Re}(s) \geq 0$. This is because the condition for pair (C, A_1) to be detectable is equivalent to the following matrix

$$\begin{bmatrix} sI - A_1 \\ C \end{bmatrix}$$

having full column rank for all s with $\text{Re}(s) \geq 0$.

From the above analysis, it can be seen that K_1 is a free matrix of parameters in the

design of a UIO. After K_1 is determined, other parameter matrices in the UIO can be computed by (4)–(8). The only restriction on the matrix K_1 is that it must stabilize the system dynamics matrix F . The matrix K_1 , which stabilizes the matrix F , is not unique due to the multivariable nature of the problem. That is to say, there is still some design freedom left in the choice of K_1 , after unknown input disturbance conditions have been satisfied. In §3, this freedom is exploited to make the diagnostic residual have directional properties that can be utilized to fulfil the fault isolation task.

3. Disturbance de-coupled fault detection filters and robust fault isolation

3.1. Basic principles of fault detection filters

The Beard fault detection filter was first developed by Beard (1971) using a matrix algebra approach and later reformed by Jones (1973) in a vector space notation. The theory of BFDFs has been extended by many researchers, for example, Massoumnia (1986) used a geometric interpretation, White and Speyer (1987) improved the design procedure using a spectral approach that is suitable for the isolation of multiple faults, and more recently Park and Rizzoni (1993) developed a closed-form expression of BFDFs using eigenstructure assignment.

In order to describe the BFDF theory, let us consider a system without disturbances in the state-space format as

$$\begin{cases} \dot{x}(t) = Ax(t) + Bu(t) + b_i \xi_i(t) \\ y(t) = Cx(t) + I_j \eta_j(t) \end{cases} \quad (14)$$

The term $b_i \xi_i(t)$ ($i = 1, 2, \dots, r$) denotes that a fault occurs in the i th actuator, $b_i \in \mathbb{R}^n$ is the i th column of the input matrix B and is defined as the fault event vector of the i th actuator fault, and $\xi_i(t)$ is an unknown scalar time-varying function that represents the evolution of the fault. The term $I_j \eta_j(t)$ ($j = 1, 2, \dots, m$) denotes that a fault occurs in the j th sensor, $I_j \in \mathbb{R}^m$ is a unit vector corresponding to a fault with the j th sensor. Note that component faults appear in the system equation in the same way as the actuator fault and, hence, are not discussed further here.

A BFDF is just a full-order observer and its structure and the residual can be described as

$$\begin{cases} \dot{\hat{x}}(t) = A\hat{x}(t) + Bu(t) + K(y(t) - C\hat{x}(t)) \\ r(t) = y(t) - C\hat{x}(t) \end{cases} \quad (15)$$

where $r \in \mathbb{R}^m$ is the residual vector, $\hat{x} \in \mathbb{R}^n$ is the state estimation, and $K \in \mathbb{R}^{m \times n}$ is the observer gain matrix that has to be specially designed to make the residual have restricted uni-directional properties in the presence of a particular fault. If the state estimation error is defined as: $e(t) = x(t) - \hat{x}(t)$, the residual and $e(t)$ will be governed by the following error system, when a fault occurs in the i th actuator

$$\begin{cases} \dot{e}(t) = (A - KC)e(t) + b_i \xi_i(t) \\ r(t) = Ce(t) \end{cases} \quad (16)$$

When a fault occurs in the j th sensor, the error system will be

$$\begin{cases} \dot{e}(t) = (A - KC)e(t) - k_j \eta_j(t) \\ r(t) = Ce(t) + I_j \eta_j(t) \end{cases} \quad (17)$$

where k_j is the j th column of the detection filter gain matrix.

The task of BFDF design is to make $Ce(t)$ have a fixed direction in the output space responding to either $b_i \xi_i(t)$ or $k_j \eta_j(t)$. Both actuator and sensor fault situations can be considered in the following general error system equation

$$\left. \begin{aligned} \dot{e}(t) &= (A - KC)e(t) + l_i \eta_i(t) \\ r(t) &= Ce(t) \end{aligned} \right\} \quad (18)$$

where $l_i \in \mathbb{R}^n$ is called the fault event direction. The definition of the ‘insolability’ of a fault with known direction l_i is given by Beard (1971) as stated below.

Definition 2—Isolability of a fault with a given direction: The fault associated with l_i in the system described by (18) is isolable if there exists a filter gain matrix K such that:

- (a) $r(t)$ maintains a fixed direction in the output space, and
- (b) $(A - KC)$ can be stabilized. □

Condition (a) guarantees that the residual has uni-directional characteristics. This condition is equivalent to the rank of the controllability matrix of the (A, l_i) pair being one, i.e.

$$\text{rank} [l_i (A - KC) l_i \cdots (A - KC)^{n-1} l_i] = 1$$

Condition (b) ensures the convergency of the filter. In the original definition of Beard (1971), condition (b) requires that the eigenvalues $(A - KC)$ can be arbitrarily assigned. This condition has been modified, as the stability requirement is sufficient if the residual response time does not need to be specified. This definition was referred to as ‘fault detectability’ by Beard (1971) and others. We feel that the ‘isolability’ is more appropriate, because the directional property of residuals is especially desirable for fault isolation purposes, although it can also be used for fault detection. The BFDF is designed to satisfy fault isolability.

A filter (or an observer) is called a BFDF if its residual has uni-directional properties. If a fault associated with the direction b_i is isolable, the residual of the BFDF will be fixed in the direction parallel to Cb_i when a fault occurs in the i th actuator. Similarly, the residual will lie somewhere in the plane defined by Ck_j and l_j when a fault occurs in the j th sensor.

In order to isolate faults associated with p isolable fault event directions l_i ($i = 1, 2, \dots, p$), the following output separability condition (Beard 1971) must be satisfied.

Definition 3—Output separability of faults: The faults associated with p fault event directions l_i ($i = 1, 2, \dots, p$) are separable in the residual space if the vectors Cl_1, Cl_2, \dots, Cl_p are linearly independent. □

Output separability is necessary for a group of faults to be isolated in the residual space according to their signature directions. The directions Cl_i ($i = 1, 2, \dots, p$) are called the fault signature directions in the residual space.

Definition 4—Mutual isolability: The faults associated with the fault event vectors l_i ($i = 1, 2, \dots, p$) are mutually isolable if there exists a filter gain matrix K that satisfies the isolability conditions of Definition 2 for all l_i ($i = 1, 2, \dots, p$), i.e.

$$\text{rank} [l_i (A - KC) l_i \cdots (A - KC)^{n-1} l_i] = 1 \quad \text{for all } i = 1, 2, \dots, p \quad \square$$

A group of mutually isolable faults can be isolated using the residual generated by a single BFDF by comparing the residual direction with fault signature directions, when there are no simultaneous faults. The condition for mutual isolability can be found in

Beard (1971), Jones (1973), White and Speyer (1987). If a group of faults is not mutually isolable, it can be divided into a number of subgroups, and each subgroup is mutually isolable. For such cases, a few fault detection filters are required to fulfil the fault isolation task. In any case, only a minimum number of filters are required for fault isolation. This is the most important and appealing advantage of the fault detection filter approaches.

In conclusion, the task of designing a fault detection filter is to make the residual have a uni-directional property by choosing the gain matrix K . Design techniques for this can be found in the classical literature on fault detection filters (Beard 1971, Jones 1973, White and Speyer 1987).

3.2. Disturbance de-coupled fault detection filters and robust fault isolation

It can be seen that uncertain factors associated with a dynamical system, such as disturbances and modelling errors, have not been considered in the design of BFDFs. This is the main disadvantages of BFDFs, because uncertain factors are unavoidable in real systems and any fault detection and isolation scheme has to be made robust against disturbances and modelling errors. Now, consider a system with disturbance term $Ed(t)$ and possible sensor and actuator faults described as

$$\left. \begin{aligned} \dot{x}(t) &= Ax(t) + Bu(t) + Ed(t) + b_i \xi_i(t) \\ y(t) &= Cx(t) + I_j \eta_j(t) \end{aligned} \right\} \quad (19)$$

If a standard BFDF described in (15) is applied to such a system, the state estimation error and residual will be

$$\left. \begin{aligned} \dot{e}(t) &= (A - KC)e(t) + Ed(t) + b_i \xi_i(t) - k_j \eta_j(t) \\ r(t) &= Ce(t) + I_j \eta_j(t) \end{aligned} \right\} \quad (20)$$

It is clear from (20) that all faults and disturbances affect the residual. It is not easy to discriminate between faults and disturbances if this residual is used to detect and isolate faults. Hence, it is necessary to de-coupled disturbance effects from the residual for reliable diagnosis.

We have shown that the disturbance can be de-coupled from the state estimation error using an unknown input observer. This inspires us to generate the residual using a UIO described in (2), the residual is thus defined as

$$r(t) = y(t) - C\hat{x}(t) = (I - CH)y(t) - Cz(t) \quad (21)$$

When this UIO-based residual generator is applied to the system described by the model of (20), the residual and the state estimation error ($e(t)$) will be

$$\left. \begin{aligned} \dot{e}(t) &= (A_1 - K_1 C)e(t) + Tb_i \xi_i(t) \\ r(t) &= Ce(t) \end{aligned} \right\} \quad (22)$$

when a fault occurs in the i th actuator.

Similarly

$$\left. \begin{aligned} \dot{e}(t) &= (A_1 - K_1 C)e(t) - k_{1j} \eta_j(t) - h_j \dot{\eta}_j(t) \\ r(t) &= Ce(t) + I_j \eta_j(t) \end{aligned} \right\} \quad (23)$$

when a fault occurs in the j th sensor, where k_{1j} is the j th column of the matrix K_1 and h_j is the j th column of the matrix H . From (22) and (23), it can be seen that the

disturbance effects have been de-coupled from the residual. This robust (in the disturbance de-coupling sense) residual can be used to detect faults according to a simple threshold logic

$$\left. \begin{array}{l} \|r(t)\| < \text{threshold} \quad \text{for no fault cases} \\ \|r(t)\| \geq \text{threshold} \quad \text{for faulty cases} \end{array} \right\} \quad (24)$$

As pointed out in the introduction, fault isolation can be facilitated using uni-directional residual vectors. Consequently, we have to make the residual generated by a UIO have the correct directional properties in order to achieve robust fault isolation. From the design of UIOs, we know that the matrix K_1 can be designed arbitrarily after the robust (in the sense of disturbance de-coupling) conditions have been satisfied. This design freedom can be exploited to make the residual have the uni-directional property.

Comparing the error system (22) with (16), it can be seen that the actuator fault is expressed in the same way for a UIO or a standard BFDF. Hence, the theory for the design of a BFDF (Beard 1971, Jones 1973, White and Speyer 1987) can be directly used to design the matrix K_1 , if the vector b_i is replaced by Tb_i and the matrix A is replaced by A_1 .

Comparing the error system (23) with (17), it can be seen that the sensor fault is also expressed in a similar way for both BFDF and UIO, except an extra term $h_j\dot{\eta}_j(t)$ occurs in the error equation of the UIO. Fortunately, this term can be treated in the same way as an actuator fault. Hence, the theory of BFDF can be adopted for the design of K_1 in the sensor isolation problem. However, it must be pointed out that the residual will lie in a subspace spanned by vectors I_j , Ck_{1j} and Ch_j when the residual uni-directional property has been satisfied. For constant sensor faults, the term $h_j\dot{\eta}_j(t)$ will disappear from the error system and the residual will lie in the plane spanned by the vectors I_j and Ck_{1j} , this is same as the BFDF.

It should be pointed out that the maximum number of faults that can be mutually isolated using an observer developed in this paper could be less than the number that can be isolated by a standard BFDF. This is because some of the design freedom has been used to achieve robustness. Hence, the number of detection filters needed to isolate the same group of faults could be increased. However, the price is worth paying if the robustness is an essential property. In fact, this is the normal case for real applications.

To combine the theory of UIOs with the theory of BFDFs, the design procedure for a robust (disturbance de-coupled) fault detection filter is summarized as follows.

- (a) Compute matrices H and T using (11) and (6), to satisfy disturbance de-coupling conditions.
- (b) Compute A_1 using (13).
- (c) Compute K_1 to satisfy the uni-direction property using the theory of Beard fault detection filters.
- (d) Compute the observer gain matrix K using (8) and (4).

The key step is then to design the matrix K_1 ; once it is available, the computation of other matrices is very straightforward. The design procedure of BFDFs can be found in the literature (Beard 1971, Jones 1973, White and Speyer 1987) and is not presented in this paper. To show the basic idea, we discuss an ideal situation, in which the number of independent measurements is equal to the number of states, i.e. $\text{rank}(C) =$

n . In this situation, all eigenvalues of the matrix $A_1 - K_1 C$ can be assigned to the same value $-\sigma < 0$, i.e.

$$A_1 - K_1 C = -\sigma I$$

This can be achieved by setting K_1 as

$$K_1 = (A_1 + \sigma I) C^* \quad (25)$$

where C^* is the pseudo-inverse of C . For this design, the residual will be

$$\begin{aligned} r(t) &= Ce(t) + I_j \eta(t) \\ &= I_j \eta(t) + Ce^{-\sigma(t-t_0)} e(t_0) + C \int_{t_0}^t e^{-\sigma(t-\tau)} [Tb_i \xi(\tau) - k_{1j} \eta(\tau) - h_j \dot{\eta}(\tau)] d\tau \\ &= Ce^{-\sigma(t-t_0)} e(t_0) + CTb_i \int_{t_0}^t e^{-\sigma(t-\tau)} \xi(\tau) d\tau \\ &\quad + I_j \eta(t) - Ck_{1j} \int_{t_0}^t e^{-\sigma(t-\tau)} \eta(\tau) d\tau - Ch_j \int_{t_0}^t e^{-\sigma(t-\tau)} \dot{\eta}(\tau) d\tau \\ &= Ce^{-\sigma(t-t_0)} e(t_0) + CTb_i \alpha(t, t_0) + I_j \eta(t) + Ck_{1j} \beta(t, t_0) + Ch_j \gamma(t, t_0) \end{aligned}$$

Clearly, the residual is parallel to CTb_i after the transient has settled down, following a fault in the i th actuator. Similarly, the residual will lie in the subspace spanned by vectors I_j , Ck_{1j} and Ch_j , when a fault occurs in the j th sensor.

Due to the residual directional property, the fault can be isolated by comparing the residual direction with the fault signature directions (or subspaces).

Definition 5: The direction of CTb_i is called as the *signature direction* of the i th actuator fault. \square

The directional relationship between two vectors CTb_i and $r(t)$ can be quantified by the correlation parameter CORR_i

$$\text{CORR}_i(t) = \frac{|(CTb_i)^T r(t)|}{\|CTb_i\|_2 \|r(t)\|_2} \quad (26)$$

If $\text{CORR}_j > \text{CORR}_k$, the fault is more likely in the j th actuator rather than in the k th actuator.

Definition 6: The *signature subspace* of the j th sensor fault is defined as

$$R_j = \text{Span} \{I_j, Ck_{1j}, Ch_j\} \quad (27)$$

\square

The relationship between the vector $r(t)$ with the subspace R_j can be measured by the relationship between the vector $r(t)$ with its projection $r_j^*(t)$ in the subspace R_j . This is quantified by

$$\text{CORR}_j(t) = \frac{|(r_j^*)^T r(t)|}{\|r_j^*\|_2 \|r(t)\|_2} \quad (28)$$

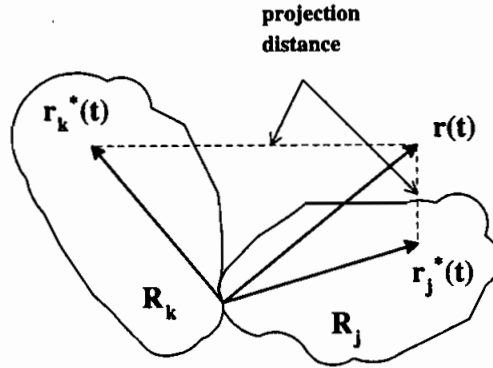


Figure 2. Fault isolation based on directional residuals.

where the projection $r_j^*(t)$ of $r(t)$ in R_j is

$$r_j^*(t) = \Phi_j (\Phi_j^T \Phi_j)^{-1} \Phi_j^T r(t) \quad (29)$$

where

$$\Phi_j = [I_j C k_{1j} C h_j]$$

If $\text{CORR}_j > \text{CORR}_k$, the fault is more likely in the j th sensor rather than in the k th sensor. The relationship between a residual vector with the signature subspace can also be judged by the normalized projection distance, which is defined as

$$\text{NPD}_j = \frac{\|r(t) - r_j^*(t)\|_2}{\|r(t)\|_2} \quad (30)$$

when NPD_j is the smallest one amongst all NPD_j ($j = 1, 2, \dots, m$), we will declare that the j *th sensor is faulty. The idea of fault isolation by comparing the residual direction with the signature subspace is shown in Fig. 2.

4. Robust isolation of faulty sensors in a jet engine system

To control a jet engine efficiently and to monitor a jet engine effectively, the sensors have to perform reliably. However, the sensors in a jet engine work in a very harsh environment and could fail during normal engine operation. Hence, the detection of fault sensors in jet engine systems is very important and has become an active research field (Merrill 1989, 1990, Meserole 1981, Patton and Chen 1992). A simplified nonlinear dynamic model of a jet engine control system can be described as

$$\left. \begin{aligned} \dot{X}_1(t) &= f_1(X_1, X_2, X_3) \\ \dot{X}_2(t) &= f_2(X_1, X_2, X_3) \\ \dot{X}_3(t) &= 10(U - X_3) \end{aligned} \right\} \quad (31)$$

where

$X_1 = n_L \mapsto$ Low pressure rotor speed

$X_2 = n_H \mapsto$ High pressure rotor speed

$X_3 = W_f \mapsto$ Main burner fuel flow

$U = W_{fe} \mapsto$ Fuel flow command

The jet engine is a very complicated nonlinear dynamic system. The nonlinear functions such as $f_1(X_1, X_2, X_3)$ and $f_2(X_1, X_2, X_3)$ cannot be written out analytically.

The system behaviour is normally expressed in a nonlinear dynamic simulation package (Merrill and Leininger 1981, Merrill 1990, Messerole 1981). This package is capable of simulating the entire operating envelope of the engine, and can also generate a linearized model for any operating points. Let us define the following non-dimensional variables

$$x_1 = \frac{X_1 - X_1^0}{X_1^0}, \quad x_2 = \frac{X_2 - X_2^0}{X_2^0}$$

$$x = \frac{X_3 - X_3^0}{X_3^0}, \quad u = \frac{U - U^0}{U^0}$$

where superscript '0' denotes the values at equilibrium. The system can be linearized around an operating point. If u is small (e.g. 1%), x_1 , x_2 and x_3 will be small, i.e. all variables have a small variation around the equilibrium and the following linear model is derived

$$\begin{cases} \dot{x}(t) = Ax(t) + Bu(t) \\ y(t) = Cx(t) \end{cases}$$

where the state is $x = [x_1 \ x_2 \ x_3]^T$ and the measurement vector is

$$y = [x_1 \ x_2 \ x_3 \ p_2 \ p_4 \ t_4]^T$$

in which

$$p_2 = \frac{P_2 - P_2^0}{P_2^0}, \quad p_4 = \frac{P_4 - P_4^0}{P_4^0}, \quad t_4 = \frac{T_4 - T_4^0}{T_4^0}$$

where

$P_2 \mapsto$ High pressure compressor discharge pressure

$P_4 \mapsto$ Turbine discharge pressure

$T_4 \mapsto$ Turbine exit temperature

When the equilibrium is set at $N_L = 450$ (RPM), the linear model matrices are

$$A = \begin{bmatrix} -1.5581 & 0.6925 & 0.3974 \\ 0.2619 & -2.2228 & 0.2238 \\ 0 & 0 & -10 \end{bmatrix} \quad B = \begin{bmatrix} 0 \\ 0 \\ 10 \end{bmatrix}$$

$$C = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0.55107 & 0.13320 & 0.30603 \\ 0.55217 & 0.13526 & 0.32912 \\ -0.25693 & -0.23625 & 0.61299 \end{bmatrix}$$

A BFDF described by (15) is designed to isolate sensor faults. If all eigenvalues of the filter are set as -3 , the gain matrix can be determined as $K = (3I + A)C^*$ because $\text{rank}(C) = 3$. The fault isolation scheme is applied to the nonlinear simulation model. A reliable diagnostic scheme should perform well for a wide range of operating conditions, hence the input is set at $u = 20\%$ in the simulation. The sensor fault is simulated as 2% offset around the normal measurement. In the simulation, we only consider the fault in sensors No. 1, No. 2 and No. 3, i.e. the low-pressure rotor speed

Faulty sensor	No. 1	No. 2	No. 3
NPD ₁	0.37090	0.77783	0.66389
NPD ₂	0.93117	0.95527	0.42455
NPD ₃	0.96529	0.71161	0.31559

Table 1. Fault isolation using Beard fault detection filter.

sensors, the high-pressure rotor speed sensor and the main burner fuel flow sensor. After the transient has settled down, the normalized projection distances for different faulty situations are shown in Table 1.

From Table 1, it can be seen that the fault in sensor No. 1 (or No. 3) can be correctly isolated as the corresponding normalized projection distance NPD₁ (or NPD₃) is the smallest one. However, the fault in sensor No. 2 will be mis-reported as a fault in sensor No. 3 as NPD₃ is the smallest one amongst all normalized projection distances. Moreover, the smallest NPD is not significantly different from other NPDs, and this could make isolation difficult when there is noise in the system.

The example in Table 1 illustrates the importance of robustness in fault isolation. The mis-isolation problem is possibly caused by the linearized errors, as the fault isolation scheme is based on the linear model and this scheme is applied to the original nonlinear system. In the model linearization, we only considered the first-order terms in the Taylor expansion. To model a system more accurately, we can consider the inclusion of second-order terms in the system dynamic equation as follows

$$\dot{x}(t) = Ax(t) + Bu(t) + Ed(x(t)) \quad (32)$$

where the matrices A and B are the same as the linear model. The term $Ed(x)$ represents modelling errors and the vector $d(x)$ consists of the second-order terms of $x(t)$ as

$$d(x) = [x_1^2 \quad x_2^2 \quad x_3^2 \quad x_1x_2 \quad x_1x_3 \quad x_2x_3]^T$$

The distribution E can be obtained using an identification procedure based on the least-squares method. Given a series of values $u^{(1)}, u^{(2)}, \dots, u^{(N)}$ for input u , we can obtain the corresponding steady responses $x^{(1)}, x^{(2)}, \dots, x^{(N)}$ and $d^{(1)}, d^{(2)}, \dots, d^{(N)}$, which are related by the following steady-state equations

$$\left. \begin{aligned} Ax^{(1)} + Bu^{(1)} + Ed^{(1)} &= 0 \\ Ax^{(2)} + Bu^{(2)} + Ed^{(2)} &= 0 \\ &\vdots \\ Ax^{(N)} + Bu^{(N)} + Ed^{(N)} &= 0 \end{aligned} \right\}$$

If N is greater than the dimension of $d(x)$, the least-squares estimate of the matrix E is given as

$$\hat{E} = (\Gamma^* \Psi)^T$$

where Γ^* is the pseudo-inverse of Γ and

$$\Gamma = \begin{bmatrix} (d^{(1)})^T \\ (d^{(2)})^T \\ \vdots \\ (d^{(N)})^T \end{bmatrix} \quad \Psi = - \begin{bmatrix} (Ax^{(1)} + Bu^{(1)})^T \\ (Ax^{(2)} + Bu^{(2)})^T \\ \dots \\ (Ax^{(N)} + Bu^{(N)})^T \end{bmatrix}$$

Faulty sensor	No. 1	No. 2	No. 3
NPD_1	0.00621	0.86727	0.90677
NPD_2	0.88625	0.00213	0.56602
NPD_3	0.89433	0.02092	0.00159

Table 2. Fault isolation using a robust fault detection filter.

From the simulation, we have

$$\hat{E} = \begin{bmatrix} 1.3293 & 3.4440 & 0.1375 & -5.1304 & -1.7826 & -1.8719 \\ 5.6812 & -0.5281 & -0.3385 & -1.6193 & 0.5229 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix}$$

The \hat{E} is not a full column matrix ($\text{rank}(\hat{E}) = 2$) and should be decomposed as $\hat{E} = E_1 E_2$. Here, E_1 is a full column matrix and will be used in the robust fault detection filter design.

$$E_1 = \begin{bmatrix} 6.2006 & 2.8639 \\ 4.1048 & -4.3262 \\ 0 & 0 \end{bmatrix}$$

All eigenvalues of the robust fault detection filter are set to -3 . Using the design procedure presented in this paper, with E replaced by E_1 , the parameter matrices of the robust fault detection filter are obtained as

$$H = \begin{bmatrix} 0.6117 & -0.1170 & 0 & 0.3215 & 0.3220 & -0.1295 \\ -0.1170 & 0.9382 & 0 & 0.0605 & 0.0623 & -0.1916 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix}$$

$$T = \begin{bmatrix} 0 & 0 & -0.1251 \\ 0 & 0 & 0.0783 \\ 0 & 0 & 1.0000 \end{bmatrix}$$

$$K = \begin{bmatrix} -0.0708 & 0.0443 & 0.5658 & 0.1400 & 0.1531 & 0.3540 \\ 0.0443 & -0.0277 & -0.3540 & -0.0876 & -0.0958 & -0.2215 \\ 0.5658 & -0.3540 & -4.5229 & -1.1193 & -1.2239 & -2.8297 \end{bmatrix}$$

This robust fault detection filter is also applied to the nonlinear simulation model to isolate faults in sensors 1, 2 and 3. To compare the isolation performance with the standard BFDF, the system and fault simulation have been set as exactly the same. The normalized projection distances for different faulty situations are shown in Table 2.

From Table 2, one can see that NPD_i ($i = 1, 2, 3$) is the smallest one amongst all normalized projection distances when a fault occurs in the i th sensor. Moreover, the smallest NPD is significantly different from other NPDs. This simulation shows that the fault can be correctly isolated using a robust fault detection filter, even in the presence of modelling errors.

5. Conclusions

This paper has studied the design of robust fault detection filters whose residuals have both disturbance de-coupling and uni-directional properties. This problem is solved using a combination of unknown input observer and Beard fault detection filter principles. To facilitate this combination, a new full-order unknown input observer structure is proposed, and the necessary and sufficient conditions are presented and proved. The design procedure for this observer is very straightforward. After satisfying the disturbance de-coupling conditions, the remaining design freedom can be utilized to make the residual have directional properties, based on Beard fault detection filter techniques. Robust and directional residuals can easily be used in reliable fault isolation. The paper has studied the sensor fault isolation problem for a jet engine control system. This is a highly nonlinear system, and the linearization error causes unreliable isolation if the robustness issues are not considered at the design stage. To cope with this problem, the paper has developed a second model to account for the linearization errors. Based on this model, a robust fault detection filter is designed and applied to the nonlinear jet engine simulation model and the results show the effectiveness of the robust fault isolation strategy developed in the paper. The technique can be applied to robust fault isolation for a wide range of systems with uncertain factors.

ACKNOWLEDGMENTS

The research on fault diagnosis of R. J. Patton and J. Chen is supported by the U.K. Engineering and Physical Science Research Council (EPSRC) under grant GR/J12758. H. Y. Zhang also acknowledges funding support from the UK EPSRC for a six month visiting fellowship (Ref. H16773) which enabled him to collaborate with R. J. Patton and J. Chen at the University of York on fault diagnosis research. H. Y. Zhang wish to thank Y. L. Wang and K. W. Du of Beijing University of Aeronautics & Astronautics for their assistance in preparing the jet engine system simulation. J. Chen values H. G. Zhang's help for reading through some of the proofs in this paper. Thanks are also expressed to the reviewers for their helpful comments in improving this paper.

Appendix A: Proof of Lemma 1

Necessity. When (5) has a solution H , we have $HCE = E$ or

$$(CE)^T H^T = E^T$$

i.e. E^T belongs to the range space of the matrix $(CE)^T$ and this leads to

$$\text{rank}(E^T) \leq \text{rank}((CE)^T)$$

i.e.

$$\text{rank}(E) \leq \text{rank}(CE)$$

However

$$\text{rank}(CE) \leq \min\{\text{rank}(C), \text{rank}(E)\} \leq \text{rank}(E)$$

Hence, $\text{rank}(CE) = \text{rank}(E)$ and the necessary condition is proved.

Sufficiency. When $\text{rank}(CE) = \text{rank}(E)$ holds true, CE is a full column rank matrix (as we have assumed that E is a full column matrix), and a left inverse of CE exists

$$(CE)^* = [(CE)^T CE]^{-1} (CE)^T$$

Clearly, $H = E(CE)^*$ is a solution to (5). \square

Appendix B: Proof of Lemma 2

If $s_1 \in \mathcal{C}$ is an unobservable mode of the pair (C_1, A) , we have

$$\text{rank} \left\{ \begin{bmatrix} s_1 I - A \\ C_1 \end{bmatrix} \right\} = \text{rank} \left\{ \begin{bmatrix} s_1 I - A \\ C \\ CA \end{bmatrix} \right\} < n$$

This means that a vector $\alpha \in \mathcal{C}^n$ will exist such that

$$\begin{bmatrix} s_1 I - A \\ C \\ CA \end{bmatrix} \alpha = 0$$

This leads to

$$\begin{bmatrix} s_1 I - A \\ C \end{bmatrix} \alpha = 0 \quad \text{or} \quad \text{rank} \left\{ \begin{bmatrix} s_1 I - A \\ C \end{bmatrix} \right\} < n$$

That is to say that s_1 is also an unobservable mode of the pair (C, A) .

If $s_2 \in \mathcal{C}$ is an unobservable mode of the pair (C, A) , we have

$$\text{rank} \left\{ \begin{bmatrix} s_2 I - A \\ C \end{bmatrix} \right\} < n$$

This means that a vector $\beta \in \mathcal{C}^n$ can always be found, such that

$$\begin{bmatrix} s_2 I - A \\ C \end{bmatrix} \beta = 0$$

This leads to

$$\begin{aligned} (s_2 I - A)\beta &= 0 & C\beta &= 0 \\ CA\beta &= Cs_2\beta = s_2 C\beta = 0 \end{aligned}$$

Hence

$$\begin{bmatrix} s_2 I - A \\ C \\ CA \end{bmatrix} \beta = \begin{bmatrix} s_2 I - A \\ C_1 \end{bmatrix} \beta = 0$$

i.e. s_2 is also an unobservable mode of the pair (C_1, A) .

As the pairs (C_1, A) and (C, A) have the same unobservable modes, their detectability is formally equivalent. \square

REFERENCES

- BEARD, R. V., 1971, Failure accommodation in linear system through self reorganization. Ph.D. thesis, Department of Aeronautics and Astronautics, MIT, Cambridge, Massachusetts, U.S.A.
- BHATTACHARYYA, S. P., 1978, Observer design for linear systems with unknown inputs. *IEEE Transactions on Automatic Control*, **23**, 483–484.
- CHEN, C. T., 1984, *Linear System Theory and Design* (New York: Holt, Rinehart and Winston).
- CHEN, J., and ZHANG, H. Y., 1991, Robust detection of faulty actuators via unknown input observers. *International Journal of System Science*, **22**, 1829–1839.
- FAIRMAN, F. W., MAHIL, S. S., and LUK, L., 1984, Disturbance decoupled observer design via singular value decomposition. *IEEE Transactions on Automatic Control*, **29**, 84–86.
- FRANK, P. M., 1990, Fault diagnosis in dynamic system using analytical and knowledge based redundancy—a survey and some new results. *Automatica*, **26**, 459–474.
- FRANK, P. M., and WÜNNENBERG, J., 1989, Robust fault diagnosis using unknown input schemes. *Fault Diagnosis in Dynamic Systems: Theory and Application*, edited by R. J. Patton, P. M. Frank and R. N. Clark (New York: Prentice Hall), Chap. 3, pp. 47–98.
- GERTLER, J., 1993, Residual generation in model-based fault diagnosis. *Control—Theory and Advanced Technology*, **9**, 259–285; 1994, Modelling errors as unknown inputs. *Preprints of IFAC Symposium on Fault Detection, Supervision and Safety for Technical Processes: SAFEPROCESS'94*, Espoo, Finland, Vol. 1, pp. 266–271.
- GERTLER, J., and KUNWER, M. K., 1993, Optimal residual decoupling for robust fault diagnosis. *Proceedings of International Conference on Fault Diagnosis: TOOLDIAG'93*, Toulouse, France, pp. 436–452.
- GUAN, Y., and SAIF, M., 1991, A novel approach to the design of unknown input observers. *IEEE Transactions on Automatic Control*, **36**, 632–635.
- HOU, M., and MÜLLER, P. C., 1992, Design of observers for linear systems with unknown inputs. *IEEE Transactions on Automatic Control*, **37**, 871–875.
- JONES, H. L., 1973, Failure detection in linear systems. Ph.D. Thesis, Department of Aeronautics and Astronautics, MIT, Cambridge, Massachusetts, U.S.A.
- KOBAYASHI, N., and NAKAMIZO, R., 1982, An observer design for linear systems with unknown inputs. *International Journal of Control*, **35**, 605–619.
- KUDVA, P., VISWANADHAM, N., and RAMAKRISHNA, A., 1980, Observers for linear system with unknown inputs. *IEEE Transactions on Automatic Control*, **25**, 113–115.
- KUREK, J., 1982, Observations of the state vector of linear multivariable systems with unknown inputs. *International Journal of Control*, **36**, 511–515.
- MASSOUMNIA, M. A., 1986, A geometric approach to the synthesis of failure detection filters. *IEEE Transactions on Automatic Control*, **31**, 839–846.
- MERRILL, W. C., 1989, Advanced detection, isolation, and accommodation of sensor failures—a real-time evaluation. *Journal of Guidance, Control and Dynamics*, **11**, 517–526; 1990, Sensor failure detection for jet engines. *Control and Dynamics—Advances in Aerospace Systems Dynamics and Control Systems*, edited by C. Leondes, Vol. 33 (New York: Academic Press), pp. 1–33.
- MERRILL, W. C., and LEININGER, G., 1981, Identification and dual adaptive control of a turbojet engine. *International Journal of Control*, **34**, 529–546.
- MESEROLE, J. S., 1981, Detection filters for fault-tolerant control of turbofan engines. Ph.D. Thesis, MIT, Cambridge, Massachusetts, U.S.A.
- MILLER, R. J., and MUKUNDAN, R., 1982, On designing reduced-order observers for linear time-invariant systems subject to unknown inputs. *International Journal of Control*, **35**, 183–188.
- PARK, J., and RIZZONI, G., 1993, A closed-form expression for the fault detection filter. *Proceedings of the 32nd IEEE Conference on Decision and Control*, San Antonio, Texas, U.S.A., pp. 259–264.

- PARK, Y., and STEIN, J. L., 1988, Closed-loop, state and inputs observer for systems with unknown inputs. *International Journal of Control*, **48**, 1121–1136.
- PATTON, R. J., and CHEN, J., 1992, Robust fault detection of jet engine sensor systems using eigenstructure assignment. *Journal of Guidance, Control and Dynamics*, **15**, 1491–1497; 1993a, Optimal unknown input distribution matrix selection in robust fault diagnosis. *Automatica*, **29**, 837–841; 1993b, A survey of robustness in quantitative model-based fault diagnosis. *Applied Mathematics and Computing Science*, **3**, 399–416; 1994, Review of parity space approaches to fault diagnosis for aerospace systems. *Journal of Guidance, Control and Dynamics*, **17**, 278–285.
- PATTON, R. J., CHEN, J., and ZHANG, H. Y., 1992, Modelling methods for improving robustness in fault diagnosis of jet engine system. *Proceedings of the 31st IEEE Conference on Control and Decision*, Tucson, Arizona, U.S.A., pp. 2330–2335.
- PATTON, R. J., FRANK, P. M., and CLARK, R. N. (editors), 1989, *Fault Diagnosis in Dynamic Systems, Theory and Application*. Control Engineering Series (New York: Prentice Hall).
- PATTON, R. J., WILLCOX, S. W., and WINTER, S. J., 1987, A parameter insensitive technique for aircraft sensor fault analysis. *Journal of Guidance, Control and Dynamics*, **10**, 359–367.
- WANG, S. H., DAVISON, E. J., and DORATO, P., 1975, Observing the states of systems with unmeasurable disturbance. *IEEE Transactions on Automatic Control*, **20**, 716–717.
- WATANABE, K., and HIMMELBLAU, D. M., 1982, Instrument fault detection in systems with uncertainties. *International Journal of System Science*, **13**, 137–158.
- WHITE, J. E., and SPEYER, J. L., 1987, Detection filter design: spectral theory and algorithm. *IEEE Transactions on Automatic Control*, **32**, 593–603.
- WILLSKY, A. S., 1976, A survey of design methods for failure detection in dynamic systems. *Automatica*, **12**, 601–611.
- YANG, F., and RICHARD, W. W., 1988, Observers for linear systems with unknown inputs. *IEEE Transactions on Automatic Control*, **33**, 677–681.