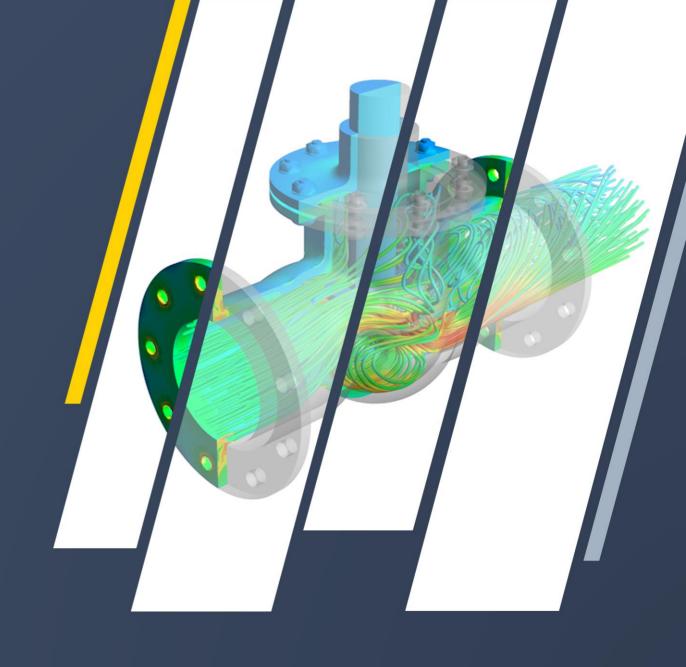


# Workshop 03.1: Pump Analysis using CFD Post

ANSYS CFX Rotating Machinery Modeling

Release 2019 R3



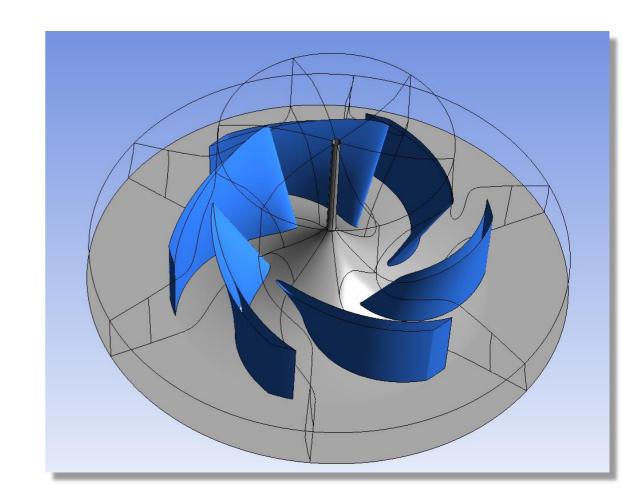
### Introduction

#### Workshop Description:

 This Workshop deals with post-processing aspect for the pump impeller solved in Workshop 1

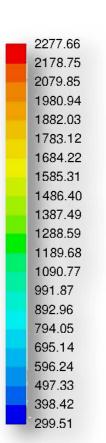
#### • Learning Aims:

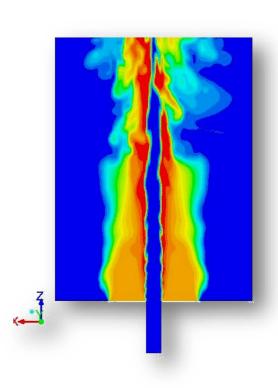
- Turbo-specific post-processing
- Creating CEL expressions to calculate various relevant quantities
- Report generation



#### **CFD-Post**

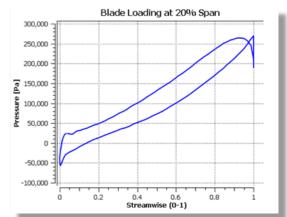
- CFD-Post includes many tools for analyzing general CFD results
  - Isosurfaces
  - Vector plots
  - Contour plots (shaded and graded)
  - Streamlines and particle tracks
  - XY plotting
  - Animation creation

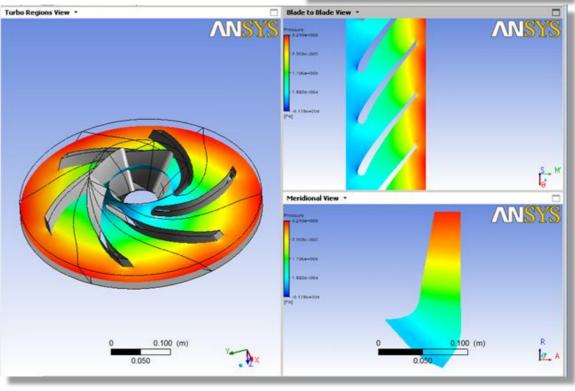




### **Post Processing**

- *CFD-Post* also has turbo specific post-processing capabilites
  - Turbo Specific Post processing views
    - Meridional
    - Blade to Blade
  - Turbo Specifc charts
    - ➤ Blade Loading
    - Circumferential
    - ➤ Inlet to Outlet
    - > Hub to Shroud
  - Turbo Specific surfaces on which to plot vectors, contours, etc.
  - Turbo Specific variables
  - Reports for different types of rotating machines



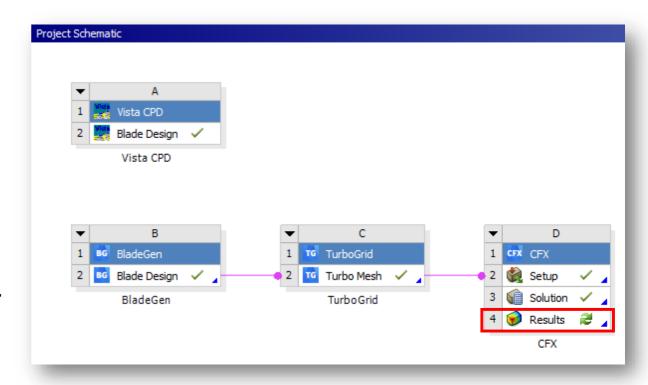


### **Post-processing Goals**

- Configure the solution in Turbo Post
- Calculate turbo velocity components
- Create plots in blade to blade and meridional views
- Create charts for blade loading
- Use CEL to calculate some global quantitative results
- Generate a pre-defined pump report
- Add some customer plots to the report

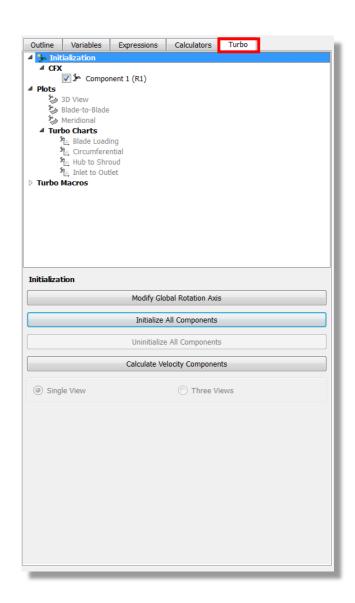
#### **Launch CFD-Post**

- Start CFD-Post
  - If you completed Workshop 1, you can double click on the Results cell to launch CFD-Post
  - Alternatively, you can load the completed Workbench archive *Pump\_solution.wbpz* (provided with the inputs of this workshop)
    - In the Workbench main menu File > Open...
      - In the *Open* dialogue box *Browse to Pump\_solution.wbpz* and click *Open*
      - In the Save As dialogue box edit the File Name to Pump\_results.wbpj and click Save



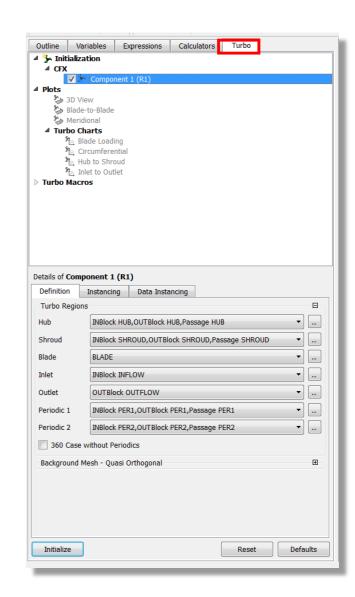
### **Turbo Post**

- Once the case is opened in *CFD-Post*, switch to the *Turbo* Tab
- If prompted to automatically initialize the case, select No



#### **Initialization**

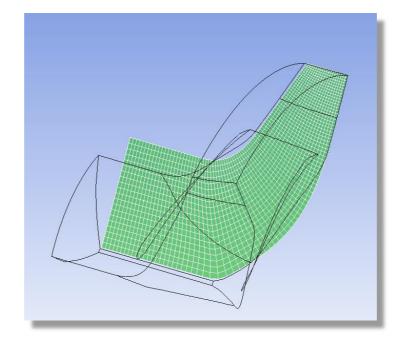
- Initializing the components is the first step
  - This process defines a new set of coordinates and variables that are useful for turbo post-processing
  - These coordinates and variables are based on the hub, shroud, inlet, outlet, blade, and periodic boundaries
  - By default CFD-Post will try to determine these boundaries,
    but they can always be input by the user

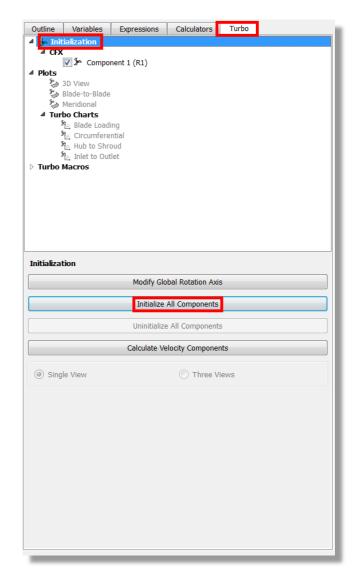




### **Initialization**

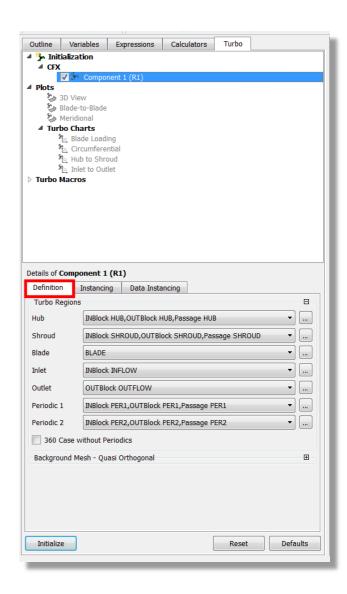
- Under *Initialization* in the *Turbo* Tree, click *Initialize All Components* 
  - This will result in a background mesh being displayed (when clicking on Component 1 in the Turbo Tree)





### **Initialization**

- In this case, all turbo regions have been automatically identified
- If they had not, the user could identify them in the *Definition* Tab

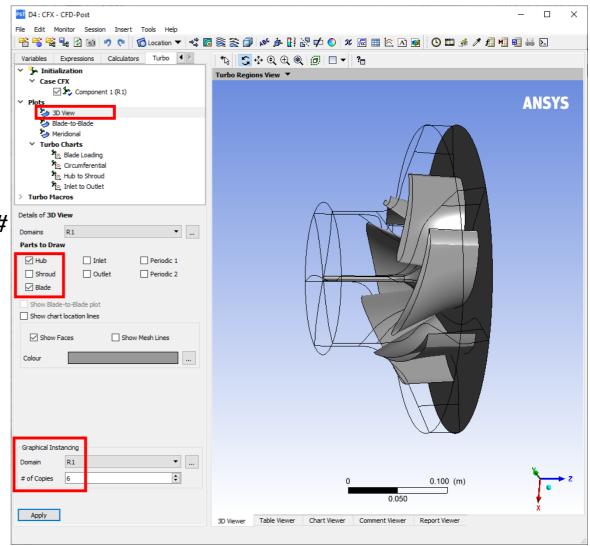




#### **Plots: 3D View**

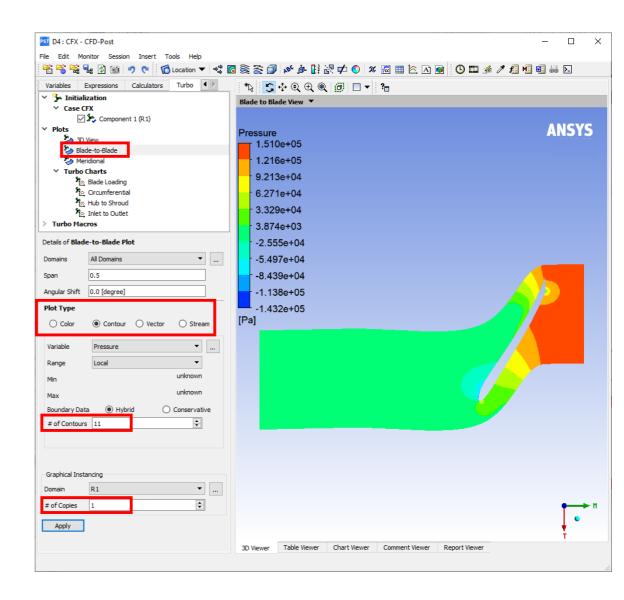
- To visualize the pump geometry (still in the *Turbo* Tab) click on *3D View* 
  - Select *Hub* and *Blade* under *Parts to Draw*
  - Select R1 under Graphical Instancing and set the # of Copies to 6
  - Click Apply

In the remaining slides numerous post-processing objects (flow visualization images, charts, tables, etc.) will be generated. Please notice that there can be differences between the indicative images & values given in the slides and the ones created by you



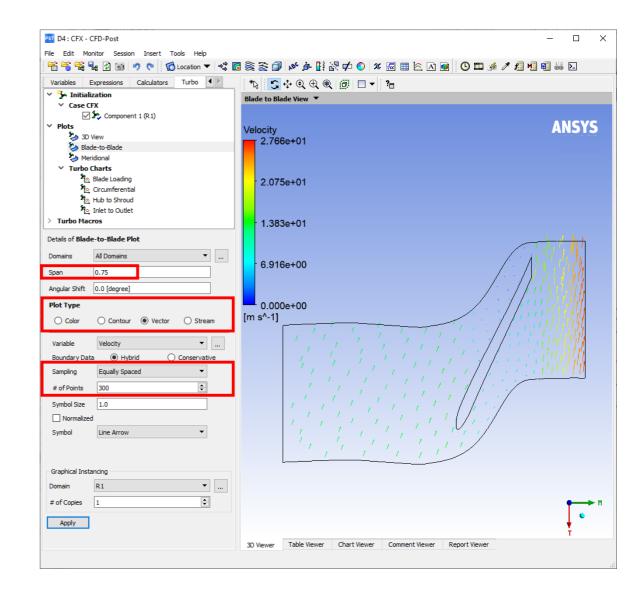
#### Plots: Blade-to-Blade

- Make an unwrapped\* blade-to-blade plot of pressure at mid-span
  - Under Blade-to-Blade in the Tree, set Contour for Plot Type
  - Set Local for Range
  - Set # of Copies back to 1
  - Click Apply
  - \* An "unwrapped" plot is one where the 3D surface contours are transformed so that they lie in a plane, similar to taking a curved piece of paper and unwrapping it to lie on a flat table



#### Blade-to-Blade

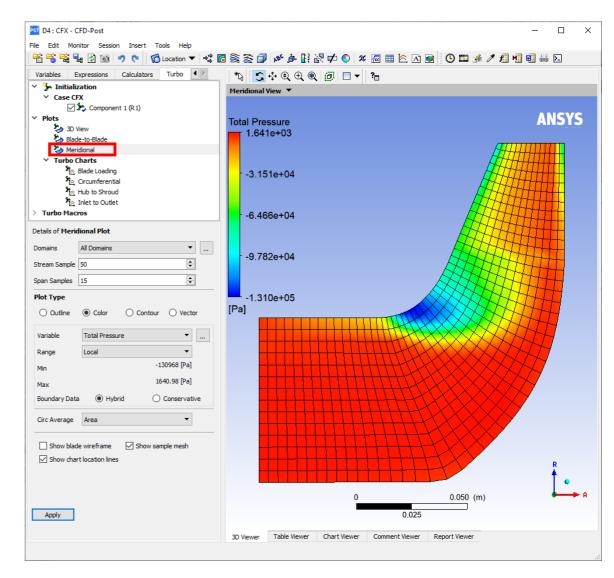
- Change this to a *Vector* plot at 75% span
  - Change *Span* to 0.75
  - Change Plot Type to Vector
  - Set Sampling to Equally Spaced
  - Set # of Points to 300
  - Click Apply





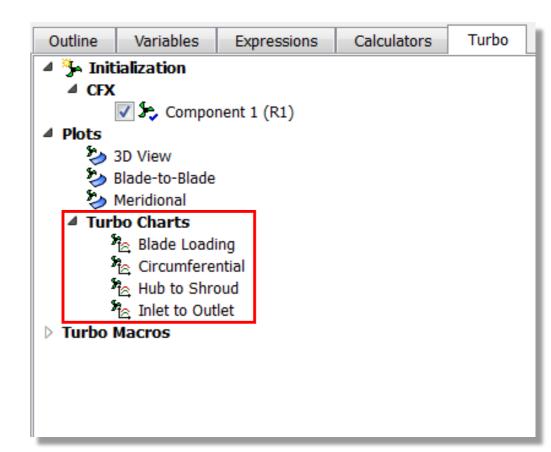
#### **Meridional Plot**

- Create a circumferentially averaged total pressure contour in the meridional\* view:
  - Click on *Meridional* in the *Tree*
  - Select Contour for Plot Type
  - Select *Total Pressure* for the *Variable*
  - Select Local for Range
  - Click Apply
  - \* A meridional view depicts the passage flow from inlet to outlet, hub to shroud using a circumferential average of the flow properties at a given position within the passage



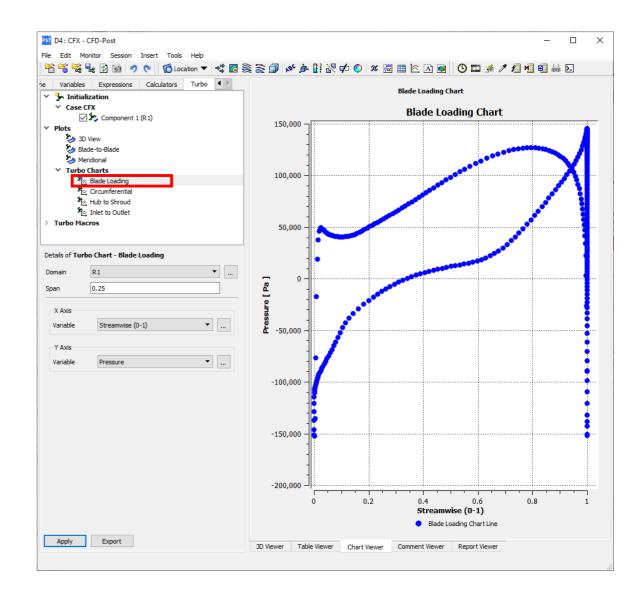
#### **Turbo Charts**

- There are four different types of 2D plots that can be created:
  - Blade Loading
  - Circumferential
  - Hub to Shroud
  - Inlet to Outlet



### **Blade Loading Chart**

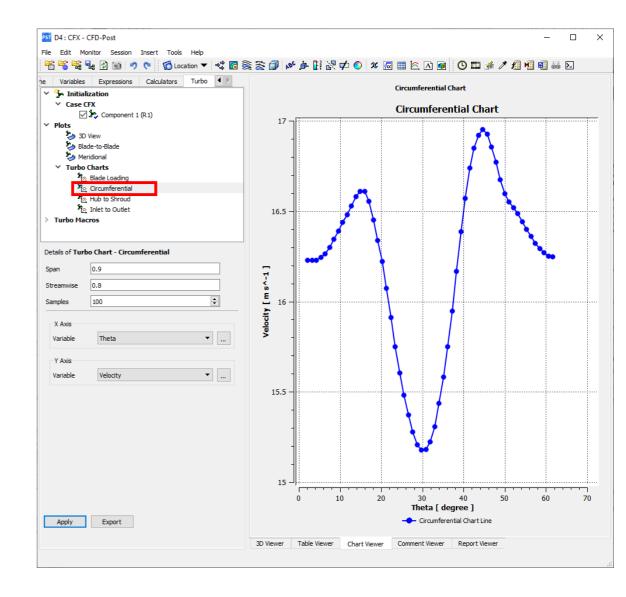
- Create a blade loading chart of pressure at 25% span
  - Click Blade Loading under Plots > Turbo Charts in the Tree
  - Set *Span* to 0.25
  - Set Y Axis Variable to Pressure
  - Click Apply





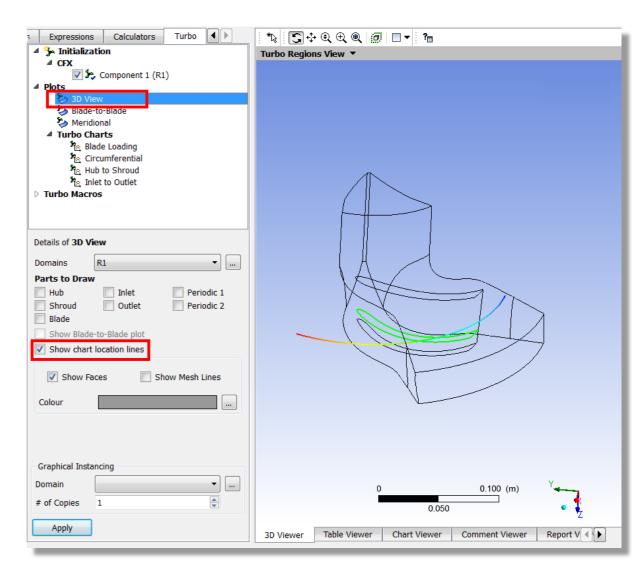
#### **Circumferential Chart**

- To examine a velocity profile in the wake of the blade, create a circumferential chart:
  - Select Circumferential under Turbo Charts
  - Set Span to 0.9
  - Set Streamwise to 0.8
  - Set Samples to 100
  - Set Y Axis Variable to Velocity
  - Click Apply



### Visualizing the location of the chart lines

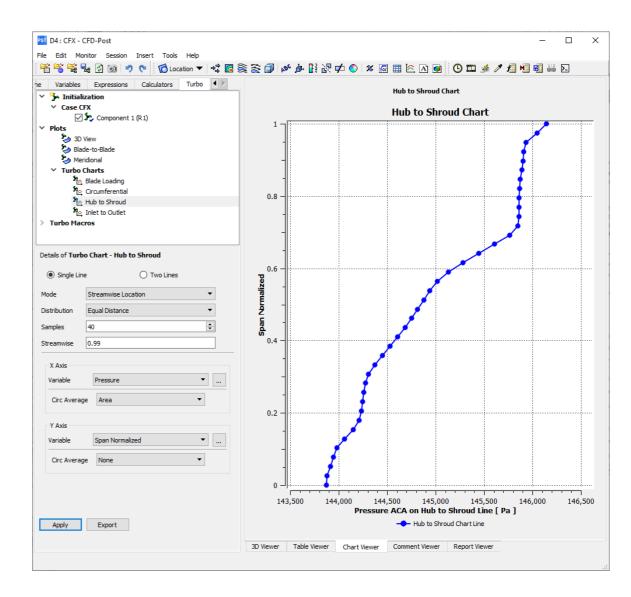
- To visualize where the previous data is being obtained:
  - Select 3D View under Plots in the Tree
  - Check the Show chart location lines checkbox
  - Click Apply
- The lines used to create the blade loading chart at 25% span and the circumferential chart at 90% span at a streamwise location of 0.8 are shown
- This enables you visualize and check where chart lines are created





#### **Hub to Shroud**

- Let's examine a circumferentially averaged pressure profile near the outlet spanwise
  - Select *Hub to Shroud* in the *Tree*
  - Set Distribution to Equal Area
  - Set Samples to 40
  - Set Streamwise to 0.99
  - Click Apply

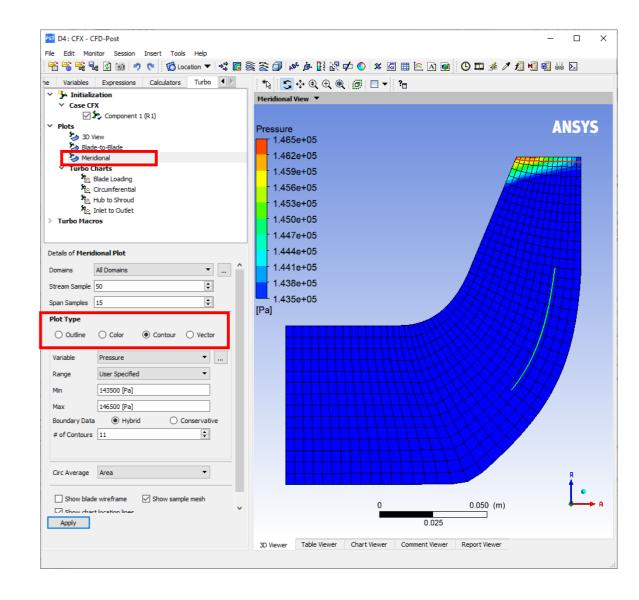




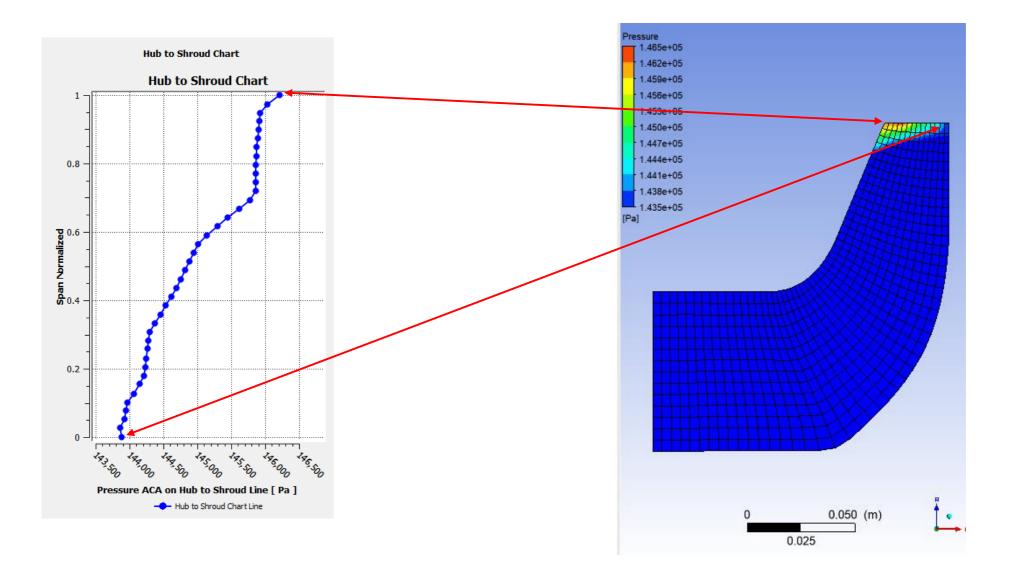
### **Compare to Meridional Plot**

- The profile just created corresponds to data shown in a meridional plot
  - Create a meridional contour as shown to the right
  - Set the Range to User Specified and set the Min and Max values to 143500 and 146500
     [Pa] respectively\*
  - Notice the pressure profile near the outlet

\*The *Min* and *Max* values of 143500 and 146500 are taken from the x-axis limits of the chart in the previous slide. You might have to give different values depending on own results



## **Compare to Meridional Plot**

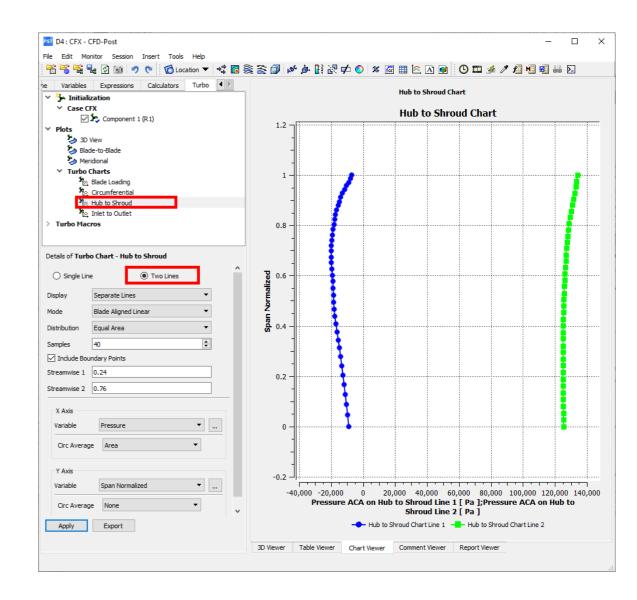


#### **Hub to Shroud**

- Create total pressure profiles at the leading and trailing edge in the same chart
  - Modify *Hub to Shroud Chart*
  - Select Two Lines
  - Set Mode to Blade Aligned Linear
  - Set Streamwise 1 to 0.24
  - Set Streamwise 2 to 0.76
  - Click Apply

The *Blade Aligned Streamwise* Variable is defined such that it equals the following values at specific locations:

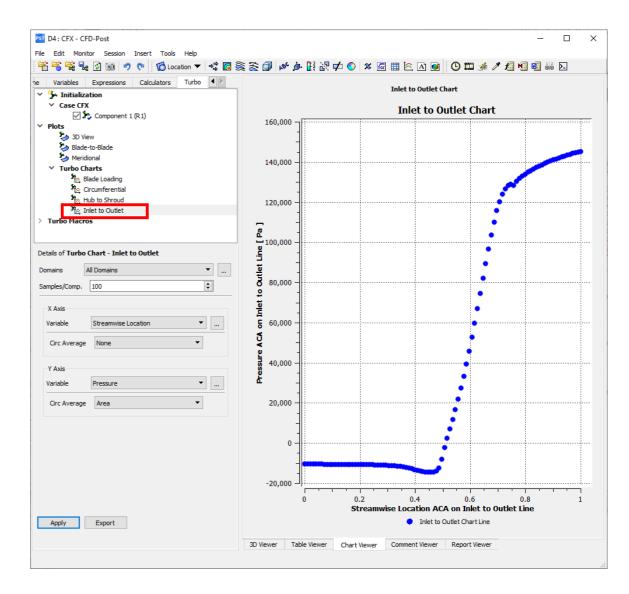
- Inlet to component = 0
- Blade leading edge = 0.25
- Blade trailing edge = 0.75
- Outlet to component = 1





#### **Inlet to Outlet**

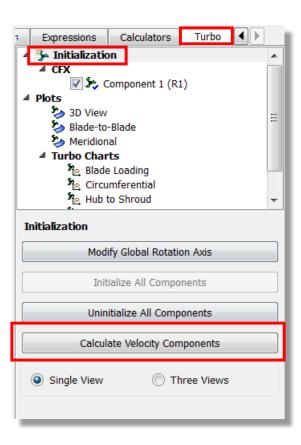
- To visualize how the pressure increases from inlet to outlet:
  - Select Inlet to Outlet under Turbo Chart
  - Set Samples/Comp. to 100
  - Set X Axis Variable to Streamwise Location
  - Set Y Axis Variable to Pressure
  - Click Apply





### **Calculate Velocity Components**

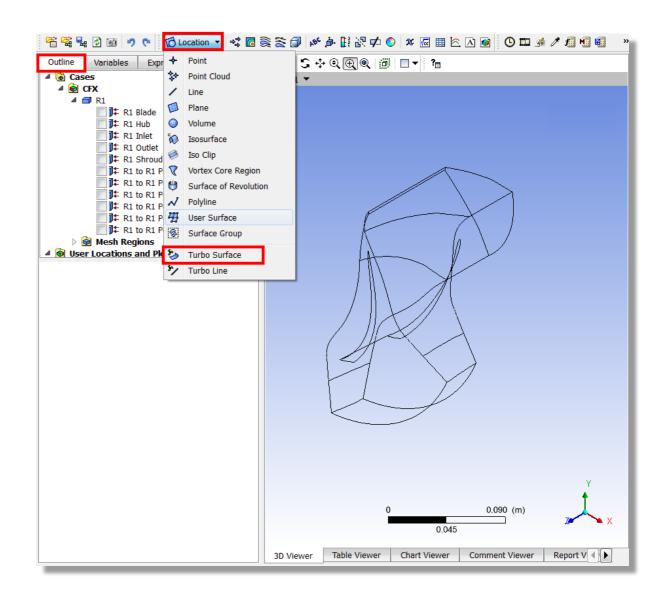
- Some additional variables useful for turbo post processing can be created
  - Still in the *Turbo* Tab
  - Double click Initialization
  - Click Calculate Velocity Components
- This creates a number of new velocity variables, such as:
  - Velocity Axial
  - Velocity Radial
  - Velocity Circumferential
  - Velocity Flow Angle
  - Velocity Spanwise
  - Velocity Streamwise
  - ...





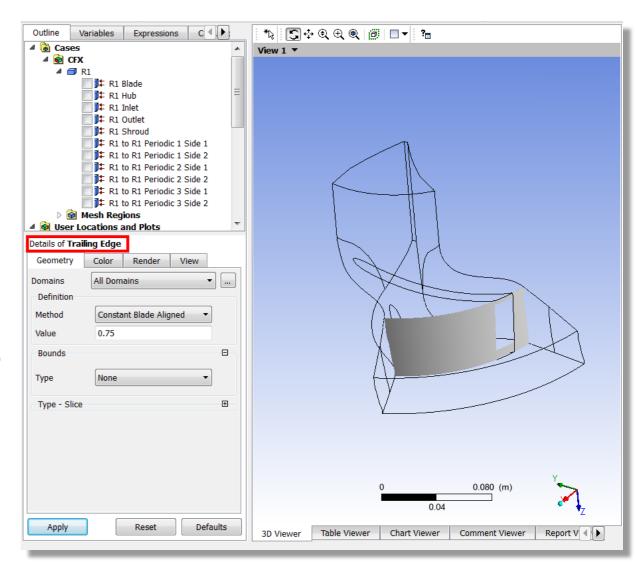
#### **Turbo Surface**

- You can create Turbo Surfaces in the standard 3D Viewer
  - Switch to the *Outline* Tab at the top left
  - Location > Turbo Surface
  - Rename the Turbo Surface to Trailing Edge



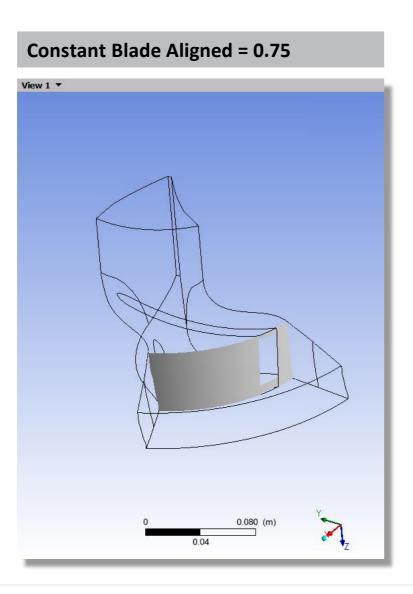
#### **Turbo Surface**

- Details of *Trailing Edge Turbo Surface* 
  - Set Method to Constant Blade Aligned
  - Set Value to 0.75
- We can see this surface intersects the trailing edge of the blade
- Move the plane slightly downstream so that it is entirely behind the trailing edge
  - Set *Value* to 0.755

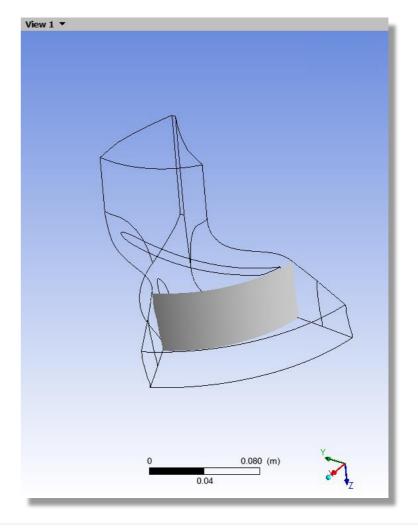




# **Constant Blade Aligned**



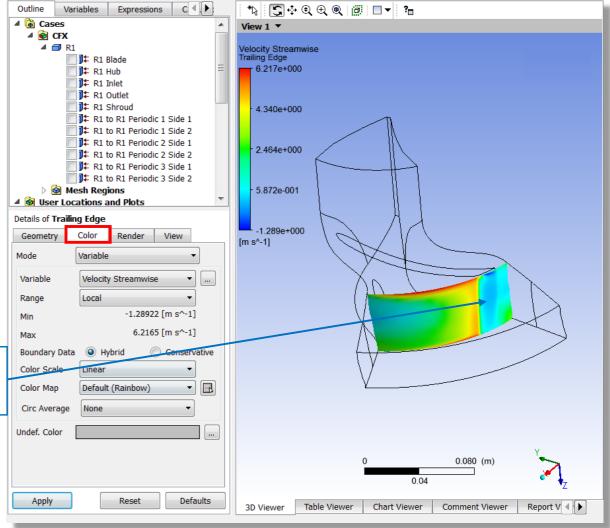
#### **Constant Blade Aligned = 0.755**



### **Colour Blade Aligned Surface**

- Color the *Turbo Surface* by the Turbo Post variable *Velocity Streamwise*
  - Switch to the Color Tab under the Details of Trailing Edge
  - Set *Mode* to *Variable*
  - Set Variable to Velocity Streamwise
  - Set Range to Local
  - Click Apply

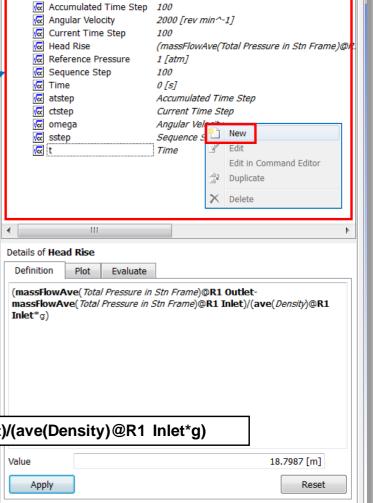
Note the low velocities in the wake of the blade



### **Quantitative Calculations**

- Create an expression to define the head generated from inlet to outlet
  - Click on the Expressions Tab
  - Right click inside the *Expressions* window and select New

Details of Head Rise Name the Expression Head Rise Evaluate (massFlowAve(Total Pressure in Stn Frame)@R1 Outlet- Type in the expression given in the box below massFlowAve(Total Pressure in Stn Frame)@R1 Inlet)/(ave(Density)@R1 Inlet\*q) Click Apply The Head is 18.8 [m] for this pump (massFlowAve(Total Pressure in Stn Frame)@R1 Outlet-massFlowAve(Total Pressure in Stn Frame)@R1 Inlet)/(ave(Density)@R1 Inlet\*g) Value 18.7987 [m] Apply



Variables

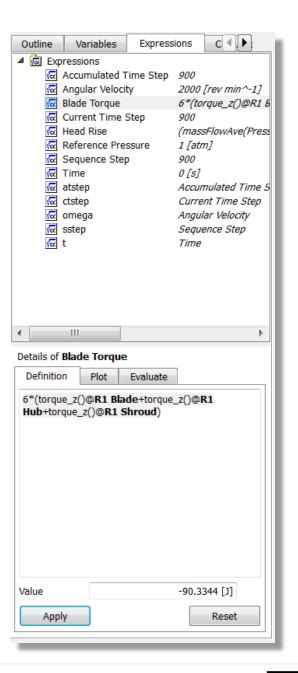
Expressions

Calculators

### **Calculate Blade Torque**

- To calculate blade torque
  - Create a new expression
  - Name Blade Torque
  - Enter expression given in the box below
  - Click Apply
- Torque is 90.33 [J] [i.e. N\*m])
- Note that we multiplied the expression by 6 which is the number of blades
- We are also accounting for torque due to shear on hub and shroud

6\*(torque\_z()@R1 Blade+torque\_z()@R1 Hub+torque\_z()@R1 Shroud)



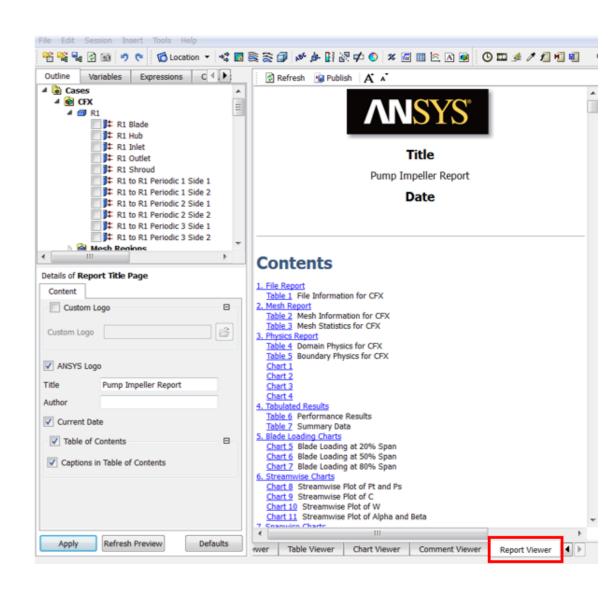
### **Quantitative Calculations - Expressions**

- You could create many expressions to evaluate quantities of interest
  - Head
  - Input Power
  - Mass or Volume Flow rate
  - Efficiency
  - Torque
  - etc.
- However, there are built in reports which can do this for you!



### **Load Pump Report**

- From the Main Menu, File > Report > Load 'Pump Impeller Report' Template
  - It will take a few minutes to generate the plots and quantitative data for the report
- Once complete, click on the Report Viewer tab
  - It will take a minute to generate the report
- Once complete, scroll through and examine the report





### **Examine Pump Report**

- Go to section *4. Tabulated Results* of the report
  - Table 6. Performance Results
    - ➤ Total Efficiency of 80.2%
  - Table 7. Summary Data
    - ➤ Flow Angle: Alpha represents the Absolute flow angle (i.e. to compare to design velocity triangles)
    - ➤ Values are calculated at inlet, outlet, leading and trailing edge
    - ➤ Ratios and differences from TE-LE are calculated

#### 4. Tabulated Results

The first table below gives a summary of the performance results for the pump impeller. The second table lists the mass or area averaged solution variables and derived quantities computed at the inlet, leading edge (LE Cut), trailing edge (TE Cut) and outlet locations. The flow angles Alpha and Beta are relative to the meridional plane; a positive angle implies that the tangential velocity is the same direction as the machine rotation.

Ta	ble	6.	Performance	Results

Rotation Speed	209.4400	[radian s^-1]
Reference Diameter	0.2081	[m]
Volume Flow Rate	0.0840	[m^3 s^-1]
Head (LE-TE)	18.7137	[m]
Head (IN-OUT)	18.7987	[m]
Flow Coefficient	0.0445	
Head Coefficient (IN-OUT)	0.0971	
Shaft Power	19252.1000	[W]
Power Coefficient	0.0054	
Total Efficiency (IN-OUT) %	80.1957	
Static Efficiency (IN-OUT) %	63.4938	

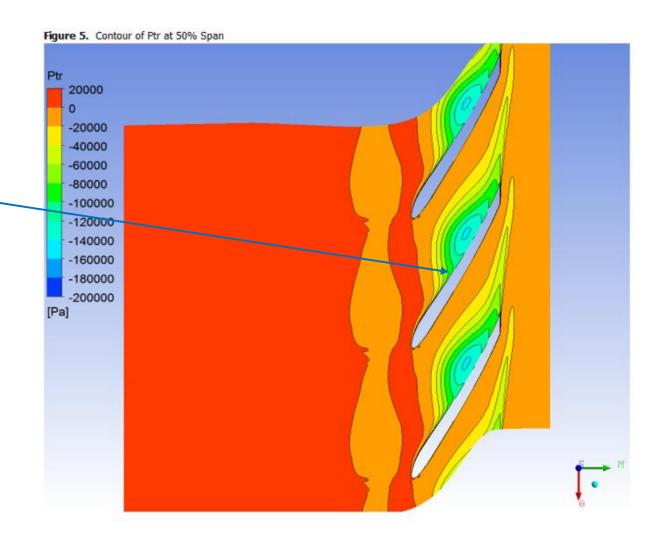
Table 7. Summary Data

Quantity	Inlet	LE Cut	TE Cut	Outlet	TE/LE	TE-LE	Units
Density	997.0000	997.0000	997.0000	997.0000	1.0000	0.0000	[kg m^-3]
Pstatic	90822.3000	83304.8000	229592.0000	246604.0000	2.7561	146287.0000	[Pa]
Ptotal	101225.0000	101174.0000	284143.0000	285024.0000	2.8085	182968.0000	[Pa]
Ptotal (rot)	100514.0000	96139.4000	76207.5000	55710.0000	0.7927	-19931.9000	[Pa]
U	10.7251	12.4750	21.7896	32.4612	1.7467	9.3146	[m s^-1]
Cm	4.5618	5.4465	3.3458	3.0825	0.6143	-2.1007	[m s^-1]
Cu	0.0362	1.8640	11.0474	6.2849	5.9268	9.1834	[m s^-1]
С	4.5678	6.7835	11.7722	7.0842	1.7354	4.9887	[m s^-1]
Distortion Parameter	1.0007	1.1947	1.3000	1.3910	1.0881	0.1053	
Flow Angle: Alpha	0.4801	25.3026	72.7069	66.3892	2.8735	47.4043	[degree]
Wu	-10.6889	-10.6113	-10.7423	-26.1764	1.0123	-0.1309	[m s^-1]
W	11.7669	12.0128	11.2978	26.4387	0.9405	-0.7150	[m s^-1]
Flow Angle: Beta	-63.8893	-46.0883	-73.1813	-83.2342	1.5878	-27.0930	[degree]

### **Examine Pump Report**

#### • Examine:

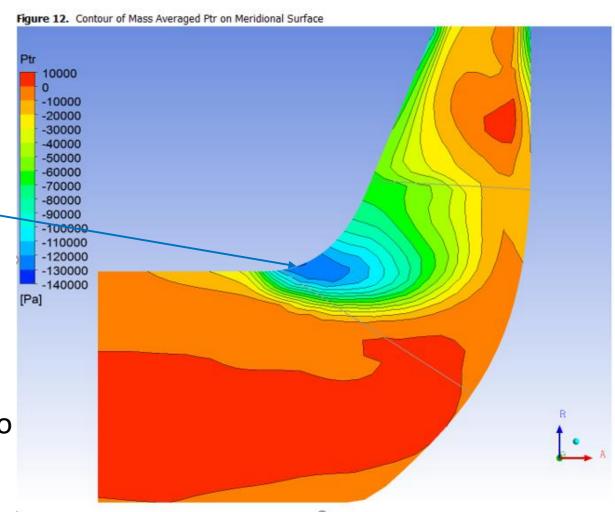
- Figure 5. Contour of Ptr at 50% span
  - There is significant flow separation from the suction side of the blade



### **Examine Pump Report**

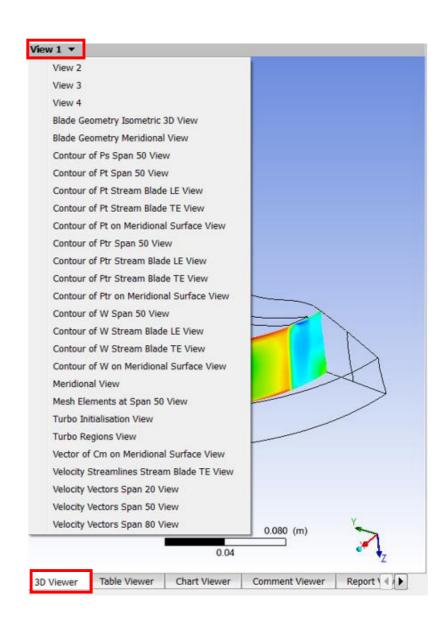
#### • Examine:

- Figure 12. Contour of Mass Averaged Ptr on Meridional Surface
  - ➤ There is significant flow separation from shroud
- This plot and the one on the previous page indicate that the shroud is not designed correctly, hence the low pump efficiency and poor convergence
- The report can provide insights on how to improve your design



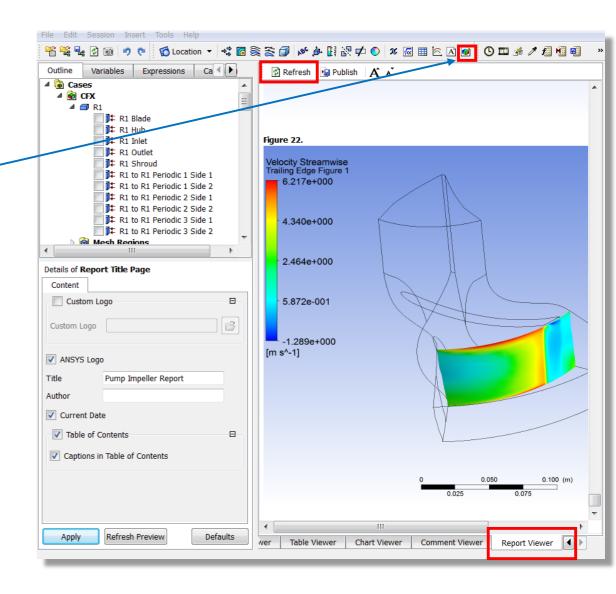
### Add a plot to the Pump Report

- Click on 3D Viewer Tab at the bottom
- Switch to View 1 at the top
- Add the colored turbo surface to the report (see next slide)



### Add a plot to the Pump Report

- Insert > Figure
  - Alternatively select Figure from the toolbar
- Name: Figure 1
  - Click OK
- Click on the Report Viewer Tab
- Click Refresh
- Scroll to bottom of the window once the report is regenerated
- The new figure has been added



### **Summary**

- This workshop demonstrated the following:
- Turbo Post Processing using the Turbo Tab
- Creating CEL expressions to evaluate global quantities for the pump
- Report Generation and customization