

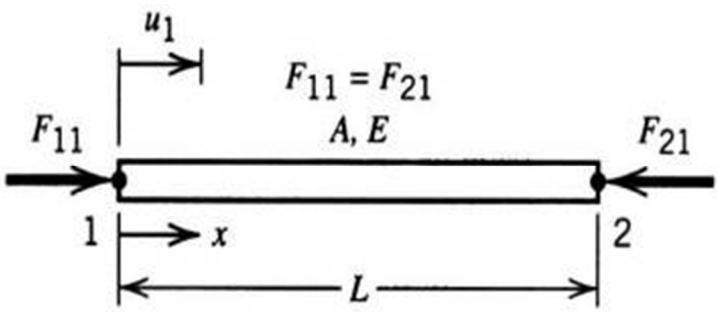
Elementos Finitos I

“Things should be described as simply as possible, but not simpler” – A. Einstein

Elementos Unidimensionales

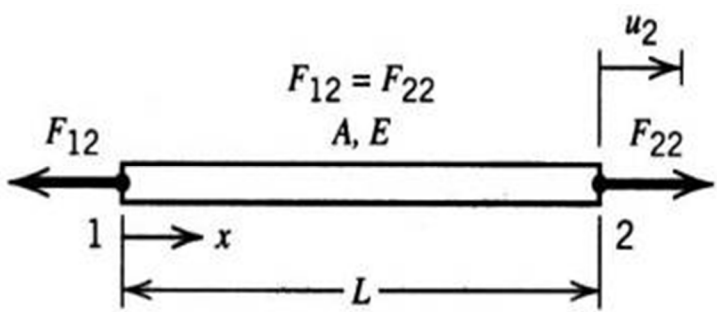
Elemento Barra

Método Directo



$$\begin{bmatrix} k^I_{11} & k^I_{12} \\ k^I_{21} & k^I_{22} \end{bmatrix} \begin{Bmatrix} u^I_1 \\ u^I_2 \end{Bmatrix} = \begin{Bmatrix} F^I_1 \\ F^I_2 \end{Bmatrix}$$

Desplazamientos Unitarios:



$$\begin{aligned} u^I_1 &= 1 ; u^I_2 = 0 \\ u^I_1 &= 0 ; u^I_2 = 1 \end{aligned}$$

$$[K_e] \{D_e\} = \{R_e\}$$

$$\begin{bmatrix} k^I_{11} & k^I_{12} \\ k^I_{21} & k^I_{22} \end{bmatrix} \begin{Bmatrix} 1 \\ 0 \end{Bmatrix} = \begin{Bmatrix} F^I_1 \\ F^I_2 \end{Bmatrix} \Rightarrow \begin{Bmatrix} k^I_{11} \\ k^I_{21} \end{Bmatrix} = \begin{Bmatrix} F^I_1 \\ F^I_2 \end{Bmatrix} = \frac{A_e E_e}{L_e} \begin{Bmatrix} 1 \\ -1 \end{Bmatrix}$$

$$\begin{bmatrix} k^I_{11} & k^I_{12} \\ k^I_{21} & k^I_{22} \end{bmatrix} \begin{Bmatrix} 0 \\ 1 \end{Bmatrix} = \begin{Bmatrix} F^I_1 \\ F^I_2 \end{Bmatrix} \Rightarrow \begin{Bmatrix} k^I_{12} \\ k^I_{22} \end{Bmatrix} = \begin{Bmatrix} F^I_1 \\ F^I_2 \end{Bmatrix} = \frac{A_e E_e}{L_e} \begin{Bmatrix} -1 \\ 1 \end{Bmatrix}$$

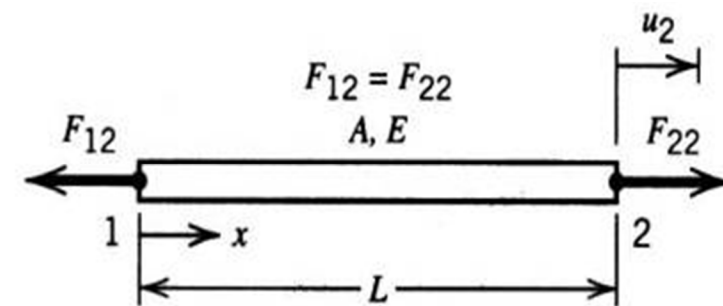
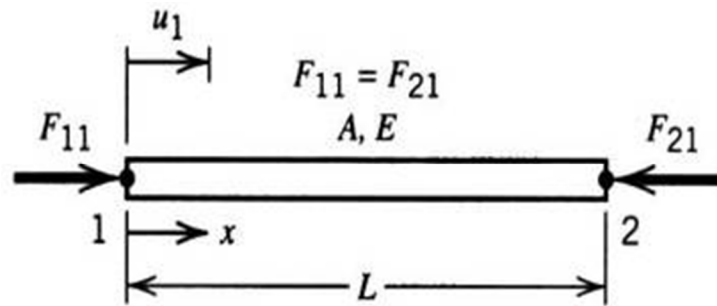
Matriz de rigidez:

$$[K_e] = \begin{bmatrix} X^I & -X^I \\ -X^I & X^I \end{bmatrix} ; X^I = \frac{A_e E_e}{L_e}$$

Elementos Unidimensionales

Elemento Barra

Método Directo

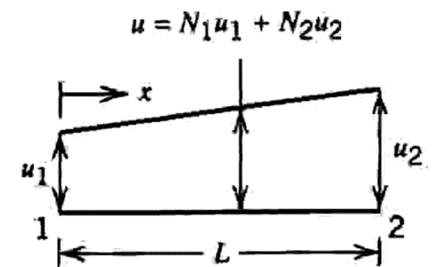
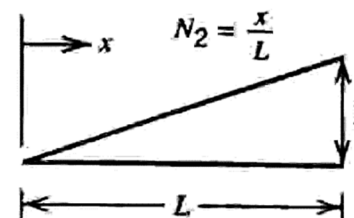
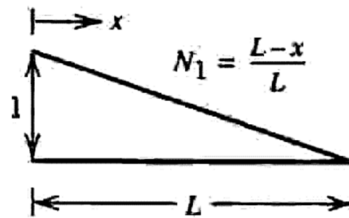


$$[K_e]\{D_e\} = \{R_e\}; \quad \begin{bmatrix} X^I & -X^I \\ -X^I & X^I \end{bmatrix} \begin{Bmatrix} u_1^I \\ u_2^I \end{Bmatrix} = \begin{Bmatrix} F_1^I \\ F_2^I \end{Bmatrix}$$

Desplazamiento

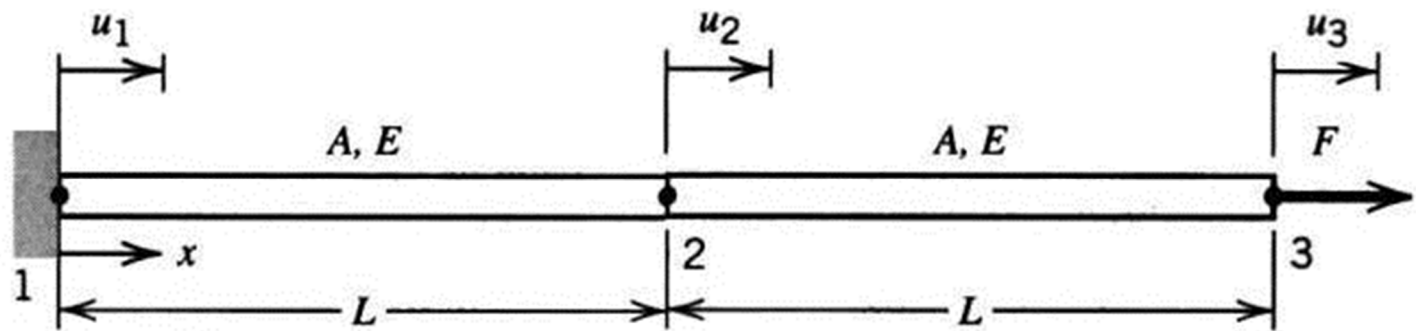
Funciones de Forma

$$u(x) = u_1 \left(\frac{L-x}{L} \right) + u_2 \left(\frac{x}{L} \right) = [N(x)]\{D_e\} = \begin{bmatrix} \frac{L-x}{L} & \frac{x}{L} \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix}$$



Estructura - Ensamblado

Dof Locales y Globales



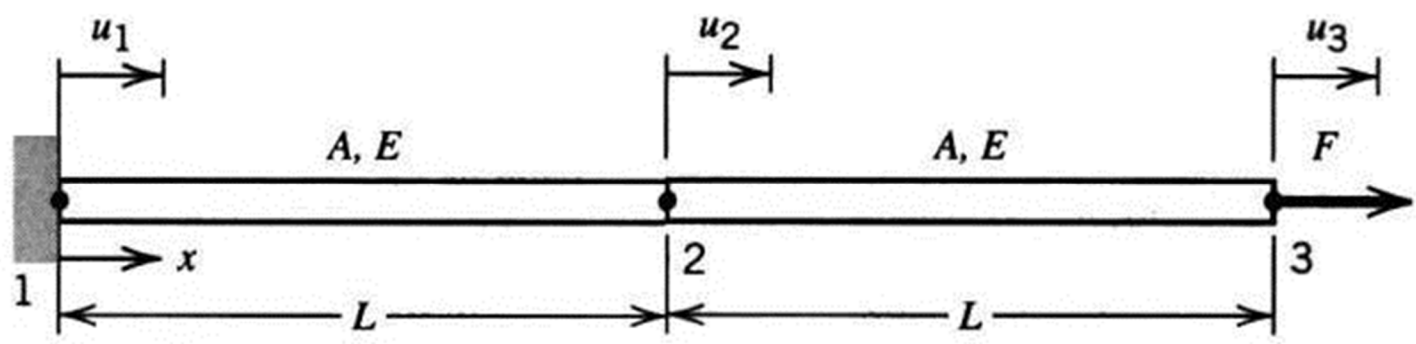
$$\begin{cases} u_1 = u^I_1 \\ u_2 = u^I_2 = u^{II}_1 \\ u_3 = u^{II}_2 \end{cases}$$

$$\begin{bmatrix} X^I & -X^I \\ -X^I & X^I \end{bmatrix} \begin{Bmatrix} u^I_1 \\ u^I_2 \end{Bmatrix} = \begin{Bmatrix} F^I_1 \\ F^I_2 \end{Bmatrix} \rightarrow \begin{bmatrix} X^I & -X^I & 0 \\ -X^I & X^I & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F^I_1 \\ F^I_2 \\ 0 \end{Bmatrix}$$

$$\underbrace{\begin{bmatrix} X^{II} & -X^{II} \\ -X^{II} & X^{II} \end{bmatrix}}_{[K_e]} \underbrace{\begin{Bmatrix} u^{II}_1 \\ u^{II}_2 \end{Bmatrix}}_{\{D_e\}} = \underbrace{\begin{Bmatrix} F^{II}_1 \\ F^{II}_2 \end{Bmatrix}}_{\{R_e\}} \rightarrow \begin{bmatrix} 0 & 0 & 0 \\ 0 & X^{II} & -X^{II} \\ 0 & -X^{II} & X^{II} \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ F^{II}_1 \\ F^{II}_2 \end{Bmatrix}$$

$$\left(\begin{bmatrix} X^I & -X^I & 0 \\ -X^I & X^I & 0 \\ 0 & 0 & 0 \end{bmatrix} + \begin{bmatrix} 0 & 0 & 0 \\ 0 & X^{II} & -X^{II} \\ 0 & -X^{II} & X^{II} \end{bmatrix} \right) \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F^I_1 \\ F^I_2 + F^{II}_1 \\ F^{II}_2 \end{Bmatrix} \rightarrow \begin{bmatrix} X^I & -X^I & 0 \\ -X^I & X^I + X^{II} & -X^{II} \\ 0 & -X^{II} & X^{II} \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix}$$

Estructura - Ensamblado



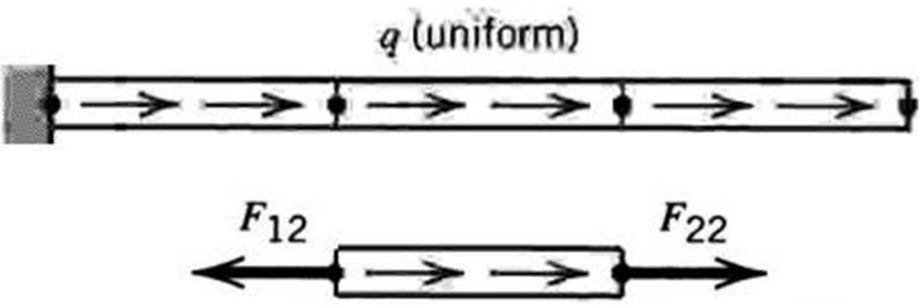
$$\underbrace{\begin{bmatrix} X^I & -X^I & 0 \\ -X^I & X^I + X^{II} & -X^{II} \\ 0 & -X^{II} & X^{II} \end{bmatrix}}_{[K]} \underbrace{\begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix}}_{\{D\}} = \underbrace{\begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix}}_{\{R\}} ; \text{ Condiciones de Borde}$$

$$\begin{cases} u_1 = 0 \\ F_2 = 0 \end{cases}$$

$$\underbrace{\begin{bmatrix} -X^I & 0 \\ X^I + X^{II} & -X^{II} \\ -X^{II} & X^{II} \end{bmatrix}}_{[K_r]} \underbrace{\begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix}}_{\{D_r\}} = \underbrace{\begin{Bmatrix} F_1 \\ 0 \\ F_3 \end{Bmatrix}}_{\{R_r\}} \therefore \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \frac{1}{X} \begin{bmatrix} 1 & 1 \\ 1 & 2 \end{bmatrix} \begin{Bmatrix} 0 \\ F_3 \end{Bmatrix} \Rightarrow \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} \frac{LF_3}{AE} \\ \frac{2LF_3}{AE} \end{Bmatrix}$$

Carga Distribuida

$$\frac{A_e E_e}{L_e} \begin{bmatrix} 1 & -1 & 0 & 0 \\ -1 & 2 & -1 & 0 \\ 0 & -1 & 2 & -1 \\ 0 & 0 & -1 & 1 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \\ F_4 \end{Bmatrix}$$



$$F^I_1 = F^I_2 = \frac{qL_e}{2} \quad \rightarrow \quad \frac{A_e E_e}{L_e} \begin{bmatrix} 1 & -1 & 0 & 0 \\ -1 & 2 & -1 & 0 \\ 0 & -1 & 2 & -1 \\ 0 & 0 & -1 & 1 \end{bmatrix} \begin{Bmatrix} 0 \\ u_2 \\ u_3 \\ u_4 \end{Bmatrix} = \begin{Bmatrix} R \\ qL \\ qL \\ 0.5qL \end{Bmatrix}$$

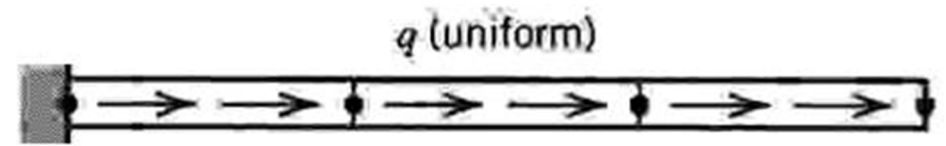
$$\begin{Bmatrix} u_2 \\ u_3 \\ u_4 \end{Bmatrix} = \frac{L_e}{A_e E_e} \begin{bmatrix} 1 & 1 & 1 \\ 1 & 2 & 2 \\ 1 & 2 & 3 \end{bmatrix} \begin{Bmatrix} qL_e \\ qL_e \\ 0.5qL_e \end{Bmatrix} = \frac{qL_e^2}{A_e E_e} \begin{Bmatrix} 2.5 \\ 4 \\ 4.5 \end{Bmatrix}$$

$$\{R\} = \frac{A_e E_e}{L_e} [-1 \quad 0 \quad 0] \frac{qL_e^2}{A_e E_e} \begin{Bmatrix} 2.5 \\ 4 \\ 4.5 \end{Bmatrix} = -2.5qL_e$$

+ 0,5 qL

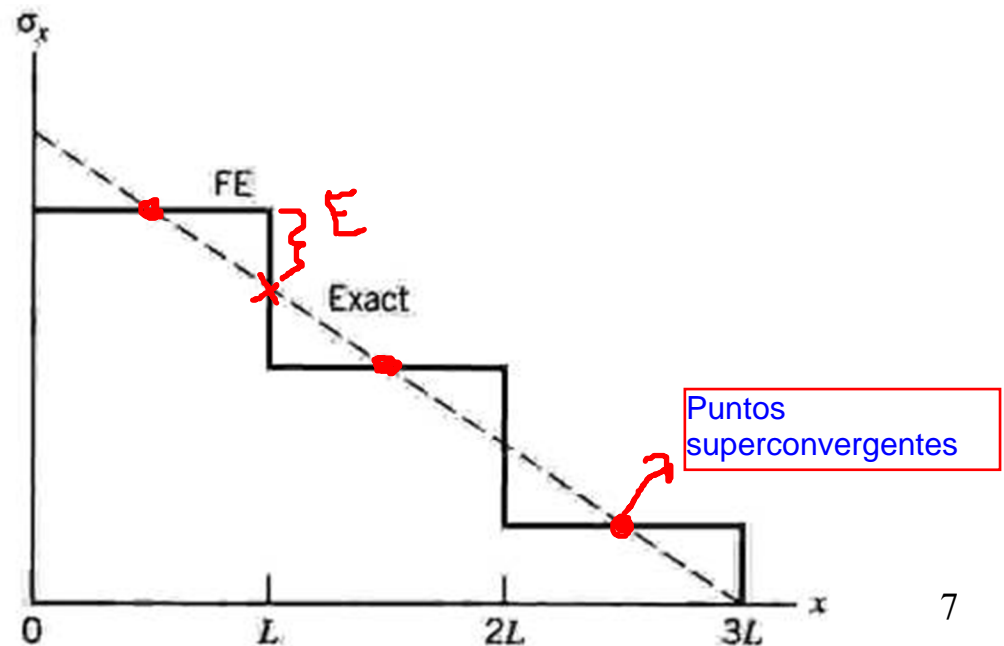
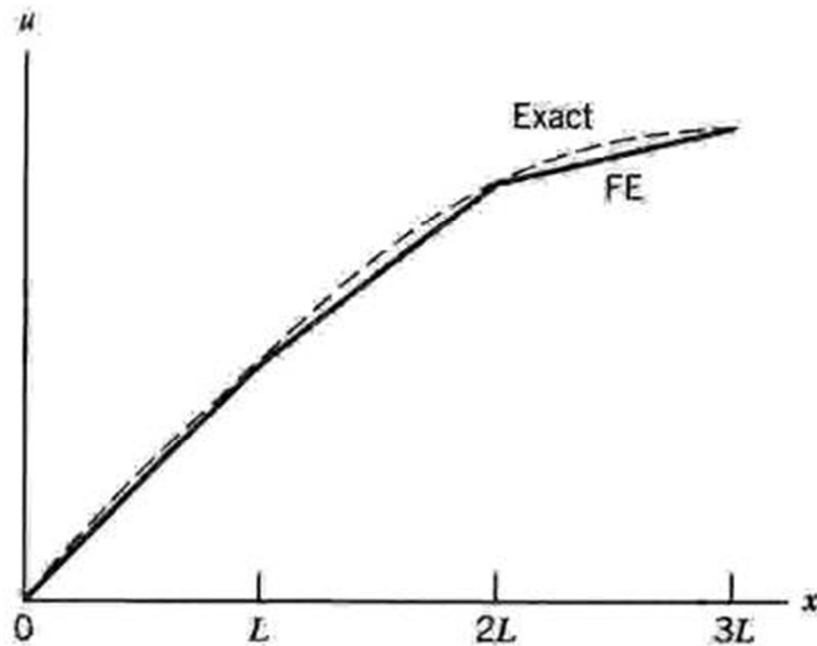
Carga Distribuida

Tensiones



$$u(x) = [N(x)]\{D_e\} = \begin{bmatrix} \frac{L_e - x}{L_e} & \frac{x}{L_e} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} ; \quad \varepsilon_x = \frac{du(x)}{dx} = \frac{\left[\frac{\partial}{\partial} \right]}{dx} [N(x)]\{d_e\} = \underbrace{\left[\frac{\partial}{\partial} [N(x)] \right]}_{[B(x)]} \{d_e\} = [B(x)]\{D_e\}$$

$$[B(x)] = \begin{bmatrix} -\frac{1}{L_e} & \frac{1}{L_e} \end{bmatrix} ; \quad \sigma_x = E_e \varepsilon_x = E_e \begin{bmatrix} -\frac{1}{L_e} & \frac{1}{L_e} \end{bmatrix} \begin{Bmatrix} 4 \\ 4.5 \end{Bmatrix} = \frac{qL_e}{2A_e}$$



Transformaciones – DOF no alineados - Barras



Matriz de transformación T

Nota: $\frac{V_{12}}{|V_{12}|} = \{l_1, m_1\}$; $\underbrace{\begin{bmatrix} u'_1 \\ v'_1 \end{bmatrix}}_{\{d'\}} = \underbrace{\begin{bmatrix} l_1 & m_1 \\ -m_1 & l_1 \end{bmatrix}}_{[T]} \underbrace{\begin{bmatrix} u_1 \\ v_1 \end{bmatrix}}_{\{d\}}$; $[T][T]^T = \begin{bmatrix} l_1 & m_1 \\ -m_1 & l_1 \end{bmatrix} \begin{bmatrix} l_1 & -m_1 \\ m_1 & l_1 \end{bmatrix}$; $[T]^T = [T]^{-1}$

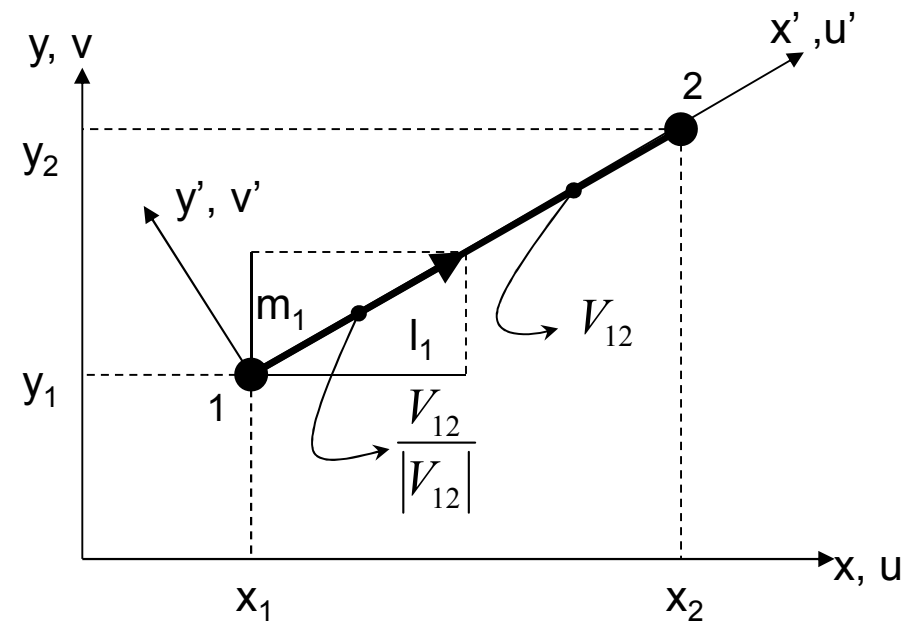
$$\{d'\} = [T]\{d\} \rightarrow \underbrace{\begin{bmatrix} u'_1 \\ u'_2 \\ v'_1 \\ v'_2 \end{bmatrix}}_{\{d'\}} = \underbrace{\begin{bmatrix} l_1 & m_1 & 0 & 0 \\ 0 & 0 & l_1 & m_1 \end{bmatrix}}_{[T]} \underbrace{\begin{bmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \end{bmatrix}}_{\{d\}}$$

$$[k']\{d'\} = \{r'\} \rightarrow [k'] [T]\{d\} = \{r'\}$$

$$[T]^T [k'] [T]\{d\} = [T]^T \{r'\}$$

$$\{r'\} = [T]\{r\} \quad ; \quad [T]^T \{r'\} = [T]^T [T]\{r\}$$

$$[T]^T [k'] [T]\{d\} = \{r\} \rightarrow [k] = [T]^T [k'] [T]$$

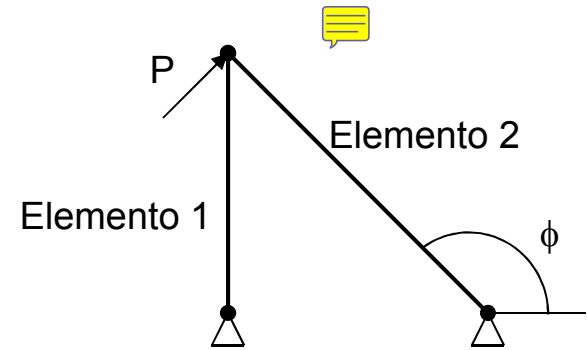
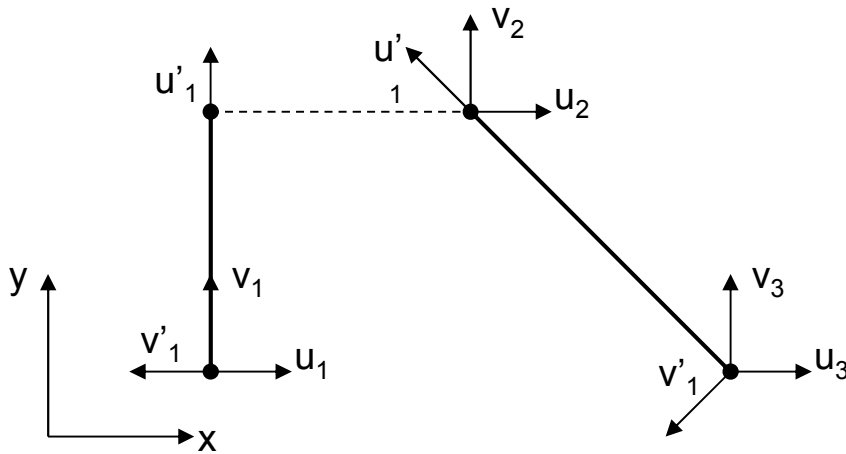


Transformaciones – DOF no alineados- Barras

Matrices de rigidez elementales

$$[K'_1] = \begin{bmatrix} k_1 & -k_1 \\ -k_1 & k_1 \end{bmatrix} \quad \begin{Bmatrix} u'_1 \\ u'_2 \end{Bmatrix} \quad k_1 = \frac{A_1 E_1}{L_1}$$

$$[K'_2] = \begin{bmatrix} k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \quad \begin{Bmatrix} u'_1 \\ u'_2 \end{Bmatrix} \quad k_2 = \frac{A_2 E_2}{L_2}$$



Matrices transformadas

$$[K_1] = [T_1]^T [K'_1] [T_1] \quad [K_1] = \begin{bmatrix} 0 & 0 \\ 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} k_1 & -k_1 \\ -k_1 & k_1 \end{bmatrix} \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

$$[K_1] = \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & k_1 & 0 & -k_1 \\ 0 & 0 & 0 & 0 \\ 0 & -k_1 & 0 & k_1 \end{bmatrix} \quad \begin{Bmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \end{Bmatrix}$$

Con $\beta=90^\circ$

$$[K_2] = [T_2]^T [K'_2] [T_2] \quad [K_2] = \begin{bmatrix} c\phi & 0 \\ s\phi & 0 \\ 0 & c\phi \\ 0 & s\phi \end{bmatrix} \begin{bmatrix} k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \begin{bmatrix} c\phi & s\phi & 0 & 0 \\ 0 & 0 & c\phi & s\phi \end{bmatrix}$$

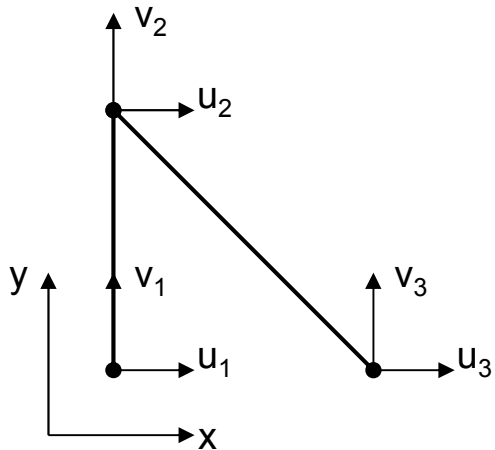
$$[K_2] = \begin{bmatrix} k_2 c^2 \phi & k_2 \cdot c\phi \cdot s\phi & -k_2 c^2 \phi & -k_2 \cdot c\phi \cdot s\phi \\ k_2 \cdot c\phi \cdot s\phi & k_2 s^2 \phi & -k_2 \cdot c\phi \cdot s\phi & -k_2 s^2 \phi \\ -k_2 c^2 \phi & -k_2 \cdot c\phi \cdot s\phi & k_2 c^2 \phi & k_2 \cdot c\phi \cdot s\phi \\ -k_2 \cdot c\phi \cdot s\phi & -k_2 s^2 \phi & k_2 \cdot c\phi \cdot s\phi & k_2 s^2 \phi \end{bmatrix} \quad \begin{Bmatrix} u_2 \\ v_2 \\ u_3 \\ v_3 \end{Bmatrix}$$

Con $\beta=\phi$

Transformaciones – DOF no alineados- Barras

Matriz global

$$[K] = \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & k_1 & 0 & -k_1 & 0 & 0 \\ 0 & 0 & k_2 c^2 \phi & k_2 \cdot c \phi \cdot s \phi & -k_2 c^2 \phi & -k_2 \cdot c \phi \cdot s \phi \\ 0 & -k_1 & k_2 \cdot c \phi \cdot s \phi & k_1 + k_2 s^2 \phi & -k_2 \cdot c \phi \cdot s \phi & -k_2 s^2 \phi \\ 0 & 0 & -k_2 c^2 \phi & -k_2 \cdot c \phi \cdot s \phi & k_2 c^2 \phi & k_2 \cdot c \phi \cdot s \phi \\ 0 & 0 & -k_2 \cdot c \phi \cdot s \phi & -k_2 s^2 \phi & k_2 \cdot c \phi \cdot s \phi & k_2 s^2 \phi \end{bmatrix} \begin{bmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{bmatrix}$$



Carga externas

$$\{R\} = \begin{Bmatrix} R_{x1} \\ R_{y1} \\ P_x \\ P_y \\ R_{x3} \\ R_{y3} \end{Bmatrix}$$

Matriz reducida

$$[K] = \begin{bmatrix} k_2 c^2 \phi & k_2 \cdot c \phi \cdot s \phi \\ k_2 \cdot c \phi \cdot s \phi & k_1 + k_2 s^2 \phi \end{bmatrix} \begin{bmatrix} u_2 \\ v_2 \end{bmatrix}$$

Desplazamientos ($\phi=45^\circ$; $A_1= A_2= 1$; $E_1= E_2= 1$; $L_1=1$; $L_2=2^{0.5}$)

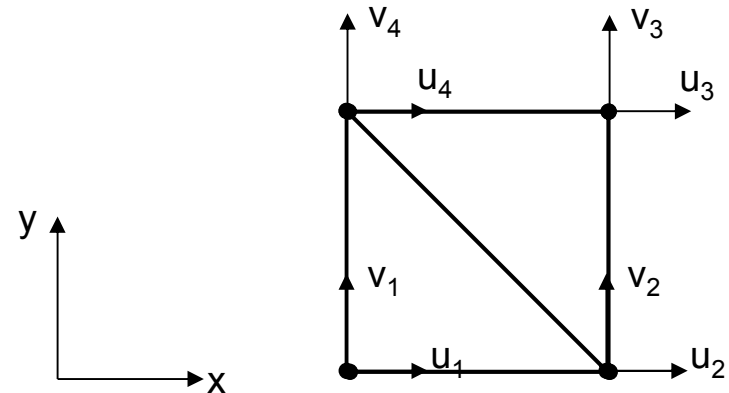
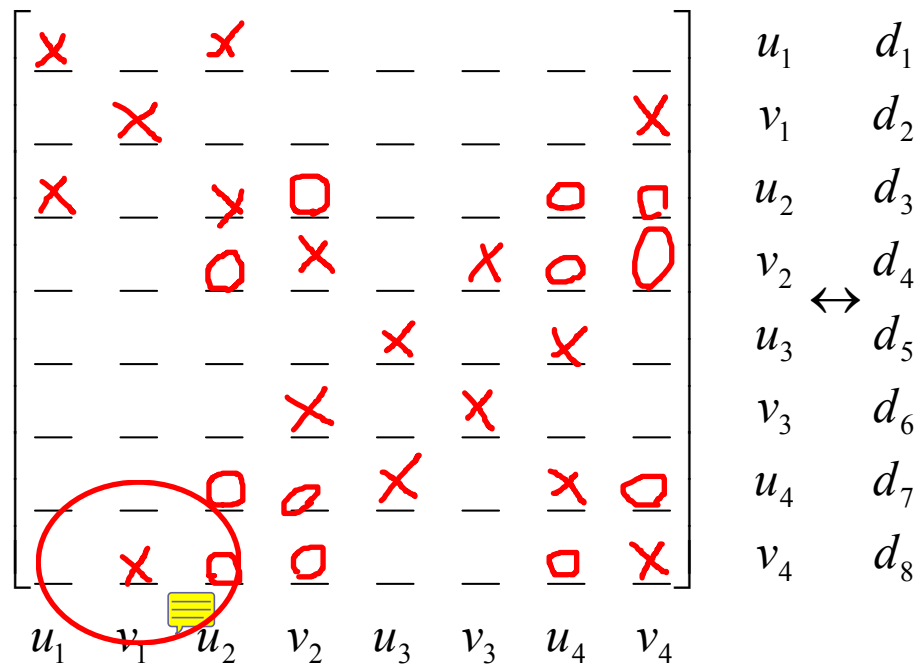


$$\{U\} = [K]^{-1}\{R\} = \begin{bmatrix} \frac{\sqrt{2}}{4} & -\frac{\sqrt{2}}{4} \\ -\frac{\sqrt{2}}{4} & 1+\frac{\sqrt{2}}{4} \end{bmatrix}^{-1} \begin{Bmatrix} P \frac{\sqrt{2}}{2} \\ P \frac{\sqrt{2}}{2} \end{Bmatrix} = P \begin{Bmatrix} 2+\sqrt{2} \\ \sqrt{2} \end{Bmatrix}$$

Reacciones Externas

$$\{R\} = [K]\{U\} = \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 \\ & 1 & 0 & -1 & 0 & 0 \\ & & \sqrt{2}/4 & -\sqrt{2}/4 & -\sqrt{2}/4 & \sqrt{2}/4 \\ & & & 1+\sqrt{2}/4 & \sqrt{2}/4 & -\sqrt{2}/4 \\ & & & & \sqrt{2}/4 & -\sqrt{2}/4 \\ & & & & & \sqrt{2}/4 \end{bmatrix} P \begin{bmatrix} 0 \\ 0 \\ 2+\sqrt{2} \\ \sqrt{2} \\ 0 \\ 0 \end{bmatrix} = P \begin{bmatrix} 0 \\ -\sqrt{2} \\ \sqrt{2}/2 \\ \sqrt{2}/2 \\ -\sqrt{2}/2 \\ \sqrt{2}/2 \end{bmatrix}$$

Ensamblado Matriz de Rigidez



Numeración Local vs Numeración Global

Práctica Preliminar

$$h = 4m ; \quad A_i = 100mm^2 \quad ; \quad A_s = 25mm^2$$

Solución Exacta:

Solución Aproximada según área:

- Máxima
- Mínima
- Media

