

Interpolación:

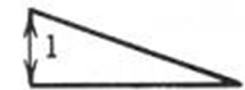
$$\phi(x) \cong \tilde{\phi}(x) = \sum_{i=0}^n a_i x^i; \quad \tilde{\phi}(x) = \underbrace{\begin{Bmatrix} 1 & x & \dots & x^n \end{Bmatrix}}_{\{X(x)\}} \begin{Bmatrix} a_0 \\ a_1 \\ \vdots \\ a_n \end{Bmatrix}; \quad \underbrace{\begin{Bmatrix} \phi_0 \\ \phi_1 \\ \vdots \\ \phi_n \end{Bmatrix}}_{\{\phi_e\}} = \underbrace{\begin{bmatrix} 1 & x_0 & x_0^2 & x_0^n \\ 1 & x_1 & x_1^2 & x_1^n \\ \vdots & \vdots & \vdots & \vdots \\ 1 & x_n & x_n^2 & x_n^n \end{bmatrix}}_{[A]} \begin{Bmatrix} a_0 \\ a_1 \\ \vdots \\ a_n \end{Bmatrix}; \quad \{a\} = [A]^{-1} \{\phi_e\}$$

$$\tilde{\phi}(x) = \underbrace{\{X(x)\}\{a\}}_{[N]} = \underbrace{\{X\}[A]^{-1}}_{[N]} \{\phi_e\} = [N(x)]\{\phi_e\}$$

Funciones de forma

Condiciones

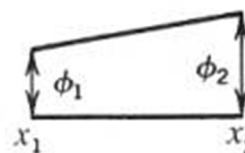
$$\begin{cases} [N_i(x_i)] = 1 \wedge [N_i(x_j)] = 0 \\ i \neq j \\ \sum_{i=0}^n N_i = 1 \end{cases}$$



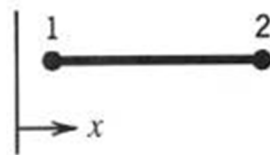
$$N_1 = \frac{x_2 - x}{x_2 - x_1}$$



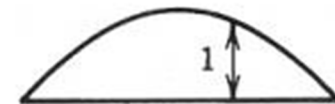
$$N_2 = \frac{x - x_1}{x_2 - x_1}$$



$$\phi = [N] \begin{Bmatrix} \phi_1 \\ \phi_2 \end{Bmatrix}$$



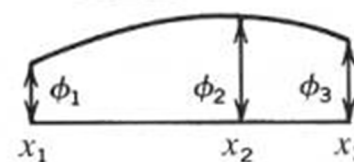
$$N_1 = \frac{(x_2 - x)(x_3 - x)}{(x_2 - x_1)(x_3 - x_1)}$$



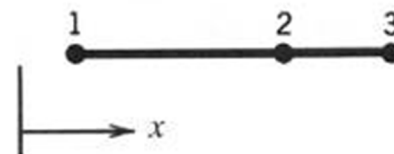
$$N_2 = \frac{(x_1 - x)(x_3 - x)}{(x_1 - x_2)(x_3 - x_2)}$$



$$N_3 = \frac{(x_1 - x)(x_2 - x)}{(x_1 - x_3)(x_2 - x_3)}$$



$$\phi = [N] \begin{Bmatrix} \phi_1 \\ \phi_2 \\ \phi_3 \end{Bmatrix}$$



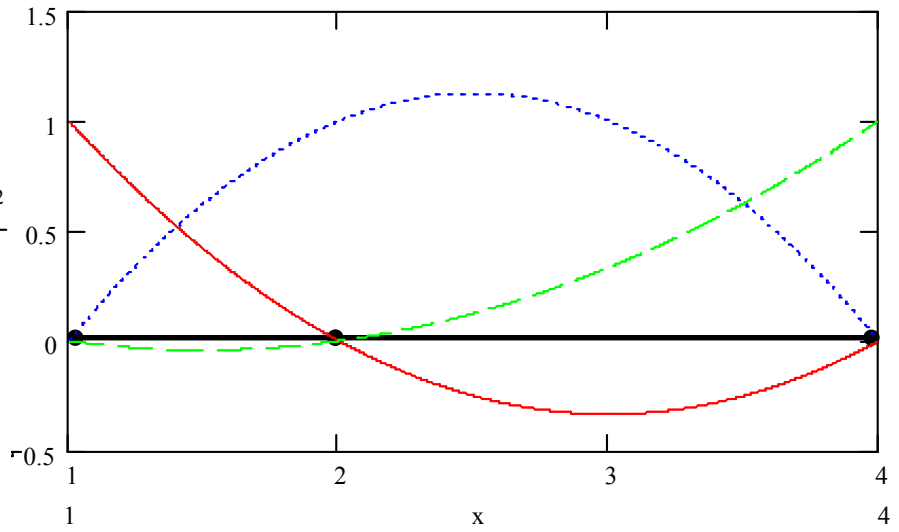
Elemento Barra 3 nodos:

$$\tilde{\phi}(x) = \begin{Bmatrix} 1 & x & x^2 \end{Bmatrix} \begin{Bmatrix} a_0 \\ a_1 \\ a_2 \end{Bmatrix}; \begin{Bmatrix} \phi_0 \\ \phi_1 \\ \phi_2 \end{Bmatrix} = \begin{bmatrix} 1 & 1 & 1 \\ 1 & 2 & 4 \\ 1 & 4 & 16 \end{bmatrix} \begin{Bmatrix} a_0 \\ a_1 \\ a_2 \end{Bmatrix};$$

$\frac{16-12 \cdot x+2 \cdot x^2}{6}$
—

$\frac{-12+15 \cdot x-3 \cdot x^2}{6}$
- - -

$\frac{2-3 \cdot x+x^2}{6}$
- - -



$$\tilde{\phi}(x) = \begin{Bmatrix} 1 \\ x \\ x^2 \end{Bmatrix}^T \frac{1}{6} \begin{bmatrix} 16 & -12 & 2 \\ -12 & 2 & -3 \\ 2 & -3 & 1 \end{bmatrix} \{\phi_e\} = \frac{1}{6} \overbrace{\begin{Bmatrix} 16-12x+2x^2 \\ -12+15x-3x^2 \\ 2-3x+x^2 \end{Bmatrix}}^{[N]^T} \{\phi_e\}$$

Matriz de Rigidez:

$$\varepsilon_x = \underbrace{\{\partial\}}_{[B(x)]} \underbrace{[N(x)]}_{[N]} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \left[\frac{1}{6} \begin{Bmatrix} -12+4x \\ 15-6x \\ -3+2x \end{Bmatrix} \right]^T \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix};$$

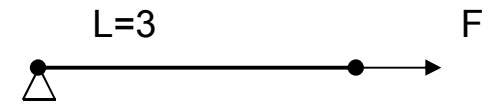
$$[K] = \int_{L_e} [B]^T E [B] A dL = \frac{AE}{36} \int_{L_e} \begin{Bmatrix} -12+4x \\ 15-6x \\ -3+2x \end{Bmatrix} \begin{Bmatrix} -12+4x \\ 15-6x \\ -3+2x \end{Bmatrix}^T dL$$

$$[K] = \frac{AE}{36} \int_{L_e} \begin{bmatrix} (-12+4x)^2 & (-12+4x)(15-6x) & (-12+4x)(-3+2x) \\ (-12+4x)(15-6x) & (15-6x)^2 & (15-6x)(-3+2x) \\ (-12+4x)(-3+2x) & (-3+2x)(15-6x) & (-3+2x)^2 \end{bmatrix} dL = \frac{AE}{36} \begin{bmatrix} 48 & -54 & 6 \\ -54 & 81 & -27 \\ 6 & -27 & 21 \end{bmatrix}$$

$\frac{-27}{68}$

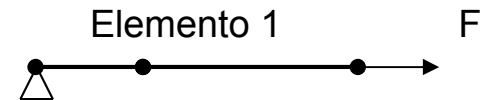
Solución Exacta:

$$u_3 = 3 \frac{F}{AE}$$



1 Elemento Barra de 3 nodos

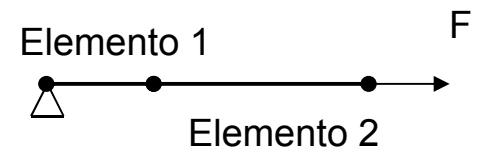
$$\frac{AE}{12} \begin{bmatrix} 24 & -18 & 2 \\ -18 & 27 & -9 \\ 2 & -9 & 7 \end{bmatrix} \begin{Bmatrix} 0 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} R \\ 0 \\ F \end{Bmatrix}$$



$$\frac{1}{9AE} \begin{bmatrix} 7 & 9 \\ 9 & 27 \end{bmatrix} \begin{Bmatrix} 0 \\ F \end{Bmatrix} = \frac{1}{AE} \begin{Bmatrix} F \\ 3F \end{Bmatrix}$$

2 Elementos Barra de 2 nodos

$$AE \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 + \frac{1}{2} & -\frac{1}{2} \\ 0 & -\frac{1}{2} & \frac{1}{2} \end{bmatrix} \begin{Bmatrix} 0 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} R \\ 0 \\ F \end{Bmatrix}$$



$$\frac{1}{AE} \begin{bmatrix} 1 & 1 \\ 1 & 3 \end{bmatrix} \begin{Bmatrix} 0 \\ F \end{Bmatrix} = \frac{1}{AE} \begin{Bmatrix} F \\ 3F \end{Bmatrix}$$

Elemento Viga:

$$\varepsilon_x(x,y)=\frac{\partial u(x,y)}{\partial x}=\frac{\partial\left(-y\frac{\partial v(x)}{\partial x}\right)}{\partial x}=-y\frac{\partial^2 v(x)}{\partial x^2}=-y\kappa$$

$$W_{int}=\int_v \varepsilon_x^T E \varepsilon_x dv = \int_0^L \int_{-\frac{b-h}{2}}^{\frac{b}{2}} \int_{-\frac{h}{2}}^{\frac{h}{2}} y^2 \kappa^T E \kappa dz dy dx = \int_0^L \kappa^T E \overbrace{\frac{h^3 b}{12}}^{I_z} \kappa dx$$

$$\kappa=\frac{\partial^2 v(x)}{\partial x^2}=\frac{\partial^2 \{N(x)\}}{\partial x^2}\{d\}=\begin{bmatrix} -\frac{6}{L^2}+\frac{12x}{L^3} \\ -\frac{4}{L}+\frac{6x}{L^2} \\ \frac{6}{L^2}-\frac{12x}{L^3} \\ -\frac{2}{L}+\frac{6x}{L^2} \end{bmatrix}^T \begin{Bmatrix} v_1 \\ \theta_{z1} \\ v_2 \\ \theta_{z2} \end{Bmatrix}$$

Matriz de Rigidez

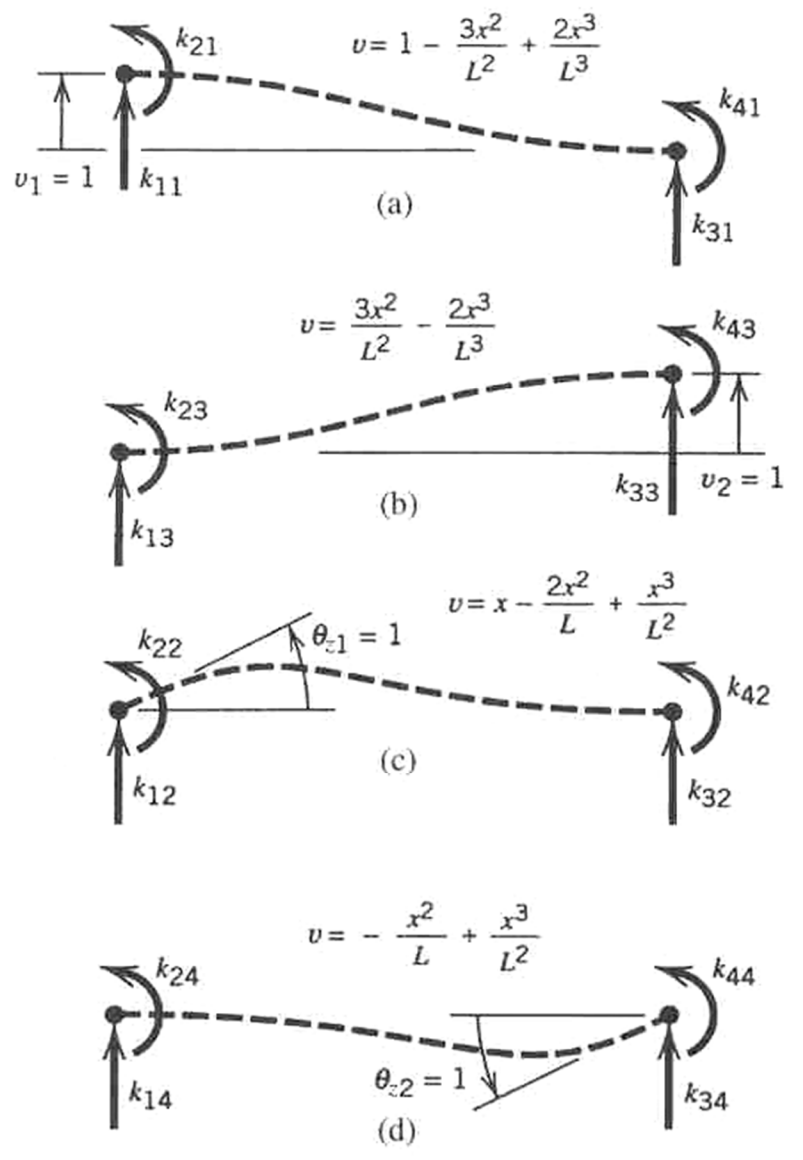
$$[K]=\int_{L_e} [B]^T E I_z [B] dx;$$

Elemento Viga:

$$v(x) = a_0 + a_1x + a_2x^2 + a_3x^3 = \underbrace{\begin{Bmatrix} 1 & x & x^2 & x^3 \end{Bmatrix}}_{\{X\}} \underbrace{\begin{Bmatrix} a_0 \\ a_1 \\ a_2 \\ a_3 \end{Bmatrix}}_{\{a\}}$$

$$\begin{Bmatrix} v_1 \\ \theta_{z1} \\ v_2 \\ \theta_{z2} \end{Bmatrix} = \begin{Bmatrix} a_0 + a_1x + a_2x^2 + a_3x^3 \\ a_1 + 2a_2x + 3a_3x^2 \\ a_0 + a_1x + a_2x^2 + a_3x^3 \\ a_1 + 2a_2x + 3a_3x^2 \end{Bmatrix} = \underbrace{\begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 1 & L & L^2 & L^3 \\ 0 & 1 & 2L & 3L^2 \end{bmatrix}}_{[A]} \begin{Bmatrix} a_0 \\ a_1 \\ a_2 \\ a_3 \end{Bmatrix}$$

$$[N] \begin{Bmatrix} v_1 \\ \theta_{z1} \\ v_2 \\ \theta_{z2} \end{Bmatrix} = \{X\} [A]^{-1} \begin{Bmatrix} v_1 \\ \theta_{z1} \\ v_2 \\ \theta_{z2} \end{Bmatrix} = \begin{bmatrix} 1 - \frac{3x^2}{L^2} + \frac{2x^3}{L^3} \\ x - \frac{2x^2}{L} + \frac{x^3}{L^2} \\ -\frac{2x^3}{L^3} + \frac{3x^2}{L^2} \\ -\frac{x^2}{L} + \frac{x^3}{L^2} \end{bmatrix}^T \begin{Bmatrix} v_1 \\ \theta_{z1} \\ v_2 \\ \theta_{z2} \end{Bmatrix}$$



Elementos Triangulares

Deformación Constante - CST

Problema Plano

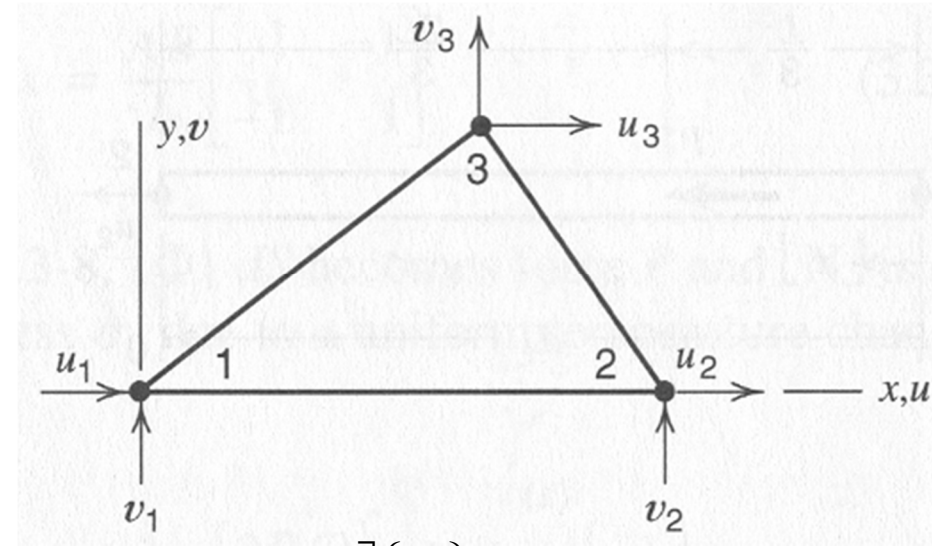
La matriz no depende de la posición (mas facil de integrar)

$$u(x, y) = \mathbf{a}_1 + \mathbf{a}_2 x + \mathbf{a}_3 y = \begin{Bmatrix} 1 & x & y \end{Bmatrix} \begin{Bmatrix} \mathbf{a}_1 \\ \mathbf{a}_2 \\ \mathbf{a}_3 \end{Bmatrix} ; \begin{cases} u(x_1, y_1) = u_1 \\ u(x_2, y_2) = u_2 \\ u(x_3, y_3) = u_3 \end{cases}$$

$$\begin{Bmatrix} \mathbf{a}_1 \\ \mathbf{a}_2 \\ \mathbf{a}_3 \end{Bmatrix} = \underbrace{\begin{bmatrix} 1 & x_1 & y_1 \\ 1 & x_2 & y_2 \\ 1 & x_3 & y_3 \end{bmatrix}}_{[A]}^{-1} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \frac{1}{\det(A)} \begin{bmatrix} x_2 y_3 - x_3 y_2 & x_3 y_1 - x_1 y_3 & x_1 y_2 - x_2 y_1 \\ y_2 - y_3 & y_3 - y_1 & y_1 - y_2 \\ x_3 - x_2 & x_1 - x_3 & x_2 - x_1 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix}$$

$$u(x, y) = \underbrace{\begin{Bmatrix} 1 & x & y \end{Bmatrix}}_{\substack{[N] \\ 1 \times 3}} [A]^{-1} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} N_1 & N_2 & N_3 \end{Bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \mathbf{a}_1 + \mathbf{a}_2 x + \mathbf{a}_3 y$$

$$v(x, y) = \underbrace{\begin{Bmatrix} 1 & x & y \end{Bmatrix}}_{[N]} [A]^{-1} \begin{Bmatrix} v_1 \\ v_2 \\ v_3 \end{Bmatrix} = \begin{Bmatrix} N_1 & N_2 & N_3 \end{Bmatrix} \begin{Bmatrix} v_1 \\ v_2 \\ v_3 \end{Bmatrix} = \mathbf{a}_4 + \mathbf{a}_5 x + \mathbf{a}_6 y$$



Elementos Triangulares

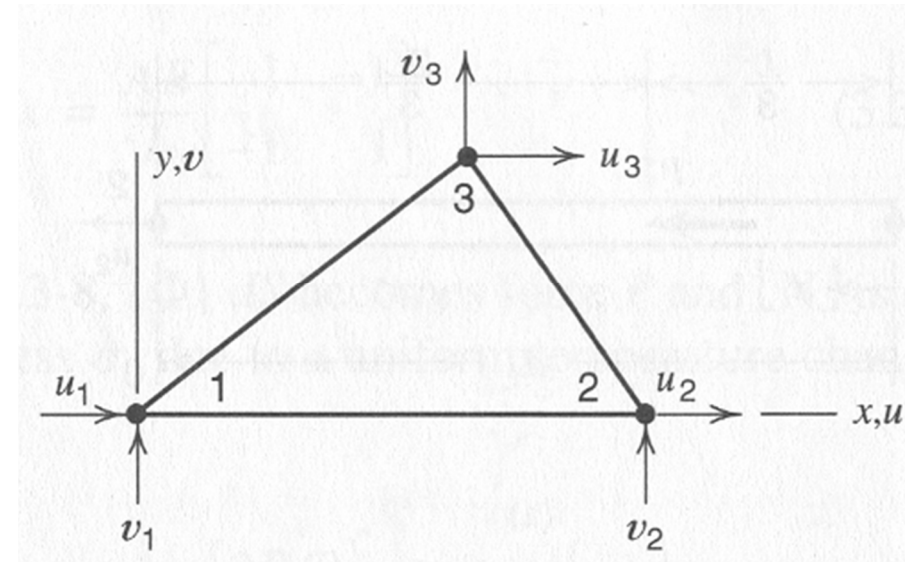
Deformación Constante - CST

Deformaciones

$$N_1 = \det(A)^{-1} ((x_2 y_3 - x_3 y_2) + x(y_2 - y_3) + y(x_3 - x_2))$$

$$N_2 = \det(A)^{-1} ((x_3 y_1 - x_1 y_3) + x(y_3 - y_1) + y(x_1 - x_3))$$

$$N_3 = \det(A)^{-1} ((x_1 y_2 - x_2 y_1) + x(y_1 - y_2) + y(x_2 - x_1))$$



$$\begin{Bmatrix} \varepsilon_x \\ \varepsilon_y \\ \gamma_{xy} \end{Bmatrix} = \begin{bmatrix} \frac{\partial}{\partial x} & 0 \\ 0 & \frac{\partial}{\partial y} \\ \frac{\partial}{\partial y} & \frac{\partial}{\partial x} \end{bmatrix} \begin{Bmatrix} u \\ v \end{Bmatrix}; \quad \begin{aligned} \varepsilon_x &= \frac{\partial u}{\partial x} = a_2 \\ \varepsilon_y &= \frac{\partial v}{\partial y} = a_6 \\ \gamma_{xy} &= \frac{\partial v}{\partial x} + \frac{\partial u}{\partial y} = a_3 + a_5 \end{aligned} \quad ; \quad [B]\{d\} = \{\partial\} \begin{bmatrix} N_1 & 0 & N_2 & 0 & N_3 & 0 \\ 0 & N_1 & 0 & N_2 & 0 & N_3 \end{bmatrix} \begin{Bmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{Bmatrix}$$

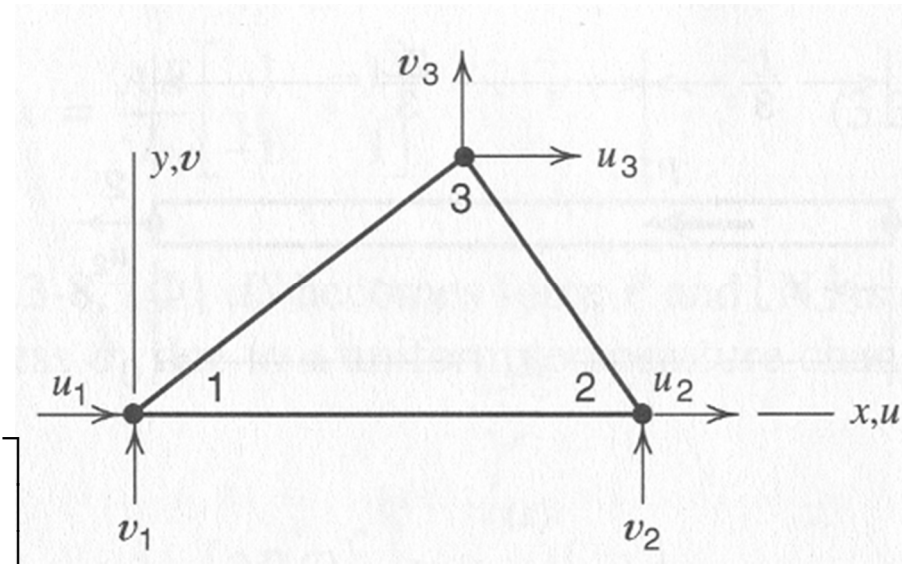
$$[B] = \begin{bmatrix} \frac{\partial N_1}{\partial x} & 0 & 0 \\ 0 & \frac{\partial N_1}{\partial y} & \dots & \frac{\partial N_3}{\partial y} \\ \frac{\partial N_1}{\partial y} & \frac{\partial N_1}{\partial x} & \dots & \frac{\partial N_3}{\partial x} \end{bmatrix} = \det(A)^{-1} \begin{bmatrix} y_2 - y_3 & 0 & y_3 - y_1 & 0 & y_1 - y_2 & 0 \\ 0 & x_3 - x_2 & 0 & x_1 - x_3 & 0 & x_2 - x_1 \\ x_3 - x_2 & y_2 - y_3 & x_1 - x_3 & y_3 - y_1 & x_2 - x_1 & y_1 - y_2 \end{bmatrix}$$

Elementos Triangulares

Deformación Constante - CST

Matriz de Rigidez

$$[B] = \frac{\begin{bmatrix} y_2 - y_3 & 0 & y_3 - y_1 & 0 & y_1 - y_2 & 0 \\ 0 & x_3 - x_2 & 0 & x_1 - x_3 & 0 & x_2 - x_1 \\ x_3 - x_2 & y_2 - y_3 & x_1 - x_3 & y_3 - y_1 & x_2 - x_1 & y_1 - y_2 \end{bmatrix}}{\det(A)}$$

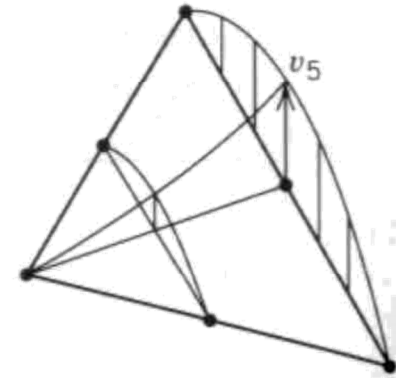
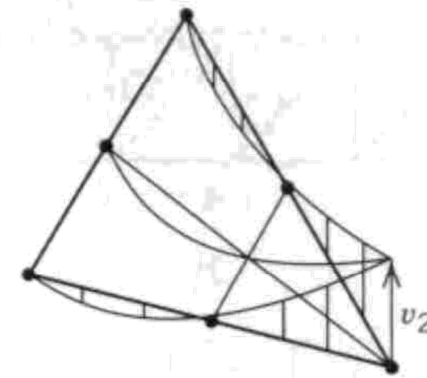
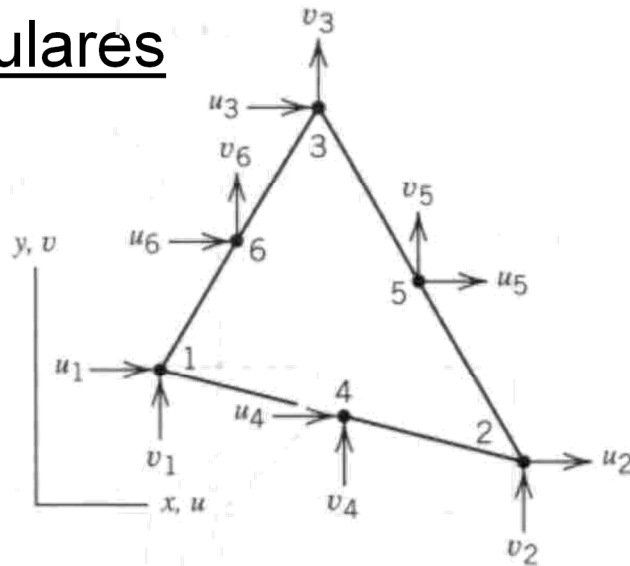


$$[K] = \int_S [B]^T [C] [B] ds$$

$$[K] = \frac{S}{\det(A)} \begin{bmatrix} y_2 - y_3 & 0 & x_3 - x_2 \\ 0 & x_3 - x_2 & y_2 - y_3 \\ y_3 - y_1 & 0 & x_1 - x_3 \\ 0 & x_1 - x_3 & y_3 - y_1 \\ y_1 - y_2 & 0 & x_2 - x_1 \\ 0 & x_2 - x_1 & y_1 - y_2 \end{bmatrix} \left(\frac{E}{(1-\nu^2)} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \right) [B]$$

Elementos Triangulares

Deformación Lineal - LST



$$u = a_1 + a_2x + a_3y + a_4x^2 + a_5xy + a_6y^2$$

$$v = a_7 + a_8x + a_9y + a_{10}x^2 + a_{11}xy + a_{12}y^2$$

La precisión del polinomio depende de si completo un piso:

$$\begin{array}{c} 1 \\ x \quad y \\ x^2 \quad xy \quad y^2 \\ x^3 \quad x^2y \quad xy^2 \quad y^3 \end{array}$$

El LST tiene el segundo piso completo

Deformaciones $\varepsilon_x = \frac{\partial u}{\partial x} = a_2 + 2a_4x + a_5y$

$$\varepsilon_y = \frac{\partial v}{\partial y} = a_9 + a_{11}x + 2a_{12}y$$

$$\gamma_{xy} = \frac{\partial v}{\partial x} + \frac{\partial u}{\partial y} = a_3 + a_8 + (a_5 + 2a_{10})x + (2a_6 + a_{11})y$$

Elementos Rectangulares

Deformación Constante - Q4

$$u = a_1 + a_2x + a_3y + a_4xy = N_1u_1 + \dots + N_4u_4$$

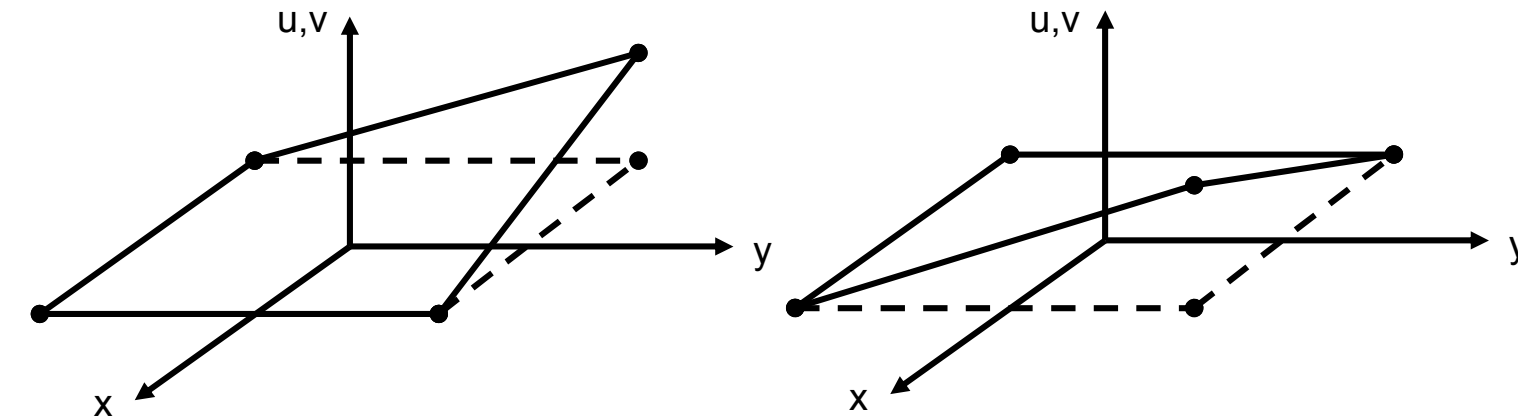
$$v = a_5 + a_6x + a_7y + a_8xy = N_1v_1 + \dots + N_4v_4$$

Deformaciones

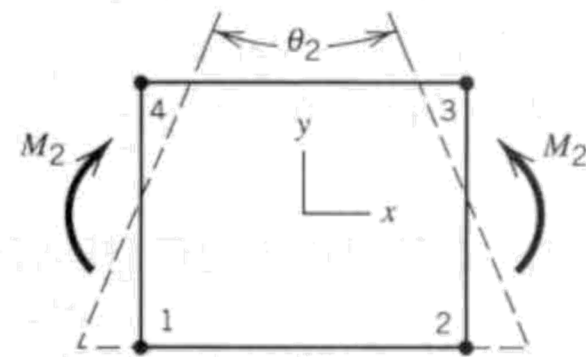
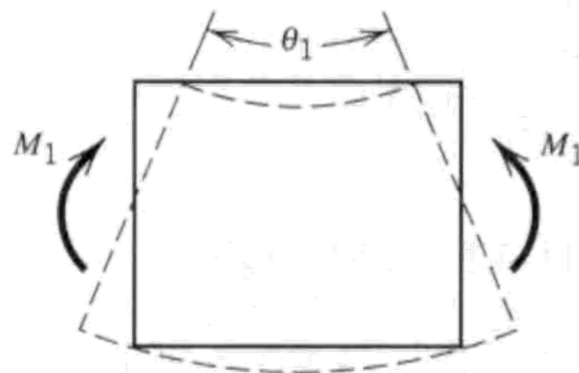
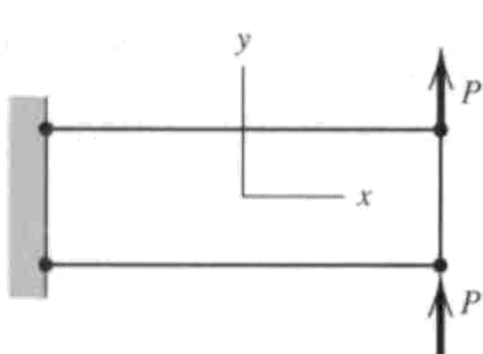
$$\varepsilon_x = \frac{\partial u}{\partial x} = a_2 + a_4y$$

$$\varepsilon_y = \frac{\partial v}{\partial y} = a_7 + a_8x$$

$$\gamma_{xy} = \frac{\partial v}{\partial x} + \frac{\partial u}{\partial y} = a_3 + a_6 + a_4x + a_8y$$



Corte Espúreo



No es bueno para simular vigas,
mala deformación por corte.

El corte espúreo se lleva la energía de deformación a corte, que para estas cuentas no existe. Quiere decir que se va a curvar menos de lo que realmente debería.

Problema Plano – Tensión Plana

Matriz Constitutiva (E=1, v=0.33):

$$C := \begin{pmatrix} 1.125 & 0.375 & 0 \\ 0.375 & 1.125 & 0 \\ 0 & 0 & 0.375 \end{pmatrix}$$

Funciones de Forma:

$$N1(x,y) = \frac{(a-x) \cdot (b-y)}{4a \cdot b}$$

$$N2 = \frac{(a+x) \cdot (b-y)}{4a \cdot b}$$

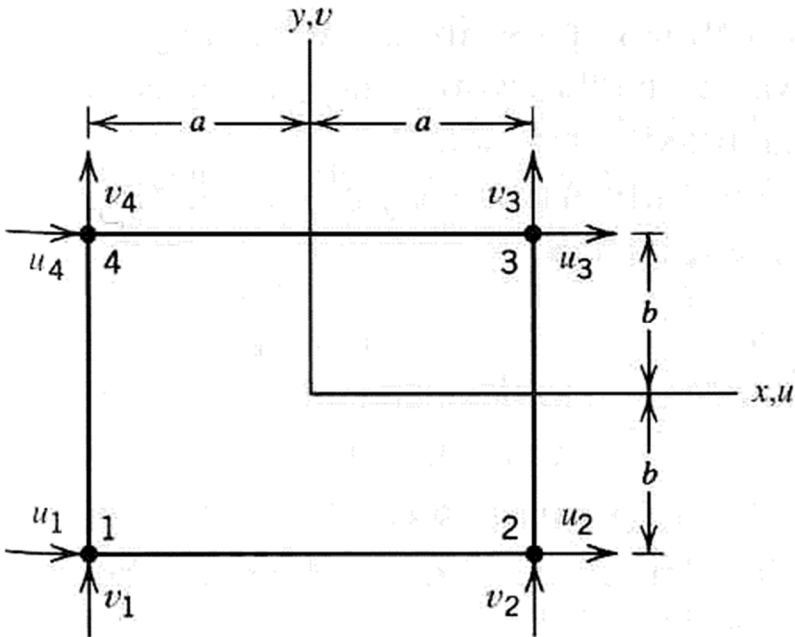
$$N3 = \frac{(a+x) \cdot (y+b)}{4a \cdot b}$$

$$N4 = \frac{(a-x) \cdot (y+b)}{4a \cdot b}$$

$$N(x,y) = \begin{pmatrix} N1(x,y) & 0 & N2(x,y) & 0 & N3(x,y) & 0 & N4(x,y) & 0 \\ 0 & N1(x,y) & 0 & N2(x,y) & 0 & N3(x,y) & 0 & N4(x,y) \end{pmatrix}$$

Matriz Desplazamiento- Deformación: [B]=[∂][N]

$$B(x,y) := \begin{pmatrix} \frac{d}{dx} N(x,y)_1 & 0 & \frac{d}{dx} N(x,y)_2 & 0 & \frac{d}{dx} N(x,y)_3 & 0 & \frac{d}{dx} N(x,y)_4 & 0 \\ 0 & \frac{d}{dy} N(x,y)_1 & 0 & \frac{d}{dy} N(x,y)_2 & 0 & \frac{d}{dy} N(x,y)_3 & 0 & \frac{d}{dy} N(x,y)_4 \\ \frac{d}{dy} N(x,y)_1 & \frac{d}{dx} N(x,y)_1 & \frac{d}{dy} N(x,y)_2 & \frac{d}{dx} N(x,y)_2 & \frac{d}{dy} N(x,y)_3 & \frac{d}{dx} N(x,y)_3 & \frac{d}{dy} N(x,y)_4 & \frac{d}{dx} N(x,y)_4 \end{pmatrix}_{77}$$



Problema Plano – Tensión Plana

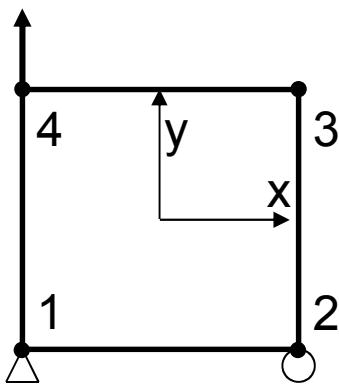
Adoptando a=1 y b=1

$$B(x,y) \rightarrow \begin{pmatrix} \frac{-1}{4} + \frac{1}{4} \cdot y & 0 & \frac{1}{4} - \frac{1}{4} \cdot y & 0 & \frac{1}{4} \cdot y + \frac{1}{4} & 0 & \frac{-1}{4} \cdot y - \frac{1}{4} & 0 \\ 0 & \frac{-1}{4} + \frac{1}{4} \cdot x & 0 & \frac{-1}{4} - \frac{1}{4} \cdot x & 0 & \frac{1}{4} + \frac{1}{4} \cdot x & 0 & \frac{1}{4} - \frac{1}{4} \cdot x \\ \frac{-1}{4} + \frac{1}{4} \cdot x & \frac{-1}{4} + \frac{1}{4} \cdot y & \frac{-1}{4} - \frac{1}{4} \cdot x & \frac{1}{4} - \frac{1}{4} \cdot y & \frac{1}{4} + \frac{1}{4} \cdot x & \frac{1}{4} \cdot y + \frac{1}{4} & \frac{1}{4} - \frac{1}{4} \cdot x & \frac{-1}{4} \cdot y - \frac{1}{4} \end{pmatrix}$$

Matriz de Rigidez

$$K = \int_{-a}^a \int_{-b}^b B(x,y)^T \cdot C \cdot B(x,y) \, dx \, dy$$

$$K = \begin{pmatrix} 0.5 & 0.188 & -0.313 & 0 & -0.25 & -0.188 & 0.063 & 0 \\ 0.188 & 0.5 & 0 & 0.063 & -0.188 & -0.25 & 0 & -0.313 \\ -0.313 & 0 & 0.5 & -0.188 & 0.063 & 0 & -0.25 & 0.188 \\ 0 & 0.063 & -0.188 & 0.5 & 0 & -0.313 & 0.188 & -0.25 \\ -0.25 & -0.188 & 0.063 & 0 & 0.5 & 0.188 & -0.313 & 0 \\ -0.188 & -0.25 & 0 & -0.313 & 0.188 & 0.5 & 0 & 0.063 \\ 0.063 & 0 & -0.25 & 0.188 & -0.313 & 0 & 0.5 & -0.188 \\ 0 & -0.313 & 0.188 & -0.25 & 0 & 0.063 & -0.188 & 0.5 \end{pmatrix}$$



Problema Plano – Tensión Plana

Matriz Reducida

$$K = \begin{bmatrix} 0.5 & 0.0625 & 0 & -0.25 & 0.1875 \\ 0.0625 & 0.5 & 0.1875 & -0.3125 & 0 \\ 0 & 0.1875 & 0.5 & 0 & 0.0625 \\ -0.25 & -0.3125 & 0 & 0.5 & -0.1875 \\ 0.1875 & 0 & 0.0625 & -0.1875 & 0.5 \end{bmatrix}$$

Desplazamientos Globales:

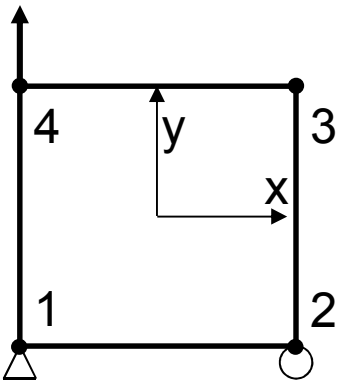
$$D = K^{-1}R \qquad D = \begin{Bmatrix} 0 \\ 0 \\ -\frac{1}{3} \\ 0 \\ \frac{5}{3} \\ -1 \\ 2 \\ 3 \end{Bmatrix}$$

Deformaciones Elementales:

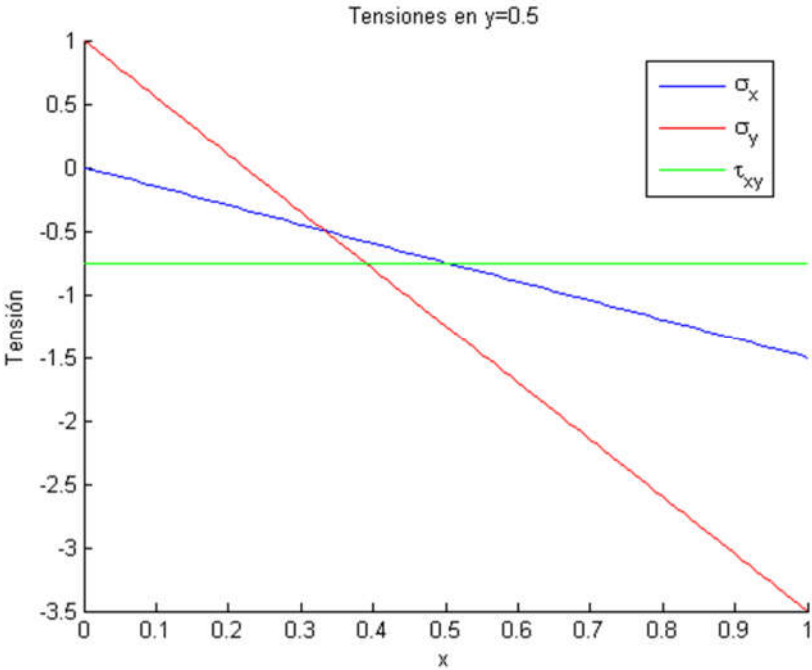
$$\varepsilon(x,y) = B(x,y)D \qquad \varepsilon(x,y) = \begin{Bmatrix} -\frac{1}{3} \\ 1-4x \\ -4y \end{Bmatrix}$$

Vector de Cargas:

$$R = \begin{Bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 1 \end{Bmatrix}$$



Variación Tensiones:

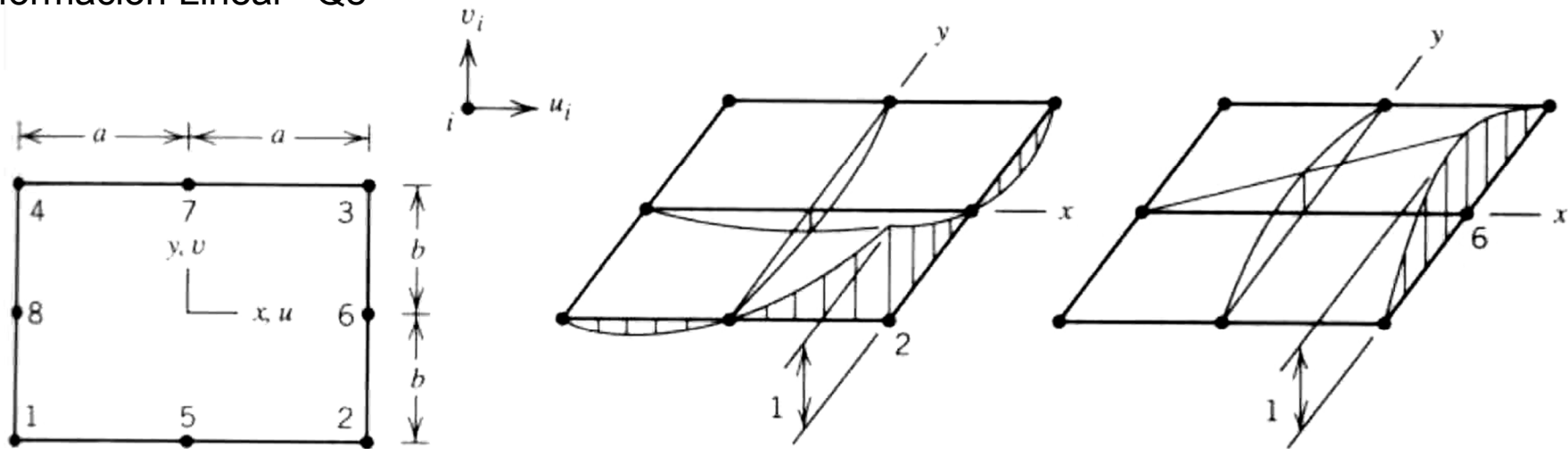


Tensiones Elementales:

$$\sigma(x,y) = CB(x,y)D \qquad \sigma(x,y) = \begin{Bmatrix} -\frac{3}{2}x \\ 1-\frac{9}{2}x \\ -\frac{3}{2}y \end{Bmatrix}$$

Elementos Rectangulares

Deformación Lineal - Q8



$$u = a_1 + a_2 x + a_3 y + a_4 x^2 + a_5 xy + a_6 y^2 + a_7 x^2 y + a_8 xy^2$$

$$v = a_9 + a_{10} x + a_{11} y + a_{12} x^2 + a_{13} xy + a_{14} y^2 + a_{15} x^2 y + a_{16} xy^2$$

Deformaciones

$$\varepsilon_x = \frac{\partial u}{\partial x} = a_2 + 2a_4 x + a_5 y + 2a_7 xy + a_8 y^2$$

$$\varepsilon_y = \frac{\partial v}{\partial y} = a_{11} + a_{13} x + 2a_{14} y + a_{15} x^2 + 2a_{16} xy$$

$$\gamma_{xy} = \frac{\partial v}{\partial x} + \frac{\partial u}{\partial y} = a_3 + a_{10} + (a_5 + 2a_{12})x + (2a_6 + a_{13})y + a_7 x^2 + 2(a_8 + a_{15})xy + a_{16} y^2$$