

# **GENERAL INSTRUCTIONS**

# **Project**



#### Auxiliar functions available in WeBeep

- For the assignments, you may use the auxiliar MATLAB functions available in WeBeep:
  - astroConstants: Use it to retrieve common astrodynamics constants (both assignments).
  - lambertMR: Use it for solving each Lambert arc (Assignment 1).
  - uplanet: Planets' ephemeris (don't propagate the planets' orbits yourself).
    - In Assignment 1, use it to compute the ephemerides of departure and flyby planets.
    - In Assignment 2, use it to compute the Sun-Earth position vector for SRP evaluation.
  - ephMoon: Analytical ephemeris of the Moon.
    - In Assignment 2, use it to compute the Moon position for third-body perturbation evaluation.
  - ephNEO: Ephemerides of several Near-Earth Objects.
    - In Assignment 1, use it to compute the ephemerides of arrival NEO.
  - timeConversion.zip: Compressed folder with several time conversion routines

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### **Teamwork**



As in a Mission Analysis team at ESA, you will also work in a group:

- Members of the group must cooperate: you are advised to share the work among the team, but
  everyone is responsible for all the work done in the project. This means the work of the team
  must be checked by the whole team.
- Make decisions towards design solutions based on numerical/analytical/physical evidence and analyses: you must always be able to motivate your design choices. You are supposed to perform the preliminary mission analysis of a real mission.
- During the final review (oral presentation), any team member can be questioned about any part of the work.



#### Overview

- Project evaluation includes:
  - Deliverables (1 submission per group)
    - Project report: A single PDF report on the assignments, of maximum 15 pages (total no exceptions)
    - Presentation slides
    - Simulation codes and results
    - Numerical results, to be submitted via a form
  - Peer evaluation
  - Oral presentation (final review)
    - 15 minutes followed by questions about the assignments and the theory of the course
    - All team members must participate in the oral presentation
    - Any student can be questioned about any part of the work



### Submission procedure and deadlines

- The deliverables must be submitted through WeBeep
  - Submit a single ZIP file with report, slides, and code, named "OrbitalMech\_group\_nnnn.zip", where nnnn is the group ID (e.g., OrbitalMech\_group\_2242.zip).
  - WeBeep submission file limit is 250MB. Larger submissions sizes are not allowed (nor needed).
  - Numerical results for specific questions and peer review submitted via forms in WeBeep.
  - Submissions via any other means will not be considered.
  - Changes to the deliverables after the deadline will not be considered.

#### Deadlines:

- Deliverables must be submitted by 7 January 2023.
  - Delivering the project is a must condition for the oral presentation and attending the written exam.
  - The delivery activity in WeBeep closes automatically on 7 January at 23:59.

#### Oral presentation

- Dates will be available during the Winter, Summer and September exam sessions. They will be notified at the beginning of each session.
- To be done before or within the same exam session (winter/summer/autumn) when the written is done.



### Report

- Single PDF of maximum 15 pages (both assignments in the same report).
- Include a front page with:
  - Title,
  - Group number, academic year,
  - For each member: full name, matriculation number, and person code.
- The report should contain explanations, data, figures, and tables supporting your design process and final solution.
  - You may follow the structure in the 'Mission analysis outputs' slides.
  - Properly indicate the units of all numerical data.
  - Include labels, legends and titles/captions in all figures.
  - No need to include theory, but properly introduce/reference all the formulas and models you use.
  - Include a 'References' section with a list of all the sources you consulted, and cite them in the text where appropriate.
  - Properly credit all images taken from other sources.



#### Code

- The codes for both assignments must be included inside a folder named Code, with two separate subfolders for each assignment as follows:
  - Assignment 1: Subfolder Code\Assignment1\ containing:
    - InterplanetaryMission\_group\_N.m: main script that reproduces your results (N is the group ID).
    - Code\Assignment1\functions\: subfolder with all the other functions you developed for the first assignment.
  - Assignment 2: Subfolder Code\Assignment2\ containing:
    - PlanetaryMission\_group\_N.m: main script that reproduces your results (N is the group ID).
    - Code\Assignment2\functions\: subfolder with all the other functions you developed for the second assignment.
- No need to upload the functions we provide to you in WeBeep, unless you modified them.



#### Code headers

- Each code file must include a header detailing:
  - Inputs and outputs (specify dimensions and units),
  - Authors,
  - Basic usage information

```
function dy = ode 2bp( t, y, muP )
%ode 2bp ODE system for the two-body problem (Keplerian motion)
% PROTOTYPE:
   dy = ode 2bp(t, y, mu)
% INPUT:
                Time (can be omitted, as the system is autonomous) [T]
    t[1]
   y[6x1]
                Cartesian state of the body ( rx, ry, rz, vx, vy, vz ) [ L, L/T ]
                Gravitational parameter of the primary [L^3/T^2]
   muP[1]
 OUTPUT:
                Derivative of the state [ L/T^2, L/T^3 ]
    dy [6x1]
 CONTRIBUTORS:
    Student 1
    Student 2
% VERSIONS
    2020-11-19: First version
```



#### Final review

The **final review** will take the form of an oral presentation, followed by several questions:

- Maximum 15 minutes for both assignments combined (not including the questions).
- All team members have to participate in the oral presentation.
- Any student can be questioned about any part of the work.
- Questions can be related to the report contents, design process, underlying theory, and final results.

### References



- Lecture notes and lab slides.
- Spacecraft orbital elements available at:
  - Space-Track: <a href="https://www.space-track.org">https://www.space-track.org</a>
  - Celestrack: <a href="https://celestrak.com/NORAD/elements/">https://celestrak.com/NORAD/elements/</a>
  - NASA/JPL's HORIZONS: <a href="https://ssd.jpl.nasa.gov/horizons/app.html">https://ssd.jpl.nasa.gov/horizons/app.html</a>

#### Books:

- D. Vallado, Fundamentals of Astrodynamics and Applications, 4th Edition, Springer, 2007, ISBN-13 978-0387718316.
   Chapters 8 and 9 (very detailed).
- R. H. Battin, An Introduction to the Mathematics and Methods of Astrodynamics, Revised Edition, AIAA Educational Series, Reston, 1999.
  - Chapter 10 (Gauss and Lagrange equations derivation).
- H. Curtis, Orbital Mechanics for Engineering Students, Second Edition, Butterworth-Heinemann, 2009, ISBN-13 978-0123747785.
  - Chapter 12 (introduction to orbit perturbations).