







AA 279 A – Space Mechanics Lecture 6: Notes

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→ Time and Reference systems

- Reading for next week (lectures 7 and 8)
 - → Bate n/a
 - Montenbruck 6.2.2, 9.1.1
 - → Vallado 6.8



Earth-Centered Earth-Fixed Coordinate System

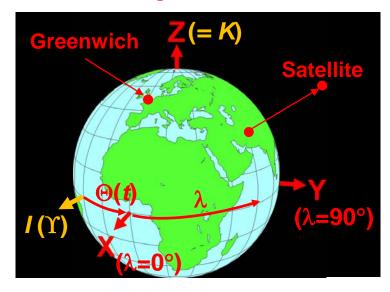
- → Solar Day (d) Familiar w.r.t. Sun

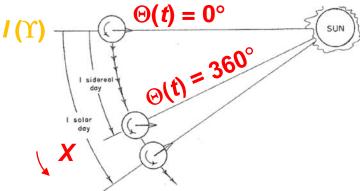
 Time between two subsequent meridian transits of sun is traditionally measured in solar days of 86400 [solar sec]
- Sidereal Day w.r.t. inertial direction
 Time taken for earth to rotate once w.r.t. inertial space, i.e. for Θ to span 360°, takes about 86164 [solar sec], i.e. about 4 solar minutes shorter than solar day
- → Neglecting small secular changes, the Greenwich mean sidereal time is given by

$$\Theta(d) = 280.4606^{\circ} + 360.9856473^{\circ} \cdot d$$

Solar time in days since January 1, 2000 at 12h

Ideal configuration ECI/ECEF

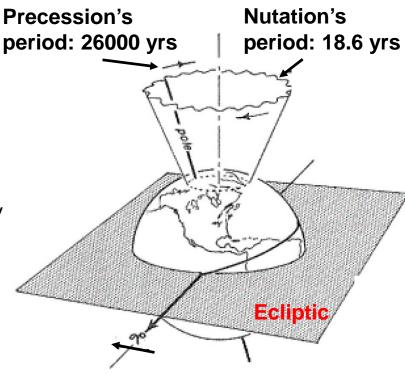




Time and Reference Systems

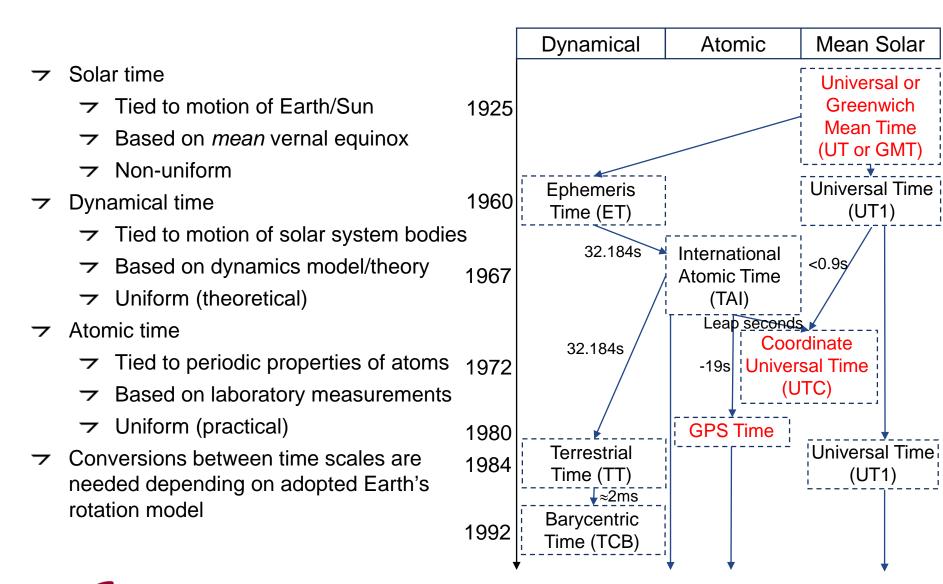
- So far we have tacitly assumed the availability of a unique time and reference system for the equation of motion (inertial)
- Traditionally, time and reference systems have been tied to the Earth's rotation and its revolution around the Sun
- ✓ In practice the Earth's orbit is slightly eccentric
 (e = 0.0167). In addition we don't live in a 2-body
 system and the rotational dynamics of the Earth
 is complex (perturbations are covered later)
- The Moon, Sun, and other planets act on a nonspherical Earth and produce torques on top of the purely radial attractive force
- This causes secular, periodic and short-term variations of the vernal equinox, equator, and Earth's rotation axis

Actual Earth's rotation dynamics



Secular drift westward of vernal equinox

Evolution of Time Scales



Simplified Calendrical Calculations

- Earth's rotation is typically expressed through Greenwich mean sidereal time Θ
- \neg Spacecraft time is typically expressed through more familiar t_{UTC} (or t_{GPS}) time
- Intermediate steps are required to convert between these time scales
- Since the civilian calendar is not well suited to handle time calculations, a Modified Julian Date (MJD) is used
- → The algorithm to compute MJD from a calendar date is given in "Montenbruck" on page 321 (A.1.1)

- 1) Obtain necessary time differences $t_{\rm UT1} t_{\rm UTC}$ (and $t_{\rm UTC} t_{\rm GPS}$) from IERS bulletin
- 2) Compute universal time $t_{\rm UT1}$
- 3) Compute Modified Julian Date (MJD) corresponding to t_{UT1}
- 4) Compute approximate Greenwich mean sidereal time in [deg]

$$\Theta(d) = 280.4606^{\circ} + 360.9856473^{\circ} \cdot d$$

$$d = MID - 51544.5$$



Conventional Reference Systems

→ Celestial Reference Frame (CRF)

- Quasi-inertial
- Origin at Earth's center of mass
- → I-axis aligned with mean equinox at J2000 epoch
- → K-axis normal to mean equator's plane at J2000 epoch
- Determined from a set of precise coordinates of extragalactic radio sources
- Mean equator and equinox of J2000 were defined by the International Astronomical Union (IAU)
- Earth Mean Equator and Equinox of J2000 (EME2000) is a practical implementation based on the FK5 star catalogue

→ Terrestrial Reference Frame (TRF)

- Co-rotating with Earth but slightly moving w.r.t. Earth's crust
- → Origin at Earth's center of mass
- Z-axis aligned with International Earth Rotation Service (IERS) pole
- X-axis defined by intersection of plane normal to Z and Greenwich mean meridian
- Determined from a set of precise coordinates of ground stations and updated every year
- World Geodetic System 1984 (WGS84) is another popular implementation based on the Global Positioning System (GPS)



Coordinate Transformation

- The transformation between CRF and TRF is accomplished by models for
 - Precession

Secular change in orientation of Earth's rotation axis and equinox

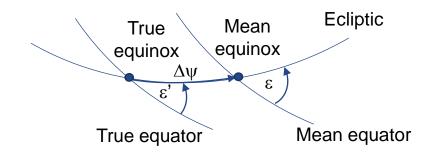
フ Nutation

Periodic and short-term variation of equator and equinox

→ Sidereal Time (w.r.t. UT1)

Earth's rotation about its axis

- These models are supplemented by the IERS Earth Observation Parameters (EOP), comprising
 - Observations of UT1-TAI difference
 - Measured coordinates of rotation axis relative to IERS pole



$$\vec{r}_{\text{MeanOfDate}} = \vec{R}_{\text{Pre}}(t) \vec{r}_{\text{CRS}}$$

$$| \approx 50"/\text{year (1.54km/year)}$$

$$\vec{r}_{\text{TrueOfDate}} = \vec{R}_{\text{Nut}}(t) \vec{r}_{\text{MeanOfDate}}$$

$$| \approx 20" \text{ (618m)}$$

$$\vec{r}_{\text{TRS'}} = \vec{R}_{\text{Rot}}(t) \vec{r}_{\text{TrueOfDate}} = \vec{R}_{\text{z}}(\Theta + \Delta \psi \cos \varepsilon) \vec{r}_{\text{TrueOfDate}}$$

$$| \text{Equation of Equinoxes: } \approx 20" \text{ (618m)}$$

$$\text{UT1-UTC: } \approx 15" \text{ (464m)}$$

$$\vec{r}_{\text{TRS}} = \vec{R}_{\text{Pol}}(t) \vec{r}_{\text{TRS'}}$$

$$| \approx 0.3" \text{ (10m)}$$

$$\vec{r}_{\text{TRS}} = \vec{R}_{\text{Pol}}(t) \vec{R}_{\text{Rot}}(t) \vec{R}_{\text{Nut}}(t) \vec{R}_{\text{Pre}}(t) \vec{r}_{\text{CRS}}$$