



Spacecraft Thermal Analysis II

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AA420 Space Design



Review

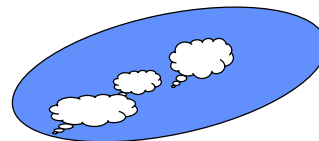
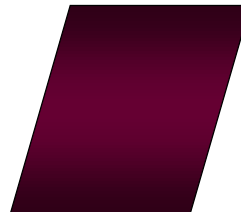
- Last time we went over the basic heat balance of objects in space.
- We showed how absorptivity and emissivity dominate heat transfer.
- We started into examples of simple passive heat transfer
- Homework added conduction and orbital environments as influences
 - Any questions regarding the homework?

Outline for Today

- A couple of additional simple objects - solar panels
- Add radiation between walls
- Set up another problem set with radiation, dissipation, and conductance

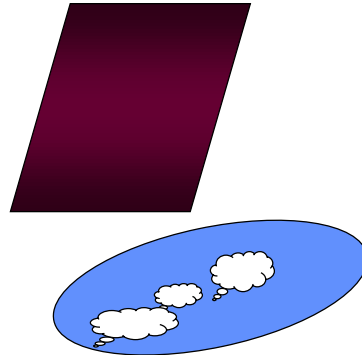
Solar Panel Examples

- An aluminum substrate solar panel is perpendicular to a vector to the center of the sun, at 1 AU. The other side is nadir pointing to a full visual field earth.
- 10 cm x 10 cm, thin
- 3 K space in the sun direction
- top: $\alpha = 0.805$, $\epsilon = 0.825$
- btm: $\alpha = 0.379$, $\epsilon = 0.0346$
- $Q_s = 1400 \text{ W/m}^2$
- $Q_e = 240 \text{ W/m}^2 + 420 \text{ W/m}^2$ solar reflection (albedo 0.3)
- $T_e = 291 \text{ K}$ (per UW School of Oceanography)
- Calculate T_{top} , T_{bottom}



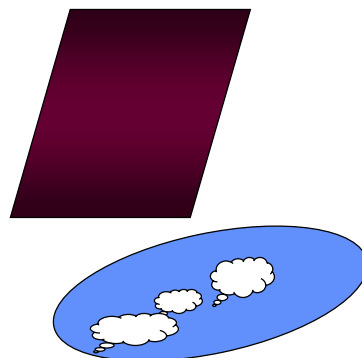
Paint the Earth Side White!

- An aluminum substrate solar panel is perpendicular to a vector to the center of the sun, at 1 AU. The other side is nadir pointing to a full visual field earth.
- 10 cm x 10 cm, thin
- 3 K space in the sun direction
- top: $\alpha = 0.805$, $\epsilon = 0.825$
- btm: $\alpha = 0.225$, $\epsilon = 0.825$
- $Q_s = 1400 \text{ W/m}^2$
- $Q_e = 240 \text{ W/m}^2 + 420 \text{ W/m}^2$ solar reflection (albedo 0.3)
- $T_e = 291 \text{ K}$
- Calculate T_{top} , T_{bottom}



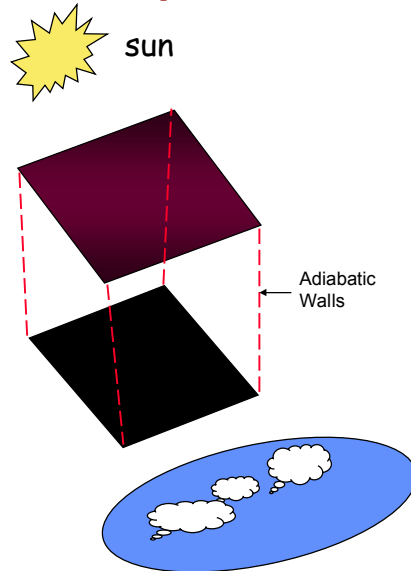
Kapton Substrate

- An Kapton blanket solar panel is perpendicular to a vector to the center of the sun, at 1 AU. The bare Kapton is nadir pointing to a full visual field earth.
- 100 cm x 100 cm, 0.01 mm Kapton
- 3 K space in the sun direction
- top: $\alpha = 0.805$, $\epsilon = 0.825$
- btm: $\alpha = 0.41$, $\epsilon = 0.86$
Note: translucency makes these values vary greatly!!!
- $Q_s = 1400 \text{ W/m}^2$
- $Q_e = 240 \text{ W/m}^2 + 420 \text{ W/m}^2$ solar reflection (albedo 0.3)
- $T_e = 291 \text{ K}$
- Calculate T_{top} , T_{bottom}



An Example More Like a Spacecraft

- A solar panel is perpendicular to a vector to the center of the sun, at 1 AU. The other side is white painted, nadir pointing to a full visual field earth. Inside surfaces are painted black.
- 10 cm x 10 cm, thin aluminum
- 3 K space in the sun direction
- top: $\alpha = 0.805$, $\epsilon = 0.825$
- inside: $\alpha = 0.95$, $\epsilon = 0.9$
- btm: $\alpha = 0.225$, $\epsilon = 0.825$
- $Q_s = 1400 \text{ W/m}^2$
- $Q_e = 240 \text{ W/m}^2 + 420 \text{ W/m}^2$ solar reflection (albedo 0.3)
- $T_e = 291 \text{ K}$
- Calculate inside surface T_{top} , T_{bottom}



Homework

1. Same as before but add Al conduction of the side walls 4 each, 0.1 cm x 10 cm x 10 cm Calculate inside surface T_{top} , T_{bottom}
2. Recalculate adding a 5 W heat dissipation on the earth facing plate to 1.
3. Calculate the same inside temperatures as 2. for eclipse

