

# PhysH308

Spinning *with* things!

Ted Brzinski, Nov. 7, 2024



Let's move the independent  
problem deadline to  
Monday?

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# Pre-registration ends this week!

There are a lot of upper-level options available this Spring!

## Physics courses:

1. **PHYSH302: Advanced Quantum**, with Walter Smith
2. Bryn Mawr is offering **PHYSB309 Advanced E&M**
3. **PHYSH304: Computational Physics** with our new faculty member Vijay Singh (whose research is in computational biological physics).
4. **PHYSH353: Topics in Soft Matter Physics**, a special topics course with visiting faculty member Vianney Gimenez-Pinto.
5. Bryn Mawr is offering **PHYSB331: Advanced Experimental Physics**

Not listed as physics, but physics, Clyde Daly in Chemistry is teaching **CHEMH350: Topics in Computational Chemistry** (time TBD), which can use PHYSH214 as a prereq.

## Astrophysics/astronomy courses:

6. **ASTR344: Topics in Astrophysics**: Gravitational Waves with Andrea Lommen.

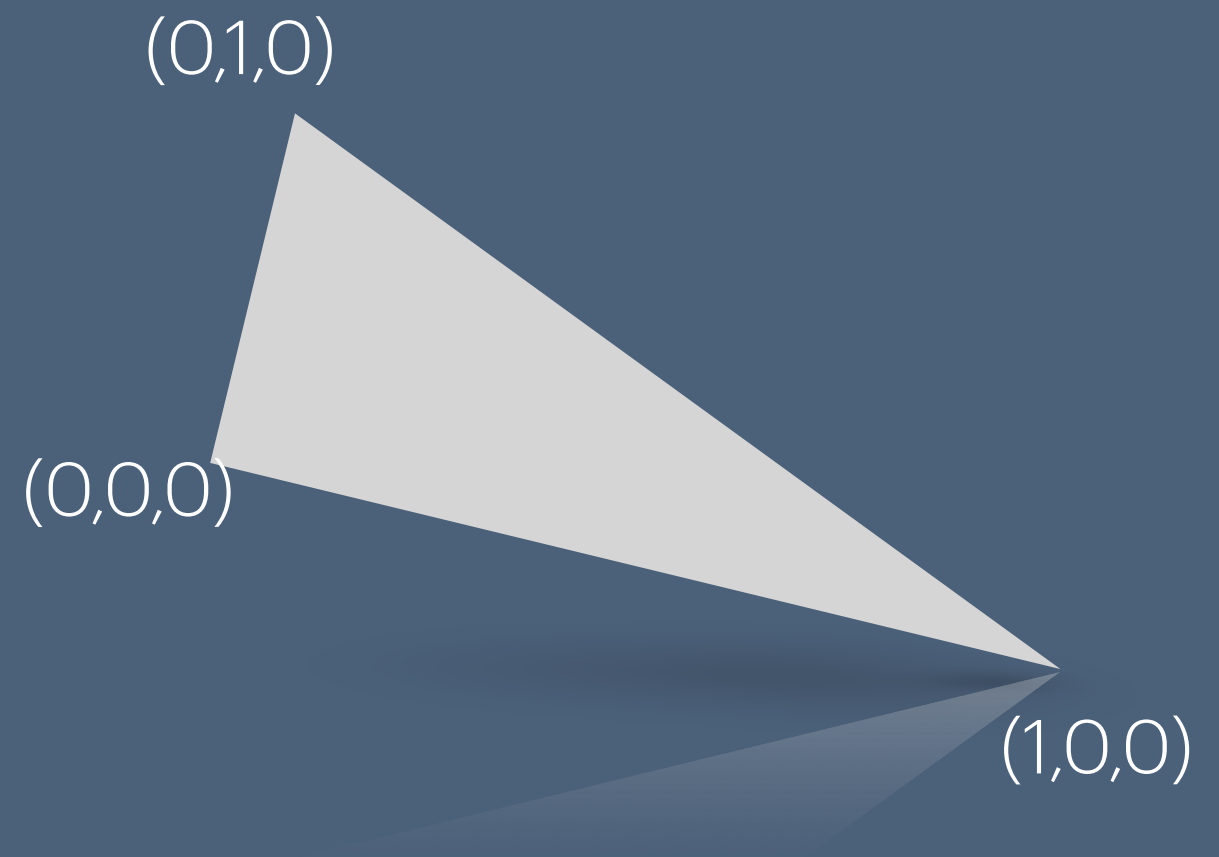
# Problem 10.37

Let's just do it — no need to turn it in!

- Given a shape, find  $\mathbf{I}$ , diagonalize it, and find the principal axes  
See that from  $\mathbf{I}$  in any coordinates you can find the principal axes

$$I_{xx} = m \sum (y^2 + z^2)$$

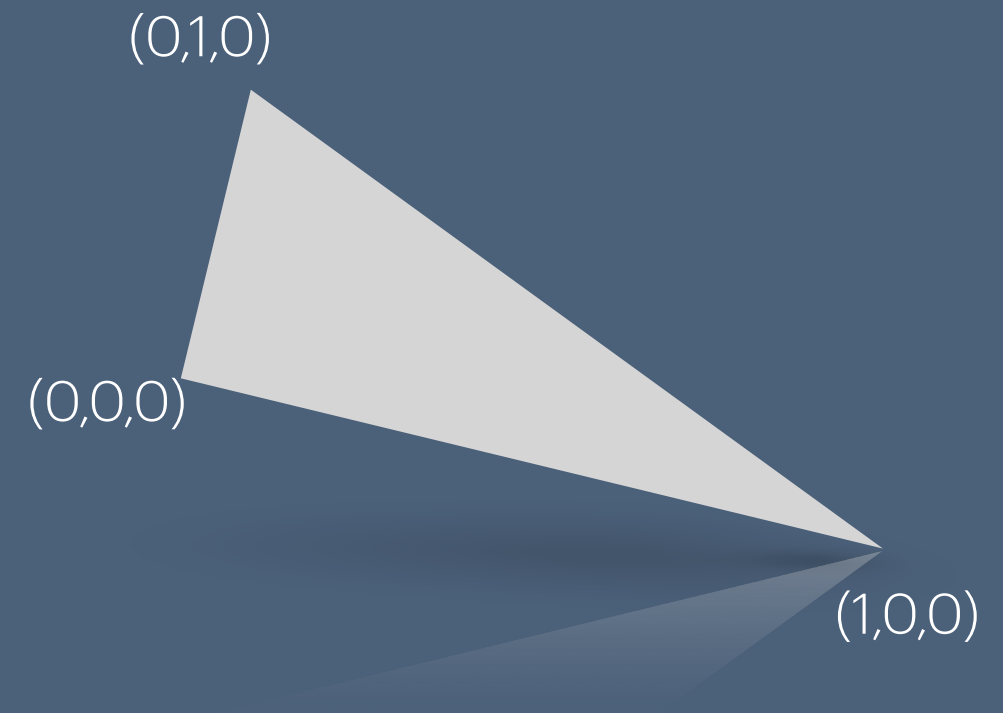
$$I_{xy} = I_{yx} = -m \sum xy$$



# Problem 10.37

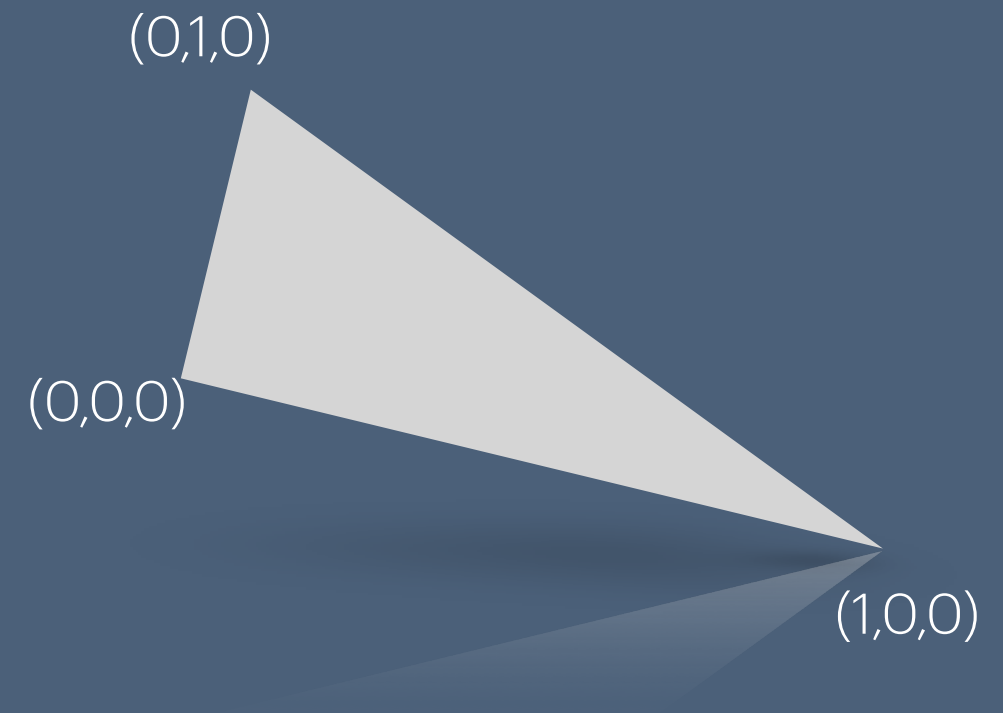
Moments of inertia

$$I_{xx} = \int dm (y^2 + z^2)$$



# Problem 10.37

Moments of inertia

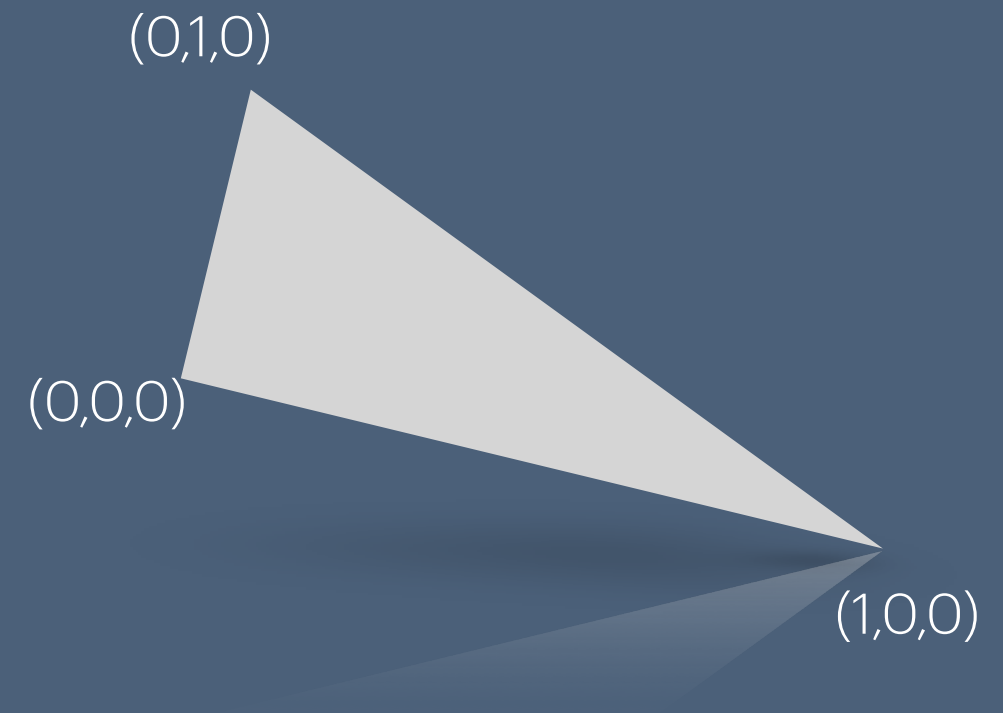


$$I_{xx} = \int dm (y^2 + z^2) = \sigma \int dA y^2$$



# Problem 10.37

Moments of inertia

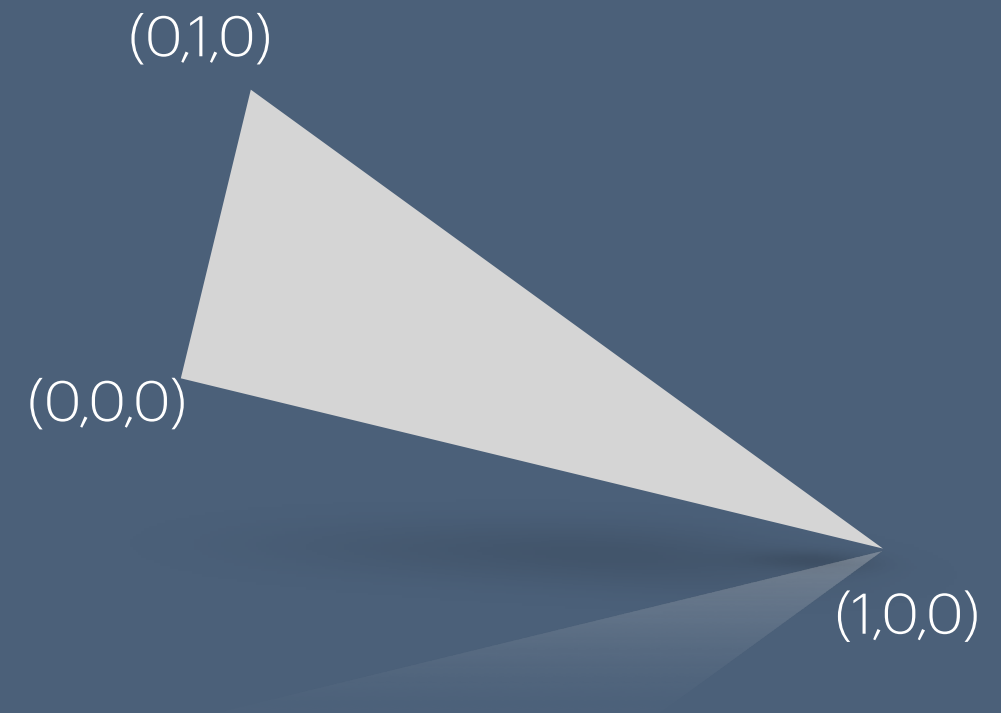


$$I_{xx} = \int dm (y^2 + z^2) = \sigma \int dA y^2 = \rho \int_0^1 dx \int_0^{1-x} dy y^2$$



# Problem 10.37

Moments of inertia



$$I_{xx} = \int dm (y^2 + z^2) = \sigma \int dA y^2 = \rho \int_0^1 dx \int_0^{1-x} dy y^2 = \sigma/12 = 2$$

$$I_{yy} = \int dm (x^2 + z^2) = \sigma \int dA x^2 = \rho \int_0^1 dy \int_0^{1-y} dx x^2 = \sigma/12 = 2$$

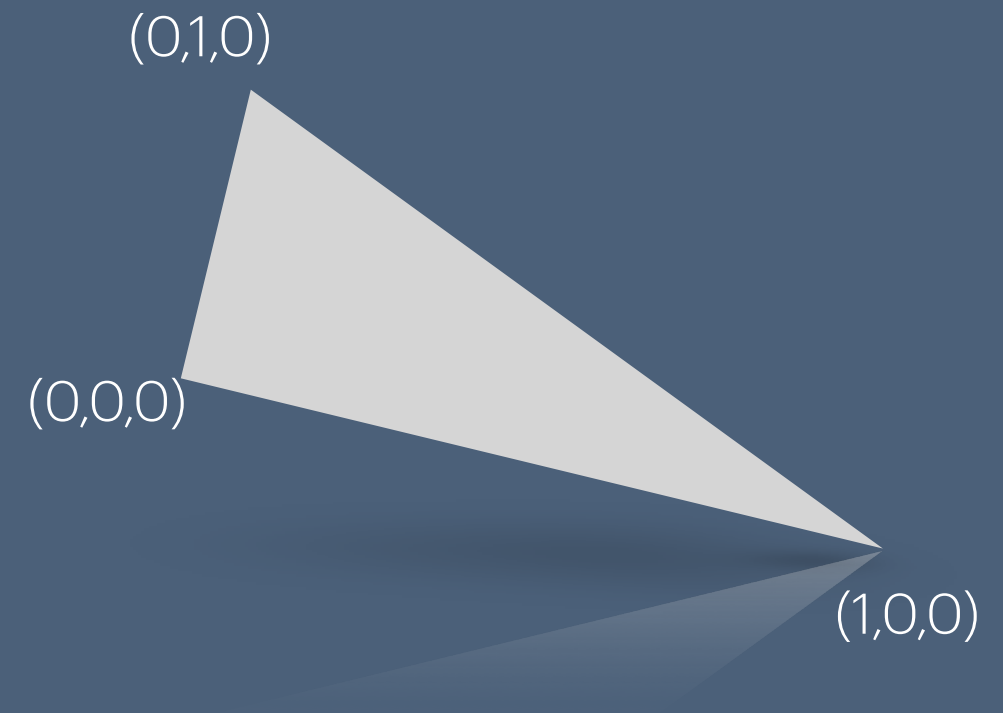
$$I_{zz} = \int dm (x^2 + y^2) = \int dm (x^2 + 2z^2 + y^2) = I_{xx} + I_{yy} = \sigma/6 = 4$$





# Problem 10.37

Products of inertia



$$I_{xz} = -\sigma \int dA \, xz = 0 = I_{yz}$$

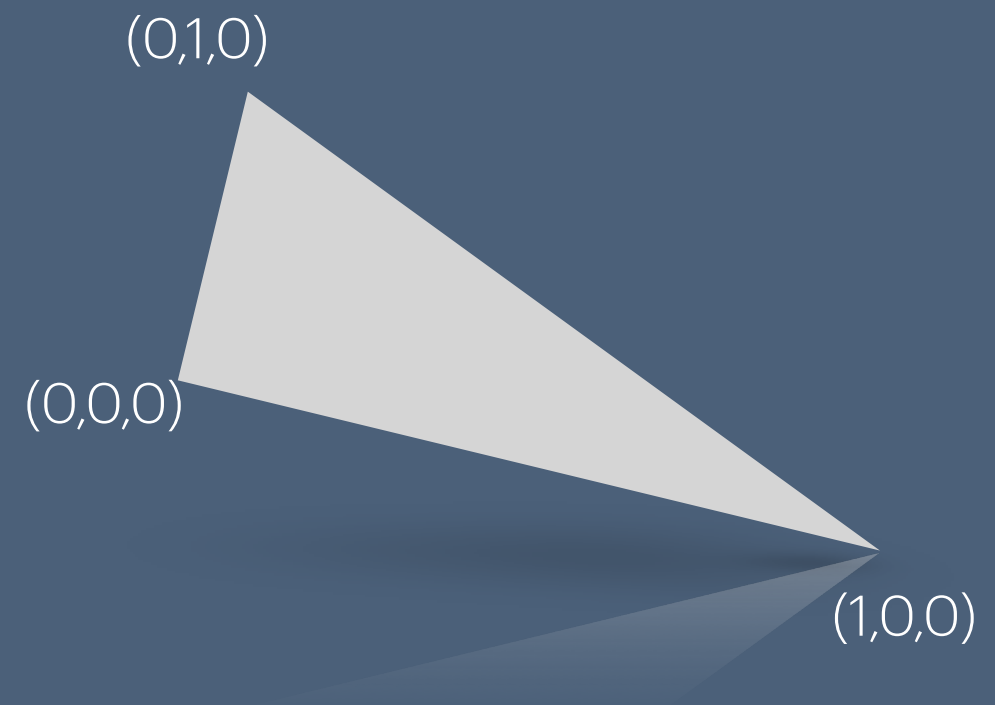
$$I_{xy} = -\sigma \int dA \, xy = -\sigma \int_0^1 dx \, x \int_0^{1-x} dy \, y = -1$$

$$\text{So, } \mathbf{I} = \begin{pmatrix} 2 & -1 & 0 \\ -1 & 2 & 0 \\ 0 & 0 & 4 \end{pmatrix}$$



# Problem 10.37

Products of inertia



$$\mathbf{I} = \begin{pmatrix} 2 & -1 & 0 \\ -1 & 2 & 0 \\ 0 & 0 & 4 \end{pmatrix}$$

$$\det(\mathbf{I} - \lambda \mathbf{1}) = 12 - 19\lambda + 8\lambda^2 - \lambda^3 = (1 - \lambda)(3 - \lambda)(4 - \lambda) = 0$$

$$\lambda_1 = 1, \lambda_2 = 3, \lambda_3 = 4$$

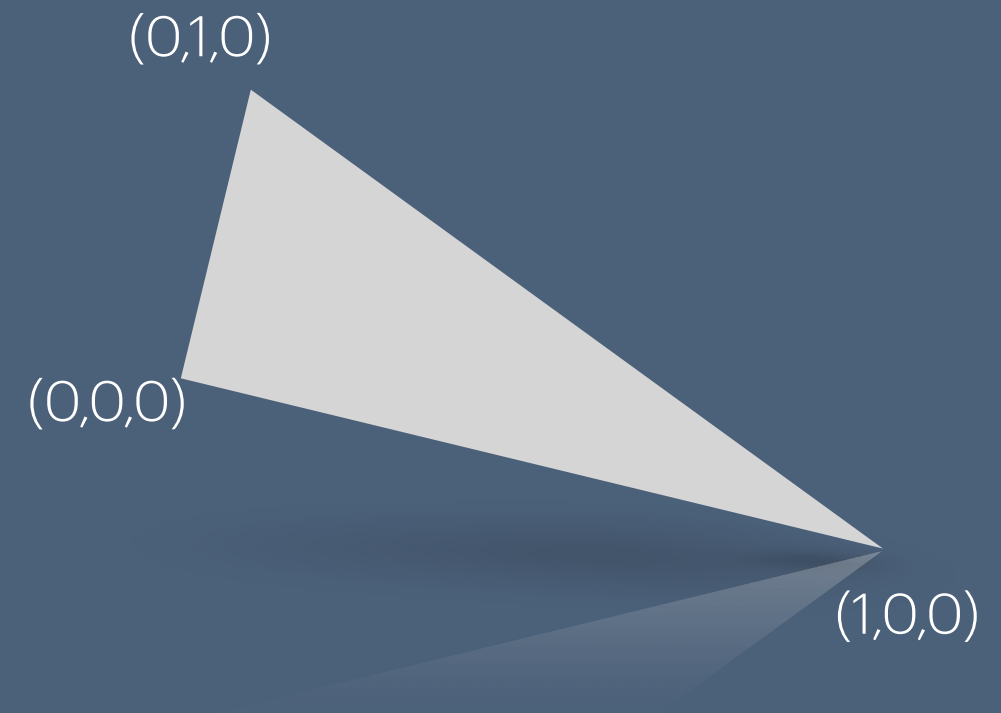
Solve  $(\mathbf{I} - \lambda_i \mathbf{1}) \hat{e}_i = 0$  to find

$$\hat{e}_1 = \frac{1}{\sqrt{2}} \begin{pmatrix} 1 \\ 1 \\ 0 \end{pmatrix}, \hat{e}_2 = \frac{1}{\sqrt{2}} \begin{pmatrix} 1 \\ -1 \\ 0 \end{pmatrix}, \hat{e}_3 = \begin{pmatrix} 0 \\ 0 \\ 1 \end{pmatrix}$$



# Problem 10.37

Products of inertia



$$\mathbf{I} = \begin{pmatrix} 2 & -1 & 0 \\ -1 & 2 & 0 \\ 0 & 0 & 4 \end{pmatrix}$$

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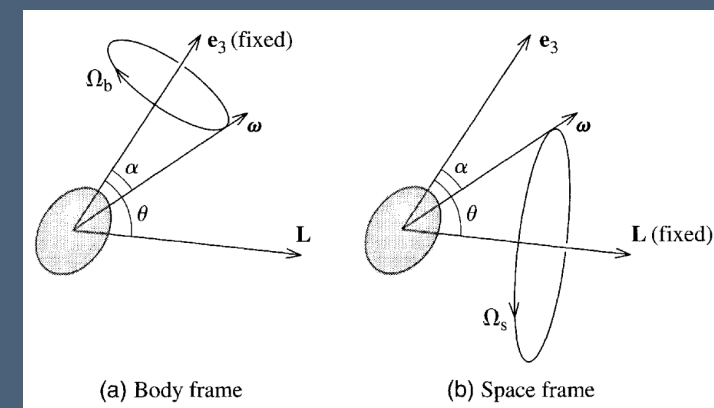
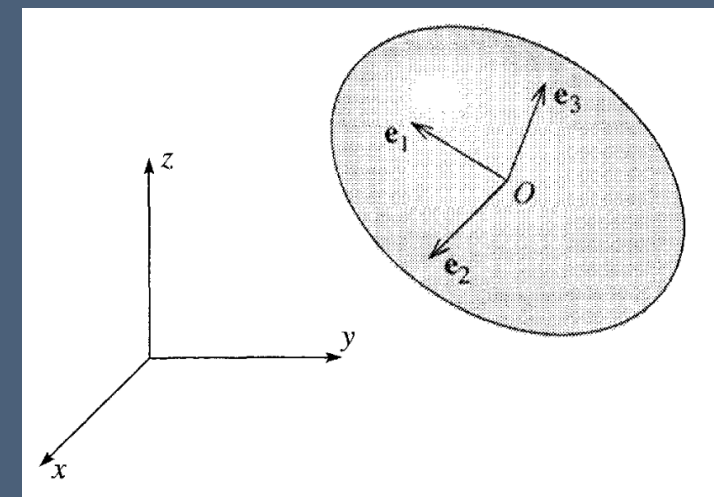
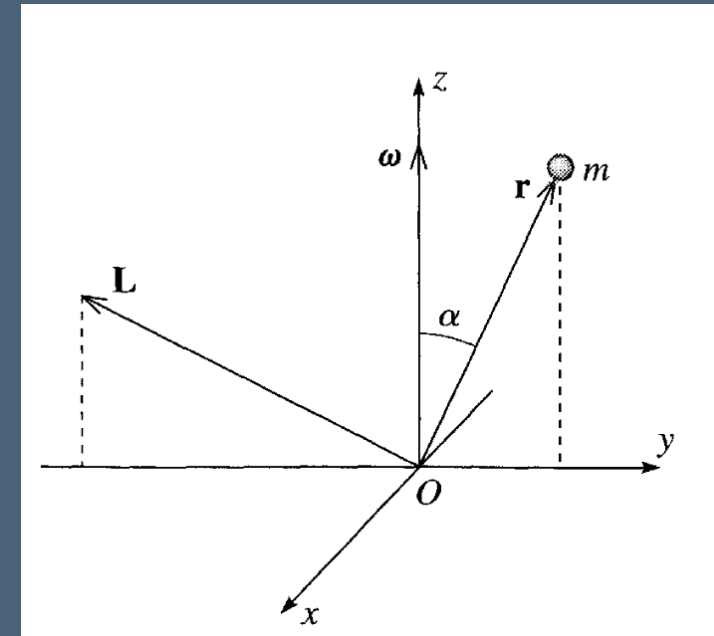
**This procedure can help with 10.42  
if you struggle to calculate  $\mathbf{I}$  at CoM**



# Euler's Angles

Return to the space frame

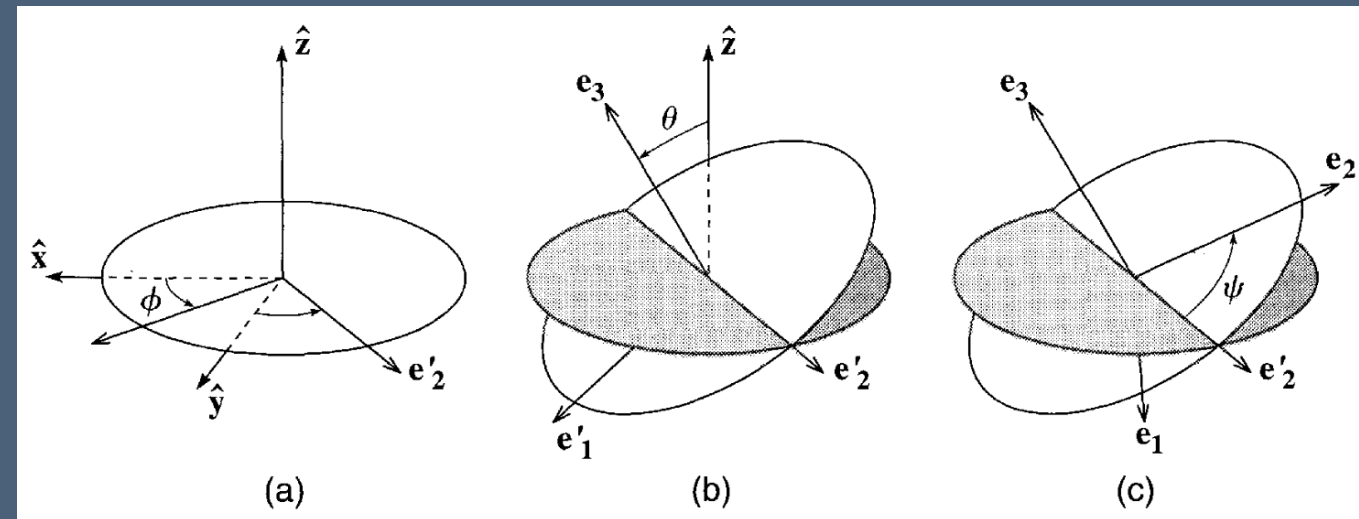
- We started with a fixed rotation in the space frame
- The inertia tensor allows us to understand the evolution of  $\vec{L}$  for fixed rotation
- Euler's Equations allowed us to solve general rotation *in the body frame*.
- Now we want to solve for general rotation *in the space frame*



# Euler's Angles

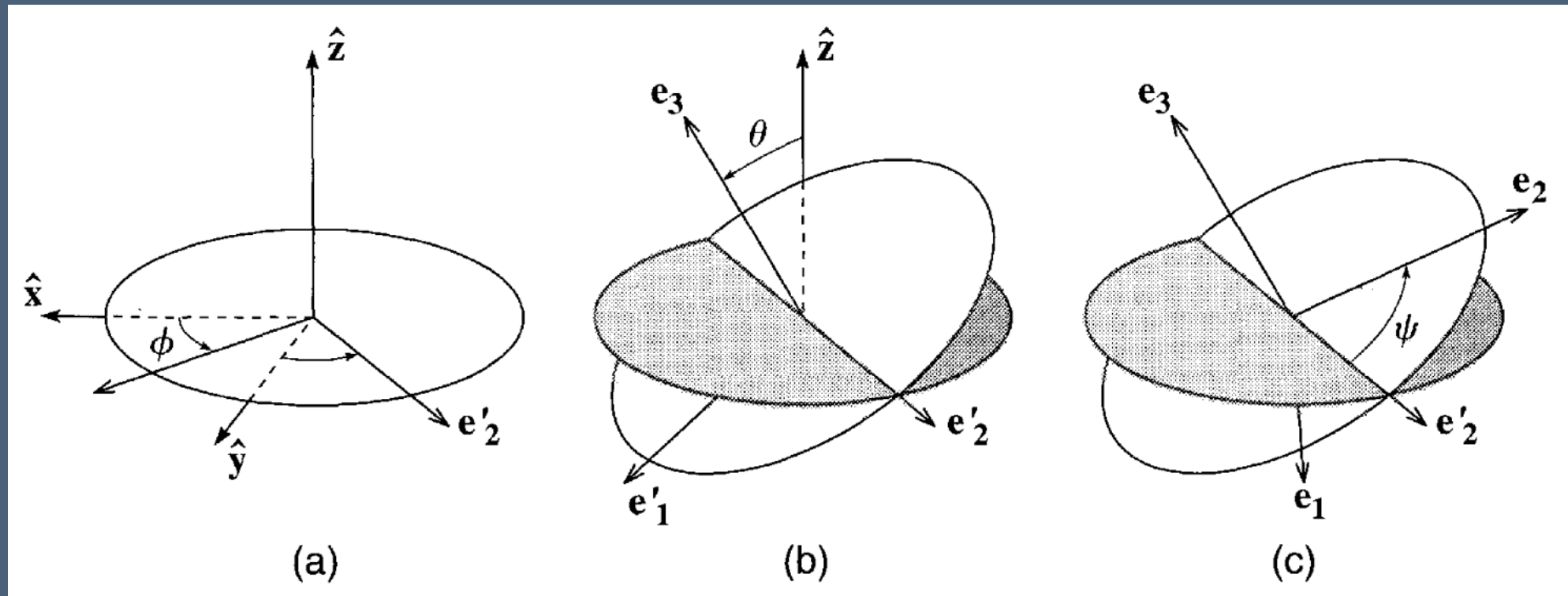
Return to the space frame

- We can describe the *orientation* of the body frame principal axes by considering rotation from space frame:
  1. Rotate by  $\phi$  about  $\hat{z}$  to align  $\hat{e}_1$  with  $\theta$
  2. Rotate by  $\theta$  around  $\hat{e}_2$  to put  $\hat{e}_3$  in its final orientation
  3. Rotate by  $\psi$  about  $\hat{e}_3$  to put  $\hat{e}_1, \hat{e}_2$  in their final orientations



# Euler's Angles

Return to the space frame

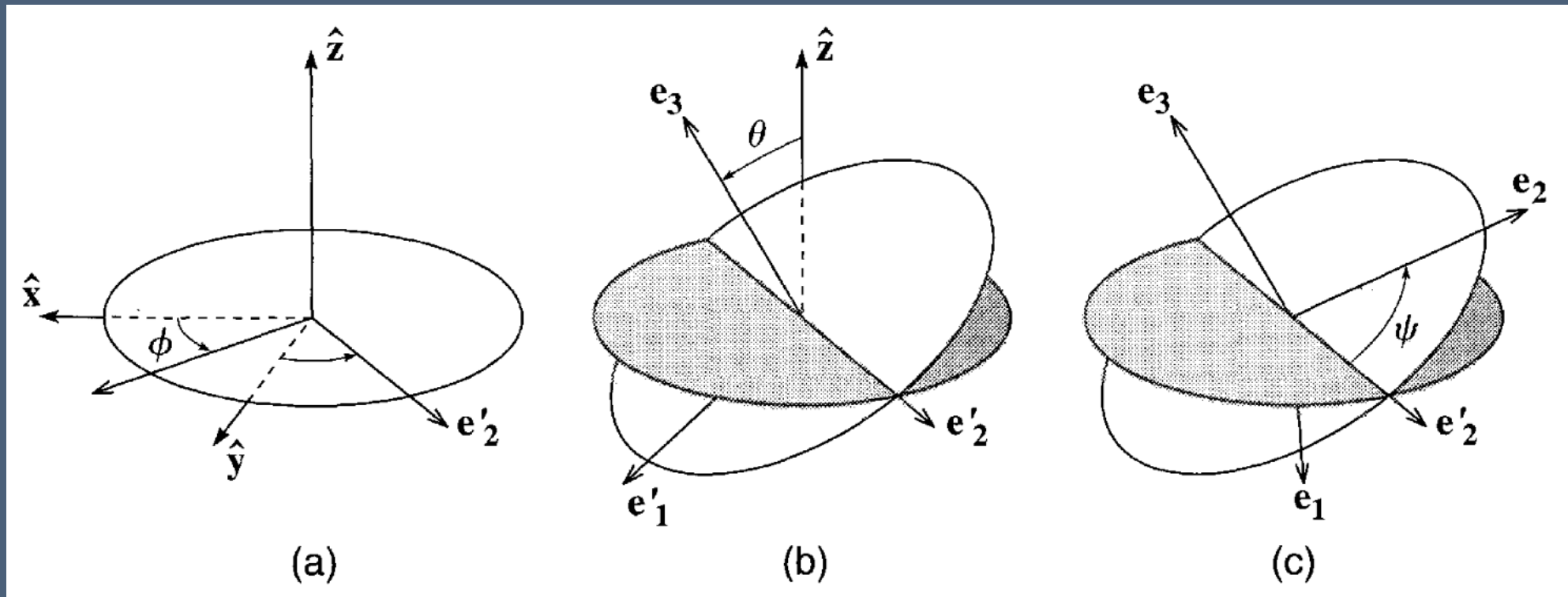


- Now we can rewrite  $\vec{\omega}$ ,  $\vec{L}$ , and thus  $T$  (see book for details; in this class we'll always assume  $\lambda_1 = \lambda_2$ , which lets us disregard the last rotation for  $\hat{e}_i$ ) in a body frame explicitly in terms of it's relation to the space frame!

$$\vec{\omega} = \left( -\dot{\phi} \sin \theta \right) \hat{e}'_1 + \dot{\theta} \hat{e}'_2 + \left( \dot{\psi} + \dot{\phi} \cos \theta \right) \hat{e}_3$$

# Euler's Angles

Return to the space frame



$$\cdot \vec{\omega} = \left( -\dot{\phi} \sin \theta \right) \hat{e}'_1 + \dot{\theta} \hat{e}'_2 + \left( \dot{\psi} + \dot{\phi} \cos \theta \right) \hat{e}_3$$

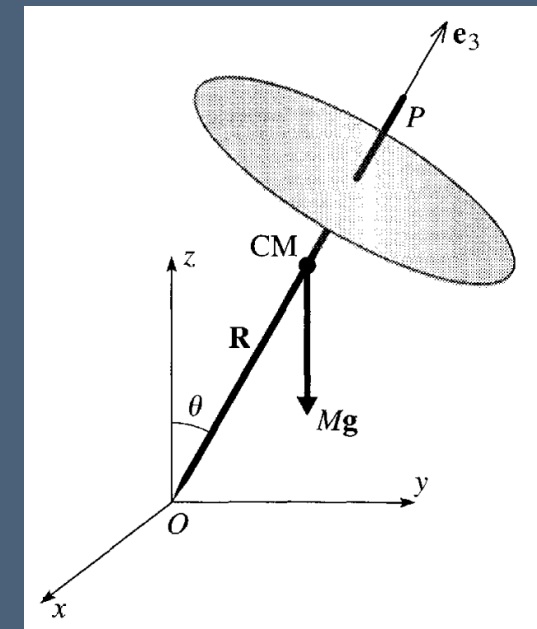
$$\cdot \vec{L} = \left( -\lambda_1 \dot{\phi} \sin \theta \right) \hat{e}'_1 + \lambda_1 \dot{\theta} \hat{e}'_2 + \lambda_3 \left( \dot{\psi} + \dot{\phi} \cos \theta \right) \hat{e}_3$$

$$\cdot T = \frac{1}{2} \lambda_i \omega_i^2 = \frac{1}{2} \lambda_1 \left( \dot{\phi}^2 \sin^2 \theta + \dot{\theta}^2 \right) + \frac{1}{2} \lambda_3 \left( \dot{\psi} + \dot{\phi} \cos \theta \right)^2$$

# Euler's Angles

Return to the space frame

- We have  $T = \frac{1}{2}\lambda_1 \left( \dot{\phi}^2 \sin^2 \theta + \dot{\theta}^2 \right) + \frac{1}{2}\lambda_3 \left( \dot{\psi} + \dot{\phi} \cos \theta \right)^2$
- For the top,  $U = MgR_{cm} \cos \theta$
- If we have  $T$  and  $U$ , we can use....





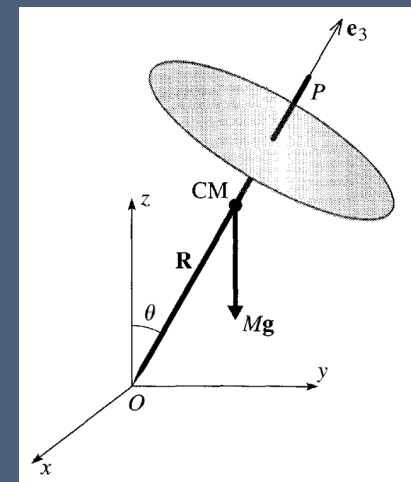
# Euler's Angles

Return to the space frame

- We have  $T = \frac{1}{2}\lambda_1 \left( \dot{\phi}^2 \sin^2 \theta + \dot{\theta}^2 \right) + \frac{1}{2}\lambda_3 \left( \dot{\psi} + \dot{\phi} \cos \theta \right)^2$

- For the top,  $U = MgR_{cm} \cos \theta$

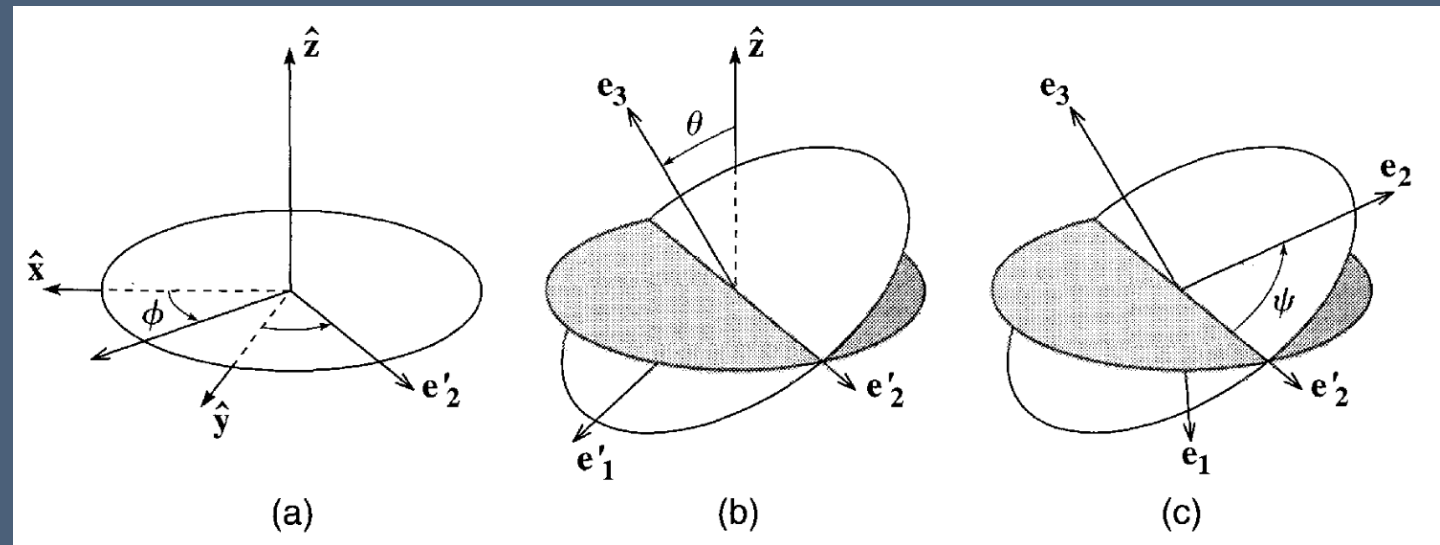
- If we have  $T$  and  $U$ , we can use....



$$\mathcal{L} = \frac{1}{2}\lambda_1 \left( \dot{\phi}^2 \sin^2 \theta + \dot{\theta}^2 \right) + \frac{1}{2}\lambda_3 \left( \dot{\psi} + \dot{\phi} \cos \theta \right)^2 - MgR_{cm} \cos \theta$$

# Euler's Angles

Return to the space frame



$$\bullet \mathcal{L} = \frac{1}{2}\lambda_1 \left( \dot{\phi}^2 \sin^2 \theta + \dot{\theta}^2 \right) + \frac{1}{2}\lambda_3 \left( \dot{\psi} + \dot{\phi} \cos \theta \right)^2 - MgR_{cm} \cos \theta$$

$$\bullet \text{ By inspection, } \frac{\partial \mathcal{L}}{\partial \psi} = \frac{\partial \mathcal{L}}{\partial \phi} = 0, \text{ so conserved momenta!}$$

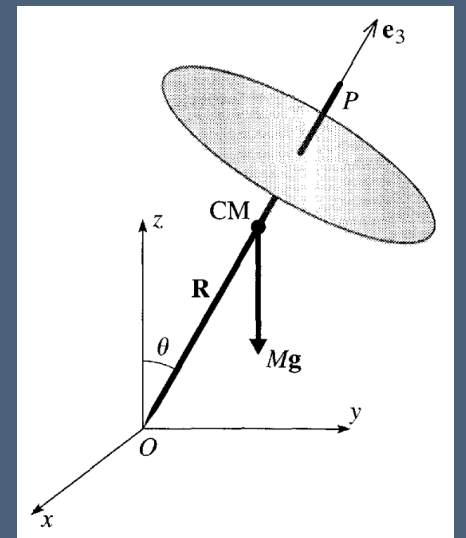
(angular momenta about  $\hat{z}$ ,  $\hat{e}_3$ : spin+precession)

• Solving for precession rate:

$$\dot{\phi} = \text{const} = \Omega = \begin{cases} \frac{MgR}{\lambda_3 \omega_3} \\ \frac{\lambda_3 \omega_3}{\lambda_1 \cos \theta} \end{cases}$$

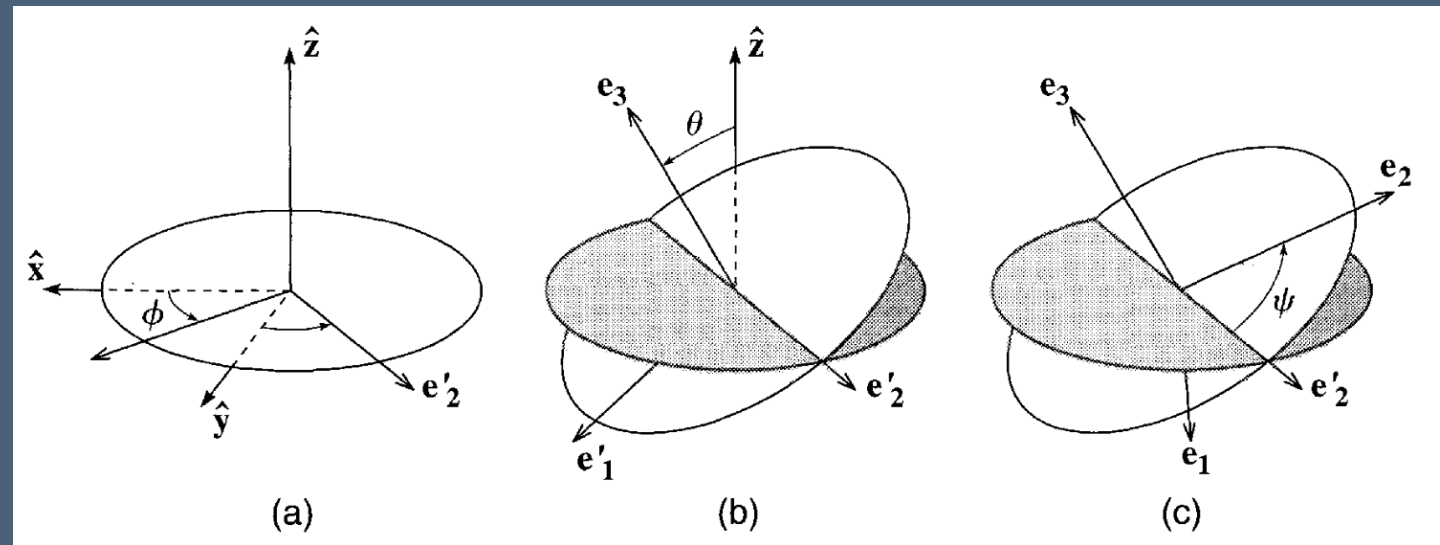
Torque due to gravity!

Like the wobbling book!



# Euler's Angles

Return to the space frame



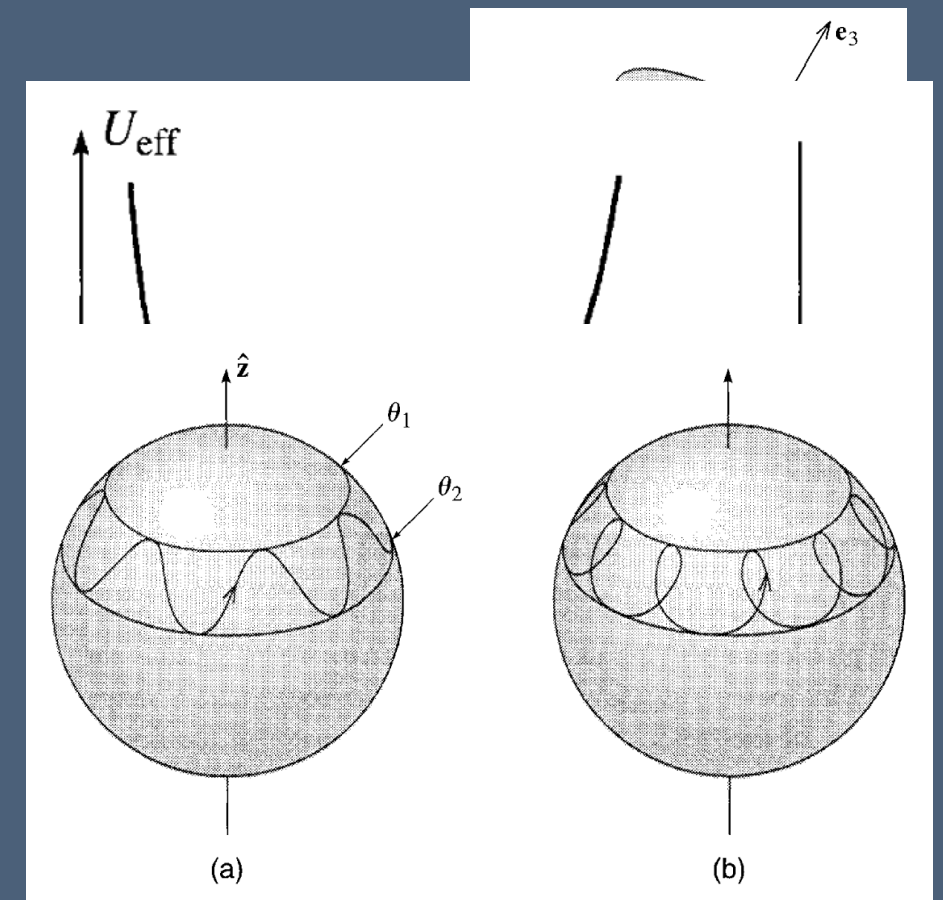
- $\mathcal{L} = \frac{1}{2}\lambda_1 \left( \dot{\phi}^2 \sin^2 \theta + \dot{\theta}^2 \right) + \frac{1}{2}\lambda_3 \left( \dot{\psi} + \dot{\phi} \cos \theta \right)^2 - MgR_{cm} \cos \theta$

- What about  $\frac{\partial \mathcal{L}}{\partial \theta} = \frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{\theta}}$ ?

- Here, it's actually easier to consider  $U_{eff}(\theta)$

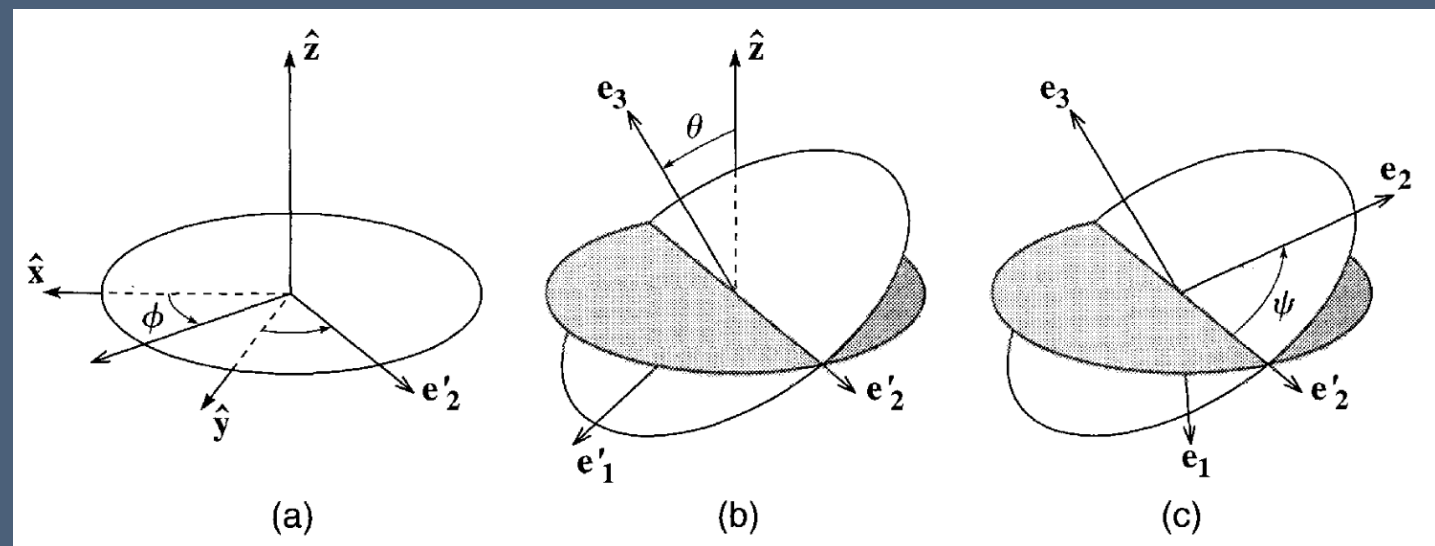
- If  $\theta$  varies, so does  $\dot{\phi} = \frac{L_z - L_3 \cos \theta}{\lambda_1 \sin^2 \theta}$

- In 10.51, you'll find  $U_{eff}(\theta)$  and see this relation directly!



# Problem 10.51

Return to the space frame



- It's actually easier to consider  $U_{eff}(\theta)$

- If  $\theta$  varies, so does  $\dot{\phi} = \frac{L_z - L_3 \cos \theta}{\lambda_1 \sin^2 \theta}$

- In **10.51**, you'll find  $U_{eff}(\theta)$  and see this relation directly!

