Reg. No.	
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## **B.Tech. DEGREE EXAMINATION, DECEMBER 2023**

Fifth Semester

## 18ASE202T - APPLIED STRUCTURAL MECHANICS

(For the candidates admitted during the academic year 2020-2021 & 2021-2022)

(i) (ii)	<b>Part - A</b> should be answered in OMR sheet within first 40 minutes and OMR sheet should be hall invigilator at the end of 40 <sup>th</sup> minute.						hande	d ove	er to
ime:	3 h	ours				Max	x. Mar	ks: 1	.00
			DADT	A (20 v. 1	20	Monko	Marks	BL	co
				A (20 × 1					
	1	Error		ver ALL (			1	1	1
	1. Every airplane is limited to the it can be flown and the					an be nown and the it can be			
			Maximum velocity, r		(B)	Maximum acceleration.			
		(11)	acceleration	1100111	(2)	maximum velocity			
		( <b>C</b> )		naximum	(D)	Maximum thrust,			
		(-)	weight			minimum acceleration			
	2.		n basic Physics, the relation on stant, is given by	nship for a	moti	ion of pure translation if the acceleration	1	1	1
			u - v = at		(B)	v + u = at			
		(C)	v - u = at		(D)	v + u = t			
	3.	The	shock struct in a landing ge	ar is comn	nonly	referred to as	1	1	1
		(A)	Oleo frame		(B)	Oleo member			
		(C)	Oleo structure		(D)	Oleo strut			
	4.	Con	vention aircraft usually con-	sist of	,		1	1	1
		(A)	Fuselage, landing gear plane	and tail	(B)	Landing gear, wings and tail plane			
		(C)	Fuselage, wings and landi	ng gear	(D)	Fuselage, wings and tail plane			
	5.	The	number of unknowns for a	beam who	se bo	th ends are fixed is	1	1	2
		(A)			(B)	4			
		(C)	6		(D)	2			
	6.		xed end beam of length 5 was the middle is given by		supp	ort moment is 6.25 kNm, carries a point	1	1	2
		(A)	0		(B)	31.25			
			62.5		(D)	10			
	7.	The	distribution factor for an ov	erhanding	bear	n is	1	12	
		(A)	3I/4L		(B)				
		(C)	I/L		(D)	0.5			

lote:

8.	Find	the horizontal deflection at C	due to P fo	or a frame shown below: Assume uniform	1	1	2
	EI.						
	A	Ł B					
	3			render and a substitution of the			
		ri signe in timber as					
		c					
		P					
	(A)	$2PL^3$	(B)	$PL^3$			
		3 <i>EI</i>		$\overline{3EI}$			
	(C)	$2PL^3$	(D)				
	(-)	$\frac{ZIL}{EI}$	(-)				
		EI		3EI			
0	Cala	ulata tha ategia anagay atagod	in a bade a	ef stance 0.04 N/mm² The same sentional	1	1	
9.			-	of stress 0.04 N/mm <sup>2</sup> . The cross sectional	•	•	٠
		is 100 m <sup>2</sup> and length of body is					
	(A)			4 Nmm			
	(C)	50 Nmm	(D)	200 Nmm			
10		1.5	•	1.70 (17 vtr)			
10.				having length 2 m, UDL of 1 kN/m over	1	1	-
		ntire span. Take $E = 2 \times 10^5 \text{ N/}$					
	(A)	75.125 mm	(B)		25		
	(C)	121.75 mm	(D)	65.125 mm			
		. 1			24		
11.	Find	the vertical force acting at the	joint as sho	own	1	1	3
		↑F <sub>PB</sub>					
	71.1	kN F					
	(4)	O1 1 INI	(70)	71 111			
	(A)	21.1 kN	(B)	71.1 kN			
	(C)	61.1 kN	(D)	0			
10	A In.		1 . C 1	at T at the control of the control	1	1	
12.				gth L, subjected to a tensile force of P.	1	1	•
				D and the remaining portion 2 having			
		eter 2D. The ratio of strain end					
	(A)		(B)				
	(C)	2	(D)	3			
4.0	****						
13.				find the buckling load in columns?	1	1	4
	(A)		beam (B)	Lagrangian formula for columns			
		columns					
	(C)	Euler formula	(D)	Buckling theorem			
				17			
14.	The	ratio of the effective length of	the column	n to the least radius of gyration is known	1	1	4
	as _	•					
	(A)	Poisson's ratio	(B)	Slenderness ratio			
	(C)	Stress ratio	(D)	Length ratio			
		3					
15.	Long	g columns have a slenderness r	atio		1	1	4
		Between 70 and 99	(B)				
	(C)	Between 51 and 69	, ,	More than 120			
	. ,		(-)	The state of the s			
16.	The	effective length of the colum	n whose or	ne end is fixed and the other end free is	1	1	4
				The same was the control of the field to			
	(A)	0.75 L	(B)	0.5 L			
	(C)	2 L	(D)	0.707 L			
	(-)		(D)	0.707			

17.	Which failure theory is used for brittle materials?	1	1	5		
	(A) Maximum principal strain theory (B) Maximum principal stress theory (C) Maximum shear stress theory (D) Maximum strain energy theory					
18.	The principal stresses are 120 MPa (tensile) and 80 MPa (tensile). According to maximum principal stress theory, find the maximum permissible stress (MPa).  (A) 40  (B) 80	1	1	5		
	(C) 200 (D) 120					
19.	The principal stresses are 200 MPa (tensile) and 150 MPa (tensile). Find the maximum shear stress (MPa)	1	1	5		
	(A) 350 (C) 25 (B) 185 (D) 200					
20.	The factor safety is given by, if the principal stresses are 300 MPa and 150 MPa, and stress at elastic limit is 350 MPa according to maximum principal stress theory.	1	1	5		
	(A) 1.285 (C) 1.167 (B) 2.333 (D) 0.777					
	(b) 0.77					
	$PART - B (5 \times 4 = 20 Marks)$	Marks	BL	CO		
	Answer ANY FIVE Questions	Maiks	DL			
21.	21. What are the factors influencing the structural failure of a unit?					
22.	Determine the carry over factor for a beam fixed at one end and simply supported at the other end.	4	2	2		
23.	Use strain energy method and find the deflection at C.	4	2	3		
	ω/Unit length					
	$\begin{array}{c} A & C \\ \downarrow L/4 & \downarrow C \\ \hline \end{array} \qquad L \longrightarrow \begin{array}{c} B \\ \downarrow L/4 & \downarrow C \\ \hline \end{array}$					
24.	How failure occurs in a short and long columns?	4	2	4		
25.	5. List down the Euler's formula for all end conditions of a column.					
26.	26. Explain the assumptions made in Euler's column theory.					
27.	27. Explain the importance of failure theories.					
	PART – C $(5 \times 12 = 60 \text{ Marks})$ Answer ALL Questions	Marks	BL	со		
28. a.	Explain a landing gear structure with a neat sketch.	12	2	1		
h	(OR) Explain in detail about loads on an aircraft structural components.	12	2	- 1		
U.						

29. a. A continuous beam ABC, where A, B and C are simply supported. The length for AB and BC are 5 m and 4 m respectively. Span AB carries a UDL of 2 kN/m over its length and span BC carries a point load of 4 kN at its middle. Find the support moments. Also plot shear force and bending moment diagrams.

3 12 3 2 i

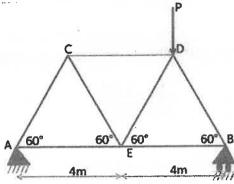
(OR)

b. A continuous beam ABC, where A is fixed, B and C are simply supported. The length for AB and BC are 6 m and 4 m respectively. Span AB carries a UDL of 3 kN/m over its length and span BC carries a UDL of 4 kN/m over its length. Draw shear force and bending moment diagrams.

12 3 2

30. a. Find the vertical deflection at D for a truss shown below. Take E = 200 GPa. Assume AE as constant and the cross sectional area of each member as 1000 sqmm.

12 3 3



(OR)

b. For a given frame below, find the forces in all the members. Assume all the members have same AE.

2 3

- 4 m 2.25 m 3.75 m
- 31. a. The external and internal diameter of a hollow cast iron column are 100 mm and 70 mm respectively. If the length of this column is 1.5 m and both ends hinged, determine the crippling load using Euler's and Rankine's formula. Take crushing stress as 500 MPa

2 3 4

- (OR)
- b. Determine the crippling load for a T-section of uniform thickness 20 mm, flange and web dimensions as 120 mm each, both ends are hinged. Take E =200 GPa.
- 12 3

32. a. Explain the five theories of failure in detail.

and Rankine's constant as 1/1600.

12 3 5

3

- (OR)
- b. Determine the diameter of a bolt which is subjected to an axial pull of 8 kN together with a transverse shear force of 4 kN using Maximum principal stress theory and Maximum principal strain theory. Given the elastic limit in tension as 220 MPa, factor of safety as 4 and Poisson's ratio as 0.28.

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