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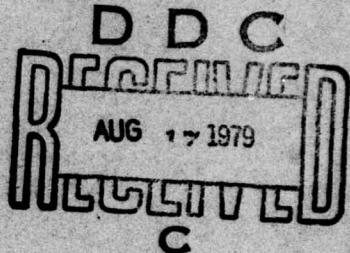
DESIGN AND ANALYSIS OF SINGLE CELL BALLOONS

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APPENDIX A

Balloon Design Program

BALDE

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LEVEL 21

MAIN

DATE = 79101

12/54/45

C - B A L D L -
C BALLOON DESIGN PROGRAM
C *****
C THIS PROGRAM CALCULATES THE SHAPE OF A FREE BALLOON GIVEN A SET OF
C INPUT CONDITIONS. IT IS FOR FULLY TAILORED BALLOONS ONLY. WITH OR
C WITHOUT AN ENDCAP. CONVERGENCE OF THE SOLUTION IS OBTAINED BY
C MATCHING THE TOP LOAD REACTION TO THE TOP LOAD. THIS PROGRAM
C INCLUDES THE EFFECTS OF THE LOAD TAPES IN THE CALCULATIONS.
C *****
C THIS PROGRAM IS BASED ON THE PROGRAMS "BALLOON" AND "FLATBALL".
C AFCLR-05-92. "DETERMINATION OF THE SHAPE OF A FREE BALLOON", BY
C J. H. SMALLEY.
C *****
C DESCRIPTION OF DATA DECK:
C * EACH SET OF DATA CONSISTS OF THREE CARDS LISTING THE VALUES
C IN THE READ STATEMENTS 500 AND 501
C ALL DATA ARE READ IN 10 WIDE FIELDS OF THE APPROPRIATE TYPE.
1 "FIRST" CARD INPUTS ARE AS FOLLOWS:
PAYLOAD IN POUNDS
ALTITUDE OPTION: 1 = ALT IN FT, 2 = ALT IN MB
ALTITUDE IN FEET OR MILLIBARS AS DESIRED
FILM TYPE: 1 = POLYETHYLENE, 2 = MYLAR
FILM THICKNESS IN INCHES
LOAD TAPE TYPE: 1 = POLYESTER, 2 = KEVLAR
TAPE LOAD RATING IN POUNDS
NUMBER OF LOAD TAPES
2 "SECOND" CARD INPUTS ARE AS FOLLOWS:
TOP LOAD IN POUNDS, (+) UP, (-) DOWN
STRESS CONSTANT TAU0 (USUALLY 0.00)
STRESS CONSTANT TAU1 (USUALLY 0.00)
SUPERPRESSURE (0.00 FOR NATURAL SHAPE)
PRINT INCREMENT N (0 FOR STANDARD OUTPUT)
NON-DIMENSIONAL GORE INCREMENT DSO
NON-DIMENSIONAL GORE LENGTH TO CAP STARTING LOCATION CSTART
FILM THICKNESS INCLUDING CAP IN INCHES
3 "THIRD" CARD INPUTS ARE AS FOLLOWS:
OUTPUT CONTROL (KEY2=2 FOR PUNCHED DECK OF SHAPE & WEIGHT
KEY2=1 FOR DISK FILE OF SHAPE & WEIGHT)
OUTPUT CONTROL (MPT IS NUMBER OF POINTS IN LOAD-ALTITUDE)
IDENTIFY LIFTING GAS (FOR HELIUM, IGAS=1)
MINIMUM RECOMMENDED PAYLOAD IN POUNDS
MAXIMUM RECOMMENDED PAYLOAD IN POUNDS
* IF NO ENDCAP, MAKE SURE THAT CSTART IS GREATER THAN ANY
ANTICIPATED GORE LENGTH.
* THE PROGRAM REQUIRES A LAST DATA SET WITH P = 0.0 TO
TERMINATE

C DESCRIPTION OF OUTPUT:
FOR EACH CASE THE INPUT DATA WILL FIRST BE PRINTED, FOLLOWED
BY A RECORD OF THE ITERATIONS REQUIRED FOR CONVERGENCE OF THE
SOLUTION. THIS RECORD CONSISTS OF A DISPLAY OF INITIAL AND
FINAL ANGLES OF THE GORE WRT THE VERTICAL AXIS OF THE BALLOON.
DURING THE FINAL ITERATION, PERTINENT VALUES ALONG THE GORE ARE
PRINTED OUT CONTINUOUSLY, FOLLOWED BY A LISTING OF FINAL VALUES.

C APPROPRIATE ERROR MESSAGES ARE SUPPLIED IF THE SOLUTION DOES NOT
CONVERGE

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***** MAIN PROGRAM *****
REAL K,KT,KTAPE,LAMDA,IAM
INTEGER CCDEF,CCDET
DIMENSION GP(100),GW(100),GWT(100),GT(100)
100 WRITE(6,600)
600 FORMAT('1')
READ(5,500) P,KEY,CCNST,CCDEF,FTHICK,CCDET,TLR,NT
500 FORMAT(F10.0,I10,F10.0,I10,F10.4,I10,F10.0,I10)
READ(5,501) TL,TAU0,TAU1,ALPHA,N,DS0,CSTART,CAP
501 FCRMAT(F10.2,F10.2,F10.2,F10.2,I10,F10.2,F10.2,F10.4)
C IF NC MCFE DATA, EXIT.
C IF(P) 200,300,200
C-----200 CONTINUE
READ(5,502)KEY2,MFT,IGAS,PMIN,PMAX
502 FCRMAT(3I10,2F10.0)
CALL DESIGN (P,KEY,CCNST,CCDEF,FTHICK,CCDET,TLR,NT,TL,TAU0,TAU1,
$ALPHA,N,DS0,CSTART,CAP,GP,GW,WBAL,KEY2,S)
C C COMPUTE LOAD - ALTITUDE CURVE
WRITE(6,600)
WRITE(6,601)
601 FORMAT(50X,'LOAD - ALTITUDE DATA',/42X,'GRUSS AIRBORN',/37X,')
1 WEIGHT -(KG)',10X,'ALTITUDE -(KM)')
GPMIN=1.+WBAL
GPMAX=GPMIN+(PMAX-PMIN)/P
DG=(GPMAX-GPMIN)/MFT
GP1=GPMIN
250 CONTINUE
CALL GLNGTH(GP1,SLMDA,1)
E1=P/(S/SLMDA)**3/.06243
CALL ECYNCY(E1,PA,IGAS,TCHIGH)
CALL ATMOS2(PA,H,TA,RHCA,B,IAM,GXPAN,TOHIGH)
GRCS=GP1*F*.454
H=H/1000.
WRITE(6,602)GRCS,H
602 FCRMAT(42X,E15.7,8X,E15.7)
GP1=GP1+DG
IF(GP1.GT.GPMAX)GO TO 100
GC TC 250
300 CONTINUE
STCH
END

```

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SUBROUTINE DESIGN (P,KEY,CCNST,CCDEF,FTHICK,CODET,TLR,NT,TL,TAU0,  
STAU1,ALPHA,N,DSC,CSTART,CAP,GP,GW,WBAL,KEY2,S)  
*****  
***** SUBROUTINE DESIGN *****  
*****  
REAL K,KT,KTAPE,NU  
REAL LAMBDA  
INTEGER CCDEF,CODET  
DIMENICK Y(6),G(6),E(4),C(4),D(4),WC RD1(3),WORD2(3)  
DIMENSION GP(100),GW(100),GWT(100),GT(100)  
DIMENSION SSTORE(300),GSTCRE(300),WSTORE(300),TSTORE(300)  
REAL F1(3)/'PCLY','ETHY','LENE'/  
REAL F2(3)/'MYLA','R ',' '/  
REAL T1(3)/'FCLY','ESTE','R '/  
REAL T2(3)/'KEVL','AR ',' '/  
REAL K1/' FT'/  
REAL K2/' MB'/  
C  
C B, C, AND D ARE CONSTANTS USED IN GILL'S MODIFIED R-K METHOD FOR  
C SOLVING THE GOVERNING DIFFERENTIAL EQUATIONS.  
DATA B/2.0,1.0,1.0,2.0/, C/0.5,0.29289,1.7071,0.5/, D/0.5,  
$0.29289,1.7071,0.16667/  
C  
CALL MATL(P,KEY,CONST,CCDEF,FTHICK,CODET,TLR,NT,CAP,SIGMA,CSIGMA,  
$TSIGMA,KT,EM,EC,ENC,LAMBDA,TSIGMA)  
NU=EMC/EM  
OMEGA1=SIGMA/1.8453  
OMEGA2=OMEGA1*(CSIGMA/SIGMA)  
OMEGA=OMEGA1+TSIGMA/1.8453  
TCNEGA=TSIGMA/6.2832  
C  
C EMPIRICAL CURVE FITS FROM SMALLEY'S REPORT. TO GET GOOD FIRST ESTIMAT  
C OF THETA0. GOOD FOR SIGMA BETWEEN 0.0 AND 0.8 .  
THETA0=1.347  
IF(CMEGA.LT.0.43354) THETA0=0.72*OMEGA+1.1015  
IF(CMEGA.LT.0.35225) THETA0=1.15*OMEGA+0.9507  
IF(CMEGA.LT.0.27096) THETA0=1.46*OMEGA+0.8687  
CNEGA=OMEGA1  
C  
IQUIT = 0  
LSTRUN=1  
IF(CODEF.EQ.2) GC TC 52  
DC 51 I=1,3  
51 WC RD1(I)=F1(I)  
GO TU 54  
52 DC 53 I=1,3  
53 WORD1(I)=F2(I)  
54 IF(CODET.EQ.2) GU TU 56  
DC 55 I=1,3  
55 WORD2(I)=T1(I)  
GC TC 58  
56 DC 57 I=1,3  
57 WORD2(I)=T2(I)  
58 IF(KEY.EC.2) GO TU 59  
WORD3=K1  
GO TL 60  
59 WC RD3=K2  
60 CONTINUE
```

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      WRITE(6,601) (WCRD1(I),I=1,3),FTHICK,SIGMA,TAU0,(WCRD2(I),I=1,3),
     $EM,TAU1,TLR,EC,ALPHA
601 FORMAT(5X,37('*')/5X,'*',35X,'*',15X,'NON-DIMENSIONAL QUANTITIES',
18X,'MISC. PARAMETERS'/5X,'*' FILM TYPE',8X,'':',A4,A4,A4,'*',5X,
2'* FILM THICKNESS : ',F7.4,' IN',4X,'*',17X,'SIGMA FILM =',F9.4
3*12X,'TAUC =',F6.3/5X,'*' TAPE TYPE',8X,':',A4,A4,A4,'*',17X,
4*EM',5X,' =',F6.3,F8.3,13X,'TAU1 =',F6.3/5X,'*' TAPE LOAD RATING : ',
5F6.0,' LBS *',17X,'EC',9X,' =',F8.3,13X,'ALPHA =',F6.3)
      WRITE(6,602) NT,ENC,DSC,F,TSIGMA,CSTART,TL,KT,N,CONST,WCRD3
602 FORMAT(' ',4X,'*' NUMBER OF TAPES : ',15X,'*',17X,'EMC',8X,
1'*',F6.3,13X,'DS0 =',F6.3/5X,'*' PAYLOAD',10X,':',F6.0,' LBS
2*',17X,'SIGMA TAPE =',F9.4,12X,'CSTART =',F5.2/5X,'*' (INCL TOPLOAD
3D CF) : ',F6.0,' LBS *',17X,'KT',9X,' =',F6.1,15X,'N =',I2/5X
4,* DESIGN ALTITUDE : ',F7.0,A4,'*',5X,'*',35X,'*',5X,37('*')
5,///5X, 'ITERATION RECORD (INITIAL AND FINAL ANGLES) //')

-----  

C IF THE SOLUTION IS NOT CONVERGING, EXIT
21 IQUIT = IQUIT+1
IF(IQUIT.LT.20) GO TO 22
      WRITE(6,603)
603 FORMAT(' -', '*' *** SOLUTION IS NOT CONVERGING - MAXIMUM NUMBER OF IT
SERATIONS HAS BEEN EXCEEDED *** ')
GU TO 5
C-----  

C LSTRUN = 0 IS THE KEY FOR LAST ITERATION. IT IS AN OUTPUT CONTROL.
22 IF(LSTRUN.NE.0)GU TO 23
C-----  

      WRITE(6,604)
604 FORMAT('0',95X,27*MANUFACTURER'S TABLE LAYOUT//10X,'S',13X,'R',
113X,'Z',13X,'T',11X,'TAUM',9X,'TAUC',8X,'GCRE POSITION (FT)',5X,
2*1/2*L GORE WIDTH (IN)'//)
23 DC 24 I=1,0
24 Q(I)=0.
231 DU 241 I=1,300
SETORE(I)=0.C
WSTORE(I)=0.
TSTORE(I)=0.
241 GSTORE(I)=0.C
JCOUNT=1
DS = DSO
SSTOP = CSTART
OMEGA=CMEGA1
TCMEGA=TSIGMA/6.2832
TW=0.
ITAG=0
RC=0.
GW2=0.
SO=0.
SOUIM=0.
R=0.
S=0.
Z=0.
WSUM=0.
ASUM=0.
TAUM=0.
TAUC=TAU0
THETA=THE TAC
RT0=1 ./6.2832/COS(THEETA)
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Y(1)=THETA
Y(2)=R
Y(3)=Z+ALPHA
Y(4)=((F+TL)/P)/6.2832/COS(THETA)
Y(5)=C.C
Y(6)=C.C
SSTOFE(JCCUNT) = SODIM
GSTOFE(JCCUNT) = GW2
2 T=6.2832*Y(4)
IF(LSTRUN)34,32,34
32 WRITE(6,605) S,R,Z,T,TAUM,TAUC,SODIM,GW2
605 FORMAT(4(4X,F10.5),4X,F8.3,5X,F8.3,12X,F7.1,17X,F7.2)
IF(KEY2.EQ.1)WRITE(3,650)Y(2),Y(3),Y(4),Y(1),Y(6)
IF(KEY2.EQ.2)WRITE(7,650)Y(2),Y(3),Y(4),Y(1),Y(6)
650 FORMAT(5E15.7)

C-----  

C I COUNT AND N ARE OUTPUT CONTROLS USED DURING THE LAST ITERATION.  

C N IS THE NUMBER OF INCREMENTS THAT WILL BE SKIPPED IN THE PRINTOUT.  

C-----  

C 34 I COUNT=N  

C-----  

C ITAG SENDS THE PROGRAM INTO ITERATIONS AND RELEASES IT WHEN THE TOP  

C OF THE BALLOON HAS BEEN REACHED  

IF(ITAG-1)4,3,4  

C-----  

3 IF(LSTRUN.EQ.0)GO TO 35
DEGO = THETA0*57.2956
THETAD = THETA0*57.2956
DS = DSC
WRITE(6,611) DEGO,THETAD
611 FORMAT(' ',5X,F10.5,5X,F10.5)
C-----  

C CONVERGENCE TEST: TOP LOAD REACTION WITHIN .5 LB OF APPLIED TOP LOAD.
F = -6.2832*Y(4)*COS(THETA)*P
IF(ABS(F+TL).LE.0.5) LSTRUN=0
C-----  

C COMPUTE CORRECTIVE TERM ON THETA0
DELTA(THETA0)/DELTA(THETA) IS GENERALLY APPROX.= (+1/-2)
DTETA IS THE DESIRED VALUE OF THETA TO MATCH THE TOP LOAD REACTION
TO THE APPLIED TCP LOAD.
DTETA = -(AFCUS(TL/(6.2832*P*Y(4))))  

DELTA = DTETA - THETA
CORR = -DELTA/2.
THETA0 = THETA0+CORR
C-----  

C GO TO 21
C-----  

C-----  

C COMPUTE AND PRINT FINAL QUANTITIES
35 VOL = 3.1416*Y(6)*(LAMEDA**3)
ASUM = (ASUM+6.2832*Y(5))*(LAMHDA**2)
WSUM = (WSUM+6.2832*UMEGA*Y(5))*P
TW=TW*6.2832*P
WEIGHT = WSUM+TW
F = -6.2832*Y(4)*COS(THETA)*P
THETAD=57.2956*THETA
DEGO = THETA0*57.2956
JMAX = JCCUNT

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PTCF=-TL/F
WBAL=WEIGHT/P
WRITE(6,606) LAMBDA,VOL,DEG0,ASUM,THETAD,WSUM,F,TW,WEIGHT
606 FFORMAT(' ',4X,'FINAL QUANTITIES (LAMDA =',F7.1,' FT):',//5X,'VOL UM
1E ',6X,'=',E10.3,' FT3',10X,'INITIAL ANGLE',5X,'=',F7.3,' DEG'/SX,
2'AREA',8X,'=',E10.3,' FT2',10X,'FINAL ANGLE',7X,'=',F7.3,' DEG'
3/SX,'FILM WEIGHT =',F7.0,' LBS',10X,'TCF LOAD REACTION =',F7.3,
4' LBS'/EX,'TAPE WEIGHT =',F7.0,' LBS'/5X,'TOT. WEIGHT =',F7.0,
5' LBS'/)
IF(KEY2.EQ.0.1)WRITE(3,650)F,PTOP,WEAL,S,LAMBDA
IF(KEY2.EQ.2)WRITE(7,650)P,PTOP,WBAL,S,LAMBDA
IF(KLY2.EQ.1)WRITE(3,651)NU,KT,EM,NT
IF(KLY2.EQ.2)WRITE(7,651)NU,KT,EM,NT
651 FFORMAT(3E15.7,I15)

C GENERATE TRANSFER MATRICES ... 100 EVENLY SPACED, UNDEFORMED
C GRID COORDINATES
C
DSNEW = SSTORE(JMAX)/SS.
J=1
SNEW=C.C
GP(J)=0.0
GW(J)=0.0
GWT(J)=TCMEGA
GT(J)=TCMEGA*6.2832/NT
WRITE(6,600) J,GP(J),J,GW(J),J,GWT(J),J,GT(J)
100 SNEW = SNEW+DSNEW
J = J+1
DO 101 I=1,JMAX
QUAN = SNEW-SSTORE(I)
IF(QUAN.LT.0.0) GC TC 102
101 CCONTINUE
102 GP(J) = SNEW
GW(J) = GSTCRE(I-1)+((SNEW-SSTCRE(I-1))*(GSTCRE(I)-GSTORE(I-1))/
$(SSTORE(I)-SSTORE(I-1)))
GWT(J) = WSTCRE(I-1)+((SNEW-SSTURE(I-1))*(WSTORE(I)-WSTORE(I-1))/
$(SSTORE(I)-SSTORE(I-1)))
GT(J) = TSTORE(I-1)+((SNEW-SSTURE(I-1))*(TSTORE(I)-TSTORE(I-1))/
$(SSTORE(I)-SSTCRE(I-1)))
WRITE(6,600) J,GP(J),J,GW(J),J,GWT(J),J,GT(J)
600 FORMAT(' ',5X,'GP(',I3,') = ',F10.3,10X,'GW(',I3,') = ',F10.3,10X,
1'GWT(',I3,') = ',F10.3,10X,'GT(',I3,') = ',F10.3)
652 FFORMAT(4E15.7)
IF(J.EQ.100) GC TC 103
GO TO 100
103 CCONTINUE
DC 300 J=1,JMAX
IF(KEY2.EQ.1)WRITE(3,652)SSTORE(J),GSTORL(J),WSTORE(J),TSTORE(J)
IF(KEY2.EQ.2)WRITE(7,652)SSTORE(J),GSTORE(J),WSTORE(J),TSTORE(J)
300 CCONTINUE
C GC TC 5
C-----COMPUTATIONAL PORTION-----
C-----CHECK IF CVER THE SHOULDER OF THE BALLOON
4 IF(SIN(THETA))6,8,8

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IF SO, CHECK IF WITHIN DS OF THE TOP
 6 IF(R-DS)7,7,8

IF SO, REDEFINE DS, KEY WITH ITAG
 7 DS=R/ABS(SIN(THETA))
 ITAG=1

 PROGRAM SOLVING STARTS HERE

C SET UP FOR GILL'S METHOD

8 DC 10 J=1,4
 G(2)=SIN(Y(1))
 G(3)=COS(Y(1))
 G(1)=(TAUC*G(3)-Y(2)*Y(3)-G(2)*Y(2)*CMEGA-TOMEGA*G(2))/Y(4)
 G(4)=TAUC*G(2)+G(3)*(OMEGA*Y(2)+TOMEGA)
 G(5)=Y(2)
 G(6)=Y(2)*Y(2)*G(3)

C GILL'S METHOD

DO 9 I=1,6
 P1 = D(J)*(G(I)-B(J)*Q(I))
 Y(I) = Y(I)+DS*F1
 9 Q(I) = Q(I)+3.0*P1-C(J)*G(I)

C 10 CONTINUE

C-----

TW=TW+TCMEGA*DS

THE TA=Y(1)

R=Y(2)

Z=Y(3)-ALPHA

S=S+DS

JCOUNT = JCOUNT +1

IF(R) 161,161,16

161 R=1.E-10

16 TAUC=TAU0+(TAU1*(Y(4)-RT0))/R
 EPSM=(Y(4)/R-EMC*TAUC/EC)/(EM-EMC*EMC/EC+KT/R/6.2832)
 EPST=EPSM
 TAUN=(Y(4)-KT*EPST/6.2832)/R
 R0=R/(1.+(TAUC-(EMC*EPSM)/EC))
 GW2=3.1416*R0*LAMBDA*12./NT
 SJ=S0+(DS/(1.+EPSM))
 SODIM=S0*LAMBDA
 SSTOPE(JCOUNT) = SODIM
 GSTORE(JCOUNT) = GW2
 WSTORE(JCOUNT)=Y(2)*GMEGA+TOMEGA
 TSTORE(JCOUNT)=TOMEGA*6.2832/NT

C-----

C IF TAUC < 0, BUCKLING OCCURS, SO SET VALUES TO 0

C-----

IF(TAUC)31,17,17

31 TAUC=0.

TAU0=C.0

TAU1=0.0

C-----

C CHECKS FOR IMPOSSIBLE BALLOONS.

C IF |THETA|>PI OR S>10.0, PRINT AN APPROPRIATE ERROR MESSAGE AND

C PROCEED TO THE NEXT SET OF DATA

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```
17 IF((ABS(THE TA)-3.1416).LT.0.0) GC TC 12
  WRITE(6,607)
607 FORMAT('---',*** SOLUTION IS NOT CONVERGING - IMPOSSIBLE BALLOON :
$ |THE TA| > PI *** ')
  GC TC 5
12 IF((S-10.0).LT.0.0) GC TC 30
  WRITE(6,608)
608 FORMAT('---',*** SOLUTION IS NOT CONVERGING - IMPOSSIBLE BALLOON :
$ NON-DIM S > 10.0 *** ')
  GO TO 5
-----
C THIS PORTION ALLOWS FOR THE CHANGE IN FILM THICKNESS DUE TO THE CAP
C
30 IF(S-SSTCF)14,13,13
13 T=6.2E32*Y(4)
131 ASUM=ASUM+6.2832*Y(5)
  WSUM=WSUM+6.2832*CMEGA*Y(5)
  Y(5)=0.
  IF(LSTRUN.NE.0) GC TC 132
609 FORMAT('0',3X,'CAP STARTS AT S =',F6.2, ' ... SIGMA CAP =',F7.4/)
  WRITE(6,609) CSTART,CSIGMA
132 UMEGA = CMEGA2
  TOMECA=TSIGMB/6.2E32
  SSTOP = 10.0
14 IF(ICOUNT)2,2,15
15 ICOUNT=ICOUNT-1
  IF(1TAG)2,4,2
5 CONTINUE
  S=S*LAMBD
  RETURN
END
```

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MATL

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SUBROUTINE MATL(P,KEY,CONST,CODEF,FTHICK,CGDET,TLR,NT,CAP,SIGMA,
\$CSIGMA,TSIGMA,KT,EM,EC,EMC,LAMECA,TSIGMB)

***** THIS SUBROUTINE CALCULATES MATERIAL PROPERTIES GIVEN INFORMATION
AS FOLLOWS:

INPUTS: PAYLOAD, ALTITUDE OPTION KEY, ALTITUDE, FILM TYPE AND
THICKNESS, TAPE TYPE AND LOAD RATING, NUMBER OF TAPES, AND
CAP THICKNESS

OUTPUTS: NON-DIMENSIONAL FILM AND TAPE WEIGHTS, TAPE STIFF-
NESS, AND FILM MODULI.

***** ALL EMPIRICAL RELATIONSHIPS USED ARE SUBJECT TO REVISION WHEN
ADDITIONAL TEST DATA BECOME AVAILABLE.

***** THIS SUBROUTINE ASSUMES, FOR LACK OF A GOOD TEMPERATURE MODEL, THAT
THE BALLOON MATERIAL TEMPERATURE IS ATMOSPHERIC TEMPERATURE.
THIS ASSUMPTION MAY NOT BE ADEQUATE.

***** REAL K,KT,KTAPE,LAMBDA
REAL NG,NA,LF,IAM
INTEGER CODEF,CGDET

100 SERIES - DEFINE NECESSARY VALUES

IF(KEY.EQ.2) GO TO 101
100 H=CONST

H=F*0.3048
CONVERSION OF H FROM FEET TO METERS FOR SUERT ATMOS

CALL ATMC(S(H,TA,PA,RHCA,E,IAM,GXPAN,TOHIGH)
GO TO 102

101 PA=CONST
CALL ATMC(S(FA,H,TA,RHCA,E,IAM,GXPAN,TOHIGH)

102 B=B*6.243E-02
CONVERSION OF B FROM KG/M3 TO LB/M/FT3

T=TA-273.16

CONVERSION OF TEMPERATURE FROM DEG(R) TO DEG(C)

LAMBDA = (P/B)**(1./3.)
K = 1. / ((6.283185)**(1./3.))

200 SERIES - COMPUTE FILM PROPERTIES

IF(CLDEF,EQ.2) GO TO 202
201 WFILM = 58.06*(FTHICK/12.)
EFILM = (0.02*(T**2)-1.60*(T)+41.95)*1000.
GO TO 203

202 WFILM = 87.06*(FTHICK/12.)

EFILM = ????? NYLAR FILM DATA IS NECESSARY

203 SIGMA = WFILM/(K*B*LAMBDA)
CSIGMA=SIGMA*(CAF/FTHICK)

THIS PROGRAM ASSUMES A POISSON'S RATIO OF 0.50

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EM = (EFILM*FTHICK*12.)/(B*(LAMBCA**2))
EC = EM
EMC = EM/2.

300 SERIES - CCMFLTE TAPE PROPERTIES

IF(CODET.EQ.2) GC TC 302
301 WTAPE = (6.1E-06)*TLR+C.CC38+.3024*(.003+FTHICK)
WTAPE2=WTAPE+.3024*(CAF-FTHICK)
KTAPE = (TLR*10.)*(1.-(6.67E-03)*(T-20.))
GO TO 303
302 WTAPE = (2.0E-06)*TLR+0.0021
KTAPE = 26.*TLR

NC INFORMATION ON LCW TEMPERATURE EFFECTS CURRENTLY AVAILABLE

303 TSIGMA = (NT*WTAPE)/(E*(LAMBDA**2))
TSIGMB=TSIGMA*WTAPE2/WTAPE
KT = (NT*KTAPE)/(B*(LAMBDA**3))
RETURN
END

LEVEL 21

ATMOS

DATE = 78226

21/29/42

SUBROUTINE ATMOS(H,TA,PA,RHOA,B,IAM,GXPAN,TOHIGH)

C SUBROUTINE FOR SOLVING FOR THE VALUES OF TEMPERATURE,
C PRESSURE, DENSITY, SPECIFIC BUOYANCY, INTEGRATED AIR
C MASS, AND "GAS EXPANSION" FOR ANY GIVEN ALTITUDE BELOW
C 61000. METERS (200131. FEET). ALL EQUATIONS HAVE BEEN
C DERIVED ACCORDING TO THE U.S. STANDARD ATMOSPHERE, 1962.
C --- ALL UNITS ARE IN THE SI SYSTEM. I.E.
C TEMPERATURE(DEG KELVIN), PRESSURE(MB), DENSITY(KG/M3),
C SPECIFIC BUOYANCY(KG/M3) INTEGRATED AIR MASS(KG/M2),
C GAS EXPANSION(DIMENSIONLESS).--- LIFTS WERE BASED ON
C GRADE A HELIUM AND PURE HYDROGEN.

C FOR SPECIFIC BUOYANCY, TO CONVERT FROM KG/M3 TO LB/FT3,
C MULTIPLY BY 0.06243

C H IS INPUT IN METERS

C
REAL MG,MA,LP,IAM
TOHIGH=1
MC=4.0026
MA=28.9644
R=8.31432E03
GY=9.80665
RHOS1F=1.2250
IGAS=1
IF HYDROGEN GAS IS USED INSTEAD OF HELIUM,IGAS MUST BE CHANGED TO
C SOMETHING OTHER THAN 1
C
IF(H.LT.61000.0) GO TO 10
WRITE(6,155) H
155 FORMAT('0',5X,'***** ALTITUDE H OUTSIDE RANGE OF SUBROUTINE ATMOS.'
1 'H =',E14.7,' METERS. *****')
TOHIGH=-1.
GO TO 28
10 IF (H.LT.52000.0) GO TO 12
HB=52000.C
TE=270.65
LP=-0.002C
PB=0.590
RHOB=0.0007594
GO TO 22
12 IF(H.LT.47000.0) GO TO 14
HB=47000.C
TB=270.65
LP=0.0
PB=1.105
RHOB=0.0014275
GO TO 22
14 IF(H.LT.32000.0) GO TO 16
HE=32000.0
TB=228.65
LP=0.0028
PB=8.680
RHOB=0.013225
GO TO 22
16 IF(H.LT. 20000.0)GC TC 18
HB=20000.C

LEVEL 21

ATMOS

DATE = 78226

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```
TB=216.65
LP=0.0010
PB=54.749
RHOB=C.088035
GO TC 22
18 IF (H.LT.11000.0)GO TC 20
HB=11000.0
TB=216.65
LP=0.0
PB=226.320
RHOB=C.36392
GO TO 22
20 HB=0.0
TB=288.15
LP=-0.0065
PB=1013.250
RHOB=1.2250
22 TA=TA+LP*(H-HB)
IF(LP.EQ.0)GC TC 24
PA=PB/(LP*(H-HB)/TB+1.0)**(GY*MA/LP/R)
GO TC 26
24 PA=PB/EXP(GY*MA/R/TB*(H-HB))
26 RHUA= PA*MA/R/TA*100.0
IF(IGAS.NE.1) MG=2.0159
B=RHUA*(1.0-MG/MA)
IAM=10.0*PA*(1.03751-0.00527*ALUG10(PA))
GXPA=RHOSTP/RHOA
28 RETURN
END
```

LEVEL 21

ATMUS2

DATE = 78226

21/29/42

SUBROUTINE ATMCS2(PA,H,TA,RHOA,B, IAM,GXFAN,TCHIGH)

C SUBROUTINE FOR SOLVING FOR THE VALUES OF TEMPERATURE,
C ALTITUDE, DENSITY, SPECIFIC BUOYANCY, INTEGRATED AIR
C MASS, AND "GAS EXPANSION" FOR ANY GIVEN ALTITUDE BELOW
C 61000. METERS (200131. FEET). ALL EQUATIONS HAVE BEEN
C DERIVED ACCORDING TO THE U.S. STANDARD ATMOSPHERE, 1962.
C --- ALL UNITS ARE IN THE SI SYSTEM. I.E.
C TEMPERATURE(DEG KELVIN), PRESSURE(MB), DENSITY(KG/M3),
C SPECIFIC BOUYANCY(KG/M3) INTEGRATED AIR MASS(KG/M2).
C GAS EXPANSION(DIMENSIONLESS).--- LIFTS WERE BASED ON
C GRADE A HELIUM AND PURE HYDROGEN.

C FOR SPECIFIC BOUYANCY, TO CONVERT FROM KG/M3 TO LB/FT3,
C MULTIPLY BY 0.06243

C PA IS INPUT IN MB

REAL MG,MA,LP,IAM
TCHIGH=1
MG=4.0026
MA=28.9664
R=8.31432E 03
GY=9.80665
RHOSTF=1.2250
IGAS=1
IF(PA.LT.226.32) GO TO 10
PB=1013.25
TB=0.0
TB=288.15
LP=-0.0065
RHCB=1.225
GO TO 22
10 IF(PA.LT.54.749) GO TO 12
PB=226.32
HE=11000.
TB=216.65
LF=0.0
RHOB=0.36392
GO TO 22
12 IF(PA.LT.8.680) GO TO 14
PB=54.749
HE=20000.
TB=216.65
LP=0.001
RHCB=0.088035
GO TO 22
14 IF(PA.LT.1.109) GO TO 16
PB=8.680
HE=32000.
TB=228.65
LP=0.0028
RHOB=0.013225
GO TO 22
16 IF(PA.LT.0.590) GO TO 18
PB=1.109
HE=47000.
TB=270.65

LEVEL 21

ATMOS2

DATE = 78226

21/29/42

```
LP=0.0
RHOB=0.0014275
GO TO 22
18 IF(PA.LT.0.1828) GO TO 20
PB=0.59
HB=52000.
TB=270.65
LP=-0.002
RHOB=0.0007594
GU TC 22
20 WRITE(6,155) PA
WRITE(6,155) PA
155 FORMAT('0',5X,'***** ATMOSPHERIC PRESSURE PA OUTSIDE RANGE OF SUBR
I0UTINE ATMOS2. PA =',E14.7,' MH. *****')
TCHIGH=-1.
GO TO 28
22 IF(LP.EQ.0.0) GO TO 24
H=HB+TB/LF*((FB/FA)**(LP*R/GY/MA)-1.0)
GO TO 26
24 H=ALCG(PB/PA)*R*TE/GY/MA+HB
26 TA=TB+LF*(H-HB)
RHOA=PA*MA/R/TA*100.0
IF(IGAS.NE.1) MG=2.0159
B=RHOA*(1.0-MG/MA)
IAM=10.0*PA*(1.03751-0.00527* ALOG10(PA))
GXFA=RHOSTP/RHCA
28 RETURN
END
```

LEVEL 21

BOUNCY

DATE = 78226

21/29/42

SUBROUTINE BOUNCY(B,PA,IGAS,TCHIGH)

C
C..... SUBROUTINE TO SOLVE FOR ATMOSPHERIC PRESSURE (PA) GIVEN THE
C..... CORRESPONDING GAS BOUYANCY (B) AT ANY ALTITUDE BELOW 61,000
C..... METERS (200,131 FEET). ALL EQUATIONS HAVE BEEN DERIVED ACCORDING
C..... TO THE U.S. STANDARD ATMOSPHERE, 1962.
C..... ---- ALL UNITS ARE IN THE SI SYSTEM:
C
C..... SPECIFIC BOUYANCY (B) IS THE INPUT IN KG/M³
C..... ATMOSPHERIC PRESSURE (PA) IS THE OUTPUT IN MB
C
C..... CALCULATIONS ARE BASED ON PURE HYDROGEN AND GRADE A HELIUM.
C..... FOR HYDROGEN, INPUT IGAS AS 0.
C..... FOR HELIUM, INPUT IGAS AS 1.
C
C..... TCHIGH WILL BE SET TO -1 IF BOUYANCY IS OUTSIDE THE PARAMETERS
C..... OF THE STANDARD ATMOSPHERE.
C

REAL MW
TCHIGH=1
MW=4.0026
B1=1.0557
B2=0.31355
B3=0.075858
B4=0.011396
B5=1.2308E-3
B6=6.5444E-4
B7=2.3635E-4
IF(IGAS.EQ.1)GO TO 90
MW=2.0159
B1=1.1397
B2=0.33853
B3=0.081899
B4=0.012303
B5=1.3288E-3
B6=7.0656E-4
B7=2.517E-4
90 IF(B.LT.B1)GO TO 1
WRITE(6,101) B
101 FORMAT('0','*****', ' VALUE OF B IS OUTSIDE LIMITS OF STANDARD AT
1MOSPHERE. B = ',E12.5,' (KG/M³) *****')
TCHIGH=-1.
GO TO 40
1 IF(B.LE.B2)GO TO 2
TB=28E.15
XLP=-0.0065
PE=1013.250
GO TO 9
2 IF(B.LE.B3)GO TO 3
TB=216.65
XLP=0.0
PE=226.320
GO TO 9
3 IF(B.LE.B4)GO TO 4
TB=216.65

LEVEL 21

BOUNCY

DATE = 78226

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```
XLP=0.0010
PB=54.749
GO TO 9
4 IF(B.LE.B5)GO TO 5
TB=22E.65
XLP=0.0028
PB=8.680
GO TO 9
5 IF(B.LE.B6)GO TO 6
TB=27C.65
XLP=0.0
PB=1.109
GO TO 9
6 IF(B.LE.B7)GO TO 7
TB=27C.65
XLP=-C.0020
PB=0.590
GO TO 9
7 WRITE(6,101) B
TCHIGH=-1.
GO TO 40
9 ITTER=0
PNEW=PB
10 POLD=PNEW
ITTER=ITTER+1
FPA=POLD-(2.870531*B*TB/(1.-MW/28.9644))*(1.+((PB/POLD)**(29.2713*
1XLP)-1.))
GPA=1.+((84.0241*B*TE*XLP*FE/((1.-MW/28.9644)*POLD**2))*(PB/POLD)***
1(29.2713*XLP-1.))
PNEW=FCLD-FPA/GPA
IF(ABS((FNEW-FCLD)/FNEW).LE..00001)GO TO 20
IF(ITTER.GE.5C)GO TO 30
GO TO 10
30 DELPA=PNEW-PCLD
WRITE(6,100) DELPA
100 FORMAT('0',10X,'SCLUTION DID NOT CONVERGE AFTER 50 ITTERATIONS.'
1/5X,'DIFFERENCE BETWEEN NEW AND OLD PRESSURE ON LAST ITTERATION WA
2S',2X,E10.3)
TCHIGH=-1.
20 PA=PNEW
40 RETURN
END
```

LEVEL 21

GLNGTH

DATE = 78226

21/29/42

SUBROUTINE GLNGTH(GP,SLMDA,INDIC)

C
C SUBROUTINE TC SOLVE FOR THE GORE LENGTH RELATIONS OF ZERO-PRESSUR-
C NATURAL-SHAPE, FLAT-TOP BALLOONS. TO OBTAIN GORE LENGTH/LAMBDA
C (SLMDA) FROM GROSS/PAYLCAD (GP) INITIALIZE INDIC TO 1. TO OBTAIN
C THE REVERSE OF THE ABOVE, INTIALIZE INDIC TO ZERO.
C
C THE FOLLOWING IS A CURVE-FIT CF FIGURE 5 IN NCAR-TN/IA-99, SECTIO
C
IF(INDIC.EQ.0)GO TO 60
IF(GP.GE.1.)GC TC 10
40 WRITE(6,100) GP
100 FORMAT('0','*****') VALUE OF GROSS/PAYOUTLOAD OUTSIDE RANGE OF SUB
ROUTINE GLNGTH. GROSS/PAYOUTLOAD(GP) = ',E12.5,'*****'
INDIC=-1
GO TO 95
10 IF(GP.GE.1.1458)GO TO 11
A=1.99442
B=0.274845
GU TO 50
11 IF(GP.GE.1.3205)GC TC 12
A=1.95287
B=0.280545
GC TO 50
12 IF(GP.GE.1.5306)GO TO 13
A=1.989503
B=0.2EE635
GU TO 50
13 IF(GP.GE.1.7838)GC TO 14
A=1.96507
B=0.291881
GO TO 50
14 IF(GP.GE.2.0897)GU TO 15
A=1.97853
B=0.267581
GC TC 50
15 IF(GP.GE.2.4586)GC TC 16
A=1.97175
B=0.302239
GU TO 50
16 IF(GP.GE.2.9035)GO TO 17
A=1.96328
B=0.307023
GO TO 50
17 IF(GP.GE.3.4380)GC TC 18
A=1.95482
B=0.311076
GC TO 50
18 IF(GP.GE.4.0778)GO TO 19
A=1.94521
B=0.315067
GO TO 50
19 IF(GP.GE.4.8380)GC TC 20
A=1.93701
B=0.318071
GC TO 50
20 IF(GP.GE.5.7363)GO TO 21

LEVEL 21

GLNGTH

DATE = 78226

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A=1.92864
B=0.320819
GU TO 50
21 IF(GP.GE.6.7898)GO TO 22
A=1.92034
B=0.323266
GO TO 50
22 IF(GP.GE.8.0155)GC TC 23
A=1.91325
B=0.325220
GC TO 50
23 IF(GP.GT.10.)GU TO 40
A=1.90667
B=C.326E75
50 SLMDA=A*GP**B
GC TO 99
60 IF(SLMDA.GE.1.99442)GU TO 70
91 WRITE(6,200) SLMDA
200 FORMAT('C','*****') VALUE OF GORE LENGTH/LAMBDA OUTSIDE RANGE
IF SUBROUTINE GLNGTH. SLMDA = ',E12.5,'*****')
INDIC=-1
GO TO 99
70 IF(SLMDA.GE.2.07044)GO TO 71
A=1.99442
B=C.274845
GO TO 95
71 IF(SLMDA.GE.2.15453)GC TC 72
A=1.95287
B=0.280545
GO TO 95
72 IF(SLMDA.GE.2.24767)GO TO 73
A=1.989503
B=0.266E35
GU TO 95
73 IF(SLMDA.GE.2.35038)GC TC 74
A=1.98507
B=0.291881
GO TO 95
74 IF(SLMDA.GE.2.46373)GU TO 75
A=1.97853
B=C.297581
GC TC 95
75 IF(SLMDA.GE.2.58781)GC TC 76
A=1.97175
B=0.302239
GC TC 95
76 IF(SLMDA.GE.2.72339)GU TO 77
A=1.96328
B=0.307023
GO TO 95
77 IF(SLMDA.GE.2.87037)GC TC 78
A=1.95482
B=0.311076
GC TO 95
78 IF(SLMDA.GE.3.02894)GO TO 79
A=1.94521
B=C.315067
GU TO 95

LEVEL 21

GLNGTH

DATE = 78226

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79 IF(SLMDA.GE.3.19819)GO TO 80
A=1.93701
B=0.318071
GC TC 95
80 IF(SLMDA.GE.3.37780)GC TC 81
A=1.92864
B=0.320819
GO TO 95
81 IF(SLMDA.GE.3.56703)GO TO 82
A=1.92034
B=0.323288
GC TC 95
82 IF(SLMDA.GE.3.76484)GC TC 83
A=1.91325
B=0.325220
GO TO 95
83 IF(SLMDA.GT.4.0)GO TO 91
A=1.90667
B=0.326875
95 GP=(SLMDA/A)**(1./E)
99 RETURN
END

```

//OPTIONS
C***** **** - B A L A N - ****
C***** **** BALLOON ANALYSIS PROGRAM ****
C***** **** THIS PROGRAM IS INTENDED TO COMPUTE THE SHAPE AND STRESS
C***** **** DISTRIBUTIONS OF A BALLOON AT ANY ALTITUDE WHERE THE FILM
C***** **** MAY BE CONSIDERED TO LOBE BETWEEN TAPES. THE MANUFACTURED
C***** **** RADIUS AND WEIGHT DISTRIBUTIONS ARE REQUIRED INPUTS. THE
C***** **** DESIGN SHAPE PARAMETERS ARE NEEDED TO START THE ITERATION
C***** **** PROCESS. THE OUTPUT CONSISTS OF THE DEFORMED SHAPE PARA-
C***** **** METERS, MERIDIONAL STRESS, CIRCUMFERENTIAL STRESS, SHEAR
C***** **** STRESS, PRINCIPAL STRESSES, AND TENSION FIELD PATTERNS
C***** **** WHERE APPLICABLE.
C***** **** ALL QUANTITIES ARE NONDIMENSIONALIZED WITH RESPECT TO
C***** **** SMAX, THE UNDEFORMED GORE LENGTH, AND PD, THE DESIGN
C***** **** PAYLOAD.
C***** ****
1      IMPLICIT REAL*8(A-H,O-Z)
2      DIMENSION A(300),C(1801),F(1801),X(1801),RU(300),WT
3           1(300),GP(300),WTAPE(300),SMP(300),SCP(300)
4      REAL*8 NU,KBAR,LAMBDA
5      ATAN(Z)=DATAN(Z)
6      ARSIN(Z)=DARSIN(Z)
7      ARCCOS(Z)=DARCCOS(Z)
8      SIN(Z)=DSIN(Z)
9      COS(Z)=DCOS(Z)
10     TAN(Z)=DTAN(Z)
11     READ(5,800) PD,PTUP,WBAL,SMAX,LAMBDA
12     READ(5,801) NU,KBAR,EBAR,N
13     READ(5,805) KMAX,ABAR,DBAR
14     READ(5,802) (GP(K),PU(K),WT(K),WTAPE(K),K=1,KMAX)
15     SMAX=GP(KMAX)
16     SOL=SMAX/LAMBDA
17     IMAX=6*KMAX+1
18     DO 10 K=1,KMAX
19     L=6*(K-1)+1
20     READ(5,803) (X(L+I), I=1,6)
21     10 CONTINUE
22     PI=4.*ATAN(1.0D0)
23     PUN=PI/N
24     X(1)=1.-PTUP
25     BBAR=BBAR*SOL**3
26     KBAR=KBAR/2./PI
27     EBAR=EBAR*SOL
28     VTUP=X(IMAX)
29     DO 20 K=1,KMAX
30     L=6*(K-1)+1
31     X(L+2)=X(L+2)/SOL
32     X(L+3)=X(L+3)/SOL
33     X(L+6)=(VTUP-X(L+6))/SOL**3*PI
34     GP(K)=GP(K)/SMAX
35     RU(K)=RU(K)*N/12./PI/SMAX
36     WT(K)=WT(K)*SOL
37     WTAPE(K)=WTAPE(K)*SOL
38     CONTINUE
39     WRITE(6,903)(K,GP(K),RU(K),WT(K),WTAPE(K),K=1,KMAX)
40
41     SPUN=SIN(PUN)
     CPUN=COS(PUN)

```

```

42      S2PON=SPUN*SPUN
43      DS=1./(KMAX-1)
44      ZPA=ABAR
45      DO 3 MNN=1,5
46 100   DO 200 K=1,KMAX
47      IF(K.EQ.50)EBAR=4.6*EBAR
48      EM=EBAR/(1.-NU**2)
49      EC=EM
50      EMC=EC*NU
51      U=EM*EC-EMC**2
52      DM=EC/D
53      DC=LM/D
54      DMC=-EMC/D
55      L=0*(K-1)+1
56      STH=SIN(X(L+5))
57      CTH=COS(X(L+5))
58      SH=SIN(X(L+1))
59      CB=COS(X(L+1))
60      ALFA=ARSIN(SPUN*CTH)
61      IF(ZPA.LT.0.)ALFA=-ALFA
62      IF(CTH.LT.0.)ALFA=-ALFA
63      ABET=ALFA+X(L+1)
64      PHI=2.*ABET
65      SPDN=SIN(ABET)
66      FIR=RU(K)*DC/DMC+(DC*DM/DMC-DMC)*KBAR
67      IF(K.EQ.1)THP=(X(12)-X(6))/DS
68      IF(K.EQ.KMAX)THP=(X(L+5)-X(L-1))/DS
69      IF(K.EQ.1) GO TO 102
70      IF(K.EQ.KMAX) GO TO 102
71      THP=(X(L+1))-X(L-1))/2./DS
72 102   CONTINUE
73      ZPA=X(L+3)+ABAR
74      SM=(X(L+4)-KBAR*DMC*ABET/PON*(RU(K)*BBAR*ZPA+STH*(WT(K)
1-WTAPE(K)/PON/2.)))/(RU(K)+KBAR*DM+KBAR*DMC*ABET/PON*
2RU(K)*THP)
75      EF=X(L+4)-RU(K)*SM
76      IF(EF.LT.0.)SM=X(L+4)/RU(K)
77      IF(EF.LT.0.)EF=0.
78      FTB=SPUN*CB*CTH+SU*(CPUN*CTH*CTH-1.+5TH*STH)
79      SC=ABET/PON*(RU(K)*SM*THP+RU(K)*BBAR
1*ZPA+STH*(WT(K)-WTAPE(K)/PON/2.))
80      SMP(K)=SM
81      A(K)=(X(L+4)-RU(K)*SM)*2.*PON
82 105   CONTINUE
83      IF(SC.LT.0.)SC=0.
84      SCP(K)=SC
85      EMP1=1.+DM*SM+DMC*SC
86      LCP1=1.+DMC*SM+DC*SC
87      F1B=SPUN*CB+SH*CTH*(CPUN-1.)
88      F2B=FTB/PUN
89      IF(LF.EQ.0.)BET=ARSIN(-PON*STH*WTAPE(K)/SC)
90      IF(LF.EQ.0.)STUFF=FAN(BET)
91      IF(LF.EQ.0.)GO TO 104
92      RATIO=RU(K)*SM/(X(L+4)-RU(K)*SM)
93      TOP=BHAR*ZPA*X(L+2)+EMP1*SPUN+PON*(WT(K)-WTAPE(K)
1/PON/2.)*STH-SC*SPUN*CTH*CB-RATIO*WTAPE(K)*STH/2.
94      BOT=CB*(RATIO*SC+SC*(CPUN*CTH*CTH+5TH*STH))
95      IF(SC.EQ.0.)STUFF=-999999999
96      IF(SC.EQ.0.)GO TO 104
97      STUFF=TOP/BOT
98 104   CONTINUE

```

```

99      ASEG=(PHI-SIN(PHI))*(SPIN/SPUN)**2*CTH
100     IF(ZPA.EQ.0.)SIGN=1.
101     IF(ZPA.EQ.0.) GO TO 101
102     SIGN=ZPA/DAHS(ZPA)
103   101 ASUBN=N*(SIN(2*PUN)+SIGN*ASEG)
104     NN=1
105     JJ=0
106     IF(K.EQ.1) GO TO 170
107     IF(K.EQ.KMAX) GO TO 180
108     IF(K.EQ.2) GO TO 160
109   159 IF(K.GE.2)JJ=0
110     IF(K.GE.2)NN=2
111   160 CONTINUE
112     C(L+1)=ATAN(STUFF)
113     IF(K.EQ.KMAX-1)GO TO 161
114     C(L+2+JJ)=X(L-4)+NN*DS*STH*EMPI
115     C(L+3+JJ)=X(L-3)+NN*DS*CTH*EMPI
116     C(L+4+JJ)=X(L-2)+NN*DS*(WT(K)*CTH+SC*STH/PUN*F1B)
117     IF(JJ.EQ.0)GO TO 161
118     IF(K.EQ.2) GO TO 159
119   161 C(L+5-JJ)=X(L+1)+NN*DS/X(L+4)*(WT(K)*STH+BBAR*ZPA*X(L+2)*EMPI
120     1*SPUN/PUN-SC/PUN*F1B)
121     C(L+6-JJ)=X(L+2)+NN*DS*EMPI*X(L+2)*X(L+2)*ASUBN*CTH/2.
122     IF(JJ.EQ.0) GO TO 200
123     IF(K.EQ.KMAX-1)NN=1
124     IF(K.EQ.KMAX-1)JJ=0
125     IF(K.EQ.KMAX-1) GO TO 161
126     GO TO 200
127   170 CONTINUE
128     C(1)=BBAR*X(7)-PTUP-WHAL
129     C(2)=ATAN(STUFF)
130     C(3)=RU(1)
131     C(4)=0.
132     C(5)=X(1)/CTH/2./PI
133     C(6)=X(12)+DS/X(5)*(STH*WT(K)+BBAR*ZPA*X(3)*EMPI*SPON/PUN
134     1-SC*F2B)
135     C(7)=X(13)+DS*EMPI*X(3)*X(3)/2.*CTH*ASUBN
136     GO TO 200
137   180 CONTINUE
138     C(L+1)=ATAN(STUFF)
139     C(L+2)=RU(KMAX)
140     C(L+3)=X(L-3)+DS*CTH*EMPI
141     C(L+4)=X(L-2)+DS*(WT(K)*CTH+SC*STH/PUN*F1B)
142     C(L+5)=-PI/2.
143     PR=BBAR*ZPA*RU(KMAX)**2*PI
144     PSOP=PTUP-PR
145     IF(PSOP.GT.0.)C(L+5)=-ARCCOS((PR-PTUP)/2./PI/X(L+4))
146     IF(PSOP.LT.0.)C(L+5)=ARCCOS((-PR+PTUP)/2./PI/X(L+4))
147   200 CONTINUE
148     EBAR=EBAR/4.0
149     RMS=0.
150     SIGMA=.50
151     DO 210 I=1,IMAX
152       F(I)=C(I)-X(I)
153       X(I)=X(I)+SIGMA*F(I)
154       RMS=RMS+F(I)*F(I)
155   210 CONTINUE
156   2 CONTINUE
157     DO 1 KL=1,KMAX
158       IF(KL.EQ.1) WRITE(6,13) KL,X(KL),F(KL)

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153      WRITE(6,7) KL,(X(I+6*KL-5),I=1,5),(F(I+6*KL-5),I=1,5),
154      1SMP(KL),SCP(KL)
155      13 FORMAT(15.2F10.6)
156      7 FORMAT(15.12F10.6)
157      1 CONTINUE
158      CHE CK=DABS(F(1)/X(1))
159      RMS=RMS**.5
160      WRITE(6,900) RMS
161      HT=BBAR/SOL**3
162      VOL=X(7)*SMAX**3
163      PAY=X(1)*PD
164      3 CONTINUE
165      WRITE(6,908) PAY,LAMBDA,E BAR,VOL,NU,ABAR,SMAX,
166      1KBAR,HT,N,KMAX
167      CALL PRINT (KMAX,A,SMP,SCP,X,WTAPE,DS)
168      C FORMAT STATEMENTS FOR INPUT
169
170      C
171      800 FORMAT(5E15.7)
172      801 FORMAT(3E15.7,115)
173      802 FORMAT(4L15.7)
174      803 FORMAT(5E15.7)
175      805 FORMAT(115.2E15.7)
176      C FORMAT STATEMENTS FOR OUTPUT
177
178      900 FORMAT(IX,*RMS VALUE = *,E20.13)
179      903 FORMAT(14.4E15.7)
180      908 FORMAT(1H1,*10X,*DESIGN VALUES*,/,*10X,*PAYLOAD=*,6X,F7.2,
181      *10X,*LAMBDA=*,F4.0,5X,*EBAR=*,F7.2,/,10X,*VOLUME=*,,
182      2L14.4,*FT(3)*,8X,*NU=*,F8.2,5X,*ABAR=*,F5.2,/,10X,*GORE LENGTH=*,,
183      3F9.2,*FT*,11X,*KBAR=*,F6.2,5X,*BBAR=*,F5.2,/,10X,
184      4*NUMBER OF TAPES=*,15,/,10X,*NUMBER OF POINTS=*,14,/)
185      999 STOP
186      END
187
188      SUBROUTINE PRINT (KMAX,F,SMP,SCP,X,WTAPE,DS)
189      IMPLICIT REAL*8(A-H,O-Z)
190      DIMENSION F(300),SCP(300),SMP(300),X(1801),WTAPE(300)
191      IMAX=6*KMAX+1
192      WRITE(6,801)
193      WRITE(6,800) (X(I),I=2,IMAX)
194      WRITE(6,900)
195      P1=4.*DATA4(1,D0)
196      DO 105 K=1,KMAX
197      C CALCULATE SHEAR STRESS
198      C
199      CTH=DCUS(X(6*K))
200      IF(K.EQ.1) GO TO 101
201      IF(K.EQ.KMAX) GO TO 102
202      TAU=(F(K+1)-F(K-1))/4./DS-WTAPE(K)*CTH/2.
203      GO TO 103
204      101 TAU=(-3.*F(K)+4.*F(K+1)-F(K+2))/4./DS-WTAPE(K)*CTH/2.
205      GO TO 103
206      102 TAU=(F(K-2)-4.*F(K-1)+3.*F(K))/4./DS-WTAPE(K)*CTH/2.
207      103 SP=SMP(K)+SCP(K)
208      C CHECK TO SEE IF A TENSION FIELD EXISTS
209      C
210      IF(ELCP(K).LE.0.) GO TO 104
211      TAUMAX=DSQRT((SMP(K)-SCP(K))/2.D0)**2+TAU**2)
212      S1=SP/2.+TAUMAX
213      S2=SP/2.-TAUMAX

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203      ANG=DATAN(2.0D0*TAU/(SMP(K)-SCP(K)))/2.
204      IF(SCP(K).GT.SMP(K)) ANG=ANG+PI/2.0D0
205      GU TU 105
206      104 ANG=DATAN(SMP(K)/TAU)-PI/2.0D0
207      S1=SMP(K)/DCOS(ANG)/DCOS(ANG)
208      S2=0.
209      TAUMAX=SMP(K)/2.0D0
210      105 WRITE(6,901) K,F(K),SMP(K),SCP(K),TAU,S1,S2,TAUMAX,ANG
C
C      FORMAT STATEMENTS FOR OUTPUT
C
211      800 FORMAT(6F10.4)
212      801 FORMAT(//,4X,'BETA',7X,'R',9X,'Z',9X,'T',8X,'THETA'
1.5X,'VOL',//)
213      900 FORMAT(1H1,43X,'MER',6X,'CIR',6X,'MER',4X,'PRINCIPAL STRESSES'
1.4X,'MAX',29X,'K',5X,'F',7X,'STRESS',4X,'STRESS',4X,'SHEAR'
2.7X,'S1',8X,'S2',6X,'SHEAR',5X,'ANGLE',//)
214      901 FORMAT(26X,14.8F10.4)
215      RETURN
216      END

```