

Model Based Optimal Tuning of Proportional Resonant Controllers

Topic number: 5

Abstract—The paper considers the optimal choice of gains of proportional-resonant controllers. A converter, controlled in closed loop, but without proportional resonant controllers, is modeled as a second order transfer function from reference to output. The closed-loop system is amended by proportional-resonant control by feeding the error between reference and output back through a set of N proportional resonant controller to subsequently alter the reference. The root locus of the closed loop system is considered as a function of the proportional-resonant gains. To find the optimal choice of gains, we consider maximizing the damping of the mode with smallest damping. This corresponds to solving a nonlinear min-max problem. After linearization and reformulation, the problem is stated as a linear program.

I. INTRODUCTION

Proportional resonant tuning methods [1]–[7]

II. CONVERTER MODEL

We consider a voltage source inverter which is assumed to operate in closed loop, but without PR controllers of integrators. The controlled voltage source inverter is modeled as a second order transfer function which maps the (sinusoidal) reference to the output,

$$y = G(s)y_{\text{ref}}, \quad G(s) = \frac{\omega^2}{s^2 + \xi\omega s + \omega^2}, \quad (1)$$

where ω is the natural frequency and ξ is the damping of the controlled inverter.

We note that, with properly designed control, most inverters are expected to behave as second order systems in closed loop. Assuming a system model on the form (1) is thus not restrictive.

TB: Harmonics produced by the converter.

A. Proportional-resonant controllers

To achieve offset free tracking of the sinusoidal reference y_{ref} and to reduce harmonics in the output, we consider adding proportional-resonant (PR) controllers [8] to the system (1). The PR controllers are added in a slower outer loop (see Fig. 1) and adjust the reference according to

$$\tilde{y}_{\text{ref}} = y_{\text{ref}} + \sum_{n \in \{1,3,5,\dots,N\}} H_n(s)(y - y_{\text{ref}}),$$

where

$$H_n(s) = \frac{\lambda_n s}{s^2 - n\omega_0^2},$$

where the fundamental frequency ω_0 is the frequency of the reference (typically corresponding to 50 or 60 Hz), and where λ_n are the feedback gains of the PR controllers. These gains are tuning parameters whose value affect the transient response of the closed loop system.

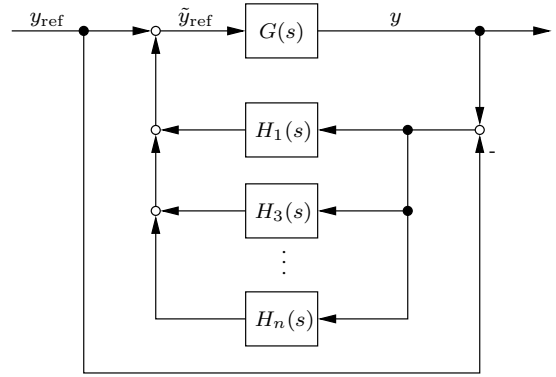


Fig. 1. Control structure of converter model with PR controllers.

B. Closed loop system

The order of the closed loop system dynamics is $2 + 2N$ where N is the number of PR controllers added in the outer loop.

TB: Transfer function of closed-loop system

III. CONTROLLER GAIN OPTIMIZATION

TB: How about Problem Formulation as section header?

The PR gains λ_n affect both the poles and zeros of the closed loop system. In the approach presented below, we focus on the poles and seek to maximize the damping of the (complex) pole pair which has the lowest damping.

To clarify the approach we consider an example: Consider the case where two PR controllers (with harmonics number 1 and 3) are included in the control loop. For this case the system has 6 poles, and their position in the complex plane is determined by two gains λ_1, λ_3 .

We enumerate combinations of gains λ_1, λ_3 and plot the resulting poles; the results are shown as the green dots in

Fig. 2. In this figure we also plot the poles obtained when both gains are small (close to zero), and the poles obtained with one particular choice of high gains. The poles obtained with low and high gains are shown by blue and red circles, respectively.

From Fig. 2 it can be seen that one pole pair moves to the right, closer to the imaginary line (and unstable domain), while the other two pole pairs move left. For sufficiently high gains, one pole pair becomes two purely real poles, one of which moves left and the other moves right, into the unstable domain.

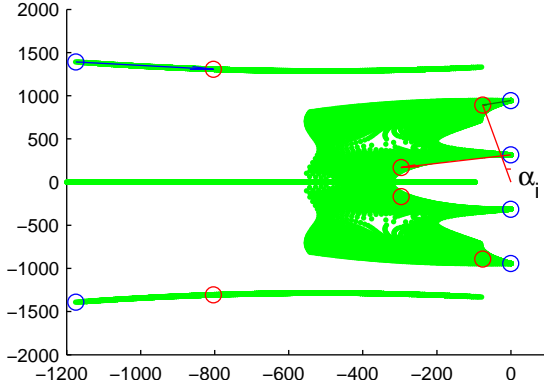


Fig. 2. Poles of the closed loop system with $N = 2$ PR controllers: The green points show poles for various combinations of gains λ_1, λ_3 . Blue circles show the poles for low gains. Red circle show poles for high gains.

Since changes in one gain affects all poles, it is not obvious how to choose the gains optimally. Increase in one gain may make one pole pair “more stable”, but may have negative effects one another pole pair.

To address the problem of how to choose the PR gains, we propose to formulate a max-min optimization problem, where we maximize the damping of the least damped pole pair. More precisely, we consider the angle between the pole pair and the imaginary axis (assuming the pole is in the open left hand plane);

$$\alpha_i = \tan^{-1}(-\text{real}(p_i)/\text{imag}(p_i))$$

where p_i is a pole in the fourth quadrant. The problem we ideally want to solve is

$$\max_{\lambda_1, \lambda_3, \dots, \lambda_N} \min_{i \in \{1, 3, \dots, N\}} \alpha_i(\lambda_1, \lambda_3, \dots, \lambda_N). \quad (2)$$

We note that the angles α_i are dependent on the PR gains λ_i , and that as the gains vary, different angles take on the role of being “the smallest”.

A. Problem approximation

To obtain a tractable solution, we proceed to approximate $\alpha_i(\lambda_1, \lambda_3, \dots, \lambda_N)$ with affine functions of the gains λ_i : That is, the angles are approximated by

$$\tilde{\alpha}_i(\lambda_1, \lambda_3, \dots, \lambda_N) = a_i^T \lambda + b_i \quad (3)$$

where

$$\lambda = [\lambda_1 \quad \lambda_3 \quad \dots \quad \lambda_N]^T$$

is a vector containing the gains and where $a_i \in \mathbb{R}^N$, $b_i \in \mathbb{R}$ are constant vectors obtained by sampling the value of the angles α_i for a number of gain combinations, and performing a least squares fit.

By describing the angles in (2) with the approximation (3), we obtain a max-min problem with affine cost function. This problem can be equivalently formulated a linear program (LP) according to

$$\min c^T x, \quad \text{s.t. } Ax \leq B \quad (4)$$

with matrices

$$A = \begin{bmatrix} 1 & -c_1^T \\ 1 & -c_2^T \\ & \vdots \\ 1 & -c_N^T \end{bmatrix}, \quad B = \begin{bmatrix} b_1 \\ b_2 \\ \vdots \\ b_N \end{bmatrix}, \quad C = \begin{bmatrix} -1 \\ 0 \\ \vdots \\ 0 \end{bmatrix}.$$

Put derivation in appendix

IV. NUMERICAL EXAMPLE

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