The Earthquake Dissipative Engine: Energy Budget and Partition

PART E: Derivation of stress pattern at the rupture tip

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- 27 Oct. 16:30-18:30 classroom 2G
- 28 Oct. 14:30-16:30 classroom 2H
- 29 Oct. 10:30-12:30 classroom Lab Paleo
- 31 Oct. 14:30-16:30 classroom 2L

Static Mode III Stress Singularity

Where does the $r^{-1/2}$ singularity in stress come from?

Perhups the simplest derivation is proposed by Williams, M. L. (1957) – On the stress distribution at the base of a stationary crack Journal of Applied Mechanics, 24, 109–114. Mode III is the simplest configuration (only two shear stresses) but it allows some

We consider an anti-plane shear (Mode III) crack in an isotropic, linear elastic solid. The out-of-plane displacement w(x,y) satisfies Laplace's equation (1)

$$\nabla^2 w = 0,$$

with shear stresses

$$\tau_{xz} = \mu \frac{\partial w}{\partial x}, \qquad \tau_{yz} = \mu \frac{\partial w}{\partial y}. \tag{2}$$
Let the crack lie along the negative reaxis, and introduce polar coordinates (r, θ) centered at the crack tip. On the crack

Let the crack lie along the negative x-axis, and introduce polar coordinates (r,θ) centered at the crack tip. On the crack faces $(\theta = \pm \pi)$ the shear traction must vanish:

$$\tau_{\theta z}(r,\theta) = \mu \frac{1}{r} \frac{\partial w}{\partial \theta} = 0 \quad \text{for} \quad \theta = \pm \pi.$$
(3)

Separable (Williams) form

Seek separable solutions of the form

$$w(r,\theta) = r^{\lambda} f(\theta). \tag{4}$$

Laplace in ploar coordinates yields: $\frac{\partial^2 w}{\partial x^2} = \frac{\partial^2 w}{\partial y} = \partial^2$	
$\nabla^2 w = \frac{\partial^2 w}{\partial r^2} + \frac{1}{r} \frac{\partial w}{\partial r} + \frac{1}{r^2} \frac{\partial^2 w}{\partial \theta^2} = 0.$	(5)
Substituting into Laplace's equation in polar coordinates gives	
$f''(\theta) + \lambda^2 f(\theta) = 0,$	(6)
whose general solution is	
$f(\theta) = A\sin(\lambda\theta) + B\cos(\lambda\theta).$	(7)
The traction-free condition requires	
$\frac{\partial w}{\partial \theta} = 0$ at $\theta = \pm \pi$,	(8)
so that $\lambda \left[A\cos(\lambda\pi) - B\sin(\lambda\pi) \right] = 0.$	(9)

Mode III symmetry and eigenvalues

Static Mode III Stress Singularity

The dominant (most singular) term near the tip corresponds to $\lambda = 1/2$, giving

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 $\cos(\lambda \pi) = 0 \quad \Rightarrow \quad \lambda = \frac{1}{2}, \frac{3}{2}, \dots$

 $w(r,\theta) = A r^{1/2} \sin\left(\frac{\theta}{2}\right).$

Mode III loading is antisymmetric about the crack plane, hence B = 0. A nontrivial solution then demands

(10)

(11)

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Stress field and singularity

Differentiating,

so that

$$\frac{\partial w}{\partial r} = \frac{A}{2} r^{-1/2} \sin\left(\frac{\theta}{2}\right),$$
$$\frac{\partial w}{\partial \theta} = \frac{A}{2} r^{1/2} \cos\left(\frac{\theta}{2}\right),$$

$$\frac{\partial w}{\partial \theta} = \frac{A}{2} r^{1/2} \cos\left(\frac{\theta}{2}\right),\,$$

 $\tau_{rz} = \mu \frac{\partial w}{\partial r} = \mu \frac{A}{2} r^{-1/2} \sin\left(\frac{\theta}{2}\right),$

For in-plane shear (Mode II) loading, the displacements u_x, u_y satisfy the Navier equations of plane elasticity,

 $\mu \nabla^2 u_i + (\lambda + \mu) \frac{\partial}{\partial r_i} (\nabla \cdot \mathbf{u}) = 0,$

(12)

(13)

(16)

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$$\tau_{\theta z} = \mu \frac{1}{r} \frac{\partial w}{\partial \theta} = \mu \frac{1}{2} r^{-1/2} \cos\left(\frac{\theta}{2}\right).$$
 Both components vary as $r^{-1/2}$, showing the characteristic inverse-square-root singularity.

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(15)

Static Mode III Stress Singularity

 $\nabla^4 \Phi = 0$.

Williams eigenfunction form

Near a traction–free crack tip, we seek homogeneous solutions of the form

 $\Phi(r, \theta) = r^{\lambda+1} f(\theta),$

where (r,θ) are polar coordinates centred at the crack tip. Substitution into the biharmonic equation yields an angular eigenvalue problem whose admissible exponents are

$$\lambda = \frac{1}{2}, \frac{3}{2}, \frac{5}{2}, \dots$$

as first shown by Williams (1957). The leading term $\lambda = \frac{1}{2}$ gives stresses varying as $r^{-1/2}$.

which may equivalently be expressed in terms of the Airy stress function $\Phi(x, y)$, defined by

(18)

(19)

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Stress field for Mode II

ahead of the tip $(\theta = 0, r = x)$,

Enforcing traction–free boundary conditions on the crack faces ($\theta = \pm \pi$) and antisymmetry of shear about the x-axis leads to the standard near-tip field

$$\sigma_{xx} = -\frac{K_{II}}{\sqrt{2\pi r}} \sin\frac{\theta}{2} \left(2 + \cos\frac{\theta}{2} \cos\frac{3\theta}{2} \right),\tag{20}$$

$$\sigma_{yy} = \frac{K_{II}}{\sqrt{2\pi r}} \sin\frac{\theta}{2} \cos\frac{\theta}{2} \cos\frac{3\theta}{2},$$

$$\tau_{xy} = \frac{K_{II}}{\sqrt{2\pi r}} \cos\frac{\theta}{2} \left(1 - \sin\frac{\theta}{2} \sin\frac{3\theta}{2}\right),$$

where
$$K_{II}$$
 is the Mode II stress intensity factor. All components exhibit the same $r^{-1/2}$ singularity. Along the crack plane ahead of the tip $(\theta = 0, r = x)$

$$\tau_{xy}(x,0) = \frac{K_{II}}{\sqrt{2\pi x}},$$

demonstrating the classical inverse–square–root stress concentration for static in–plane shear.

Remarks

The functional form of the angular terms differs from Mode III, but the scaling of stress with distance from the crack tip is identical, following from the same eigenvalue $\lambda = 1/2$ in the Williams expansion. Equivalent derivations can be obtained through the complex–potential (Muskhelishvili) formulation, using analytic functions $\phi(z)$ and $\psi(z)$ for which the near-tip behaviour again yields the $1/\sqrt{r}$ singularity.

(21)

(22)

(23)

References

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