

Laminar Burning Speed of *n*-Hexane–Air Mixtures

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8th US National Combustion Meeting

University of Utah
May 19 - 22, 2013



Summary

1. Motivation

- Accidental Ignition

2. Previous Work

3. Materials and Methods

4. Results

5. Conclusions

Accidental Ignition

■ Accidental ignition

- electrostatic ignition of fuel
- lightning strike
- electrical faults in pumps, fuel quantity instrumentation
- hot surface ignition

■ Characterize fuel-oxidizer properties (*n*-hexane)

- ignition delay time (Burcat et al. and Zhukov et al.)
- heating rate on the low temperature oxidation of hexane by air (Boettcher et al.)
- minimum ignition temperature (Boettcher)
- minimum ignition energy (Bane)
- laminar burning speed



TWA 800, NY 747-100, July 17, 1996



China Air Flight 120 caught fire in Okinawa Japan (BBC News, August 20, 2007)

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Laminar Burning Speed

- Davis and Law :
 - $T_0 = 296$ K and $P_0 = 100$ kPa
- Farrell et al. :
 - $T_0 = 450$ K and $P_0 = 304$ kPa
- Kelley et al. :
 - $T_0 = 353$ K and $P_0 = 100\text{-}1000$ kPa
- Ji et al. :
 - $T_0 = 353$ K and $P_0 = 100$ kPa

Laminar Burning Speed

$P = 0.2 \text{ atm}$

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n-hexane-air
 $P_0 \leq 100 \text{ kPa}$
 $T_0 = 296\text{-}380 \text{ K}$

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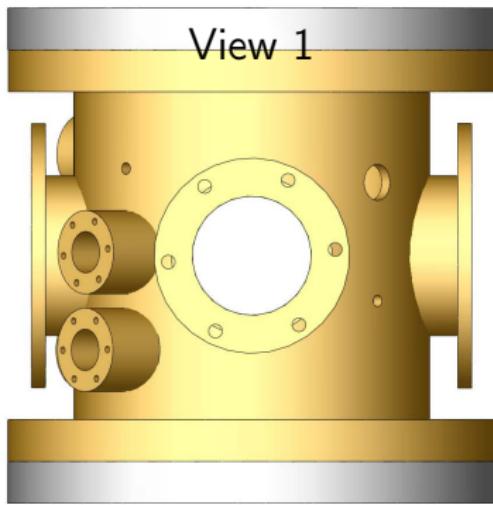
- Experimental Setup
- Burning Speed Measurements

4. Results

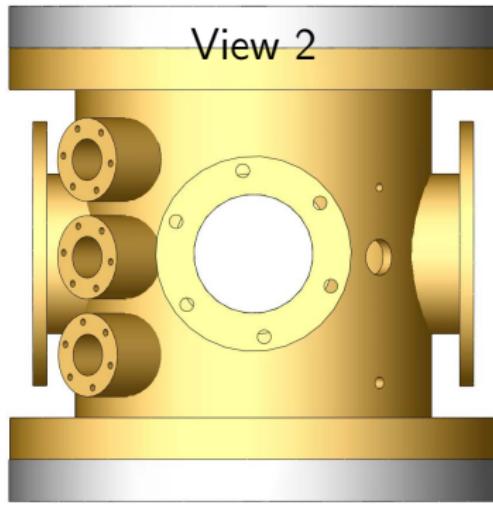
5. Conclusions

Experimental Setup : Combustion Vessel

View 1

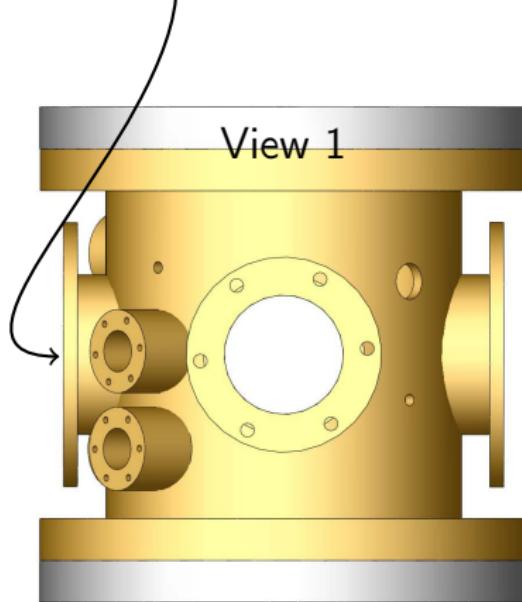


View 2

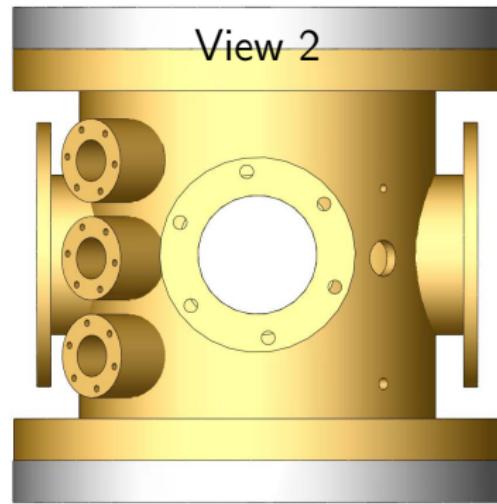


Experimental Setup : Combustion Vessel

11.7 cm diameter windows



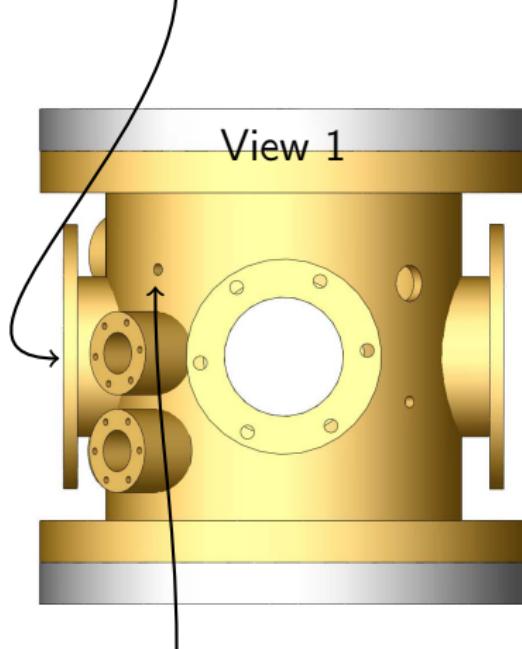
View 1



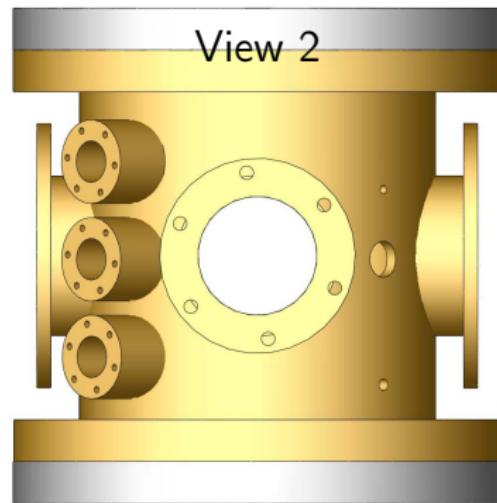
View 2

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View 1



View 2

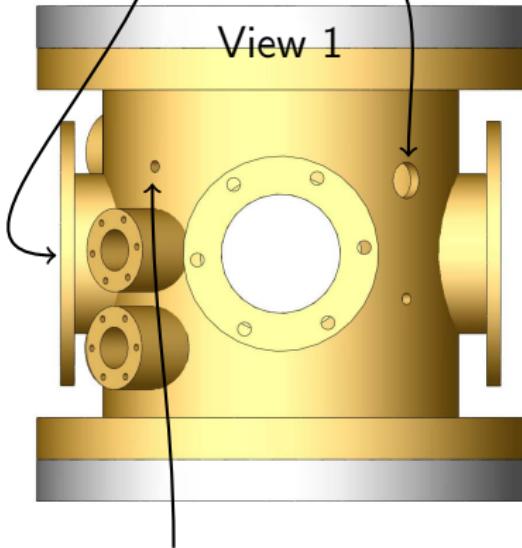
pressure manometer

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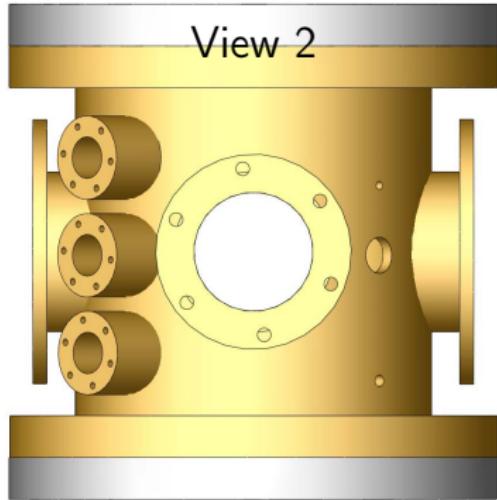
11.7 cm diameter windows

vacuum

View 1



View 2



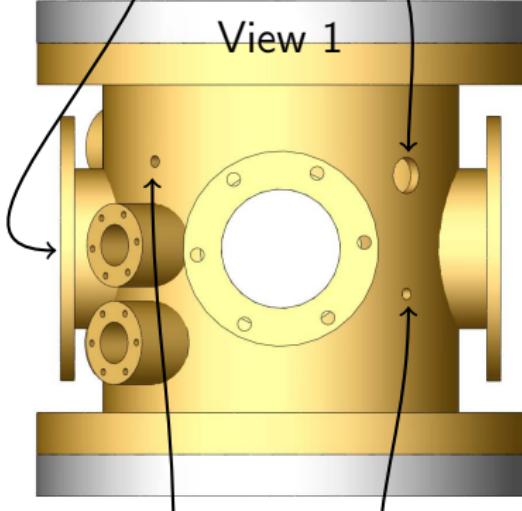
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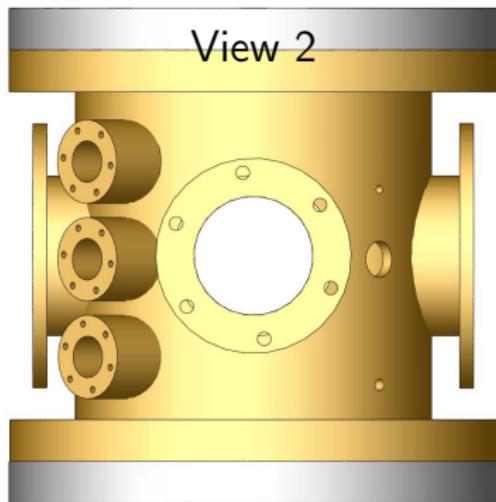
View 1



pressure manometer

gas fill line

View 2

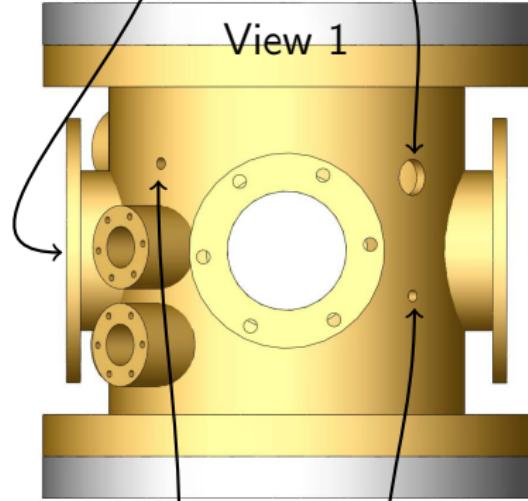


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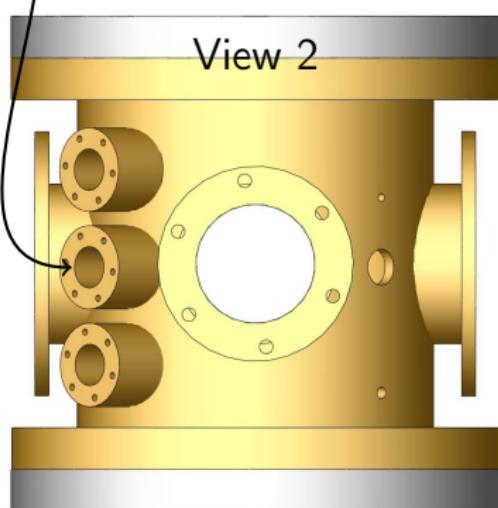


pressure manometer

gas fill line

fan mixer

View 2

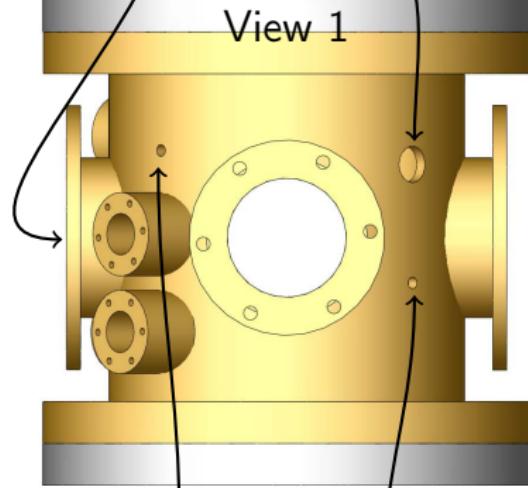


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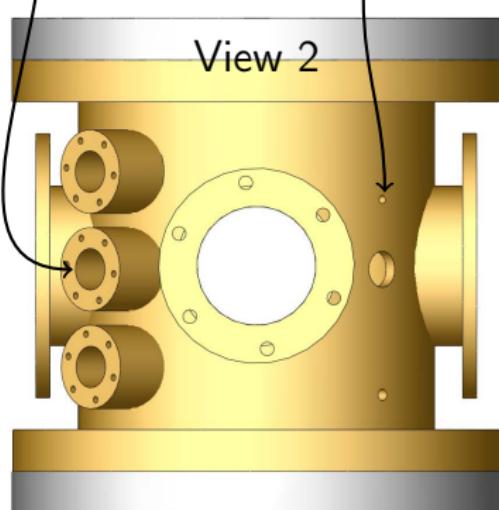
pressure manometer

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septum

View 2

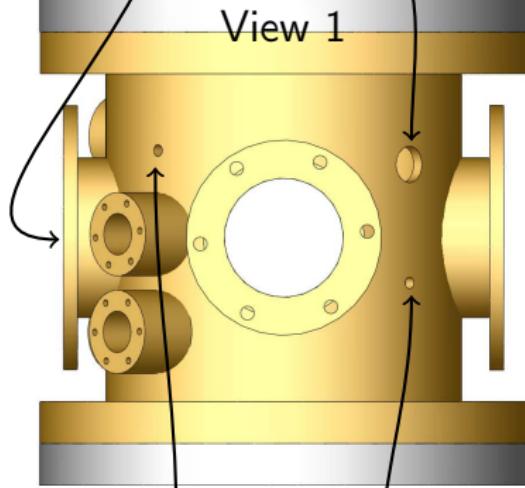


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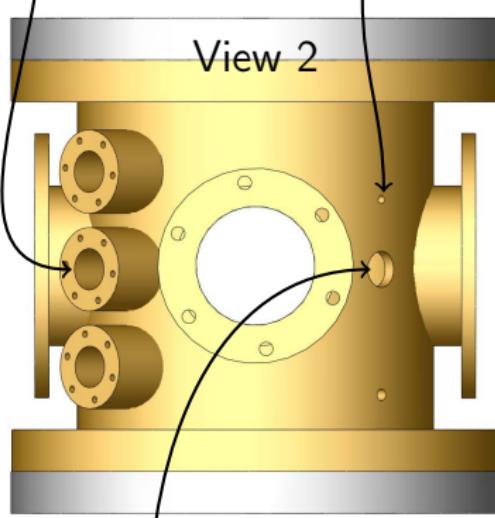
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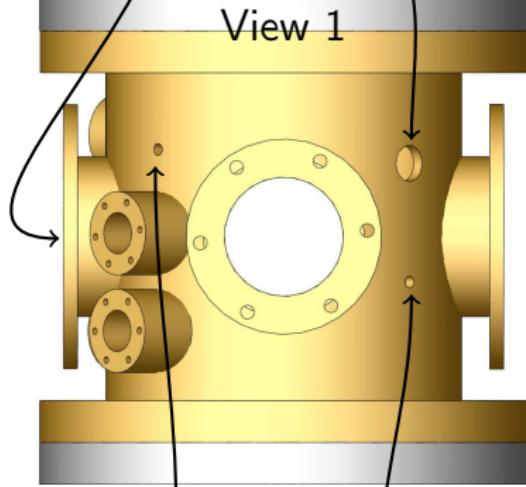
piezoresistive pressure transducer

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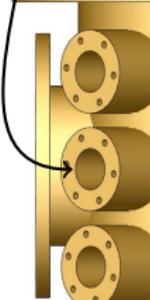
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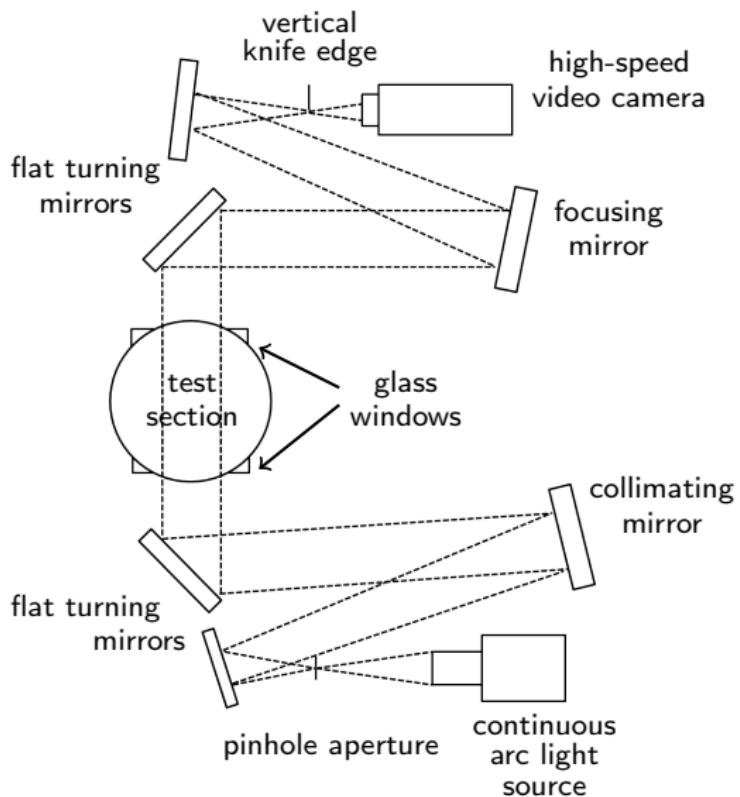


thermocouple

piezoresistive pressure transducer

Experimental Setup : Schlieren Setup

- Observe changes in the density gradient of the fluid due to variations in the refractive index
- Visualize flame :
 - very hot flame propagating into cold unburned reactants
- High speed camera :
 - 10,000 frames per second
 - 512×512 resolution

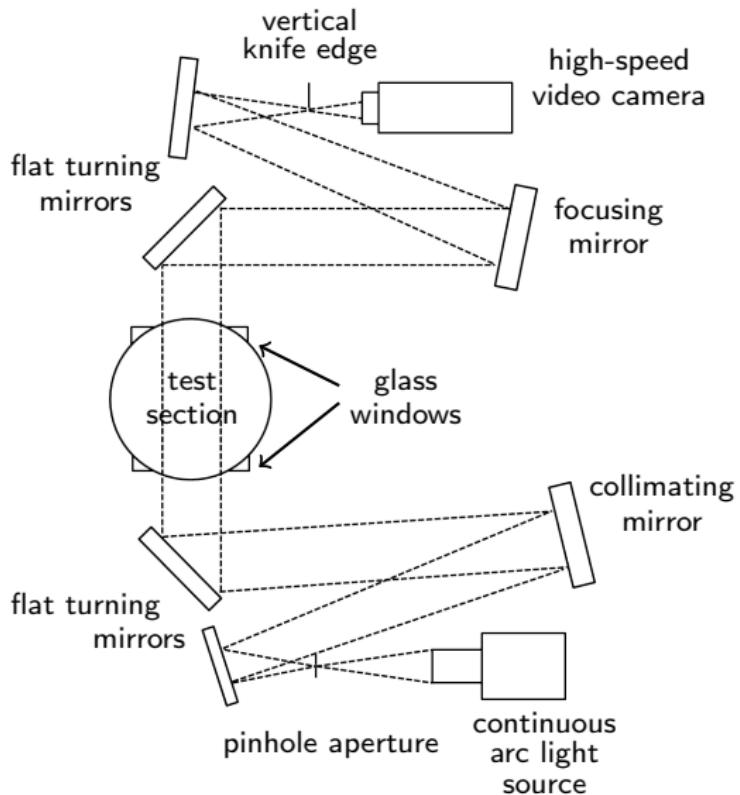


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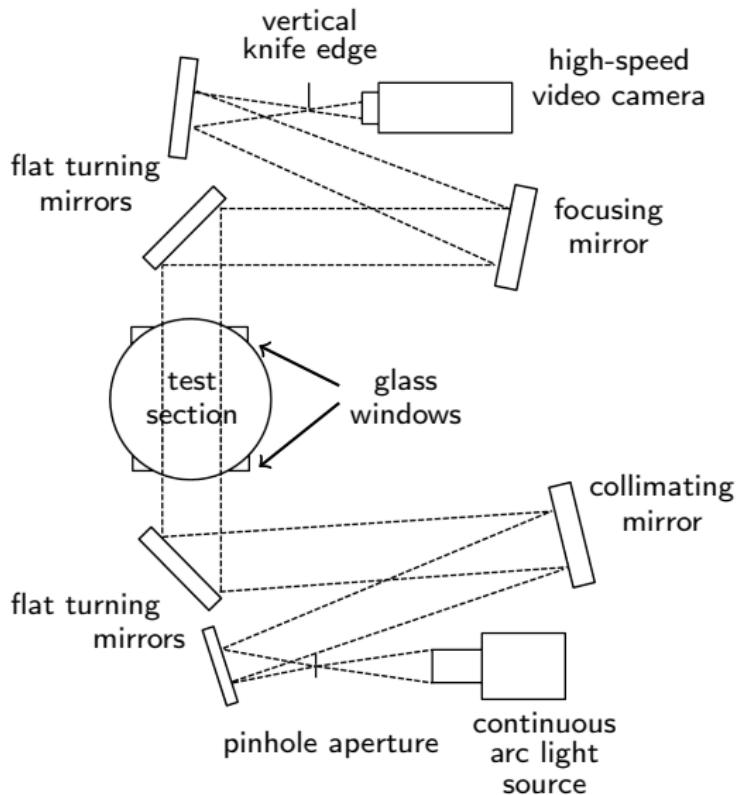
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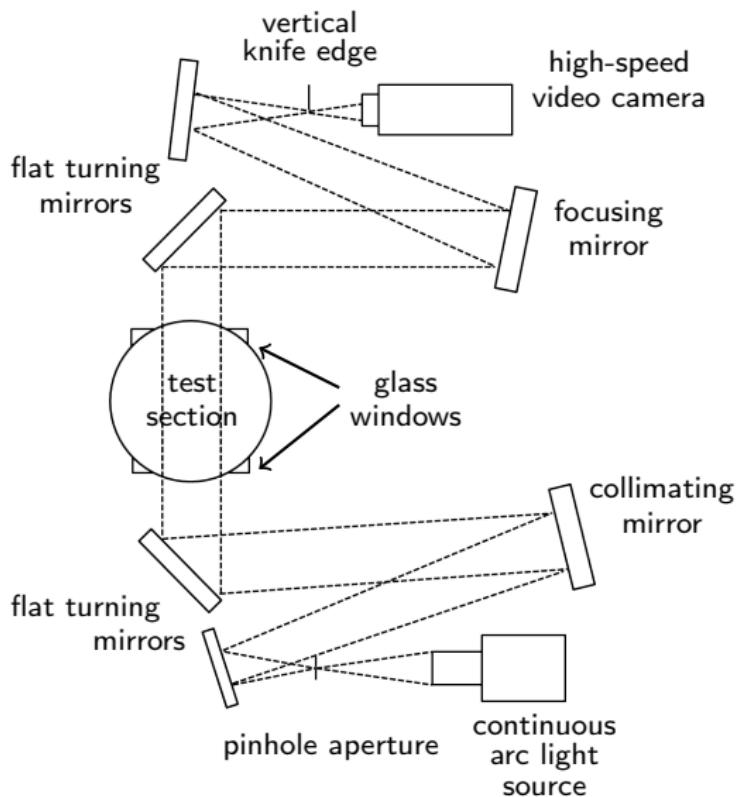
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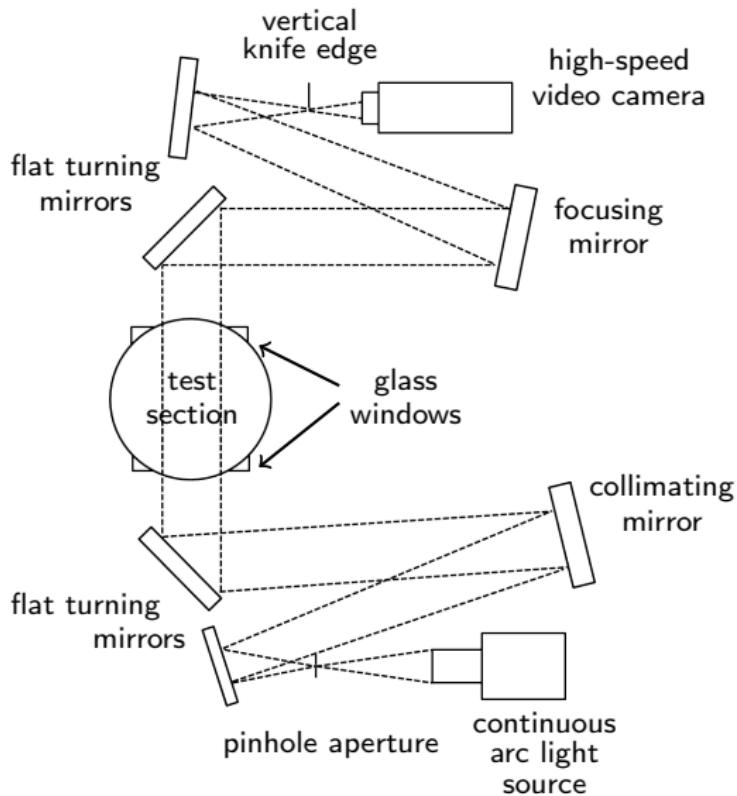
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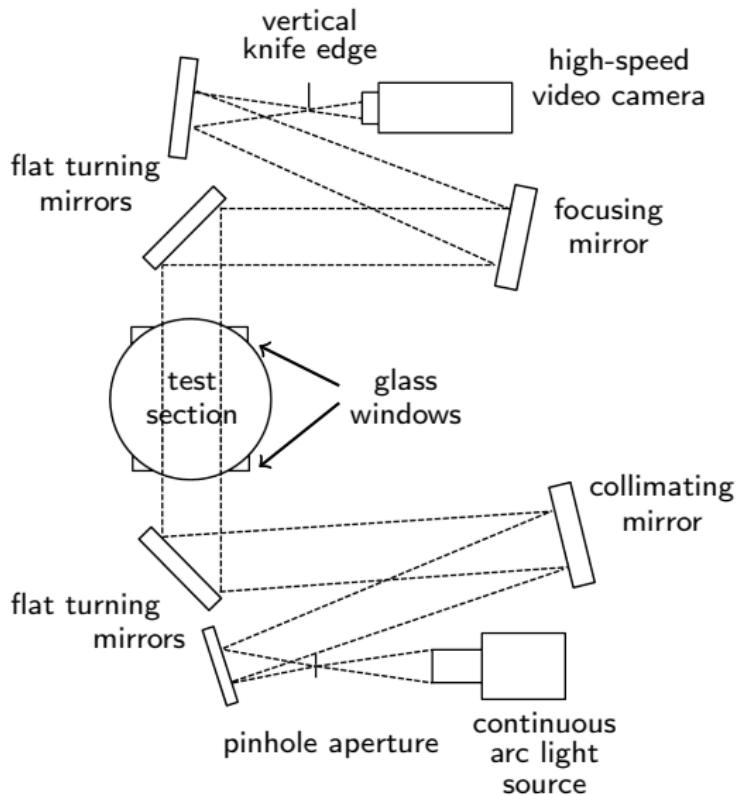
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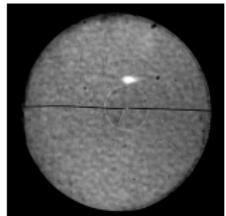


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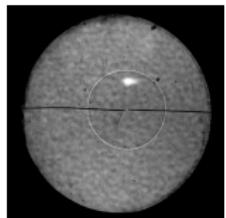


Burning Speed Measurements

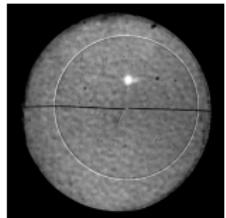


$t = 5.0 \text{ ms}$

- Edge detection using the Canny method (MATLAB)
- Fit ellipse to detected edge
 - use area of ellipse to find an equivalent radius
- Linear extrapolation to unstretched flame speed

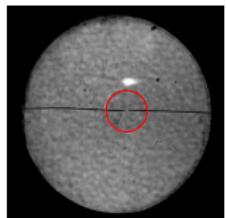


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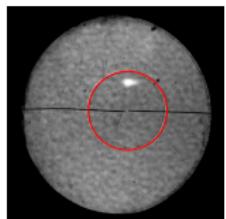
$t = 17.1 \text{ ms}$

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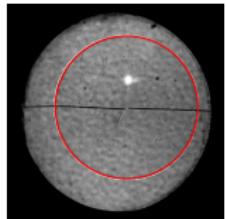


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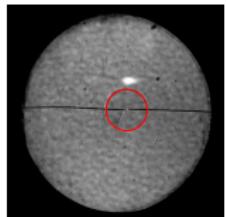


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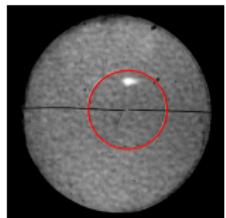


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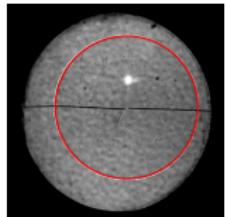
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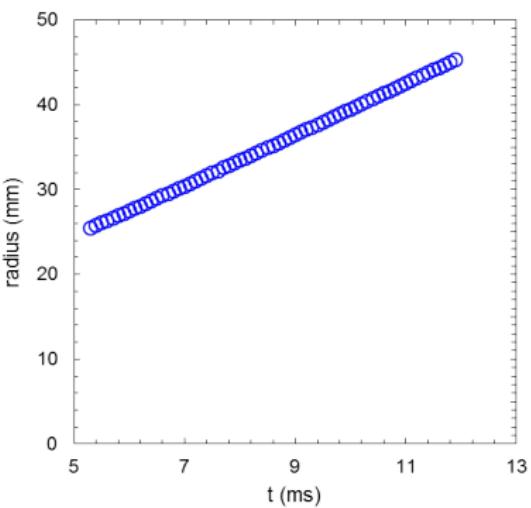


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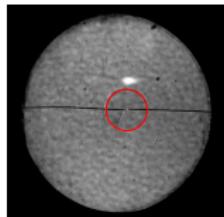


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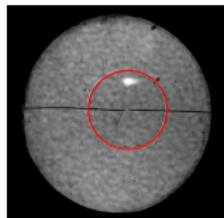
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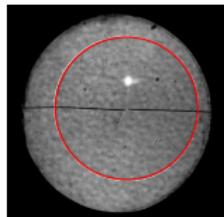
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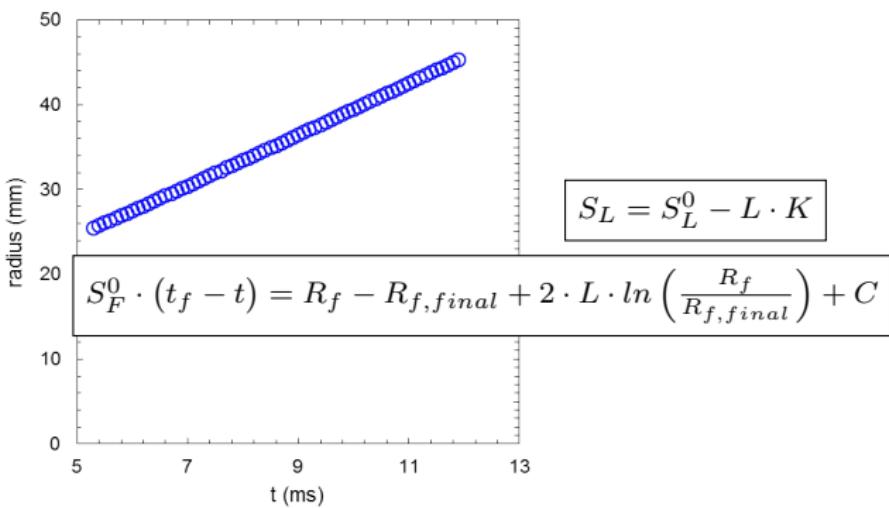


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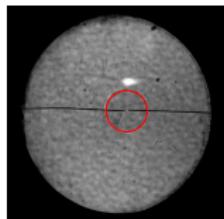


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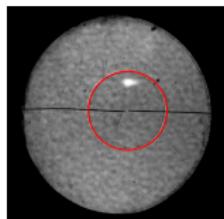
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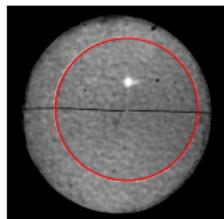
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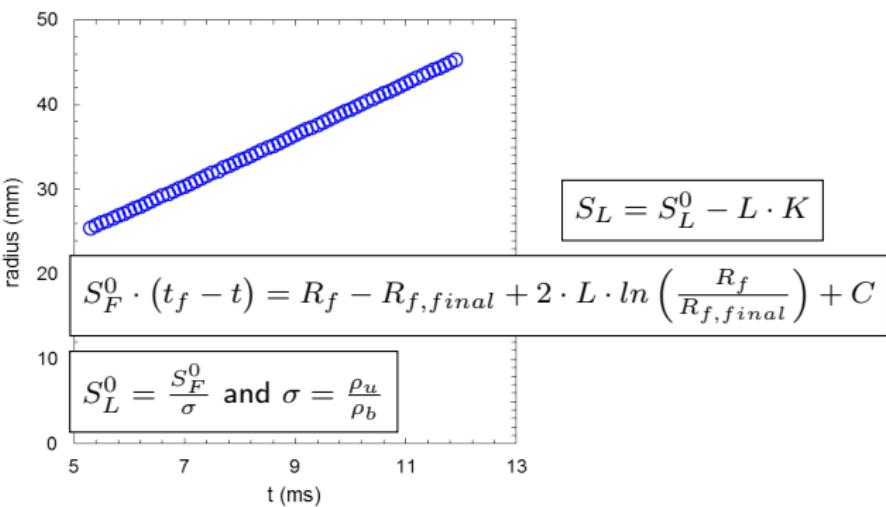


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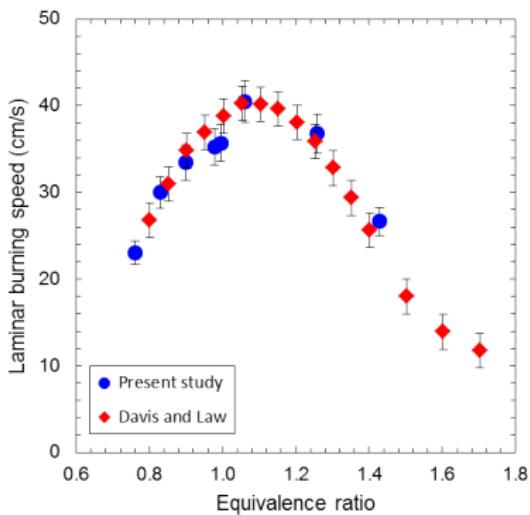


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Validation of Burning Speed Measurements



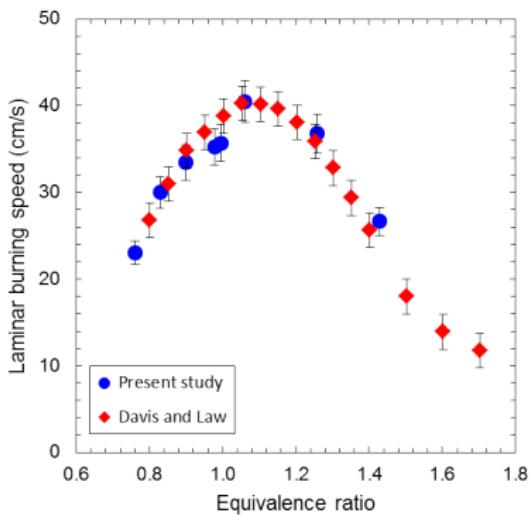
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- Two-tailed z-test ($\phi = 0.8-1.4$)

$H_0 : \mu_1 = \mu_2$ and $H_a : \mu_1 \neq \mu_2$
 μ_1 = present study mean
 μ_2 = Davis and Law mean

- Null hypothesis, H_0 cannot be rejected
- Difference between the two data sets is zero ($\alpha = 0.02$ confidence level)

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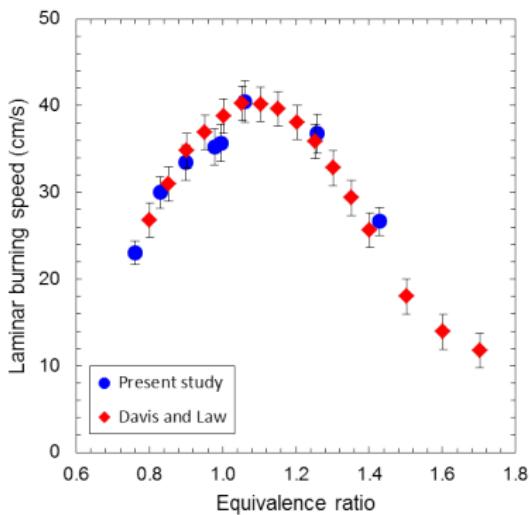
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1. Motivation

2. Previous Work

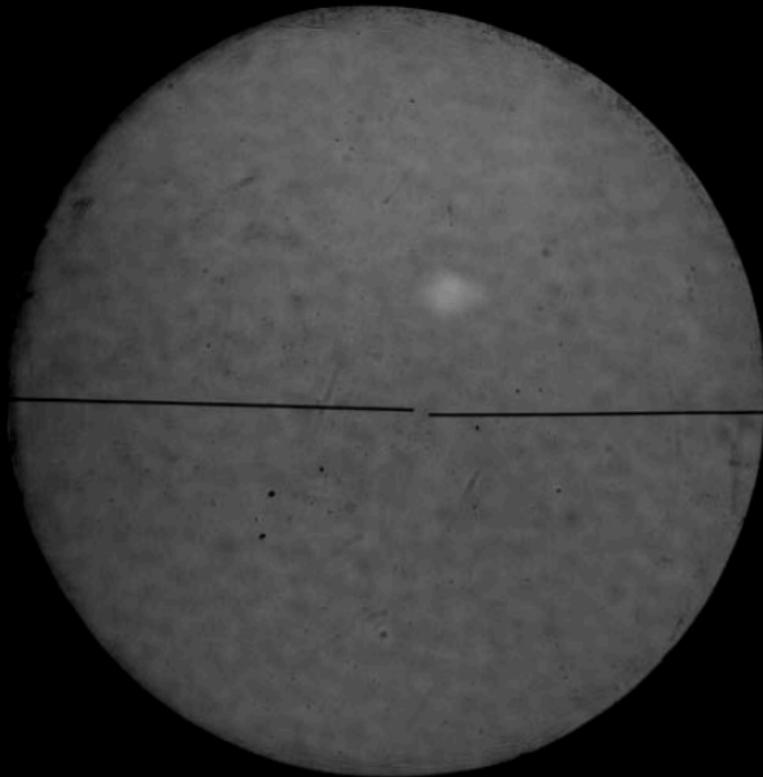
3. Materials and Methods

4. Results

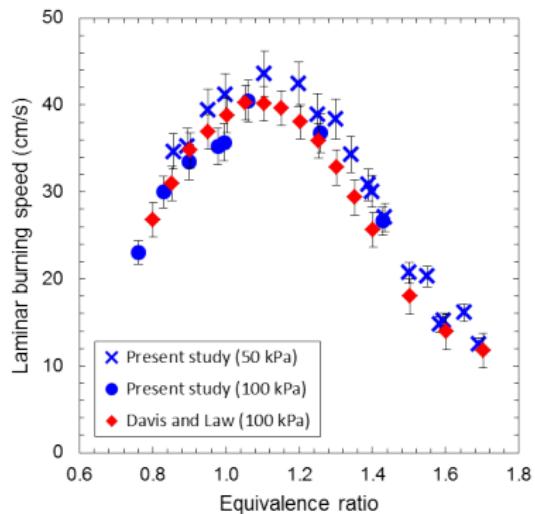
- Experimental Results
- Modeling Results

5. Conclusions

$T_0 = 380$ K, $P_0 = 50$ kPa, $\phi = 1.10$

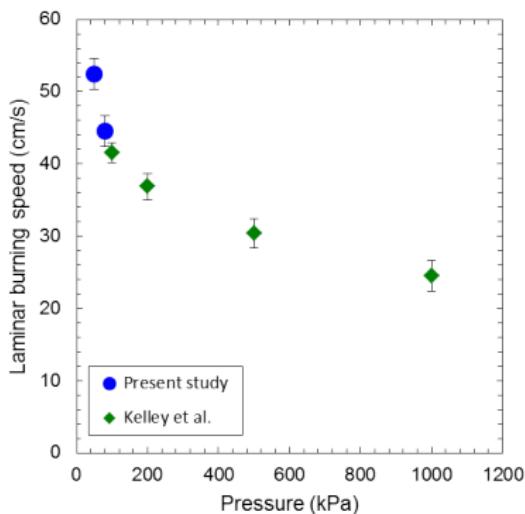


Pressure Effect



$$T_0 = 296 \text{ K}$$

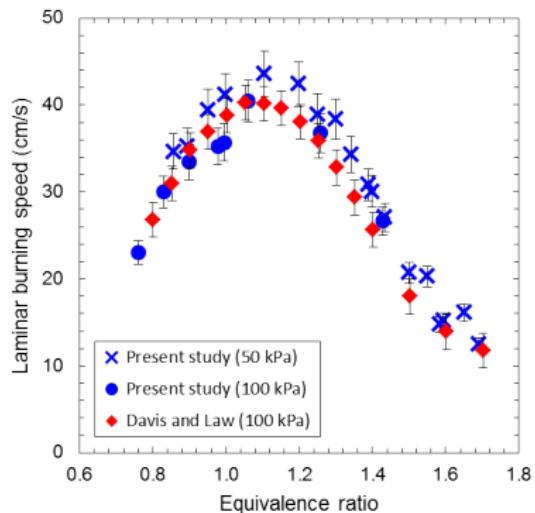
- Uncertainty at 50 kPa $\approx 5\%$
- t-test ($\alpha = 0.2$ confidence level)
 - statistically significant difference



$$T_0 = 353 \text{ K and } \phi = 0.9$$

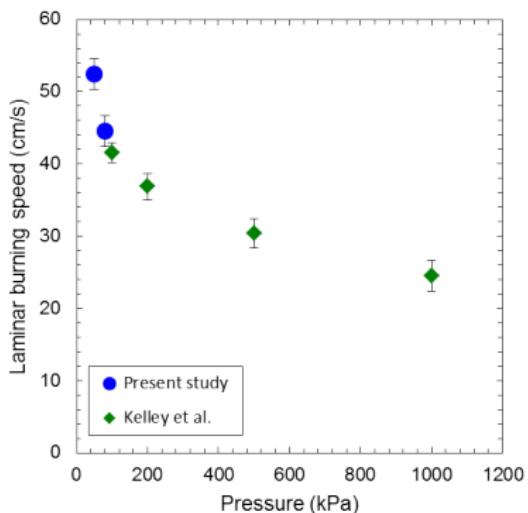
- Decrease in burning speed with increase in pressure
 - increase in the upstream gas density

Pressure Effect



$$T_0 = 296 \text{ K}$$

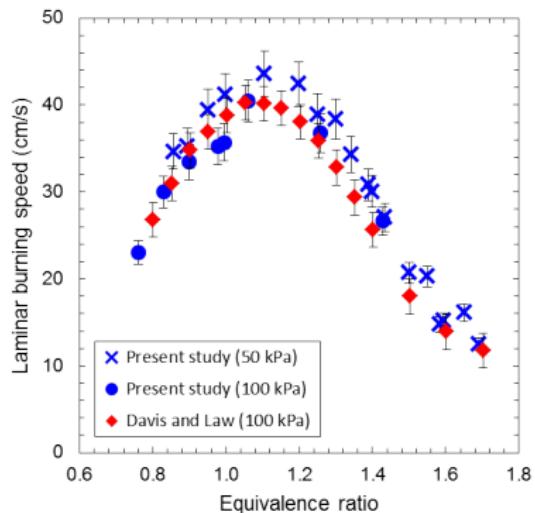
- Uncertainty at 50 kPa $\approx 5\%$
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 - statistically significant difference



$$T_0 = 353 \text{ K and } \phi = 0.9$$

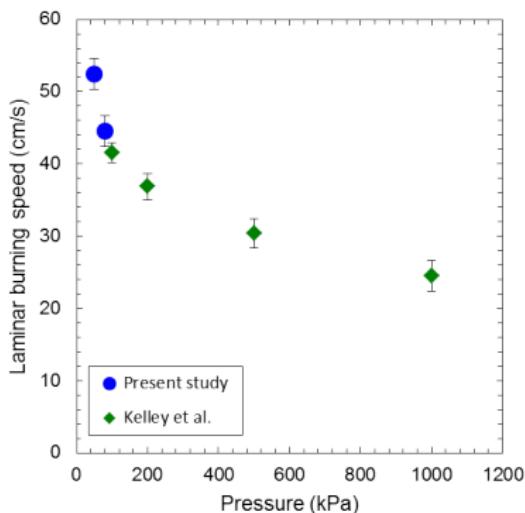
- Decrease in burning speed with increase in pressure
 - increase in the upstream gas density

Pressure Effect



$$T_0 = 296 \text{ K}$$

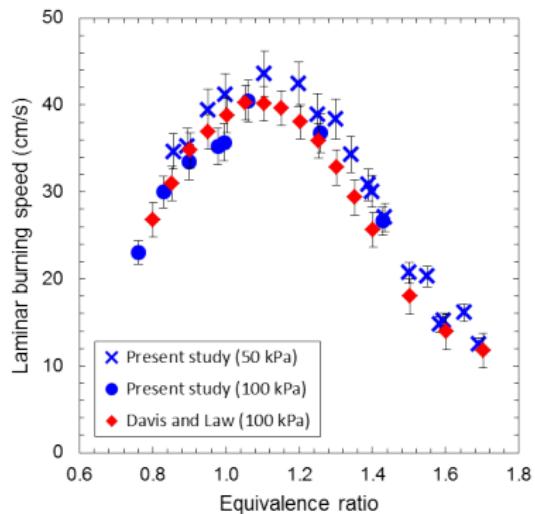
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$$T_0 = 353 \text{ K and } \phi = 0.9$$

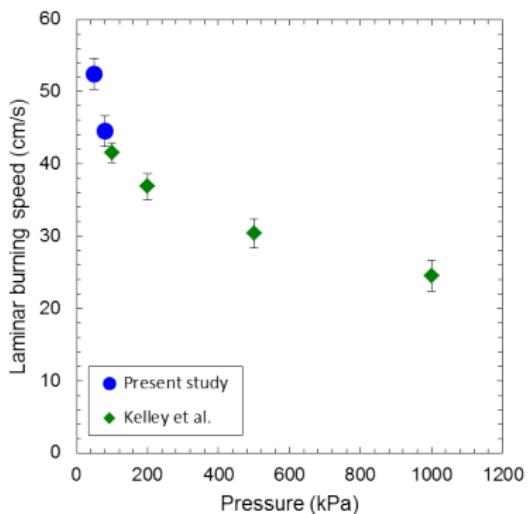
- Decrease in burning speed with increase in pressure
 - increase in the upstream gas density

Pressure Effect



$$T_0 = 296 \text{ K}$$

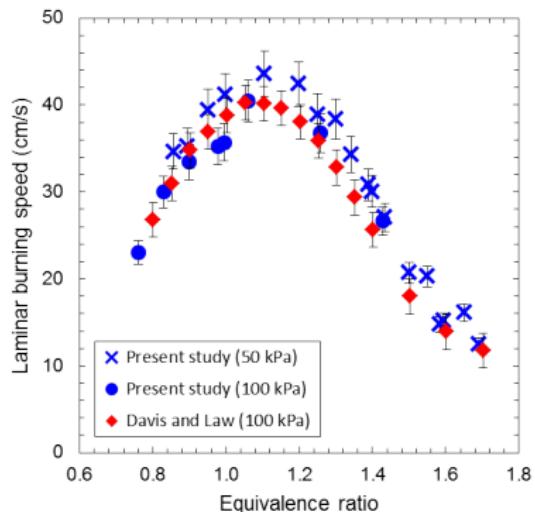
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 - statistically significant difference



$$T_0 = 353 \text{ K and } \phi = 0.9$$

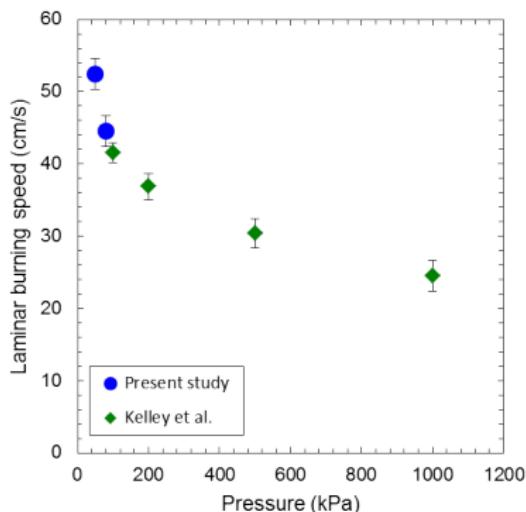
- Decrease in burning speed with increase in pressure
 - increase in the upstream gas density

Pressure Effect



$$T_0 = 296 \text{ K}$$

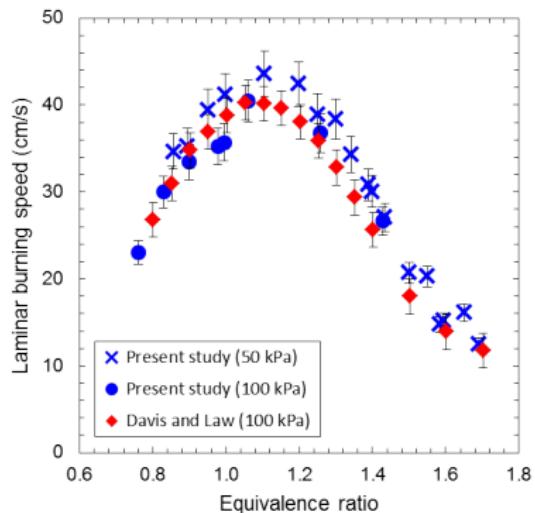
- Uncertainty at 50 kPa $\approx 5\%$
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 - statistically significant difference



$$T_0 = 353 \text{ K and } \phi = 0.9$$

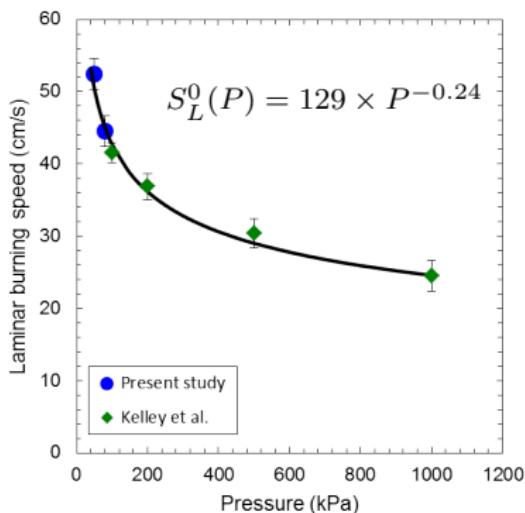
- Decrease in burning speed with increase in pressure
 - increase in the upstream gas density

Pressure Effect



$$T_0 = 296 \text{ K}$$

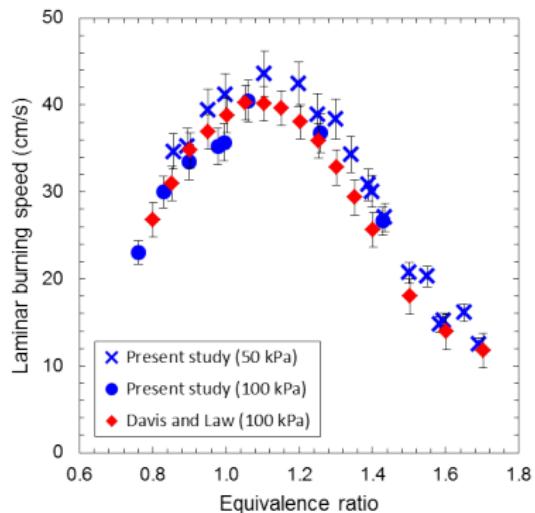
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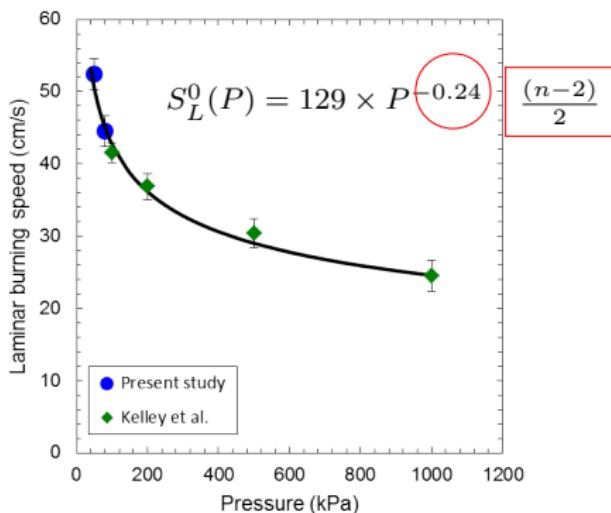
- Decrease in burning speed with increase in pressure
 - increase in the upstream gas density

Pressure Effect



$$T_0 = 296 \text{ K}$$

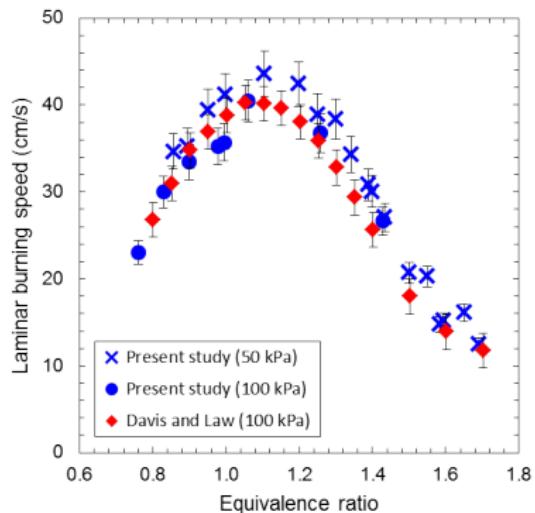
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$$T_0 = 353 \text{ K and } \phi = 0.9$$

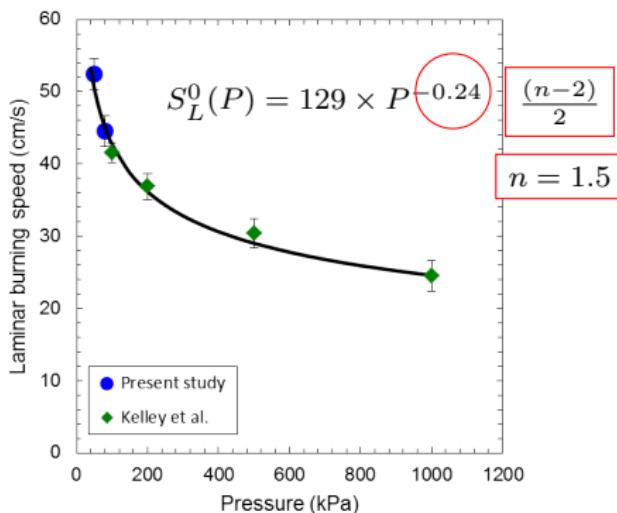
- Decrease in burning speed with increase in pressure
 - increase in the upstream gas density

Pressure Effect



$$T_0 = 296 \text{ K}$$

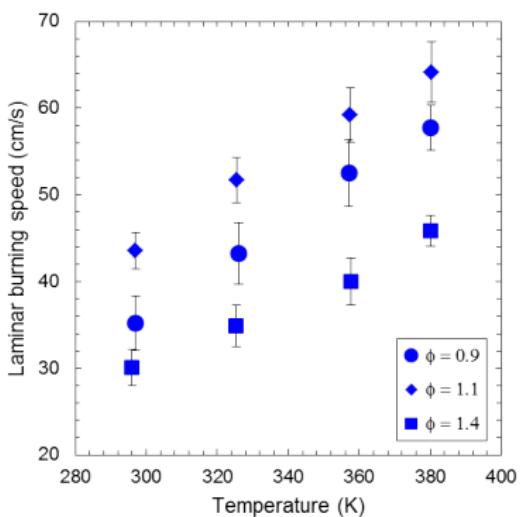
- Uncertainty at 50 kPa $\approx 5\%$
- t-test ($\alpha = 0.2$ confidence level)
 - statistically significant difference



$$T_0 = 353 \text{ K and } \phi = 0.9$$

- Decrease in burning speed with increase in pressure
 - increase in the upstream gas density

Temperature Effect



$$P_0 = 50 \text{ kPa}$$

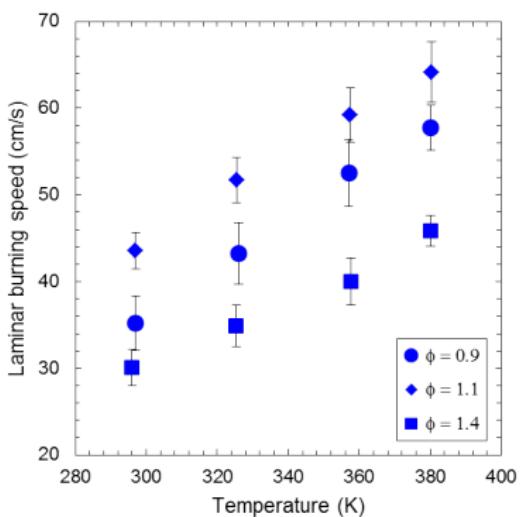
■ From $T_0 = 296\text{--}380 \text{ K}$

- 64% increase at $\phi = 0.9$
- 47% increase at $\phi = 1.1$
- 53% increase at $\phi = 1.4$

■ Rate of burning speed increase with temperature for fixed ϕ

- 0.27 cm/s/K for $\phi = 0.9$
- 0.25 cm/s/K for $\phi = 1.1$
- 0.19 cm/s/K for $\phi = 1.4$

Temperature Effect



$$P_0 = 50 \text{ kPa}$$

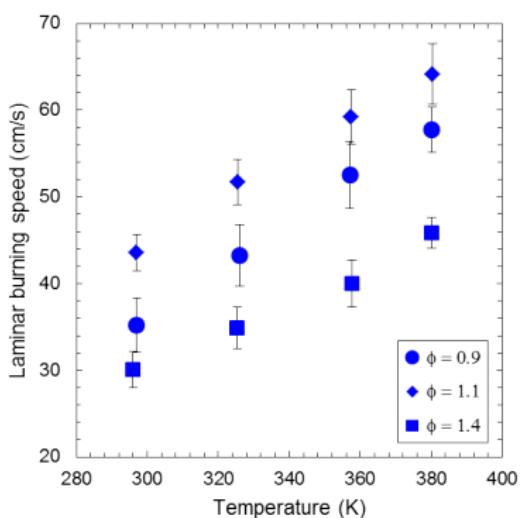
■ From $T_0 = 296\text{--}380 \text{ K}$

- 64% increase at $\phi = 0.9$
- 47% increase at $\phi = 1.1$
- 53% increase at $\phi = 1.4$

■ Rate of burning speed increase with temperature for fixed ϕ

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Temperature Effect



$$P_0 = 50 \text{ kPa}$$

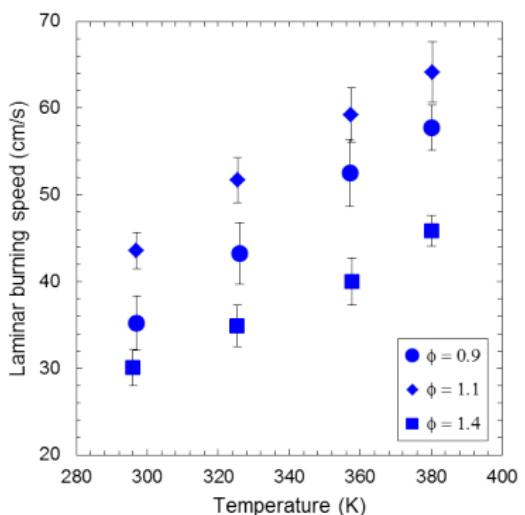
■ From $T_0 = 296\text{--}380 \text{ K}$

- 64% increase at $\phi = 0.9$
- 47% increase at $\phi = 1.1$
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Temperature Effect



$$P_0 = 50 \text{ kPa}$$

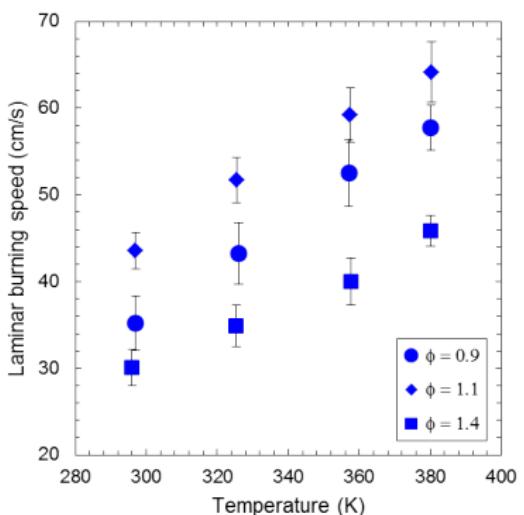
■ From $T_0 = 296\text{--}380 \text{ K}$

- 64% increase at $\phi = 0.9$
- 47% increase at $\phi = 1.1$
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■ Rate of burning speed increase with temperature for fixed ϕ

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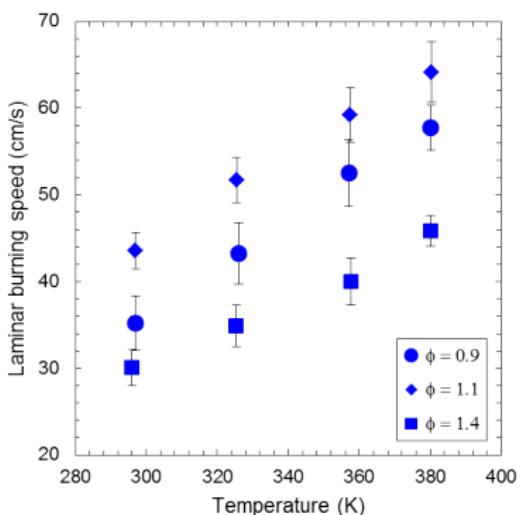
Temperature Effect



$$P_0 = 50 \text{ kPa}$$

- From $T_0 = 296\text{--}380 \text{ K}$
 - 64% increase at $\phi = 0.9$
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 - 0.27 cm/s/K for $\phi = 0.9$
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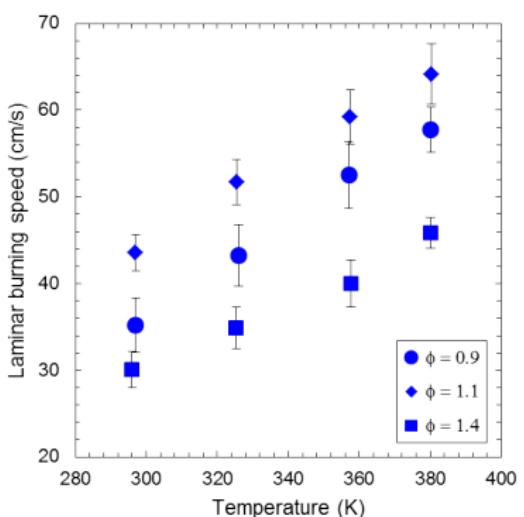
Temperature Effect



$$P_0 = 50 \text{ kPa}$$

- From $T_0 = 296\text{--}380 \text{ K}$
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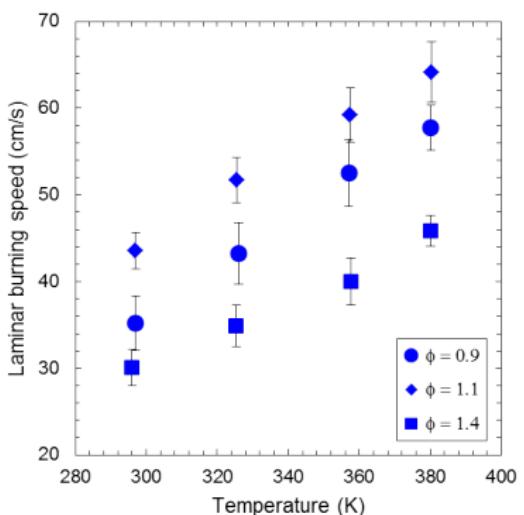
Temperature Effect



$$P_0 = 50 \text{ kPa}$$

- From $T_0 = 296\text{--}380 \text{ K}$
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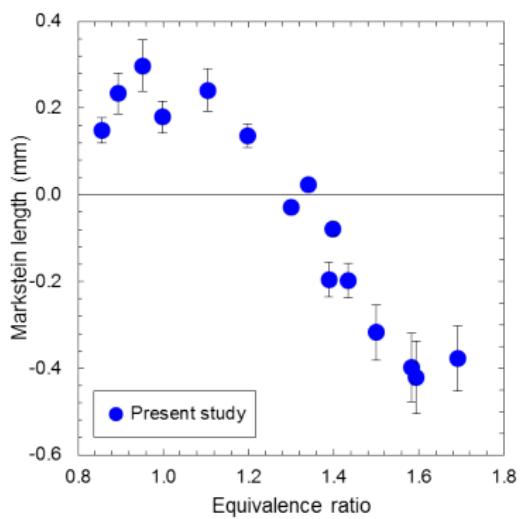
Temperature Effect



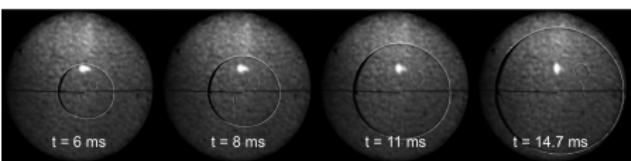
$$P_0 = 50 \text{ kPa}$$

- From $T_0 = 296\text{--}380 \text{ K}$
 - 64% increase at $\phi = 0.9$
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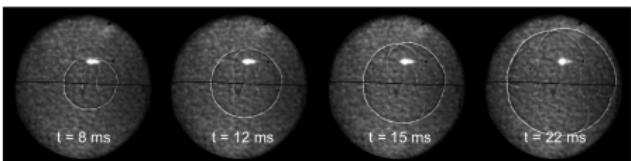
Markstein Length



$T_0 = 296$ K and $P_0 = 50$ kPa

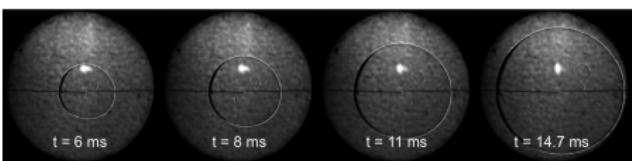
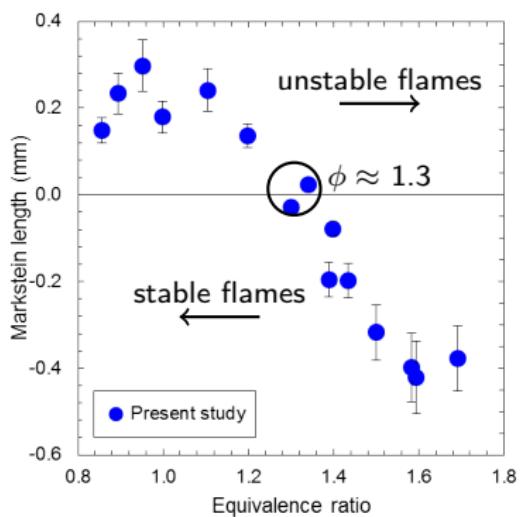
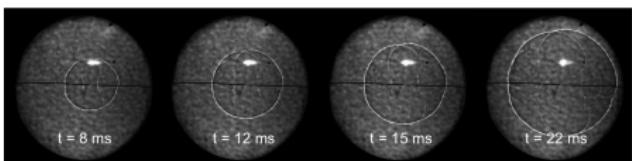


$$\phi = 0.91$$



$$\phi = 1.65$$

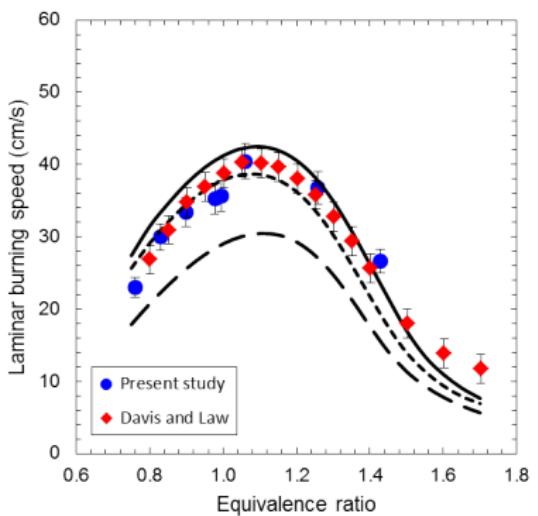
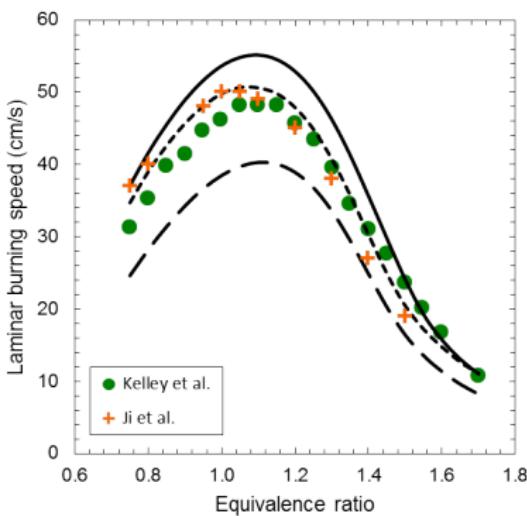
Markstein Length


 $\phi = 0.91$

 $\phi = 1.65$
 $T_0 = 296\text{ K}$ and $P_0 = 50\text{ kPa}$

Reaction Models

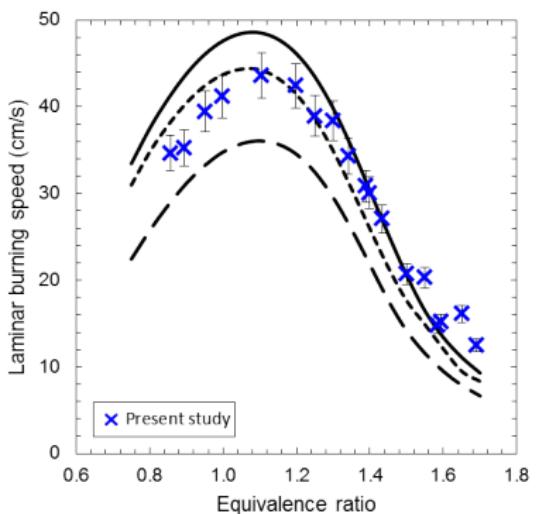
- JetSURF model
 - 2163 reactions
 - 348 species
- Ramirez et al. model
 - 1789 reactions
 - 401 species
- Blanquart (CIT) model
 - 1119 reactions
 - 155 species
- Regath software
 - FORTRAN 90 package
 - thermodynamics and chemical routines
- Results
 - 1D freely propagating flame
 - mixture averaged transport
 - no thermal diffusion

Equivalence Ratio Effect

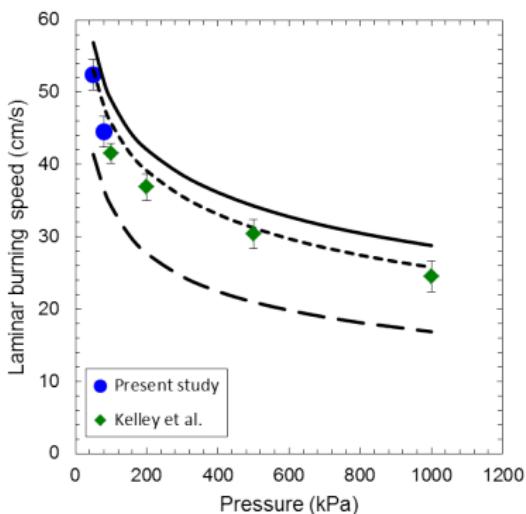

 $T_0 = 296\text{ K}$ and $P_0 = 100\text{ kPa}$

 $T_0 = 353\text{ K}$ and $P_0 = 100\text{ kPa}$

JetSurf : - - - ; Ramirez et al. : - - - ; Blanquart : - - -

Pressure Effect



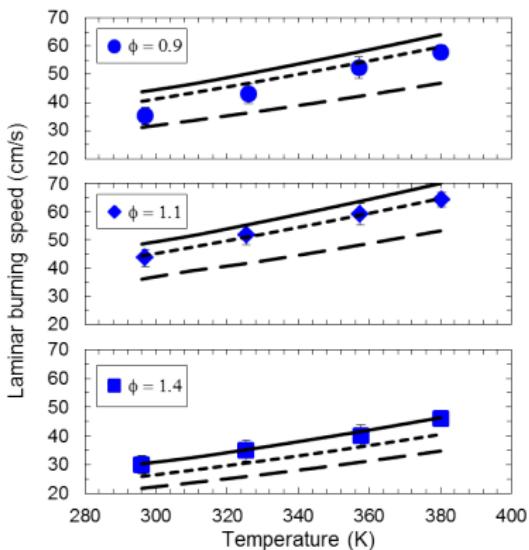
$T_0 = 296 \text{ K}$ and $P_0 = 50 \text{ kPa}$



$T_0 = 353 \text{ K}$ and $\phi = 0.9$

JetSurf : ----- ; Ramirez et al. : - - - ; Blanquart : —————

Temperature Effect



$$P_0 = 50 \text{ kPa}$$

JetSurf : ----- ; Ramirez et al. : - - - ; Blanquart : —————

Summary

1. Motivation

2. Previous Work

3. Materials and Methods

4. Results

5. Conclusions

- Experiments
- Reaction Models

Experimental Conclusions

- Increase in the laminar burning speed from $P_0 = 100 \text{ kPa}$ to $50 \text{ kPa} \rightarrow \alpha = 0.2$ confidence level
- Highest rate of burning speed increase with temperature \rightarrow lean mixtures
- Lowest rate of burning speed increase with temperature \rightarrow rich mixtures
- Pressure dependency agreement with thermal flame theory of Mallard and Le Chatelier $\rightarrow n = 1.5$
- Transition from positive to negative Markstein lengths consistent with Kelley et al. data

Comparison of JetSURF, Ramirez et al., and Blanquart Models

- At $T_0 = 296$ K, the JetSURF model prediction is $<12\%$ at approximately $\phi \leq 1.30$
- At $T_0 = 353$ K, the JetSURF model prediction is $<10\%$ at approximately $\phi \leq 1.45$
- At $T_0 = 296$ K, the Blanquart model prediction is $<12\%$ at $\phi \approx 1.30-1.60$
- At $T_0 = 353$ K, the Blanquart model prediction is $<10\%$ at $\phi \approx 1.45-1.70$
- The Ramirez et al. model systematically underestimates the laminar burning speed

Acknowledgements

The present work was carried out in the Explosion Dynamics Laboratory of the California Institute of Technology and was supported by The Boeing Company through a Strategic Research and Development Relationship Agreement CT-BA-GTA-1.

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Thank You