

Statistical Analysis of Spark Ignition of Kerosene-Air Mixtures

S. A Coronel¹, S. P. M. Bane², P. A. Boettcher¹
and J. E. Shepherd¹

¹*California Institute of Technology*, ²*Purdue University*

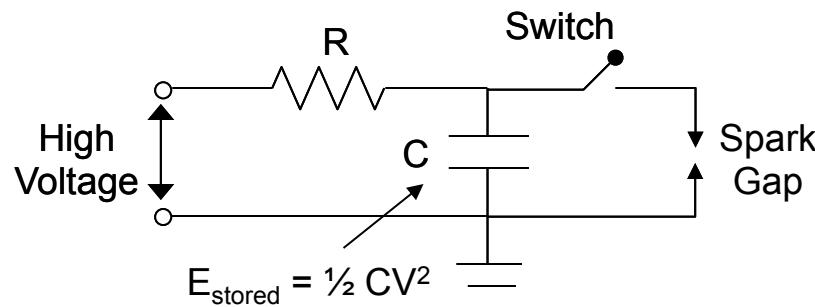
Fall Technical Meeting of the Western States Section of the
Combustion Institute

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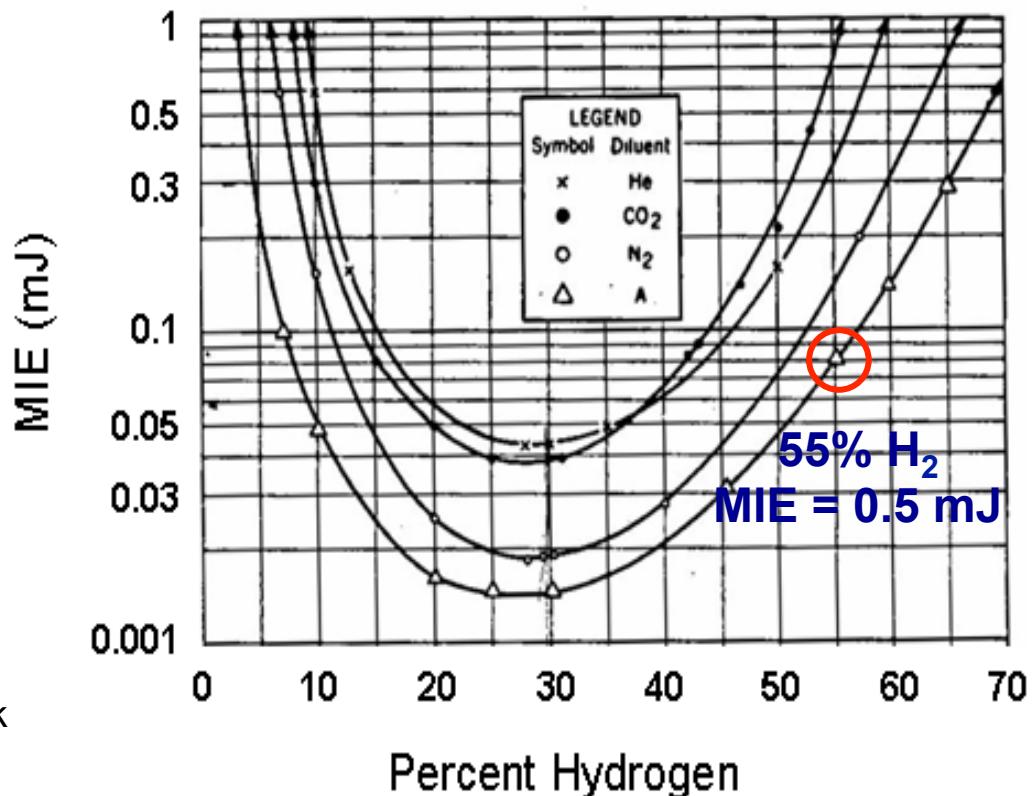


Spark Ignition and Minimum Ignition Energy

- Risk of accidental ignition in industry and aviation
- **Minimum Ignition Energy** (MIE): traditional basis for quantifying ignition hazards
- Capacitive spark discharge as ignition source
- Pioneering work – Blanc, Guest, Lewis & von Elbe at Bureau of Mines (1940s)

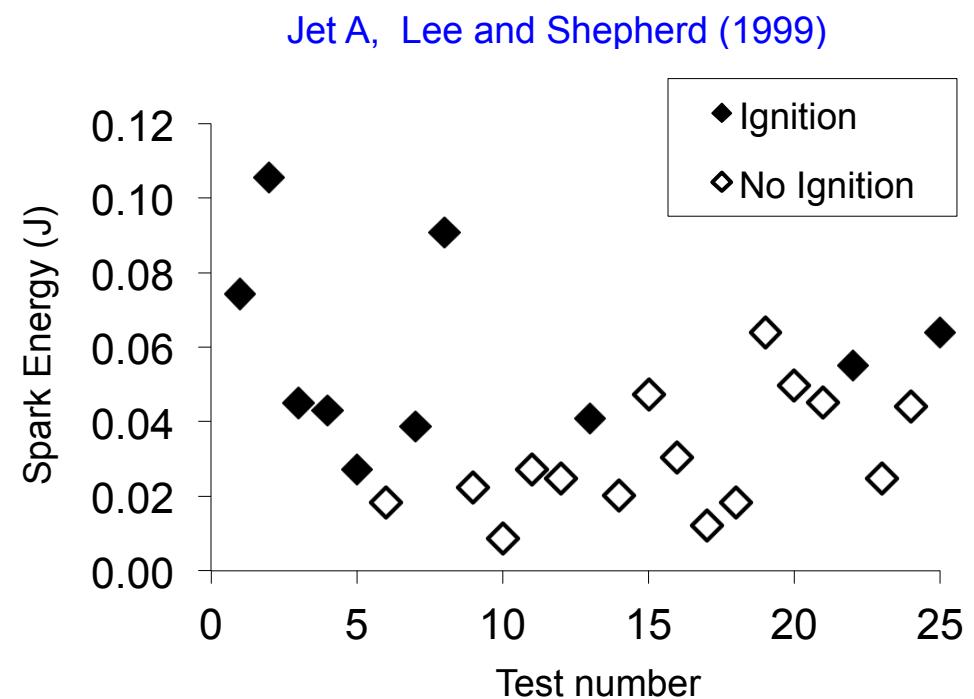
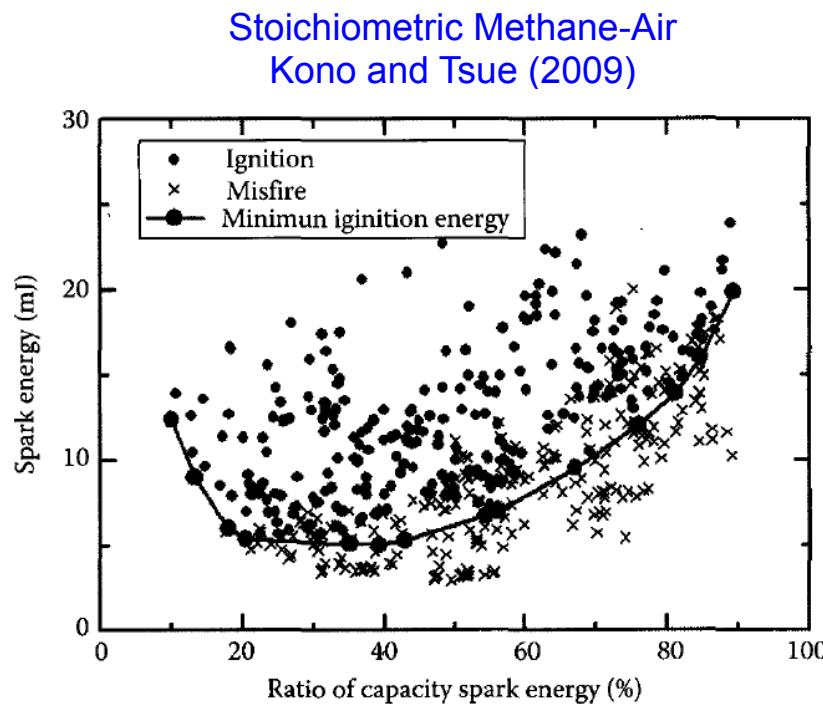


MIE curves for hydrogen mixtures,
Lewis and von Elbe (1961)



Statistical Analysis of Ignition Test Data

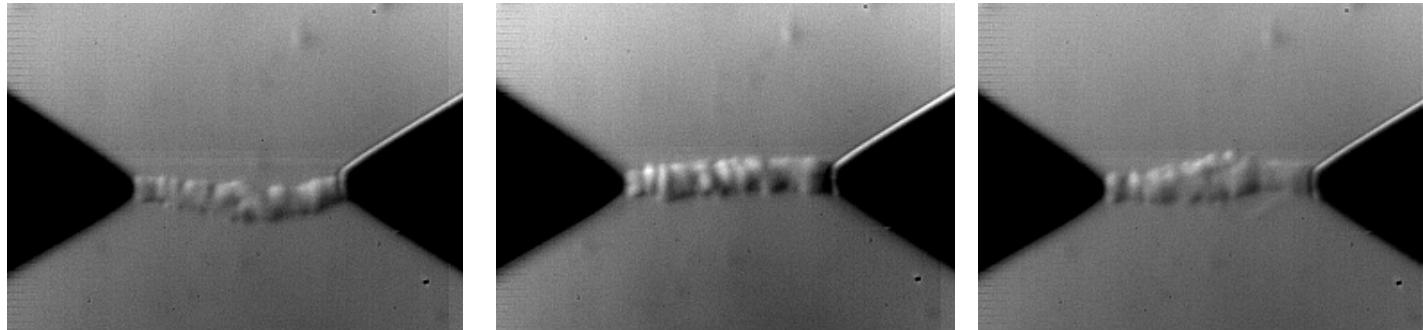
- **New viewpoint** – ignition as statistical phenomenon
- More consistent with test data
- Little work done on statistics of ignition of other flammable mixtures
- Can't assign a probability to historical MIE data



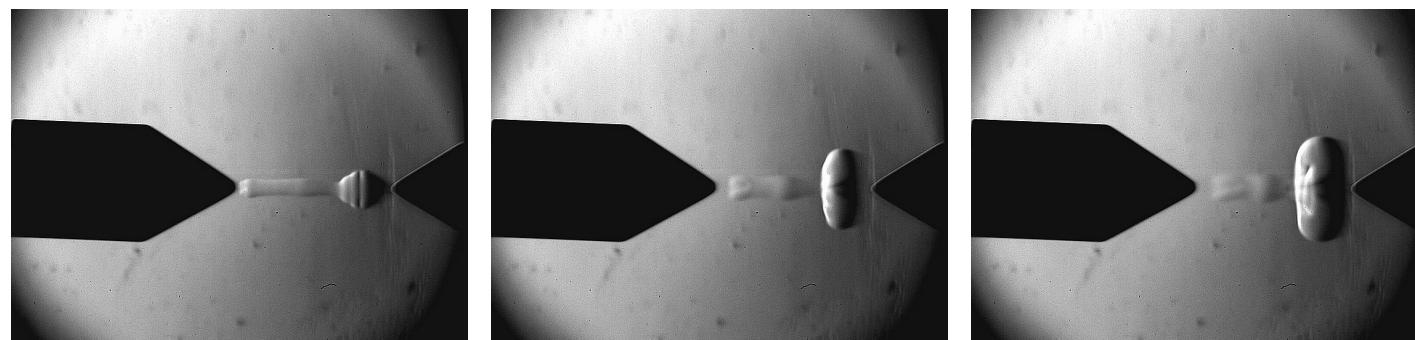
Spark Breakdown and Spark Channel Formation



Unpredictable Plasma Instabilities



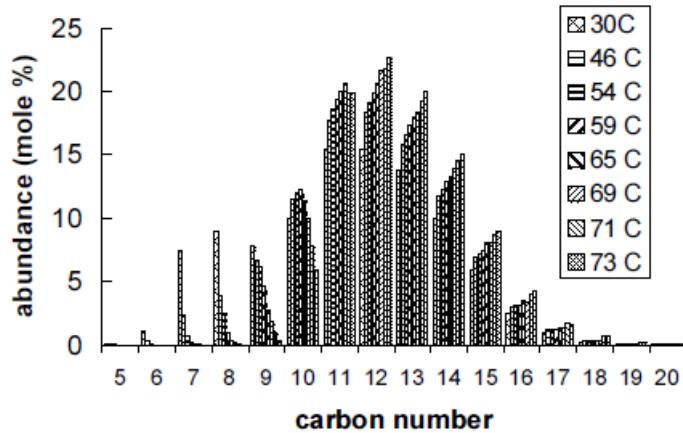
Localized Ignition



Kerosene Tests: Experimental Considerations

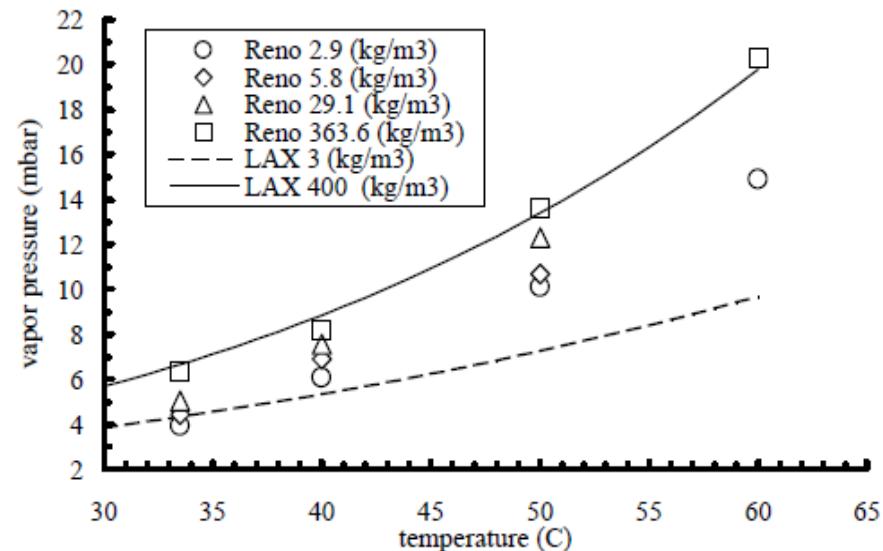
Ignition hazard in aircraft is due to much more complex fuels, that is a kerosene-based fuel

- Ignition testing: low vapor pressure must heat significantly or decrease pressure
- vapor pressure depends on many variables



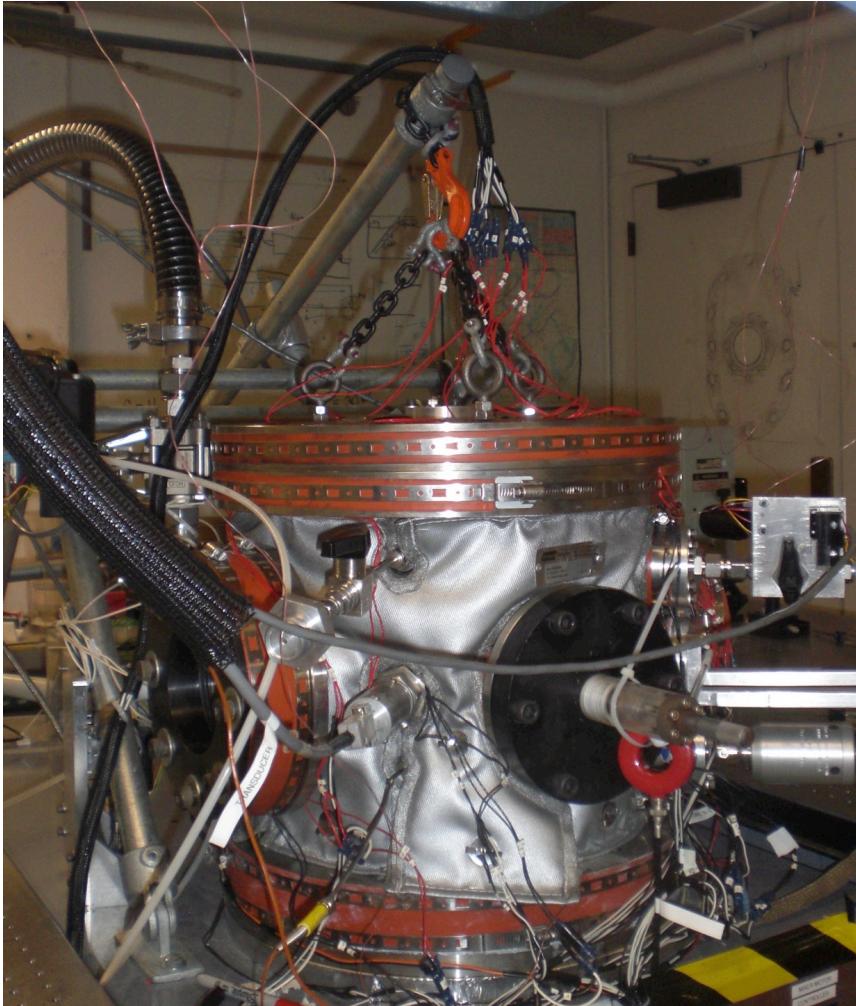
Gas chromatograph (Woodrow, 2000)

Fuel vapor pressure vs. temperature and fuel mass loading (Lee and Shepherd, 1999)



- Liquid fuel composition usually not known for commercial fuels → several 100 components
- composition changes from batch to batch, can be affected by history, transport, etc.
- composition of liquid not the same as fuel vapor

Experimental Setup



22 L, stainless steel, cylindrical combustion vessel

✧ Ignition Detection

- flame visualization
- pressure transducer
- thermocouple

✧ Schlieren visualization

- high-speed camera (10,000+ frames per second)

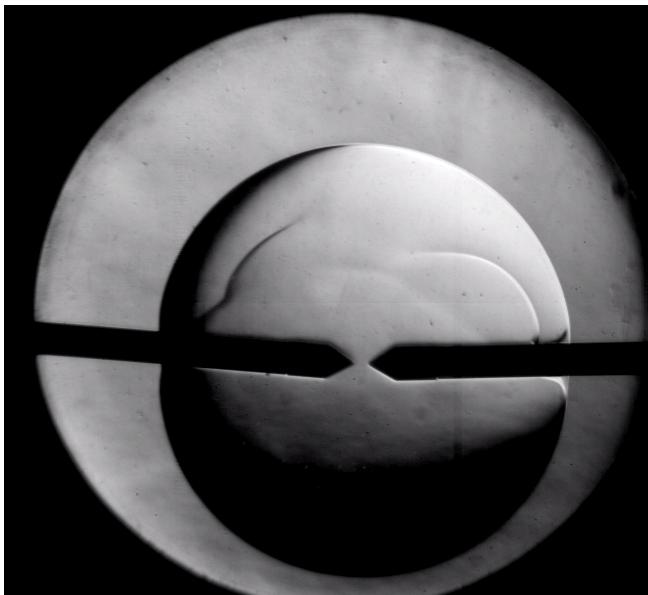
✧ Vessel Heating System

- silicone heaters, 4 zones
- high-current heater control unit
- up to $\sim 150^{\circ}\text{C}$



Results: Kerosene Ignition

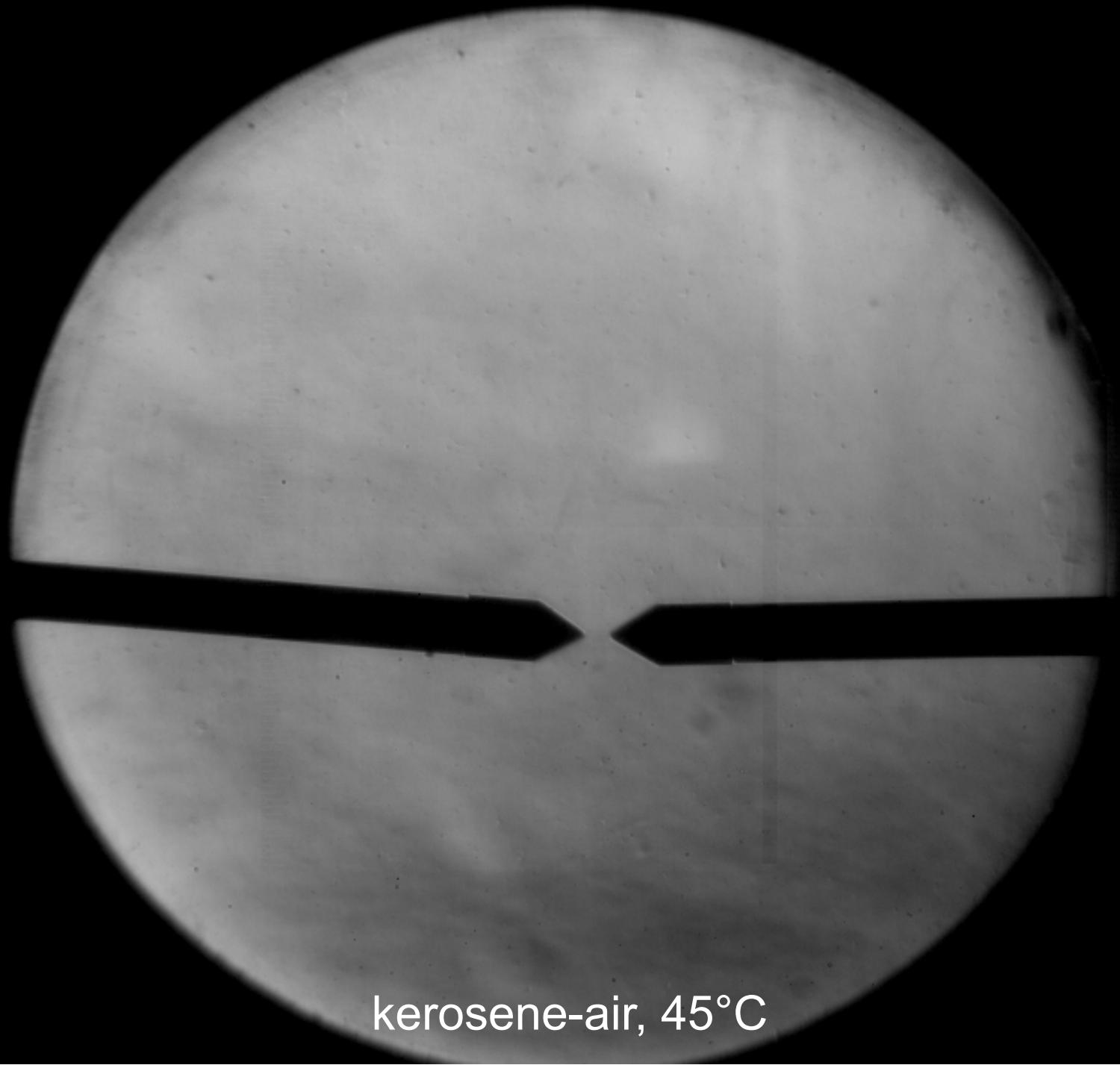
- 1-K kerosene at 45-62°C, 100 kPa
- fixed 3.3 mm spark gap
- 50 kg/m³ fuel mass loading
- C ~ 11 – 68 pF, V ~ 6.4 – 11.4 kV → E_{spark} ~ 0.3 – 2.3 mJ



kerosene-air, 45°C



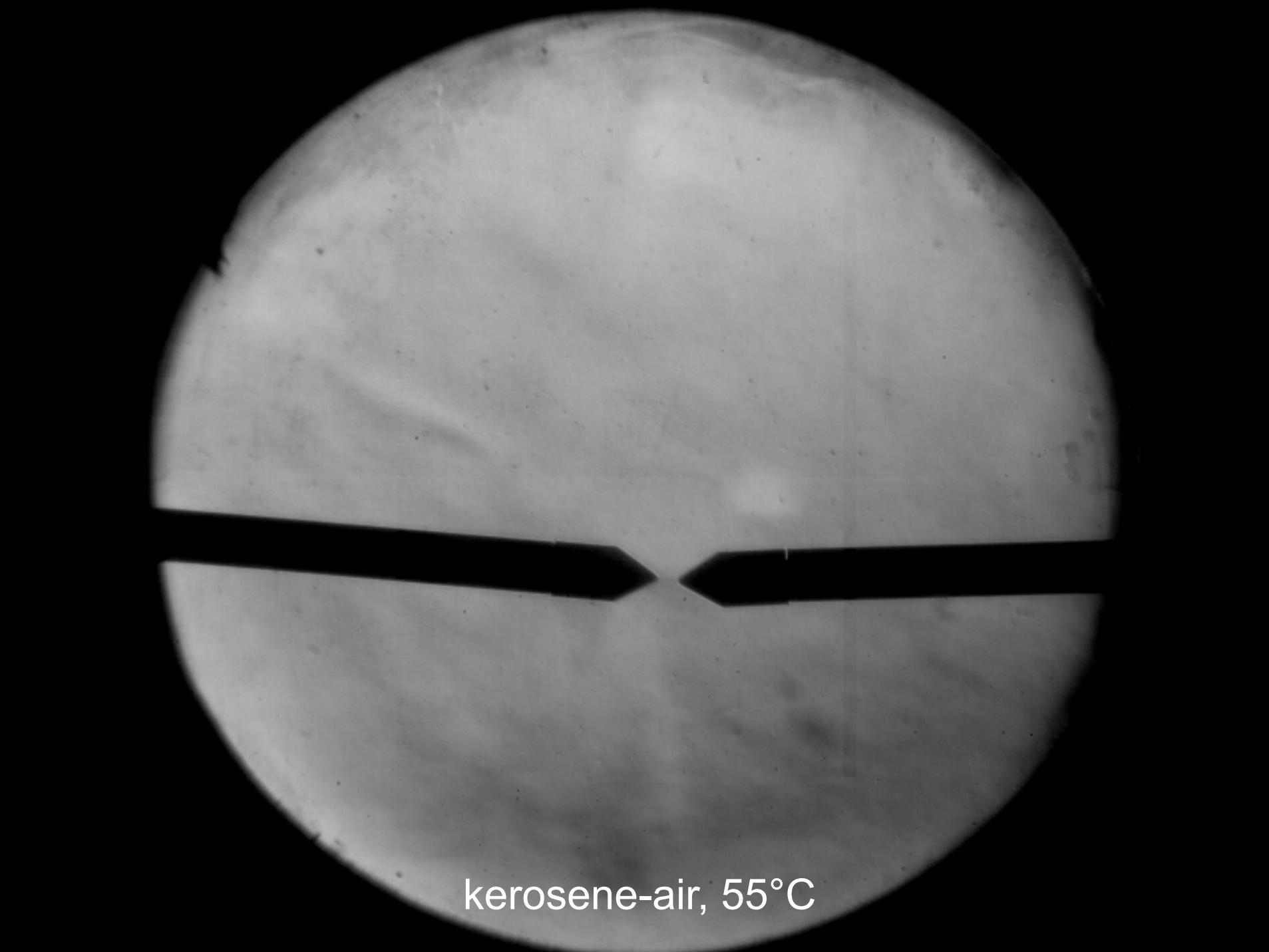
kerosene-air, 55°C



kerosene-air, 45°C



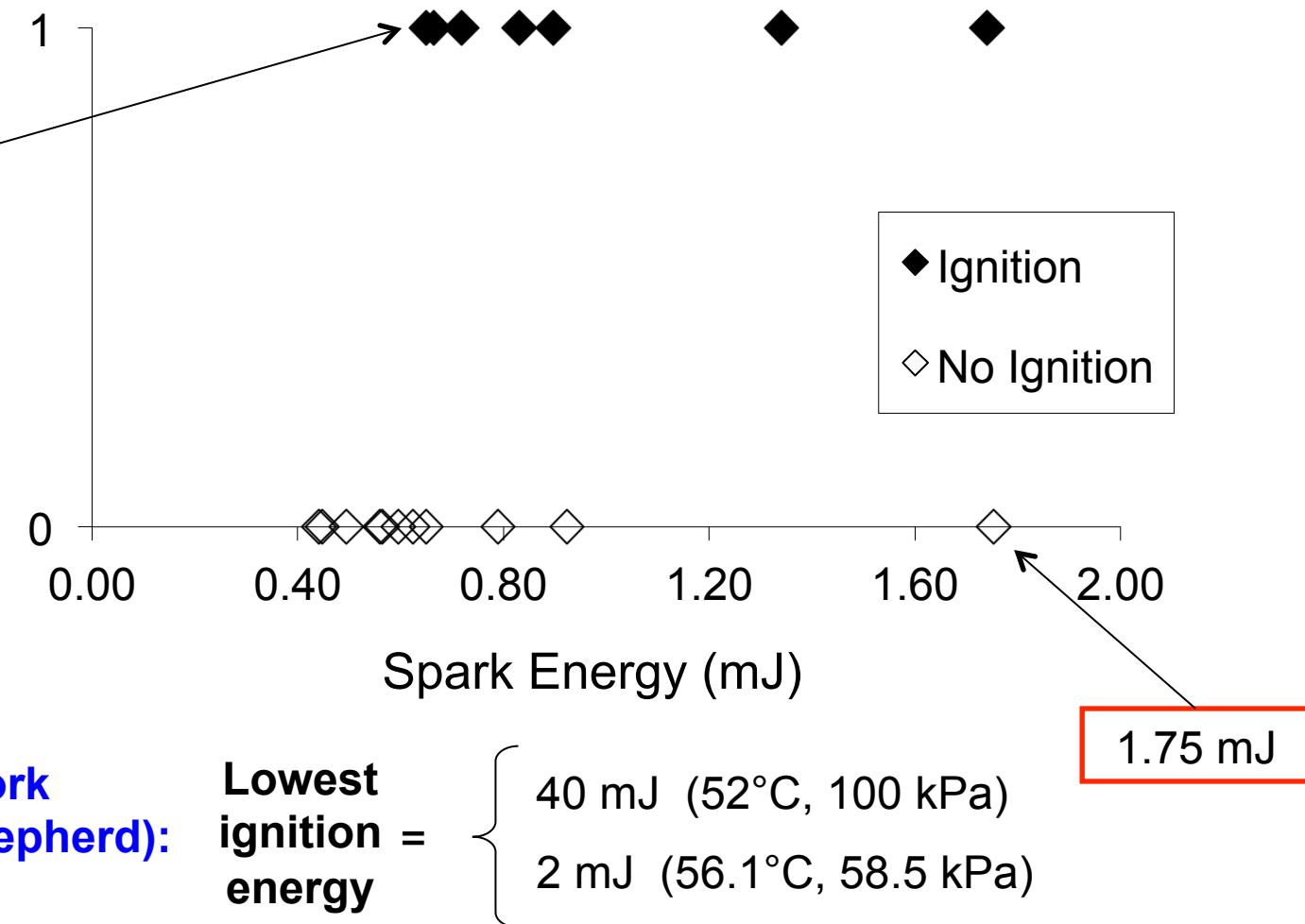
kerosene-air, 50°C



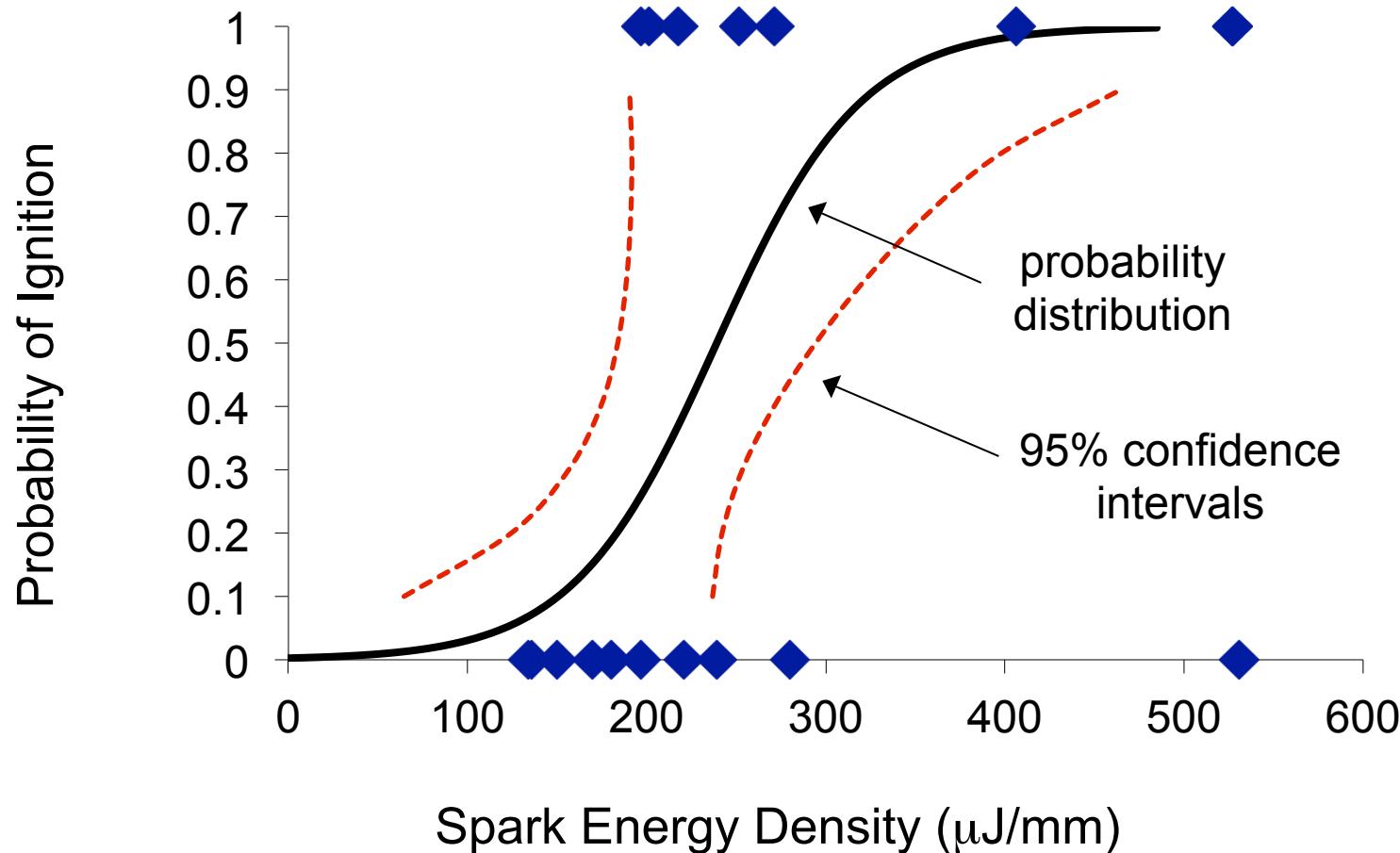
kerosene-air, 55°C



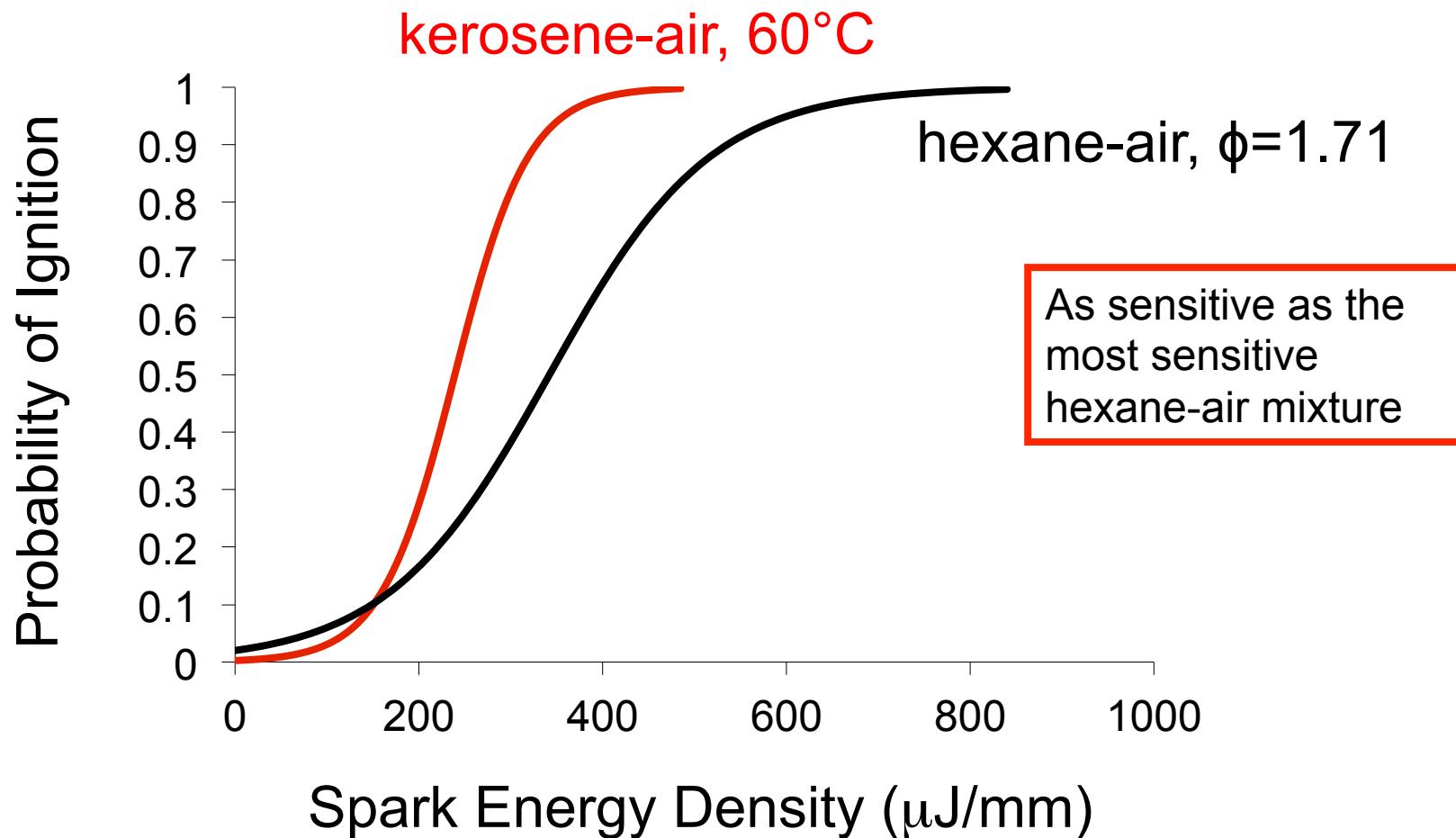
Results: Kerosene-Air at 60°C



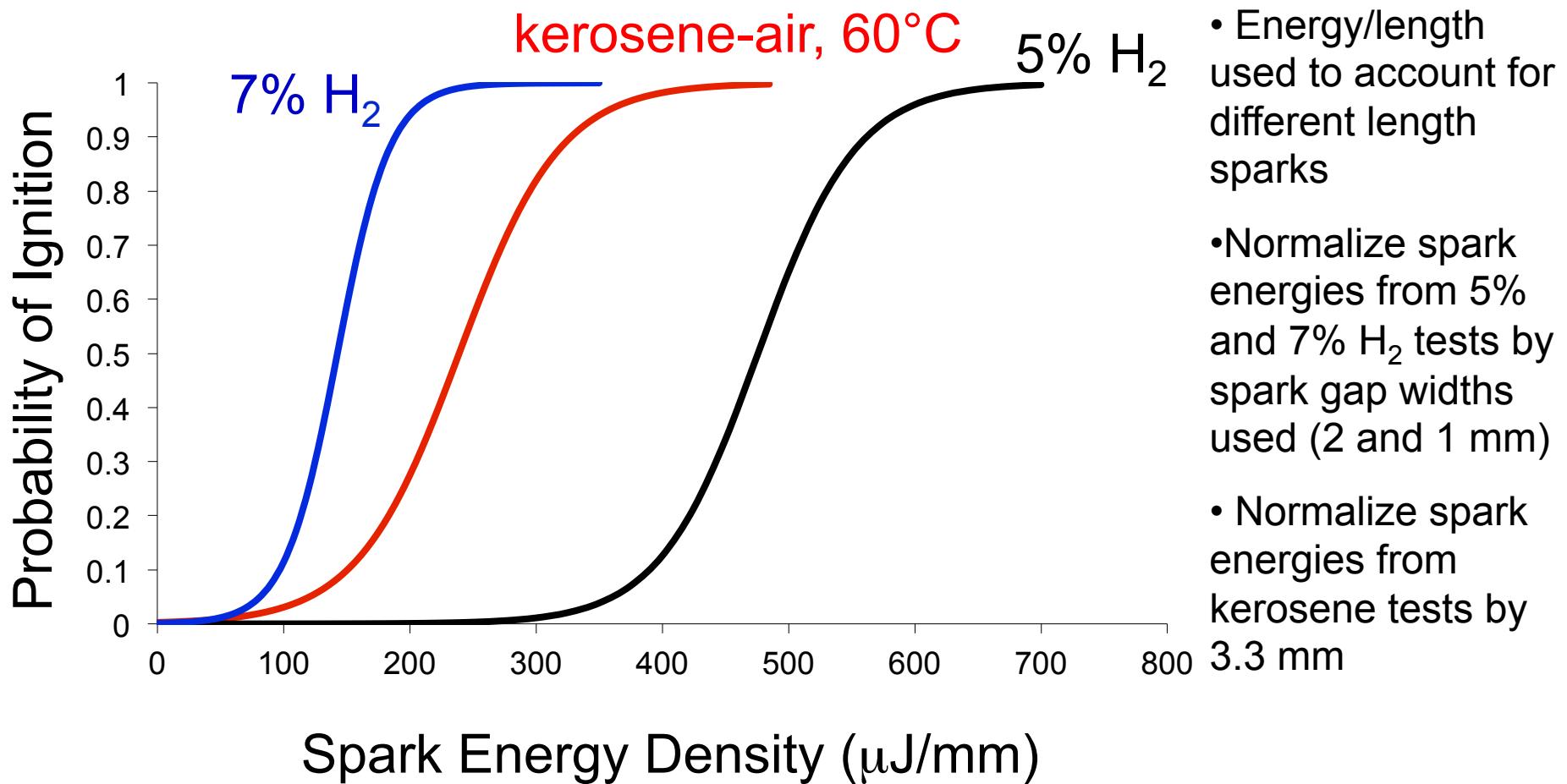
Results: Kerosene-Air at 60°C



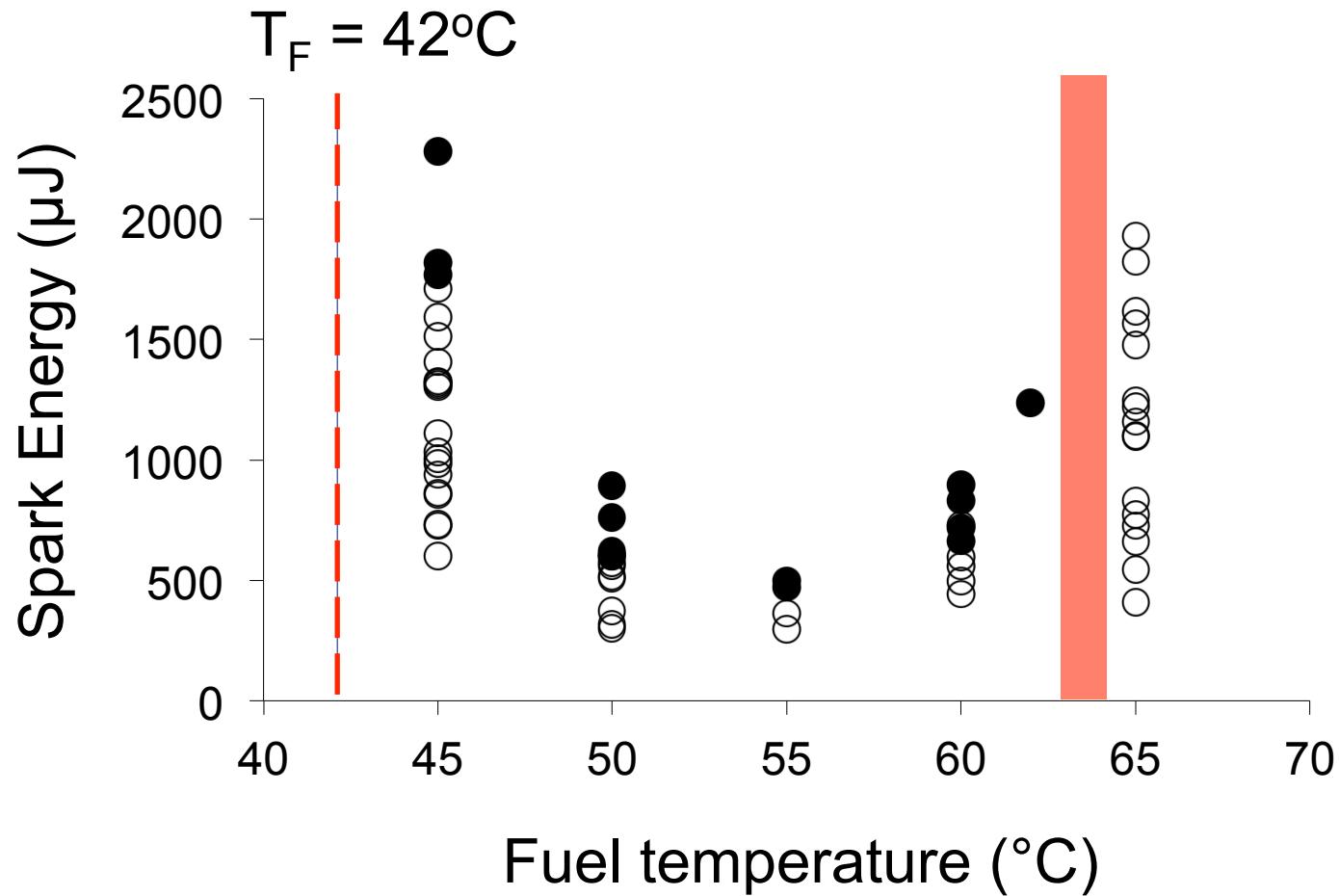
Results: Comparison with Hexane



Results: Comparison with H₂



Results: Varying Kerosene Fuel Temperature





Parametric Study

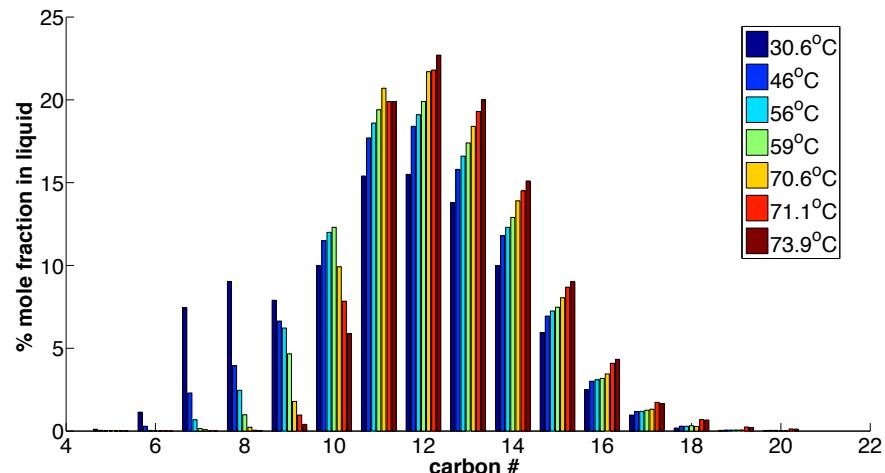
- Ignition energy model
- Mixture composition is required
- Raoult's Law

$$P_i = X_i P_{vp,i} \text{ for } i = 5, 6, \dots, 20$$

liquid composition

saturation pressure

- Obtain X_i through a distillation curve or gas chromatography



Gas chromatograph (Woodrow, 2000)



Parametric Study

- Raoult's law does not take into consideration the fuel mass loading
- "headspace equation" → conservation of moles
- K is the hydrocarbon liquid-vapor distribution coefficient

$$C_G = \frac{C_L^o}{K + V_V/V_L}$$

initial liquid composition

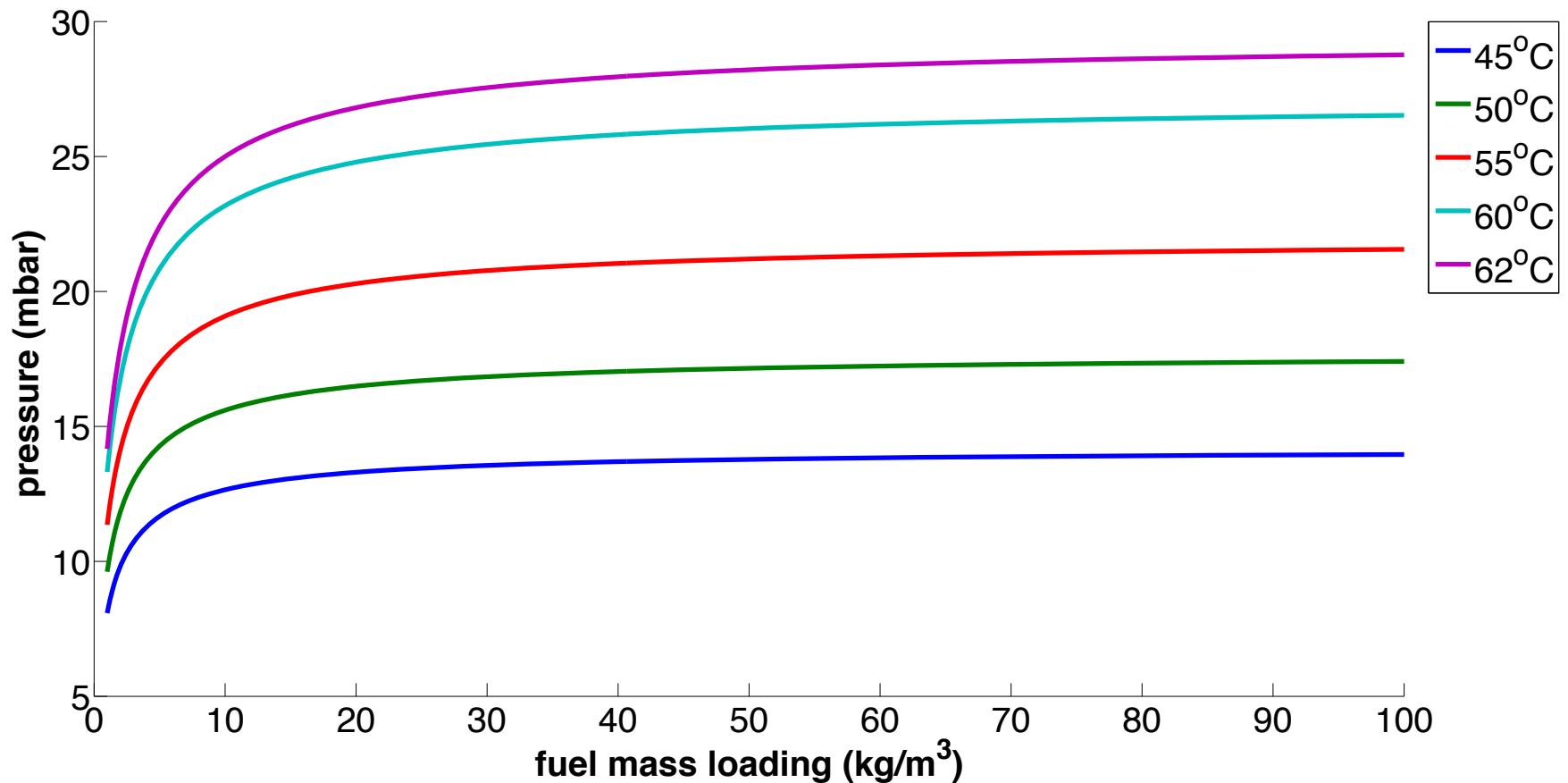
$$K_i = \frac{N}{W_L} \frac{RT}{P_{vp,i}}$$

fuel vapor composition

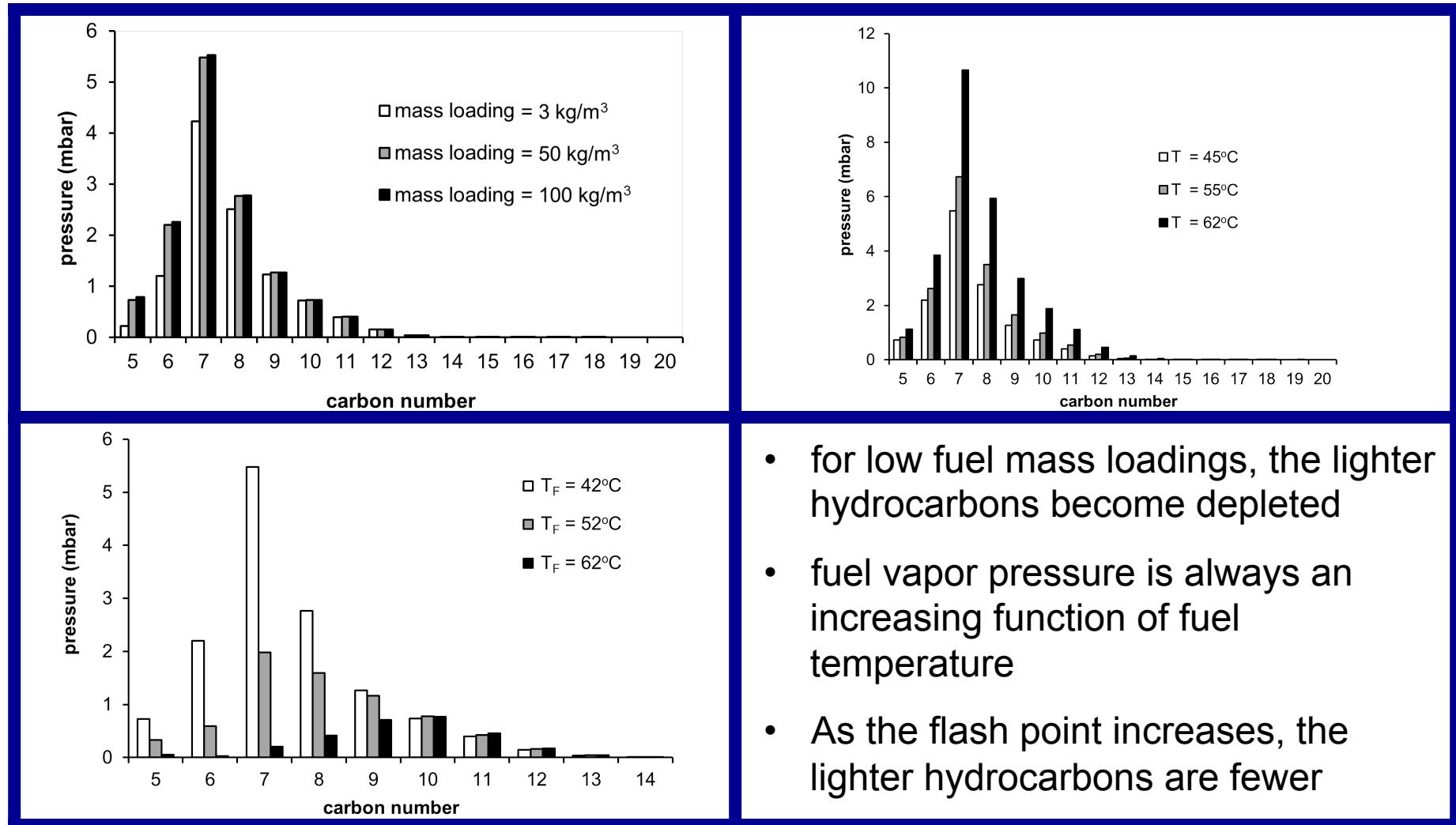


Parametric Study Results

- Effect of fuel mass loading on pressure for a kerosene-based fuel



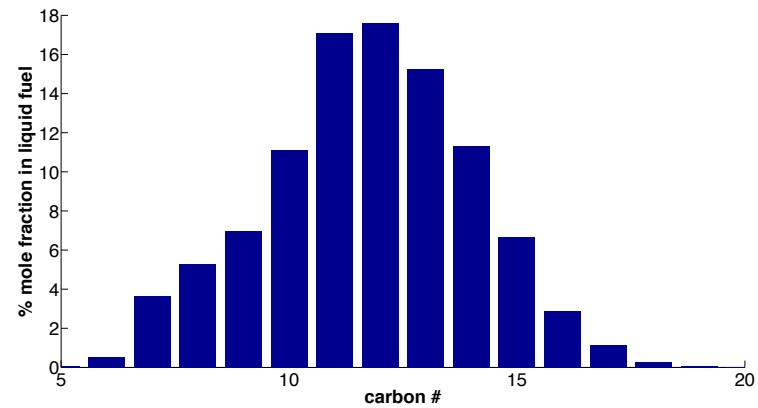
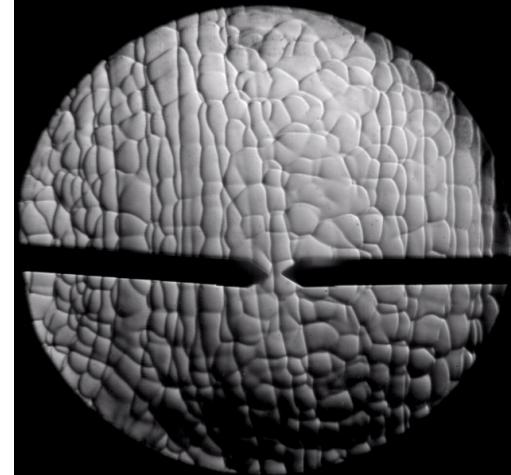
Parametric Study Results





Conclusions

- Electrostatic spark ignition tests using kerosene over a range of temperatures at atmospheric pressure.
- Statistical analysis of probability of ignition vs. spark energy density at 60°C
- Kerosene-air mixture has a lower ignition energy at 60°C than H₂-O₂-Ar mixture previously used in aircraft certification (5% H₂)
- Model to obtain fuel vapor composition given changes in flash point, temperature, pressure and fuel mass loading
- From vapor composition predictions, lighter mass hydrocarbons are more appropriate for low temperature fuel vapor surrogates.
 - ❖ The experimental work was carried out in the Explosion Dynamics Laboratory of the California Institute of Technology and was supported by the Boeing Company through a Strategic Research and Development Relationship Agreement CT-BA-GTA-1



Questions

