



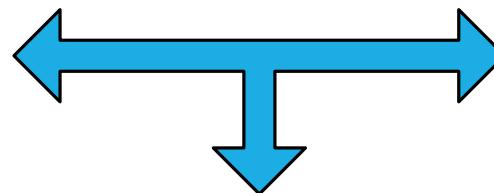
DEFLECTION OF BEAMS

MECHANICS OF MATERIAL
(SOLID MECHANICS)

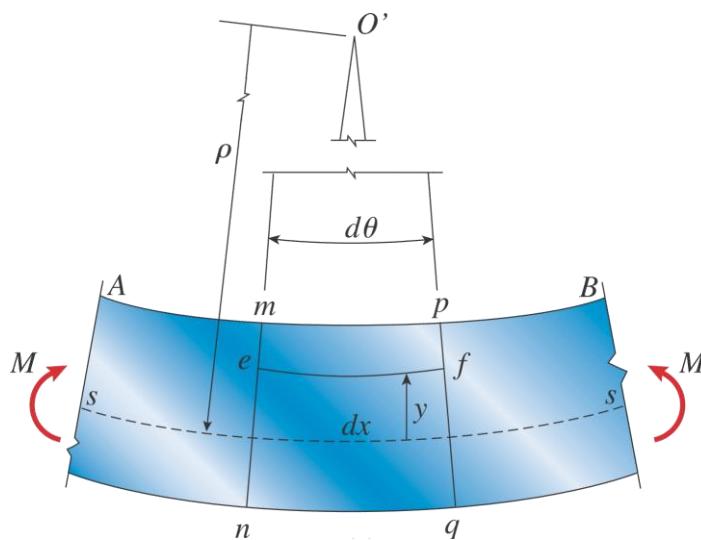


PURE BENDING: FLEXURE FORMULA

$$\sigma_x = -E\kappa y$$

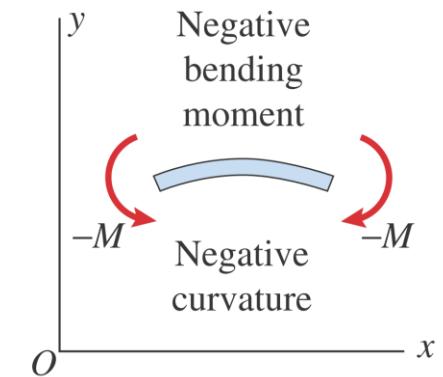
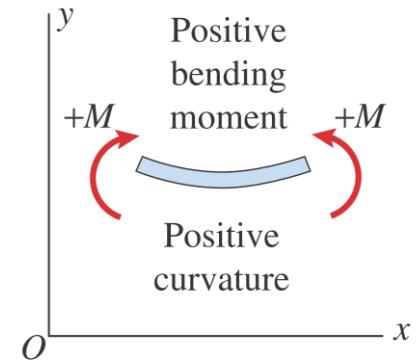


$$\kappa = \frac{M}{EI}$$



$$\sigma_x = -\frac{My}{I}$$

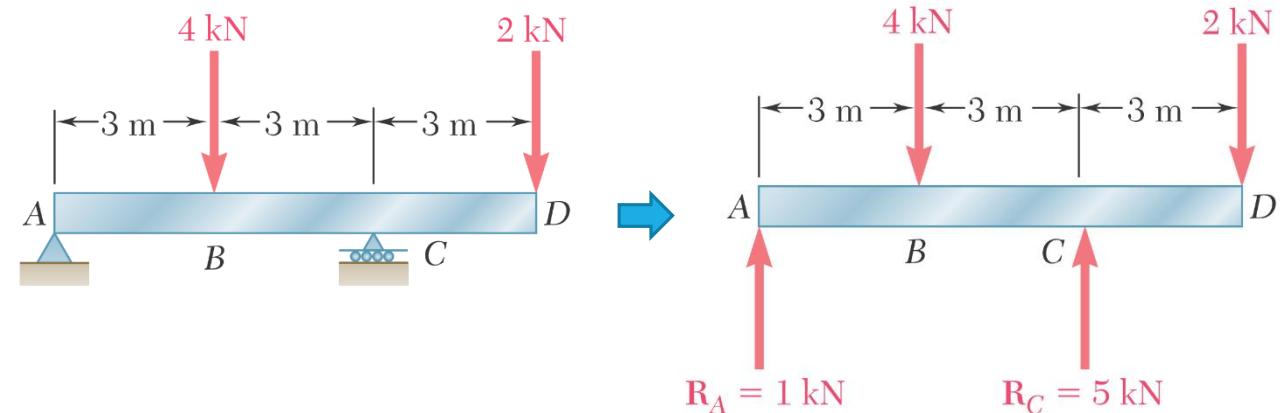
FLEXURE FORMULA!



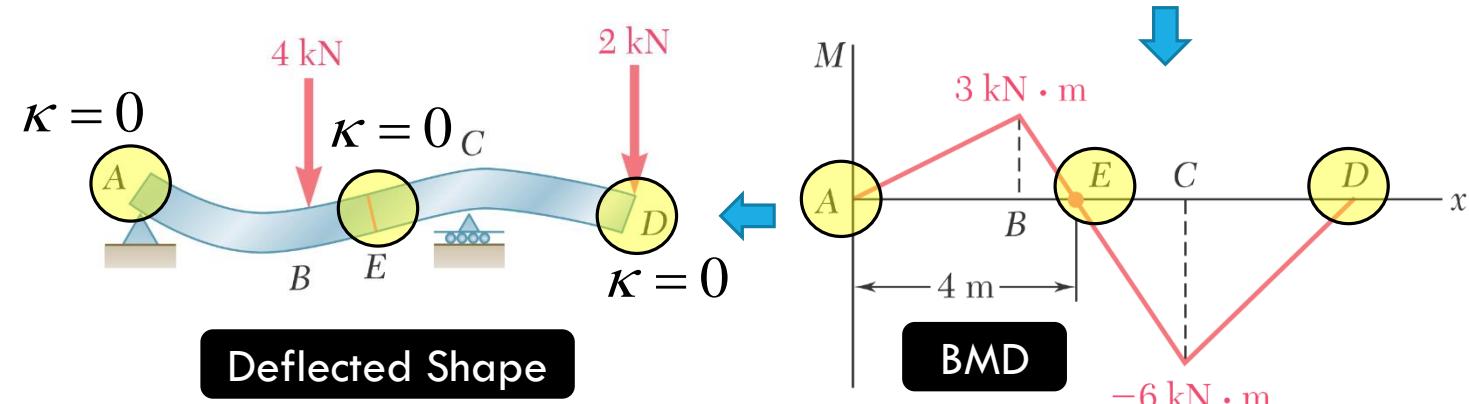
CURVATURE AND MOMENT RELATIONSHIP

- For a general case of loading, bending moment M varies from one section to another

$$\therefore \kappa = \frac{1}{\rho} = \frac{M(x)}{EI}$$



- Zero curvature (κ) at points of zero bending moment

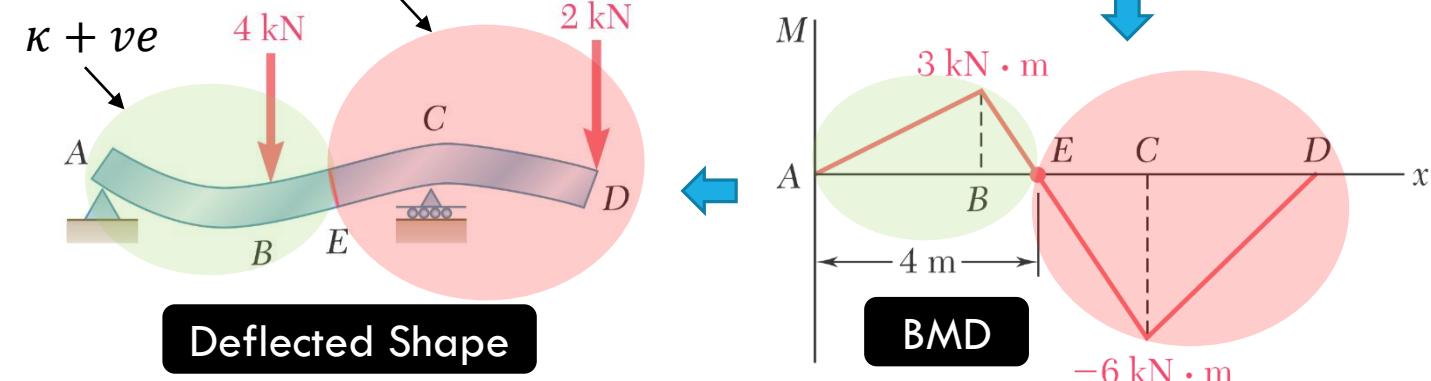
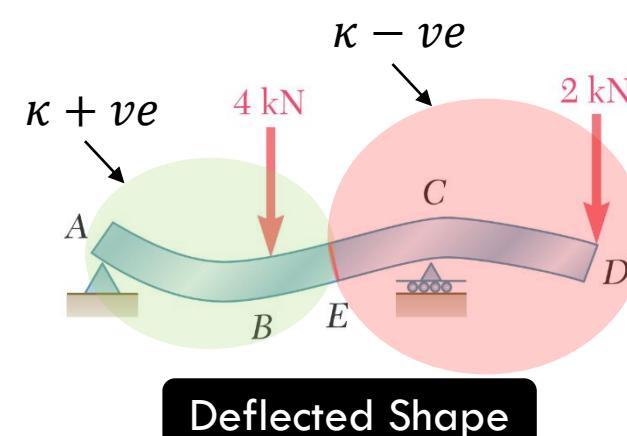
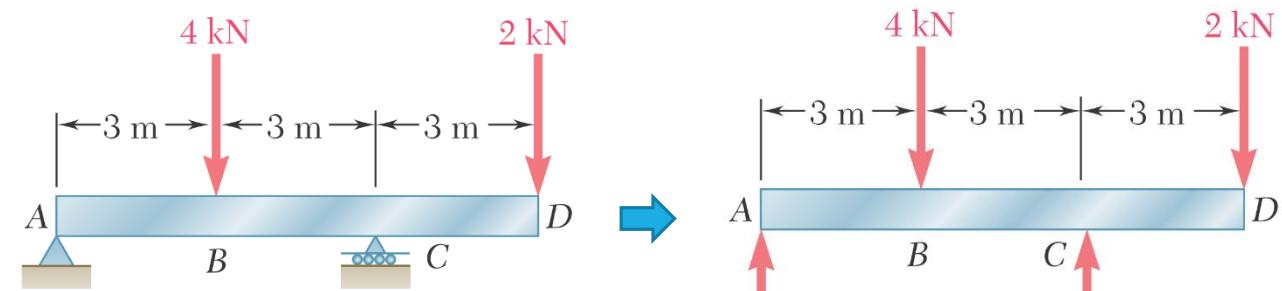


CURVATURE AND MOMENT RELATIONSHIP

- For a general case of loading, bending moment M varies from one section to another

$$\therefore \kappa = \frac{1}{\rho} = \frac{M(x)}{EI}$$

- Zero curvature (κ) at points of zero bending moment
- Beam is concave upward where M and κ is +ve
- Beam is concave downward where M and κ is -ve

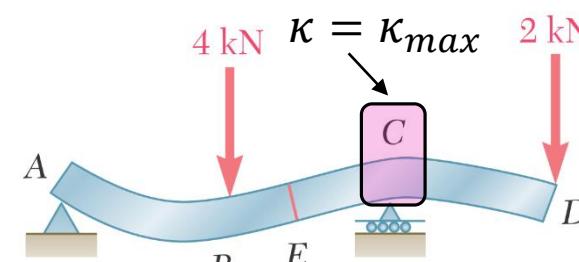
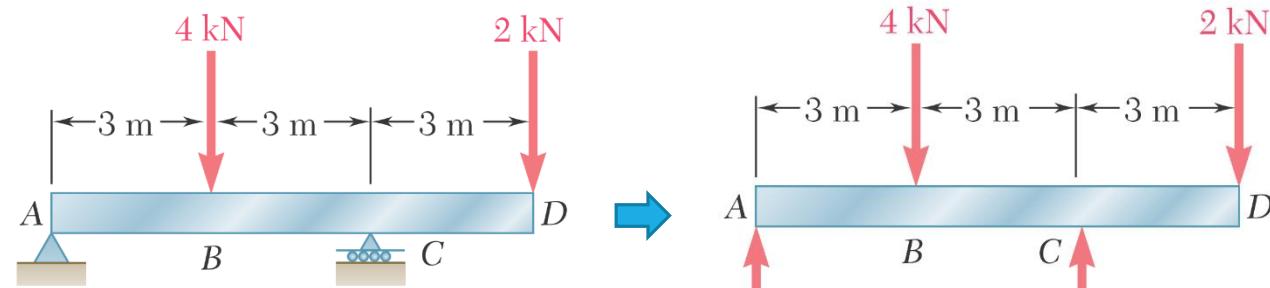


CURVATURE AND MOMENT RELATIONSHIP

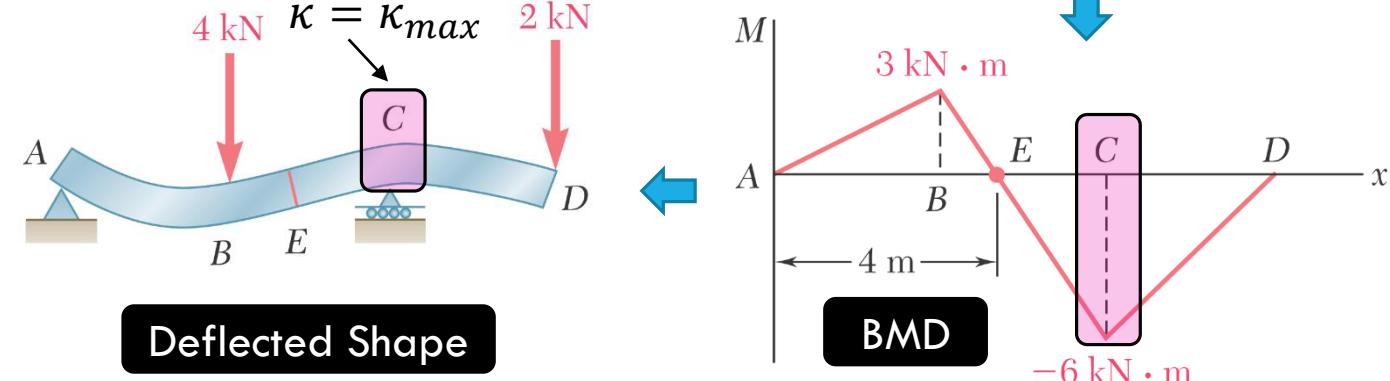
- For a general case of loading, bending moment M varies from one section to another

$$\therefore \kappa = \frac{1}{\rho} = \frac{M(x)}{EI}$$

- Zero curvature (κ) at points of zero bending moment
- Beam is concave upward where M and κ is +ve
- Beam is concave downward where M and κ is -ve
- Maximum curvature occurs where absolute M is max.
- Curvature helps establish a sense of shape of the beam



Deflected Shape



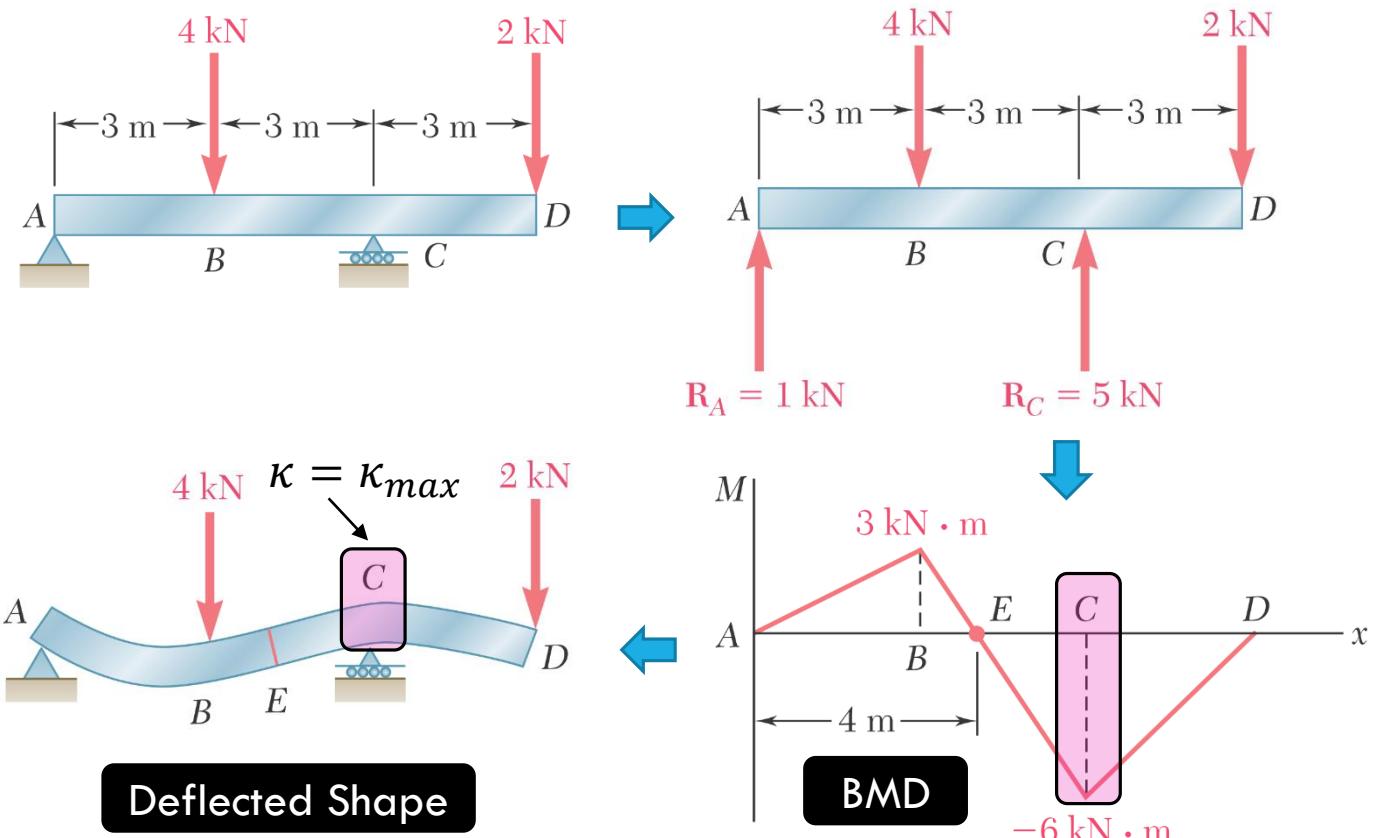
BMD

CURVATURE AND MOMENT RELATIONSHIP

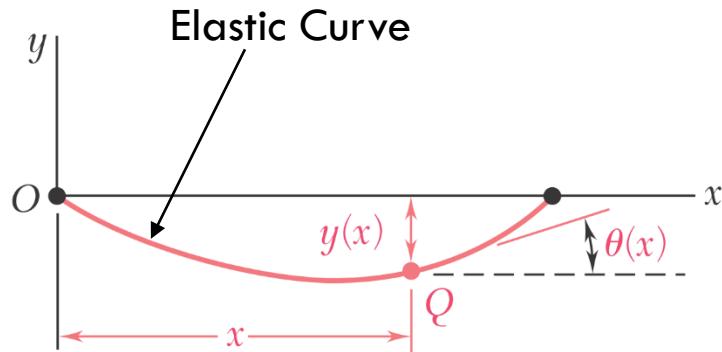
- For a general case of loading, bending moment M varies from one section to another

$$\therefore K = \frac{1}{\rho} = \frac{M(x)}{EI}$$

- Zero curvature (κ) at points of zero bending moment
- Beam smiles where M and κ is +ve
- Beam cries where M and κ is -ve
- Maximum curvature occurs where absolute M is max.
- Curvature helps establish a sense of shape of the beam
- We need an equation of the beam shape (elastic curve) to find deflections and slope!



EQUATION OF ELASTIC CURVE

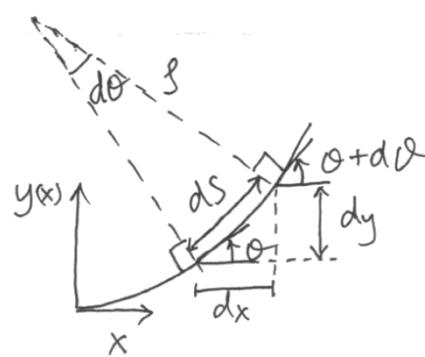


Recall from elementary calculus

$$\frac{dy}{dx} = \tan \theta, \quad \frac{1}{\rho} = \frac{d\theta}{ds} = \frac{d\theta}{dx} \sqrt{1 + \left(\frac{dy}{dx}\right)^2}$$

$$\frac{d(\tan \theta)}{dx} = \frac{d^2 y}{dx^2} = \frac{d(\tan \theta)}{d\theta} \frac{d\theta}{dx} = (1 + \tan^2 \theta) \frac{d\theta}{dx} = \left(1 + \left(\frac{dy}{dx}\right)^2\right) \frac{d\theta}{dx}$$

$$\frac{d\theta}{dx} = \frac{\frac{d^2 y}{dx^2}}{1 + \left(\frac{dy}{dx}\right)^2}, \quad \kappa = \frac{1}{\rho} = \frac{\frac{d^2 y}{dx^2}}{\left(1 + \left(\frac{dy}{dx}\right)^2\right)^{3/2}}$$



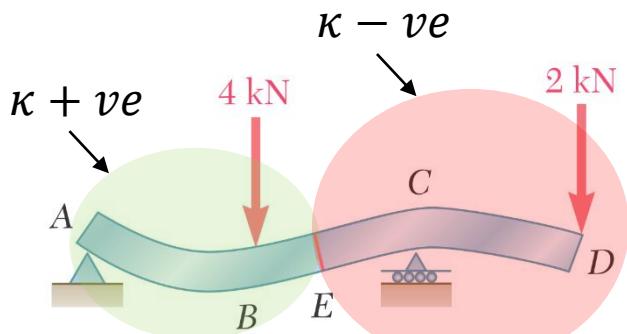
- However, for small angles of rotation:

$$\kappa = \frac{1}{\rho} = \frac{\frac{d^2 y}{dx^2}}{\left[1 + \left(\frac{dy}{dx}\right)^2\right]^{3/2}} \approx \frac{d^2 y}{dx^2}$$

This is the **exact expression** for curvature-deflection relationship

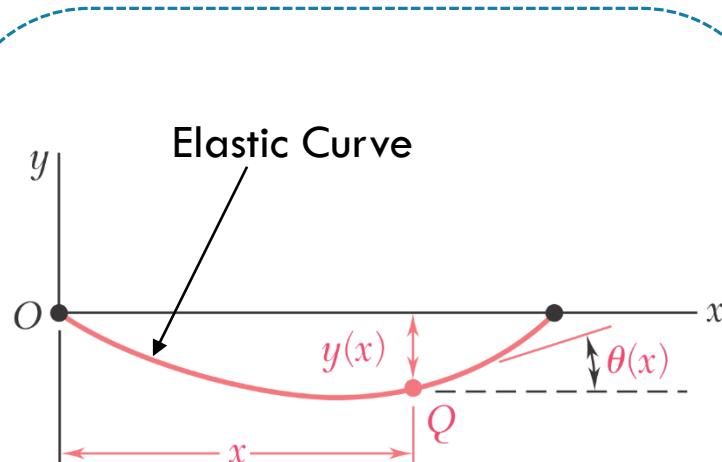
[Holds good for small or large deflections/rotations]

DEFLECTION OF BEAMS



$$\kappa = \frac{M(x)}{EI}$$

From “Pure Bending”

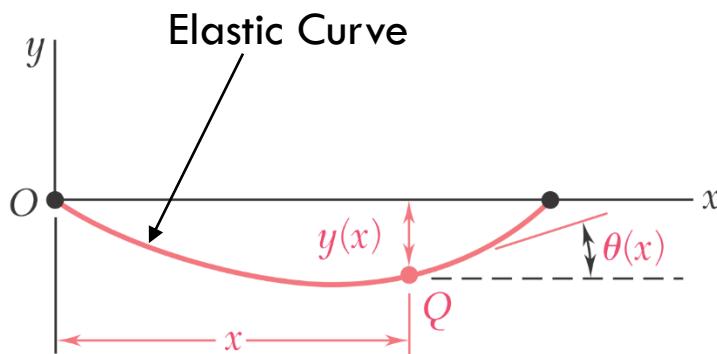


$$\kappa = \frac{1}{\rho} = \frac{\frac{d^2 y}{dx^2}}{\left[1 + \left(\frac{dy}{dx}\right)^2\right]^{3/2}} \approx \frac{d^2 y}{dx^2}$$

From Calculus

$$\frac{d^2 y}{dx^2} = \frac{M(x)}{EI}$$

EQUATION OF ELASTIC CURVE



$$EI \frac{d^2y}{dx^2} = M(x)$$

$$\Rightarrow EI \frac{dy}{dx} = \int_0^x M(x) dx + C_1$$

↪ beam is prismatic $\rightarrow I = I(x)$

For small angles, $\frac{dy}{dx} = \tan\theta = \theta(x)$

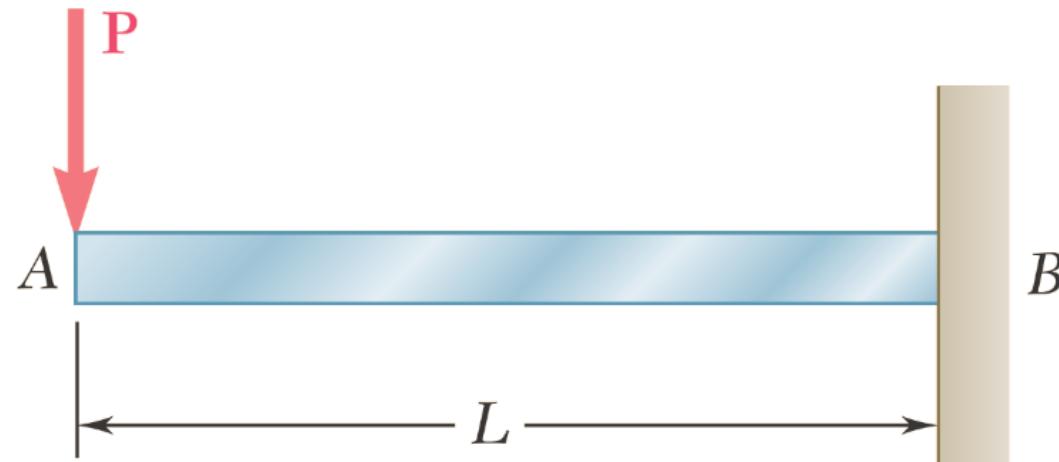
$$\Rightarrow EI \theta(x) = \int_0^x M(x) dx + C_1 \quad \text{— gives slope!}$$

$$\Rightarrow EI y = \int_0^x \left[\int_0^x M(x) dx + C_1 \right] dx + C_2 \quad \text{— gives deflection!}$$

$C_1, C_2 \rightarrow$ Constants of integration

\curvearrowleft Determined from boundary conditions.

EXAMPLE PROBLEM



Find deflection and slope at A

(Solution – next page)

$$EI \frac{d^2y}{dx^2} = M(x)$$

$$\therefore EI \frac{d^2y}{dx^2} = -Px$$

$$\Rightarrow EI \frac{dy}{dx} = -\frac{Px^2}{2} + C_1 \quad \text{Integrating once}$$

$$\Rightarrow EI y = -\frac{Px^3}{6} + C_1 x + C_2 \quad \text{Integrating twice.}$$

Boundary Conditions

$$@ x=L, \theta_B = 0 \quad (\frac{dy}{dx})_B = 0$$

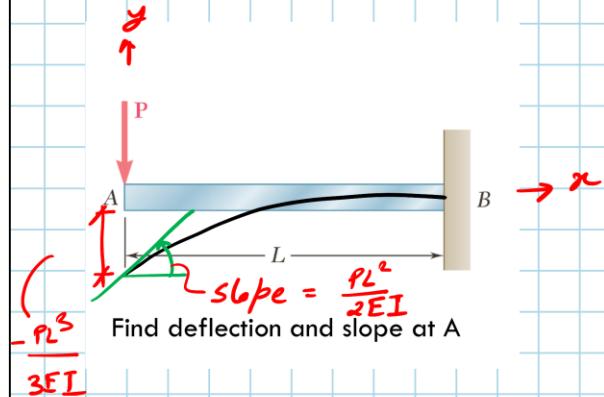
$$@ x=L, y_B = 0$$

After substitution:

$$C_1 = \frac{PL^2}{2}; \quad C_2 = -\frac{PL^3}{3}$$

$$\therefore \text{Slope} \rightarrow EI \frac{dy}{dx} = -\frac{Px^2}{2} + \frac{PL^2}{2}$$

$$\therefore \text{Deflection} \rightarrow EI y = -\frac{Px^3}{6} + \frac{PL^2 x}{2} - \frac{PL^3}{3}$$

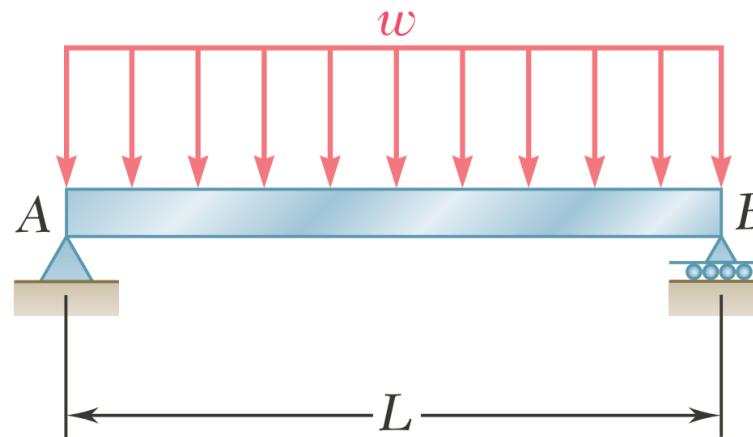


At A : $x=0$

$$\therefore \frac{dy}{dx} \Big|_{x=0} = \boxed{\frac{dy}{dx} \Big|_A} = \frac{PL^2}{2EI}$$

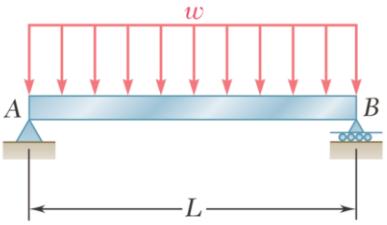
$$\therefore \boxed{y_A = -\frac{PL^3}{3EI}}$$

EXAMPLE PROBLEM



Find maximum deflection in the beam

(Solution – next page)



Find elastic curve equation and maximum deflection of the beam

Using method of sections, $M(x)$ at any distance x from left end:

$$M(x) = -\frac{wx^2}{2} + \frac{wlx}{2}$$

$$\therefore EI \frac{d^2y}{dx^2} = -\frac{wx^2}{2} + \frac{wlx}{2}$$

Integrating twice again

$$EIy = -\frac{\omega x^4}{24} + \frac{\omega lx^3}{12} + C_3x + C_4$$

$$@A: x=0; y_A=0$$

$$@B: x=L; y_B=0$$

$$\Rightarrow C_3 = -\frac{\omega L^3}{24}, C_4 = 0$$

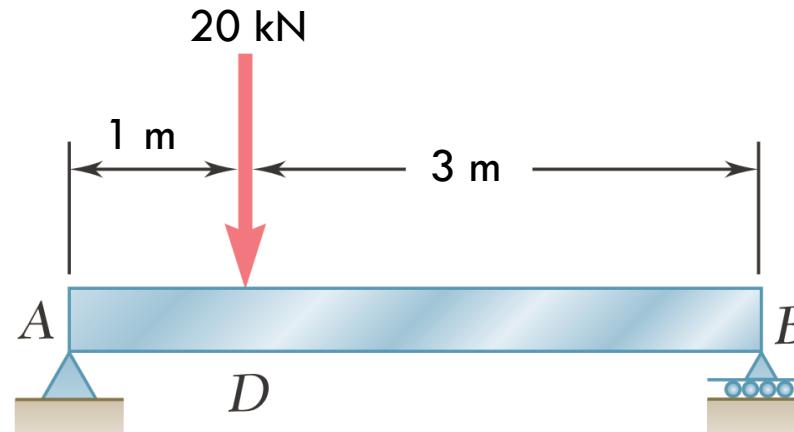
Equation of elastic curve

$$y = \frac{\omega}{24EI} (-x^4 + 2Lx^3 - L^3x)$$

Max deflection @ $x = \frac{L}{2}$

$$y_{\max} = -\frac{5}{384} \frac{\omega L^4}{EI}$$

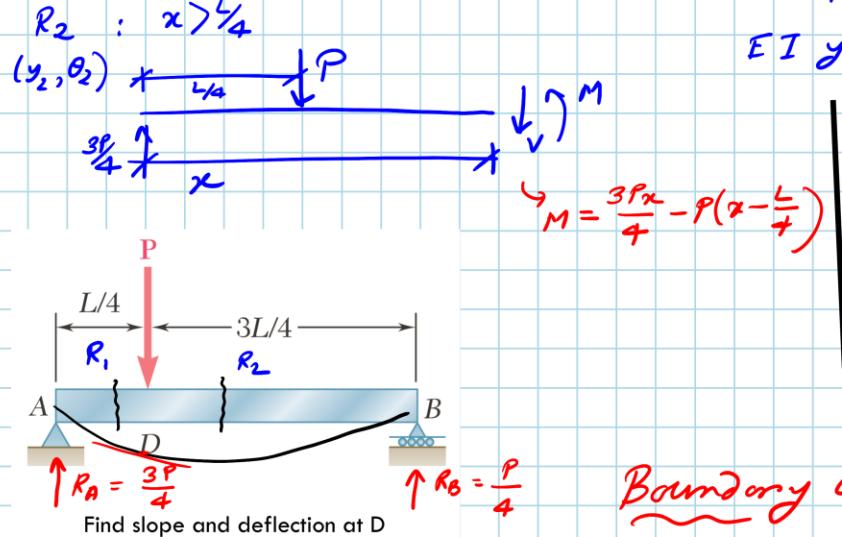
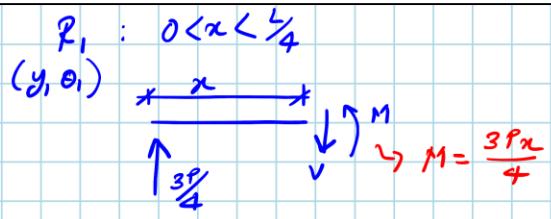
EXAMPLE PROBLEM



Find slope and deflection at D

[modulus of elasticity of $E = 10 \text{ GPa}$ and a rectangular cross section of $b = 60 \text{ mm}$ and $h = 125 \text{ mm}$.]

(Solution – next page)



Region #1 (R_1)

$$EI \frac{d^2y}{dx^2} = \frac{3Px}{4}$$

Integrating once & twice:

$$EI \theta_1 = \frac{3Px^2}{8} + C_1 \quad \text{---(1)}$$

$$EI y_1 = \frac{Px^3}{8} + C_1 x + C_2 \quad \text{---(2)}$$

Region #2 (R_2)

$$EI \frac{d^2y}{dx^2} = \frac{3Px}{4} - P(x - \frac{L}{4})$$

Integrating

$$EI \theta_2 = -\frac{Px^2}{8} + \frac{PLx}{4} + C_3 \quad \text{---(3)}$$

$$EI y_2 = -\frac{Px^3}{24} + \frac{PLx^2}{8} + C_3 x + C_4 \quad \text{---(4)}$$

Boundary Conditions:

$$[x=0, y_1=0]$$

$$[x=L, y_2=0]$$

$[x=\frac{L}{4}, \theta_1=\theta_2]$ — Slope Compatibility

$[x=\frac{L}{4}, y_1=y_2]$ — Displacement Compatibility

$$C_1 = -\frac{7PL^2}{128}; C_2 = 0; C_3 = -\frac{11PL^2}{128}; C_4 = \frac{PL^3}{384}$$

$$\theta_D = -\frac{PL^2}{32EI}$$

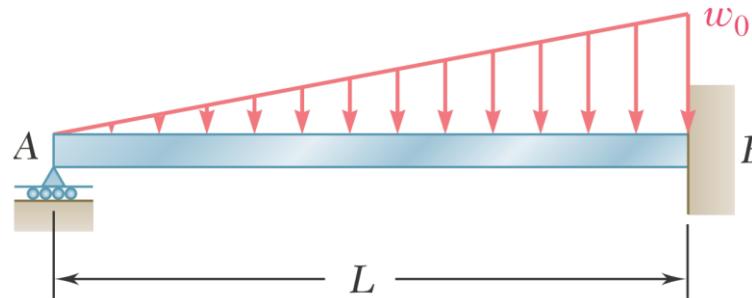
$$y_D = -\frac{3PL^3}{256EI}$$

For given values,
we get:

$$\theta_D = -0.10$$

$$y_D = 153 \text{ mm}$$

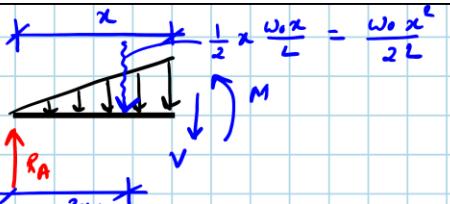
STATICALLY INDETERMINATE BEAMS!



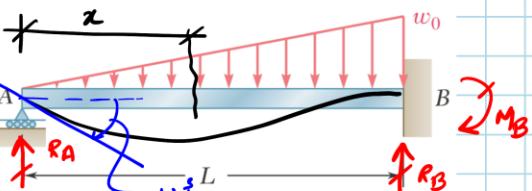
[Beam Statically Indeterminate to First Degree]

Find reaction at A, derive equation for elastic curve, and find slope at A.

(Solution – next page)



$$M = R_A x - \frac{w_0 x^3}{6L}$$



Find reaction at A, derive equation for elastic curve, and find slope at A.

o° Equation of Elastic Curve:

$$EIy = \frac{1}{6} \left(\frac{w_0 L}{10} \right) x^3 - \frac{w_0 x^5}{120L} - \frac{w_0 L^3 x}{120}$$

o° Slope @ A $\hat{x}=0$:

$$EI \frac{dy}{dx} \Big|_{x=0} \rightarrow \theta_A = -\frac{w_0 L^3}{120EI}$$

We know

$$EI \frac{d^2y}{dx^2} = M(x) = R_A x - \frac{w_0 x^3}{6L}$$

Integrating twice

$$EI \frac{dy}{dx} = \frac{R_A x^2}{2} - \frac{w_0 x^4}{24L} + C_1 \quad \leftarrow \text{First integration}$$

$$EIy = \frac{R_A x^3}{6} - \frac{w_0 x^5}{120L} + C_1 x + C_2$$

Boundary Conditions

- ① AT A: $x=0, y=0$
- ② AT B: $x=L, y=0$
- ③ AT B: $x=L, \theta = \frac{dy}{dx} = 0$

On Solving

$$R_A = \frac{w_0 L}{10}; C_1 = -\frac{w_0 L^3}{120}; C_2 = 0$$



DEFLECTION OF BEAMS

[Using Singularity Functions]
Amazing Method

MECHANICS OF MATERIAL
(SOLID MECHANICS)



Singularity Functions or Macaulay Functions (Mathematical Rules)

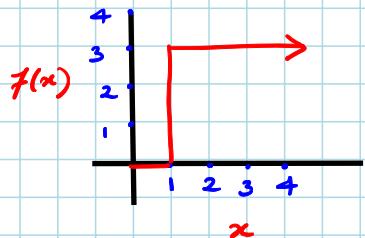
$\langle \rangle$ - Macaulay Brackets

A singularity function is defined as:

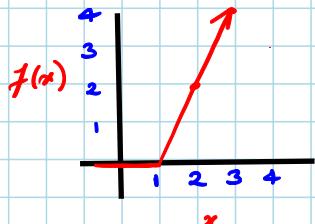
$$f(x) = \langle x-a \rangle^n = \begin{cases} (x-a)^n & \text{for } x > a \\ 0 & \text{for } x < a \end{cases} \quad \text{Valid for } n > 0$$

Example:

$$f(x) = 3 \langle x-1 \rangle^0$$



$$f(x) = 2 \langle x-1 \rangle^1 \quad \begin{array}{l} \xleftarrow{x=2} 2(2-1)^1 = 2 \\ \xleftarrow{x=3} 2(3-1)^1 = 2 \times 2 = 4 \end{array}$$



Integration Rules for
Macaulay Functions

$$\int \langle x-a \rangle^n dx = \frac{\langle x-a \rangle^{n+1}}{n+1} \quad \text{for } n > 0$$

$$= \langle x-a \rangle^{n+1} \quad \text{for } n < 0$$

SINGULARITY FUNCTIONS

Sign Convention

P, w positive upwards,
M positive clockwise

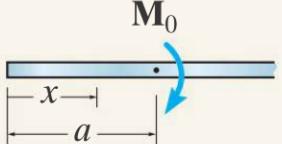
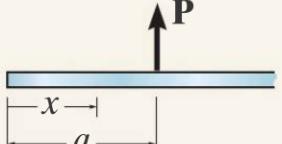
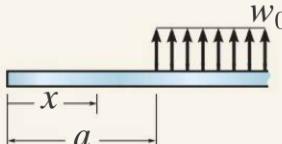
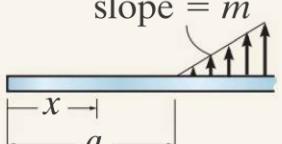
Macaulay Functions

$$\langle x - a \rangle^n = \begin{cases} 0 & \text{for } x < a \\ (x - a)^n & \text{for } x \geq a \end{cases}$$

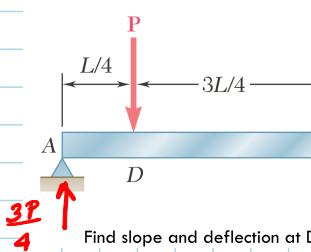
$n \geq 0$

Integration rules

$$\int \langle x - a \rangle^n dx = \begin{cases} \frac{\langle x - a \rangle^{n+1}}{n + 1} & \text{for } n \geq 0 \\ \langle x - a \rangle^{n+1} & \text{for } n < 0 \end{cases}$$

Loading	Loading Function $w = w(x)$	Shear $V = \int w(x)dx$	Moment $M = \int Vdx$
	$w = M_0 \langle x - a \rangle^{-2}$	$V = M_0 \langle x - a \rangle^{-1}$	$M = M_0 \langle x - a \rangle^0$
	$w = P \langle x - a \rangle^{-1}$	$V = P \langle x - a \rangle^0$	$M = P \langle x - a \rangle^1$
	$w = w_0 \langle x - a \rangle^0$	$V = w_0 \langle x - a \rangle^1$	$M = \frac{w_0}{2} \langle x - a \rangle^2$
	$w = m \langle x - a \rangle^1$	$V = \frac{m}{2} \langle x - a \rangle^2$	$M = \frac{m}{6} \langle x - a \rangle^3$

Loading	Loading Function $w = w(x)$	Shear $V = \int w(x) dx$	Moment $M = \int V dx$
	$w = M_0(x-a)^{-2}$	$V = M_0(x-a)^{-1}$	$M = M_0(x-a)^0$
	$w = P(x-a)^{-1}$	$V = P(x-a)^0$	$M = P(x-a)^1$
	$w = w_0(x-a)^0$	$V = w_0(x-a)^1$	$M = \frac{w_0}{2}(x-a)^2$
	$w = m(x-a)^1$	$V = \frac{m}{2}(x-a)^2$	$M = \frac{m}{6}(x-a)^3$



$$M(x) = \frac{3P}{4} \langle x-0 \rangle^1 - P \langle x-\frac{L}{4} \rangle^1$$

$$EI \frac{d^2y}{dx^2} = M(x)$$

$$\Rightarrow EI \frac{d^2y}{dx^2} = \frac{3P}{4} \langle x-0 \rangle^1 - P \langle x-\frac{L}{4} \rangle^1$$

$$\Rightarrow EI \frac{dy}{dx} = \frac{3P}{4x^2} \langle x-0 \rangle^2 - \frac{P}{2} \langle x-\frac{L}{4} \rangle^2 + C_1$$

$$\Rightarrow EIy = \frac{3P}{8x^3} \langle x-0 \rangle^3 - \frac{P}{2x^3} \langle x-\frac{L}{4} \rangle^3 + C_1x + C_2$$

Boundary Conditions

$[x=0, y=0] ; [x=L, y=0]$

$$EI \times 0 = \frac{3P}{24} \langle 0-0 \rangle^3 - \frac{P}{6} \langle 0-\frac{L}{4} \rangle^3 + C_1 \times 0 + C_2$$

$$\therefore C_2 = 0$$

$$EI \times 0 = \frac{3P}{24} \langle L-0 \rangle^3 - \frac{P}{6} \langle L-\frac{L}{4} \rangle^3 + C_1 L + 0$$

$$\therefore C_1 = -\frac{7PL^2}{128}$$

$$EIy = \frac{3P}{24} \langle x-0 \rangle^3 - \frac{P}{6} \langle x-\frac{L}{4} \rangle^3 - \frac{7PL^2x}{128}$$

Substituting $x = \frac{L}{4}$ i.e. at D

$$y_D = -\frac{3PL^3}{256EI} ; \theta_D = -\frac{PL^2}{32EI}$$

Loading	Loading Function $w = w(x)$	Shear $V = \int w(x) dx$	Moment $M = \int V dx$
	$w = M_0(x-a)^{-2}$	$V = M_0(x-a)^{-1}$	$M = M_0(x-a)^0$
	$w = P(x-a)^{-1}$	$V = P(x-a)^0$	$M = P(x-a)^1$
	$w = w_0(x-a)^0$	$V = w_0(x-a)^1$	$M = \frac{w_0}{2}(x-a)^2$
	$w = m(x-a)^1$	$V = \frac{m}{2}(x-a)^2$	$M = \frac{m}{6}(x-a)^3$

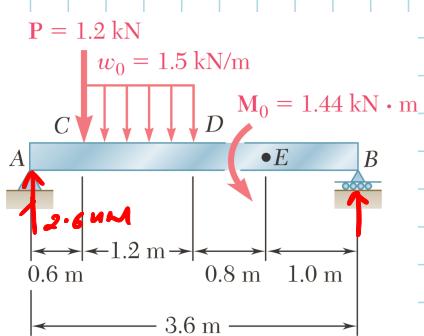
$$EI \frac{d^3y}{dx^3} = M(x) = 2.6(x-0)' - 1.2(x-0.6)' - \frac{1.5}{2}(x-0.6)^2 + \frac{1.5}{2}(x-1.8)^2 - 1.44(x-2.6)''$$

Integrating twice: $EIy = \frac{2.6}{6}(x-0)^3 - \frac{1.2}{6}(x-0.6)^3 - \frac{1.5}{24}(x-0.6)^4 + \frac{1.5}{24}(x-1.8)^4 - \frac{1.44}{2}(x-2.6)^2 + C_1x + C_2$

Boundary Conditions:

$$x=0, y=0 \quad ; \quad x=6, y=0 \rightarrow C_2 = 0; C_1 = -2.692$$

$$\therefore \text{At } D (x = 1.8m) \rightarrow y_D = -\frac{2.794}{EI}$$



Find deflection at D

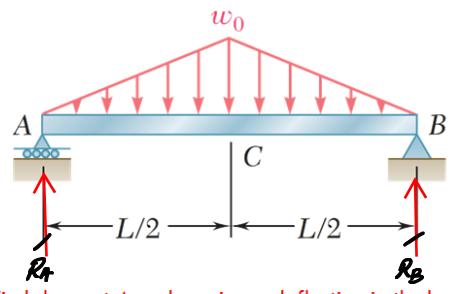
$$1.2 m$$

$$2.6 m$$

$$1.5 m$$

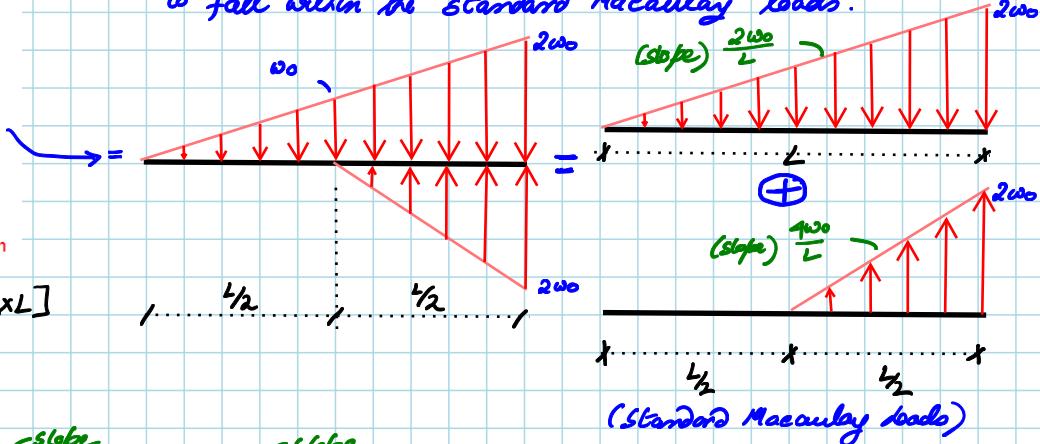
$$1.44 m$$

$$1.5 m$$



Find slope at A and maximum deflection in the beam

To use Macaulay Functions, we must adjust the loading to fall within the standard Macaulay loads.



$$\text{From symmetry: } R_A = R_B = \frac{1}{2} \left[\frac{1}{2} \times w_0 \times L \right] = \frac{w_0 L}{4}$$

$$EI \frac{d^2y}{dx^2} = R_A \langle x - 0 \rangle^1 - \frac{2w_0}{6} \langle x - 0 \rangle^3 + \frac{4w_0}{6} \langle x - \frac{L}{2} \rangle^3$$

$$\Rightarrow EI \frac{dy}{dx} = \frac{w_0 L}{4x^2} \langle x - 0 \rangle^2 - \frac{w_0}{3 \times 4L} \langle x - 0 \rangle^4 + \frac{2w_0}{3L \times \frac{L}{2}} \langle x - \frac{L}{2} \rangle^4 + C_1 \quad \text{slope}$$

$$\Rightarrow EI y = \frac{w_0 L}{24} \langle x - 0 \rangle^3 - \frac{w_0}{60L} \langle x - 0 \rangle^5 + \frac{w_0}{30L} \langle x - \frac{L}{2} \rangle^5 + C_1 x + C_2 \quad \text{deflection}$$

Boundary Conditions: $\begin{cases} @ x=0, y=0 \\ @ x=L, y=0 \end{cases}$ Substitute in deflection equation.

$$EI x 0 = 0 - 0 + 0 + C_1 x 0 + C_2 \Rightarrow C_2 = 0$$

$$\text{and } EI x 0 = \frac{w_0 L}{24} (L)^3 - \frac{w_0}{60L} (L)^5 + \frac{w_0}{30L} (\frac{L}{2})^5 + C_1 x L$$

$$\Rightarrow C_1 = \frac{w_0}{60} L^3 - \frac{w_0 L^2}{960} - \frac{w_0 L^3}{24} = -\frac{5}{192} w_0 L^3 \quad \therefore C_1 = -\frac{5}{192} w_0 L^3$$

$$\therefore \text{slope } @ A : EI \frac{dy}{dx} = 0 - 0 + 0 + C_1 = -\frac{5}{192} w_0 L^3$$

$$\therefore \boxed{\frac{dy}{dx} \Big|_{x=A} = -\frac{5}{192EI} w_0 L^3}$$

\therefore Maximum deflection (by observation @ $x = \frac{L}{2}$)

$$\therefore EI y \Big|_{x=\frac{L}{2}} = \frac{w_0 L}{24} \left(\frac{L}{2}\right)^3 - \frac{w_0}{60L} \times \left(\frac{L}{2}\right)^5 - \frac{w_0}{30L} \times 0 - \frac{5}{192} w_0 L^4$$

$$= -\frac{w_0 L^4}{120} \quad \Rightarrow \boxed{y @ x = \frac{L}{2} = -\frac{w_0 L^4}{120EI}}$$