

Lu Sun, and many more.

A Notebook on Control System



*To my family, friends and communities members who
have been dedicating to the presentation of this
notebook, and to all students, researchers and faculty
members who might find this notebook helpful.*



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Foreword

If software or e-books can be made completely open-source, why not a notebook?

This brings me back to the summer of 2009 when I started my third year as a high school student in Harbin No. 3 High School. In around the end of August when the results of Gaokao (National College Entrance Examination of China, annually held in July) are released, people from photocopy shops would start selling notebooks photocopies that they claim to be from the top scorers of the exam. Much curious as I was about what these notebooks look like, never have I expected myself to actually learn anything from them, mainly for the following three reasons.

First of all, some (in fact many) of these notebooks were more difficult to understand than the textbooks. I guess we cannot blame the top scorers for being so smart that they sometimes make things extremely brief or overwhelmingly complicated.

Secondly, why would I want to adapt to notebooks of others when I had my own notebooks which in my opinion should be just as good as theirs.

And lastly, as a student in the top-tier high school myself, I knew that the top scorers of the coming year would probably be a schoolmate or a classmate. Why would I want to pay that much money to a complete stranger in a photocopy shop for my friend's notebook, rather than requesting a copy from him or her directly?

However, things had changed after my becoming an undergraduate student in 2010. There were so many modules and materials to learn in a university, and as an unfortunate result, students were often distracted from digging deeply into a module (For those who were still able to do so, you have my highest respect). The situation became even worse as I started pursuing my Ph.D. in 2014. As I had to focus on specific research areas entirely, I could hardly split much time on other irrelevant but still important and interesting contents.

This motivated me to start reading and taking notebooks for selected books and articles, just to force myself to spent time learning new subjects out of my comfort zone. I used to take hand-written notebooks. My very first notebook was on *Numerical Analysis*, an entrance level module for engineering background graduate students. Till today I still have on my hand dozens of these notebooks. Eventually, one day it suddenly came to me: why not digitalize them, and make them accessible online and open-source, and let everyone read and edit it?

As most of the open-source software, this notebook (and it applies to the other notebooks in this series as well) does not come with any “warranty” of any kind, meaning that there is no guarantee for the statement and knowledge in this notebook to be absolutely correct as it is not peer reviewed. **Do NOT cite this notebook in your academic research paper or book!** Of course, if you find anything helpful with your research, please trace back to the origin of the citation and double confirm it yourself, then on top of that determine whether or not to use it in your research.

This notebook is suitable as:

- a quick reference guide;
- a brief introduction for beginners of the module;
- a “cheat sheet” for students to prepare for the exam (Don’t bring it to the exam unless it is allowed by your lecturer!) or for lecturers to prepare the teaching materials.

This notebook is NOT suitable as:

- a direct research reference;
- a replacement to the textbook;

because as explained the notebook is NOT peer reviewed and it is meant to be simple and easy to read. It is not necessary brief, but all the tedious explanation and derivation, if any, shall be “fold into appendix” and a reader can easily skip those things without any interruption to the reading experience.

Although this notebook is open-source, the reference materials of this notebook, including textbooks, journal papers, conference proceedings, etc., may not be open-source. Very likely many of these reference materials are licensed or copyrighted. Please legitimately access these materials and properly use them.

Some of the figures in this notebook is drawn using Excalidraw, a very interesting tool for machine to emulate hand-writing. The Excalidraw project can be found in GitHub, [excalidraw/excalidraw](https://github.com/excalidraw/excalidraw).

Preface

Control System



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Model predictive controller belongs to the family of optimal controllers. Given the plant model and a control signal, the behavior of the system can be foreseen. MPC finds the control signal with the optimal forecast.

Comparing with other optimal controllers such as LQG, MPC has a better adaption to different kinds of restrictions. This makes MPC extremely popular in industry. From this perspective, many optimal controllers can be taken as special cases of MPC.

The references of this chapter include:

- Rawlings, J.B., Mayne, D.Q. and Diehl, M., 2017. Model predictive control: theory, computation, and design (Vol. 2). Madison, WI: Nob Hill Publishing. [2].

6.1 Introduction

The MPC uses system dynamic model to forecast system behavior, and optimize the forecast to produce the best decision at the current moment.

From the above description, we can see that there are at least 3 key factors of MPC:

- Plant model and constraints, to be derived from the plant.
- Plant model parameters, to be obtained via parameter estimation.
- Plant current state, to be obtained via state estimation.

- Optimal control, to be obtained by defining the performance index and solving the functional.

where the model derived from assumptions, its parameters obtained by parameter estimation. Plant current state is obtained by state estimation. Finally, an optimization problem with varieties of restrictions is proposed to generate the control signal.

6.1.1 Model and Constraints

The model in an MPC plays the most important role. Many types of models are used to describe the plant in the MPC scope. Depending on the plant, the model can be either in continuous time domain or in discrete time domain; either state-space or input-output; either centralized or discrete (discrete models are used when the behavior of the system is not spatially uniform); and either deterministic or stochastic. Different mathematics tools are used to describe and solve MPC problems of different types of models.

MPC has a good adaption to different types of constraints. The constraints can be largely divided into two categories, namely physical constraints and performance constraints. Input signal $u(k)$ related constraints are often physical constraints. It describes physical limits of the system. If the controller does not respect these constraints, the physics enforce them. An example of a physical constraint is the maximum input power to a motor.

State vector $x(k)$ and output $y(k)$ related constraints, on the other hand, are often performance constraints. They describe the desirable performance of the system, and they may or may not be achievable in practice. MPC is able to find out whether the performance constraints are achievable or not. Should the performance constraints not be achievable, they need to be modified and downgraded.

Input signal related constraints are often given in the following format.

$$\begin{bmatrix} -I \\ I \end{bmatrix} u(k) \leq \begin{bmatrix} -u_{\min} \\ u_{\max} \end{bmatrix}$$

State vector related constraints are often given in the following format.

$$Fx(k) \leq f$$

When augmented state vector is used, it is possible to combine the input and the state vector together to form constraints with more flexibility. For example, let

$$\tilde{x}(k) = \begin{bmatrix} x(k) \\ u(k-1) \end{bmatrix}$$

Then the following constraint

$$\begin{bmatrix} 0 & -I \\ 0 & I \end{bmatrix} \tilde{x}(k) + \begin{bmatrix} I \\ -I \end{bmatrix} u(k) \leq \begin{bmatrix} \Delta_{\max} \\ -\Delta_{\min} \end{bmatrix}$$

essentially translates to

$$\Delta_{\min} \leq u(k) - u(k-1) \leq \Delta_{\max}$$

which can become handy sometimes.

It is also possible to limit the state vector and/or the input to be integers or discrete values, which is not commonly seen in other optimal controllers.

6.1.2 Parameter Estimation and State Estimation

The most commonly used parameter estimation method is the least squares estimation. The most commonly used state estimation method is the Kalman filter. Both least squares estimator and Kalman filter are linear filters and they are designed optimal for Gaussian noise. Least squares estimator can be implemented in a recursive manner, in which case it becomes a special case of Kalman filter mathematically.

It is possible to add “forget factor” to least squares estimator or Kalman filter. A window with fixed length is specified. When a new measurement set becomes available, the earliest measurement in the window will be removed, thus its impact on the state estimate eliminated. This leads to the moving horizon estimation (in contrast, the conventional Kalman filter implementation is also known as growing-memory estimation), which is less efficient but more robust to time-varying in the system.

6.1.3 Optimal Control

The optimal control to the plant vary with the definition of the cost function and constraints. In this section, for simplicity, a linear quadratic regulator problem is used to demonstrate MPC. Notice that although LQR is often regarded as a stand alone optimal control problem, there is no harm to introduce LQR as a simplified MPC problem, just for demonstration purpose.

Finite LQR Formulation

Consider the following dynamic model in discrete time domain. Current timestamp is $k = 0$. For simplicity, it is assumed that the model is known and the state is measurable precisely, thus there is no need to build an estimator for the model or the state.

$$x(k+1) = Ax(k) + Bu(k) \quad (6.1)$$

$$y(k) = x(k) \quad (6.2)$$

With the above model, it is possible to predict how the state evolves given a sequence of inputs as below.

$$u = [u(0) \quad u(1) \quad \dots \quad u(N-1)]$$

where N is the largest time scale considered in this problem.

The purpose of the control is to regulate non-zero initial state $x(0)$ to zero while keeping the magnitude of $u(k)$ small. Therefore, define the cost function as follows.

$$V(x(0), u) = \frac{1}{2} \sum_{k=0}^{N-1} (x(k)^T Q x(k) + u(k)^T R u(k)) + \frac{1}{2} x(N)^T P_f x(N) \quad (6.3)$$

where $Q \geq 0$, $P_f \geq 0$ and $R > 0$ are symmetric matrices of user's choice. It can be proved that the above problem formulation must have a unique solution.

A Brief Introduction to Dynamic Programming

The aforementioned optimization problem in (6.3) can be solved via variety of ways. Consider using dynamic programming (DP).

A brief introduction to DP is given as follows. For illustration of DP, consider the following simple optimization problem.

$$x^*, y^* = \arg \min_{x, y} f(x, y) \quad (6.4)$$

Since x, y are coupled in $f(x, y)$, an intuitive way of solving this optimization problem is to let

$$\begin{cases} \frac{\partial}{\partial x} f(x, y) = 0 \\ \frac{\partial}{\partial y} f(x, y) = 0 \end{cases}$$

and if $f(x, y)$ is quadratics of x, y , the above should generate a unique solution.

Alternatively, consider solving (6.4) using the following approach. Think of (6.4) as a two-stage optimization problem. Given any constant value of y , there is an associated optimized $x^*(y)$ which minimizes (6.4). Using this method, we can transform $f(x, y)$ into $f(x^*(y), y)$, a function of y alone. From there, the optimal y can be found easily. Subsequently, $x^*(y)$ can be obtained as well. To summarize, (6.4) is equivalent of

$$\begin{aligned} y^* &= \arg \min_y \left(\min_x f(x, y) \right) \\ x^* &= \arg \min_x f(x, y^*) \end{aligned}$$

Solving using DP as above can simplify the problem, especially when only part of the optimal variable values is required (for example, only y^* is required).

Solution to the Finite LQR Problem

The LQR problem in (6.3) is essentially something similar with (6.4). The

initial state $x(0)$ is a constant in the optimization problem, the control signals $u(0), \dots, u(N-1)$ to be optimized, and the coupling introduced by the system dynamics (6.1). The ultimate goal of the optimization problem is to determine $u(0)$.

Similarly, the optimization is divided into multiple stages, where in each stage only one instant $u(k)$ is considered, and other variables leading to that stage is considered as constant. We start with the last control input $u(N-1)$. The optimization problem is formulated as follows.

$$\begin{aligned} V^{\text{sub}}(N-1) &= \frac{1}{2} (x(N-1)^T Q x(N-1) + u(N-1)^T R u(N-1)) \\ &\quad + \frac{1}{2} x(N)^T P_f x(N) \\ \text{s.t.} \quad &x(N) = A x(N-1) + B u(N-1) \end{aligned}$$

where $x(N-1)$ is taken as a constant from another stage.

Solving $u(N-1)$ as a function of $x(N-1)$ gives

$$\begin{aligned} u^*(N-1) &= K_{N-1} x(N-1) \\ x(N) &= (A + B K_{N-1}) x(N-1) \\ V^{\text{sub}}(N-1) &= \frac{1}{2} x(N-1)^T \Pi_{N-1} x(N-1) \end{aligned}$$

where K_{N-1} and Π_{N-1} are two intermediate parameters calculated as follows.

$$\begin{aligned} K_{N-1} &= -(B^T P_f B + R)^{-1} B^T P_f A \\ \Pi_{N-1} &= Q + A^T P_f A - A^T P_f B (B^T P_f B + R)^{-1} B^T P_f A \end{aligned}$$

Consider the next stage optimization problem as follows.

$$\begin{aligned} V^{\text{sub}}(N-2) &= \frac{1}{2} (x(N-2)^T Q x(N-2) + u(N-2)^T R u(N-2)) \\ \text{s.t.} \quad &x(N-1) = A x(N-2) + B u(N-2) \end{aligned}$$

where $x(N-2)$ is taken as constant and $u^*(N-2)$ is to be found. The solution is given by

$$\begin{aligned} u^*(N-2) &= K_{N-2} x(N-2) \\ x(N-1) &= (A + B K_{N-2}) x(N-2) \\ V^{\text{sub}}(N-2) &= \frac{1}{2} x(N-2)^T \Pi_{N-2} x(N-2) \end{aligned}$$

where

$$\begin{aligned} K_{N-2} &= -(B^T \Pi_{N-1} B + R)^{-1} B^T \Pi_{N-1} A \\ \Pi_{N-1} &= Q + A^T \Pi_{N-1} A - A^T \Pi_{N-1} B (B^T \Pi_{N-1} B + R)^{-1} B^T \Pi_{N-1} A \end{aligned}$$

It is clear by now that the above calculations can be done recursively to obtain $u^*(0)$. Notice that all the control signals have the following form

$$u^*(k) = K_k x(k)$$

which implies that the optimal control can be taken as a classic state-space feedback controller with time-varying control gains.

Infinite LQR Problem

Since the recursive calculation is done N times to calculate K_{N-1}, \dots, K_0 , apparently different value of N results in different K_0 . It is worth mentioning that optimal control does not suggest stable control. In other words, it is possible for a specific system plant, specific choice of Q , R , P_f , and specific N , to lead to an unstable closed loop system, i.e., $A + BK_0$ with eigenvalues larger than 1. However, it is guaranteed that when $N \rightarrow \infty$, K_0 converges, and $A + BK_0$ is always stable. This introduces the infinite LQR problem, where the cost function corresponding with (6.3) is changed to

$$V(x(0), u) = \frac{1}{2} \sum_{k=0}^{\infty} (x(k)^T Q x(k) + u(k)^T R u(k))$$

Infinite LQR can be taken as a special case of finite LQR where $N \rightarrow \infty$.

We have been making an underlying assumption that the original plant is controllable. This means that the system can be driven from any initial state to $x = 0$ in finite time. Apply this concept to infinite LQR problem. This implies that in the beginning finite steps of this infinite control sequence, the state should have already been regulated to zero.

Since the control gain $K(k)$, $k = N-1, N-2, \dots$ converges to K , the first infinite control gains would be K . Therefore, it can be concluded that the infinite LQR suggests the control gain to remain K until all states becomes zero. Therefore, the infinite LQR problem results in constant control gain rather than time-varying control gains as in the finite LQR problem. The control gain can be obtained by solving the Riccati equation of Π as follows.

$$\begin{aligned} K &= -(B^T \Pi B + R)^{-1} B^T \Pi A \\ \Pi &= Q + A^T \Pi A - A^T \Pi B (B^T \Pi B + R)^{-1} B^T \Pi A \end{aligned}$$

6.1.4 Commonly Seen MPC Objectives

The most commonly seen MPC objectives are tracking, state regulation, and disturbance rejection.

- Tracking: the output of the plant shall follow the given reference point (not necessarily to be a constant) as close as possible.

- State regulation: the state vector of the plant shall converge to a given steady-state as soon as possible, while not violating system restrictions.
- Disturbance rejection: the plant shall maintain its operating point under unknown stochastic noise and disturbances.



7

Adaptive Control System

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Adaptive controller is generally referred to a controller with adjustable parameters and associated mechanism to adjust such parameters, to conform to new circumstances.

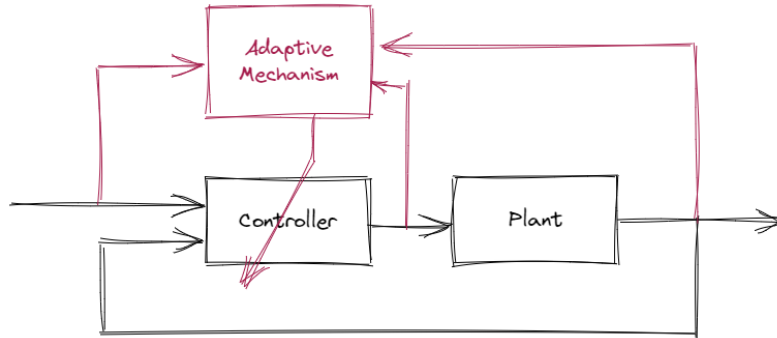
Comparing with most of the other control systems, an adaptive control system does not require a very precise modeling of the plant. More precisely, an adaptive control system can adapt the control gains to the unknown or time-varying plant and maintain the stability of the closed-loop system.

The references of this chapter include:

- Åström, K.J. and Wittenmark, B., 2013. Adaptive control. Courier Corporation [1].

7.1 Introduction

Adaptive controller is generally referred to a controller with adjustable parameters and associated mechanism to adjust such parameters, to conform to new circumstances.

**FIGURE 7.1**

Adaptive control system general schema.

Due to the parameter adjustment mechanism, the controller becomes non-linear. For example, while a conventional PID controller is linear, an adaptive PID controller is nonlinear since the $P(t)$, $I(t)$ and $D(t)$ are variables instead of constant gains.

In practice, an adaptive controller is often a modified version of a conventional controller just like the adaptive PID controller example. There are often 2 types of feedback loops in an adaptive control system: the original loops that comes with the conventional controller, and the additional loops to adjust the parameters of the conventional controller. A demonstrative figure is given in Fig. 7.1.

7.1.1 A Brief History of Adaptive Control System

The first adaptive control system application was designed for autopilots of a high-performance aircraft, where the conventional control system was found to work for one condition but not for the other. Gain scheduling was adopted to solve this problem.

Later on more sophisticated adaptive control systems that utilizes state-space model and stability theory were introduced. Stochastic control theory and system identification techniques were developed side-by-side.

The proof of stability of the adaptive control system, especially universal stability of the system, started to draw attention. The proof can be done often only with strong restrictions. People started to investigate the connection and difference of robust control and system identification with adaptive control and system identification. Computer controlled system and artificial neural network started to emerge and people realized the connection between adaptive control and computer learning.

Adaptive control system commercialization started in 1980s, and are used

in handling process dynamics and disturbances, and to provide automatic tuning of controller parameters.

7.1.2 Conventional Control System

The most conventional way of designing a control system is as follows.

1. Assume a linear model of the plant at the operating point.
2. Calculate or estimate the parameters of the plant model.
3. Design a closed-loop control system for the plant.

The control system designed following the above approach is usually intrinsically insensitive to modeling error and disturbance to some extent (due to the closed-loop implementation). But there are difficulties that could cause variation and troubles.

- **Nonlinear actuator** is one of the sources of modeling uncertainty, making the control system work only on/near the pre-determined operating point, but not universally.
- **Flow and speed variations** in the pipes of a system, if exist, can change from time to time. The different flow and speed can cause changes in the process dynamics, thus change the plant model.
- **Environmental dynamics and uncertainties** often draw significant concerns in control. For example, the aircraft flying at different height and speed suffers from different environmental conditions. The same applies to ship steering at different speed.

7.1.3 Adaptive Control Schema

Different adaptive control schemes have been designed to tackle the above issues. The commonly used ones are briefly introduced here.

Gain Scheduling

Measurement feedback may correlate well with changes in the process dynamics. The idea of gain scheduling is to map the process dynamic parameters with the control parameters, by either a function manner or a table lookup manner, i.e., schedule the control parameters to compensate the process dynamics.

A demonstrative figure is given in Fig. 7.2.

Gain scheduling is one of the most intuitive adaptive control schema, and has the earliest use case among all.

Model-Reference Adaptive System (MRAS)

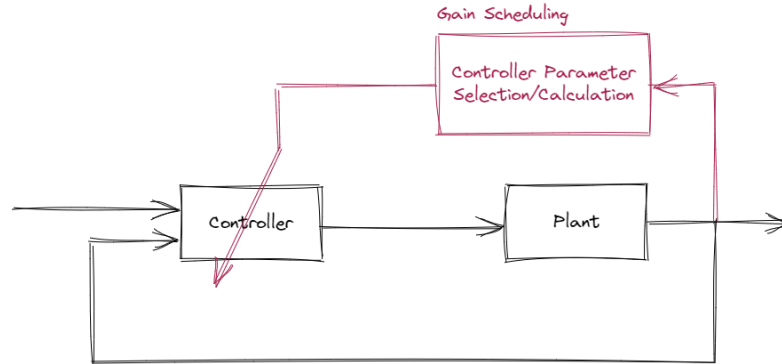


FIGURE 7.2
Gain scheduling schema.

In this scenario, a reference model is built that generates the “ideal” output of the closed-loop system given the reference point. This is often a virtual model that does not suffer from disturbances and error.

The MRAS then uses the output of both the actual plant and the reference model to calculate the controller parameter adjustment mechanism to minimize the difference between the outputs of the two models.

A demonstrative figure is given in Fig. 7.3.

The MRAS is somewhat like a learning system. Usually, there is a “adaptation rate” (just like the learning rate) that controls the adapting speed.

Self-Tuning Regulator (STR)

The idea of STR is to estimate the system process and design and change the whole control system in real-time. Unlike the gain scheduling approach and MRAS where parameter adjustment mechanism is calculated in real-time, in STR approach, the controller designing problem is solved in real-time. From this sense, STR can be taken as the “high-end” gain scheduling problem.

A demonstrative figure is given in Fig. 7.4.

STR can be made very flexible, because it is essentially a new control system design from scratch each time the environment changes.

Dual Control

The aforementioned adaptive control schemas do not taken into account the stochastics and statistics of the measurement and estimation uncertainty.

To analyze the performance of the system under estimation uncertainty and to develop control algorithm that can optimize the system performance under such uncertainty, nonlinear stochastic control theory needs to be used, which leads to the notion of dual control.

A demonstrative figure is given in Fig. 7.5.

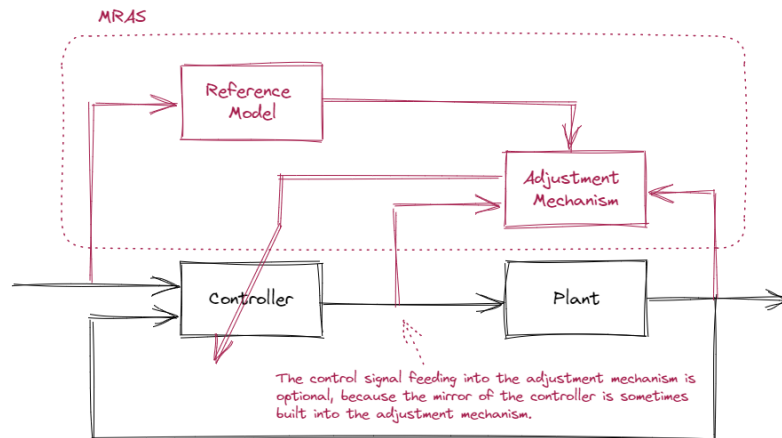


FIGURE 7.3
MRAS schema.

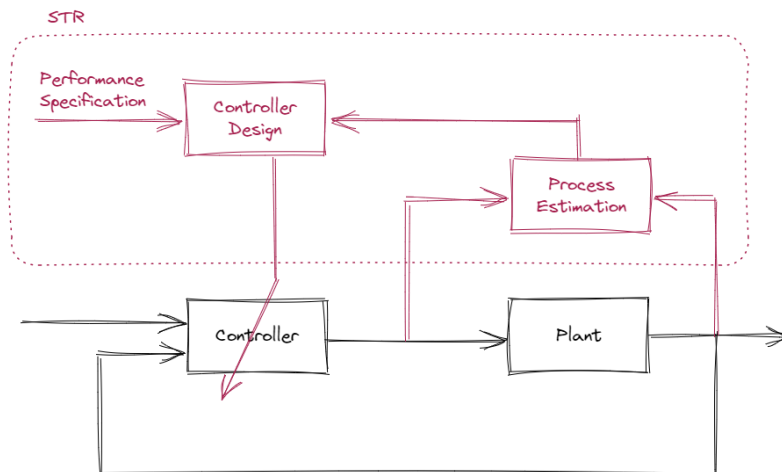


FIGURE 7.4
STR schema.

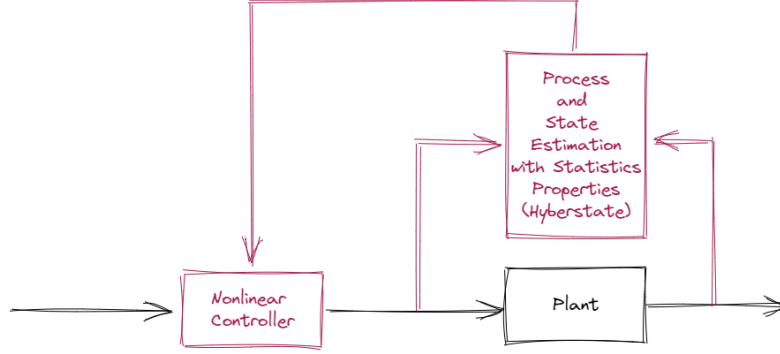


FIGURE 7.5
Dual control schema.

This control system is essentially a nonlinear controller with a robust/adaptive filter that estimates the process dynamics and the states of the plant, together with statistics information of the estimates.

7.1.4 A Typical Adaptive Control Problem Formulation

For simplicity, consider the following LTI plant realizations.

Continuous time domain state space model:

$$\begin{aligned}\dot{x}(t) &= Ax(t) + Bu(t) \\ y(t) &= Cx(t)\end{aligned}$$

Continuous time domain transfer function model:

$$G(s) = \frac{B(s)}{A(s)} = \frac{b_0 s^m + b_1 s^{m-1} + \dots + b_m}{s^n + a_1 s^{n-1} + \dots + a_n}$$

Continuous time domain input-output model (with p the differential operator, $p = \frac{d}{dt}$):

$$y(t) = G(p)u(t)$$

Discrete time domain state-space model:

$$\begin{aligned}x(k+1) &= Ax(k) + Bu(k) \\ y(k) &= Cx(k)\end{aligned}$$

Discrete time domain transfer function:

$$G(z) = \frac{B(z)}{A(z)} = \frac{b_0 z^m + b_1 z^{m-1} + \dots + b_m}{z^n + a_1 z^{n-1} + \dots + a_n}$$

Discrete time domain input-output model (with q the forward shift operator, $qy(k) = y(k+1)$):

$$y(k) = H(q)u(k)$$

The above models may be describing the same plant, with different level of details. For example, when dealing with transfer function and input-output models, the initial condition of the system is often ignored. When dealing with discrete time domain plant model, it is open a sampled system derived from the original system.

If the parameters in the models are known, it is straight forward to design a closed-loop controller for them. This is called the underlying design problem. The adaptive control problem, on the other hand, is to design a controller without knowing the parameters. In response, the controller comes with adjustable control gains and associated adjustment mechanism.

A typical adaptive controller design shall include the following procedures:

1. Characterize the desired closed-loop behavior.
2. Determine a suitable control law with adjustable parameters.
3. Design adjustment mechanism.
4. Implement the control law with the parameters adjusted using the adjustment mechanism.

Some successful use cases for adaptive control systems include

- Automatic tuning controller, such as the PID automatic tuning controller. In practice, all conventional control systems can benefit from automatic parameter tuning, if done correctly.
- Gain scheduling, which is the standard technique used in flight control system for high-performance aircraft, and is becoming increasingly popular in industrial process control.
- Continuous adaptation control for continuously varying system.
- “Human-in-the-loop” control system.

Since adaptive control system is often more costly and complicated than a conventional constant-gain controller, it should be used wisely and not abused. If the dynamics of the system are manageable, consider using parametric model based control algorithm rather than the adaptive control system, the later of which is usually more complicated.

7.2 Parameter Estimation

Parameter estimation (or more generally, system identification) is widely used, either directly or indirectly, in many adaptive control schemas. An obvious example is STR, where the process dynamics estimation is built into the controller. Parameter estimation also happens implicitly in MRAS.

The key factors in system identification include the following.

- Selection of model structure.
- Experiment design.
- Parameter and state estimation.
- Validation.

One thing to notice is that in adaptive control scope, the plant is assumed to change continuously, thus the parameter estimation needs to be done in a recursive manner.

7.2.1 Least Squares Estimation

Karl Friedrich Gauss defines LS estimation problem as follows.

LS estimation:

The parameters in the model should be chose such that the sum of the squares of the differences between the observed and the computed values, multiplied by numbers that measure the degree of precision, is a minimum.

Notice that from the above definition, if all observations share the same degree of precision, the multipliers to all observations shall be the same, otherwise they should be different. Conventionally, the later case is referred as “Weighted LS (WLS) estimation” where each multiplier is called the “weight” associated with the observation. The more precise an observation, the higher its associated weight. On the other hand, the former case where all observations share the same precision is referred as LS estimation.

To reduce confusion, from now on LS estimation and WLS estimation are used to refer to the former and later cases, respectively.

Formulate LS estimation problem on a LTI system as follows. Let $\theta_i, i = 1, \dots, n$ be n unknown parameters in the model and $y(j), j = 1, \dots, m$ be m

measurements collected from the mode. It is required that $m \geq n$. Variable $\phi_i(j)$ is called a regression variable (or a regressor) that links the measurements with the model parameters, so that

$$y(j) = \phi_1(j)\theta_1 + \phi_2(j)\theta_2 + \dots + \phi_n(j)\theta_n + \epsilon(j) \quad (7.1)$$

where $\epsilon(j)$ is assumed to be independent measurement noise associated with $y(j)$. Equation (7.1) is called a regression model. Putting (7.1) into matrix form gives

$$y = \Phi\theta + \epsilon \quad (7.2)$$

where

$$\begin{aligned} y &= [y(1) \dots y(m)]^T \\ \Phi &= \begin{bmatrix} \phi_1(1) & \dots & \phi_n(1) \\ \vdots & \ddots & \vdots \\ \phi_1(m) & \dots & \phi_n(m) \end{bmatrix} \\ \theta &= [\theta_1 \dots \theta_n]^T \\ \epsilon &= [\epsilon(1) \dots \epsilon(m)]^T \end{aligned}$$

with y the measurement vector, Φ the regression model matrix, and θ the unknown system parameter vector.

LS estimation of θ in (7.2) is formulated as follows. Select estimation $\hat{\theta} = [\hat{\theta}_1, \dots, \hat{\theta}_n]^T$ so that the following cost function $V(\hat{\theta})$ is minimized.

$$V(\hat{\theta}) = \frac{1}{2} \sum_{i=1}^m e(i)^2 \quad (7.3)$$

where

$$\begin{aligned} e(i) &= y(i) - (\phi_1(i)\hat{\theta}_1 + \phi_2(i)\hat{\theta}_2 + \dots + \phi_n(i)\hat{\theta}_n) \\ &= y(i) - \phi(i)^T \hat{\theta} \end{aligned} \quad (7.4)$$

with

$$\phi(i)^T = [\phi_1(i) \dots \phi_n(i)] \quad (7.5)$$

which is the i th row of Φ . Variable $e(i)$ is called the residual.

Put everything into matrix format. Equation (7.3) becomes

$$\begin{aligned} V(\hat{\theta}) &= \frac{1}{2} \sum_{i=1}^m (y(i) - \phi(i)^T \hat{\theta})^2 \\ &= \frac{1}{2} (y - \Phi\hat{\theta})^T (y - \Phi\hat{\theta}) \end{aligned} \quad (7.6)$$

Denote $e = [e(1), \dots, e(m)]^T$. From (7.4),

$$e = y - \Phi \hat{\theta} \quad (7.7)$$

and (7.6) becomes

$$V(\hat{\theta}) = \frac{1}{2} e^T e \quad (7.8)$$

The quadratic optimization problem given in (7.8) has the following unique analytical solution. Differentiating (7.8) w.r.t. θ gives (using numerator-layout notation)

$$\frac{dV(\hat{\theta})}{d\theta} = \frac{dV(\hat{\theta})}{de} \frac{de}{d\theta} \quad (7.9)$$

$$= -e^T \Phi \quad (7.10)$$

Equating (7.10) to zero gives the minimum of (7.8) as follows.

$$-e^T \Phi = 0 \quad (7.11)$$

Substituting (7.7) into (7.11) gives

$$\begin{aligned} -(y - \Phi \hat{\theta})^T \Phi &= 0 \\ -\Phi^T (y - \Phi \hat{\theta}) &= 0 \\ \Phi^T \Phi \hat{\theta} &= \Phi^T y \\ \hat{\theta} &= (\Phi^T \Phi)^{-1} \Phi^T y \end{aligned} \quad (7.12)$$

provided that $\Phi^T \Phi$ is nonsingular, which is known as the excitation condition. Equation (7.12) is the solution to the LS estimation problem in (7.3).

A Quick Note on Matrix Differentiation

In numerator-layout notation, the scalar $V(\hat{\theta})$ differentiated w.r.t. the vector e in (7.9) is given by

$$\begin{aligned} \frac{dV(\hat{\theta})}{de} &= \begin{bmatrix} \frac{dV(\hat{\theta})}{de(1)} & \cdots & \frac{dV(\hat{\theta})}{de(m)} \end{bmatrix} \\ &= \begin{bmatrix} e(1) & \cdots & e(m) \end{bmatrix} \\ &= e^T \end{aligned}$$

The vector e differentiated w.r.t. the vector θ is given by

$$\begin{aligned}\frac{de}{d\theta} &= \begin{bmatrix} \frac{de(1)}{d\theta_1} & \cdots & \frac{de(m)}{d\theta_n} \\ \vdots & \ddots & \vdots \\ \frac{de(m)}{d\theta_1} & \cdots & \frac{de(m)}{d\theta_n} \end{bmatrix} \\ &= \begin{bmatrix} -\phi_1(1) & \cdots & -\phi_n(1) \\ \vdots & \ddots & \vdots \\ -\phi_1(m) & \cdots & -\phi_n(m) \end{bmatrix} \\ &= -\Phi\end{aligned}$$

where

$$\frac{de(i)}{d\theta_j} = -\phi_j(i)$$

because of (7.4).

Therefore,

$$\frac{dV(\hat{\theta})}{d\theta} = \frac{dV(\hat{\theta})}{de} \frac{de}{d\theta} = -e^T \Phi$$

which is (7.10).

A WLS estimation problem can be formulated similarly, by changing (7.3) with

$$V(\hat{\theta}) = \frac{1}{2} \sum_{i=1}^m w(i)e(i)^2$$

The solution to a WLS estimation can be similarly derived and the result is

$$\hat{\theta} = (\Phi^T W \Phi)^{-1} \Phi^T W y$$

where $W = \text{diag}(w(1), \dots, w(m))$.

7.2.2 Statistics Properties of LS Estimation

In the discussions below, We will assume that all measurement noise $\epsilon(i)$, $i = 1, \dots, m$ are i.i.d. noise, whose mean and variance are zero and σ^2 respectively.

Substituting (7.2) into (7.12) gives

$$\begin{aligned}\hat{\theta} &= (\Phi^T \Phi)^{-1} \Phi^T (\Phi \theta + \epsilon) \\ &= \theta + (\Phi^T \Phi)^{-1} \Phi^T \epsilon\end{aligned}\tag{7.13}$$

Using (7.13), the following statistics properties of LS estimation can be drawn.

$$\begin{aligned} E[\hat{\theta}] &= \theta \\ \text{Cov}[\hat{\theta}] &= E[\hat{\theta}\hat{\theta}^T] \\ &= \sigma^2 (\Phi^T \Phi)^{-1} \end{aligned}$$

which suggests that the estimator is zero mean. With the fixed number n of parameters to be estimated, the estimation variance decreases as the number m of observations increases, and the variance converges to zero eventually. The variance of the noise σ^2 can potentially be estimated from the state estimate $\hat{\theta}$. Details are given as follows. Substituting (7.2), (7.12) into (7.7) gives

$$e = (I - \Phi (\Phi^T \Phi)^{-1} \Phi^T) \epsilon \quad (7.14)$$

Using (7.14),

$$E[ee^T] = \sigma^2 (I - \Phi (\Phi^T \Phi)^{-1} \Phi^T) (I - \Phi (\Phi^T \Phi)^{-1} \Phi^T)^T \quad (7.15)$$

Taking the trace of matrix from both sides of (7.15) gives

$$E[e^T e] = (m - n) \sigma^2 \quad (7.16)$$

From (7.8) and (7.16),

$$\sigma^2 = \frac{2}{m - n} E[V(\hat{\theta})]$$

can be used to estimate the variance of the noise.

7.2.3 Recursive LS Estimation

Recursive state estimation allows “updating” the state estimate each time a new measurement set becomes available. It takes advantages of previously calculated state estimate with less measurements, instead of calculating everything from scratch, which is also known as the batch estimation. Recursive estimation saves computational and memory burden than the batch estimation. Given that observations are often naturally obtained sequentially in real time, recursive estimation has a large demand in practice.

Let each measurement be obtained associated with a incremental time stamp $k = 1, 2, \dots$. While the number of parameters to be estimated remains constant n , the number of measurement m is dynamic and replaced by the notion k .

It is worth mentioning that the size of Φ in (7.2) grows with k , in the sense that each time a new measurement becomes available, Φ grows by one

row. For convenience, $\Phi(k)$ is used to describe Φ in (7.2) when there are k measurements.

As more and more measurements come in, the state estimate $\hat{\theta}$ also evolves and become more and more accurate. Denote $\hat{\theta}(k)$ the state estimate when up to k measurements are available, $k \geq n$.

In recursive estimation schema, the estimator is formulated as a Markov process, and the state estimate $\hat{\theta}(k+1)$ is formulated as a function of $\hat{\theta}(k)$ and $y(k+1)$, and does not directly involve the earlier state estimates $\hat{\theta}(k-1)$, $\hat{\theta}(k-2)$, ... or measurements $y(k)$, $y(k-1)$, This is possible because the information of $\hat{\theta}(k-1)$, $\hat{\theta}(k-2)$, ... and $y(k)$, $y(k-1)$, ... are integrated into $\hat{\theta}(k)$.

The derivation of the recursive estimation is as follows. Notice that $\phi(i)^T$ in (7.5) is the i th row of $\Phi(k)$, hence

$$\Phi(k)^T \Phi(k) = \sum_{i=1}^k \phi(i) \phi(i)^T \quad (7.17)$$

Assume that the excitation condition holds. Denote $P(k) = (\Phi(k)^T \Phi(k))^{-1}$, or equivalently $P(k)^{-1} = \Phi(k)^T \Phi(k)$. Use (7.17),

$$P(k+1)^{-1} = P(k)^{-1} + \phi(k+1) \phi(k+1)^T \quad (7.18)$$

which can be calculated recursively.

Rewrite (7.12) as follows.

$$\begin{aligned} \hat{\theta}(k) &= \left(\sum_{i=1}^k \phi(i) \phi(i)^T \right)^{-1} \sum_{i=1}^k \phi(i) y(i) \\ &= P(k) \sum_{i=1}^k \phi(i) y(i) \\ \hat{\theta}(k+1) &= \left(\sum_{i=1}^{k+1} \phi(i) \phi(i)^T \right)^{-1} \sum_{i=1}^{k+1} \phi(i) y(i) \\ &= P(k+1) \left(\sum_{i=1}^k \phi(i) y(i) + \phi(k+1) y(k+1) \right) \\ &= P(k+1) \left(P(k)^{-1} \hat{\theta}(k) + \phi(k+1) y(k+1) \right) \\ &= P(k+1) \left((P(k+1)^{-1} - \phi(k+1) \phi(k+1)^T) \hat{\theta}(k) \right. \\ &\quad \left. + \phi(k+1) y(k+1) \right) \\ &= \hat{\theta}(k) \\ &\quad + P(k+1) \phi(k+1) \left(y(k+1) - \phi(k+1)^T \hat{\theta}(k) \right) \end{aligned} \quad (7.19)$$

Equation (7.18) and (7.19) give the recursive LS estimation. Notice that the (7.18) gives the recursive calculation of $P(k)^{-1}$. The recursive calculation of $P(k)$ can be easily derived from (7.18) using matrix inversion lemma as follows.

$$\begin{aligned}
 P(k+1) &= P(k) - P(k)\phi(k+1) \\
 &\quad \times (I + \phi(k+1)^T P(k)\phi(k+1))^{-1} \phi(k+1)^T P(k) \\
 P(k+1)\phi(k+1) &= P(k)\phi(k+1) \\
 &\quad \times (I - (I + \phi(k+1)^T P(k)\phi(k+1))^{-1} \\
 &\quad \times \phi(k+1)^T P(k)\phi(k+1)) \\
 &= P(k)\phi(k+1) (I + \phi(k+1)^T P(k)\phi(k+1))^{-1} \\
 &\quad \times ((I + \phi(k+1)^T P(k)\phi(k+1)) \\
 &\quad - \phi(k+1)^T P(k)\phi(k+1)) \\
 &= P(k)\phi(k+1) (I + \phi(k+1)^T P(k)\phi(k+1))^{-1}
 \end{aligned}$$

where notice that in the above equations the identity matrix I under the scope of our discussion should be 1, as the measurement at any instant, $y(k)$, is a scalar. We followed the convention using I because in practice, the measurement at any instant can be a vector. Hopefully this explanation helps to clear some confusion.

Denote $K(k) = P(k)\phi(k)$ the Kalman gain, and residual $\varepsilon(k+1) = y(k+1) - \phi(k+1)^T \hat{\theta}(x)$ the innovation vector, where $\phi(k+1)^T \hat{\theta}(x)$ can be interpreted as a “guess” of $y(k+1)$ using the state estimate $\hat{\theta}(x)$. Recursive LS estimation can be taken as a special case of Kalman filter where (a) measurement $y(k)$ is a scalar; (b) process matrix is unit matrix; (c) input and process noise are both zero.

7.2.4 LS Estimation with Time-Varying Parameters

If the parameter is time varying and has a solid dynamic model, considering using Kalman filter and other parametric model based estimators. If not, consider two scenarios: (a) parameters changes abruptly and infrequently; (b) parameters changes slowly and continuously.

Consider case (a). In this case, the abrupt changes can be detected from the magnitude of the innovation vector. The innovation vector covariance is jointly decided by the measurement noise and the state estimate precision. When the parameter is constant, the covariance matrix converges to a certain value that can be calculated analytically. A simple test can be used to detect when the innovation vector violates the statistics assumption, indicating that there is an abrupt parameters change.

In such case, just “reset” the system by assuming the new measurement is the first set of measurement collected by the system. In practice, this can

be achieved equivalently by reset $P(k-1)$ and/or $P(k)$ to a diagonal matrix with large elements.

Consider case (b). One way to handle this parameter change is to use moving horizon estimation, or introducing a forgetting factor into the cost function. An example of introducing forgetting factor is given as follows. In (7.6), set

$$V(\hat{\theta}) = \frac{1}{2} \sum_{i=1}^k \lambda^{k-i} \left(y(i) - \phi(i)^T \hat{\theta} \right)^2 \quad (7.20)$$

where $0 < \lambda < 1$ is a forgetting factor, and the method given in (7.20) is called exponential forgetting. By introducing the forgetting factor, the contribution of innovation introduced by early measurements and state estimates to the cost function is reduced, thus reducing their impact on the state estimates.

The batch and recursive algorithms to calculate the state estimates for (7.20) can be derived similarly. Some key derivations are given below.

Equation (7.12) becomes

$$\begin{aligned} \hat{\theta} &= (\Phi^T \Lambda \Phi)^{-1} \Phi^T \Lambda y \\ \Lambda &= \text{diag} \left(\lambda^{m-1} \quad \dots \quad \lambda \quad 1 \right) \end{aligned}$$

Equation (7.18) becomes

$$\begin{aligned} P(k)^{-1} &= \Phi(k)^T \Lambda \Phi(k) \\ P(k+1)^{-1} &= \lambda P(k)^{-1} + \phi(k+1)\phi(k+1)^T \end{aligned}$$

Applying matrix inversion lemma to the above equation gives

$$\begin{aligned} P(k+1) &= \frac{1}{\lambda} P(k) - \frac{1}{\lambda} P(k) \phi(k+1) \\ &\quad \times (\lambda I + \phi(k+1)^T P(k) \phi(k+1))^{-1} \phi(k+1)^T P(k) \\ K(k+1) &= P(k+1) \phi(k+1) \\ &= P(k) \phi(k+1) (\lambda I + \phi(k+1)^T P(k) \phi(k+1))^{-1} \end{aligned}$$

Equation (7.19) has the same eventual form of

$$\hat{\theta}(k+1) = \hat{\theta}(k) + P(k+1) \phi(k+1) \left(y(k+1) - \phi(k+1)^T \hat{\theta}(k) \right)$$

7.2.5 Simplified LS Estimation: Projection Algorithm, SA, and LMS

The recursive LS algorithm (7.18) and (7.19) requires the updates of $P(k)$ which involves matrix inversion, where $P(k)$ has a size of $n \times n$, and n is the

number of parameters to be estimated. When n is large, this calculation can be time consuming.

Consider interpreting the cost function (7.6) as follows.

$$\hat{\theta}(k) = \arg \min_{\hat{\theta}} V_k(\hat{\theta}) = \arg \min_{\hat{\theta}} \frac{1}{2} \sum_{i=1}^k \left(y(i) - \phi(i)^T \hat{\theta} \right)^2 \quad (7.21)$$

$$\hat{\theta}(k+1) = \arg \min_{\hat{\theta}} V_{k+1}(\hat{\theta}) = \arg \min_{\hat{\theta}} \frac{1}{2} \sum_{i=1}^{k+1} \left(y(i) - \phi(i)^T \hat{\theta} \right)^2 \quad (7.22)$$

where

$$V_i(\hat{\theta}) = \frac{1}{2} \sum_{j=1}^i \left(y(j) - \phi(j)^T \hat{\theta} \right)^2$$

is the cost function formulated using the first i measurements $y(1), \dots, y(i)$.

Formulating (7.22) as a function of $V_k(\hat{\theta})$ gives

$$\hat{\theta}(k+1) = \arg \min_{\hat{\theta}} \left[V_k(\hat{\theta}) + \frac{1}{2} \left(y(k+1) - \phi(k+1)^T \hat{\theta} \right)^2 \right] \quad (7.23)$$

We already know that $\hat{\theta}(k)$ minimizes $V_k(\hat{\theta})$ from (7.21), and $V_k(\hat{\theta})$ happens to be the first term in (7.23). Therefore, (7.23) is essentially an optimization problem that concerns the following terms

$$\|\hat{\theta} - \hat{\theta}(k)\| \quad (7.24)$$

and

$$\|y(k+1) - \phi(k+1)^T \hat{\theta}\| \quad (7.25)$$

where $\|\cdot\|$ denotes the Euclidean norm of a vector, and the weights associated with the two terms in the optimization problem are determined by the covariance of the state estimate and measurement noise.

Consider a modification to the optimization problem in (7.23) as follows. Let (7.24) be the only term in the optimization, and (7.25) be a equality constraint, i.e.,

$$\min \quad \frac{1}{2} \left(\hat{\theta} - \hat{\theta}(k) \right)^T \left(\hat{\theta} - \hat{\theta}(k) \right) \quad (7.26)$$

$$\text{s.t.} \quad y(k+1) - \phi(k+1)^T \hat{\theta} = 0 \quad (7.27)$$

Notice that this modified optimization problem given by (7.26) does not necessarily give the same result as the original problem in (7.23). This is because $\hat{\theta}(k+1)$ in (7.23) may not meet the equality constraint. Nevertheless, (7.26) is an approximation to $\hat{\theta}(k+1)$ in (7.23).

Optimization (7.26) can be solved using Lagrange multiplier method. Details are neglected here, and the result is given by

$$\hat{\theta}(k+1) = \hat{\theta}(k) + \frac{\phi(k+1)}{\phi(k+1)^T \phi(k+1)} \left(y(k+1) - \phi(k+1)^T \hat{\theta}(k) \right) \quad (7.28)$$

Parameters $0 < \gamma < 2$ and $\alpha > 0$ are added to (7.28) as shown below. The motivations of adding γ and α are to make it possible to change the step length of the parameter adjustment and to avoid a potential problem what would occur when $\phi(k+1) = 0$.

$$\hat{\theta}(k+1) = \hat{\theta}(k) + \frac{\gamma \phi(k+1)}{\alpha + \phi(k+1)^T \phi(k+1)} \left(y(k+1) - \phi(k+1)^T \hat{\theta}(k) \right) \quad (7.29)$$

which is known as the projection algorithm. The choice $0 < \gamma < 2$ guarantees the convergency of (7.29).

One may think the quality constraint (7.27) is too strong. When measurement error is considered, a modified version of (7.28) is given below.

$$\hat{\theta}(k+1) = \hat{\theta}(k) + \frac{\phi(k+1)}{\sum_{i=1}^{k+1} \phi(i)^T \phi(i)} \left(y(k+1) - \phi(k+1)^T \hat{\theta}(k) \right) \quad (7.30)$$

which is known as the stochastic approximation (SA) algorithm.

A further simplification to (7.29) and (7.30) is given below.

$$\hat{\theta}(k+1) = \hat{\theta}(k) + \gamma \phi(k+1) \left(y(k+1) - \phi(k+1)^T \hat{\theta}(k) \right)$$

which is known as the least mean square (LMS) algorithm.

7.2.6 Models

The LS estimation algorithm can be applied to the models introduced in this section. By assuming a model, experiments can be designed and empirical data can be collected which serves as the input to the LS estimation algorithm. The parameters of the model can then be decided.

The initial states of the plant is ignored for now. In practice, with the model and parameters known, state estimation is carried out to determine the state of the system.

Finite Impulse Response (FIR) Model

An LTI system can be uniquely characterized by its impulse response. A demonstrative figure is given in Fig. 7.6. For an asymptotically stable system, the transient will converge to zero exponentially fast, depending on the position of the poles in the transfer function.

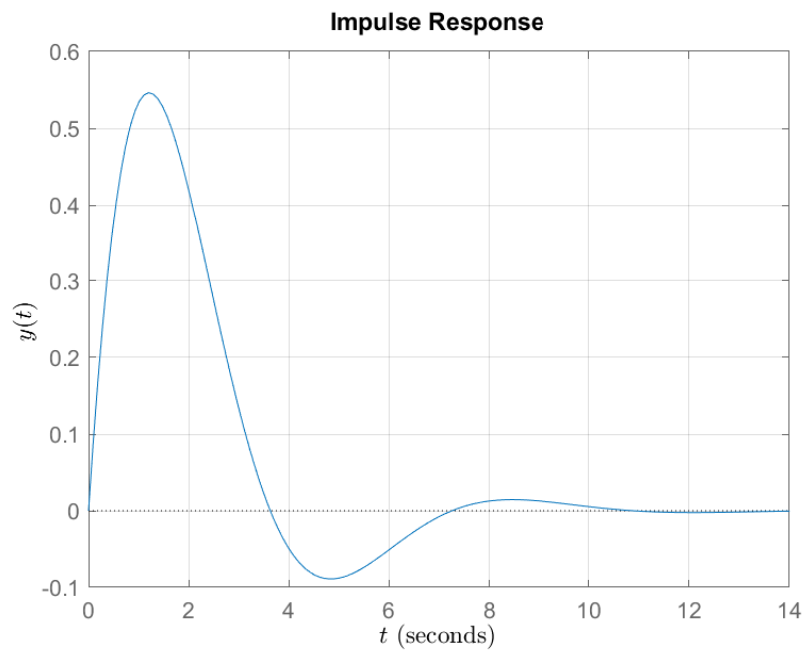


FIGURE 7.6
LTI system impulse response demonstration.

Consider sampling the outputs of Fig. 7.6 in discrete time domain. The input at k , i.e., $u(k)$, would affect the consequent outputs $y(k+1), y(k+2), \dots$. This sequence can go infinitely long in theory, but in practice we only consider the first finite number of outputs before y settles to zero. The length of the sequence is determined by the sampling period and the time constant of the system.

Due to the superposition of the LTI system, the above also indicates that output of the system at k , i.e., $y(k)$, is affected by the past input signals $u(k-1), u(k-2), \dots$, as given by the following equation. Likewise, only finite number of inputs are included in the equation.

$$y(k) = b_1 u(k-1) + b_2 u(k-2) + \dots + b_n u(k-n) \quad (7.31)$$

Equation (7.31) is known as the FIR model of a system. It has n parameters to be estimated, and it fits well into the LS estimation schema described by (7.1).

General Transfer Function Model

Consider a more general LTI model described by the following input-output relations.

$$\begin{aligned} A(q)y(k) &= B(q)u(k) \\ A(q) &= q^n + a_1 q^{n-1} + \dots + a_n \\ B(q) &= b_1 q^{m-1} + b_2 q^{m-2} + \dots + b_m \end{aligned}$$

where q is the forward shift operator $qy(k) = y(k+1)$. The goal of the parameter estimation is to estimate parameters $a_i, i = 1, \dots, n$ and $b_i, i = 1, \dots, m$.

Rewrite (7.32) as follows.

$$\begin{aligned} y(k) &= -a_1 y(k-1) - a_2 y(k-2) - \dots - a_n y(k-n) \\ &\quad + b_1 u(k-1) + b_2 u(k-2) + \dots + b_m u(k-m) \end{aligned} \quad (7.32)$$

In (7.32), treat $y(k-1), \dots, y(k-n)$ and $u(k-1), \dots, u(k-m)$ as regressors and a LS estimation problem can be formulated.

Nonlinear Model

LS estimation can be applied to nonlinear models as long as the output to the parameters to be estimated are of linear relationship. An example is given as follows. Consider

$$y(k) + ay(k-1) = b_1 u(k-1) + b_2 \sin(u(k-2))$$

This is a nonlinear model due to the sine function applied to $u(k-2)$. However, it has no impact on parameter estimation, as $y(k)$ is linear to a, b_1 and b_2 . The nonlinear portion is wrapped into the regressor.

Stochastic Model

LS estimation is designed to handle independent and identically distributed Gaussian noise. This indicates that the noise should be white noise, i.e., $\epsilon(i)$ and $\epsilon(j)$, $i \neq j$ shall be independent. In addition, the noise shall not correlate with the regressor by any means.

In the case where the noise are not white, i.e., $\epsilon(i)$ and $\epsilon(j)$ correlates for some $i \neq j$, it is possible to get the parameters estimate as follows. Formulate the system dynamic model as

$$A(q)y(k) = B(q)u(k) + C(q)\varepsilon(k)$$

where $\varepsilon(k)$ are re-formulated i.i.d. noise and $C(q)$ is used to describe the correlation of the noise. The parameters in $C(q)$ will be estimated together with $A(q)$ and $B(q)$ to get a sense of the bandwidth of the original noise $\epsilon(k)$.

Part IV

System Identification



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Classic System Identification and State Estimation

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