

# **EE 4323 – Nonlinear Control Systems**

## **Module 10: Sinusoidal-Input Describing Function Methods – SIDF Analysis**

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## Topic Outline

- Introduction
- Basic Concepts
- SIDF Calculations
- Qualitative Behavior of SIDFs; Heuristics
- Classical Harmonic Balance (Limit Cycle Conditions)
- Classical SIDF I/O Analysis (“Transfer Functions”)
- Examples
- Systems with Multiple Nonlinearities
- Modern Algebraic SIDF Methods
- Examples

### References:

- D. P. Atherton, *Nonlinear Control Engineering*, Van Nostrand, 1975 (Reprinted as Student Edition, 1982).
- A. Gelb & W. Vander Velde, *Multiple-Input DF's and Nonlinear System Design*, McGraw-Hill, 1968.
- J. H. Taylor, *Describing Functions*, an article in the *Electrical Engineering Encyclopedia*, John Wiley & Sons, Inc., New York, 1999.

# Introduction

- Problem to be addressed: Analyzing Periodic Phenomena
  - **Nonlinear oscillations** (limit cycles)
  - **Response of a nonlinear system to sinusoidal inputs**
- Considerations:
  - Simulation is often too time-consuming and cumbersome, especially for parametric trade-off studies.
  - There are situations in which simulation is almost useless for studying periodic behavior (e.g., flight control systems)
  - It's the only method that handles **high-order systems** and/or **systems with multiple nonlinearities** with ease

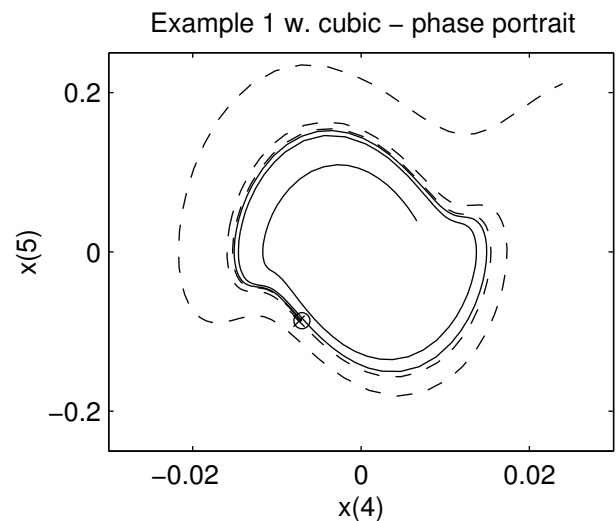
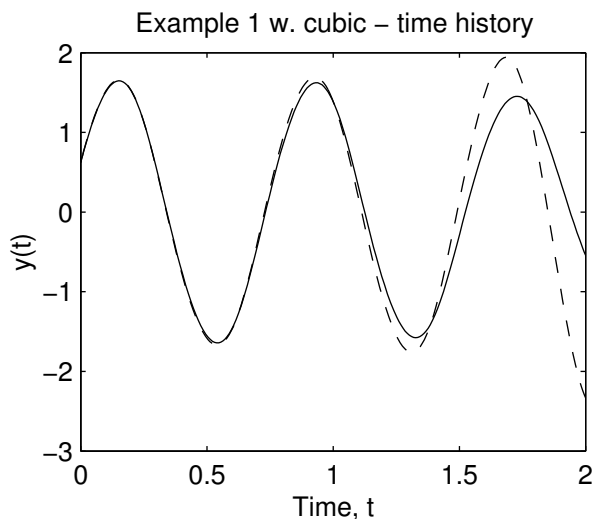
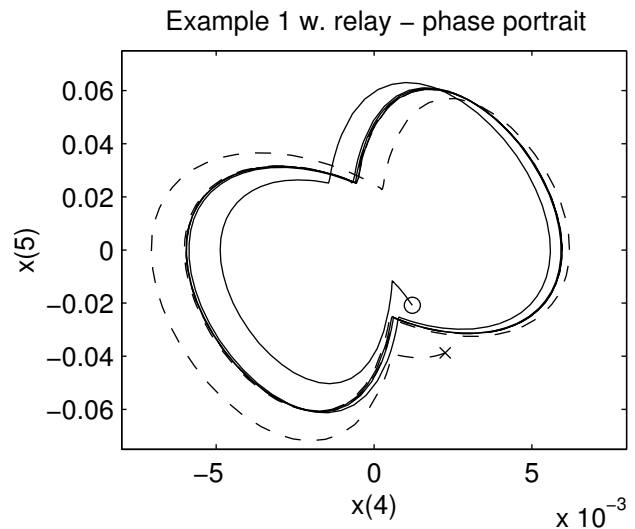
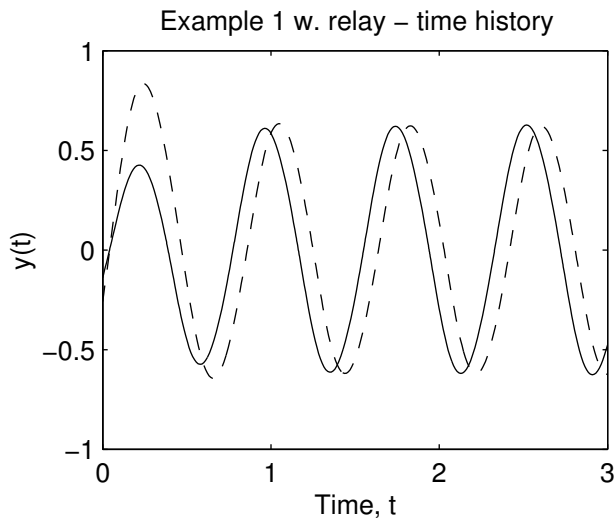
## Importance of Periodic Effects

- A limit cycle may be **desired**, with a specified frequency and amplitude – can you design the system? This is the only practical way to build an oscillator ...
- A limit cycle may be **unwanted but unavoidable** – is it small enough or slow enough to be acceptable?
- An unstable limit cycle is a **stability boundary** – is it large enough?
- A nonlinear system may be driven by **sinusoidal inputs** – how will it respond?
  - SIDF I/O models for different amplitudes → diagnosis
  - SIDF I/O models exhibit interesting phenomena, e.g., “jump resonance”
  - SIDF I/O models form an excellent basis for control system design – later

## Definition of Limit Cycles

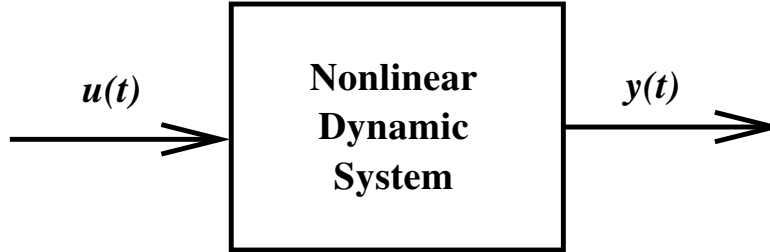
A simple limit cycle is a periodic trajectory in the state space,  $x^*(t + T) = x^*(t)$ ,  $\forall t$  where  $T$  is the period, such that all nearby trajectories:

- asymptotically approach  $x^*(t)$  (a **stable** limit cycle) **or**
- diverge from  $x^*(t)$  (an **unstable** limit cycle)



Note that **limit cycles** are not generally **ellipses**

## Definition of Sinuoidal-Input Response



- Input:  $u(t) = u_0 + a \cos(\omega t)$
- Output: **may be** periodic; to be safe, run a simulation:

$$y = \sum_{k=0}^{\infty} b_k \cos(k\omega t + \psi_k)$$

- “Transfer function” for the fundamental component:

$$G(j\omega; u_0, a) = \frac{b_1}{a} \exp(j\psi_1) \quad (1)$$

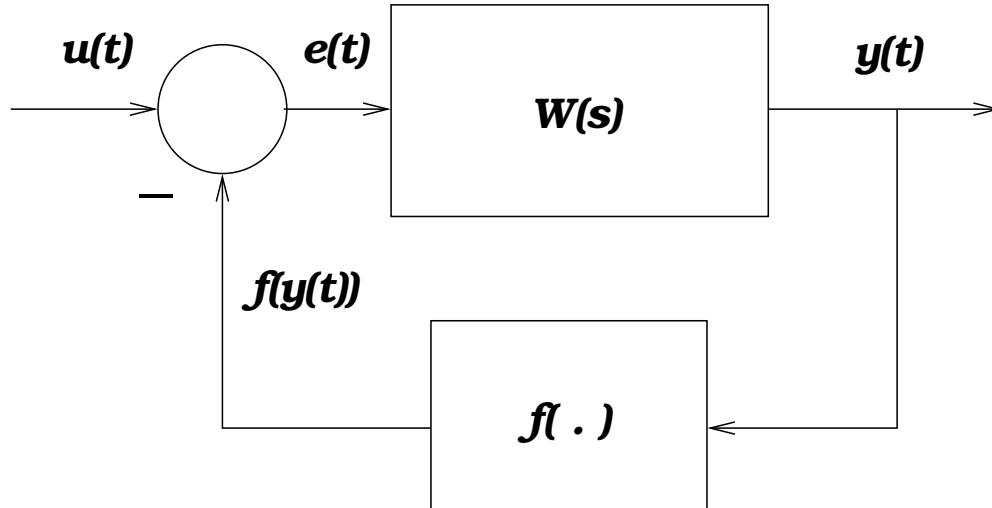
- Operating point (“DC level”):  $b_0(u_0, a)$

Note that the “transfer function” and operating point are coupled

Hereafter we will call  $G(j\omega; u_0, a)$  an **SIDF Input/Output Model** (SIDF I/O Model)

## Basic System Models

Classical Case:



$$\begin{aligned}
 Y(s) \triangleq \mathcal{L}(y(t)) &= \frac{p(s)}{q(s)} E(s) \\
 &\triangleq W(s) \mathcal{L}(e(t)) \\
 e(t) &= u(t) - f(y(t))
 \end{aligned} \tag{2}$$

Multivariable Case:

$$\begin{aligned}
 \dot{x} &= f(x, u(t)) \\
 y(t) &= h(x, u(t))
 \end{aligned} \tag{3}$$

For **limit cycle analysis**:  $u(t) = u_0$

For **forced response**:  $u(t) = u_0 + \text{Re}[a \exp(j\omega t)]$

## Basic System Model in MATLAB

- Given:  $W(s) = 2/(s^2 + 3s + 7)$  and  $f(y) = 4y^3$
- In ODE form:  $\ddot{y} + 3\dot{y} + 7y + 8y^3 = 2u(t)$
- In state-space form (one realization):  $x^T = [y \ \dot{y}]$

$$\dot{x}_1 = x_2$$

$$\dot{x}_2 = -7x_1 - 3x_2 - 8x_1^3 + 2u(t)$$

- In MATLAB:

```
function xdot = basic(t,x)
% Example in controllable canonical form:
% JH Taylor - 9 July 2002

num = 2; den = [ 1 3 7 ];
u = 3.5 * sin(10*t);
xdot(1) = x(2);
xdot(2) = - den(3)*x(1) - den(2)*x(2) + num*(u - 4*x(1)^3);
xdot = xdot(:);
```

- To run a simulation:

```
tspan = [ 0 6*pi/10 ]; % three cycles of sin(10*t)
x0 = [ 1.2 -3.4 ]; % arbitrary initial condition
[t,x] = ode45('basic',tspan,x0); % model is in basic.m
plot(t,x(:,1)); % plot first state only
```



## Basic Idea of the Describing Function Method

- Knowledge of signal **form** and **amplitude** is essential in understanding the behavior of a nonlinear system
- Linear system approaches are the most powerful tools we have for analysis
- Replacing nonlinearities with **signal-dependent linear gains (“describing functions”)** provides the best way to take advantage of linear system approaches to understand the behavior of a nonlinear system; this process is called “quasilinearization”
- You will see examples that use the *machinery* of Nyquist plots and root locus plots, ... but the **underlying theory** is entirely different. The Routh-Hurwitz technique can also be exploited, but it is difficult compared to the others

## Classical Definition of a SIDF

- Given: a specific nonlinearity  $f(v)$  and an input signal form,  
 $v(t) = v_0 + \text{Re}(a \exp(j\omega t))$
- Find: the quasilinear model

$$f(v) \approx f_0(v_0, a) + N(v_0, a) \cdot a \exp(j\omega t) \quad (4)$$

such that **mean square approximation error** is minimized

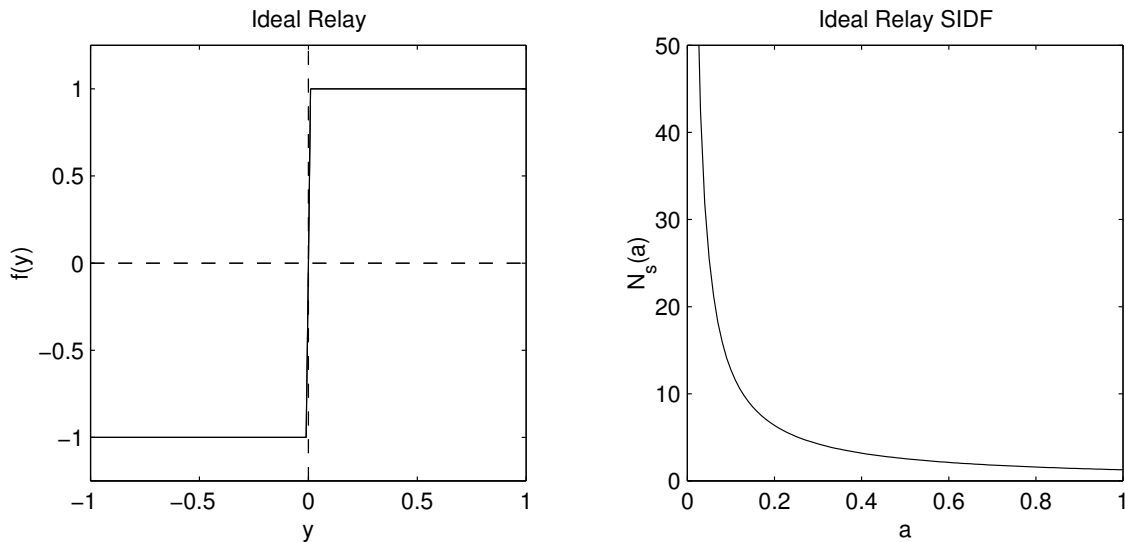
- Method 1:  $f_0(v_0, a)$  and  $N(v_0, a)$  are determined by **Fourier analysis** (constant plus first harmonic terms)
- Method 2:  $f_0(v_0, a)$  and  $N(v_0, a)$  are determined by using **trigonometric identities** (for power-law and product-type nonlinearities)

# Calculating SIDFs – Piece-Wise-Linear Case

**Ideal relay:**  $f(y) = D \cdot \text{sgn}(y)$  where we assume no DC level,  
 $y(t) = a \cos(\omega t)$

Set up and evaluate the integral for the first Fourier coefficient divided by  $a$  as follows:

$$\begin{aligned}
 N_s(a) &= \frac{1}{\pi a} \int_0^{2\pi} f(a \cos(x)) \cdot \cos(x) dx \\
 &= \frac{4 D}{\pi a} \int_0^{\pi/2} \cos(x) dx \quad (\text{by symmetry}) \\
 &= \frac{4 D}{\pi a}
 \end{aligned} \tag{5}$$



**This makes good, intuitive sense;** this SIDF was used to generate the **stable limit cycle** on slide 5

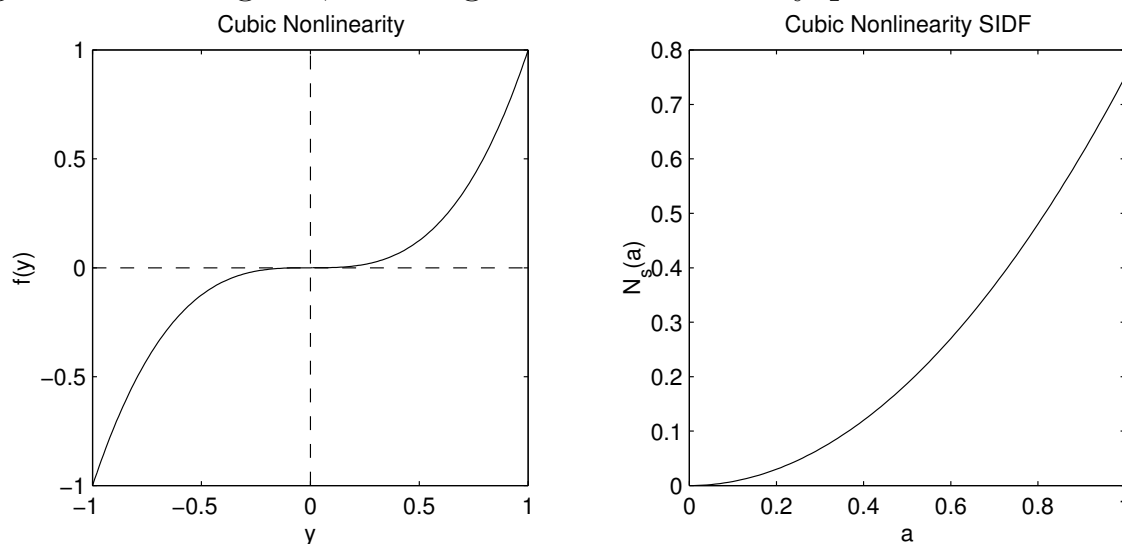
## Calculating SIDs – Trig. Identity Case

**Cubic nonlinearity:**  $f(y) = K y^3(t)$ ; again, assume  $y(t) = a \cos(\omega t)$

Directly write the Fourier expansion using trigonometric identities:

$$\begin{aligned}
 f(a \cos(\omega t)) &= K [a \cos(\omega t)]^3 \\
 &= K a^3 \left[ \frac{3}{4} \cos(\omega t) + \frac{1}{4} \cos(3\omega t) \right] \\
 &\approx \frac{3 K a^2}{4} \cdot a \cos(\omega t)
 \end{aligned} \tag{6}$$

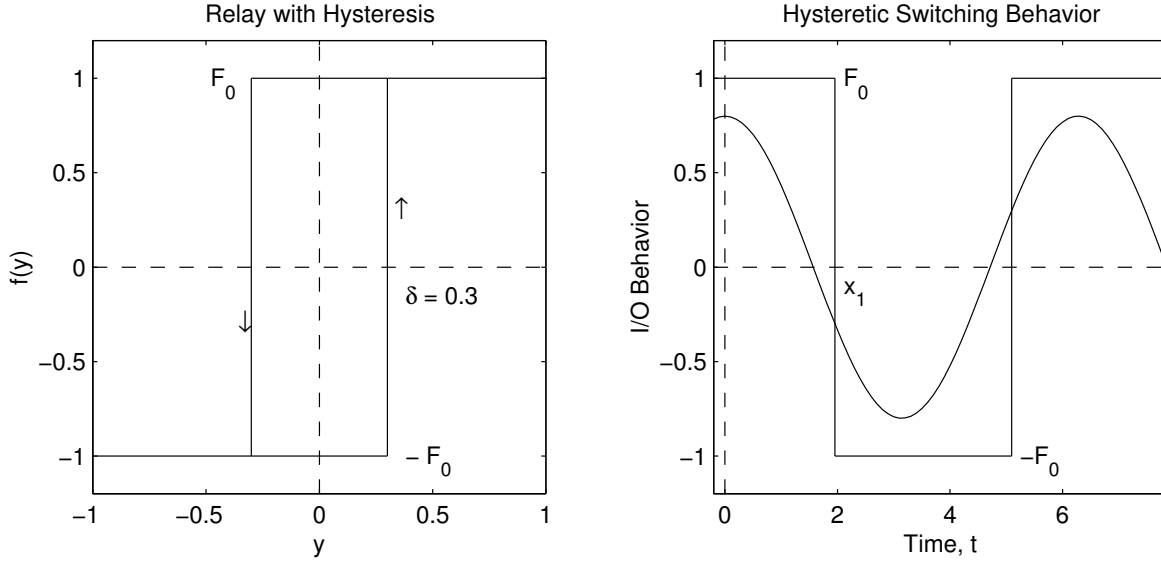
so  $N_s(a) = 3 K a^2/4$ . Trigonometric identities are a shortcut to using Fourier integrals; use trig. identities for any power-law element.



**This also makes good, intuitive sense;** this SIDF was used to generate the **unstable limit cycle** on slide 5

# Calculating SIDFs – Multi-valued Case

Setting up the Fourier integrals requires care:



$$\begin{aligned}
 N(a) &= \frac{1}{\pi a} \int_0^{2\pi} f(a \cos(x)) \cdot \exp(-jx) dx \\
 &= \frac{2 F_0}{\pi a} \left\{ \int_0^{x_1} \exp(-jx) dx - \int_{x_1}^{\pi} \exp(-jx) dx \right\} \quad (\text{by symmetry})
 \end{aligned}$$

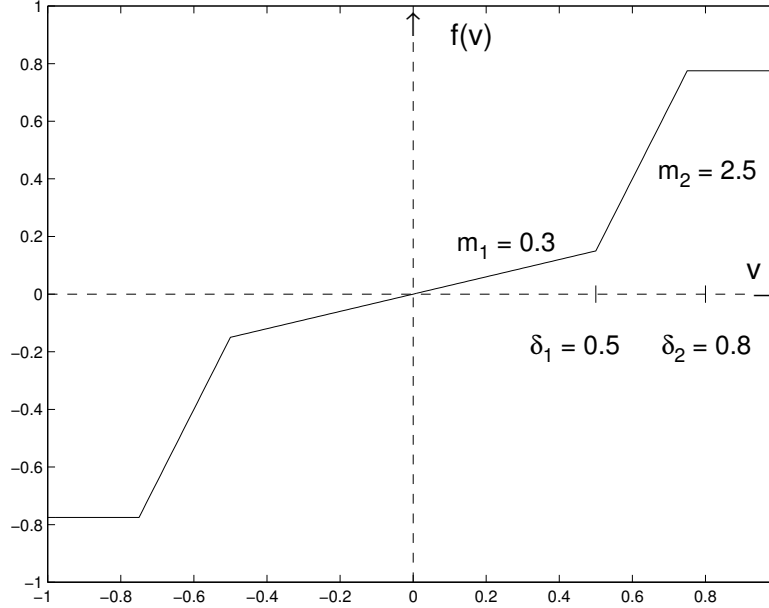
where  $x_1 = \cos^{-1}(-\delta/a)$ ;

$$= \begin{cases} \frac{4 F_0}{\pi a} \left\{ \sqrt{1 - (\delta/a)^2} - j \delta/a \right\} & a > \delta \\ 0 & a \leq \delta \end{cases} \quad (7)$$

Note that  $N(a) \triangleq 0$  if  $a \leq \delta$  ... the relay **does not switch**  $\Rightarrow$  the output is not periodic; more importantly, **this SIDF is complex-valued**, because it causes **lag**

## Calculating SIDFs – Details

Given: a typical piecewise linear characteristic  $f(v)$ ,



Set up the Fourier integral for this function for  $a > \delta_2$ :

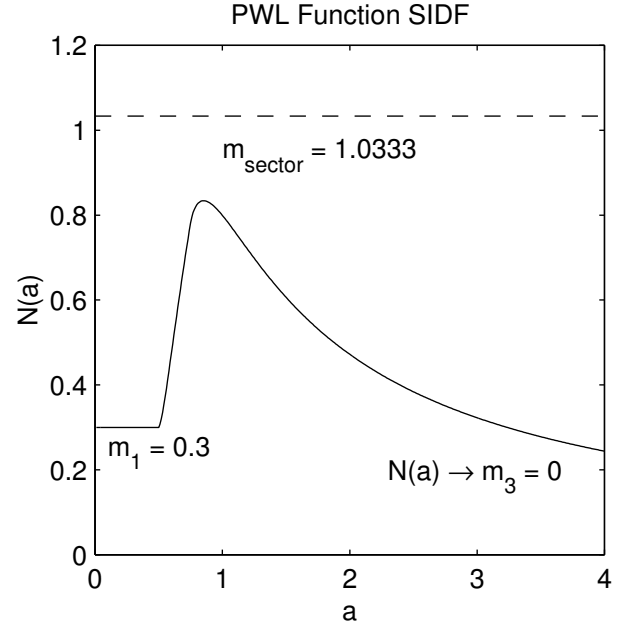
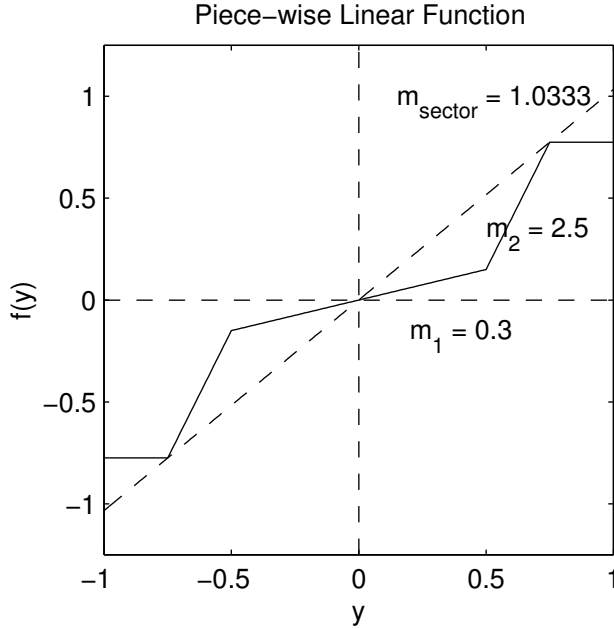
- Integrate from 0 to  $\pi/2$ , by symmetry, and break the integral into three intervals,  $[0, x_1]$ ,  $[x_1, x_2]$ ,  $[x_2, \pi/2]$  where  $x_1 = \cos^{-1}(\delta_2/a)$  and  $x_2 = \cos^{-1}(\delta_1/a)$
- Be sure to use the correct function for  $f(v)$  in each interval,  $f(v) = m_1 v$  in interval 3,  $f(v) = m_2(v - \delta_1) + m_1 \delta_1$  in interval 2,  $f(v) = m_2 \delta_2 - (m_2 - m_1) \delta_1 \triangleq D_0$  in interval 1

- Therefore the complete SIDF Fourier integral is:

$$N(a) = \frac{4}{\pi a} \left[ \int_0^{x_1} D_0 \cos(x) dx + \int_{x_1}^{x_2} [m_2 a \cos(x) - (m_2 - m_1) \delta_1] \cos(x) dx + \int_{x_2}^{\pi/2} m_1 a \cos^2(x) dx \right]$$

# Qualitative Behavior of SIDFs

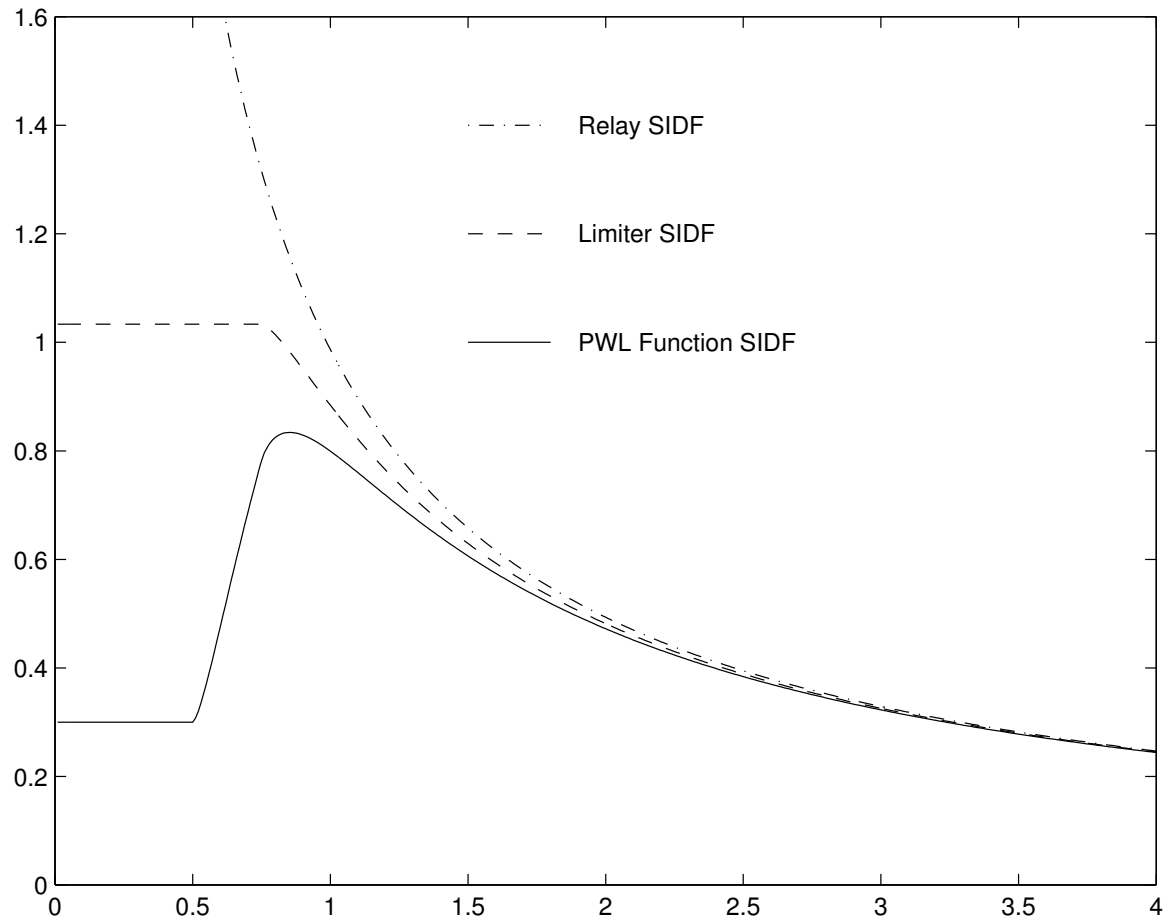
Considering the same nonlinearity  $f(v)$  –



- For small signals  $N(a) \approx N(0) = [df/dv]_{v=0} = m_1$  (if the derivative exists)
- The SIDF cannot lie outside the slopes of the enclosing sector
- The SIDF is always continuous, even though the nonlinearity derivative is discontinuous
- The SIDF always approaches the ultimate slope of the nonlinearity as  $a \rightarrow \infty$  (zero for this example)

## Qualitative Behavior of SIDFs (Cont'd)

- For large signals ( $a > 2$ ) the “details” near the origin do not make much difference



**Conclusion:** In many cases we don't need to evaluate  $N(a)$  exactly for a **qualitative analysis** ...



## Calculating SIDFs in MATLAB

- First, define the basic “saturation function” used in calculating SIDFs for piece-wise-linear functions:

$$f_{\text{sat}} = \begin{cases} \text{sign}(x), & |x| \geq 1 \\ 2 [\sin^{-1}(x) + x\sqrt{1-x^2}] / \pi, & |x| < 1 \end{cases} \quad (8)$$

- The SIDF for a general limiter is  $N_{LIM}(a) = m f_{\text{sat}}(\delta/a)$
- The SIDF for the piece-wise-linear example is  $N_{PWL}(a) = m_1 f_{\text{sat}}(\delta_1/a) + m_2 [f_{\text{sat}}(\delta_2/a) - f_{\text{sat}}(\delta_1/a)]$
- Therefore the previous plots are obtained as follows:

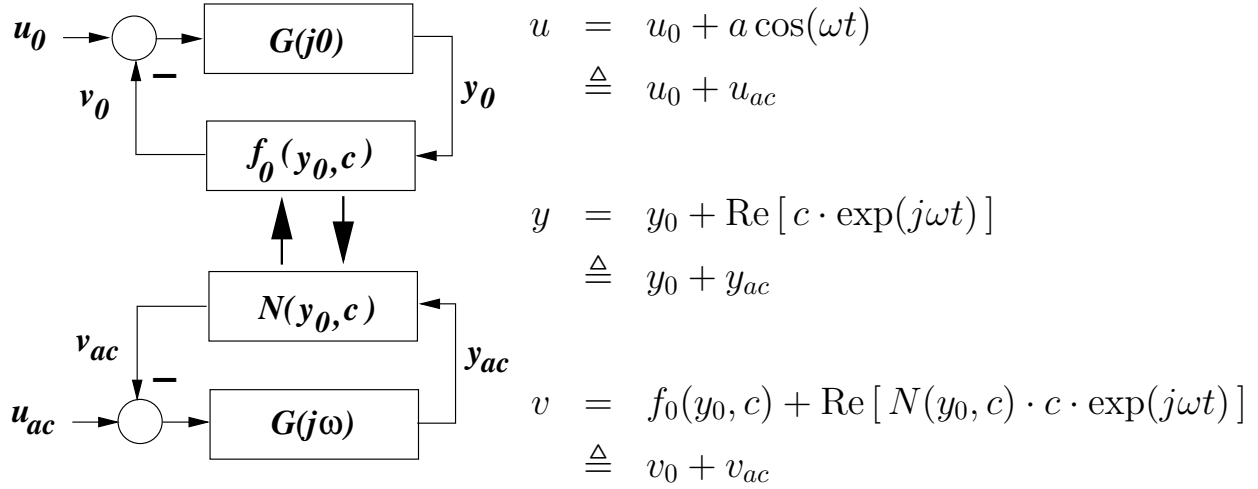
```
D = m1*d1 + m2*(d2 - d1); m_sect = D/d2;
av = 0.01:0.01:4.0;
for i = 1:length(av);
    DFqual(i) = m1*f_sat(d1/av(i)) + m2*(f_sat(d2/av(i))-f_sat(d1/av(i)));
    DFlim(i) = m_sect*f_sat(d2/av(i)); % limiter
    DFrel(i) = 4*D / (pi*av(i)); % relay
end
plot(av,DFqual,av,DFlim,'--',av,DFrel,'-.' );
axis([0 4 0 1.6]);
```

where:

```
function f_sat = f_gv dv(x)
% saturation function "f" for calculating SIDFs for PWL functions
% Gelb & Vander Velde, Appendix B, p. 519
% JH Taylor - 18 June 2002
if abs(x) >= 1,
    fdf = sign(x);
else
    fdf = 2*(asin(x) + x*sqrt(1 - x*x))/pi;
end
```

# Harmonic Balance – Limit Cycle Conditions

## 1. Classical Case:



**DC Harmonic Balance:**  $y_0 = G(j0)[u_0 - f_0(y_0, c)]$  ... must be solved simultaneously with an AC harmonic balance relation (below) to obtain  $y_0$  and  $c$

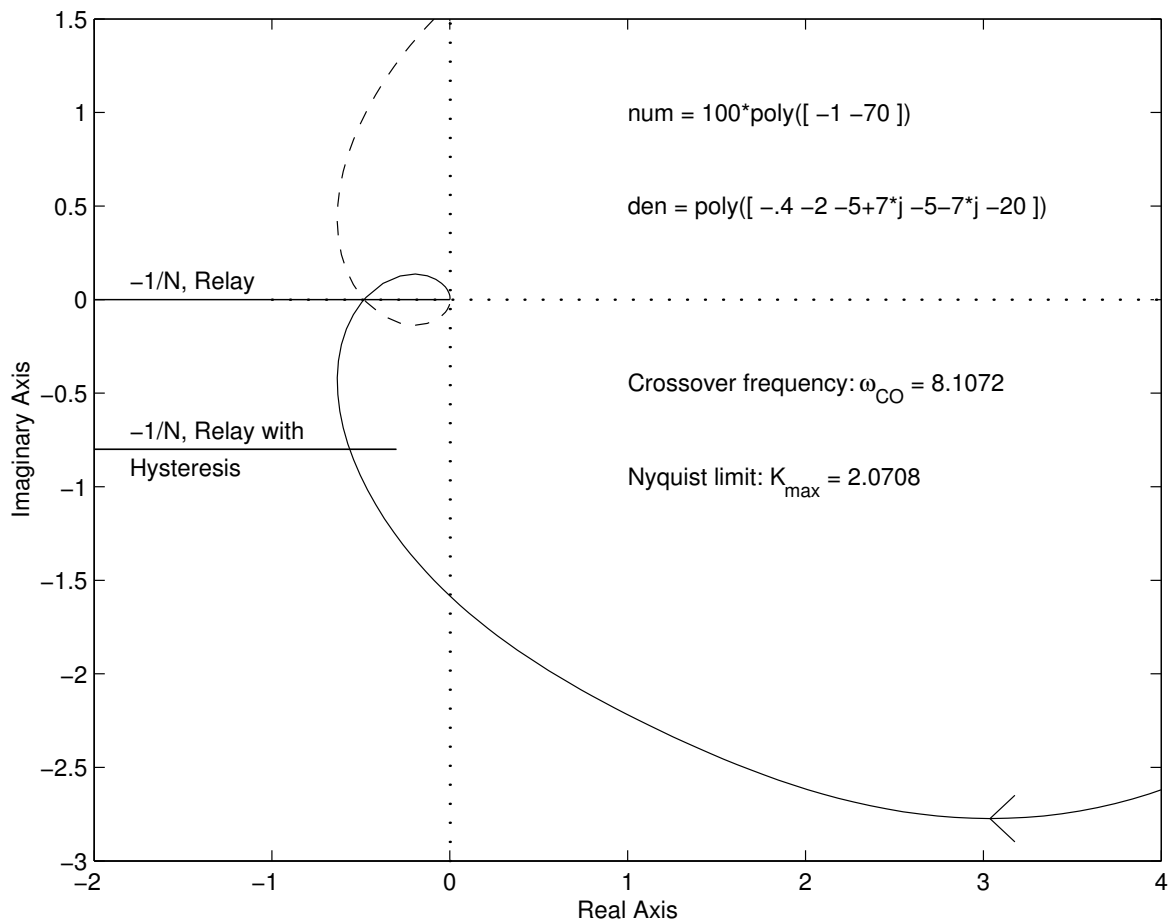
## AC Harmonic Balance:

- Limit Cycles:  $a = 0$ ;  $G(j\omega) \cdot N(y_0, c) = -1$  must be satisfied for some  $\{y_0, c, \omega\}$  for limit cycle prediction
- Forced Response:  $a \neq 0$ ;  $c = \frac{G(j\omega)}{1+N(y_0, c) \cdot G(j\omega)} \cdot a$ ; solve for  $c(j\omega; u_0, a)$  then  $N(y_0, c)$  to obtain the overall “transfer function”  $W(j\omega; u_0, a) = \frac{G(j\omega)}{1+N(y_0, c) \cdot G(j\omega)}$  (the frequency response from  $u_{ac}$  to  $y_{ac}$ )

# Classical Limit Cycle Analysis Using a Nyquist-plot Approach

Given  $G(s) = \frac{s^2 + 71s + 70}{10s^5 + 32.4s^4 + 3468s^3 + 21616s^2 + 37712s + 11840}$ :

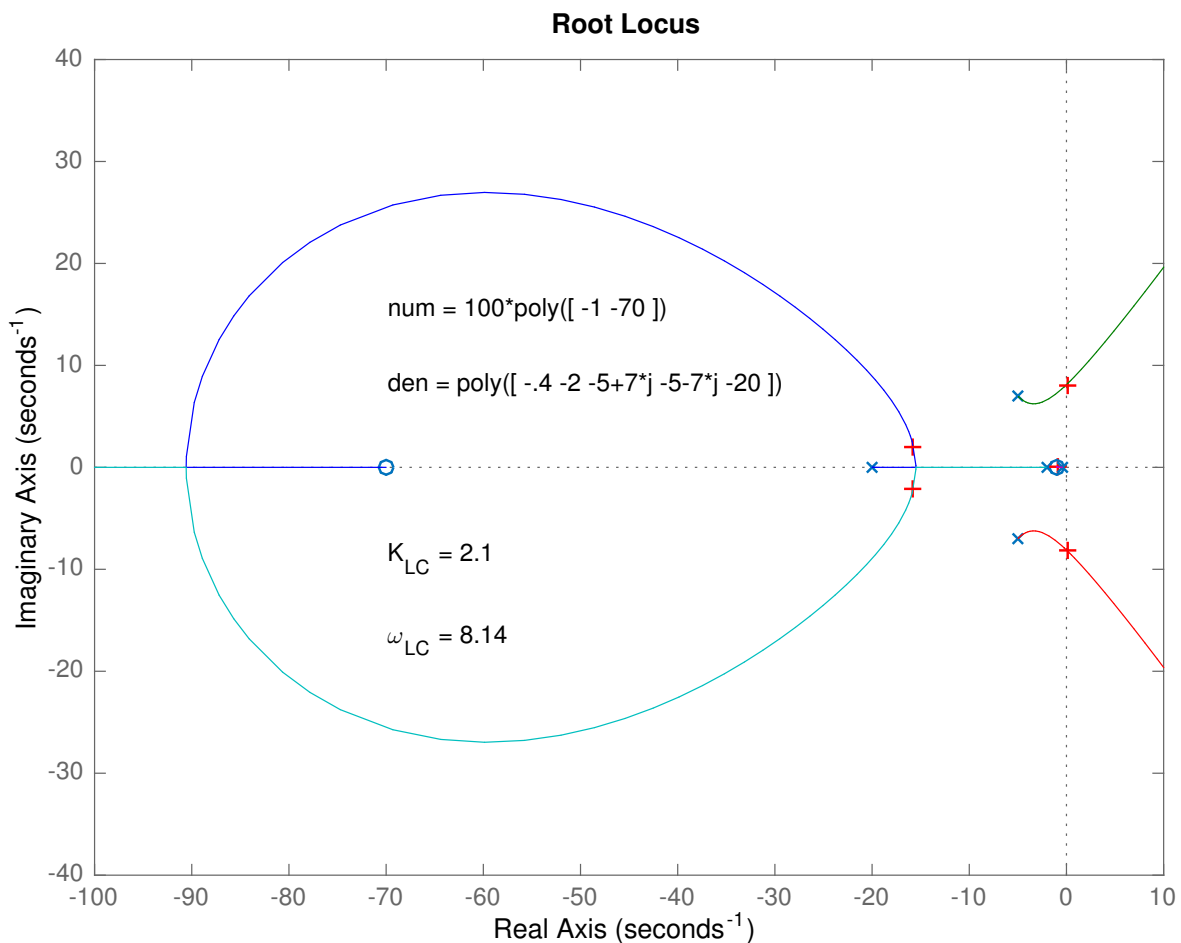
The condition  $G(j\omega) \cdot N = -1$  (or  $G(j\omega) = -1/N$ ) is easily investigated on a Nyquist plot:



Note: This is **not** the Nyquist test for stability!

# Classical Limit Cycle Analysis Using a Root Locus-plot Approach

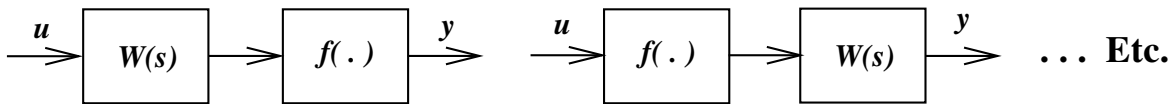
The condition that the root locus of  $G(j\omega)$  has poles on the imaginary axis is easily investigated on a root locus plot:



```
clear; close all
den = poly([ -.4 -2 -5+7*j -5-7*j -20 ]);
num = 100*poly([-1 -70]);
rlocus(num,den)
[K,poles] = rlocfind(num,den);
```

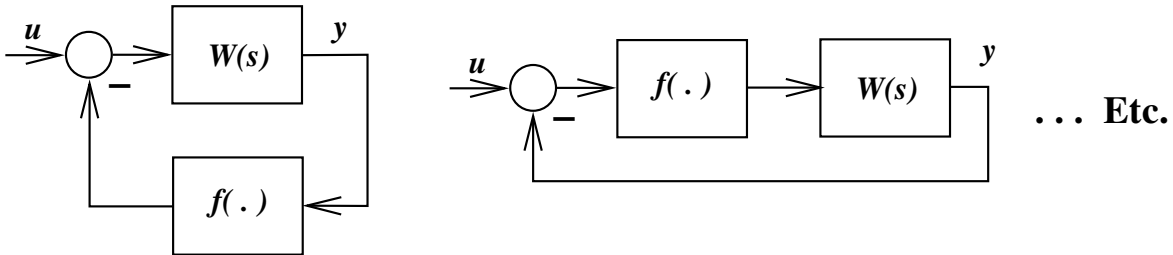
# Limitations of SIDF Analysis

- Situations when SIDFs are exact:



**The first harmonic of  $y$  is correct**

- Situations when SIDFs are **not** exact:



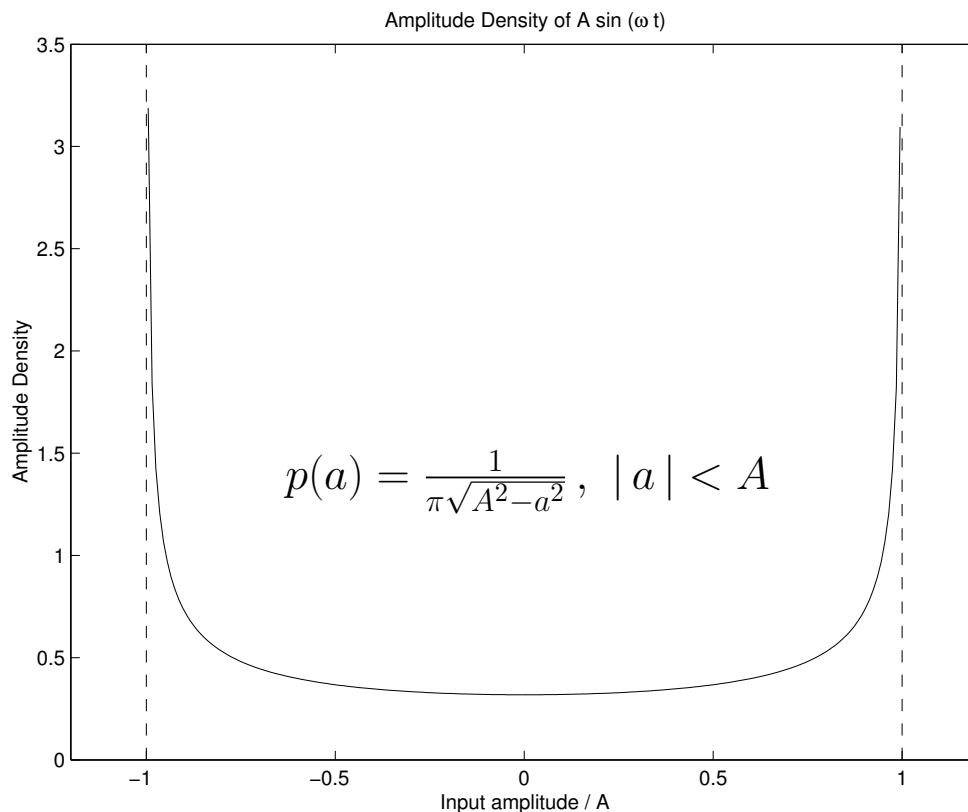
**The first harmonic of  $y$  is approximate**

- How to deal with inexact situations:
  - Consider the validity of the “low-pass filter hypothesis”: The nonlinearity input is essentially sinusoidal due to the filtering of higher harmonics by  $W(j\omega)$  – if true, SIDF results should be accurate.
  - Consider how well-behaved the system nonlinearity is
  - Look at simulation results, assess the importance of higher harmonics (distortion)

## Limitations of SIDF Analysis (Cont'd)

Except for multi-valued nonlinearities (hysteresis, backlash etc.) the DF is not dependent on the assumption of periodicity – only the **amplitude distribution** matters

- For a triangular (“saw-tooth”) wave the DF is the same as that for a uniformly distributed random variable
- In many control applications the sine-wave distribution is a good approximation:



## Limitations of SIDF Analysis (Cont'd)

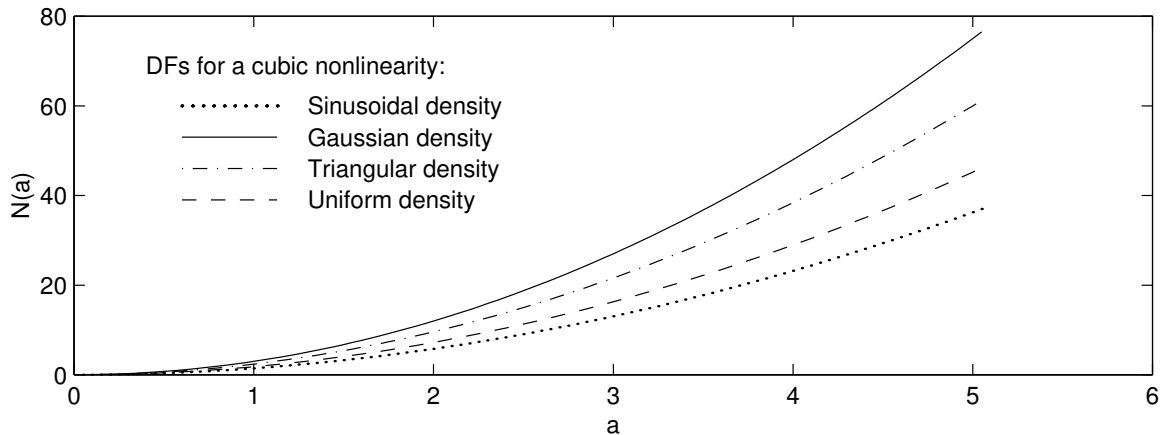
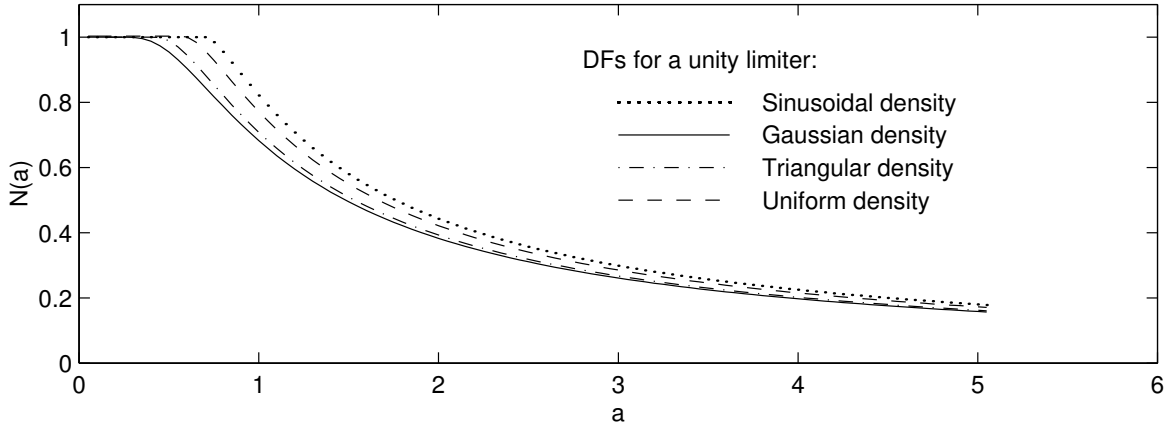
- The basic DF definition for  $f(v)$  where  $v$  has a bias  $b$  is:

$$F_0(\sigma, b) = \int_{-\infty}^{\infty} f(b+z) p(z) dz \quad (9)$$

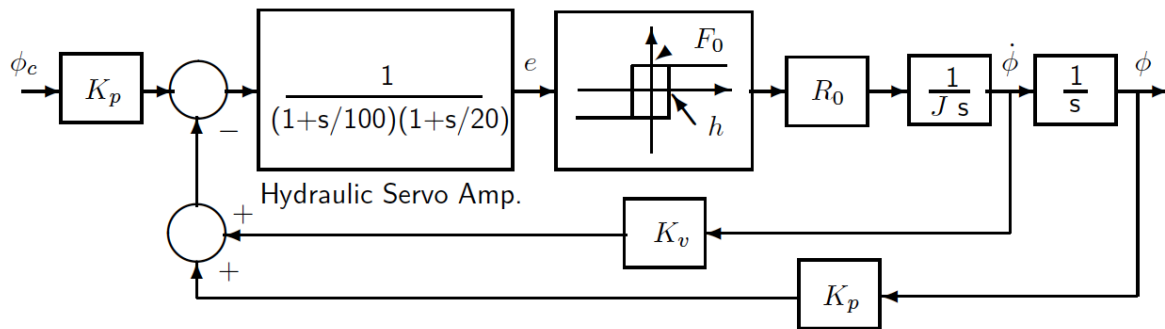
$$N_z(\sigma, b) = \frac{1}{\sigma^2} \int_{-\infty}^{\infty} z f(b+z) p(z) dz \quad (10)$$

where  $\sigma$  is the generalized input signal amplitude,  $\sigma = \sqrt{E[(v-b)^2]}$  which for a sinusoidal signal is  $\sigma_s = a/\sqrt{2}$

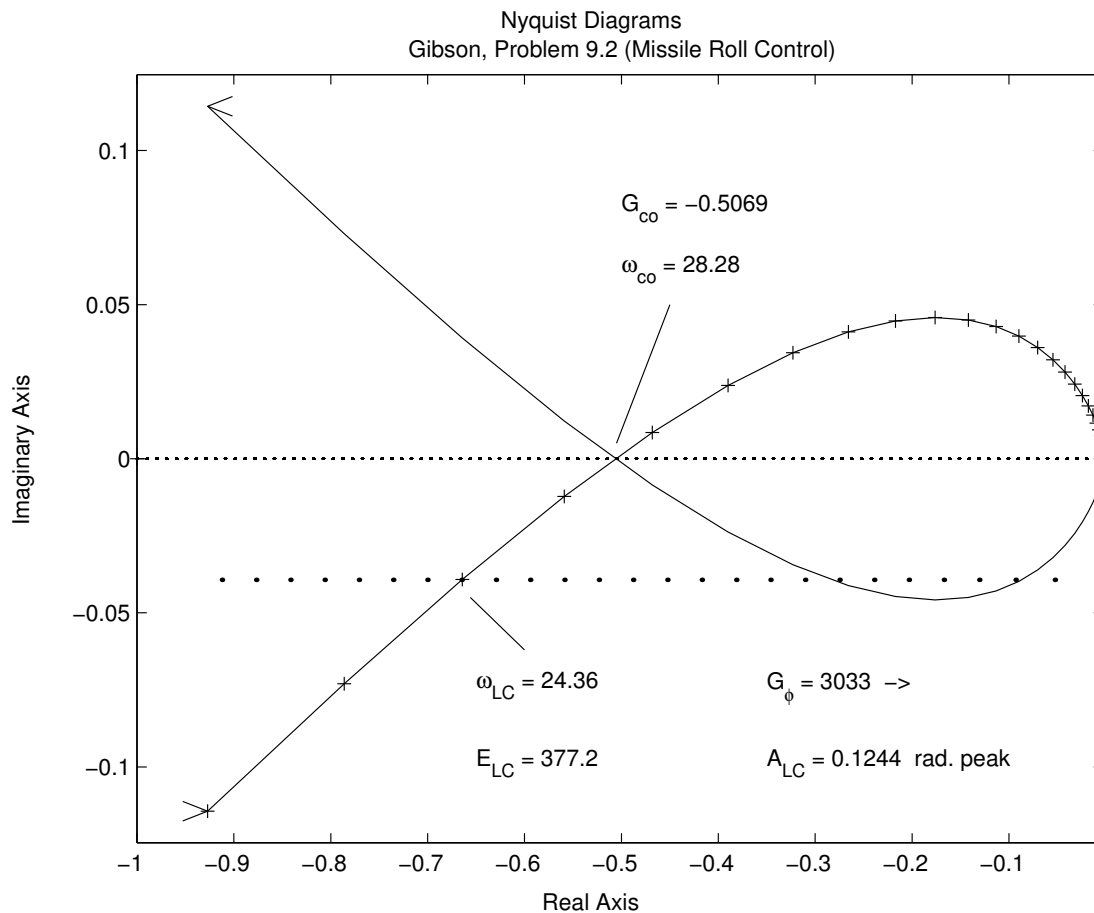
- For many nonlinearities the DF is not particularly sensitive to the amplitude distribution:



# Example: Limit Cycle Analysis, Missile Roll-Control Loop



The relay-with-hysteresis implements “bang-bang” control, a simple robust method that is easy to implement

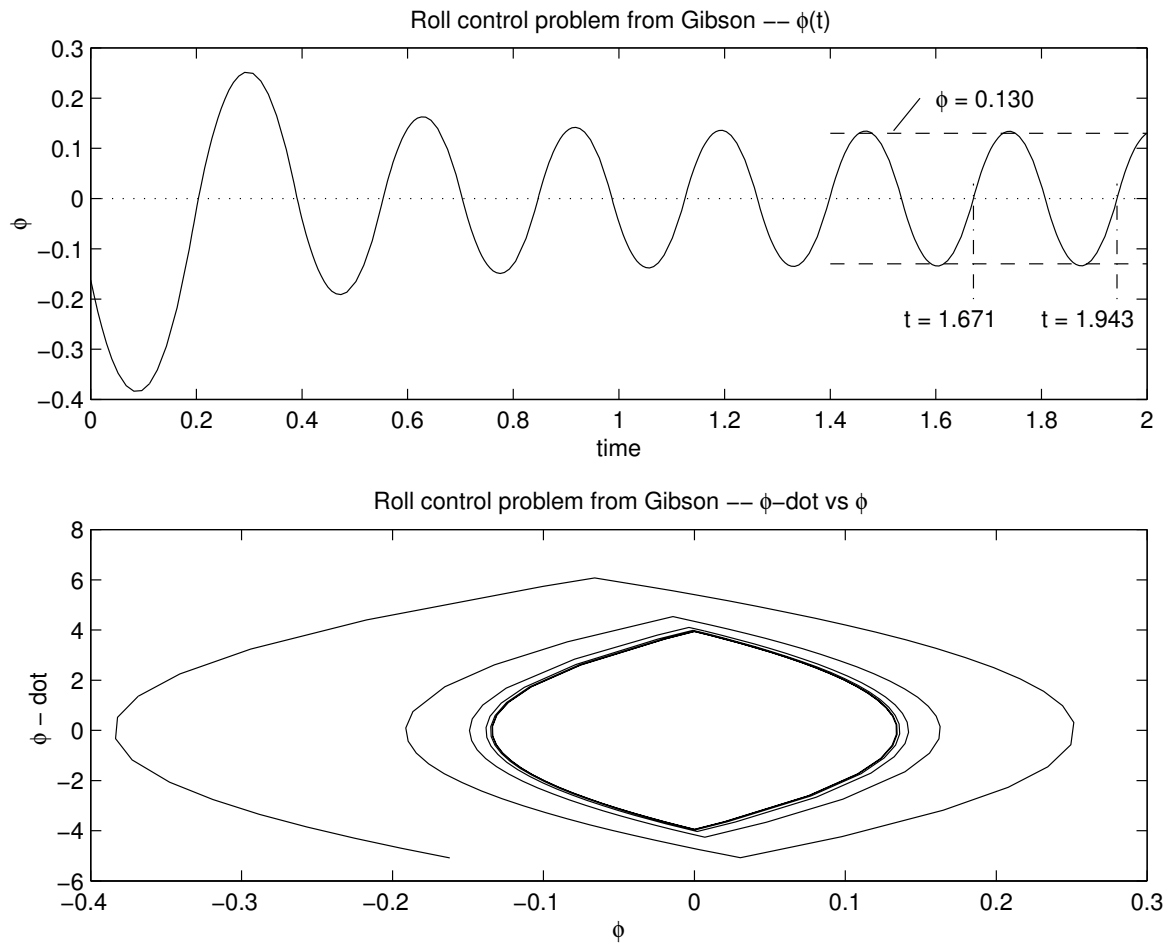


Note:  $G_\phi$  = gain from  $e$  to  $\phi$



# Limit Cycle Verification

Simulation provides a good verification:



$$T = 0.272 \text{ sec} \rightarrow \omega_{LC} = 23.1 \text{ rad/sec}$$

Compare with the SIDf result on the previous slide:

$$T = 0.258 \text{ sec} \rightarrow \omega_{LC} = 24.36 \text{ rad/sec}$$

The SIDF approach is usually this accurate – excellent!

# Harmonic Balance “Transfer Functions”

Two methods for generating the SIDF I/O model  $G(j\omega; u_0, a)$ :

1. Analytic approach: solve the AC Harmonic Balance equation for  $c(j\omega; u_0, a)$ , divide by  $a$ 
  - (a) Advantage: you can tell, for example, when solutions do not exist
  - (b) Disadvantage: it's difficult to carry out if the nonlinear system is at all complicated
2. Simulation approach: develop a simulation model for the nonlinear dynamic system with a sinusoidal input, simulate to obtain the steady-state response, perform Fourier analysis of the result
  - (a) Advantages: No need to assume that the input to each nonlinearity is sinusoidal, the number of system states and nonlinearities is relatively unimportant
  - (b) Disadvantages: May be quite time consuming, may be difficult to interpret the results

# Harmonic Balance “Transfer Function” – Classical Duffing’s Equation

**Duffing’s Equation:**  $\ddot{x} + 2\zeta\dot{x} + x + x^3 = a \cos(\omega t)$

This represents, for example, a normalized mass-spring-damper system with a hardening spring; in the standard form  $W(s) = 1/(s^2 + 2\zeta s + 1)$ ,  $u(t) = a \cos(\omega t)$  and  $f(\cdot) = x^3$

Let  $b$  be the amplitude of the fundamental component of  $x$ ; then quasilinearize Duffing’s equation to obtain:

$$b^2 \left[ \left(1 + \frac{3}{4}b^2 - \omega^2\right)^2 + (2\zeta\omega)^2 \right] = a^2$$

or, if we let  $B = b^2$ ,

$$B \left[ \left(1 + \frac{3}{4}B - \omega^2\right)^2 + (2\zeta\omega)^2 \right] = a^2$$

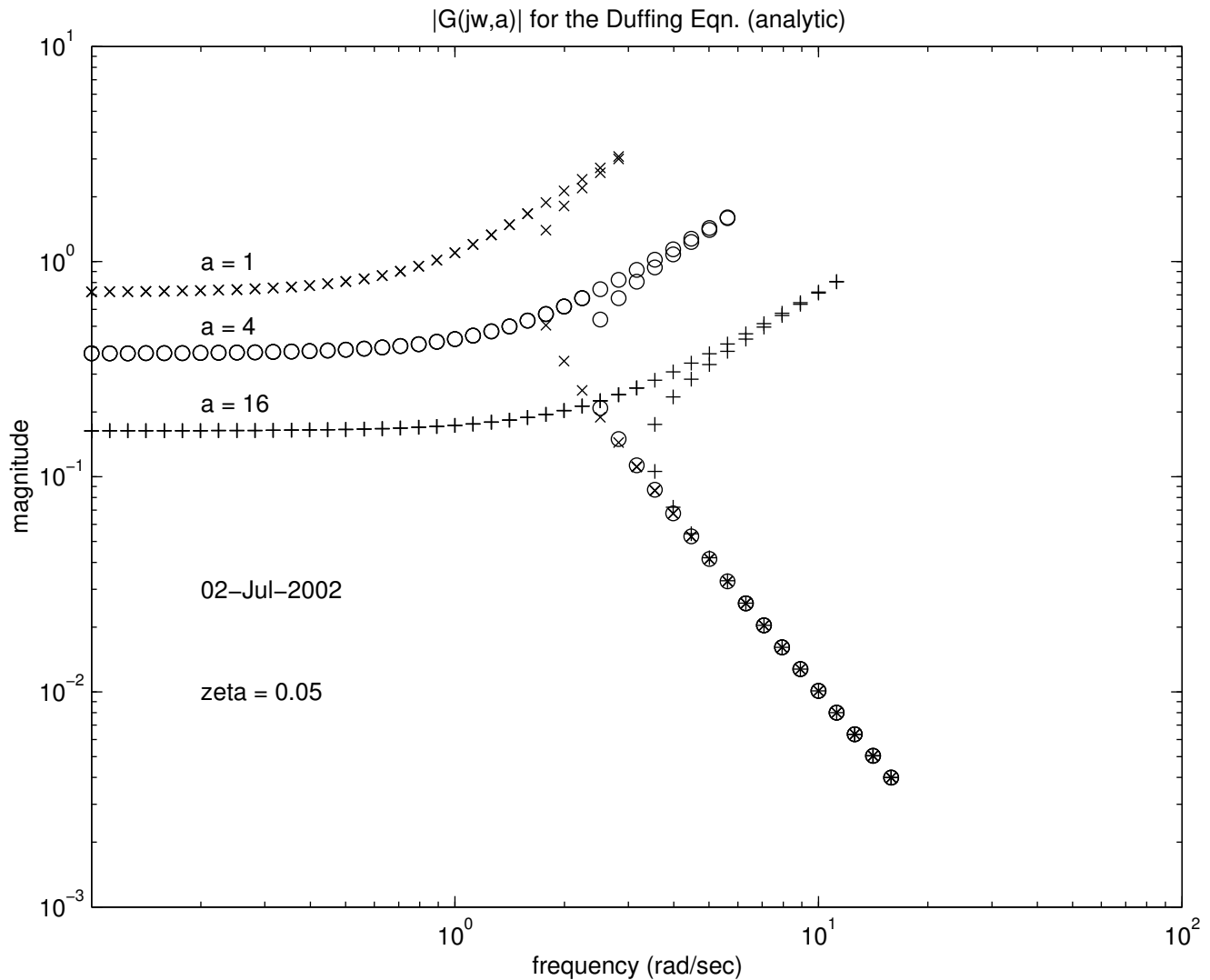
finally,

$$\frac{9}{16}B^3 + \frac{3}{2}(1 - \omega^2)B^2 + [(1 - \omega^2)^2 + (2\zeta\omega)^2] B - a^2 = 0 \quad (11)$$

The last simple polynomial equation may have 1 or 3 real roots, depending on  $a$  and  $\omega$ :

# Duffing's Equation “Transfer Function”

The results for several values of  $a$  are as follows:



Here we see a **jump resonance phenomenon**, which can be a real surprise to an experimentalist!

# Solving the Duffing Problem in MATLAB

- First, define the polynomial in Eqn. 11 multiplied by  $16/3$ :

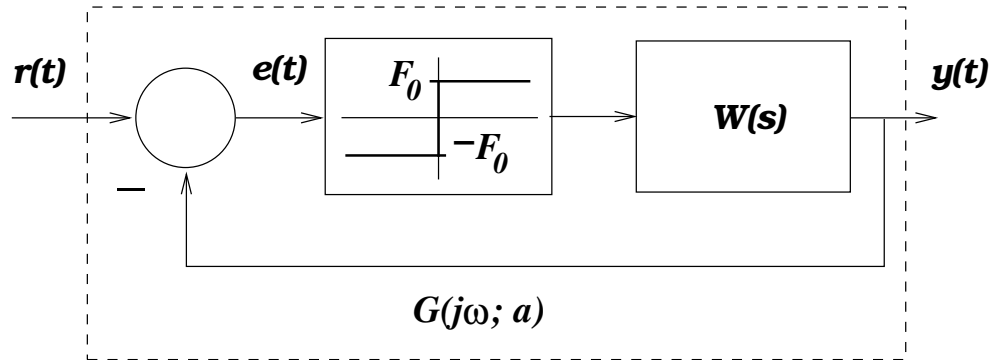
```
function soln = duff_poly(a,w,zeta)
% polynomial to be solved for Duffing's Equation
beta = 1 - w*w; gamma = 2*zeta*w; K = 16/3;
C(1) = 3; C(2) = 8*beta; C(3) = K*(beta^2 + gamma^2);
C(4) = - K*a*a;
soln = sqrt(roots(C)./(a*a));
```

- Now, set up loops for 3 amplitudes and 45 frequencies:

```
zeta = 0.050;
for jj=1:3 %% amplitude loop
    a = 4^(jj-1) %% a = 1, 4, 16
    av(jj) = a;
    for ii=1:45 %% frequency loop
        w = 10^((ii-21)/20) %% w_min = 0.1, w_max = 10
        wv(ii) = w;
        G = duff_poly(a,w,zeta);
        % discard any complex conjugate
        if imag(G(1)) ~= 0 | imag(G(2)) ~= 0,
            for iii=1:3
                if imag(G(iii)) == 0, RG = G(iii); end
            end
            G(1) = RG; G(2) = RG; G(3) = RG; % only one real root exists
        end
        GM(ii,3*jj-2) = G(1); GM(ii,3*jj-1) = G(2); GM(ii,3*jj) = G(3);
    end % frequency loop
end % amplitude loop
%% plotting
loglog(wv,GM(:,1),'x',wv,GM(:,2),'x',wv,GM(:,3),'x', ...
        wv,GM(:,4),'o',wv,GM(:,5),'o',wv,GM(:,6),'o', ...
        wv,GM(:,7),'+',wv,GM(:,8),'+',wv,GM(:,9),'+');
title('|G(jw,a)| for the Duffing Eqn. (analytic)')
xlabel('frequency (rad/sec)');
ylabel('magnitude');
```

# Harmonic Balance “Transfer Functions” (Cont’d)

Closed-loop system with relay:



$$r(t) = a \cos(\omega t)$$

$$y(t) = \text{Re} [ c \exp(j\omega t) ]$$

**Harmonic Balance Relation:**

$$c = (a - c) \cdot \frac{4F_0}{\pi |a - c|} W(j\omega)$$

- Magnitude part:

$$\begin{aligned} M(j\omega) &\triangleq |W(j\omega)|; \\ |G(j\omega; a)| &\triangleq \frac{|c|}{a} = \frac{4F_0}{\pi a} M(j\omega) \end{aligned}$$

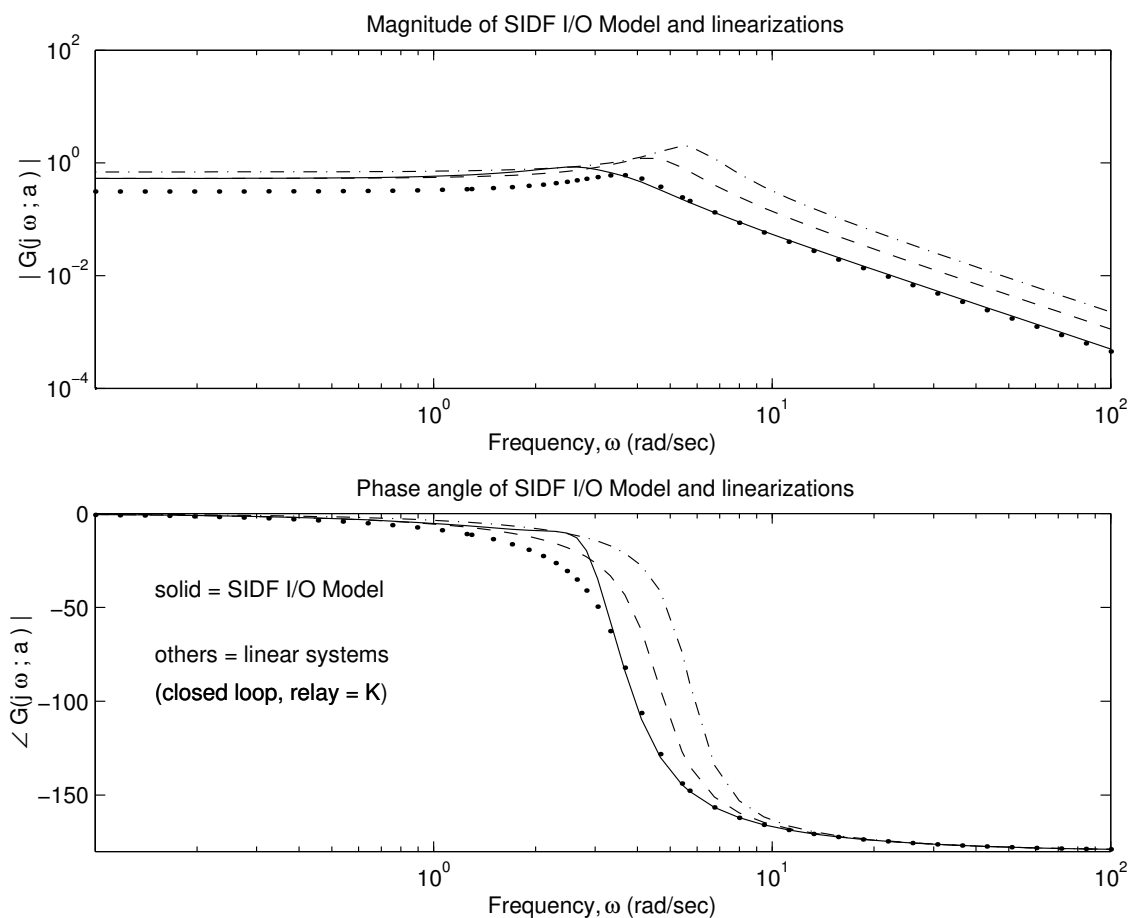
- Phase part:

$$\begin{aligned} \psi &\triangleq \angle W(j\omega); \\ \angle G(j\omega; a) &= \psi - \sin^{-1} \left( \frac{4F_0}{\pi a} M(j\omega) \sin(\psi) \right) \end{aligned}$$

# Harmonic Balance “Transfer Functions” (Cont’d)

Closed-loop system with relay (cont’d)

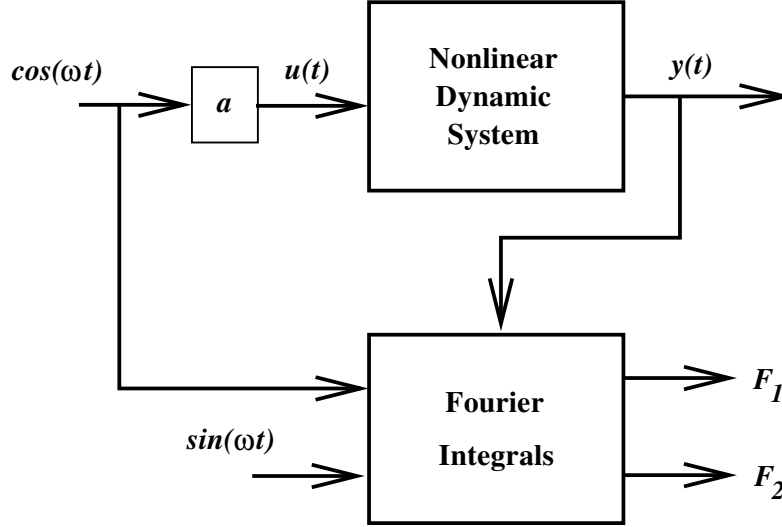
- The magnitude relation is quite straightforward (but it appears that the feedback disappears!)
- The phase relation can only be met if the input amplitude  $a$  is large enough that the argument of  $\sin^{-1}$  is less than one at all frequencies
- Example:  $W(s) = 45/(s^2 + 2s + 9)$ ,  $F_0 = \pi/2$ ,  $a = 18 \rightarrow$



Note that the SIDA transfer functions has characteristics that cannot be duplicated by any linear  $G(j\omega)$

# SIDF I/O Models by Simulation

The most efficient approach is to simulate and perform Fourier analysis simultaneously:



$$F_1^k = \int_{(k-1)T}^{kT} y(t) \cdot \cos(\omega t) dt$$

$$F_2^k = \int_{(k-1)T}^{kT} y(t) \cdot \sin(\omega t) dt$$

from which we obtain:

$$\begin{aligned} \operatorname{Re} G(j\omega; u_0, a) &= \frac{\omega}{\pi a} F_1^k \\ \operatorname{Im} G(j\omega; u_0, a) &= -\frac{\omega}{\pi a} F_2^k \end{aligned}$$

Integrate for  $k$  cycles where  $k$  is sufficiently large that the magnitude and phase of  $G(j\omega; u_0, a)$  have converged to your satisfaction.



# SIDF I/O Model by Simulation in MATLAB

1. Add the Fourier integral states to your model:

```
function xdot = lim_filt2(t,x)
% Second-order linear model with limiter; model is
% augmented with Fourier integrals, to obtain G(jw,a)
% JH Taylor, 10 July 2002
%
zeta = 0.15; global Ampl Freq
u = Ampl*sin(Freq*t);
xdot(1) = x(2);
xdot(2) = u - x(1) - 2*zeta*x(2);
%% define Y and set up the Fourier integrals:
if abs(x(1)) < 1
    y = x(1);
else
    y = sign(x(1));
end
xdot(3) = y*sin(Freq*t);
xdot(4) = y*cos(Freq*t);
xdot = xdot(:); %% end of model lim_filt2
```

2. Run a simulation to steady state and extract  $G(j\omega)$ :

```
function [mag,phase] = ggen(Model,MAGTOL,PHASETOL)
%% ggen(model,MAGTOL,PHASETOL) returns the magnitude
%% and phase of an ODE model defined in the file 'Model'.m
%% JH Taylor - University of New Brunswick - 7 July 2002

% Initialize:
global Ampl Freq Xdim;
k = 0; T = 2*pi/Freq; tspan = [ 0 T ]; x0 = zeros(Xdim,1);
[t,x] = ode45(Model,tspan,x0);
[nrows,ncols] = size(x);
xf = x(nrows,:);
mag0 = Freq/(pi*Ampl)*abs(xf(ncols-1)+j*xf(ncols));
```

```

phase0 = atan2(xf(ncols),xf(ncols-1));
% Simulate cycle-by-cycle until convergence obtained:
while (k >= 0)
    k = k+1;
    x0 = xf; % initial condition from last cycle
    x0(ncols-1) = 0; % reset the Fourier states
    x0(ncols) = 0;
    [t,x] = ode45(Model,tspan,x0);
    [nrows,ncols] = size(x);
    xf = x(nrows,:);
    mag = Freq/(pi*Ampl)*abs(xf(ncols-1)+j*xf(ncols));
    phase = atan2(xf(ncols),xf(ncols-1));
    magdiff = abs(20*log10(mag/mag0));
    phasediff = (180/pi)*abs(phase-phase0);
    if ((magdiff >= MAGTOL) | (phasediff >= PHASETOL))
        mag0 = mag;
        phase0 = phase;
    else
        k = -1;
    end
end;

```

3. Here is the main executive:

```

%% script for generating a set of G(jw,a) for model "mdl"
%% JH Taylor 5 July 2002

global Ampl Freq Xdim; dpr = 180/pi; % degrees/radian
mtol = 1; % magnitude tolerance (dB)
ptol = 5; % phase tolerance (deg)
mdl = 'lim_filt2' % model = lim_filt2.m (2nd order filter + limiter)
Xdim = 4; % # states, **including Fourier integrals**
%
% amplitude loop
for jj=1:3
    Ampl = 4^(jj-2) %% Ampl = .25, 1, 4
    av(jj) = Ampl;

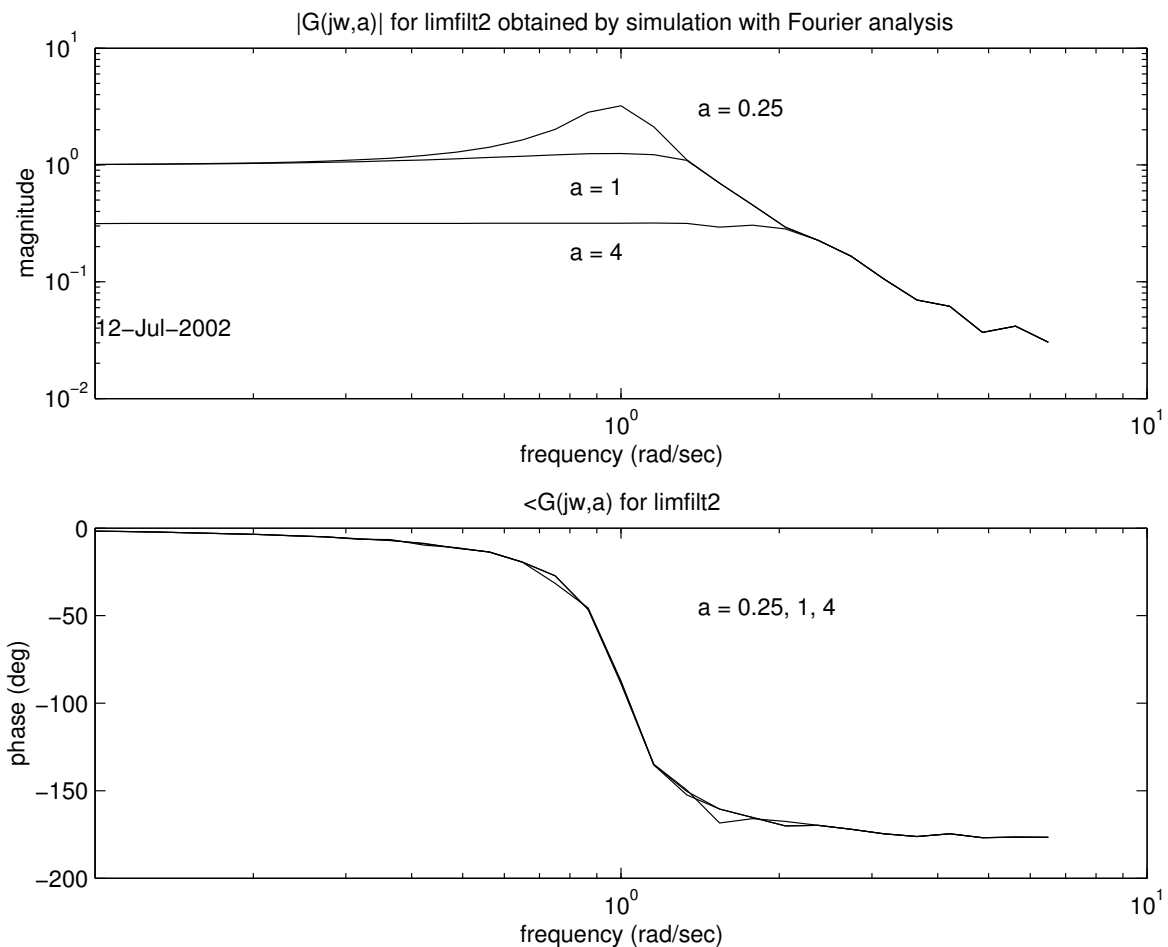
```

```

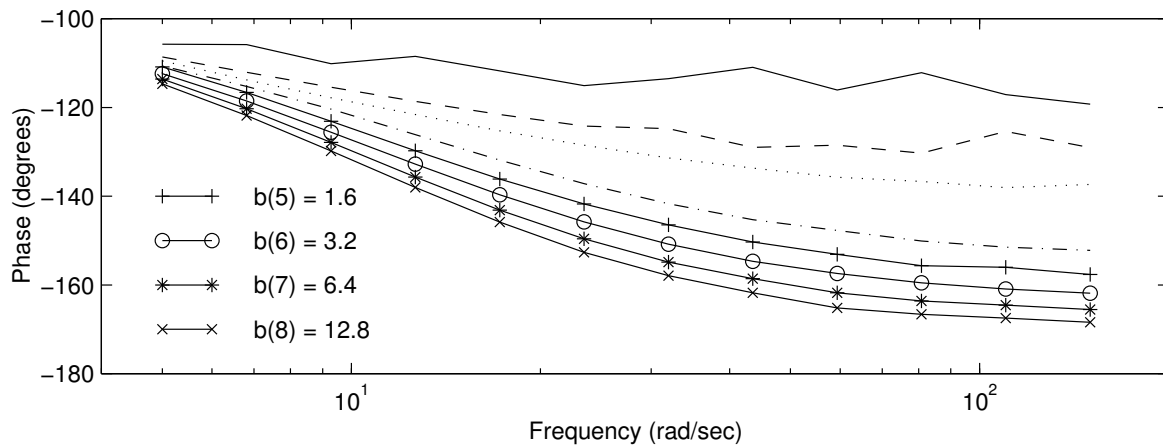
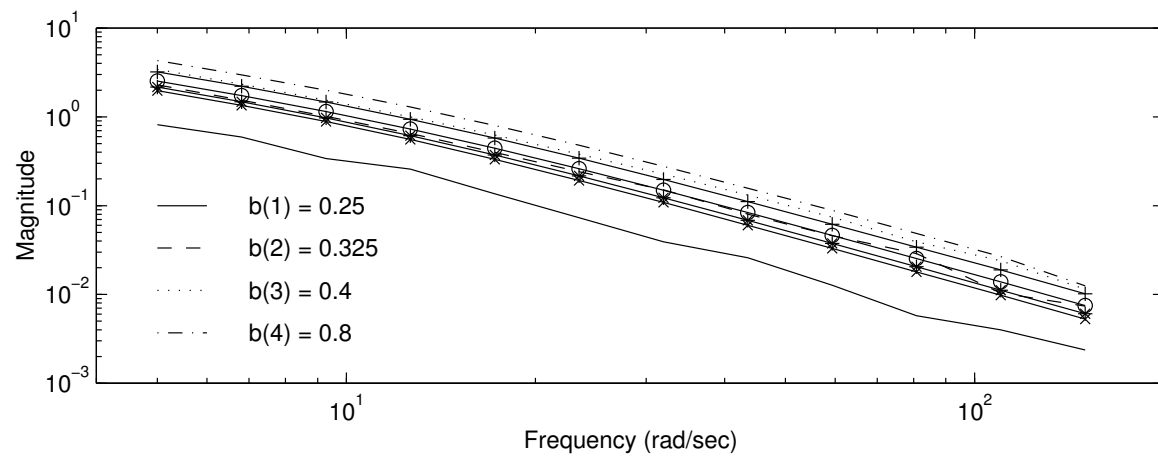
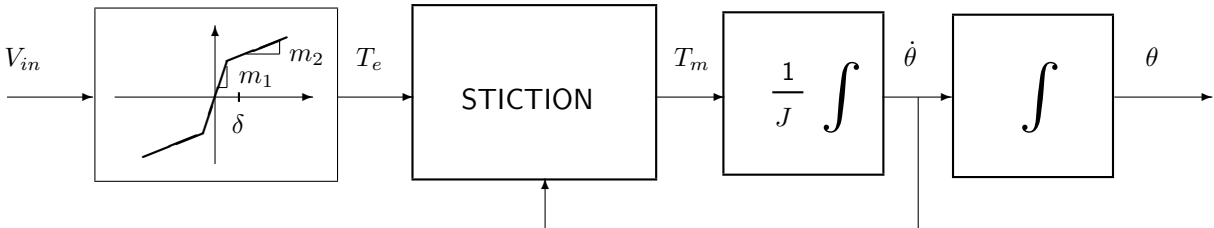
% frequency loop
for ii=1:30
    Freq = 10^((ii-17)/16)  %% w_min = 0.1, w_max ~= 6.5
    wv(ii) = Freq;
    [mag(ii,jj),phase(ii,jj)] = ggen mdl,mtol,ptol);
end % frequency loop
end % amplitude loop
phase = phase .*dpr;  %% change radians to degrees
%% routine plotting commands for "Bode plots" omitted

```

4. Finally, here is the main result:



# Example: SIDF I/O Model, Electromechanical System, by Simulation



## Power of Classical SIDF Approach

When will SIDF limit cycle predictions be “good”?

- When  $-1/N(a)$  definitely cuts  $G(j\omega)$  (not a near miss or near hit)
- When only one limit cycle is predicted (no “nesting”)
- When  $G(3j\omega_{LC})$  is far from  $-1/N(a)$  where  $\omega_{LC}$  is the predicted limit cycle frequency

When will SIDF I/O models be “good”?

- When the nonlinear system is not highly resonant
- When higher harmonics are not dominant predicted limit cycle frequency

## Modern SIDF Analysis

- Given:  $\dot{x} = f(x, u)$  with  $u(t) = u_0 + \text{Re} [a \exp(j\omega t)]$
- Assume:  $x(t) \approx x_c + \text{Re} [b \exp(j\omega t)]$
- Quasilinearize the entire state-space system:

$$\begin{aligned}
 f(x, u) = & f_B(u_0, a, x_c, b) \\
 & + \text{Re} [A_{DF}(u_0, a, x_c, b) \cdot b \exp(j\omega t)] \\
 & + \text{Re} [B_{DF}(u_0, a, x_c, b) \cdot a \exp(j\omega t)] \quad (12)
 \end{aligned}$$

- Therefore DC harmonic balance is given by  $0 = f_B(u_0, a, x_c, b)$
- ...and AC harmonic balance is given by:
  - Nonlinear Oscillations:  $a = 0$ , find  $b \neq 0$  such that  $[j\omega_{LC}I - A_{DF}]^{-1}b = 0$  (“ $A_{DF}$  has pure imaginary eigenvalues and  $b$  is the corresponding eigenvector”), i.e., limit cycles are predicted if solutions  $b, \omega_{LC}$  exist
  - Forced Response:  $b = [j\omega I - A_{DF}]^{-1}B_{DF} \cdot a$

## SIDFs for Multivariable Functions

- Single-input nonlinearities  $f(v)$  are quasilinearized as before
- Multi-variable nonlinearities  $f(v_1, v_2, \dots)$  are more complicated; products and powers of states are easiest to do:

$$\begin{aligned}
 \text{Given: } f(x) &= x_1 x_2^2 \\
 &= (x_{10} + \text{Re}[a_1 \exp(j\omega t)])(x_{20} + \text{Re}[a_2 \exp(j\omega t)])^2 \\
 &= \dots \\
 &\simeq \left[ x_{10} x_{20}^2 + \frac{1}{2} x_{10} |a_2|^2 + x_{20} a_1 \bullet a_2 \right. \\
 &\quad \left. + \left[ x_{20}^2 + \frac{1}{4} |a_2|^2 \right] \cdot x_{1,AC} \right. \\
 &\quad \left. + \left[ 2x_{10} x_{20} + \frac{1}{2} a_1 \bullet a_2 \right] \cdot x_{2,AC} \right] \quad (13)
 \end{aligned}$$

(via trigonometric identities and eliminating higher harmonic terms), where  $\bullet$  denotes dot product,  $a_1 \bullet a_2 = \text{Re } a_1 \cdot \text{Re } a_2 + \text{Im } a_1 \cdot \text{Im } a_2$

Handling multivariable functions represents a **significant generalization** over the classical approach

# Multivariable Limit Cycle Example

The following second-order differential equation has been derived to describe the local behavior of solutions to a two-mode panel flutter model:

$$\ddot{\chi} + (\alpha + \chi^2) \dot{\chi} + (\beta + \chi^2) \chi = 0 \quad (14)$$

Heuristically, it is reasonable to predict that limit cycles may occur for negative  $\alpha$  (so the second term provides damping that is negative for small values of  $\chi$  but positive for large values). Observe also that there are three singularities if  $\beta$  is negative:  $\chi_0 = 0, \pm\sqrt{-\beta}$ . Making the usual choice of state vector,  $x = [\chi \ \dot{\chi}]^T$ , the corresponding state vector differential equation is

$$\dot{x} = \begin{bmatrix} \dot{\chi} \\ \ddot{\chi} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\beta & -\alpha \end{bmatrix} x - \begin{bmatrix} 0 \\ x_1^2(x_1 + x_2) \end{bmatrix} \quad (15)$$

The SIDF assumption for this system of equations is that

$$\begin{aligned} x_1 = \chi &\approx \chi_c + a_1 \cos(\omega t) \\ x_2 = \dot{\chi} &\approx -a_1 \omega \sin(\omega t) \end{aligned}$$

(  $x_2 = \dot{\chi}$  so  $x_2$  has a center value of 0 and  $a_2 = -j \omega a_1$  )

Therefore, the combined nonlinearity in Eqn. 15 may be quasilinearized to obtain

$$\begin{aligned} x_1^2(x_1 + x_2) &= (\chi_c + a_1 \cos(\omega t))^2(\chi_c + a_1 \cos(\omega t) - a_1 \omega \sin(\omega t)) \\ &\approx (\chi_c^3 + \frac{3}{2}\chi_c a_1^2) + (3\chi_c^2 + \frac{3}{4}a_1^2) a_1 \cos(\omega t) \\ &\quad + (\chi_c^2 + \frac{1}{4}a_1^2) (-a_1 \omega \sin(\omega t)) \\ &\triangleq f_B + N_1 \cdot a_1 \cos(\omega t) + N_2 \cdot (-a_1 \omega \sin(\omega t)) \end{aligned} \quad (16)$$

This result is obtained by expanding the first expression using trigonometric identities and discarding all terms except the fundamental ones ( $k = 0, 1$ ).

Therefore, we require that

$$F_{DF} = \begin{bmatrix} 0 \\ -\chi_c(\beta + \chi_c^2 + \frac{3}{2}a_1^2) \end{bmatrix} = 0 \quad (17)$$

$$A_{DF} = \begin{bmatrix} 0 & 1 \\ -(\beta + 3\chi_c^2 + \frac{3}{4}a_1^2) & -(\alpha + \chi_c^2 + \frac{1}{4}a_1^2) \end{bmatrix} \triangleq \begin{bmatrix} 0 & 1 \\ -\omega_{LC}^2 & 0 \end{bmatrix} \quad (18)$$

(so  $A_{DF}$  has imaginary eigenvalues  $\pm j \omega_{LC}$ ; again, the canonical form of  $A_{DF}$  ensures harmonic balance, not “pure imaginary eigenvalues”).



# Multivariable Limit Cycle Example (Cont'd)

Relation 17 shows two possibilities:

- **Case 1:**  $\chi_c = 0$ , in which case Eqn. 18 yields

$$a_1 = 2\sqrt{-\alpha}, \quad \omega_{LC} = \sqrt{\beta - 3\alpha} \quad (19)$$

The amplitude  $a_1$  and frequency  $\omega_{LC}$  must be real for limit cycles to exist. Thus, as conjectured,  $\alpha < 0$  is required. for a LC to exist centered about the origin, and  $\beta$  must satisfy  $\beta > 3\alpha$ , so  $\beta$  can take on any positive value but cannot be more negative than  $3\alpha$ .

- **Case 2:**  $\chi_c = \pm\sqrt{(\beta - 6\alpha)/5} \triangleq \pm\chi_{c0}$ , yielding

$$a_1 = 2\sqrt{(\alpha - \beta)/5}, \quad \omega_{LC} = \sqrt{\beta - 3\alpha} \quad (20)$$

For the two limit cycles in Case 2 to exist, centered at  $\pm\chi_{c0}$ , it is necessary that  $3\alpha < \beta < \alpha$ , so again limit cycles cannot exist unless  $\alpha < 0$ . One additional constraint must be imposed:  $|\chi_c| > a_1$  must hold, or the two limit cycles will “overlap”; this condition reduces the permitted range of  $\beta$  to  $2\alpha < \beta < \alpha$ .

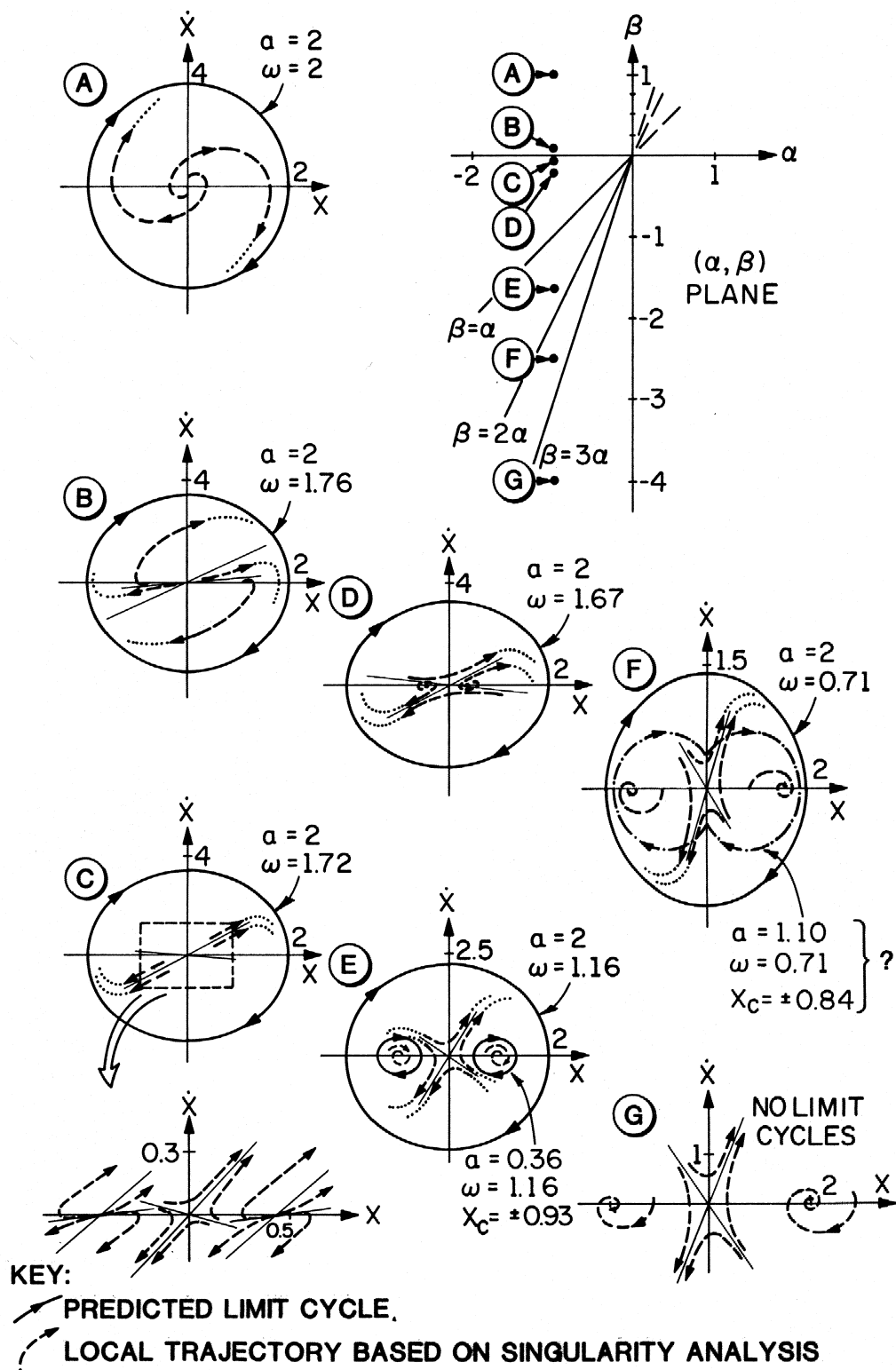
The stability of the Case 1 LC can be determined as follows: Take any  $\epsilon > 0$  and perturb the LC amplitude to a slightly larger value, e.g.,  $a_1^2 = -4\alpha + \epsilon$ . Substituting into Eqn. 18 yields

$$A_{DF} = \begin{bmatrix} 0 & 1 \\ -(\beta - 3\alpha + \frac{3}{4}\epsilon) & -\frac{1}{4}\epsilon \end{bmatrix} \quad (21)$$

which for  $\epsilon > 0$  has “slightly stable eigenvalues”. Thus a trajectory perturbed just outside the LC will decay, indicating that the Case 1 LC is stable. A similar analysis of the Case 2 LC is more complicated (since a perturbation in  $a_1$  produces a shift in  $\chi_c$  that must be considered), and thus is omitted.

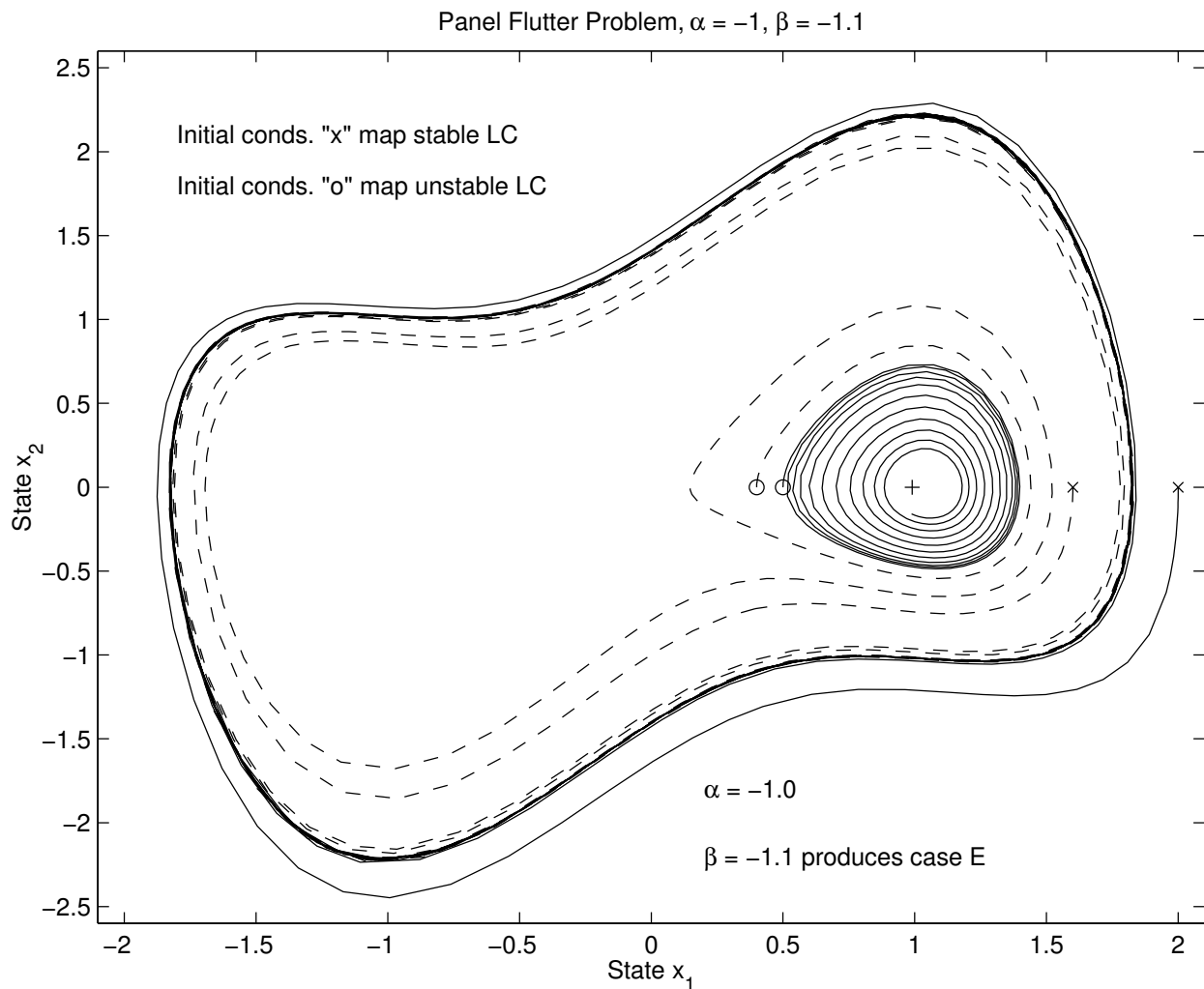
Based on the SIDF-based LC analysis outlined above, the behavior of the original system Eqn. 15 is portrayed for  $\alpha = -1$  and seven values of  $\beta$  below:

# Multivariable Limit Cycle Example (Cont'd)



## Multivariable Limit Cycle Example (Cont'd)

Here is one simulation confirmation of these SIFD results, corresponding to Case E:



The outer limit cycle is clearly stable, while the inner limit cycle is unstable; the equilibrium inside the inner limit cycle is stable. There is another unstable limit cycle centered (approximately) at  $x_{10} \approx -1$ ,  $x_{20} = 0$

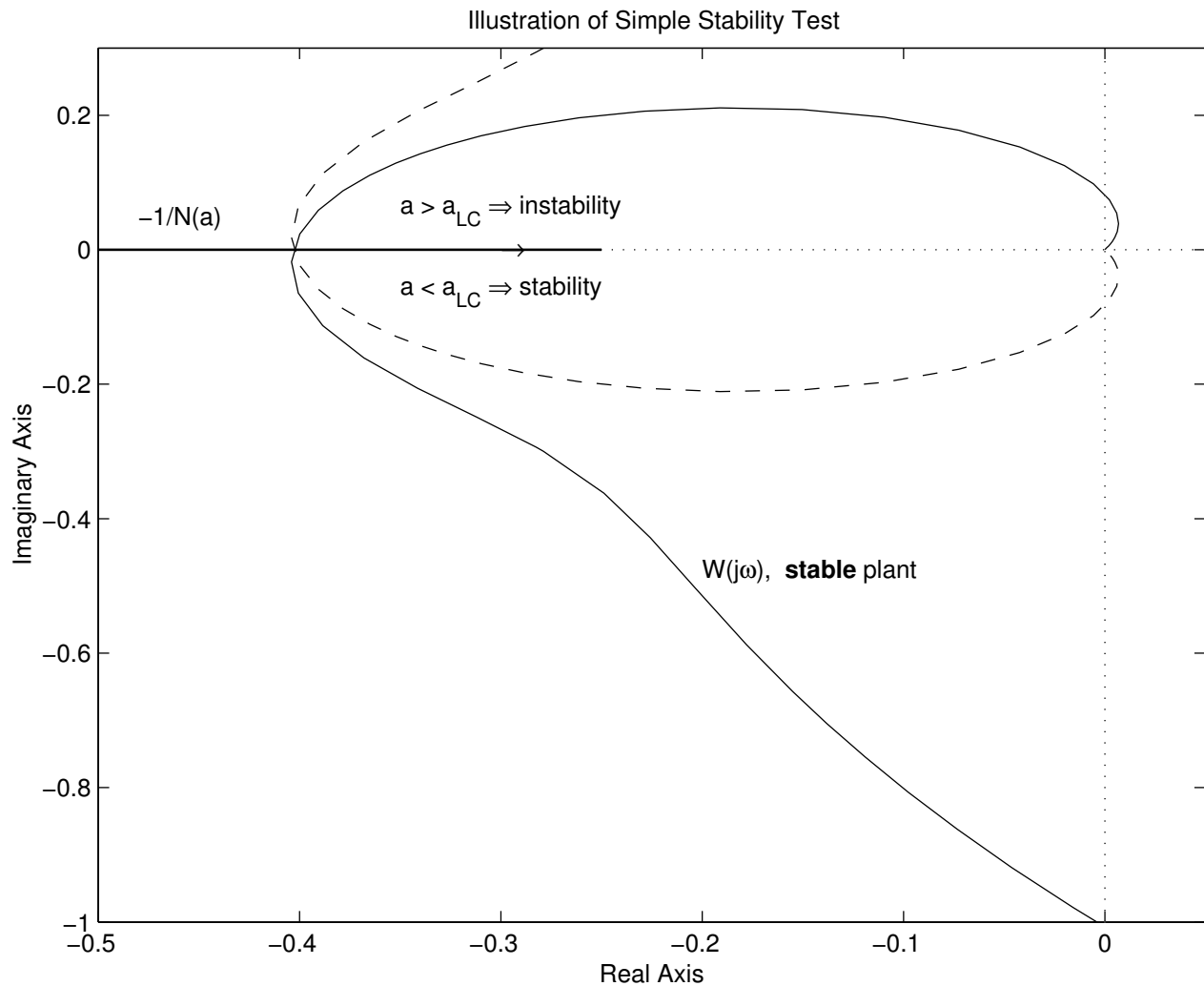
## Limit Cycle Stability

The SIDF approach can also yield information on **limit cycle stability** – assume a limit cycle is predicted with amplitude  $a_{LC}$ , then:

- The limit cycle is **stable** if  $a > a_{LC}$  moves the pure imaginary eigenvalues into the *left* half plane and  $a < a_{LC}$  moves the pure imaginary eigenvalues into the *right* half plane
- The limit cycle is **unstable** if the converse is true
- Otherwise the limit cycle is **structurally unstable** (this is an uncommon “borderline” case)
- These conditions are easy to check in cases where there is no bias (DC level), otherwise the coupling between the center value and amplitude  $(y_c, a)$  must be taken into account

## Limit Cycle Stability (Cont'd)

Here is a limit cycle stability test in the no bias case:



In other words: if  $-1/N(a)$  for  $a > a_{LC}$  moves inside the RHP map of  $W(s)$  the limit cycle is unstable, and conversely.

Another test works if there is no bias and there is only one limit cycle predicted: The limit cycle is stable if the enclosed equilibrium is unstable, and conversely.

## SIDF Methods: Conclusions

- SIDF techniques are very powerful for studying periodic behavior (nonlinear oscillations, forced response), even in high order and highly nonlinear dynamic system models, even where discontinuous and multi-valued functions exist
- One of the key uses of this approach is exploration:
  - Finding areas in parameter space where limit cycles exist and boundaries where bifurcations occur
  - Determining how a nonlinear system's response to sinusoidal inputs changes as model parameters change
- SIDF analysis and simulation are highly complementary; both have important roles to play