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# SMALL-SCALE TRANSONIC INVESTIGATION OF THE EFFECTS OF PARTIAL-SPAN LEADING-EDGE CAMBER ON THE AERODYNAMIC CHARACTERISTICS OF A 50 DEG 38 SWEPTBACK WI



Small-Scale Transonic Investigation of the Effects of Partial-Span Leading-Edge Camber on the Aerodynamic Characteristics of a 50 Deg 38 Sweptback Wing of Aspect Ratio 2.98

NASA Technical Reports Server (NTRS)

Bibliogov. Paperback. Book Condition: New. This item is printed on demand. Paperback. 34 pages. Dimensions: 9.7in. x 7.4in. x 0.1in. A small-scale transonic investigation of two semispan wings of the same plan form was made in the Langley high-speed 7- by 10-foot tunnel through a Mach number range of 0.70 to 1.10 and a mean-test Reynolds number range of 745,000 to 845,000 to determine the effects of partial-span leading-edge camber on the aerodynamic characteristics of a swept-back...

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