



The Hong Kong Polytechnic University

The Department of Land Surveying and Geo-informatics

2021 - 2022 Semester 1

LSGI3322: Satellite Positioning Systems

Subject Lecturer: Prof. George Liu & Prof. Wu Chen

Final Individual Project: GPS Positioning (30%)

Intermediate Report 1: Reading Two GPS RINEX Files (7%)

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Expected Grade: A Range



1 Objective

To complete the final project which counts for 30% scores of LSGI3322, it is then divided into three sections:

- 7 % - Reading Two GPS RINEX data files (6/10)
- 8 % - Computing GPS Satellite Positioning (3/11)
- 15% - Final Production: Single Point Positioning using Pseudo-range Measurement (12/12)

In the **First Intermediate Report**, the mission is to read two GPS RINEX data files, in terms of navigation and observation files (the given data: site0900.01n and site0900.01o) by using self-developed programs is conducted. I have chosen MATLAB program software to complete this individual project.



2 RINEX Data Structure

2.1 RINEX Navigation File

The below displays the navigation file "site0900.01n" in RINEX format. It is used the first section as an example

site0900(1).01n											
NAVIGATION DATA											
1	2	RINEX VERSION / TYPE									
2	CCRINEXN V1.5.9 UX CDDIS	PGM / RUN BY / DATE									
3	IGS BROADCAST EPHEMERIS FILE	COMMENT									
4	0.4191D-07 0.1490D-07 -0.2384D-06 -0.5961D-07	ION ALPHA									
5	0.1495D+06 0.0000D+00 -0.3932D+06 0.3932D+06	ION BETA									
6	-0.239808173319D-13-0.139698386192D-07	61440	1108	DELTA-UTC: A0,A1,T,W							
7	13			LEAP SECONDS							
8				END OF HEADER							
9	2 01 3 31 0 0 0.0-0.360158272088D-03-0.534328137292D-11	0.000000000000D+00									
10	0.520000000000D+02 0.781250000000D+01 0.497556439509D-08	0.187238074216D+01									
11	0.162050127983D-06 0.207882055547D-01 0.959262251854D-05	0.515368904686D+04									
12	0.518400000000D+06-0.221654772758D-06 0.551731029172D+00-0.441446900368D-06										
13	0.933503344596D+00 0.181281250000D+03-0.205664455792D+01-0.816105422638D-08										
14	-0.263225250103D-09 0.000000000000D+00 0.1107000000000D+04	0.000000000000D+00									
15	0.100000000000D+01 0.000000000000D+00-0.186264514923D-08 0.564000000000D+03										
16	0.511530000000D+06 0.000000000000D+00 0.000000000000D+00	0.000000000000D+00									
17	3 01 3 31 0 0 0.0 0.162767246366D-04 0.318323145621D-11	0.000000000000D+00									
18	0.530000000000D+02 0.863125000000D+02 0.498020744563D-08	0.349439502758D+00									
19	0.443309545517D-05 0.209663005080D-02 0.779703259468D-05	0.515368713570D+04									
20	0.518400000000D+06 0.000000000000D+00 0.163186669490D+01	0.596046447754D-07									
21	0.937298053788D+00 0.218531250000D+03 0.714973987799D+00-0.833034699239D-08										
22	0.957182727646D-10 0.100000000000D+01 0.110700000000D+04	0.000000000000D+00									
23	0.400000000000D+01 0.000000000000D+00-0.465661287308D-08 0.565000000000D+03										
24	0.518148000000D+06 0.000000000000D+00 0.000000000000D+00	0.000000000000D+00									
25	4 01 3 31 0 0 0.0 0.700922217220D-03 0.250111042988D-11	0.000000000000D+00									

Table 1. An Interpretation of GPS RINEX Navigation File.

Explanation of GPS Navigation Data Record									
2	01	3	31	0	0	0	-0.36158272088D-3	-0.534328137292D-11	0.000000000000D+00
Recorder identifier	Year (2-digit)	Month	Day	Hour	Minute	Second (2 digits)	SV clock bias (seconds)	SV clock drift (sec/sec)	SV clock drift rate (sec/sec2)
0.520000000000D+02	0.781250000000D+01	0.497556439509D-08	0.187238074216D+01						
Ephemeris	Crs (meters)	Delta n (radian/sec)	M0 (radians)						
0.162050127983D-06	0.207882055547D-01	0.959262251854D-05	0.515368904686D+04						
Cuc (radians)	e Eccentricity	Cus (radians)	sqrt(A) (sqrt(m))						
0.518400000000D+06	-0.221654772758D-06	0.551731029172D+00	-0.441446900368D-08						
Toe Time of Ephemeris (Sec of BDT week)	Cic (radians)	OMEGA0 (radians)	Cis (radians)						
0.933503344596D+00	0.181281250000D+03	-0.205664455792D+01	-0.816105422638D-08						
i0	Crc (meters)	omega (radians)	OMEGA DOT (radians/sec)						
-0.263225250103D-09	0.000000000000D+00	0.110700000000D+04	0.000000000000D+00						
IDOT (radians/sec)	Codes on L2 channel	GPS week	L2 P Data Flag						
0.100000000000D+01	0.000000000000D+00	-0.1862624514923D-08	0.564000000000D+03						
SV Accuracy (meters)	SV Health	TGD (seconds)	IODC Issue of Date, Clock						
0.511530000000D+06	0.000000000000D+00	Spare	Spare						
Transmission Time of Message	Fit interval (hours)	Spare	Spare						

2.1 RINEX Observation File

The below shows the observation data file "site0900.01o" in RINEX format.

```

site0900(1).01o

1 2.10      OBSERVATION DATA G (GPS)      RINEX VERSION / TYPE
2 teqc 2000Jul20 WCDA Data collecting20010401 00:12:37UTCPGM / RUN BY / DATE
3 Solaris 2.7|Ultra 2|cc SC5.0|=+-|*Sparc COMMENT
4 teqc 2000Jul20 WCDA Data collecting20010401 00:12:31UTCCCOMMENT
5 teqc 2000Jul20 WCDA Data collecting20010331 01:12:17UTCCCOMMENT
6 site      xxxxxxxx MARKER NAME
7 xxxxxxxx   xxxxxxxxxxxxxxxxxxxx MARKER NUMBER
8 xxxxxxxx   OBSERVER / AGENCY
9 2025       AOA BENCHMARK ACT 3.3.32.2N 1k99/07/28REC # / TYPE / VERS
10 95174     AOAD/M_T EMRA ANT # / TYPE
11 00000001.0000 00000001.0000 00000001.0000 APPROX POSITION XYZ
12 0.1000    0.0000 0.0000 ANTENNA: DELTA H/E/N
13 1 1       1 1 WAVELENGTH FACT L1/2
14 7 L1      L2 C1 P1 P2 S1 S2 # / TYPES OF OBSERV
15 30.0000
16 STATION INFORMATION EFFECTIVE FROM: 2000/03/15 18:18:00 INTERVAL
17 TO: CURRENT DATE (above) COMMENT
18 BIT 2 OF LLI FLAGS (+4) DATA COLLECTED UNDER 'AS' CONDITION COMMENT
19 L1 PHASE CENTRE 0.110m ABOVE ARP COMMENT
20 L2 PHASE CENTRE 0.128m ABOVE ARP COMMENT
21 where ARP is the Antenna Reference Point for HI measurement COMMENT
22 CONTACT xxxx@pgc.nrcan.gc.ca FOR ADDITIONAL INFORMATION COMMENT
23 P1 = P1 TurboRogue; = Y1 Benchmark COMMENT
24 L1 = L1(CA) COMMENT
25 P2 = P2 TurboRogue; = Y2 Benchmark COMMENT
26 L2 = L2(P2) TurboRogue; = L2(Y2) Benchmark COMMENT
27 SNR is mapped to RINEX snr flag value [1,4-9] COMMENT
28 SNR: >316 >100 >31.6 >10 >3.2 >0 bad=0 COMMENT
29 L1 & L2: 9 8 7 6 5 4 1 COMMENT
30 2001 3 31 0 0 0.0000000 GPS TIME OF FIRST OBS
31 32 01 3 31 0 0 0.0000000 0 10G10G22G17G15G18G 6G26G28G23G 3 END OF HEADER
33 -4571014.571 4 -3561795.634 4 23688534.679 23688535.001 23688542.297
34 37.000 27.000
35 -8749437.103 5 -6817728.119 5 22540847.552 22540847.117 22540852.845
36 66.000 48.000

```

Table 2. An Interpretation of GPS RINEX Observation File.

Explanation of GPS Observation Data Record			
Line 1	RINEX VERSION / TYPE	2.10	Format version
		OBSERVATION DATA	File type: O for Observation Data
		G (GPS)	Satellite System
Line 2,4,5	PGM / RUN BY / DATE	teqc	Name of program creating current file
		WCDA Data	Name of agency creating current file
		2000 July 20	Date and time of file creation
Line 3	COMMENT	Solaris 2.7 Ultra 2 cc SC5.0 =+- *Sparc	Solaris System
Line 6	MARKER NAME	site xxxxxxxx	The name of the marker
Line 7	MARKER NUMBER	xxxxxxxxxx	The number of the marker
Line 8	OBSERVER / AGENCY	xxxxxxxxxx xxxxxxxxxxxxxxxxxxxx	The name of the observer or agency
Line 9	REC # / TYPE/ VERS	2025	Receiver number
		AOA BENCHMARK ACT	Type
		3.3.32.2N 1k99/07/28REC	Version
Line 10	ANT # / TYPE	SG6085133254175	Antenna number
		STHG6SG6X-T970A	Type
Line 11	APPROX POSITION XYZ	X: 00000001.0000 Y: 00000001.0000 Z: 00000001.0000	Geocentric approximate marker position
Line 12	ANTENNA: DELTA H/E/N	0.100	- Antenna height: Height of the antenna reference point (ARP) above the marker
		East: 0.0000 West: 0.0000	- Horizontal eccentricity of ARP relative to the marker (east/north)
Line 13	WAVELENGTH FACT L1/ L2	1 1	The wavelength of L1 and L2
Line 14	#/ TYPES OF OBSER	-Satellite system code: G R E C S -Number of different observation types for the specified satellite system: 9 11 9 5 5 -Observation descriptors: 1 2 5 7 8	- Satellite system code (G/R/E/J/C/I/S) - Number of different observation types for the specified satellite system - Observation descriptors: Type: C = Code / Pseudorange L = Phase D = Doppler Band: 1=L1 2=L2 5=L5 7=E5b 8=E5a+b

		C1C L1C D1C C2W L2W C2X L2X C5X L5X C1C L1C D1C C1P L1P C2C L2C C2P L2P C3X C1X L1X D1X C8X L8X C7X L7X C5X L5X C1I L1I D1I C7I L7I C1C L1C D1C C5I L5I	Attribute: P: P code-based C: C code-based X: I+Q channels W: based on Z-tracking I: I channel
Line 15	INTERVAL	5.000	Observation interval in seconds
Line 16-17	STATION INFORMATION EFFECTIVE DAY	2000/03/15 18:18:00	The effective day of the station information
Line 18	COMMENT	BIT 2 OF LLI FLAGS (+4) DATA COLLECTED UNDER 'AS' CONDITION	The information of Loss-of-lock indicator
Line 19	L1 PHASE CENTRE	0.110 M ABOVE ARP	The information of L1 phase centre
Line 20	L2 PHASE CENTER	0.118 M ABOVE ARP	The information of L2 phase centre
Line 21-29	COMMENT	/	The Information of Sound Noise Ratio (SNR)
Line 30	TIME OF FIRST OBS	Year:2001 Month: 03 Day: 31 Hour: 0 Min: 0 Sec: 0.0000000 Time System: GPS	Time of first observation record (4-digit-year, month, day, hour, min, sec)
Line 31	END OF HEADER	/	Last record in the header section

3 Programming Script

In this section, MATLAB program is used to first read navigation and observation files. Afterwards, MATLAB program is also used for the upcoming orbit computation as well as the pseudo-range measurement.

3.1 Reading RINEX Navigation File

MATLAB programme script for reading the RINEX navigation file. The result is shown in Section 4.1. To get a better readability, the general steps are also commented.

```
%=====
%=====Read RINEX Navigation File=====
%
navigation_path='site0900.01n';
Nav_Data=fopen(navigation_path,'r');
%=====Find out the header section "END OF HEADER"=====
Read_Nav_Line=fgetl(Nav_Data);
while ischar(Read_Nav_Line)
if contains(Read_Nav_Line, 'END OF HEADER')
break;
end

Read_Nav_Line=fgetl(Nav_Data);
end
%=====Read Navigation Data=====
Epoch_Nav=1;
while ischar(Read_Nav_Line)
Read_Nav_Line=fgetl(Nav_Data);
if (Read_Nav_Line == -1)
Epoch_Nav=Epoch_Nav-1;
break;
end
%=====CV/ Epoch/ SV CLK=====
nav(Epoch_Nav).PRN=str2num(Read_Nav_Line(1:2));
nav(Epoch_Nav).Year=str2num(Read_Nav_Line(4:5));
nav(Epoch_Nav).Month=str2num(Read_Nav_Line(7:8));
nav(Epoch_Nav).Day=str2num(Read_Nav_Line(10:11));
nav(Epoch_Nav).Hour=str2num(Read_Nav_Line(13:14));
nav(Epoch_Nav).Minute=str2num(Read_Nav_Line(16:17));
nav(Epoch_Nav).Second=str2num(Read_Nav_Line(18:22));
nav(Epoch_Nav).Date_Numerical=datenum(nav(Epoch_Nav).Year,nav(Epoch_Nav).Month,nav(Epoch_Nav).Day,
nav(Epoch_Nav).Hour,nav(Epoch_Nav).Minute,nav(Epoch_Nav).Second);
No_of_weekday_In_nav=weekday(nav(Epoch_Nav).Date_Numerical)-1;
nav(Epoch_Nav).Time_in_GPS=No_of_weekday_In_nav*60*60*24+nav(Epoch_Nav).Hour*60*60+nav(Epoch_Nav).Minute*60+nav(Epoch_Nav).Second;
nav(Epoch_Nav).SV_Clock_Bias=str2num(Read_Nav_Line(23:41));
```

```

nav(Epoch_Nav).SV_Clock_drift=str2num(Read_Nav_Line(42:60));
nav(Epoch_Nav).SV_Clock_drift_rate=str2num(Read_Nav_Line(61:79));
Read_Nav_Line=fgetl(Nav_Data);
%=====Broadcast Orbit - 1=====
nav(Epoch_Nav).IODE=str2num(Read_Nav_Line(4:22));
nav(Epoch_Nav).Crs=str2num(Read_Nav_Line(23:41));
nav(Epoch_Nav).Delta_N=str2num(Read_Nav_Line(42:60));
nav(Epoch_Nav).M0=str2num(Read_Nav_Line(61:79));
Read_Nav_Line=fgetl(Nav_Data);
%=====Broadcast Orbit - 2=====
nav(Epoch_Nav).Cuc=str2num(Read_Nav_Line(4:22));
nav(Epoch_Nav).e=str2num(Read_Nav_Line(23:41));
nav(Epoch_Nav).Cus=str2num(Read_Nav_Line(42:60));
nav(Epoch_Nav).sqrt_a=str2num(Read_Nav_Line(61:79));
Read_Nav_Line=fgetl(Nav_Data);
%=====Broadcast Orbit - 3=====
nav(Epoch_Nav).Toe_time=str2num(Read_Nav_Line(4:22));
nav(Epoch_Nav).Cic=str2num(Read_Nav_Line(23:41));
nav(Epoch_Nav).Omega_0=str2num(Read_Nav_Line(42:60));
nav(Epoch_Nav).CIS=str2num(Read_Nav_Line(61:79));
Read_Nav_Line=fgetl(Nav_Data);
%=====Broadcast Orbit - 4=====
nav(Epoch_Nav).i0=str2num(Read_Nav_Line(4:22));
nav(Epoch_Nav).Crc=str2num(Read_Nav_Line(23:41));
nav(Epoch_Nav).Omega=str2num(Read_Nav_Line(42:60));
nav(Epoch_Nav).Omega_dot=str2num(Read_Nav_Line(61:79));
Read_Nav_Line=fgetl(Nav_Data);
%=====Broadcast Orbit - 5=====
nav(Epoch_Nav).IDOT=str2num(Read_Nav_Line(4:22));
nav(Epoch_Nav).weekNO=str2num(Read_Nav_Line(42:60));
Read_Nav_Line=fgetl(Nav_Data);
%=====Broadcast Orbit - 6=====
nav(Epoch_Nav).SV_accuracy=str2num(Read_Nav_Line(4:22));
nav(Epoch_Nav).SV_health=str2num(Read_Nav_Line(23:41));
nav(Epoch_Nav).TGD=str2num(Read_Nav_Line(42:60));
nav(Epoch_Nav).IODC=str2num(Read_Nav_Line(61:79));
Read_Nav_Line=fgetl(Nav_Data);
%=====Broadcast Orbit - 7=====
nav(Epoch_Nav).Transmission_timeofMessage=str2num(Read_Nav_Line(4:22));
nav(Epoch_Nav).Fit_interval=str2num(Read_Nav_Line(23:41));
nav(Epoch_Nav).Spare_1=str2num(Read_Nav_Line(42:60));
nav(Epoch_Nav).Spare_2=str2num(Read_Nav_Line(61:79));

Epoch_Nav=Epoch_Nav+1;
%====The END of Reading Navigation Data=====
end

```

3.2 Reading RINEX Observation File

MATLAB programme script for reading the RINEX observation file. The result is shown in Section 4.2. To get a better understanding, the general steps are also commented.

```

%=====
%=====Read RINEX Observation File=====
%=====

Observation_path='site0900.01o';
ObsData=fopen(Observation_path,'r');

%=====Find the number of Satellites=====
Read_Obs_C_Line=fgetl(ObsData);
while ischar(Read_Obs_C_Line)
if contains(Read_Obs_C_Line, '# / TYPES OF OBSERV');
break;
end

Read_Obs_C_Line=fgetl(ObsData);
end

```

```

No_of_Satellite=str2num(Read_Obs_C_Line(4:6));
%=====Find the C Position Index of Satellites=====
for i=1>No_of_Satellite
Types_of_observation{i}=Read_Obs_C_Line(4+6*i:6+6*i);
end

C_position_index = find(contains(Types_of_observation,'C1'));

%=====Find the Wording: "TIME OF FIRST OBS"=====
Read_ObsLine=fgetl(ObsData);
while ischar(Read_ObsLine)
if contains(Read_ObsLine, 'TIME OF FIRST OBS');
Obs_Time_of_first_obs=string(Read_ObsLine(5:6));
break;
end

Read_ObsLine=fgetl(ObsData);
end
%=====Find the Wording: "END OF HEADER"=====
Read_ObsLine=fgetl(ObsData);
while ischar(Read_ObsLine)
if contains(Read_ObsLine, 'END OF HEADER');
break;
end

Read_ObsLine=fgetl(ObsData);
end
%=====Read Navigation Data=====
Epoth_Obs=0;
while ischar(Read_ObsLine)
Read_ObsLine=fgetl(ObsData);
if (Read_ObsLine == -1)
Epoth_Obs=Epoth_Obs-1;
break;
end

if isequal(string(Read_ObsLine(2:3)),Obs_Time_of_first_obs)
No_of_PRN=str2num(Read_ObsLine(31:32));

for j=1: No_of_PRN
Obs(Epoth_Obs + j).PRN=str2num(Read_ObsLine(31+3*j:32+3*j));
Obs(Epoth_Obs + j).Year=str2num(Read_ObsLine(1:3))+2000;
Obs(Epoth_Obs + j).Month=str2num(Read_ObsLine(5:6));
Obs(Epoth_Obs + j).Day=str2num(Read_ObsLine(7:9));
Obs(Epoth_Obs + j).Hour=str2num(Read_ObsLine(11:12));
Obs(Epoth_Obs + j).Minute=str2num(Read_ObsLine(14:15));
Obs(Epoth_Obs + j).Second=str2num(Read_ObsLine(17:26));
Obs(Epoth_Obs + j).Epoch_Flag=str2num(Read_ObsLine(28:29));
Obs(Epoth_Obs + j).Date_numerical=datenum(Obs(Epoth_Obs + j).Year,Obs(Epoth_Obs + j).Month,Obs(Epoth_Obs + j).Day,Obs(Epoth_Obs + j).Hour,Obs(Epoth_Obs + j).Minute,Obs(Epoth_Obs + j).Second);
No_of_weekday_In_Obs=weekday(Obs(Epoth_Obs + j).Date_numerical)-1;
Obs(Epoth_Obs + j).Time_in_GPS=No_of_weekday_In_Obs*60*60*24+Obs(Epoth_Obs + j).Hour*60*60+Obs(Epoth_Obs + j).Minute*60+Obs(Epoth_Obs + j).Second;
end
%=====Read C1 Epoch's GPS Data=====
for k=1>No_of_PRN
Read_ObsLine=fgetl(ObsData);
Obs(Epoth_Obs + k).C1 = str2num(Read_ObsLine(2+16*C_position_index:14+16*C_position_index));
Read_ObsLine=fgetl(ObsData);
end

Epoth_Obs=Epoth_Obs + No_of_PRN;
end
%=====The END of Reading Observation Data=====
end

```

4 Output

4.1 Navigation File

The below screenshots shows the output of the above-mentioned MATLAB navigation scripts. The output is a 1x381 data structure, which contains: *PRN, year, month, day, hour, minute, second, date, GPS time, SV clock bias, SV clock drift and rate, IODE, Crs, delta n, m0, Cuc, e, Cus, sqrt(a), toe time, Cie, omega_0, Cis, i0, Crc, omega, omega dot, IDOT, week number, SV accuracy and SV health, TGD, IODC, Transmission of time mess, fit interval and spares.*

Figure 1 shows the MATLAB Editor window with the code 'LSGI3322_20016345D_Read_NavFile.m'. The variable 'nav' is a 1x381 struct containing 38 fields. The table displays the values for these fields across 25 rows, with columns 2 through 38 being collapsed. The visible fields include PRN, Year, Month, Day, Hour, Minute, Second, Date_Numerical, Time_in_GPS, SV_Clock_Bias, SV_Clock_drift, and SV_Clock_drift_rate.

Figure 1. Screenshot 1 of Reading RINEX Navigation Data

Figure 2 shows the MATLAB Editor window with the code 'LSGI3322_20016345D_Read_NavFile.m'. The variable 'nav' is a 1x381 struct containing 38 fields. The table displays the values for these fields across 25 rows, with columns 2 through 38 being collapsed. The visible fields include IODE, Crs, Delta_N, M0, Cuc, e, Cus, sqrt_a, Toe_time, Cic, Omega_0, Cis, i0, Crc, and Omega.

Figure 2. Screenshot 2 of Reading RINEX Navigation Data

The screenshot shows the MATLAB Editor window with the title 'Editor - LSGI3322_20016345D_Read_NavFile.m'. The current tab is 'nav'. Below it, a table displays a 1x381 struct with 38 fields. The first few columns show field names like Omega_dot, IDOT, weekNO, SV_accuracy, SV_health, TGD, IODC, Transmission_timeofMess, Fit_interval, Spare_1, and Spare_2. The data rows contain numerical values such as -8.1611e-09, -2.6323e-10, 1107, 1, 0, -1.8626e-09, 564, 511530, 0, 0, 0, and 0.

Fields	Omega_dot	IDOT	weekNO	SV_accuracy	SV_health	TGD	IODC	Transmission_timeofMess	Fit_interval	Spare_1	Spare_2
1	-8.1611e-09	-2.6323e-10	1107	1	0	-1.8626e-09	564	511530	0	0	0
2	-8.3303e-09	9.5718e-11	1107	4	0	-4.6566e-09	565	518148	0	0	0
3	-8.2207e-09	2.2858e-11	1107	2	0	-6.0536e-09	234	511218	0	0	0
4	-8.2671e-09	1.5536e-10	1107	2.8000	0	-5.1223e-09	569	511218	0	0	0
5	-8.4407e-09	2.0358e-11	1107	0	0	-2.3283e-09	59	511410	0	0	0
6	-8.2264e-09	-1.3572e-11	1107	2	0	-4.6566e-09	30	515538	0	0	0
7	-8.1336e-09	-4.3930e-11	1107	0	0	-6.0536e-09	170	511350	0	0	0
8	-8.8243e-09	6.6074e-11	1107	1	0	-8.3819e-09	790	512100	0	0	0
9	-7.8625e-09	2.0715e-10	1107	2	0	-1.1642e-08	529	511218	0	0	0
10	-7.9896e-09	2.7715e-10	1107	2	0	-1.0245e-08	272	511740	0	0	0
11	-8.3743e-09	7.6075e-11	1107	4	60	-2.7940e-09	98	511218	0	0	0
12	-8.2346e-09	1.7286e-10	1107	2.8000	0	-2.3283e-09	364	511218	0	0	0
13	-8.1228e-09	-9.9647e-11	1107	4	0	-7.4506e-09	261	514278	0	0	0
14	-8.3618e-09	3.0358e-11	1107	6.8500	60	-3.2596e-09	404	511206	0	0	0
15	-7.8600e-09	-7.9289e-11	1107	1	0	-1.8626e-09	228	511800	0	0	0
16	-8.3671e-09	-4.4859e-10	1107	2.8000	0	-4.1910e-09	310	511878	0	0	0
17	-7.6878e-09	-1.3929e-10	1107	2.8000	0	-2.7940e-09	450	513078	0	0	0
18	-8.1282e-09	2.9287e-11	1107	4	0	-9.3132e-10	550	511218	0	0	0
19	-8.5436e-09	1.7858e-11	1107	4	0	-7.4506e-09	276	511218	0	0	0
20	-8.1221e-09	1.9465e-10	1107	2.8000	0	-6.0536e-09	391	514278	0	0	0
21	-8.1468e-09	1.1786e-10	1107	3.4000	0	-4.1910e-09	539	511206	0	0	0
22	-7.7957e-09	-2.2680e-10	1107	4	0	-1.0710e-08	769	516948	0	0	0
23	-8.1146e-09	1.9108e-10	1107	0	0	-6.9849e-09	362	511410	0	0	0
24	-7.9600e-09	-2.8751e-10	1107	2.8000	0	-7.4506e-09	452	511218	0	0	0
25	-8.4079e-09	1.4179e-10	1107	2	0	-5.5879e-09	156	511380	0	0	0

Figure 3. Screenshot 3 of Reading RINEX Navigation Data

4.2 Observation File

The below screenshots shows the output of the above-mentioned MATLAB observation scripts. The output is a 1x23209 data structure, which contains: *PRN, year, month, day, hour, minute, second, epoch flag, data, time in GPS and the selected C1 epoch's GPS data* (it is required to use program code to read one epoch's GPS data).

The screenshot shows the MATLAB Editor window with the title 'Editor - LSGI3322_20016345D_Read_ObsFile.m'. The current tab is 'Obs'. Below it, a table displays a 1x23209 struct with 11 fields. The first few columns show field names like PRN, Year, Month, Day, Hour, Minute, Second, Epoch_Flag, Date_numerical, Time_in_GPS, and C1. The data rows contain numerical values such as 10, 2001, 3, 31, 0, 0, 0, 0, 730941, 518400, and 2.3689e+07.

Fields	PRN	Year	Month	Day	Hour	Minute	Second	Epoch_Flag	Date_numerical	Time_in_GPS	C1
1	10	2001	3	31	0	0	0	0	730941	518400	2.3689e+07
2	22	2001	3	31	0	0	0	0	730941	518400	2.2541e+07
3	17	2001	3	31	0	0	0	0	730941	518400	2.0429e+07
4	15	2001	3	31	0	0	0	0	730941	518400	2.0782e+07
5	18	2001	3	31	0	0	0	0	730941	518400	2.1938e+07
6	6	2001	3	31	0	0	0	0	730941	518400	2.1540e+07
7	26	2001	3	31	0	0	0	0	730941	518400	2.2252e+07
8	28	2001	3	31	0	0	0	0	730941	518400	2.4135e+07
9	23	2001	3	31	0	0	0	0	730941	518400	2.1490e+07
10	3	2001	3	31	0	0	0	0	730941	518400	2.4464e+07
11	10	2001	3	31	0	0	30	0	7.3094e+05	518430	2.3708e+07
12	22	2001	3	31	0	0	30	0	7.3094e+05	518430	2.2545e+07
13	17	2001	3	31	0	0	30	0	7.3094e+05	518430	2.0426e+07
14	15	2001	3	31	0	0	30	0	7.3094e+05	518430	2.0772e+07
15	18	2001	3	31	0	0	30	0	7.3094e+05	518430	2.1921e+07
16	6	2001	3	31	0	0	30	0	7.3094e+05	518430	2.1554e+07
17	26	2001	3	31	0	0	30	0	7.3094e+05	518430	2.2243e+07
18	28	2001	3	31	0	0	30	0	7.3094e+05	518430	2.4123e+07
19	23	2001	3	31	0	0	30	0	7.3094e+05	518430	2.1478e+07
20	3	2001	3	31	0	0	30	0	7.3094e+05	518430	2.4444e+07
21	10	2001	3	31	0	1	0	0	7.3094e+05	518460	2.3728e+07
22	22	2001	3	31	0	1	0	0	7.3094e+05	518460	2.2549e+07
23	17	2001	3	31	0	1	0	0	7.3094e+05	518460	2.0423e+07
24	15	2001	3	31	0	1	0	0	7.3094e+05	518460	2.0762e+07
25	18	2001	3	31	0	1	0	0	7.3094e+05	518460	2.1904e+07

Figure 4. Screenshot of Reading RINEX Observation Data

5 Conclusion and Future Planning

In this first intermediate period, it has completed the readings of both navigation and observation RINEX data, by opening their data structure in MATLAB workplace, in table format with the corresponding headings.

For the upcoming progress, it is required to complete the calculation of GPS satellite position. As long as both navigation and observation data are read, the components regarding orbit and satellite positioning computation can be carried out. For instance, time ephemerides (tk), time correction (t), mean anomaly (M), true anomaly (θ), mean motion (n), corrected inclination (i), corrected radius (r), corrected argument of latitude (ϕ), second order of Harmonic perturbation and terrestrial coordinates transformation.

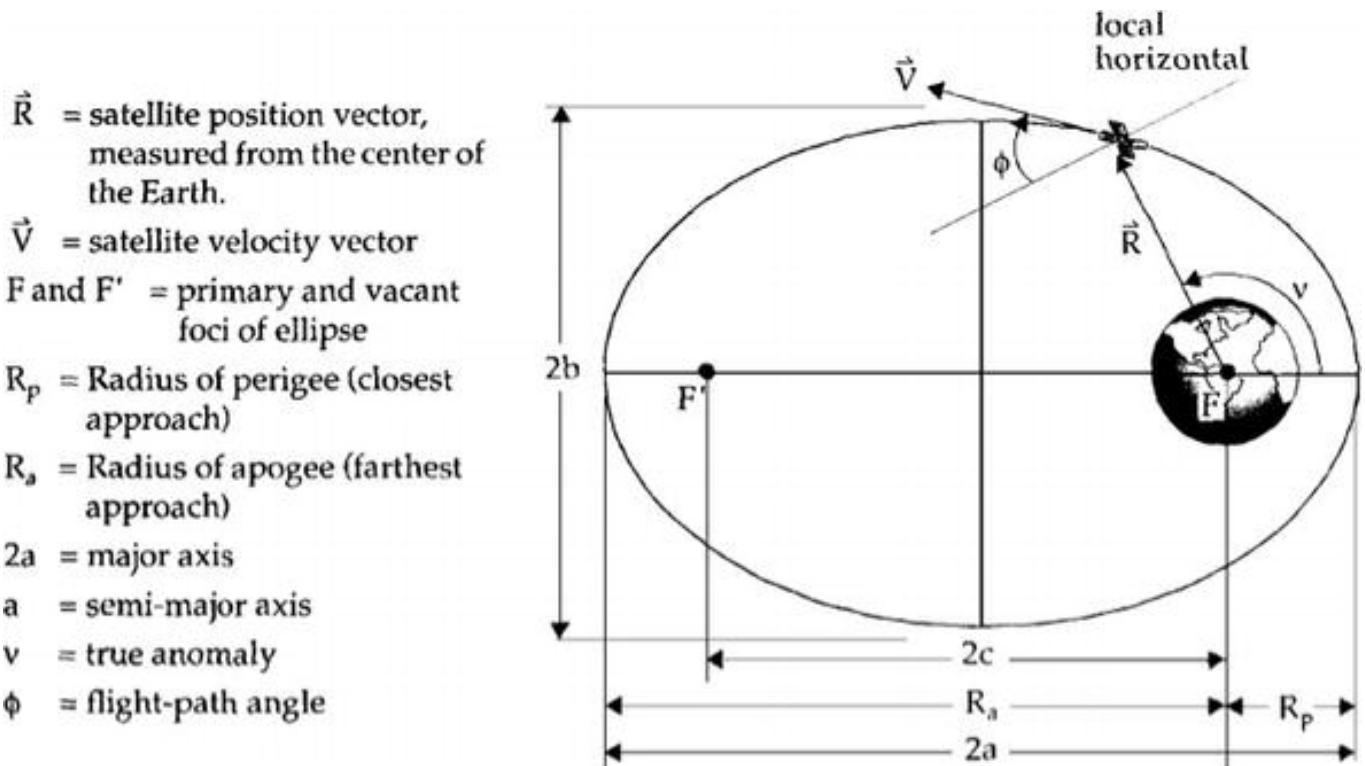


Figure 5. Several components regarding the orbit and satellite positioning components.

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--- THIS IS THE END OF THE FIRST INTERMEDIATE REPORT. THANK YOU. ---